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CROSSED-MOLECULAR BEAM APPARATUS
TIME-OF-FLIGHT COMPUTER INTERFACE

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Date Transmitted: January 1977

PREPARED FOR THE U. S. ENERGY RESEARCH AND DEVELOPMENT ADMINISTRATION
UNDER CONTRACT W-7405-eng-82

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TABLE OF CONTENTS

	iv
	Page
ABSTRACT	iv, v
INTRODUCTION	vii
APPARATUS AND EXPERIMENTAL	1
APPENDIX I.	39
APPENDIX II.	41
APPENDIX III.	42
REFERENCES	56
PROGRAM LISTING	57

ABSTRACT

A variety of the gas phase reactions of barium and tin atoms have been studied using a crossed molecular beams machine with a universal quadrupole mass filter based detection system incorporating a time-of-flight (TOF) spectrometer for velocity analysis. Center-of-mass (CM) reactive product recoil angle and relative recoil energy distributions have been measured for the following reactions: 1) $\text{Ba} + \text{N}_2\text{O} \rightarrow \text{BaO} + \text{N}_2$ 2) $\text{Ba} + \text{CO}_2 \rightarrow \text{BaO} + \text{CO}$ 3) $\text{Ba} + \text{O}_2 \rightarrow \text{BaO} + \text{O}$ 4) $\text{Sn} + \text{Cl}_2 \rightarrow \text{SnCl} + \text{Cl}$ 5) $\text{Sn} + \text{CH}_3\text{I} \rightarrow \text{SnI} + \text{CH}_3$ 6) $\text{Sn} + n\text{-C}_3\text{H}_7\text{I} \rightarrow \text{SnI} + \text{C}_3\text{H}_7\text{I}$ and 7) $\text{Sn} + \text{CCl}_4 \rightarrow \text{SnCl} + \text{CCl}_3$. At relative collision energies below 15kJmole^{-1} , the BaO product of reaction 1) is seen to be sharply backwards scattered with only 10 to 20% of the total available energy in product recoil energy. These results are consistent with near 100% efficiency for production of electronically excited BaO. Evidence of a significant effect of N_2O vibrational energy on the total reactive cross section is presented. Reaction 2), at collision energies as high as 31kJmole^{-1} , and reaction 3), at a collision energy of 13kJmole^{-1} , both seem to proceed via a long-lived complex, and their product recoil energy distributions are reasonably well fit by the transition state theory of Herschbach et. al. No evidence of a CO_2 internal energy effect on the total reactive cross section for reaction 2) was found. The SnCl product

of reaction 4) was predominately forward scattered with low product recoil energy; the SnI product of reactions 5) and 6) were sharply backwards scattered with relatively high recoil energies; the SnCl product of reaction 7) was sideways scattered. Thus the reactions of tin atoms studied here are very similar to the corresponding alkali and alkaline earth metal atom reactions. A minicomputer interface for collection and display of TOF spectrums and product angular distributions was built using a Digital Equipment Corp. PDP8/e computer and is described here in detail.

INTRODUCTION

This report describes a time-of-flight spectrometer system for a crossed molecular beams machine utilizing a PDP8/e computer for on-line collection and display of data. Detailed descriptions, circuit diagrams, and program listings are presented. This report constitutes a chapter from the authors thesis, and is printed as an IS report so that the system, remaining at ISU, may be understood. The abstract lists those chemical reactions that have been studied with this system while at ISU. Further detail on these studies may be found in the author's thesis, from the University of California, Berkeley.

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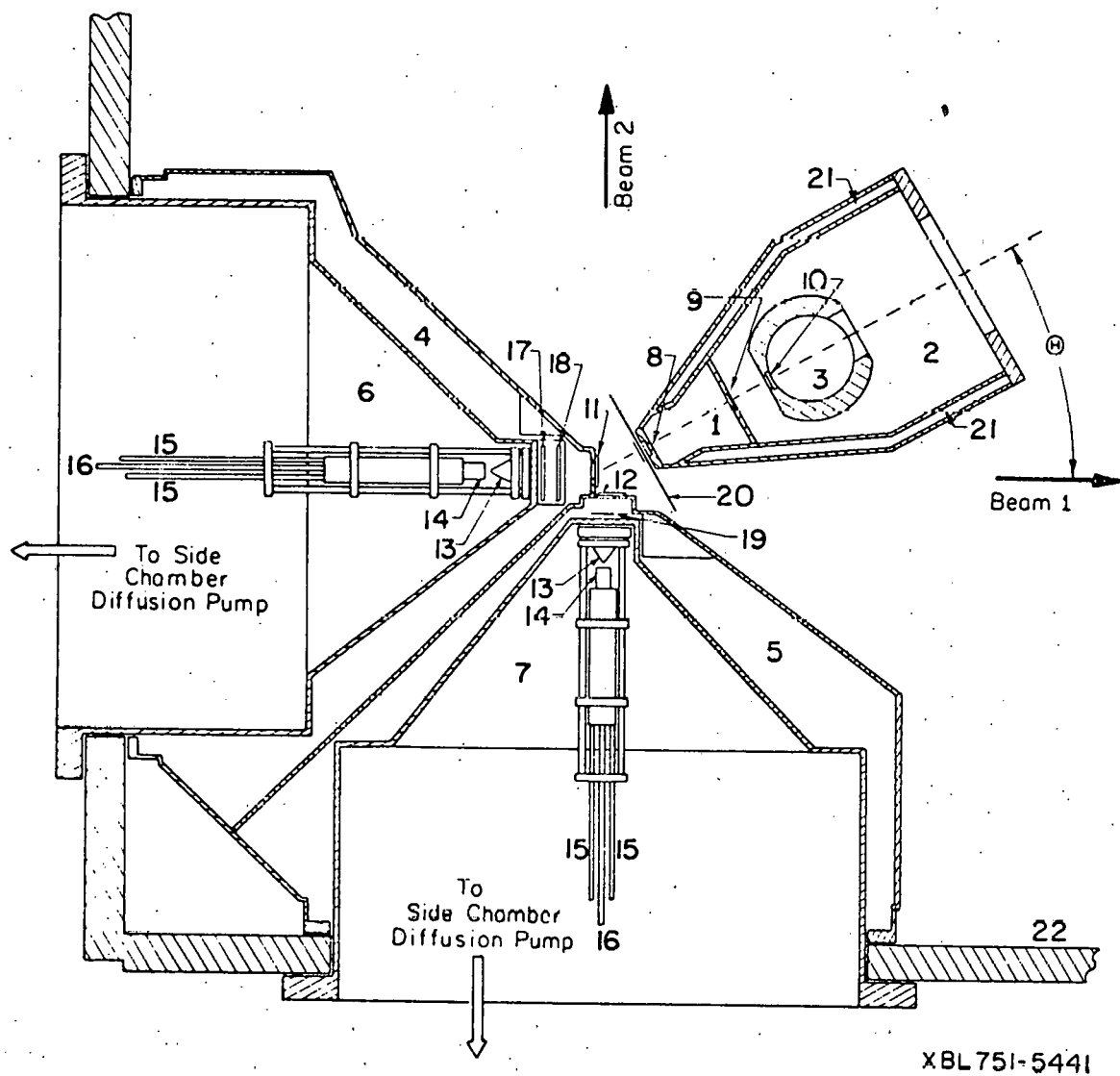
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APPARATUS AND EXPERIMENTAL

A. General Description

The crossed molecular beam machine used in this investigation is a modification of one designed by Y. T. Lee¹, and the details of its design criteria and construction have already been discussed⁵ by R. Behrens², who was responsible for the building of this apparatus. For the benefit of the reader, a brief description of the machine will be given here along with a more detailed description of the portions of the experimental set-up for which the author was responsible.

The two molecular beams are produced in two source chambers as shown in figure 1 with the following exception; the source in source chamber 2 is not a nozzle gas source as shown but a high temperature oven designed by A. Freedman, described in his thesis³, and shown in figure 2.

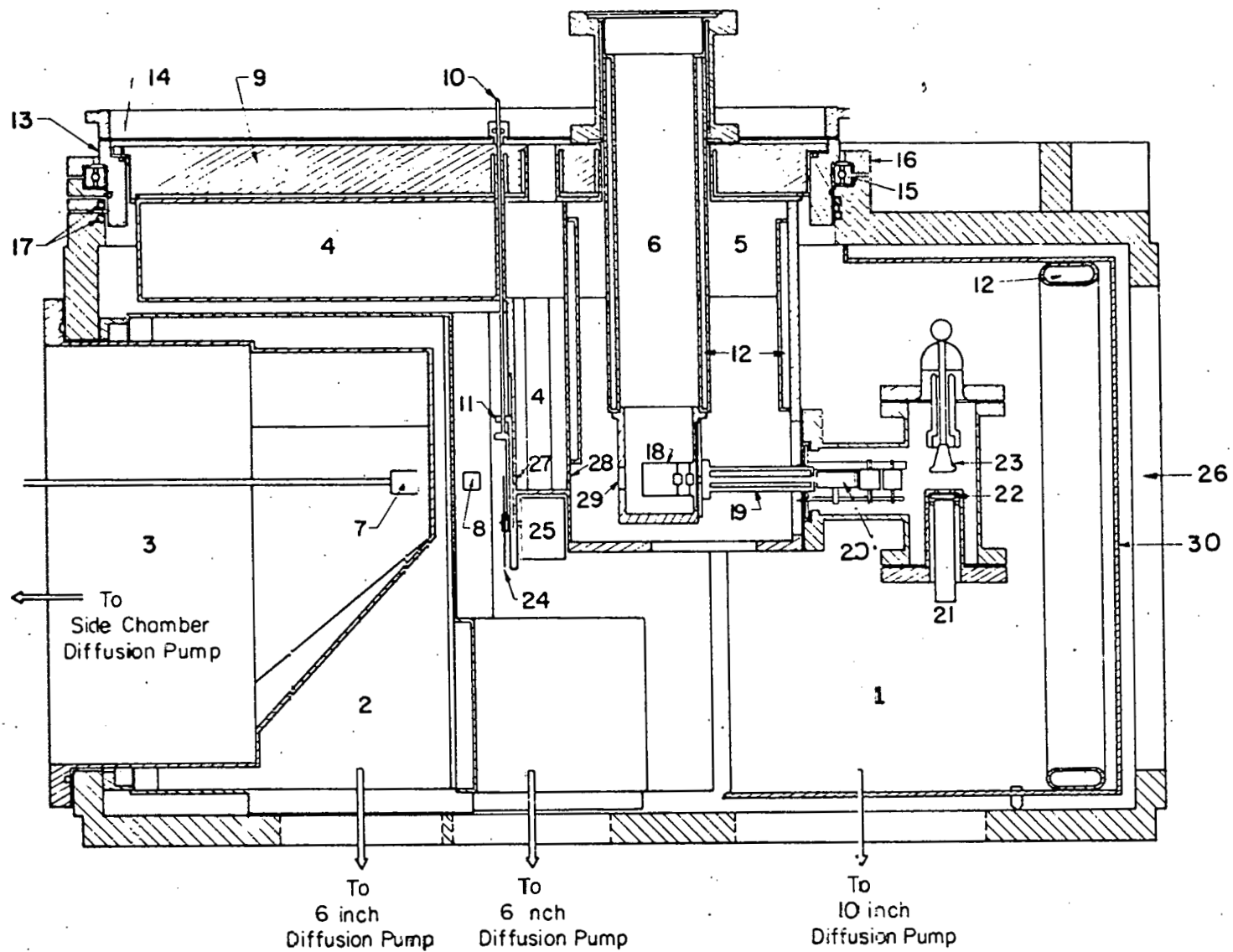


XBL75I-544I

Fig. 1 A. Top Cross Sectional View of the Apparatus

Figure IA. Top Cross Sectional View of the Apparatus

1. Detection Chamber No. 1.
2. Detection Chamber No. 2.
3. Ionization Chamber.
4. Collimation Chamber 1.
5. Collimation Chamber 2.
6. Source Chamber 1.
7. Source Chamber 2.
8. Detector Chamber 1 entrance slit.
9. Detector Chamber 2 entrance slit.
10. Ionization Chamber entrance slit.
11. Beam 1 Collimating slit.
12. Beam 2 Collimating slit.
13. Skimmers
14. Nozzles
15. Heating or cooling fluid tubes.
16. Gas inlet.
17. Beam Flag 1.
18. Beam chopper
19. Beam Flag 2.
20. Time-of-flight wheel.
21. Liquid nitrogen cooling.
22. Main Chamber.



XBL 751-5449

Fig. 1 B. Cross Sectional View of the Apparatus

Figure IB. Cross Sectional View of the Apparatus

1. Main chamber.
2. Collimation chamber No. 2.
3. Source chamber No. 2.
4. Detector chamber No. 1.
5. Detector chamber No. 2.
6. Ionization chamber.
7. Nozzle source No. 2.
8. Beam source No. 1.
9. Rotating detector lid.
10. Gate valve drive rod.
11. Detector gate valve.
12. Liquid nitrogen cooling reservoirs.
13. Rotating ring.
14. Rotating lid O-ring seal.
15. Kaydon KG350XPO 35 inch bearing.
16. Bearing retainer ring.
17. Tec-Ring seals No. A-01926.
18. Ionizer
19. EAI Quad 250 quadrupole mass filter.
20. Quadrupole exit lenses.
21. EMI 9524S photomultiplier tube.
22. Pilot B plastic scintillator.
23. High voltage cathode.
24. Time of flight chopping wheel.
25. IMC time of flight motor.

Figure IB (cont'd).

- 26. Access port.
- 27. Detector chamber 1 entrance slit.
- 28. Detector chamber 2 entrance slit.
- 29. Ionization chamber entrance slit.
- 30. Cold shield.

HIGH TEMPERATURE OVEN

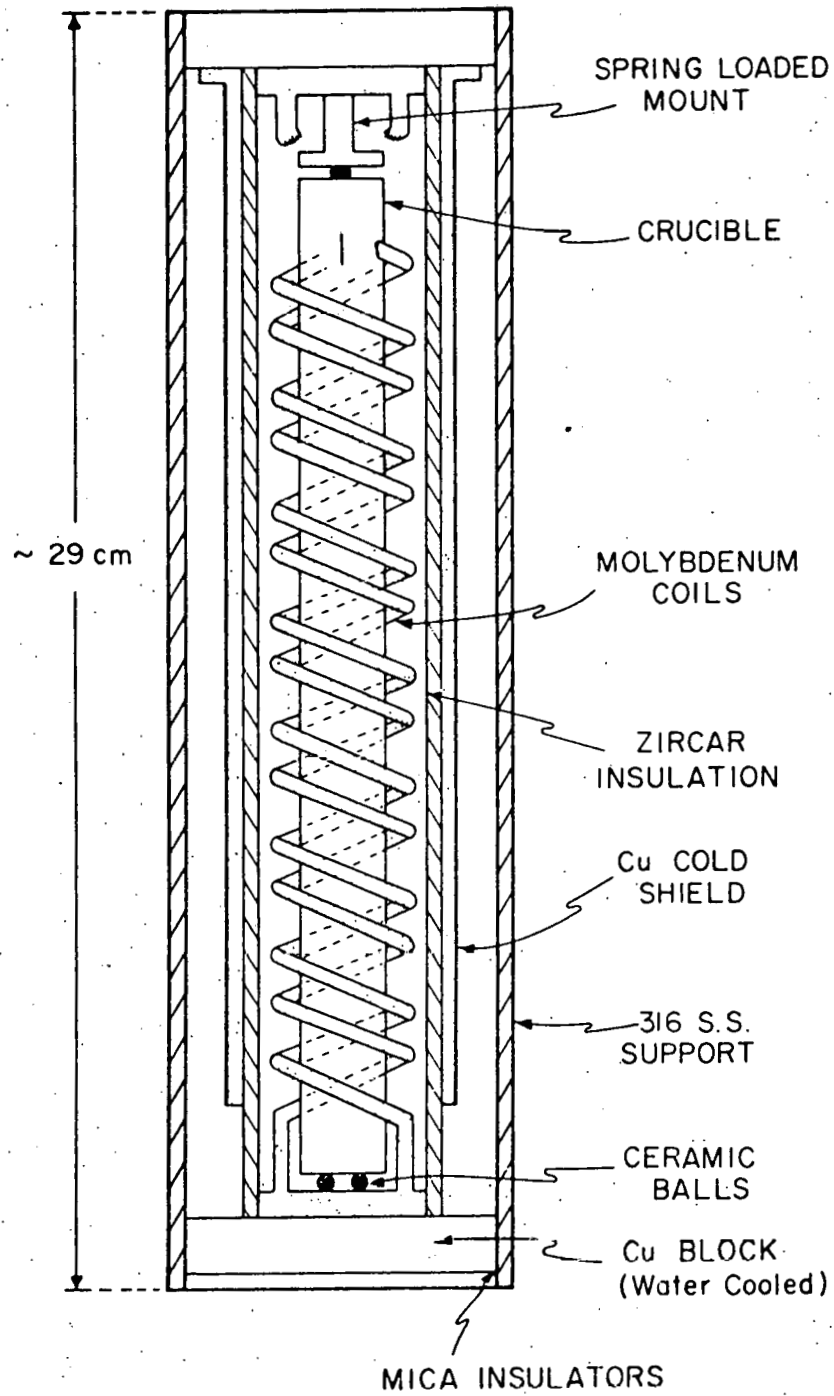


Fig. 2

Briefly it consists of , for the barium experiments, a molybdenum crucible, 1.9 cm in dia and 25 cm long, that holds the charge, with a 0.025 cm slit 0.5 cm long cut in the face as shown. The lid of the crucible is welded shut to prevent any barium effusion or creepage from the top of the crucible, a problem with press fit lids. The crucible is heated radiatively by molybdenum resistance heating coils wound around it operating at typical power levels of between 1 and 2 kW depending on the desired temperature. Wound around, but not touching, the heating coils is a layer of Zircor⁴ insulation, which is contained by a copper cold shield connected to water cooling. In front of the oven is a solenoid operated beam flag. For the tin experiments the molybdenum crucible had to be replaced because the molten tin literally dissolved and alloyed with the molybdenum. At first a quartz liner for the molybdenum was tried, but the tin reacted with the quartz, turning it white and crystalline and producing a strong tin oxide component to the beam. Finally a crucible almost identical in design to the molybdenum one but made from graphite was tried with great success. This crucible, shown in figure 3, employed a press fit lid which showed no signs of leaking, and could be easily reopened after a run without damaging either the lid or the crucible. Furthermore the graphite showed no signs of being attacked or penetrated by the tin (the old crucible showed signs of the tin diffusing through the moly so that the whole surface acted as a source rather than just the slit). The slit in the graphite crucible was cut by mechanical means, rather than the electron beam discharge cutter used on the molybdenum; for this reason it is wider - 0.5 cm versus 0.25 cm.

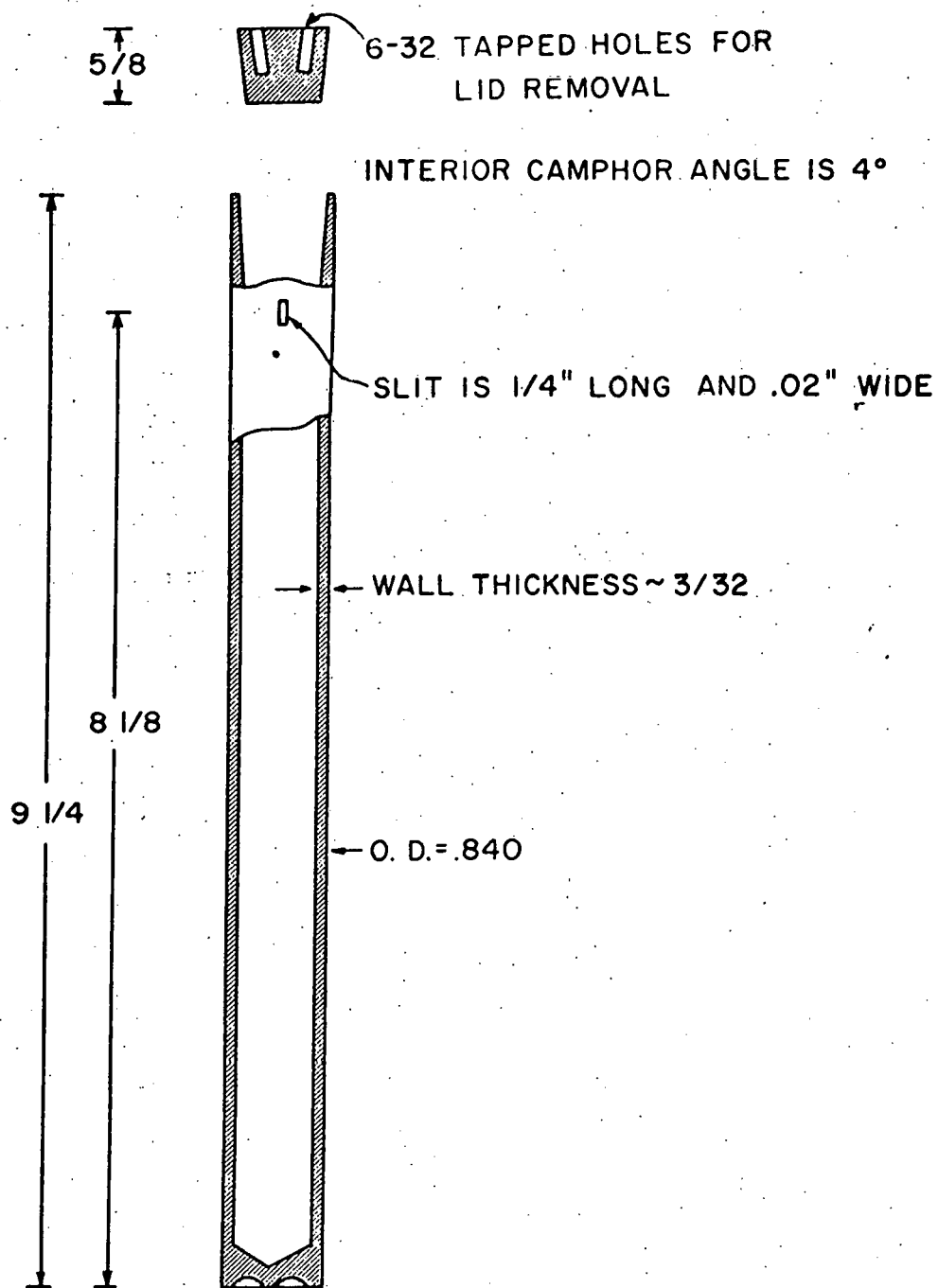


Fig. 3. Graphite Crucible

Since this oven certainly doesn't need the benefits of a differently pumped collimation chamber, the inside source chamber part 7 of figure 1 is removed and the oven source is mounted in the collimation chamber 2 which becomes the source chamber now pumped by the 10" side chamber oil diffusion pump through a 20" dia stainless steel el shaped chamber. The mounting is such that the slit in the crucible is 16.2 cm from the collimating slits (12 of figure 1) and 8.5 cm from the beam collision region. The collimating slits are two strips of Ta ribbon 0.2 cm apart and the geometry is such that the beam has an angular FWHM of 1.83 degrees and a spatial FWHM at the collision region of 0.27 cm for the barium experiments and the same for the tin experiments. (For the tin experiments the umbra is smaller and the penumbra larger, giving the same FWHM.

The other change involves the removal of the inside source chamber 1 (part 6 figure 1) and the remounting of the supersonic nozzle skimmer assembly (for a description see ref 2) on the collimation chamber on the back face of the now removed collimating slits (part 11). This was done in order to increase the intensity of the nozzle beam at the collision zone since it now placed the skimmer opening only 2.76 cm and the nozzle (which can be moved back and forth with respect to the skimmer) at a typical 4.11 cm from the collision region, compared to previous skimmer and nozzle distances of 10.5 and 11.9 cm respectively. Since the intensity of the beam falls off as the inverse of its distance to the collision zone squared one would expect an 8 fold increase in signal and alternations in the new configuration. The best way to compare beam intensity in the two configurations is to look at how much the nozzle beam attenuates the oven beam. In the old configuration we could never really measure any substantial attenuation, less than .1%, unless the source

pressure was turned up so high as to swamp the diffusion pump or the nozzle moved very close (less than 2 nm) to the skimmer (both procedures causing adverse broadening of the beam speed distribution). Furthermore in the old configuration the nozzle beam never made any measurable rise in the main chamber pressure suggesting that not much beam was getting into the main chamber. In the new configuration we typically run with between 1 and 4% attenuations with reasonable source pressures (between 200 and 300 torr) and turning on the beam would cause the main chamber pressure to rise from a typical background pressure of 4×10^{-7} torr to between 1 and 2×10^{-6} torr. This is just the pressure rise expected according to a simple calculation based on the measured attenuation and main chamber pumping speed estimates. Some of this improvement was due to the use of larger diameter nozzle and skimmer orifices, and was made at the expense of beam angular resolution; the new configuration was calculated to have a FWHM angular spread of 7° compared to the old configurations 0.8° . The 7° wide beam is suitable for reactive scattering where fine angular resolution is not needed, but is unsuitable for some elastic scattering studies where poor angular resolution would wash out the details of the angular distribution (see ref. 5). Another disadvantage of this configuration is that there is now no room for the beam flag and beam chopper as shown in figure 1 (parts 17 and 18), and to chop the beam a solenoid, (actually a relay), with a metal flag attached, is mounted on the main chamber face of the collimation chamber so that when not energized it blocks the beam's path to the collision zone and when energized it allows the beam to pass. The disadvantage is that when blocked the deflected beam still passes into the main chamber in the vicinity of the collision zone and might react with the barium or tin on surfaces surrounding the collision zone, either the collimation chambers or the detector. The

products of these surface reactions might be picked up by the detector and cause negative signals i.e. signals larger when the beam is off than on, that would invalidate the single collision reactive signal desired. No evidence of this problem was observed except with the bromine time of flight experiments discussed later. Also discussed later, in Appendix I, are the details of the nozzle beam geometry.

With the inside source chamber 1 removed, the nozzle beam is pumped by both the 10" CVC oil diffusion pump via the el chamber and the 6" NRC oil diffusion pump at the bottom of the collimation chamber. (The 10" pump is backed by a KDM-65 mechanical and a KMB 230 booster pump; the 10" diffusion pump on chamber 2 by a KDM-80, and all other diffusion pumps by KC-15 mechanical pumps manufactured by Kinney vacuum.) Typical source chamber 1 pressure while operating the nozzle is 2×10^{-4} torr.

The two beams pass into and collide in the main chamber made, as are almost all components of this machine that are in vacuum, of stainless steel, and pumped by a cryogenically baffled 10" CVC oil diffusion pump and indirectly by a 6" NRC pump. Typical main chamber background pressure is 5×10^{-7} torr. The size of the main chamber, 132 cm x 130 cm x 84 cm, was dictated by the size of the rotating lid, housing the detector. Inside the main chamber, encompassing the region that the detector rotates in, is a cryogenically cooled copper cold shield, affording large pumping speed for condensable gases (LN_2 temperatures). (The area between the cold shield and main chamber is a semi-sealed region pumped by the aforementioned 6" NRC oil diffusion pump and the nozzle beam strikes a beam deflector and is pumped out through this region unless the detector is in the way.)

The detector is housed in a triply differentially pumped cryogenically cooled vacuum chamber made of stainless steel and mounted on an 89 cm dia lid in a horizontal plane defined by the two beams, from -8° to 105° where 0° is defined as the nozzle beam (source 1). The rotating seal consists of two Teflon Tec-Ring gaskets (Tec Seal Corp). The detector entrance slit (figure 1 part 8) is 0.380 cm square and 4.44 cm from the collision zone, (for measuring the beams another slit is available which is only 0.008 cm in diameter to decrease the intensity of incoming gas and prevent overload of the detector). This chamber is pumped by an Ultek 150 l/sec ion pump, backed by a titanium sublimator, and the next chamber (2 on figure 1) surrounds the ionizing chamber and is pumped by a Veeco 225 l/sec ion pump, backed by a titanium electron beam sublimator. The third chamber (3 of figure 1) houses the ionizing portion of the mass spectrographic detection system, is liquid nitrogen cooled, and pumped by a Veeco 100 l/sec noble gas pump and by a liquid He pump not used in these experiments... typical background pressures in this chamber are 2×10^{-9} torr. The detection system consists of an axial electron bombardment ionizer housed in chamber 3 and constructed so that particles from the collision region which enter and are not ionized pass out through the back, are pumped away in chamber 2, and do not contribute to the background gas near the ionizer. Particles that are ionized are drawn out and focussed by an electrostatic lens system to enter a quadrupole mass filter and ions of the selected mass are passed out the back of the quadrupole and accelerated by a pair of lenses to strike an aluminum plated cathode held at minus 30 kV with respect to the energy of the ions. The resulting shower of ejected electrons are accelerated and strike an aluminum coated plastic scintillator causing

a shower of photons which are coupled to a photomultiplier tube (EMI 95245) producing a current burst at the output for each ion produced in the ionizer. The advantage of this system over using an electron multiplier to measure the ions is twofold: first the PM is sealed and therefore not subject to rather rapid degradation (by the corrosive gases used in some experiments) as open electron multipliers are. Second, each ion that strikes the Al cathode produces many electrons (estimated 2 to 6) which, when they strike the plastic scintillator, (Pilot B) produce about 12 photons; therefore each ion produces a burst of about 50 photons, and the problem of discriminating ion counts from stray light photon background counts becomes very easy. A disadvantage was that the high voltage needed for the cathode would cause electrical interference, by some form of non-sustaining discharge phenomenon, that would cause the computer interface to malfunction. This was completely solved by electrostatically shielding all conductors of the high voltage inside the vacuum chamber. The current bursts from the PM tube are counted electronically by a system described later. The ionizer used for the barium experiments, similar in design to that of Brink⁶ and designed and described by Behrens², was normally operated at an electron energy of 60 V, electron current of between 10 and 15 mA and had an estimated ionizing region length of about 4 cm. The ionizer for the tin experiments, an Extranuclear Labs Model 041-1, was normally operated at 100 V electron energy and 5mA emission, and had an estimated ionizing length of 0.5 cm. For looking at the beams the emission was dropped to between $10\mu\text{A}$ and $100\mu\text{A}$ for both ionizers. Although not carefully measured (a difficult procedure), the typical ionizing efficiency of this type of ionizer lies between 0.01 and 0.1 so the probability of ionizing a particle is directly proportional to the length

of time it spends in the ionizer and therefore inversely proportional to the particles velocity causing the detector to measure a number density as opposed to a flux. (Particles per sq. cm per sec. times inverse velocity bias equals particles per cubic cm.) The quadrapole mass filter used for the barium experiment was an Electronics Assoc. Inc. Model Quad 250 along with its associated driving electronics. The quadrapole used in the tin experiments was an Extranuclear Labs Model 4-270-9 with Models 11, 13 and 15 Hi-Q heads and Models 020-2, and 011-15 driving electronics. The motivation for the change was hoped increased sensitivity of the companion ionizer, better resolution of the mass filter, and more accurate and stable mass selection (the EAI mass filter's calibration was non-linear and changed frequently enough to make hunting for a mass peak an annoying procedure).

B. AC Synchronous Detection

Along with the reactive signal coming from the collision region there is also present at the output a background signal due to background gas in the detector or in the main chamber. One way to subtract out this background, or baseline, is to measure it before the beams are on and subtract that quantity from all later measurements, but this assumes that the baseline is constant with time. A better method, the one used here is to constantly measure this baseline by periodically interrupting the nozzle beam and accumulating the baseline count in one set of counters and the signal plus baseline count, when the beam is not interrupted, in a different set of counters. The difference between the two will then be that portion of the signal that depends on the collision of the beams without any added baseline. (An even better method, judged too complicated in this application, would be to chop both beams and measure the signal

at the beat of the two chopping frequencies; this would subtract all components of the signal not dependent on having both the beams on and colliding.) The duty cycle (time of beam on/total time) was chosen to be 50% for reasons of simplicity. (50% is not always the optimum duty cycle: if the baseline is very small or zero there is of course no point in measuring it so the optimum duty cycle would be 100%. It can be shown, appendix II, that the optimum duty cycle is equal to $(S+B)/(S+2B)$ where S and B are the signal and baseline rates respectively.) Although one would like to have a high chopping frequency so as to average out fast drifts in the baseline, the frequency is limited by the spread in flight times of the beam particles from the chopping mechanism to the collision region and to the detector. There may be a sharply defined burst of particles at the chopping mechanism but since some travel faster than others, both in the beam and after they have reacted at the collision region, they may get to the detector before the slow particles from the previous pulse arrive; the phase information has been lost and synchronous detection won't work. For an idea of the time scale involved we make the more stringent requirement that we will discard all counts obtained for that amount of time past each beam on or off transition such that all but the slowest few percent of the particles in the beam have time to get to the detector. With a typical distance of 7 cm and typical slow velocity of 5×10^3 cm/sec we get 1.4 msec, so a chopping frequency of 180 Hz would be wasting half the counting time on waiting for this "dead time".

In the original configuration of this machine, the beam was chopped using a motor driven wheel with slots in it that interrupted the beam in the collimation chamber as shown in figure 1 (part 18). An advantage of this system was that when the beam was chopped off all of it was pumped away in the collimation chamber and none got into the main

chamber to cause the previously discussed negative signals, from surface reactions. (A lesser advantage is that only half the total beam winds up in the main chamber as background gas.) A disadvantage was that the motor was inside the source chamber and the relatively high concentration of corrosive gases (when running such gasses in the nozzle) caused frequent failure of the motor bearings. Another disadvantage was, as previously mentioned, the nozzle had to be further away from the collision zone causing less beam intensity, so it was moved up to the collimation chamber face and the new beam chopper, mounted in the main chamber, was used. This beam chopper (the metal flag mounted on a relay) is much slower than the rotating wheel because, mechanically, the relay won't move much faster than a few cycles per second. A chopping period of 200 msec on and 200 msec off was chosen with 20 msec thrown out at each transition to allow the flag to move into position and the transients (the flag oscillates slightly) to die down. (The chopping frequency could be made much slower, all the way to DC, but problems would develop with the computer interface to be discussed later.)

C. Time of Flight Spectrometry

The system described so far measures the reactive signal as a function of scattered angle. It is also useful to measure it as a function of the velocity of its constituent particles. For inelastic scattering studies this velocity distribution, giving a translational energy distribution, would be helpful in elucidating the energy transfer mechanism being studied. The translational energy distribution of reactive scattering would also be helpful in elucidating the total reactive product energy partitioning, an important part of microscopic kinetics. Finally the velocity distribution of the beams and scattered signal are needed to form the Newton diagram

and transform the differential cross section from the lab to the more meaningful center of mass frame of reference (see chap. 3). Therefore it would be advantageous to have some device that specifies or selects velocity in conjuncture with the rotatable lid that selects angle.

Conventional velocity selectors operate on a time of flight principle and consist of arrays of slotted disks, arranged in series on a common shaft, and spun by a variable RPM motor. The disks are precisely aligned so that only those particles within a certain small velocity range will traverse through the slots without hitting any of the tines and being deflected out of the path to the detector. The selected velocity range is varied by changing the motor RPM and the resolution is given by the construction of the slotted disks. These selectors are however, too big to incorporate in most rotatable detector molecular beam machines, and furthermore since it takes considerable time to complete a velocity sweep, changing beam conditions could distort the measured velocity distributions. **A much simpler way to obtain velocity information is by means of a single slotted disk, spun so that it allows short bursts of molecules to enter the detector. Each of the molecules in the burst travels the distance, d , between the slotted disk and the ionizer portion of the detector, where they are ionized and quickly detected. It is obvious that the faster ones get there sooner than the slow ones. The signal at the detector therefore consists of a burst of counts, where each count is delayed from the initial pulse by a time $t = d/v$, v being the velocity of the particular molecule being counted and d being the distance from the slotted disk and the detector. This "time of flight spectrum", i. e. the intensity of counts as a function of flight time, can be inverted to obtain a velocity distribution, since the flight distance, d ,**

is known. See eq 1 where $I(t)$ and $I(v)$ are intensities as a function of time and velocity respectively.

$$\text{eq 1) } \left| \dot{I}(v)dv \right| = \left| I(t)dt \right| \Rightarrow I(v) = I(t) \left| \frac{dt}{dv} \right| \Rightarrow I(v=\frac{d}{t}) = \frac{I(t)t^2}{d}$$

Notice that the resolution for this method of velocity measurement is poorest for highest velocities i. e. $dv = \frac{v^2}{d} dt$

The slotted disk, shown in figure 1 as part 20 and called the "time-of-flight" wheel (TOF wheel) is mounted below the detector so that the slots pass directly in front of the detector opening. (The wheel is spun by an AC synchronous hysteresis motor.) It is 14 cm in diameter and the diameter to the center of the detector opening is 13.2 cm. The perfect TOF wheel would let through an infinitely narrow (both time and therefore special) burst of particles for maximum possible time, and therefore velocity, resolution. It would also have an unusable zero transmission, defined as percentage open time. If the fractional time resolution is defined as the open time for each slot (being equal to the uncertainty in starting time for each burst of particles) divided by the period between slots (being the total time scale of the TOF spectrum), it is obvious that the fractional time resolution is equal to the transmission. It can also be seen that the absolute resolution is given by the fraction transmission times the wheel period divided by the number of slots. It is also given by wheel period times slot width divided by total wheel circumference. If the fractional resolution is made better by decreasing the slot size then the transmission will suffer; if the transmission is then made larger by increasing the number of slots then another problem, wrap-around, will occur. Wrap-around occurs when the burst of particles

from one slot comes before all the particles from the previous burst have gone the flight distance and been counted by the detector, thus falsely showing up as very high velocity particles rather than their true slow velocity. The actual design of the wheel was chosen as follows. First the gate function, the transmission of the wheel as a function of time, actually depends on the size of the hole in the detector as well as the slot size of the wheel. In fact the convolution of the square slot passing in front of the square detector hole produces a trapezoidal gate function. In the case of equal sizes, the design chosen was both equal to 0.38 cm, the gate function is a triangle. If the wheel slot size is made much smaller than this the transmission (for a given wrap-around condition) will suffer while the fractional resolution will now, due to the convolution with the hole in the detector, not get much better. (Example: if the slot is halved in size the transmission is halved but the gate function total width is only decreased by 25%). If the slot is made much larger the resolution will suffer, so equal sizes were chosen as a compromise. The number of slots was governed by the maximum RPM of the motor and the requirements of wrap-around problems. The maximum motor RPM was about 12500 because above that vibration of the motor was transmitted through the lid to the ion pumps causing pressure bursts and instabilities and increasing the background count rate. Thus at maximum RPM, and therefore maximum absolute resolution, the wheel period was 4.8 msec. In order to give a TOF spectrum length of 1.2 msec, very adequate protection against wrap-around even for the heaviest and therefore slowest molecules such as tin iodide, a wheel with four slots was chosen, each slot being the previously mentioned 0.38 cm wide. Another good choice would have been six slots giving a minimum TOF spectrum length of 0.8 msec, but this would have necessitated slowing the wheel down and decreasing the absolute resolution when wraparound became a problem. (The wheel must have an even number of slots for reason of balance and for reasons pertaining

to the interface to the computer.) From the above design it can be seen that the wheel has a transmission of 3.7% and a minimum gate function of 0.44 μ secs FWHM. (For a velocity of 5×10^4 cm/sec this is a resolution of $(5 \times 10^4)^2 (4.4 \times 10^{-6}) / 19.3 = 5.7 \times 10^2$ cm/sec, which is using the previously given equation $dv = V^2 dt/d$.

Not yet mentioned in this discussion is the flight distance d , which is the distance from the TOF wheel to the center of the ionizing region of the ionizer. There is an uncertainty in the flight distance given by $\pm 1/2$ the length of the ionizing region. For the barium experiments the average d and uncertainty are 19.3 and ± 2 cm and for the tin experiments 18.85 and ± 0.250 cm owing to the different ionizers used³. The gate function (uncertainty in flight time due to uncertainty in starting time) and the length of the ionizing region (uncertainty in flight distance) cause the measured TOF spectrum to be broader than the true spectrum. In fact the measured spectrum is a complicated two dimensional convolution of the true spectrum with the gate function and the ionizer length. A program to deconvolute the measured to obtain the true spectrum has been written by A. Freedman and is discussed in his thesis³.

A TOF parameter of lesser importance is the delay time from ionizing a particle and detecting it as a pulse from the photomultiplier, basically the time it takes an ion to drift through the quadrupole at some given ion energy V (measured as the difference in potential between the ionizer grid and the quadrupole or ground). Since the actual fields involved are not known, the delay time can be given approximately by the equation $t(m) = k \sqrt{M/V}$ ($Mv^2 \propto V$, $t \propto 1/v$, so $t \propto \sqrt{M/V}$) where M is the ion mass (gm molecular weight) and V is the ion energy (volts). If the TOF spectrum of species

of two different mass, yet assumed same velocity distributions, are taken, then the constant k can be determined:

$$t(M_2) - t(M_1) = k \{ \sqrt{M_2/V} - \sqrt{M_1/V} \} = \Delta t \quad \text{so}$$

$k = \Delta t \{ \sqrt{M_2/V} - \sqrt{M_1/V} \}^{-1}$ where Δt is the difference in time between the peaks of the two TOF distributions. For example the TOF spectrums of SF^+ and SF_5^+ , both from fragmentation of SF_6 in the ionizer, were measured and the peaks differed by 15 μ secs. Therefore $k = 15 \{ \sqrt{127/23} - \sqrt{51/23} \}$ or $k = 2 \times 10^{-5} \text{ sec } V^{1/2} \text{ gm}^{-1/2}$ valid for the tin experiments (done on the ENI ionizer and quadrupole.) For the barium experiments $k = 1 \times 10^{-5}$. For a typical ion energy of 60V and mass of 153 (BaO) this would give an ion flight time of 16 μ secs a small correction.

D. Cross Correlation TOF Spectrometry

Another TOF technique, not used here due to lack of time for conversion, is cross correlation TOF spectrometry, with the added advantage that the transmission is always 50% regardless of the resolution^{7,8}. (50% transmission, compared to 3.7%, would require only 7% of the counting time.)

This method would also involve a slotted disk spun in front of the detector opening, as before, except that the disk would be divided into 127 equally spaced regions and the presence of a slot or a time would be given by a pseudo-random maximal length binary sequence. These sequences of binary numbers are finite in length, $N=2^n - 1$, where n is an integer, and possess many of the properties of random white noise. The auto correlation function for each sequence, over the length of the sequence, approximates a delta function ie is equal to -1 for all time shifts other than zero for which it is equal to N (discrete sum rather than integral. The autocorrelation of true random noise is a delta function.) Although the output of the detector will now look like random noise, the time of flight spectrum can be extracted

by cross correlating the output signal with the pseudorandom gating function.

Suppose we have a constant incoming signal of K gated by a gate function

$A_n = 1/2 (G_n + 1)$ where the subscripts denote discrete steps in time equal to the time for one time or slot to pass the detector hole and also equal to the bin time of the measuring system (discussed below).

G_n is the pseudorandom sequence taking on values of $+1$ or -1 ;
therefore A_n , the gate function, takes on values of $+1$ or 0 (open or closed).

The signal measured by the detector will be the convolution of the gate function and the response function of the particles i.e. their TOF spectrum h_n .

$$S_n = K \sum_{m=1}^N 1/2(G_{n-m} + 1)h_m \text{ which is to say that the signal at some time } n$$

is equal to the gate function at time n times the response function for zero time (infinite velocity) plus the gate function for previous times multiplied by the response function for non-zero flight time. Now to extract the TOF spectrum, h_m , we cross correlate S_n with G_n :

$$\begin{aligned} R_{SG} &= \sum_{n=1}^N S_{n+p} G_n = \frac{K}{2} \sum_{n=1}^N \left[\sum_{m=1}^N (G_{n-m+p} + 1)h_m \right] G_n \\ &= \frac{K}{2} \sum_{m=1}^N \left[\sum_{n=1}^N (G_{n-m+p} + 1) G_n \right] h_m \end{aligned}$$

$$\text{but } \sum_{n=1}^N G_{a-b} G_a = N \text{ if } b=0, = -1 \text{ if } b \neq 0, \text{ and } \sum_{n=1}^N G_n = +1$$

$$\text{so } R_{SG} = \frac{K}{2} \sum_{m=1}^N [(N+1) \delta_{m,p} - 1 + 1] h_m$$

$$R_{SG} = \frac{K}{2} (N+1) \sum_{m=1}^N \delta_{m,p} h_m = \frac{K}{2} (N+1) h_p$$

so TOF spectrum $hp = 2R_{SG}/K(N+1)$

The advantage in cross correlation TOF spectrometry are mainly the high transmission, 50%, and the variable resolution set by the motor RPM (and theroretically arbitrarily small if it weren't for the finite size of the detector opening and finite ionizer length.) Another advantage can be seen by example when the cross correlation of the pseudorandom sequence and true random noise is taken and found to be proportional to the square root of the sequence length, N . As seen above the cross correlation of a true signal is proportional to the sequence length + 1. Thus there is an improvement in signal to noise ratio, proportional to N . A disadvantage, also seen by example, is that statistical noise from a large peak in the TOF spectrum will be spread out by the cross correlation to all regions of the spectrum and possibly mask small ancillary peaks or at least make their signal to noise ratio worse. Thus this technique is most attractive when colorless TOF spectrum are anticipated, and least attractive when small peaks of interest are dominated by large peaks. Furthermore there is the problem of the design of the wheel. If the slot size is taken as 0.38 cm for reasons given before, then the wheel must have 127 "slots" since this number must be equal to $(2^n - 1) \cdot m$ where n and m are integers. If a sequence length of 127 ($n=7$) is chosen then only one sequence per circumference is allowed ($m=1$), and this would mean that, at the maximum RPM of 12,500, the TOF spectrum would be taken out to an unrealistically long 4.8 msec. If a sequence length of 31 ($n=5$) is chosen then four sequences can be included ($m=4$, No. slots=124) and the TOF period is 1.2 msec, but the added signal to noise improvement of longer sequences is lost. A solution to this would be to remount the motor to one side of the detector and use a larger diameter wheel. (The current 14 cm dia. wheel is mounted below the detector

and restricts angular movement between 0° and 90° . Mounting to the side would allow a 28 cm dia wheel and would restrict angular movement to one half the total range so that, if the wheel could be mounted on either side, the full range of angles could be reached by doing each experiment in two sections.) A good compromise design would be a 28 cm dia wheel with two sequences of length 63 giving a minimum TOF spectrum length of 12 msec and a minimum resolution twice as good as before (21 μ secs).

E. TOF Computer Interface

The time of flight (TOF) spectrum could be displayed on an oscilloscope; the horizontal sweep would be synchronized with the pulses from the slotted wheel, the horizontal axis would be the flight time t , and the vertical axis would be the intensity (signal from the detector). This will not work at all for scattered species because the signal is buried in too much noise. Some sort of signal averaging is needed, where many bursts of molecules are let into the detector and the time of flight spectrums from each burst are averaged together. This is best done by using a computer. The flight time is compartmentalized into "bins" where each bin has a width of Δt and the n^{th} bin corresponds to a flight time of $n \cdot \Delta t$. In reality these bins are locations in the computer memory. Counters count the signal occurring during the first bin's time $t = 0$ to Δt , and add these counts to those already in the first bin; then they reset and count the signal occurring in the next Δt segment in time and add these to the contents of the next bin etc.; whenever the wheel opens up to let in another burst, the counters go back to the first bin and start over again. After a while the TOF spectrum builds up in the computer memory, where consecutive locations in memory hold data from consecutive flight times.

A block diagram of the time of flight system used by our group is shown in figure 4. (The interface is built with standard TTL circuitry and detailed schematics are shown in figures 5 to 13.) The computer used is Digital Equipment Corp.'s PDP-8/e. It is operated in the Direct Memory Access (DMA) mode using the one cycle data break for collection of data. Thus the collection of data is totally transparent to the program being run on the computer, which is usually a display routine that shows the developing TOF spectrum on an oscilloscope. A programmed interrupt system is used to transfer commands to the interface and to receive status signals from it.

As shown in the diagram, the experiment consists of two beams that are crossed to form a scattered signal which is passed through the rotating slotted disk to form short bursts of molecules. (A light bulb-photo diode assembly is mounted 180° away from the detector opening so as to generate an electrical signal that tells when the wheel is "open". A discriminator-line driver circuit shapes this pulse and sends it via coax to the computer as the "FAST WHEEL SIGNAL" (positive going pulses).) The molecules traverse the flight distance, d , and are ionized by the ionizer, selected as to mass by the quadrupole mass filter, and accelerated to strike the aluminum cathode by the exit lens. The Al cathode then emits a shower of electrons which, when they hit the scintillator, create a burst of photons, which is detected as a pulse by the photomultiplier tube. (Each of these pulses corresponds to a single ion from the ionizer.) The pulses are detected by the PM discriminator-line driver circuit and sent via coax to the computer as "DATA PULSES" (negative going pulses.). After the slotted wheel lets in a burst of molecules the fast wheel signal pulse resets the DMA address counters to zero. The contents of these counters specify, partially, what memory address in the computer will receive the data being transferred. (The rest of this address is given by the field control which sets bits 0 to 3 and is discussed later. A total of 512 locations are reserved for single precision data;

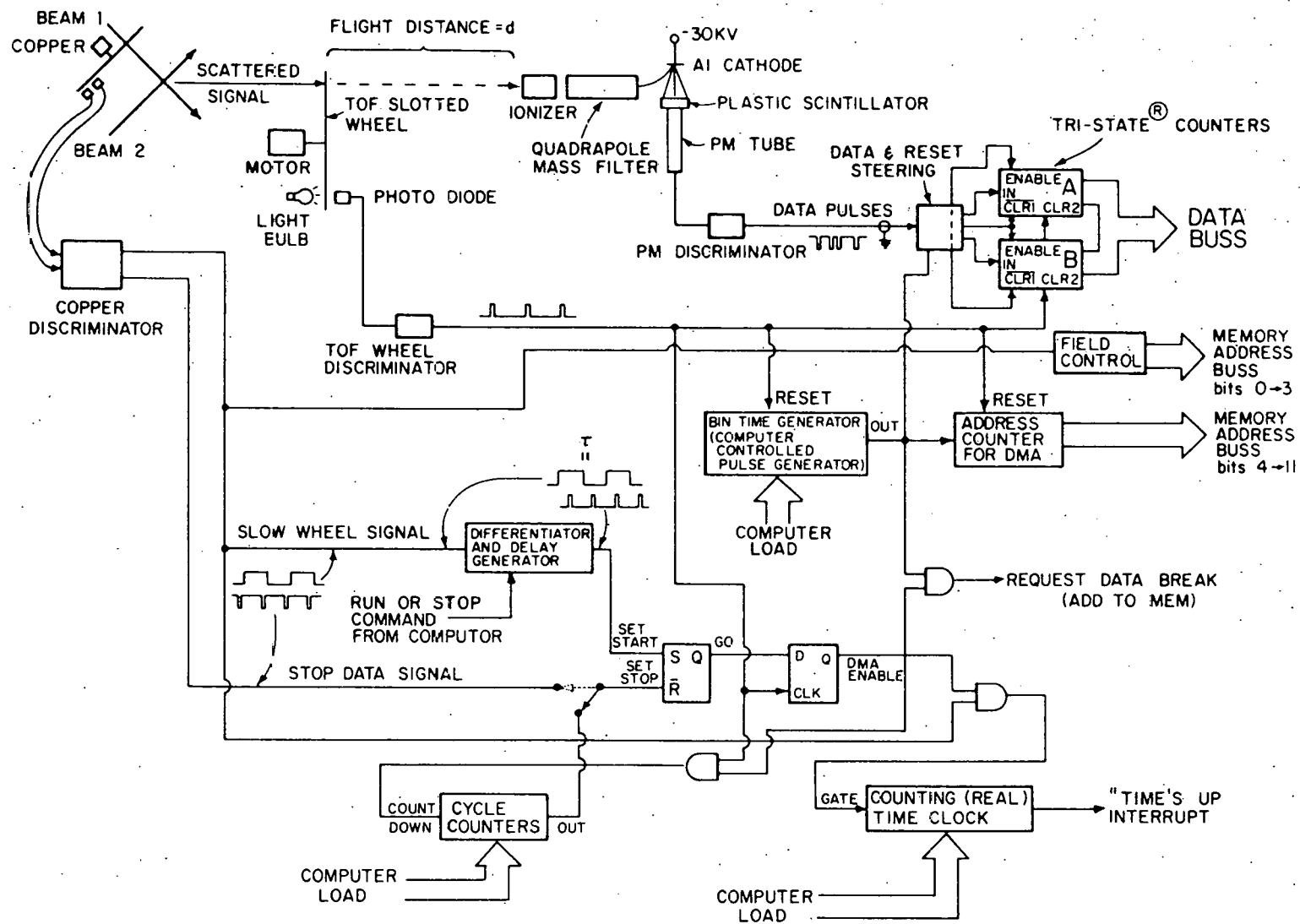


Fig. 4

256 for signal plus noise and 256 for noise.) The fast wheel signal pulse also reset plus noise and 256 for noise.) The fast wheel signal pulse also resets the data counters A and B and the bin time generator (BTGO which is a computer setable pulse generator. (The operator can load from the teletype keyboard whatever "bin time", Δt , discussed before, that he wants the BTG to generate.) After the fast wheel pulse the data gate sends the data pulses to data counter A. (These are Schottky TTL counters and can count up to 100 MHz although the PM tube in use can't get much above 5 MHz). After a time Δt , the BTG generates a pulse that initiates a number of things. First the data pulses from the PM disc are routed to the B counter (identical to A counter). Also the address counter is incremented by one (address now equal to 2001_8 if beam on, 2401_8 if beam off), and a data break is requested of the DMA circuit. When the DMA circuit grants this request the contents of the address counter (and field control) are gated onto the memory address lines and the contents held in data counter A, those counts that occurred in the first Δt of the flight time, are added to the contents of memory location specified by the address buss. When this transfer is over, the A counter is cleared. Thus those counts destined for "bin number one" are loaded into the computer while the B counter is accumulating the counts to be added to bin number two. When the nex BTG pulse comes, the process repeats except that the contents of the B counter will be added to bin number two (location 2002_8 or 2402_8) while the A counter accumulates bin number three's counts. This continues until the next fast wheel pulse*, when the address is again set to zero and the cycle repeats. (note that this process of counting has zero dead time since one counter is accumulating data while the other is sending data to the computer.)

* If the address counters go through all 256 locations before the next fast wheel pulse, they stop the data taking until the next fast wheel pulse comes.

From the above description it can be seen that the TOF spectrum will reside in the computer memory with the contents of each location being the intensity and the address of each location being directly related to the flight time t .

The majority of remaining circuitry of the interface is concerned with controlling the above data collection process i. e. telling the computer when to start and stop taking data. This is done synchronously with the "slow wheel signal", named for the beam chopping wheel used in the old configuration of the machine. As mentioned before, in the new configuration, the nozzle beam, beam 1, is chopped by a flag type chopper and the electronics that drives this relay chopper also generates the slow wheel signal, a 50 ohm TTL level signal that is 2.5V when the beam is open and OV when its blocked. The sequence of events is as follows: the operator types an "R" for run and the computer sends a RUN command to the interface. The next positive going transition of the slow wheel signal (i.e. beam turns on) is then passed through the differentiator (which generates a short positive going pulse for every transition of the slow wheel signal, either positive or negative) and then is delayed by a time τ (internally set in the computer by a 20 msec one-shot). The delay time τ is required to allow time for the beam flag to move since the slow wheel signal is coincident with the relay driving signal but the relay can't move instantaneously. (This delay time also allows time for the chopped beam to reach the collision zone and the detector as previously mentioned, thus eliminating transients in the TOF spectrum.) The delayed pulse then sets the GO flip-flop so that the next TOF wheel pulse will set the DMA ENABLE true allowing the BTG to generate data break requests and, in effect, initiating TOF data taking. Also enabled are the TOF wheel pulse CYCLE COUNTERS which, having been preset by the operator (through the keyboard) to some number N , now start to count down to zero.

When they reach zero they reset themselves to the number N and generate a narrow pulse that clears the GO flip-flop allowing the (negative going transition of the) current TOF wheel pulse to clear the DMA ENABLE flip-flop thereby halting the data breaking process and stopping the TOF spectrum. On the next transition of the slow wheel signal (which would now be the beam turning off) the process repeats itself identically except that the field control is now set such that data are being added to the beam off field. (When the slow wheel signal is low the data field control output is 0101 for address bits 0 to 3 i.e. locations 2401_8 to 2777_8 . When high, (beam on), the data field is 0100 or locations 2001_8 to 2377_8 for the beam on field.) The process continues to repeat itself until either 1) the operator types "S" for stop 2) the counting time clock times out and generates a TIME'S UP interrupt or 3) the overflow circuit (discussed later) generates interrupt. In all of these cases the computer issues a stop command to the interface and the control circuitry is set such that the next positive going transition of the slow wheel signal is not allowed to restart data breaking. Therefore the computer stops taking data after having executed the same number of beam on as beam off cycles.

The TOF wheel CYCLE COUNTERS must be preset by the operator who must calculate the value of N beforehand as follows: the beam chopper is driven so that the open (and closed) time is 200 msec, subtracting the delay time of 20 msec gives 180 msec into which N TOF wheel cycles must fit. If the TOF period (from one pulse to the next) is 1.2 msec then $N = 180/1.2 = 150$ or 135 for a safety factor. Notice that the system is constrained to count exactly equal amounts of time with the beam on as off since the CYCLE COUNTERS always allow only N cycles of the TOF period per each half cycle of the beam chopper. (The counting time scale during each TOF cycle is governed by the crystal controlled BTG and so the beam

on and beam off counting times are precisely equal to each other to within almost the limits of reasonable measurement.) This feature was not operational for some of the early studies of this group.¹³

Note that the COUNTING TIME CLOCK, a real time crystal controlled clock that can be set, as an alarm clock, to generate an interrupt after any given amount of time between 0.2 and 200000 seconds, is only enabled during beam on data counting so that the count rate is the number of counts divided by the counting time.

The DMA system is used in the add to memory mode, i. e. the 12 bit content of the data counters is added to the 12 bit content of the selected memory location in the computer. If this sum should overflow into the 12th bit the hardware has been designed to generate an interrupt telling the computer that this has occurred and that further data taking would result in non-retrievable overflow errors. On receiving this interrupt the computer clears the run flip-flop so that data taking will stop on the next slow wheel on transition. After DMA'ing has stopped, the computer goes to a software double precision add routine that takes data existing in the single precision data fields and adds them to double precision data fields (locations 3002 to 3777 for signal plus noise and locations 4002 to 4777 for noise). After this is complete the computer restarts the data taking, with the single precision data fields now empty and ready to accept more counts.

The above description is for the mode in which the "fast wheel" is installed and the computer is taking time of flight data. It is also useful to remove the "fast wheel" and take angular distribution data. The advantage to this is higher total signal (increased by a factor of the inverse of the resolution of the particular "fast wheel"), and the ability to rotate the lid through a larger range of angles. The simple way to get the computer interface to operate in an angular distribution mode is simply to fool it into thinking it's taking TOF data. "Fake" fast wheel pulses are generated by a pulse generator set by the operator, (who then also recalculates and reloads N.) The computer is set up to run exactly as in the TOF mode. The only difference is that the display and printout routines are told to show

the total sum of all the addressed bins*, which is the total signal (or signal plus noise or noise, depending on display mode) at the particular angle being looked at. (The bin number, bin time, velocity, and flux columns are ignored in this mode.)

As mentioned before, the computer system operates in both the direct memory access mode and the programmed interrupt mode. The DMA process is totally transparent to the program being run. The software written for this system consists of a background program that displays the developing TOF spectrum on an X-Y oscilloscope and a foreground program that services the interrupts generated by the interface and also transfers commands to the interface. A complete description of these programs can be found in the comment portion of the program listing. Briefly, the display routine can display either the single precision or the double precision data fields, either signal plus noise (beam on), noise (beam off), or the difference (signal). It can be told to display any number of points and can be told to average over (sum up) any number of consecutive points. This last feature is useful for very noisy signals where structure may become apparent if portions of the signal are averaged together, i. e. resolution is sacrificed for better signal to noise ratio. The display routine automatically scales both axis (to within a factor of 2). It can also inform the operator what full scale corresponds to in number of counts. The display routine is always running, even while data are being taken, so that the TOF spectrum can be seen to "grow".

The foreground program handles interrupts both from the interface and from the teletype keyboard. If the interface wishes to inform the computer

*Actually the number of addressed bins minus 2 or 3 for a safety factor since the last bin is interrupted by the TOF pulse and is not always valid.

of something, it generates an interrupt and then the computer generates on Input Output Transfer that transfers a STATUS WORD into the accumulator. The computer then "interrogates" this status word where each bit has the following meanings. Bit zero active (equal to 1) means the bin time is too small i. e. the DMA system is attempting to data break faster than the computer can handle it. The computer will type "BIG TOO FAST" and the operator must reload the bin time generator (see Table I) with a larger time and hit "R" for run again. A reasonable minimum bin time is 5 μ secs (bintime must be in whole number μ secs only) although the computer can go as fast as 3 μ secs if everything stays in sync.

Bit one active means that the prescribed counting time has elapsed. The computer stops the data taking and types out "TIME'S UP". The operator may then do one of several things. He may print out the data (see below). He may continue data taking for another prescribed counting time by pressing "R". If the present data do not look good the operator may erase them by pressing "K" and start over. Both the counting time clock and the bin time generator "remember" their values unless reloaded or unless the computer is turned off. The contents of the counting time clock can be read even while the experiment is running so the operator can see how much longer he has to wait. Also he may stop the experiment at any time (press "S") and read the CTC to see how long the experiment has been counting for. Loading the CTC entails an inaccuracy of about $\pm 3\%$, but reading the CTC gives the counting time exactly.

Bit two active signifies an overflow into the 12th bit while DMA adding to memory. The computer stops the data taking and then executes the previously described double precision add routine. After this is complete the computer restarts the data taking.

The computer can also send commands to the interface. They can be in the form of an IOT that transfers a command work to the interface. The details are shown in Table II.

The machine-operator interface consists of a teletype for input and output and the previously mentioned X-Y oscilloscope for output. The operator initiates computer operations by pressing a single key on the teletype keyboard. Some of these keys are "non printing" i. e. they transfer commands to the computer but the computer does not echo back anything so the Teletype does not print anything. Table I gives a list of all the keyboard commands. Any character other than those listed causes the computer to echo a carriage return and line-feed back to the Teletype. When loading numbers the format should be as described in "Introduction to Programming" Vol II floating point package DEC-08-YQ2B-B. (When loading the BTG only whole μ secs are loaded. Thus 13.8 will be loaded as 13 μ secs.) Note that while the experiment is running, the computer should not be in the comment mode since the interrupt system is off in this mode and a computer won't be able to process a "TIME'S UP" or "OVERFLOW" interrupt. Note also that single precision data can be added to the double precision data at any time while the data taking is in process by pressing "L" (or "W"). If the experiment is "STOPPED", however, pressing either L or W will result in the computer adding single precision to the double precision data and printing out the double precision data. Note also the discrepancies listed under "N" (NUM. BINS) and "[" (NS).

The "delay time" mentioned under "LOAD PARA" and "NS" is merely a constant that is subtracted off the accumulated bin time in the "WRITE DATA" program. For example, if the bin time is 10 μ secs and the delay time is -50 μ secs then the first bin is at 55 μ secs, the second at 65 μ secs etc. The sense of this delay time is such that positive delay

Table I.

OPERATING INSTRUCTIONS FOR THE EXPERIMENT CONTROL PROGRAM

KEY	NAME	DESCRIPTION
A	NCYCLE	LOADS TIME TOF WHEEL PULSE COUNTER. HIT A, COMPUTER WILL TYPE "#CYCLES=", TYPE AND INTEGER HIT RETURN
B	BIN TIME	LOADS THE BIN TIME GENERATOR. HIT 'B', COMPUTER WILL TYPE "BT=", THEN OPERATOR CAN TYPE IN THE VALUE HE WANTS (IN MICRO-SECONDS). SEE 'INPUT FORMAT'.
C	CONTINUE	TELLS THE COMPUTER TO CONTINUE COUNTING AFTER HAVING BEEN STOPPED. DOES NOT RESET THE COUNTING TIME CLOCK. (IF CT=100 AND EXP. STOPPED AT 70, THEN HITTING 'C' WILL CAUSE COMPUTER TO CONTINUE COUNTING FOR 30 SEC.)
E	READ CTC	CAUSES COMPUTER TO PRINT OUT THE CURRENT CONTENTS OF THE COUNTING TIME CLOCK, I. E. HOW LONG THE COMPUTER HAS BEEN COUNTING FOR.
F	FULL SCALE	CAUSES THE COMPUTER TO PRINT WHAT FULL SCALE (10 VOLTS) ON THE DISPLAY CORRESPONDS TO IN TERMS OF COUNTS. WORKS FOR ANY MODE OF THE DISPLAY ROUTINE.
K	KILL	ERASES DATA FIELD. 'RUN' DOES NOT DO THIS.
L	LOAD	CURRENTLY IDENTICAL TO 'W'.
N	NUM. BINS	LOADS THE NUMBER OF 'BINS' THAT THE COMPUTER DISPLAYS OR PRINTS OUT. HIT 'N', COMPUTER WILL NOT TYPE ANYTHING, OPERATOR CAN THEN TYPE IN WHAT HE WANTS. CURRENT DISCREPENCY: N IS THE NUMBER OF POINTS THAT THE DISPLAY WILL SHOW BUT THE WRITE OUT ROUTINE WILL ONLY PRINT UP TO BIN NUMBER N. THESE ARE NOT EQUIVALENT UNLESS 'NS' IS EQUAL TO ONE.
Q	QUEARY	CAUSES THE COMPUTER TO PRINT OUT THE STATUS OF THE EXPERIMENT: RUNNING OR STOPPED.
U	SINGLE PRE.	CAUSES THE DISPLAY ROUTINE TO LOOK AT THE SINGLE PRECISION DATA FIELD:
V	DOUBLE PRE.	CAUSES THE DISPLAY ROUTINE TO LOOK AT THE DOUBLE PRECISION DATA FIELD.
X	SIG. + NOISE	CAUSES THE DISPLAY TO LOOK AT BEAM ON DATA FIELD.
Y	NOISE	CAUSES THE DISPLAY TO LOOK AT THE BEAM OFF DATA FIELD.
Z	SIGNAL	CAUSES THE DISPLAY TO SHOW THE ARITHMETICAL DIFFERENCE OF THE BEAM ON AND BEAM OFF DATA FIELDS.
R	RUN	RESETS THE COUNTING TIME CLOCK AND THEN STARTS COUNTING.
S	STOP	STOPS THE EXPERIMENT.
T	COUNT TIME	LOADS THE COUNTING TIME. HIT 'T', COMPUTER WILL TYPE "CT=", THEN OPERATOR CAN ENTER THE AMOUNT OF TIME HE WANTS THE EXPERIMENT TO COUNT FOR. THIS TIME MUST BE BETWEEN 2049 AND 204799 SECONDS. SECONDS IS THE ONLY UNIT TO BE USED. THIS TIME IS ACCURATE TO ABOUT + OR - 3%. THE TIME READ WHEN 'E' IS HIT IS, HOWEVER, EXACT.
W	WRITE DATA	CAUSES COMPUTER TO PRINT OUT THE CONTENTS OF THE DATA FIELD IN THE FOLLOWING FORMAT. BIN NUMBER, TIME IN MICRO-SECONDS THAT THIS BIN CORRESPONDS TO, TOTAL SIGNAL PLUS NOISE (BEAM ON DATA FIELD), TOTAL NOISE (BEAM OFF DATA FIELD), SIGNAL (BEAM ON - BEAM OFF), STATISTICAL NOISE (SQUARE ROOT(S+N + N)), VELOCITY IN CM/SEC, AND FLUX AT THIS VELOCITY.

← COMMENT ALLOWS OPERATOR TO TYPE COMMENTS. HIT '←' (SHIFT O),
COMPUTER WILL TYPE '/', TYPE ANYTHING YOU WANT. RETURN
WILL CAUSE CARRIAGE RETURN LINE FEED AND WILL GET
COMPUTER OUT OF COMMENT ROUTINE. TO TYPE FURTHER
COMMENTS, OPERATOR MUST HIT '←' AGAIN.

↑ LOAD PARA. LOADS EXPERIMENTAL PARAMETERS. HIT 'N', (SHIFT N),
COMPUTER WILL DO NOTHING, TYPE IN THE DELAY TIME IN
MICRO-SECONDS (PLUS OR MINUS), TYPE SPACE, TYPE IN THE FLUX
FUDGE FACTOR, (USUALLY 10000), TYPE RETURN. (WHERE
RETURN AND SPACE ARE THE SINGLE KEYS ON THE KEYBOARD.)

[NS TELLS COMPUTER HOW MANY BINS YOU WANT IT TO SUM UP,
(AVERAGE OVER). HIT '[', (SHIFT K), COMPUTER WILL PRINT
'NS= ', TYPE IN AN INTERGER, TYPE RETURN.
EXAMPLE: IF NS=3, THEN COMPUTER WILL, (WHEN TOLD), TYPE
THE SUM OF THE CONTENTS OF BINS 1, 2, AND 3 AND CALL IT
BIN NUMBER 3. CONTINUING, IT WILL SUM BINS 4, 5, AND 6 AND
CALL IT BIN NUMBER 6. ETC. CURRENT DISCREPENCY.
THE PROPER AVERAGE TIME FOR THE SUM OF THE FIRST 3
BINS WOULD BE HALF WAY THROUGH THE SECOND ONE.
I. E. IF THE BIN TIME WERE 10 μ SECS, (AND THE DELAY
TIME ZERO), THIS PROGRAM TYPES OUT THE TIME HALF-WAY
THROUGH THE LAST BIN OF THE SUM, I. E. IN THIS EXAMPLE
25 μ SECS.

Table II

OP-CODE	INSTRUCTION
6170	LOAD COMMAND WORK
6171	SEND STATUS WORD TO ACC
6172	SEND COUNTING TIME CLOCK TO ACC
6173	LOAD COUNTING TIME CLOCK (CTC)
6174	LOAD BIN TIME GENERATOR
6175	LOAD MODE

DATA BIT	COMMAND	DATA BIT	STATUS
0		0	BIN TIME TOO FAST FLAG
1		1	TIME'S UP FLAG
2	CLEAR TIME'S UP FLAG	2	OVERFLOW FLAG
3	CLEAR OVERFLOW FLAG	3	
4	RESET CTC	4	
5	SET RUN	5	
6	SET STOP	6	
7	CLEAR BIN TIME TOO FAST	7	
8		8	
9		9	
10		10	
11		11	"GO" FLAG (1 = running, 0 = stopped)

corresponds to the detected ions somehow being delayed i. e. by the transit time through the quadrapole. Negative delay corresponds to the fast wheel pulse being "late" as in a mispositioned photo diode assembly or an externally applied delay of the fast wheel signal for purposes of "throwing away" the beginning portions of a TOF spectra. The latter process concerns the use of a delay generator between the fast wheel signal line driver and the computer.

The problem mentioned under "NS", i. e. that the printed accumulated bin time for an average over NS bins is not the proper average time, may be corrected by temporarily loading a new delay time = old delay time + NS x BIN TIME / 2. Example: if delay time setting is - 20, NS = 4, and BIN TIME = 20 μ secs then temporary new delay time, for this NS, is +20 μ secs.

In the WRITE DATA routine the last column is called the FLUX. This is a crude inversion of the time of flight data to give a velocity distribution. As shown in Eq. 1) the "FLUX" is equal to the measured intensity as a function of time multiplied by a Jacobian of t^2/d , which is also equal to d/v^2 . Our detector, however, is not a flux detector; it is a number density detector because of the following. Normally the intensity would be given by the number of counts per sec per sq cm of detector area, a flux. However, the ionization probability, at low ionization efficiency, is inversely proportional to particle velocity because those molecules going faster spend less time in the detector and therefore have less probability of being ionized. Thus we measure flux x $1/v$ or number density, No. /cm³. The last column, called FLUX, therefore prints out F x intensity x d/v , instead of divided by v^2 . F is a fudge factor, normally 10^4 , to make the flux come out to reasonable numbers.

All of the above program fits into the 4K memory space of the computer with only a few scattered locations left. Even minor additions would require purchasing more memory. Such additions could include a program to punch data onto paper tape in a format that could be read by the Computation Center's computer, or possibly a program to time normalize angular distributions and plot them out on an X-Y recorder. Finally the existing hardware would be usable, without modification, with a pseudo random TOF technique but the program would have to be expanded to include the cross correlation (actually a relatively simple procedure).

Other TOF systems, described in references 9, 10 and 11, use interfaces similar to the one presented here.

APPENDIX I: Nozzle Geometry.

The geometry of the nozzle beam source, as modified for these experiments, is as follows. The nozzle has an orifice 0.02 cm in diameter and can be moved back and forth with respect to the skimmer to maximize the beam intensity.... the normal operating distance was 1.35 cm. The conical shaped skimmer has an orifice of 0.10 cm, is 2.76 cm from the center of the collision zone (CZ), and "empties" directly into the main chamber. The beam chopper is directly in front of the downstream side of the skimmer, about 1 cm from the CZ center. This geometry was calculated so as to not produce a viewing factor if the skimmer is assumed to act as a collimator. The viewing factor is simply the fraction of the total collision region that the detector can see as a function of the laboratory angle. The slit geometry of the detector (0.38 cm at entrance 4.44 cm from CZ and 0.3 cm at the ionizer entrance 18.44 cm from CZ) gives an umbra that subtends 0.4 cm at the CZ. If the collision zone, which is trapezoidal in shape, is larger than 0.4 cm then the detector will be

looking at varying volumes of reaction region as the lid is rotated causing an experimental bias in angular distribution measurements for certain angular regions over others. The easiest way to avoid this is to make sure that the largest dimension of the collision zone is less than 0.4 cm so that the detector is always viewing the whole collision zone (plus certain areas around it) at all angles.

If the skimmer was acting as a collimator, i.e. "casting" the standard umbra and penumbra - "shadows", then there would be no viewing factor. However if the nozzle expansion has not frozen by the time it reaches the skimmer (i. e. the particles are still interacting with and deflecting each other) then the skimmer might act as a source giving a more divergent beam and possibly leading to a viewing factor. In fact, measurements made on the beam in this configuration, implied that this might partially be the case. Because of the geometry of the detector and because the detector rotates about the collision zone center and not about the beam radiant point the detection system can't see the beam beyond about $\pm 0.4^\circ$ (assuming the skimmer as the source). Therefore another method of measuring the beam profile was used: the background pressure in detector chamber one, which should be proportional to the amount of beam entering the detector, was monitored as a function of the lid angle which could be back calculated to the detector opening distance from the beam centerline. With the nozzle at 1.35 cm from the skimmer, and if the skimmer is assumed to collimate, then the beam at the detector opening should have an umbra of 1.07 and penumbra of 1.23 cm. Approximating this as a rectangle of width 1.15 cm and convoluting with the detector opening of 0.38 cm we get an expected FWHM of 1.53 cm. The measured distribution, with 150 torr in the nozzle, had a FWHM at the detector of 1.6 cm but contained a weak tail that went out to 2.7 cm (FW1/10M), so that the skimmer might be partially operating as a source.

In order to assure freedom from a viewing factor a beam collimator/ deflector was spot welded to the face of the collimating chamber directly in front of the beam chopper. It consisted of a 0.5 cm wide horizontal strip of metal with a centrally located 1 cm diameter deflector containing a 0.23 cm wide by 0.38 cm high slot for the beam to pass through. The top and bottom of the deflector portion were angled away from the beam to deflect it above and below the region of the collision zone. For a proof of the absence of a viewing factor with this collimator installed, refer to figure 14 to which the following equations apply:

$$d_1 = 8.47(.1125/6.25) - .0125 = .140 \text{ cm}$$

$$d_2 = (2.76 + .115)(.12/1.81) - .05 = .140 \text{ cm}$$

Now we approximate the CZ by the dotted rectangle 0.28 x 0.28 cm, an overestimate, and calculate its longest dimension, the dotted line, to be .39 cm which is less than the 0.4 cm that the detector sees. Therefore, according to this calculation, there is no viewing factor. This was confirmed when the beam and detector geometries for both the Ba and Sn experiments were fed into a computer program, written by C. A. Mims¹², that does a detailed calculation of the viewing factor taking all considerations into account. The factor was equal to 1.0 to within 0.002 for all angles. This was also confirmed experimentally by a repeat of the aforementioned beam profile measurement method.

APPENDIX II Duty Cycle

Let S and B be the signal and background count rates respectively and let T on and T off be the time counted with the beam on and off. Then the absolute counts accumulated in the on channel is N on = T on (S+B) and off channel N off = T off (B). Now the desired quantity is the total signal counts N_s given by the equation

$$N_s = N \text{ on } (T \text{ on} + T \text{ off})/2 T \text{ on} - N \text{ off } (T \text{ on} + T \text{ off})/2 T \text{ off}$$

$$(N_s = ((S+B)-B)(T_{on} + T_{off})/2 = S(T_{on} + T_{off})/2)$$

Now we make the condition that statistical noise in each of the two terms of N_s be equal so as to not put too much influence on one or the other. The statistical noise of the first term is $\sqrt{N_{on}} (T_{on} + T_{off})/2 T_{on}$ since N_{on} is an absolute number of counts with, therefore, an error of $\sqrt{N_{on}}$, and $(T_{on} + T_{off})/2 T_{on}$ is just a multiplying factor. Therefore we have the equation:

$$\sqrt{N_{on}} (T_{on} + T_{off})/2 T_{on} = \sqrt{N_{off}} (T_{on} + T_{off})/2 T_{off} \quad \text{or:}$$

$$\sqrt{T_{on}(S+B)}/T_{on} = \sqrt{T_{off}(B)}/T_{off} \quad \text{or:}$$

$$\sqrt{(S+B)/B} = \sqrt{T_{on}/T_{off}} \quad \text{so}$$

$$T_{on}/T_{off} = (S+B)/B \quad \text{or duty cycle } T_{on}/(T_{on} + T_{off}) = D$$

$$D = [(S+B)/B] / [(S+B)/B + 1] = (S+B)/(S+2B)$$

APPENDIX III Circuit Diagrams

The detailed circuit diagrams for the TOF computer interface presented here are shown in figures 5 through 13. Only a brief description of each figure will be given. The circuits are built using mostly standard TTL circuits assembled on wire-wrap boards that plug into the PDP8/e Unibus. Not shown are the two D to A circuits that control the X-Y oscilloscope.

Fig. 5: Experiment control board including the GO flip flop (E30), the differentiator (E38), the delay generator (E26), the DMA ENABLE (E21), the DMA ADDRESS counters (E41 and E20) and the BTG too fast interrupt flag (E29). The differentiated slow wheel signal is gated into the delay generator by the GO flip-flop. The RUN command (data G valid) is loaded into E22 by IOTO and enables GO on the rising edge of SLOW WHEEL signal. The delayed signal (E26 pin 6 and 12) is passed to E18/E22 which is a glorified R/S flip flop. The rn signal enables (E21 pin 8) the DMA

enable (E21 pin 5) on the next falling edge of the TOF wheel pulse (E17 pin 10 from E39-6). This gates the BTG pulses (HA1-18) into the address counters (E41, E20). If the address counters time out before the next TOF pulse (256 cycles) the data breaking is stopped via the action of E21 pin 2 on the clear (pin 10). The BREAK REQUEST flag (E29) initiates DMA request via the action of the BTG (2, E25 pin 6). If another such request should come before the previous one is granted (CLR REQUEST) the BTG TOO FAST flag (E29 pin 3) will be set (on the coincidence of BRK REQUEST and BTG). The steering logic (E18 pins 5 and 6) is toggled by BTG.

Fig. 6: CYCLE COUNTERS. The cycle counters are loaded via IOTS. Upon being enable by the delayed slow wheel signal (E18 pin 3) the leading edges of TOF pulses start decrementing the cycle counters (E03, E04, E09). Upon time out the counters generate a pulse at E09 pin 13 which is passed through a one shot (E28) to generate the stop data pulse which disables the DMA ENABLE flip flop on the falling edge of the present TOF pulse (fig 6 E21). This pulse also presets the cycle counters to the value stored in the cycle count memory (E07, E02), and the process continues as described in the text of this chapter. The detailed timing diagram of this system is shown in figure 13. (The circuit works for any case of DSW coincident with FW).

Fig. 7: DATA COUNTERS. The signal counts from the PM discriminator (TTL level, negative going) are passed via the steering logic (C7-1, C7-2) to the A or B counters (A, B, C, 5 or A, B, C, 6). The 82591 is manufactured by Signetics Corp. The steering logic also directs which counters are to be read by the computer (C3; A, B, D3; A, B, D, 4; the 74126's are TRI-STATE buffers), and also which are to be

cleared by CLEAR REQUEST (B8). The TOF pulses (C7-6), from the TOF wheel discriminator (B8 pin 9) also clear the counters.

Fig. 8: BIN TIME GENERATOR (BTG). The bin time (μ secs) is loaded by IOT4 into the 74174's. The crystal oscillator (E34) toggles the counters (E29, E25, E22) until their out matches the value held in the 74174's at which time the comparators (E28, E24, E20) bring the one shot's input up (E18 pin 2) causing a BTG pulse at (HA1-18) and resetting the counters for another cycle. (The counters are also reset by the TOF pulse (MA1-17))

Fig. 9: COUNTING TIME CLOCK (CTC). The binary mantissa of the desired count time is loaded by IOT3 from DATA lines 0 through 7 and this number is stored in 74174's and compared with the contents of counters much the same way the BTG works. In order to have a large range of count times, the CTC is designed to accept a binary exponent, in addition to the mantissa, loaded the same way from bits 7 through 11. This exponent is read by the MK5009 which is a MDS crystal oscillator and programmable divider. The MK5009 produces an out-put count at the rate of $1\text{MHz} \div 2^n$ where n is the binary exponent. These output pulses (E37 pin 1) are counted by the binary counters (E33, E21, E08) and when the output of the last two (E21, E08) is equal to the stored mantissa the comparators trigger the TIME'S up interrupt flag (74107 pin 3). The count time can be read by IOT2... notice that while the "alarm" feature triggers off only the last two counters the read feature reads all three counters and gives the exact time.

Fig. 10: DMA CONTROL and Fig. 11: IOT CONTROL see the DEC manual "Small Computer Handbook" for an explanation of these circuits.

(Fig. 10 is correct.... DEC's figure contains an error.)

Fig. 13: The circuits for the small auxillary electronic devices are self explanatory. The following power supply voltages apply to fig. 13: NES29K: pin 10 = +6V, pin 3 = -6V, pin 9 = +5V; SN 75451P pin 8 = + 5V, pin 4 = GND. Use ground plane PC construction and bypass all power supplies with .05 μ F at 20 V ceramic capacitors.

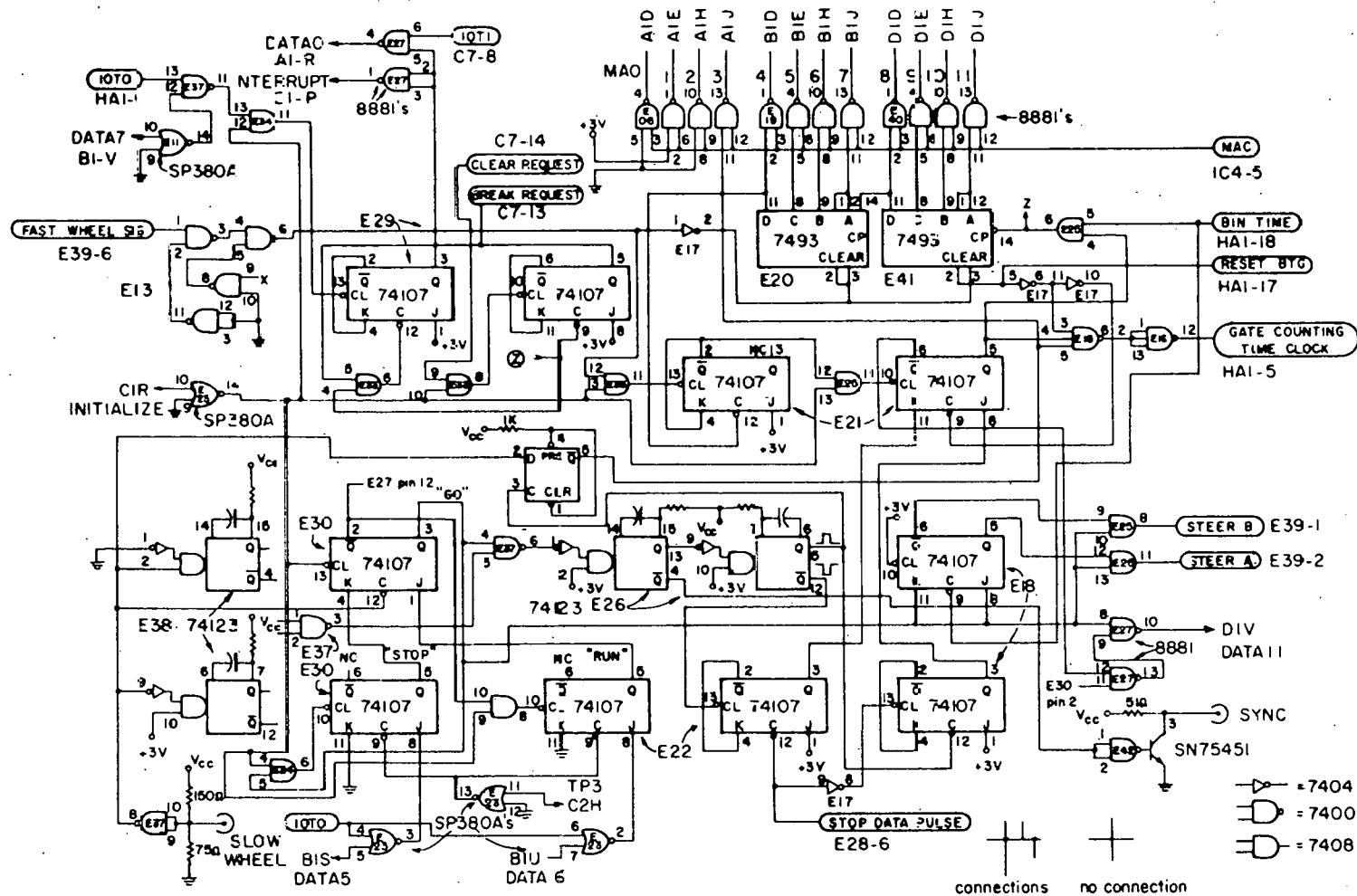


Fig. 5

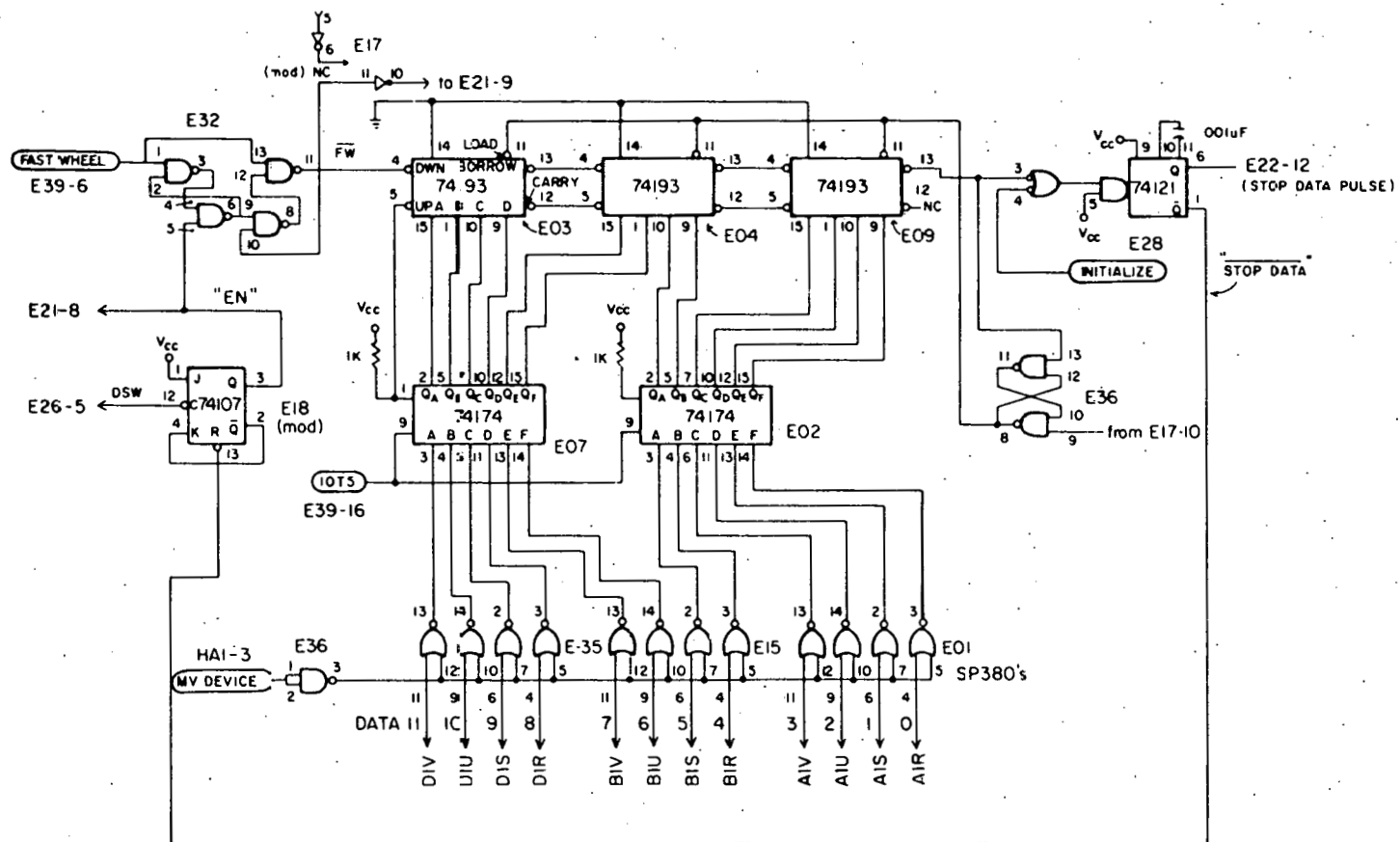


Fig. 6

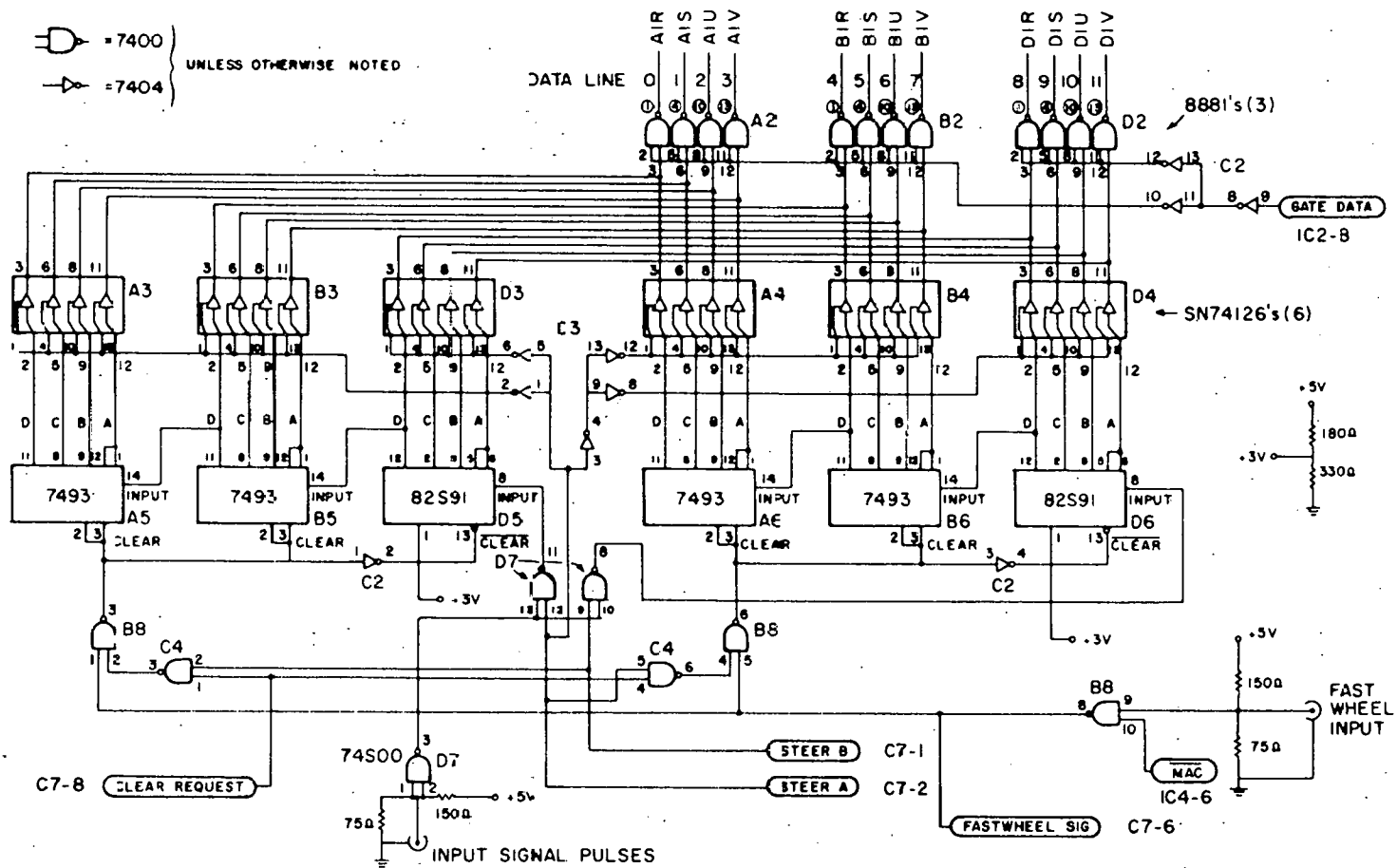


Fig. 7

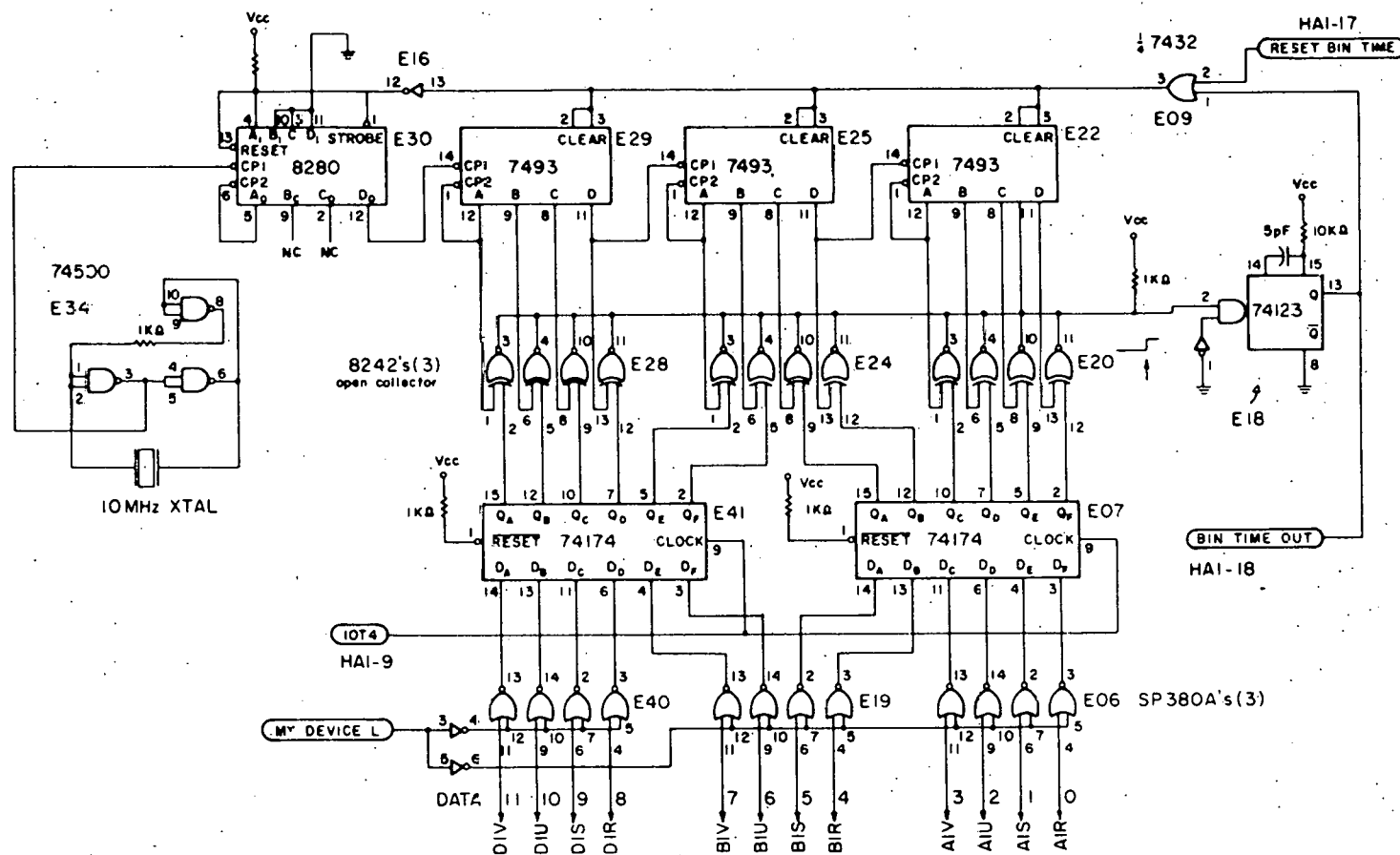


Fig. 8

(COUNTING TIME CLOCK)

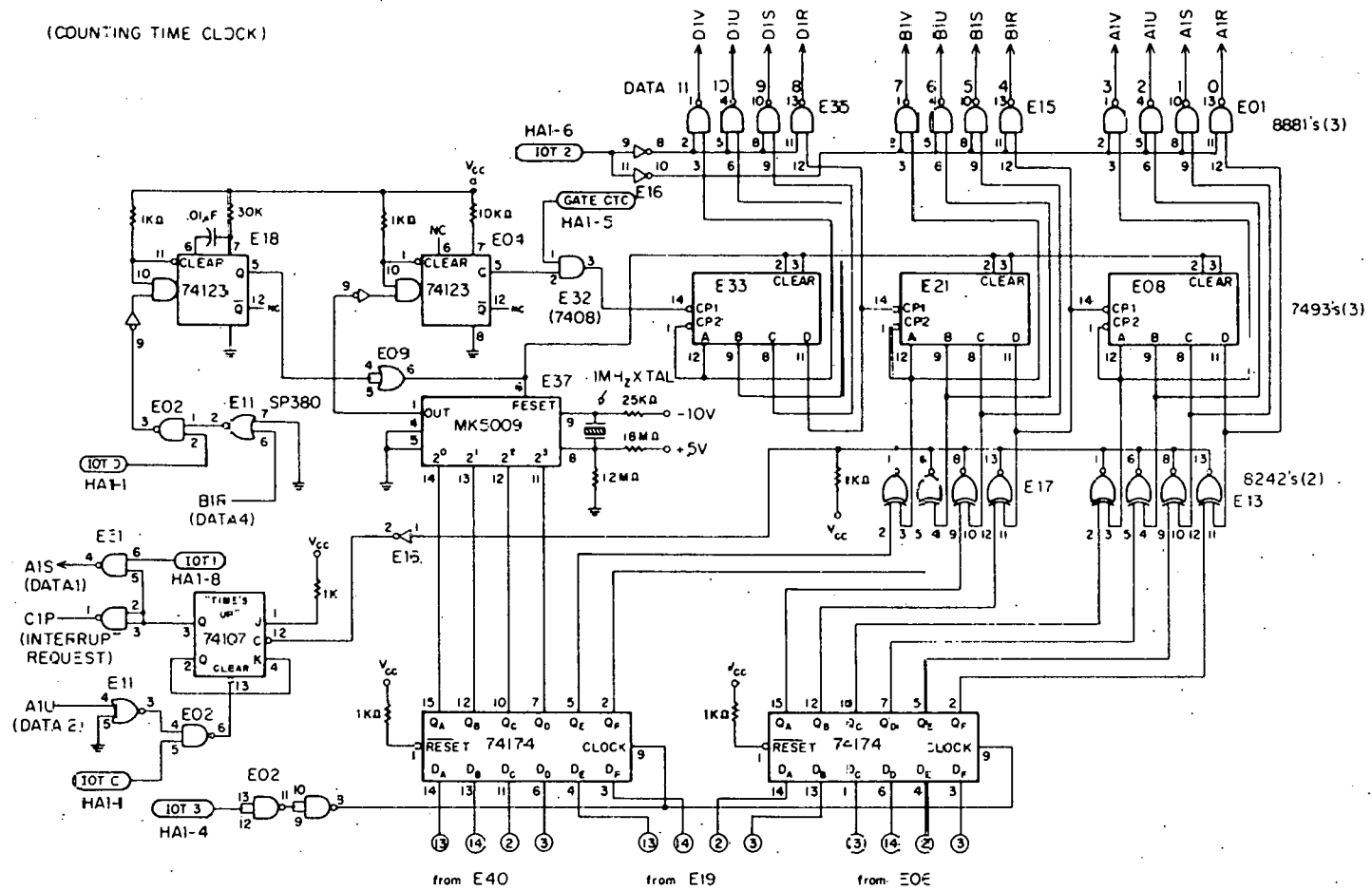


Fig. 9

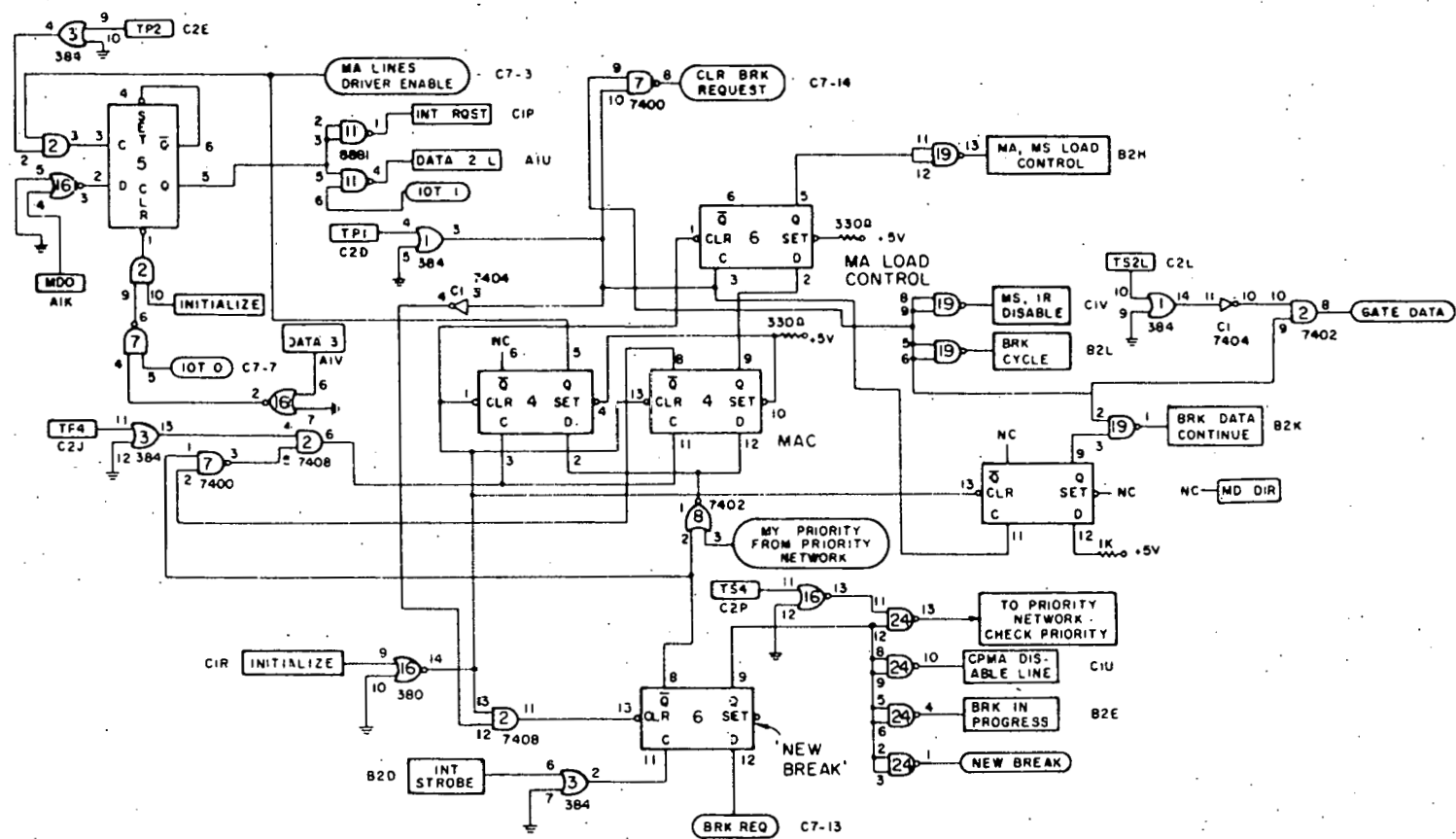
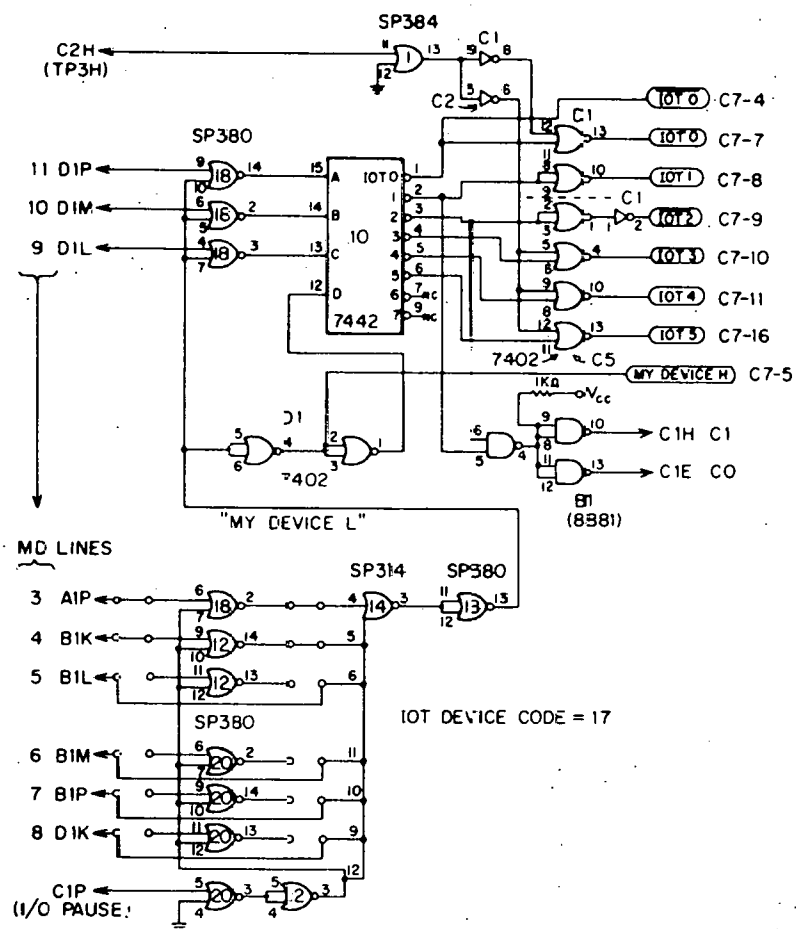


Fig. 10

IOT CONTROL



DATA BREAK PRIORITY NETWORK

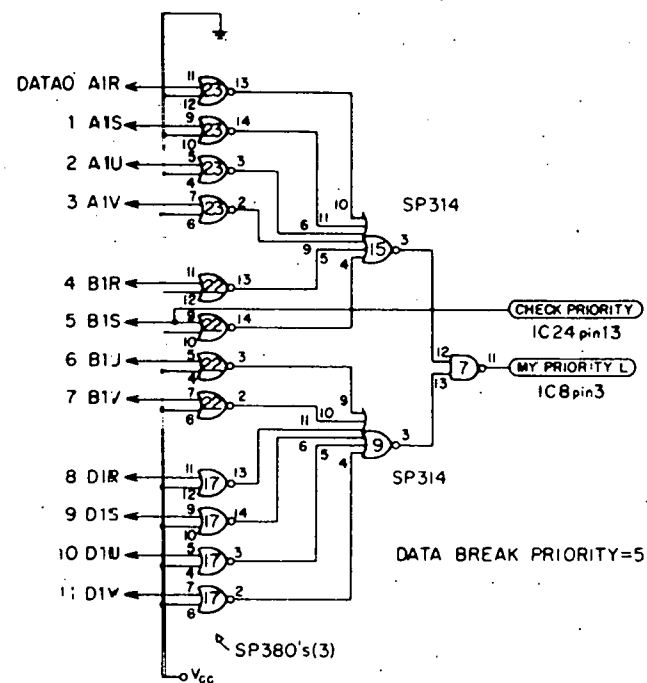
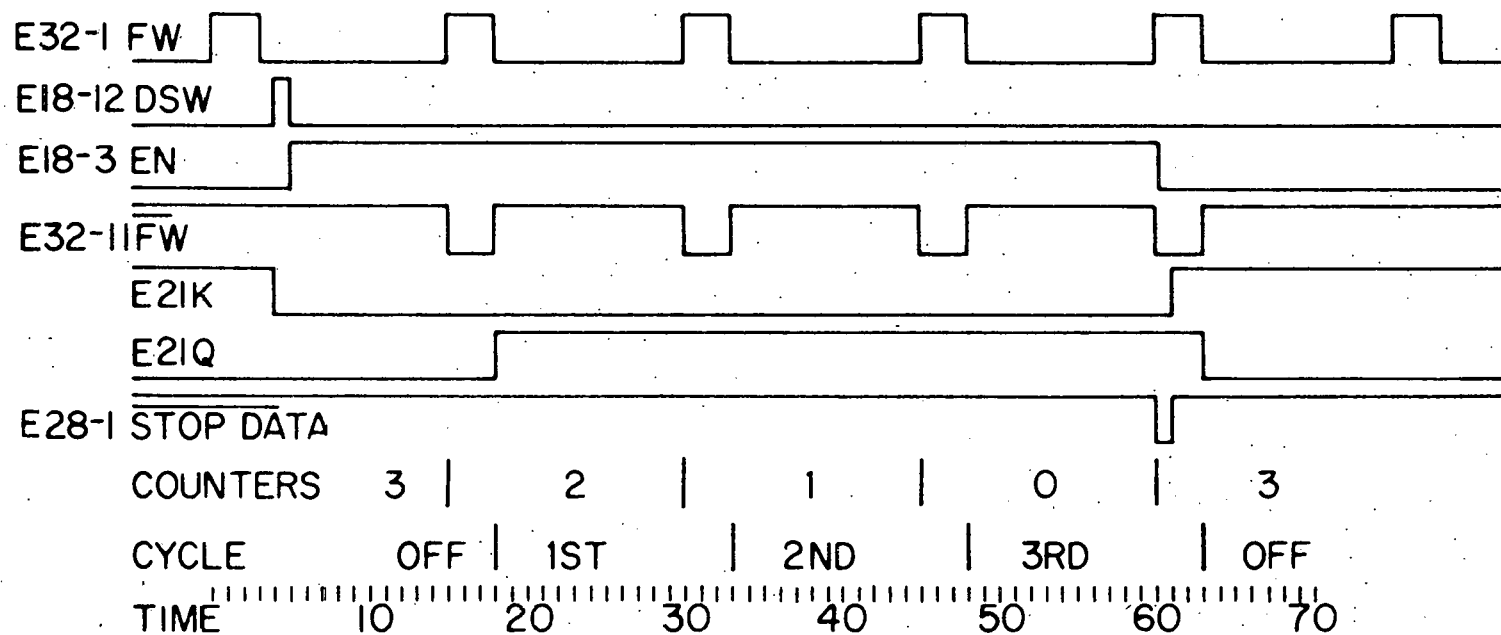


Fig. 11



CYCLE COUNTER TIMING DIAGRAM
(NCYCLE=3)

Fig. 12

PM AND TOF DISCRIMINATORS
(TOF VALUES IN PARENTHESES)

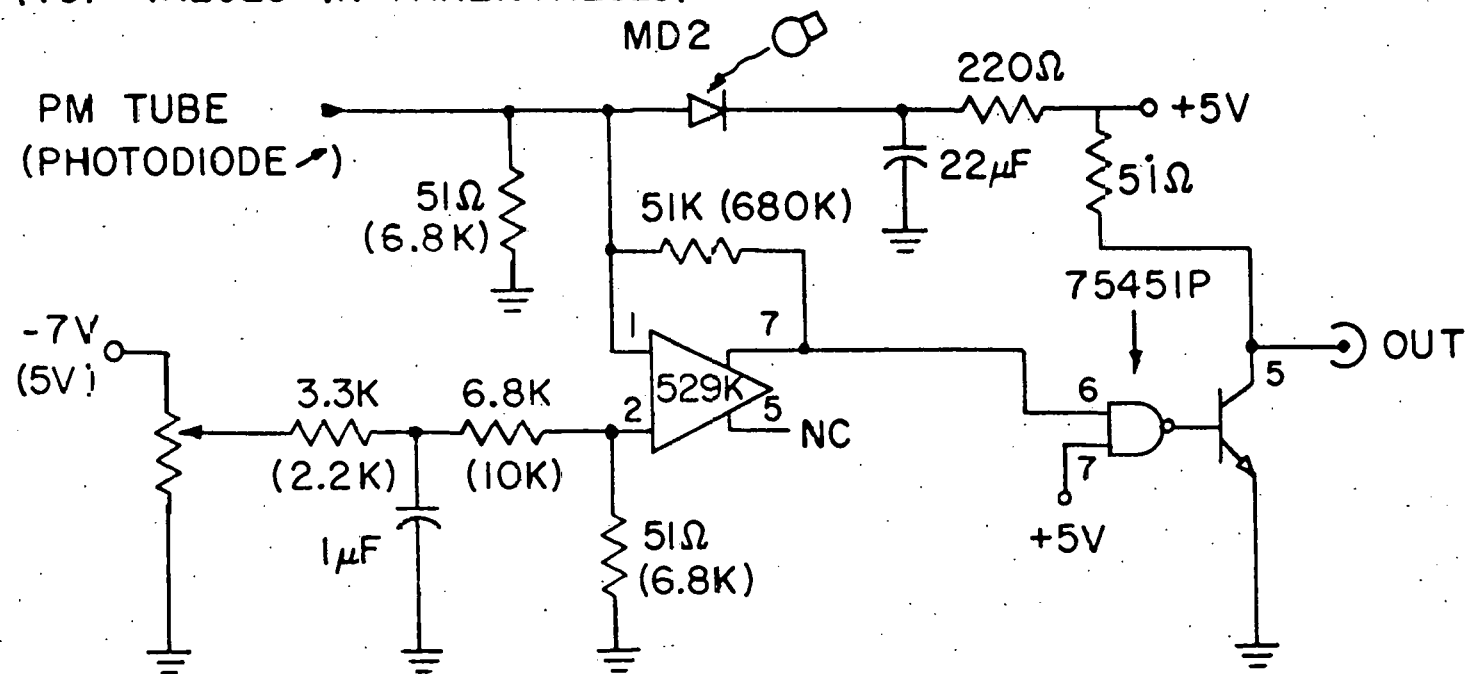


Fig. 13

BEAM GEOMETRY (ALL DIMENSIONS IN cm)

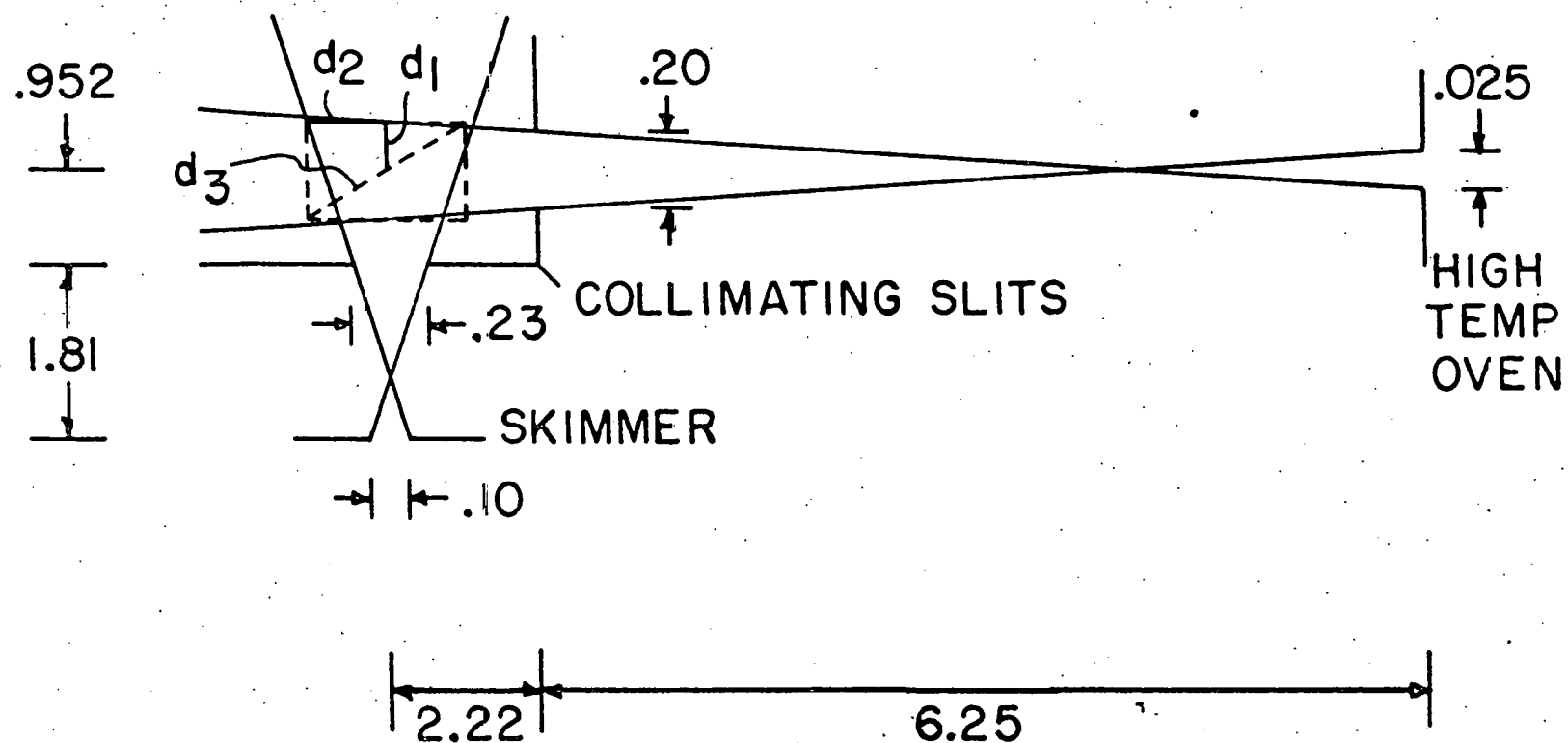


Fig. 14

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COMPUTER PROGRAM LISTING

0000	0000	0000	
0001	3030	DCA AC	
0002	7004	RAL	
0003	5010	JMP .+5	
0004	0000	0000	
0005	7400	7400	
0006	7200	7200	
0007	5600	5600	
0010	3031	DCA L	
0011	6171	6171	/READ STATUS WORD
0012	7100	CLL	
0013	7010	RAR	
0014	7440	SZA	
0015	5100	JMP EXPS	
0016	6031	KSF	
0017	5022	JMP RESTOR	
0020	5777	JMP KEYFLG	
	7402	HLT	
	7300	RESTOR, CLA CLL	
	1031	TAD L	
	7010	RAR	
	1030	TAD AC	
	6001	ION	
	5400	JMP I O	
0030	0000	AC, O	
	0000	L, O	
	0000	CRLF, O	
	7300	CLA CLL	
	6046	TLS L	
	1076	TAD LF	
	5063	JMP 63	
	1000	FLAGR, 1000R	/FLAGR=0, RUNNING
0063	6046	TLS	/FLAGR≠0, STOPPED
	6041	TSF	
	5064	JMP.-1	
	7200	CLA	
	1077	TAD CR	
0070	6046	TLS	
	6041	TSF	
	5071	JMP.-1	
	6042	TCF	
	7300	CLA CLL	
	5432	JMP I CRLF	
	0212	LF, 212	
	0215	CR, 215	
		*100	
0100	7300	EXPS, CLA CLL	
	6171	6171	
	7006	RTL	
	7004	RAL	

0104	7420		SNL
	5710		JMP I 110
	5507		JMP I GRAND
	1000	GRAND,	1000
0110	1140		1140
		*111	
0111	0000	SPACE,	0
	7300		CLA CLL
	1511		TAD I SPACE
	7041		CIA
	3126		DCA NSPCS
	1124		TAD TK240
	4525		JMS I ATLSX
0120	2126		ISZ NSPCS
	5116		JMP.-3
	2111		ISZ SPACE
	5511		JMP I SPACE
	0240	TK240,	240
	1242	ATLSX,	1242
	0000	NSPCS,	0
		*127	
0127	4533	KILL,	JMS I CLRK
0130	2000		2000
	3000		3000
	5144		JMP 144
	1546	CLRK,	1546
	0000	INPUT,	0
	7300		CLA CLL
	6046		TLS
	4405		JMS I 5
0140	6041		TSF
	5140		JMP.-1
	6042		TCF
	5157		JMP FIX
	6001		ION
	5546		JMP I DISPZ
	1600	DISPZ,	1600
	0000	OUTPUT,	0
0150	7000		NOP
	6046		TLS
	4406		JMS I 6
	6041		TSF
	5153		JMP.-1
	6042		TCF
	5547		JMP I OUTPUT
	7300	FIX,	CLA CLL
0160	1171		TAD THIR
	1044		TAD 144
	3170		DCA NFIX
	1045		TAD 45
	7010		RAR
	2170		ISZ NFIX

00166	5164		JMP, -2
00167	5534		JMP I INPUT
00170	0000	NFIX,	O
00171	7765	TMIR,	(-13)
00172	0000	CTHOLD	O
00173	0000		O
00174	0000		O
00175	7600	ABSTOP,	7600
00176	0365		
*200			
00200	6002		IOF
00201	5602		IMP I 202
00202	1300		1300
*203			
00203	4134		JMS INPUT
00204	4407		JMS I 7
00205	6362		FPUT 362
00206	0000		FEXT
00207	5022		JMP RESTOR
*400			
00400	7300	KEYFLG,	CLA CLL
00401	6036		KR2
00402	1223		TAD TZO
00403	7510		SPA
00404	5777		JMP QM
00405	3221		DCA OFFSET
00406	1221		TAD OFFSET
00407	1224		TAD TF
00410	7700		SMA CLA
00411	5777		JMP QM
00412	1260		TAD SLIST
00413	1221		TAD OFFSET
00414	3222		DCA INSTRR
00415	1622		TAD I INSTRR
00416	3222		DCA INSTRR
00417	5622		JMP I INSTRR
00420	2402		HCT
00421	0000	OFFSET,	0000
00422	0000	INSTRR,	0000
00423	7477	TZO,	-301
00424	7741	TF,	-37
*425			
00425	4631	STOPX,	JMS SSTOP
00426	7240		CLA CMA
00427	3037		DCA FLAGR
00430	5022		JMP RESTOR
00431	7600		SSTOP
00432	1237	RNX,	TAP K17
00433	6170		6170

00433	6170		6170
00434	7300		CLA CLL
00435	3037		DCA FLAGR
00436	5022		JMP RESTOR
00437	1700	147,	1700
00440	0000		
00441	0000		
00442	7000	STATUS,	NOP
00443	6171		6171
00444	7010		RAR
00445	7430		SZL
00446	5252		JMP 52
00447	4656		JMS I ATYPX
00450	5251		CMSTOP
00451	5254		JMP 54
00452	4656		JMS I ATYPX
00453	5245		CMRUN
00454	4032		JMS CRLF
00455	5022		JMS RESTOR
00456	1200	ATYPX,	1200
00457	0005		

*460

00460	0461	SLIST,	0461	/ADDR. 1ST PNTR.
00461	1063		NCYCLE	/A
00462	5000		BTX	/B
00463	1307		CONTX	/C
00464	1251		QM	/D
00465	5146		CTR	/E
00466	0532		FS	/F
00467	1251		QM	/G
00470	1251		QM	/H
00471	1251		QM	/I
00472	1251		QM	/J
00473	0127		KILL	/K
00474	1000		OVERF	/L
00475	1251		QM	/M
00476	1316		NBINX	/N
00477	1251		QM	/O
00500	1251		QM	/P
00501	0442		STATUS	/Q
00502	0432		RUNX	/R
00503	0425		STOPX	/S
00504	5142		CTLX	/T
00505	1353		SP	/U
00506	1360		DP	/V
00507	1000		WRITED	/W
00510	1341		SPN	/X
00511	1344		N	/Y
00512	1347		SIG	/Z
00513	1251		NS	/[
00514	1253		QM	/\
00515	1251		QM	/]
00516	1325		PAPAL	/f
00517	1255		COMM	/-

00520	0000		O	1 ISZ XN
00521	7650		SNA CLA	1 IFF D. P.
00522	5720		JMP I 520	
00523	1731		TAD I 531	
00524	7710		SPA CLA	
00525	2730		ISZ I 530	1 ISZ XN
00528	2320		ISZ 520	
00527	5720		JMP I 520	1 RETURN
00530	1756		1756	1 AXN
00531	1545		1545	1 ASORD
00532	1775	FS,	TAD I AFI	1 FULL SCALE ROUTINE
00533	1364		TAD KF1	
00534	7041		CIA	
00535	3365		DCA KF2	
00536	4407		JMS I 7	1 EMTER F. P.
00537	5366		FGET KF3	
00540	0000		FEXT	1 EXIT F. P.
00541	4407		JMS I 7	1 ENTER F. P. (SEE 546)
00542	3371		FMULT KF4	
00543	7000		FNOR	
00544	0000		FEXT	
00545	2365		ISZ KF2	
00546	5341		JMP. -5	
00547	7300		CLA CLL	
00550	7000		NOP	
00551	4776		JMS I AF2	
00552	5255		CMFS	
00553	6040		SPF	
00554	7300		CLA CU	
00555	1374		TAD FF8	
00556	3062		DCA 62	
00557	4406		JMS OUTPUT	
00560	6041		TSF	
00561	5360		JMP. -1	
00562	4032		JMS CRLF	
00563	5022		JMP RESTOR	
00564	0000	KF1,	6	
00565	0000	JF2,	0	
00566	0007	KF0,	FLTG 128	
00567	2000			
00570	0000			
00571	0001	KF4,	FLTG 2	
00572	2000			
00573	0000			
00574	0010	KF8,	10	
00575	1762	AF1,	1762	
00576	1200	AF2,	1200	
00577	1251	QM,	1251	
00600	4756	WRITED,	JMS SSTOP	
00601	4487		JMS I 7	/SETS STOP

00602	5772	FGET
00603	6771	FPUT FBINN
00604	0000	FEXT
00605	5210	JMP 13
00606	5210	JMS 32 / EXPECTS CRLF TO BE AT 0032
00607	4032	JMS 32
00610	5611	JMP 1611
00611	1371	1371
00612	0260	260
00613	4753	JMS I ATYPX / EXPECTS TYPX TO BE AT 1200
00614	5300	CMBIN
00615	4111	JMS SPACE
00616	0005	5
00617	4753	JMS I ATYPX
00620	5302	CMTIME
00621	4111	JMS SPACE
00622	0006	6
00623	4753	JMS I ATYPX
00624	5305	CMSPEN
00625	4111	JMS SPACE
00626	0010	10
00627	4753	JMS I ATYPX
00630	5307	CMN
00631	7300	CLACLL
00632	3045	DCA95
00633	3046	DCA96
00634	4407	JMS I 7
00635	6779	FPUT FSF
00636	6773	FPUT FSFV
00637	0000	FEXT
00640	7300	CLA CLL
00641	1376	TAD (3002)
00642	3354	DCA BINNUM
00643	1375	TAD (4002)
00644	3355	DCA ZINNUM
00645	7000	NOP
00646	7300	CLA CU
00647	3045	DCA 95
00650	3046	DCA 46
00651	4407	JMS I 7
00652	6767	FPUT SPNREG
00653	6766	FPUT NR EG
00654	0000	FEXT
00655	7300	CLA CLL
00656	1754	TAD I BINNUM
00657	3046	DCA 46
00660	2354	I SZ BINNUM
00661	1754	TAD I BINNUM
00662	3045	DCA 45
00663	1370	TAD (27)
00664	3044	DCA 44
00665	4407	JMS I 7
00666	1767	FADD SPNREG
00667	6767	FPUT SPNREG / LOADS S+N REG
00670	0000	FEXT
00671	1755	TAD I ZINNUM
00672	3046	DCA 46
00673	2355	I SZ ZINNUM
00674	1755	TAD I ZINNUM
00675	3045	DCA 45
00676	1370	TAD (27)
00677	3044	DCA 44

RETWED,

00701	1760	FADD NREG	
00702	6766	FPUT NREG	/LOADS N REG
00703	5767	FGET SPNREG	
00704	2766	FSUB NREG	
00705	6765	FPUT SREG	/LOADS SIG (DIFF) REG
00706	57 71	FGET FBINN	
00707	3764	FMPY FTWO	
00710	2772	FSUB FONE	
00711	3763	FMPY FBINT	
00712	4764	FDIV FTWO	
00713	2762	FSUB FDT	
00714	6761	FPUT FTIME	/TIME=BINTIME*(2BIN#-1)/2 -DEL
00715	0000	FEXT	
00716	5612	JMP I612	
00717	4760	JMS TYPE	
00720	0003	3	
00721	4407	JMS I 7	
00722	57 61	FGET FTIME	
00723	0000	FEXT	
00724	4760	JMS TYPE	
00725	0006	6	
00726	4407	JMS I 7	
00727	5767	FGET SPNREG	
00730	0000	FEXT	
00731	4760	JMS TYPE	
00732	0007	7	
00733	4407	JMS I 7	
00734	57 66	FGET NREG	
00735	0000	FEXT	
00736	47 60	JMS TYPE	
00737	0007	7	
00740	4407	JMS I 7	
00741	57 65	FGET SREG	
00742	0000	FEXT	
00743	47 60	JMS TYPE	
00744	0007	7	
00745	4407	JMS I 7	
00746	5767	FGET SPNREG	
00747	1766	FADD NREG	
00750	0002	SQROOT	
00751	0000	FEXT	
00752	5757	JMP 210	
00753	1200	ATYPX, 1200	
00754	0000	BINNUM, 0	
00755	0000	ZINNUM, 0	
00756	7600		
00757	0210		
00760	0302		
00761	0332		
00762	0351		
00763	0346		
00764	0340		
00765	0327		
00766	0324		
00767	0321		
00770	0027		
00771	0335		
00772	0343		
00773	0316		
00774	0313		

00777	0040		
		*210	
00210	4302	JMS TYPE	
00211	0004	4	
00212	5613	JMS I 213	/FOR FLUX 212=4407
00213	0233	233	/AND 213-5354
00214	3357	FMPY FMILL	
00215	4332	FDIV FTIME	
00216	6321	FPUT SPNREG	/SPNREG NOW USED TO HOLD VELOCITY
00217	0000	FEXT	
00220	4302	JMS TYPE	
00221	0007	7	
00222	4407	JMS I 7	
00223	5327	FGET SREG	
00224	3354	FMPY FDIST	
00225	4321	FDIV SPNREG	/NREG NOW USED FOR FLUX
00226	6324	FPUT NREG	
00227	3362	FMPY FTTHOU	
00230	0000	FEXT	
00231	4302	JMS TYPE	
00232	0007	7	
00233	4407	JMS I 7	
00234	5324	FGET NREG	
00235	1313	FADD FSF	
00236	6313	FPUT FSF	
00237	5324	FGET NREG	
00240	3321	FMPY SPNREG	
00241	1316	FADD FSFV	
00242	6316	FPUT FSFV	
00243	5335	FGET FBINN	
00244	1343	FADD FONE	
00245	6335	FPUT FBINN	
00246	2365	FSUB FNBINS	
00247	0000	FEXT	
00250	4032	JMS CRLF	
00251	1045	TAD 45	
00252	7550	SPA SNA	
00253	5274	TMP KEYQ	
00254	5022	JMP RESTOR	
00255	4032	JMS CRLF	
00256	4032	JMS CRLF	
00257	5022	JMP RESTOR	
00260	2111	ISZ BINNUM	
00261	2776	ISZ ZINNUM	
00262	2370	ISZ NSP	
00263	5773	JMP I 373	
00264	7300	CLA CLL	
00265	1271	TAD NBIND	
00266	7041	CIA	
00267	3370	PCA NSP	
00270	4407	JMS I 7	
00271	5335	FGET FBINN	
00272	0000	FEXT	
00273	5772	JMP I 372	
00274	6031	KEYQ,	
00275	5774	KSF	
00276	6030	JMP RETUR	
00277	4775	KCF	
00300	5351	JMS TYPX	
00301	5255	CMTERM	
00302	0000	JMP 255	
00303	6046	0	
		TYPE,	
		TLS	

00777 0040

*210

-65-

00210 4302 JMS TYPE
 00211 0004 4
 00212 5613 JMS I 213
 00213 0233 233
 00214 3357 FMPY FMILL
 00215 4332 FDIV FTIME
 00216 6321 FPUT SPNREG
 00217 0000 FEXT
 00220 4302 JMS TYPE
 00221 0007 7
 00222 4407 JMS I 7
 00223 5327 FGET SREG
 00224 3354 FMPY FDIST
 00225 4321 FDIV SPNREG
 00226 6324 FPUT NREG
 00227 3362 FMPY FTTHOU
 00230 0000 FEXT
 00231 4302 JMS TYPE
 00232 0007 7
 00233 4407 JMS I 7
 00234 5324 FGET NREG
 00235 1313 FADD FSF
 00236 6313 FPUT FSF
 00237 5324 FGET NREG
 00240 3321 FMPY SPNREG
 00241 1316 FADD FSFV
 00242 6316 FPUT FSFV
 00243 5335 FGET FBINN
 00244 1343 FADD FONE
 00245 6335 FPUT FBINN
 00246 2365 FSUB FNBINS
 00247 0000 FEXT
 00250 4032 JMS CRLF
 00251 1045 TAD 45
 00252 7550 SPA SNA
 00253 5274 JMP KEYQ
 00254 5022 JMP RESTOR
 00255 4032 JMS CRLF
 00256 4032 JMS CRLF
 00257 5022 JMP RESTOR
 00260 2777 ISZ BINNUM
 00261 2776 ISZ ZINNUM
 00262 2370 ISZ NSP
 00263 5773 JMP I 373
 00264 7300 CLA CLL
 00265 1271 TAD NBIND
 00266 7041 CLA
 00267 3370 PCA NSP
 00270 4407 JMS I 7
 00271 5335 FGET FBINN
 00272 0000 FEXT
 00273 5772 JMP I 372
 00274 6031 KEYQ, KSF
 00275 5774 JMP RETUR
 00276 6030 KCF
 00277 4775 JMS TYPX
 00300 5351 CMTERM
 00301 5255 JMP 255
 00302 0000 TYPE, O
 00303 6046 TLS
 00304 1702 TAD I TYPE

/FOR FLUX 212 = 4407

I AND 213 = 5354

/SPNREG NOW USED TO HOLD VELOCITY

/NREG NOW USED FOR FLUX

00306	4406	JMS I 6
00307	4111	JMS SPACE
00310	0003	3
00311	2302	ISZ TYPE
00312	5702	JMP I TYPE

00313	0000	FSF,	FLTG 0
00314	0000		
00315	0000		
00316	0000	FSFV,	FLTG 0
00317	0000		
00320	0000		
00321	0000	SPNREG,	FLTG 0
00322	0000		
00323	0000		
00324	0000	NREG,	FLTG 0
00325	0000		
00326	0000		
00327	0000	SREC,	FLTG 0
00330	0000		
00331	0000		
00332	0000	FTIME,	FLTG 0
00333	0000		
00334	0000		
00335	0000	FBINN,	FLTG 0
00336	0000		
00337	0000		
00340	0002	FTWO,	FLTG 2
00341	2000		
00342	0000		
00343	0001	FONE,	FLTG 1
00344	2000		
00345	0000		
00346	0007	FBINT,	FLTG 100
00347	3100		
00350	0000		
00351	0000	FDT,	FLTG 0
00352	0000		
00353	0000		
00354	0004	FDIST,	FLTG 10
00355	2400		
00356	0000		
00357	0024	FMILL,	FLTG 1.0E06
00360	3641		
00361	1000		
00362	0016	FTTHOU,	FLTG 1.0E04
00363	2342		
00364	0000		
00365	0006	FNBINS,	FLTG 32
00366	2000		
00367	0000		
370	7777		
371	1764		
372	0717		
373	5334		

```

00375 1200
00376 0755
00377 0754
...
05300 0211 *5300
05301 1600 CMBIN, TEXT "BI
N"
05302 2411 CMTIME, TEXT "TI
05303 1505 ME
05304 0000 "
05305 2353 CMSPN, TEXT "S+
05306 1600 N"
05307 1600 CMN, TEXT "N
05310 4040
05311 4040
05312 4040
05313 0411 CMD
05314 0606 FF
05315 4040
05316 4040
05317 2321 SQ
05320 4022 R
05321 2440 T
05322 4026 V
05323 0514 EL
05324 1703 OC
05325 1124 IT
05326 3140 Y
05327 4040
05330 4006 F
05331 1425 LU
05332 3037 X
05333 3700 "
05334 4407 JMS I 7
05335 5742 FGET FBINN
05336 1743 FADD FONE
05337 6742 FPUT FBINN
05340 0000 FEXT
05341 5744 JMP RETWED
05342 0335 FBINN, 335
05343 0343 FONE, 343
05344 0655 RETWED, 655
05345 0000 0
05346 0000 0
05347 0000 0
05350 0000 0
05351 4024 CMTERM, TEXT " T
05352 0522 ER
05353 1511 MI
05354 1601 NA
05355 2405 TE
05356 0400 D "
*1000
01000 7300 OVERF, CLA CLL
01001 1377 TAD ( 40)
01002 6170 LCW
/ STOPS EXP.

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01005	3333	DCA AOV	
01006	1330	TAD BIOV	
01007	3334	DCA BOV	
01010	3336	DCA FLGOV	
01011	6171	RSW	
01012	7010	RAR	
01013	7430	SZL	
01014	5211	JMP, -3	
01015	7300	CLA CLL	
01016	1332	GOAUNO, TAD KOV2	
01017	6170	LSW	
01020	7000	NOP	
01021	7300	RETOV, CLA CLL	
01022	2333	ISZ AOV	
01023	2334	ISZ BOV	
01024	1734	TAD I BOV	
01025	1733	TAD I AOV	
01026	3734	DCA I BOV	
01027	3733	DCA I AOV	
01030	7004	RAL	
01031	2334	ISZ BOV	
01032	1734	TAD I DOV	
01033	7430	SZL	
01034	4255	JMS TYPQVR	
01035	7510	SPA	
01036	2336	ISZ FLGOV	
01037	3734	DCA I BOV	
01040	5243	JMP 1 to 43	
01041	1337	TAD KOV1	
01042	5314	JMP 1114	
01043	1330	TAD BIOV	
01044	7041	CIA	
01045	1333	TAD AOV	
01046	7710	SPA CLA	
01047	5221	JMP RETOV	
01050	5307	JMP GOBCK	
01051	1037	TAD FLAGR	
01052	7440	SZA	
01053	5727	JMP 11127	
01054	5241	JMP 1041	
01055	0000	TYPQVR, 0	
01056	7300	CLA CLL	
01057	7604	LAS	/SET SWITCH REG. BIT 11 TO 1
01060	7010	RAR	/TO KILL MESSAGES
01061	7430	SZL	
01062	5655	JMP I TYPQVR	
01063	4032	JMS CRLF	
01064	7000	NOP	
LG 01065	4776	JMS TYPX	
01066	5233	CMOVER	/TYPES ERROR MESSAGE
01067	1326	TAD AIOV	
01070	7041	CIA	
01071	1333	TAD AOV	
01072	3045	DCA 45	
01073	3046	DCA 46	
01074	1375	TAD (13)	
01075	3044	DCA 44	
01076	4407	JMS I 7	
01077	7000	FNOR	
01100	0000	FEXT	
01101	7240	CLA CMA	
01102	3055	DCA 55	
01103	1374	TAD (3)	

01104	3062		DCA 62
01105	4547		JMS OUTPUT
01106	5655		JMP I TYPOVR
01107	7300	GOBCK,	CLA CLL
01110	1336		TAD FLGOV
01111	7440		SZA
01112	5317		JMP ,+5
01113	5251		JMP 1051
01114	6170		6170
01115	5022		JMP RESTOR
01116	7600		ASTOP
01117	4716		JMS SSTOP
01120	4032		JMS CRLF
LG 01121	4776		JMSTYPX
01122	5200		CMFULL
01123	7201		CLA IAC
01123	3037		DCA FLAGR
01125	5022		JMP RESTOR
01126	1777	AIOV,	1777
01127	0600		0600
01130	2777	BIOV,	2777
01131	0000		0
01132	0440		440
01133	0000	AOV,	0
01134	0000	BOV,	0
01135	0000	NOV,	0
01136	0000	FLGOV,	0
01137	0500	KOVI,	0500
01140	4575		JMS I 175
01141	4575		JMS I 175
01142	6171		RSW
01143	7004		RAL
01144	7430		SZL
01145	5353		JMP 1153
01146	7004		RAL
01147	7430		SZL
01150	5361		JMP 1161
01151	7402		HLT
01152	5022		JMP RESTOR
01153	4767		JMS TYPX
01154	5263		CMBTG
01155	4032		JMS CRLF
01156	1370		TAD KE20
01157	6170		LSW

1160	5022	JMP RESTOR	
1160	5022	JMP RESTOR	
1161	4767	JMS TYPX	
1162	5272	CMTIME	
1163	4032	JMS CRLF	
1164	1371	TAD KEIK	
1165	6170	LSW	
1166	5372	JMP 1172	
1167	1200		
1170	0020		
1170	0020	KEZO,	20
1171	1000	KEIK,	1000
1172	3037		DCA FLAGR
1173	5022		JMP RESTOR
01174	0003		
01175	0013		
01176	1200		
01177	0400		
		*1200	
01200	0000	TYPX,	0
01201	7300		CLA CLL
01202	1600		TAD I TYPX
01203	3214		DCA TYPNT
01204	2200		ISZ TYPX
01205	1614	TYPXI,	TAD I TYPNT
01206	7002		7002
01207	4215		JMS TYPY
01210	1614		TAD I TYPNT
01211	2214		ISZ TYPNT
01212	4215		JMS TYPY
01213	5205		JMP TYPXI
01214	0000	TYPNT,	0
01215	0000	TYPY,	0
01216	0234		AND TK77
01217	7450		SNA
01220	5600		JMP I TYPX
01221	1235		TAD TKM37
01222	7440		SZA
01223	522u		JMP TYPY1
01224	1236		TAD TK215
01225	4242		JMS TLSX
01226	1237		TAD TKM125
01227	7510	TYPY1,	SPA
01230	1240		TAD TK100
01231	1241		TAD TK237
01232	4242		JMS TLSX
01233	5615		JMP I TYPY
01234	0077	TK77,	77
01235	7741	TKM37,	-37
01236	0215	TK215,	215
01237	7653	TKM125,	-125
01240	0100	TK100,	100
01241	0237	TK237,	237

01243	6046		TLS
01244	6041		TSF
01245	5244		JMP .-1
01246	6042		TCF
01247	7200		CLA
01250	5642		JMP I TLSX
01251	4654	QM,	JMS I QMK
01252	7300		CLA CLL
01253	5022		JMP RESTOR
01254	0032	QMK,	CRLF

*QMK+1

01255	7300	COMM,	CLA CLL
01256	1275		TAD TK334
01257	4242		JMS TLSX
01260	6031		KSF
01261	5260		JMP .-1
01262	6036		KRB
01263	7041		CIA
01264	1236		TAD TK215
01265	7440		SZA
01266	5271		JMP ECHO
01267	4032		JMS CRLF
01270	5022		JMP RESTOR
01271	7041	ECHO,	CIA
01272	1236		TAD TK215
01273	4242		JMS TLSX
01274	5260		JMP COMM+3
01275	0257	TK257	257

01400	0000	*1400	
01401	7300	DPDADD,	0
01402	1345	CLA	CLL
01403	7710	TAD	SORD
01404	1377	SPA	CLA
01405	1377	TAD	(400)
01406	1776	TAD	(400)
01407	3261	TAD	XN
01410	1776	DCA	YN
01411	3260	TAD	XN
01412	1373	DCA	PADD
01413	3257	TAD	AA001
01414	7100	DCA	AADD
01415	1660	RETADD,	CLL
01416	1657	TAD	I PADD
01417	3657	TAD	I AADD
01420	1345	DCA	I AADD
01421	7700	TAD	SORD
01422	5227	SMA	CLA
01423	7004	JMP	..+5
01424	2260	RAL	
01425	1660	ISZ	PADD
01426	7410	TAD	I PADD
01427	7004	SKP	
01430	2257	RAL	
01431	1657	ISZ	AADD
01432	3657	TAD	I AADD
01433	7004	DCA	I AADD
01434	2257	RAL	
01435	1657	ISZ	AADD
01436	3657	TAD	I AADD
01437	1257	DCA	I AADD
01440	7041	TAD	AADD
01441	1371	CIA	
01442	4770	TAD	AA0013
01443	5600	JMS	I 1570
01444	1361	JMP	I DPDADD
01445	3260	TAD	YN
01446	1372	DCA	PADD
01447	3257	TAD	AA0012
01450	5214	DCA	AADD
01451	0000	JMP	RETADD
01452	0000	ALSDSN,	0
01453	0000	AMSDSN,	0
01454	0000	AHSDSN,	0
01455	0000	ALSDN,	0
01456	0000	AMSDN,	0
01457	0000	AHSDN,	0
01460	0000	AADD,	0
01461	0000	PADD,	0
01462	0000	YN,	0
01463	7300	DPDSUB,	0
01464	1254	CLA	CLL
01465	7041	TAD	ALSDN
01466	1251	CIA	
01467	3307	TAD	ALSDSN
01470	7004	DCA	LSDH
01471	3312	RAL	
01472	1255	DCA	KEEPD
01473	7040	TAD	AMSDN
01474	1252	CMA	
01475	1312	TAD	AMSDSN
		TAD	KEEPD

/S.P. DATA MUST BE IN N
/D.P. DATA MUST BE IN M

/ADDS PRESENT DATA
/TO HOLDING REG (ALSDSN ETC.)

/TAKES SUM(S+N)-SUM(N) AND PUTS
/IT IN LSDH ETC. ('TRIPLE PREC')

01477	7004	RAL	
01500	3312	DCA	KEEPD
01501	1256	TAD	AHSDN
01502	7040	CMA	
01503	1253	TAD	AHSDSN
01504	1312	TAD	KEEPD
01505	3311	DCA	HSDH
01506	5662	JMP	I DPDSUB
01507	0000	LSDH,	0
01510	0000	MSDH,	0
01511	0000	HSDH,	0
01512	0000	KEEPD,	0
01513	7200	CHECKD,	CLA
01514	1344	TAD	SORN
01515	7510	SPA	
01516	5325	JMP	NOISE
01517	7640	SZA	CLA
01520	5335	JMP	SIGPN
01521	4262	JMS	DPDSUB
01522	7420	SNL	
01523	5775	JMP	1704
01524	5774	JMP	SROT
01525	7300	NOISE,	CLA CLL
01526	1254	TAD	ALSDN
01527	3307	DCA	LSDH
01530	1255	TAD	AMSDN
01531	3310	DCA	MSDH
01532	1256	TAD	AHSDN
01533	3311	DCA	HSDH
01534	5774	JMP	SROT
01535	1251	SIGPN,	TAD ALSDSN
01536	3307	DCA	LSDH
01537	1252	TAD	AMSDSN
01540	3310	DCA	MSDH
01541	1253	TAD	AHSDSN
01542	3311	DCA	HSDH
01543	5774	JMP	SROT
01544	0001	SORN,	1
01545	0001	SORD,	1
01546	0000	CLEAR,	0
01547	7200	CLA	
01550	1746	TAD	I CLEAR
01551	3365	DCA	STADD
01552	2346	ISZ	CLEAR
01553	1746	TAD	I CLEAR
01554	7041	CIA	
01555	3366	DCA	COUNTC
01556	3765	DCA	I STADD
01557	2365	ISZ	STADD
01560	2366	ISZ	COUNTC
01561	5356	JMP	--3
01562	2346	ISZ	CLEAR
01563	7300	CLA	CLL
01564	5746	JMP	I CLEAR
01565	0000	STADD,	0
01566	0000	COUNTC,	0
01570	0520	/DISPLAY ROUTINE FOR TIME OF FLIGHT.	
01571	1456	/AUTOMATICALLY SCALES BOTH AXES. TAKES SUM OF N BINS.	
01572	1454	/TAKES S+N-N, LOOKS AT ORIGINAL DATA FIELD (SING PREC)	
01573	1451	/OR AT DOUBLE PRECISION STORAGE LOCATIONS.	
01574	1647		

/CHECKS WHETHER YOU WANT TO DIS
/S+N, N, OR S+N-N=S AND LOADS L
/ACCORDINGLY

/SORN>0, S+N; <0, N; =0, S+N-N=S
/SORD>0, DMA SINGLE PREC.
/CLEARS MEMORY LOCATIONS STADD
/PLUS N. ENTERED THUSLY:
/JMS CLEAR; STADD; N

01575	1704		
01576	1756		
01577	0400		
		*1600	
01600	7300	DISPLA,	CLA CLL
01601	3355		DCA NDIS
01602	3362		DCA ROTF
01603	3365		DCA NA
01604	1354		TAD NBINA
01605	2355		ISZ NOIS
01606	7004		RAL
01607	4720		SNL
01610	5205		JMP. -3
01611	7300		CLA CLL
01612	3365		DCA NA
01613	7200	STAR,	CLA
01614	7000		NOP
01615	7000		NOP
01616	5222		JMP. +4
01617	7240		CLA CMA
01620	1362		TAD ROTF
01621	3362		DCA ROTF
01622	1362		TAD ROTF
01623	1377		TAD (+6)
01624	7500		SMA
01625	5231		JMP. +4
01626	7200		CLA
01627	1376		TAD (-5)
01630	3362		DCA ROTF
01631	7240		CLA CMA
01632	1353		TAD START2
01633	3356		DCA XN
01634	4775	RET,	JMS CLEAR
01635	1451		ALSDSN
01636	0006		6
01637	1364		TAD NBIND
01640	7041		CIA
01641	3360		DCA NBIN2
01642	2356		ISZ XN
01643	4774		JMS DPDADD
01644	2360		ISZ NBIN2
01645	5242		JMP. -3
01646	5773		JMP CHECKD
01647	1362	SROT,	TAD ROTF
01650	7500		SMA
01651	7041		CIA
01652	7450		SNA
01653	5260		JMP. +5
01654	3363		DCA ROTF2
01655	4320		JMS TROT
01656	2363		ISZ ROTF2
01657	5255		JMP. -2
01660	7200		CLA
01661	1772	TAD MSDH	
01662	7440		SZA
01663	5350		JMP BIG
01664	1771		TAD MSDH
01665	7440		SZA
01666	5350		JMP BIG
01667	1770		TAD LSDH
01670	7510		SPA
01671	7000		NOP
01672	6063		DYL

/SETS ROTF=ROTF-1 IF NO VALUE >
/TWO OF FULL SCALE
/WON'T ROTATE ANY MORE THAN 5
/XN=START2-1
/ADDS UP N BINS TO TRIPLE PRE
/MAKES ROTF2=-ABS(ROTF)
/SKIPS ALL ROTATING IF ROTF=0
/ROTATES TRIPLE PRECISION
/RIGHT (OR LEFT IF ROTF<0
/IF ANYTHING LEFT IN MSDH OR MSB
/THEN DIDN'T ROTATE RIGHT ENOUGH
/IF ANY VALUE IS WITHIN 2 OF FS
/FLAGD IS SET NON-ZERO

/PREVIOUS STEP LOADS Y-AXIS

01673	7300		CLA CLL
01674	1355		TAD NDIS
01675	7041		CIA
01676	3357		DCA NN
01677	1365		TAD NA
01700	2357		ISZ NN
01701	5315		JMP ROT
01702	6053		DXL
01703	6054		DIX
01704	7300		CLA CLL
01705	1354		TAD NBINA
01706	7041		CIA
01707	1365		TAD NA
01710	2365		ISZ NA
01711	7710		SPA CLA
01712	5234		JMP RET
01713	5211		JMP 1611
01714	7000		NOP
01715	7100	ROT,	CLL
01716	7004		RAL
01717	5300		JMP SKIP
01720	0000	TROT,	0
01721	7200		CLA
01722	1362		TAD ROTF
01723	7500		SMA
01724	5326		JMP TRR
01725	5341		JMP TRL
01726	7300	TRR,	CLA CLL
01727	1771		TAD HSDH
01730	7010		RAR
01731	3771		DCA HSDH
01732	1772		TAD MSDH
01733	7010		RAR
01734	3772		DCA MSDH
01735	1770		TAD LSDH
01736	7010		RAR
01737	3770		DCA LSDH
01740	5720		JMP I TROT
01741	7300	TRL,	CLA CLL
01742	1770		TAD LSDH
01743	7004		RAL
01744	7430		SZL
01745	5350		JMP BIG
01746	3770		DCA LSDH
01747	5720		JMP I TROT
01750	2362	BIG,	ISZ ROTF
01751	7000		NOP
01752	5213		JMP STAR
01753	2001	START2,	2001
01754	0062		0062
01755	0000	NDIS,	0000
01756	0000	XN,	0000
01757	0000	NN,	0000

01760	0000	NBIN2,	0
01761	0000	FLAGD,	0
01762	0000	ROTF,	0
01763	0000	ROTF2,	0
01764	0004	NBIND,	0004

01770	1507
01771	1511
01772	1510
01773	1513
01774	1400
01775	1546
01776	7773
01777	0006

05000	4777	*5000	JMS TYPX	
05001	5357	BTX,	CMBINT	
05002	4134		JMS INPUT	
05003	6174		6174	
05004	4407		JMS I 7	
05005	6776		FPUT FBINT	
05006	0000		FEXT	
05007	7300		CLA CLL	
05010	5022		JMP RESTOR	
05011	0000	NBTL,	0	
05012	0000	CTL,	0	
05013	7300		CLA CLL	
05014	3340		DCA NCTL	
05015	4134		JMS 134	
05016	4407		JMS I 7	
05017	6172		FPUT CTHOLD	
05020	3264		FMPY MILL	/MILL=MILLION, 10 ⁶
05021	0000		FEXT	
05022	4407	CTLRET,	JMS I 7	
05023	4267		FDIV TEN	/TEN=10 (10)
05024	0000		FEXT	
05025	2340		ISZ NCTL	
05026	1375		TAD (-7)	
05027	1044		TAD 44	
05030	7510		SPA	
05031	5272		JMP CTLERR	
05032	1374		TAD (-4)	
05033	7540		SMA SZA	/ 7<=BIN EXP<=11 (10)
05034	5222		JMP CTLRET	
05035	7450	CTLAGN,	SNA	
05036	5247		JMP CTLYES	
05037	2044		ISZ 44	
05040	7300		CLA CLL	
05041	1045		TAD 45	
05042	7010		RAR	
05043	3045		DCA 45	
05044	1373		TAD (-13)	
05045	1044		TAD 44	
05046	5235		JMP CTLAGN	
05047	1372	CTLYES,	TAD (-3)	
05050	1340		TAD NCTL	
05051	7510		SPA	
05052	5272		JMP CTLERR	
05053	1371		TAD (-5)	/ 3<=NCTL<=8?? NO: ERROR MESS
05054	7540		SMA SZA	
05055	5272		JMP CTLERR	
05056	7200		CLA	
05057	1045		TAD 45	
05060	0370		AND (7760)	/THROW AWAY 4 LSB
05061	1340		TAD NCTL	
05062	6173		6173	
05063	5612		JMP I CTL	
05064	0024	MILL,	FLTG +100.00E+04	
05065	3641			
05066	1000			
05067	0004	TEN,	FLTG +100.00E-01	
05070	2400			
05071	0000			
05072	4777	CTLERR,	JMS TYPX	
05073	5365		CTERR	

05075	0000	CTR,	0	/COUNTING TIME READ
05076	7200		CLA	
05077	1340		TAD NCTL	
05100	7041		CIA	
05101	3341		DCA NCTLL	
05102	4407		JMS I 7	
05103	5267		FGET TEN	
05104	0000		FEXT	
05105	2341	CTRRET,	ISZ NCTLL	
05106	7410		SKP	
05107	5315		JMP CTROUT	
05110	4407		JMS I 7	
05111	3267		FMPY TEN	
05112	7000		FNOR	
05113	0000		FEXT	
05114	5305		JMP CTRRET	
05115	4407	CTRROUT,	JMS I 7	
05116	6335		FPUT NCTLF	
05117	0000		FEXT	
05120	7300		CLA CLL	
05121	6172		6170	
05122	3045		DCA 45	
05123	1367		TAD (13)	
05124	3044		DCA 44	
05125	3046		DCA 46	
05126	4407		JMS I 7	
05127	7000		FNOR	
05130	4264		FDIV MILL	
05131	3335		FMPY NCTLF	
05132	7000		FNOR	
05133	0000		FEXT	
05134	5675		JMP I CTR	
05135	0000	NCTLF,	0	
05136	0000		0	
05137	0000		0	
05140	0000	NOTL,	0	
05141	0000	NCTLL,	0	
05142	4777	CTLX,	JMS TYPX	
05143	5372		CMCT	
05144	4212		JMS CTL	
05145	5022		JMP RESTOR	
05146	4777	CTRX,	JMS TYPX	
05147	5375		CMCTR	
05150	4275		JMS CTR	
05151	1366		TAD (6)	
05152	3062		DCA 62	
05153	1365		TAD (3)	
05154	4147		JMS OUTPUT	
05155	5022		JMP RESTOR	
05156	4777	SHIFTK,	JMS TYPX	
05157	1314		CMNS	
05160	4134		JMS INPUT	
05161	3764		DCA NBIND	
05162	5144		JMP 144	

NBIND=1764
TYPX=1200
FBINT=346
RESTOR=22
CTHOLD=172

05163 1070
05164 1764
05165 0003
05166 0006
05167 0013
05170 7760
05171 7773
05172 7775
05173 7765
05174 7774
05175 7771
05176 0346
05177 1200

*5357

05357 0211 CMBINT,
05360 1640 N
05361 2411 TI
05362 1505 ME
05363 7540 =
05364 4000 "

TEXT "BI

05365 2422 CTERR,
05366 3140 Y
05367 0107 AG
05370 1641 NI
05371 3700 -"

TEXT "TR

05372 0324 CMCT,
05373 7540 =
05374 4000 "

TEXT "CT

05375 2411 CMCTR,
05376 1505 ME
05377 7500 ="

TEXT "TI

```

*1300
01300 4705 4705
01301 2000 2000
01302 3000 3000
01303 6001 ION
01304 5706 5706
01305 1546 1546
01306 1600 1600
01307 7300 CLA CLL
01310 1313 TAD KCONTI
01311 5712 JMP I1312
01312 0432 0432
01313 7600 -200
01314 1623 CMNS,
01315 7500 =" TEXT "NS

*1316
01316 4134 NBINX, JMS INPUT
01317 3724 DCA I NBINAX
01320 4407 JMS I 7
01321 6576 FPUT I FNBINX
01322 0000 FEXT
01323 5144 JMP 144
01324 1754 NBINAX, 1754

01325 4134 PARALD, JMS INPUT
01326 4407 JMS I 7
01327 6736 FPUT I FDISTX
01330 0000 FEXT
01331 4134 JMS INPUT
01332 4407 JMS I 7
01333 6737 FPUT I FDTX
01334 0000 FEXT
01335 5740 JMP I TOT
01336 0354 FDISTX, 354
01337 0351 FDTX, 351
01340 0203 TOT, 203
01341 7201 SPN, CLA IAC
01342 3752 DCA I SORNX
01343 5144 JMP 144
01344 7 240 NOISE, CLA CMA
01345 3752 DCA I SORNX
01346 5144 JMP 144
01347 7300 SIGNAL, CLA CLL
01350 3752 DCA I SORNX
01351 5144 JMP 144
01352 1544 SORNX, 1544
01353 7201 SP, CLA IAC
01354 3765 DCA I SORDX
01355 1367 TAD T000
01356 3766 DCA I STARTX
01357 5144 JMP 144
01360 7240 DP, CLA CMA
01361 3765 DCA I SORDX
01362 1370 TAD T00T
01363 3766 DCA I STARTX
01364 5144 JMP 144
01365 1545 SORDX, 1545
01366 17 53 STARTX, 1753
01367 2001 T000, 2001
01370 3002 T00T, 3002

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*5200
05200 0401 CMFULL, TEXT /DA
05201 2401 TA
05202 4010 H
05203 0123 AS
05204 4022 R
05205 0501 EA
05206 0310 CH
05207 0504 ED
05210 4006 F
05211 2514 UL
05212 1440 L
05213 0417 DO
05214 2502 UB
05215 1405 LE
05216 4020 P
05217 2205 RE
05220 0311 CI
05221 2311 SI
05222 1716 ON
05223 7340 ;
05224 0417 DO
05225 1624 NT
05226 4003 C
05227 1716 ON
05230 2411 TI
05231 1625 NU
05232 0500 E/

05233 1726 CMOVER, TEXT /OV
05234 0522 ER
05235 0614 FL
05236 1727 OW
05237 3700 -
05240 4340 #
05241 0331 CY
05242 0315 CL
05243 0523 ES
05244 7500 =

RSW=6171
LCW=6170