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(54) **SYSTEMS AND METHODS FOR
INTEGRATED EMISSIVITY AND
TEMPERATURE MEASUREMENT OF A
SURFACE**

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patent is extended or adjusted under 35
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4, 2002.

(51) **Int. Cl.**⁷ **G01K 11/30; G06F 15100**

(52) **U.S. Cl.** 7021134

(58) **Field of Search** 702140, 127, 130,
702/134, 85, 99; 374/1, 2, 9, 121, 129, 126,
3741100

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Primary Examiner—Bryan Bui

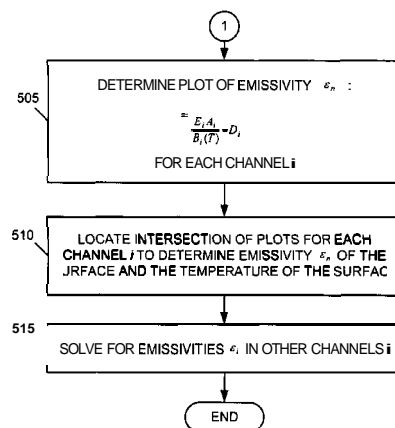
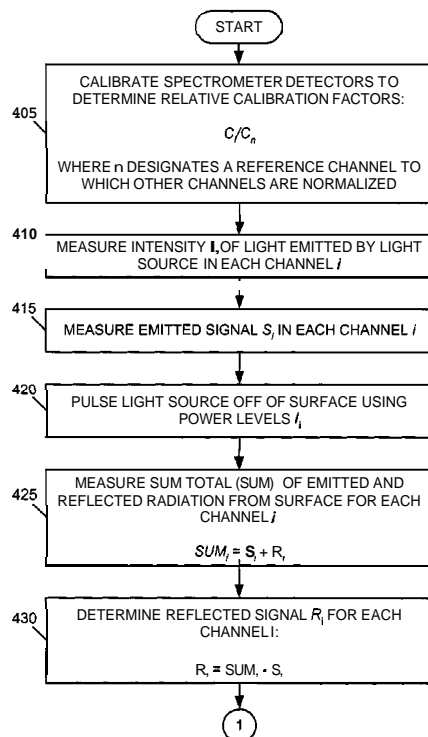
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(57) **ABSTRACT**

A multi-channel spectrometer and a light source are used to measure both the emitted and the reflected light from a surface which is at an elevated temperature relative to its environment. In a first method, the temperature of the surface and emissivity in each wavelength is calculated from a knowledge of the spectrum and the measurement of the incident and reflected light. In the second method, the reflected light is measured from a reference surface having a known reflectivity and the same geometry as the surface of interest and the emitted and the reflected light are measured for the surface of interest. These measurements permit the computation of the emissivity in each channel of the spectrometer and the temperature of the surface of interest.

4 Claims, 9 Drawing Sheets



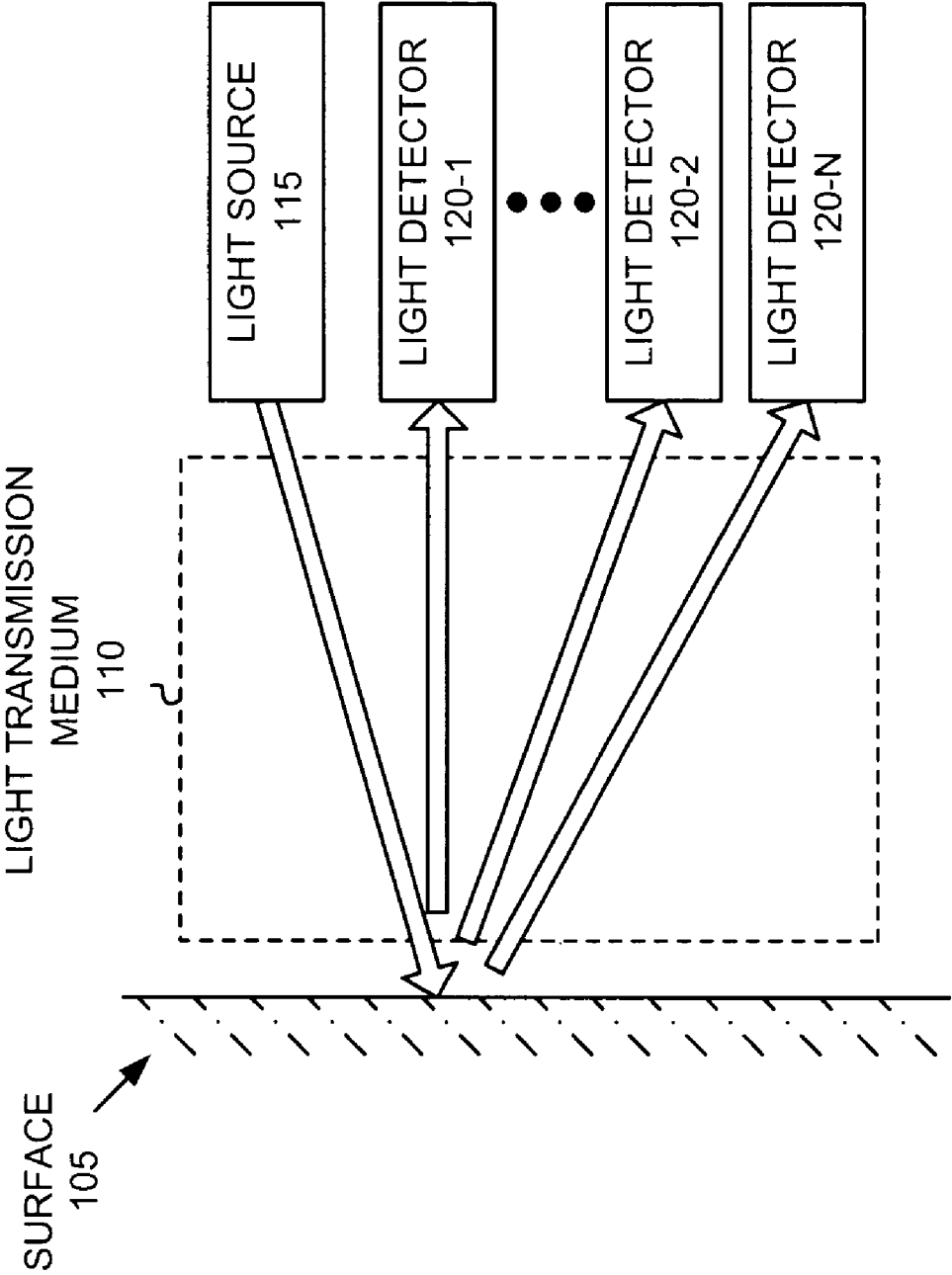


FIG. 1

200

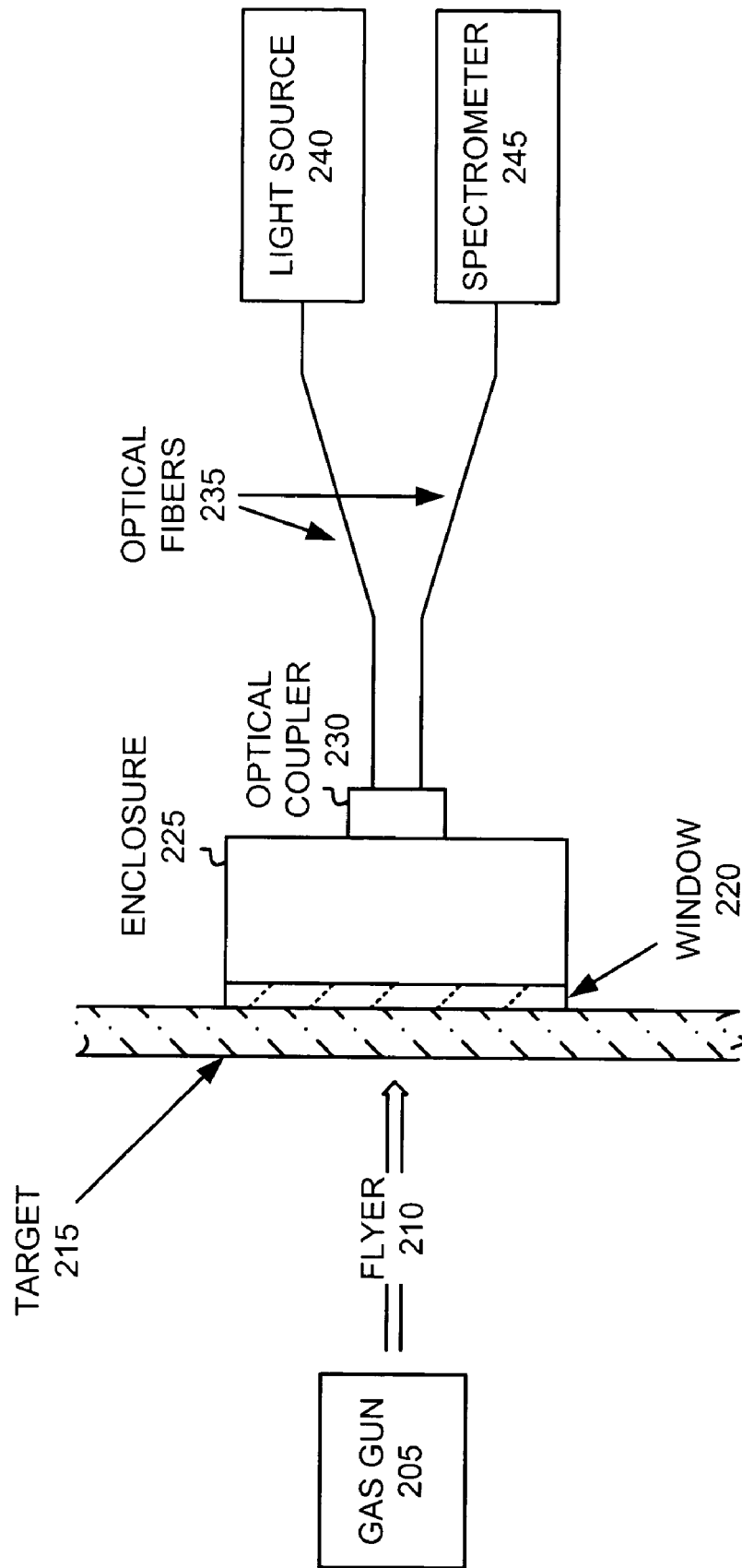


FIG. 2

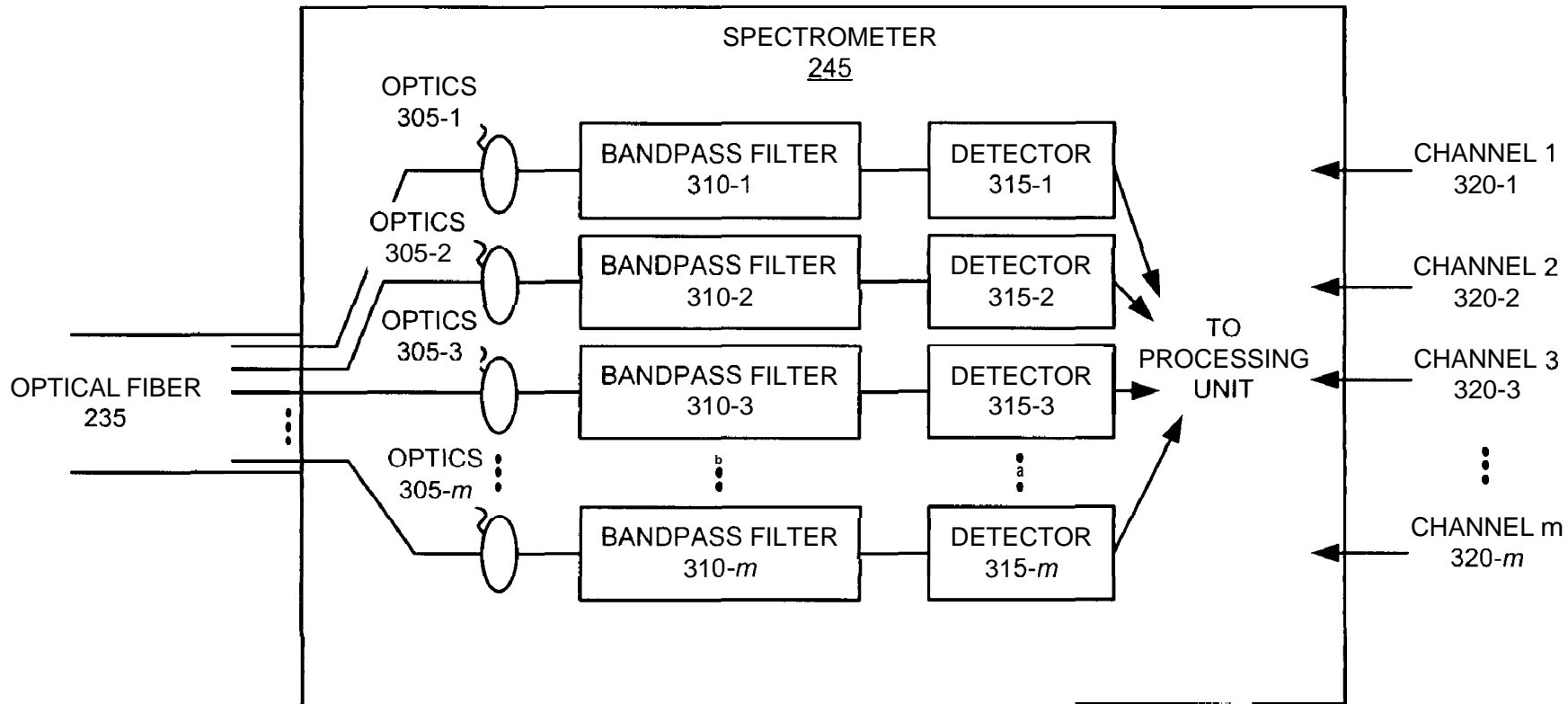


FIG. 3

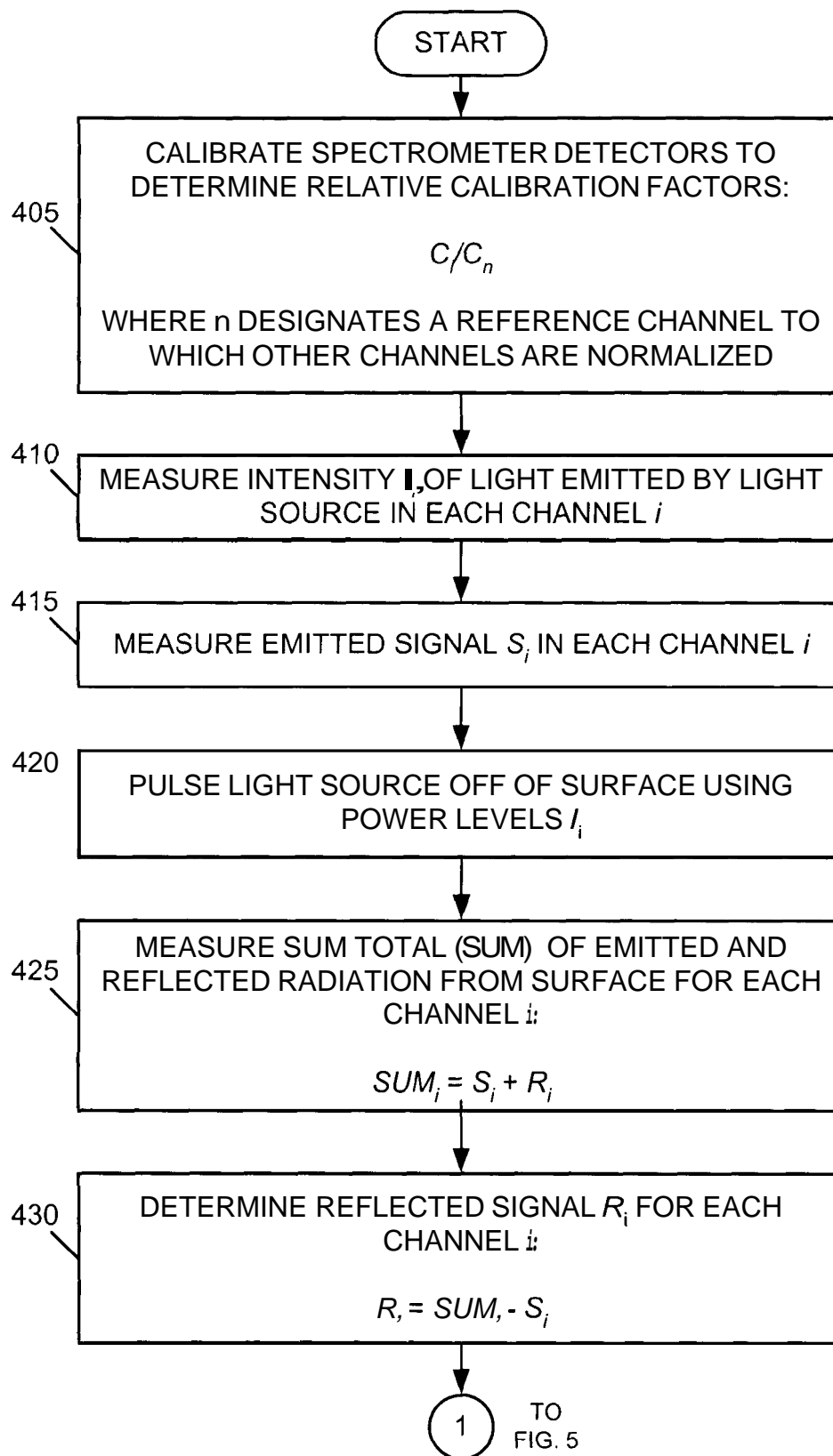
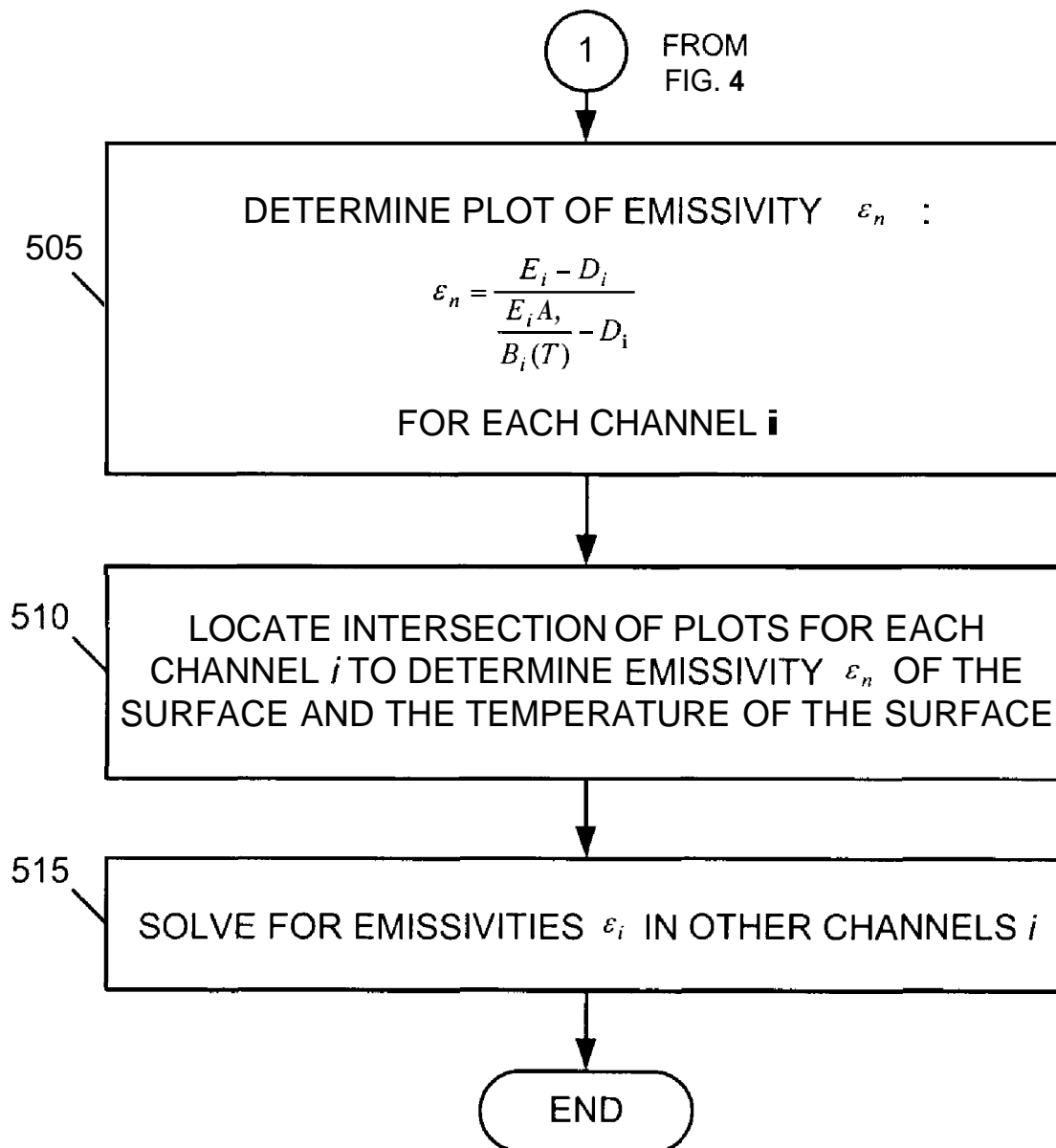


FIG. 4

**FIG. 5**

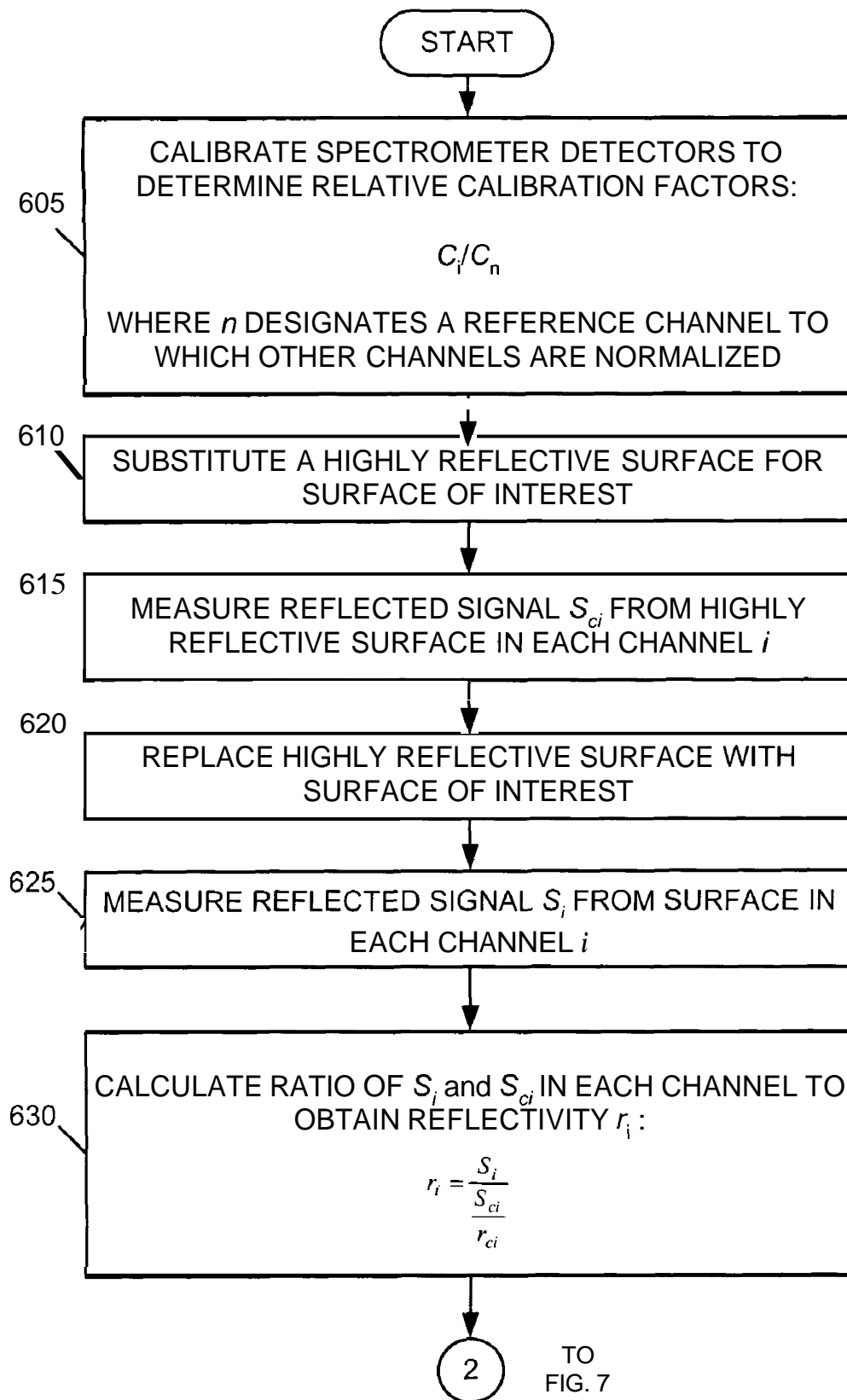
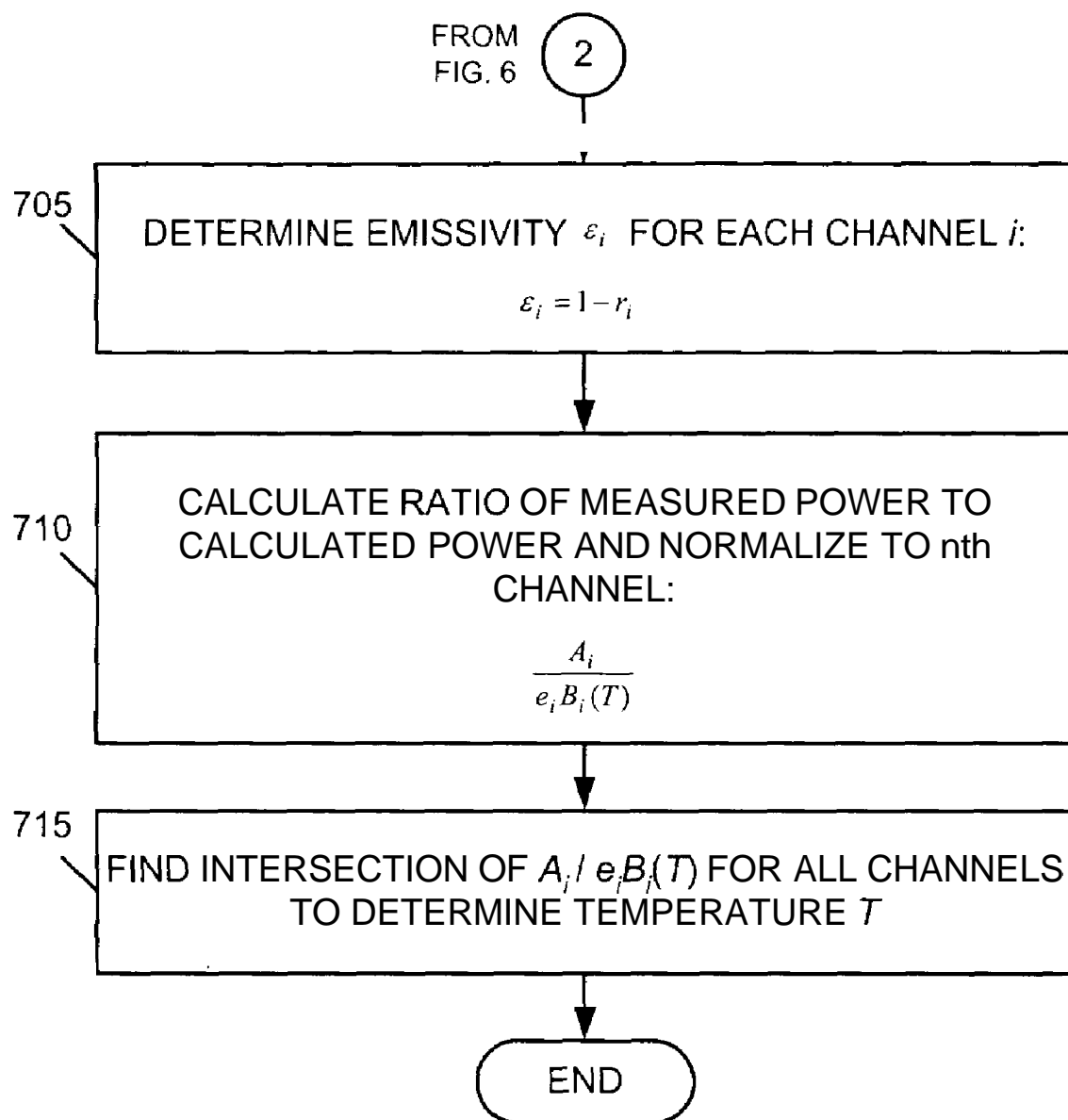


FIG. 6

**FIG. 7**

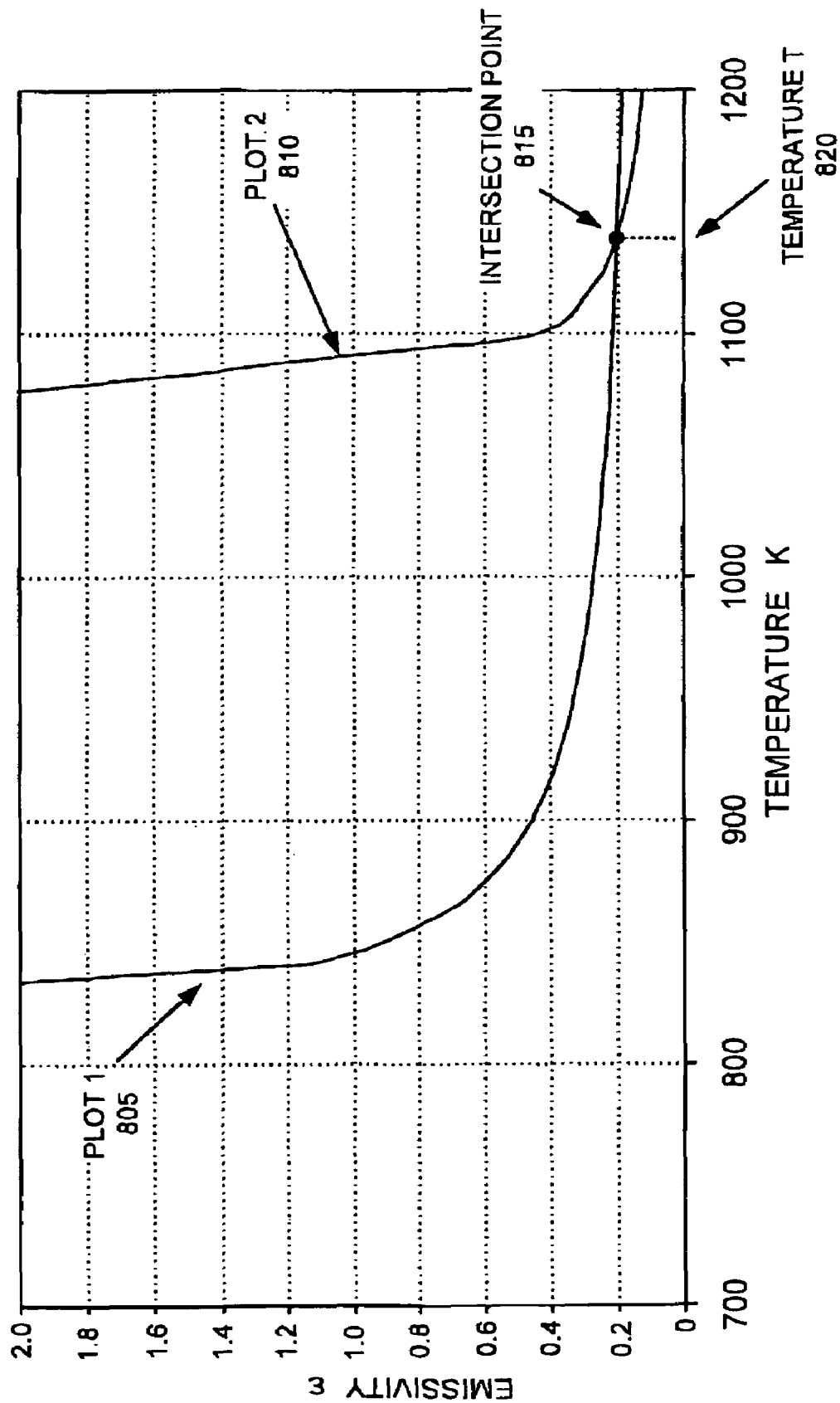


FIG. 8

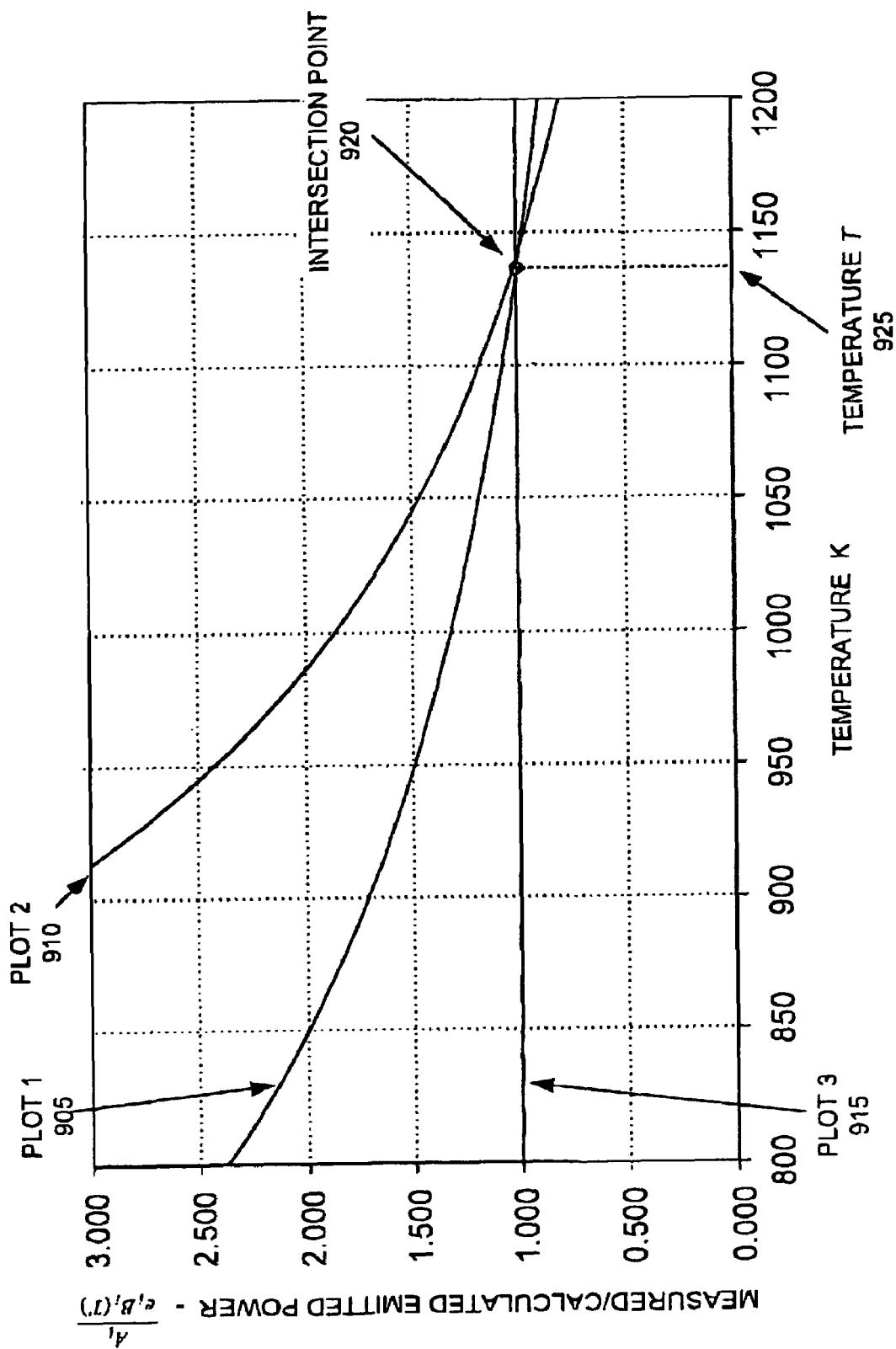


FIG. 9

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SYSTEMS AND METHODS FOR INTEGRATED EMISSIVITY AND TEMPERATURE MEASUREMENT OF A SURFACE

CROSS REFERENCE TO RELATED APPLICATIONS

The instant application claims priority from provisional application No. 601409,419, filed Sep. 4, 2002, the disclosure of which is incorporated by reference herein in its entirety.

GOVERNMENT CONTRACT

The United States Government has rights in this invention pursuant to Contract No. W-7405-ENG-48 between the United States Department of Energy and the University of California for the management and operation of Lawrence Livermore National Laboratory.

FIELD OF THE INVENTION

The present invention relates generally to pyrometry and, more particularly, to systems and methods for implementing pyrometric measurements of simple or complex surfaces of interest.

BACKGROUND OF THE INVENTION

It is relatively easy to measure the radiation spectrum from a target surface. The spectrum itself, however, does not yield information about the temperature without knowledge or information about the emissivity of the surface as a function of wavelength and possibly angle. The emissivity is often assumed to be constant (i.e., the gray body assumption) or to have a prescribed variation with wavelength. However, in actuality, the emissivity can vary strongly and without regularity as a function of wavelength. Hence, a lack of knowledge of the emissivity leads to a large uncertainty in the temperature obtained from the radiation measurement. In some present applications, the emitted light is compared to a surface having a known temperature, however, the validity of this comparison hinges on the assumption that the surface being measured and the reference temperature have the same emissivity.

SUMMARY OF THE INVENTION

Systems and methods consistent with the present invention add a measurement of reflected and/or incident light to the information used to determine surface emissivity and temperature. Either knowledge of the spectrum of the incident light, or a measurement of the light reflected from a known surface in addition to a measurement of the emitted and the sum of the emitted and the reflected light from the subject surface, allow the computation of the emissivity as a function of wavelength and the temperature of the surface. One feature of the present invention is that it does not use absolute values of the measurement. Instead, it uses relative values between channels. The use of relative values cancels out geometry factors and eliminates the need for absolute calibrations. Another feature of the present invention is that it is fast enough to be applied to rapidly moving or changing surfaces, such as surfaces of materials that are deformed or shocked by a high pressure impact or explosive. A further feature of the present invention is that the fiberoptics for the

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sensors and fiberoptics for the incoming light are located very nearly at the same point in space to automatically correct for the angle of dependence of the emissivity. An example of application of the method to a small and uniform spatial area at one point in time may be given by describing an experiment in which a flat plate is shocked. The method can easily be replicated both in space and time to produce a record of temporal and spatial variations in temperature by using arrays of sensors, such as can be provided by digital video cameras.

Other advantages and features of the present invention will become readily apparent to those skilled in this art from the following detailed description. The invention is capable of modifications in various obvious respects, all without departing from the invention. Accordingly, the drawings are to be regarded as illustrative in nature, and not as restrictive.

BRIEF DESCRIPTION OF THE DRAWINGS

The accompanying drawings, which are incorporated in and constitute a part of this specification, illustrate exemplary embodiments of the invention and, together with the description, explain the invention. In the drawings,

FIG. 1 illustrates an exemplary system for measuring the emissivity and temperature of a surface consistent with the present invention;

FIG. 2 illustrates an exemplary implementation of the system of FIG. 1 that measures the emissivity and temperature of a shocked surface consistent with the present invention;

FIG. 3 illustrates exemplary components of the spectrometer of FIG. 2 consistent with the invention;

FIGS. 4-5 illustrate an exemplary process that utilizes the relative magnitude of the emitted light in each channel and the incident light spectrum measured in the same channels for measuring the emissivity and temperature of a surface;

FIGS. 6-7 illustrate an exemplary process that utilizes a measurement of reflected power from a surface having a known reflectivity as a basis for determining the emissivity and temperature of a surface of interest;

FIG. 8 illustrates exemplary plots of emissivity versus temperature that may be employed with the exemplary process of FIGS. 4-5 for determining the emissivity and temperature of a surface; and

FIG. 9 illustrates exemplary plots of normalized ratios of measured to calculated emitted power as a function of temperature that may be employed with the exemplary process of FIGS. 6-7 for determining the temperature of a surface.

DETAILED DESCRIPTION

The following detailed description of the invention refers to the accompanying drawings. The same reference numbers in different drawings identify the same or similar elements. Also, the following detailed description does not limit the invention. Instead, the scope of the invention is defined by the appended claims.

FIG. 1 illustrates an exemplary surface emissivity and temperature measurement system **100** consistent with the present invention. System **100** may measure the temperature and emissivity of any object or surface using emitted and reflected light from a small area of the object or surface. System **100** may include a surface **105**, a light transmission medium **110**, a light source **115** and multiple light detectors **120-1** through **120-N**. Surface **105** may include any type of simple or complex surface such, for example, a room,

people, the ground, etc. Light transmission medium **110** may include any type of medium for conveying light from light source **115** to surface **105**, and from surface **105** to light detectors **120-1** through **120-N**. Light transmission medium **110** may include, for example, one or more optical fibers, lenses, etc. Light source **115** may include any type of pulsed light source such as, for example, a xenon high pressure lamp with a sapphire window. One skilled in the art will recognize, however, that other pulsed light sources may equivalently be used. Each of light detectors **120-1** through **120-N** may include a device for measuring the intensity of light over a given wavelength range (i.e., a channel), such as, for example, a spectrometer or a still or video digital camera. Light detectors **120-1** through **120-N**, and light source **115**, may be aimed to cover the same small area of surface **105**. Though not shown in FIG. 1, an array of light sources **115** and light detectors **120** may be arranged to produce a temperature map of complex surfaces or objects having different temperatures and emissivities over their surfaces.

Light detectors **120-1** through **120-N** may be aimed to cover the same small area of a surface or object and the emitted light from the surface or object may be measured. Light source **115** may then be activated such that light detectors **120-1** through **120-N** may measure the sum of the emitted and reflected light from the surface or object. The emissivity and temperature of the small area covered by light detectors **120-1** through **120-N** may then be calculated. In some exemplary implementations, light source **115** can be continuously pulsed to give a continuous reading of a temperature of surface **105**. System **100** may, thus, be replicated in time by pulsing light continuously, and in space by using an array of light detectors **120**. Light detectors **120-1** through **120-N** may be flashed on and off at a frequency related to the desired time resolution of the temperature measurement, so that (after processing the data) a continuous record of the temperature distribution of a surface or object as a function of time may be obtained.

If light detectors **120-1** through **120-N** include cameras, a bandpass filter may be placed in front of each camera so that each camera only measures a portion of the light spectrum. System **100** may use at least three cameras, each having a different bandpass filter. All of the cameras may be aimed to cover the same surface or object and the light emitted from the object may be measured. Light source **115** may then be activated and the cameras may measure the sum of the emitted light and the reflected light. Using the measured emitted and reflected light, the emissivity and temperature of each small area covered by a pixel in each of the three cameras may be calculated.

FIG. 2 illustrates a specific application **200** of system **100**, consistent with the present invention, for measuring the emissivity and temperature of a shocked surface. System **200** may include a gas gun **205**, a flyer **210**, a target **215**, a window **220**, an enclosure **225**, an optical coupler **230**, optical fibers **235**, a light source **240** and a spectrometer **245**. Gas gun **205** may include conventional driving mechanisms for driving flyer **210** to a predetermined velocity (e.g., 3 km/s) to impact target **215**. Flyer **210** may include a high density material such as, for example, copper, though other high density materials may be used. Target **215** may include, for example, a plate of material that serves as a target for flyer **210** when driven by gas gun **205**. Target **215** may include, for example, a plate of polished molybdenum (though other materials may be used) having a thickness of approximately 6 mm. Target **215** may be backed by window **220** that may include a sheet of, for example, lithium fluoride crystal. Window **220** may serve to increase the

pressure at the interface between target **215** and window **220** relative to a completely released condition. Window **220** may further prevent the creation of ejected micro-particles at the surface of target **215** when flyer **210** impacts target **215**.

Enclosure **225** may completely enclose the impact point of the flyer **210** upon target **215**, including enclosing window **220**. Enclosure **225** may include a cavity with a non-reflecting surface, such as, for example, a cavity darkened by soot to minimize reflections. Enclosure **225** may include a "light tight" cavity in which the only entry or egress of light into enclosure **225** is through optical coupler **230**. Optical coupler **230** may include a conventional coupling device for connecting the fibers from detectors (not shown) in spectrometer **245** and from light source **240** through enclosure **225**. Optical fibers **235** may serve as an optically transmissive medium for conveying light to enclosure **225** from light source **240** and from a surface of target **215** to detectors in spectrometer **245**. Light source **240** may include a pulsed light source such as, for example, an xenon high pressure lamp with a sapphire window. One skilled in the art will recognize, however, that other pulsed light sources may equivalently be used. Spectrometer **245** may include multiple channels for detecting and measuring signals with specified optical wavelengths from a surface of target **215**. The optical measurements from spectrometer **245** may be supplied to a processing device (not shown), such as a conventional computer, for subsequent processing.

Consistent with the exemplary process of the present invention, target **215** may be shocked by impact of flyer **210**, with the shock reaching the interface between target **215** and window **220**. The radiation from the shocked surface of target **215** emits through window **220** and travels through optical fibers **235** to spectrometer **245**. Once a steady state has been reached and measured on the surface of target **215** (steady state is reached in approximately 0.1 microseconds), pulsed light source **240** may be fired and radiation may be reflected off of the shocked surface of target **215**. The sum of the emitted and the reflected radiation may be measured by spectrometer **245**.

EXEMPLARY SPECTROMETER

FIG. 3 illustrates an exemplary spectrometer **245** that may measure radiation emitted or reflected from a target surface consistent with the present invention. Spectrometer **245** may include multiple optics **305-1** through **305-m**, multiple bandpass filters **310-1** through **310-m**, and multiple detectors **315-1** through **315-m**. Each optics **305**, bandpass filter **310** and detector **315** may be associated with a given channel **320-1** through **320-m** (i.e., a given wavelength interval). For example, optics **305-1**, bandpass filter **310-1** and detector **315-1** may be associated with channel **1 320-1**. Each optics **305** may receive signals from optical fiber **235** and may include, for example, a lens. Each bandpass filter **310** may receive signals from optics **305** and pass only those signals within its bandpass region to a respective detector **315**. Each detector **315** may convert the signals received from a respective bandpass filter **310** to an electrical signal that can be supplied to a processing unit, such as, for example, a personal computer (PC) for subsequent processing.

FIRST EXEMPLARY PROCESS FOR MEASURING EMISSIVITY AND TEMPERATURE OF A SURFACE

FIGS. 4 5 illustrate a first exemplary process for measuring the emissivity and temperature of a surface that

utilizes relative magnitudes of emitted light in each of several channels and the incident light spectrum measured in the same channels. In this first exemplary process, knowledge of the geometry of the surface and absolute determination of the emitted and reflected power is not required as long as the light is measured at the same point wherein the incident light originates. The exemplary process of FIGS. 4-5 uses the relative magnitude of the emitted light in each of the light detector **120** (or spectrometer **245**) channels and the incident light spectrum measured in each of the light detector **120** (or spectrometer **245**) channels to compute the channel emissivities and the surface temperature.

In accordance with this first exemplary process, the relation for the light emitted from surface **105** or target **215** is the following:

$$S_i = \epsilon_i F_i(T) K C_i \quad \text{Eqn. (1)}$$

where

- S_i is the surface emission signal in channel i ;
- ϵ_i is the emissivity of the surface or target in the wavelength interval of channel i ;
- F_i is the known black body power in channel i as a function of temperature T ;
- K is a geometry factor which is assumed to be independent of the wavelength channel; and
- C_i is a sensor calibration factor for channel i .

The relation for the light reflected from surface **105** or target **215** is the following:

$$R_i = (1 - \epsilon_i) I_i H C_i \quad \text{Eqn. (2)}$$

where

- R_i is the signal corresponding to the reflected light from light source **115** or light source **240** in channel i ;
- I_i is the power of the light from light source **115** or light source **240** in channel i ; and
- H is a geometry factor that may be assumed to be independent of the wavelength of the channel.

In Eqn. (2) above, it has been assumed that the reflectivity is one minus the emissivity for each channel. There are conditions relating to the angular independence of the emissivity that must be satisfied for this relation to be applicable.

To eliminate the geometry factors K and H , relative values may be used (i.e., the spectrum instead of absolute flux). Geometry factors K and H may be eliminated by choosing an arbitrary channel n and normalizing the equations. Typically the channel chosen for channel n is the one with good signal resolution for both the emitted and the reflected signal. Eqns. (1) and (2) become, when normalized to an arbitrary channel n for each channel i , the following:

$$A_i = \epsilon_i B_i(T) \quad \text{Eqn. (3)}$$

$$D_i = E_i(1 - \epsilon_i \epsilon_n) / (1 - \epsilon_n) \quad \text{Eqn. (4)}$$

$$\text{where } \epsilon_i \epsilon_n / \epsilon_n; \quad \text{Eqn. (5)}$$

$$A_i = S_i C_n / S_n C_i \quad \text{Eqn. (6)}$$

$$B_i = F_i / F_n \quad \text{Eqn. (7)}$$

$$D_i = R_i C_n / R_n C_i; \text{ and} \quad \text{Eqn. (8)}$$

$$E_i = I_i / I_n \quad \text{Eqn. (9)}$$

The values of A_i , D_i , E , and the functional dependence of B_i on T are known. Eqns. (3) and (4) may be solved for the emissivity ϵ_i as a function of the temperature T :

$$\epsilon_n = \frac{E_i - D_i}{\frac{E_i A_i}{B_i(T)} - D_i} \quad \text{Eqn. (10)}$$

The exemplary process may begin with the calibration of light detectors **120-1** through **120-N** (or spectrometer **145** detectors **215-1** through **215-m**) to determine relative calibration factors C_i/C_n , where n designates a reference channel to which all other channels are normalized [step **405**]. The intensity I_i of light emitted by light source **115** or **240** in each channel i may then be measured [act **410**]. In exemplary system **200**, subsequent to spectrometer calibration, flyer **210** may then be driven into a surface of target **215** using, for example, gas gun **205** to shock the surface of target **115**. An emitted signal S_i may be measured in each channel i associated with light detectors **120-1** through **120-N** [step **415**]. After measuring the emitted signal S_i in each channel i , light source **115** or **240** may then be pulsed off of surface **105** or the shocked surface of target **215** using the measured power level I_i for each channel i of light detectors **120-1** through **120-N** [step **420**]. During pulsing of light source **115** or **240** off of surface **105** or target **215**, a sum total (SUM.) of the emitted (S_i) and reflected (R_i) radiation from surface **105** or target **215** for each channel i may be measured [step **425**]:

$$SUM_i = S_i + R_i \quad \text{Eqn. (11)}$$

The reflected signal R_i for each channel i may be determined [step **430**] according to the following:

$$R_i = SUM_i - S_i \quad \text{Eqn. (12)}$$

A plot of emissivity ϵ_n of the reference channel n for each channel i may then be determined [step **505**] using the following relation (e.g., Eqn. (10)):

$$\epsilon_n = \frac{E_i - D_i}{\frac{E_i A_i}{B_i(T)} - D_i}$$

where:

$$A_i = S_i C_n / S_n C_i$$

$$B_i = F_i / F_n$$

$$D_i = R_i C_n / R_n C_i; \text{ and}$$

$$E_i = I_i / I_n.$$

An intersection of the all the plots of emissivity ϵ_n versus temperature for each channel i may be located to determine the actual emissivity ϵ_n and the temperature T of surface **105** or target **215** [step **510**]. For example, as shown in FIG. 8, plot **805** represents a plot of ϵ_n vs. T for channel **1**. Plot **2 810** represents a plot of ϵ_n vs. T for channel **2**. The intersection point **815**, where plot **1 805** and plot **2 810** intersect, represents the solution for the emissivity and temperature of surface **105** or target **215**. In the example shown, surface **105** or target **215** has an emissivity ϵ_n of 0.2 at a temperature (T) of 1140 K. The emissivities ϵ_i for all the other channels i may then be solved using the determined emissivity ϵ_n for channel n and Eqns. (3)-(9) [step **515**].

SECOND EXEMPLARY PROCESS FOR MEASURING EMISSIVITY AND TEMPERATURE OF A TARGET SURFACE

FIGS. 6–7 illustrate a second exemplary process that utilizes measurements of reflected power from a surface having a known reflectivity as a basis for determining the emissivity and temperature of a given surface. In this second exemplary process, a surface having a known reflectivity may be substituted for surface **105** or target **215** and a direct measurement of the reflected signal in each channel *i* from the substituted surface may be made. Surface **105** or target **215** may be substituted back into system **100** or **200**, respectively, and a measurement of the reflected signal in each channel *i* from surface **105** or target **215** may be made. The reflectivity in each channel may be obtained by calculating the ratio of the measured reflected signals.

The exemplary process of FIGS. 6–7 may begin with the calibration of light detectors **120-1** through **120-N** (or spectrometer **145** detectors **215-1** through **215-m**) to determine relative calibration factors C_i/C_n , where *n* designates a reference channel to which all other channels are normalized [step **605**]. A highly reflective surface, with a precisely known reflectivity r_i , may be substituted for the surface of interest (e.g., surface **105** or target **215** in systems **100** or **200**, respectively) [step **610**]. A reflected signal S_{ci} may be measured from the highly reflective surface in each channel *i* by pulsing light source **115** or **240** and measuring the reflected signal **S1** using light detectors **120-1** through **120-N** (or spectrometer **145**) [act **615**]. The highly reflective surface may then be replaced with the surface of interest (e.g., surface **105** or target **215**) [step **620**]. In the exemplary implementation of FIG. 2, flyer **210** may then be driven into target **215**, using gas gun **205**, to shock the surface of target **215**. The reflected signal S_i from the shocked target surface may be measured in each channel *i* [step **625**]. The ratio of S_i and S_{ci} in each channel *i* may be calculated [step **630**] to obtain the reflectivity r_i according to the following relation:

$$r_i = \frac{S_i}{\frac{S_{ci}}{r_{ci}}} \quad \text{Eqn (13)}$$

The emissivity ϵ_i for each channel *i* may be determined [step **705**] according to the following:

$$\epsilon_i = 1 - r_i \quad \text{Eqn (14)}$$

A ratio of measured power to calculated power may be calculated and normalized to a channel *n* for each channel *i* [step **710**]

$$\frac{A_i}{\epsilon_i B_i(T)} \quad \text{Eqn. (15)}$$

The intersection of the plots of $A_i/\epsilon_i B_i(T)$ for each channel *i* may be found to determine the target surface temperature *T* [step **715**]. For example, as shown in FIG. 9, plot **1 905** represents a plot of the ratio of measured power to calculated power (normalized to channel *n*) for channel **1**. Plot **2 910** represents a plot of the ratio of measured power to calculated power (normalized to channel *n*) for channel **2**. Plot **3 915**

represents a plot of the ratio of measured power to calculated power (normalized to channel *n*) for channel **3**. Plot **1 905**, plot **2 910** and plot **3 915** intersect at intersection point **920** at a temperature **T 925** of **1140 K**. Surface **105** or target **115**, thus, is **1140 K** in the example of FIG. 9.

The foregoing description of exemplary embodiments of the present invention provides illustration and description, but is not intended to be exhaustive or to limit the invention to the precise form disclosed. Modifications and variations are possible in light of the above teachings or may be acquired from practice of the invention. For example, while certain components of the invention have been described as implemented in software and others in hardware, other configurations may be possible. While series of steps have been described with regard to FIGS. 4–7, the order of the steps may vary in other implementations consistent with the present invention. Also, non-dependent steps may be performed in parallel. No element, act, or instruction used in the description of the present application should be construed as critical or essential to the invention unless explicitly described as such. Also, as used herein, the article "a" is intended to include one or more items. Where only one item is intended, the term "one" or similar language is used. The scope of the invention is defined by the following claims and their equivalents.

What is claimed is:

1. A method of obtaining an emissivity and temperature of a surface of interest, comprising:

substituting the surface of interest with a highly reflective surface having a known reflectivity (r_{ci});
measuring a reflected signal S_{ci} from the highly reflective surface in each wavelength interval *i* of a plurality of wavelength intervals;
replacing the highly reflective surface with the surface of interest;
measuring a reflected signal S_i from surface of interest in each wavelength interval *i*;
obtaining a reflectivity r_i for each wavelength interval *i* using the following relationship:

$$r_i = \frac{S_i}{\frac{S_{ci}}{r_{ci}}}$$

determining an emissivity (ϵ_i) for each wavelength interval *i* according to the following relationship:

$$\epsilon_i = 1 - r_i$$

plotting, for each wavelength interval *i*, a ratio of measured power to calculated power normalized to an *n*th wavelength interval; and

obtaining the temperature of the surface of interest based on the plots for each wavelength interval *i*.

2. The method of claim **1**, wherein obtaining the temperature of the surface of interest comprises locating an intersection of the plots with one another.

3. The method of claim **1**, further comprising:

shocking the surface of interest prior to measuring a reflected signal S_i from the surface of interest.

4. The method of claim **3**, wherein shocking the surface comprises:

driving a flyer into the surface of interest at a given velocity.

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