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Columbia University
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NUMERICAL SIMULATION STUDIES OF **MASTER**
TOROIDAL PINCHES

by

Wonchull Park

1978



Work Supported by DOE Contract EY-76-S-02-2456

Plasma Laboratory

School of Engineering and Applied Science

Columbia University

New York, N.Y. 10027

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ABSTRACT

A time dependent two-dimensional MHD calculation for pinch experiments, previously developed at Columbia¹, is improved and modified so that very long calculations (until the plasma reaches a quasi-equilibrium state) are possible. The calculation method is presented, and the simulation results of the Jutphaas screw pinch SP-1 and the Los Alamos reverse-field pinches ZT-I, ZT-S and ZT-40 are studied.

A set of single fluid resistive MHD equations are solved using standard alternate direction implicit method in an Eulerian difference scheme. Initial conditions correspond to the start of the main discharge in the experiment, with a homogeneous fully ionized plasma at rest. The equations are driven by time dependent boundary conditions : the toroidal magnetic field and the poloidal flux at the wall. The poloidal flux is indirectly extrapolated from the prescribed total toroidal plasma current. Most of other boundary values both at the wall and at the coordinate origin are obtained from the conservation laws.

The anomaly of the electric resistivity is studied by comparing the experimental results with the calculated results of different resistivity coefficients. The initial setting up phase of ZT-I and ZT-S are well simulated using a semi-empirical anomalous resistivity algorithm due to R. Chodura. In the later quasi-equilibrium phase however,

the resistivity appears to be classical.

The toroidal curvature effects are automatically obtained in this two-dimensional calculation. The calculation shows a strong toroidal plasma flow with an antisymmetric toroidal velocity profile against the midplane of the torus. In the later phase, this toroidal flow is balanced and sustained by the azimuthal asymmetry of the dissipative effects due to toroidal curvature, thus resulting in a steady-state quasi-equilibrium. The physical origin and the consequences of this toroidal flow are discussed (e.g., the toroidal shift of the magnetic axis is nearly doubled due to this flow).

A simulation predicting ZT-40 behavior is presented. The basic differences of ZT-40 from smaller ZT-I and ZT-S are discussed from both physical and numerical view points. The scaling law of the dissipative MHD equations is derived. The important time scales, i.e., Alfvén, field diffusion, thermal diffusion, viscous dissipation and numerical viscous dissipation time scales, are studied.

The overall good agreement between experiment and calculation presented assures that the plasma model used (i.e., single fluid dissipative MHD) gives a good representation of the physical phenomena, at least in the parameter regime of the smaller experiments SP-1, ZT-I and ZT-S.

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I. INTRODUCTION

It is well known that the efficiency of a magnetic fusion reactor will largely depend on the plasma β , since fusion power density goes as β^2 . Thus, much effort has been concentrated on the attainment of high- β plasma confinement. For tokamaks, which currently operate at $\beta \lesssim 1\%$, there are plans to increase the plasma β by a fast auxiliary heating (e.g., neutral beam injection, rf heating, etc.) as in the flux conserving tokamak³ concept, or by elongation of the poloidal cross section.

Meanwhile, stable high β ($\sim 50\%$) plasmas have already been attained in various toroidal pinch experiments (e.g., reverse-field pinch, screw pinch, belt pinch, etc.). But the current pinch machines are generally quite small devices of short duration and thus, the obtained plasma parameters are quite low in view of the ignition temperature and the Lawson criterion. Therefore, larger pinch experiments are planned in several laboratories around the world to scale up the plasma parameters. The scaling character of the pinch discharges is one of the crucial-questions to be investigated by these larger devices.

In this thesis, we study the plasma behavior of the toroidal pinch experiments using a two-dimensional resistive magneto-hydrodynamic (MHD) calculation. Lui and Chu¹ reported a two-dimensional MHD simulation of the initial phase of the Jutphaas SP-1 screw pinch.⁴ In this, the single-fluid MHD equations with finite electrical and thermal conductivities were solved by an Eulerian finite difference scheme in a toroidally

axisymmetric geometry. The plasma was initially at rest, and it was driven by currents and fields prescribed as functions of time. The gross motions of the plasma, namely pinching, bouncing and toroidal shift, were all obtained as part of the solution. In the present study, this code has been modified, so that very long calculations beyond the initial phase are possible.² An initially quiet homogeneous plasma, driven by rising external currents which are then crowbarred, can be followed through many pinching and bouncing motions until a quasi-equilibrium dissipation state is reached. The plasma behavior during this process is discussed in Section (4.1). The method of the calculation is described in Chapters (II and III). (In Sections (2.1, 3.1, and 3.2.D), we follow Lui and Chu¹.)

The same code is also used to simulate the Eta-Beta reverse-field pinch at the University of Padua and ZT-I, ZT-S, and ZT-40 reverse-field pinches⁵ at Los Alamos. (The Eta-Beta results have been reported in ref. (6).) The ZT-40 device which is currently being constructed, is a larger device compared to the others mentioned here. Comparing the ZT-40 results with others, the scaling behavior of the pinch discharge is studied in Section (4.4). For this, the scaling law of the MHD equations is derived in Section (4.4.A). In Section (4.4.C), we discuss some general confinement characteristics of large pinch experiments.

A strong toroidal flow is seen in the calculation. This flow is sustained while the poloidal flow is damped until it becomes

a diffusion velocity. In other words, the quasi-equilibrium state reached is a steady state rather than a static quasi-equilibrium state. This and other toroidal curvature effects are discussed in Section (4.3.B).

The question of the anomaly of the electric resistivity is also studied. For this, the observed field profiles are compared using two separate simulation results, one using the classical resistivity and the other using an anomalous resistivity algorithm (the Chodura algorithm described in Section (2.3)). In the setting-up phase of the discharge, the anomalous algorithm gives somewhat better results than the classical resistivity does. But the difference of the two results is not very significant and both results compare with the experiment quite well. In the later dissipation phase however, the anomalous resistivity algorithm is found not valid and the experimental resistivity seems to be quite close to the classical one. These are discussed in Sections (4.2 and 4.3.A).

We now briefly describe the toroidal reverse-field pinch. The reverse-field pinch (RFP) relies on wall and magnetic shear stabilization for its stability. The reversal of the toroidal magnetic field at the wall is necessary to have nonzero magnetic shear everywhere, when a vacuum region or a low current region exists outside the pinch. Due to shear stabilization, RFP operates with the safety factor $q < 1$.⁷ Thus, stable high β plasmas can be produced with an arbitrary aspect ratio device. (cf. The β -values of the tokamaks are limited by the two conditions, poloidal beta $\beta_p \lesssim$ aspect ratio $\frac{R}{a}$, and $q = \frac{aB_t}{RB_p} > \sim 1$.) Due to

the large toroidal current allowed, it may be also possible to heat the plasma to ignition by Joule heating alone.⁸

Figure 1 shows the basic geometry of the torus and the coordinate systems. The highly conducting copper shell is slotted poloidally and toroidally to permit field penetration. Inside, there is a dielectric discharge tube usually made of quartz or alumina. The imposed external toroidal (I_ϕ) and poloidal (I_θ) currents drive the plasma implosion. Figure 2(a) shows time traces of the driving boundary conditions of RFP in the calculation, i.e., the total toroidal plasma current I_{zp} [†] and the toroidal field at the wall, B_{zw} . These functions are directly related to the external currents (cf. Sections 3.2.B and 3.2.C). The plasma current I_{zp} is not strictly a boundary condition of the calculation, but rather it is used as a constraint for an actual boundary value, volt-seconds around the torus (cf. Section 3.2.B). Figure 2(b) shows typical toroidal (B_z) and poloidal field (B_θ) profiles. The direction of B_z is reversed at the wall.

There are two different methods of producing the RFP configuration, i.e., one by a fast field programming; the other from self reversal. The self reversal⁹ occurs due to a helical instability which relaxes the discharge to a minimum energy state as can be explained by Taylor's theory in ref. (10). Thus, to study the self-reversal mode, a three-dimensional calculation⁹ is necessary. On the other hand, the driven reversal mode is an axisymmetric process and thus, will be studied in the present thesis.

[†] Subscript z represents ϕ direction, not Z direction.

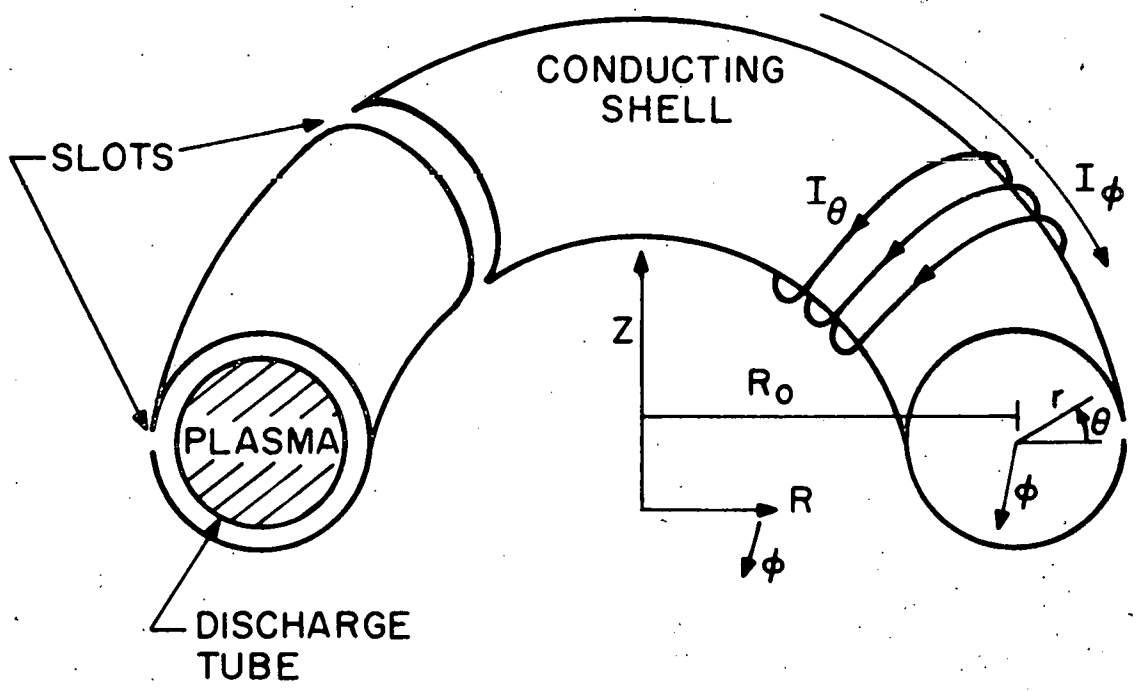


FIG. 1. BASIC GEOMETRY OF A TOROIDAL PINCH AND COORDINATE SYSTEMS

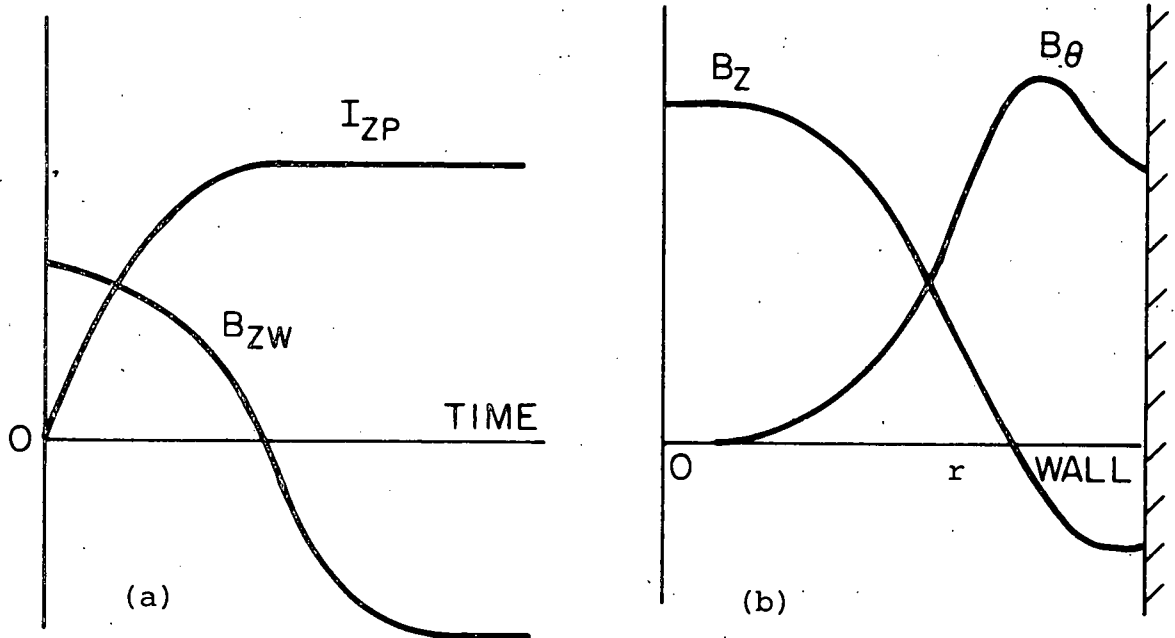


FIG. 2. TYPICAL REVERSE-FIELD PINCH CONDITIONS: (a) TIME TRACES OF THE TOROIDAL PLASMA CURRENT I_{zP} AND THE TOROIDAL MAGNETIC FIELD AT THE WALL, B_{zw} ; (b) PROFILES OF THE TOROIDAL FIELD B_z AND THE POLOIDAL FIELD B_θ .

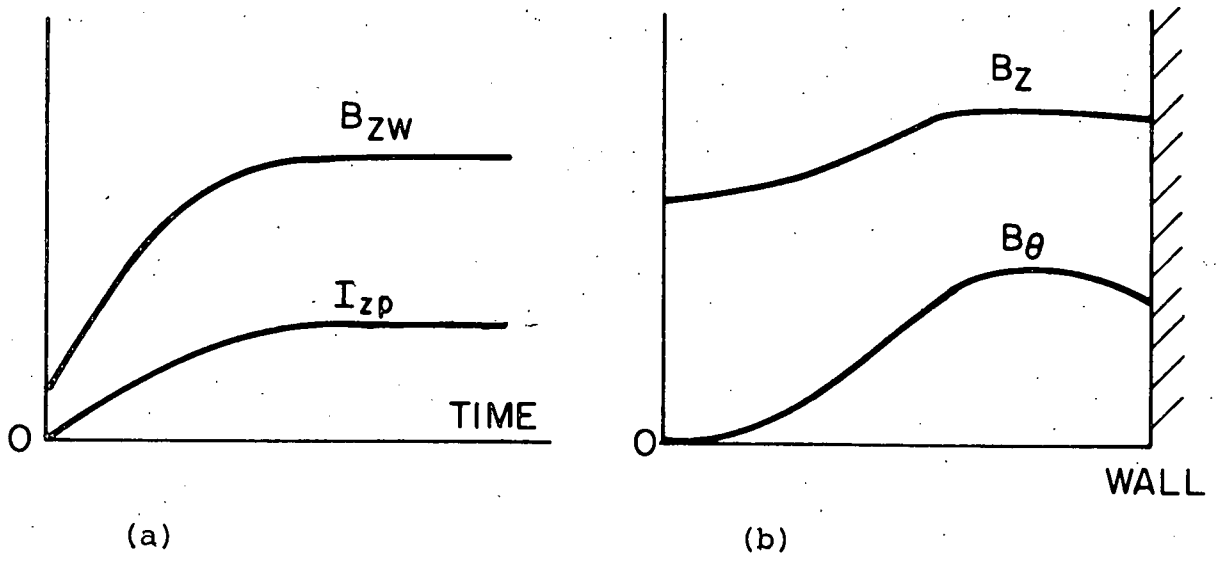


FIG. 2'. TYPICAL SCREW PINCH CONDITIONS.

The screw pinch is similar to RFP in geometry. But instead of the magnetic shear, screw pinch relies on the stabilizing effect of $q > \sim 1$, and on force-free currents.⁴ Due to force-free currents, some reduction of q -value is possible.⁴ Figure 2'(a) shows the driving boundary conditions of screw pinch in the calculation, and should be compared to Figure 2(a) of the RFP case. Typical field profiles are shown in Figure 2'(b). Unlike in the RFP case, the toroidal field contributes to plasma confinement. But this contribution will be only comparable to the poloidal field contribution (i.e., $\beta_p \sim 1$), even when $B_z \gg B_\theta$, especially in a large experiment, as explained in Section (4.4.C).

II. FORMULATION OF THE PROBLEM

The calculation will follow plasma behavior starting from its initial state at rest, through many pinching and bouncing motions until it reaches a quasi-equilibrium dissipation state (where the plasma oscillation has been damped away and thus, diffusion mechanisms determine the radial plasma velocity). The equations to be solved are derived in the following section. We adopt the single fluid resistive magnetohydrodynamic (MHD) model of the plasma. The usual criteria¹¹ for the applicability of the MHD equations are well satisfied in the smaller experiments, SP-1, ZT-I, and ZT-S, due to high density and low temperature in the discharges. In fact, the calculation results agree well with the experimental measurements as will be shown in Chapter IV. For the larger ZT-40 experiment, however, the mean free path will become comparable to the torus dimension and much larger than the ion gyroradius due to higher temperature, and thus, the applicability of the MHD equations is questionable. Therefore, the present model of the plasma should be regarded as a crude approximation in the ZT-40 case. (e.g., The neoclassical effect might become important.)

Toroidal symmetry is assumed. The implication and the validity of this assumption are discussed in Section (2.2). Finite electrical and thermal conductivities are included but the viscosity is neglected. These transport coefficients are explained in Section (2.3).

According to the geometry of the boundary (Figure 1), toroidal polar coordinates (r, θ, ϕ) are used throughout, rather than cylindrical coordinates (R, z, ϕ) . The distance R from the major axis, $R = R_0 + r \cos \theta$, is often used in the equation for brevity. The units are Gaussian unless otherwise specified.

2.1 Derivation of the Basic Equations

The differential equations of single fluid magnetohydrodynamics with finite electrical and thermal conductivities are

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \underline{u}) = 0, \quad (1a)$$

$$\frac{\partial \rho \underline{u}}{\partial t} + \nabla \cdot (\rho \underline{u} \underline{u}) = -\nabla p + \underline{j} \times \underline{B}/c, \quad (1b)$$

$$\rho C_v \left(\frac{\partial T}{\partial t} + \underline{u} \cdot \nabla T \right) = -p \nabla \cdot \underline{u} + j^2/\sigma + \nabla \cdot (K \nabla T), \quad (1c)$$

$$\frac{1}{c} \frac{\partial \underline{B}}{\partial t} = -\text{curl } \underline{E}, \quad (1d)$$

$$\underline{j} = \frac{c}{4\pi} \text{curl } \underline{B}, \quad (1e)$$

$$\text{div } \underline{B} = 0, \quad (1f)$$

$$\begin{aligned} \underline{E} &= \underline{j}/\sigma - \underline{u}_e \times \underline{B}/c \\ &= \underline{j}/\sigma - \underline{u} \times \underline{B}/c + \frac{\underline{j}}{ne} \times \underline{B}/c, \end{aligned} \quad (1g)$$

where σ and K are electric and thermal conductivities, C_v is the specific heat at constant volume, n is the electron number density ($n = \rho/M_{\text{atom}}$, for a hydrogen or deuterium filling gas),

and $\underline{u}_e = \underline{u} - \frac{\underline{j}}{ne}$ is the electron fluid velocity. (\underline{u}_e will be used in the equations for brevity instead of using $(\underline{u} - \frac{\underline{j}}{ne})$.) The Hall term is included in the Ohm's law (1g), because the drift velocity, $\underline{V}_d = \underline{j}/ne$, is often of the same order of magnitude as the fluid velocity \underline{u} . (In other words, the condition¹¹ $\frac{L\omega_{pe}^2}{\omega_{ce}} \frac{V_0}{c^2} \gg 1$ for neglecting the Hall term is not satisfied.)

We now derive the equations to be solved in (r, θ, ϕ) coordinates, assuming toroidal axisymmetry ($\frac{\partial}{\partial \phi} = 0$). Since $\text{div } \underline{B} = 0$, we define as usual a poloidal flux function ψ as

$$\psi(r, \theta) = \frac{1}{2\pi} \int_{F(r, \theta)} \underline{B} \cdot d\underline{s} \quad (2)$$

where F represents any open surface bounded by a large circle around the torus at (r, θ) . Differentiating each side of equation (2), and using axisymmetry, we obtain the usual representation for \underline{B} ,

$$\underline{B} = -\frac{\hat{\phi}}{R} \times \nabla \psi + B_\phi \hat{\phi}.$$

The toroidal field B_ϕ is more conveniently represented by introducing a variable $\chi = RB_\phi$. Then each component of \underline{B} is represented by ψ , χ , and their derivatives, as

$$B_r = \frac{1}{Rr} \frac{\partial \psi}{\partial \theta}, \quad B_\theta = -\frac{1}{R} \frac{\partial \psi}{\partial r}, \quad B_\phi = \frac{\chi}{R}. \quad (3)$$

The current density components are also so expressed from Ampere's Law (1e),

$$j_r = \frac{c}{4\pi Rr} \frac{\partial \chi}{\partial \theta}, \quad j_\theta = -\frac{c}{4\pi R} \frac{\partial \chi}{\partial r},$$

$$j_\phi = -\frac{c}{4\pi} \left\{ \frac{1}{r} \frac{\partial}{\partial r} \left(\frac{r}{R} \frac{\partial \psi}{\partial r} \right) + \frac{1}{r} \frac{\partial}{\partial \theta} \left(\frac{1}{Rr} \frac{\partial \psi}{\partial \theta} \right) \right\}. \quad (4)$$

Integrating the induction equation (1d) over the surface $F(r, \theta)$, we obtain

$$-\frac{1}{cR} \frac{\partial \psi}{\partial t} = E_\phi. \quad (5)$$

Substituting equations (3, 4, and 5) into Ohm's Law (1g), we obtain the differential equation for ψ ,

$$\begin{aligned} \frac{\partial \psi}{\partial t} + u_{er} \frac{\partial \psi}{\partial r} + u_{e\theta} \frac{\partial \psi}{r \partial \theta} \\ = \frac{c^2}{4\pi \sigma} \left\{ \frac{R}{r} \frac{\partial}{\partial r} \left(\frac{r}{R} \frac{\partial \psi}{\partial r} \right) + \frac{R}{r} \frac{\partial}{\partial \theta} \left(\frac{1}{R} \frac{\partial \psi}{r \partial \theta} \right) \right\}. \end{aligned} \quad (6a)$$

Substituting equations (3 and 4) first into Ohm's Law (1g) and then into the induction equation (1d), we obtain the equation for χ

$$\begin{aligned} \frac{\partial \chi}{\partial t} = \frac{c^2}{4\pi} \left\{ \frac{R}{r} \frac{\partial}{\partial r} \left(\frac{r}{\sigma R} \frac{\partial \chi}{\partial r} \right) + \frac{R}{r} \frac{\partial}{\partial \theta} \left(\frac{1}{\sigma Rr} \frac{\partial \chi}{\partial \theta} \right) \right\} \\ - \frac{R}{r} \frac{\partial}{\partial r} \left(\frac{r}{R} u_{er} \chi \right) - \frac{R}{r} \frac{\partial}{\partial \theta} \left(\frac{1}{R} u_{e\theta} \chi \right) \\ + \frac{R}{r} \frac{\partial}{\partial r} (r B_r u_{e\phi}) + \frac{R}{r} \frac{\partial}{\partial \theta} (B_\theta u_{e\phi}). \end{aligned} \quad (6b)$$

The energy equation (1c) gives the equation for temperature T which is of the same form in these coordinates,

$$\begin{aligned}
\rho C_v \left(\frac{\partial T}{\partial t} + u_r \frac{\partial T}{\partial r} + u_\theta \frac{\partial T}{r \partial \theta} \right) = \\
\frac{j_r^2 + j_\theta^2 + j_\phi^2}{\sigma} + \frac{1}{Rr} \frac{\partial}{\partial r} (KRr \frac{\partial T}{\partial r}) \\
+ \frac{1}{Rr} \frac{\partial}{\partial \theta} (KR \frac{\partial T}{r \partial \theta}) \\
- (\gamma-1) \rho C_v T \left\{ \frac{1}{Rr} \frac{\partial}{\partial r} (Rr u_r) + \frac{1}{Rr} \frac{\partial}{\partial \theta} (Ru_\theta) \right\}, \quad (6c)
\end{aligned}$$

where the ratio of specific heats γ is 5/3 for three degrees of freedom for each particle. Equations (6a, 6b, and 6c), which govern the time evolution of ψ , χ and T form a group of three parabolic equations. On the other hand, the mass and momentum equations, (1a) and (1b), form an essentially hyperbolic system* of four equations. These are in (r, θ, ϕ) coordinates

$$\begin{aligned}
\frac{\partial}{\partial t} \rho + \frac{1}{r} \frac{\partial}{\partial r} (r \rho u_r) + \frac{\partial}{r \partial \theta} (\rho u_\theta) \\
= - \frac{\rho (u_r \cos \theta - u_\theta \sin \theta)}{R}, \quad (7a)
\end{aligned}$$

$$\begin{aligned}
\frac{\partial}{\partial t} (\rho u_r) + \frac{1}{r} \frac{\partial}{\partial r} r (\rho u_r^2 + (\gamma-1) \rho C_v T) + \frac{1}{r} \frac{\partial}{\partial \theta} (\rho u_r u_\theta) \\
= - \frac{\rho u_r}{R} (u_r \cos \theta - u_\theta \sin \theta) + \frac{\rho u_\theta^2}{r} + \frac{\rho u_\phi^2 \sin \theta}{R} \\
+ \frac{1}{c} (j_\theta B_\phi - j_\phi B_\theta), \quad (7b)
\end{aligned}$$

* Although in actuality, these 4 equations together with the 3 parabolic equations (6) form a parabolic system of 7 equations since they are coupled together, we will refer to these 4 equations as a hyperbolic system for convenience.

$$\begin{aligned}
& \frac{\partial}{\partial t} (\rho u_\theta) + \frac{1}{r} \frac{\partial}{\partial r} (\rho r u_r u_\theta) + \frac{1}{r} \frac{\partial}{\partial \theta} (\rho u_\theta^2 - (\gamma-1) \rho C_V T) \\
& = - \frac{\rho u_\theta}{R} (u_r \cos \theta - u_\theta \sin \theta) - \frac{\rho u_r u_\theta}{r} - \frac{\rho u_\theta^2}{R} \sin \theta \\
& \quad + \frac{1}{c} (j_\phi B_r - j_r B_\phi) , \tag{7c}
\end{aligned}$$

$$\begin{aligned}
& \frac{\partial}{\partial t} (\rho u_\phi) + \frac{1}{r} \frac{\partial}{\partial r} (\rho r u_r u_\phi) + \frac{1}{r} \frac{\partial}{\partial \theta} (\rho u_\theta u_\phi) \\
& = - \frac{2\rho u_\phi}{R} (u_r \cos \theta - u_\theta \sin \theta) + \frac{1}{c} (j_r B_\theta - j_\theta B_r) . \tag{7d}
\end{aligned}$$

For convenience in later explanation, we represent these four hyperbolic equations as a vector equation,

$$\begin{aligned}
& \frac{\partial \underline{W}}{\partial t} + \frac{1}{r} \frac{\partial}{\partial r} (r \underline{A}(\underline{W}, T) \cdot \underline{W}) + \frac{\partial}{r \partial \theta} (\underline{B}(\underline{W}, T) \cdot \underline{W}) \\
& = \underline{S}(\underline{W}, T, r, \theta) + \underline{F}(\underline{j}, \underline{B}) , \tag{8}
\end{aligned}$$

where $\underline{W} = (\rho, \rho u_r, \rho u_\theta, \rho u_\phi)$, \underline{A} and \underline{B} are 4×4 matrices,

$$\underline{A} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ (\gamma-1)C_V T & u_r & 0 & 0 \\ 0 & 0 & u_r & 0 \\ 0 & 0 & 0 & u_r \end{bmatrix}, \quad \underline{B} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & u_\theta & 0 & 0 \\ (\gamma-1)C_V T & 0 & u_\theta & 0 \\ 0 & 0 & 0 & u_\theta \end{bmatrix} ,$$

and \underline{S} contains all the undifferentiated terms involving \underline{W} , T , and the coordinate variables and represents the curvature effects, while \underline{F} represents $\underline{j} \times \underline{B}/c$ term.

2.2 A Remark on the Toroidal Axisymmetry

Toroidal axisymmetry is a basic assumption in our study. If the initial conditions and the driving boundary conditions are all axisymmetric, the plasma will develop in an axisymmetric way, unless axially asymmetric instabilities occur in which case our calculation is no longer applicable. While most of the conditions are rather obviously axisymmetric, the fact that the electric field which penetrates through the slots also gives axisymmetric effect on the plasma, would require some explanation, since this E field itself is much asymmetric in fact.

In general, the E field can be represented uniquely as a superposition of two fields,¹³

$$\underline{E} = \underline{E}_B + \underline{E}_\sigma,$$

where \underline{E}_B is a divergence free field generated by $\text{curl } \underline{E}_B = -\frac{\partial \underline{B}}{c \partial t}$, while \underline{E}_σ is a curl free field generated by $\text{div } \underline{E}_\sigma = 4\pi\sigma$. In our case, it is easy to see that \underline{E}_B is axisymmetric while \underline{E}_σ is axially asymmetric. Thus, at the conducting wall (inside surface of the shell), $E_{B\phi}$ is constant around the torus. $E_{\sigma\phi}$ is also constant as long as one stays away from the slot, since $E_{\sigma\phi} = -E_{B\phi}$ at the wall to satisfy the boundary condition $E_\phi = 0$ at a conducting surface. However at the slot, $E_{\sigma\phi}$ becomes very large representing large Poynting's vector into the torus. This asymmetric \underline{E}_σ field however, will have little effect on the plasma behavior, since the plasma will react immediately to shield out the curl free \underline{E}_σ field by charge separation on the plasma surface.

More precisely, \underline{E}_σ generated by σ on both conducting shell surface and the plasma surface will be such that $E_{\sigma\phi} = -E_{B\phi}$ on the conducting shell surface (both inside and outside) away from the slot, $\underline{E}_\sigma = 0$ inside the plasma, $E_{\sigma\phi} = 0$ on the plasma surface and

$$\int_0^{\Delta x} E_{\sigma\phi} \cdot dx = \frac{E_{B\phi} 2\pi R}{N} \quad \text{at the slot} (\Delta x \text{ is the width of the slot, } N \text{ the number of slots}).$$

It is easy to verify that these conditions together with the fact that the total charge in the plasma and the total charge in each shell segment are all zero, determine a unique solution of σ . We note that the charge density $\sigma(\underline{r}, t)$ specified in this argument, together with the solution of the problem posed in Section (2.1), gives a solution of the actual problem in the MHD limit.

In short, the \underline{E} field inside the plasma will remain axisymmetric although the \underline{E} field outside the plasma is highly asymmetric. This fact has been explained concerning the poloidal slots. The effect of the toroidal slots can also be explained in a similar way. These are also the reason why we can use later, two seemingly contradictory boundary conditions, $B_n = 0$ and $E_\phi, E_\theta \neq 0$ on the conducting surface. In fact, the experimental measurements in nearly all of the axisymmetric toroidal experiments, show close toroidal symmetry of the plasma behavior.

2.3 Transport Coefficients

The classical transport coefficients¹⁴ which are derived from binary encounter theory sometimes do not satisfy the experimental measurements. Although other reasons may well be important,¹⁵ this is generally attributed to turbulence created by plasma instabilities. Thus, some simulation studies on toroidal pinches reported the transport to be significantly anomalous while other studies on different toroidal pinches reported the transport to be classical. One anomalous case was reported by Krall and Liewer¹⁶ who used an electric resistivity algorithm where the effects of various instabilities are taken into account self-consistently, using the known properties of the instabilities. The results thus obtained agreed with the theta pinch experiment at the University of Texas. A more simple semi-empirical resistivity algorithm has been also reported to describe satisfactorily the behavior of the plasma in the Garching theta pinch (by Chodura¹⁷) and ZT-1 reverse field pinch (by Sgro and Nielson¹⁸), when it is used in a hybrid model (Vlasov ions, fluid electrons) of the plasma in one dimensional calculations.

On the other hand, the Culham group reported that the HBTX-1 reverse-field pinch had been successfully simulated using the classical transport coefficients.¹⁹ Another study had been done by the University of Padua group who reported that "No clear indication about the classical or anomalous

nature of the resistivity appears from the simulated gross plasma behavior of Eta-Beta' reverse-field pinch discharge."²⁰

In the present study, we test both the classical resistivity formula and the anomalous resistivity algorithm of Chodura¹⁷, comparing the obtained results with the experiments. In the Chodura anomalous resistivity algorithm, the resistivity is found as follows:

$$v = c\omega_{pi}[1 - \exp(-|V_d|/fV_s)],$$

$$\eta = \text{Max}(vM_e/ne^2, \eta_{\text{classical}}), \quad (9)$$

where the ion acoustic speed $V_s = (\gamma kT_e/M_i)^{1/2}$, the drift velocity $V_d = j/ne$, and the coefficients $c = 0.5$ and $f = 5.0$. These same numerical coefficients are used in the present study. This algorithm gives a classical resistivity when the ratio of the electron drift velocity to the ion acoustic speed is small, but an empirically enhanced value will be switched on when the ratio becomes large. The density dependence of this formula is also important especially in the low density region outside the main plasma column, since the classical formula does not have much dependence on density. Here, we note that the resistivity η is used as a scalar. Some efforts were made initially, to use this resistivity coefficient in a tensor form as used by Sgro and Nielson¹⁸. But test runs indicate that the anisotropy of the resistivity is not important enough in our simulation to warrant the necessary complications and loss in numerical stability.

It is shown in Section (4.2) and (4.3) that for the initial setting up phase of the ZT-I and ZT-S experiments, the Chodura anomalous algorithm gives somewhat better results than the pure classical formula. But the difference is not very significant and both results agree with the experiment quite well so that a conclusion similar to that of the Padua group stated above is drawn. It is noted that the Eta-Beta experiment is in a similar parameter range as ZT-I and ZT-S experiments. In the later dissipation phase however, the Chodura resistivity algorithm is found not valid and the resistivity seems to be quite close to the classical one.

As for the thermal conductivity, if the classical formula is used, the results did not compare with the experiment. This is because the classical formula gives no conduction across the wall, as explained in the following, while the experiment shows significant heat loss. In the tenuous plasma region outside the pinched main plasma column (cf. Section 3.3), the plasma density is negligible while the magnetic field and temperature are substantial. Thus, $\omega\tau \rightarrow \infty$, where ω and τ are the cyclotron frequency and the collision time, respectively. When $\omega\tau \rightarrow \infty$, the perpendicular conductivity¹⁴ becomes $K_{\perp} \sim K_{\perp}^{\perp} \rightarrow \frac{2nk^2 T_i \tau_i}{m_i (\omega\tau)_i^2}$, where the subscript i represents the ion species. Since this value will be negligible near the wall and the magnetic field is parallel to the wall, there will be zero net energy loss

from the torus system in the calculation. On the other hand, in the experiments, the energy loss due to the radiation and conduction is generally significant. In some discharges (e.g., some of the Eta-Beta discharges)²², the radial loss was found negligible, even though the total energy loss was significant. Therefore, it seems that in general, some anomalous conduction loss is present together with the radiation loss.

In the present study, we simply set the thermal conductivity coefficient at a constant $K = 2 \times 10^6$ erg/cm · sec · °K (which corresponds to the classical parallel conductivity at ~ 4 eV), which was found satisfactory in our previous simulation of the Eta-Beta reverse-field pinch. This value is again found acceptable for the simulation of ZT-I, ZT-S and SP-1, all of which are in a similar parameter range as Eta-Beta. In our calculation where the radiation loss is neglected, this would imply that the net conduction loss in the calculation is on the same order of magnitude as the net energy loss in the experiment through both conduction and radiation.

The other transport mechanism, i.e., the viscosity, is neglected in the present calculation. The classical viscosity, is mainly due to the ion species and the parallel and perpendicular coefficients are on the same order of magnitude. The kinematic viscosity coefficient²¹ is roughly

$$\nu \approx \nu_i \sim \frac{kT_i \tau_i}{M_i}$$

For the smaller machines, i.e., SP-1, ZT-I, and ZT-S, the viscous

dissipation time scale $\tau_v = \frac{r_0^2}{\nu}$ is on the order of 1 ms. (It will be shown in Section (3.1) that the numerical viscosity is also on this order.) This is much longer than the time of interest in our calculation since the Alfvén time scale for these smaller machines is $\sim 1 \mu\text{s}$, and the field and heat diffusion times are both $\sim 10 \mu\text{s}$. (cf. Section (4.1)). This justifies the assumption of no viscosity in the calculation. On the other hand, in a larger ZT-40 experiment, the viscous dissipation time becomes $\sim 10 \mu\text{s}$ due to higher temperature, assuming the classical viscosity. Thus, the viscosity might be important in this case. A further discussion on this subject is made in Section (4.4.B).

2.4 Initial Conditions

Although our code can start from any initial condition, the initial conditions for the present simulation are very simple. We start the calculation from an ionized plasma at rest which corresponds to the beginning of the main discharge in the experiment. We assume initial full ionization. A given toroidal bias field $B_{\phi_{\text{bias}}}$ is present without any current. Therefore the initial conditions are

$$\rho = \text{Const.}, \quad \rho U_r, \quad \rho U_\theta, \quad \rho U_\phi = 0.$$

$$T = \text{Const. (typically } 1 \sim 10 \text{ eV depending on the pre-} \\ \text{ionization.)}$$

$$\psi = 0, \quad \chi = RB_{\phi_{\text{bias}}}.$$

III. NUMERICAL SOLUTION PROCEDURE

3.1 Basic Numerical Procedure

We use an Eulerian finite difference scheme to solve numerically the seven nonlinear equations (6 and 7) formulated in the previous chapter. The standard alternating direction implicit scheme¹² (implicit in r or θ direction, alternatively) is used for both parabolic (6) and hyperbolic equations (7). The solution procedure is explained in the following.

We first calculate the predicted value $\underline{W}^* = (\rho, \rho U_r, \rho U_\theta, \rho U_\phi)^*$ for the coefficients of the nonlinear terms from \underline{W}^n of the old time step by an explicit scheme (as usual, n denotes the time step, and i, j denote the indices in the r and θ direction respectively, i.e.,

$$F_{i,j}^n = F(i\Delta n, j\Delta\theta, n\Delta t),$$

$$\begin{aligned} \underline{W}_{i,j}^* = & \underline{W}_{i,j}^n - \Delta t (H_r (A^n \cdot \underline{W}^n)_{i,j} + H_\theta (B^n \cdot \underline{W}^n)_{i,j}) \quad (10) \\ & + \Delta t (S_{i,j}^n + F_{i,j}^n) \end{aligned}$$

where, for any vector \underline{V} ,

$$\begin{aligned} H_r \underline{V}_{i,j} & \equiv (r_{i+1} \underline{V}_{i+1,j} - r_{i-1} \underline{V}_{i-1,j}) / r_i \Delta r, \\ H_\theta \underline{V}_{i,j} & \equiv (\underline{V}_{i,j+1} - \underline{V}_{i,j-1}) / r_i \Delta 2\theta. \end{aligned}$$

This \underline{W}^* is then used in the coefficients, e.g., as

$$U_{er}^* = U_r^* - \left(\frac{j_r}{n_e} \right)^n, \text{ to calculate the new time step}$$

values for the three variables, ψ^{n+1} , χ^{n+1} and T^{n+1} , using an alternate direction implicit scheme.

Finally, the new time step value \underline{W}^{n+1} is calculated by solving the hyperbolic equations (6) with an alternate direction implicit scheme. Here \underline{W}^* is used for the nonlinear coefficients and \underline{S} is approximated by $\frac{1}{2}(\underline{S}^n + \underline{S}^*)$, \underline{F} by $\frac{1}{2}(\underline{F}^n + \underline{F}^{n+1})$. The difference equation for \underline{W} , at the r direction implicit time step for example, is as follows:

$$\begin{aligned} \underline{W}_{i,j}^{n+1} + \Delta t H_r (A_{i,j}^* \underline{W}_{i,j}^{n+1}) \\ = \underline{W}_{i,j}^n - \Delta t H_\theta (B_{i,j}^n \underline{W}_{i,j}^n) \\ + \frac{1}{2} (\underline{S}^* + \underline{S}^n) + \frac{1}{2} (\underline{F}^n + \underline{F}^{n+1}). \end{aligned} \quad (11)$$

This difference scheme, together with the boundary conditions explained later in this chapter, results in a 4 x 4 block tridiagonal matrix equation $\underline{P} \underline{x} = \underline{f}$, where \underline{P} and \underline{f} are given. Likewise in the solution procedure for ψ , χ and T , we encounter a similar tridiagonal matrix equation. These matrix equations are efficiently solved by Crout reduction.

Of course, if we iterate back and forth between the parabolic and the hyperbolic equations, the results will be more accurate than the results from the present method. But this would require much more computer time.

To enhance numerical stability, a Lapidus type numerical viscosity term ^{12,24} is added in the right side of the equation (11). This viscosity term is

$$\begin{aligned} & \kappa \left\{ \frac{\Delta t}{\Delta r} \delta_r (|\delta_r U_r^n| |\delta_r W^n)_{i,j} \right. \\ & \left. + \frac{\Delta t}{r_i \Delta \theta} \delta_\theta (|\delta_\theta U_\theta^n| |\delta_\theta W^n)_{i,j} \right\} \end{aligned} \quad (11+)$$

where

$$\delta_r V_{i,j} \equiv \frac{V_{i+1/2,j}}{2} - \frac{V_{i-1/2,j}}{2},$$

$$\delta_\theta V_{i,j} \equiv \frac{V_{i,j+1/2}}{2} - \frac{V_{i,j-1/2}}{2},$$

and the coefficient of the viscosity term κ is typically 0.1.

For the velocity of $10^5 \sim 10^6$ cm/sec and $\Delta r, r\Delta\theta \sim 1$ cm, we estimate the effective viscous dissipation time $\tau_v = \frac{r_0^2}{\nu}$

of this dissipation term to be 100 \sim 1000 μ s. Since, this time scale is considerably longer than the time scale of the calculation (cf. Chapter IV), this artificial viscosity term can be justified.

We note here that using the predictor value \underline{W}^* as given in equation (10) sometimes causes a numerical instability in the simulation of a very violent process. In such a case, we use a slightly less accurate procedure which gives better numerical stability, that is, we simply find \underline{W}^* without the predicting terms in equation (10) but, adding a dissipation

term, i.e.,

$$\underline{W}^* = (1 - \theta) \underline{W}_{i,j}^n + \frac{1}{4} \theta (\underline{W}_{i+1,j}^n + \underline{W}_{i-1,j}^n + \underline{W}_{i,j+1}^n + \underline{W}_{i,j-1}^n), \quad (10')$$

where the dissipation constant $\theta = 0.1$.

Due to the complex nonlinearity of the equations, a rigorous overall error analysis for the solution of the difference scheme described in this section would be exceedingly difficult, if not impossible. Therefore, we took the usual approach of comparing results of the code obtained with different grid sizes and time steps. If the solutions obtained under these conditions are converging to a solution, it is assumed that the solution obtained is correct. This is because the truncation errors are at least first order both in the grid size and in the time step. In this way (e.g., by noticing that the results do not change significantly when the grid and time step sizes are reduced to a half), it is also checked that the effective dissipation time due to the overall numerical dissipation is much longer than the calculation time scale which is in the order of 10 μ s.

For the boundary conditions of the system of seven differential equations (6) and (7), we first consider the dissipation-free case, where the dissipation coefficients are all zero. The boundary conditions for these hyperbolic equations can be determined by examining the characteristics²⁵. Since there is no plasma flow across the wall and also since we require that no vacuum region appears near the wall (cf. Section (3.3)), a boundary condition $U_r = 0$ is necessary. This condition $U_r = 0$ and the fact that the magnetic fields are parallel to the wall, allow only one characteristic (a fast (magnetosonic) wave) to point into the domain. Therefore, $U_r = 0$ is a sufficient boundary condition for this case. (If $U_r < 0$, all six boundary values except the one for the density ρ would be necessary.)

On the other hand, for the present dissipative magneto-hydrodynamic equations with finite electric and thermal dissipation coefficients, the system become parabolic and additional boundary conditions for the variables ψ , χ and T are necessary. For these variables, either a Dirichlet boundary condition (e.g., $T = \text{Const.}$) or a Neumann boundary condition (e.g., $\frac{\partial T}{\partial r} = 0$) for each variables, is sufficient both for the differential equations and for the difference equations. The boundary conditions used are explained in detail in Section (3.2.B).

Furthermore, for the variables $\underline{W} = (\rho, \rho U_r, \rho U_\theta, \rho U_\phi)$, the difference scheme may need additional extraneous boundary conditions, and careless imposing of these conditions may cause various difficulties, e.g., numerical boundary layers,

numerical instability, etc. (cf. Ref. (26)). In the following section, these extraneous boundary conditions for \underline{W} are derived from the constraint that there is no mass and momentum flow across the wall. For this procedure, we should first define the mass and momentum flow between computational grids, consistently with the difference scheme used. This is also shown in the following section.

The boundary conditions for the θ direction are given in Section (3.2.D). Finally the treatment in the coordinate origin $r = 0$ is discussed in Section (3.2.E). This procedure is necessary because the origin $r = 0$ is a boundary point for the coordinates (r, θ, ϕ) even though it is an interior point of the plasma.

3.2.A. Interpretation of the Basic Difference Scheme and the Boundary Conditions

We first explain the definition of the mass flow between computational grids, using a one dimensional analogy of the mass conservation equation, that is,

$$\rho^{n+1} - \rho^n + \frac{\Delta t}{2\Delta x} (\rho U_{i+1}^{n+1} - \rho U_{i-1}^{n+1}) = 0 \quad (12)$$

Fig. 3 shows the computational mesh scheme in one dimensional analogy. (The two dimensional case is shown in Fig. 4 on page ().) All the variables ρ and U are defined at the integral points I 's. Naturally, the mass inside the unit grid I is defined as $\rho_i \Delta x$.

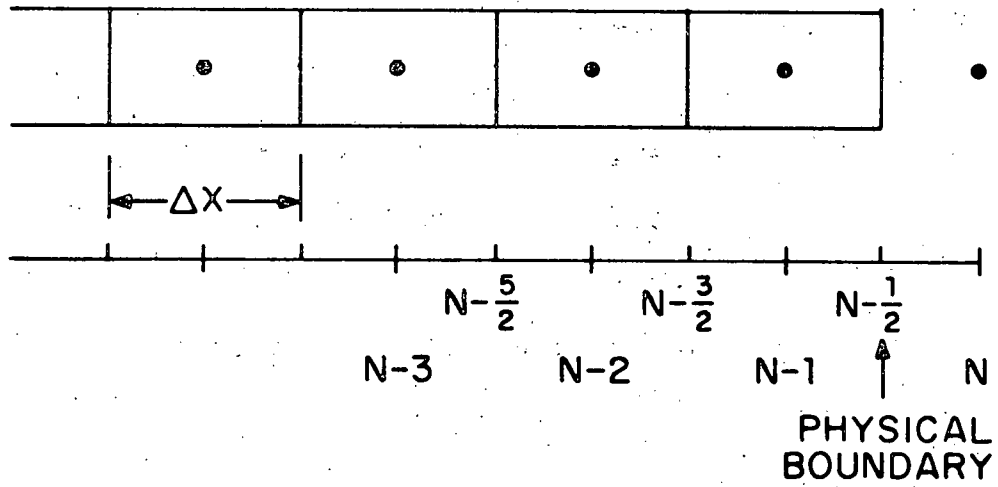


FIG. 3, COMPUTATIONAL MESH SCHEME IN ONE DIMENSIONAL ANALOGY.

If we calculate the mass change in the grid N_1 through N_2 using equation (12),

$$\begin{aligned}
 \Delta M &= \Delta t \sum_{i=N_1}^{N_2} \frac{1}{2} (\rho U_{i-1} - \rho U_{i+1}) \\
 &= \Delta t \left(\frac{\rho U_{N_1} + \rho U_{N_1}}{2} - \frac{\rho U_{N_2} + \rho U_{N_2+1}}{2} \right) \\
 &= \Delta t \left((\rho U)_{N_1 + \frac{1}{2}} - (\rho U)_{N_2 + \frac{1}{2}} \right) \quad (13)
 \end{aligned}$$

Thus, the mass flux density ρU through the point $N_2 + \frac{1}{2}$ can be defined as

$$(\rho U)_{N_2 + \frac{1}{2}} = \frac{\rho U_{N_2} + \rho U_{N_2+1}}{2},$$

and the symmetry of this equation shows that the difference scheme (12) exactly conserves mass if the boundary conditions are accordingly defined. If we set the position of the actual physical boundary at the point $N - \frac{1}{2}$ (Fig. 3), the zero mass flow condition at the wall, i.e., $\rho U_{N - \frac{1}{2}} = \frac{\rho U_{N-1} + \rho U_N}{2} = 0$ gives the boundary condition

$$(\rho U)_N = -(\rho U)_{N-1} \quad (14a)$$

Such a specification would conserve total mass exactly.

From a similar reasoning for the momentum equation, the zero momentum flow condition determines another boundary condition,

$$(\rho U U)_N = -(\rho U U)_{N-1} \quad (14b)$$

Although setting the boundary conditions as $\rho_N = -\rho_{N-1}$ and $U_N = U_{N-1}$ would satisfy the above two conditions, the condition $\rho_N = -\rho_{N-1}$ can not be used due to the ∇P term in the momentum equation. (The boundary condition for the ∇P term will be explained later in this section.)

Therefore, we can either use the boundary conditions (14a) and (14b) incorporating them directly into the solution procedure, or we can subtract at each time step, the artificial flow passing through the physical boundary point $N-\frac{1}{2}$, that is $\Delta t \rho U_{N-\frac{1}{2}} = \Delta t \frac{\rho U_{N-1}}{2}$, if U is set at zero for example in the mass $\frac{1}{2}$ equation. These two methods yield nearly identical results.

In an analogous way to the above, we can easily find the boundary conditions for the difference equation (11) in our two dimensional case. (The mesh scheme is shown in Fig. 4.) For example, the mass flux density through $r_{i+\frac{1}{2}}$ in r direction is defined as

$$\frac{(\rho U r)_i + (\rho U r)_{i+1}}{r_i + r_{i+1}} \quad (15)$$

From this equation and the similar equations for the ρU_r , ρU_θ and ρU_ϕ flux densities, we find the boundary conditions,

$$\begin{aligned} (r\rho U_r)_N &= -(r\rho U_r)_{N-1}, \\ (r\rho U_r U_r)_N &= -(r\rho U_r U_r)_{N-1}, \\ (r\rho U_r U_\theta)_N &= -(r\rho U_r U_\theta)_{N-1}, \\ (r\rho U_r U_\phi)_N &= -(r\rho U_r U_\phi)_{N-1}, \end{aligned} \quad (16)$$

where N denotes the boundary index (i_{\max} in Fig. 4) for the r direction, and the zero ϕ momentum flow condition across the wall is used for the fourth equation, for example.

The necessity of these 'correct' boundary conditions can be seen, in a one dimensional analogy (Fig. 3), from a case where the boundary value $(\rho U)_N$ is given incorrectly as $(\rho U)_N = 0$ which seems to represent zero mass flow, therefore often used. Then, because of the artificial mass flow through the wall, i.e., $(\rho U)_{N-\frac{1}{2}} = \frac{1}{2}(\rho U)_{N-1}$, twice the mass in the cell $(N-1)$ will be swept in at each inward pinching of the plasma. (Here $(\rho U)_{N-1} \approx (\rho U)_{N-2}$ is used.) This error is quite severe in a polar mesh scheme (Fig. 4) for a circular cross section torus. For example, with 16 grid points in the r direction, this error will cause an artificial increase of $\sim 15\%$ of the

total plasma mass at the end of the first pinching. This large error is due to the fact that this error is of zeroth order in Δr . We note that the boundary conditions (16) do not take into account the toroidal curvature. But this error is of first order in Δr with very small coefficients. This method of treatment of mass and momentum flow will also be used for the treatment of the low density region described in Section (3.3).

For the hyperbolic equations (7), the boundary condition for ∇P term, $\nabla(\gamma-1)C_V\rho T$, still remains to be determined. This will be determined as before, from the meaning of the term. From a similar reasoning as in the mass flow case, the pressure at the wall is represented by

$$(\rho T)_{N-\frac{1}{2}} = \frac{(\rho T)_N + (\rho T)_{N-1}}{2}.$$

But unlike in the previous situation, there is no constraint which can be used to determine the boundary value exactly.

This time we can simply set the boundary condition as $(\rho T)_N = (\rho T)_{N-1}$, allowing an error of the first order in Δr . Obviously, a small change of pressure at the wall would have little effect on the system as a whole.

Using similar arguments, the necessary boundary conditions of the fluid velocities which appear in the parabolic equations (6), can be determined as

$$\begin{aligned} (U_r, U_\theta, U_\phi)_N &= (U_r, U_\theta, U_\phi)_{N-1}, \\ (U_{er}, U_{e\theta}, U_{e\phi})_N &= (U_{er}, U_{e\theta}, U_{e\phi})_{N-1}. \end{aligned} \quad (17)$$

3.2.B. Boundary Conditions at the Wall

Many of the boundary conditions at the wall for the basic equations (6 and 7) are already determined in the previous section. The only remaining boundary conditions to be determined are the ones for ψ , χ and T for the parabolic equations (6).

For the temperature equation, the boundary value of the temperature T_w , is set at a constant which is typically 10^{10} ev depending on preionization. The condition $T_w = T_{w-1}$ can also be used if an adiabatic wall is assumed.

The wall values of ψ and χ are both pure functions of time. Let I_θ be the total external poloidal current passing through the hole of the torus (cf. Fig. 1). Then, from Ampere's law,

$$\chi_w(t) = 2I_\theta(t)/c \quad (18)$$

Since the function $I_\theta(t)$ is readily measurable and programmable in the experiment, the boundary value $\chi_w(t)$ is well determined by this equation.

Due to the assumption of perfect conductivity for the outer shell, ($B_n = 0$), ψ is a constant at the wall at a given time. This boundary value ψ_w is proportional to the total magnetic flux through the hole of the torus (or to the volt-seconds around the torus). The flux in the hole is the algebraic sum of the primary transformer flux and the flux induced by the

plasma current and the current on the outer shell. Therefore in general, $\psi_w(t)$ is a complicated function of time. For example, it will depend on the plasma motion, imploding or expanding, and the plasma position. Hence, it is very difficult to predict or program $\psi_w(t)$ in the experiment. On the other hand, the total toroidal plasma current $I_{\phi p}(t)$ (which is a sufficient boundary condition for an infinite aspect ratio case, but cannot be used directly as a boundary condition in our two dimensional case of a finite aspect ratio torus), is readily measurable and programmable in experiment. Therefore, for the simulation of given experiments, we use $\psi_w(t)$ as a boundary condition in such a way that the resulting total plasma current $I_{\phi p}(t)$ matches the experimental measurements. The value of $\psi_w(t)$ is adjusted at each time step to give this result. This is done by extrapolating the ratio of the increments of ψ_w and $I_{\phi p}$ at the previous time step to the new time step values.

The reason for the previously stated fact that $I_{\phi p}(t)$ is a simple function of time without much dependence on the plasma motion or position, and also that it is easily predictable and programmable, will be found in the next section where the relationship between $\psi_w(t)$ and $I_{\phi p}(t)$ is explained.

3.2.C. Flux in the Hole and the Total Toroidal Plasma Current

In this section, we study the relation between flux in the hole of the torus and total plasma current to explain further the necessity of adopting the latter as an effective boundary condition. It is noted here that all the equations derived in this section except equation(22) are not actually used in our calculation since the circuit equation is not incorporated. However in cases where the incorporation of the circuit equation become necessary, this could be easily done using the equations in this section.

The magnetic flux in the hole of the torus, $F_w = 2\pi\psi_w$, can be expressed as a linear function of the external toroidal circuit current $I_{\phi ex}$ and the total toroidal plasma current $I_{\phi p}$, with constant coefficients M and L which are determined by the physical geometry of the system alone,* independent of plasma motion or position, i.e.,

$$F_w = MI_{\phi ex} + LI_{\phi p}, \quad (19)$$

with constant coefficients M and L.

* B_θ and B_r outside the torus ($\text{div } \underline{B} = 0$ and $\text{curl } \underline{B} = 0$) at a given time, are determined by the unique solution of a problem posed by given values of $I_{\phi ex}$ and $I_{\phi p}$ and the boundary condition $B_n = 0$ on the outer surface of the shell. This solution is in turn, a superposition of the two solutions, one posed by a given value of $I_{\phi ex}$, $I_{\phi p} = 0$ and $B_n = 0$ on the shell, and the other by $I_{\phi ex} = 0$, a given value of $I_{\phi p}$ and $B_n = 0$ on the shell. These two solutions are each linear in $I_{\phi ex}$ and $I_{\phi p}$, respectively. Therefore, equation(19) results, with constant coefficients M and L. The linearity of the magnetic medium outside the torus is assumed, in this proof.

For a sufficiently well conducting plasma (more quantitatively) $R \ll \omega L/c$ where ω is the characteristic frequency of the system, R the effective total toroidal resistivity of the plasma and c the light speed), the left side of equation (19), i.e., the flux F_w will be negligible compared to the terms on the right side, during the time before the crowbar time $t_{cb} \sim 1/\omega$. (The procedure after the crowbar time will be explained later in this section.) This can be readily seen by noting that the flux

F_w is[†]

$$F_w \sim \int_0^t cR I_{\phi p} dt, \quad \text{where } t < t_{cb}.$$

Therefore, we obtain equation (19),

$$I_{\phi p} \approx - \frac{M}{L} I_{\phi ex}. \quad (20)$$

Applying similar reasoning to the external circuit, we find that the external current $I_{\phi ex}$ can be evaluated by introducing simple equivalent circuit equations whose effective inductance is a constant L_{eff} given by

$$L_{eff} = L_{ex} - \frac{M^2}{L}, \quad (21)$$

where L_{ex} is the self-inductance of the external primary loop circuit.

[†] Here, it is also assumed that the gap between the conducting wall and the surface of the plasma column is small compared to the major radius of the torus.

The external current $I_{\phi_{ex}}$ found in this way, determines the plasma current I_{ϕ_p} by equation (20). Therefore, the plasma current I_{ϕ_p} has little dependence on the plasma parameters such as its motion or position, and is easily programmable in the experiment. Of course, the actual measurement of the plasma current I_{ϕ_p} shows some residual oscillations which represent mainly the coupling effect to the plasma implosion or expansion phase due to finite conductivity of the plasma.

We note here that even though the two terms on the right side of equation (19), i.e., $I_{\phi_{ex}}$ and I_{ϕ_p} term, are programmable functions, the sum of these two functions, i.e., $F_w (=2\pi\psi_w)$ is not. This is not a contradiction, because F_w is represented as a sum of two big quantities with opposite signs, which becomes zero in the case of a perfectly conducting plasma.

We also note that using equation (19) (which is not an approximation but an exact equation), the circuit equations can be easily incorporated into the problem and solved simultaneously with the plasma equations, even in the present two dimensional case. Although this procedure would somewhat improve the accuracy of the solution, the difference would not be important in view of our rather crude modeling of the plasma.

Until now, we have been concerned with the boundary condition ψ_w before the crowbar time. After the crowbar time, the method of following the plasma current by adjusting ψ_w

is also used in the case of the power crowbar, with $I_{\phi p}(t) =$ constant. Or, if the experiment is crowbarred directly on the torus shell without power crowbar, as in some of ZT-S discharges, $\psi_w(t)$ is given by

$$\Delta\psi_w = \mathcal{L} \Delta I_{\phi p} \quad (22)$$

after the crowbar, where \mathcal{L} is a constant representing the flux in between the outer shell and the discharge tube wall, which is approximately proportional to the plasma current.

3.2.D. Boundary Conditions for the θ Direction and Computational Mesh Scheme

In general, the boundary condition for the θ direction will be simply a periodic boundary condition from the fact that $f(\theta + 2\pi) = f(\theta)$ for any variable f . However, to save computer time, we used the symmetry condition against the horizontal mid-plane of the torus. Therefore, only those toroidally symmetric instabilities which are also symmetric against the mid-plane, if any, will appear in our calculation.

It is easy to see that the following symmetry conditions against the mid-plane satisfy equations (6) and (7).

Even symmetry: ψ, χ, T

$$B_{\theta}, B_{\phi}, j_{\theta}, j_{\phi}, \rho, U_r, \quad (24)$$

Odd symmetry: $B_r, j_r, U_{\theta}, U_{\phi}$.

Fig. 4 shows the computational mesh scheme used. We use polar meshes with constant Δr and $\Delta\theta$, which cover the upper half of the torus cross section with additional $\Delta\theta$ at each side (i.e., $j = 1$ and $j = j_{\max}$). In this mesh scheme, the symmetry condition (24) determines the boundary conditions for the θ direction as

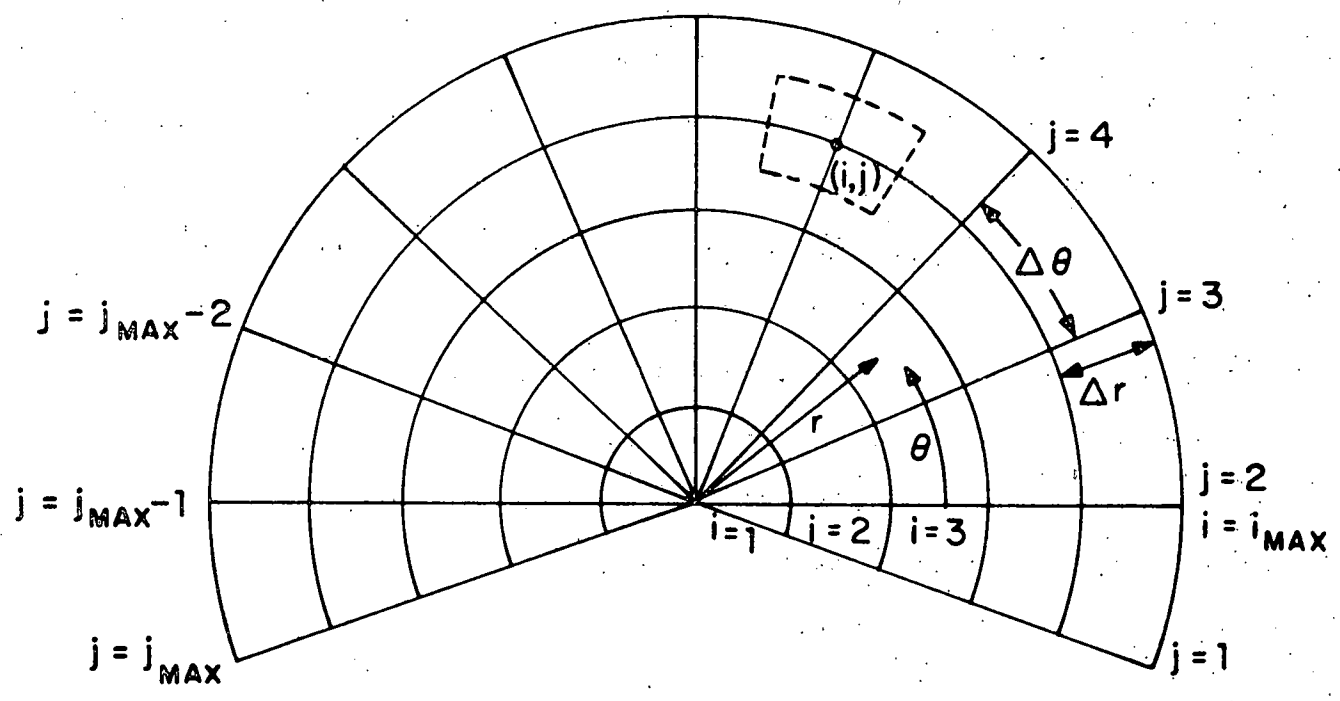


FIG. 4. SCHEMATIC DIAGRAM OF THE COMPUTATIONAL MESHES

$$f_{i,1} = \pm f_{i,3}$$

$$f_{i,j_{\max}} = \pm f_{i,j_{\max}-2}$$

+ for the variables of
even symmetry,

(25)

- for the variables of
odd symmetry.

3.2.E. Treatment of the Coordinate Origin

Although the coordinate origin $r = 0$ is an interior point of the plasma, it is a boundary point of the coordinate system (r, θ, ϕ) . Therefore, appropriate treatment of the origin is necessary. We will describe two separate methods for the treatment of the coordinate origin. The first method is simple and easy to use but good only for a relatively gentle process (e.g., a low pinch ratio discharge). The other method is rather complicated to use but found to be satisfactory in general.

The first method exploits the fact that the value of $\underline{W} = (\rho, \rho U_r, \rho U_\theta, \rho U_\phi)$ at the origin does not appear in the solution procedure, if we adopt the following difference scheme of the r derivation, i.e.,

$$\frac{1}{r} \frac{\partial r f}{\partial r} \rightarrow \frac{1}{r_i} \frac{(r f)_{i+1} - (r f)_{i-1}}{2\Delta r}, \quad (26)$$

where f is any function of \underline{W} . All the terms which require boundary values of \underline{W} in the equations (6) and (7) can be naturally represented in this way.

Among the remaining boundary values for the variables T, ψ, χ , the boundary value T_0^{n+1} of the variable T at the next time step $t = (n+1)\Delta t$, can be given simply as the average value of the old time step value $T_{i=2,j}^n$ of the grid points at $r = \Delta r$ (see Fig. 4), i.e.,

$$T_0^{n+1} = \frac{1}{N} \sum_{j=1}^N \sum_{i=2,j}^n T_{i=2,j}^n \quad (27)$$

where $N = \frac{2\pi}{\Delta\theta}$, and the values at $j > j_{\max}$ are found from the symmetry condition (25).

The boundary value χ_0^{n+1} is found by applying Faraday's law on the center grid (the region $r < \Delta r$), i.e.,

$$\frac{(\Delta\chi_0)^n}{\Delta t} = \frac{R_0 (\Delta B_\phi)_0}{\Delta t} = \frac{R_0 c}{\pi (\Delta r)^2} \sum_{j=1}^N (E_\theta)_{i=2,j}^n \Delta r \Delta\theta,$$

$$(E_\theta = \frac{j_0}{\sigma} + \frac{1}{c} (U_{B_\phi r} - U_{r B_\phi})), \quad (28)$$

$$\chi_0^{n+1} = \chi_0^n + (\Delta\chi_0)^n.$$

The boundary value ψ_0^{n+1} is determined from $(E_\phi)_0^n$ which is set at the average value of $(E_\phi)_{i=1,j}^n$ at $r = \Delta r$, i.e.,

$$(E_\phi)_0^n = \frac{1}{N} \sum_{j=1}^N (E_\phi)_{i=1,j}^n,$$

$$\frac{1}{cR_0} \frac{(\Delta\psi_0)^n}{\Delta t} = (E_\phi)_0^n, \quad (29)$$

$$\psi_0^{n+1} = \psi_0^n + (\Delta\psi_0)^n.$$

This simple method just described, is found satisfactory when the plasma behavior is rather gentle. But when the plasma behavior is more violent (e.g., for high pinch ratio discharge),

the \underline{W} values tend to oscillate near the origin and the calculation tends to break down.

This instability is probably due to the fact that the difference scheme (26) does not conserve mass and momentum near the origin. For example, the mass flow $\frac{(r\rho U_r)_{i=2}}{2}$ through the circle $r = \frac{1}{2} \Delta r$ is lost, since the value $\rho_{i=1}$ at the origin does not appear in the equation. (cf. Section 3.2.A). Although the resulting error in the total is rather small due to small size of the cells near the origin, this error can result in a severe numerical strain in these small cells. Therefore, the following more elaborate treatment of the origin, which conserves mass and momentum near the origin, is adopted, and it is found satisfactory for any cases treated thus far.

Instead of the difference scheme (26), we use the following scheme for the r derivatives, i.e.,

$$\frac{1}{r} \frac{\partial r f}{\partial r} = \frac{f}{r} + \frac{\partial f}{\partial r} \rightarrow \frac{\frac{1}{2} f_i + \frac{1}{4} (f_{i+1} + f_{i-1})}{r_i} + \frac{f_{i+1} - f_{i-1}}{2\Delta r} \quad (30)$$

In cylindrical coordinates (r, θ) , using a similar argument as in Section (3.2.A), we find that the total mass change in the grid (i, j) due to the U_r term $\frac{1}{r} \frac{\partial}{\partial r} (r\rho U_r)$ is

$$\begin{aligned}
(\Delta M)_{ij} &= \Delta\rho\Delta r r\Delta\theta \\
&= - \left(\dot{r}_i \Delta\theta + \frac{\Delta r \Delta\theta}{2} \right) \left(\frac{\rho U_{ri+1} + \rho U_{ri}}{2} \right) \Delta t \\
&\quad + \left(r_i \Delta\theta - \frac{\Delta r \Delta\theta}{2} \right) \left(\frac{\rho U_{ri+1} + \rho U_{ri-1}}{2} \right) \Delta t \\
&= - r_{i+\frac{1}{2}} \Delta\theta (\rho U_r)_{i+\frac{1}{2}} \Delta t \\
&\quad + r_{i-\frac{1}{2}} \Delta\theta (\rho U_r)_{i-\frac{1}{2}} \Delta t \tag{31}
\end{aligned}$$

Therefore, the difference scheme (30) conserves mass. The mass flux density through a point $i+\frac{1}{2}$ is defined as*

$$\frac{(\rho U_r)_i + (\rho U_r)_{i+1}}{2}$$

The definition of the momentum flux density can also be found in a similar way (e.g., for the θ component, as $\frac{(\rho U_\theta U_r)_i + (\rho U_\theta U_r)_{i+1}}{2}$). Using these definitions, we determine the boundary value \underline{W}_0 at the origin ($i=1$) such that total mass and momentum are conserved.

For the value ρ_0 , we apply the mass conservation equation

$$\frac{d}{dt} \int \rho dV = \int \rho \underline{U} \cdot d\underline{s}$$

on the region $r \leq r/2$, and find

* The difference between this and definition (15) in Section (3.2.A), is due to the different r direction difference schemes used, and results in a minor change in the boundary conditions at the wall. (The boundary conditions for the present scheme are $(\underline{W}_r)_N = -(\underline{W}_r)_{N-1}$, compared to the condition (16), i.e., $(r\underline{W}_r)_N = -(r\underline{W}_r)_{N-1}$, for the previous difference scheme (11).) But near the origin, these two definitions give quite different results.

$$\frac{\Delta \rho}{\Delta t} = \frac{-4}{N \Delta r} \sum_{j=1}^N (\rho U_r)_{i=3/2} S_j,$$

where $(\rho U_r)_{i=3/2} = ((\rho U_r)_{i=1} + (\rho U_r)_{i=2})/2$, and $S_j = 1 + \frac{\Delta r}{2R_0} \cos \theta_j$

appearing due to the toroidal curvature.

For the momentum values at the origin, the momentum equation

$$\frac{\partial}{\partial t} \int \rho \underline{U} dv = - \int (\rho \underline{U} (\underline{U} \cdot d\underline{s}) + p \cdot d\underline{s}) + \frac{1}{c} \int (\underline{j} \times \underline{B}) dv$$

is applied in an infinitesimal sector (since \underline{U} is a vector), $r \leq \frac{\Delta r}{2}$ and $\phi_c \leq \phi \leq \phi_c + d\phi$. This is calculated first in (R, Z, ϕ) coordinates (cf. Fig. 1) which are uniquely defined at the origin, unlike (r, θ, ϕ) coordinates. From the symmetry condition (24), we readily find that U_z and U_ϕ are zero at the origin. The nonzero R component ρU_R is found as

$$\begin{aligned} \frac{\Delta (\rho U_R)_0}{\Delta t} = & - \frac{4}{N \Delta r} \sum_{j=1}^N ((\rho U_r U_r)_{i=3/2, j} S_j + P_{i=3/2, j} \cos \theta_j) \\ & + \frac{1}{c} (\underline{j}_0 \times \underline{B}_0)_R, \end{aligned} \quad (33)$$

where

$$(\rho U_r U_r)_{i=3/2, j} = (\rho U_r U_r \cos \theta - \rho U_\theta U_r \sin \theta)_{i=3/2, j},$$

$$\left[\begin{aligned} (\rho U_r U_r)_{i=3/2, j} &= \frac{1}{2} \{ (\rho U_r U_r)_{i=1, j} + (\rho U_r U_r)_{i=2, j} \}, \\ (\rho U_\theta U_r)_{i=3/2, j} &= \frac{1}{2} \{ (\rho U_\theta U_r)_{i=1, j} + (\rho U_\theta U_r)_{i=2, j} \} \end{aligned} \right],$$

$$P_{i=3/2, j} = \frac{1}{2} (\gamma - 1) \{ (\rho T)_{i=2, j} + (\rho T)_{i=1, j} \},$$

and the values j_0 and B_0 at the origin will be found later in this section. Then, the values $(\rho U_r)_{i=1,j}$ and $(\rho U_\theta)_{i=1,j}$ at the origin are found from the value $(\rho U_R)_{i=1}$ given above. Since in general for a vector $\underline{V} = V_R \hat{R} + V_Z \hat{Z} + V_\phi \hat{\phi}$,

$$(V_r)_{i=1,j} = (V_R)_{i=1} \cos \theta_j + (V_Z)_{i=1} \sin \theta_j,$$

$$(V_\theta)_{i=1,j} = -(V_R)_{i=1} \sin \theta_j + (V_Z)_{i=1} \cos \theta_j,$$

we find

$$(\rho U_r)_{i=1,j} = (\rho U_R)_{i=1} \cos \theta_j, \quad (34)$$

$$(\rho U_\theta)_{i=1,j} = -(\rho U_R)_{i=1} \sin \theta_j.$$

For the temperature T_0 at the origin, we simply average $T_{i=2,j}$ as the previous method (27) instead of using the energy conservation. This is for simplicity, because the temperature equation is not written in a conservative form so that incorporating the exact energy conservation would be complicated. Also, the value T_0 will not much affect the overall stability, due to the large conduction term in the temperature equation.

The boundary value χ_0 is found exactly the same as in the previous method (Eq. 28). The value ψ_0 is also found in a similar way as in the previous method (Eq. 29), except this time the value $(E_\phi)_0^n$ is found directly from $E_{\phi 0} = \frac{j\phi_0}{\sigma_0} + \frac{1}{c}$

$(U_{r0} B_{\theta 0} - U_{\theta 0} B_{r0})$ without the averaging step.

The boundary values \underline{j}_0 and \underline{B}_0 are found using Maxwell's equations in (R, Z, ϕ) coordinates and then transforming them into (r, θ, ϕ) coordinates in the same way as the procedure for the value $\rho \underline{U}_0$. The values in (R, Z, ϕ) coordinates are

$$\begin{aligned}
 j_{R0} &= 0, \\
 j_{Z0} &= - \frac{c}{2\pi R_0 N \Delta r} \sum_{j=1}^N \chi_{i=2,j} \cos \theta_j, \\
 j_{\phi 0} &= \frac{c}{2\pi N \Delta r} \sum_{j=1}^N (B_{\theta})_{i=2,j}, \quad (35)
 \end{aligned}$$

$$B_{R0} = 0$$

$$B_{Z0} = - \frac{2}{R_0 N \Delta r} \sum_{j=1}^N \psi_{i=2,j} \cos \theta_j,$$

$$B_{\phi 0} = \frac{\chi_0}{R_0}.$$

3.3 Treatment of the Low Density Region

When the plasma pinches inward, the density near the wall drops quickly resulting in a very low density plasma region (or maybe a vacuum) outside the main pinched plasma column. Treatment of this low density region has been one of the major difficulties since the early work of Hain et al.²⁸

It is debatable whether we should treat this region as a true vacuum or as a very low density plasma. Even if the mathematical model (i.e., the dissipative MHD model, in the present case) indicate this region to be a vacuum, the actual situation in the experiment may be quite different, e.g., due to turbulent diffusion, emission from the wall, etc. Aside from this fact, even mathematically, the answer is not clear. If a Lagrangian difference scheme is used, the plasma boundary will part from the wall and create a vacuum region. But this may not represent a correct picture even if a smooth exact solution exists with density $\rho > 0$ in the outer region, because the conductivity σ can be still finite. Moreover, the exact solution of the differential equation may possess singularities in it,²⁷ which are removed by the finite spacing of the Lagrangian scheme.

Often, there is some advantage in assuming the outer region to be a vacuum, because the numerical calculation of the low density plasma possesses difficulties which will be discussed later. Although it is very difficult to incorporate

the vacuum region in our two dimensional calculation due to the complexity of the vacuum-plasma boundary, it is easily done in one dimensional ($\frac{\partial}{\partial \phi} = 0$, $\frac{\partial}{\partial \theta} = 0$) calculation with a Lagrangian scheme or even a Eulerian scheme. Therefore, we tested this idea using a one dimensional Eulerian code and found that the comparison between the calculation and the experimental results was not satisfactory. This large discrepancy is not surprising because the partial shielding of the applied fields by the low density region plasma is totally neglected. This effect is more important in our case of a circular cross section torus due to the large cross sectional area of the low density region.

To solve the problem incorporating the low density region plasma, it is necessary to devise a method to go around the numerical difficulties due to a high Alfvén speed. Since the plasma density is very low and the magnetic field is strong in the low density region, the Courant condition for stability requires prohibitively small time step for an explicit difference scheme. One thus improves this situation by using an implicit scheme as Hain et al.,²⁸ or by substituting an artificial signal speed for the speed of light as Boris²⁹. These methods will extend the solvable range of the density ρ , but of course, there is a limit for this range because of nonlinear instabilities or inaccuracy of the solution, etc. Therefore, it seems that one must also use a procedure which keeps the density from falling below a fixed floor value, rather than simulate the real low density in this region.

Hain et al.²⁸ assumed the low density plasma to be created continuously at the wall to maintain the plasma density above a floor value. For some experimental parameters, however, this is not enough. Therefore the injection of plasma at individual cells when the density falls below a floor value may be necessary as in Lui and Chu.¹ Although these procedures can be thought of as analogies of the ionization process, these methods produce largely artificial effects such as artificial increase of the mass, momentum and energy. These artificial effects become severe when the calculation is run for a long time as in our calculation. Lindemuth³⁰ assumes the wall to be a gate which opens when the plasma pinches inward, remains open to let the plasma flow out when the plasma expands outward, and then closes when the total mass of the plasma returns to the original value. This procedure would somewhat improve the situation in the direction of the mass conservation. But for example, at the end of the first pinching, the artificial mass increase will be the same as the previous method. To conserve the total mass, some codes prevent the mass flow by setting the fluid velocity to zero when the density drops below a floor value. But in this method, the plasma in the low density region will see the whole electric field such that the current flow ($\underline{j} = \sigma \underline{E}$ with $\underline{U} = 0$) will be artificially high, especially when the crowbar is not yet applied. In spite of this error, we found this method useful for its good stability character, but only in limited cases when a small region of low density plasma is created until the crowbar time.

The method we use is modified but similar to the ones used by Hain et al.²⁸ and Lui and Chu¹, with a floor value of $\sim 5\%$. To correct the artificial increase of the mass and momentum, we adopt the following procedure. The main idea of this procedure is that we picture the low density region as a very tenuous plasma of finite conductivity, and also that the flow of the plasma mass and momentum from the low density region into the main plasma column is negligible due to the negligible density of the outer region.

1) Define the low density region-main plasma column boundary. Most of the time, this boundary is well defined and easy to find, because there is a sharp density increase at this boundary.

2) Subtract the mass and momentum flow into the main plasma column from the low density region, at each time step. For this procedure, we use the definition of the mass and momentum flow given in Section (3.2.A). We will explain this step using a one-dimensional analogy (Figure 3 on page). When the flow is inward to the main plasma column, the artificial increase of the mass and momentum at the first cell (cell M) in the main plasma column, is subtracted, i.e.,

$$\rho_M^{\text{new}} = \rho_M + \frac{\Delta t}{2\Delta x} \left((\rho u)_{M+1} + (\rho u)_M \right);$$

$$(\rho u)_M^{\text{new}} = (\rho u)_M + \frac{\Delta t}{2\Delta x} \left((\rho u u)_{M+1} + (\rho u u)_M \right) \quad (36)$$

when the flow is inward (i.e., $(\rho u)_{M+1} + (\rho u)_M < 0$). Here, the point $M + \frac{1}{2}$ corresponds to the low density - main plasma boundary.

When the flow is outward to the low density region (i.e., $(\rho u)_{M+1} + (\rho u)_M > 0$), a detailed adjustment can be made to ensure the conservation of the total mass. But this procedure can be usually omitted, because the difference resulting from this procedure is small, due to the fact that the plasma outflow from the main plasma column usually dominates the flow in the low density region. Nevertheless, if a certain situation demands, this procedure can be derived in an analogous way as the inward flow case, except that this time, the boundary cell (cell L) in the low density region which makes contact with the main plasma column, should be treated as a cell inside the main plasma, thereby inducing an outward movement of the low density - main plasma boundary.

We note that no correction is made to the flow of energy, because the thermal energy is the least accurate variable in our calculation (cf. Section (2.3)).

IV. RESULTS OF THE SIMULATION

In this chapter, we present the calculated results of SP-1 screw pinch and ZT-I, ZT-S and ZT-40 reverse-field pinch experiments. (For the description of the experiments, see chapter I.) The calculated results are compared to the experimental values when they are available.

Typical operating parameters in various reverse field pinches are compared in table 1. The starred items have been simulated with the present code. The four devices on the upper columns, including ZT-S, are currently operating experiments. They are similarly on a quite low parameter range. The next three larger devices, including ZT-40, are either on a planning or construction stage. The scaling character of the pinch discharges is the crucial question to be investigated by these larger devices. Each of these devices is envisioned as an intermediate step for the next even larger experiment which would reach $n\tau \sim 10^{14} \text{ sec} \cdot \text{cm}^{-3}$ range. The old ZETA parameters and the ZT-I which was the previous version of ZT-S, and also the SP-1 screw pinch parameters are shown on the last columns for comparison.

Since the three experiments, SP-1, ZT-I and ZT-S are on a similar parameter range, discussions made for one experiment would similarly apply to the others, most of the time. Therefore, according to the availability of the

TABLE 1

Typical Operating Parameters in Various Reverse-Field Pinches

	<u>Minor Rad.</u>	<u>Major Rad.</u>	<u>Filling Gas</u>	<u>Lifetime</u>	<u>Temperature</u>	<u>Location</u>
HBTX-I	6.5cm	100cm	$\sim 2 \times 10^{15} \text{cm}^{-3} \text{D}_2$	$\sim 15 \mu\text{s}$	$\sim 50 \text{ev}$	Culham
* ETA-BETA	5	40	$\sim 10^{15} \text{H}_2$	~ 15	~ 10	Padua
ETL-TPE-1	5	40	$\sim 2 \times 10^{15}$	~ 20	~ 20	Tokyo
* ZT-S	7.7	40	$\sim 2 \times 10^{15} \text{D}_2$	~ 30	~ 20	LASL
HBTX-II	60	180	$\sim 10^{14}$	$\sim 10,000$	~ 300	Culham
ETA-BETA-II	24					Padua
* ZT-40	20	114	$\sim 2 \times 10^{15}$	~ 200	~ 200	LASL
ZETA	50	150	$\sim 10^{14}$	$\sim 10,000$	~ 200	Harwell
* ZT-I	5.2	38	$\sim 2 \times 10^{15} \text{D}_2$	~ 15	~ 10	LASL
* SP-1 Screw Pinch	6	36	$\sim 2 \times 10^{15} \text{H}_2$	~ 10	~ 20	Jutphaas

experimental data, a different emphasis is given on the discussion of each experiment. The gross motion of the plasma and its approach to the quasi-equilibrium dissipation state are discussed in detail for the SP-1 simulation (Section 4.1). The comparison between the given anomalous resistivity algorithm and the classical resistivity is emphasized in the discussion of the ZT-I results (Section 4.2). This comparison is extended to a much longer time through the dissipation phase for the ZT-S experiment (Section 4.3.A). The effects of toroidal curvature, i.e., toroidal shift, toroidal plasma flow, etc., are discussed at length in Section (4.3.B).

For the larger ZT-40 experiment, the effects due to a bigger machine size are studied. For this, we first study the scaling law of the dissipative magnetohydrodynamic equations (Section 4.4.A). Finally, in the last section, we discuss some confinement characteristics of large pinch machines.

4.1 SP-1 Screw Pinch

The SP-1 toroidal screw pinch³¹ at Jutphaas has a major radius of 36cm and a minor radius of 6cm. We will present the simulation results of a discharge with $n_0 = 4 \times 10^{-9}$ gm/cm³ hydrogen filling gas, initial temperature of 1 ev and a toroidal bias field of +100G. Figure 5 shows the time traces of the toroidal magnetic field applied at the wall and the total toroidal plasma current. Although,

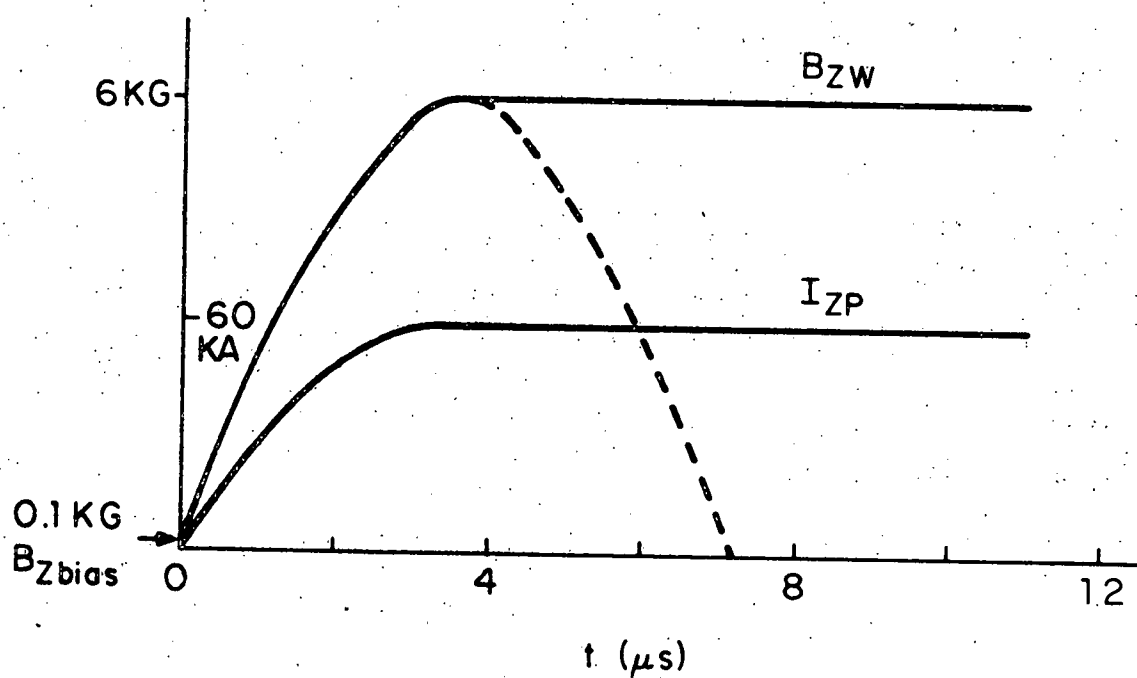


FIG. 5. TIME TRACES OF THE TOROIDAL MAGNETIC FIELD B_{zw} AT THE WALL AND THE TOTAL TOROIDAL PLASMA CURRENT I_{zp} . THE DASHED LINE INDICATES THAT THE DISCHARGE IS NOT CROWBARRED IN THE EXPERIMENT.

in the experiment, the discharge is allowed to ring as indicated by the dashed experimental curve, we effectively crowbarred the discharge at $3.6\mu\text{s}$ in our simulation to study the transition toward a quasi-equilibrium dissipation state. The following results are obtained using the classical resistivity formula. The Chodura anomalous resistivity formula described in Section (2.3) gives essentially the same results until the crowbar time. A detailed comparison between the classical resistivity and the anomalous algorithm is made in the following two sections on ZT-I and ZT-S results.

Figure 6 shows the toroidal field at the center of the poloidal plane as a function of time. The toroidal field at the wall is also shown for a comparison. The solid curve represents the calculated results and the dashed curve the experimental results which is measured only until the crowbar time ($3.6\mu\text{s}$). This figure shows at once the pinching and bouncing motions of the plasma being damped due to the dissipation mechanisms and reaching a quiet phase which will be shown later to be a quasi-equilibrium dissipation state. Therefore, the present simulation is successful in that we could follow the plasma behavior through the pinching and bouncing motions all the way to the quasi-equilibrium state. The previous simulation of this same discharge could follow the plasma behavior until about $2.5\mu\text{s}$, but it broke down due to a numerical instability. A careful treatment of the boundary conditions is found to be mainly responsible for

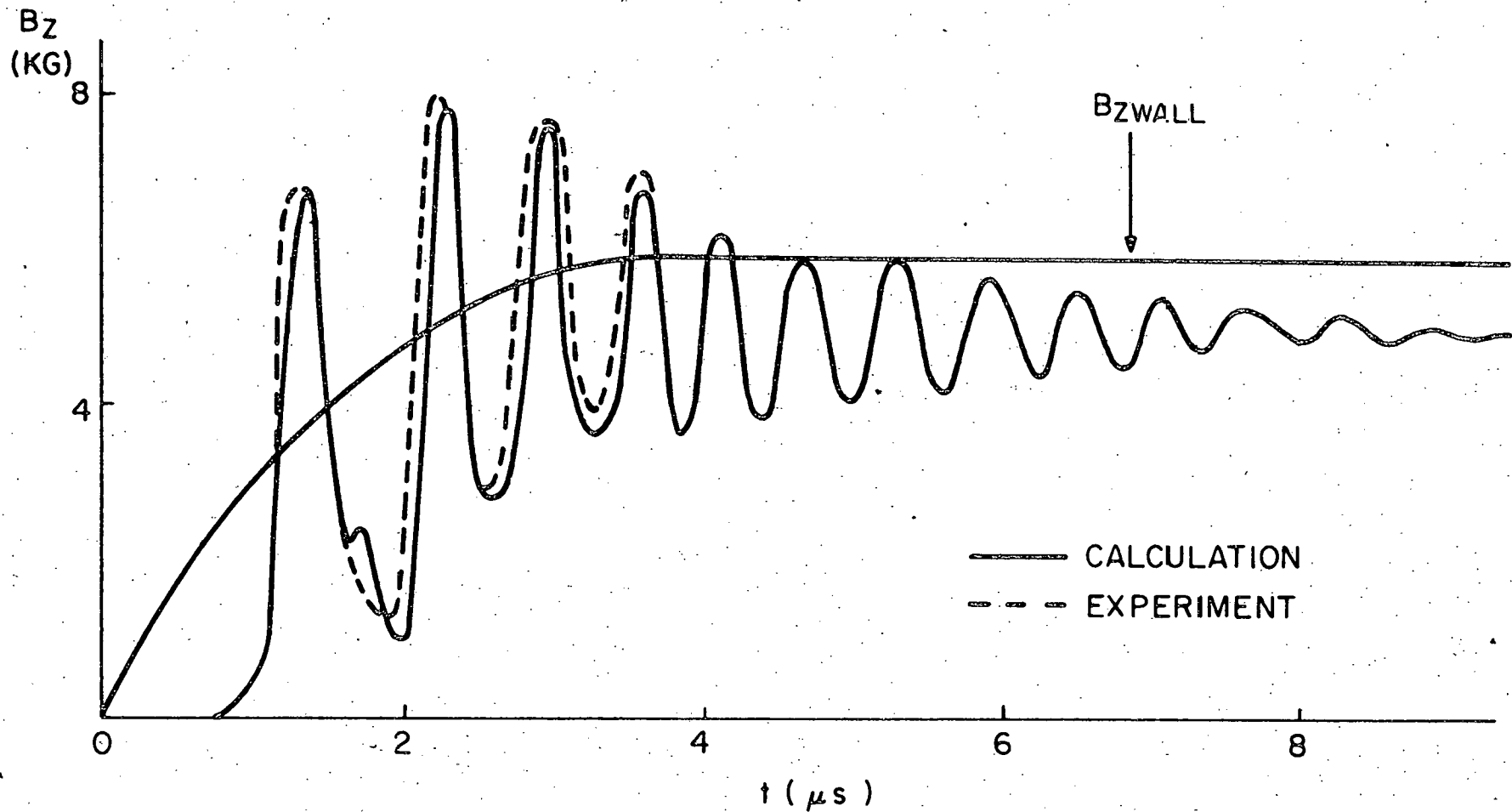


FIG. 6. TIME TRACE OF THE TOROIDAL FIELD AT THE MINOR AXIS.
THE TOROIDAL FIELD AT THE WALL IS ALSO SHOWN.

this successful simulation. The very close agreement between the calculation and the experiment shown in Figure 6 is somewhat optimistic. As we will see later in this chapter, all the simulation results show good agreement with the experiment, but not in general as close as this result.

In Figure 7, the profiles of the density and temperature on the horizontal diameter ($Z = 0$) of the poloidal plane are shown at various instants of time during the initial two bounces. The density profile at $1.8\mu\text{s}$ corresponds to the maximum outward expansion after the first pinching, and it shows that the main plasma column stays well away from the wall. Both the density and the temperature are higher on the inside of the torus since the magnetic field and the current density are stronger inside. A toroidal shift of $\sim 1\text{cm}$ is seen at $2.2\mu\text{s}$. Since the pinching time is much shorter than the heat diffusion time (which will be discussed later), the compressional heating is clearly seen.

After several more successively smaller bounces, the density profile reaches a nearly stationary state as shown in the next figure. Figure 8 shows the density and temperature profiles at 6, 7, and $8\mu\text{s}$. The bouncing motion is largely damped away and the toroidal shift motion of the plasma column seems to be the dominant movement. This toroidal shift motion is found to be oscillatory as shown in the next figure.

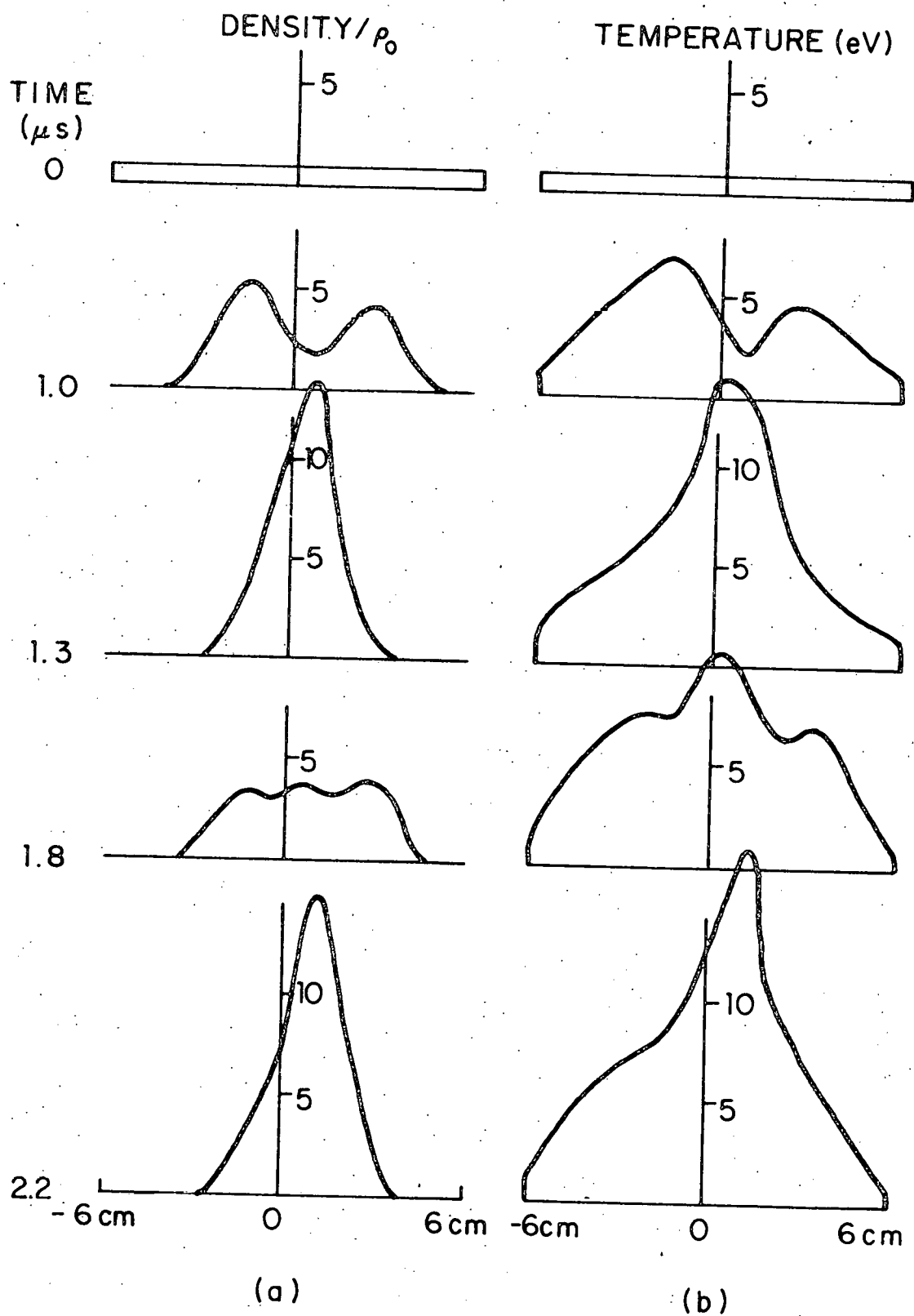


FIG. 7. PROFILES OF THE DENSITY AND THE TEMPERATURE ON THE HORIZONTAL DIAMETER OF THE POLOIDAL PLANE.

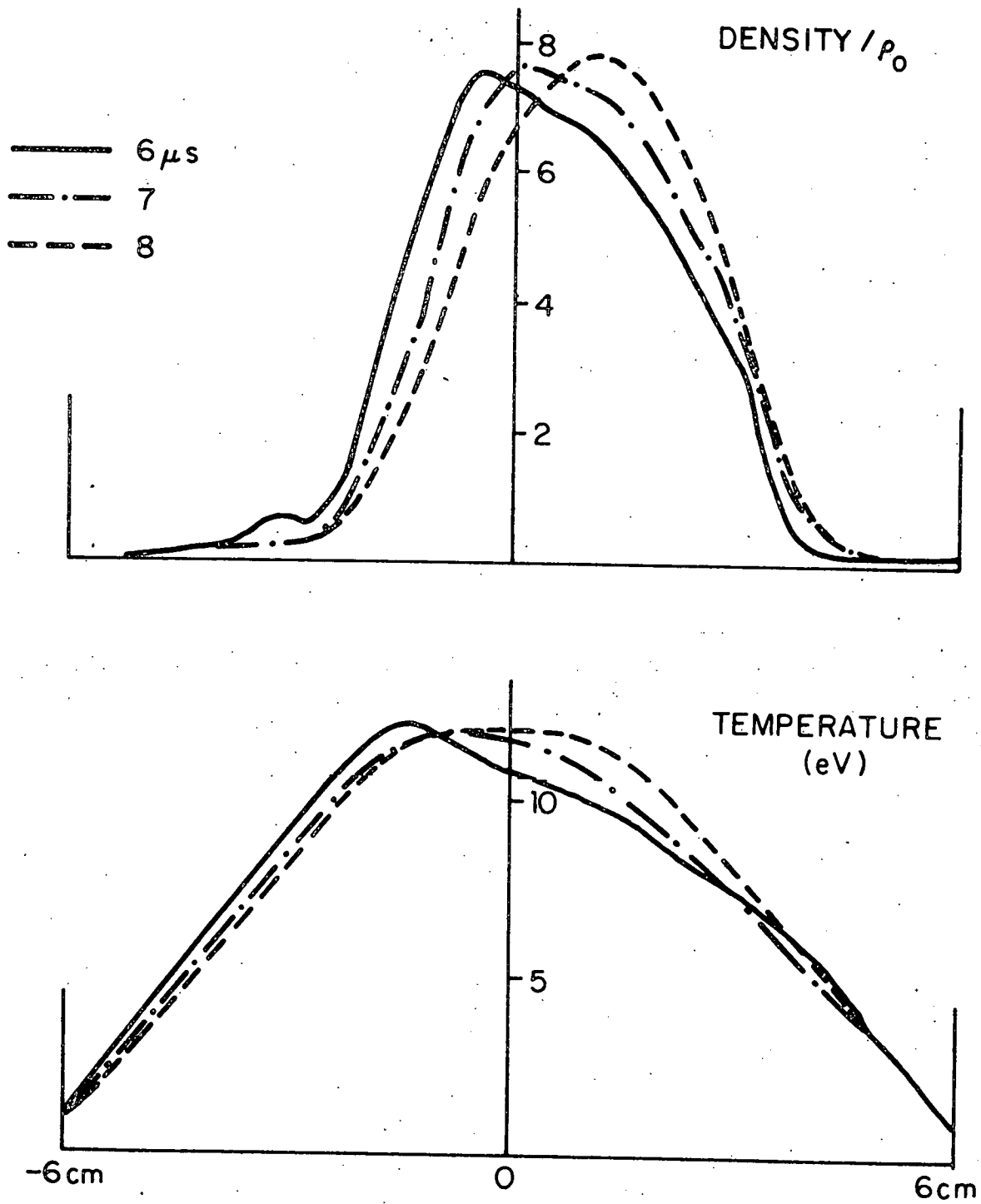


FIG. 8. DENSITY AND TEMPERATURE PROFILES
AT 6, 7 and 8 μs .

In Figure 9, the dotted curve represents the time trace of the magnetic axis, and the two semi-broken curves represent the time traces of the positions of the two density peaks. The major axis is on the lower side. The traces of the density peaks show the pinching and bouncing motions of the plasma. This bouncing motion is mostly damped away at about $5\mu\text{s}$ and the density peak subsequently follows closely the trace of the magnetic axis. The motion of the toroidal shift shows a damped oscillation with a period of $\sim 5\mu\text{s}$ and a dissipation time of $\sim 10\mu\text{s}$. The equilibrium position of this toroidal shift is $\sim 1.2\text{cm}$ which compares with the experimental measurement of $\sim 1.5\text{cm}$.

The time scales of these oscillations and the damping process can be understood from the basic time scales (cf. Section (4.4.A)) of this diffuse pinch discharge. The Alfvén time scale $\tau_A = r^0 \sqrt{4\pi\rho^0} / B^0$ (the superscript 0 denotes a representative value of the variable) remains at $\sim 1\mu\text{s}$ throughout the discharge. The field diffusion time

$\tau_B = \frac{4\pi r^0{}^2}{c \eta^0}$ is initially $\sim 5\mu\text{s}$ then increases to $\sim 20\mu\text{s}$ after the initial phase. The heat diffusion time $\tau_H = \frac{C_V \rho^0 r^0{}^2}{K^0}$

is $\sim 30\mu\text{s}$ throughout the discharge. The bouncing motions in Figure 9 are on the Alfvén time scale as expected. The toroidal shift oscillation, which will be also on the Alfvén time scale, is several fold slower than the bouncing motion. One of the main reasons for this would be that the half

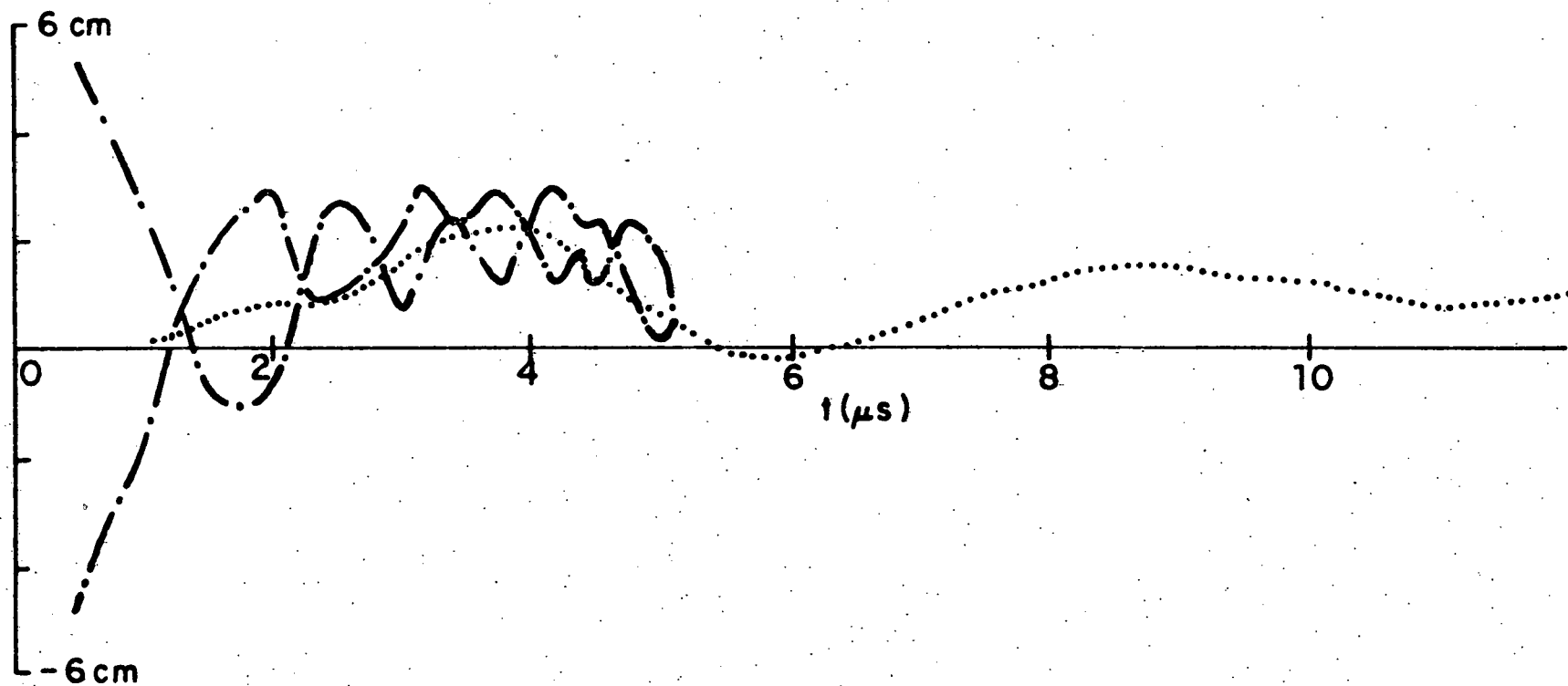


FIG. 9. TIME TRACES OF THE MAGNETIC AXIS (.....) AND THE TWO DENSITY PEAKS (- . - . -).

period of the toroidal shift oscillation corresponds to the time during which the overcompression of the magnetic field travels from one wall to the other, while the bouncing motions occur between the center and the wall.

The damping time scale of the bouncing motion in Figure 9 is $\sim 5\mu\text{s}$, since a heavy field diffusion occurs during the initial pinching when the field diffusion time τ_B is short. On the other hand, the damping time of the toroidal shift oscillation is somewhat longer with a damping time scale of $\sim 10\mu\text{s}$ which is on the field diffusion time scale of the post implosion phase. Therefore, the damping time scale of the plasma oscillation is on the field diffusion time τ_B , as expected. One can ask, however, whether the heat diffusion time τ_H , if $\tau_H \ll \tau_B$, would determine the damping time scale. We tested this question by simulating discharges with parameters such that $\tau_H \ll \tau_B$, and found that τ_H does not determine the damping time scale, even when $\tau_H \ll \tau_B$. This fact can be easily understood by considering the limiting case where $\tau_H \rightarrow 0$ (i.e., the thermal conductivity $K \rightarrow \infty$). The temperature T becomes constant in this case, but the mechanical oscillation will still remain due to the density compression (i.e., $\nabla p = kT\nabla n$). In general, if the viscosity is included, the damping time will be determined by the smaller between τ_B and the viscous dissipation time $\tau_v = \frac{r^2}{\nu}$ where ν is the kinematic viscosity coefficient. Assuming the viscosity is classical (the

viscosity coefficient²¹ $\nu \approx \nu_i \sim \frac{kT_i \tau_i}{M_i}$, where the subscript

i represents the ion species and τ the collision time), the viscous dissipation time τ_ν becomes ~ 1 ms. This long time scale compared to other time scales τ_A , τ_B , and τ_H justifies the neglect of the viscosity in the calculation. The numerical viscosity in the calculation also results in an effective dissipation time of $100 \sim 1000 \mu\text{s}$ (as discussed in Section 3.1), which is much longer than other time scales.

Figure 10 shows the radial velocity profiles on the horizontal diameter of the poloidal plane at various instants of time. The major axis is on the left and the plus sign of the velocity indicates outward (toward right) direction. At $1 \mu\text{s}$, we see imploding velocities of $\sim 5 \times 10^6$ cm/sec which is of the same magnitude as Alfvén velocity. At $5 \mu\text{s}$, we see only small imploding velocities imbedded in a larger velocity of an inward toroidal shift. At $10 \mu\text{s}$, the fluid velocities dropped by orders of magnitude from the velocities of the imploding stage. These velocities still represent a small remaining oscillation of the toroidal shift (cf. Figure 9), and they are still somewhat larger than the diffusion velocity $v_D \sim \frac{c^2 n \nabla p}{B^2}$ which is $\sim 5 \times 10^4$ cm/sec. Nevertheless, for most practical purposes, this plasma state could be thought of as a quasi-equilibrium dissipation state. In other words, the remaining small toroidal shift oscillation will have little influence on the subsequent plasma evolution since the kinetic energy associated

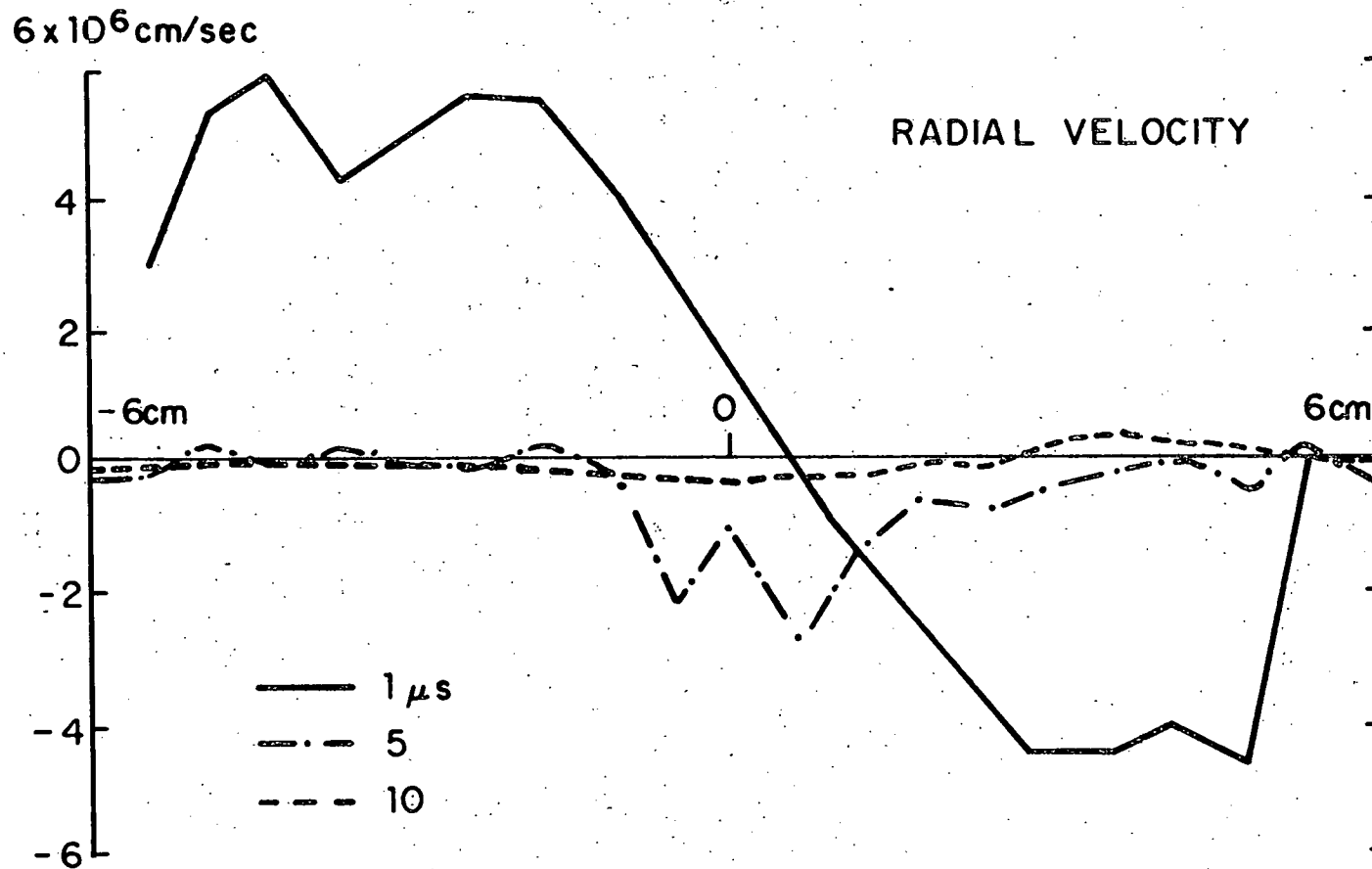


FIG. 10. RADIAL VELOCITY PROFILES ON THE SYMMETRY PLANE.

with this oscillation is many orders of magnitude smaller than both the magnetic field energy and thermal energy. The plasma evolution thereafter will be determined mainly by the dissipation mechanisms.

From the discussions on Figures (9 and 10), we can see that in a class of discharges as the present one, there will be three phases of the plasma state, namely, the initial phase where the radial pinching and bouncing motion is dominant, the second phase where the toroidal drifting motion is dominant and the final quasi-equilibrium dissipation phase. We note here that a strong toroidal flow is still remaining in the final phase. The magnitude of this velocity is comparable to the Alfvén velocity. Therefore, the final plasma state is actually a steady quasi-equilibrium state rather than a static quasi-equilibrium state. The effect of the toroidal plasma flow will be discussed in detail in Section (4.3.B) where the toroidal curvature effects are explained.

Figure 11 shows the calculated toroidal (B_z) and poloidal (B_θ)* field profiles on the horizontal diameter ($Z=0$) during the initial two bounces. Since the bouncing

* Although the poloidal field actually contains a small B_r component due to the toroidal curvature, we will frequently let B_θ represent the poloidal field for convenience, when the meaning is apparent from the context.

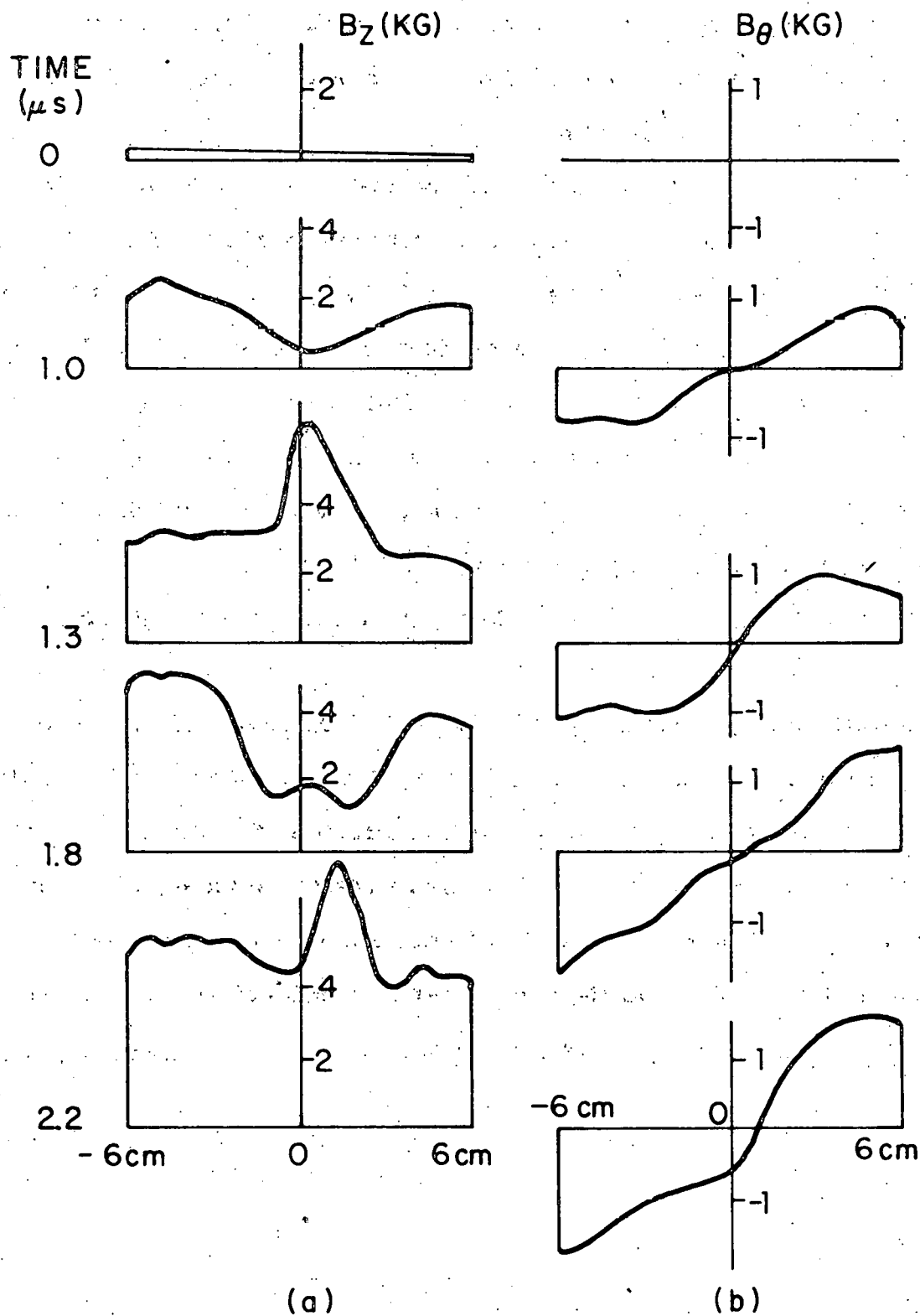


FIG. 11. TOROIDAL (a) AND POLOIDAL (b) FIELD PROFILES DURING THE INITIAL PHASE.

time scale is much shorter than the field diffusion time, we see a clear convection of the fields according to the pinching and expansion.

Figure 12 shows the field profiles at much later instants (6, 8, and 10 μ s). At 10 μ s, most of the toroidal field has been convected and diffused in. The toroidal current also largely penetrated into the center and its profile became flat as readily seen from the almost straight B_θ profile. The toroidal field and the poloidal field each balance a pressure of $\sim \frac{(2kG)^2}{4\pi}$. (The field curvature accounts a half of the total B_θ magnetic pressure.) Therefore, the plasma beta $\beta = \frac{8\pi P_0}{B_0^2}$ is $\sim 40\%$ at 10 μ s with similar contributions from

toroidal and poloidal fields. This fact that the toroidal field, even when it is much stronger than the poloidal field, contributes only similar amount of plasma pressure confinement as the poloidal field does, seems to be a general characteristic for the ohmically heated pinch experiments. The reason for this will be explained in Section (4.4.C) where we discuss confinement characteristics of the pinch experiments.

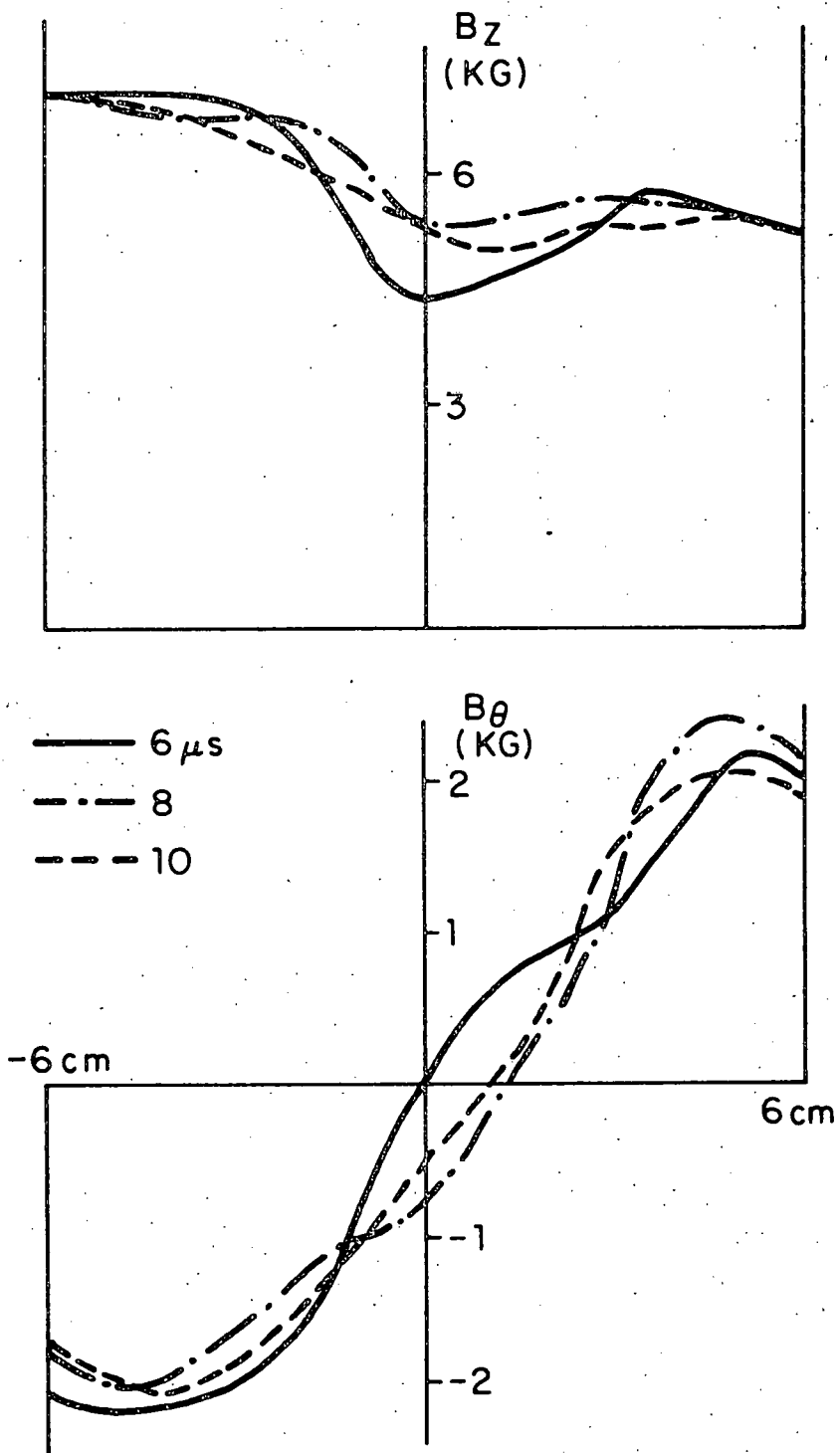


FIG. 12. TOROIDAL AND POLOIDAL FIELD PROFILES
AT 6, 8 AND 10 μs .

4.2 ZT-I Reverse-Field Pinch

The ZT-I toroidal reverse-field Z pinch (which has been converted to a modestly larger ZT-S experiment) in Los Alamos had a major radius of 38.2cm and a minor radius of 5.2cm. We will present a simulation result of a discharge with $n = 8.2 \times 10^{-9} \text{ gm/cm}^3$ deuterium filling gas and initial temperature of 1 ev. Figure 13 shows two driving boundary conditions for this discharge, namely, the toroidal magnetic field at the wall and the total toroidal plasma current, as functions of time. The initial bias field is 1.6kG and the circuits are crowbarred at $t = 2.4 \mu\text{s}$. It is noted that in ZT-I and ZT-S experiments, the magnetic fields are measured on the vertical radius ($\theta = \frac{\pi}{2}$) of the poloidal plane. Thus, whenever we compare the field profiles with the experiment, it is on the vertical radius.

In Figure 14, we present the toroidal and poloidal field profiles on the vertical radius ($\theta = \frac{\pi}{2}$) at 1.2, 2.0, and 2.8 μs . The upper figure (a) shows the results from two different calculations, namely, one using the pure classical resistivity (the dashed curves) and the other using the Chodura anomalous resistivity algorithm (cf. Section (2.3)). The lower figure shows the experimental results and these results agree well with the calculated results. Although the results obtained using the anomalous resistivity show somewhat better agreement with the experiment than the classical resistivity results do, the difference is not definitive enough to conclude that the classical resistivity is invalid.

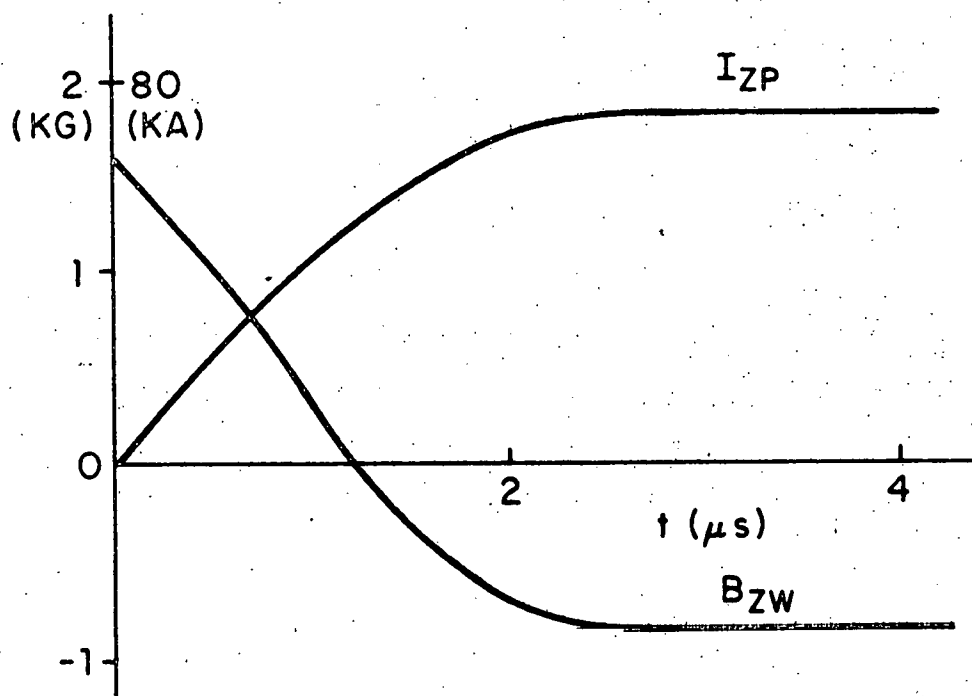


FIG. 13. TIME TRACES OF THE TOROIDAL MAGNETIC FIELD AT THE WALL AND THE TOTAL TOROIDAL PLASMA CURRENT IN A ZT-I DISCHARGE.

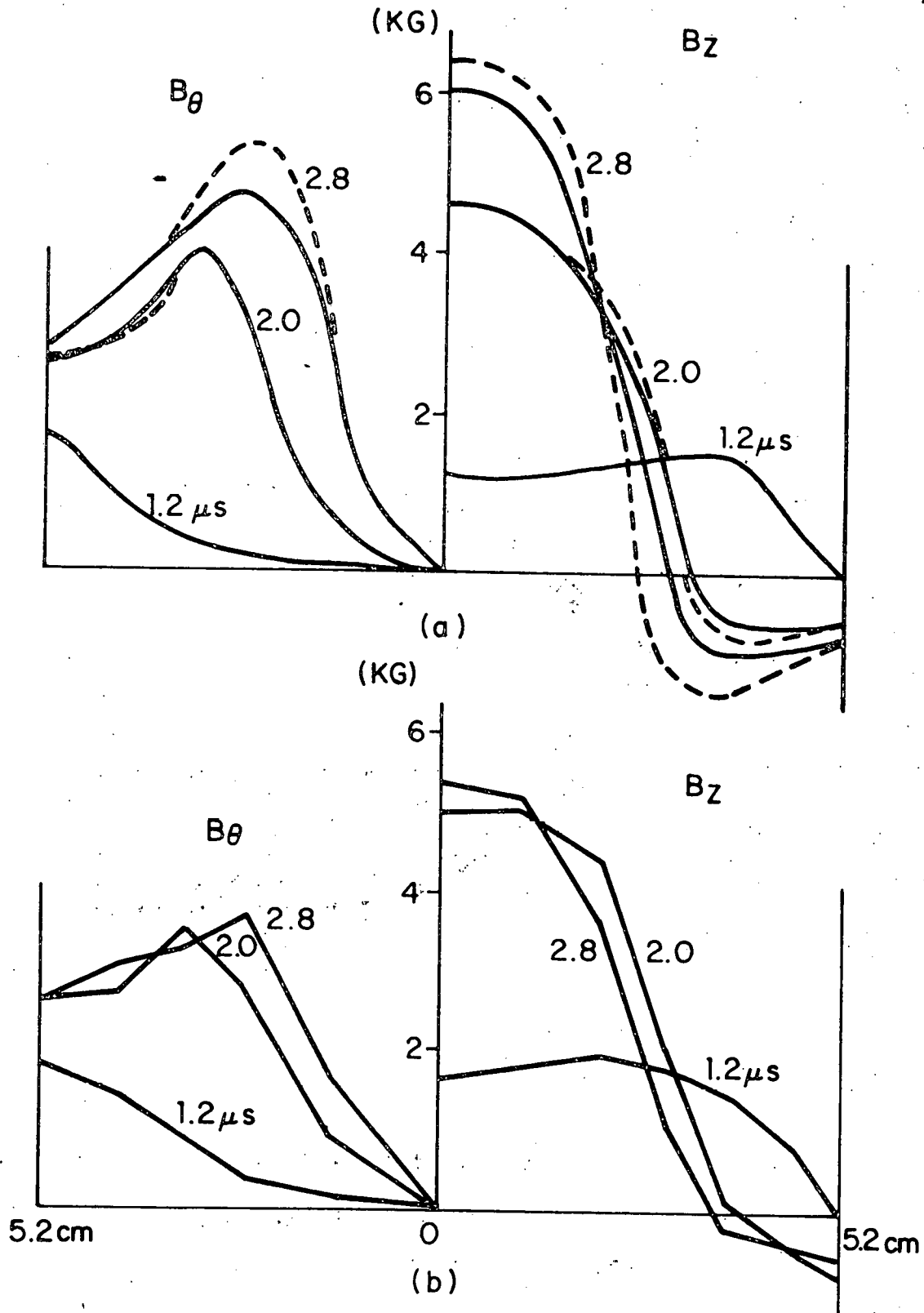


FIG.14. TOROIDAL AND POLOIDAL FIELD PROFILES: (a) CALCULATION RESULTS WITH CLASSICAL RESISTIVITY (-----), AND WITH THE ANOMALOUS RESISTIVITY (———); (b) EXPERIMENTAL RESULTS.

We also compared these calculation results with the simulation results of the same discharge due to Sgro and Nielson¹⁸ who used a one-dimensional hybrid (fluid electrons and Vlasov ions) code. The results of the two simulations using anomalous resistivity agreed with each other quite well. However, the results of the hybrid simulation using the pure classical resistivity show a marked difference, as shown in Figure 15. This figure shows the field profiles at $1.2\mu\text{s}$ and the profiles do not compare with the experiment. These field profiles with very thin current sheath seem to be due to near zero thermal conductivity used in the hybrid simulation. If there is no thermal conduction, the initial current flow on the plasma skin will result in a local overheating of the skin. This high temperature skin will in turn results in a very

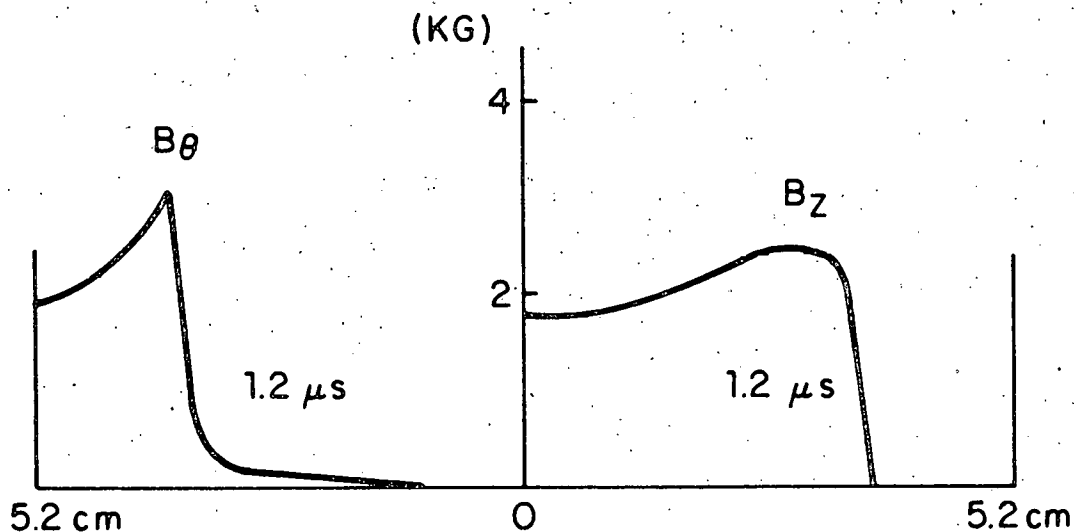


FIG. 15. POLOIDAL AND TOROIDAL FIELD PROFILES FROM THE HYBRID CALCULATION USING CLASSICAL RESISTIVITY

thin current sheath, since the classical resistivity decreases at $\propto T^{-3/2}$. On the other hand, for the anomalous resistivity, the temperature dependence is much weaker and, moreover, there is additional dependence on the current density and the plasma density. ($\eta_a \propto \frac{j}{T^{1/2} n^{3/2}}$, in the high temperature limit.) Therefore, the current sheath will spread, even when there is no thermal conduction as in the hybrid simulation case. This explains the similar results obtained from the two simulations both using the same anomalous resistivity. Other reasons for the similarity of the two simulation results should be also mentioned, that is, the ion reflection from the magnetic piston which can be seen in the hybrid code and not in our fluid code, is small due to the high bias field, and also the toroidal curvature effect which is naturally taken into account in our two dimensional code gives small difference in the overall shape of the magnetic fields due to the large aspect ratio of ZT-I. (Nevertheless, this toroidal curvature effect gives rise to some important phenomena as explained in Section (4.3.B).) As a conclusion for this paragraph, we note that the calculation results with the Chodura anomalous resistivity are relatively insensitive to the thermal conductivity used, while the results with the classical resistivity are more sensitively dependent on the thermal conductivity.

Figure 16 shows the profiles of the plasma density, toroidal current density and the ratio of the anomalous resistivity to the classical resistivity η_a/η_{cl} on the vertical radius ($\theta = \frac{\pi}{2}$) at 1.6, 2.0 and 2.8 μs . Values of η_a/η_{cl} are

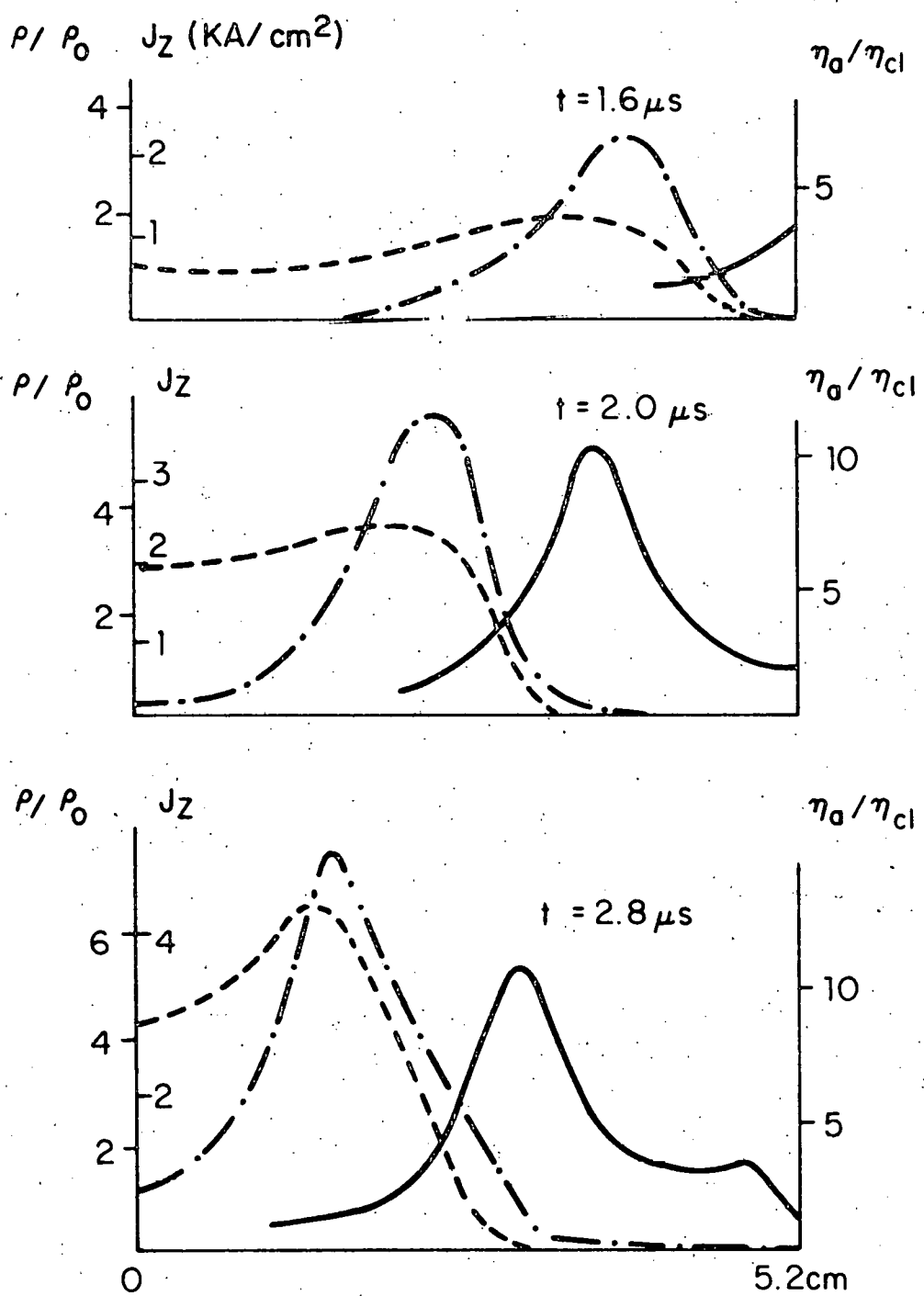


FIG. 16. PROFILES OF THE PLASMA DENSITY (-----), TOROIDAL CURRENT DENSITY (- . - . -) AND THE RATIO OF THE ANOMALOUS RESISTIVITY TO CLASSICAL ONE (————)

P.M.E. - 3400 J.D.

shown only where the anomalous resistivity is turned on (i.e., $\eta_a > \eta_{cl}$). Inside the main plasma column, the resistivity remains essentially classical. The anomalous resistivity affects mainly on the outside low density region. This effect on the low density region would not much influence the overall plasma behavior, since the current in this region is already weak. This is because the pressure gradient is small in this region and therefore, the fluid velocity approximately cancels out the electric field (i.e., $\frac{\mathbf{u} \times \mathbf{B}}{c} \sim -\mathbf{E}$). Thus, we note that the fluid velocity in the calculation should be given correctly in the low density region especially when the classical resistivity is used (cf. Section (3.3)).

The discussion on Figure 16 explains the fact that the field profiles, one obtained using the anomalous resistivity and the other using the classical resistivity, were quite close (Figure 14). Also, the temperature profiles which would be strongly affected by the resistivity coefficient are quite close for the two cases as shown in Figure 17. The solid curves are obtained using the anomalous resistivity and the dashed curves using the classical one.

All the above discussions are true only in the initial setting up phase of the discharge, of which the time scale is much shorter than the dissipation time scales. Quite different results are obtained in the later dissipation phase when the dissipation process becomes more important. This later phase will be discussed in Section (4.3.A). It will be shown that the anomalous resistivity is not valid for this

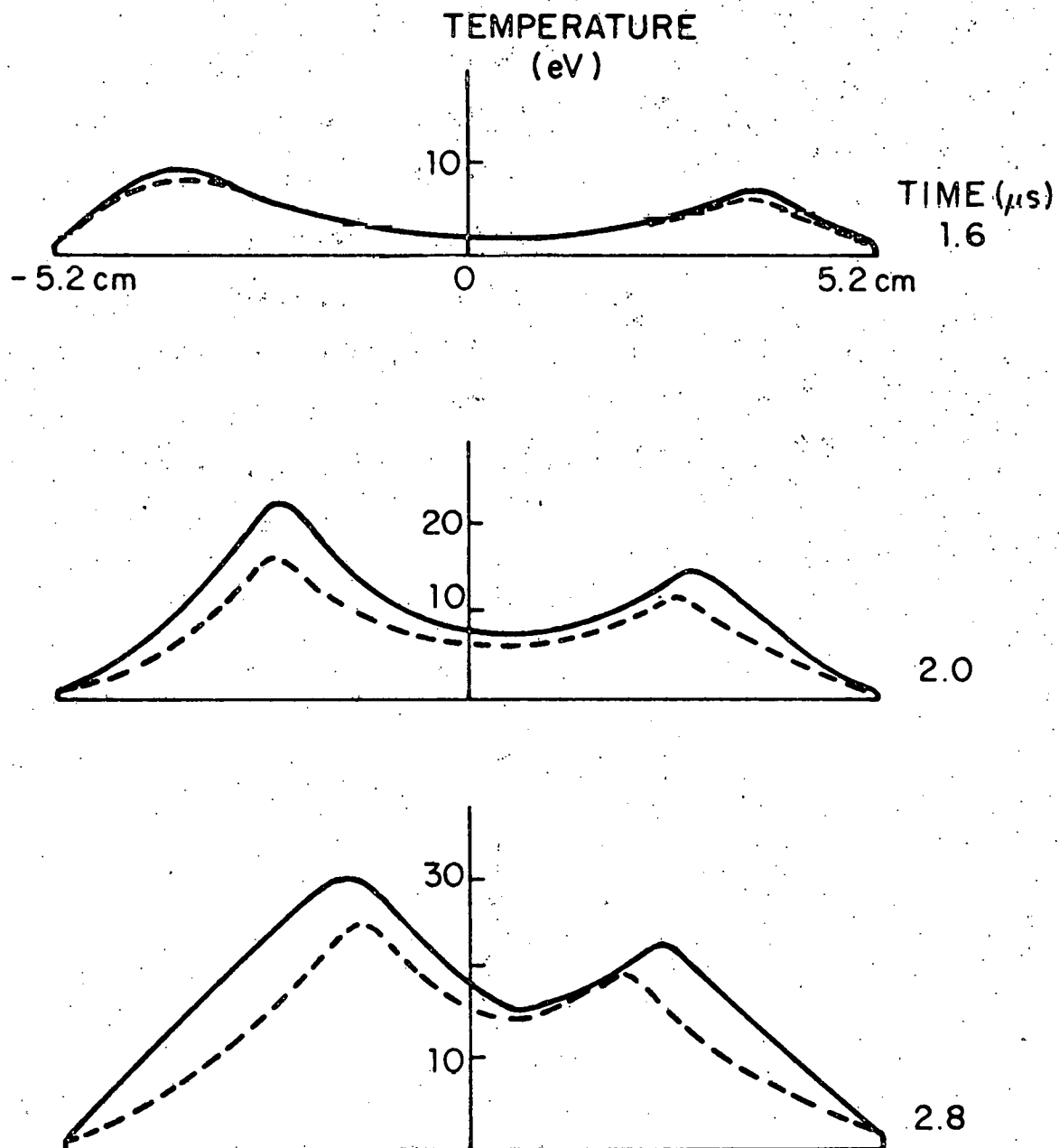


FIG.17. TEMPERATURE PROFILES AT 1.6, 2.0 AND 2.8 μ s.

SOLID CURVES ARE OBTAINED USING THE ANOMALOUS RESISTIVITY AND DASHED CURVES THE CLASSICAL ONE.

phase and instead, the classical resistivity gives better comparison with the experiment.

The results discussed in this section are consistent with the simulation results of HBTX-1 reverse-field pinch by the Culham group and also with that of Eta-Beta reverse-field pinch by the University of Padua group. The Culham group reported that 'Both the setting up and the decay phase of HBTX-1 can be simulated numerically using classical coefficients.' The Padua group reported²⁰ that 'No clear indication about the classical or anomalous nature of the resistivity appears from the simulated gross plasma behavior of Eta-Beta experiment.'

4.3 ZT-S Reverse-Field Pinch

4.3.A. General Behavior of the Plasma

ZT-S reverse-field pinch⁵ is an enlarged version of the previous ZT-I experiment. The minor radius has increased modestly from 5.2cm to 7.7cm while the major radius remained about the same. One of the main purposes of this larger size is to study the scaling behavior of reverse-field pinches. This would also give more information for the design of the larger ZT-40 experiment.

Figure 18 shows the two field boundary conditions of the discharge we simulate. The toroidal plasma current rises to 55kA in 1.6 μ s, and the initial toroidal bias field of 0.5kG is reversed at 3.6 μ s. Initially, $\rho_0 = 2.2 \times 10^{-9} \text{ gm}\cdot\text{cm}^{-3}$ deuterium filling gas with $T_0 = 2\text{ev}$ is present.

In Figure 19, the density profiles on the poloidal plane are shown at various instants of time. The initial pinching reaches its maximum at 2.1 μ s, and then the plasma expands until 2.6 μ s. Several successively smaller pinching and bouncing motions follow until a relatively quiet phase is reached at 4.6 μ s (cf. discussions on Figure 21). The plasma stays well away from the wall in the calculation throughout the discharge.

Figure 20 shows the calculated toroidal and poloidal field profiles (shown in solid lines) on the vertical radius ($\theta = \frac{\pi}{2}$) of the poloidal plane, compared with the experimental values (shown in dashed lines), at 2.6 and 4.6 μ s. Chodura anomalous resistivity is used for the results shown. These results do not change significantly when classical resistivity is used instead, in this initial setting-up phase, as discussed

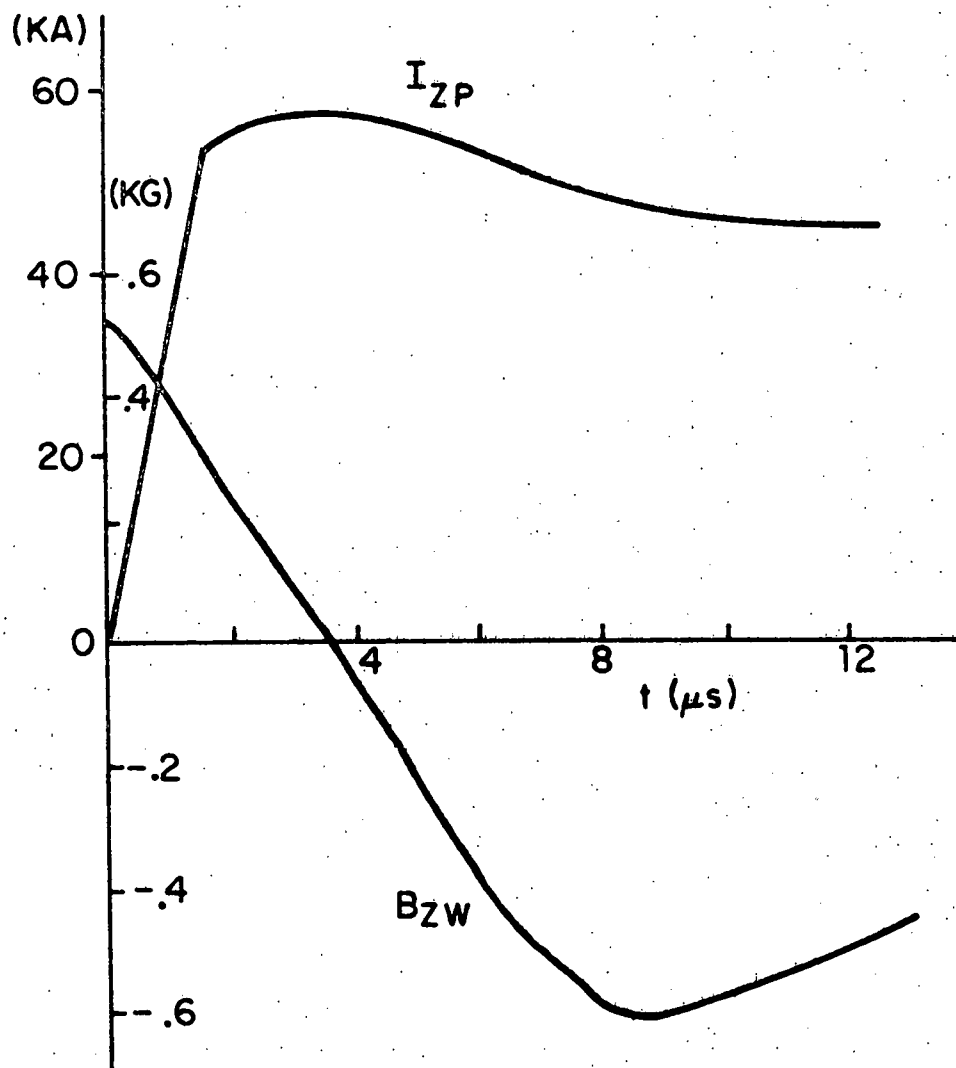


FIG.18. TIME TRACES OF THE TOTAL TOROIDAL PLASMA CURRENT I_{zP} AND THE TOROIDAL FIELD AT THE WALL, B_{zw} .

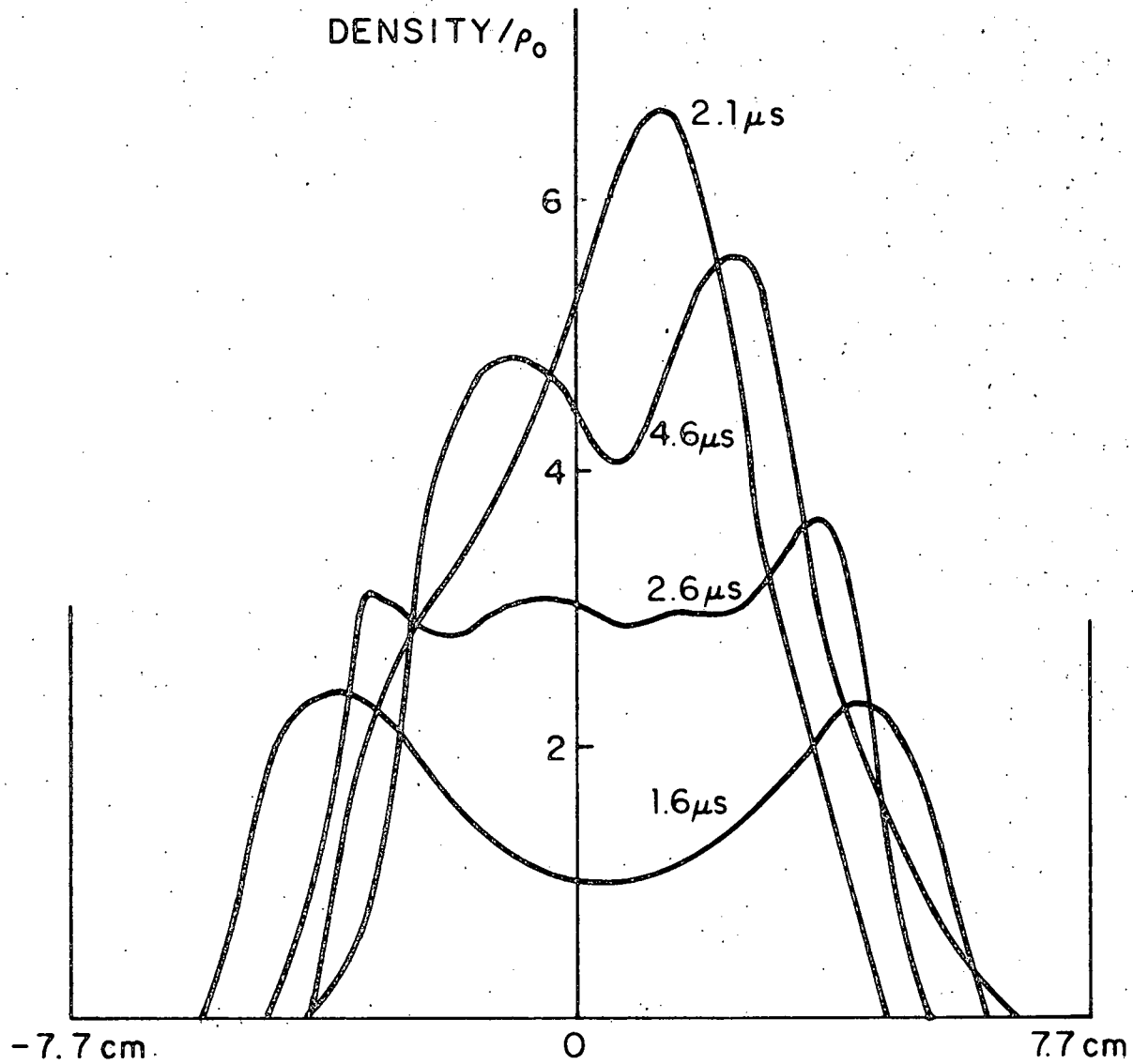


FIG. 19. DENSITY PROFILES NORMALIZED TO THE INITIAL DENSITY

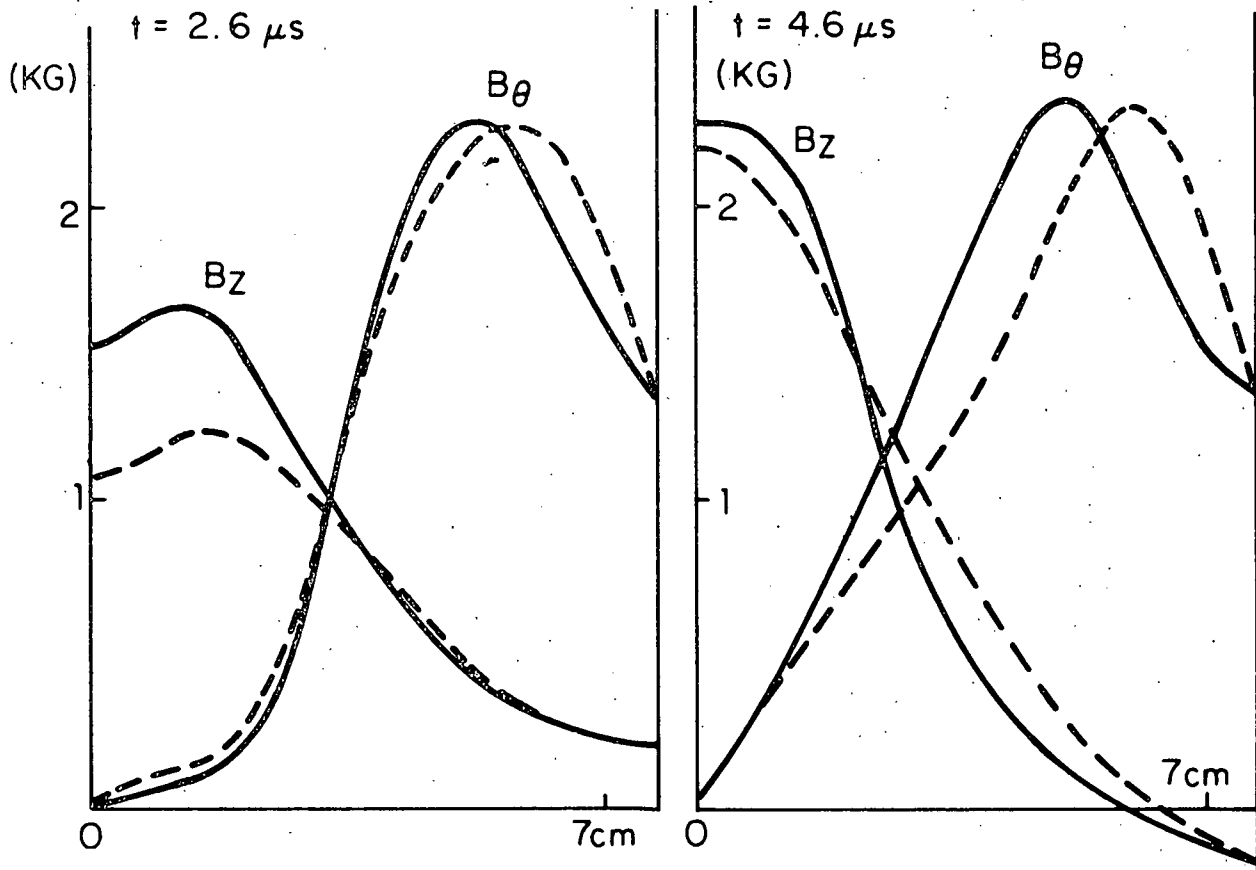


FIG. 20. TOROIDAL AND POLOIDAL FIELD PROFILES ON THE VERTICAL RADIUS ($\theta = \pi/2$) OF THE POLOIDAL PLANE ARE COMPARED BETWEEN THE CALCULATION (—) AND EXPERIMENT (-----) AT 2.6 AND 4.6 μs .

in the case of the ZT-I experiment (Section 4.2). The agreement shown in Figure 20 is quite satisfactory. The large difference between the two toroidal field values at the center ($r = 0$) at $2.6\mu\text{s}$ is due to a small time lag of the calculation, which can be seen in the next figure.

In Figure 21, the toroidal field at the minor axis is shown as a function of time. The points represent the experimental values. The second and third points correspond to $2.6\mu\text{s}$ and $4.6\mu\text{s}$ depicted in the previous Figure 20. The observed point at $2.6\mu\text{s}$ shows that the large discrepancy between two field values at the center is due to a small time lag of the calculation. This figure also shows that a quiet phase is reached at about $5\mu\text{s}$, as discussed before for Figure 19. This quiet phase actually corresponds to the second phase of the pinch when the toroidal drifting motion is dominant, as explained in Section (4.1) for SP-1 screw pinch. This can be readily seen from Figure 24 in the next section, which shows the time trace of the toroidal shift of the same discharge, and this figure also shows that the toroidal drifting oscillation is much smaller in this reverse-field pinch case than in the previous SP-1 screw pinch discharge. The good agreement shown in Figure 21 in the later quiet phase, i.e., 6.6 and $8.6\mu\text{s}$, is found misleading, since the field profiles as a whole do not compare each other between calculation and experiment, as will be seen in Figure 23. Also, we will see that the anomalous resistivity and the pure classical resistivity give quite different results at this later time. This is because the

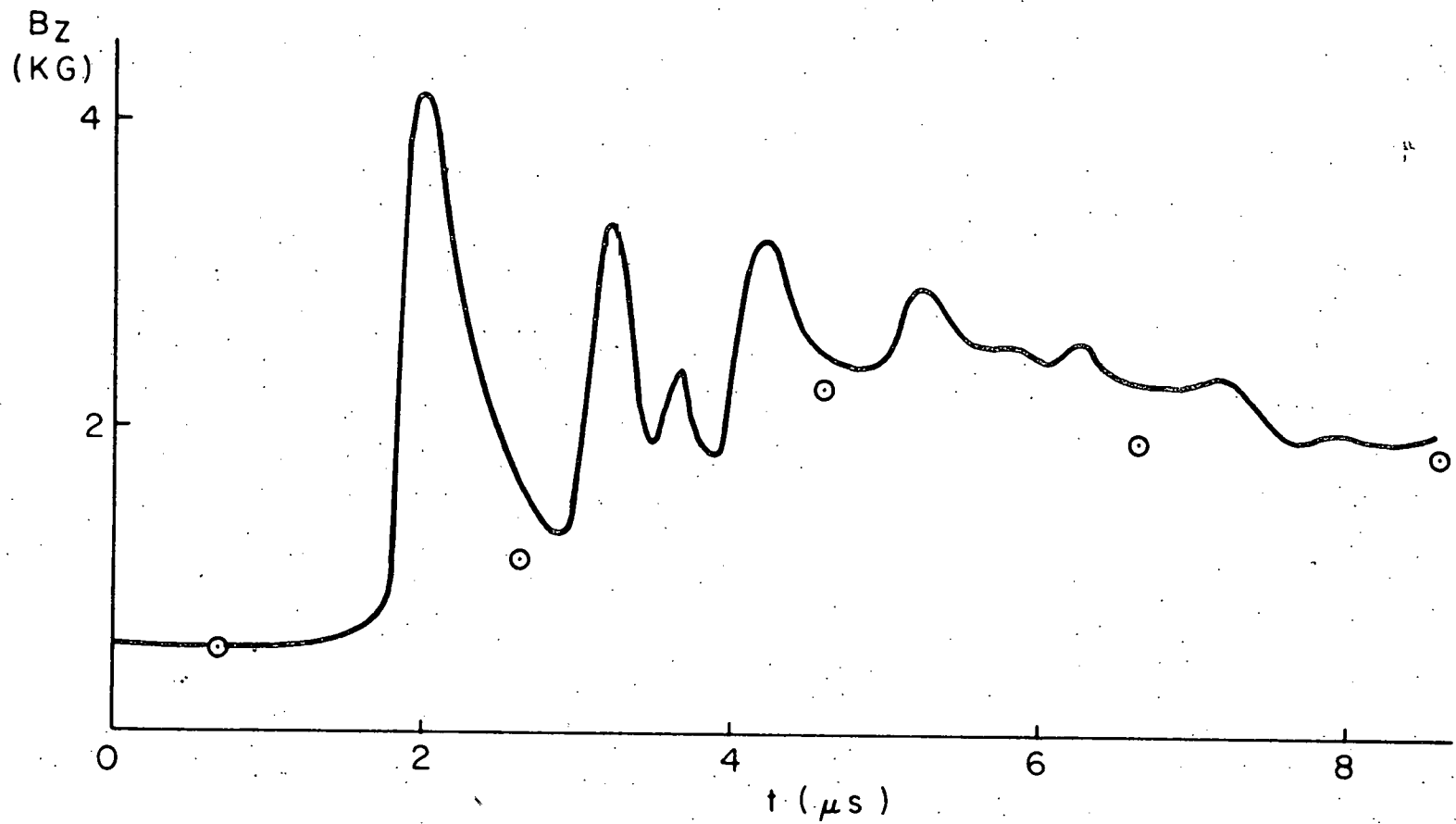


FIG. 21. TIME TRACE OF THE CALCULATED TOROIDAL FIELD AT THE MINOR AXIS.
THE POINTS REPRESENT THE EXPERIMENTAL VALUES.

elapsed time is by now comparable to the field diffusion time of $\sim 10\mu\text{s}$, and more importantly, because the anomalous resistivity is turned on throughout the plasma by $6\mu\text{s}$, as shown in Figure 22. This figure shows the profiles of the plasma density, toroidal current density and the ratio of the anomalous resistivity to the classical resistivity, and should be compared to Figure 16 of the ZT-I case. The anomalous resistivity is turned on throughout since the current and temperature are high throughout the plasma.

Figure 23 compares the toroidal and poloidal field profiles between the experiment and the two separate calculations each using different resistivity coefficient. The broken lines represent the experimental values and the semi-broken lines the calculation results using the anomalous resistivity coefficient. The semi-broken lines indicate much more compressed plasma profiles than the observed results indicate. This over-compression is probably due to the large loss of trapped toroidal flux in the calculation. The trapped flux in this calculation appears to be less than half of the observed. (Recall that the cross section is circular.) This strongly suggests that the real resistivity is much less than the anomalous resistivity used. Therefore, another calculation was made where the anomalous resistivity was replaced by pure classical resistivity after $5\mu\text{s}$ and the thermal conductivity coefficient was also reduced by a factor of 10 at the same time. The solid lines in Figure 23 represent this result, and they show much better agreement with the experiment. The trapped flux in this

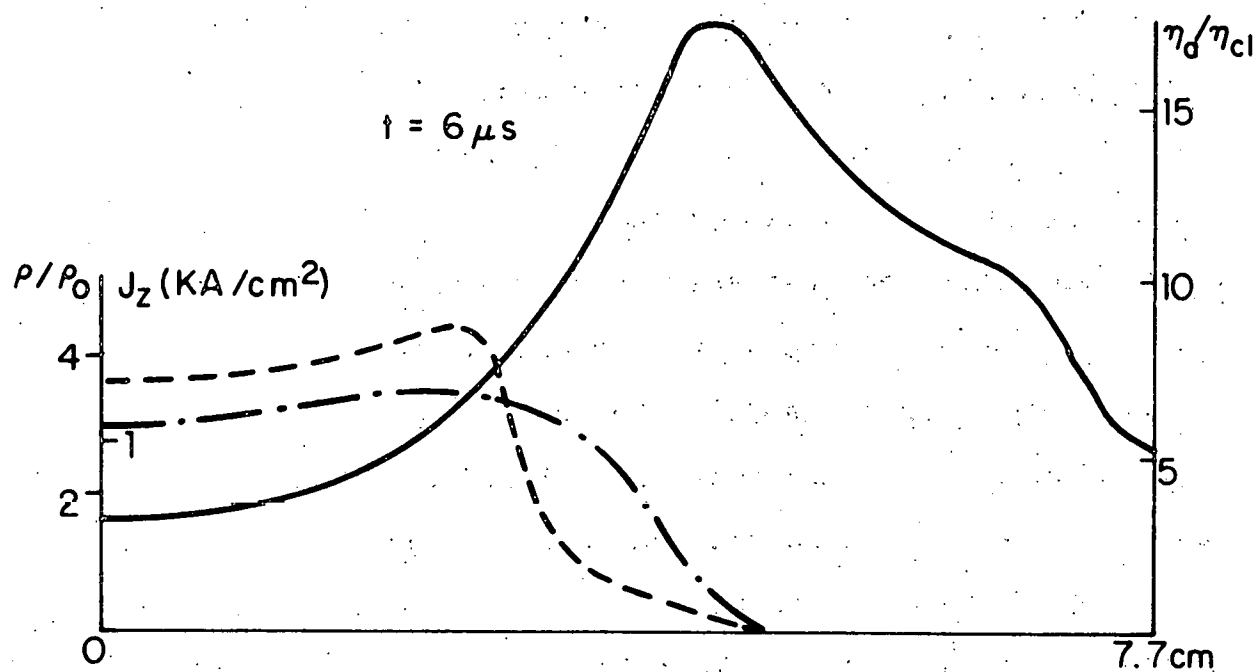


FIG. 22. PROFILES OF THE PLASMA DENSITY (-----), TOROIDAL CURRENT DENSITY (- . - . -) AND THE RATIO OF THE ANOMALOUS RESISTIVITY TO THE CLASSICAL ONE (————) AT 6 μs .

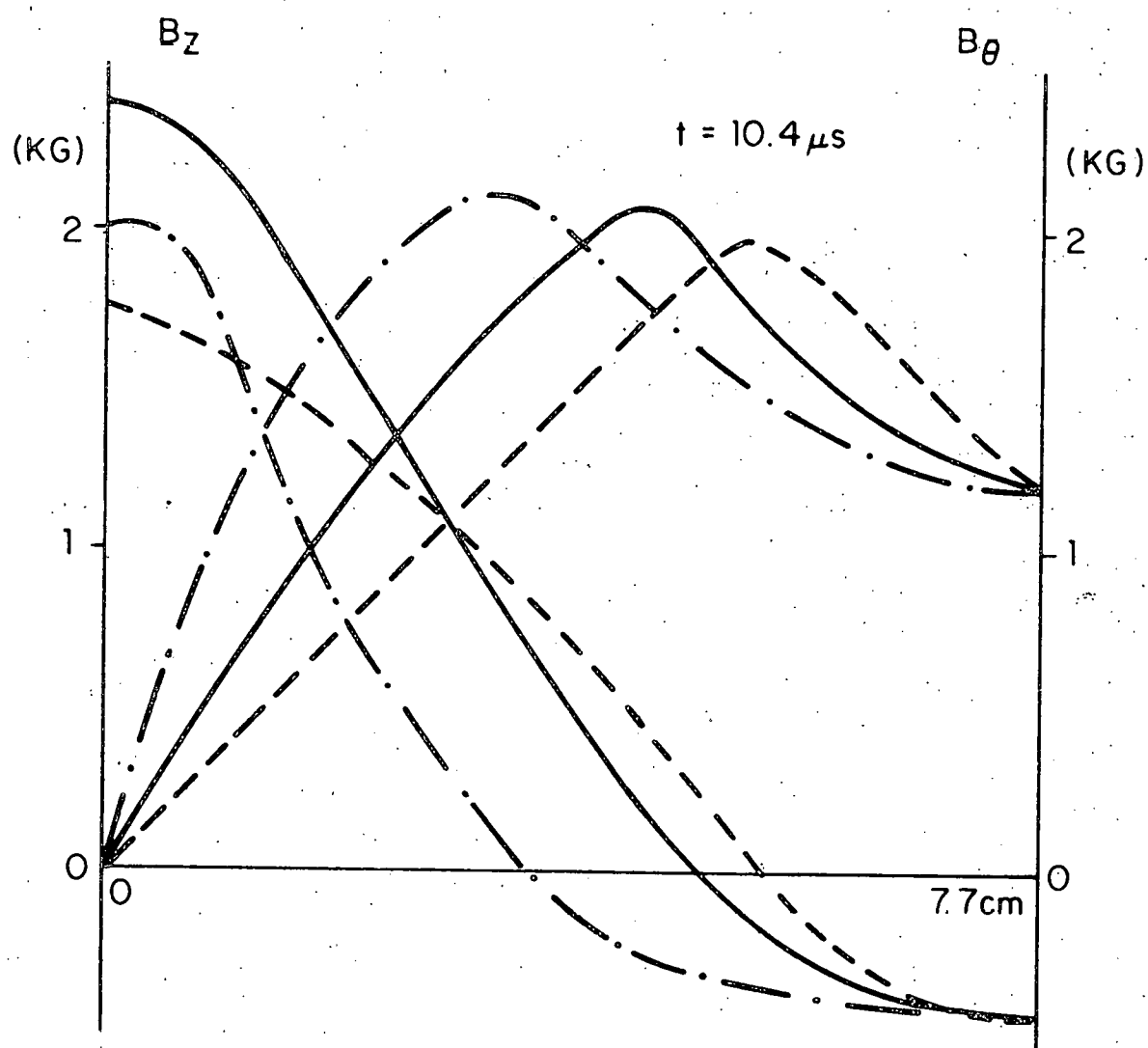


FIG.23. TOROIDAL AND POLOIDAL FIELD PROFILES AT $10.4 \mu s$ ARE COMPARED BETWEEN THE EXPERIMENT (-----) AND TWO SEPARATE CALCULATION RESULTS, ONE OBTAINED USING THE ANOMALOUS RESISTIVITY THROUGHOUT (- · - · -) AND THE OTHER BY TURNING IT OFF AT $5 \mu s$ (————).

calculation is about the same as the observed one. If the thermal conductivity is kept the same instead of reducing 10-fold, the plasma collapses a little more but the trapped flux remains about the same.

From these discussions, we conclude that, for the simulation of the later, quiet phase, the Chodura anomalous resistivity algorithm is invalid, and the real resistivity seems to be quite close to the classical one. This is consistent with a concept of the anomalous resistivity being the effect of the plasma instabilities generated in the initial violent phase. (See also the related discussions at the end of Section (4.2).) Whether this will be still true for the higher temperature and higher current density regime is an important question which should be investigated, for example, by ZT-40 experiment.

Now the scaling behavior of ZT-S is briefly discussed. Since diffusion times $\propto a^2/D$ (a = plasma radius, D = diffusion coefficient) and the stable period in a reverse-field pinch discharge is found to be limited by field diffusion,³² one expects that a ZT-S discharge would have at least twice longer stable time compared to a similar ZT-I discharge. This was confirmed in the experiment, which measured a stable time of $\sim 30\mu\text{s}$ in ZT-S compared to $\sim 15\mu\text{s}$ in ZT-I. It is also reported that a significant current penetration into the center of the plasma occurs at $\sim 5\mu\text{s}$ in ZT-S compared to $\sim 2\mu\text{s}$ in ZT-I. This was also seen in the calculation. Here, we just refer to

Figure 14 on page and Figure 20 on page . From the poloidal field profiles of ZT-I in Figure 14, we see that the toroidal current largely penetrated to the center at $2\mu\text{s}$. On the other hand, Figure 20 shows negligible current penetration to the center at $2.6\mu\text{s}$ in ZT-S. A significant current penetration occurs at $\sim 4\mu\text{s}$ and the field profiles at $4.6\mu\text{s}$ becomes as shown. A more detailed discussion on the scaling behavior is made in Section (4.4) where the larger ZT-40 experiment is studied.

4.3.B. Toroidal Curvature Effects

In this section, we study the toroidal curvature effects, using simulation results of the same discharge studied in the previous section. These toroidal effects are automatically incorporated in our two space dimensional code. In Figure 24, the calculated toroidal shift of the magnetic axis is shown as a function of time. This shows an outward toroidal shift of ~ 1 cm. (The minor radius of the torus is 7.7cm.) From this figure and Figure 21 on page , we can see that the three stages of the discharge, i.e., the bouncing, toroidal drift and diffusion phase occur as in the case of SP-1 screw pinch. The time scales of these three phases can be similarly explained as for the SP-1 case in Section (4.1). However, the toroidal drift oscillation appears to be much smaller compared to the SP-1 case in Figure 9.

Although this toroidal effect does not much influence the overall plasma profiles, it nevertheless, gives rise to many important phenomena such as the toroidal plasma flow which will be discussed later. Some diagnostic interpretations are also influenced by the toroidal effect, as shown in the next figure. Figure 25 shows two calculated current density profiles at $6.5\mu\text{s}$. The solid line represents the current density on the vertical diameter ($\theta = \frac{\pi}{2}$) of the poloidal plane, while the dashed line on the horizontal diameter. We note that the vertical profile appears almost flat in the main plasma region, while the horizontal profile is still hollow, revealing that the whole two-dimensional current density profile is actually

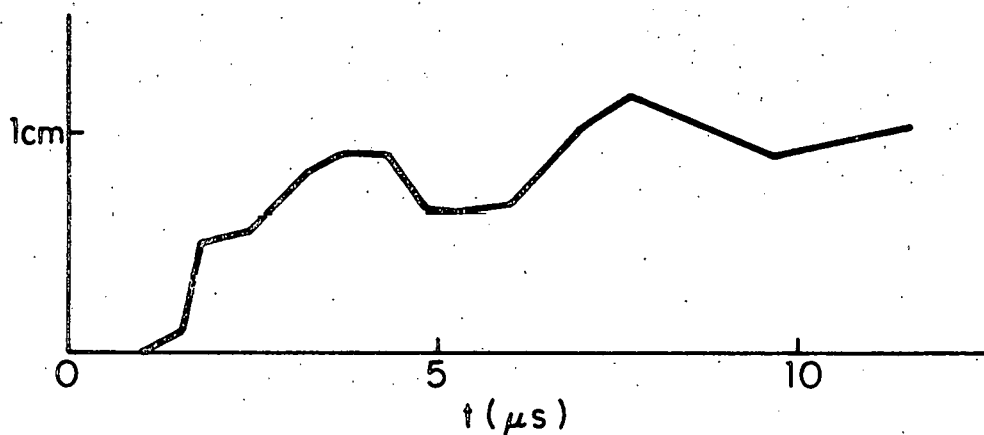


FIG. 24. TIME TRACE OF THE TOROIDAL SHIFT OF THE MAGNETIC AXIS.

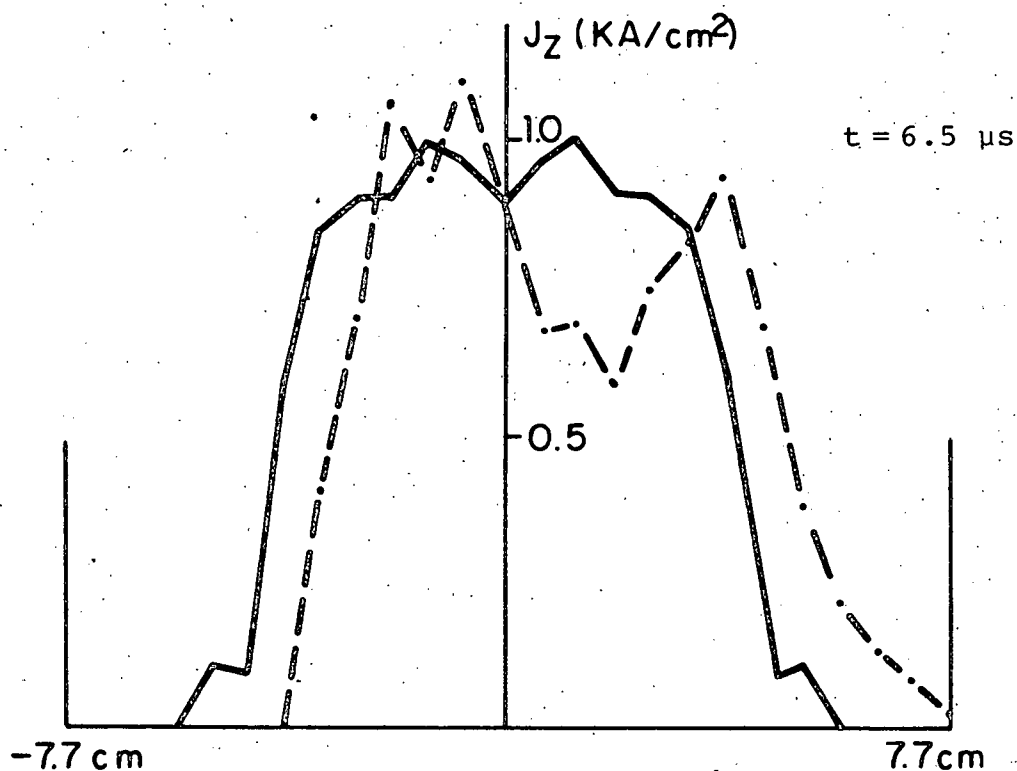


FIG. 25. CALCULATED TOROIDAL CURRENT DENSITY PROFILES AT 6.5 μ s. THE SOLID LINE REPRESENTS THE VALUES ON THE VERTICAL DIAMETER ($\theta = \pm\pi/2$) OF THE POLOIDAL PLANE, AND THE DASHED LINE ON THE HORIZONTAL DIAMETER.

annular. Since the ZT-S machine has only vertical diagnostic ports, the experiment, at this instant, would measure only the flat vertical profile, and may give wrong conclusion that the whole current profile is already flat. The resistivity so deduced would be erroneously high. This shows an example which indicates that a horizontal diagnostic port would be necessary due to the toroidal curvature effect, for accurate diagnostics.

The large difference between the vertical and horizontal profiles shown in Figure 25 is, of course, directly related to the toroidal shift shown in Figure 24. This shift of $\sim 1\text{cm}$ is about twice larger than the toroidal shift predicted from an equilibrium calculation of the Los Alamos group.³³ This difference is likely due to a strong toroidal plasma flow which is seen in our calculation but not included in the equilibrium calculation. Figure 26(a) shows the time traces of the three computed velocity components, i.e., the radial (U_r), azimuthal (U_θ), and toroidal (U_ϕ) velocities, at a point depicted in Figure 26(b). The toroidal velocity is antisymmetric against the midplane as depicted in Figure 26(b). Therefore, the angular momentum is conserved. The oscillations of the radial and azimuthal velocities are largely damped away at $5\mu\text{s}$, but the toroidal velocity persists and increases until it reaches a steady magnitude at $\sim 10\mu\text{s}$. Therefore, the final plasma state is a steady quasi-equilibrium state rather than a static one, as in the SP-1 case. The steady equilibrium is a very complex problem which is largely remained to be investigated,

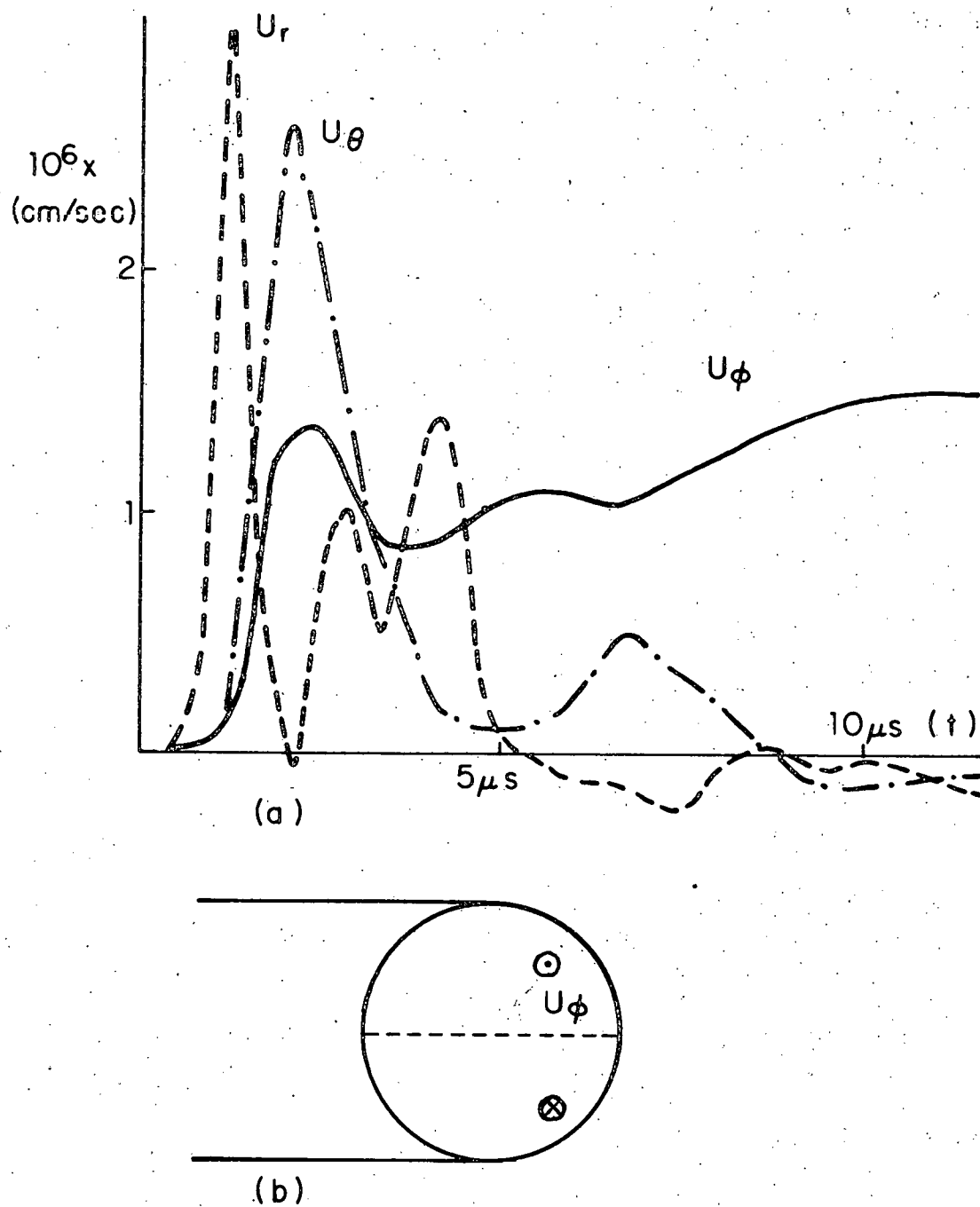


FIG.26. TIME TRACES OF THE RADIAL (U_r), AZIMUTHAL (U_θ) AND TOROIDAL (U_ϕ) VELOCITIES AT THE POINT DEPICTED IN FIGURE (b).

especially in a finite beta case. (For a review, see for example, Green (34)). Here we just estimate the effect of the toroidal plasma flow on the toroidal shift. For a steady equilibrium, the convection term should be added to the usual pressure balance equation of the static equilibrium, so that

$$\rho \underline{u} \cdot \nabla \underline{u} + \nabla p = \frac{\underline{j} \times \underline{B}}{c}$$

For the present case, $\rho \underline{u} \cdot \nabla \underline{u} = -\rho \frac{u_\phi^2}{R} \hat{R}$, with $u_\phi \sim 1.5 \times 10^6$ cm/sec.

This 'centrifugal pseudo pressure gradient' is an order of magnitude smaller than the average magnetic pressure gradient $(\frac{\underline{j} \times \underline{B}}{c})$. Therefore, we can expect that an additional toroidal shift of about one-tenth of the torus minor radius is necessary to balance this centrifugal force.

We are not aware of any toroidal plasma flow measurements which can be checked with the present results. Although a toroidal plasma flow of $u_\phi = \sim 10^6$ cm/sec has been reported in the ORMAK experiment in Oak Ridge and also in the PLT experiment in Princeton³⁵, both under a neutral beam injection heating, these results were attributed to the momentum transfer from the neutral beam. The toroidal velocities, therefore, did not show antisymmetric profiles against the midplane. Nevertheless, we believe that a careful measurement will reveal a strong toroidal flow with antisymmetric profile against the midplane, especially in the reverse-field pinches. A physical reasoning for the behavior of this toroidal flow is given in the following.

In an axisymmetric pinch, the toroidal plasma flow is

generated only by $\frac{j \times B}{c}$ force of the poloidal plane (i.e., $\frac{j_{\text{pol}} \times B_{\text{pol}}}{c}$), since the v_p term does not have a toroidal component. The behavior of the toroidal flow shown in Figure 26(a) is directly connected to the fact that the current surfaces tend to shift more outward than the magnetic field surfaces as shown in the next figure. (In the infinite aspect ratio limit, these two surfaces would coincide. Thus, $j_{\text{pol}} \times B_{\text{pol}} = 0$ and no toroidal flow will occur.) Figure 27 shows the magnetic field lines (the solid lines) and the current lines (the dashed lines) on the poloidal plane at $9\mu\text{s}$. The current lines appear to be more shifted outward than the magnetic field lines, and thus, $\frac{j_{\text{pol}} \times B_{\text{pol}}}{c}$ force antisymmetric against the midplane results. During the initial phase of the discharge, the poloidal current lines will be shifted approximately together with the plasma density, since the pinch time ($\sim 1\mu\text{s}$) is much shorter than the field diffusion time ($\sim 10\mu\text{s}$) and thus the toroidal magnetic field will be almost frozen in the plasma. On the other hand, the toroidal shift of the poloidal field lines will be slower, since the toroidal currents are continuously created at the wall and then carried in by convection and diffusion. These would be the reasons for the steep increase of the toroidal velocity at $\sim 2\mu\text{s}$ (Figure 26(a)) when the first pinching reaches its maximum.

At a later time, the toroidal velocity reaches a steady magnitude, indicating that the field and current lines on the poloidal plane approximately coincide (i.e., $j_p \times B_p \approx 0$). The field convection due to the velocity is

$$\nabla \times (\underline{u} \times \underline{B}) = (B \cdot \nabla) \underline{u} - (\underline{u} \cdot \nabla) \underline{B} - \underline{B} (\nabla \cdot \underline{u}).$$

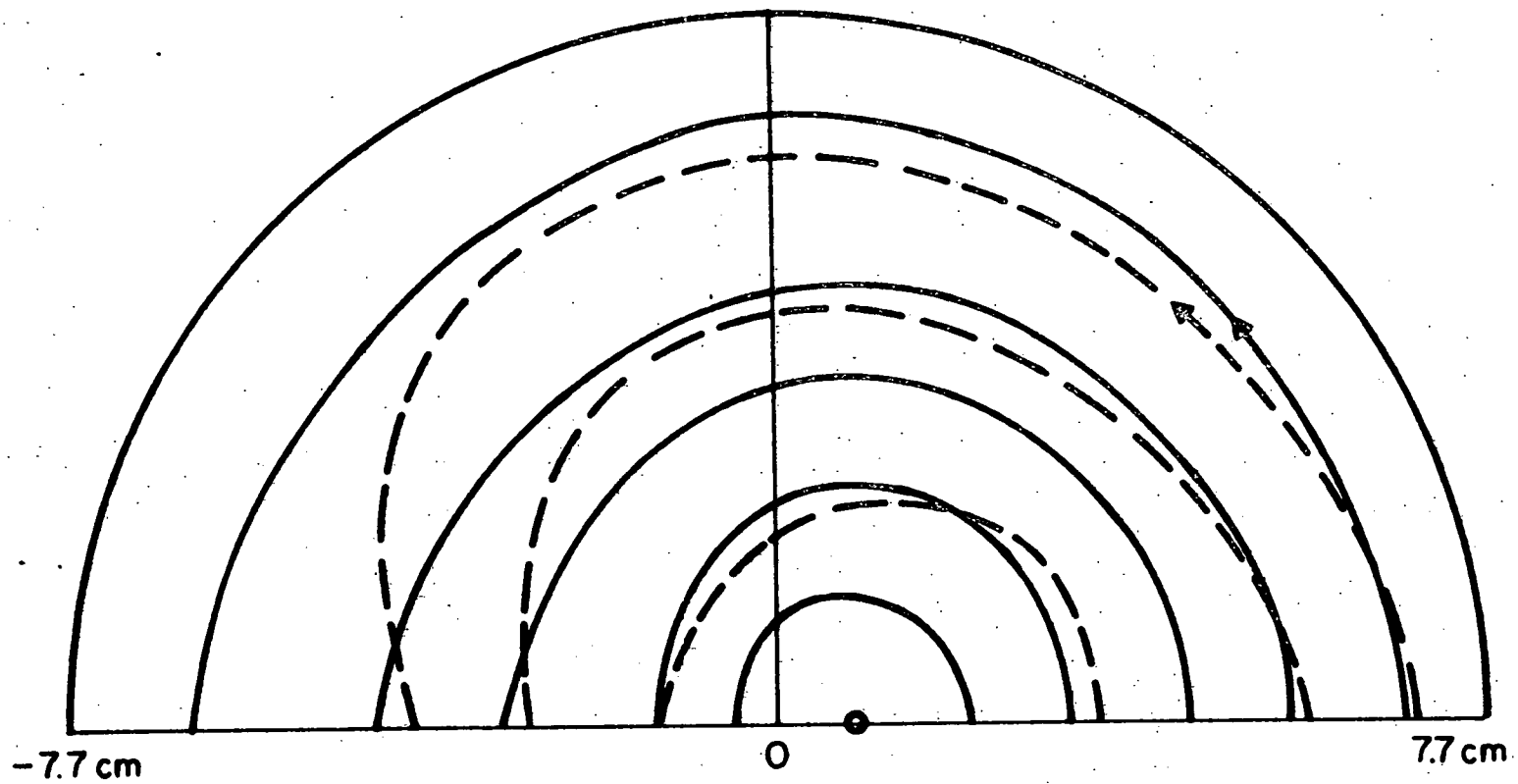


FIG. 27. MAGNETIC FIELD LINES (————) AND THE CURRENT LINES (-----)
ON THE POLOIDAL PLANE AT 9 μ s.

With $\underline{u} = u_\phi \hat{\phi}$, this is $(B_p \cdot \nabla u_\phi - u_\phi B_R/R) \hat{\phi}$. Thus, the convection due to the toroidal flow will tend to shift the current surfaces more inward than the field surfaces. This effect must be balanced by some opposing effects to keep the field and current lines on the poloidal plane coincide. One of these opposing effects comes from the fact that much more toroidal currents are generated at the inside wall ($r = r_0$, $\theta = \pi$) than at the outside wall ($r = r_0$, $\theta = 0$). This is because the electric field at the inside wall is $\sim 50\%$ stronger than at the outside wall due to $\sim 50\%$ shorter loop length (aspect ratio = ~ 5) and the same loop voltage. The generation of the toroidal current at the wall will occur throughout the discharge to offset the dissipation loss of total current, if the discharge is power-crowbarred. For a simply-crowbarred discharge, this effect will be also present in a lesser extent. The other opposing effect would come from the fact that the toroidal field reversal is $\sim 50\%$ larger at the inside wall than at the outside wall. Thus, more dissipation of the toroidal field will occur at the inner part of the poloidal plane. Above two effects together will act to shift the current surfaces more outward than the magnetic field surfaces, thereby, balancing the field convection effect of the toroidal velocity.

A simpler explanation which is parallel to the previous one can be made as follows. The two effects described above, i.e., more toroidal currents generated at the inside wall, and more toroidal field dissipation at the inner part of the poloidal

plane, give a tendency to steepen the pitch of the magnetic field in the inner part of the poloidal plane while making the pitch less steep in the outer part. This tendency will be balanced by the field convection of the toroidal flow, as explained in the following figure. Figure 28 shows a helical magnetic field line and the directions of the toroidal plasma flow (by \rightarrow or \leftarrow) as seen in the simulation. It is readily seen that the field convection of this toroidal flow would steepen the pitch of the magnetic field at the outside (drawn in dashed lines) while making the pitch less steep at the inside. This convection effect will balance the opposite effects previously described, so that the pitch angle remains nearly stationary.

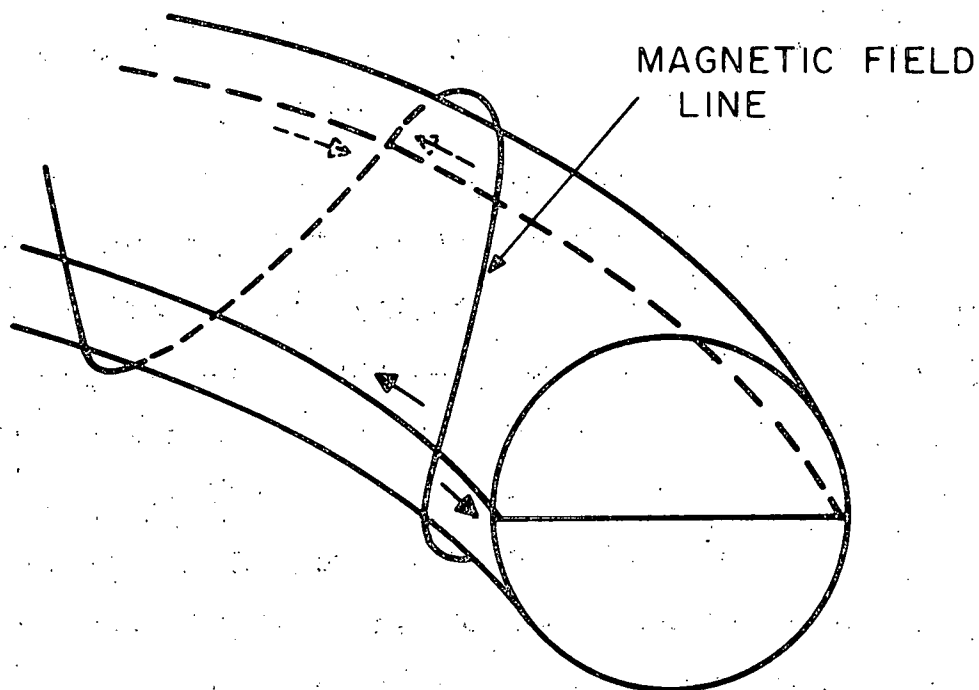


FIG. 28. A HELICAL MAGNETIC FIELD LINE AND THE DIRECTIONS OF THE VELOCITY (\rightarrow or \leftarrow) are depicted. THE VELOCITY IS ANTISYMMETRIC AGAINST THE MIDPLANE. FIELD CONVECTION DUE TO THE VELOCITY WILL STEEPEN THE FIELD PITCH AT THE OUTSIDE (DRAWN IN DASHED LINES), WHILE MAKE IT LESS STEEP AT THE INSIDE.

4.4 ZT-40 Reverse-Field Pinch

The ZT-40 experiment is a next generation reverse-field pinch, now being constructed in Los Alamos. The size will be about three times larger than ZT-S, with a 20cm minor radius and a 114cm major radius. Previous ZT-I and ZT-S experiments showed that nonideal effects due to plasma resistivity limit the stable time of the discharge. Thus, it is expected that the confinement time will be extended at least an order of magnitude since diffusion time scales as $\propto a^2/D$ (a = plasma radius; D = diffusion coefficient). This, and other scaling behaviors, will be studied with ZT-40 experiment in a size range unexplored by previous reverse-field pinches. This experiment will also yield information for design of a larger ZT-II experiment which will be projected to take the reverse-field pinch concept to $n\tau = 10^{13} - 10^{14} \text{ sec}\cdot\text{cm}^{-3}$ range.

Before studying the ZT-40, we will first examine the scaling of the mathematical model used (i.e., the dissipative magnetohydrodynamic equations (Eqs. 1)). This would give more insight to the scaling behavior of the experiments and also the three time scales, Alfvén, field diffusion, and heat diffusion times, will be clearly defined in the course.

4.4.A. Scaling of the Equations

From the resistive magnetohydrodynamic equation (Eqs. 1), we obtain the following equations by neglecting the Hall term (which is a first-order effect in the inverse aspect ratio), eliminating the variables \underline{E} and \underline{j} and denoting $C_V T$ as \tilde{T} .

($\frac{D}{Dt}$ denotes $\frac{\partial}{\partial t} + (\mathbf{u} \cdot \nabla)$ in the following.)

$$\partial \rho / \partial t + \text{div}(\rho \mathbf{u}) = 0,$$

$$\rho \frac{D\mathbf{u}}{Dt} = -(\gamma - 1) \nabla(\rho \tilde{T}) + \frac{1}{4\pi} \text{curl} \mathbf{B} \times \mathbf{B},$$

$$\rho \frac{D\tilde{T}}{Dt} = -(\gamma - 1) \rho \tilde{T} \text{div} \mathbf{u} + \frac{1}{4\pi} (\text{curl} \mathbf{B})^2 \frac{c^2}{4\pi\sigma} + \text{div} \frac{K}{C_V} \nabla \tilde{T},$$

$$\frac{\partial \mathbf{B}}{\partial t} = -\text{curl} \left(\frac{c^2}{4\pi\sigma} \text{curl} \mathbf{B} - \mathbf{u} \times \mathbf{B} \right). \quad (37)$$

The solution of these equations will be determined by the two coefficients $\frac{c^2}{4\pi\sigma}$ and $\frac{K}{C_V}$ in addition to the appropriate initial and boundary conditions. Thus, if the values of these two coefficients and the initial and boundary conditions are numerically identical in two different sets of length, mass and time units, the solutions will also be identical in the respective units. In other words, the problems so posed are mathematically equivalent.

The initial and boundary conditions of the present pinch experiments are easily scaled if we use the following length, mass and time units $[\ell_0, m_0, t_0]$, i.e., the torus minor radius r_0 as a length unit ℓ_0 , $\rho_0 \times r_0^3$ (ρ_0 : initial density) as a mass unit m_0 and the Alfvén time $\frac{r_0}{u_A} = \frac{r_0}{B_0/\sqrt{4\pi\rho_0}}$ (B_0 : bias field) as a time unit t_0 (which is equivalent to adopting $\frac{B_0}{\sqrt{4\pi}}$ as a magnetic field unit). For example, the wall is at $r = 1$, initial density 1 and bias field $\sqrt{4\pi}$ in these units. By inspecting the dimensionalities of the two coefficients $\frac{c^2}{4\pi\sigma}$ and $\frac{K}{C_V}$, we readily find the numeral values of these coefficients in

the given units, to be $\frac{c^2}{4\pi\sigma} \times \frac{t_0}{l_0^2} = \frac{t_0}{\tau_B}$ (the field diffusion time $\tau_B = \frac{4\pi\sigma r_0^2}{c^2}$) and $\frac{K}{C_V} \times \frac{l_0 t_0}{m_0} = \frac{t_0}{\tau_H}$ (the heat diffusion time $\tau_H = \frac{C_V \rho_0 r_0^2}{K}$).

As an example, the following table shows parameters for ZT-40 and ZT-S discharges which are mathematically equivalent assuming the dissipative MHD model with classical resistivity and constant thermal conductivity.

TABLE 2

	<u>ZT-40</u>	<u>ZT-S</u>	<u>Typical ZT-S Operating Parameters</u>
Filling gas	Deuterium	Deuterium	Deuterium
r_0	20cm	7.7	7.7
ρ_0	$5 \times 10^{-9} \text{ gm/cm}^3$	10×10^{-9}	5×10^{-9}
B_0	3kG	6	1
Izp max.	500kA	400	50
Rise time	10 μ s	2.5	2
T_0	10ev	15	2

The ZT-40 values are the expected typical operating parameters, while the ZT-S values are found from the equivalence condition discussed above. A typical operating parameter of ZT-S is also shown for comparison. The values ρ_0 and B_0 of ZT-S are found from the condition that $\frac{t_0}{\tau_B}$ and $\frac{t_0}{\tau_H}$ be the same as in the ZT-40 case. The values, r_0 , ρ_0 and B_0 , then, define the units $[l_0, m_0, t_0]$ as described before. From these units, all other

parameters are found such that their numeral values in these units are the same as in the ZT-40 case. The two solutions for ZT-40 and ZT-S, thus, will be identical in the respective sets of units $[\ell_0, m_0, t_0]$. Further, we can expect that the experimental results of these two discharges would also be approximately equivalent, assuming the mathematical model is roughly valid for both ZT-40 and ZT-S. The information thus gained in the ZT-S experiment would be helpful to anticipate the behavior of a bigger experiment and be valuable for the design of the ZT-40. But, the ZT-S discharge with the above shown parameters would probably become unstable, due to the high total plasma current (thus, high drift velocity). It is reported that in the high current regime, Eta-Beta discharges became unstable.²² In any case, by increasing the total current and the preionization temperature as much as possible and also appropriately adjusting the other parameters, one could study experimentally, the transition to a bigger size experiment.

4.4.B. Simulation Results

In this section, we present simulation results of a ZT-40 discharge with the parameters shown in table 2 of the previous section. The most important difference of the ZT-40 simulation from the previous smaller experiments (SP-1, ZT-I, and ZT-S) is the fact that the two ratios τ_A/τ_B and τ_A/τ_H (τ_A : Alfvén time = t_0 in the previous notation. τ_B , τ_H : field and heat diffusion time, respectively) are much smaller, due to the larger machine size and higher temperature. For this ZT-40 discharge, τ_A , τ_B , and τ_H are $\sim 1\mu\text{s}$, $\sim 1000\mu\text{s}$ and $\sim 1000\mu\text{s}$ respectively, compared to $\tau_A = \sim 1\mu\text{s}$, $\tau_B = \sim 10\mu\text{s}$ and $\tau_H = \sim 10\mu\text{s}$ in the previous experiments.

Since the allowable time steps in the simulation with a fixed number of grids are roughly proportional to the Alfvén time ($\Delta t \sim \frac{1}{500}\tau_A$ for the present code with 17×12 grids in r and θ directions), it is almost prohibitive to simulate this ZT-40 discharge to $t = \sim \tau_B$, although it was acceptable for the previous cases. Thus, the present code should be used to simulate the discharge until the plasma reaches a quasi-equilibrium diffusion state (the damping time will be found later to be $\sim 10\mu\text{s}$ due to viscosity), and then a diffusion code^{36,37} (where the inertia is neglected so that the allowable time steps are determined by the diffusion time) can take over and study the diffusion phase of the discharge. However, diffusion codes generally use a static equilibrium state as the initial condition, and may not be able to incorporate the convection effects due to the large toroidal flow present in the steady equilibrium

state obtained from the present code (cf. Section 4.3.B).

One possible solution in this case would be to use a difference scheme which heavily damps the fast time oscillations but gives small dissipation to the slow time oscillations.

As discussed in Section (4.1), the time needed to reach a quasi-equilibrium state (the damping time) is $\sim \tau_B$, when the viscosity is small. Although in the previous smaller experiments, the viscous dissipation time $\tau_v = \frac{r_0^2}{\nu}$ was much longer than other time scales, τ_v in the present case is much shorter than τ_B with $\tau_v = \sim 10\mu\text{s}$, due to higher temperature assuming classical viscosity (cf. Section 2.3). Thus, in the experiment, the damping time would be $\sim \tau_v$ ($\sim 10\mu\text{s}$). Therefore, the viscosity effect should be incorporated into the simulation for more accurate results. However, the error resulted from the neglect of the viscosity in the present code should be quite small, since the kinetic energy remains several orders of magnitude smaller than both the field energy and the thermal energy. (This is generally true in a reverse-field pinch where a strong bias field is present.) Thus, a small oscillation is still present at $t = 30\mu\text{s}$ in the following simulation, but the plasma state at this instant would be quite close to the actual quasi-equilibrium state.

Figure 29 shows the two field boundary conditions for the calculation. The toroidal plasma current rises to 500kA in $10\mu\text{s}$ and a bias field of 3kG is reversed at $10\mu\text{s}$. The filling gas is $\rho_0 = 5 \times 10^{-9} \text{gm/cm}^3$ deuterium and the initial temperature 10ev. In Figure 30, the density profiles at 10, 20 and $30\mu\text{s}$

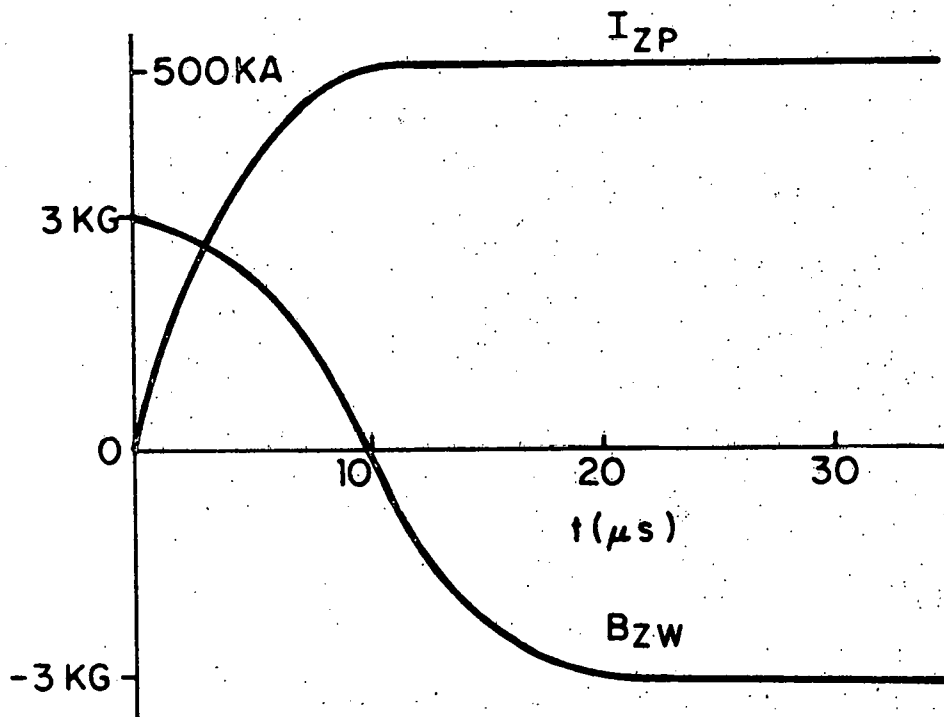


FIG. 29. TIME TRACES OF THE TOROIDAL MAGNETIC FIELD AT THE WALL AND THE TOROIDAL PLASMA CURRENT.

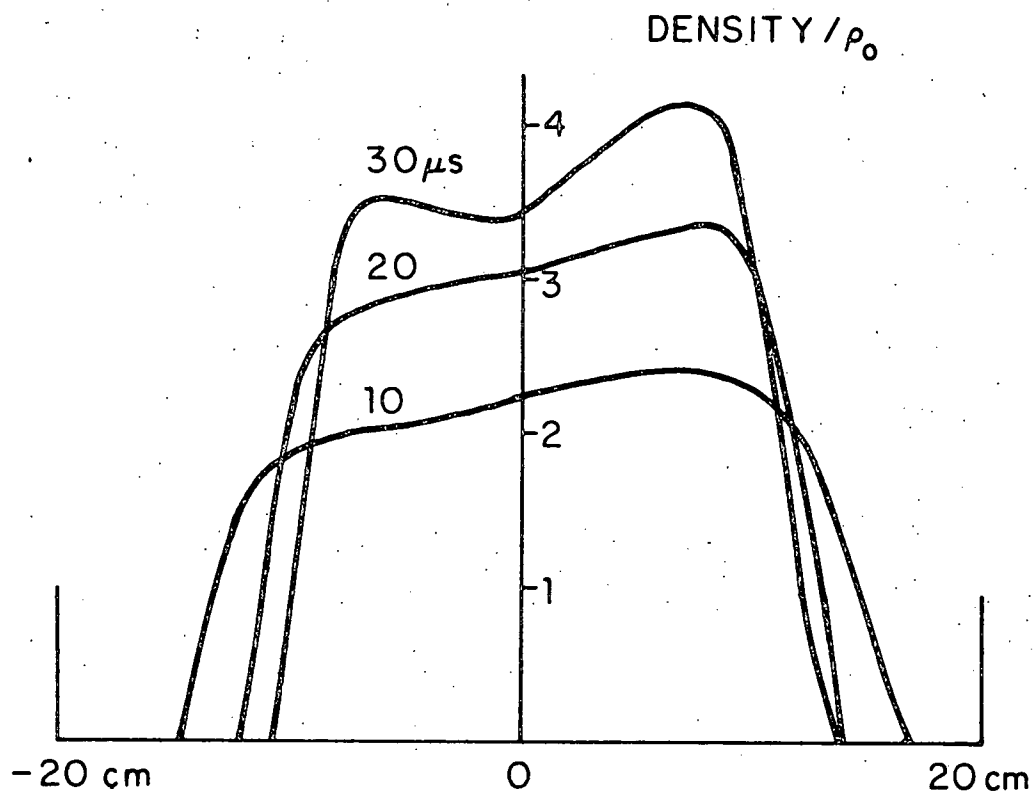


FIG. 30. DENSITY PROFILES (NORMALIZED TO THE INITIAL VALUE) AT 10, 20 and 30 μs .

are shown. As discussed previously, a small oscillation still remains at $30\mu\text{s}$, but we will regard the plasma state at this instant as a close approximation of the actual quasi-equilibrium state. The density profile at $30\mu\text{s}$ shows a compression ratio of ~ 2 as in the previously studied ZT-S discharge, but the present profile shows a much steeper transition between low density region to main plasma column compared to a corresponding profile of ZT-S (e.g., $t = 4.6\mu\text{s}$ profile in Figure 19). This is because the current sheath is very thin compared to the machine radius in the present case as will be shown later.

Figure 31 shows the toroidal and poloidal field profiles at 10, 20, and $30\mu\text{s}$. Most of the main plasma region remains essentially current-free, and a slow penetration of thin current sheath is seen. This is more obvious in Figure 32 where the toroidal and poloidal current density profiles are plotted. Considerable amount of current is present outside the main plasma region until the respective crowbar times (for the toroidal circuit at $10\mu\text{s}$ and for the poloidal circuit at $20\mu\text{s}$), due to the applied loop voltage. After the circuits are crowbarred, the current sheath becomes very thin. This result is expected because the field diffusion time is $\sim 1\text{ms}$ in this discharge, and thus, the current penetration would be still very shallow at $t = 30\mu\text{s}$. (Small poloidal currents deep inside the main plasma region are visible at times, due to the toroidal field in this region perturbed by a compressional wave.)

One important question then, is how this thin current profile will change at a much later time when $t \rightarrow \tau_B$. The current profile may broaden subsequently and eventually become

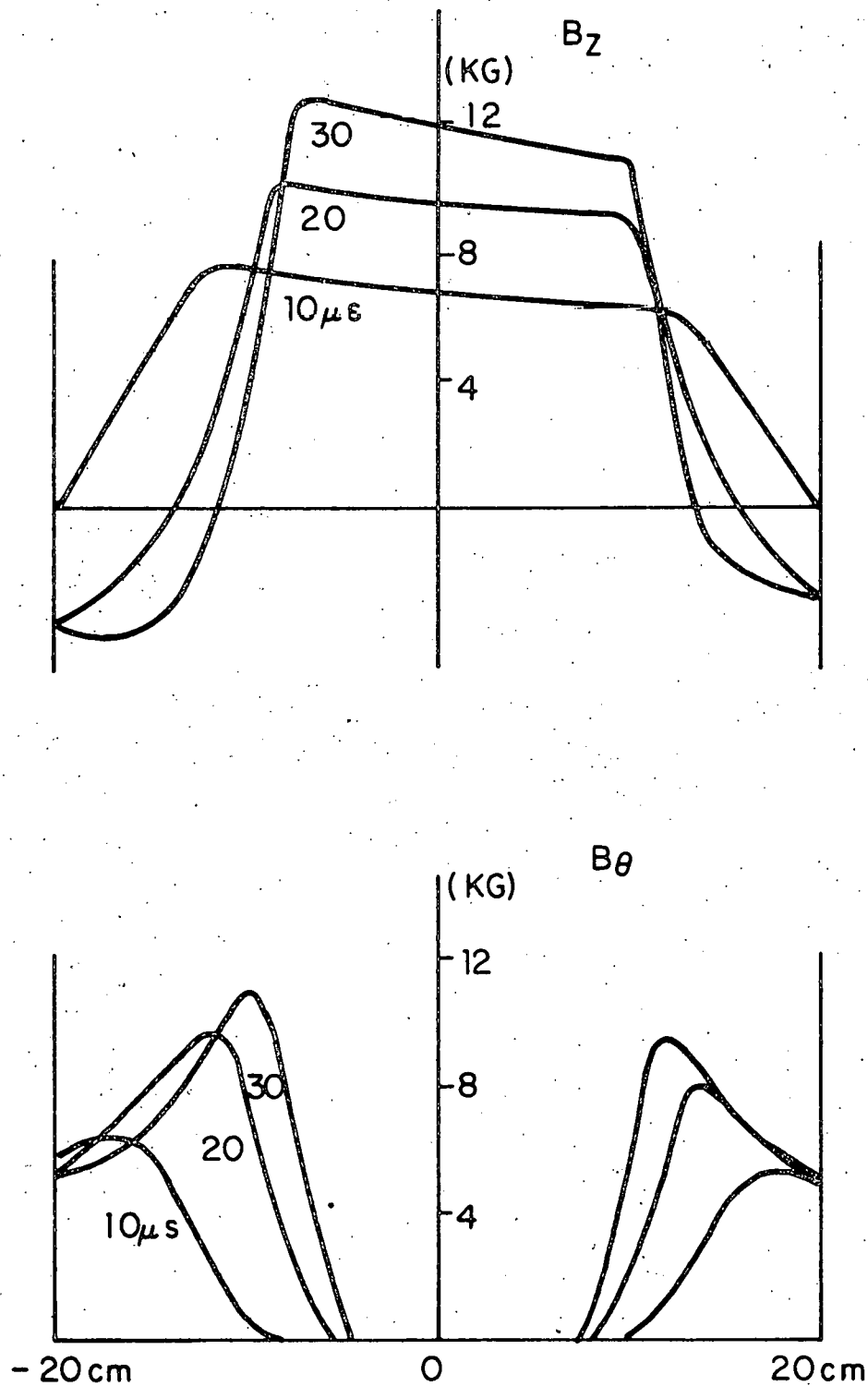


FIG. 31. TOROIDAL AND POLOIDAL FIELD PROFILES
 AT 10, 20 AND 30 μs .

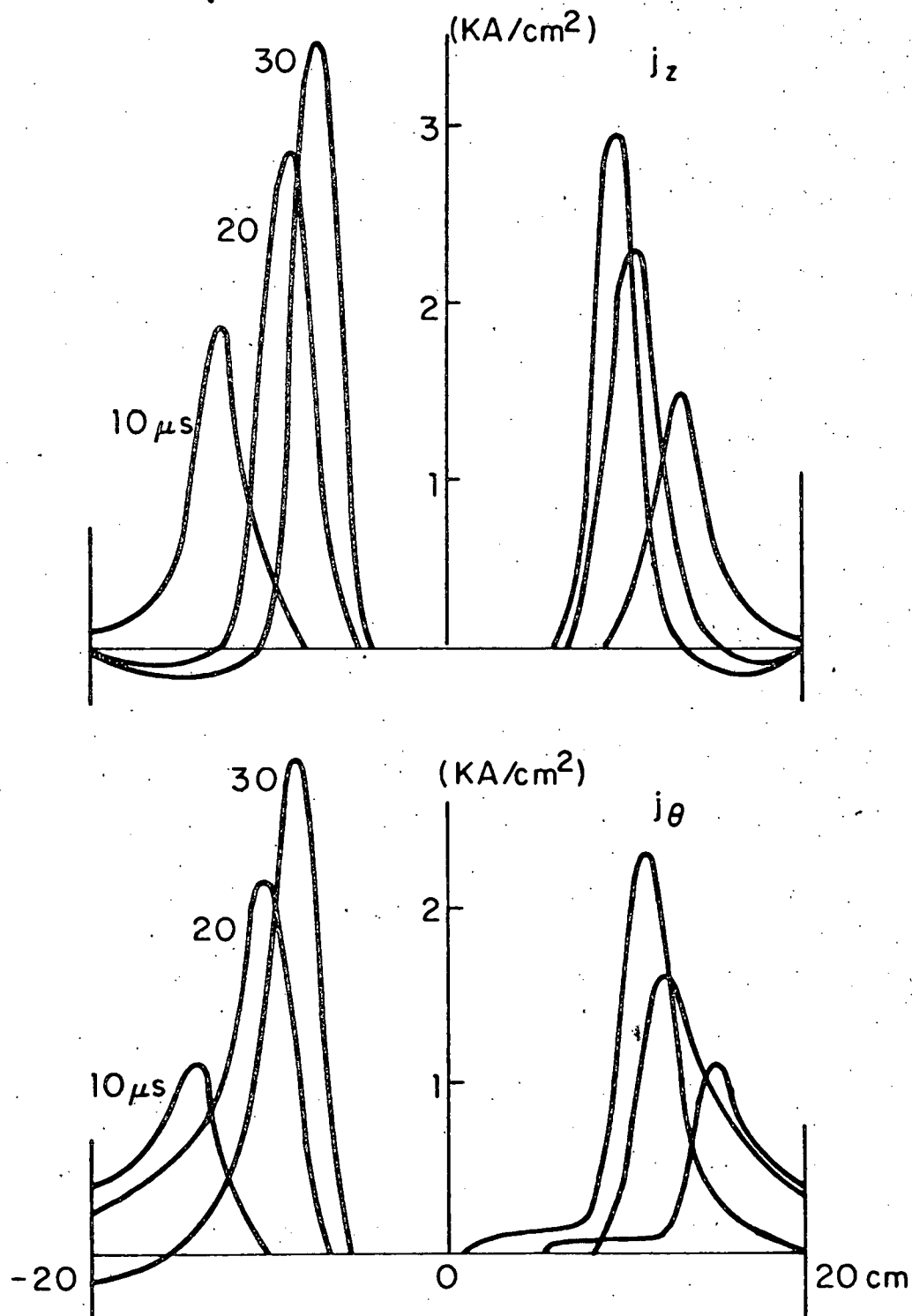


FIG. 32. TOROIDAL (j_z) AND POLOIDAL (j_θ) CURRENT DENSITY PROFILES AT 10, 20 and 30 μs .

a flat profile as seen in the ZT-S case (Figure 25). Or, the current profile may remain rather thin while it penetrates into the plasma. This question can be studied using a diffusion code. In any case, since the stability of a reverse-field pinch sensitively depends on the field profile, it would be highly valuable to be able to program desired field profiles by controlling the boundary conditions in the experiment. For a substantial programming of the field, it would be necessary to control the boundary conditions at least $\sim \frac{1}{10} \tau_B$. This is $\sim 100 \mu s$ in the present case, and some difficulties may arise due to this long time scale. If we assume a simple field programming where the rise time of the current and the reversal time of the toroidal field are extended to $\sim 100 \mu s$, helical instabilities might develop and may result in a self-reversal of the toroidal field before the intended reversal time. Thus, the study of the self-reversal mode of operation might be more important than otherwise expected.

It is noted that a full dynamic code such as the present one would be necessary to study the field programming, even though this would require quite long computing time. This is because the convection effects will be substantial during the field programming, and thus, usual diffusion codes would not be suitable.

4.4.C. Confinement Characteristics

In this final section, we discuss some confinement characteristics of pinch experiments in general, when the size of the machine becomes large and the temperature high. (We mean by this, at least ZT-40 parameter range or higher.) The simulation results previously shown will be used for comparison.

As discussed in the previous section, one of the basic differences of the ZT-40 experiment from other smaller experiments is that the field diffusion time τ_B becomes much larger than Alfvén time τ_A . The condition, $\tau_B \gg \tau_A$, is a general character associated with a large size and high temperature, and thus, will be more pronounced when the experiment approaches the reactor regime. Similarly, the other fact that the damping time τ_{Damp} is much smaller than τ_B will be also true in general. For the ohmic heating to have any significance, the heat loss time τ_H must be at least comparable to τ_B . This would be accomplished in a large experiment by reduction of the line radiation, as a result of high temperature, and also by an improved vacuum technique. In the following study, we additionally assume a rather obvious fact that the rise time t_{rise} is also much shorter than τ_B in a large experiment. In short, the following conditions will hold, i.e., $\tau_A, \tau_{\text{Damp}}, t_{\text{rise}} \ll \tau_B \lesssim \tau_H$.

Therefore, a quasi-equilibrium dissipation state will be reached in a much shorter time scale (τ_{QE}) than τ_B . We now examine the mechanisms of force balance and the plasma β in this

initial quasi-equilibrium state. The poloidal field generally confines, while the toroidal field can either confine or repel. The other repelling force is the plasma pressure due to adiabatic compression of the initial pressure. The pressure contribution due to ohmic dissipation, at this moment, will be negligible because $\tau_{QE} \ll \tau_B$. Since the wall stabilization is generally important, the compression ratio is usually limited to ~ 2 . This limits the adiabatically compressed pressure at about ten times the initial pressure, independent of the machine size. Therefore, the plasma β at the initial quasi-equilibrium state of a large experiment will be quite low, unless the preionization temperature is made very high. For example, in the ZT-40 discharge studied in the previous section with the initial temperature of 10ev, the adiabatically compressed pressure accounts for less than 10% of the magnetic pressure. Therefore, the field profiles at 30 μ s in Figure 31 show quite low β , compared to the ZT-S profiles in Figure 23 which shows $\sim 50\%$ β . This tendency will become more pronounced as the size of the machine becomes larger and the temperature higher, unless the preionization temperature can be made very high, which is an exceedingly difficult task. Then, we can conclude that at the initial quasi-equilibrium state of a large pinch experiment, the confining poloidal field should be nearly balanced by the repelling toroidal field. This low β initial quasi-equilibrium state will evolve to a later higher β quasi-equilibrium state as $t \rightarrow \tau_B$, due to ohmic dissipation. The thermal energy of

this later stage will be, at most, comparable to the poloidal field energy (i.e., the poloidal plasma beta $\beta_p \lesssim 1$), since the initial quasi-equilibrium condition (i.e., $P_{\text{poloidal field}} \approx -P_{\text{toroidal field}}$) limits the dissipatable energy of the toroidal field at about the same level as the poloidal field energy.

Thus, it is found that the condition $\beta_p \lesssim 1$ which is rather obvious for the reverse-field pinch is also true for other pinch experiments in general. This condition $\beta_p \lesssim 1$ is more stringent than the usual criterion from the equilibrium consideration, i.e., $\beta_p \lesssim$ aspect ratio. If $B_t \gg B_p$, the plasma beta is thus limited at $\beta \sim \frac{B_p^2}{B_t^2}$. Therefore, to achieve a high

β , the toroidal field must be comparable to the poloidal field, (i.e., the safety factor $q < 1$ for a reasonable aspect ratio).

One special feature of the reverse-field pinch concept is that this condition is possible due to the magnetic shear stabilization, and this character, thus, will be more important as the size of the experiment becomes large.

It is noted that the above discussion will not hold when an auxiliary heating, such as neutral injection or rf heating is used with a heating time scale much shorter than τ_B . In such a case, the plasma confinement due to the toroidal field could be made much larger than the poloidal field contribution (i.e., $\beta_p \gg 1$).

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APPENDIX : Computer Program Listing

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1          BLOCK      DATA
2  CC
3  CC----- INPUT PARAMETERS (CGS GAUSSIAN UNITS) -----
4  CC
5  CC....EXECUTION, PLOTTING AND PRINTING
6  CC  MS : STOP EXEC IF CYCLE#>MS
7  CC  MW : PLOT AND PRINT IF MOD(CYCLE#,MW)=0
8  CC  MST : NO PLOT OR PRINT IF CYCLE#<MST
9  CC  MAA,MBB : IF MAA<CYCLE#<MBB, PRINT INFORM. ON
10 CC      PROCESS(IZP TO PSI(ID))(CF.SECTION 3.2.B)
11 CC  MBB : IF MBB=ODD, REDUCED PRINTING ON VARIABLES
12 CC
13 CC....MACHINE PARAM. AND GRID AND STEP SIZE
14 CC  RB : MAJOR RADIUS
15 CC  RSMA : MINOR RADIUS
16 CC  ID : # OF GRIDS IN R
17 CC  JD : # OF GRIDS IN THETA
18 CC  DDDT : TIME STEP SIZE
19 CC  ADDT,CHTIME : IF TIME>CHTIME, TIME STEP=DDDT+ADDT
20 CC  ADADT,CHCHTM : IF TIME>CHCHTM, TIME STEP=DDDT+ADDT+ADADT
21 CC
22 CC....INITIAL CONDITIONS
23 CC  DN : INITIAL DENSITY
24 CC  TEMINI : INITIAL TEMP. IN KELVIN
25 CC
26 CC....B. C. ; TOROIDAL CURRENT IZP
27 CC***** (TMTM = TIME * 1.0E6 )*****
28 CC  CRTIMP, PETIMP, CURRR, PEP1, FS1, TMP1, CUR1, CU1BA :
29 CC  IZP (IN AMP.) = CURRR*SIN(PI*TMTM/PETIMP)
30 CC  IF TMTM>TMP1, IZP=CU1BA+CUR1*SIN(PI*(TMTM-FS1)/PEP1)
31 CC  IF TMTM>CRTIMP, IZP=CONSTANT
32 CC
33 CC....B. C. ; TOROIDAL MAGNETIC FIELD AT WALL, BZW
34 CC  BTOR, BMIN, CRTIMX, PETIMX, PEPEX, OTIM, FSTIMX, BIAS, TMX2, OTM2 :
35 CC  BZW = BIAS + BTOR*COS(PI*(TMTM+FSTIMX)/PETIMX)
36 CC  IF TMTM>OTIM, BZW=-BMIN*SIN(PI*(TMTM-OTIM)/PEPEX)
37 CC  IF TMTM>TMX2, BZW=-BMIN*EXP(-(TMTM-TMX2)*OTM2)
38 CC  IF TMTM>CRTIMX, BZW=CONSTANT
39 CC
40 CC....TRANSPORT COEFFICIENTS (CF. SECTION 2.3)
41 CC  ECONC : CLASSICAL ELECTRIC CONDUCTIVITY AT 1 EV
42 CC  CHOF : F*1.51E-8/SQRT(MASS OF ATOM) (CF. EQ.9, SECTION 2.3)
43 CC  CHOC : C*6.7E-18/SQRT(MASS OF ATOM) (CF. EQ.9, SECTION 2.3)
44 CC  CNCON : CONSTANT THERMAL CONDUCTIVITY (IN ERG.CM**2/KELVIN)
45 CC  BOWJ,POW : IF TMTM>BOWJ, ANOMAL.CONDUC.=POW*ANOMAL.CONDUC.
46 CC              (WITH POW>>1, ELECTRIC CONDUCT. BECOME PURE CLASSICAL)
47 CC  BOWJ,GAP : IF TMTM>BOWJ, THERMAL CONDUCT.=GAP
48 CC
49 CC....NUMERICAL DISSIPATION
50 CC  DISS : KAPPA IN EQ.(11+), SECTION 2.3
51 CC  PPOWJ : THETA IN EQ.(10'), SECTION 3.1
52 CC  OPERA : IF TMTM>OPERA, DISS AND PPOWJ BECOME 0.0
53 CC
54 CC....LOW DENSITY REGION
55 CC  HU : FLOOR DENSITY = HU*DN
56 CC
57 CC....FILLING GAS PARAM.
58 CC  FR : BOLTZMANN CONST./AVERAGE MASS PER PARTICLE(INCLUDING ELECTRONS)
59 CC  FK : GAS CONST. GAMMA
60 CC  DNUM : # OF ATOMS PER 1 GM

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61 CC
62 CC....ALGORITHM FOR DETERMINING PSI(ID) FROM TOROIDAL PLASMA CURRENT
63 CC (CF. SECTION 3.2.B)
64 CC FFLUXS : INITIAL START FOR PSI(ID) (FIRST 10 TIME STEPS)
65 CC PSI(ID) = FFLUXS*SIN(PI*TMTM/PETIMP)
66 CC BOW, PPOW, SDMM, PPOW, CURL, CURLL :
67 CC (AFTER 10 TIME STEPS)
68 CC (NOTATIONS : PSINC(INCREMENT OF PSI(ID) AT NEW TIME STEP),
69 CC PSIPRE(INCREMENT AT OLD TIME STEP),
70 CC TCUR(TORO CURRENT AT OLD STEP), TTCUR(CURRENT AT 2 TIME
71 CC STEPS BEFORE), TTTCUR(CURRENT AT 3 TIME STEPS BEFORE),
72 CC SURCUR((TCUR+TTCUR)/2-TARGET CURRENT), DT(NEW TIME STEP SIZE),
73 CC DTPRE(PREVIOUS TIME STEP SIZE),
74 CC PSIN10(PSINC AT 10TH TIME STEP))
75 CC SDM = ABS(SURCUR/CURRR)**0.5*PSIN10*SDMM
76 CC IF TMTM>1.5*CRTIMP, SDM=SDM*PPOW
77 CC IF SURCUR>CURL OR <CURLL, SDM=SDM*BBOW
78 CC PSINC = SURCUR/ABS(SURCUR)*ABS(PSIPRE)*
79 CC ABS(SURCUR/(TTTCUR-TCUR)/BOW)**PPOW*DT/DTPRE
80 CC PSIMAX=PSIPRE*DT/DTPRE+SDM*DT
81 CC PSIMIN=PSIPRE*DT/DTPRE-SDM*DT
82 CC PSINC = AMIN1(PSINC,PSIMAX)
83 CC PSINC = AMAX1(PSINC,PSIMIN)
84 CC
85 CC....NOT USED
86 CC DIVID, GAPTM
87 CC
88 CC
89 COMMON/YY/MS,MJ,MST, ID, JD, MAA, MBB, PPOW, SDMM, CNCON, CURRR, BOW
90 1, BMIN, BTOR, RB, RSMA, HU, DN, DDDT, FFLUXS, DIVID, OPERA, FR, FK
91 2, CURL, PPOW, BBOW, CURLL, PPOW, BOW, ECONC, DNUM, CHOF, CHOC
92 3, ADDT, ADADT, CHTIME, CHCHTM, GAP, TEMINI, GAPTM
93 4, CRTIMP, CRTIMX, PETIMP, PETIMX, PEPEX, OTIM, FSTIMX
94 5, PEP1, FS1, TMP1, CUR1, CU1BA, TMX2, OTM2, DISS, BIAS
95 DATA MS, MJ, MST/0200, 100, 001/
96 1, ID, JD, DDDT/16, 9, 010, 0E-9/
97 2, GAP, ADDT, ADADT, CHTIME, CHCHTM/2.0E5, -5.E-9, 20.E-9, 6.E-6, 50.E-6/
98 2, BMIN, BTOR, RB, RSMA/0620.0, 0050.0, 40.0, 07.7/
99 3, MAA, MBB/0001, 1500/
100 4, CNCON, SDMM, CURRR, PPOW, BOW/2.0E6, 0.6000, 6.0E4, 0.30, 2.0/
101 5, CURL, PPOW, BBOW, POW/10.0E2, 0.5, 2.000, 100000.0/
102 5, CURLL, PPOW, BOW, ECONC/-10.0E2, 0.1, 6.0, 1.30E13/
103 5, DIVID, DN, FFLUXS, FR, FK/1.00000, 4.3E-9, 11.00E4, 08.310E7, 1.666666/
104 5, DNUM, CHOF, CHOC/3.00E23, 4.15E4, 1.84E-6/
105 6, HU, OPERA/0.10, 60.00/
106 7, CRTIMP, CRTIMX/9.0, 9.0/
107 7, PETIMP, PETIMX, PEPEX/18.0, 6.0, 10.0/
108 7, OTIM, FSTIMX, BIAS, TEMINI, GAPTM/100.0, 9.0, 400.0, 1.2E4, 100.0/
109 7, PEP1, FS1, TMP1, CUR1/6.00, 0.6, 100.0, 6.0E3/
110 7, CU1BA, TMX2, OTM2, DISS/5.1E4, 100.0, 0.073, 0.125/
111 END
112 C
113 C
114 C....MAIN PROGRAM .....
115 C
116 DIMENSION PSI(20, 16, 2), B1(20, 16, 2), B2(20, 16, 2), XI(20, 16, 2),
117 IC1(20, 16, 2), C2(20, 16, 2), C3(20, 16, 2), T(20, 16, 2),
118 2W(4, 20, 16, 3), F(4, 20, 16, 3), E1(20, 16, 2), B3(20, 16, 2), SR(20),
119 4GA(20, 16), GB(20, 16), GC(20, 16), GD(20, 16), CN(20, 16, 2),
120 SAA(20, 12), BB(4, 4), ANG(16), BATB(20), ALAM(20, 16)

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121 6,VELD(20,16),AJOR(20,16),AJPE(20,16),VINT(20,16)
122 COMMON PSI,B1,B2,B3,XI,C1,C2,C3,T,W,F,SR,ANG,BATB
123 1,GA,GB,GC,GD,CN,EI,AA,BB,AJCON
124 COMMON/YY/MS,MW,MST,ID,JD,MAA,MBB,PPOW,SDMM,CNCON,CURRR,BOW
125 1,BMIN,BTOR,RB,RSMA,HU,DN,DDDT,FFLUXS,DIVID,OPERA,FR,FK
126 2,CURL,PPOW,BBOW,CURLL,PPOWJ,BOWJ,POW,ECONC,DNUM,CHOF,CHOC
127 3,ADDT,ADADT,CHTIME,CHCHTM,GAP,TEMINI,GAPTM
128 4,CRTIMP,CRTIMX,PETIMP,PETIMX,PEPEX,OTIM,FSTIMX
129 5,PEP1,FS1,TMP1,CUR1,CU1BA,TMX2,OTM2,DISS,BIAS
130 COMMON/WX/X(40),FVL(40,6)
131 DIMENSION SJA(16),THJ(16),SEQ1(16),SEQ2(16)
132 1 ,SEQ5(16),SEQ6(16),SEQ7(16),SEQ8(16)
133 2 ,SEQ10(16),BMS(16),URR(16),UTH(16),UFI(16),WCO(16),WSI(16)
134 3 ,TVA1(100),TVA2(100),TVA3(100),XNL(100),TVA4(100),TVA5(100)
135 4 ,TVA6(100)
136 COMMON/XY/NUG,DX,JE,JB,PI,IE,THJ
137
138 CC
139 CC...VARIABLES .....
140 CC W(1,I,J,K) : DENSITY/'DN'
141 CC W(N,I,J,K) : RADIAL(N=2), THETA(N=3) AND TOROIDAL(N=4)
142 CC VELOCITIES/'UN' (UN=BTOR/SQRT(4*PI*DN))
143 CC EI : TEMPERATURE /'UN**2/CV' (CV:SPECIFIC HEAT)
144 CC PSI : POLOIDAL FLUX FUNCTION /'BTOR*RSMA**2'
145 CC XI : R*83 /'RSMA*BTOR'
146 CC B1,B2,B3 : RADIAL, THETA AND TOROIDAL MAG. FIELD /'BTOR'
147 CC C1,C2,C3 : RADIAL, THETA AND TOROIDAL CURRENT DENSITIES /
148 CC '4*PI*RSMA/C/BTOR'
149 CC T : ELECTRICAL CONDUCTIVITY / INITIAL VALUE
150 CC CN : THERMAL CONDUCTIVITY /'DN*RSMA*UN*CV'
151
152 OPTIMIZE
153 WZIN=0.
154 WTFIN=0.
155 WRRR=0.
156 FBKP CONTINUE
157 CALL KEEP80(1,2)
158 CALL DD80ID(9HPLOT TES ,1)
159 IS=1
160 JS=1
161 IB=IS+1
162 JB=JS+1
163 IE=ID-1
164 JE=JD-1
165 JP=JE-JS
166 IP=IE-IS
167 RRSM=RSMA
168 RSMA=RSMA*2.0*IE/(2.0*IE-1.0)
169 TCUR=1.0
170 TTCUR=2.0
171 TTTTUR=3.0
172 PI=3.1415926536
173 UN=BTOR/SQRT(4.*PI*DN)
174 DDDDT=DDDT *UN/RSMA
175 ANORTM= UN/RSMA
176 ADDT=ADDT *UN/RSMA
177 ADADT=ADADT *UN/RSMA
178 CHTIME=CHTIME *UN/RSMA
179 CHCHTM=CHCHTM *UN/RSMA
180 FLUXS=FFLUXS /BTOR/RSMA/RSMA
FLNOR= BTOR*RSMA*RSMA

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181 PSINC=0.0
182 PSIN10=0.0
183 SMALL=1.0E-16
184 TMALL=SMALL**2
185 CV=FR/(FK-1.)
186 TNOR=UN*UN/CV
187 C=2.9980E+10
188 AJNOR=C*BTOR/(4.0*PI*RSMA)
189 K=1
190 Q=RSMA/RB
191 DCA=PI /((JD-3.0)
192 HJCON=C*1.67E-24*UN/BTOR/RSMA/4.0E-10
193 DDT=0.1E-10 *UN/RSMA
194 M=0
195 DX=1./((ID-1.0)
196 DO 333 I=IS, ID
197 SI=I
198 SR(I)=(SI-1.)*DX
199 333 CONTINUE
200 SGO=ECONC *(TEMINI/11600.)*1.5
201 101 FORMAT(1H , "I=", I4, "RADIUS=", E14.7)
202 DO 4 J=JS, JD
203 SJ=J
204 ANG(J)=(SJ-2.)*DCA
205 WRITE (6,102) J,ANG(J)
206 4 CONTINUE
207 102 FORMAT (1H , "J=", I4, "ANGLE=", E14.7)
208 COK=DN*RSMA*UN*CV
209 CC
210 CC.... INITIAL CONDITIONS
211 DO 5 K=1,2
212 DO 5 I=IS, ID
213 DO 5 J=JS, JD
214 PSI(I,J,K)=0.0
215 XI(I,J,K)=COS(PI*FSTIMX/PETIMX)/Q+BIAS/BTOR/Q
216 EI(I,J,K)=TEMINI*CV/UN/UN
217 CN(I,J,K)=CNCON /COK
218 T(I,J,K)=1.
219 B1(I,J,K)=0.
220 B2(I,J,K)=0.
221 B3(I,J,K)=XI(I,J,K)/(1./Q+SR(I)*COS(ANG(J)))
222 C1(I,J,K)=0.
223 C2(I,J,K)=0.
224 C3(I,J,K)=0.
225 W(1,I,J,K+1)=1.0
226 W(2,I,J,K+1)=0.
227 W(3,I,J,K+1)=0.
228 W(4,I,J,K+1)=0.
229 W(1,I,J,K)=1.0
230 W(2,I,J,K)=0.
231 W(3,I,J,K)=0.
232 W(4,I,J,K)=0.
233 5 CONTINUE
234 CC
235 CC
236 NUG=2*(JE-JB)
237 DO 901 J=JS, JD
238 THJ(J)=(J-JS-1)*2.0*PI/NUG
239 SJA(J)=1.0+ Q*DX *COS(THJ(J))/2.0
240 901 CONTINUE

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241      DO 621 I=IS, ID
242      DO 621 J=JS, JD
243      VINT(I, J)=(B1(I, J, 1)**2+B2(I, J, 1)**2+B3(I, J, 1)**2)/8./PI
244      621 CONTINUE
245      CALL VOLIN(VINT, WBOR, RB, RSMA)
246      WTIW=WBOR*BTOR*BTOR
247      DO 630 I=IS, ID
248      DO 630 J=JS, JD
249      VINT(I, J)=1.0
250      630 CONTINUE
251      CALL VOLIN(VINT, VOLU, RB, RSMA)
252      DO 625 J=JS, JD
253      UTH(J)=1.0
254      625 CONTINUE
255      CALL SURIN(UTH, SURA, RB, RSMA)
256      WRITE(6, 631) VOLU, SURA, AJCON
257      631 FORMAT(1P7E14.6)
258      TIME=0.
259      RMTM=0.0
260      6 M=M+1
261      IF(RMTM.GT.OPERA)DISS=0.0
262      IF(RMTM.GT.OPERA)PPOWW=0.0
263      IF(RMTM.LE.BOWW)GO TO 38
264      DO 37 I=1, ID
265      DO 37 J=1, JD
266      CN(I, J, 1)=GAP/COK
267      CN(I, J, 2)=GAP/COK
268      37 CONTINUE
269      38 CONTINUE
270      TIMPRE=TIME
271      K=1
272      CC
273      CC....B. C. ; TOROIDAL CURRENT (CF. SECTION 3.2.B)
274      IF(M.EQ. 10)PSINC=FLUXS*SIN(PI*RMTM/PETIMP)
275      I -FLUXS*SIN(PI*(RMTM-DDDDT*1.0E6/ANORTM)/PETIMP)
276      PSIPR=PSINC
277      DT=DDDDT
278      DTPRE=DT
279      IF (TIME .GT. CHTIME) DT=DDDDT+ADDT
280      IF (TIME .GT. CHCHTM) DT=DDDDT+ADDT+ADADT
281      DO 3 I=IS, ID
282      SI=I
283      SR(I)=(SI-1.)*DX
284      IF (I.EQ. IS) GO TO 3
285      BATB(I)=DT/(2.*SR(I)*DCA)
286      3 CONTINUE
287      BATA=DT/(2.*DX)
288      COE= C*C*DT/(4.*PI*RSMA*UN*SGO)
289      TIME=TIME+DT
290      RMTM=TIME/ANORTM *1.0E6
291      TIMINC=(TIME-TIMPRE)/ANORTM
292      IF (RMTM .GT. CRTIMP) GO TO 990
293      SURCUR=(CURRR *SIN(PI*RMTM/PETIMP)+(TCUR+TTCUR)*0.5)
294      IF (RMTM.GT. TMP1) SURCUR=CU1BA+(CUR1*SIN(PI*(RMTM-FS1)/PEP1)+(TCU
295      IR+TTCUR)*0.5)
296      PSINC=SIGN(1.0, SURCUR)*ABS(PSINC)*ABS(SURCUR/(TTCUR-TCUR)/BOW)
297      I **PPOW*DT/DTPRE
298      TCURMA=TCUR
299      GO TO 809
300      990 CONTINUE

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301      SURCUR=-TCURMA      +(TCUR+TTCUR)*0.5
302      PSINC=SIGN(1.0,SURCUR)*ABS(PSINC)*ABS(SURCUR/(TTCUR-TCUR)/BOW)
303      1 **PPDW*DT/DTPRE
304      809 CONTINUE
305      IF(M.EQ. 10)PSIN10=PSIPR/DDDT
306      SDM=ABS(SURCUR/CURRR)*0.5*SDMM*PSIN10
307      IF(RMTM.GT.1.5*CRTIMP)SDM=ABS(SURCUR/CURRR)*0.5*PPDW*SDMM*PSIN10
308      IF(SURCUR.GT.CURL .OR. SURCUR.LT.CURLL)SDM=SDM*BBOW
309      PSIMA=PSIPR*DT/DTPRE+SDM*DT/ANORTM
310      PSIMI=PSIPR*DT/DTPRE-SDM*DT/ANORTM
311      BZAX=B3(1,2,1)*BTOR
312      IF(M.EQ. MAA+1)WRITE(G,550)
313      550 FORMAT(1H0,4X,"CYCLE",2X,"PSI(ID)",7X,"EST PSINC",1X,"ACT PSINC"
314      1 ,1X,"TOR CUR",2X,"CUR ERROR",1X,"PSIMAX",4X,"PSIMIN",4X,
315      2 "DENS AXIS",1X,"BZ AXIS",3X,"TIME")
316      IF(M.GT.MAA.AND.M.LT.MBB)WRITE(6,807)M,PSI(ID,5,K),PSINC,PSIPR
317      1,TCUR,SURCUR,PSIMA,PSIMI,W(1,1,2,1),BZAX,RMTM
318      PSINC=AMIN1(PSINC,PSIMA)
319      PSINC=AMAX1(PSINC,PSIMI)
320      807 FORMAT("PSINC",15,1PE14.6,10E10.2)
321      DO 9 J=JS,JD
322      PSI(ID,J,K+1)=PSI(ID,J,K)+PSINC
323      IF(M.LT.10)PSI(ID,J,K+1)=FLUXS*SIN(PI*RMTM/PETIMP)
324      9 CONTINUE
325      991 CONTINUE
326      CC
327      CC....B. C. ; TOROIDAL FIELD AT WALL
328      IF(RMTM.GT.CRTIMX)GO TO 900
329      DO 999 J=JS,JD
330      XI(ID,J,K+1)=
331      1 COS(PI*(RMTM +FSTIMX)/PETIMX)/Q +BIAS/BTOR/Q
332      IF (RMTM.GT. OTIM ) XI(ID,J,K+1)=- (BMIN/BTOR)*
333      1 SIN(PI*(RMTM-OTIM)/PEPEX)/Q
334      IF (RMTM.GT. TMX2) XI (ID, J, K+1)=- (BMIN/BTOR)*EXP(-(RMTM-TMX2)*OTM2)/Q
335      998 CONTINUE
336      999 CONTINUE
337      CC
338      CC....COORDINATE ORIGIN TREATMENT (CF. SECTION 3.2.B)
339      900 SUM=0.
340      SUN=0.
341      TUM=0.
342      VELO=0.0
343      CL=0.
344      DO 23 J=JB,JE
345      DL=2.
346      IF (J.EQ. JB .OR. J.EQ. JE) DL=1.
347      CL=CL+DL
348      SUM=SUM+(C*C/(4.*PI*RSMA*UN*SGO)*C3(IB,J,K)/T(IB,J,K)+
349      1(W(2,IB,J,K)/W(1,IB,J,K)*B2(IB,J,K)-W(3,IB,J,K)/W(1,IB,J,K)*
350      2B1(IB,J,K)))*DL
351      SUN=SUN+(C*C/(4.*PI*RSMA*UN*SGO)*C2(IB,J,K)/T(IB,J,K)+
352      1(W(4,IB,J,K)/W(1,IB,J,K)*B1(IB,J,K)-W(2,IB,J,K)/W(1,IB,J,K)*
353      2B3(IB,J,K)))*DL
354      TUM=TUM+EI(IB,J,K)*DL
355      VELO=VELO+(W(2,IB,J,K)*COS(ANG(J))-W(3,IB,J,K)*SIN(ANG(J)))*
356      1 SJA(J)*DL
357      23 CONTINUE
358      SUM=SUM/CL
359      TUM=TUM/CL
360      VELO=VELO/CL

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361      DO 30 I=IS, ID
362      W(1, I, JE+1, K) = W(1, I, JE-1, K)
363      W(2, I, JE+1, K) = W(2, I, JE-1, K)
364      W(3, I, JE+1, K) = -W(3, I, JE-1, K)
365      W(4, I, JE+1, K) = -W(4, I, JE-1, K)
366      W(1, I, JB-1, K) = W(1, I, JB+1, K)
367      W(2, I, JB-1, K) = W(2, I, JB+1, K)
368      W(3, I, JB-1, K) = -W(3, I, JB+1, K)
369      W(4, I, JB-1, K) = -W(4, I, JB+1, K)
370 30 CONTINUE
371      DO 902 J=JS, JD
372      WCO(J)=COS(THJ(J))
373      WSI(J)=SIN(THJ(J))
374      URR(J)=W(2, IB, J, K)/W(1, IB, J, K)
375      ALF=1.0/Q
376      SEQ1(J)=- (W(2, IS, J, K)+W(2, IB, J, K))*SJA(J)
377      SEQ2(J)=- (W(2, 1, J, K)**2/W(1, 1, J, K)+W(2, 2, J, K)**2/W(1, 2, J, K))*
378 1 WCO(J)- (W(2, 1, J, K)*W(3, 1, J, K)/W(1, 1, J, K)+W(2, 2, J, K)*W(3, 2, J, K)/
379 1 W(1, 2, J, K))*WSI(J))*SJA(J)
380 2 - (FK-1.0)*W(1, IB, J, K)*EI(IB, J, K)*WCO(J)
381      SEQ6(J)= XI(IB, J, K)*WCO(J) /ALF
382      SEQ7(J)=-B2(IB, J, K)
383      SEQ8(J)=-EI(IB, J, K)*W(2, IS, J, K)/W(1, IS, J, K)
384 1 - (FK-1.0)*EI(IS, J, K)*URR(J)*SJA(J)
385 2 + (CN(IS, J, K)+CN(IB, J, K))*(EI(IB, J, K)-EI(IS, J, K))*(SJA(J)+1)
386 2 / (W(1, IS, J, K)*DX*2.0)
387      SEQ10(J)=PSI(IB, J, K) *WCO(J)/ALF
388 902 CONTINUE
389      CALL AVERA(SEQ1, AVV1)
390      CALL AVERA(SEQ2, AVV2)
391      CALL AVERA(SEQ6, AVV6)
392      CALL AVERA(SEQ7, AVV7)
393      CALL AVERA(SEQ8, AVV8)
394      CALL AVERA(SEQ10, AVV10)
395      DO 905 J=JS, JD
396      C1(IS, J, K) = AVV6*WSI(J)
397      C2(IS, J, K) = AVV6*WCO(J)
398      C3(IS, J, K) = AVV7
399      B1(IS, J, K) = AVV10*WSI(J)
400      B2(IS, J, K) = AVV10*WCO(J)
401      B3(IS, J, K) = XI(IS, J, K)*Q
402 905 CONTINUE
403      AVV2=AVV2+C3(1, 2, K)*B2(1, 2, K)-C2(1, 2, K)*B3(1, 2, K)
404      W(2, IS, 2, K+1)=W(2, 1, 2, K)+DT*AVV2
405      DO 24 J=JS, JD
406      XI(IS, J, K+1)=XI(IS, J, K)+DT*DCA*SUN/(PI*DX)/Q
407      W(1, IS, J, K+1)=W(1, 1, J, K)+DT*AVV1
408      W(2, IS, J, K+1)=W(2, IS, 2, K+1)*WCO(J)
409      W(3, IS, J, K+1)=-W(2, IS, 2, K+1)*WSI(J)
410      EI(IS, J, K+1)=COE*(C1(IS, J, K)**2+C2(IS, J, K)**2+C3(IS, J, K)**2)
411 1 / (DT*T(IS, J, K)*W(1, IS, J, K))
412      EI(IS, J, K+1)=AVV8+EI(IS, J, K+1)
413      EI(IS, J, K+1)=TUM
414      PSI(IS, J, K+1)=W(2, IS, J, K)*B2(IS, J, K)/W(1, IS, J, K)
415 1-W(3, IS, J, K)*B1(IS, J, K)/W(1, IS, J, K)
416      PSI(IS, J, K+1)=-COE*C3(IS, J, K)/(DT*T(IS, J, K))*ALF-ALF*PSI(IS, J, K+1)
417      PSI(IS, J, K+1)=PSI(IS, J, K)+DT*PSI(IS, J, K+1)
418 24 CONTINUE
419 CC
420 CC

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421      DO 28 J=JB,JE
422      DO 29 L=1,4
423      W(L, ID, J, K )= W(L, IE, J, K )
424 29  CONTINUE
425      B1(ID, J, K)=0.0
426      C1(ID, J, K)=-C1(IE, J, K)
427      C2(ID, J, K)= C2(IE, J, K)
428      C3(ID, J, K)= C3(IE, J, K)
429 28  CONTINUE
430      K=1
431      K2=1
432      K3=1
433      K4=1
434  CC
435  CC.... FIND (S+F)/2 (S AND F IN EQ.8, SECTION 2.1)
436      DO 433 J=JB,JE
437      DO 434 I=IB,IE
438      F(1, I, J, K2)=(-(W(1, I, J, K2)/W(1, I, J, K2)*W(2, I, J, K2)*
439      1 COS(ANG(J))-W(1, I, J, K2)/W(1, I, J, K2)*W(3, I, J, K2)*SIN(
440      1 ANG(J)))/(1./Q+SR(I)*COS(ANG(J))) *DT/2. -(W(1, I, J, K2)/
441      1 W(1, I, J, K2)*W(2, I, J, K2)*0.5+0.25*(W(2, I+1, J, K2)+W(2, I-1, J, K2)))
442      1 /SR(I)*DT/2.0
443      F(2, I, J, K2)=(-(W(2, I, J, K2)**2/W(1, I, J, K2)*0.5+0.25*(W(2, I+1, J, K2)
444      1 **2/W(1, I+1, J, K2)+W(2, I-1, J, K2)**2/W(1, I-1, J, K2)))/SR(I)
445      1+W(3, I, J, K2)**2/(W(1, I, J, K2)*SR(I))-W(2, I, J, K2)*
446      2(W(2, I, J, K2)*COS(ANG(J))-W(3, I, J, K2)*SIN(ANG(J)))/(W(1, I, J, K2)
447      3*(1./Q+SR(I)*COS(ANG(J)))+(W(4, I, J, K2)**2)*COS(ANG(J))
448      4/(W(1, I, J, K2)*(1./Q+SR(I)*COS(ANG(J)))+(C3(I, J, K2)*
449      5B2(I, J, K2)-C2(I, J, K2)*B3(I, J, K2)))*DT/2.
450      F(3, I, J, K2)=(
451      1-(2.0*W(2, I, J, K2)*W(3, I, J, K2)/W(1, I, J, K2)*0.75+0.25*(W(2, I+1, J, K2)
452      1 *W(3, I+1, J, K2)/W(1, I+1, J, K2)+W(2, I-1, J, K2)*W(3, I-1, J, K2)
453      1 /W(1, I-1, J, K2)))/
454      1SR(I)-(W(4, I, J, K2)**2)*SIN(ANG(J))/(W(1, I, J, K2)*
455      2(1./Q+SR(I)*COS(ANG(J)))-W(3, I, J, K2)*(W(2, I, J, K2)*
456      3COS(ANG(J))-W(3, I, J, K2)*SIN(ANG(J)))/(W(1, I, J, K2)*(1./Q+
457      4SR(I)*COS(ANG(J))))+(C1(I, J, K2)*B3(I, J, K2)-C3(I, J, K2)*
458      5B1(I, J, K2))*DT/2.
459      F(4, I, J, K2)=(-2.*W(4, I, J, K2)*(W(2, I, J, K2)*COS(ANG(J))
460      1-W(3, I, J, K2)*SIN(ANG(J)))/(W(1, I, J, K2)*(1./Q+SR(I)*
461      2COS(ANG(J)))+(C2(I, J, K2)*B1(I, J, K2)-C1(I, J, K2)*B2(I, J, K2))
462      3)*DT/2. -(W(2, I, J, K2)*W(4, I, J, K2)/W(1, I, J, K2)
463      1 *0.5+0.25*(W(2, I+1, J, K2)*W(4, I+1, J, K2)/W(1, I+1, J, K2)
464      1 +W(2, I-1, J, K2)*W(4, I-1, J, K2)/W(1, I-1, J, K2)))/SR(I)*DT/2.0
465  CC
466  CC.... INTERMEDIATE VALUES W(M, I, J, 2) (EQ. (10'), SECTION 2.1)
467      W(1, I, J, K+1)=0.125*(W(1, I+1, J, K)+W(1, I-1, J, K)+W(1, I, J+1, K)+
468      1W(1, I, J-1, K))*2.0*PPOW+(1.0-PPOW)*W(1, I, J, K)
469      W(2, I, J, K+1)=0.125*(W(2, I+1, J, K)+W(2, I-1, J, K)+W(2, I, J+1, K)+
470      1W(2, I, J-1, K))*2.0*PPOW+(1.0-PPOW)*W(2, I, J, K)
471      W(3, I, J, K+1)=0.125*(W(3, I+1, J, K)+W(3, I-1, J, K)+W(3, I, J+1, K)+
472      1W(3, I, J-1, K))*2.0*PPOW+(1.0-PPOW)*W(3, I, J, K)
473      W(4, I, J, K+1)=0.125*(W(4, I+1, J, K)+W(4, I-1, J, K)+W(4, I, J+1, K)+
474      1W(4, I, J-1, K))*2.0*PPOW+(1.0-PPOW)*W(4, I, J, K)
475      DO 435 L=1,4
476      IF (ABS(W(L, I, J, K+1)) .LT. SMALL) W(L, I, J, K+1)=0.
477      435 CONTINUE
478      434 CONTINUE
479  CC
480  CC.... B. C. ; W

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481      433 CONTINUE
482      DO 280 J=JB,JE
483      DO 281 L=1,4
484      W(L, ID, J, K+1)=W(L, IE, J, K+1)
485      281 CONTINUE
486      W(2, ID, J, K+1)=-W(2, IE, J, K+1)
487      W(2, ID, J, K )=-W(2, IE, J, K )
488      280 CONTINUE
489      CC
490      CC....B. C. IN THETA DIRECTION ; W
491      DO 35 I=IS, ID
492      W(1, I, JE+1, K+1)= W(1, I, JE-1, K+1)
493      W(2, I, JE+1, K+1)= W(2, I, JE-1, K+1)
494      W(3, I, JE+1, K+1)=-W(3, I, JE-1, K+1)
495      W(4, I, JE+1, K+1)=-W(4, I, JE-1, K+1)
496      W(1, I, JB-1, K+1)= W(1, I, JB+1, K+1)
497      W(2, I, JB-1, K+1)= W(2, I, JB+1, K+1)
498      W(3, I, JB-1, K+1)=-W(3, I, JB+1, K+1)
499      W(4, I, JB-1, K+1)=-W(4, I, JB+1, K+1)
500      W(1, I, JE+1, K)= W(1, I, JE-1, K)
501      W(2, I, JE+1, K)= W(2, I, JE-1, K)
502      W(3, I, JE+1, K)=-W(3, I, JE-1, K)
503      W(4, I, JE+1, K)=-W(4, I, JE-1, K)
504      W(1, I, JB-1, K)= W(1, I, JB+1, K)
505      W(2, I, JB-1, K)= W(2, I, JB+1, K)
506      W(3, I, JB-1, K)=-W(3, I, JB+1, K)
507      W(4, I, JB-1, K)=-W(4, I, JB+1, K)
508      35 CONTINUE
509      CC
510      CC      CALL ADI ; FIND NEXT TIME STEP VALUES PSI, XI AND EI(I,J,2)
511      CALL      ADI(K, M, DT, DX, DCA, CV, FK, COE, RB, UN, Q, SGO,
512      1IS, IB, IE, ID, JS, JB, JE, JD, C, IP, JP)
513      118 FORMAT (1H0, 3I12)
514      DO 906 J=JS, JD
515      W(2, ID, J, 1)=-W(2, IE, J, 1)
516      W(2, ID, J, 2)=-W(2, IE, J, 2)
517      B3(ID, J, K+1)=(      XI(ID, J, K+1))/(1./Q+SR(ID)*COS(ANG(J)))
518      B1(IS, J, K+1)=B1(IS, J, K)
519      B2(IS, J, K+1)=B2(IS, J, K)
520      B3(IS, J, K+1)=B3(IS, J, K)
521      C1(IS, J, K+1)=C1(IS, J, K)
522      C2(IS, J, K+1)=C2(IS, J, K)
523      C3(IS, J, K+1)=C3(IS, J, K)
524      906 CONTINUE
525      CC
526      CC....EVALUATE ELECTRIC CONDUCTIVITY, T (CF. SECTION 2.3)
527      DO 86 J=JB,JE
528      DO 86 I=IS,IE
529      IF (EI( I, J, K+1) .LT. 0.) WRITE (6, 118) I, J ,M
530      AJPE(I, J)=(C1(I, J, K+1)*B1(I, J, K+1)+C2(I, J, K+1)*B2(I, J, K+1)+C3(I, J,
531      1K+1)*B3(I, J, K+1))/SQRT(B1(I, J, K+1)**2+B2(I, J, K+1)**2+B3(I, J, K+1)**
532      22)
533      VELD(I, J)=SQRT(C1(I, J, K+1)**2+C2(I, J, K+1)**2+C3(I, J, K+1)**2)
534      AJOR(I, J)= ABS(VELD(I, J)**2-AJPE(I, J)**2)**0.5
535      VELD(I, J)=0.5*(ABS(AJOR(I, J))+ABS(AJPE(I, J)))
536      VELD(I, J)=VELD(I, J)*AJNOR
537      1/(W(1, I, J, K+1)*DN*DNUM*4.8E-10)
538      VELD(I, J)=ABS(VELD(I, J))/CHOF/SQRT(EI(I, J, K+1)*TNOR)
539      VELD(I, J)=CHOC*(1.0-EXP(-VELD(I, J)))/SQRT((W(1, I, J, K+1)-0.00*HU)*
540      1 DN*DNUM)

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541      IF (VELD(I,J).LT.1.0E-55)VELD(I,J)=1.0E-55
542      VELD(I,J)=1.0/VELD(I,J)/SGO
543      AJPE(I,J)=CHOC /SQRT((W(1,I,J,K+1)-0.00*HU)*
544      1 DN*DNUM)
545      IF (AJPE(I,J).LT.1.0E-55)AJPE(I,J)=1.0E-55
546      AJPE(I,J)=1.0/AJPE(I,J)/SGO
547      T(I,J,K+1)=(ECONC*(UN*UN*EI(I,J,K+1)/CV/11600.))*1.5/SGO
548      IF (RMTM.GT.80*W)VELD(I,J)=POW*VELD(I,J)
549      T(I,J,K+1)=AMIN1(T(I,J,K+1),VELD(I,J))
550
551 CC      86 CONTINUE
552 CC
553      DO 78 J=JS,JD
554      C3(ID,J,2)=C3(IE,J,2)
555      T(ID,J,K+1)=T(IE,J,2)
556      78 CONTINUE
557      DO 99 I=IS, ID
558      T(I,JE+1,K+1)=T(I,JE-1,K+1)
559      T(I,JB-1,K+1)=T(I,JB+1,K+1)
560      C3(I,JE+1,2)=C3(I,JE-1,2)
561      C3(I,1,2)=C3(I,3,2)
562      99 CONTINUE
563      K2=2
564      DO 66 I=2, IE
565      DO 66 J=2, JE
566      VINT(I,J)=(1.0-PPOWW)*T(I,J,2)+0.125*(T(I+1,J,2)+T(I-1,J,2)+
567      1 T(I,J+1,2)+T(I,J-1,2))*2.0*PPOWW
568      T(I,J)=(1.0-PPOWW)*C3(I,J,2)+PPOWW*0.25*(C3(I+1,J,2)+
569      1 C3(I-1,J,2)+C3(I,J+1,2)+C3(I,J-1,2))
570      66 CONTINUE
571      DO 67 I=2, IE
572      DO 67 J=2, JE
573      T(I,J,2)=VINT(I,J)
574      C3(I,J,2)=ALAM(I,J)
575      67 CONTINUE
576      DO 68 I=IS, ID
577      T(I,JE+1,K+1)=T(I,JE-1,K+1)
578      T(I,JB-1,K+1)=T(I,JB+1,K+1)
579      C3(I,JE+1,2)=C3(I,JE-1,2)
580      C3(I,1,2)=C3(I,3,2)
581      68 CONTINUE
582      DO 69 J=1, JD
583      C3(ID,J,2)=0.0
584      T(ID,J,2)=0.0
585      69 CONTINUE
586 CC
587 CC....EVALUATE (S+F)/2 USING W(M,I,J,2) (S AND F IN EQ.8,SECTION2.1)
588      DO 33 J=JB,JE
589      DO 34 I=IB,IE
590      F(1,I,J,K2)=(-(W(1,I,J,K2)/W(1,I,J,K2))*W(2,I,J,K2)*
591      1 COS(ANG(J))-W(1,I,J,K2)/W(1,I,J,K2)*W(3,I,J,K2)*SIN(
592      1 ANG(J)))/(1./Q+SR(I)*COS(ANG(J)))*DT/2.-(W(1,I,J,K2)/
593      1 W(1,I,J,K2))*W(2,I,J,K2)*0.5+0.25*(W(2,I+1,J,K2)+W(2,I-1,J,K2))
594      1 /SR(I)*DT/2.0
595      F(2,I,J,K2)=(-(W(2,I,J,K2)**2/W(1,I,J,K2)*0.5+0.25*(W(2,I+1,J,K2)
596      1 **2/W(1,I+1,J,K2)+W(2,I-1,J,K2)**2/W(1,I-1,J,K2)))/SR(I)
597      1+W(3,I,J,K2)**2/(W(1,I,J,K2)*SR(I))-W(2,I,J,K2)*
598      2(W(2,I,J,K2)*COS(ANG(J))-W(3,I,J,K2)*SIN(ANG(J)))/(W(1,I,J,K2)
599      3*(1./Q+SR(I)*COS(ANG(J)))+(W(4,I,J,K2)**2)*COS(ANG(J))
600      4/(W(1,I,J,K2)*(1./Q+SR(I)*COS(ANG(J)))+(C3(I,J,K2)*

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601      5B2(I,J,K2)-C2(I,J,K2)*B3(I,J,K2))*DT/2.
602      F(3,I,J,K2)=(
603      1-(2.0*W(2,I,J,K2)*W(3,I,J,K2)/W(1,I,J,K2)*0.75+0.25*(W(2,I+1,J,K2)
604      1 *W(3,I+1,J,K2)/W(1,I+1,J,K2)+W(2,I-1,J,K2)*W(3,I-1,J,K2)
605      1 /W(1,I-1,J,K2)))/
606      1SR(I)-(W(4,I,J,K2)**2)*SIN(ANG(J))/W(1,I,J,K2)*
607      2(1./Q+SR(I)*COS(ANG(J)))=W(3,I,J,K2)*(W(2,I,J,K2)*
608      3COS(ANG(J))-W(3,I,J,K2)*SIN(ANG(J)))/W(1,I,J,K2)*(1./Q+
609      4SR(I)*COS(ANG(J)))+(C1(I,J,K2)*B3(I,J,K2)-C3(I,J,K2)*
610      5B1(I,J,K2))*DT/2.
611      F(4,I,J,K2)=(-2.*W(4,I,J,K2)*(W(2,I,J,K2)*COS(ANG(J))
612      1-W(3,I,J,K2)*SIN(ANG(J)))/W(1,I,J,K2)*(1./Q+SR(I)*
613      2COS(ANG(J)))+(C2(I,J,K2)*B1(I,J,K2)-C1(I,J,K2)*B2(I,J,K2))
614      3)*DT/2.-(W(2,I,J,K2)*W(4,I,J,K2)/W(1,I,J,K2)
615      1 *0.5+0.25*(W(2,I+1,J,K2)*W(4,I+1,J,K2)/W(1,I+1,J,K2)
616      1 +W(2,I-1,J,K2)*W(4,I-1,J,K2)/W(1,I-1,J,K2))/SR(I)*DT/2.0
617      34 CONTINUE
618      33 CONTINUE
619      CC
620      CC.... FIND NEXT TIME STEP VALUES W(M,I,J,3) ;
621      CC.... YIMCRO(THETA IMPLICIT), XIMCRO(R DIRECTION IMPLICIT)
622      IF (MOD(M,2) .EQ. 1) GO TO 55
623      GO TO 52
624      55 CALL      YIMCRO(K,M,DT,DX,DCA,CV,FK,COE,RB,Q,UN,BATA,SGO,
625      1IS,IB,IE,ID,JS,JB,JE,JD,C,IP,JP,TMALL,DISS)
626      GO TO 253
627      52 CALL      XIMCRO(K,M,DT,DX,DCA,CV,FK,COE,RB,Q,UN,BATA,SGO,
628      1IS,IB,IE,ID,JS,JB,JE,JD,C,IP,JP,TMALL,DISS)
629      253 CONTINUE
630      TTCUR=TTCUR
631      TTCUR=TCUR
632      TCUR=0.0
633      CC      FIND TOTAL TOROIDAL PLASMA CURRENT, TCUR(IN AMP.)
634      CALL ARAIN(C3(1,1,K+1),TCUR,RB,RSMA)
635      TCUR=TCUR/3.0E9*C*BTOR/(4.0*PI*RSMA)
636      DO 903 J=JS,JD
637      DO 903 I=1,4
638      W(I,IS,J,K+2)=W(I,IS,J,K+1)
639      903 CONTINUE
640      DO 620 J=JS,JD
641      UTH(J)=(EI(ID,J,K+1)-EI(ID-1,J,K+1))/(1.*DX)
642      UTH(J)= UTH(J)*UN*UN/CV/RSMA*CNCON
643      620 CONTINUE
644      CC      TOTAL CONDUCTION LOSS WTFIN
645      CALL SURIN(UTH,WTFINC,RB,RSMA)
646      WTFIN=WTFIN+WTFINC*TIMINC
647      WINTOT=WIN+WTFIN
648      DO 607 J=JS,JD
649      UTH(J)=0.5*DN*(W(2,IE,J,K+2)**2+W(3,IE,J,K+2)**2+
650      1 W(4,IE,J,K+2)**2)/W(1,IE,J,K+2)*UN*UN
651      UTH(J)=UTH(J)+2.5*W(1,IE,J,K+2)*DN*2.*DNUM*1.38E-16*
652      1 EI(IE,J,K+1)/CV*UN*UN
653      UTH(J)=UTH(J)*W(2,IE,J,K+2)*UN/W(1,IE,J,K+2)
654      607 CONTINUE
655      CALL SURIN(UTH,WRRINC,RB,RSMA)
656      WRRR=WRRR+WRRINC*TIMINC
657      IF (MOD(M,MW) .EQ. 0 .AND. M.GE.MST) GO TO 89
658      GO TO 91
659      89 CONTINUE
660      SIME=TIME*RSMA/UN

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661      WRITE (6,104) M,SIME,DT,UN,COE
662      104 FORMAT(1H1,"CYCLE=",15,2X,"TIME=",1PE14.7,2X,"DT=",E14.7,
663      1"UN=",E14.7,E16.7)
664      CC
665      CC....CHANGE TO CGS GAUSSIAN UNITS FOR PRINTING
666      DO 92 I=IS, ID
667      DO 92 J=JS, JD
668      XI (I,J,K)=XI (I,J,K+1)*BTOR *RSMA
669      PSI(I,J,K)=PSI(I,J,K+1)*BTOR*RSMA**2
670      B1 (I,J,K)=B1 (I,J,K+1)*BTOR
671      B2 (I,J,K)=B2 (I,J,K+1)*BTOR
672      B3 (I,J,K)=B3 (I,J,K+1)*BTOR
673      C1 (I,J,K)=C1 (I,J,K+1)*C*BTOR/(4.*PI*RSMA)
674      C2 (I,J,K)=C2 (I,J,K+1)*C*BTOR/(4.*PI*RSMA)
675      C3 (I,J,K)=C3 (I,J,K+1)*C*BTOR/(4.*PI*RSMA)
676      92 CONTINUE
677      106 FORMAT (1H ,2I3,8E15.7)
678      WRITE (6,107)
679      107 FORMAT (1H0,1X,"I",1X,"J",2X,"DENSITY",2X,"RAD VEL",2X,
680      1 "TEMP",5X,"THET MAG",1X,"TORO MAG",1X,"THET CUR",1X,"TORO CUR"
681      2 ,1X,"ANOM RES",1X,"CLAS RES",1X,"THET VEL",1X,"TORO VEL"
682      3 ,1X,"RAD MAG")
683      DO 93 I=IS, ID
684      DO 93 J=JS, JD
685      F(2,I,J,K+1)=W(2,I,J,K+2)/W(1,I,J,K+2)*UN
686      F(3,I,J,K+1)=W(3,I,J,K+2)/W(1,I,J,K+2)*UN
687      F(4,I,J,K+1)=W(4,I,J,K+2)/W(1,I,J,K+2)*UN
688      F(1,I,J,K+1)=EI(I,J,K+1)/CV*UN*UN
689      T (I,J,K)=T (I,J,K+1)
690      CN(I,J,K)=CN(I,J,K+1)*DN*CV*RSMA*UN
691      ALAM(I,J)=(F(1,I,J,K+1)/TEMINI)**1.5
692      IF (MOD(MBB,2).EQ.0)WRITE (6,108) I,J,W(1,I,J,K+2),F(2,I,J,K+1),
693      1 F(1,I,J,K+1), B2(I,J,K),B3(I,J,K)
694      2,C2(I,J,K),C3(I,J,K),VELD(I,J),ALAM(I,J),F(3,I,J,K+1)
695      3,F(4,I,J,K+1),B1(I,J,K)
696      93 CONTINUE
697      CC
698      CC....FOR PLOTTING
699      DO 551 I=1,10
700      IF(I.GT.6) GO TO 552
701      L=I-7
702      IF(B2(-L,JE,1).LT.0.0)BAX=L-B2(-L+1,JE,1)/(B2(-L,JE,1)-B2(-L+1,JE,
703      1))
704      IF(B2(-L,JE,1).LT.0.0)GO TO 553
705      552 CONTINUE
706      L=I-5
707      IF(B2(L,JB,1).GT.0.0)BAX=L-2-B2(L-1,JB,1)/(B2(L,JB,1)-B2(L-1,JB,1)
708      1)
709      IF(B2(L,IB,1).GT.0.0)GO TO 553
710      551 CONTINUE
711      553 BAX=BAX*DX
712      IDD=ID-2
713      DO 1200 I=1,IDD
714      J=ID-I
715      X(I)=(I-IE)
716      FVL(I,1)=W(1,J,JE,3)
717      FVL(I,2)=F(1,J,JE,2)
718      FVL(I,3)=B3(J,JE,1)
719      FVL(I,4)=-B2(J,JE,1)
720      FVL(I,5)=C3(J,JE,1)

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721      FVL(1,6)=(SR(J)+BAX)*Q*B3(J,JE,1)/B2(J,JE,1)
722 1200 CONTINUE
723      IDS=ID-1
724      IDN=2*ID-3
725      DO 1300 I=IDS, IDN
726      X(I)=(I-IE)
727      J=I-ID+2
728      FVL(I,1)=W(1,J,JB,3)
729      FVL(I,2)=F(1,J,JB,2)
730      FVL(I,3)=B3(J,JB,1)
731      FVL(I,4)=B2(J,JB,1)
732      FVL(I,5)=C3(J,JB,1)
733      FVL(I,6)=(SR(J)-BAX)*Q*B3(J,JB,1)/B2(J,JB,1)
734 1300 CONTINUE
735      CALL WPLOTT(X,FVL(1,1),FVL(1,2),FVL(1,3),FVL(1,4),FVL(1,5),
736      1 FVL(1,6),IDN,SIME)
737 CC
738 CC
739      100 FORMAT (1H ,2I2,1P14E9.1)
740      DO 802 I=1, ID
741      DO 802 J=1, JD
742      VINT(I,J)=AMAX1(B3(I,J,K),0.0)
743 802 CONTINUE
744 CC TRAPED POSITIVE TOROIDAL FLUX, TRFL
745      CALL ARAIN(VINT,TRFL,RB,RSMA)
746      DO 805 I=1, ID
747      DO 805 J=1, JD
748      VINT(I,J)=AMIN1(B3(I,J,K),0.0)
749      EI(I,J,1)=-XI(I,J,1)
750      T(I,J,1)=-C2(I,J,1)
751 805 CONTINUE
752 CC TOTAL TOROIDAL FLUX, TOTFL
753      CALL ARAIN(VINT,ANFL,RB,RSMA)
754      TOTFL=TRFL+ANFL
755      WRITE(6,306)TCUR
756      WRITE(6,807)M, PSI(ID,5,K+1),PSINC,PSIPP,TCUR,SURCUR,TRFL,TOTFL
757      IF(ITURN.EQ. 1)CALL WWPLOTT(XNL,TVA1,TVA2,TVA3,NLL)
758      ITURN=0
759 306 FORMAT(1H , "*** TOTAL TOROID CURR(IN AMP.)=",1PE12.3)
760      WINTOT=WINTOT+1.5*DN*DNUM*2.0*VOLU*1.38E-16*TEMINI
761      DO 602 I=1S, ID
762      DO 602 J=JS, JD
763      VINT(I,J)=(B1(I,J,K)**2+B2(I,J,K)**2+B3(I,J,K)**2)/B./PI
764 CCCC WRITE(6,108)I,J,F(3,I,J,2),F(4,I,J,2),B1(I,J,K),C1(I,J,K)
765 602 CONTINUE
766 CC TOTAL MAGNETIC ENERGY, WBE
767      CALL VOLIN(VINT,WBE,RB,RSMA)
768      DO 603 I=1S, ID
769      DO 603 J=JS, JD
770      VINT(I,J)=1.5*W(1,I,J,K+2)*DN*2.*DNUM*1.38E-16*F(1,I,J,K+1)
771 603 CONTINUE
772 CC TOTAL THERMAL ENERGY, WIE
773      CALL VOLIN(VINT,WIE,RB,RSMA)
774      DO 604 I=1S, ID
775      DO 604 J=JS, JD
776      VINT(I,J)=0.5*W(1,I,J,K+2)*DN*(F(2,I,J,K+1)**2+F(3,I,J,K+1)**2+
777      1 F(4,I,J,K+1)**2)
778 604 CONTINUE
779 CC TOTAL KINETIC ENERGY, WME
780      CALL VOLIN(VINT,WME,RB,RSMA)

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781      WTOTR=WBE+WIE
782      WTOT=WTOTR+WME
783      DO 632 I=IS, ID
784      DO 632 J=JS, JD
785      VINT(I, J)=W(1, I, J, K+2)
786      632 CONTINUE
787      CC      TOTAL MASS, TOMA
788      CALL VOLIN(VINT, TOMA, RB, RSMA)
789      CC      POLOIDAL BETA(PLBT), AVERAGE TEMP.(AVT), TOTAL ENERGY INPUT(WIINT)
790      TOMA=TOMA/VOLU
791      PLBT=WIE*2.0/TCUR/TCUR/1.5
792      TNM=VOLU*DN*DNUM*2.0
793      AVT=WIE/INM/1.4E-16/1.5/11600.0
794      WIINT=WTOT-WTFIN
795      WRITE(6, 606)M, AVT, PLBT, WIINT
796      NLL=NLL+1
797      IL=ID*3/5
798      JL=JD/4+1
799      TVA1(NLL)=AVT
800      TVA2(NLL)=PLBT
801      TVA3(NLL)=WIINT
802      XNL(NLL)=NLL
803      JM=JD/2.0+0.5
804      TVA4(NLL)=Q*B3(ID, JM, 1)/B2(IE, JM, 1)
805      TVA5(NLL)=B3(1, 2, 1)
806      TVA6(NLL)=W(1, 1, 2, 3)
807      WRITE(6, 606)M, SIME, WZIN, WTIN, WBE, WIE, WME, WRRR, WTFIN, WIN, WINTOT
808      1, WTOT, TOMA
809      606 FORMAT(1H0, I5, 1P12E10.2)
810      91 CONTINUE
811      IF(M.GT. MBK) GO TO ABKP
812      GO TO PASS
813      ABKP CONTINUE
814      QQQQ=PI
815      PASS CONTINUE
816      IF (M.GE. MS) GO TO 94
817      CC
818      CC      INITIALIZATION FOR NEXT CYCLE
819      DO 96 I=IS, ID
820      DO 96 J=JS, JD
821      T (I, J, K)=T (I, J, K+1)
822      EI(I, J, K) = EI(I, J, K+1)
823      PSI(I, J, K)=PSI(I, J, K+1)
824      B3(I, J, K) = B3(I, J, K+1)
825      B1 (I, J, K)=B1 (I, J, K+1)
826      B2 (I, J, K)=B2 (I, J, K+1)
827      XI (I, J, K)=XI (I, J, K+1)
828      C1 (I, J, K)=C1 (I, J, K+1)
829      C2 (I, J, K)=C2 (I, J, K+1)
830      C3 (I, J, K)=C3 (I, J, K+1)
831      CN (I, J, K)=CN (I, J, K+1)
832      DO 97 L=1, 4
833      W(L, I, J, K) = W(L, I, J, K+2)
834      IF (ABS(W(L, I, J, K)) .LT. SMALL) W(L, I, J, K)=0.
835      97 CONTINUE
836      CC
837      CC.....LOW DENSITY REGION TREATMENT (CF. SECTION 3.3)
838      IF(I.EQ. 1 .OR. I.EQ.ID)GO TO 81
839      IF(W(1, I, J, K).LT.1.2*HU.AND.W(1, I-1, J, K).GT.1.9*HU.AND.
840      1 W(2, I-1, J, K).LT.0.0)W(1, I-1, J, K)=W(1, I-1, J, K)+BATA*(W(2, I-1, J, K)

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841      1 +W(2,I,J,K))
842      IF(W(1,I,J,K).LT.1.2*HU.AND.W(1,I-1,J,K).GT.1.8*HU.AND.
843      1 W(2,I-1,J,K).GT.0.0)W(1,I-1,J,K)=W(1,I-1,J,K)+BATA*W(2,I-1,J,K)
844      IF(W(1,I-1,J,K).LT.1.2*HU.AND.W(1,I,J,K).GT.1.8*HU.AND.W(2,I,J,K)
845      1 .GT.0.0)W(1,I,J,K)=W(1,I,J,K)-BATA*(W(2,I,J,K)+W(2,I-1,J,K))
846      IF(W(1,I,J,K).LT.1.2*HU.AND.W(1,I-1,J,K).GT.1.8*HU.AND.W(2,I-1,J,
847      1 K).LT.0.0)W(2,I-1,J,K)=W(2,I-1,J,K)+BATA*(W(2,I-1,J,K)**2/
848      1 W(1,I-1,J,K)+W(2,I,J,K)**2/W(1,I,J,K))
849      IF(W(1,I,J,K).LT.1.2*HU.AND.W(1,I-1,J,K).GT.1.8*HU.AND.W(2,I-1,J,
850      1 K).GT.0.0)W(2,I-1,J,K)=W(2,I-1,J,K)+BATA*W(2,I-1,J,K)**2/
851      1 W(1,I-1,J,K)
852      IF(W(1,I-1,J,K).LT.1.2*HU.AND.W(1,I,J,K).GT.1.8*HU.AND.W(2,I,J,K)
853      1 .GT.0.0)W(2,I,J,K)=W(2,I,J,K)-BATA*(W(2,I,J,K)**2/W(1,I,J,K)
854      1 +W(2,I-1,J,K)**2/W(1,I-1,J,K))
855  81  IF(W(1,I,J,K).LT.HU) W(1,I,J,K)=HU
856      96 CONTINUE
857      GO TO 6
858      94 CONTINUE
859      CALL UJWPLT(XNL,TVA1,TVA2,TVA3,NLL)
860      CALL UJWPLT(XNL,TVA4,TVA5,TVA6,NLL)
861      WRITE(6,109) DT
862  109 FORMAT(1H,"END OF THE SCREW PINCH",4X,"DT=",E14.7)
863      STOP
864      END
865      SUBROUTINE WPLT(X,A,B,C,D,E,F, IDN,SIME)
866      DIMENSION X(40),A(40),B(40),C(40),D(40),E(40),F(40)
867      COMMON /HEADERS/ LABEL(8),TIME(2),JOBNM,DATE, HOUR
868      REWIND 63
869      WRITE(63,2)SIME
870      2  FORMAT(2HT=,1PE9.2,4H SEC)
871      READ(63,3)TIME
872      3  FORMAT(2A10)
873      CALL LPLT(-2,4, IDN,X,A,1,1,1H,1HR,4HDENS)
874      CALL LPLT(-3,4, IDN,X,B,1,1,1H,1HR,4HTEMP)
875      CALL LPLT(-2,5, IDN,X,C,1,1,1H,1HR,2HBZ)
876      CALL LPLT(-2,6, IDN,X,D,1,1,1H,1HR,3HOTH)
877      CALL LPLT(-3,5, IDN,X,E,1,1,1H,1HR,2HJZ)
878      CALL LPLT(-3,6, IDN,X,F,1,1,1H,1HR,1HQ)
879      RETURN
880      END
881      SUBROUTINE UJWPLT(XNL,TVA1,TVA2,TVA3,NLL)
882      DIMENSION XNL(100),TVA1(100),TVA2(100),TVA3(100)
883      CALL DDERS(-1)
884      CALL FRAME
885      CALL MAPPARAM(2,0,15,0,1)
886      AMY=AMAXAF(TVA1,1,NLL,1)
887      AMIY=AMINAF(TVA1,1,NLL,1)
888      CALL MAPS(XNL(1),XNL(NLL),AMIY,AMY,0.0,1.0,0.76,1.0)
889      CALL SETPCH(1,0,1,0,20)
890      CALL TRACEC(1H,XNL,TVA1,NLL)
891      AMY=AMAXAF(TVA2,1,NLL,1)
892      AMIY=AMINAF(TVA2,1,NLL,1)
893      CALL MAPS(XNL(1),XNL(NLL),AMIY,AMY,0.0,1.0,0.44,0.68)
894      CALL SETPCH(1,0,1,0,20)
895      CALL TRACEC(1H,XNL,TVA2,NLL)
896      AMY=AMAXAF(TVA3,1,NLL,1)
897      AMIY=AMINAF(TVA3,1,NLL,1)
898      CALL MAPS(XNL(1),XNL(NLL),AMIY,AMY,0.0,1.0,0.12,0.36)
899      CALL SETPCH(1,0,1,0,20)
900      CALL TRACEC(1H,XNL,TVA3,NLL)

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901 RETURN
902 END
903 SUBROUTINE YIMCRO(K,M,DT,DX,DCA,CV,FK,COE,RB,Q,UN,BATA,SGO,
904 IIS,IB,IE,ID,JS,JB,JE,JD,C,IP,JP,TMALL,DISS)
905 CC
906 CC----- SOLVE NEW TIME STEP VALUES W(M,I,J,3) IN THETA DIREC. IMPLICIT
907 CC
908 DIMENSION PSI(20,16,2),B1(20,16,2),B2(20,16,2),XI(20,16,2),
909 IC1(20,16,2),C2(20,16,2),C3(20,16,2),T(20,16,2),
910 2W(4,20,16,3),F(4,20,16,3),EI(20,16,2),B3(20,16,2),SR(20),
911 4GA(20,16),GB(20,16),GC(20,16),GD(20,16),CN(20,16,2),
912 5AA(20,12),BB(4,4),ANG(16),BATB(20)
913 COMMON PSI,B1,B2,B3,XI,C1,C2,C3,T,W,F,SR,ANG,BATB
914 1,GA,GB,GC,GD,CN,EI,AA,RR,AICCN
915 K1=1
916 K2=3
917 K3=1
918 K=1
919 CC
920 CC.... FIND F IN THE MATRIX EQ. PW = F
921 DO 38 J=JB,JE
922 DO 39 I=IB,IE
923 F(1,I,J,K2)=W(1,I,J,K1)
924 1-BATA *( W(2,I+1,J,K3)-
925 2 W(2,I-1,J,K3))+F(1,I,J,K)+F(1,I,J,K+1)
926 6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
927 7/W(1,I,J,K))*W(1,I+1,J,K)-W(1,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
928 8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(1,I,J,K)-W(1,I-1,J,K)))
929 6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
930 7/W(1,I,J,K))*W(1,I,J+1,K)-W(1,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
931 8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(1,I,J,K)-W(1,I,J-1,K)))
932 F(2,I,J,K2)=W(2,I,J,K1)
933 1-BATA *( (W(2,I+1,J,K3)**2
934 2/W(1,I+1,J,K3)+(FK-1.)*W(1,I+1,J,K3)*EI(I+1,J,K3))-
935 3(W(2,I-1,J,K3)**2/W(1,I-1,J,K3) +(FK-1.)*W(1,I-1,J,K3)
936 4*EI(I-1,J,K3)))+F(2,I,J,K)+F(2,I,J,K+1)
937 6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
938 7/W(1,I,J,K))*W(2,I+1,J,K)-W(2,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
939 8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(2,I,J,K)-W(2,I-1,J,K)))
940 6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
941 7/W(1,I,J,K))*W(2,I,J+1,K)-W(2,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
942 8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(2,I,J,K)-W(2,I,J-1,K)))
943 F(3,I,J,K2)=W(3,I,J,K1)
944 1-BATA *( W(2,I+1,J,K3)*
945 2W(3,I+1,J,K3)/W(1,I+1,J,K3)- W(2,I-1,J,K3)*W(3,I-1,J,K3)
946 3/W(1,I-1,J,K3))+F(3,I,J,K)+F(3,I,J,K+1)
947 6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
948 7/W(1,I,J,K))*W(3,I+1,J,K)-W(3,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
949 8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(3,I,J,K)-W(3,I-1,J,K)))
950 6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
951 7/W(1,I,J,K))*W(3,I,J+1,K)-W(3,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
952 8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(3,I,J,K)-W(3,I,J-1,K)))
953 F(4,I,J,K2)=W(4,I,J,K1)
954 1-BATA *( W(2,I+1,J,K3)
955 2*W(4,I+1,J,K3)/W(1,I+1,J,K3)- W(2,I-1,J,K3)*W(4,I-1,J,K3)
956 1/W(1,I-1,J,K3))+F(4,I,J,K)+F(4,I,J,K+1)
957 6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
958 7/W(1,I,J,K))*W(4,I+1,J,K)-W(4,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
959 8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(4,I,J,K)-W(4,I-1,J,K)))
960 6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)

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961      7/W(1,I,J,K))*W(4,I,J+1,K)-W(4,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
962      8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(4,I,J,K)-W(4,I,J-1,K)))
963      39 CONTINUE
964      F(2,IE,J,K2)=F(2,IE,J,K2)+2.0*BATA*W(2,ID,J,K)**2/W(1,ID,J,K)
965      38 CONTINUE
966      CC
967      CC.... FIND ELEMENTS AA OF MATRIX P IN THE MATRIX EQ. PW=F.
968      DO 113 I=IB,IE
969      DO 114 J=JS,JP
970      IF (J.EQ. JS) GO TO 115
971      AA(4*J-3,1)=0.
972      AA(4*J-3,2)=0.
973      AA(4*J-3,3)=-BATB(I)
974      AA(4*J-3,4)=0.
975      AA(4*J-2,1)=0.0
976      AA(4*J-2,2)=-BATB(I)*W(3,I,J,K+1)/W(1,I,J,K+1))
977      AA(4*J-2,3)=0.0
978      AA(4*J-2,4)=0.
979      AA(4*J-1,1)=-BATB(I)*(FK-1.)*EI(I,J,K+1)
980      AA(4*J-1,2)=0.
981      AA(4*J-1,3)=-BATB(I)*( W(3,I,J,K+1)/W(1,I,J,K+1))
982      AA(4*J-1,4)=0.
983      AA(4*J ,1)=0.0
984      AA(4*J ,2)=0.
985      AA(4*J ,3)=0.0
986      AA(4*J ,4)=-BATB(I)*W(3,I,J,K+1)/W(1,I,J,K+1))
987      115 CONTINUE
988      IF (J.EQ. JP) GO TO 116
989      AA(4*J-3,9)=0.
990      AA(4*J-3,10)=0.
991      AA(4*J-3,11)=+BATB(I)
992      AA(4*J-3,12)=0.
993      AA(4*J-2,9)=0.0
994      AA(4*J-2,10)=+BATB(I)*W(3,I,J,K+1)/W(1,I,J,K+1))
995      AA(4*J-2,11)=0.0
996      AA(4*J-2,12)=0.
997      AA(4*J-1,9)=+BATB(I)*(FK-1.)*EI(I,J,K+1)
998      AA(4*J-1,10)=0.
999      AA(4*J-1,11)=+BATB(I)*( W(3,I,J,K+1)/W(1,I,J,K+1))
1000     AA(4*J-1,12)=0.
1001     AA(4*J ,9)=0.0
1002     AA(4*J ,10)=0.
1003     AA(4*J ,11)=0.0
1004     AA(4*J ,12)=+BATB(I)*W(3,I,J,K+1)/W(1,I,J,K+1))
1005     116 CONTINUE
1006     AA(4*J-3,5)=1.
1007     AA(4*J-3,6)=0.
1008     AA(4*J-3,7)=0.
1009     AA(4*J-3,8)=0.
1010     AA(4*J-2,5)=0.
1011     AA(4*J-2,6)=1.
1012     AA(4*J-2,7)=0.
1013     AA(4*J-2,8)=0.
1014     AA(4*J-1,5)=0.
1015     AA(4*J-1,6)=0.
1016     AA(4*J-1,7)=1.
1017     AA(4*J-1,8)=0.
1018     AA(4*J ,5)=0.
1019     AA(4*J ,6)=0.
1020     AA(4*J ,7)=0.

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1021      AA(4*J ,8)=1.
1022      114 CONTINUE
1023      DO 207 L=1,2
1024      DO 208 LL=9,12
1025      AA(L ,LL)= 2.*AA(L ,LL)
1026      AA(L+2,LL)=0.
1027      208 CONTINUE
1028      207 CONTINUE
1029      JP1=4*(JP-1)+1
1030      JP2=JP1+1
1031      DO 307 L=JP1,JP2
1032      DO 308 LL=1,4
1033      AA(L ,LL)= 2.*AA(L ,LL)
1034      AA(L+2,LL)=0.
1035      308 CONTINUE
1036      307 CONTINUE
1037      CC
1038      CC....CROUT REDUCTION
1039      JPP=4*JP
1040      DO 117 J=5,JPP,4
1041      DO 188 L=1,4
1042      DO 189 LL=1,4
1043      BB(L,LL)=0.
1044      DO 121 LLL=1,4
1045      IF (ABS(AA(J+L-1,LLL)) .LT. TMALL) AA(J+L-1,LLL)=0.
1046      IF (ABS(AA(J+LLL-5,8+LL)) .LT. TMALL) AA(J+LLL-5,8+LL)=0.
1047      121 BB(L,LL)=BB(L,LL) + AA(J+L-1,LLL)*AA(J+LLL-5,8+LL)
1048      189 CONTINUE
1049      188 CONTINUE
1050      DO 122 L=1,4
1051      122 AA(J-1+L,5)=AA(J-1+L,5)-BB(L,1)
1052      DO 123 L=1,3
1053      123 AA(J,5+L)=(AA(J,5+L)-BB(1,L+1))/AA(J,5)
1054      IF (J .EQ. (JPP-3)) GO TO 924
1055      DO 124 L=1,4
1056      124 AA(J,8+L)=AA(J,8+L)/AA(J,5)
1057      924 CONTINUE
1058      DO 125 L=1,3
1059      125 AA(J+L,6)=AA(J+L,6)-BB(L+1,2)-AA(J+L,5)*AA(J,6)
1060      DO 126 L=1,2
1061      126 AA(J+1,6+L)=(AA(J+1,6+L)-BB(2,2+L)-AA(J+1,5)*AA(J,6+L))/AA(J+1,6)
1062      IF (J .EQ. (JPP-3)) GO TO 127
1063      DO 128 L=1,4
1064      AA(J+L+3,2)=AA(J+L+3,2)-AA(J+L+3,1)*AA(J,6)
1065      AA(J+1,8+L)=(AA(J+1,8+L)-AA(J+1,5)*AA(J,8+L))/AA(J+1,6)
1066      128 CONTINUE
1067      127 CONTINUE
1068      DO 129 L=1,2
1069      129 AA(J+L+1,7)=AA(J+L+1,7)-BB(2+L,3)-AA(J+L+1,5)*AA(J,7)-
1070      1AA(J+L+1,6)*AA(J+1,7)
1071      AA(J+2,8)=(AA(J+2,8)-BB(3,4)-AA(J+2,5)*AA(J,8)-AA(J+2,6)*AA(J+1,8)
1072      2)/AA(J+2,7)
1073      IF (J .EQ. (JPP-3)) GO TO 130
1074      DO 131 L=1,4
1075      DO 132 LL=1,2
1076      AA(J+L+3,3)=AA(J+L+3,3)-AA(J+L+3,LL)*AA(J+LL-1,7)
1077      AA(J+2,8+LL)=AA(J+2,8+LL)-AA(J+2,4+LL)*AA(J+LL-1,8+LL)
1078      132 CONTINUE
1079      AA(J+2,8+L)=AA(J+2,8+L)/AA(J+2,7)
1080      131 CONTINUE

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1081      130 CONTINUE
1082      AA(J+3,8)=AA(J+3,8)-BB(4,4)-AA(J+3,5)*AA(J,8)-AA(J+3,6)*AA(J+1,8)
1083      1-AA(J+3,7)*AA(J+2,8)
1084      IF (J .EQ. (JPP-3)) GO TO 133
1085      DO 134 L=1,4
1086      DO 135 LL=1,3
1087      AA(J+L+3,4)=AA(J+L+3,4)-AA(J+L+3,LL)*AA(J+LL-1,8)
1088      AA(J+3,8+L)=AA(J+3,8+L)-AA(J+3,4+LL)*AA(J+LL-1,8+L)
1089      135 CONTINUE
1090      AA(J+3,8+L)=AA(J+3,8+L)/AA(J+3,8)
1091      134 CONTINUE
1092      133 CONTINUE
1093      DO 136 L=1,4
1094      DO 137 LL=1,4
1095      137 F(L,I,(J+7)/4,K+2)=F(L,I,(J+7)/4,K+2)-AA(J+L-1,LL)*F(LL,I,(J+7)/4
1096      1-1,K+2)
1097      LE=L-1
1098      IF (L .EQ. 1) GO TO 148
1099      DO 147 LL=1,LE
1100      F(L,I,(J+7)/4,K+2)=F(L,I,(J+7)/4,K+2)-AA(J+L-1,4+LL)*F(LL,I,(J+7)
1101      1/4,K+2)
1102      147 CONTINUE
1103      148 CONTINUE
1104      F(L,I,(J+7)/4,K+2)=F(L,I,(J+7)/4,K+2)/AA(J+L-1,5+L-1)
1105      136 CONTINUE
1106      117 CONTINUE
1107      W(4,I,JE,K+2)=F(4,I,JE,K+2)
1108      W(3,I,JE,K+2)=F(3,I,JE,K+2)-AA(JPP-1,8)*W(4,I,JE,K+2)
1109      W(2,I,JE,K+2)=F(2,I,JE,K+2)-AA(JPP-2,8)*W(4,I,JE,K+2)
1110      1-AA(JPP-2,7)*W(3,I,JE,K+2)
1111      W(1,I,JE,K+2)=F(1,I,JE,K+2)-AA(JPP-3,8)*W(4,I,JE,K+2)
1112      1-AA(JPP-3,7)*W(3,I,JE,K+2)-AA(JPP-3,6)*W(2,I,JE,K+2)
1113      DO 138 J=5,JPP,4
1114      JJ=JPP+5-J-4
1115      DO 139 L=1,4
1116      BB(1,L)=0.
1117      DO 941 LL=1,4
1118      941 BB(1,L)=BB(1,L)+AA(JJ+1-L,8+LL)*W(LL,I,(JJ+4)/4+1,K+2)
1119      139 CONTINUE
1120      W(4,I,(JJ+4)/4,K+2)=F(4,I,(JJ+4)/4,K+2)-BB(1,1)
1121      W(3,I,(JJ+4)/4,K+2)=F(3,I,(JJ+4)/4,K+2)-BB(1,2)
1122      1-AA(JJ-1,8)*W(4,I,(JJ+4)/4,K+2)
1123      W(2,I,(JJ+4)/4,K+2)=F(2,I,(JJ+4)/4,K+2)-BB(1,3)
1124      1-AA(JJ-2,8)*W(4,I,(JJ+4)/4,K+2)-AA(JJ-2,7)*W(3,I,(JJ+4)/4,K+2)
1125      W(1,I,(JJ+4)/4,K+2)=F(1,I,(JJ+4)/4,K+2)-BB(1,4)
1126      1-AA(JJ-3,8)*W(4,I,(JJ+4)/4,K+2)-AA(JJ-3,7)*W(3,I,(JJ+4)/4,K+2)
1127      2-AA(JJ-3,6)*W(2,I,(JJ+4)/4,K+2)
1128      138 CONTINUE
1129      113 CONTINUE
1130      RETURN
1131      END
1132      SUBROUTINE XIMCRO(K,M,DT,DX,DCA,CV,FK,COE,RB,Q,UN,BATA,SGO,
1133      1 IS,IB,IE,ID,JS,JB,JE,JD,C,IP,JP,TMALL,DISS)
1134      CC
1135      CC---- SOLVE NEW TIME STEP VALUES W(M,I,J,3) IN R DIRECTION IMPLICIT
1136      CC
1137      DIMENSION PSI(20,16,2),B1(20,16,2),B2(20,16,2),XI(20,16,2),
1138      1C1(20,16,2),C2(20,16,2),C3(20,16,2),T(20,16,2),
1139      2W(4,20,16,3),F(4,20,16,3),E1(20,16,2),B3(20,16,2),SR(20),
1140      4GA(20,16),GB(20,16),GC(20,16),GD(20,16),CN(20,16,2),

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1141      5AA(90,12),BB(4,4),ANG(16),BATB(20)
1142      COMMON PSI,B1,B2,B3,XI,C1,C2,C3,T,W,F,SR,ANG,BATB
1143      I,GA,GB,GC,GD,CN,EI,AA,BB,A,ICON
1144      K=1
1145      K1=1
1146      K2=3
1147      K3=1
1148      CC
1149      CC....FIND F IN THE MATRIX EQ. PW=F
1150      DO 41 J=JB,JE
1151      DO 42 I=IB,IE
1152      F(1,I,J,K2)=W(1,I,J,K1)
1153      2-BATB(I)*(W(3,I,J+1,K3)-W(3,I,J-1,K3))
1154      3+(F(1,I,J,K)+F(1,I,J,K+1))
1155      6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1156      7/W(1,I,J,K))*W(1,I+1,J,K)-W(1,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
1157      8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(1,I,J,K)-W(1,I-1,J,K))
1158      6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1159      7/W(1,I,J,K))*W(1,I,J+1,K)-W(1,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
1160      8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(1,I,J,K)-W(1,I,J-1,K))
1161      F(2,I,J,K2)=W(2,I,J,K1)
1162      4-BATB(I)*(W(2,I,J+1,K3)*W(3,I,J+1,K3)
1163      5/W(1,I,J+1,K3)-W(2,I,J-1,K3)*W(3,I,J-1,K3)/W(1,I,J-1,K3))
1164      6+(F(2,I,J,K)+F(2,I,J,K+1))
1165      6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1166      7/W(1,I,J,K))*W(2,I+1,J,K)-W(2,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
1167      8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(2,I,J,K)-W(2,I-1,J,K))
1168      6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1169      7/W(1,I,J,K))*W(2,I,J+1,K)-W(2,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
1170      8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(2,I,J,K)-W(2,I,J-1,K))
1171      F(3,I,J,K2)=W(3,I,J,K1)
1172      3-BATB(I)*(W(3,I,J+1,K3)**2/W(1,I,J+1,K3)+
1173      4(FK-1.)*W(1,I,J+1,K3)*EI(I,J+1,K))-W(3,I,J-1,K3)**2/
1174      5W(1,I,J-1,K3)+(FK-1.)*W(1,I,J-1,K3)*EI(I,J-1,K))
1175      6+(F(3,I,J,K)+F(3,I,J,K+1))
1176      6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1177      7/W(1,I,J,K))*W(3,I+1,J,K)-W(3,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
1178      8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(3,I,J,K)-W(3,I-1,J,K))
1179      6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1180      7/W(1,I,J,K))*W(3,I,J+1,K)-W(3,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
1181      8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(3,I,J,K)-W(3,I,J-1,K))
1182      F(4,I,J,K2)=W(4,I,J,K1)
1183      3-BATB(I)*(W(3,I,J+1,K3)*W(4,I,J+1,K3)
1184      4/W(1,I,J+1,K3)-W(3,I,J-1,K3)*W(4,I,J-1,K3)/W(1,I,J-1,K3))
1185      5+(F(4,I,J,K)+F(4,I,J,K+1))
1186      6+DISS*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1187      7/W(1,I,J,K))*W(4,I+1,J,K)-W(4,I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
1188      8-W(2,I-1,J,K)/W(1,I-1,J,K))*W(4,I,J,K)-W(4,I-1,J,K))
1189      6+DISS*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1190      7/W(1,I,J,K))*W(4,I,J+1,K)-W(4,I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
1191      8-W(3,I,J-1,K)/W(1,I,J-1,K))*W(4,I,J,K)-W(4,I,J-1,K))
1192      42 CONTINUE
1193      41 CONTINUE
1194      DO 153 J=JB,JE
1195      F(2,IE,J,K+2)=F(2,IE,J,K+2)-(FK-1.)*W(1,ID,J,K+1)*EI(ID,J,K+1)
1196      1*BATA+BATA*W(2,ID,J,2)**2/W(1,ID,J,2)
1197      F(1,IE,J,3)=F(1,IE,J,3)-BATA*W(2,ID,J,2)
1198      F(3,IE,J,3)=F(3,IE,J,3)-BATA*W(2,ID,J,2)*W(3,ID,J,2)/W(1,ID,J,2)
1199      F(4,IE,J,3)=F(4,IE,J,3)-BATA*W(2,ID,J,2)*W(4,ID,J,2)/W(1,ID,J,2)
1200      F(1,IB,J,K+2)=F(1,IB,J,K+2)+BATA*W(2,IS,J,K+1)

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1201      F(2, IB, J, K+2) = F(2, IB, J, K+2) + BATA * W(2, IS, J, K+1) ** 2 / W(1, IS, J, K+1)
1202      1 + BATA * (FK-1.0) * W(1, IS, J, K+1) * EI( IS, J, K+1)
1203      F(3, IB, J, K+2) = F(3, IB, J, K+2) + BATA * W(2, IS, J, K+1) * W(3, IS, J, K+1)
1204      1 / W(1, IS, J, K+1)
1205      F(4, IB, J, K+2) = F(4, IB, J, K+2) + BATA * W(2, IS, J, K+1) * W(4, IS, J, K+1)
1206      1 / W(1, IS, J, K+1)
1207      CC
1208      CC.... FIND ELEMENTS AA OF MATRIX P IN THE MATRIX EQ. PW=F
1209      DO 154 I=IS, IP
1210      IF (I .EQ. IS) GO TO 155
1211      AA(4*I-3, 1) = 0.
1212      AA(4*I-3, 2) = -BATA
1213      AA(4*I-3, 3) = 0.
1214      AA(4*I-3, 4) = 0.
1215      AA(4*I-2, 1) = -BATA * (FK-1.) * EI(I, J, K+1)
1216      AA(4*I-2, 2) = -BATA * ( W(2, I, J, K+1) /
1217      1W(1, I, J, K+1))
1218      AA(4*I-2, 3) = 0.
1219      AA(4*I-2, 4) = 0.
1220      AA(4*I-1, 1) = 0.0
1221      AA(4*I-1, 2 ) = 0.0
1222      AA(4*I-1, 3 ) = -BATA * (W(2, I, J, K+1) / W(1, I, J, K+1))
1223      AA(4*I-1, 4 ) = 0.
1224      AA(4*I , 1) = 0.0
1225      AA(4*I , 2) = 0.0
1226      AA(4*I , 3) = 0.
1227      AA(4*I , 4) = -BATA * (W(2, I, J, K+1) / W(1, I, J, K+1))
1228      155 CONTINUE
1229      IF (I .EQ. IP) GO TO 156
1230      AA(4*I-3, 9) = 0.
1231      AA(4*I-3, 10) = +BATA
1232      AA(4*I-3, 11) = 0.
1233      AA(4*I-3, 12) = 0.
1234      AA(4*I-2, 9) = +BATA * (FK-1.) * EI(I, J, K+1)
1235      AA(4*I-2, 10) = +BATA * ( W(2, I, J, K+1) /
1236      1W(1, I, J, K+1))
1237      AA(4*I-2, 11) = 0.
1238      AA(4*I-2, 12) = 0.
1239      AA(4*I-1, 9) = 0.0
1240      AA(4*I-1, 10) = 0.0
1241      AA(4*I-1, 11) = +BATA * (W(2, I, J, K+1) / W(1, I, J, K+1))
1242      AA(4*I-1, 12) = 0.
1243      AA(4*I , 9) = 0.0
1244      AA(4*I , 10) = 0.0
1245      AA(4*I , 11) = 0.
1246      AA(4*I , 12) = +BATA * (W(2, I, J, K+1) / W(1, I, J, K+1))
1247      156 CONTINUE
1248      AA(4*I-3, 5) = 1.
1249      AA(4*I-3, 6) = 0.
1250      AA(4*I-3, 7) = 0.
1251      AA(4*I-3, 8) = 0.
1252      AA(4*I-2, 5) = 0.
1253      AA(4*I-2, 6) = 1.
1254      AA(4*I-2, 7) = 0.
1255      AA(4*I-2, 8) = 0.
1256      AA(4*I-1, 5) = 0.
1257      AA(4*I-1, 6) = 0.
1258      AA(4*I-1, 7) = 1.
1259      AA(4*I-1, 8) = 0.
1260      AA(4*I , 5) = 0.

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1261      AA(4*I ,6)=0.
1262      AA(4*I ,7)=0.
1263      AA(4*I ,8)=1.
1264      154 CONTINUE
1265      CC
1266      CC....CROUT REDUCTION
1267      IPP=4*IP
1268      DO 165 I=5, IPP, 4
1269      DO 166 L=1, 4
1270      DO 167 LL=1, 4
1271      BB(L,LL)=0.
1272      DO 168 LLL=1, 4
1273      IF (ABS(AA(I+L-1,LLL)) .LT. TMALL) AA(I+L-1,LLL)=0.
1274      IF (ABS(AA(I+LLL-5,8+LL)) .LT. TMALL) AA(I+LLL-5,8+LL)=0.
1275      168 BB(L,LL)=BB(L,LL) + AA(I+L-1,LLL)*AA(I+LLL-5,8+LL)
1276      167 CONTINUE
1277      166 CONTINUE
1278      DO 169 L=1, 4
1279      169 AA(I-1+L,5)=AA(I-1+L,5)-BB(L,1)
1280      DO 170 L=1, 3
1281      170 AA(I,5+L)=(AA(I,5+L)-BB(1,L+1))/AA(I,5)
1282      IF (I .EQ. (IPP-3)) GO TO 971
1283      DO 171 L=1, 4
1284      171 AA(I,8+L)=AA(I,8+L)/AA(I,5)
1285      971 CONTINUE
1286      DO 172 L=1, 3
1287      172 AA(I+L,6)=AA(I+L,6)-BB(L+1,2)-AA(I+L,5)*AA(I,6)
1288      DO 173 L=1, 2
1289      173 AA(I+1,6+L)=(AA(I+1,6+L)-BB(2,2+L)-AA(I+1,5)*AA(I,6+L))/AA(I+1,6)
1290      IF (I .EQ. (IPP-3)) GO TO 191
1291      DO 174 L=1, 4
1292      AA(I+L+3,2)=AA(I+L+3,2)-AA(I+L+3,1)*AA(I,6)
1293      AA(I+1,8+L)=(AA(I+1,8+L)-AA(I+1,5)*AA(I,8+L))/AA(I+1,6)
1294      174 CONTINUE
1295      191 CONTINUE
1296      DO 175 L=1, 2
1297      175 AA(I+L+1,7)=AA(I+L+1,7)-BB(2+L,3)-AA(I+L+1,5)*AA(I,7)-
1298      1AA(I+L+1,6)*AA(I+1,7)
1299      AA(I+2,8)=(AA(I+2,8)-BB(3,4)-AA(I+2,5)*AA(I,8)-AA(I+2,6)*AA(I+1,8)
1300      2)/AA(I+2,7)
1301      IF (I .EQ. (IPP-3)) GO TO 190
1302      DO 176 L=1, 4
1303      DO 177 LL=1, 2
1304      AA(I+L+3,3)=AA(I+L+3,3)-AA(I+L+3,LL)*AA(I+LL-1,7)
1305      AA(I+2,8+L)=AA(I+2,8+L)-AA(I+2,4+LL)*AA(I+LL-1,8+L)
1306      177 CONTINUE
1307      AA(I+2,8+L)=AA(I+2,8+L)/AA(I+2,7)
1308      176 CONTINUE
1309      190 CONTINUE
1310      AA(I+3,8)=AA(I+3,8)-BB(4,4)-AA(I+3,5)*AA(I,8)-AA(I+3,6)*AA(I+1,8)
1311      1-AA(I+3,7)*AA(I+2,8)
1312      IF (I .EQ. (IPP-3)) GO TO 180
1313      DO 178 L=1, 4
1314      DO 179 LL=1, 3
1315      AA(I+L+3,4)= AA(I+L+3,4)-AA(I+L+3,LL)*AA(I+LL-1,8)
1316      AA(I+3,8+L)= AA(I+3,8+L)-AA(I+3,4+LL)*AA(I+LL-1,8+L)
1317      179 CONTINUE
1318      AA(I+3,8+L)= AA(I+3,8+L)/AA(I+3,8)
1319      178 CONTINUE
1320      180 CONTINUE

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1321      DO 181 L=1,4
1322      DO 182 LL=1,4
1323      IF (ABS(AA(I+L-1,LL)).LT.TMALL) AA(I+L-1,LL)=0.
1324      IF (ABS(F(LL,(I+7)/4-1,J,K+2)) .LT. TMALL)
1325      IF(LL,(I+7)/4-1,J,K+2)=0.
1326      182 F(L,(I+7)/4,J,K+2)=F(L,(I+7)/4,J,K+2)-AA(I+L-1,LL)*F(LL,(I+7)/4-1
1327      1,J,K+2)
1328      LE=L-1
1329      IF (L .EQ. 1) GO TO 184
1330      DO 183 LL=1,LE
1331      F(L,(I+7)/4,J,K+2)=F(L,(I+7)/4,J,K+2)-AA(I+L-1,4+LL)*F(LL,(I+7)/4
1332      1,J,K+2)
1333      183 CONTINUE
1334      184 CONTINUE
1335      F(L,(I+7)/4,J,K+2)=F(L,(I+7)/4,J,K+2)/AA(I+L-1,5+L-1)
1336      181 CONTINUE
1337      165 CONTINUE
1338      W(4,IE,J,K+2)=F(4,IE,J,K+2)
1339      W(3,IE,J,K+2)=F(3,IE,J,K+2)-AA(IPP-1,8)*W(4,IE,J,K+2)
1340      W(2,IE,J,K+2)=F(2,IE,J,K+2)-AA(IPP-2,8)*W(4,IE,J,K+2)
1341      1-AA(IPP-2,7)*W(3,IE,J,K+2)
1342      W(1,IE,J,K+2)=F(1,IE,J,K+2)-AA(IPP-3,8)*W(4,IE,J,K+2)
1343      1-AA(IPP-3,7)*W(3,IE,J,K+2)-AA(IPP-3,6)*W(2,IE,J,K+2)
1344      DO 192 I=5,IPP,4
1345      II=IPP+5-I-4
1346      DO 193 L=1,4
1347      BB(1,L)=0.
1348      DO 194 LL=1,4
1349      IF (ABS(AA(II+1-L,8+LL)) .LT. TMALL) AA(II+1-L,8+LL)=0.
1350      IF (ABS(W(LL,(II+4)/4+1,J,K+2)) .LT. TMALL)
1351      1W(LL,(II+4)/4+1,J,K+2)=0.
1352      194 BB(1,L)=BB(1,L)+AA(II+1-L,8+LL)*W(LL,(II+4)/4+1,J,K+2)
1353      193 CONTINUE
1354      W(4,(II+4)/4,J,K+2)=F(4,(II+4)/4,J,K+2)-BB(1,1)
1355      W(3,(II+4)/4,J,K+2)=F(3,(II+4)/4,J,K+2)-BB(1,2)
1356      1-AA(II-1,8)*W(4,(II+4)/4,J,K+2)
1357      W(2,(II+4)/4,J,K+2)=F(2,(II+4)/4,J,K+2)-BB(1,3)
1358      1-AA(II-2,8)*W(4,(II+4)/4,J,K+2)-AA(II-2,7)*W(3,(II+4)/4,J,K+2)
1359      W(1,(II+4)/4,J,K+2)=F(1,(II+4)/4,J,K+2)-BB(1,4)
1360      1-AA(II-3,8)*W(4,(II+4)/4,J,K+2)-AA(II-3,7)*W(3,(II+4)/4,J,K+2)
1361      2-AA(II-3,6)*W(2,(II+4)/4,J,K+2)
1362      192 CONTINUE
1363      153 CONTINUE
1364      RETURN
1365      END
1366      SUBROUTINE ADI(K,M,DT,DX,DCA,CV,FK,COE,RB,UN,Q,SGO,
1367      1IS,IB,IE,ID,JS,JB,JE,JD,C,IP,JP)
1368      DIMENSION PSI(20,16,2),B1(20,16,2),B2(20,16,2),XI(20,16,2),
1369      1C1(20,16,2),C2(20,16,2),C3(20,16,2),T(20,16,2),
1370      2W(4,20,16,3),F(4,20,16,3),EI(20,16,2),B3(20,16,2),SR(20),
1371      4GA(20,16),GB(20,16),GC(20,16),GD(20,16),CN(20,16,2),
1372      5AA(90,12),BB(4,4),ANG(16),BATB(20)
1373      COMMON PSI,B1,B2,B3,XI,C1,C2,C3,T,W,F,SR,ANG,BATB
1374      1,GA,GB,GC,GD,CN,EI,AA,BB,AJCON
1375      CC
1376      CC---- USING ADI METHOD. SOLVE NEW TIME STEP VALUES PSI, XI, EI(I,J,2)
1377      CC
1378      CC....SOLVE PSI
1379      K=1
1380      DO 770 I=IS,ID

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1381 DO 770 J=JS,JD
1382 F(2,I,J,3)=W(2,I,J,2)
1383 F(3,I,J,3)=W(3,I,J,2)
1384 F(4,I,J,3)=W(4,I,J,2)
1385 W(2,I,J,K+1)=W(2,I,J,K)-AJCON*C1(I,J,K)
1386 W(3,I,J,K+1)=W(3,I,J,K)-AJCON*C2(I,J,K)
1387 W(4,I,J,K+1)=W(4,I,J,K)-AJCON*C3(I,J,K)
1388 W(2,I,J,K)=W(2,I,J,K)-AJCON*C1(I,J,K)
1389 W(3,I,J,K)=W(3,I,J,K)-AJCON*C2(I,J,K)
1390 W(4,I,J,K)=W(4,I,J,K)-AJCON*C3(I,J,K)
1391 770 CONTINUE
1392 DO 772 J=1,JD
1393 W(2,ID,J,2)=-W(2,ID,J,2)-2.0*AJCON*C1(ID,J,1)
1394 W(2,ID,J,1)=W(2,ID,J,2)
1395 772 CONTINUE
1396 IF (MOD(M,2) .EQ. 1) GO TO 11
1397 GO TO 12
1398 11 CONTINUE
1399 DO 14 J=JB,JE
1400 DO 13 I=IB,IE
1401 GC(I,J)=COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K+1)*((SR(I)*DCA)**2)*
1402 1(1./Q+SR(I)*COS(ANG(J)-DCA/2.)))+DT*W(3,I,J,K+1)/W(1,I,J,K+1)
1403 2/(2.*SR(I)*DCA)
1404 GA(I,J)=COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K+1)*((SR(I)*DCA)**2)*
1405 1(1./Q+SR(I)*COS(ANG(J)+DCA/2.)))-DT*W(3,I,J,K+1)/W(1,I,J,K+1)
1406 2/(2.*SR(I)*DCA)
1407 GB(I,J)=1.+COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K+1)*SR(I)*SR(I)
1408 1*DCA*DCA)*(1./(1./Q+SR(I)*COS(ANG(J)+DCA/2.))+1./(1./Q+SR(I)*
1409 2COS(ANG(J)-DCA/2.)))
1410 GD(I,J)=PSI(I,J,K)+COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K)*SR(I)*
1411 1DX*DX)*((SR(I+1)+SR(I))/2.*(PSI(I+1,J,K)-PSI(I,J,K)))/(1./Q+
1412 2(SR(I+1)+SR(I))/2.*COS(ANG(J)))-(SR(I-1)+SR(I))/2.*(PSI(I,J,K)-
1413 3PSI(I-1,J,K)))/(1./Q+(SR(I-1)+SR(I))/2.*COS(ANG(J))))
1414 4-DT*W(2,I,J,K)/W(1,I,J,K)*(PSI(I+1,J,K)-PSI(I-1,J,K))/(2.*DX)
1415 13 CONTINUE
1416 14 CONTINUE
1417 CC USE CROUT REDUCTION
1418 CALL CROUT(M,K,IB,IE,JB,JE,IP,JP,GA,GB,GC,GD,PSI,JD, ID)
1419 GO TO 17
1420 12 CONTINUE
1421 DO 19 J=JB,JE
1422 DO 18 I=IB,IE
1423 GC(I,J)=COE*(1./Q+SR(I)*COS(ANG(J)))*(SR(I-1)+SR(I))/2.
1424 1/(T(I,J,K+1)*SR(I)*DX*DX*(1./Q+(SR(I-1)+SR(I))/2.*COS(ANG(J))))
1425 2+DT*W(2,I,J,K+1)/W(1,I,J,K+1)/(2.*DX)
1426 GA(I,J)=COE*(1./Q+SR(I)*COS(ANG(J)))*(SR(I+1)+SR(I))/2.
1427 1/(T(I,J,K+1)*SR(I)*DX*DX*(1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J))))
1428 2-DT*W(2,I,J,K+1)/W(1,I,J,K+1)/(2.*DX)
1429 GB(I,J)=1.+COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K+1)*SR(I)*DX*DX)
1430 1*((SR(I+1)+SR(I))/2./(1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J))))
1431 2+(SR(I-1)+SR(I))/2./(1./Q+(SR(I-1)+SR(I))/2.*COS(ANG(J))))
1432 GD(I,J)=PSI(I,J,K)+COE*(1./Q+SR(I)*COS(ANG(J)))/(T(I,J,K)*SR(I)*
1433 1SR(I)*DCA*DCA)*((PSI(I,J+1,K)-PSI(I,J,K))/(1./Q+SR(I)*COS(ANG(J))+
1434 2DCA/2.))-(PSI(I,J,K)-PSI(I,J-1,K))/(1./Q+SR(I)*COS(ANG(J)-DCA/2.))
1435 3)-DT*W(3,I,J,K)/W(1,I,J,K)/(SR(I)*2.*DCA)*(PSI(I,J+1,K)-
1436 4PSI(I,J-1,K))
1437 18 CONTINUE
1438 19 CONTINUE
1439 CALL CROUT(M,K,IB,IE,JB,JE,IP,JP,GA,GB,GC,GD,PSI,JD, ID)
1440 17 CONTINUE

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1441      DO 25 I=IS, ID
1442      PSI(I, JE+1, K+1)=PSI(I, JE-1, K+1)
1443      PSI(I, JB-1, K+1)=PSI(I, JB+1, K+1)
1444      25 CONTINUE
1445      DO 27 J=JB, JE
1446      DO 26 I=IB, IE
1447      C3(I, J, K+1)=-(((SR(I+1)+SR(I))/2.*(PSI(I+1, J, K+1)
1448      1-PSI(I, J, K+1)))/(1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J)))-(SR(I-1)
1449      2+SR(I))/2.*(PSI(I, J, K+1)-PSI(I-1, J, K+1))/(1./Q+(SR(I-1)+SR(I))/2.
1450      3*COS(ANG(J))))/(SR(I)*DX*DX)+ ((PSI(I, J+1, K+1)-PSI(I, J, K+1))
1451      4/(1./Q+SR(I)*COS(ANG(J)) + DCA/2. ))-(PSI(I, J, K+1)-
1452      5PSI(I, J-1, K+1))/(1./Q+SR(I)*COS(ANG(J) - DCA/2. ))
1453      6/((SR(I)*DCA)**2))
1454      26 CONTINUE
1455      27 CONTINUE
1456      DO 32 J=JB, JE
1457      DO 31 I=IB, IE
1458      B1(I, J, K+1)=-((PSI(I, J+1, K+1)-PSI(I, J-1, K+1))/(SR(I)*2.*DCA*
1459      1(1./Q+SR(I)*COS(ANG(J))))
1460      B2(I, J, K+1)= (PSI(I+1, J, K+1)-PSI(I-1, J, K+1))/(2.*DX*
1461      1(1./Q+SR(I)*COS(ANG(J))))
1462      31 CONTINUE
1463      32 CONTINUE
1464      CC
1465      CC.... SOLVE XI
1466      IF (MOD(M, 2) .EQ. 1) GO TO 311
1467      GO TO 312
1468      311 CONTINUE
1469      DO 16 J=JB, JE
1470      DO 15 I=IB, IE
1471      GC(I, J)=(COE/(SR(I)*SR(I)*DCA*DCA*(T(I, J-1, K+1)+T(I, J, K+1))/2.*(
1472      11./Q+SR(I)*COS(ANG(J) - DCA/2. )))+DT*W(3, I, J-1, K+1)/
1473      2W(1, I, J-1, K+1)/(2.*SR(I)*DCA*(1./Q+SR(I)*COS(ANG(J-1))))
1474      3*(1./Q+SR(I)*COS(ANG(J)))
1475      GA(I, J)=(COE/(SR(I)*SR(I)*DCA*DCA*(T(I, J+1, K+1)+T(I, J, K+1))/2.*(
1476      11./Q+SR(I)*COS(ANG(J) + DCA/2. )))-DT*W(3, I, J+1, K+1)/
1477      2W(1, I, J+1, K+1)/(2.*SR(I)*DCA*(1./Q+SR(I)*COS(ANG(J+1))))
1478      3*(1./Q+SR(I)*COS(ANG(J)))
1479      GB(I, J)=1.+COE*(1./Q+SR(I)*COS(ANG(J)))/((SR(I)*DCA)**2)*
1480      1(2./((T(I, J+1, K+1)+T(I, J, K+1))*(1./Q+SR(I)*COS(ANG(J) + DCA/2.)))
1481      2+2./((T(I, J-1, K+1)+T(I, J, K+1))*(1./Q+SR(I)*COS(ANG(J) - DCA/2.))))
1482      GD(I, J)=XI(I, J, K)+(COE/(SR(I)*DX*DX)*((SR(I+1)+SR(I))/2.*
1483      1(XI(I+1, J, K)-XI(I, J, K))/((T(I+1, J, K)+T(I, J, K))/2.*(1./Q+(SR(I+1)
1484      2+SR(I))/2.*COS(ANG(J)))) - (SR(I-1)+SR(I))/2.*(XI(I, J, K)-
1485      3XI(I-1, J, K))/((T(I-1, J, K)+T(I, J, K))/2.*(1./Q+(SR(I-1)+SR(I))/2.
1486      4*COS(ANG(J)))) -DT*(SR(I+1)*W(2, I+1, J, K)/W(1, I+1, J, K)*
1487      5XI(I+1, J, K)/(1./Q+SR(I+1)*COS(ANG(J)))-SR(I-1)*W(2, I-1, J, K)/
1488      6W(1, I-1, J, K)*XI(I-1, J, K)/(1./Q+SR(I-1)*COS(ANG(J))))/(2.*SR(I)
1489      7*DX))*((1./Q+SR(I)*COS(ANG(J))) +
1490      7((SR(I+1)*B1(I+1, J, K)*W(4, I+1, J, K)/W(1, I+1, J, K)
1491      8 -SR(I-1)*B1(I-1, J, K)*W(4, I-1, J, K)/W(1, I-1, J, K))/(2.*SR(I)*DX)
1492      9+(B2(I, J+1, K+1)*W(4, I, J+1, K+1)/W(1, I, J+1, K+1)
1493      1 -B2(I, J-1, K+1)*W(4, I, J-1, K+1)/W(1, I, J-1, K+1))/(2.*SR(I)*DCA))
1494      2*DT*(1./Q+SR(I)*COS(ANG(J)))
1495      15 CONTINUE
1496      16 CONTINUE
1497      CALL      CROUT(M, K, IB, IE, JB, JE, IP, JP, GA, GB, GC, GD, XI, JD, ID)
1498      GO TO 317
1499      312 CONTINUE
1500      DO 22 J=JB, JE

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1501      DO 21 I=IB, IE
1502          GC(I, J)=(COE*(SR(I-1)+SR(I))/2./(SR(I)*DX*DX*(T(I-1, J, K+1)+
1503      1T(I, J, K+1))/2.*(1./Q+(SR(I-1)+SR(I))/2.*COS(ANG(J)))) +DT*SR(I-1)
1504      2*W(2, I-1, J, K+1)/W(1, I-1, J, K+1)/(2.*SR(I)*DX*(1./Q+SR(I-1)
1505      3*COS(ANG(J))))*(1./Q+SR(I)*COS(ANG(J)))
1506          GA(I, J)=(COE*(SR(I+1)+SR(I))/2./(SR(I)*DX*DX*(T(I+1, J, K+1)+
1507      1T(I, J, K+1))/2.*(1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J)))) -DT*SR(I+1)
1508      2*W(2, I+1, J, K+1)/W(1, I+1, J, K+1)/(2.*SR(I)*DX*(1./Q+SR(I+1)
1509      3*COS(ANG(J))))*(1./Q+SR(I)*COS(ANG(J)))
1510          GB(I, J)=1.+COE/(SR(I)*DX*DX)*((SR(I+1)+SR(I))/2./((T(I+1, J, K+1)
1511      1+T(I, J, K+1))/2.*(1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J))))+(SR(I-1)+
1512      2SR(I))/2./((T(I-1, J, K+1)+T(I, J, K+1))/2.*(1./Q+(SR(I-1)+SR(I))/2.*
1513      3COS(ANG(J))))*(1./Q+SR(I)*COS(ANG(J)))
1514          GD(I, J)=XI(I, J, K)+(COE/((SR(I)*DCA)**2)*((XI(I, J+1, K)-XI(I, J, K))/
1515      1((T(I, J+1, K)+T(I, J, K))/2.*(1./Q+SR(I)*COS(ANG(J)) + DCA/2. )))
1516      2-(XI(I, J, K)-XI(I, J-1, K))/((T(I, J-1, K)+T(I, J, K))/2.*(1./Q+SR(I)*
1517      3COS(ANG(J) - DCA/2. ))) -DT/(2.*SR(I)*DCA)*(W(3, I, J+1, K)
1518      4/W(1, I, J+1, K)*XI(I, J+1, K)/(1./Q+SR(I)*COS(ANG(J+1))) -
1519      5W(3, I, J-1, K)/W(1, I, J-1, K)*XI(I, J-1, K)/(1./Q+SR(I)*COS(ANG(J-1)))
1520      6*(1./Q+SR(I)*COS(ANG(J))) +
1521      7((SR(I+1)*B1(I+1, J, K+1)*W(4, I+1, J, K+1)/W(1, I+1, J, K+1)
1522      8 -SR(I-1)*B1(I-1, J, K+1)*W(4, I-1, J, K+1)/W(1, I-1, J, K+1))/(2.*SR(I)
1523      9*DX)+(B2(I, J+1, K)*W(4, I, J+1, K)/W(1, I, J+1, K)
1524      1 -B2(I, J-1, K)*W(4, I, J-1, K)/W(1, I, J-1, K))/(2.*SR(I)*DCA))
1525      2*DT*(1./Q+SR(I)*COS(ANG(J)))
1526      21 CONTINUE
1527      22 CONTINUE
1528      CALL      CROUT(M, K, IB, IE, JB, JE, IP, JP, GA, GB, GC, GD, XI, JD, ID)
1529      317 CONTINUE
1530      DO 325 I=IS, ID
1531          XI(I, JE+1, K+1)= XI(I, JE-1, K+1)
1532          XI(I, JB-1, K+1)= XI(I, JB+1, K+1)
1533      325 CONTINUE
1534      DO 332 J=JB, JE
1535      DO 331 I=IB, IE
1536          B3(I, J, K+1)=(      XI(I, J, K+1))/(1./Q+SR(I)*COS(ANG(J)))
1537          C1(I, J, K+1)=-((XI(I, J+1, K+1)-XI(I, J-1, K+1))/(
1538      1(1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DCA)
1539          C2(I, J, K+1)= ((XI(I+1, J, K+1)-XI(I-1, J, K+1))/(
1540      1(1./Q+SR(I)*COS(ANG(J)))*2.*DX)
1541      331 CONTINUE
1542      332 CONTINUE
1543      DO 771 I=IS, ID
1544      DO 771 J=JS, JD
1545          W(2, I, J, K+1)=F(2, I, J, 3)
1546          W(3, I, J, K+1)=F(3, I, J, 3)
1547          W(4, I, J, K+1)=F(4, I, J, 3)
1548          W(2, I, J, K)=F(2, I, J, 3)
1549          W(3, I, J, K)=F(3, I, J, 3)
1550          W(4, I, J, K)=F(4, I, J, 3)
1551      771 CONTINUE
1552      CC
1553      CC.... SOLVE TEMPERATURE EI
1554      IF (MOD(M, 2) .EQ. 1) GO TO 211
1555      GO TO 212
1556      211 CONTINUE
1557      DO 214 J=JB, JE
1558      DO 213 I=IB, IE
1559          GC(I, J)=DT*(1./Q+SR(I)*COS(ANG(J)-DCA/2.))*CN(I, J, K+1)+
1560      1CN(I, J-1, K+1))/(2.*W(1, I, J, K+1)      *(1./Q+SR(I))*

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1561 2COS(ANG(J))*SR(I)*SR(I)*DCA*DCA) +DT*W(3,I,J,K+1)/
1562 3W(1,I,J,K+1)/(2.*SR(I)*DCA)
1563 GA(I,J)=DT*(1./Q+SR(I)*COS(ANG(J)+DCA/2.))*CN(I,J,K+1)+
1564 1CN(I,J+1,K+1)/(2.*W(1,I,J,K+1) *(1./Q+SR(I)*
1565 2COS(ANG(J))*SR(I)*SR(I)*DCA*DCA) -DT*W(3,I,J,K+1)/
1566 3W(1,I,J,K+1)/(2.*SR(I)*DCA)
1567 GB(I,J)=1.+DT/(W(1,I,J,K+1) )*((1./Q+SR(I)*
1568 1COS(ANG(J)+DCA/2.))*CN(I,J,K+1)+CN(I,J+1,K+1))/2.+(1./Q+SR(I)*
1569 2COS(ANG(J)-DCA/2.))*CN(I,J,K+1)+CN(I,J-1,K+1))/2.)/( (1./Q+SR(I)*
1570 3COS(ANG(J))*SR(I)*SR(I)*DCA*DCA) ) +DT*(FK-1.)/2.*
1571 4(((1./Q+SR(I+1)*COS(ANG(J)))*SR(I+1)*W(2,I+1,J,K+1)/
1572 5W(1,I+1,J,K+1)-(1./Q+SR(I-1)*COS(ANG(J)))*SR(I-1)*W(2,I-1,J,
1573 6K+1)/W(1,I-1,J,K+1))/(1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DX) +
1574 7((1./Q+SR(I)*COS(ANG(J+1)))*W(3,I,J+1,K+1)/W(1,I,J+1,K+1)-
1575 8(1./Q+SR(I)*COS(ANG(J-1)))*W(3,I,J-1,K+1)/W(1,I,J-1,K+1))/
1576 9((1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DCA)
1577 PASS=
1578 6+0.5/4.*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1579 7/W(1,I,J,K))* ( EI(I+1,J,K) - EI(I,J,K) ) - ABS(W(2,I,J,K)/W(1,I,J,K)
1580 8-W(2,I-1,J,K)/W(1,I-1,J,K))* ( EI(I,J,K) - EI(I-1,J,K) ) )
1581 6+0.5/4.*BATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1582 7/W(1,I,J,K))* ( EI(I,J+1,K) - EI(I,J,K) ) - ABS(W(3,I,J,K)/W(1,I,J,K)
1583 8-W(3,I,J-1,K)/W(1,I,J-1,K))* ( EI(I,J,K) - EI(I,J-1,K) ) )
1584 GD(I,J)=EI(I,J,K) +DT/(W(1,I,J,K))*
1585 1((1./Q+(SR(I+1)+SR(I))/2.*COS(ANG(J)))*0.5*(SR(I)+SR(I+1))*0.5*
1586 2(CN(I,J,K)+CN(I+1,J,K))* (EI(I+1,J,K)-EI(I,J,K)) -
1587 3(1./Q+(SR(I-1)+SR(I))/2.*COS(ANG(J)))*0.5*(SR(I)+SR(I-1))*0.5*
1588 4(CN(I,J,K)+CN(I-1,J,K))* (EI(I,J,K)-EI(I-1,J,K)))/(1./Q+
1589 5SR(I)*COS(ANG(J)))*SR(I)*DX*DX) -DT*W(2,I,J,K)/W(1,I,J,K)
1590 6*(EI(I+1,J,K)-EI(I-1,J,K))/(2.*DX) -0.5*DT*(FK-1.)*EI(I,J,K)*
1591 4(((1./Q+SR(I+1)*COS(ANG(J)))*SR(I+1)*W(2,I+1,J,K)/W(1,I+1,J,K)
1592 5 -(1./Q+SR(I-1)*COS(ANG(J)))*SR(I-1)*W(2,I-1,J,K)/W(1,I-1,J,K)
1593 6)/(1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DX) +((1./Q+SR(I)*
1594 7COS(ANG(J+1)))*W(3,I,J+1,K)/W(1,I,J+1,K)-(1./Q+SR(I)*
1595 8COS(ANG(J-1)))*W(3,I,J-1,K)/W(1,I,J-1,K))/(1./Q+
1596 9SR(I)*COS(ANG(J)))*SR(I)*2.*DCA)
1597 8+COE*(C1(I,J,K+1)**2+C2(I,J,K+1)**2+C3(I,J,K+1)**2)/
1598 9(T(I,J,K+1)*W(1,I,J,K+1)). +PASS
1599 213 CONTINUE
1600 214 CONTINUE
1601 CALL CROUT(M,K,IB,IE,JB,JE,IP,JP,GA,GB,GC,GD, EI,JD, ID)
1602 GO TO 217
1603 212 CONTINUE
1604 DO 235 J=JB,JE
1605 DO 234 I=IB,IE
1606 GC(I,J)=DT*(1./Q+(SR(I)+SR(I-1))/2.*COS(ANG(J)))*0.5*(SR(I)+
1607 1SR(I-1))*0.5*(CN(I,J,K+1)+CN(I-1,J,K+1))/W(1,I,J,K+1) *
1608 2(1./Q+SR(I)*COS(ANG(J)))*SR(I)*DX*DX) +DT*W(2,I,J,K+1)/
1609 3W(1,I,J,K+1)/(2.*DX)
1610 GA(I,J)=DT*(1./Q+(SR(I)+SR(I+1))/2.*COS(ANG(J)))*0.5*(SR(I)+
1611 1SR(I+1))*0.5*(CN(I,J,K+1)+CN(I+1,J,K+1))/W(1,I,J,K+1) *
1612 2(1./Q+SR(I)*COS(ANG(J)))*SR(I)*DX*DX) -DT*W(2,I,J,K+1)/
1613 3W(1,I,J,K+1)/(2.*DX)
1614 GB(I,J)=DT/(W(1,I,J,K+1) )*((1./Q+(SR(I)+SR(I+1))/2.*
1615 1COS(ANG(J)))*0.5*(SR(I)+SR(I+1))*0.5*(CN(I,J,K+1)+CN(I+1,J,K+1))
1616 2+(1./Q+(SR(I)+SR(I-1))/2.*COS(ANG(J)))*0.5*(SR(I)+SR(I-1))*
1617 30.5*(CN(I,J,K+1)+CN(I-1,J,K+1)))/(1./Q+SR(I)*COS(ANG(J)))*SR(I)
1618 1*DX*DX) +1. + DT*(FK-1.)/2. *
1619 4(((1./Q+SR(I+1)*COS(ANG(J)))*SR(I+1)*W(2,I+1,J,K+1)/
1620 5W(1,I+1,J,K+1)-(1./Q+SR(I-1)*COS(ANG(J)))*SR(I-1)*W(2,I-1,J,

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1621      6K+1)/W(1,I-1,J,K+1))/((1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DX) +
1622      7((1./Q+SR(I)*COS(ANG(J+1)))*W(3,I,J+1,K+1)/W(1,I,J+1,K+1)-
1623      8(1./Q+SR(I)*COS(ANG(J-1)))*W(3,I,J-1,K+1)/W(1,I,J-1,K+1))/
1624      9((1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DCA))
1625      PASS=
1626      6+0.5/4.*0.5*DT/DX*(ABS(W(2,I+1,J,K)/W(1,I+1,J,K)-W(2,I,J,K)
1627      7/W(1,I,J,K))* (EI(I+1,J,K)-EI(I,J,K))-ABS(W(2,I,J,K)/W(1,I,J,K)
1628      8-W(2,I-1,J,K)/W(1,I-1,J,K))* (EI(I,J,K)-EI(I-1,J,K)))
1629      6+0.5/4.*8ATB(I)*(ABS(W(3,I,J+1,K)/W(1,I,J+1,K)-W(3,I,J,K)
1630      7/W(1,I,J,K))* (EI(I,J+1,K)-EI(I,J,K))-ABS(W(3,I,J,K)/W(1,I,J,K)
1631      8-W(3,I,J-1,K)/W(1,I,J-1,K))* (EI(I,J,K)-EI(I,J-1,K)))
1632      GD(I,J)=EI(I,J,K) +DT/(W(1,I,J,K))*(((1./Q+SR(I)*
1633      1COS(ANG(J)+DCA/2.))*0.5*(CN(I,J,K)+CN(I,J+1,K))* (EI(I,J+1,K)-
1634      2EI(I,J,K))-(1./Q+SR(I)*COS(ANG(J)-DCA/2.))*0.5*(CN(I,J,K)+
1635      3CN(I,J-1,K))* (EI(I,J,K)-EI(I,J-1,K))))/((1./Q+SR(I)*COS(ANG(J)))*
1636      4SR(I)*SR(I)*DCA*DCA))-DT*W(3,I,J,K)/W(1,I,J,K)*(EI(I,J+1,K)-
1637      5EI(I,J-1,K))/(2.*SR(I)*DCA)-DT*(FK-1.)*EI(I,J,K)*0.5*
1638      4((1./Q+SR(I+1)*COS(ANG(J)))*SR(I+1)*W(2,I+1,J,K)/W(1,I+1,J,K)
1639      5 -(1./Q+SR(I-1)*COS(ANG(J)))*SR(I-1)*W(2,I-1,J,K)/W(1,I-1,J,K)
1640      6)/((1./Q+SR(I)*COS(ANG(J)))*SR(I)*2.*DX) +((1./Q+SR(I)*
1641      7COS(ANG(J+1)))*W(3,I,J+1,K)/W(1,I,J+1,K)-(1./Q+SR(I)*
1642      8COS(ANG(J-1)))*W(3,I,J-1,K)/W(1,I,J-1,K))/((1./Q+
1643      9SR(I)*COS(ANG(J)))*SR(I)*2.*DCA))
1644      8+COE*(C1(I,J,K+1)**2+C2(I,J,K+1)**2+C3(I,J,K+1)**2)/
1645      9(T(I,J,K+1)*W(1,I,J,K+1)) +PASS
1646      234 CONTINUE
1647      235 CONTINUE
1648      CALL      CROUT(M,K,IB,IE,JB,JE,IP,JP,GA,GB,GC,GD,EI,JD,ID)
1649      217 CONTINUE
1650      DO 225 I=IS,ID
1651      EI(I,JE+1,K+1)=EI(I,JE-1,K+1)
1652      EI(I,JB-1,K+1)=EI(I,JB+1,K+1)
1653      225 CONTINUE
1654      RETURN
1655      END
1656      SUBROUTINE CROUT(M,K,IB,IE,JB,JE,IP,JP,GA,GB,GC,GD,PSI,JD,ID)
1657      CC
1658      CC-----CROUT REDUCTION METHOD -----
1659      DIMENSION AA(90,3),E(90),X(90),GA(20,16),GB(20,16),GC(20,16),
1660      1GD(20,16),PSI(20,16,2)
1661      SMALL=1.0E-70
1662      IF (MOD(M,2) .EQ. 1) GO TO 3
1663      GO TO 2
1664      3 CONTINUE
1665      JS=1
1666      DO 123 I=IB,IE
1667      DO 122 J=JS,JP
1668      AA(J,2)=GB(I,J+1)
1669      IF (J .EQ. JP) GO TO 124
1670      IF (J .EQ. JS) GO TO 125
1671      AA(J,3)=-GA(I,J+1)
1672      GO TO 125
1673      124 AA(JP,1)=- (GC(I,JP+1)+GA(I,JP+1))
1674      125 CONTINUE
1675      IF (J .EQ. JS) GO TO 126
1676      IF (J .EQ. JP) GO TO 127
1677      AA(J,1)=-GC(I,J+1)
1678      GO TO 127
1679      126 AA(JS,3)=- (GA(I,JS+1)+GC(I,JS+1))
1680      127 CONTINUE

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1681 E(J)=GD(I,J+1)
1682 122 CONTINUE
1683 AA(JS,3 )=AA(JS,3 )/AA(JS,2 )
1684 E(JS)=E(JS)/AA(JS,2 )
1685 DO 130 J=JB,JP
1686 IF (ABS(AA(J, 1)) .LE. SMALL) AA(J, 1)=0.
1687 IF (ABS(AA(J-1,3)) .LE. SMALL) AA(J-1,3)=0.
1688 AA(J,2)=AA(J,2)-AA(J,1 )*AA(J-1,3)
1689 IF (ABS(E(J-1)) .LE. SMALL) E(J-1)=0.
1690 E(J)=(E(J)-AA(J, 1)*E(J-1))/AA(J,2)
1691 IF (J .EQ. JP) GO TO 131
1692 AA(J,3 )=AA(J,3 )/AA(J,2)
1693 131 CONTINUE
1694 130 CONTINUE
1695 X(JP)=E(JP)
1696 JQ=JP-1
1697 DO 133 J=JS,JQ
1698 K=JQ+JS-J
1699 IF (ABS(AA(K,3 )) .LE. SMALL) AA(K,3 )=0.
1700 X(K)=E(K)-AA(K,3 )*X(K+1)
1701 133 CONTINUE
1702 DO 128 J=JS,JP
1703 PSI(I,J+1,2)=X(J)
1704 IF (ABS(PSI(I,J+1,2)) .LT. SMALL) PSI(I,J+1,2)=0.
1705 128 CONTINUE
1706 123 CONTINUE
1707 GO TO 4
1708 2 CONTINUE
1709 IS=1
1710 DO 23 J=JB,JE
1711 DO 22 I=IS,IP
1712 E(I)=GD(I+1,J)
1713 AA(I,2)=GB(I+1,J)
1714 IF (I .EQ. IP) GO TO 24
1715 AA(I,3 )=-GA(I+1,J)
1716 GO TO 25
1717 24 E(IP)=GD(IP+1,J)+GA(IP+1,J)*PSI(IP+2,J,2)
1718 25 CONTINUE
1719 IF (I .EQ. IS) GO TO 26
1720 AA(I, 1)=-GC(I+1,J)
1721 GO TO 27
1722 26 CONTINUE
1723 E(IS)=GD(IS+1,J)+GC(IS+1,J)*PSI(IS,J,2)
1724 27 CONTINUE
1725 22 CONTINUE
1726 AA(IS,3 )=AA(IS,3 )/AA(IS,2 )
1727 E(IS)=E(IS)/AA(IS,2 )
1728 DO 30 I=IB,IP
1729 IF (ABS(AA(I, 1)) .LE. SMALL) AA(I, 1)=0.
1730 IF (ABS(AA(I-1,3)) .LE. SMALL) AA(I-1,3)=0.
1731 AA(I,2)=AA(I,2)-AA(I, 1)*AA(I-1,3)
1732 IF (ABS(E(I-1)) .LE. SMALL) E(I-1)=0.
1733 E(I)=(E(I)-AA(I, 1)*E(I-1))/AA(I,2)
1734 IF (I .EQ. IP) GO TO 30
1735 AA(I,3 )=AA(I,3 )/AA(I,2)
1736 30 CONTINUE
1737 X(IP)=E(IP)
1738 IQ=IP-1
1739 DO 33 I=IS,IQ
1740 K=IQ+IS-I

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1741       IF (ABS(AA(K,3 )) .LE. SMALL) AA(K,3 )=0.
1742       X(K)=E(K)-AA(K,3 )*X(K+1)
1743   33 CONTINUE
1744       DO 28 I=IS,IP
1745       PSI(I+1,J,2)=X(I)
1746       IF (ABS(PSI(I+1,J,2)) .LT. SMALL) PSI(I+1,J,2)=0.
1747   28 CONTINUE
1748   23 CONTINUE
1749       4 CONTINUE
1750       K=1
1751       RETURN
1752       END
1753       SUBROUTINE AVERA(SEQ,AVV)
1754   CC
1755   CC....AVERAGING IN THETA DIRECTION
1756       DIMENSION SEQ(16)
1757       COMMON /XY/NUG,DX,JE,JB,PI ,IE,THJ
1758       AVV=0.0
1759       JBB=JB+1
1760       JEE=JE-1
1761       DO 1 J=JBB ,JEE
1762       AVV=AVV+SEQ(J)
1763   1 CONTINUE
1764       AVV=2.0*AVV+SEQ(JB)+SEQ(JE)
1765       AVV=2.0* AVV/(NUG*DX)
1766       RETURN
1767       END
1768       SUBROUTINE SURIN(A,B,RB,RSMA)
1769   CC
1770   CC....SURFACE INTEGRATION (ON THE WALL)
1771       DIMENSION A(16),THJ(16)
1772       COMMON /XY/NUG,DX,JE,JB,PI,IE,THJ
1773       A(JE)=A(JE)/2.
1774       A(JB)=A(JB)/2.
1775       B=0.0
1776       DO 1 J=JB,JE
1777       WUG=2.*PI*RSMA*(IE-0.5)/IE /NUG
1778       B=B+A(J)*2.*PI*(RB+RSMA*(IE-0.5)/IE*COS(THJ(J)))*WUG
1779   1 CONTINUE
1780       B=2.*B/2.0/PI/RB*100.0
1781       RETURN
1782       END
1783       SUBROUTINE VOLIN(A,B,RB,RSMA)
1784   CC
1785   CC....VOLUME INTEGRATION
1786       DIMENSION A(20,16),THJ(16)
1787       COMMON /XY/NUG,DX,JE,JB,PI,IE,THJ
1788       B=0.0
1789       IDD=IE
1790       DO 1 I=1,IDD
1791       ARA=PI*RSMA*RSMA*DX*DX*((I-0.5)**2-(I-1.5)**2)/NUG
1792       IF (I.EQ.1) ARA=PI*RSMA*RSMA*DX*DX*(I-0.5)**2/NUG
1793       A(I,JE)=A(I,JE)/2.
1794       A(I,JB)=A(I,JB)/2.
1795       DO 1 J=JB,JE
1796       B=B+A(I,J)*2.*PI*(RB+RSMA*DX*(I-1)*COS(THJ(J)))*ARA
1797   1 CONTINUE
1798       B=2.*B/2.0/PI/RB*100.0
1799       RETURN
1800       END

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1801 SUBROUTINE ARAIN(A,B,RB,RSMA)
1802 CC
1803 CC... AREA INTEGRATION (ON POLOIDAL PLANE)
1804 DIMENSION A(20,16),THJ(16)
1805 COMMON /XY/NUG,DX,JE,JB,PI,IE,THJ
1806 B=0.0
1807 IDD=IE
1808 DO 1 I=1,IDD
1809 ARA=PI*RSMA*RSMA*DX*DX*((I-0.5)**2-(I-1.5)**2)/NUG
1810 IF(I.EQ.1)ARA=PI*RSMA*RSMA*DX*DX*(I-0.5)**2/NUG
1811 A(I,JE)=A(I,JE)/2.
1812 A(I,JB)=A(I,JB)/2.
1813 DO 1 J=JB,JE
1814 B=B+A(I,J)*ARA
1815 1 CONTINUE
1816 DO 2 I=1,IDD
1817 A(I,JE)=A(I,JE)*2.
1818 A(I,JB)=A(I,JB)*2.
1819 2 CONTINUE
1820 B=2.0*B
1821 RETURN
1822 END
1823
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