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**OAK
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**Steady-State Axial Pressure Losses
Along the Exterior of Deformed
Fuel Cladding: Multirod Burst
Test (MRBT) Bundles B-1 and B-2**

J. F. Mincey

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Prepared for the U.S. Nuclear Regulatory Commission
Office of Nuclear Regulatory Research
Under Interagency Agreements DOE 40-551-75 and 40-552-75

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Printed in the United States of America. Available from
National Technical Information Service
U.S. Department of Commerce
5285 Port Royal Road, Springfield, Virginia 22161

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NUREG/CR-1011
ORNL/NUREG/TM-350
Dist. Category R3

Contract No. W-7405-eng-26

Engineering Technology Division

STEADY-STATE AXIAL PRESSURE LOSSES ALONG THE EXTERIOR
OF DEFORMED FUEL CLADDING: MULTIROD BURST
TEST (MRBT) BUNDLES B-1 AND B-2

J. F. Mincey

Manuscript Completed -- December 20, 1979
Date Published -- January 1980

NOTICE: This document contains information of a preliminary nature. It is subject to revision or correction and therefore does not represent a final report.

Prepared for the
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NRC FIN No. B0120

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FOREWORD

This report summarizes results and analyses of flow characterization tests conducted with Multirod Burst Test Program (MRBT) bundles B-1 and B-2. Summary results of these tests have been reported routinely in the MRBT progress reports. While this procedure disseminates the information as it is developed, there are drawbacks in that the results and any revisions made after further analysis are scattered among several reports. In order to alleviate these shortcomings and because these results are believed to be of sufficient importance to warrant it, they are being published in a single document.

The MRBT Program at Oak Ridge National Laboratory is sponsored by the Division of Reactor Safety Research of the Nuclear Regulatory Commission. Progress reports issued by this program are listed below:

<u>NUREG Report No.</u>	<u>ORNL Report No.</u>	<u>Period Covered</u>
	ORNL/TM-4729	July-September 1974
	ORNL/TM-4805	October-December 1974
	ORNL/TM-4914	January-March 1975
	ORNL/TM-5021	April-June 1975
	ORNL/TM-5514	July-September 1975
	ORNL/NUREG/TM-10	October-December 1975
	ORNL/NUREG/TM-36	January-March 1976
	ORNL/NUREG/TM-74	April-June 1976
	ORNL/NUREG/TM-77	July-September 1976
	ORNL/NUREG/TM-95	October-December 1976
	ORNL/NUREG/TM-108	January-March 1977
	ORNL/NUREG/TM-135	April-June 1977
NUREG/CR-0103	ORNL/NUREG/TM-200	July-December 1977
NUREG/CR-0225	ORNL/NUREG/TM-217	January-March 1978
NUREG/CR-0398	ORNL/NUREG/TM-243	April-June 1978
NUREG/CR-0655	ORNL/NUREG/TM-297	July-December 1978
NUREG/CR-0817	ORNL/NUREG/TM-323	January-March 1979
NUREG/CR-1023	ORNL/NUREG/TM-351	April-June 1979

Topical reports pertaining to research and development carried out as a part of this program are:

1. R. H. Chapman (Comp.), *Characterization of Zircaloy-4 Tubing Procured for Fuel Cladding Research Programs*, ORNL/NUREG/TM-29 (July 1976).

2. W. E. Baucum and R. E. Dial, *An Apparatus for Spot Welding Sheathed Thermocouples to the Inside of Small-Diameter Tubes at Precise Locations*, ORNL/NUREG/TM-33 (August 1976).
3. W. A. Simpson, Jr., et al., *Infrared Inspection and Characterization of Fuel Pin Simulators*, ORNL/NUREG/TM-55 (November 1976).
4. R. H. Chapman et al., *Effect of Creep Time and Heating Rate on Deformation of Zircaloy-4 Tubes Tested in Steam with Internal Heaters*, NUREG/CR-0343 (ORNL/NUREG/TM-245) (October 1978).

The following limited-distribution quick-look and data reports have been issued by this program:

1. R. H. Chapman, (Comp.), *Quick-look Report on MRBT No. 1 4 x 4 Bundle Burst Test*, Internal Report ORNL/MRBT-2 (September 1977).
2. R. H. Chapman, (Comp.), *Quick-look Report on MRBT No. 2 4 x 4 Bundle Burst Test*, Internal Report ORNL/MRBT-3 (November 1977).
3. R. H. Chapman, *Quick-look Report on MRBT No. 3 4 x 4 Bundle Burst Test*, Internal Report ORNL/MRBT-4 (August 1978).
4. R. H. Chapman et al., *Bundle B-1 Test Data*, ORNL/NUREG/TM-322 (June 1979).
5. R. H. Chapman et al., *Bundle B-2 Test Data*, ORNL/NUREG/TM-337 (August 1979).

STEADY-STATE AXIAL PRESSURE LOSSES ALONG THE EXTERIOR
OF DEFORMED FUEL CLADDING: MULTIROD BURST
TEST (MRBT) BUNDLES B-1 AND B-2

J. F. Mincey

ABSTRACT

The experimental and COBRA-IV computational data presented in this report confirm that increased pressure losses, induced by the steady-state axial flow of water exterior to deformed Multirod Burst Test (MRBT) bundles B-1 and B-2, may be closely predicted using a bundle-averaged approach for describing flow channel restrictions. One anomaly that was encountered using this technique occurred while modeling the B-2 flow test data near a severe channel restriction: the COBRA-IV results tended to underestimate experimental pressure losses.

Keywords: Hydraulics, pressure losses, reactor fuel, cladding, experimental data, computer modeling.

1. INTRODUCTION

The general guidelines¹ for the Multirod Burst Test (MRBT) Program allude to a series of flow tests to experimentally determine pressure losses induced by post-burst-test cladding deformation. Accordingly, the following objectives were established for the MRBT flow test effort:

- Provide an experimental data base in which the pressure losses of a burst bundle are compared with those of an undeformed bundle under similar hydraulic conditions.
- Determine the geometrical details necessary for modeling these experimental data with COBRA-IV (ref. 2).

Inasmuch as additional MRBT bundle tests are in progress at this writing, this report will concentrate on B-1 and B-2 bundle data in meeting these objectives.

Since the MRBT Program is primarily concerned with the cladding deformation associated with a loss-of-coolant accident (LOCA) and not with a detailed investigation of the effect of such deformation on the actual thermal-hydraulic performance of the bundles, it was considered outside the scope of the program to construct a flow facility designed to produce

thermal-hydraulic conditions typical of a LOCA. Instead, it was judged sufficient to characterize the flow behavior of both deformed and undeformed bundles in a flow facility used for another research program. Hence, the single-phase, steady-state water flow conditions were dictated in part by the available equipment rather than by technical considerations. These restraints had two detrimental effects on the data to be presented in this report. First, the quality and calibration of the available equipment were limited because the equipment was not available to the MRBT Program continuously. Second, the flow test data are limited in range and are directly applicable to only a very small class of postulated LOCA conditions.

However, with the demonstration of COBRA-IV's ability to predict measured pressure losses to within experimental uncertainty, a computational technique has been identified which may be validated against more applicable, non-MRBT experimental data in analyzing other categories of hydraulic systems. The intent of this report, then, is to document the uncertainties in the experimental data with which COBRA-IV results are compared, to describe how the MRBT COBRA-IV models were developed, and to compare COBRA-IV predictions with measured pressure losses.

2. EXPERIMENTAL FLOW FACILITY

The experimental flow facility is a closed flow loop circulating room-temperature (20 to 35°C) water at pressures less than 300 kPa above atmospheric. Water flow along the bundle axes is directed vertically upwards toward the flanged end of the burst bundles (i.e., opposite the direction of steam flow during the burst tests). The flow test facility, which is entirely independent of the burst test facilities, is shown in Fig. 2.1, with a view of the MRBT test stand in the foreground; the flow loop piping is in the background.

The test stand was designed in such a way that it could be quickly disconnected from the main flow loop piping. In fact, the main obstacle encountered in converting the loop back to its non-MRBT functions was associated with instrumentation modifications. The instrumentation used for these tests consisted of three pressure transducers and associated electronics, a turbine meter (with its electronics) downstream from the test stand, and a magnetic flow meter upstream of the test stand. The test stand and instrumentation of the test facility are examined in detail in the remaining sections of this chapter.

2.1 MRBT Flow Test Stand

For discussional purposes, the test stand will be broken down into eight types of components, seven of which are shown in Fig. 2.1: the lower transition section (A), the lower flow shroud (B), the bundle extension (C), the manifolds and connecting tubing (D), the upper flow shroud (E), the upper transition section (F), and the jackscrew retaining mechanism (G). These seven items are also illustrated in Figs. 2.2 to 2.10. The eighth item, the reference bundle, is shown in Fig. 2.11. These eight items, along with the support structure, constitute the MRBT flow test stand; all other items necessary for the flow tests were part of the flow loop. The MRBT equipment is described in detail below, starting with item A in Figs. 2.1 and 2.2 and then following the upward water flow direction.

Water enters the test stand via a tee in the 10.16-cm-ID polyvinyl chloride (PVC) flow loop piping. The flow first encounters the lower transition section (item A in Figs. 2.1 and 2.2). One function of this transition section is to provide a structural support (visible on the right side of Fig. 2.2) for the bundle extensions (item C). This support is a 1.9-cm-thick Plexiglas disk with holes to accommodate the bundle extensions and to permit water flow. The support represents a 45% reduction in flow area compared with the flow area of the 10.16-cm-ID flow loop piping. The main function of the lower transition section is to provide a gradual hydraulic transition downstream of the structural support and into a rectangular flow channel with a bundle in the middle. This transition is achieved by means of four sloping inserts fastened to the inside of the lower transition section, which form a rectangular flow channel about the centrally located bundle extension.

The flow next enters one of two lower flow shrouds (Figs. 2.2 and 2.3; item B in both figures), which have channel dimensions appropriate for the two different upper shrouds (housing the burst bundles) used in the flow tests. Notice that four pressure taps are located at the same positions on both lower shrouds (two taps on each side). The two rightmost taps are used for pressure references, since measurements on the corresponding shroud sides are made directly against them. There are a few exceptions: the B-1/shroud 1* and some reference bundle/shroud 1* data were measured relative to atmospheric pressure.

These lower shrouds are 30.5 cm in length and were constructed from 1.9-cm-thick Plexiglas. They are bolted to the lower transition section with a neoprene gasket (having a cutout matching the rectangular flow channel) trapped between the two flanges. The bundle extension also passes through these lower shrouds and has a grid spacer (not shown on the data plots) roughly centered at the joint between the lower shroud and transition section. Adapters are fastened to the upper (left) end of the bundle extension to fit inside the burst and reference bundles. These adapters were centerless-ground to a diameter of 1.092 ± 0.005 cm

*In this context the diagonal (or solidus) means "in" — that is, "B-1/shroud 1" means "B-1 in shroud 1" and "reference bundle/shroud 1" means "reference bundle in shroud 1." This construction is used throughout this report.

to match the outside diameter of the Zircaloy cladding used to construct MRBT bundles and the bundle extension. The Inconel grid spacer is also from standard MRBT stock.

Summarizing, the flow enters the test stand through the support plate and passes into the lower transition section where the flow channel gradually becomes rectangular about a central bundle extension. The flow then enters the extension bundle/lower flow shroud assembly, which serves as a hydraulic entrance region.

Next, the water enters upper flow shrouds 1 and 2, shown in Figs. 2.4 and 2.5, respectively. Pressure taps are located on two sides of each shroud, the tap numbers decreasing in the flow direction. The shrouds are built in such a fashion that the sides with odd-number pressure taps are removable for easy insertion of the bundle for flow testing. The interface between the actual bundle and the bundle extension occurs roughly at the corresponding interface between the upper and lower shrouds. The differences between shrouds 1 and 2 are discussed next.

Design calculations indicated and experimental data confirmed that trade-offs exist pertaining to how much bypass flow area should be allowed between a bundle's periphery and the flow shroud walls. When axial pressure loss profiles for the burst bundle are compared with those for the reference bundle, the pressure loss differences between the two data sets should be made as large as possible to minimize uncertainties resulting from the precision limits of the data (this topic is discussed in depth in Chap. 3). This consideration favors a small bypass flow area. Hence, shroud 1 was originally built with a 6.35-cm by 6.35-cm channel, representing the smallest area the then unburst B-1 bundle could reasonably be construed to fit inside. However, after the B-1 burst test, flare-outs at the cladding rupture points required that three relief grooves be milled into the walls of shroud 1 and a thicker sealing gasket used to increase its flow channel area. A schematic of the modified shroud 1 flow channel, with the orientation of the B-1 and reference bundle tubes, is shown in Fig. 2.6. Figure 2.7 depicts the orientation of B-2 in shroud 1; Fig. 2.8 is a photograph of the three relief grooves in shroud 1.

It was recognized that shroud 1 might be too small to accommodate future burst bundles, and for this reason a second shroud with a larger flow channel was designed (shroud 2, shown in Figs. 2.5 and 2.9). The construction of shroud 2 was completed in time for the B-2 tests; hence, a direct comparison is possible between the sets of flow test data for the reference and B-2 bundles in both flow shrouds. Although B-2 was flow tested in shroud 2 without any shroud modifications being necessary, the resolution of increased B-2 pressure losses over those of the reference bundle is not as precise as for the comparable shroud 1 data. This undesirable feature is the result of pressure losses having the expected inverse square dependence on the flow channel area. Hence, there is a trade-off between fitting the burst bundles easily into the shrouds and minimizing the bypass flow (keeping the shrouds close fitting enough to ensure large pressure losses). If the pressure losses are not kept large, precision limitations on the data (due to instrumentation limits) can represent an intolerably large percentage of uncertainty.

A second trade-off also exists. The pressure taps cannot distinguish between reductions in the system pressure due to piezometric losses (the effects of friction, drag, and flow area lumped together for convenience into one term) and pressure reductions caused by diverted water flowing in a nonperpendicular fashion relative to the taps. Thus, it is imperative that the water velocity be parallel to the shroud walls (perpendicular to the pressure taps) so that the taps will be sensitive only to piezometric pressure effects. Otherwise, a mixture of piezometric and diverted-flow-induced pressure effects will be measured, which makes the mathematical modeling of such data extremely difficult. Hence, a significant bypass area between deformed bundles and the shroud walls ensures that any deformation-induced flow disturbances are dissipated and the flow is redirected parallel to the walls and is not impinging toward, or away from, the pressure taps.

The pressure tap holes for both shrouds were 0.159 cm in diameter through the 1.9-cm-thick shroud walls. Every effort was made to ensure that these tap holes were perpendicular to the shroud walls and that no rounded corners or burrs were present. The tap fittings were of the nylon Swagelok variety and were sealed to the shroud wall. Eighty taps

were installed on the upper and lower shrouds — 40 taps on each side. These taps are located at matching locations on opposite walls for both shrouds; the axial spacing between taps for shroud 1 and shroud 2 differ slightly, however, due to improvements in the shroud 2 design.

Manifolds (item D in Fig. 2.1) were constructed to feed PVC pressure tap lines into the pressure transducer ports via valves. With this arrangement it is possible to directly measure the pressure "seen" by any tap against that "seen" by another tap. Further, no static pressure head corrections were made, since both transducer pressure ports were always under the same effective column height of water no matter which two manifold valves were opened. This column is essentially the height of water in the upper transition section (F in Fig. 2.1) above the transducer ports, routed through the shroud, shroud taps, and into the two manifolds; hence, the same column height is always "seen" from both transducer ports because it is independent of manifold valve operation. The flow test procedures used in conducting these tests have been documented and are available in the MRBT files.

After leaving the upper flow shroud, the water flow is directed back into the flow loop piping via the upper transition section (see Fig. 2.10). This fitting is designed with a raised ring to seal against the MRBT bundle flange, the bundle passing through the central hole. One of the fitting's two arms is sealed with a blank flange, the discharge being directed through the other arm back into the loop. The bundle and upper transition assembly bolts onto the upper flow shrouds.

The jackscrew retaining mechanism (item G) may be seen in Fig. 2.1. This mechanism serves to prevent any upward shifting of the burst bundle tubes during the flow tests. Also, an adapter on the jackscrews provides a waterproof seal to protect against any water flowing up through the bursts in the tubes.

Figure 2.11 is a photograph of the reference bundle. The bundle rods were constructed of stainless steel rods centerless-ground to a 1.092 ± 0.005 -cm diameter. They are mounted in a Plexiglas flange, and the spacing is maintained by three standard MRBT spacer grids to a 1.443 ± 0.010 -cm square pitch. The placement of these grids matches the posttest B-1 and B-2 locations.

2.2 Differential Pressure Transducers

The following three transducers were used:

1. 0- to 25-kPa strain-gage transducer (BLH Electronics, Model No. 1100)
2. 0- to 75-kPa strain-gage transducer (BLH Electronics, Model No. 1300)
3. 0- to 200-kPa strain-gage transducer (BLH Electronics, Model No. 1800)

Endevco signal-conditioning and voltage-regulated bridge conditioner units (Models 4470 and 4471) were used in conjunction with these transducers. The transducer outputs were displayed on a Hewlett-Packard integrating digital voltmeter (Model 2401C), manually integrated over a 10-sec interval, and then hand recorded. The millivolt readings were then converted to kilopascal (kPa) units with transducer calibration curves and plotted via a computer program.

The transducers were checked against a water manometer before and after the flow tests for four or five reference pressures to ensure the integrity of their calibration curves. Data typical of these checks are given in Table 2.1. The upper pressure ranges of transducers 2 and 3 were not checked during the actual flow testing because they exceeded the maximum range of the manometer. However, recent full-range checks of these two transducers have confirmed the linearity of their pressure responses, as noted in Table 2.1. Pressure measurements made with the system herein described fall within a precision of about $\pm 5\%$.

2.3 Turbine Meter

As stated earlier, a turbine meter (when operational and available) was located in the discharge line downstream of the MRBT test stand. A Flow Technology rotor and flow rate monitor (Model PRI-102FR), along with the previously described digital voltmeter, comprised the turbine meter instrumentation.

This assembly was checked against the upstream magnetic flow meter before and after each flow test to provide a precision estimate. Hence,

Table 2.1. Results from pressure transducer calibration checks

Numerical error limits represent one standard deviation of data used in the checks; percentage limits represent precision estimates

Flow or calibration test description	BLH Electronics HLD transducer conversion factors (kPa/mV)		
	Model 1100	Model 1300	Model 1800
B-1 in shroud 1	Unavailable	2.082 ± 0.007	5.378 ± 0.069
Reference bundle in shroud 1	Unavailable	2.082 ± 0.007	5.378 ± 0.069
B-2 in shroud 1	Unavailable	2.068 ± 0.007	5.543 ± 0.021
B-2 in shroud 1	0.807 ± 0.021 ^a	2.420 ± 0.014 ^a	6.385 ± 0.034 ^a
Reference bundle in shroud 1	0.696 ± 0.007	2.082 ± 0.014	5.495 ± 0.034
Grid test 1 (reference bundle in shroud 1)	0.703 ± 0.007	2.089 ± 0.007	5.730 ± 0.048
Grid test 2 (reference bundle in shroud 2)	0.696 ± 0.007	2.096 ± 0.007	5.736 ± 0.076
Calibration tests	0.710 ± 0.007	2.110 ± 0.007	5.654 ± 0.028
Average of above data	0.699 ± 3%	2.087 ± 2%	5.551 ± 4%

^aThese values are larger because a temporary circuit modification was used during this flow test; to make the values directly comparable, multiply by 0.861.

the turbine meter was primarily used as a precision check on the volumetric flow rate measurements, the magnetic flow meter being used to determine absolute magnitudes.

2.4 Magnetic Flow Meter

The determination of absolute volumetric flow rate constitutes the least-explored facet of the experimental data. Due to the fact that instrumentation for these tests was borrowed for a limited time only (as

noted earlier), a Foxboro 9600 series magnetic flow meter was used to determine absolute volumetric flow rates without any calibration checks being conducted by MRBT personnel.

Precision estimates were made against the turbine meter, however, and were found to be in reasonable agreement with the $\pm 1\%$ precision limit at full scale (31.5 liters/sec) listed by the magnetic flow meter manufacturer. Over the volumetric flow rate range used for the flow tests, the precision limit, then, ranges from $\pm 5\%$ at 6.3 liters/sec to $\pm 2\%$ at 25.0 liters/sec. As pressure losses depend roughly on the square of the volumetric flow rate, these percentages reflect uncertainties of about $\pm 10\%$ (at lower flow rates) to $\pm 5\%$ (at higher flow rates) by which calculated and measured pressure profiles could conceivably differ.

Consequently, while reported volumetric flow rates and Reynolds number are very important to the reproducibility of the pressure loss measurements, only precision estimates are available (no data on absolute accuracy) for the reported flow rate data. There are two consolations concerning this matter, however: (1) the regular operators of the flow loop believe that the magnetic flow meter is within the manufacturer's tolerances, and (2) turbine meter checks tend to confirm that opinion.

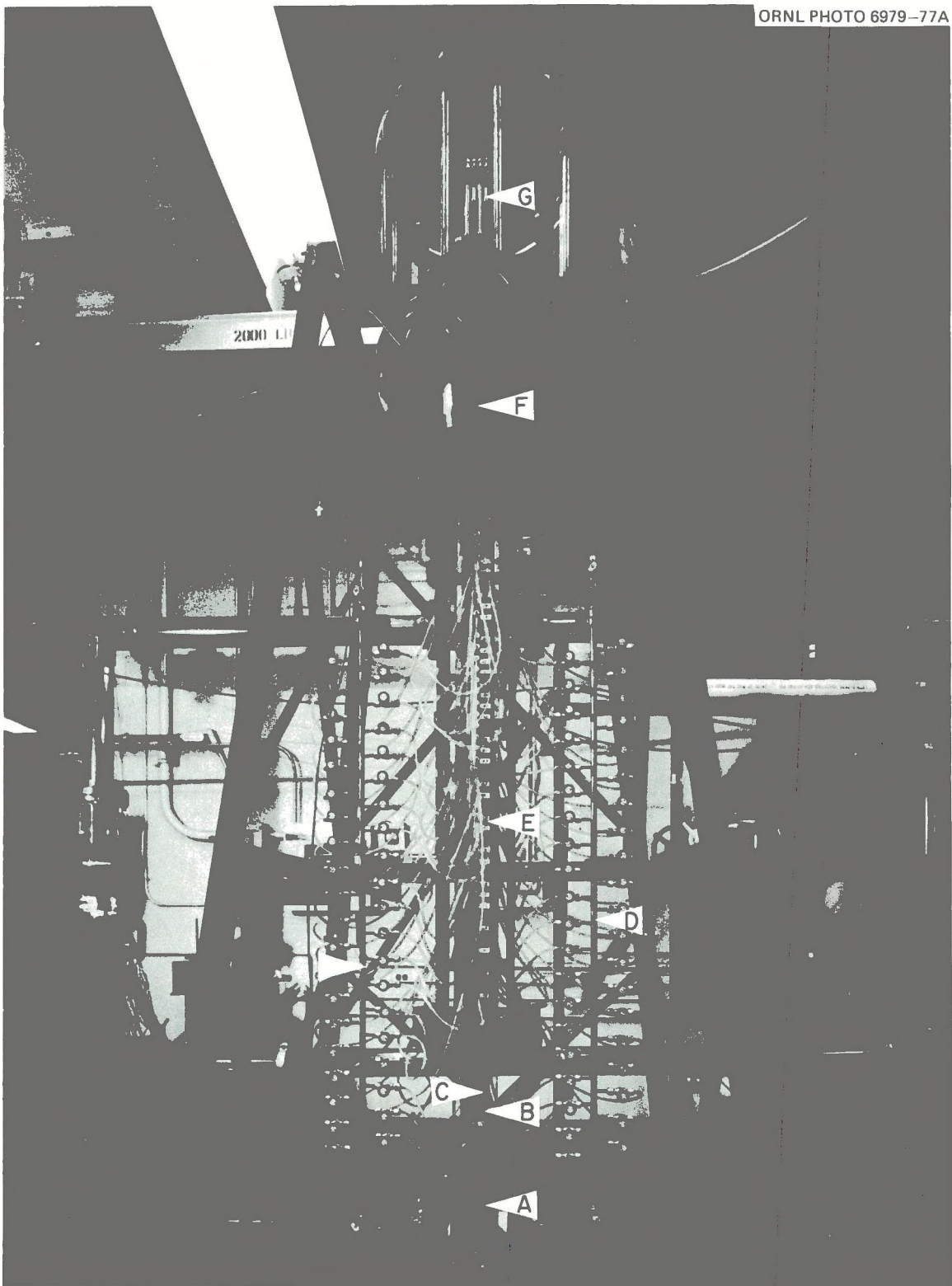


Fig. 2.1. MRBT flow test stand: (A) lower transition section, (B) lower flow shroud, (C) bundle extension, (D) manifolds and connecting tubing, (E) upper flow shroud, (F) upper transition section, (G) jackscrew retaining mechanism.

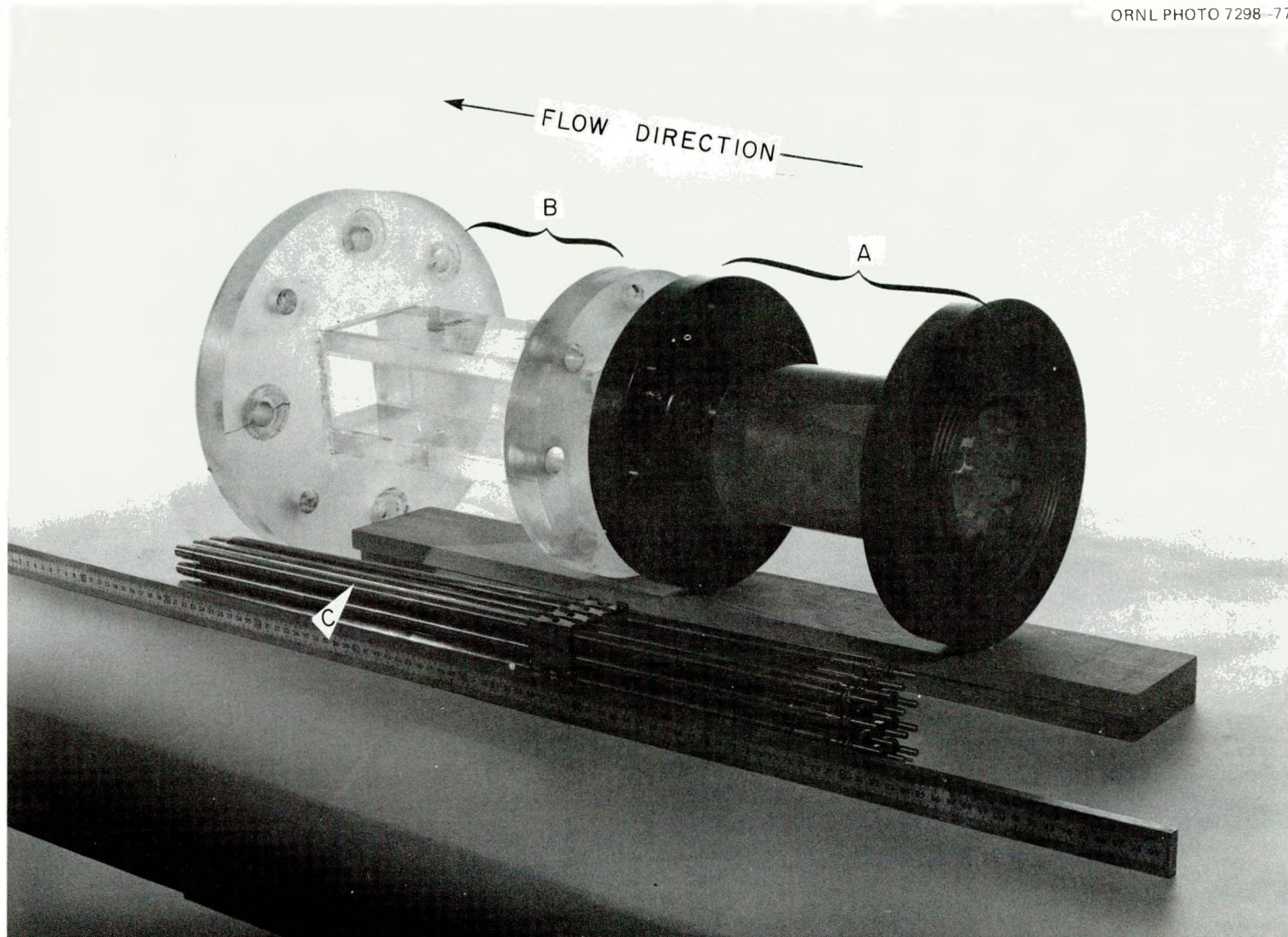


Fig. 2.2. Disassembled view of MRBT flow test stand: (A) lower transition section, (B) lower flow shroud 1, and (C) bundle extension.

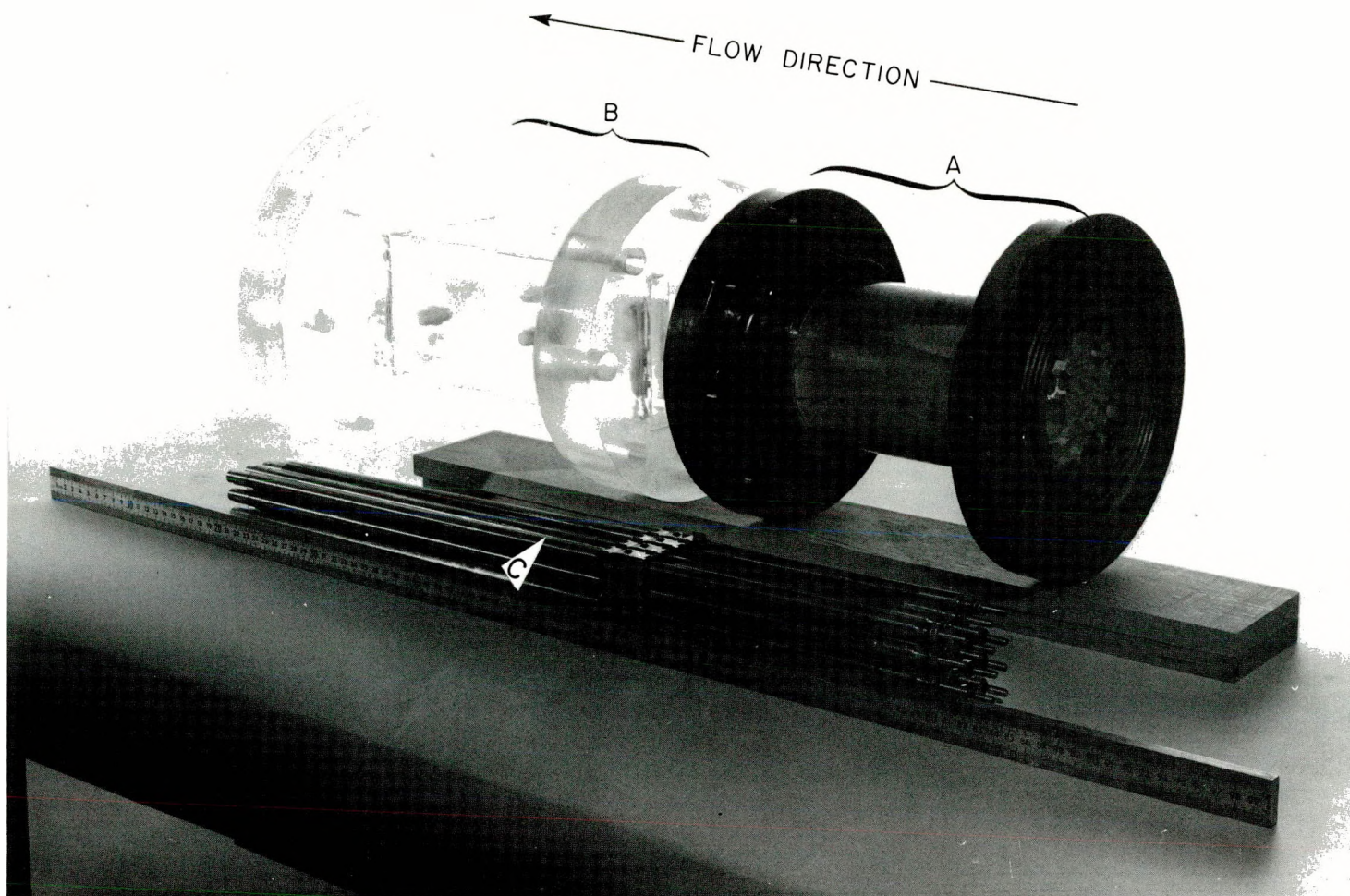


Fig. 2.3. Disassembled view of MRBT flow test stand: (A) lower transition section, (B) lower flow shroud 2, and (C) bundle extension.

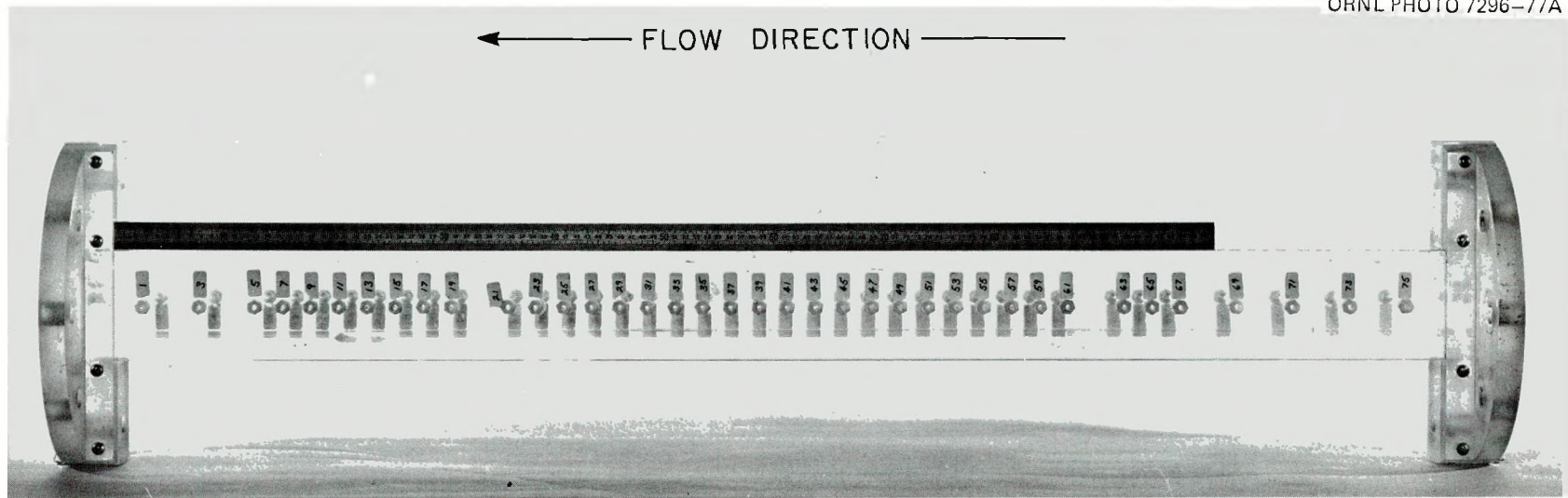


Fig. 2.4. Upper flow shroud 1.

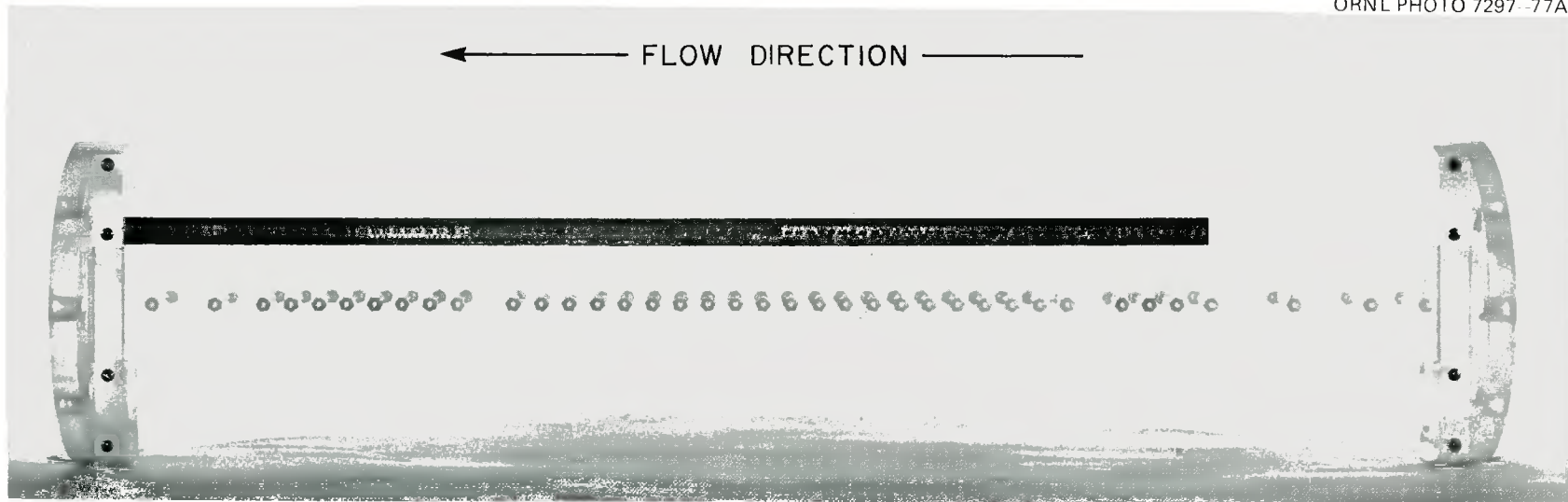
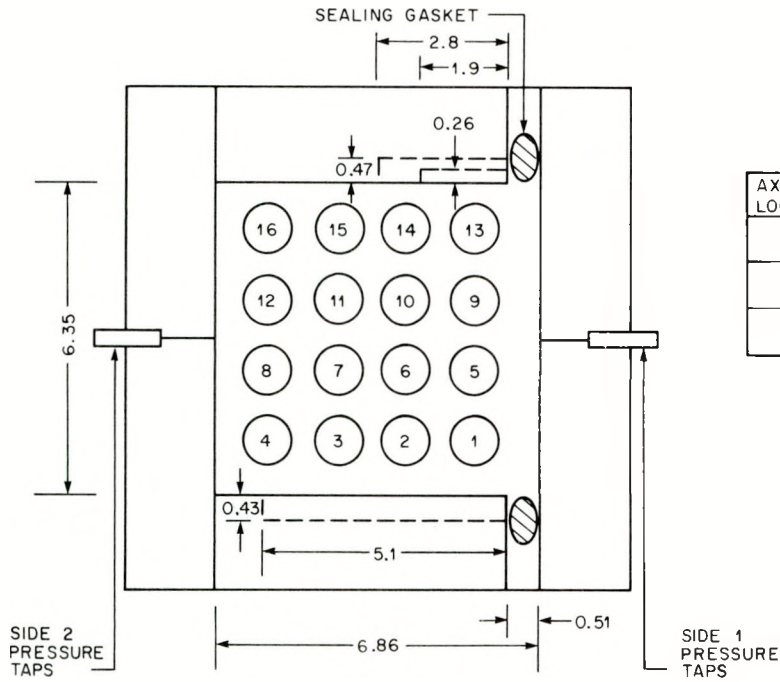


Fig. 2.5. Upper flow shroud 2.



AREA INFORMATION

UNDEFORMED BUNDLE AREA
= $14.99 \pm 0.15 \text{ cm}^2$

UNDEFORMED WATER FLOW AREA
= $28.8 \pm 0.5 \text{ cm}^2$

RELIEF GROOVES

AXIAL B-1 LOCATION	WIDTH	DEPTH	LENGTH
40.3	3.0	0.26	1.9
46.3	3.5	0.47	2.8
76.6	4.9	0.43	5.1

ALL DIMENSIONS IN CENTIMETERS

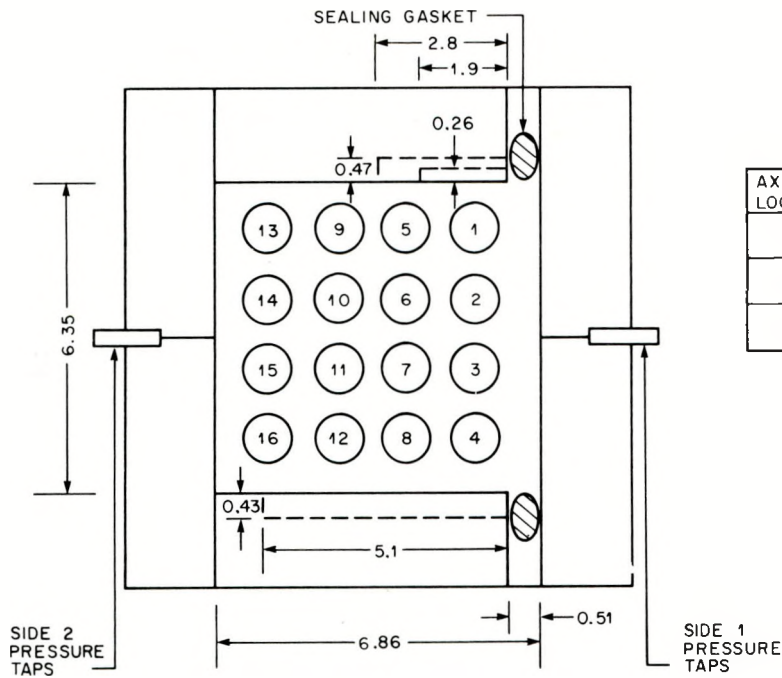
Fig. 2.6. General flow shroud 1 configuration with the reference bundle and B-1 present.

AREA INFORMATION

UNDEFORMED BUNDLE AREA
= $14.99 \pm 0.15 \text{ cm}^2$

UNDEFORMED WATER FLOW AREA
= $28.8 \pm 0.5 \text{ cm}^2$

RELIEF GROOVES



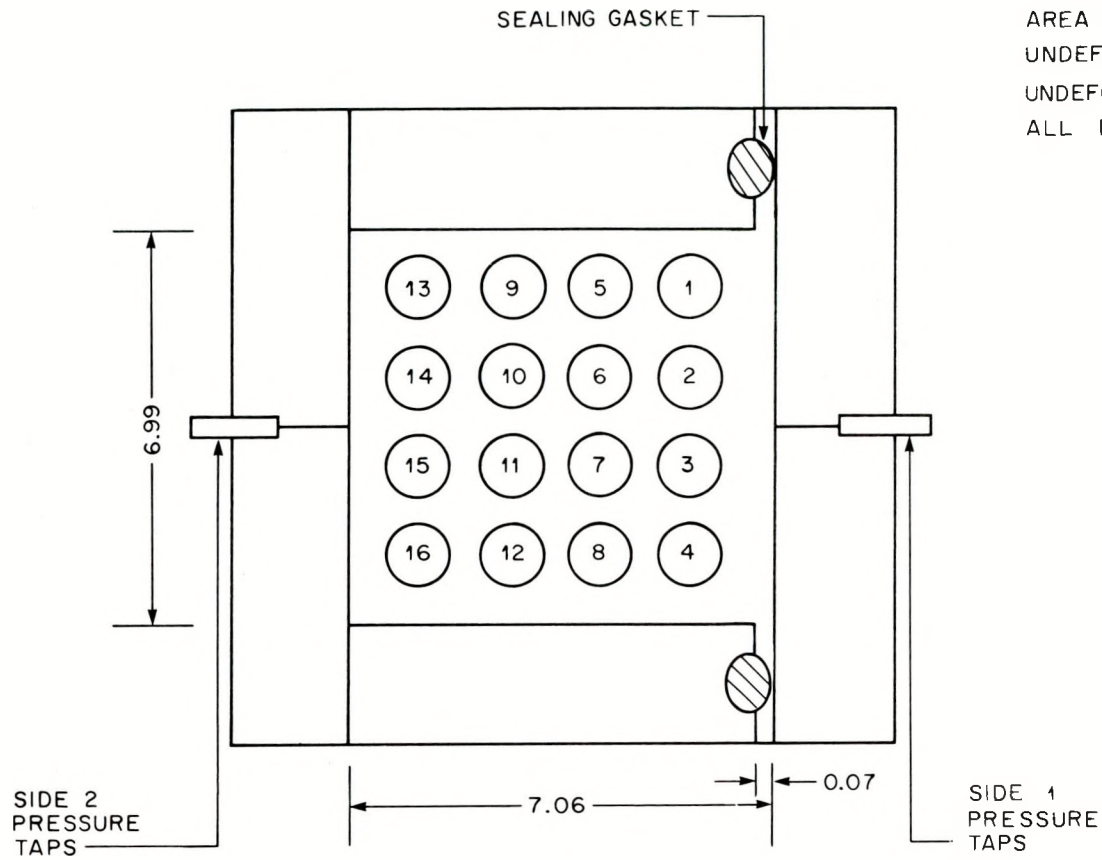
AXIAL B-1 LOCATION	WIDTH	DEPTH	LENGTH
40.3	3.0	0.26	1.9
46.3	3.5	0.47	2.8
76.6	4.9	0.43	5.1

ALL DIMENSIONS IN CENTIMETERS

Fig. 2.7. General flow shroud 1 configuration with B-2 present.



Fig. 2.8. View of the three relief grooves in upper shroud 1.



AREA INFORMATION
 UNDEFORMED BUNDLE AREA = $14.99 \pm 0.15 \text{ cm}^2$
 UNDEFORMED WATER FLOW AREA = $34.3 \pm 0.5 \text{ cm}^2$
 ALL DIMENSIONS IN CENTIMETERS

Fig. 2.9. General flow shroud 2 configuration with B-2 present.

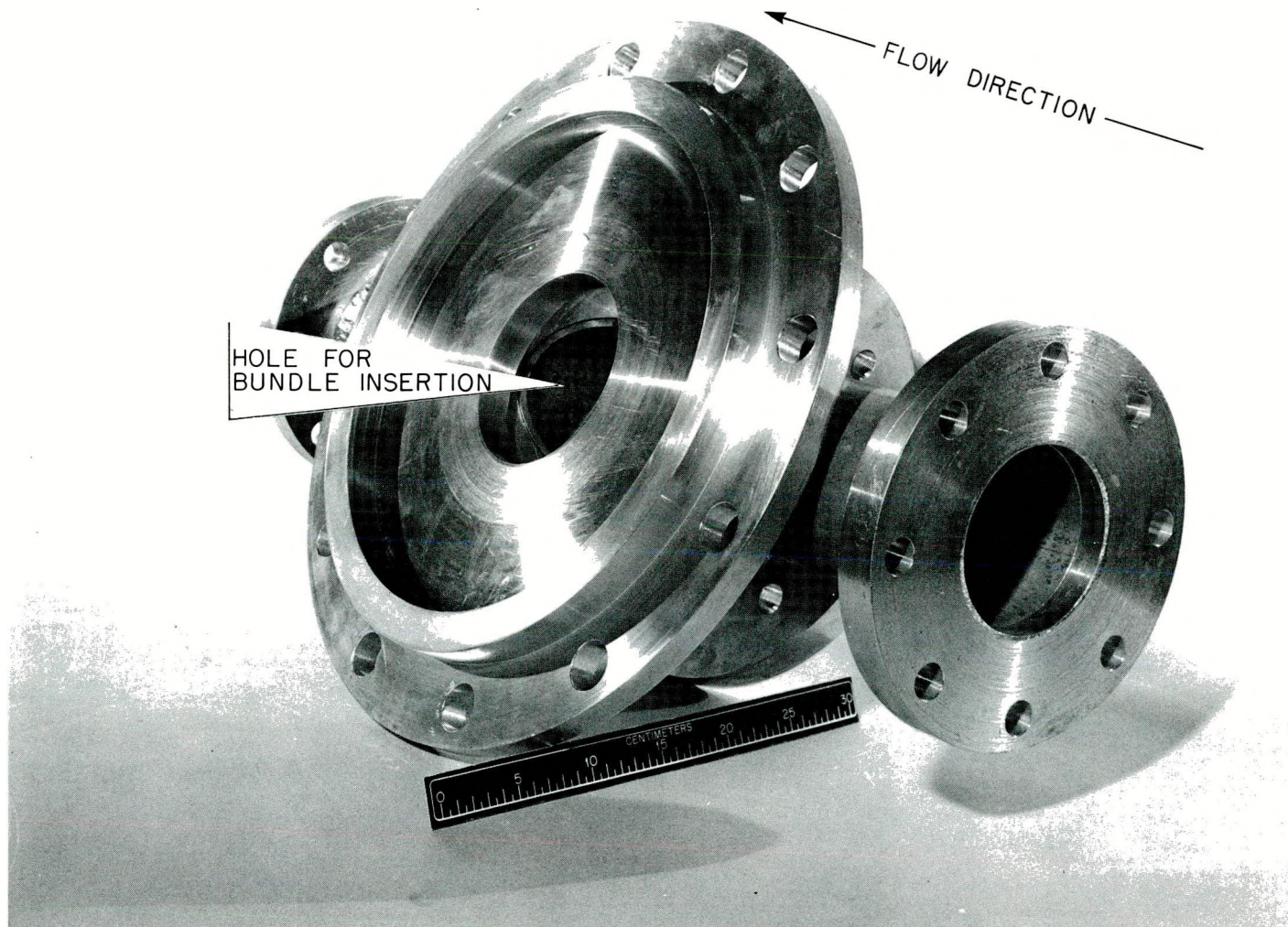


Fig. 2.10. Upper transition section.

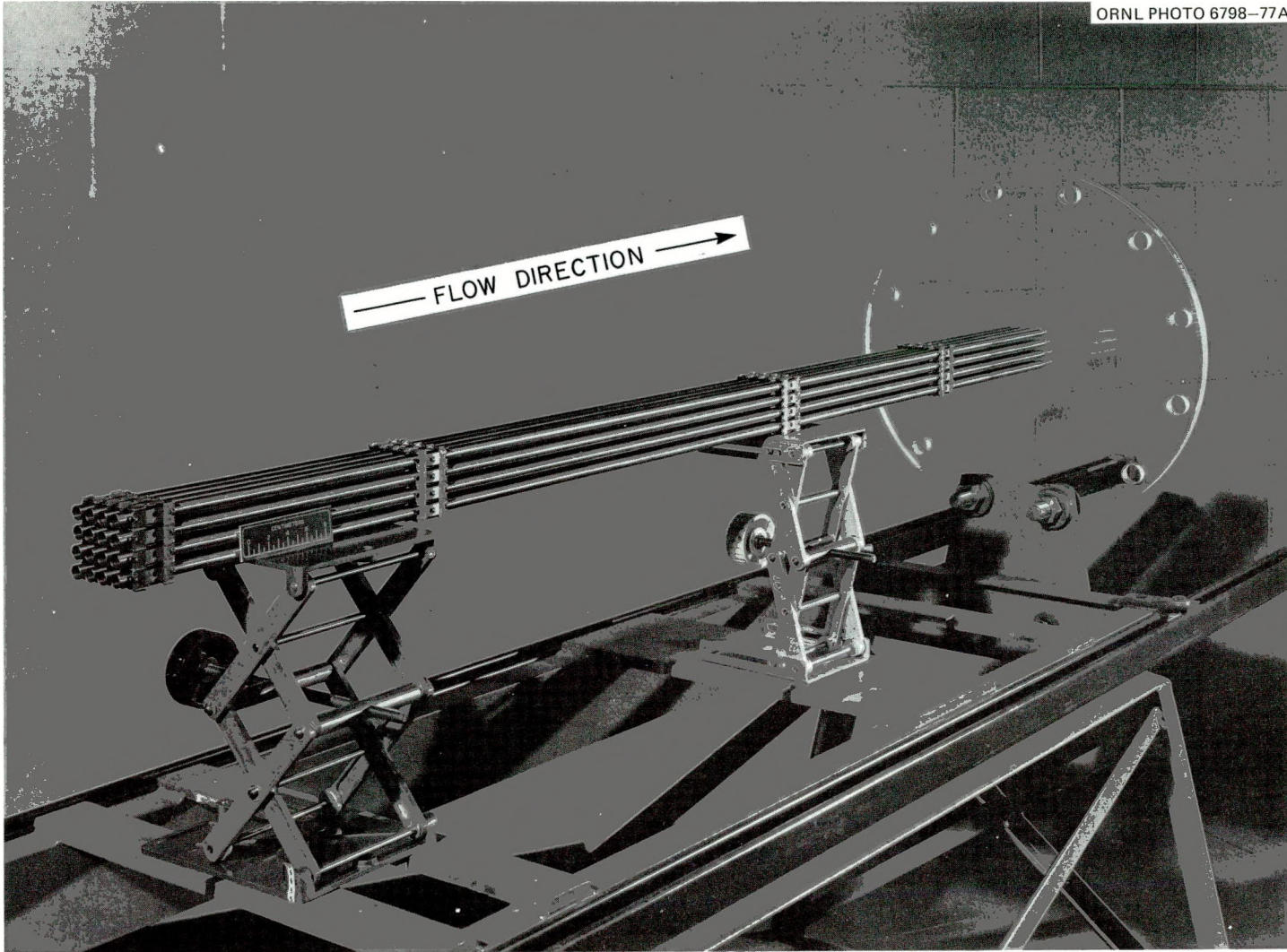


Fig. 2.11. Reference bundle for flow tests.

3. EXPERIMENTAL DATA

This chapter will deal mainly with the graphical presentation of experimental MRBT flow test data; the same data are available without interpretation in both graphical and tabulated formats in the MRBT B-1 and B-2 data reports.^{3,4} Consequently, all comments made in this chapter on the mathematical modeling of such data will be of a qualitative nature, the details being deferred until Chaps. 4 and 5. The primary intent of this chapter is to provide the reader with some feeling for the precision to which these data are known.

Two general types of data plots were generated with the previously described procedures and equipment. The first type is concerned with the differential pressure loss performance (the pressure loss profile) along a bundle's axis for a fixed volumetric flow rate. In generating these data, differential pressure loss measurements were made directly against the two reference pressure taps, using both sets of wall-mounted taps during the flow tests. The two resulting sets of data may then be compared separately, against one another, or averaged together for comparison against similar sets of data generated with other bundles.

The second type of data plot displays the integrated pressure loss over the instrumented portion of the bundle/shroud assembly as a function of Reynolds number. This type of comparison provides some insight into how well the differential pressure loss measurements may be interpolated to other Reynolds numbers in the $3 \times 10^4 \leq Re \leq 2 \times 10^5$ flow regime. (It should be noted that this integrated pressure loss is the average differential loss measured between the uppermost and reference shroud taps.) A simplified representation of various friction factor correlations for this flow regime indicates that this functional dependence of pressure loss measurements on the Reynolds number is closely approximated for isothermal systems by:

$$\Delta P \propto (Re)^n, \quad (3.1)$$

where n varies from about 1.7 to 1.8. Hence, by statistically fitting the data points in this second type of plot to such a power curve (and

ignoring the details of the grid spacers' Reynolds-number-dependent contribution to the integrated pressure loss, ΔP_I), additional insight is gained as to how well the data conform with existing empirical correlations (which suggest a Reynolds number, power-type dependence).

The Reynolds numbers used for these plots are averaged values in that the flow channel dimensions of the entire shroud were used in computing equivalent hydraulic diameters. The use of the adjective "maximum" in this context refers to the fact that this average Reynolds number is at a maximum in undeformed regions of the bundle, deformities increasing the wetted perimeter and hence lowering the Reynolds number for an isothermal, constant, volumetric flow rate. Such Reynolds numbers are referred to in the plots as "maximum Reynolds numbers" and are given to two significant figures, indicative of the precision to which they could be calculated with experimental flow channel dimensions, water temperature, and volumetric flow rate measurements.

3.1 First Single-Grid Tests in Shroud 1

Two series of flow tests were conducted with the reference bundle in shroud 1, using only one grid spacer at different locations in the instrumented portion of the reference bundle/shroud 1 assembly. The motivation for conducting these two experimental test series is to discriminate between pressure loss effects induced by friction in zones without spacer grids and those effects induced solely by the grids.

Results from the first single-grid test series begin with Figs. 3.1 to 3.6. The purpose of this series is to see what the pressure loss profile looks like for regions far away from grid spacers. Note that as the flow rate increases, two features emerge. First, a disturbance zone extends about 30 cm (20 hydraulic diameters) downstream of the spacer grid. Second, a significant nonlinearity exists in the pressure loss profile downstream of 65 cm. The persistence of these disturbed zones over several flow rates suggests these effects are real and not merely due to statistical error. The nonlinearity downstream of 65 cm is speculated to be induced by shroud exit effects and/or the third relief groove.

At one time some concern existed as to whether or not the three shroud 1 relief grooves (Figs. 2.6 to 2.8) adversely affected the flow tests. These grooves are located between 40 to 50 cm and 70 to 80 cm. For the 40- to 50-cm zone, some nonlinearities are visible but are relatively small and are considered of secondary importance when compared with data uncertainties. Hence, while the 70- to 80-cm relief groove zone roughly matches the location of the nonlinearity downstream of 65 cm, it is doubtful that the relief groove is solely responsible, since no similar effects of comparable magnitude appear in the 40- to 50-cm zone. Instead, shroud exit effects are suspected to have played the major role in producing this nonlinearity.

Figure 3.7 presents the average integrated pressure losses of this assembly as a function of Reynolds number. The power curve fit (solid line) was obtained using a standard linear regression analysis with the six data points shown on the plot. This power curve fit demonstrates the expected functional dependence of pressure loss measurements on the Reynolds number. Hence, interpolation of measured pressure losses to other Reynolds numbers in this range should be straightforward; extrapolations below or beyond this range must, of course, be subject to changes in the functional behavior of pressure losses with Reynolds number.

3.2 Second Single-Grid Tests in Shroud 1

A second flow test series, utilizing the same grid spacer further downstream, was conducted in shroud 1 to explicitly determine grid-spacer-induced pressure losses. These test results begin with Figs. 3.8 through 3.13 and are somewhat similar to those for the first single-grid tests. The actual modeling of grid-spacer-induced effects will be deferred until Chaps. 4 and 5.

Recalling that a zone of nonlinearity existed downstream of the grid in the first tests, it is somewhat surprising that the disturbed zone behind the grid in Figs. 3.8 to 3.13 is not as pronounced. Since the 40- to 50-cm relief groove zone also falls within the region, anomalies

due to the two grooves and the grid are very small as any effects induced by them are not easily discernible.

Perhaps the most significant anomaly occurring in this set of plots is the separation of pressure tap measurements made on opposite sides of the shroud in Fig. 3.13. After measurements were made on one side of the shroud, a documented 4% flow rate variation occurred while the remainder of the data was being taken on the other side. A less-pronounced variation of the same type is suspected to have occurred while Fig. 3.12 data were being taken. Inasmuch as a 4% variation in the flow rate induces about a 7% variation in pressure losses (the flow test water temperature staying essentially constant) due to the $\Delta P \propto Re^n$ dependence, the anomalies in Figs. 3.12 and 3.13 are considered understood.

The sensitivity of such pressure loss measurements on other experimental parameters (e.g., flow rate variations) is a topic to consider carefully. In mathematically modeling such flow test data, one generally takes the reported flow rates, water temperature, and flow channel geometry data and tries to calculate the corresponding pressure loss profiles. In comparing these calculated and experimentally determined pressure loss profiles, there is a tendency to concentrate on how well two such profiles agree and to neglect the significance of flow rate or Reynolds number and flow area uncertainties. From Figs. 2.6, 2.7, and 2.9, the precision of the flow channel area determinations may be seen to be about 2%. The precision of the flow rate or Reynolds number determinations may be assumed to follow the behavior described in Sect. 2.3, ranging from $\pm 5\%$ at very low flow rates to $\pm 2\%$ at higher flow rates. Consequently, when an attempt is made to mathematically model the pressure profiles, portions of the calculated pressure profiles may conceivably be varied roughly on the order of $\pm 10\%$ due to the precision limits on the flow channel area and the flow rate or Reynolds number determinations [the $\pm 10\% = 1 - (1.05)^2$]. In addition, individual pressure measurements may be off by as much as $\pm 5\%$ of the reported values. Note that these comments are based solely on precision estimates; they do not reflect any systematic errors which might exist due to undetected instrument inaccuracies (e.g., the magnetic flow meter which MRBT personnel were unable to calibrate).

Figure 3.14 presents a plot of the average integrated pressure losses as a function of Reynolds number for this test series. The differences in the data fit correlations given in Figs. 3.7 and 3.14 result from the statistical nature of the technique used; both correlations can be shown to predict about the same values. The conclusions to be drawn are identical to those stated for Fig. 3.7. A discussion of the pressure loss characteristics of the grid spacer will be deferred until Chaps. 4 and 5.

3.3 Reference Bundle Tests in Shroud 1

A third series of flow tests was conducted with the reference bundle/shroud 1 assembly. Four standard MRBT Inconel grid spacers, placed at posttest B-1 grid locations, were used to determine undeformed pressure losses with the reference bundle. Data from this test series begin with Figs. 3.15 to 3.20. In these plots the disturbance zones behind the spacer grids are fairly visible (except for lower flow rates); the nonlinearity downstream of 65 cm is masked, however, by the grid spacer near that position. Overall, the data are reasonably good; the slopes between grids are relatively linear, and separation between the two pressure tap sets is minimal.

3.4 B-1 Tests in Shroud 1

Figures 3.21 to 3.24 are plots of differential pressure loss data taken with the B-1/shroud 1 assembly for four flow rates. Tables 3.1 and 3.2 list the B-1 deformed tube areas, using the minimum and maximum flow restriction definitions given in an earlier report.⁵ Figures 3.25 to 3.28 represent averaged pressure loss profiles for B-1 (using the data in Figs 3.21 to 3.24) and the reference bundle against the overall B-1/shroud 1 flow restriction profiles (derived with minimum and maximum restriction data from Tables 3.1 and 3.2). By examining the behavior of the B-1 pressure loss profiles against the corresponding flow-channel-restriction profiles (in Figs. 3.25 to 3.28), one notices that the restriction profiles appear to be inverse, scaled-down representations of

Table 3.1. Lower limit of B-1 deformed tube areas (mm²)

Elevation (cm)	Tube No.																Total
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	
0.0	96	94	98	97	96	96	95	96	96	96	97	96	98	97	96	96	1549
1.8	104	109	108	109	105	113	108	109	109	111	105	109	111	109	110	105	1742
3.3	108	131	113	113	110	125	120	125	121	135	115	113	114	117	135	118	1919
5.2	108	129	108	114	109	119	121	128	123	130	120	111	110	119	150	118	1926
8.9	100	102	101	103	102	103	104	105	103	106	105	103	100	102	103	103	1651
11.8	102	104	101	104	103	105	105	107	103	106	107	104	101	102	105	103	1671
14.1	119	126	110	117	128	135	129	126	131	137	135	125	146	127	151	130	2080
15.4	117	124	114	119	144	137	128	125	134	140	135	125	175	132	174	139	2167
17.3	119	123	114	119	157	135	131	126	136	141	146	122	146	133	173	147	2175
18.8	124	130	112	124	149	140	135	131	153	149	195	121	127	146	179	153	2273
20.1	121	132	108	123	138	143	138	130	158	157	217	121	118	147	187	156	2301
20.6	121	132	108	122	136	142	137	128	157	156	215	118	116	142	180	154	2274
22.3	121	132	109	118	146	143	143	129	153	176	207	122	117	138	159	159	2279
23.9	120	134	111	117	151	140	146	130	154	195	195	127	121	141	153	179	2324
25.5	115	140	109	118	155	137	153	130	172	177	196	127	124	147	139	165	2312
26.5	116	144	110	120	172	144	158	127	177	175	190	126	129	150	133	151	2330
28.1	117	141	109	122	194	173	155	128	155	172	164	126	140	144	127	138	2311
29.7	116	136	109	117	170	190	146	129	140	153	141	123	146	135	126	133	2218
30.7	113	136	110	116	154	176	144	131	136	144	135	122	153	132	131	133	2173
33.2	109	136	109	112	141	145	152	137	130	137	133	128	133	130	139	128	2107
34.5	109	134	111	114	145	150	149	136	130	139	134	129	127	126	137	129	2108
36.6	110	130	113	117	147	143	145	141	125	134	136	125	127	128	133	140	2100
38.1	113	136	113	118	148	135	141	150	125	132	140	125	138	132	136	143	2131
39.7	124	144	109	117	149	132	141	156	133	134	141	130	221	143	139	139	2259
40.8	125	143	109	117	152	130	140	156	141	140	141	130	236	149	135	136	2288
42.9	119	138	112	120	169	125	138	170	163	143	134	136	150	156	133	137	2251
44.7	122	140	117	122	177	124	144	188	182	138	133	149	143	180	142	142	2352
46.7	123	139	113	124	188	124	156	171	193	137	134	162	155	187	147	134	2395
47.5	122	137	112	124	190	128	158	160	201	141	135	171	150	178	147	130	2391
48.6	119	133	118	125	176	126	151	153	180	140	130	176	143	161	142	132	2314
50.4	123	129	141	122	166	124	140	145	177	140	130	161	144	152	135	133	2270
52.4	130	131	113	116	148	123	137	143	175	141	131	142	143	148	135	128	2195
54.0	128	132	108	115	143	120	134	133	163	137	129	130	138	139	141	124	2121
55.6	120	125	106	115	149	117	129	131	153	132	126	124	128	129	137	122	2052
57.8	116	122	106	116	154	111	128	137	138	124	121	123	124	125	128	123	2004
60.1	122	123	105	123	135	109	141	130	132	123	121	116	126	129	124	114	1979
61.7	115	117	103	116	124	107	127	119	119	121	115	113	122	120	120	111	1875
64.1	100	101	97	101	104	98	102	102	101	102	101	101	103	102	103	102	1630
66.9	102	102	97	101	104	97	104	103	104	102	102	101	103	102	104	100	1636
68.8	124	118	103	117	136	102	126	117	121	120	120	117	117	118	120	113	1898
70.3	140	126	109	122	162	103	133	122	128	128	121	122	119	127	119	118	2006
72.7	147	139	112	127	192	102	149	128	145	128	124	122	119	142	126	120	2129
74.2	163	153	113	140	197	103	174	128	160	134	128	121	123	153	135	122	2254
76.5	172	159	113	173	196	103	172	125	147	135	126	131	136	150	137	127	2309
77.3	156	164	113	153	182	102	166	126	135	128	123	134	134	141	131	125	2221
80.2	147	161	111	137	149	104	143	130	140	125	128	129	137	141	130	119	2140
81.6	136	153	110	131	140	105	134	118	129	122	126	123	127	135	134	117	2047
83.5	132	152	109	120	142	108	128	114	130	122	124	122	129	129	137	119	2026
85.1	120	148	103	109	132	103	116	113	133	115	118	118	133	121	128	119	1935
86.5	118	129	101	105	119	101	110	111	133	114	116	115	132	114	119	115	1860
87.9	112	108	100	103	104	99	104	107	116	108	108	108	116	105	109	112	1726
90.0	102	98	96	100	97	96	99	98	96	98	99	94	96	98	101	100	1576
92.5	93	93	94	95	93	93	93	93	94	93	94	94	93	94	94	94	1503

Table 3.2. Upper limit of B-1 deformed tube areas (mm²)

Elevation (cm)	Tube No.																Total
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	
0.0	96	94	98	97	96	96	95	96	96	96	97	96	98	97	96	96	1549
1.8	104	109	108	109	105	113	108	109	109	111	105	109	111	109	110	105	1742
3.3	108	131	113	113	110	125	120	125	121	135	115	113	114	117	135	118	1919
5.2	108	129	108	114	109	119	121	128	123	130	120	111	110	119	150	118	1926
8.9	100	102	101	103	102	103	104	105	103	106	105	103	100	102	103	103	1651
11.8	102	104	101	104	103	105	105	107	103	106	107	104	101	102	105	103	1671
14.1	119	126	110	117	128	135	129	126	131	137	135	125	146	127	151	130	2080
15.4	117	124	114	119	144	137	128	125	154	140	135	125	175	132	174	139	2167
17.3	119	123	114	119	157	135	131	126	136	141	146	122	146	133	173	147	2175
18.8	124	130	112	124	149	140	135	131	153	149	195	121	127	146	179	153	2273
20.1	121	132	108	123	138	143	138	130	158	157	217	121	118	147	189	156	2303
20.6	121	132	108	122	136	142	137	128	157	156	215	118	116	142	180	154	2274
22.3	121	132	109	118	146	143	142	129	153	176	269	122	117	138	159	227	2409
23.9	120	134	111	117	151	140	146	130	154	209	217	127	121	141	153	283	2464
25.5	115	140	109	118	155	137	153	130	172	177	196	127	124	147	139	165	2312
26.5	116	144	110	120	172	144	158	127	177	175	190	126	129	150	133	151	2330
28.1	117	141	109	122	194	173	155	128	155	172	164	126	140	144	127	138	2311
29.7	116	136	109	117	170	253	146	129	140	153	141	123	146	135	126	133	2281
30.7	113	136	110	116	154	176	144	131	136	144	135	122	153	132	131	133	2173
33.2	109	136	109	112	141	145	152	137	130	137	133	128	133	130	139	128	2107
34.5	109	134	111	114	145	150	149	136	130	139	134	129	127	126	137	129	2108
36.6	110	130	113	117	147	143	145	141	125	134	136	125	127	128	133	140	2100
38.1	113	136	113	118	148	135	141	150	125	132	140	125	138	132	136	143	2131
39.7	124	144	109	117	149	132	141	156	133	134	141	130	221	143	139	139	2259
40.8	125	143	109	117	152	130	140	156	141	140	141	130	281	149	135	136	2333
42.9	119	138	112	120	169	125	138	170	163	143	134	136	150	156	133	137	2251
44.7	122	140	117	122	177	124	144	244	182	138	133	149	143	180	142	142	2409
46.7	123	139	113	124	188	124	156	171	291	137	134	162	155	262	147	134	2568
47.5	122	137	112	124	193	128	158	160	201	141	135	210	150	178	147	130	2433
48.6	119	133	118	125	176	126	151	153	180	140	130	222	143	161	142	132	2359
50.4	123	129	160	122	166	124	140	145	177	140	130	161	144	152	135	133	2289
52.4	130	131	113	116	148	123	137	143	175	141	131	142	143	148	135	128	2195
54.0	128	132	108	115	143	120	134	133	163	137	129	130	138	139	141	124	2121
55.6	120	125	106	115	149	117	129	131	153	132	126	124	128	129	137	122	2052
57.8	116	122	106	116	154	111	128	137	138	124	121	123	124	125	128	123	2004
60.1	122	123	105	123	135	109	141	130	132	123	121	116	126	129	124	114	1979
61.7	115	117	103	116	124	107	127	119	119	121	115	113	122	120	120	111	1875
64.1	100	101	97	101	104	98	102	102	101	102	101	101	103	102	103	102	1630
66.9	102	102	97	101	104	97	104	103	104	102	102	101	103	102	104	100	1636
68.8	124	118	103	117	136	102	126	117	121	120	120	117	117	118	120	113	1898
70.3	140	126	109	122	162	103	133	122	128	128	121	122	119	127	119	118	2006
72.7	147	139	112	127	192	102	149	128	145	128	124	122	119	142	126	120	2129
74.2	163	153	113	140	197	103	177	128	160	134	128	121	123	153	135	122	2257
76.5	243	181	113	258	196	103	225	125	147	135	126	131	136	150	137	127	2540
77.3	215	214	113	230	182	102	166	126	135	128	123	134	134	141	131	125	2406
80.2	147	161	111	137	149	104	143	130	140	125	128	129	137	141	130	119	2140
81.6	136	153	110	131	140	105	134	118	129	122	126	123	127	135	134	117	2047
83.5	132	152	109	120	142	108	128	114	130	122	124	122	129	129	137	119	2026
85.1	120	148	103	109	132	103	116	113	133	115	118	118	133	121	128	119	1935
86.5	118	129	101	105	119	101	110	111	133	114	116	115	132	114	119	115	1860
87.9	112	108	100	103	104	99	104	107	116	108	108	108	116	105	109	112	1726
90.0	102	98	96	100	97	96	99	98	96	98	99	94	96	98	101	100	1576
92.5	93	93	94	95	93	93	93	93	94	93	94	94	93	94	94	94	1503

the pressure loss profiles, neither restriction definition being obviously more representative.

Figures 3.21 to 3.28 suggest that several generalizations may be made about burst bundle experimental data. With the exception of a few data points, the two sets of differential pressure loss profiles taken on opposing sides of the B-1/shroud 1 assembly appear to be equivalent within the data precision limits (see Figs. 3.21 to 3.24). Further, by a somewhat tedious comparison with Tables 3.1 and 3.2, it may be observed that the axial points at which differences between opposing pressure measurements appear (outside experimental uncertainty limits) are generally covariant with the occurrence of larger tube areas along only one shroud wall which houses a set of pressure taps. Hence, while localized pressure profile features may be induced by deformed tubes close to the pressure taps, the general shape of the pressure loss profiles is primarily dependent upon the gross axial-flow-area behavior of the burst bundle/shroud assembly. By averaging the two differential profiles together for each flow test (as in Figs. 3.25 to 3.28), a fairly good approximation to the bundle-averaged pressure loss profile may be derived from this experimental data.

These overlain averaged pressure loss profiles also provide a visual comparison between B-1 and reference bundle pressure losses as a function of axial position. Caution must be taken, however, in the resolution of B-1 deformation-induced losses over the pressure losses of an undeformed bundle. The effect of precision limitations on comparisons between two such sets of data is important yet subtle. In Fig. 3.25, the B-1 and reference bundle profiles are essentially the same below 0 cm; similarly, the water temperature for the two tests is about the same (24°C). Yet the B-1/reference bundle flow rate ratio is about 0.93, implying that the B-1 pressure profile should be uniformly lower, by about a factor of 14%, than that of the reference bundle under these conditions. This anomaly is indicative of the precision to which flow rates and Reynolds numbers could be calculated for these tests (i.e., the profiles might indeed agree below 0 cm if the reported B-1 flow rate was higher by 3.5% and the reference bundle flow rate lower by 3.5%). Consequently, uncertainties

in the determination of deformation-induced pressure losses and the normal pressure losses an unburst bundle would experience are rather severe due to existing Reynolds number and pressure measurement precision limits.

The following scenario illustrates the previous discussion. Since the pressure loss profiles in Fig. 3.25 agree below 0 cm despite flow rate or Reynolds number differences, some assumptions must be made to determine the increased nonrecoverable B-1 losses due to bundle deformation over the normal, permanent losses of an unburst bundle. First, assume that the flow channel geometries are identical below 0 cm and water temperatures are the same for both bundles. Now two sources of uncertainty exist: uncertainties in flow rate or Reynolds number determinations and in pressure measurements. These assumptions and the agreement of the pressure loss profiles below 0 cm imply the same flow rate for both tests. If an average is taken to determine a Reynolds number corresponding to both profiles, no flow-rate-type corrections need be made to the two pressure loss profiles. At 96 cm, the B-1 data point is 17.2 kPa; for reference bundle, it is 12.1 kPa. Assume that the $\pm 5\%$ precision limit on differential pressure measurements mentioned in Sect. 2.2 applies to these values. Hence, the uncertainty for the 17.2-kPa measurement is ± 0.9 kPa, and that for 12.1 kPa is ± 0.6 kPa. The difference between the B-1 and reference pressures at 96 cm is then 5.1 ± 1.1 kPa, or $\pm 22\%$ for a Reynolds number of $3.5 \pm 0.3 \times 10^4$. Further, if one desires to compare such experimentally derived deformation-induced pressure differences with their calculational counterparts, the effects of flow rate uncertainties (i.e., the $\pm 0.3 \times 10^4$) on such comparisons must be considered. While the preceding is admittedly an extreme scenario (the precision limits on these two particular measurements may well be ± 2 or $\pm 3\%$ instead of $\pm 5\%$), the point must be strongly made that uncertainties in the flow test data severely limit how precisely deformation-induced pressure losses between burst and reference bundle pressure loss profiles may be resolved.

Figures 3.26, 3.27, and 3.28 give B-1 to reference bundle flow rate ratios of 1.02, 1.01, and 1.00, respectively (because of comparable fluid

temperatures, the corresponding Reynolds number ratios are the same). Notice that the agreement of the two profiles below 0 cm generally improves with increasing flow rate magnitudes. This tendency, along with the discrepancy discussed for Fig. 3.25, is indicative of the fact that the larger flow rates and Reynolds numbers are known more precisely on a percentage basis.

By an examination of Figs. 3.25 through 3.28, an estimation of the permanent pressure losses induced by the three deformation zones (12 to 32 cm, 38 to 63 cm, and 68 to 96 cm) may be obtained in terms of losses induced by a spacer grid. Bundle deformities induce two types of pressure losses: a permanent nonrecoverable loss and a nonpermanent loss which is recovered shortly downstream of the restriction. At 96 cm in Figs. 3.25 through 3.28, virtually all of the difference in pressure between the B-1 and reference bundle profiles results from permanent losses induced by B-1 deformities. If this pressure difference is divided by 3 (for the three burst zones) and then by the permanent loss associated with a spacer grid, these B-1 deformation zones are seen to have permanent losses of about 1 1/2 times those typical of MRBT grid spacers for this flow regime.

Figure 3.29 presents average integrated pressure loss measurements as a function of Reynolds number for the B-1 and reference bundles in shroud 1. As seen, these two sets of pressure data follow the expected functional dependence on the Reynolds number.

3.5 B-2 Tests in Shroud 1

Figures 3.30 to 3.33 are plots of differential pressure loss data taken with the B-2/shroud 1 assembly for four flow rates. Tables 3.3 and 3.4 list B-2 deformed tube areas based on the minimum and maximum flow restriction definitions. Figures 3.34 to 3.37 compare averaged B-2 and reference bundle differential pressure loss profiles against the fractional flow restriction profiles for the B-2/shroud 1 assembly.

Note (in Figs. 3.30 to 3.33) that, as for the similar set of B-1 plots, pressure measurements made on opposing shroud walls tend to agree

Table 3.3. Lower limit of B-2 deformed tube areas (mm²)

Elevation (cm)	Tube No.																TOTAL
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	
0.0	92	93	93	94	93	92	93	94	92	92	92	93	91	92	92	93	1490
1.8	102	104	105	104	101	107	107	106	102	103	103	107	101	105	102	102	1670
3.4	114	115	117	114	114	132	121	116	109	114	115	119	111	123	112	108	1862
5.0	114	116	115	116	113	128	119	115	110	114	115	115	111	119	114	109	1850
6.6	110	112	109	108	108	120	114	108	107	111	111	110	107	112	108	106	1767
8.8	102	103	101	101	100	103	101	101	99	99	101	100	100	100	99	99	1616
11.5	100	101	101	102	100	104	101	102	100	100	102	102	100	101	99	101	1622
13.3	110	108	111	108	106	123	117	112	111	112	119	114	109	113	103	106	1791
15.1	120	116	116	112	110	136	124	119	114	118	127	125	116	128	106	107	1902
16.8	124	126	122	116	116	141	125	126	115	126	131	129	122	154	115	114	2009
18.1	125	130	129	119	122	142	125	131	115	127	128	130	125	177	118	116	2065
19.5	133	133	134	121	131	140	130	137	123	138	127	136	136	188	118	119	2150
21.4	134	129	134	114	130	138	130	141	121	140	124	145	133	164	115	117	2115
23.2	133	126	138	112	131	145	131	151	123	142	123	139	131	148	119	120	2118
25.0	126	121	144	115	129	142	137	152	124	137	120	131	126	140	124	123	2096
26.6	124	116	148	120	122	134	142	136	122	137	120	132	123	125	128	117	2054
28.5	127	118	157	127	123	137	145	129	123	143	122	133	123	124	131	114	2083
30.0	123	118	164	131	123	138	142	126	120	142	124	132	118	125	129	116	2076
32.0	118	124	153	134	123	136	138	126	120	137	122	135	119	128	121	120	2060
34.0	116	124	139	129	123	143	133	123	121	131	123	136	126	129	119	119	2042
35.6	119	130	138	121	124	194	129	126	120	134	127	130	127	140	121	116	2105
37.7	120	131	143	120	130	233	125	136	123	146	126	124	122	158	126	118	2190
39.5	117	123	135	121	125	182	131	142	120	164	122	122	118	155	125	120	2128
41.2	132	125	143	125	126	171	141	157	126	191	127	132	129	150	128	127	2237
43.3	139	125	156	128	125	194	133	177	122	165	127	137	128	137	124	131	2256
44.7	142	123	177	131	123	185	128	165	122	151	126	131	125	132	119	138	2225
46.2	152	128	179	134	126	174	128	142	128	142	126	130	123	130	117	155	2218
47.7	166	132	160	134	132	176	135	130	137	145	129	132	125	131	120	160	2253
49.7	155	133	138	132	136	170	136	129	132	135	134	135	122	133	127	139	2196
51.6	136	134	126	142	151	155	128	131	129	136	131	139	122	136	129	133	2166
53.5	130	134	133	156	170	148	134	129	133	138	130	147	139	142	136	153	2261
54.5	132	139	133	151	155	148	140	129	130	140	129	151	158	144	135	175	2296
56.2	136	146	137	151	146	147	143	139	129	145	129	153	187	152	142	188	2378
57.6	128	140	138	153	138	137	133	144	123	138	120	142	169	146	135	164	2254
59.8	120	123	131	143	123	129	127	127	119	128	115	129	126	130	121	128	2027
61.8	115	113	118	118	114	120	123	117	115	120	114	121	119	117	114	114	1881
63.8	103	103	103	105	103	104	104	105	100	103	103	104	105	104	103	105	1665
66.5	102	102	103	106	102	104	105	104	101	103	104	103	104	103	101	103	1658
68.4	113	115	120	114	115	129	130	116	118	121	123	115	115	114	109	111	1886
70.1	123	130	134	125	126	151	141	127	128	125	136	127	112	132	118	118	2065
71.6	123	139	137	138	136	151	138	129	133	130	135	127	112	148	119	122	2124
73.1	126	146	145	159	139	147	141	133	141	131	136	129	114	150	120	129	2195
74.6	138	165	174	177	142	159	152	139	156	135	149	134	127	145	128	135	2366
76.2	142	179	184	188	139	171	162	139	143	146	168	141	128	140	139	140	2459
78.0	144	155	166	166	143	155	168	139	135	152	183	149	128	141	161	141	2432
79.5	141	136	140	151	143	140	182	138	134	153	170	156	127	139	170	139	2366
81.6	128	130	127	149	132	143	226	126	132	154	148	182	122	131	146	126	2311
83.8	116	125	125	132	121	143	164	119	126	137	135	138	114	126	122	116	2066
86.0	109	121	118	119	124	123	136	116	121	141	122	117	108	122	112	114	1930
88.1	106	109	105	108	113	111	117	109	108	117	108	107	103	112	106	109	1758
89.5	55	96	96	99	97	97	98	95	96	100	94	99	93	98	94	95	1550
91.5	54	94	95	95	95	93	95	93	94	96	94	95	93	96	94	94	1518

Table 3.4. Upper limit of B-2 deformed tube areas (mm²)

Elevation (cm)	Tube No.																TOTAL
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	
0.0	93	93	93	94	93	92	93	94	92	92	92	93	91	92	92	93	1490
1.8	102	104	105	104	101	107	107	106	102	103	103	107	101	105	102	102	1670
3.4	114	115	117	114	114	132	121	116	109	114	115	119	111	123	112	108	1862
5.0	114	116	115	116	113	128	119	115	110	114	115	115	111	119	114	109	1850
6.9	110	112	109	108	108	120	114	108	107	111	111	110	107	112	108	106	1767
8.8	102	103	101	101	100	103	101	101	99	95	101	100	100	100	99	99	1616
11.5	100	101	101	102	100	104	101	102	100	100	102	102	100	101	99	101	1622
13.3	110	108	111	108	106	123	117	112	111	112	119	114	109	113	103	106	1791
15.1	120	116	116	112	110	136	124	119	114	118	127	125	116	128	106	107	1902
16.8	124	126	122	116	116	141	125	126	115	126	131	129	122	154	115	114	2009
18.1	125	130	129	119	122	142	125	131	115	127	128	130	125	177	118	116	2065
19.5	133	133	134	121	131	140	130	137	123	138	127	136	136	233	118	119	2195
21.4	134	129	134	114	130	138	130	141	121	140	124	145	133	164	115	117	2115
23.2	133	126	138	112	131	145	131	151	123	142	123	139	131	148	119	120	2118
25.0	126	121	144	115	129	142	137	152	124	137	120	131	126	140	124	123	2096
26.9	124	116	148	120	122	134	142	136	122	137	120	132	123	125	128	117	2054
28.5	127	118	157	127	123	137	145	129	123	143	122	133	123	124	131	114	2083
30.0	123	118	164	131	123	138	142	126	120	142	124	132	118	125	129	116	2076
32.0	118	124	153	134	123	136	138	126	120	137	122	135	119	128	121	120	2060
34.0	116	124	139	129	123	143	133	123	121	131	123	136	126	129	119	119	2042
35.9	119	130	138	121	124	194	129	126	120	134	127	130	127	140	121	116	2105
37.7	120	131	143	120	130	248	125	136	123	146	126	124	122	158	126	118	2205
39.5	117	123	135	121	125	182	131	142	120	164	122	122	118	155	125	120	2128
41.2	132	125	143	125	126	171	141	157	126	256	127	132	129	150	128	127	2302
43.3	139	125	156	128	125	194	133	259	122	165	127	137	128	137	124	131	2338
44.7	142	123	177	131	123	185	128	165	122	151	126	131	125	132	119	138	2225
46.2	152	128	179	134	126	174	128	142	128	142	126	130	123	130	117	155	2218
47.7	204	132	160	134	132	176	135	130	137	145	129	132	125	131	120	160	2288
49.7	155	133	138	132	136	170	136	129	132	135	134	135	122	133	127	139	2196
51.6	136	134	126	142	151	155	128	131	129	136	131	139	122	136	129	133	2166
53.5	130	134	133	156	261	148	134	129	133	138	130	147	139	142	136	153	2352
54.9	132	139	133	151	155	148	140	129	130	140	129	151	158	144	135	175	2296
56.2	136	146	137	151	146	147	143	139	129	145	129	153	253	152	142	271	2527
57.6	128	140	138	153	138	137	133	144	123	138	120	142	169	146	135	164	2254
59.8	120	123	131	143	123	129	127	127	119	128	115	129	126	130	121	128	2027
61.8	115	113	118	118	114	120	123	117	115	120	114	121	119	117	114	114	1881
63.8	103	103	103	105	103	104	104	105	100	103	103	104	105	104	103	105	1665
66.5	102	102	103	106	102	104	105	104	101	102	104	103	104	103	101	103	1658
68.4	113	115	120	114	115	129	130	116	118	121	123	115	115	114	109	111	1886
70.1	123	130	134	125	126	151	141	127	128	125	136	127	112	132	118	118	2065
71.6	123	139	137	138	136	151	138	129	133	130	135	127	112	148	119	122	2124
73.1	126	146	145	159	139	147	141	133	141	131	136	129	114	150	120	129	2195
74.6	138	165	174	177	142	159	152	139	184	139	149	134	127	145	128	135	2394
76.2	142	203	264	288	139	171	162	139	143	146	168	141	128	140	139	140	2663
78.0	144	155	166	166	143	155	168	139	135	152	228	149	128	141	161	141	2478
79.5	141	136	140	151	143	140	182	138	134	153	170	156	127	139	189	139	2384
81.6	128	130	127	149	132	143	295	126	132	154	148	228	122	131	146	126	2425
83.8	116	125	125	132	121	143	164	119	126	137	135	138	114	126	122	116	2066
86.0	109	121	118	119	124	123	136	116	121	141	122	117	108	122	112	114	1930
88.1	106	109	105	108	113	111	117	109	108	117	108	107	103	112	106	109	1758
89.9	95	96	96	99	97	97	98	95	96	100	94	99	93	98	94	95	1550
91.5	94	94	95	95	95	93	95	93	94	96	94	95	93	96	94	94	1518

well. The exceptional points, as for B-1, closely correlate with the occurrence of significantly larger tube areas adjacent to the taps registering the higher pressures (i.e., tube 15 at 19.5 cm; tubes 2, 3, and 4 at 76.2 cm; see Tables 3.3 and 3.4). Hence, in comparing B-2 flow channel restriction profiles with averaged pressure losses (Figs. 3.34 through 3.37), one concludes that while local deformation features cause local pressure perturbations, the general shape of the pressure profile follows the gross flow channel restriction profiles. And again, as for B-1, no preference for either restriction definition can be made by visual comparisons of Figs. 3.34 to 3.37.

When the B-2 and reference bundle axial pressure loss profiles below 0 cm in Figs. 3.34 to 3.37 are compared, both profiles are seen to closely agree in this zone. For such pressure measurements, made under geometrically identical conditions but with different flow rates and fluid temperatures, the ratio of the burst to reference bundle pressure may be approximated as:

$$\frac{\Delta P_B}{\Delta P_R} = \left(\frac{(Re_R)^{2-n_R}}{(Re_B)^{2-n_B}} \right) \left(\frac{V_B}{V_R} \right)^2, \quad (3.2)$$

where

$\Delta P \equiv$ pressure loss,

$Re \equiv$ Reynolds number,

$V \equiv$ volumetric flow rate,

$n \equiv$ exponent in $\Delta P \propto Re^n$ proportionality, Eq. (3.1),

$B \equiv$ burst bundle subscript,

$R \equiv$ reference bundle subscript.

An estimate of the precision of reported flow rates and Reynolds numbers may be made by computing the $\Delta P_B/\Delta P_R$ ratio, since the pressure profiles indicate this ratio to be about unity below 0 cm. The four B-2 to reference bundle volumetric flow rate ratios, in order of ascending flow rates, are 1.03, 1.01, 1.00, and 1.00. The ratios of reference to B-2

Reynolds numbers are, respectively, 0.98, 0.96, 1.00, and 1.00. Assuming that $N_B \cong N_R \cong 1.8$, the corresponding $\Delta P_B/\Delta P_R$ ratios are 1.06, 1.01, 1.00, and 1.00, respectively. As significant uncertainties exist in the lower-magnitude flow rates, this trend of $\Delta P_B/\Delta P_R$ ratios approaching unity with ascending flow rates is consistent with previous comments made on this topic in Sect. 2.4. These rough comparisons between the $\Delta P_B/\Delta P_R$ ratios of unity predicted by the pressure profile data against the corresponding calculated ratios (predicted with Reynolds number and flow rate data) point out that such discrepancies exist, are significant, and are on an order of magnitude consistent with data precision limits. (See Sect. 3.2.)

When the general axial averaged pressure profiles of B-2 in Figs. 3.34 through 3.37 are compared against flow area restriction profiles, two features are striking relative to similar B-1 comparisons in Figs. 3.25 through 3.28. First is the smoothness of B-2's axial pressure and restriction profiles (contrasted with B-1's) in the 20- to 55-cm zone. Up to about 60 cm (comparing Figs. 3.26 and 3.35, 3.27, and 3.36), both bundles seem to cause about the same permanent pressure losses to the system. However, the average integrated pressure losses (i.e., the differential losses at 96 cm) for B-2 are significantly larger than the corresponding B-1 measurements. These and other differential pressure measurements above 60 cm indicate that the severe B-2 restriction at about 76 cm caused the corresponding shroud 1 taps to register significantly larger pressures than they did for the B-1 tests. Also, while Tables 3.1 to 3.4 indicate that the B-2 restriction at 76 cm was somewhat larger than its B-1 counterpart, the difference is not enough to suggest that the B-2 profile should be more difficult to predict than the B-1 profile in this zone. Yet, as will be shown in Sect. 5.5, this was the case. The underestimation of pressures in this zone by the B-2 COBRA-IV model, as well as the fact that the same B-2 pressure measurements are larger relative to those for B-1, naturally leads to a questioning of the validity of these particular pressure measurements. This type of behavior can be construed to be characteristic of the pressure taps recording a nonpiezometric component; the shape of the B-2 blockage at 76 cm may

well have diverted flow to such a degree as to jeopardize measurements made there and downstream.

Figure 3.38 plots average integrated pressure losses against the corresponding Reynolds numbers for B-2 and the reference bundle, both in shroud 1. As was the case for the similar B-1 comparison (Fig. 3.29), the B-2 pressure data appear to show the expected functional dependence on Reynolds numbers.

3.6 Reference Bundle Tests in Shroud 2

Differential axial pressure loss profiles for the reference bundle in shroud 2 are given in Figs. 3.39 through 3.45. Overall, these data appear to be of about the same quality as the corresponding shroud 1 data. The slopes between the grid spacers are relatively linear; some slight nonlinearities exist immediately downstream from the grids and are, of course, expected. Figure 3.45 suggests that this grid disturbance zone may persist for about 20 cm (12 hydraulic diameters) downstream of the grid. Also visible is some slight separation of pressure measurements made on opposing sides of the shroud. This separation is attributed to the drifting of flow rates as the data were taken.

3.7 B-2 Tests in Shroud 2

Figures 3.46 through 3.50 are plots of differential pressure loss profiles taken with the B-2/shroud 2 assembly. As may be seen by comparing these figures with their shroud 1 counterparts (Figs. 3.30 through 3.33), the shapes of the plots are similar. Localized profile features due to deformities near the pressure taps are even reproduced with fairly consistent detail (i.e., pressure spikes due to deformities in tube 15 at 19.5 cm and in tubes 2, 3, and 4 at 76.2; see Tables 3.3 and 3.4). This result is not totally unexpected, since the shroud 2 walls are not much further away from B-2 than was the case for shroud 1.

Figures 3.51 through 3.55 present comparisons between B-2 and reference bundle axial pressure loss profiles (averaged) against B-2 flow channel restriction profiles. As shown, the agreement between the B-2 and reference bundle profiles below 0 cm is very good, implying a $\Delta P_B/\Delta P_R$ ratio of unity [using Eq. (3.2) notation]. The five B-2 to reference bundle volumetric flow rate ratios corresponding to Figs. 3.51 to 3.55 are 1.00, 0.98, 1.01, 1.01, and 1.01. The respective reference bundle to B-2 Reynolds number ratios are 1.02, 1.02, 1.00, 0.96, and 1.00. Computing the $\Delta P_B/\Delta P_R$ ratios for $N_B \cong N_R \cong 1.8$ [see Eq. (3.2)] corresponding to Figs. 3.51 to 3.55 with these values, one obtains $\Delta P_B/\Delta P_R$ ratios of 1.00, 0.96, 1.02, 1.01, and 1.02. While these calculated $\Delta P_B/\Delta P_R$ ratios do not approach unity with increasing flow rates as closely as did similar ratios for previous data sets, these results are still explainable in terms of the precision limits given in Sect. 2.4. Hence, when the $\Delta P_B/\Delta P_R$ ratios of unity implied by the overlain B-2 and reference bundle pressure loss profiles are compared with corresponding $\Delta P_B/\Delta P_R$ ratios based on flow rate and Reynolds number determinations, discrepancies are observable which are indicative of the precision to which volumetric flow rates were measured.

Figure 3.56 compares the average integrated pressure losses of the B-2 and reference bundles in shroud 2 against the corresponding Reynolds numbers; the curve fits to these data are reasonably good.

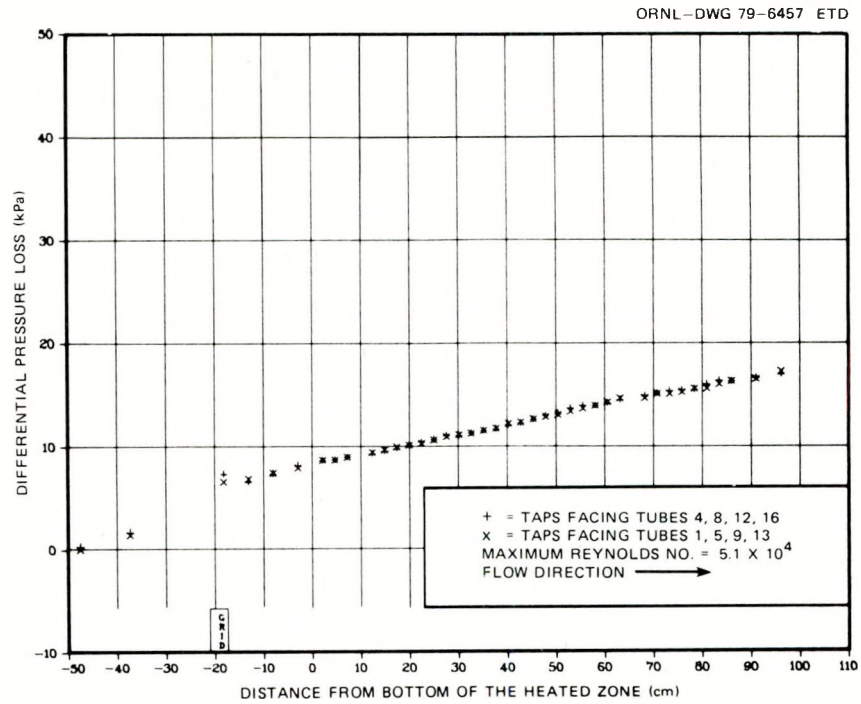


Fig. 3.1. First single-grid test axial pressure loss profiles;
flow rate = 9.2 liters/sec.

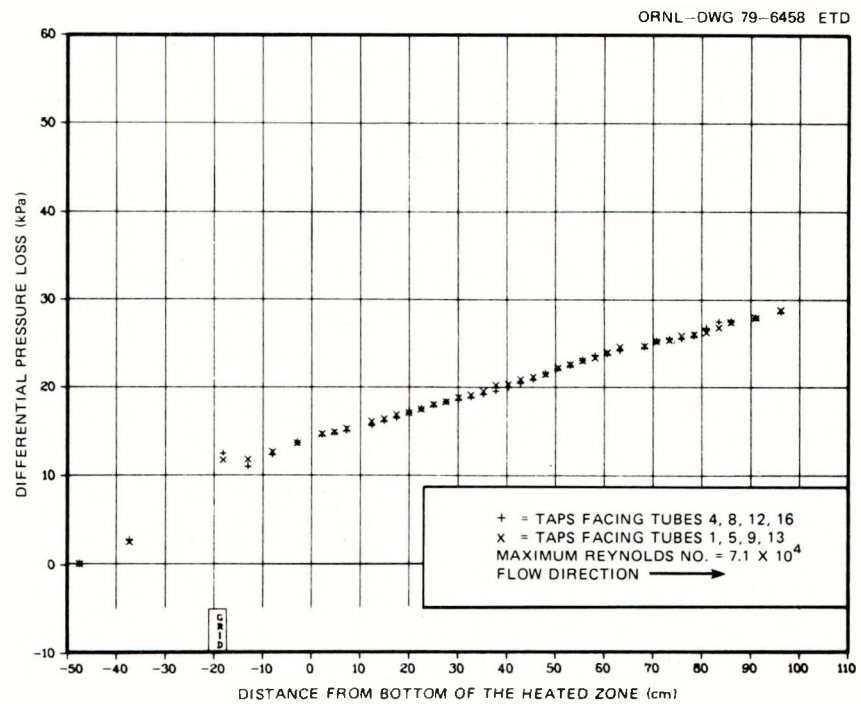


Fig. 3.2. First single-grid test axial pressure loss profiles;
flow rate = 12.3 liters/sec.

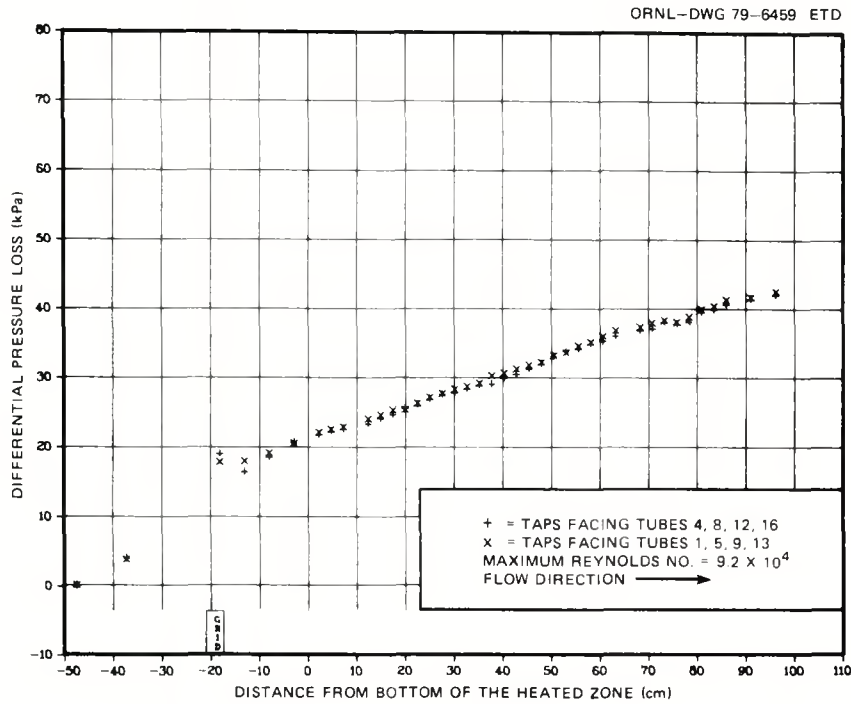


Fig. 3.3. First single-grid test axial pressure loss profiles;
flow rate = 15.5 liters/sec.

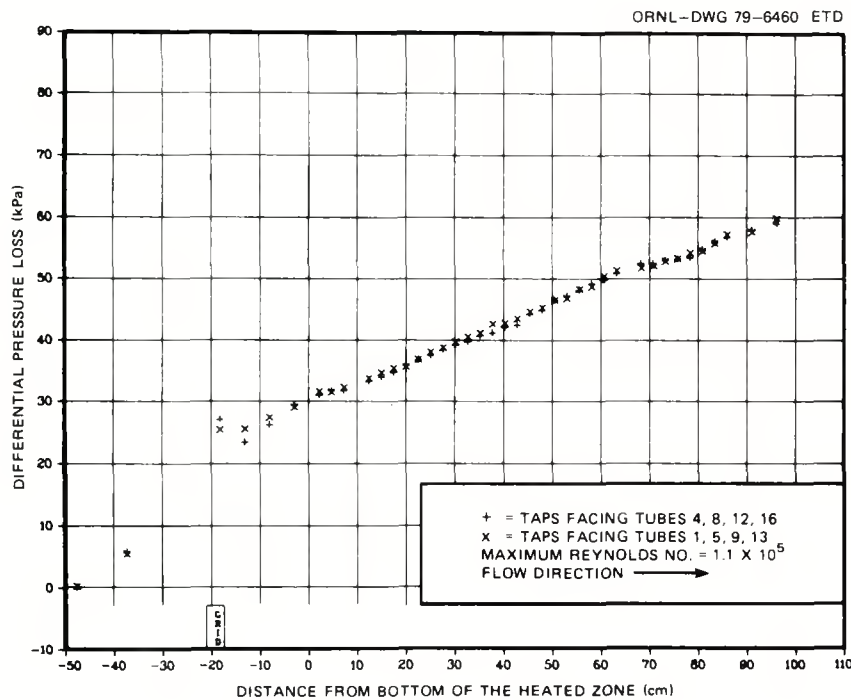


Fig. 3.4. First single-grid test axial pressure loss profiles;
flow rate = 18.6 liters/sec.

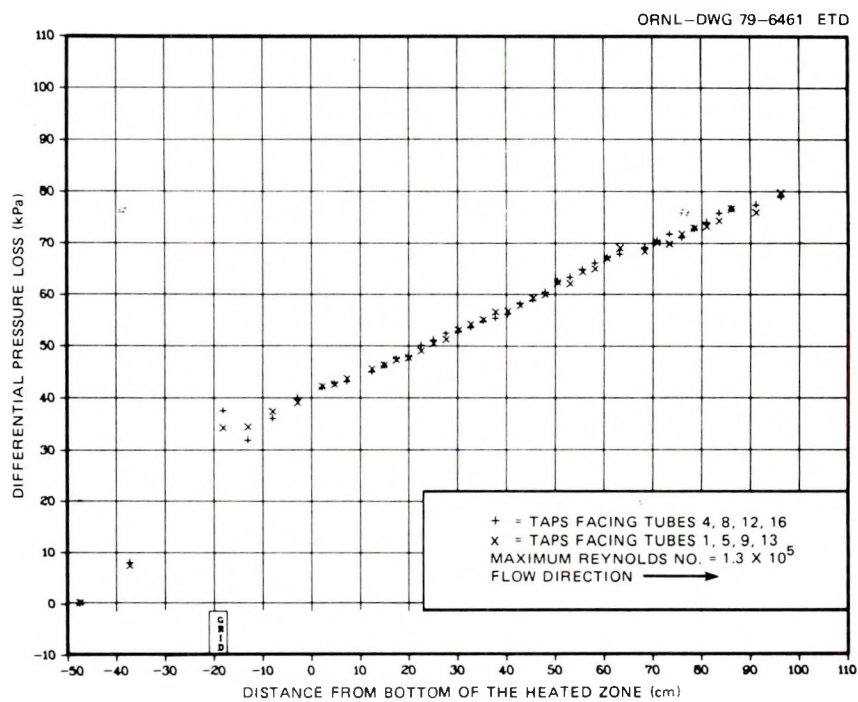


Fig. 3.5. First single-grid test axial pressure loss profiles; flow rate = 21.9 liters/sec.

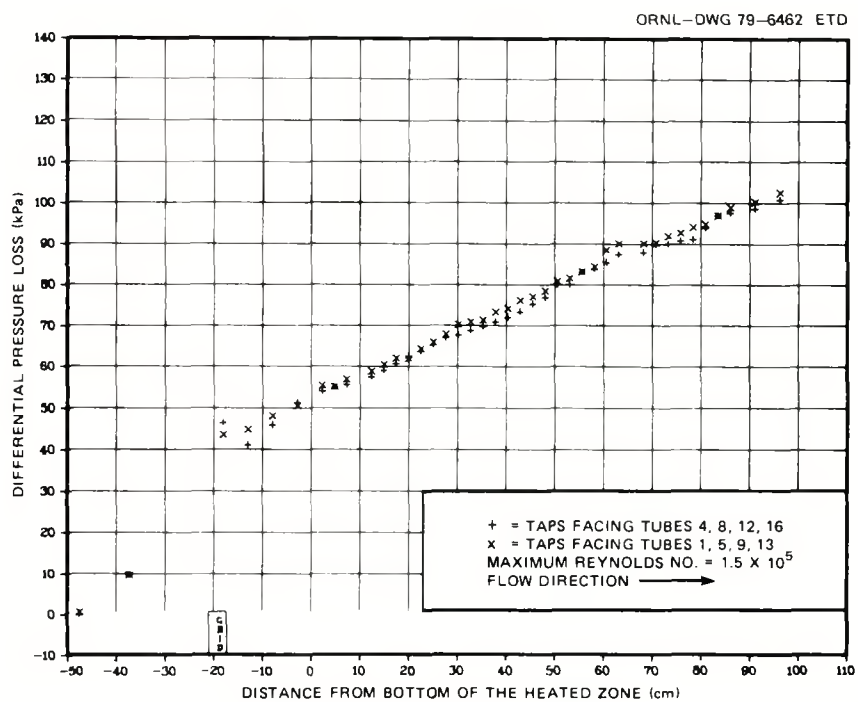


Fig. 3.6. First single-grid test axial pressure loss profiles; flow rate = 25.0 liters/sec.

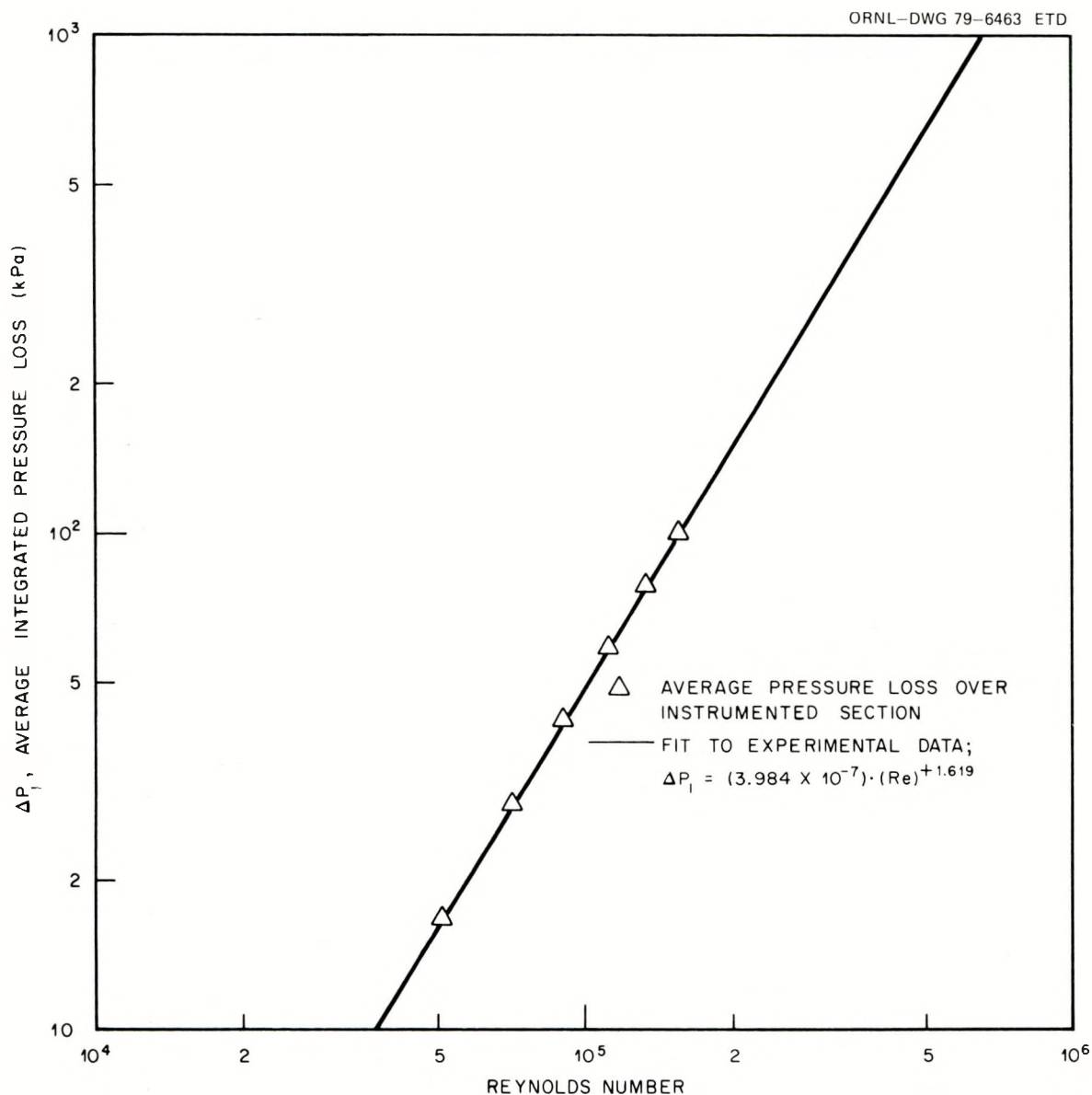


Fig. 3.7. Average integrated pressure losses vs Reynolds number for the first single-grid tests. (Note: The Reynolds numbers used to generate this plot were not limited to two significant figures in contrast with the previously reported values.)

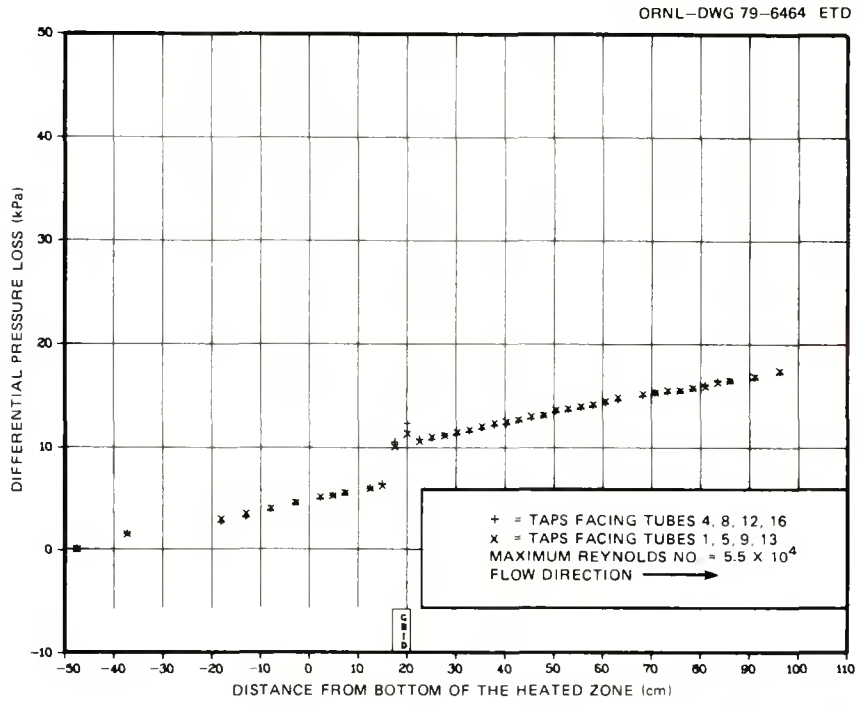


Fig. 3.8. Second single-grid test axial pressure loss profiles; flow rate = 9.36 liters/sec.

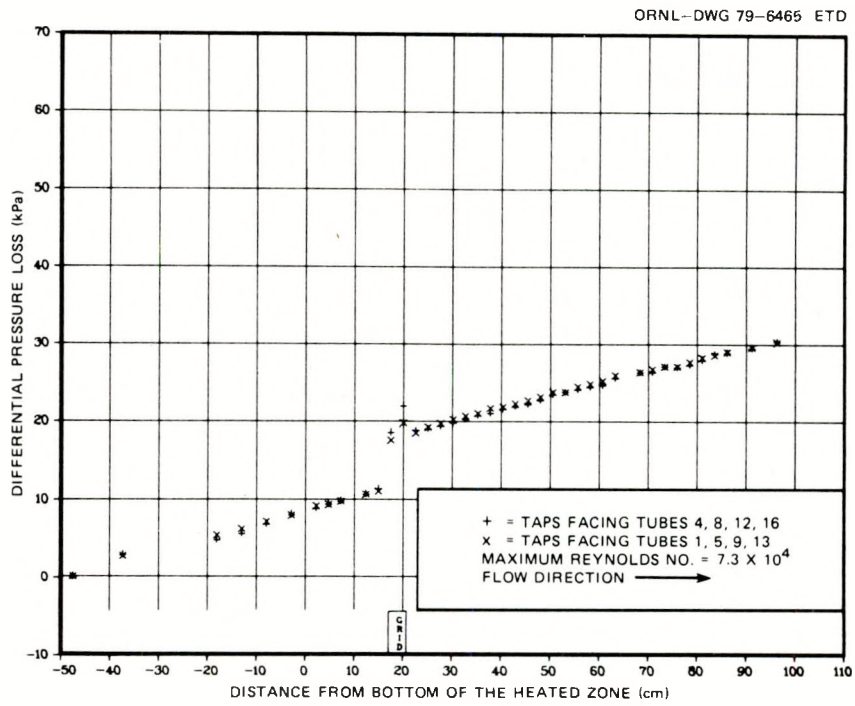


Fig. 3.9. Second single-grid test axial pressure loss profiles; flow rate = 12.7 liters/sec.

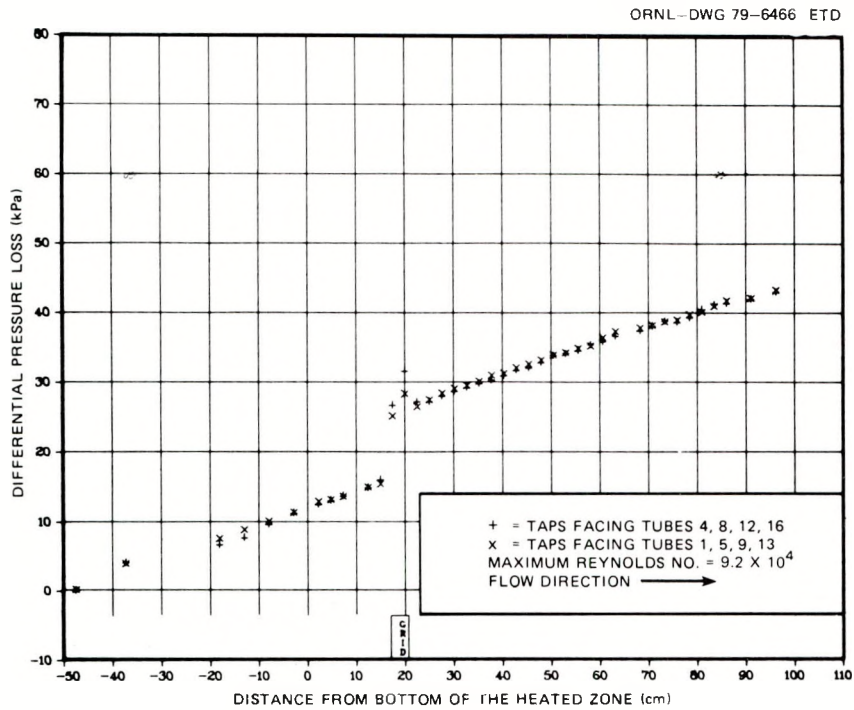


Fig. 3.10. Second single-grid test axial pressure loss profiles;
flow rate = 15.6 liters/sec.

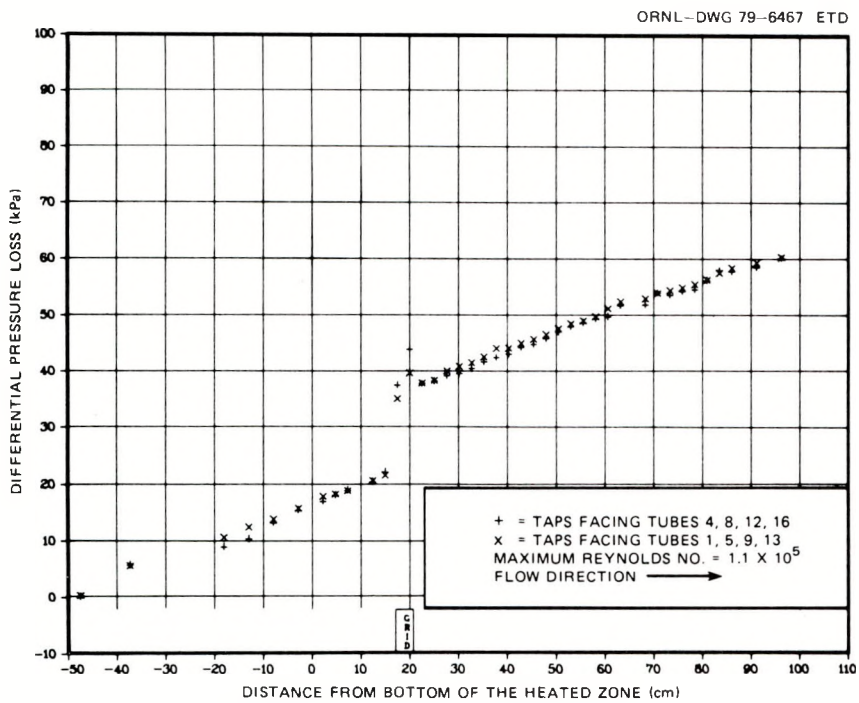


Fig. 3.11. Second single-grid test axial pressure loss profiles;
flow rate = 18.8 liters/sec.

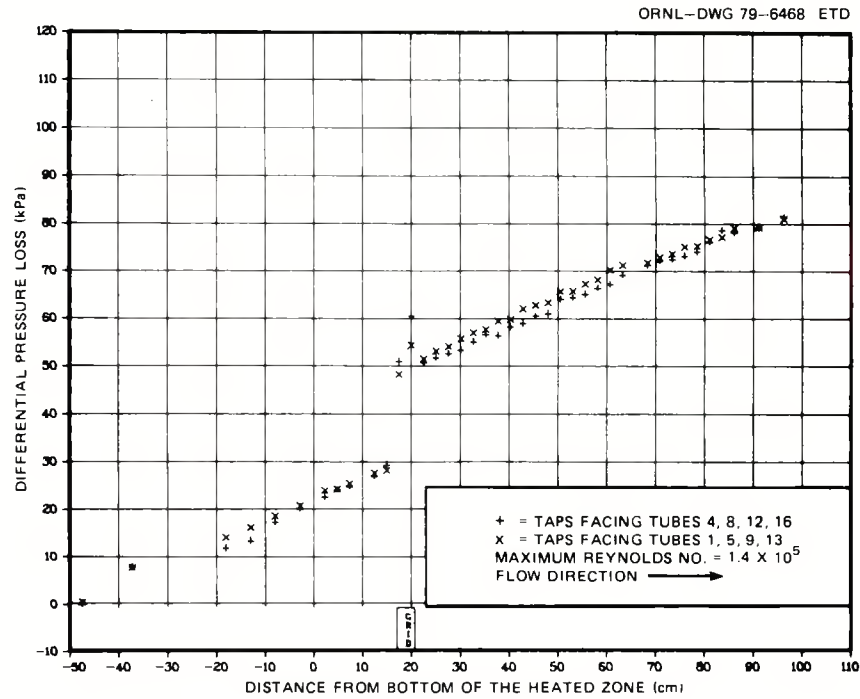


Fig. 3.12. Second single-grid test axial pressure loss profiles; flow rate = 22.1 liters/sec.

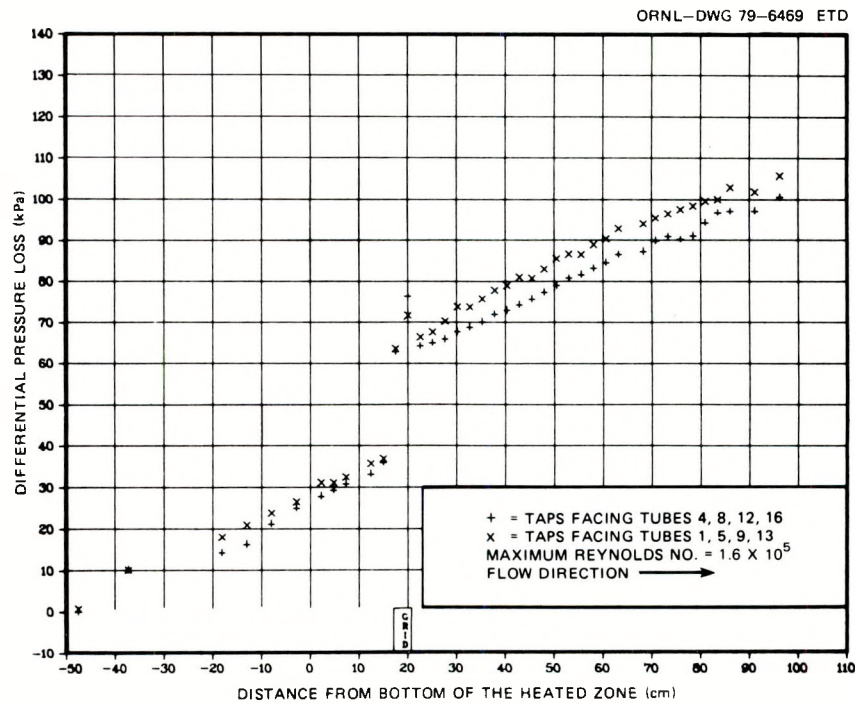


Fig. 3.13. Second single-grid test axial pressure loss profiles; flow rate = 25.3 liters/sec.

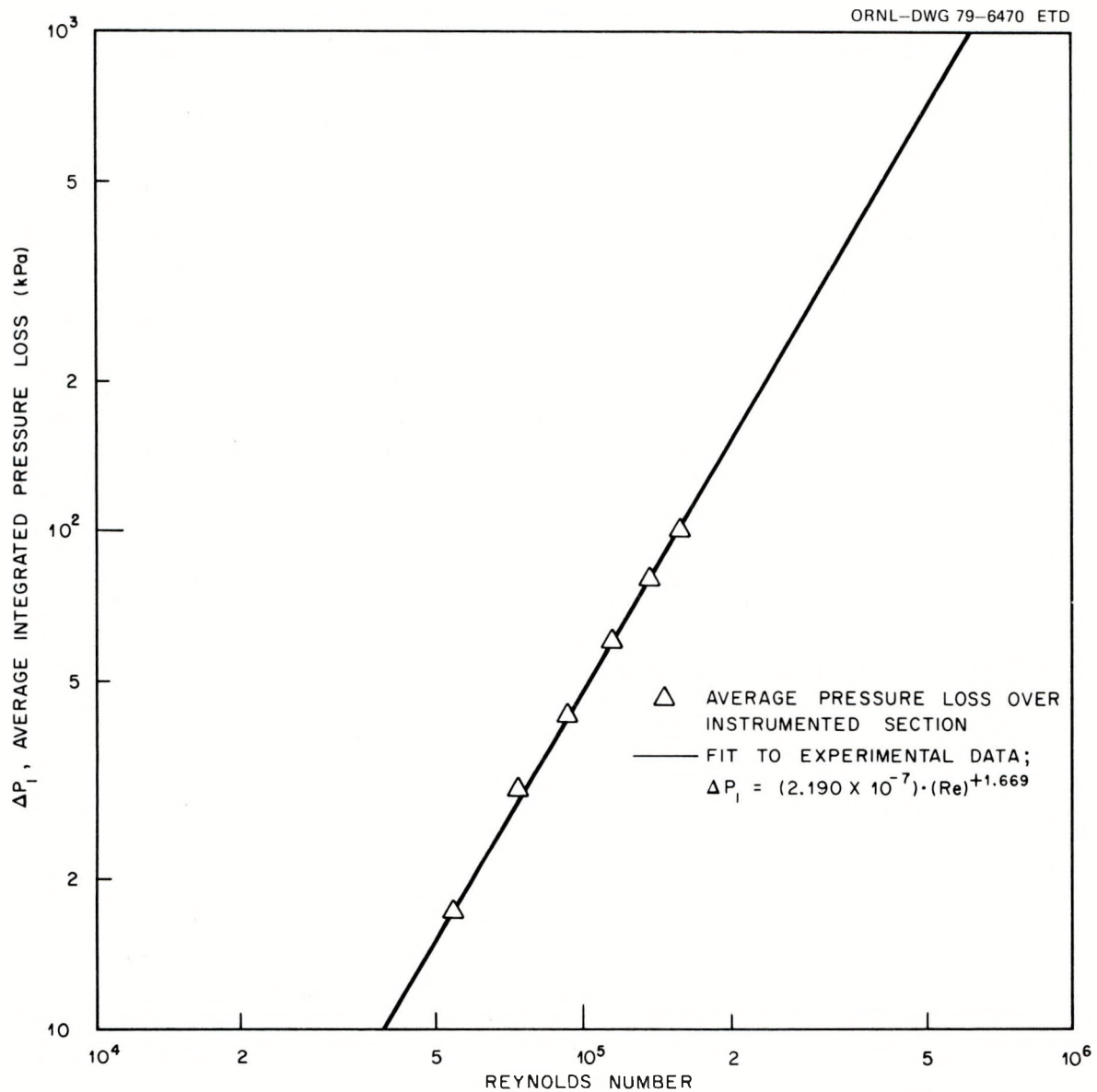


Fig. 3.14. Average integrated pressure losses vs Reynolds number for the second single-grid tests. (Note: The Reynolds numbers used to generate this plot were not limited to two significant figures in contrast with the previously reported values.)

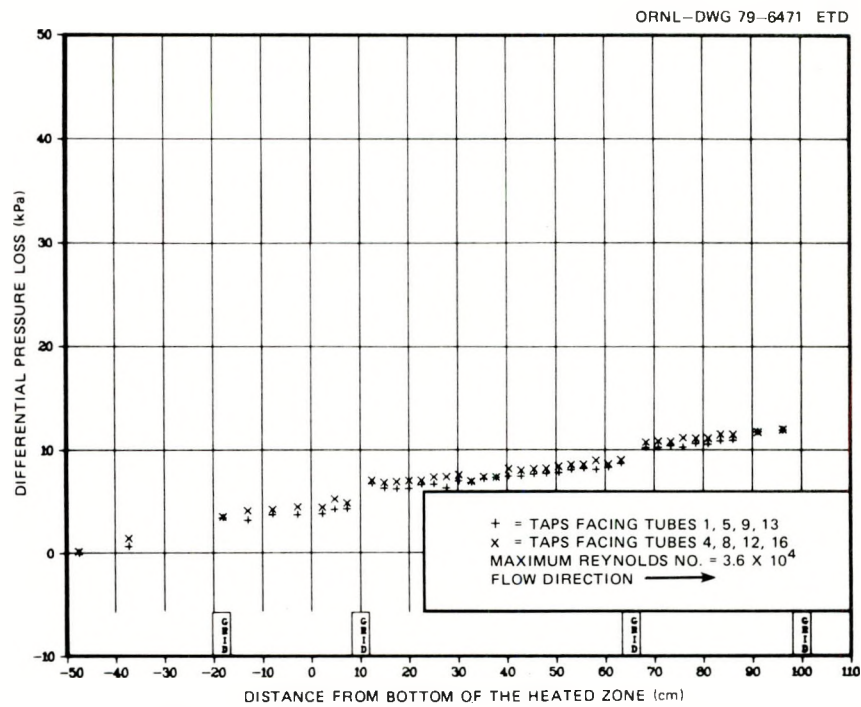


Fig. 3.15. Reference bundle/shroud 1 axial pressure loss profiles;
flow rate = 6.77 liters/sec.

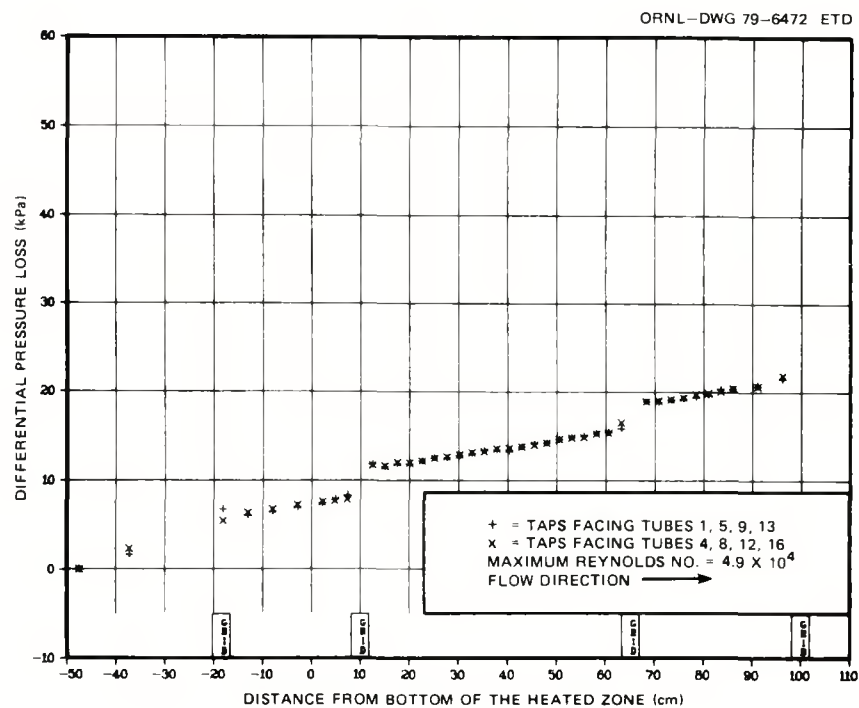


Fig. 3.16. Reference bundle/shroud 1 axial pressure loss profiles;
flow rate = 9.22 liters/sec.

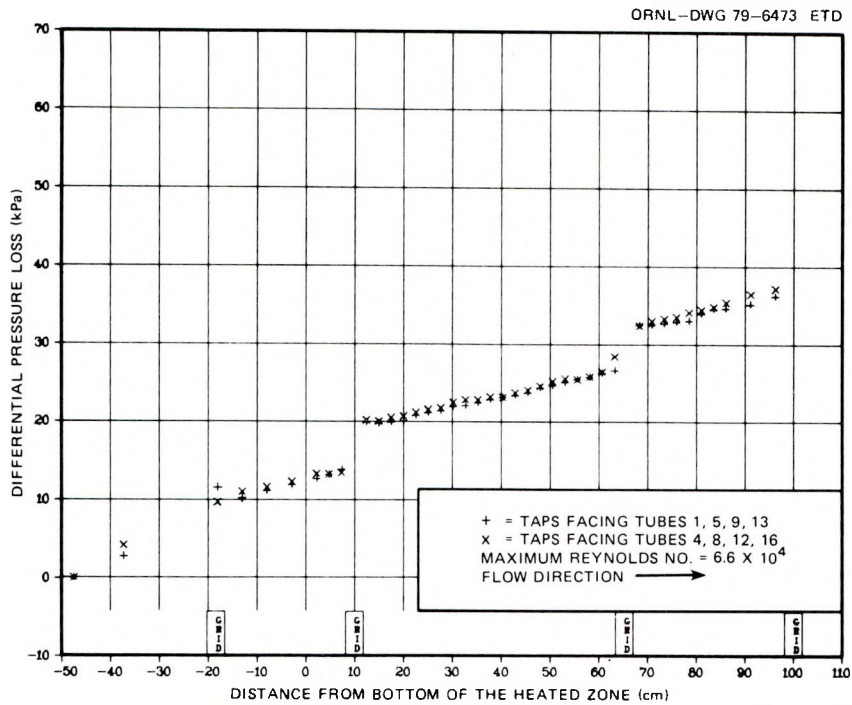


Fig. 3.17. Reference bundle/shroud 1 axial pressure loss profiles; flow rate = 12.5 liters/sec.

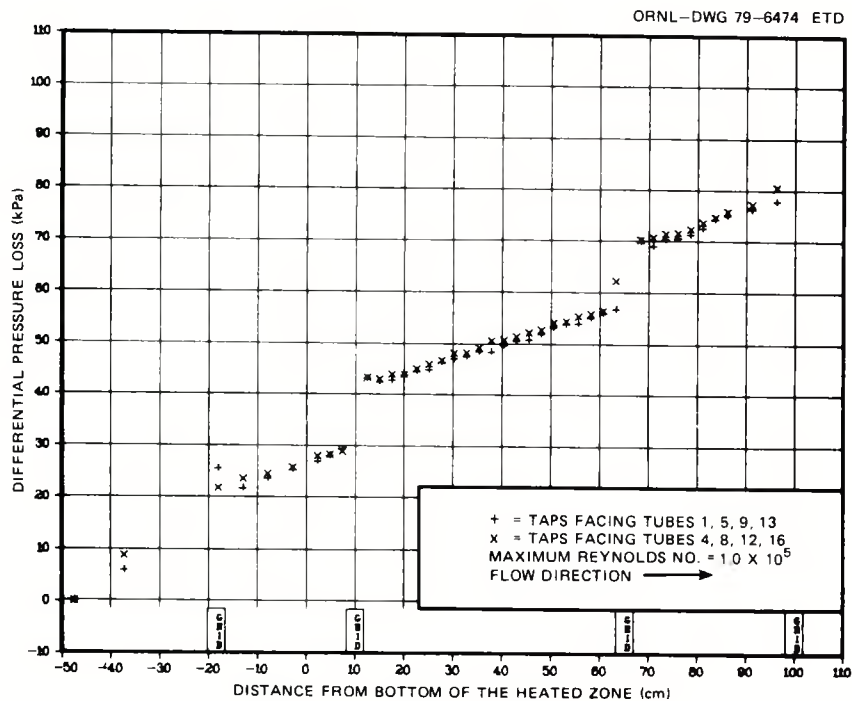


Fig. 3.18. Reference bundle/shroud 1 axial pressure loss profiles; flow rate = 18.9 liters/sec.

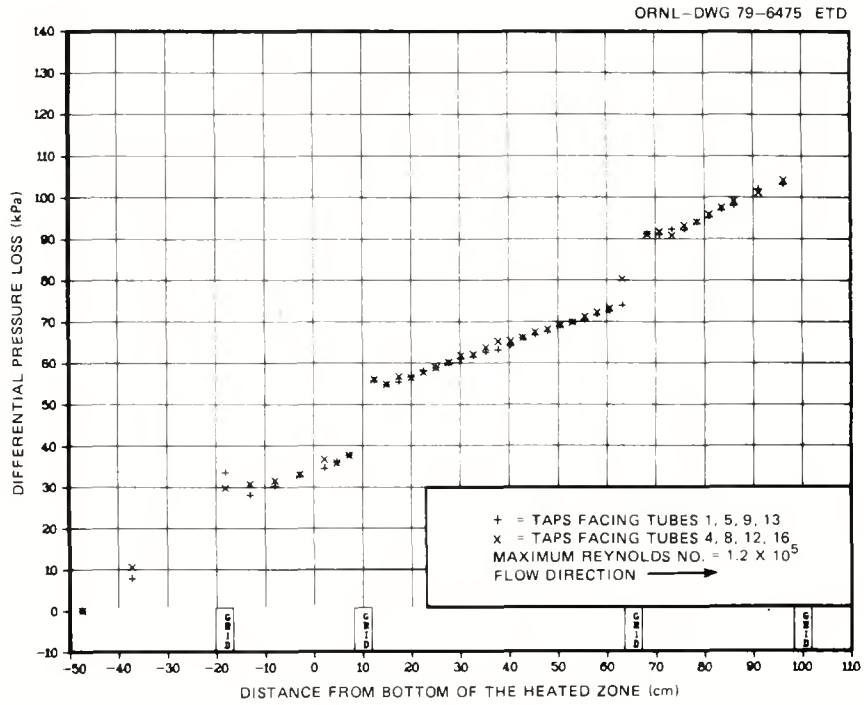


Fig. 3.19. Reference bundle/shroud 1 axial pressure loss profiles; flow rate = 22.1 liters/sec.

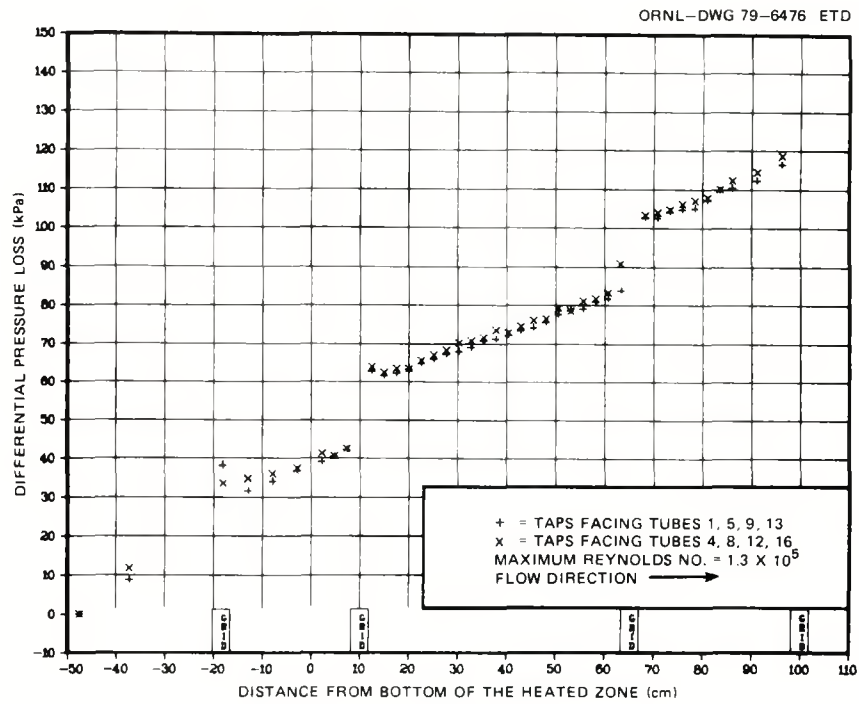


Fig. 3.20. Reference bundle/shroud 1 axial pressure loss profiles; flow rate = 23.9 liters/sec.

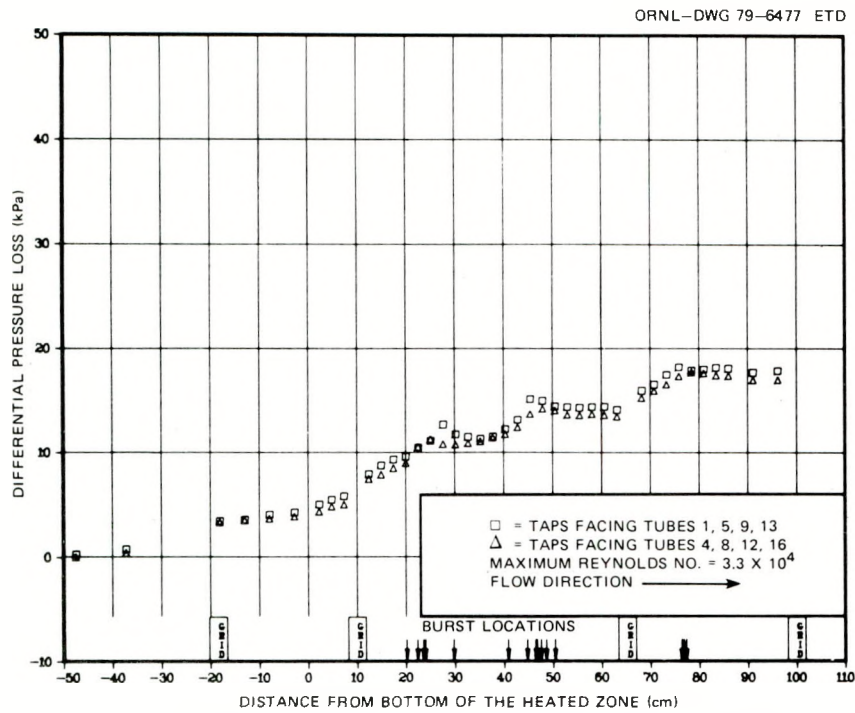


Fig. 3.21. B-1/shroud 1 axial pressure loss profiles; flow rate = 6.31 liters/sec.

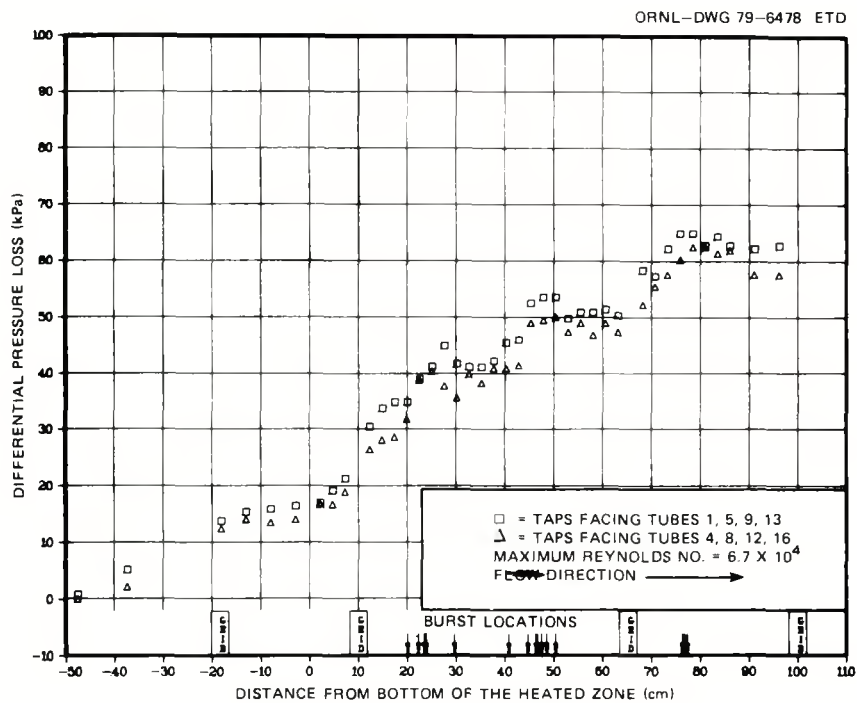


Fig. 3.22. B-1/shroud 1 axial pressure loss profiles; flow rate = 12.8 liters/sec.

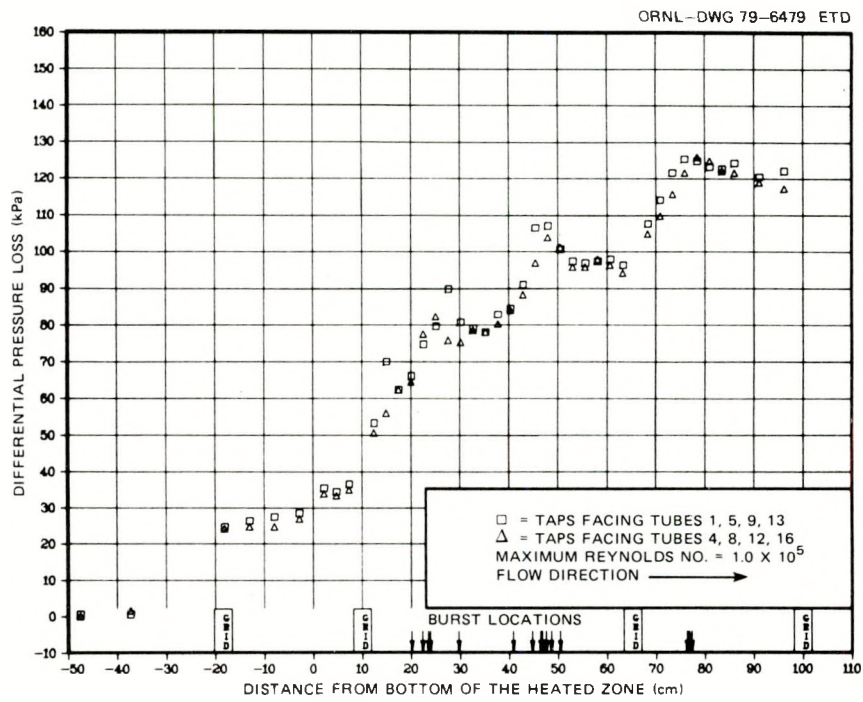


Fig. 3.23. B-1/shroud 1 axial pressure loss profiles; flow rate = 19.1 liters/sec.

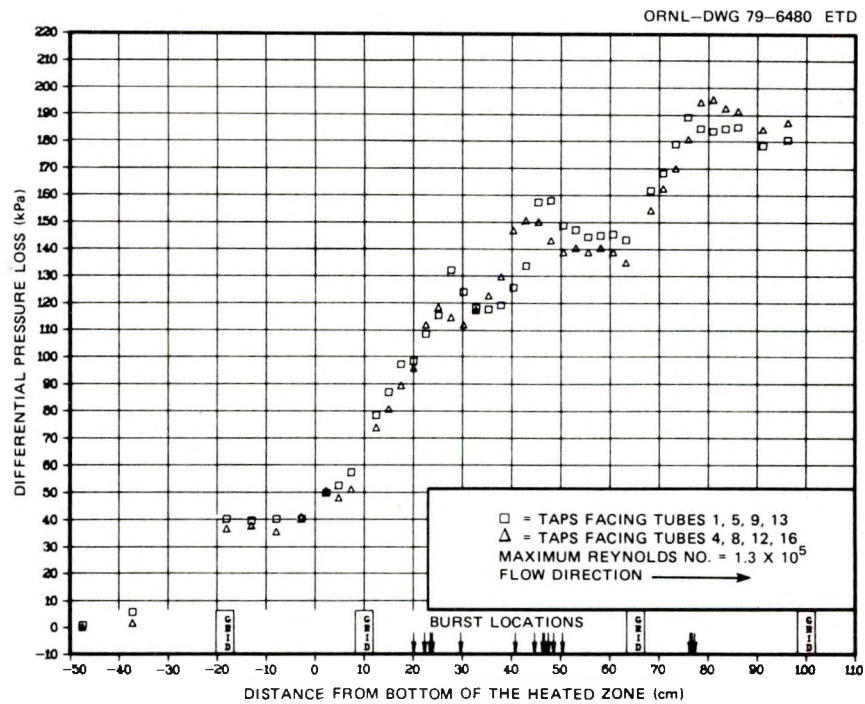


Fig. 3.24. B-1/shroud 1 axial pressure loss profiles; flow rate = 23.9 liters/sec.

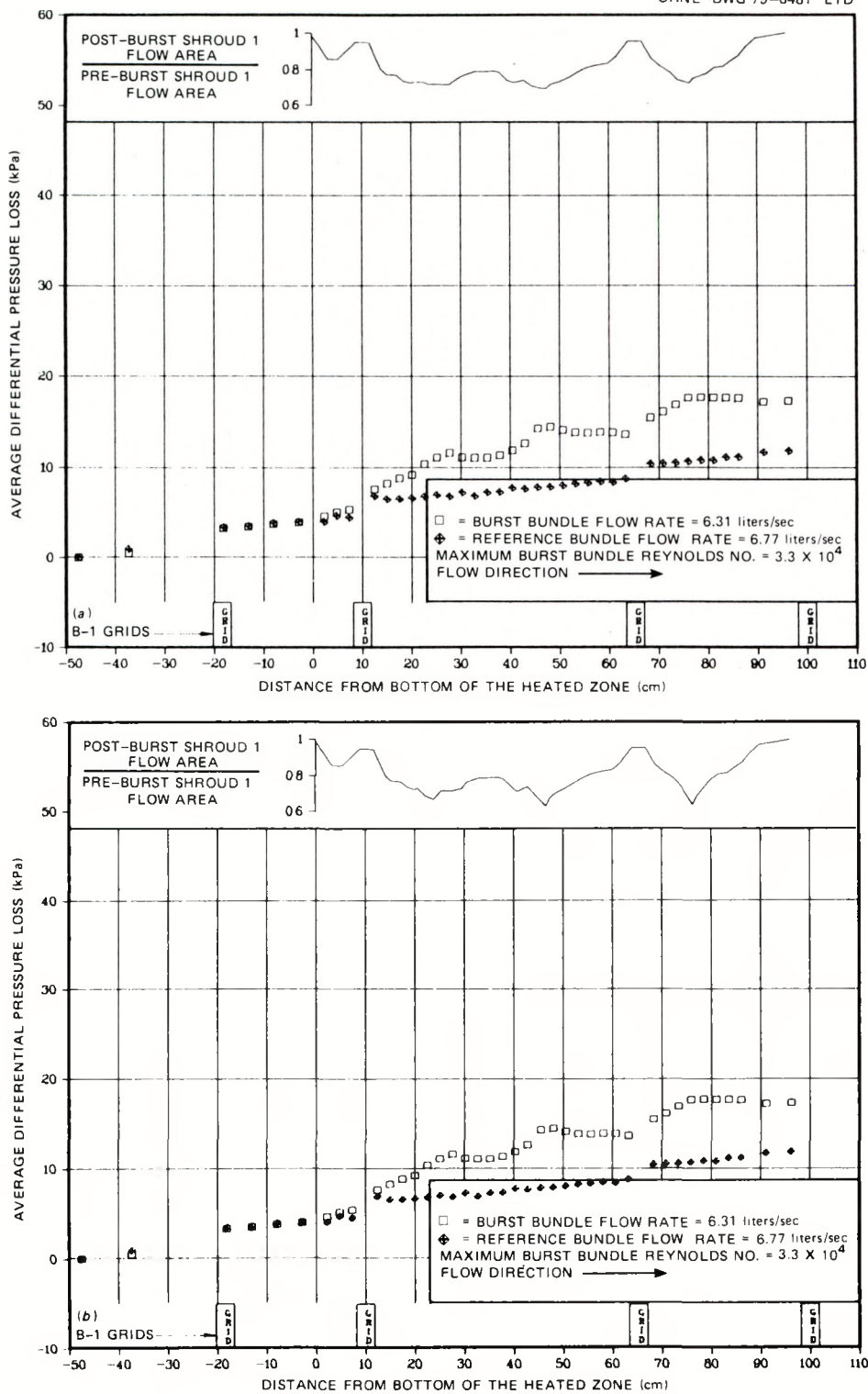


Fig. 3.25. B-1/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 6.31 liters/sec vs: (a) the minimum B-1 restriction profile (b) the maximum B-1 restriction profile.

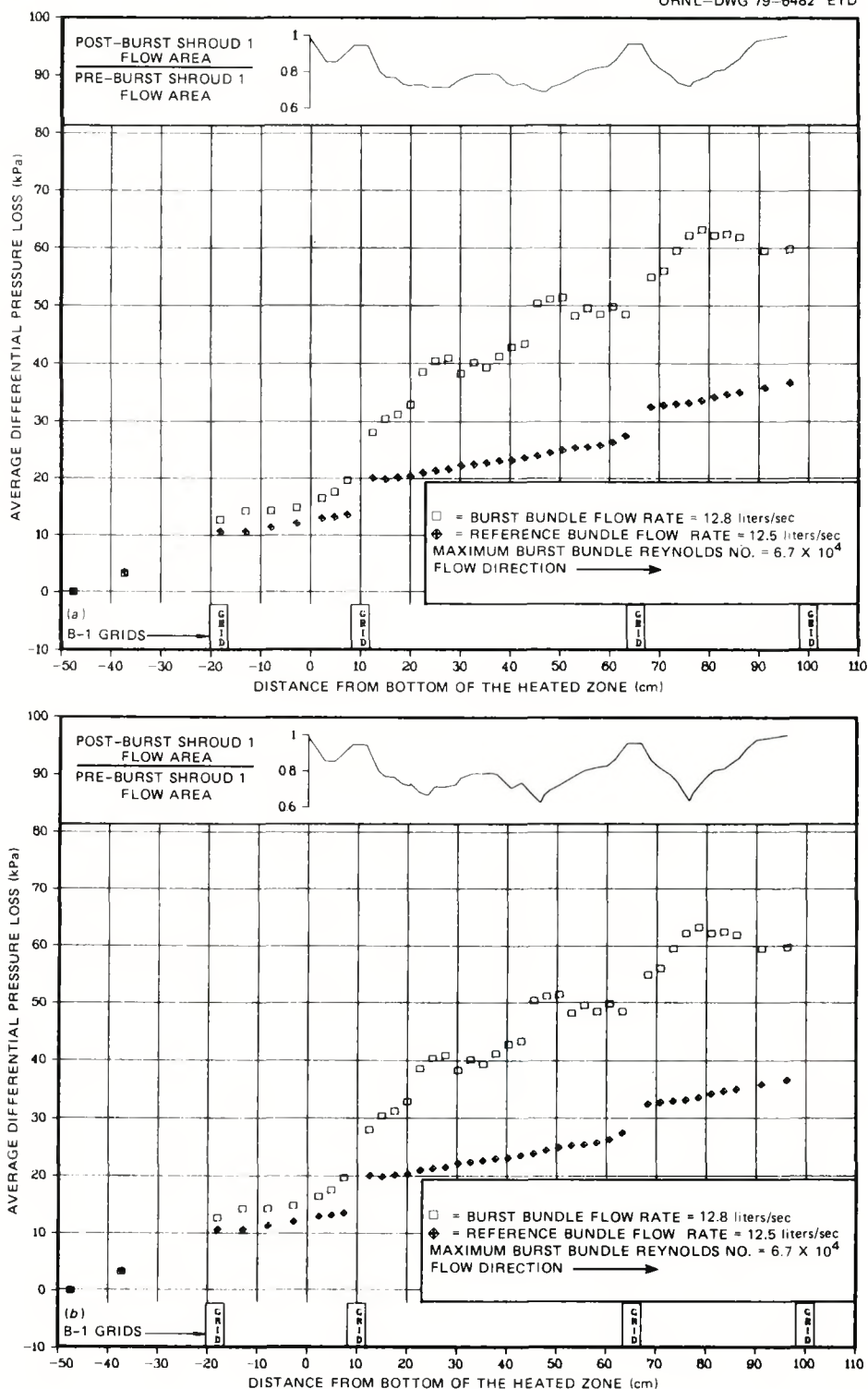


Fig. 3.26. B-1/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 12.8 liters/sec vs: (a) the minimum B-1 restriction profile (b) the maximum B-1 restriction profile.

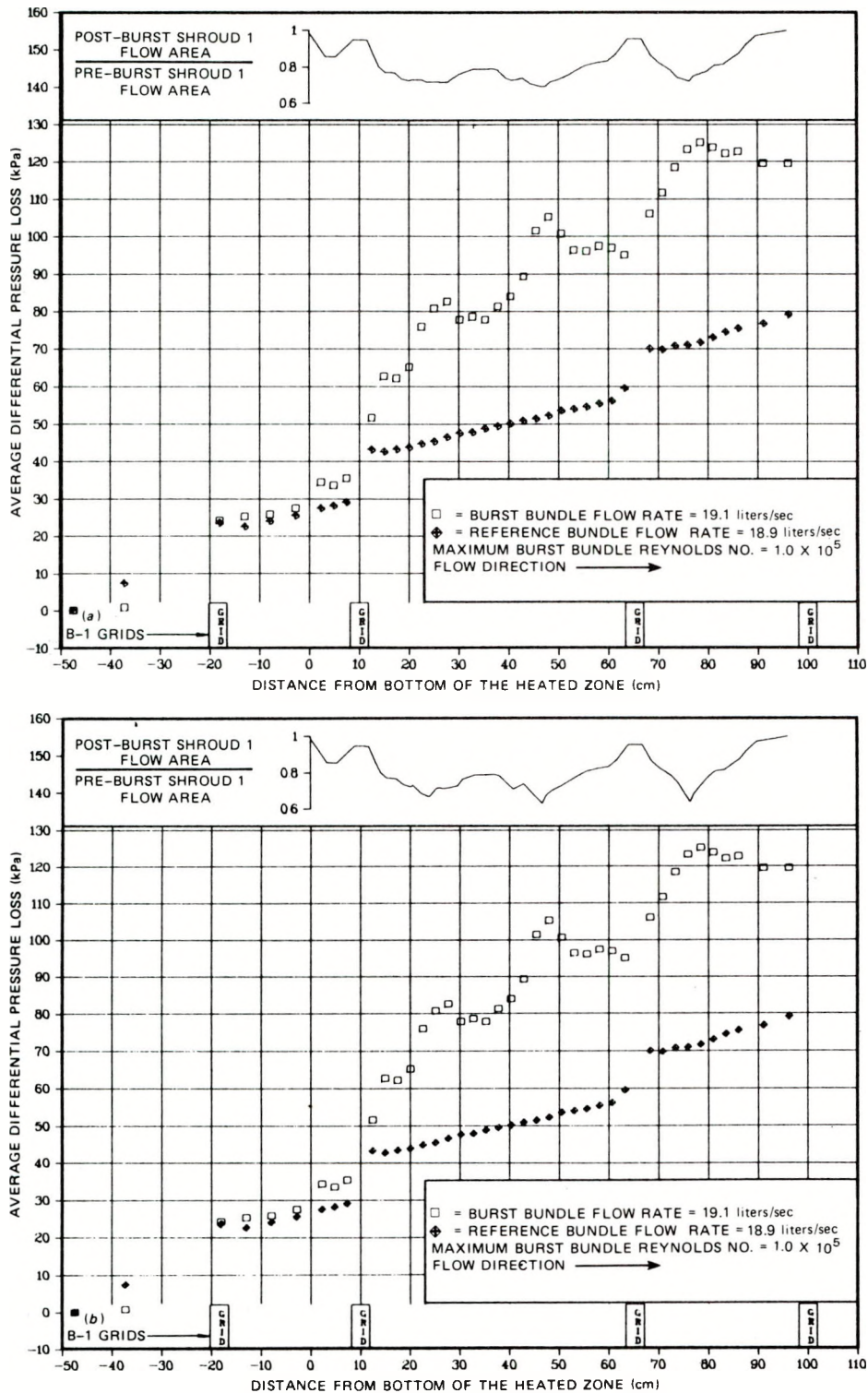


Fig. 3.27. B-1/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 19.1 liters/sec vs: (a) the minimum B-1 restriction profile (b) the maximum B-1 restriction profile.

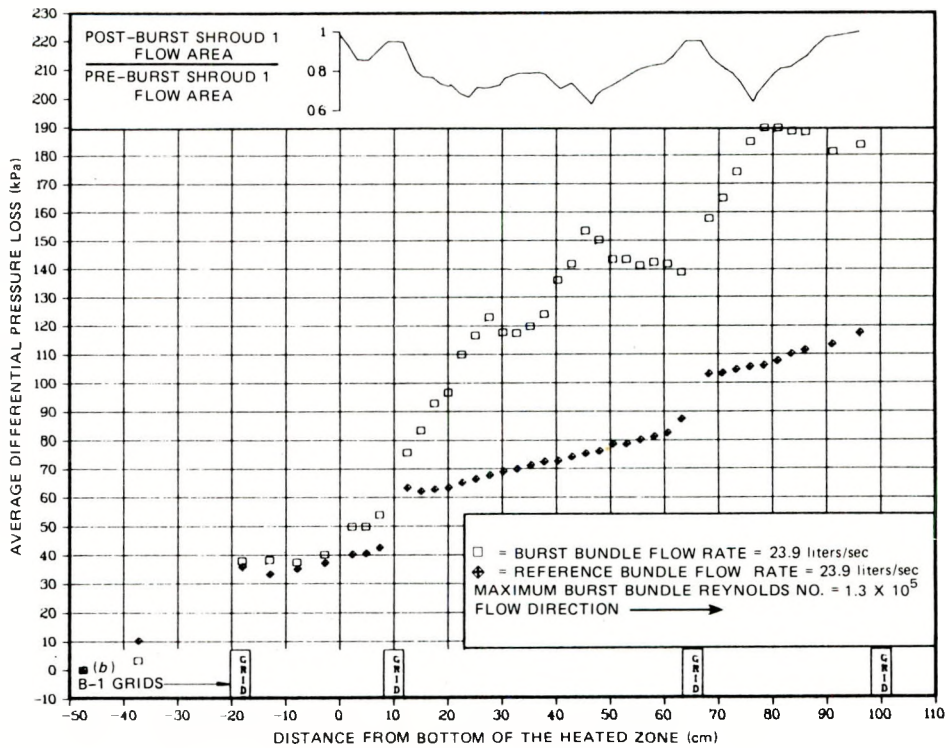
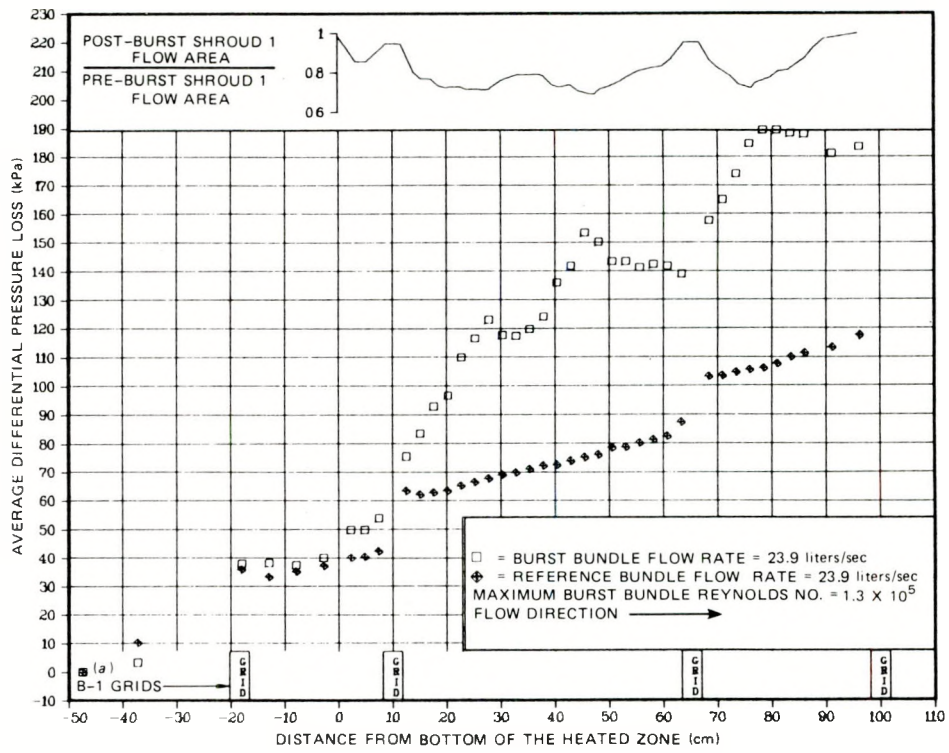


Fig. 3.28. B-1/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 23.9 liters/sec vs: (a) the minimum B-1 restriction profile (b) the maximum B-1 restriction profile.

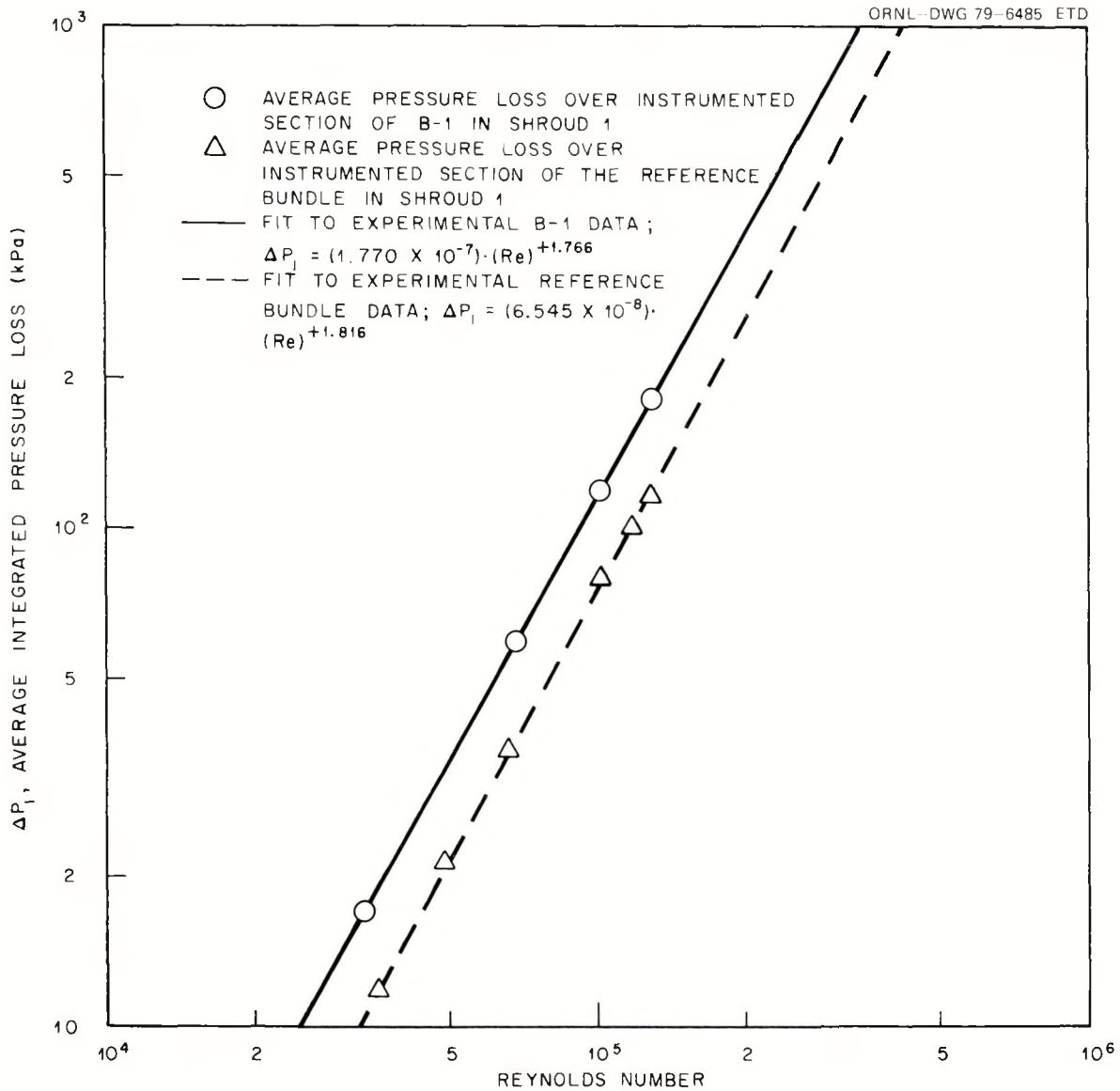


Fig. 3.29. Average integrated pressure losses vs Reynolds number for the B-1 and reference bundle tests in shroud 1. (Note: The Reynolds numbers used to generate this plot were not limited to two significant figures in contrast with the previously reported values.)

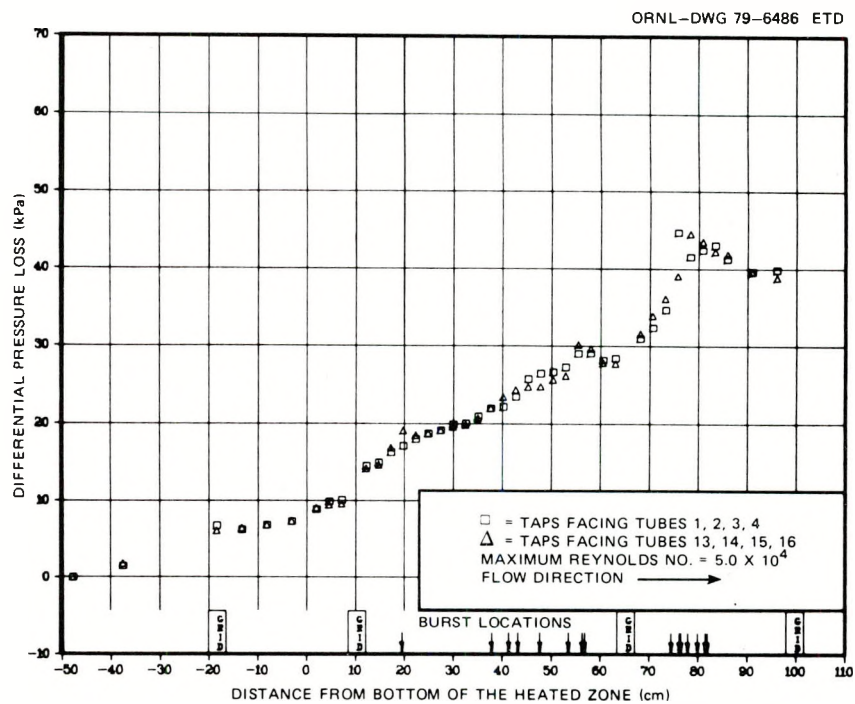


Fig. 3.30. B-2/shroud 1 axial pressure loss profiles; flow rate = 9.46 liters/sec.

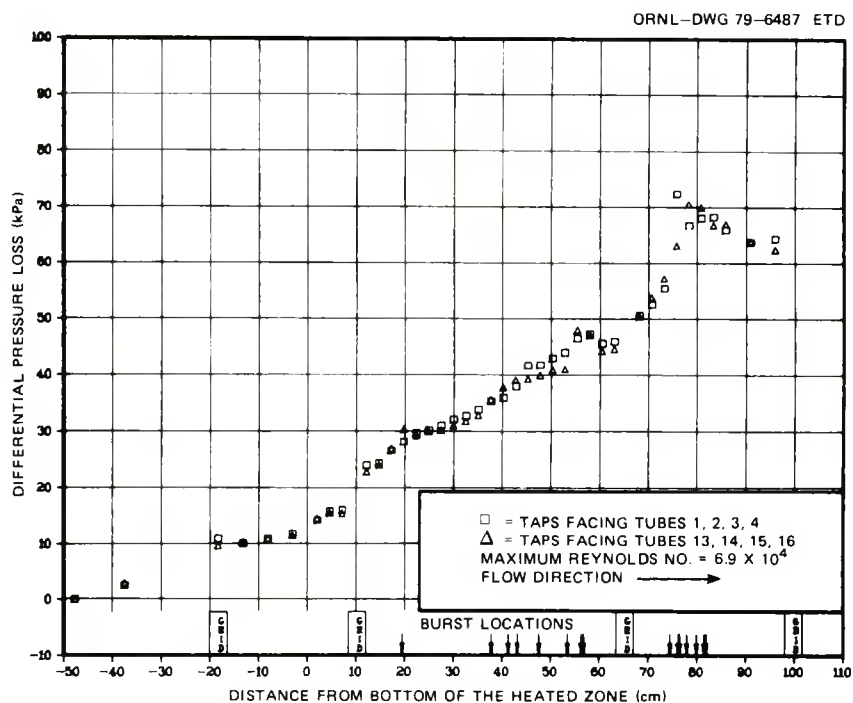


Fig. 3.31. B-2/shroud 1 axial pressure loss profiles; flow rate = 12.6 liters/sec.

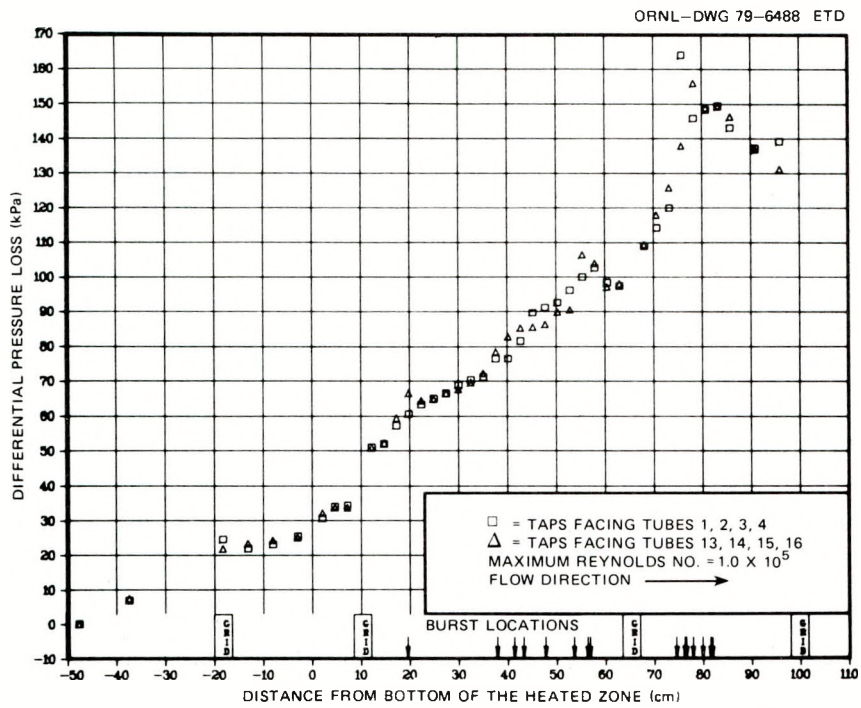


Fig. 3.32. B-2/shroud 1 axial pressure loss profiles; flow rate = 18.9 liters/sec.

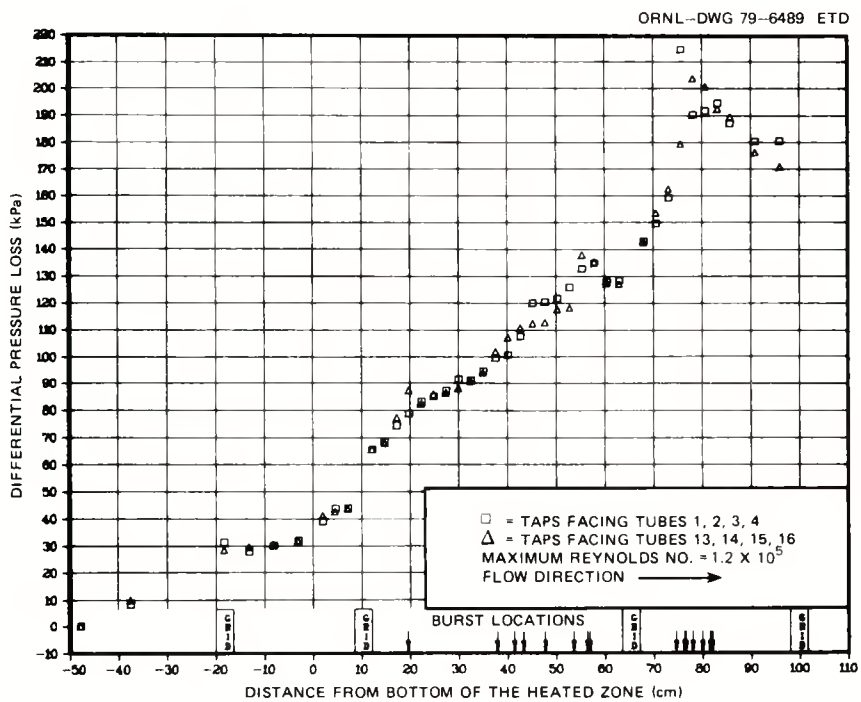


Fig. 3.33. B-2/shroud 1 axial pressure loss profiles; flow rate = 22.0 liters/sec.

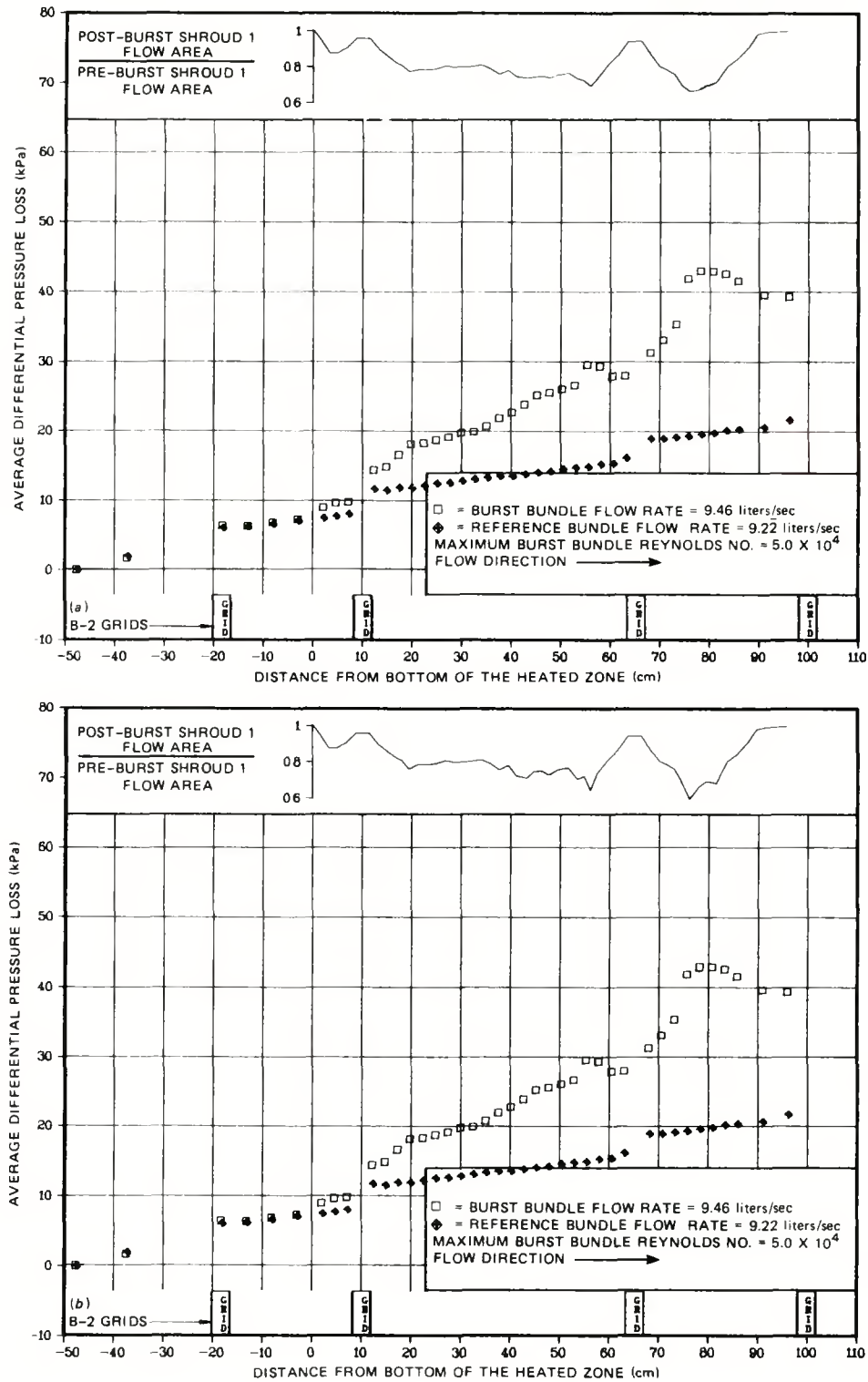


Fig. 3.34. B-2/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 9.46 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

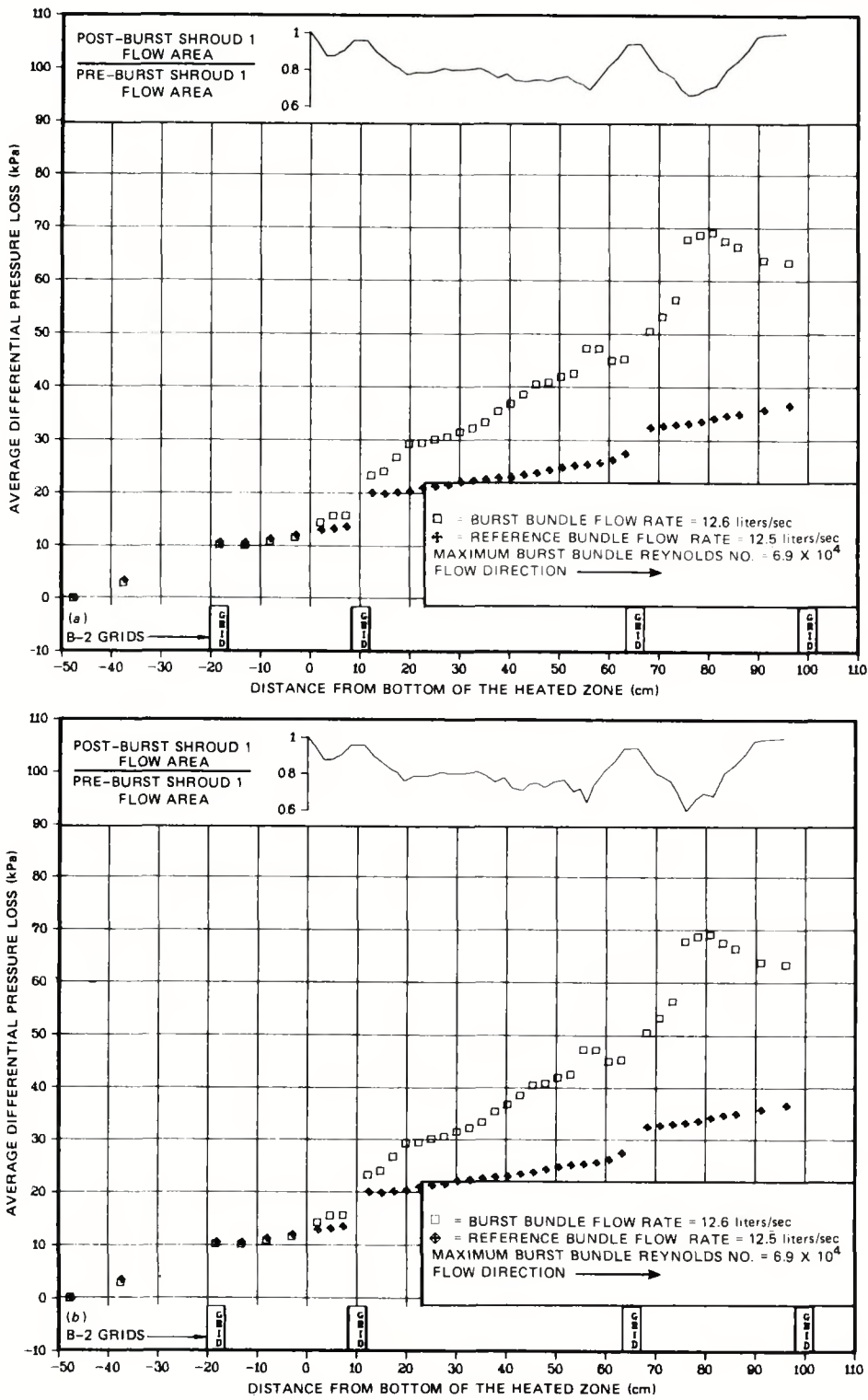


Fig. 3.35. B-2/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 12.6 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

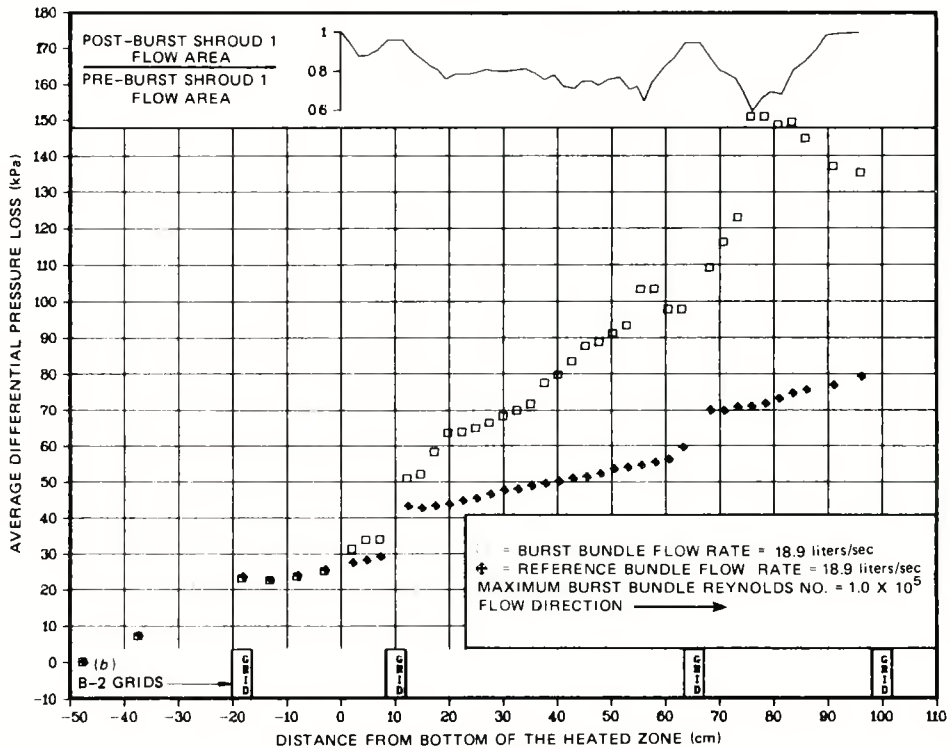
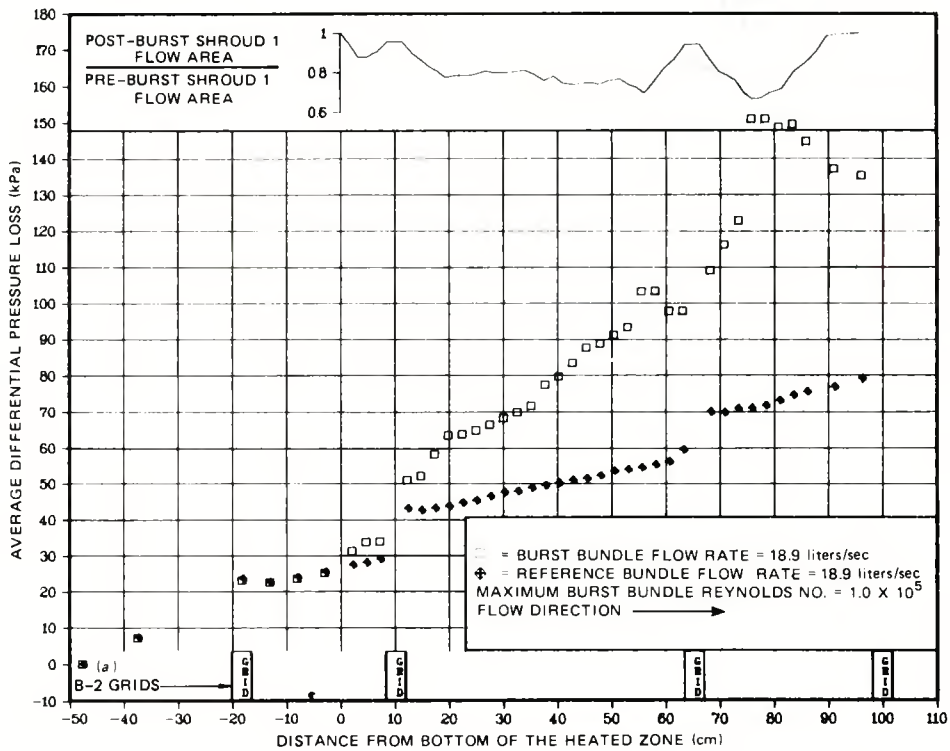


Fig. 3.36. B-2/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 18.9 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

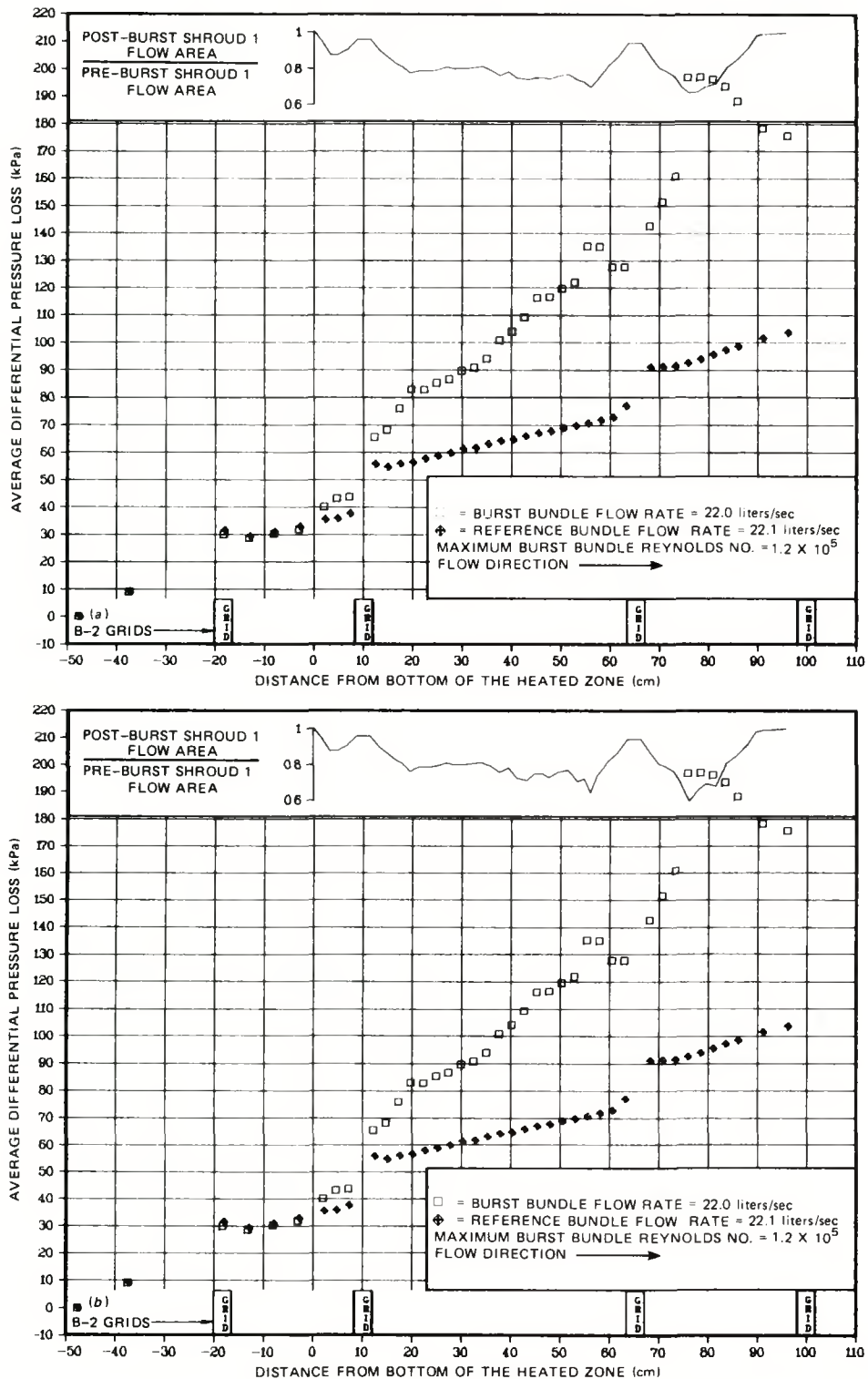


Fig. 3.37. B-2/shroud 1 and reference bundle/shroud 1 axial pressure loss profiles; burst bundle flow rate = 22.0 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

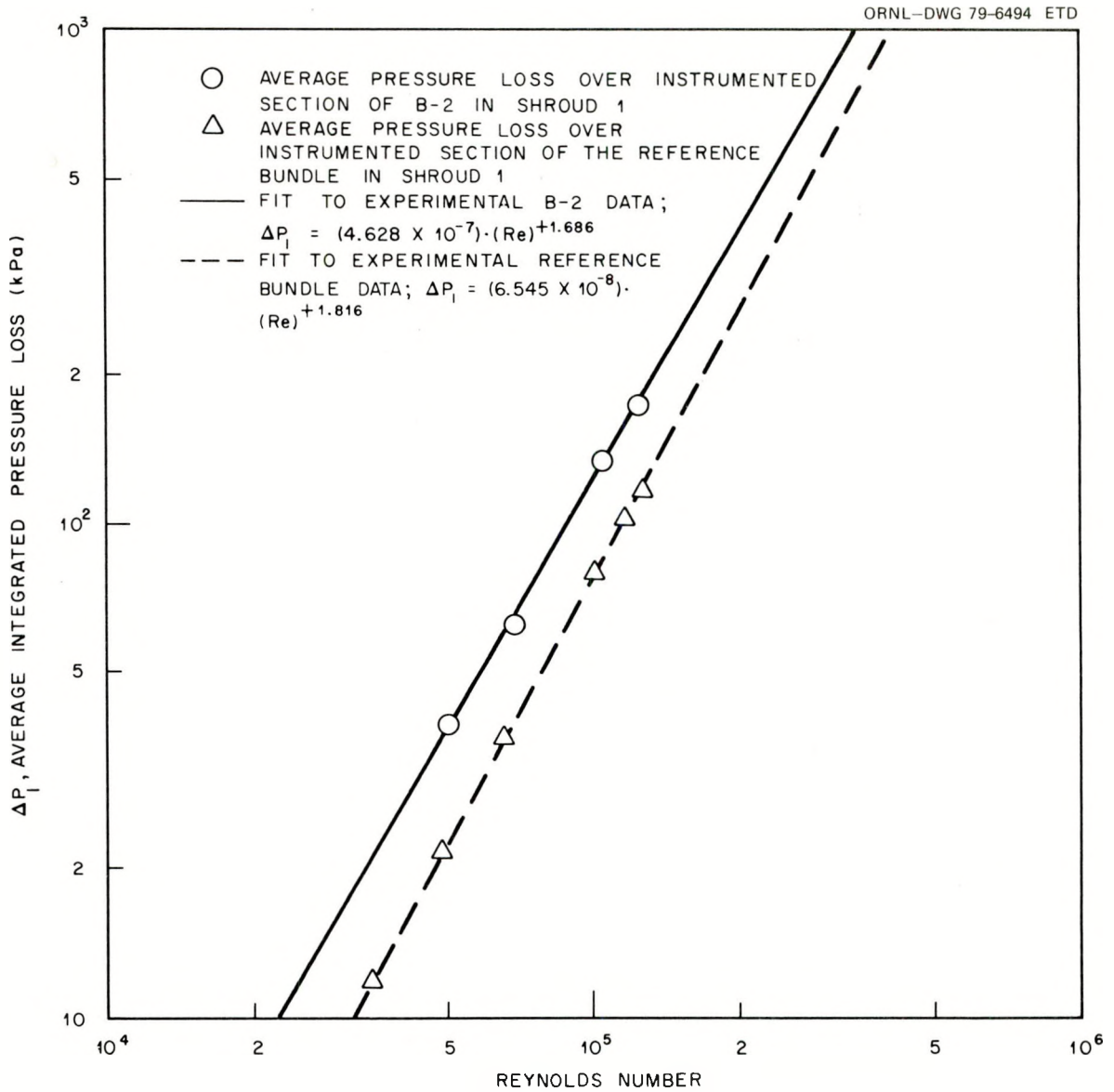


Fig. 3.38. Average integrated pressure losses vs Reynolds number for the B-2 and reference bundle tests in shroud 1. (Note: The Reynolds numbers used to generate this plot were not limited to two significant figures in contrast with the previously reported values.)

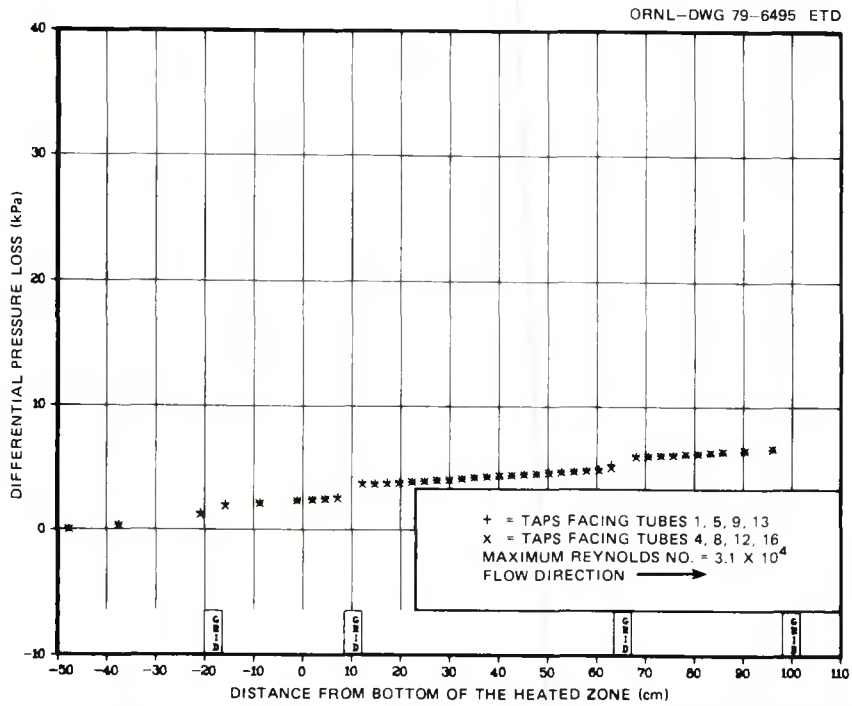


Fig. 3.39. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 6.31 liters/sec.

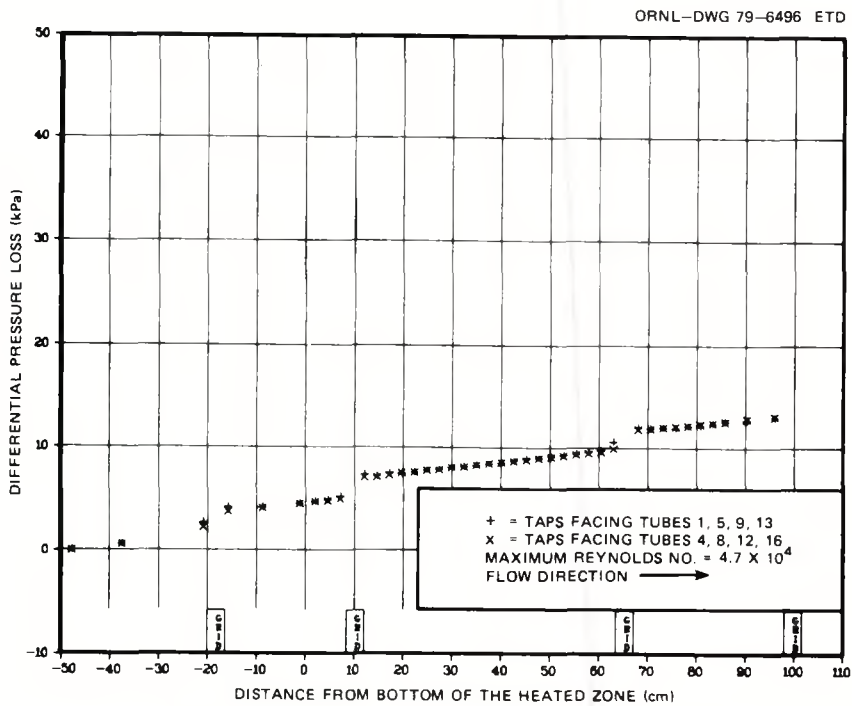


Fig. 3.40. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 9.45 liters/sec.

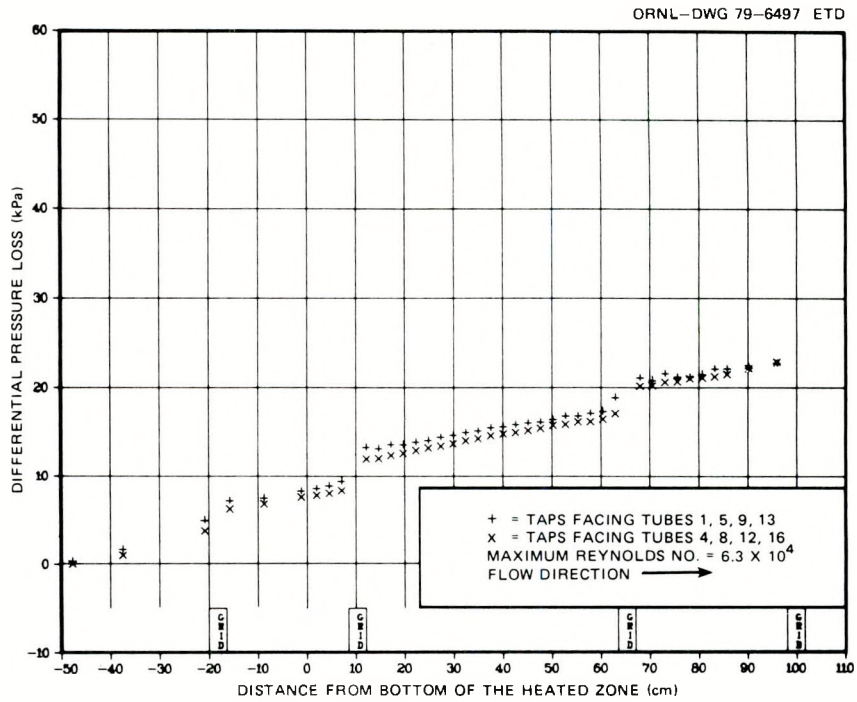


Fig. 3.41. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 12.8 liters/sec.

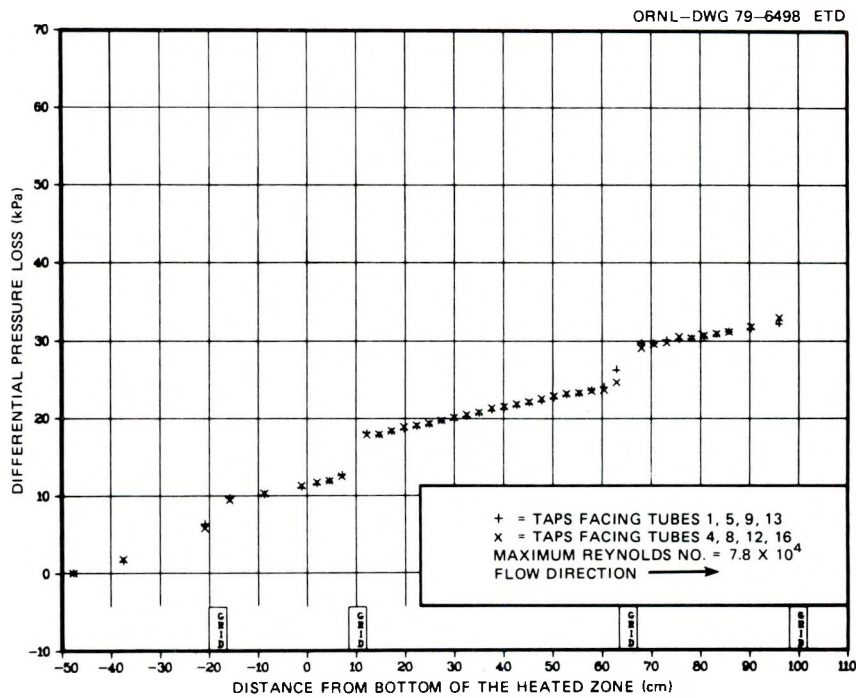


Fig. 3.42. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 15.7 liters/sec.

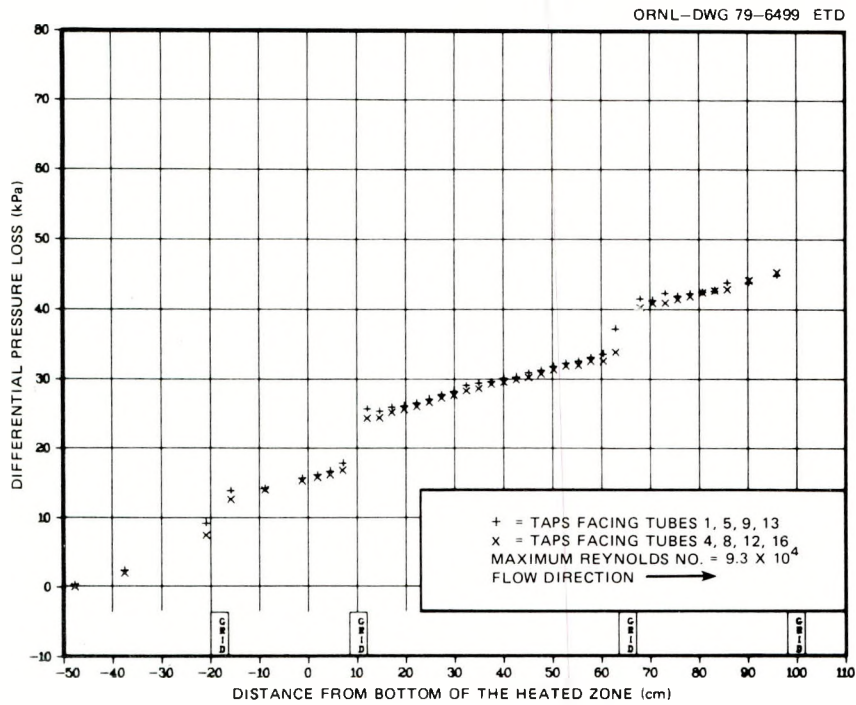


Fig. 3.43. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 18.9 liters/sec.

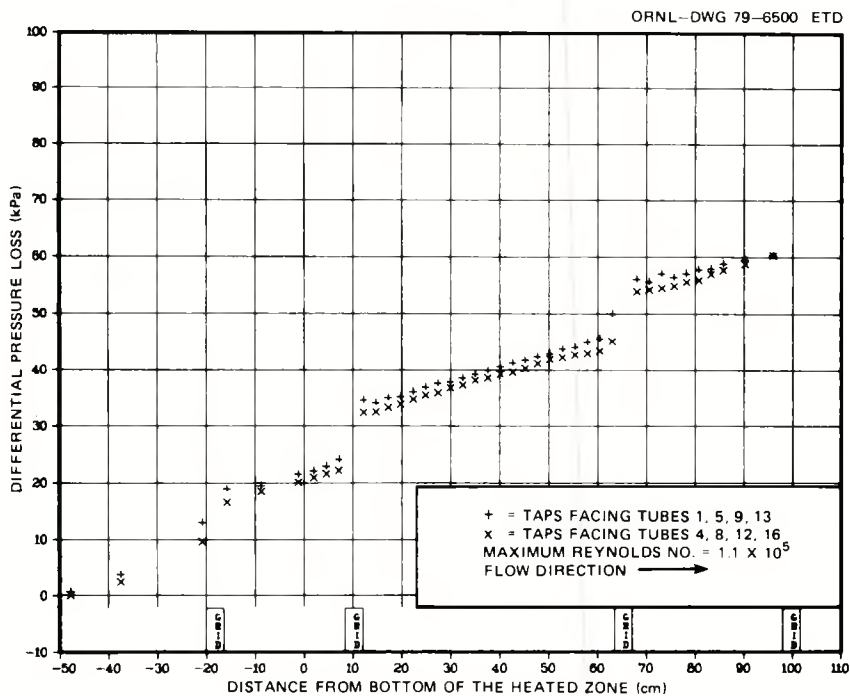


Fig. 3.44. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 22.0 liters/sec.

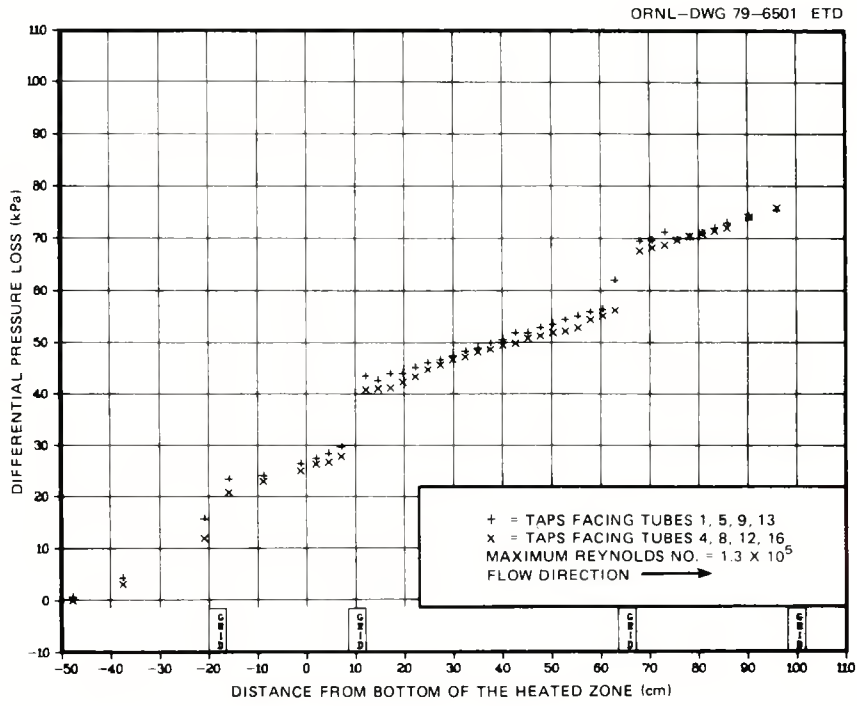


Fig. 3.45. Reference bundle/shroud 2 axial pressure loss profiles; flow rate = 25.3 liters/sec.

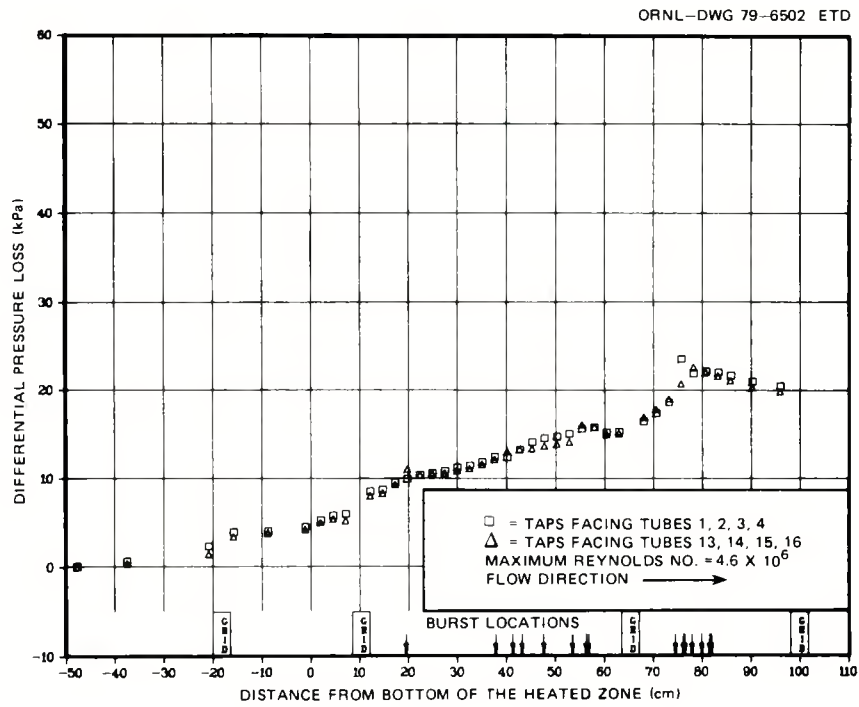


Fig. 3.46. B-2/shroud 2 axial pressure loss profiles; flow rate = 9.47 liters/sec.

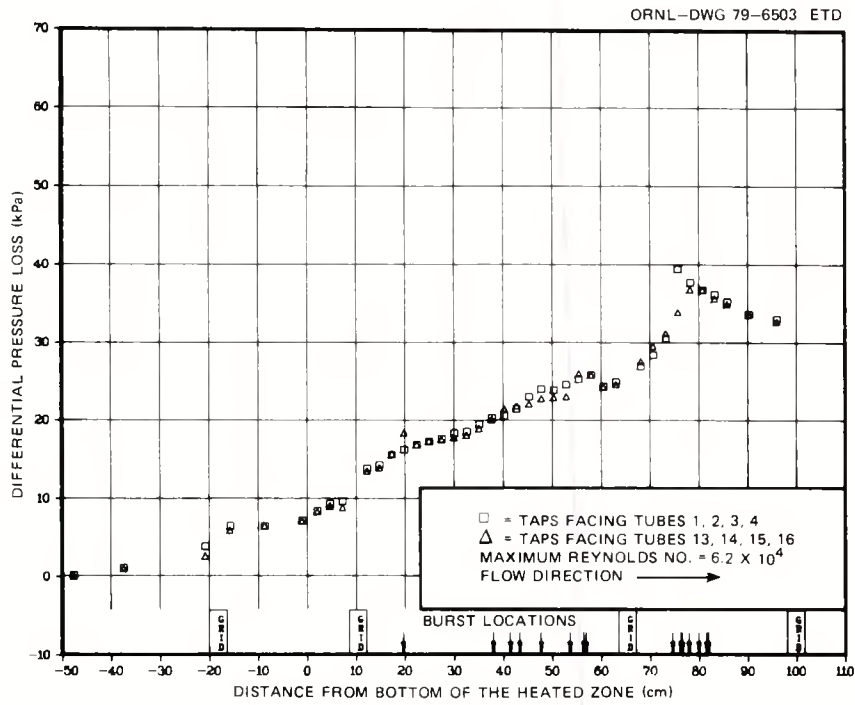


Fig. 3.47. B-2/shroud 2 axial pressure loss profiles; flow rate = 12.5 liters/sec.

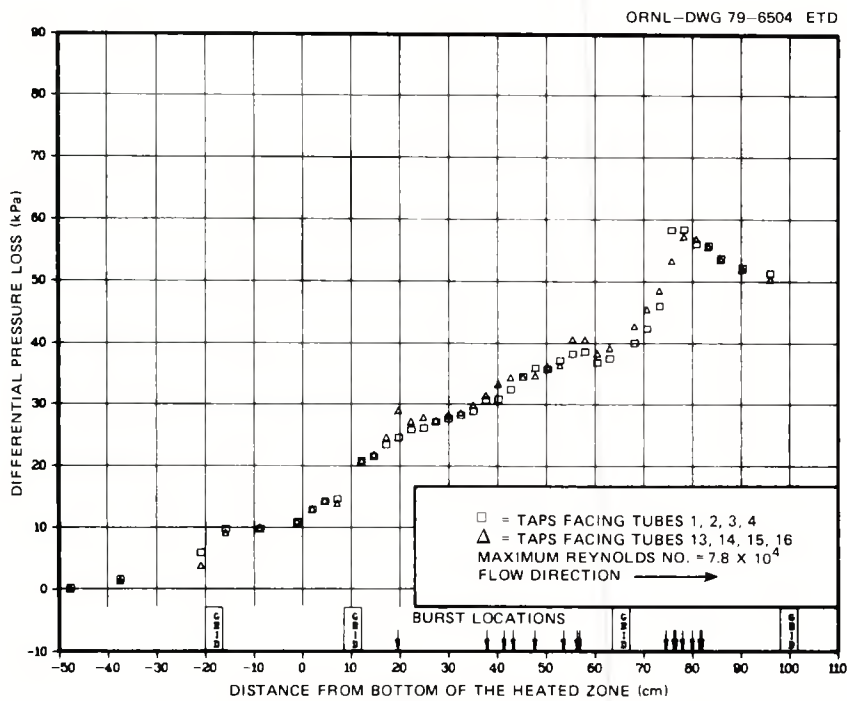


Fig. 3.48. B-2/shroud 2 axial pressure loss profiles; flow rate = 15.8 liters/sec.

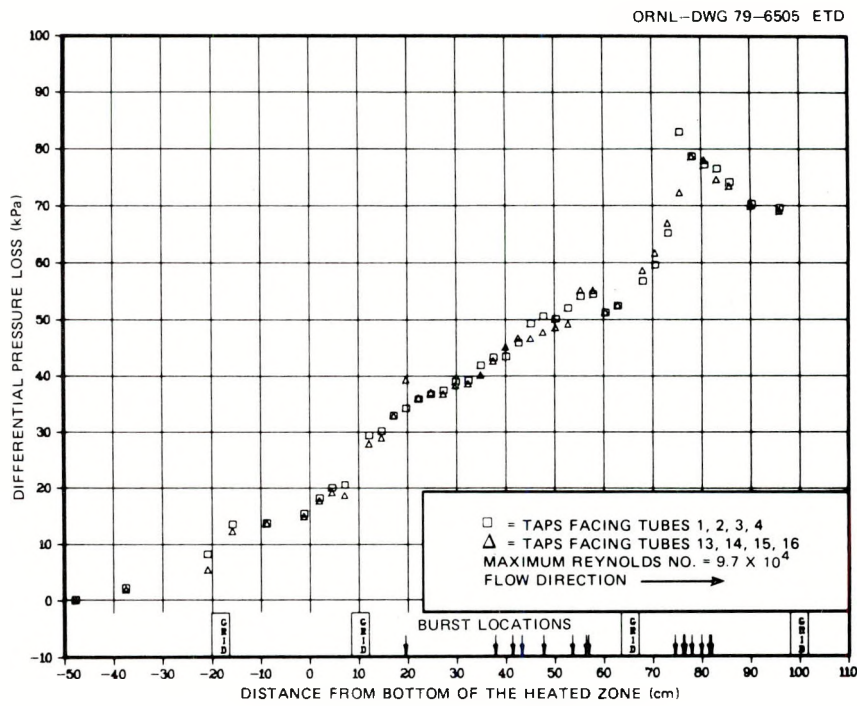


Fig. 3.49. B-2/shroud 2 axial pressure loss profiles; flow rate = 19.0 liters/sec.

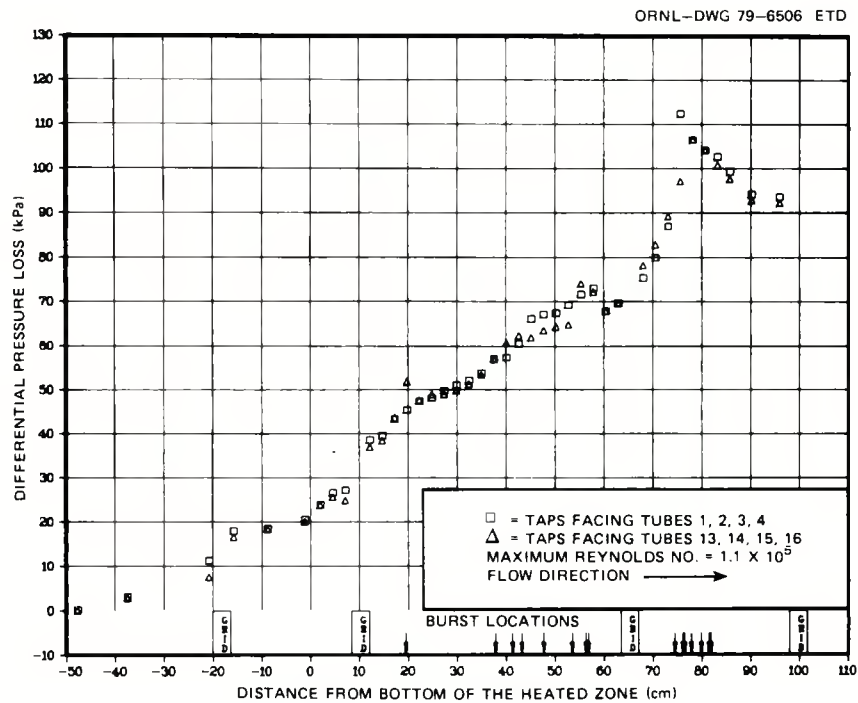


Fig. 3.50. B-2/shroud 2 axial pressure loss profiles; flow rate = 22.2 liters/sec.

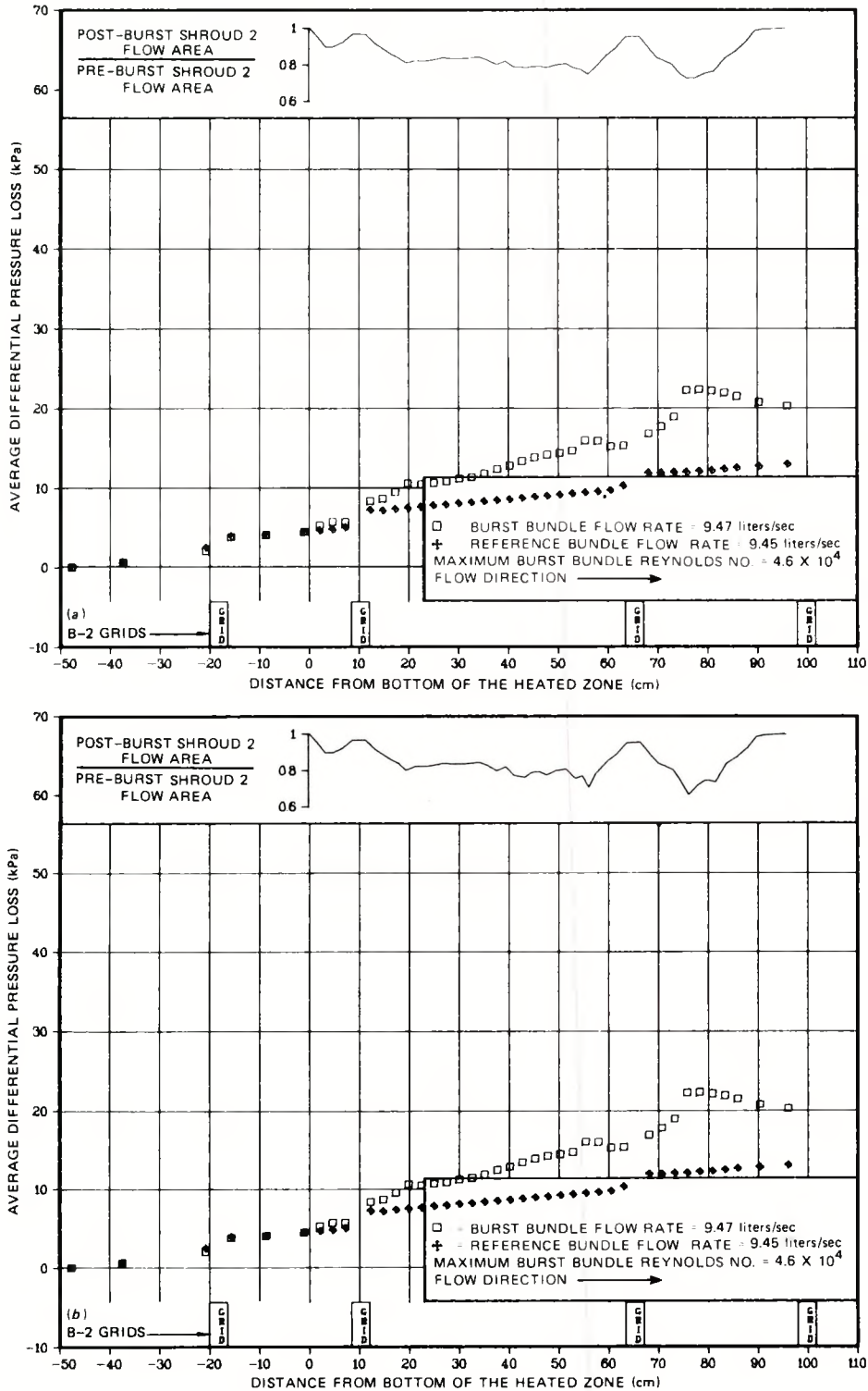


Fig. 3.51. B-2/shroud 2 and reference bundle/shroud 2 axial pressure loss profiles; burst bundle flow rate = 9.47 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

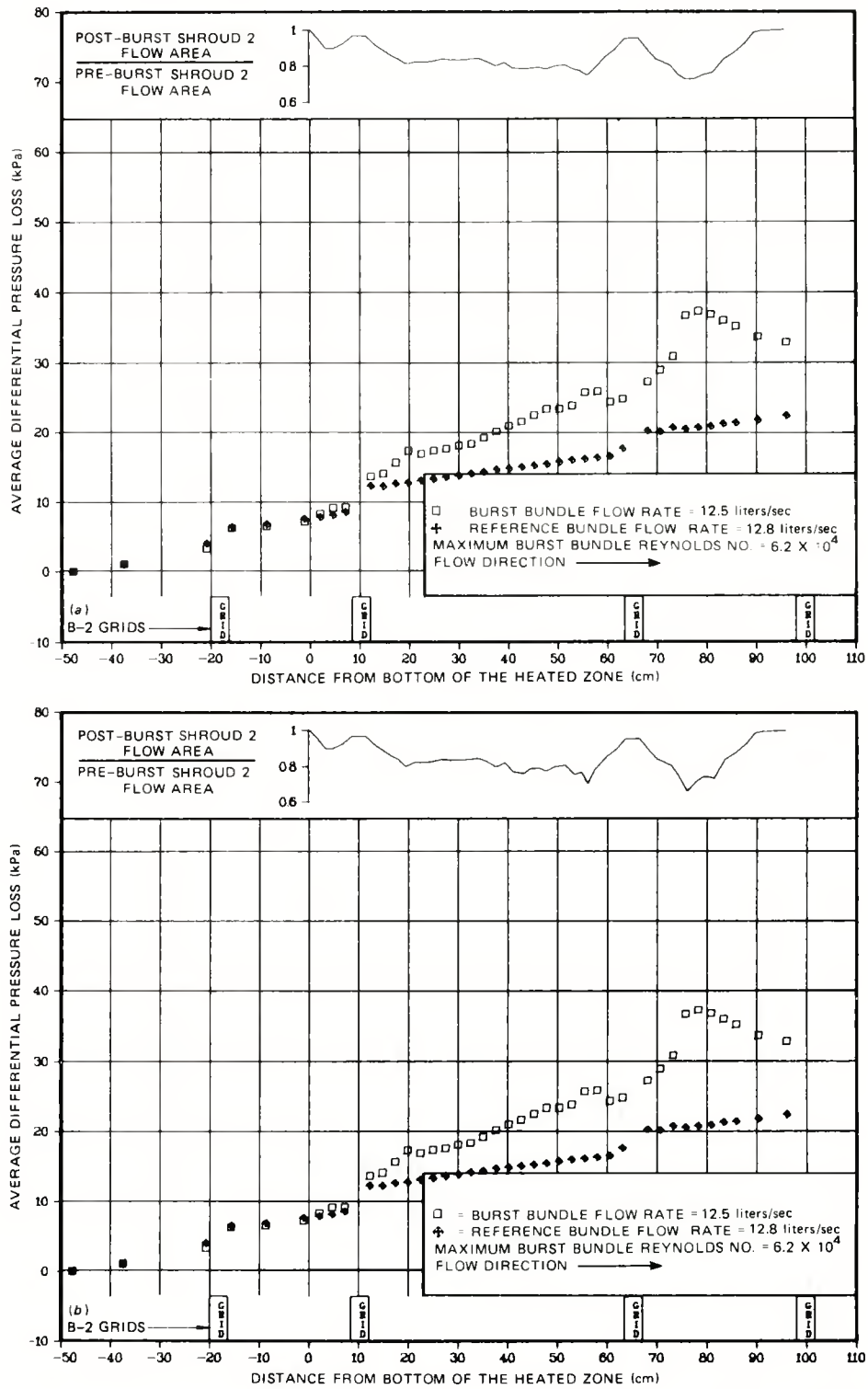


Fig. 3.52. B-2/shroud 2 and reference bundle/shroud 2 axial pressure loss profiles; burst bundle flow rate = 12.5 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

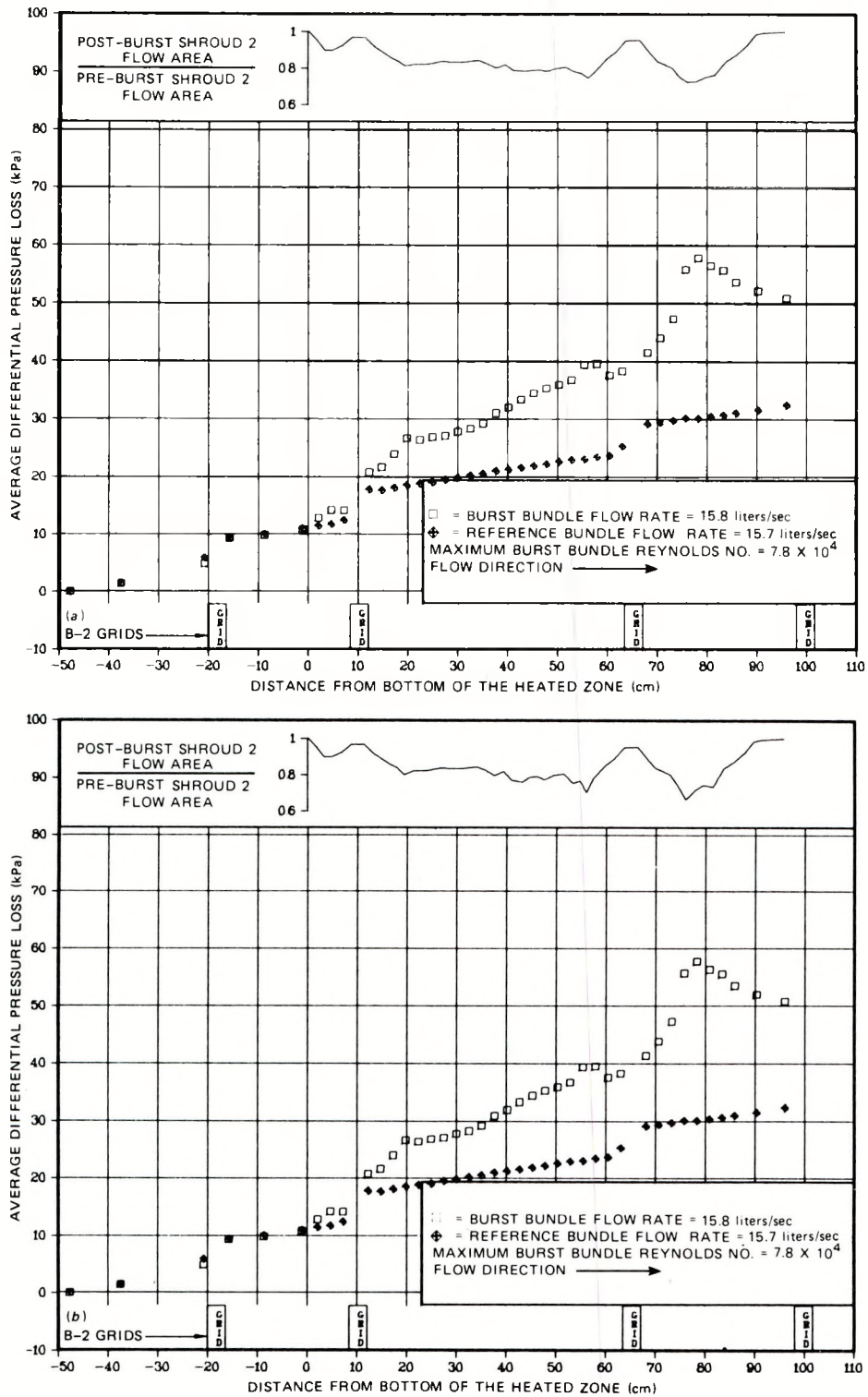


Fig. 3.53. B-2/shroud 2 and reference bundle/shroud 2 axial pressure loss profiles; burst bundle flow rate = 15.8 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

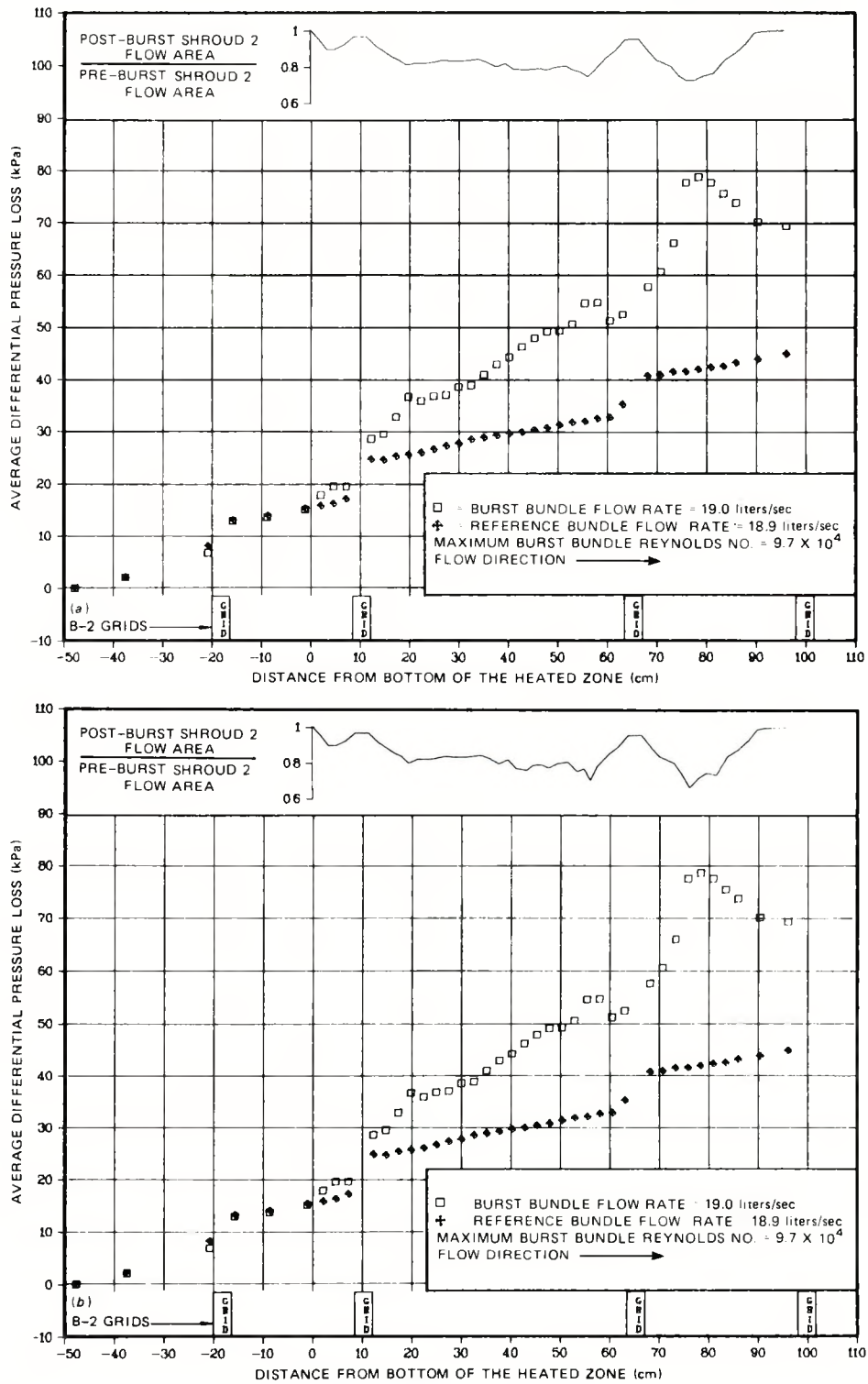


Fig. 3.54. B-2/shroud 2 and reference bundle/shroud 2 axial pressure loss profiles; burst bundle flow rate = 19.0 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

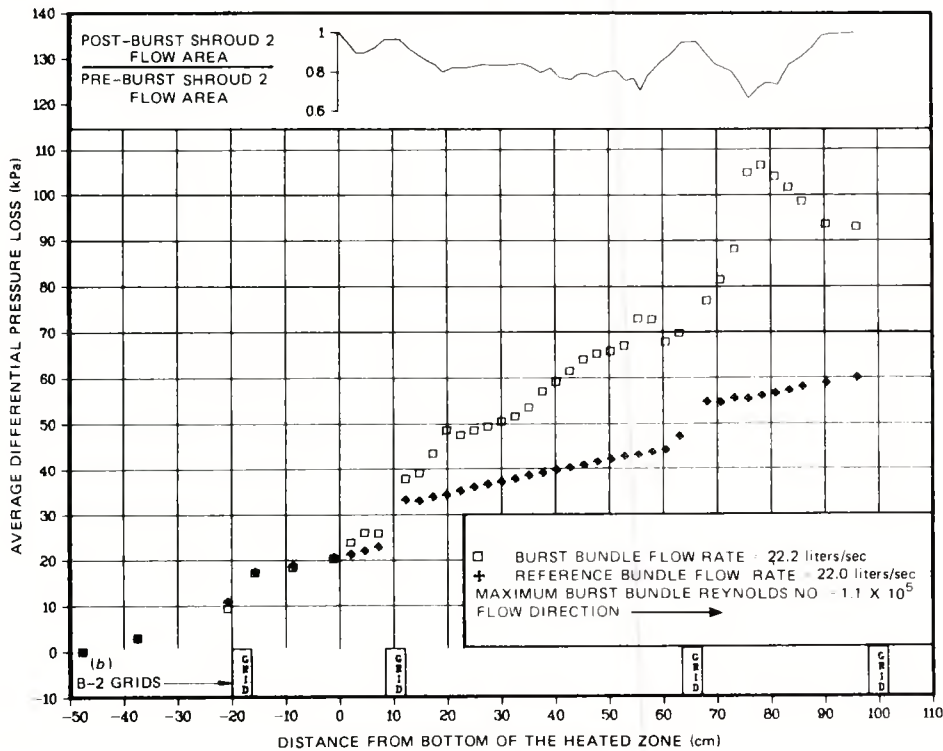
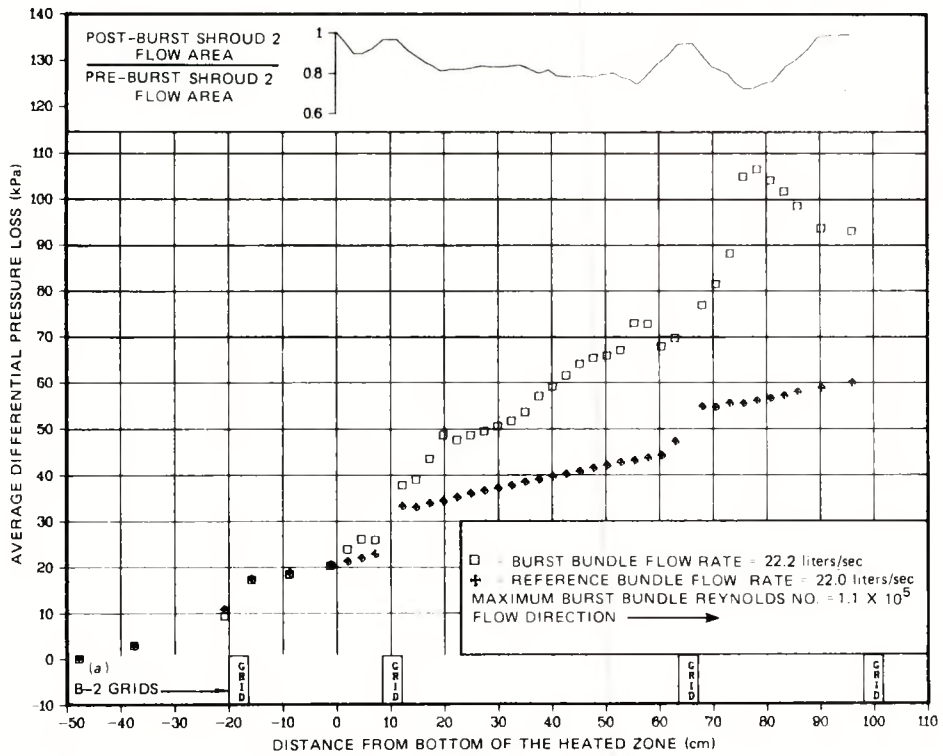


Fig. 3.55. B-2/shroud 2 and reference bundle/shroud 2 axial pressure loss profiles; burst bundle flow rate = 22.2 liters/sec vs: (a) the minimum B-2 restriction profile (b) the maximum B-2 restriction profile.

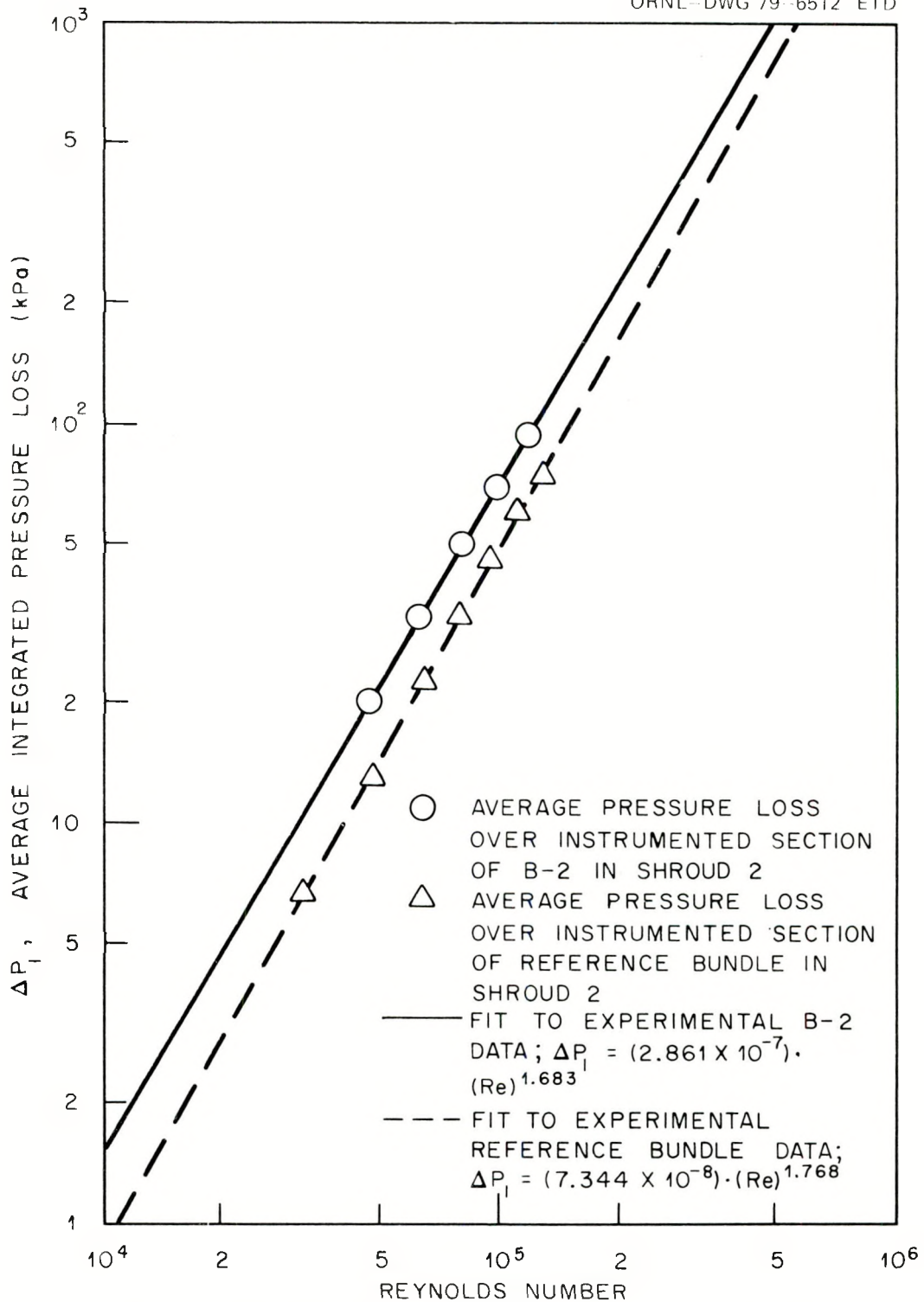


Fig. 3.56. Average integrated pressure losses vs Reynolds number for the B-2 and reference bundle tests in shroud 2. (Note: The Reynolds numbers used to generate this plot were not limited to two significant figures in contrast with the previously reported values.)

4. COBRA-IV COMPUTATIONAL MODELS

The MRBT flow tests were conducted with the goal of conforming to the benchmark experiment philosophy: they were experimentally well-defined (as much as equipment limitations allowed) and hydraulically similar, but not necessarily physically identical, to a class of LOCA conditions. By mathematically modeling these experimental data, one can obtain some idea of a particular model's ability to predict pressure losses for systems that are hydraulically similar and more representative of reactor systems. The COBRA-IV thermal-hydraulics code² was chosen for the following reasons: (1) the widespread usage and acceptance of the COBRA codes; (2) their capability for handling severe flow restrictions; and (3) the ease with which the empirical correlations used as input can be modified.

4.1 Sensitivity Studies

The first step taken in developing the computer models was to conduct a sensitivity study, using various values for the COBRA-IV input parameters which govern the dynamics of the numerical computational schemes. These studies were conducted for the reference and B-2 bundle geometries in shroud 1. The findings indicated that while some slight improvements (computational time savings) are possible without affecting the accuracy of calculated pressures, the default values are generally quite adequate.

The card group 9 COBRA-IV parameters which do affect calculated pressure losses are somewhat self-evident. Briefly, the bundle orientation (THETA in card group 9), the fluid temperature, the flow rate, the flow geometry, the grid spacer pressure loss coefficient, and the friction factor correlation were found to be important. As pointed out in Sect. 2.1, the flow test equipment was so designed to eliminate the static pressure head contribution (described by THETA) from the measurements. Hence, THETA was set equal to 90° from vertical to preclude this type of contribution from piezometric pressures calculated by COBRA-IV. The fluid temperature and flow rate parameters are, of course, dependent upon data measured for each test.

4.2 Bundle-Averaged Geometries

Figures 4.1 and 4.2 show COBRA-IV models of the reference bundle/shroud 1 and reference bundle/shroud 2 assemblies. To calculate burst bundle geometries with COBRA-IV, card groups 5 and 6 require subchannel area and gap ratios relating burst to reference bundle values by axial bundle level. The preparation of card group 5 and 6 data could conceivably become very involved if burst bundle geometries had to be exactly modeled in terms of individual subchannel restrictions and gap variations. Instead of taking this route, a bundle-averaged approach was adopted in producing the calculational results given in this report.

The subchannel restrictions, denoted as AFACT data in ref. 2, were calculated with the following formula:

$$\text{AFACT}_{zi} = \frac{C_{zi} + \left[A_0 - \left(\sum_{j=1}^{16} \text{AF}_{jz} \right) / 16 \right] F_i}{C_{zi}}, \quad (4.1)$$

where

C_{zi} = undeformed-bundle subchannel i flow area at elevation z ,

A_0 = cross-sectional area occupied by an undeformed-bundle tube,

AF_{jz} = cross-sectional area occupied by the deformed-bundle tube j at elevation z ,

F_i = fraction of one tube area which occupies subchannel i before the burst test (1/4, 1/2, or 1).

The C_{zi} parameter used for modeling tests in shroud 1 included the appropriate relief groove (see Fig. 2.6) contributions to the subchannel areas.

This formulation essentially calculates an average tube area based on the 16 measured tube areas per axial level. This average tube area then replaces each of the 16 tubes when calculating AFACT values for the 25 subchannels. Consequently, AFACT data generated in this way represent a smeared or bundle-averaged restriction, which preserves the gross flow

channel area but ignores the twists and turns individual tubes might make in and out of the surrounding subchannels.

The deformed-to-undeformed gap ratios, or GFACT (ref. 2) data, were computed with the following formula:

$$\text{GFACT}_{zk} = \left\{ \frac{2G_{zk} - \left[\sqrt{\left(\sum_{j=1}^{16} \text{AF}_{jz} \right) / \pi} \right] P_k}{2G_{zk} - \left[4\sqrt{(A_o/\pi)} \right] P_k} \right\} \quad (4.2)$$

(if the formula above yields a negative value, $\text{GFACT}_{zk} = 0$),

where

G_{zk} = distance from the undeformed tube center to the adjacent tube center or shroud wall for gap k at elevation z ,

P_k = 1/2 for the 16 gaps between the tubes and shroud, 1 for the 24 tube-to-tube gaps.

This formulation calculates an average burst tube area based on the 16 tube areas at a given axial location, as was done for generating AFACT data. The radius of this average burst tube area is then determined and used to generate the GFACT data corresponding to the 40 gaps per axial location. No corrections to GFACT data were made to reflect the presence of shroud relief grooves. AFACT and GFACT data for the B-1 and B-2 COBRA-IV models may be found in Appendices A through F.

4.3 Grid Spacer Pressure Loss Coefficients

COBRA-IV requires as input an effective grid spacer loss coefficient, K_c , implicitly defined in the following formula:

$$\Delta P_s = \left(\frac{K_c}{2\rho g_c} \right) \left(\frac{M}{A_f} \right)^2, \quad (4.3)$$

where

ΔP_S = nonrecoverable pressure loss induced by the grid spacer,

K_C = COBRA-IV spacer loss coefficient,

ρ = fluid density,

g_C = a unit conversion factor,

M = fluid mass flow rate,

A_f = total flow area upstream of the grid spacer.

Using an alternate formulation by de Stordeur,⁶ a drag coefficient, C_S , is implicitly defined:

$$\Delta P_S = \left(\frac{C_S}{2\rho g_C} \right) \left(\frac{1 - \sigma}{\sigma^2} \right) \left(\frac{M}{A_f} \right)^2, \quad (4.4)$$

where C_S is the spacer drag coefficient and σ is the ratio of the flow area in the spacer grid region to A_f . Equating Eqs. (4.3) and (4.4), K_C may be expressed as

$$K_C = C_S \left(\frac{1 - \sigma}{\sigma^2} \right). \quad (4.5)$$

The determination of K_C then hinges on knowing σ for the system under investigation.

For MRBT reference bundle flow tests, the projected grid frontal area [$= A_f(1 - \sigma)$] was estimated to be $3.8 \pm 0.6 \text{ cm}^2$ by means of linear measurements. The single-grid flow tests (Sects. 3.1 and 3.2) indicated this value to be more on the order of $5.8 \pm 0.2 \text{ cm}^2$ (based on COBRA-IV data fits to be discussed later). This discrepancy is the result of uncertainties induced by miscellaneous features on the spacer grids such as welding beads, dimples, etc. As the determination of frontal areas for such features is difficult, the approach using linear grid dimensions (metal thicknesses and lengths) understandably underestimated the total effective frontal area of the grid. The determination of an effective projected grid frontal area ($5.8 \pm 0.2 \text{ cm}^2$) derived from the single-grid

flow test data circumvents this difficulty and is believed to be a more accurate representation.

Based on the $5.8 \pm 0.2 \text{ cm}^2$ area, values of σ may be calculated for reference bundle/shroud geometries. For shroud 1, σ is estimated to be 0.82 ± 0.02 . Values for K_c may be estimated, using a value of 1.7 (see ref. 5) for C_s [Eq. (4.5)]. For shroud 1, $K_c = 0.53 \pm 0.07$; for shroud 2, $K_c = 0.46 \pm 0.07$. As will be shown in Chap. 5, these values were found to be in good agreement with most of the flow test data.

4.4 Friction Factor Correlations

A similar empirical approach was taken in determining an appropriate friction correlation for the turbulent friction factor. After finding a fairly good estimate for the spacer loss coefficient via the second single-grid test series, the turbulent friction factor correlation was varied in the COBRA-IV single-grid test models until good agreement was achieved between calculational and experimental pressure loss profiles. This topic will be discussed in more detail later in Chap. 5.

McAdams⁷ proposed the following correlation, which approximates the Darcy-Weisbach friction factor, f , for turbulent flow in smooth tubes:

$$f = 0.184(\text{Re})^{-0.2} \text{ for } 3 \times 10^4 < \text{Re} < 2 \times 10^6 . \quad (4.6)$$

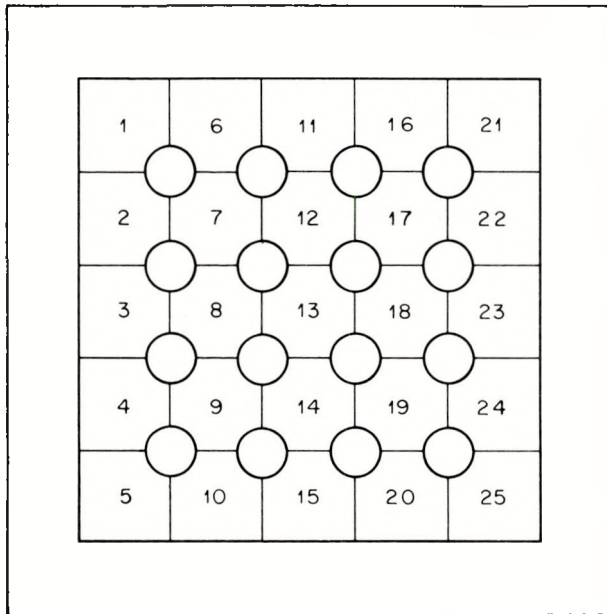
Because MRBT flow tests involved flow exterior to tube bundles, some modifications to the Darcy-Weisbach factor for flow inside a smooth-walled tube are required to represent bundle frictional flow losses. Data given by Tong,⁸ though inconclusive, indicate that the reference (preburst) MRBT bundle's pitch-to-diameter ratio of 1.32 reflects about a 30% increase in the smooth-tube Darcy-Weisbach friction factor. A more detailed study by Marek, Maubach, and Rehme⁹ suggests only a 6% increase. As will be shown in Chap. 5, the correlations chosen for MRBT COBRA-IV models reflect increases on the order of 20 to 30%. For both shrouds and all bundles, the following modified McAdams correlation was found to fit well:

$$f = (0.225 \pm 0.010)(\text{Re})^{-0.2} \text{ for } 3 \times 10^4 < \text{Re} < 2 \times 10^6 . \quad (4.7)$$

4.5 Summary

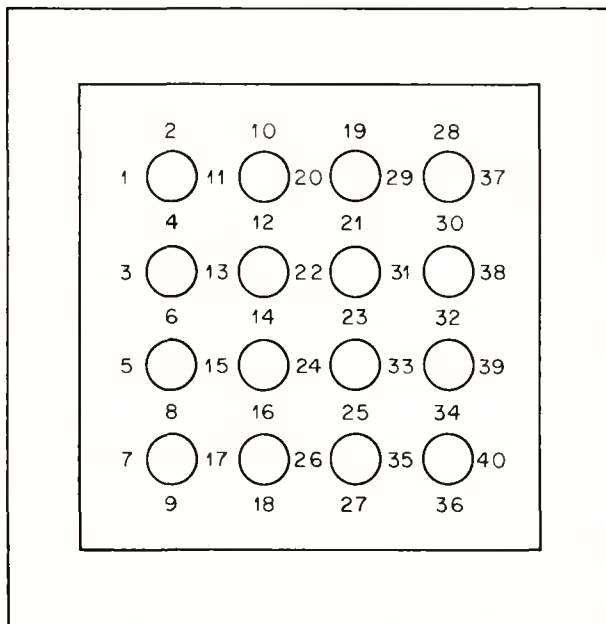
Two points should be reiterated concerning the COBRA-IV models. The first is the use of bundle-averaged geometry representations for burst bundles, a practice which considerably simplified COBRA-IV input preparation. The validity of this approximation will be demonstrated when computational and experimental pressure loss profiles are compared in Chap. 5.

The second point is the fact that uncertainties in reported experimental data (especially flow rates), friction factors, and spacer loss coefficients are intertwined; hence, the friction factors and spacer loss coefficients cannot be determined any more precisely than experimental precision limits allow. This point must be kept in mind when comparing friction factor correlations and spacer loss coefficients found in the literature with those used for COBRA-IV MRBT models.



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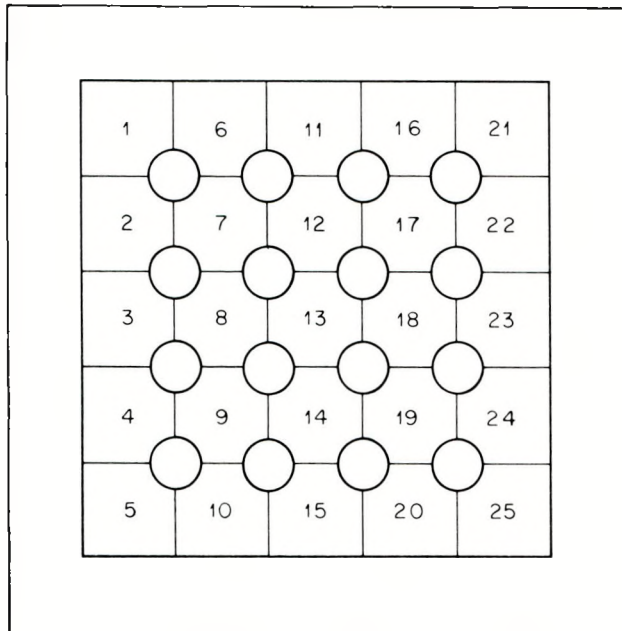
<u>SUBCHANNEL NUMBER</u>	<u>SUBCHANNEL AREA* (cm²)</u>
1, 5, 21, 25	1.045
2, 3, 4, 22, 23, 24	1.357
6, 10, 11, 15, 16, 20	0.990
7, 8, 9, 12, 13, 14, 17, 18, 19	1.145



<u>GAP NUMBER</u>	<u>GAP WIDTH* (cm)</u>
1, 3, 5, 7, 37, 38, 39, 40	0.719
2, 9, 10, 18, 19, 27, 28, 36	0.465
4, 6, 8, 11, 12, 13, 14, 15, 16, 17, 20, 21, 22, 23, 24, 25, 26, 29, 30, 31, 32, 33, 34, 35	0.351

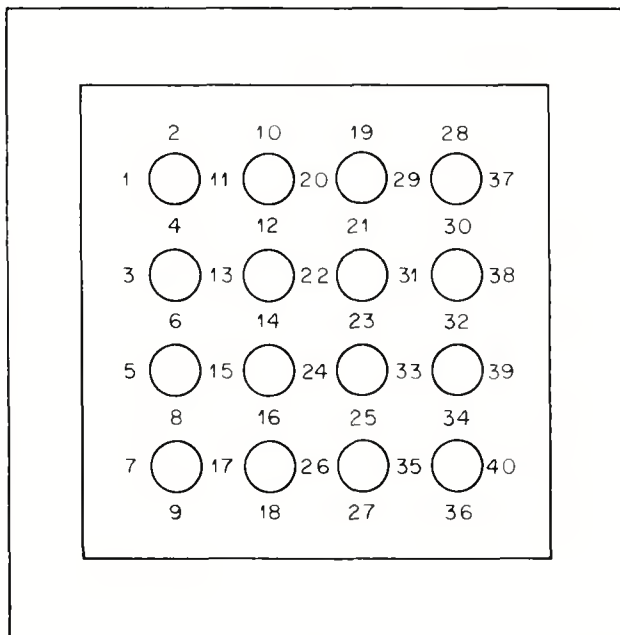
*FOR REGIONS AWAY FROM SHROUD DEFORMITIES IN SUBCHANNELS 1, 5, 6, 10, 11, 15, AND 20

Fig. 4.1. COBRA-IV model of the reference bundle in shroud 1; subchannel and gap dimensions.



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<u>SUBCHANNEL NUMBER</u>	<u>SUBCHANNEL AREA (cm²)</u>
1, 5, 21, 25	1.581
2, 3, 4, 22, 23, 24	1.503
6, 10, 11, 15, 16, 20	1.448
7, 8, 9, 12, 13, 14, 17, 18, 19	1.145



<u>GAP NUMBER</u>	<u>GAP WIDTH (cm)</u>
1, 3, 5, 7, 37, 38, 39, 40	0.820
2, 9, 10, 18, 19, 27, 28, 36	0.782
4, 6, 8, 11, 12, 13, 14, 15, 16, 17, 20, 21, 22, 23, 24, 25, 26, 29, 30, 31, 32, 33, 34, 35	0.351

Fig. 4.2. COBRA-IV model of the reference bundle in shroud 2; subchannel and gap dimensions.

5. COMPARISONS BETWEEN EXPERIMENTAL AND COMPUTATIONAL RESULTS

In this chapter, the data generated with the previously described COBRA-IV models are compared with the experimental data presented in Chap. 3. An attempt has been made to arrange these analyses such that all calculational and experimental data available for a particular comparison are generally presented in the same figure; comparisons with calculational results based on different friction factors and spacer loss coefficients will be denoted "a," "b," or "c." The figure numbers are arranged in order of ascending flow rates. The Reynolds numbers reported in these figures are based solely on the reported experimental flow rates, temperature, and flow area data, as discussed in Chaps. 2 and 3.

5.1 First Single-Grid Tests in Shroud 1

Averaged experimental flow test data from the first single-grid tests in shroud 1 are compared with COBRA-IV bundle-averaged pressure loss calculations in Figs. 5.1 through 5.6. The "a" comparisons reflect COBRA-IV bundle-averaged pressure losses based on a friction factor correlation of $0.220(\text{Re})^{-0.2}$ and a grid spacer pressure loss coefficient of 0.50. The "b" comparisons represent calculations performed with a friction factor correlation of $0.235(\text{Re})^{-0.2}$ and a grid loss coefficient of 0.60.

As shown in these figures, the COBRA-IV calculations based on the $0.235(\text{Re})^{-0.2}$ friction factor correlation and 0.60 grid loss coefficient agree significantly better with the experimental data. The flow rates used for these calculations are +3% to +1% above the reported experimental values. This discrepancy is the result of a deliberate effort to determine the minimum best-fit values for friction factors and grid loss coefficients based on the effects of reasonable uncertainties (see Sect. 2.4) in the reported experimental flow rates. Hence, the "b" friction factor correlation and grid spacer loss coefficient represent approximate minimum limits on these empirical quantities for good agreement with COBRA-IV pressure loss calculations. The nonlinearities downstream of the spacer grid and 65 cm discussed in Sect. 3.1 are very evident in Fig. 5.6b.

5.2 Second Single-Grid Tests in Shroud 1

Figures 5.7 through 5.12 present a similar analysis for the second single-grid test experimental data. Again the "b" comparisons with COBRA-IV bundle-averaged pressures based on a $0.235(\text{Re})^{-0.2}$ friction factor correlation and a 0.60 grid loss coefficient match the experimental data significantly better than the "a" comparisons. Note that, for these calculations, flow rates were closely matched, unlike the first single-grid test series. Hence, the $0.235(\text{Re})^{-0.2}$ friction factor and 0.60 grid loss coefficient are evidently good representations for the two single-grid test series.

The nonlinearity downstream of 65 cm is again visible (see Fig. 5.12b). However, the disturbed zone downstream of the grid in these tests is almost nonexistent relative to effects seen in Fig. 5.6b. No good explanation exists at present for this anomaly. Finally, Fig. 5.12b indicates that while some flow rate drifting occurred (see Sect. 3.2), COBRA-IV results still match experimental pressure loss data reasonably well and in a manner consistent with the other second single-grid flow tests.

5.3 Reference Bundle Tests in Shroud 1

Figures 5.13 through 5.18 compare COBRA-IV-derived pressure loss profiles with the corresponding experimental data presented in Sect. 3.3. As noted earlier, the "a" comparisons use a friction correlation of $0.220(\text{Re})^{-0.2}$ compared with $0.235(\text{Re})^{-0.2}$ for the "b" comparisons and a grid coefficient of 0.50 compared with 0.60 for "b". As shown, the calculated pressure loss profiles using both friction factor correlations and grid loss coefficients tend to overestimate the experimental pressure loss profiles, especially at the higher flow rates. If the "a" comparisons are examined more closely, it can be seen that the discrepancies between calculated and experimental pressure loss profiles could be explained by about a 3% decrease in the reported volumetric flow rates. Hence, the $0.220(\text{Re})^{-0.2}$ friction factor and 0.50 grid loss coefficient fit the experimental results better than the "b" correlations and represent approximate maximum correlation limits, as the discrepancies in "a"

data might be explained in terms of the lower-bound precision limits of the experimental data. This finding is inconsistent with the minimum best-fit correlations recommended for the single-grid test data, the "b" values; that is, the maximum "a" correlations are lower in magnitude than the minimum single-grid values.

5.4 B-1 Tests in Shroud 1

Because of the wide range of variation in friction factor correlations and grid loss coefficients which best fit the reference and single-grid experimental data, no attempt was made to tailor the friction factor correlations to reflect the effect of lower pitch/tube diameter ratios characterizing burst bundles. Instead, the previously used empirical correlations were employed in generating calculated B-1/shroud 1 pressure losses for comparison with experimentally derived data. These comparisons are shown in Figs. 5.19 through 5.22; the calculations used were based on the minimum flow restriction definition (see Sect. 3.4, especially Table 3.1). Similar comparisons, with calculational results based on the maximum flow restriction definition (see Sect. 3.4, especially Table 3.2), are given in Figs. 5.23 through 5.26. As seen, the calculations based on the minimum flow restriction definition appear to fit the experimental pressure loss profiles best. Also evident in Figs. 5.19b and 5.20b is the fact that the $0.235(\text{Re})^{-0.2}$ friction factor correlation and the 0.60 grid loss coefficient represent the better fit to experimental pressure loss profiles up to a $1.28 \times 10^{-2} \text{ m}^3/\text{sec}$ flow rate. Above this flow rate, as shown in Figs. 5.21a and 5.22a, the calculations based on the $0.220(\text{Re})^{-0.2}$ friction factor correlation and 0.50 grid loss coefficient represent the better fit with experimental data.

Hence, as was the case in the comparison of COBRA-IV and experimentally derived pressure loss profiles for single-grid and reference bundle tests, no single friction factor correlation or grid loss coefficient was found to be adequate. The fact that the predicted shapes of the pressure loss profiles agree well with experimental data for a judicious choice of friction factor correlations and grid loss coefficients suggests that COBRA-IV provides a very adequate tool for modeling such

pressure losses. Difficulties in adjusting absolute values of calculated pressure loss profiles to agree with experimental values are indicative of the precision limits on the experimental data describing the experiment (i.e., fluid temperatures, flow rates, and flow areas) and not of COBRA-IV limitations.

5.5 B-2 Tests in Shroud 1

Comparisons of COBRA-IV and experimental pressure loss profiles were made for the B-2/shroud 1 flow tests. These comparisons are shown in Figs. 5.27 through 5.30; the calculational results are based on the minimum restriction definition (see Table 3.3) and the two previously used correlation sets. Similar comparisons based on the maximum restriction definition (see Table 3.4) for B-2 tube areas are shown in Figs. 5.31 through 5.34.

Determining which restriction definition produces COBRA-IV pressure loss profiles that are more representative of experimentally derived data is impossible. This dilemma results from the fact that both restriction definitions are virtually identical except around 42 and 76 cm above the bottom of the heated zone. The differences at 42 cm between the calculated profiles based on these two definitions and the experimental data are inconsequential. On the other hand, both definitions underestimate the experimental data at 76 cm. As stated in Sect. 3.5, there is reason to believe that the experimental data in the 76-cm region are invalid — it is suspected that they do not strictly represent piezometric pressures. Hence, no determination can really be made as to the better restriction definition.

One may conclude, however, that once again no single best correlation for the friction factor or grid loss coefficient can be established. As shown in Figs. 5.27 and 5.31, the calculational data based on the $0.235(\text{Re})^{-0.2}$ friction factor correlation and 0.60 grid loss coefficient fit the experimental data best, consistent with the B-1 comparisons at similar Reynolds numbers. At the higher flow rates, the alternate set of correlations produced results in better agreement with the experimental data, again consistent with the B-1 findings. Hence, for shroud 1,

regardless of the bundle, the grid loss coefficients given in Sect. 4.3 and the friction factor correlation given in Sect. 4.4 [Eq. (4.5)] are recommended in conjunction with this data. The particular values chosen (especially for grid coefficients) from these ranges should generally follow this rule-of-thumb: choose larger values for low flow rates and smaller values for higher flow rates. This trend is consistent with behavior observed previously⁶ — that is, the observed dependence of grid spacer loss coefficients on Reynolds number.

5.6 Reference Bundle Tests in Shroud 2

Similar analyses of reference bundle/shroud 2 tests are found in Figs. 5.35 through 5.41. Three sets of computational results, labeled "a," "b," and "c," are given in each figure, representing three sets of empirical friction factor and grid loss coefficient correlations. The "a" correlations are $0.215(\text{Re})^{-0.2}$ and 0.40 for the friction factor and grid loss coefficient, respectively; the "b" correlations are $0.220(\text{Re})^{-0.2}$ and 0.45; the "c" correlations are $0.235(\text{Re})^{-0.2}$ and 0.50. The friction factor was not expected to be shroud-dependent and, as shown, this proved to be the case. The grid loss coefficient and its estimated uncertainty for shroud 2, derived in Sect. 4.3, were found adequate.

As shown in these and previous figures, the best agreement between calculated and experimental results again occurs when the larger correlation values are used at low flow rates and the smaller values are used at higher flow rates. Overall, the agreement between calculated and experimentally derived results is considered good.

5.7 B-2 Tests in Shroud 2

Comparisons of calculated B-2/shroud 2 pressure loss profiles, based on the minimum restriction definition (see Table 3.3), are made with experimental data in Figs. 5.42 through 5.46. Similar comparisons are made of pressure-loss profiles based on the maximum restriction definition

(see Table 3.4) in Figs. 5.47 through 5.51. As for the B-2/shroud 1 comparisons, no clear choice can be made as to which restriction definition is the better.

The $0.235(\text{Re})^{-0.2}$ friction factor and 0.50 grid loss correlation results fit better in Figs. 5.42, 5.47, 5.43, and 5.48. The "b" fits were better for 5.44, 5.49, 5.45, and 5.50. The $0.220(\text{Re})^{-0.2}$ friction factor and 0.45 correlation tend to fit better with increasing flow rate magnitudes. This observation is consistent with the earlier rule-of-thumb as to the choice of the empirical correlation values as a function of flow rate.

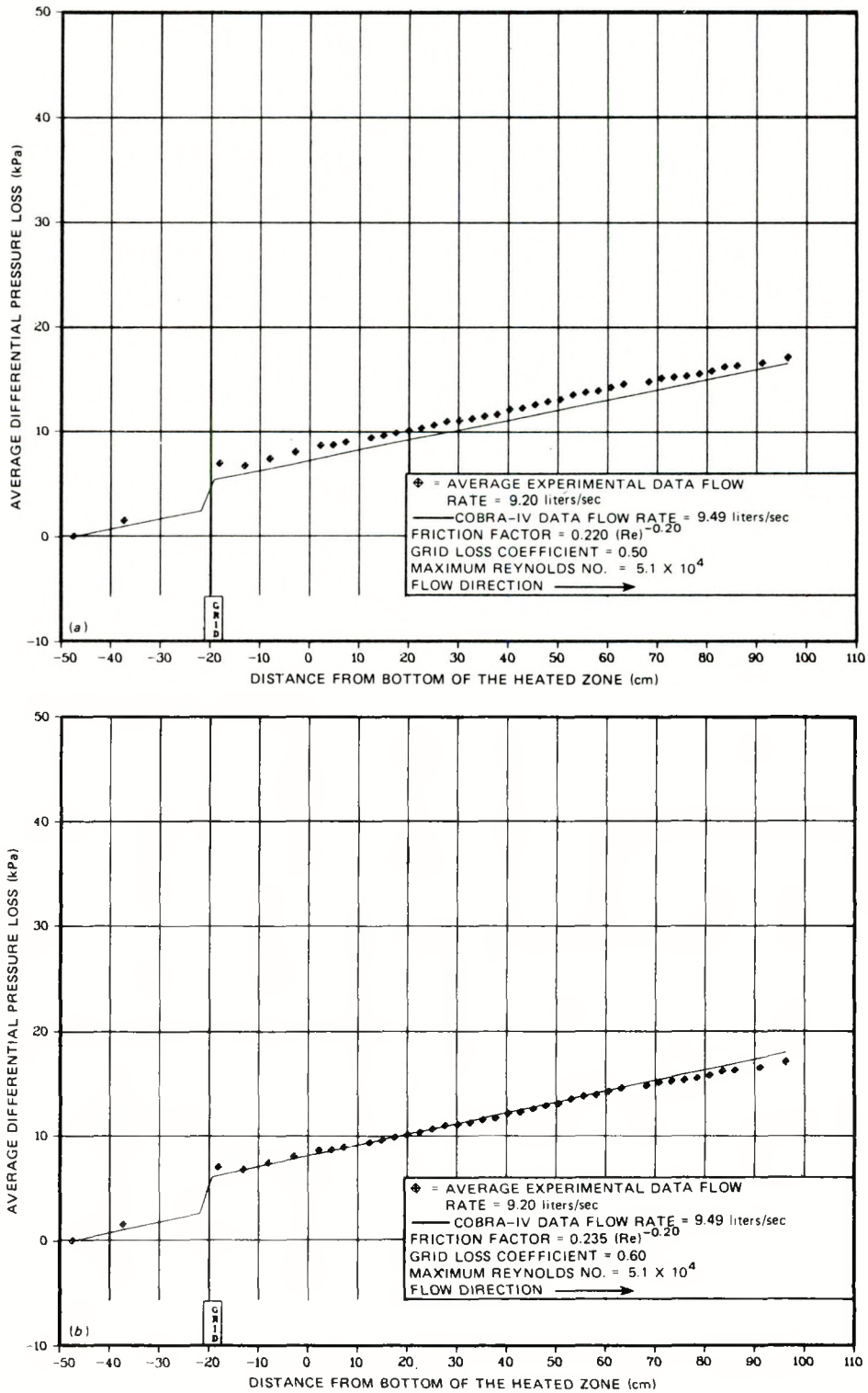


Fig. 5.1. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.20 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

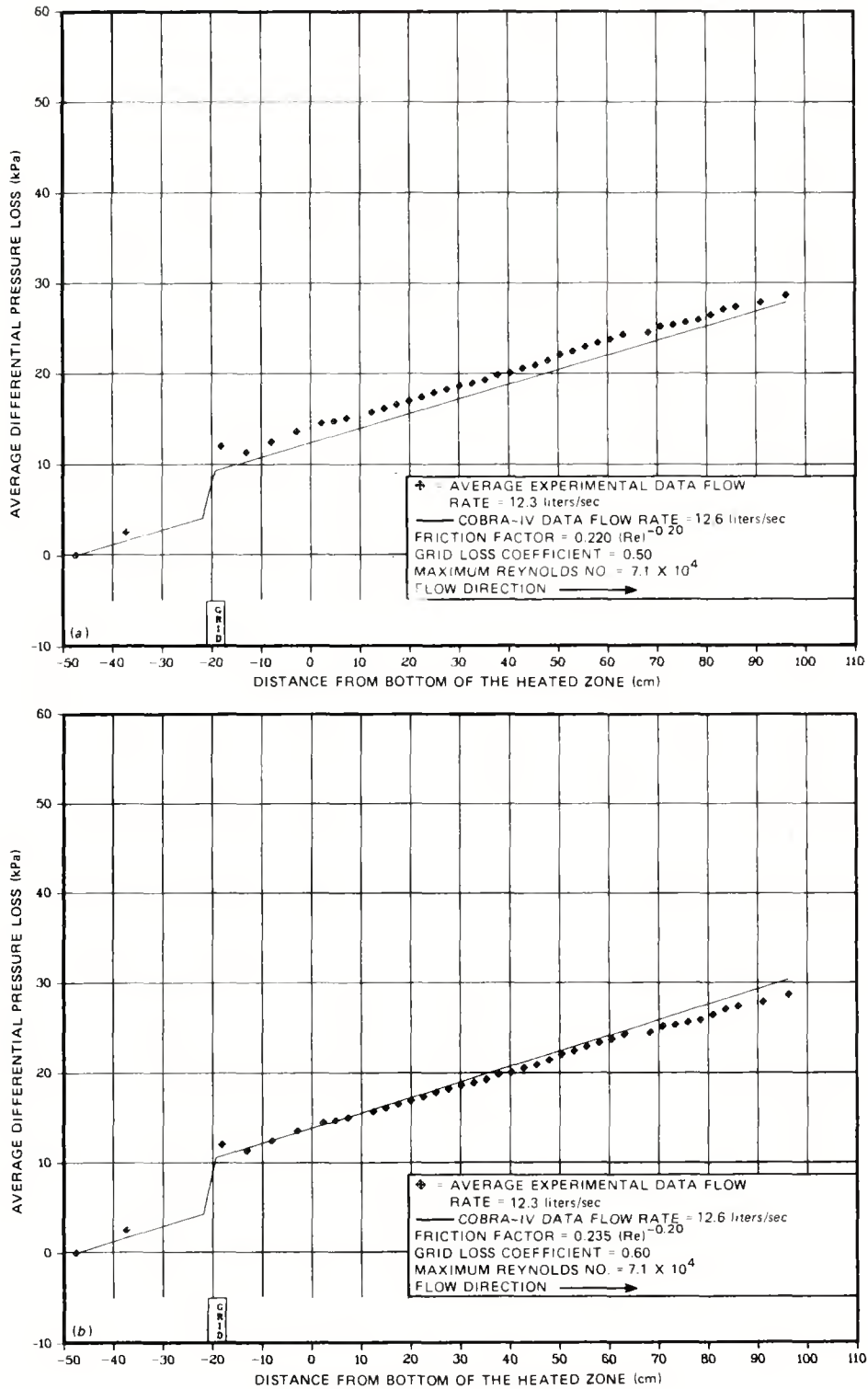


Fig. 5.2. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.3 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

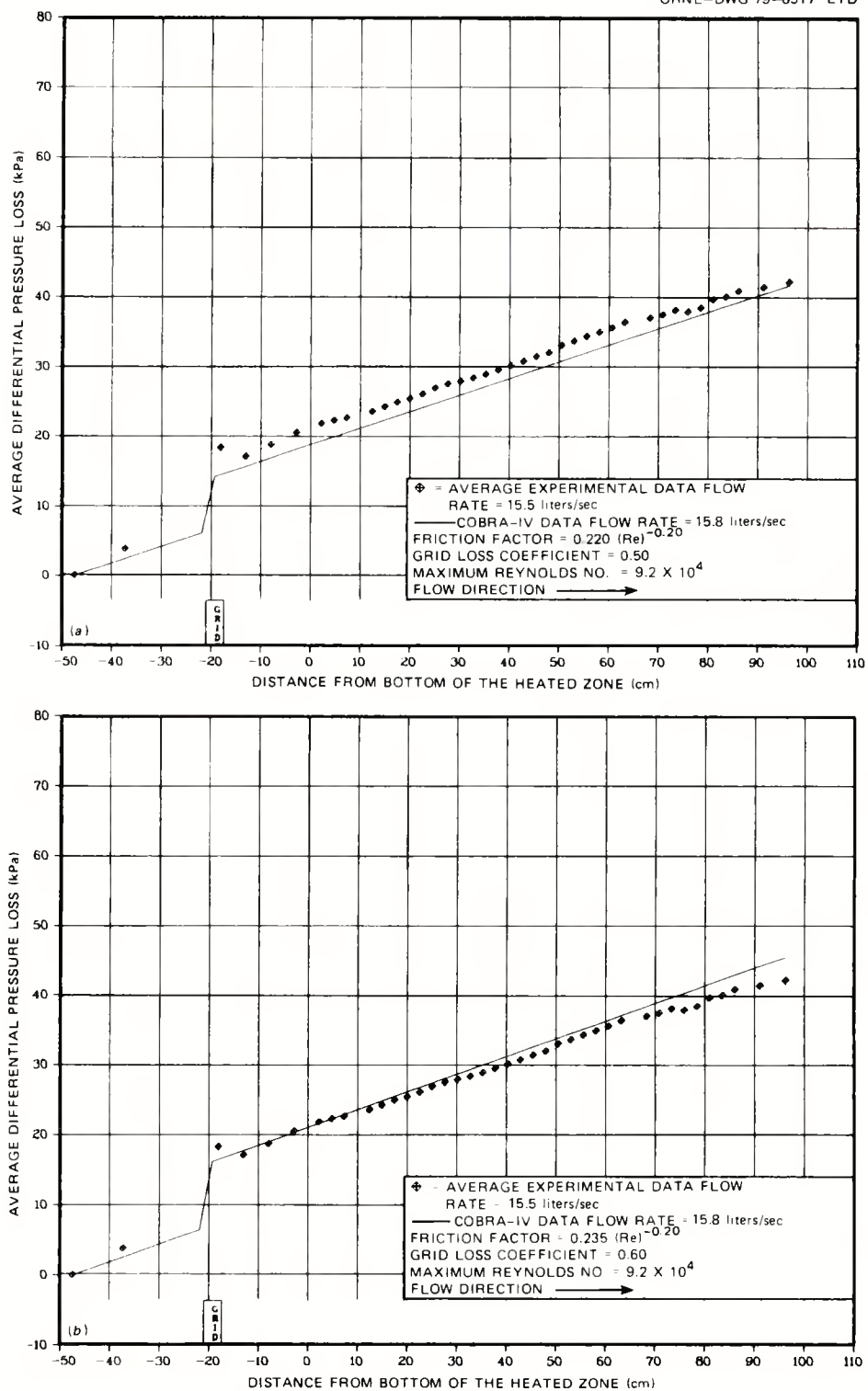


Fig. 5.3. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 15.5 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

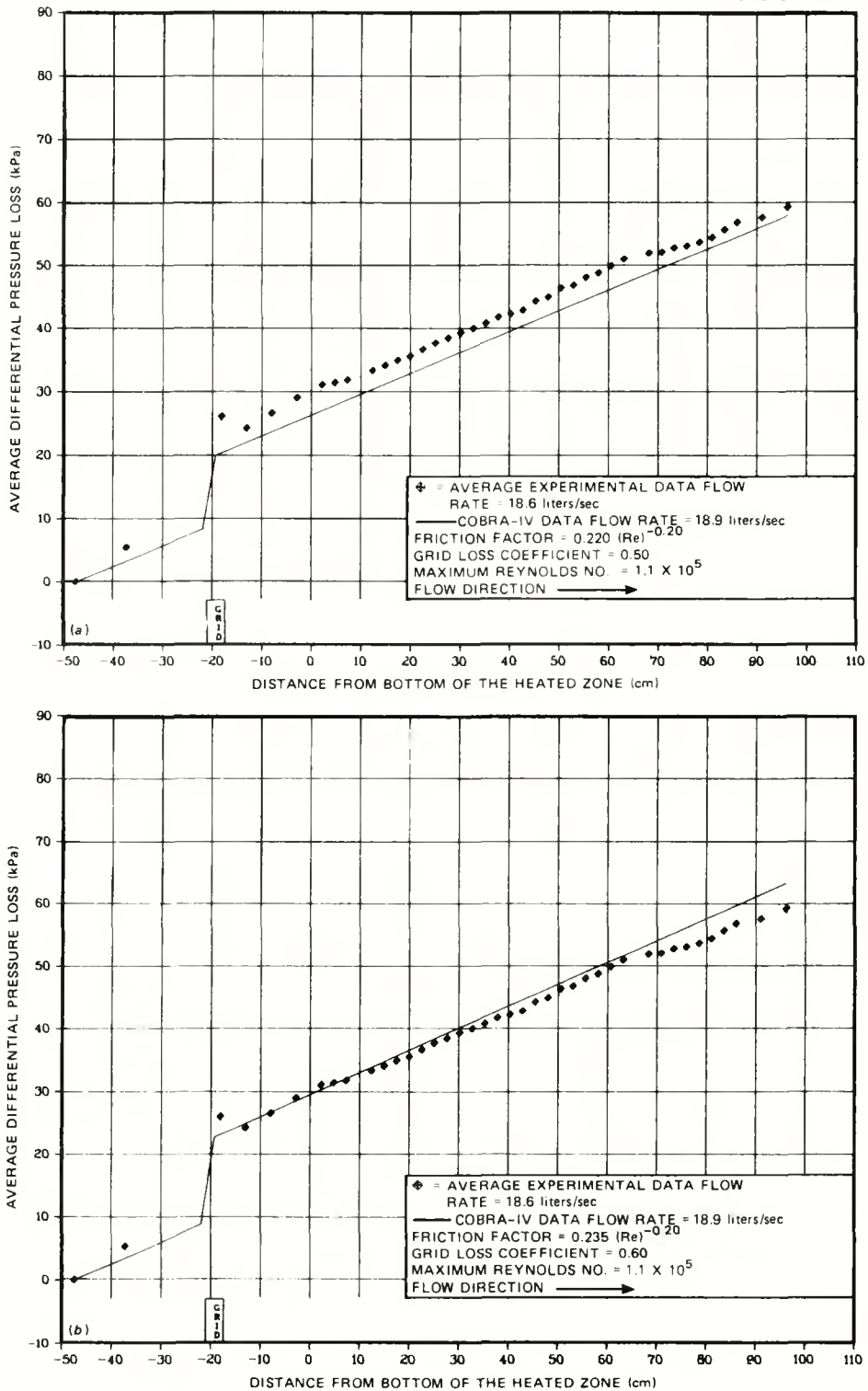


Fig. 5.4. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.6 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

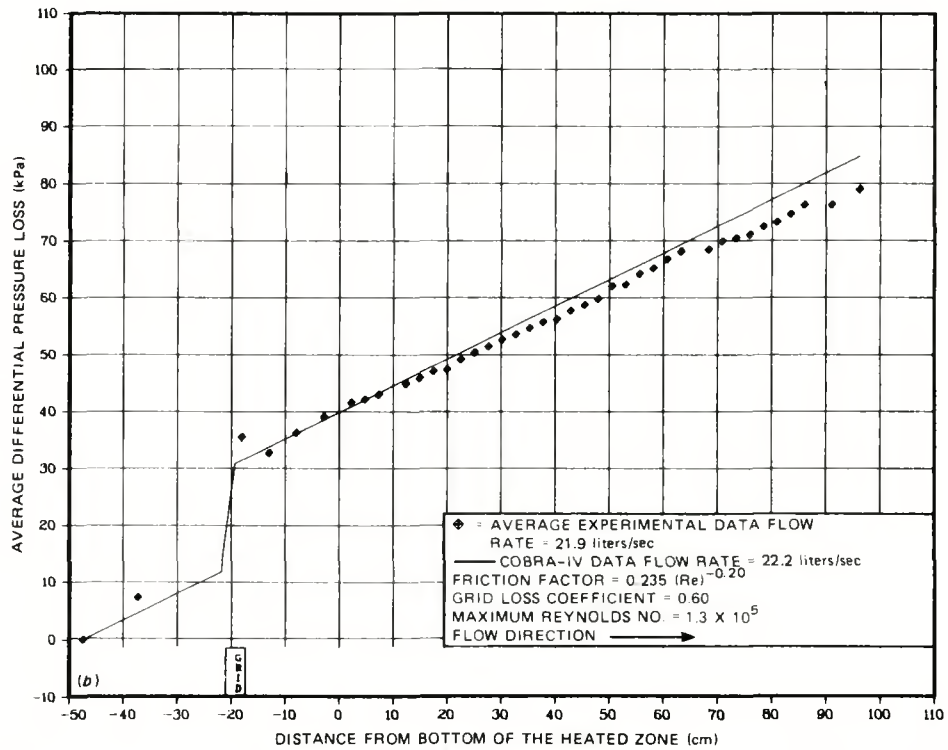
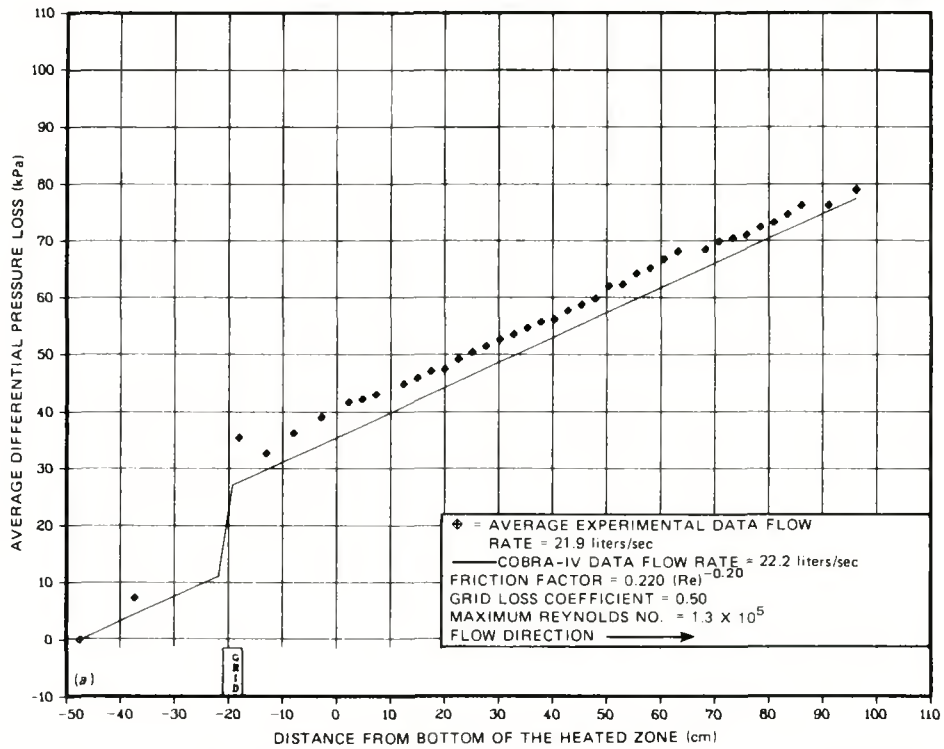


Fig. 5.5. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 21.9 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

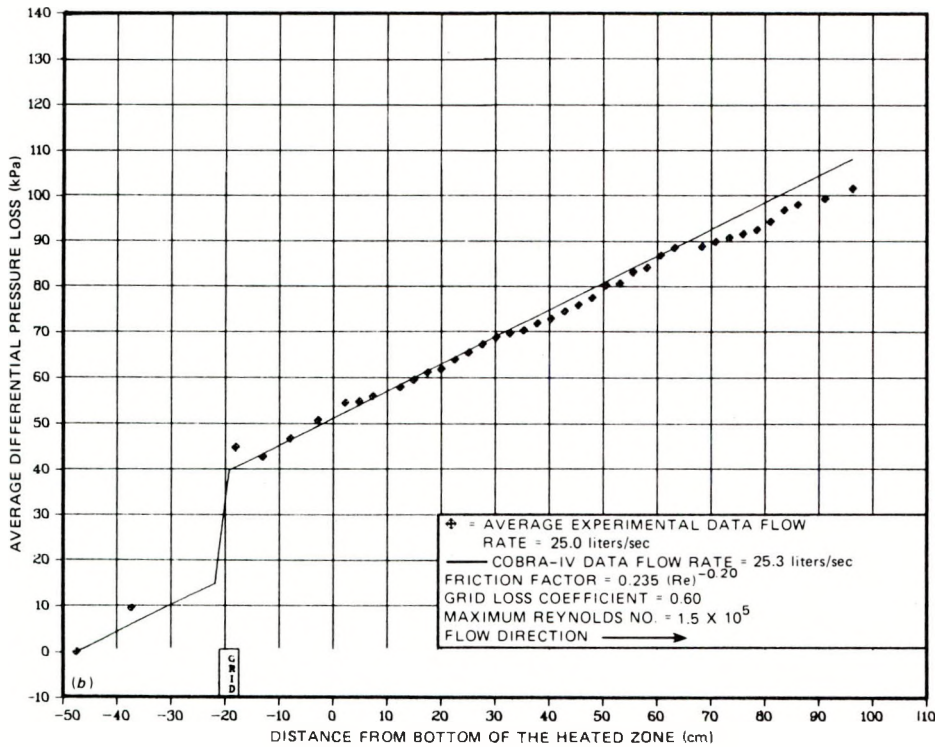
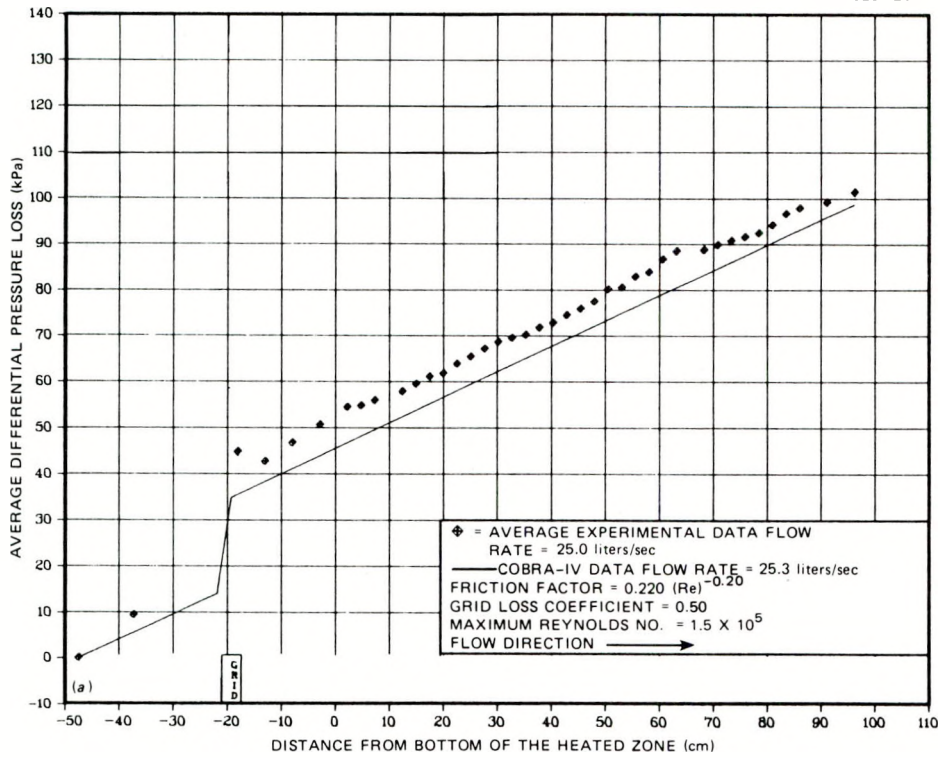


Fig. 5.6. Comparison of the first single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 25.0 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

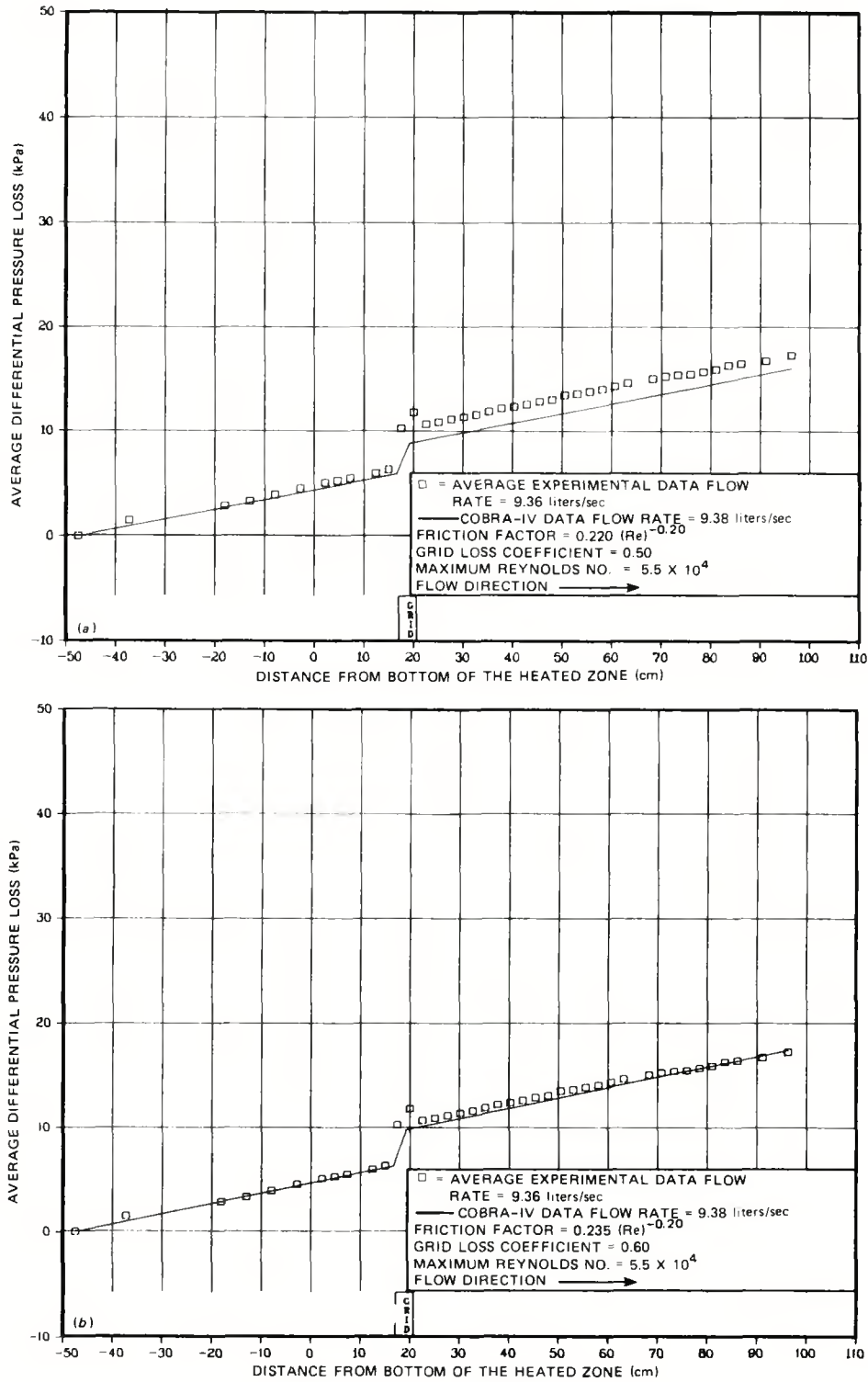


Fig. 5.7. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.36 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

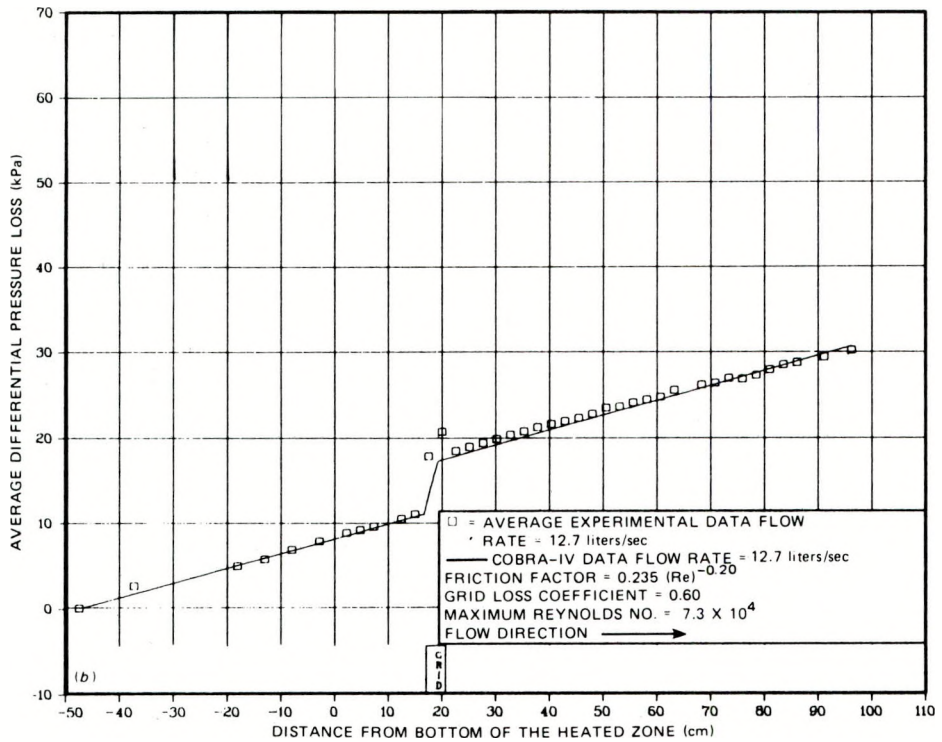
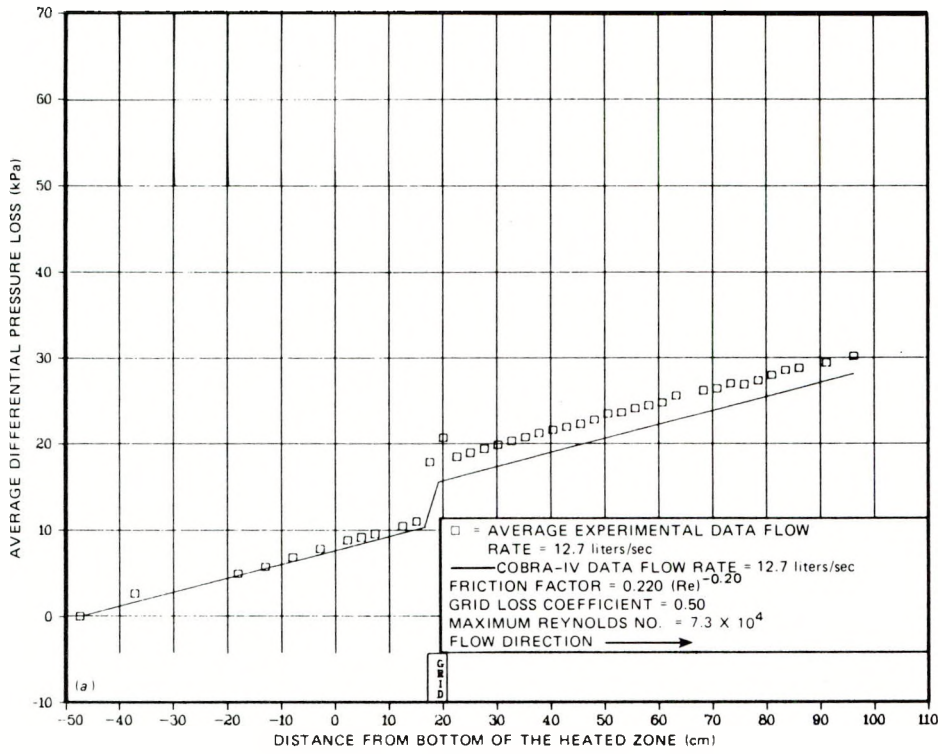


Fig. 5.8. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.7 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

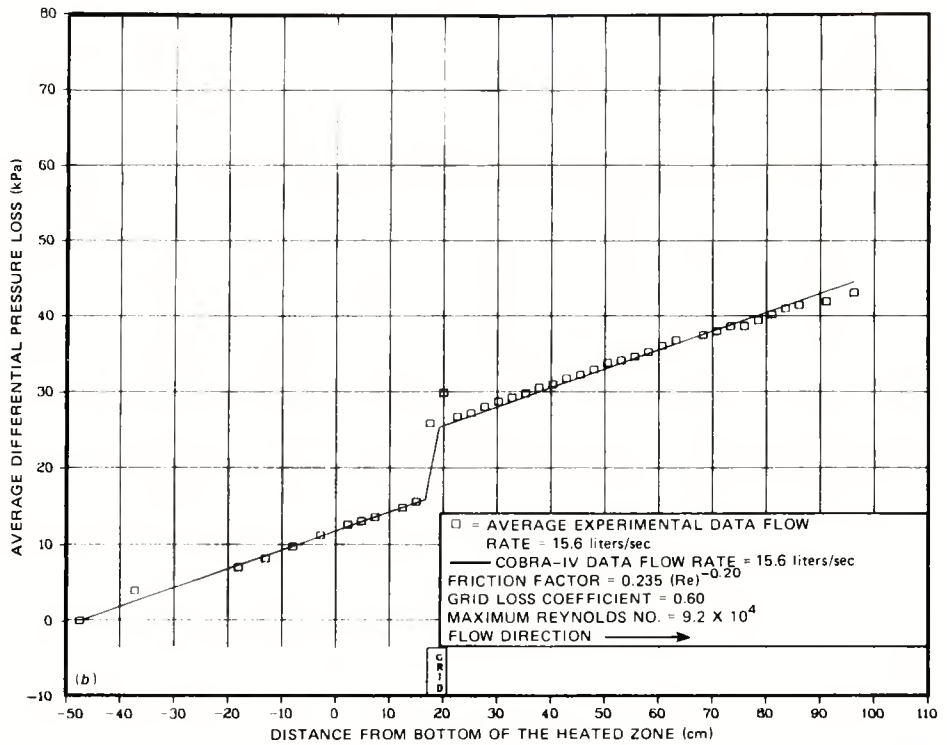
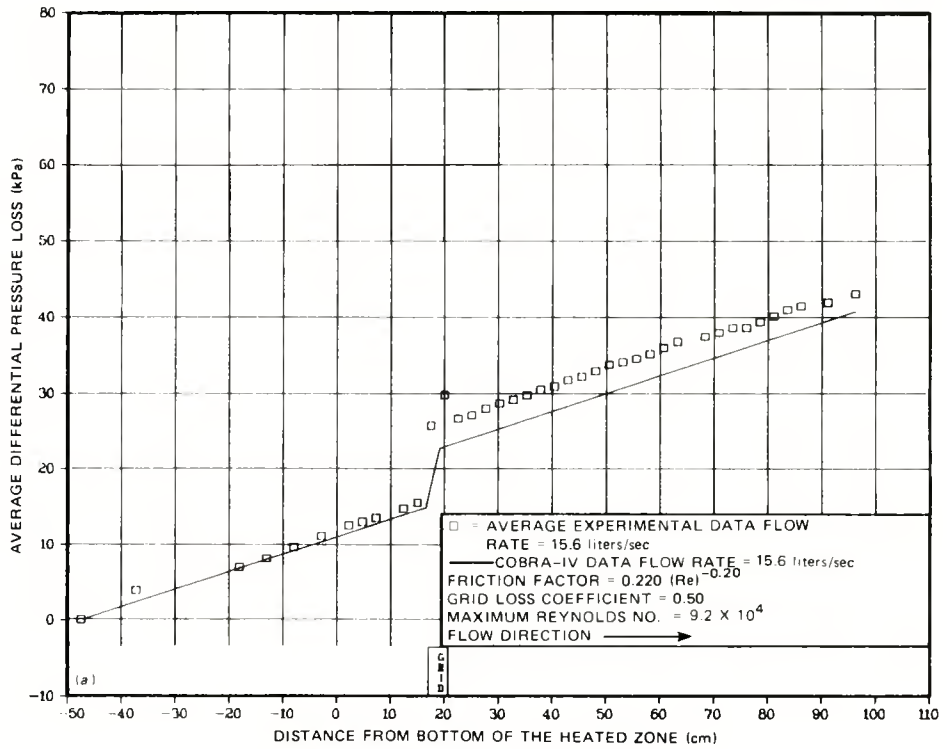


Fig. 5.9. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 15.6 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

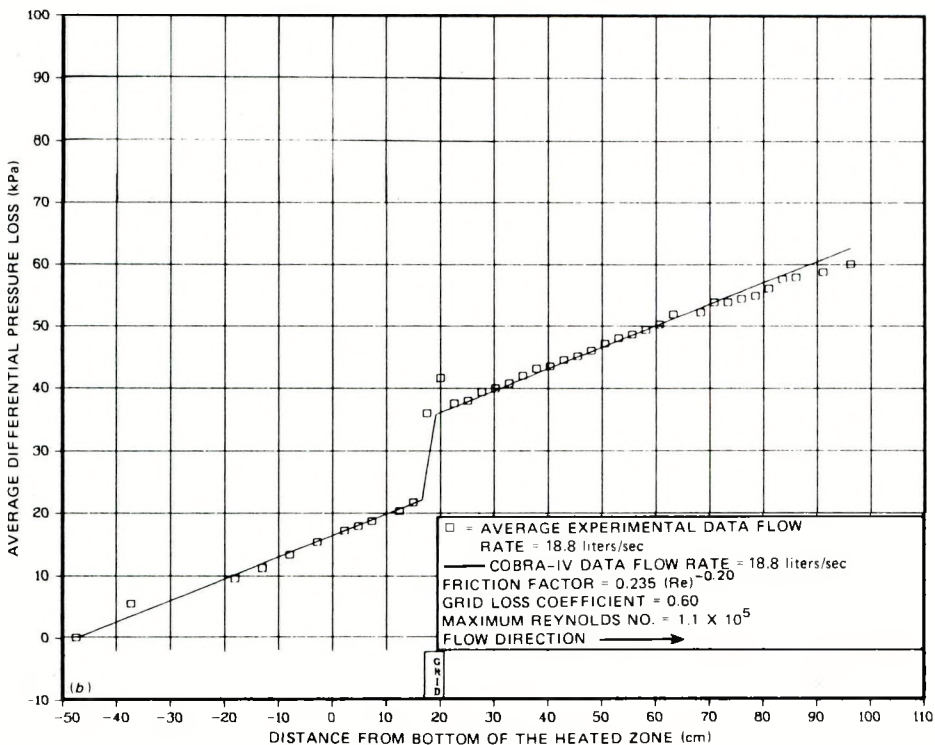
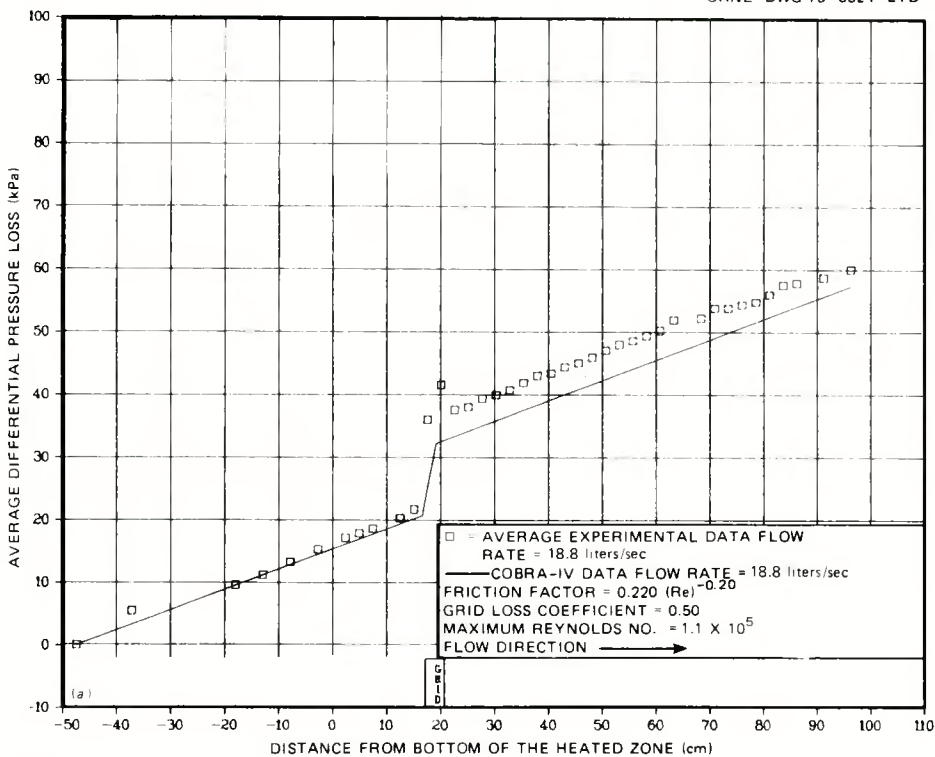


Fig. 5.10. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.8 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

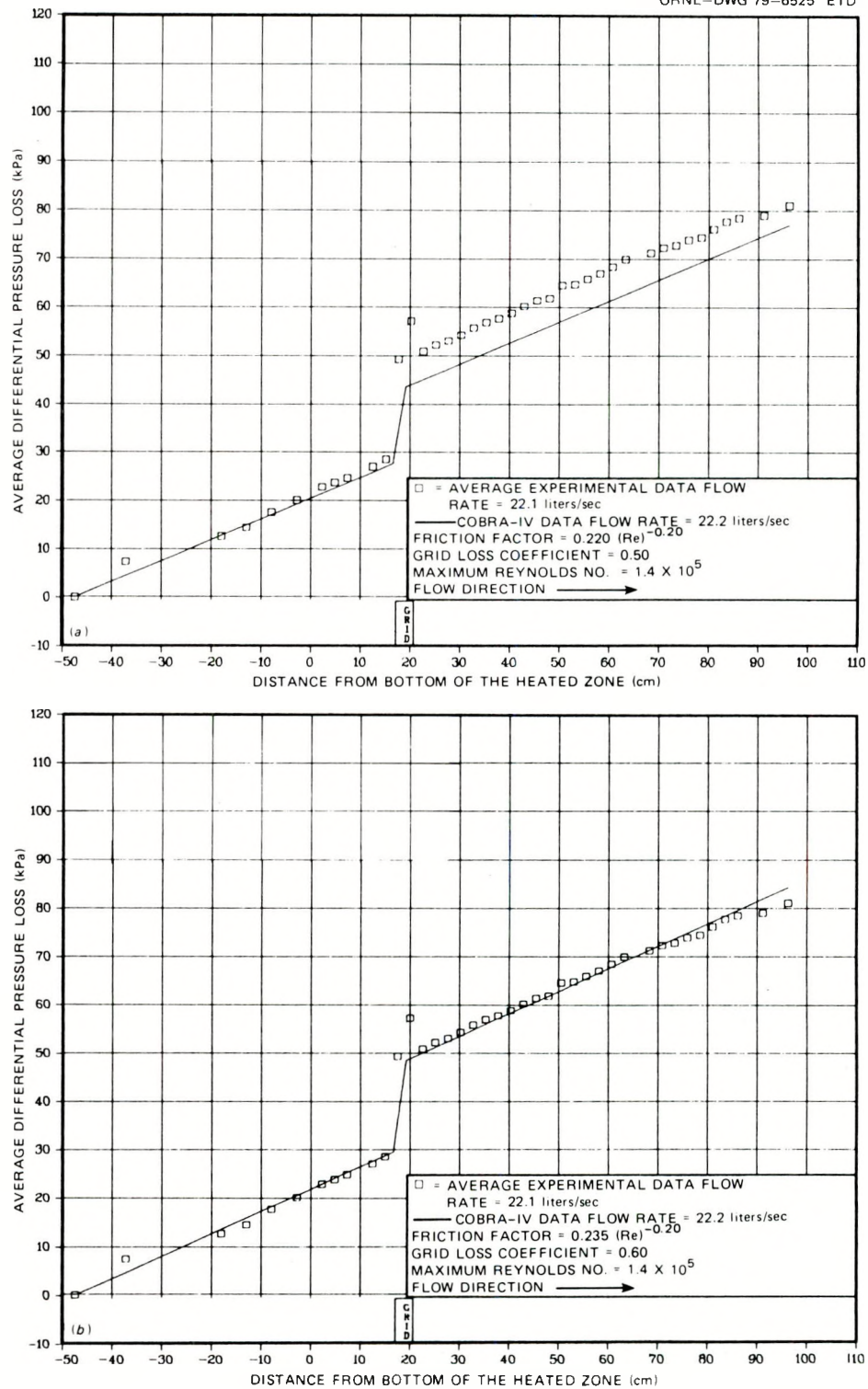


Fig. 5.11. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 22.1 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

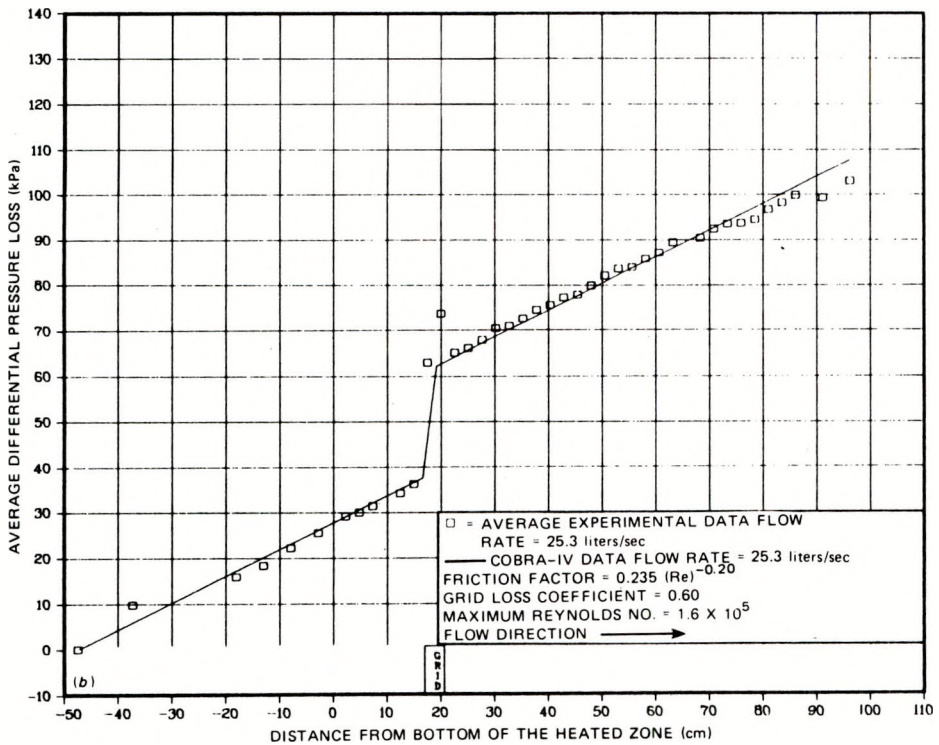
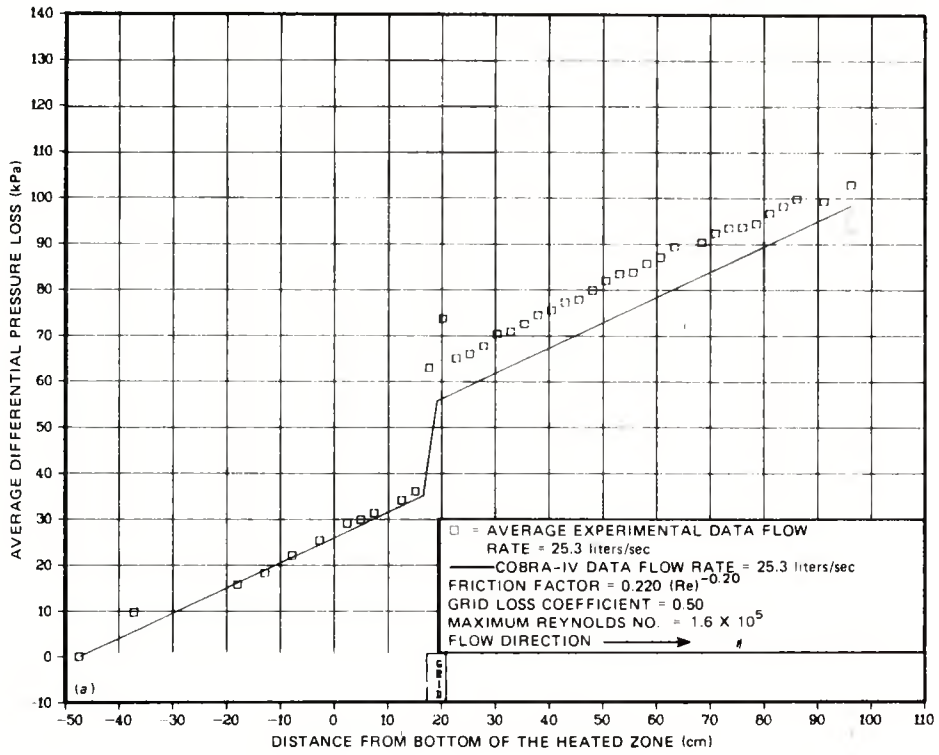


Fig. 5.12. Comparison of the second single-grid test experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 25.3 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

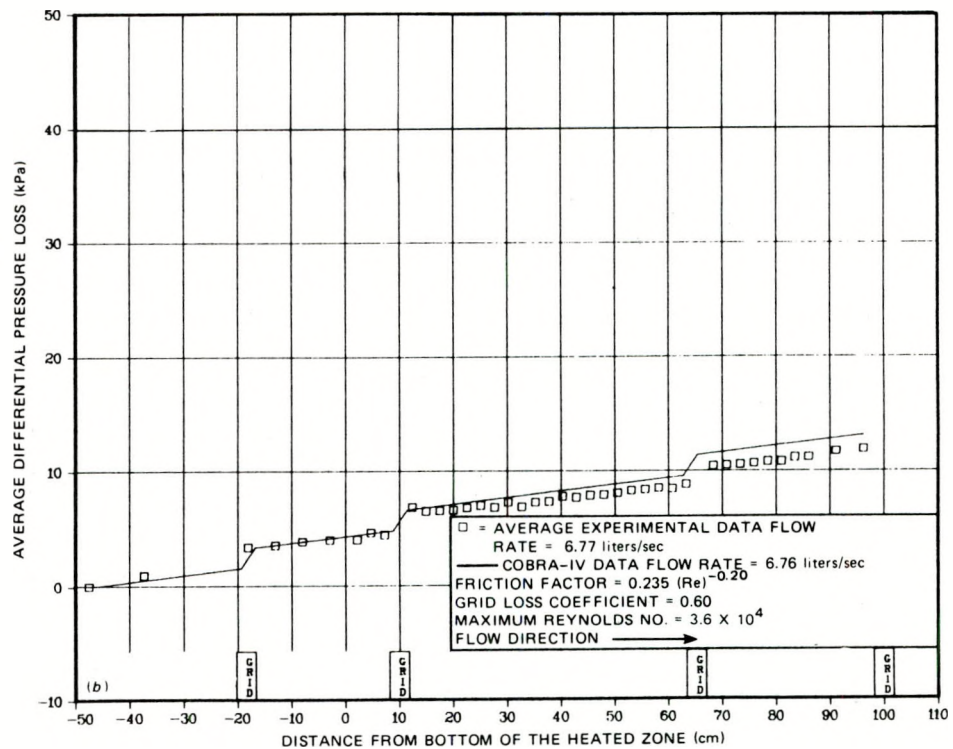
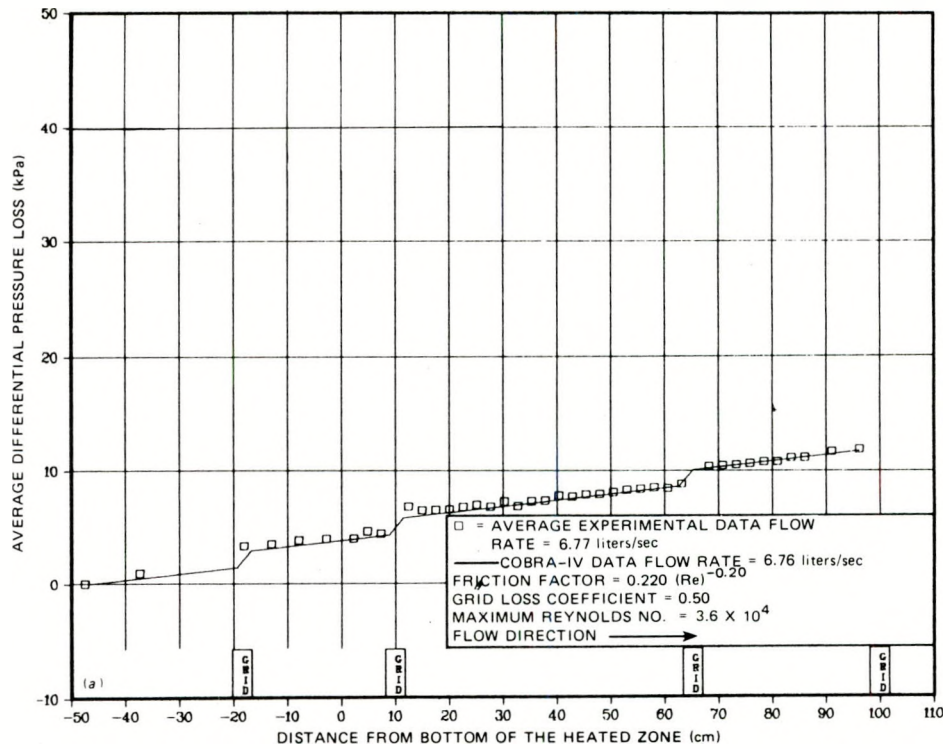


Fig. 5.13. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 6.77 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

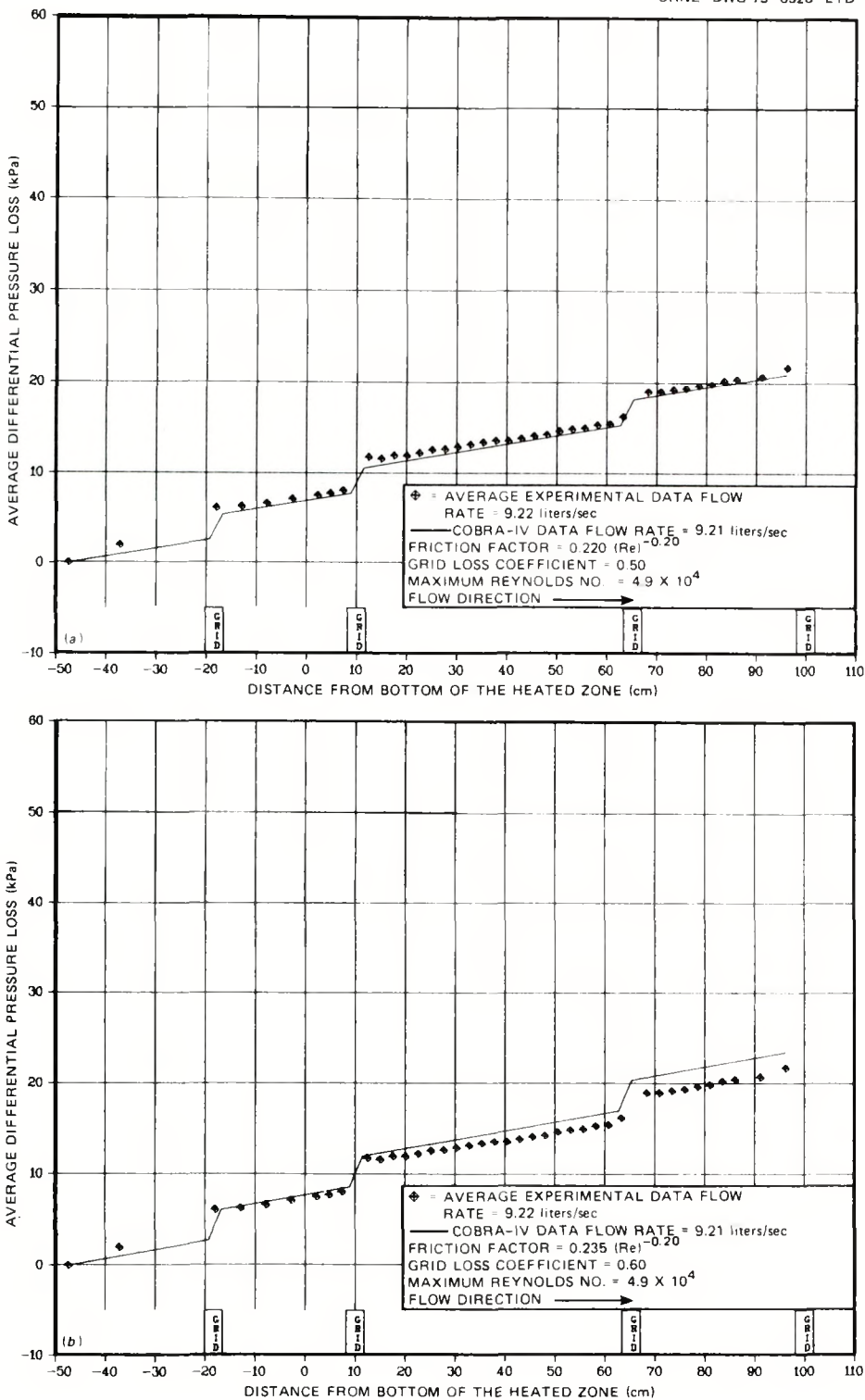


Fig. 5.14. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.22 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

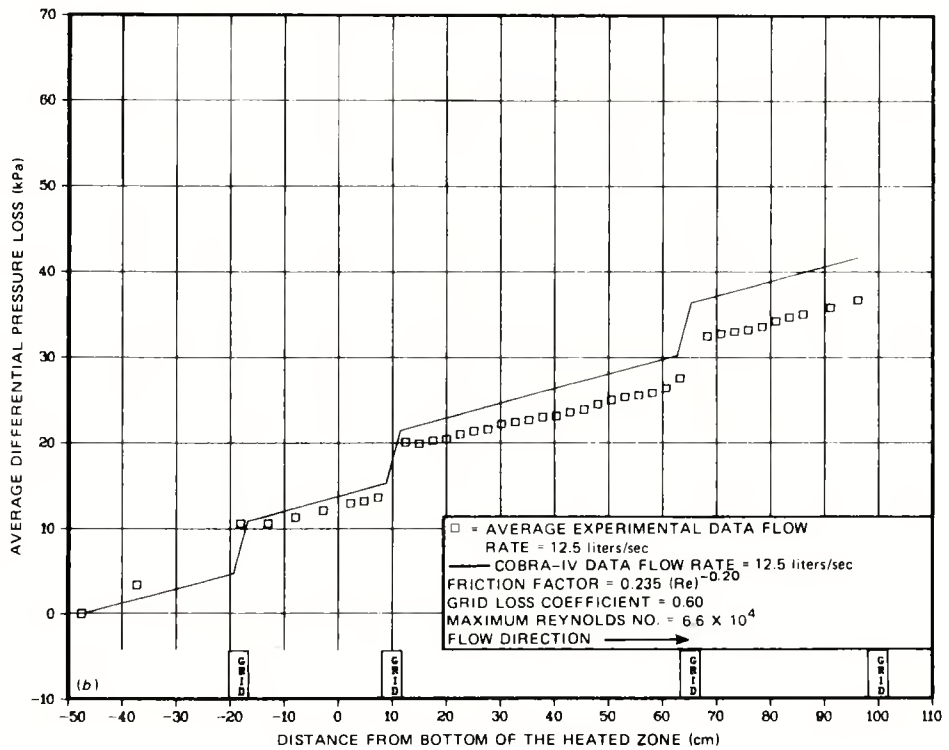
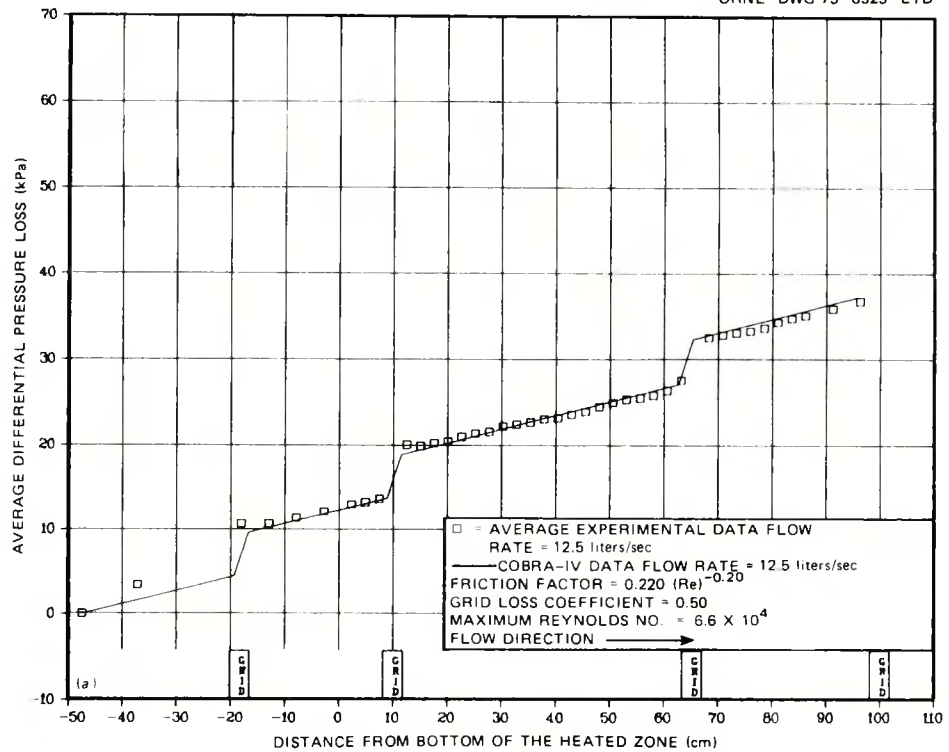


Fig. 5.15. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.5 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

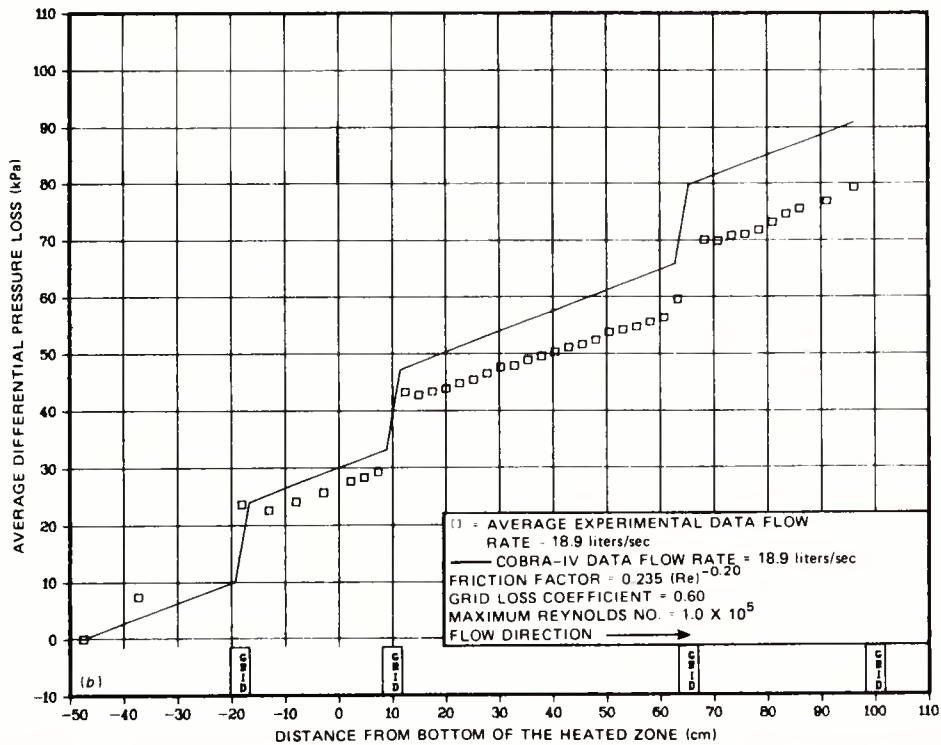
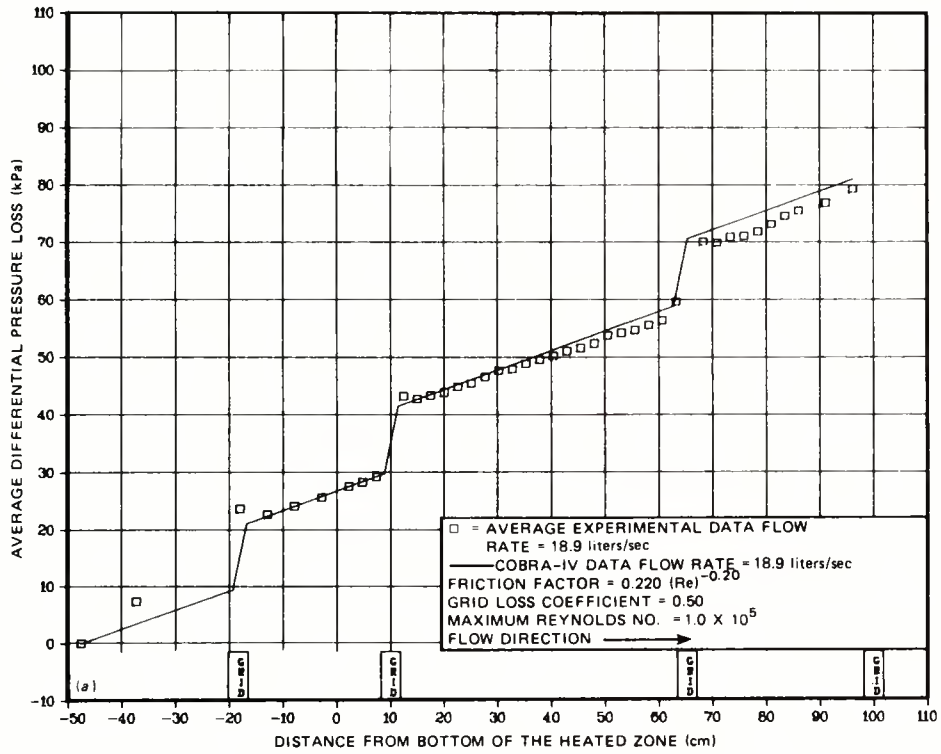


Fig. 5.16. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.9 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

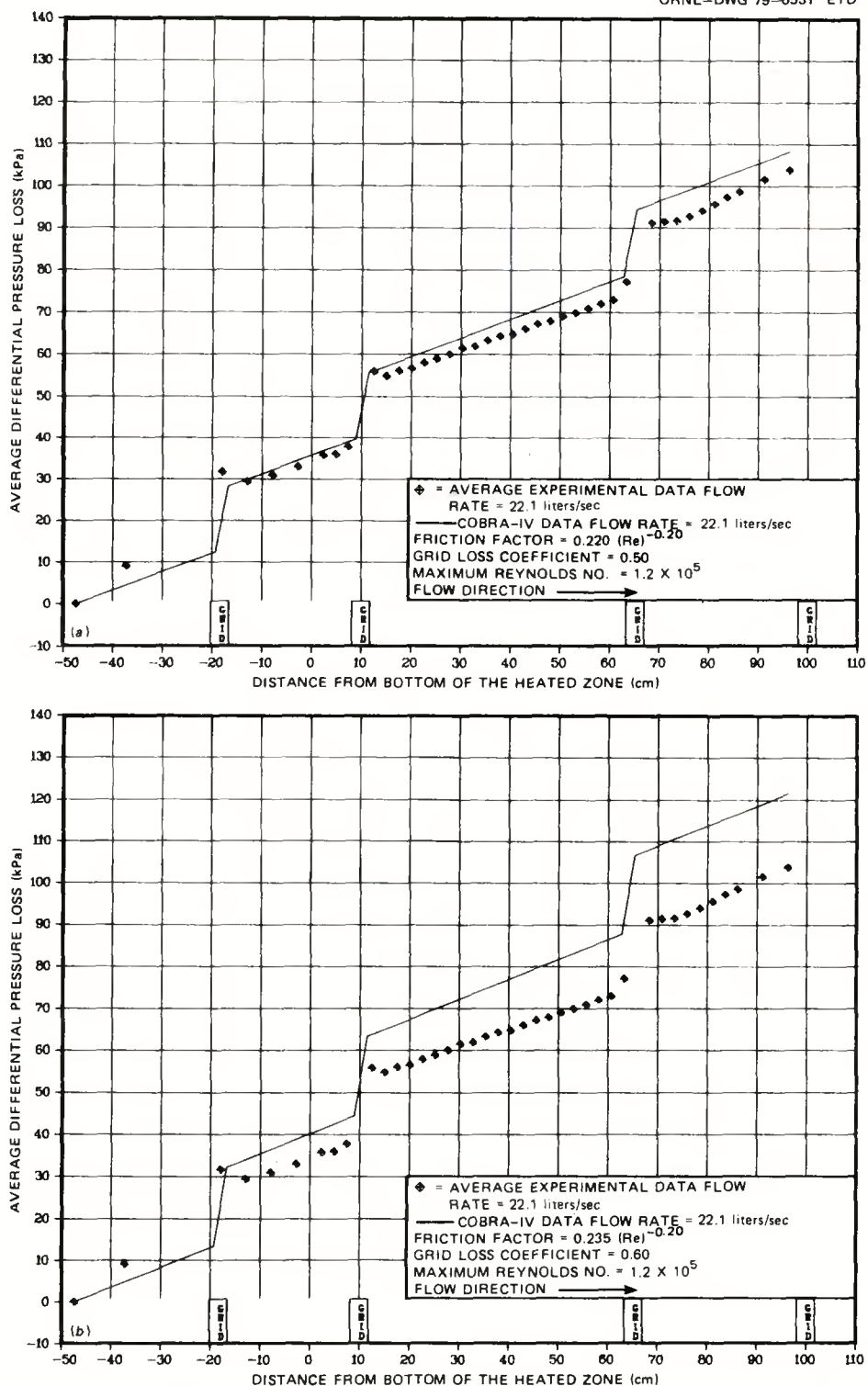


Fig. 5.17. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 22.1 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

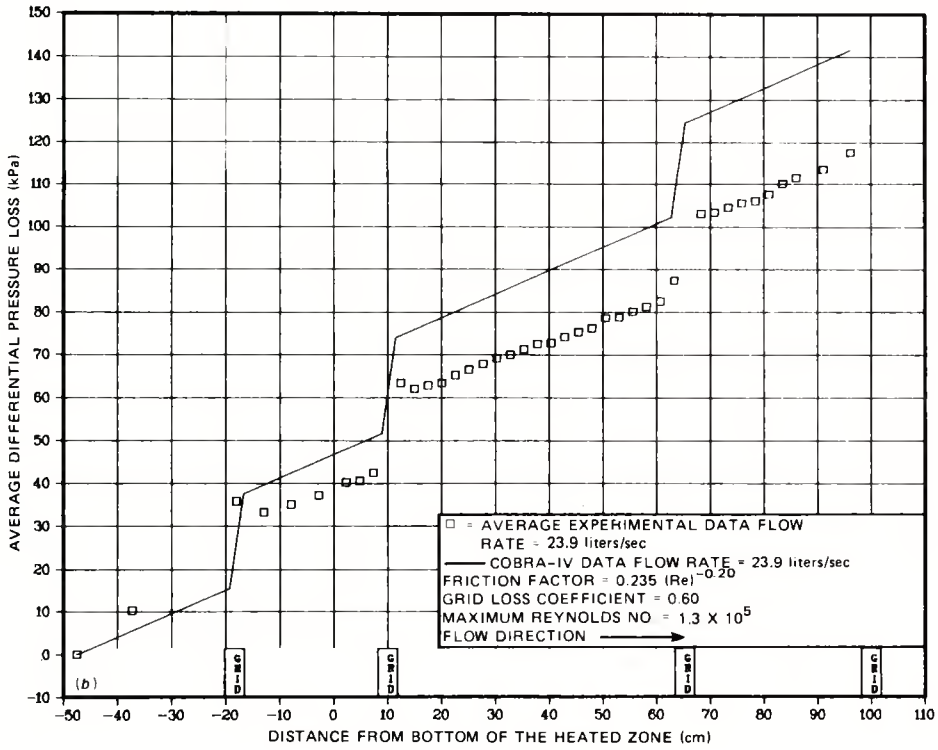
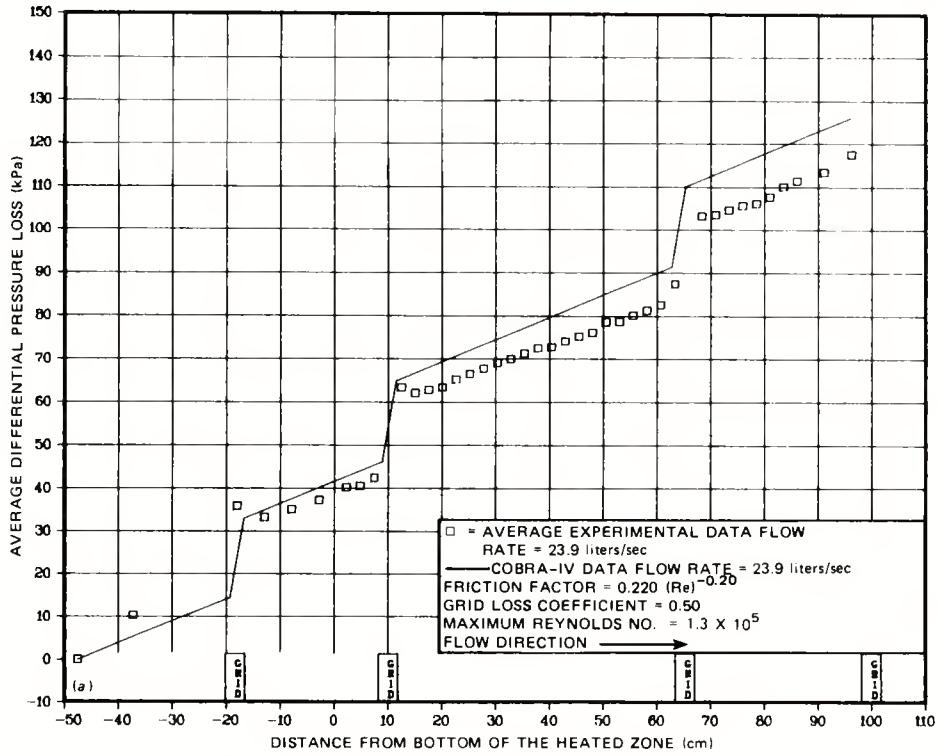


Fig. 5.18. Comparison of reference bundle/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 23.9 liters/sec. (a) Lower-limit; (b) upper-limit correlation values.

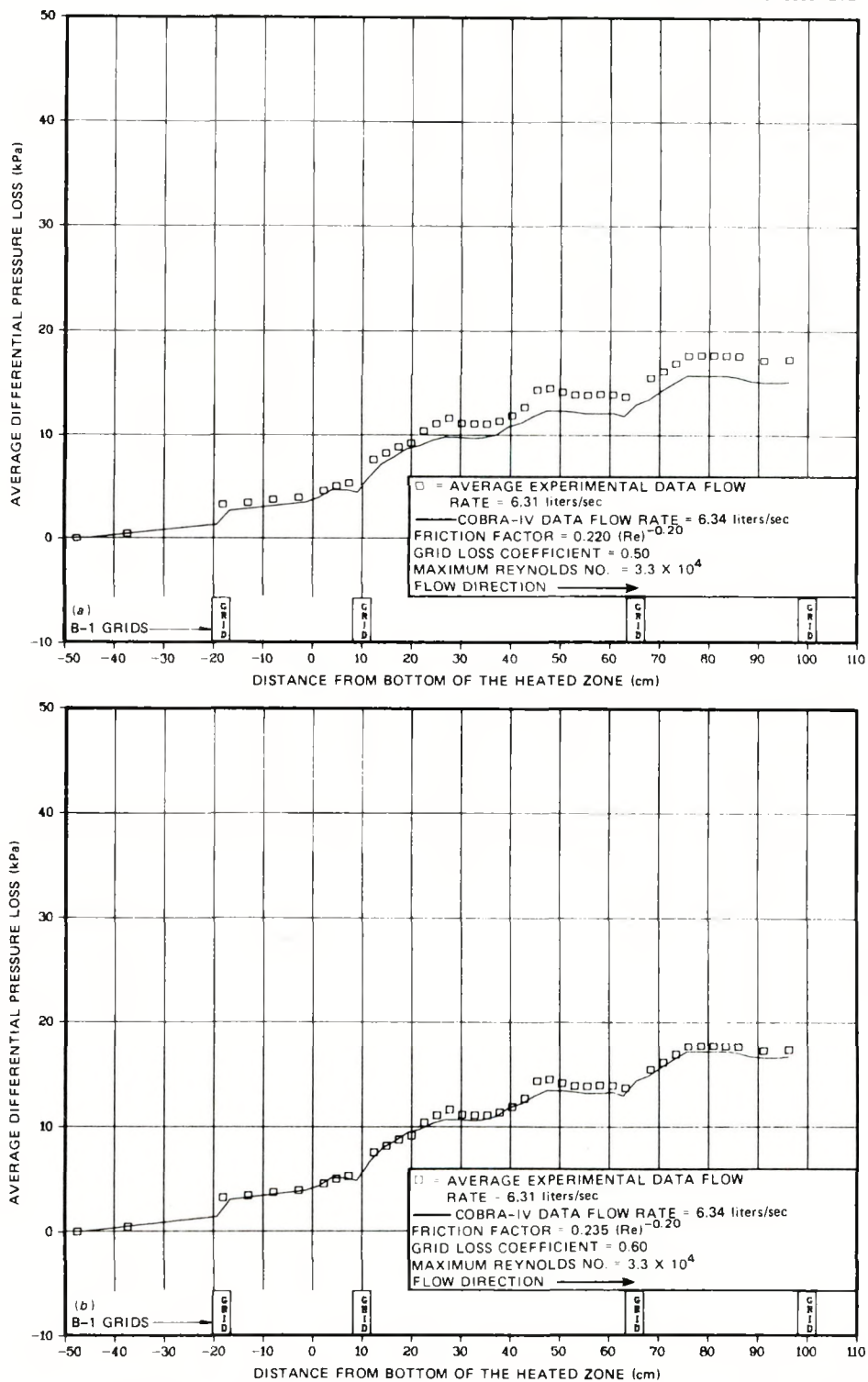


Fig. 5.19. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 6.31 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

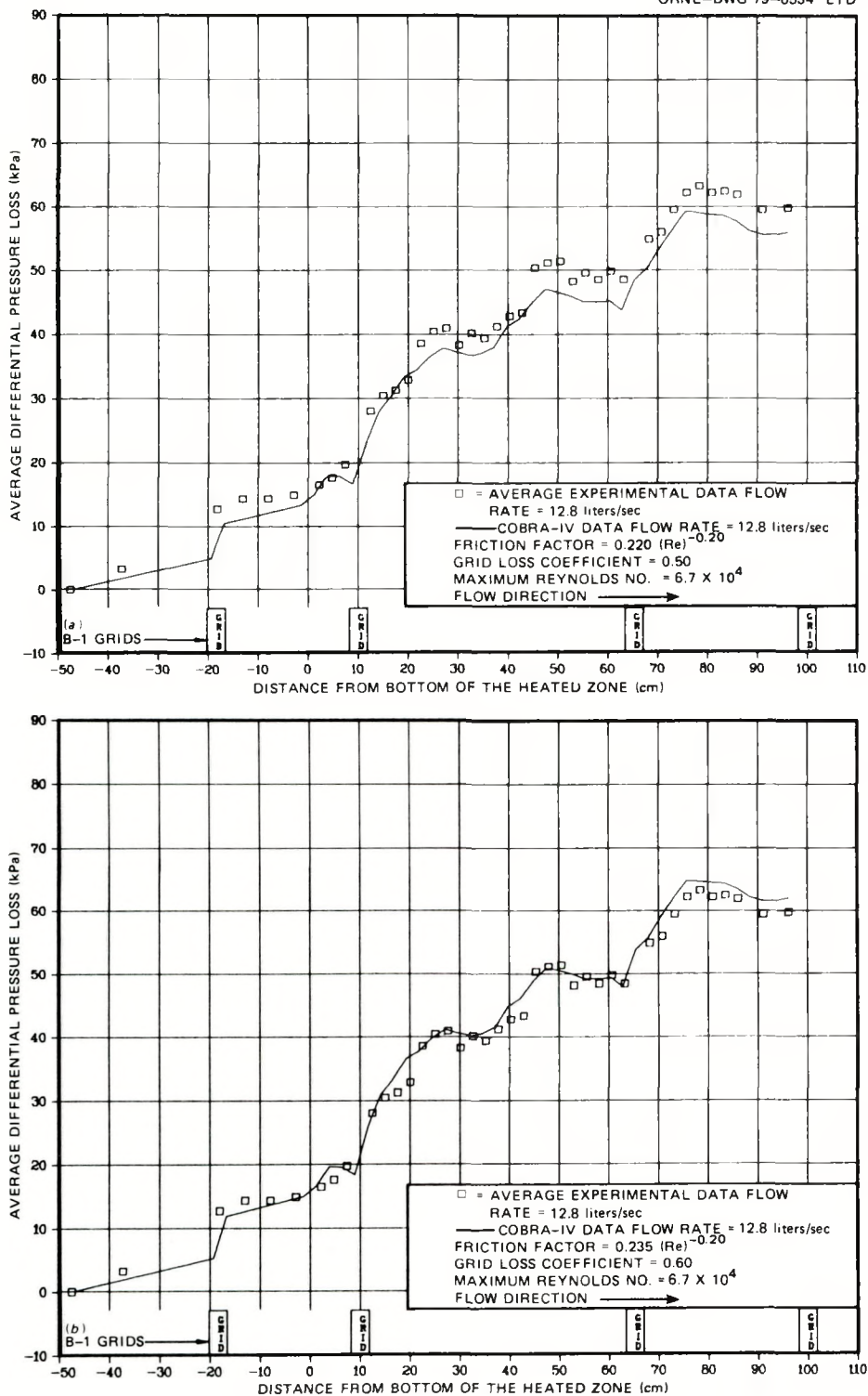


Fig. 5.20. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.8 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

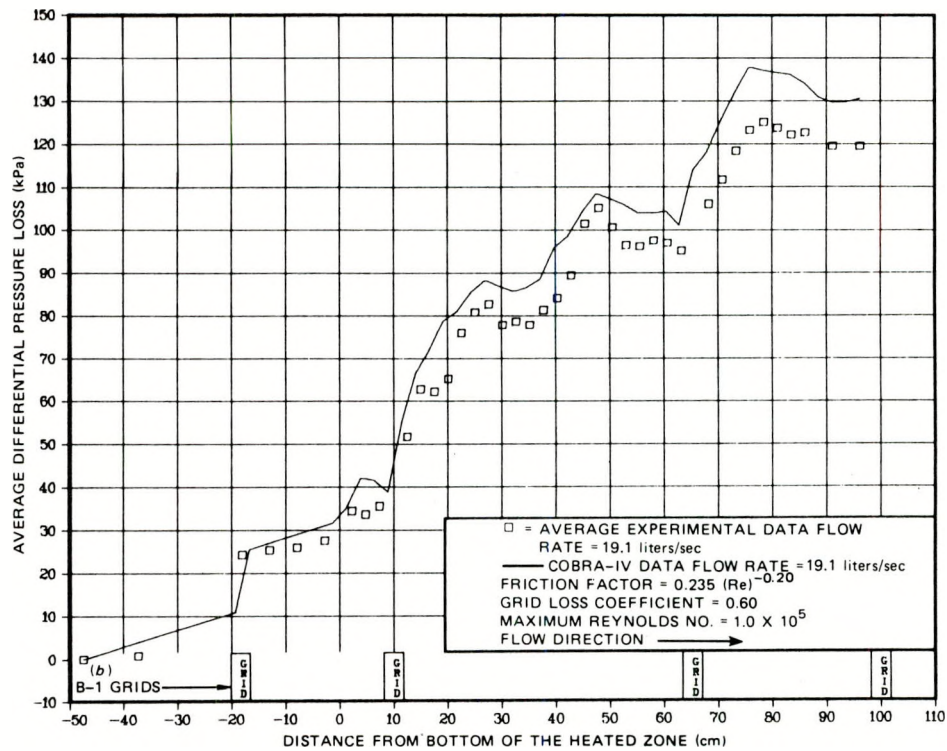
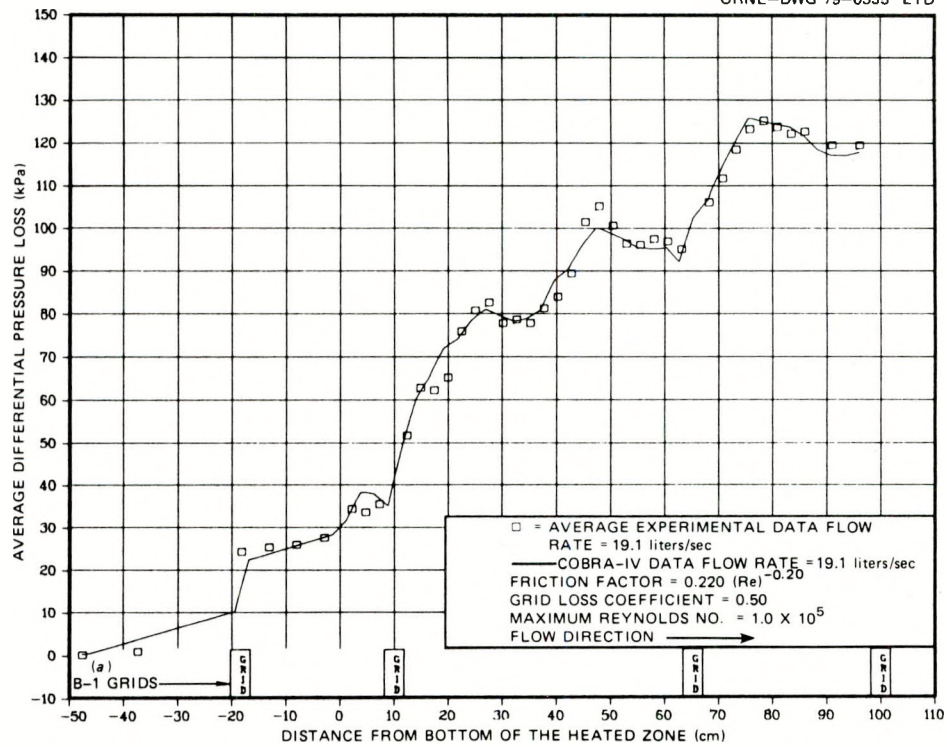


Fig. 5.21. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 19.1 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

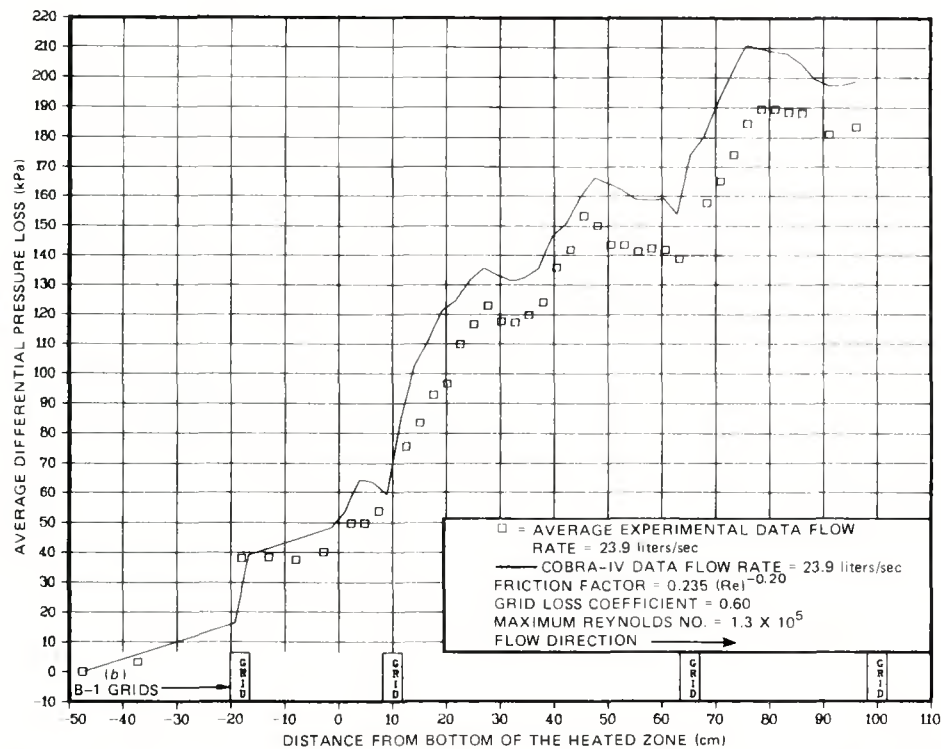
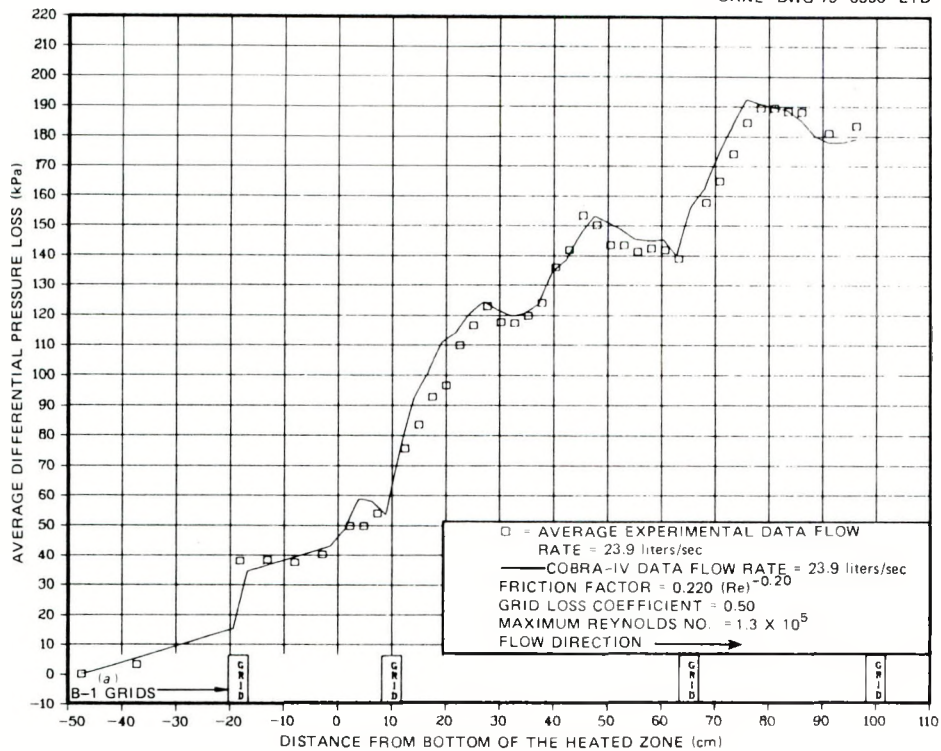


Fig. 5.22. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 23.9 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

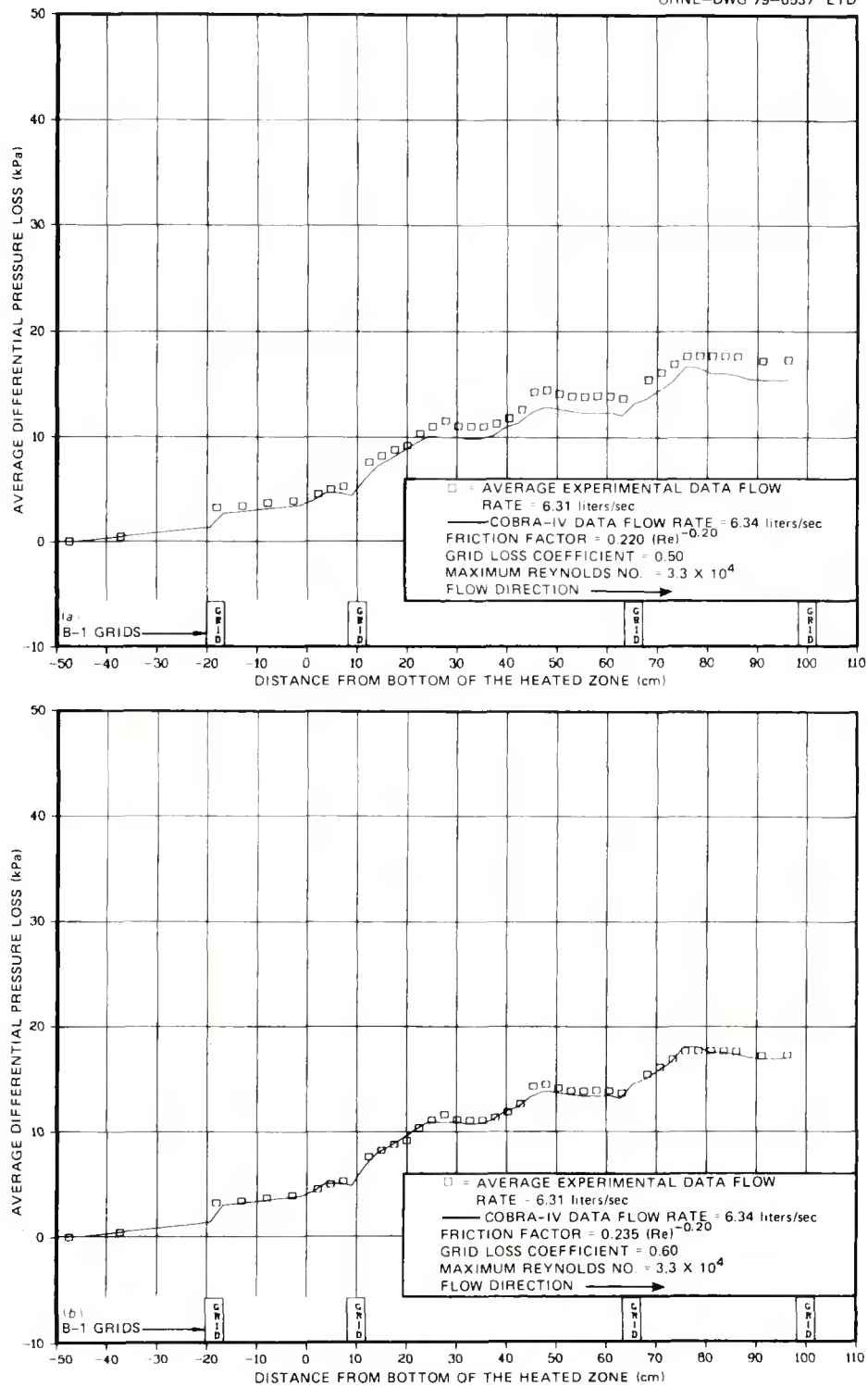


Fig. 5.23. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 6.31 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

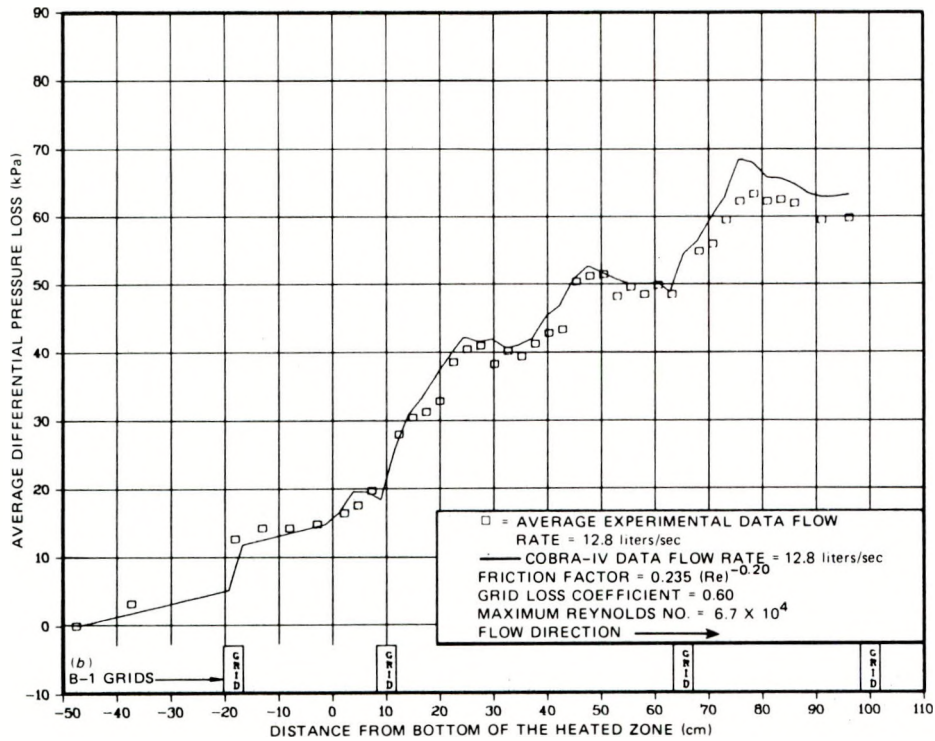
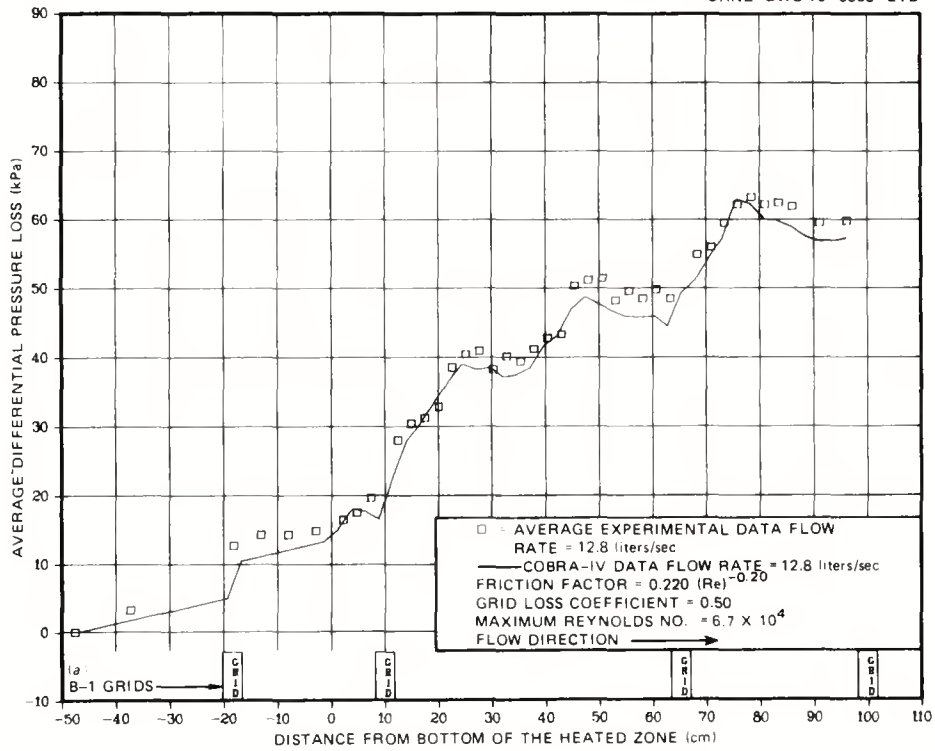


Fig. 5.24. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.8 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

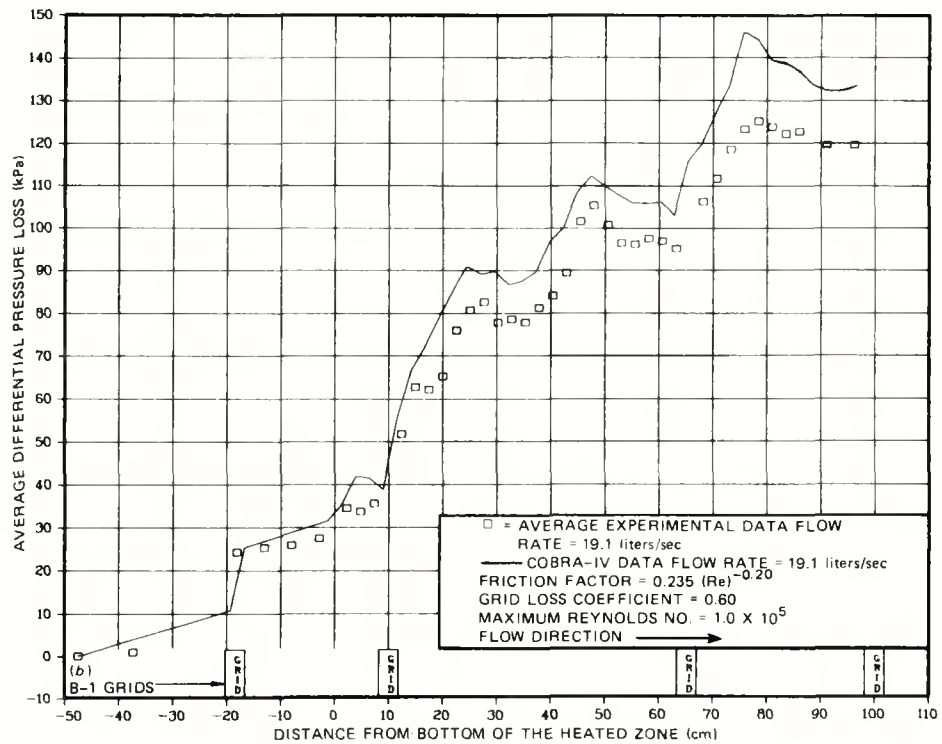
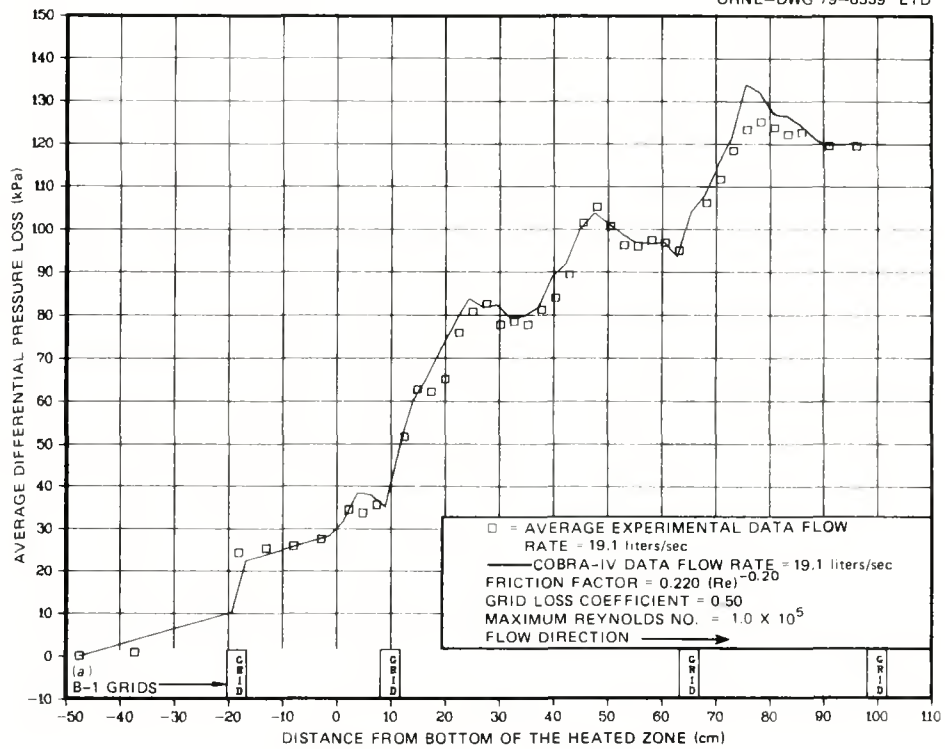


Fig. 5.25. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 19.1 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

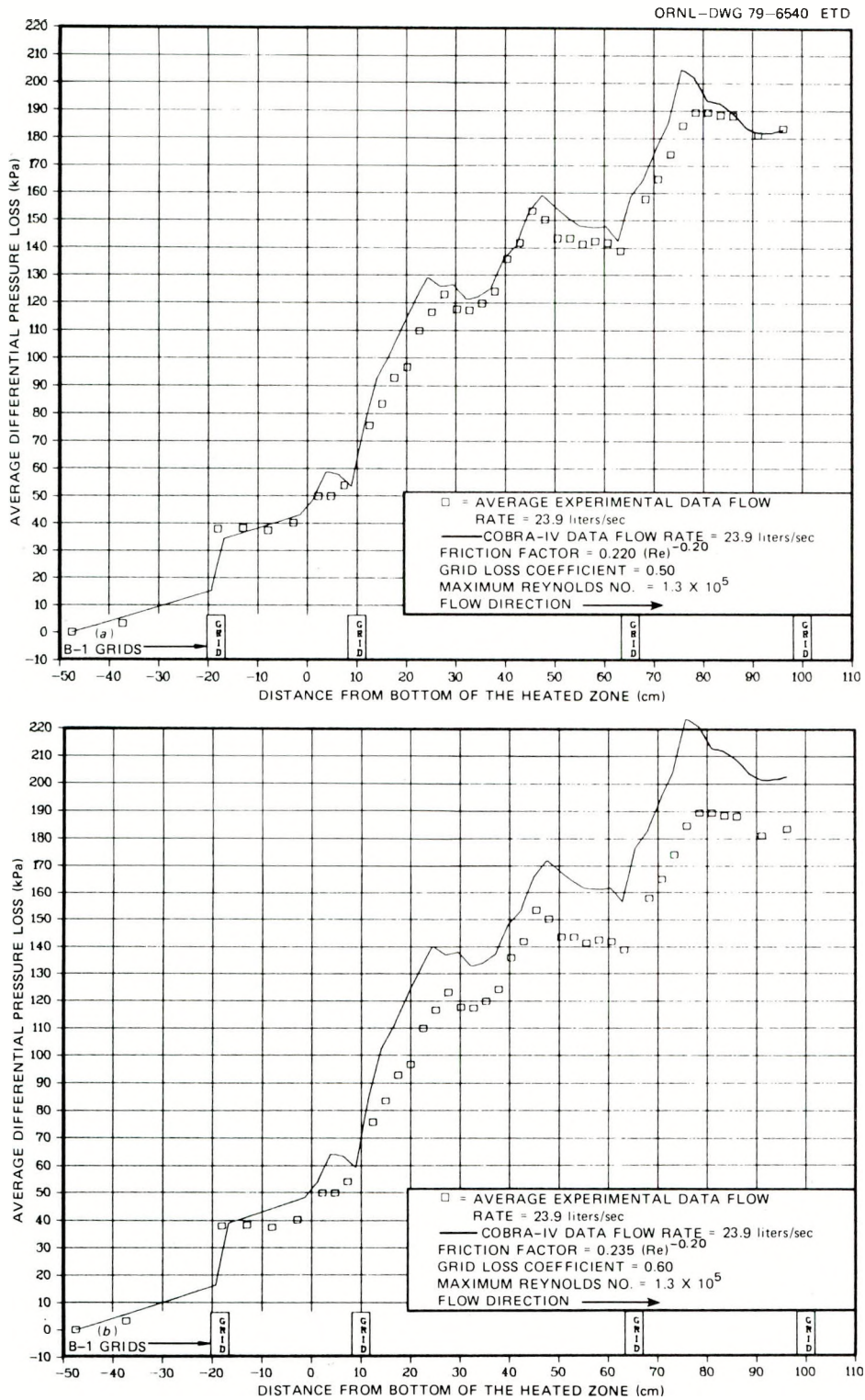


Fig. 5.26. Comparison of B-1/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 23.9 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

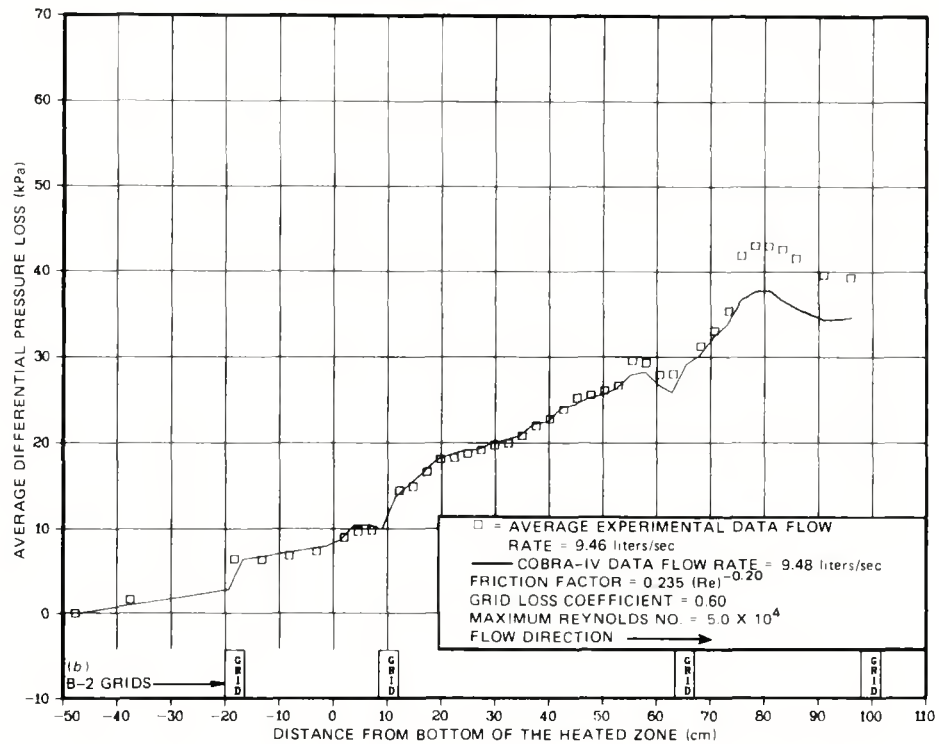
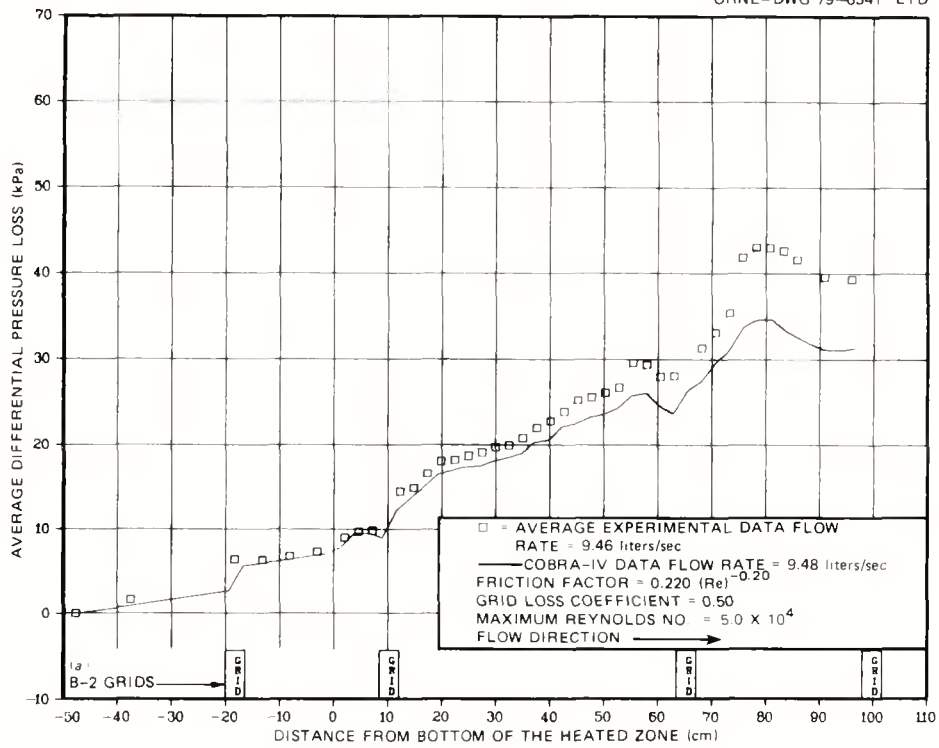


Fig. 5.27. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.46 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

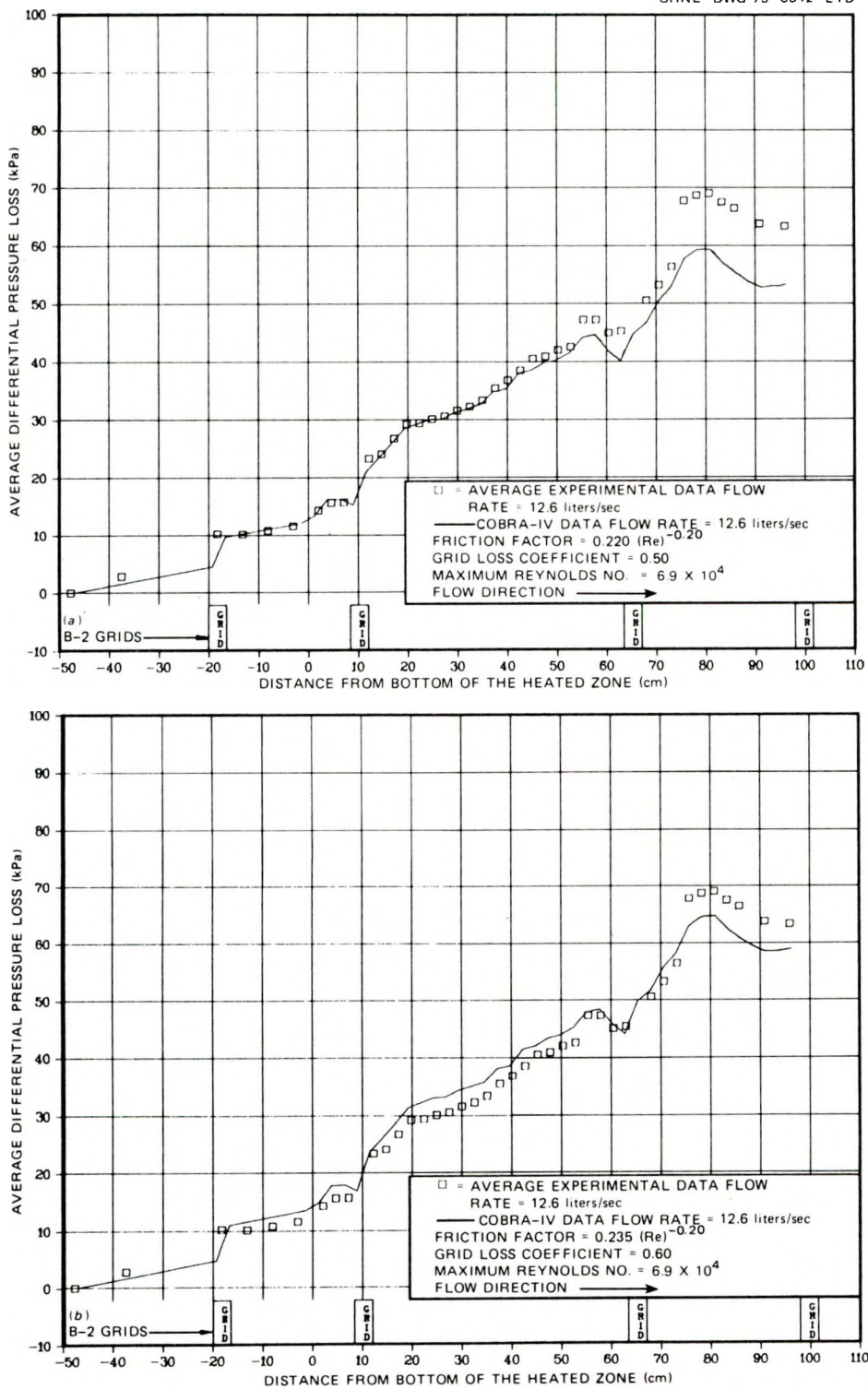


Fig. 5.28. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.6 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

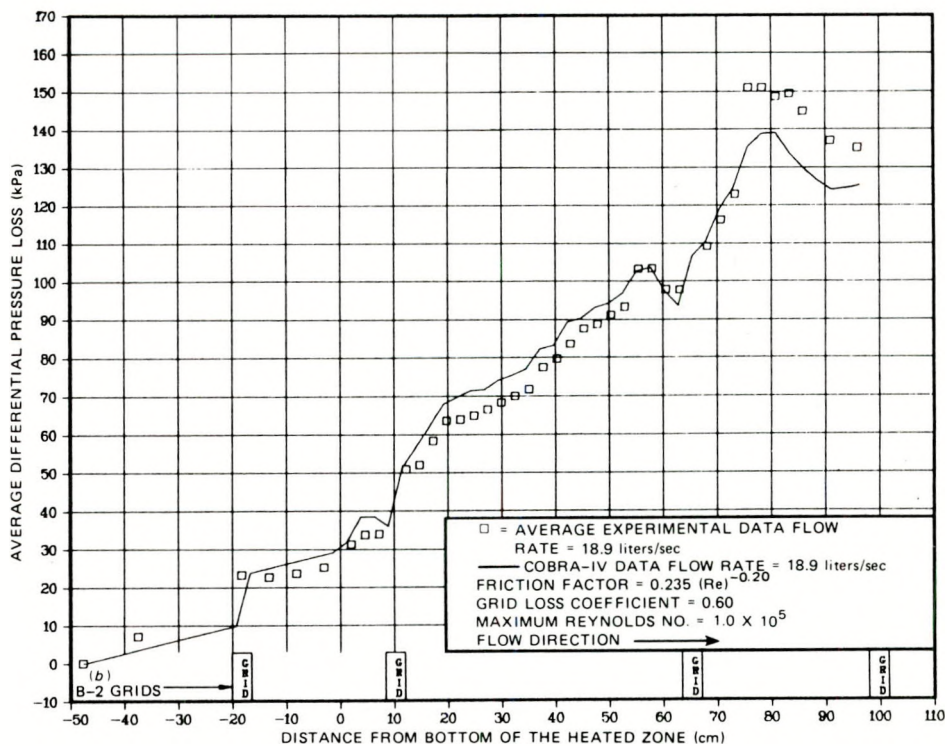
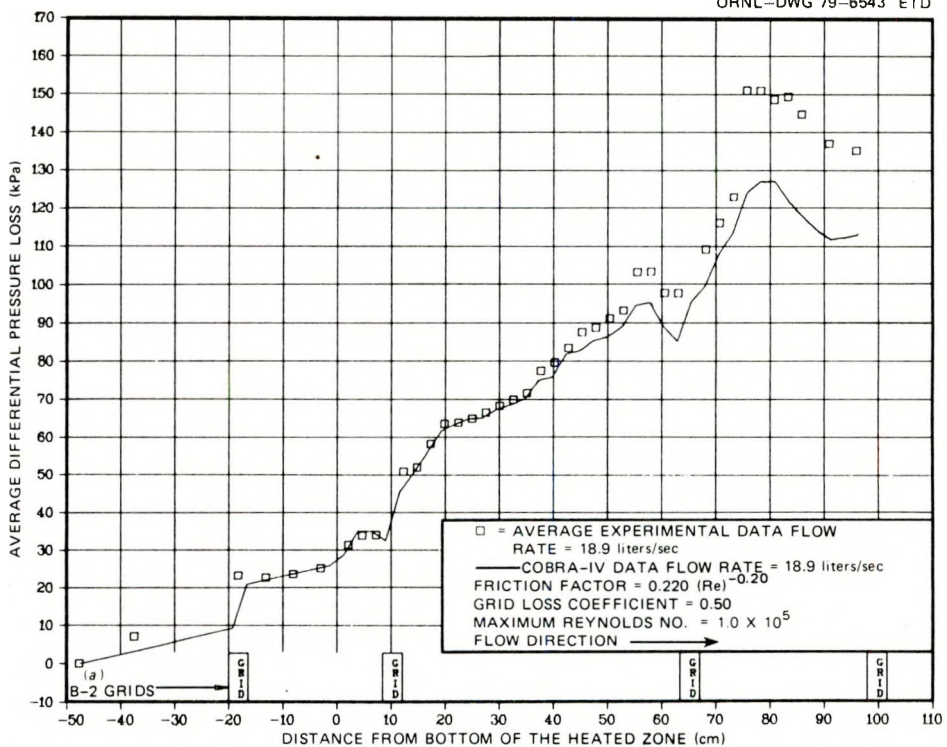


Fig. 5.29. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.9 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

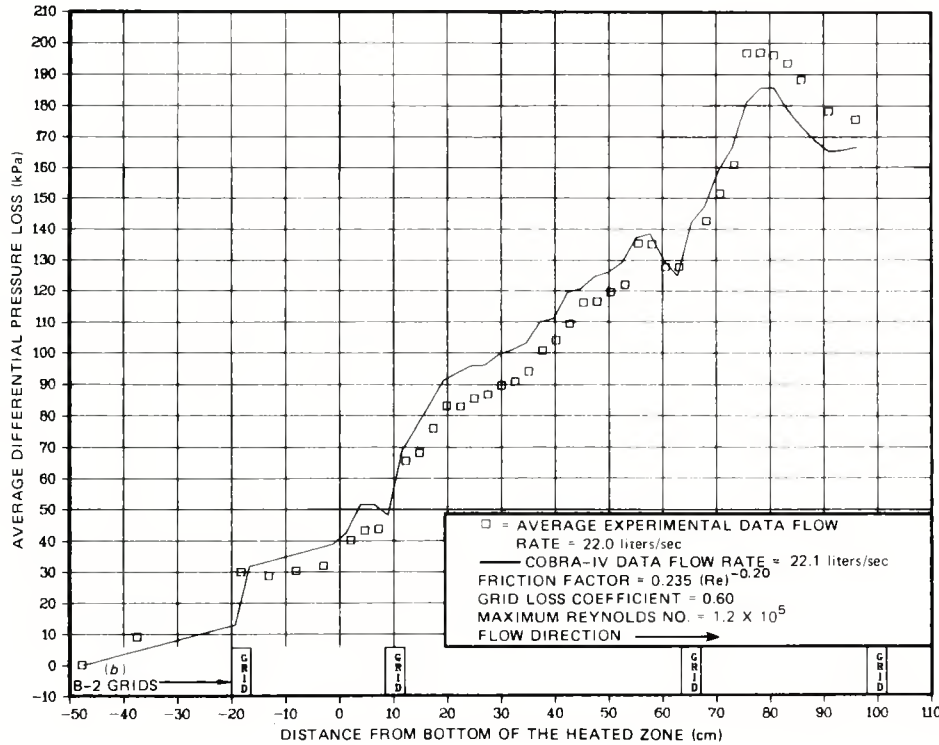
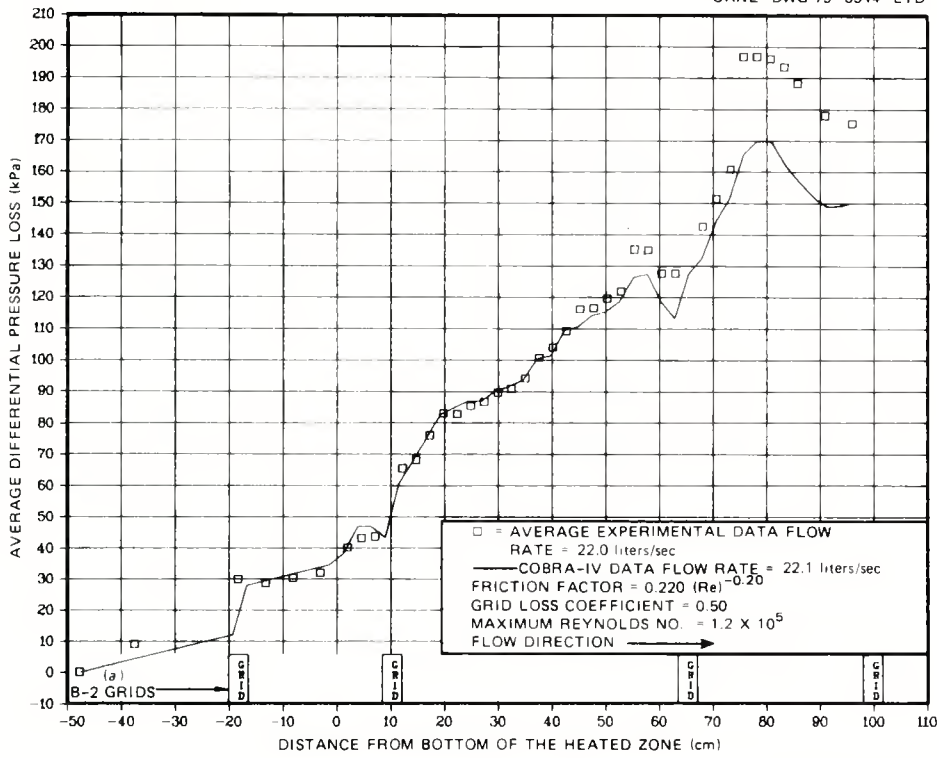


Fig. 5.30. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 22.0 liters/sec; minimum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

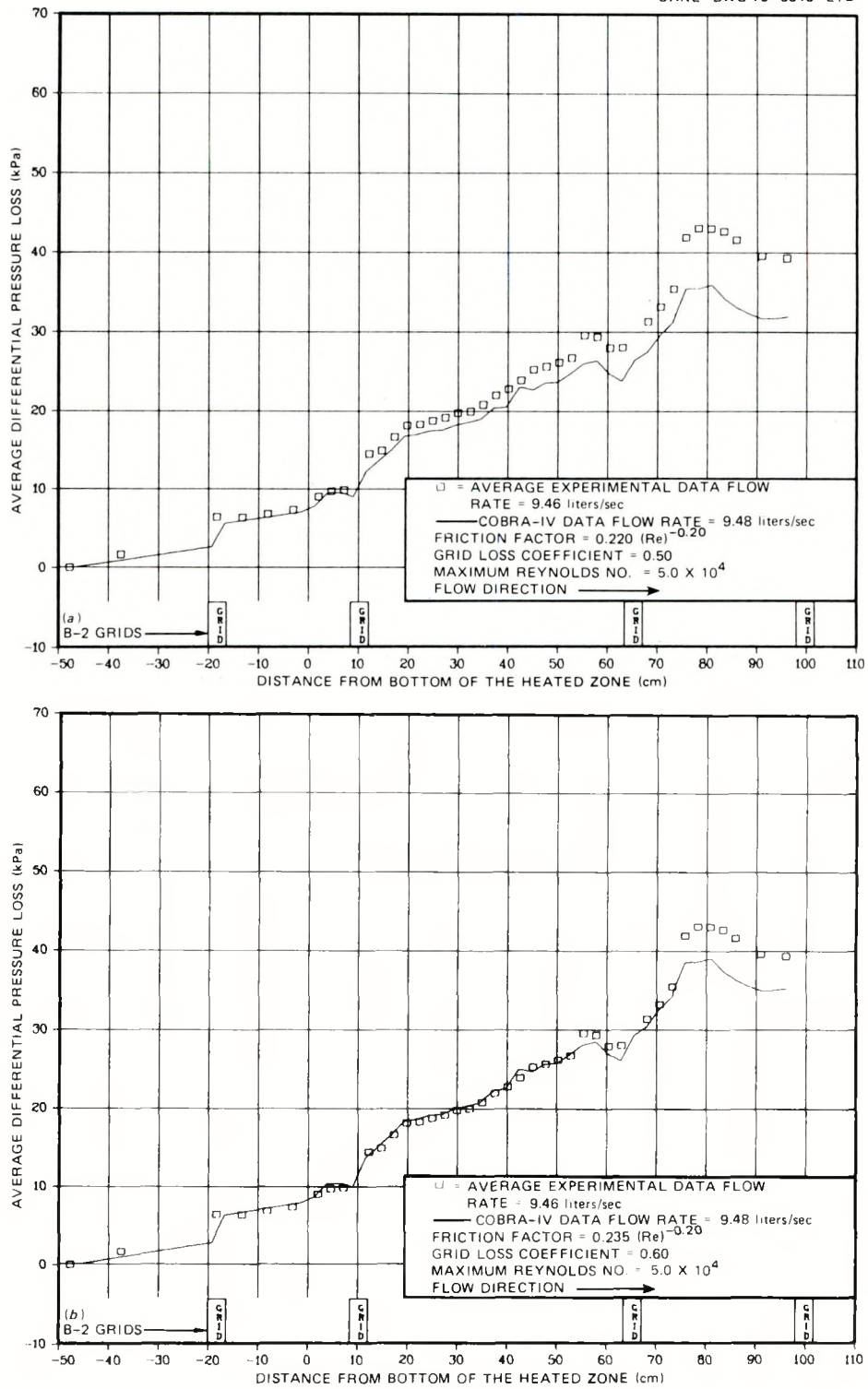


Fig. 5.31. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.46 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

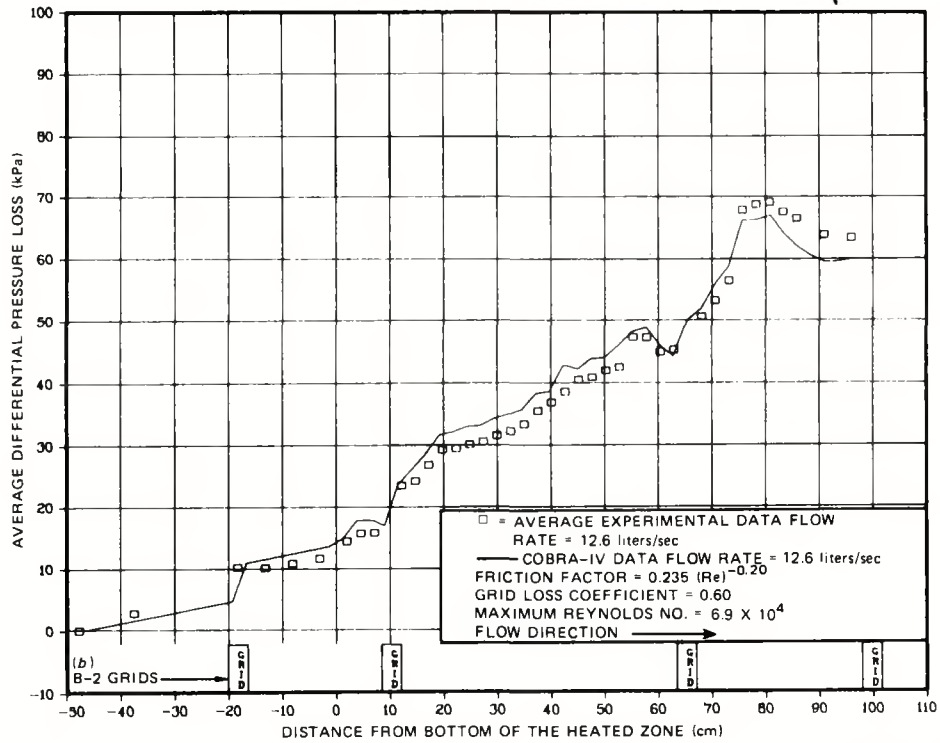
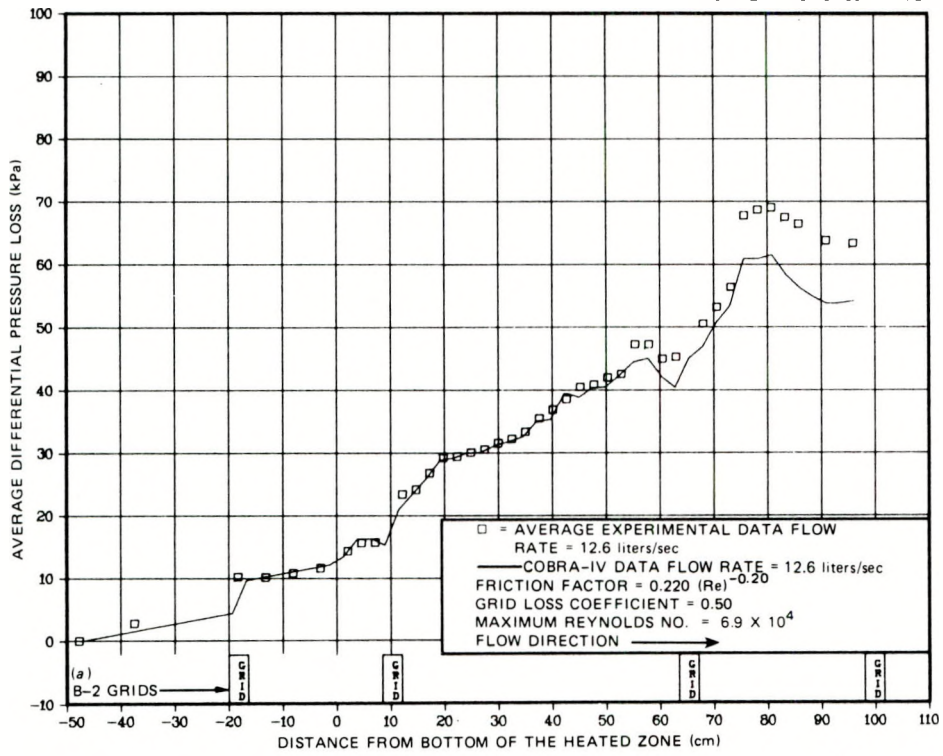


Fig. 5.32. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.6 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

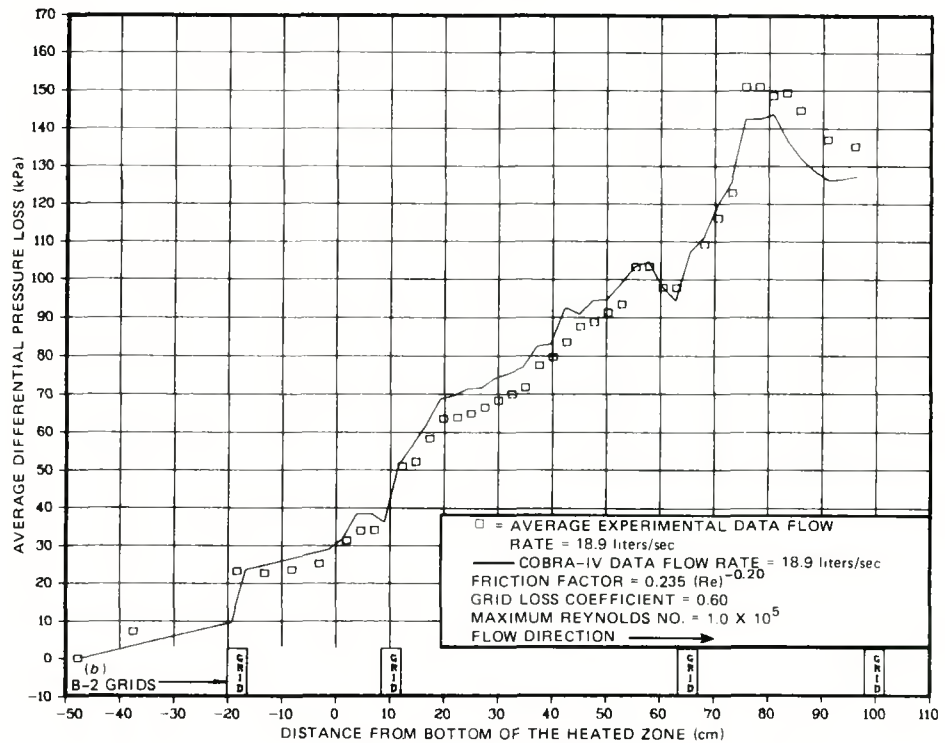
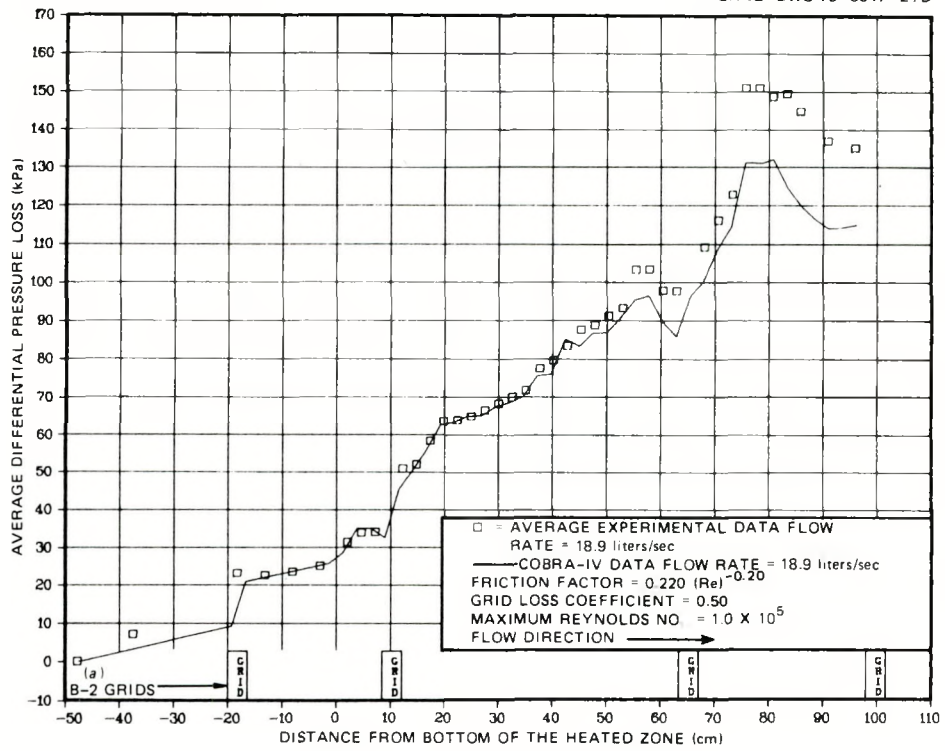


Fig. 5.33. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.9 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

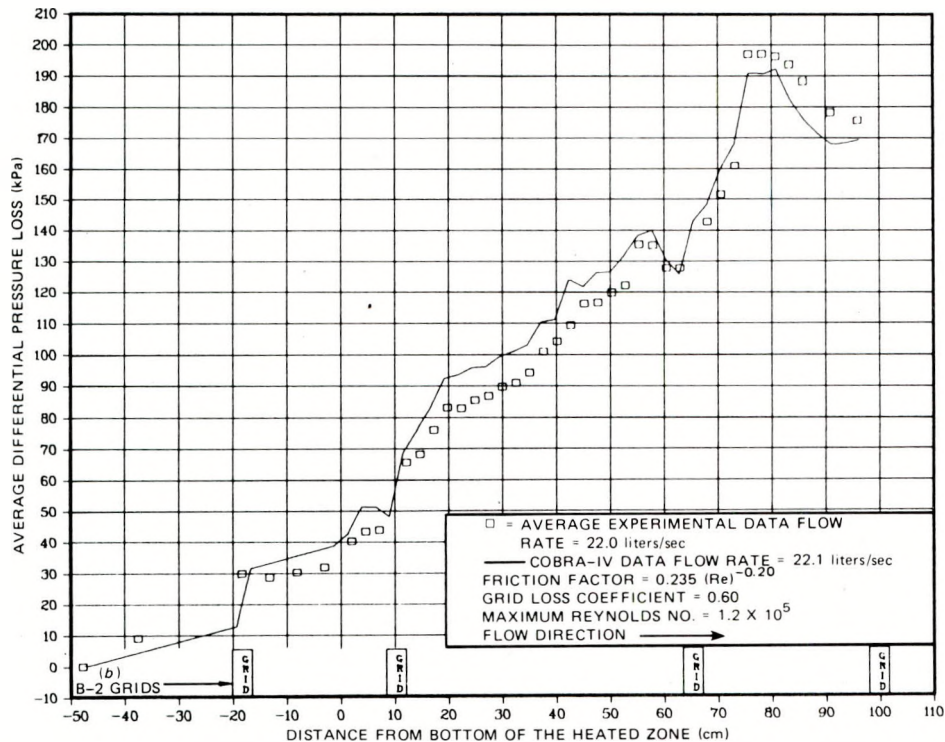
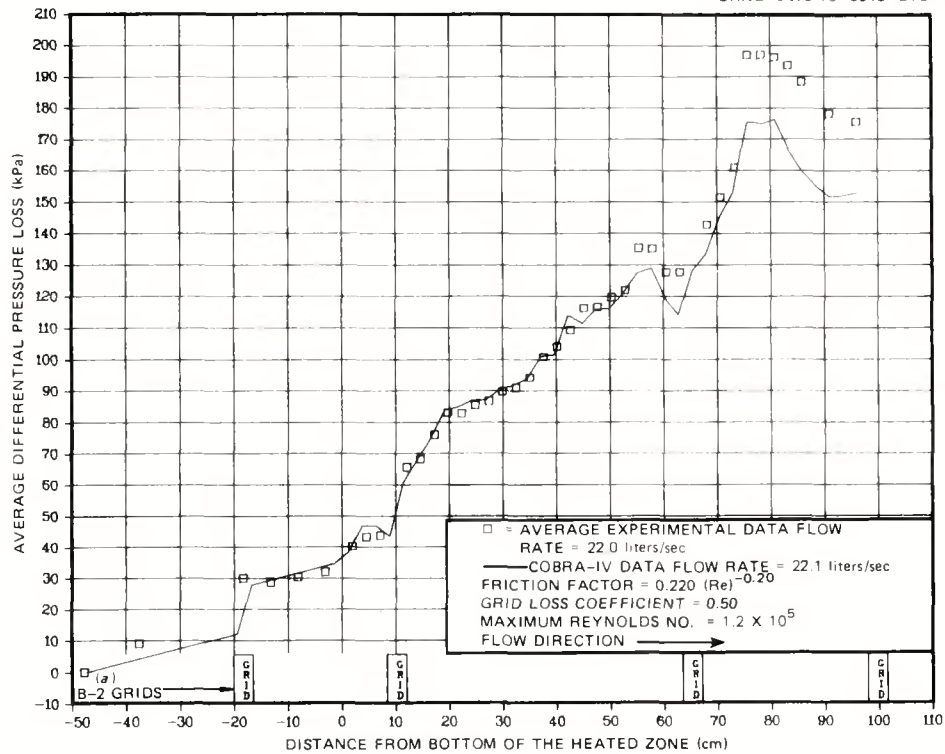


Fig. 5.34. Comparison of B-2/shroud 1 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 22.0 liters/sec; maximum restriction definition. (a) Lower-limit; (b) upper-limit correlation values.

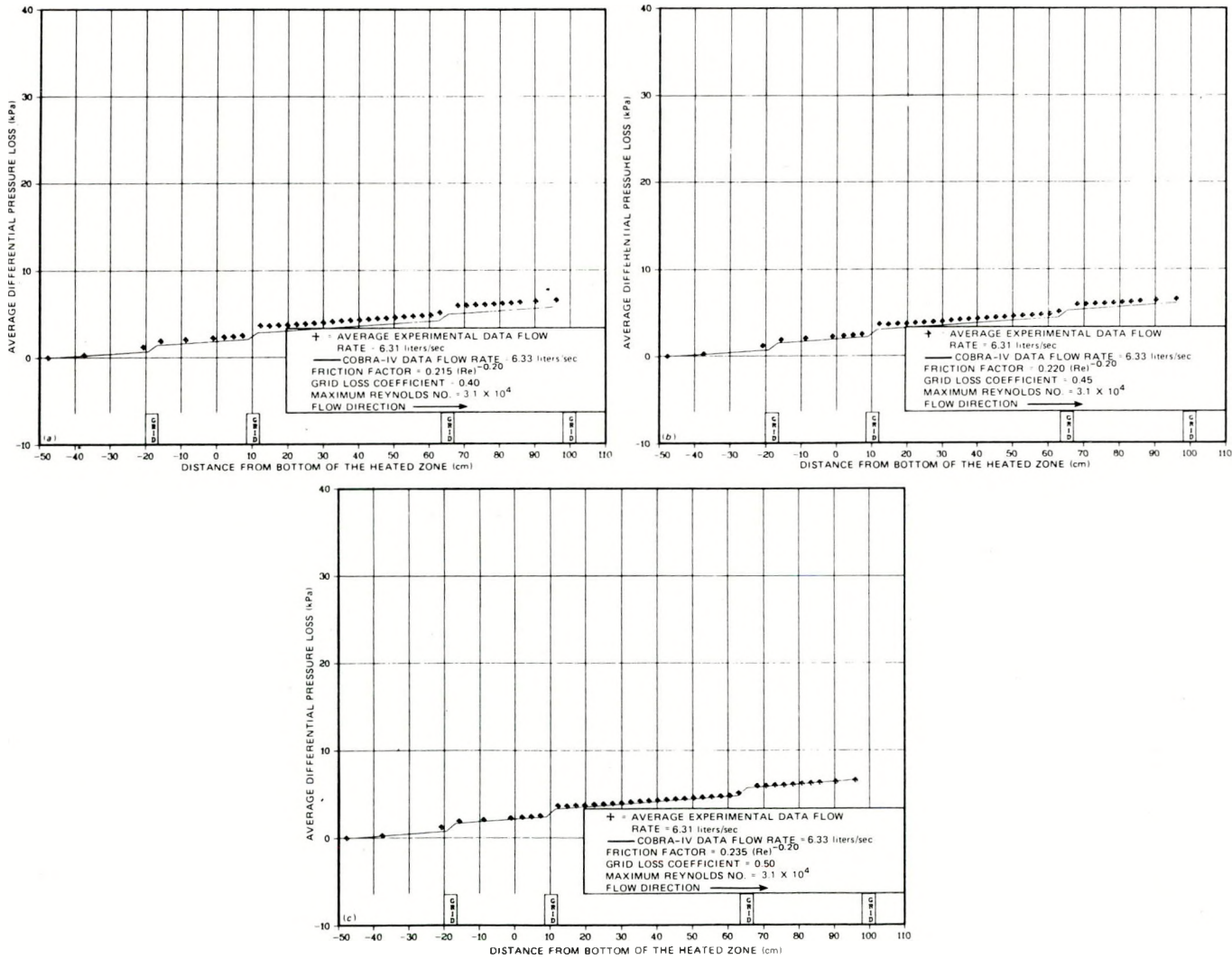


Fig. 5.35. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 6.31 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

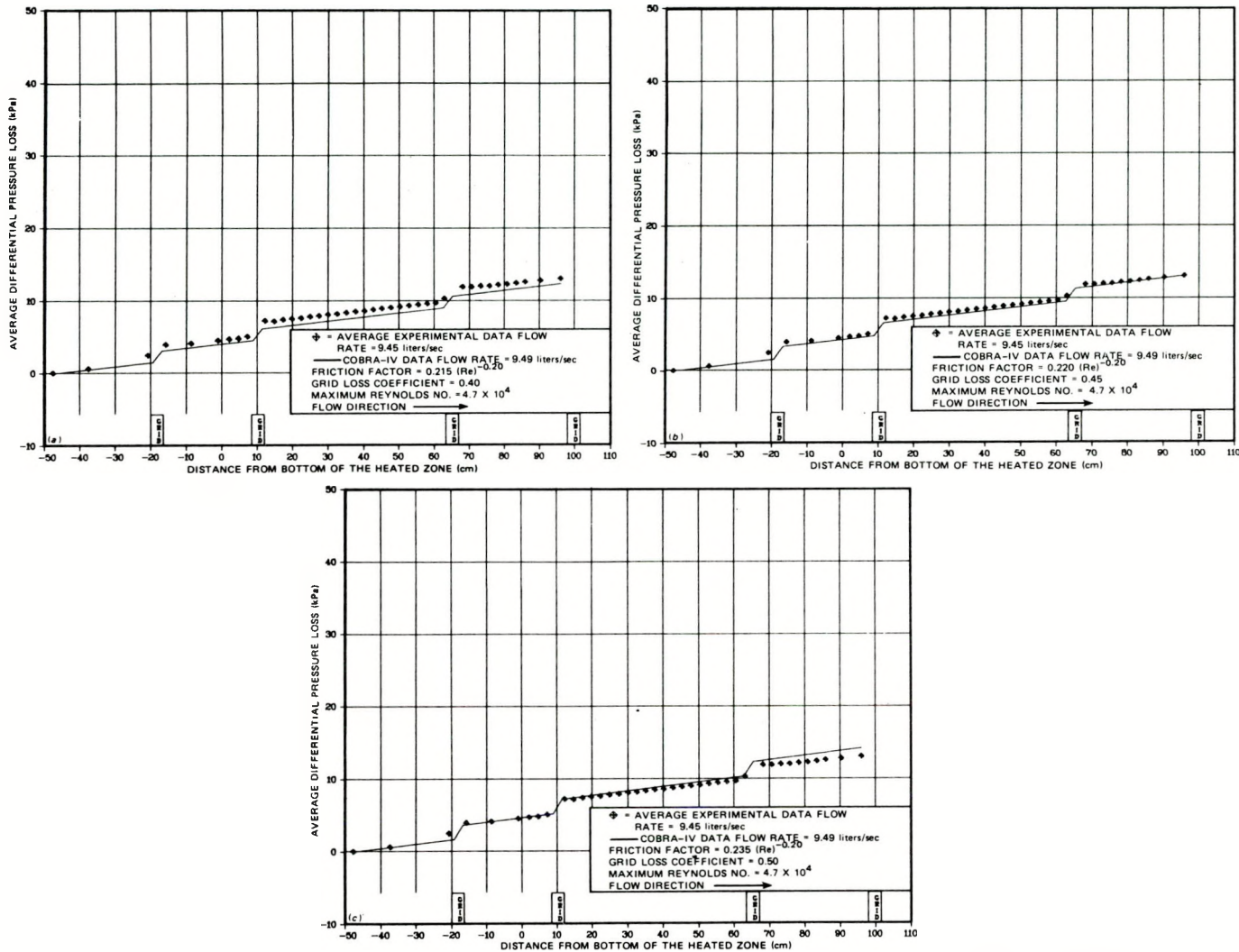


Fig. 5.36. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 9.45 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

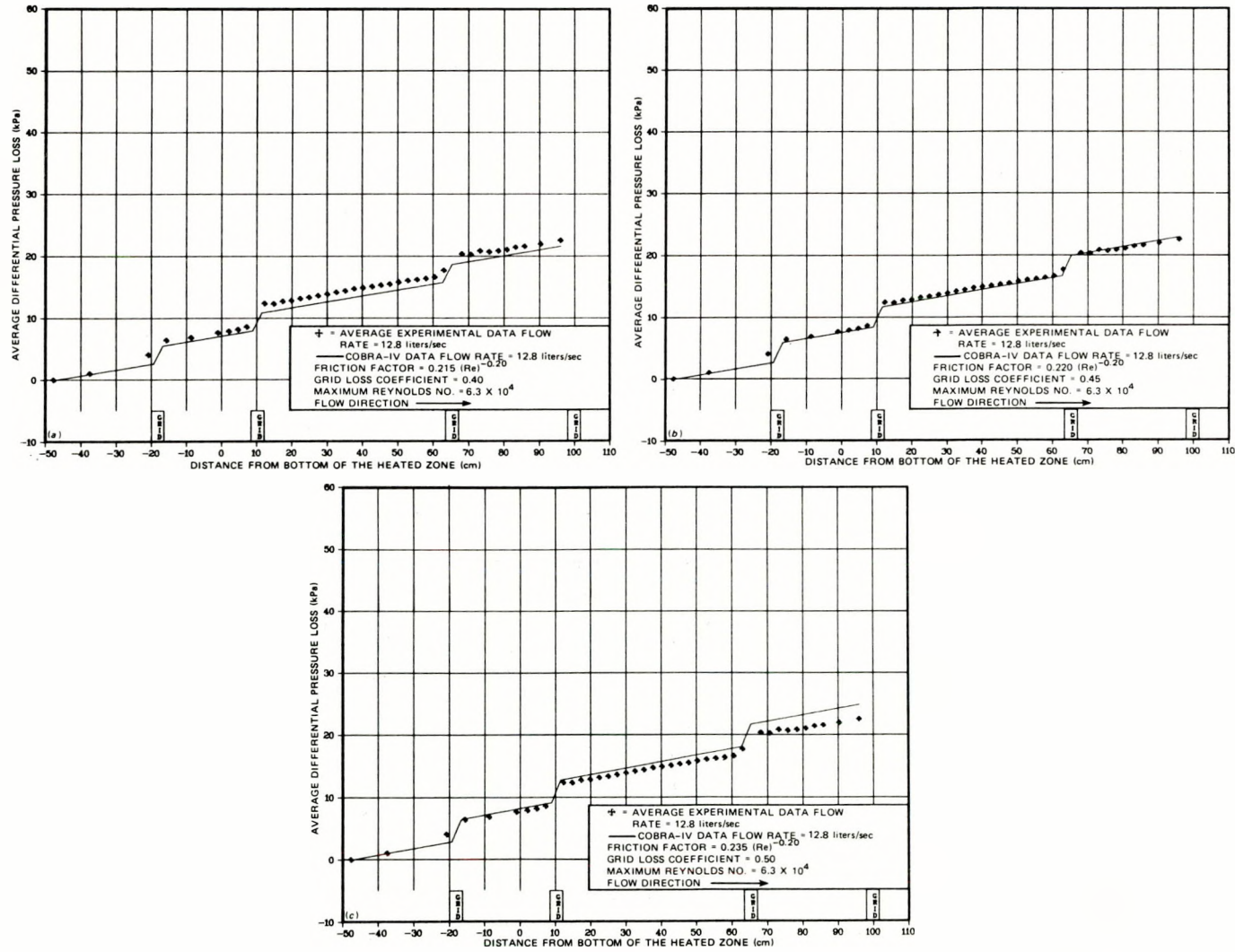


Fig. 5.37. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 12.8 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

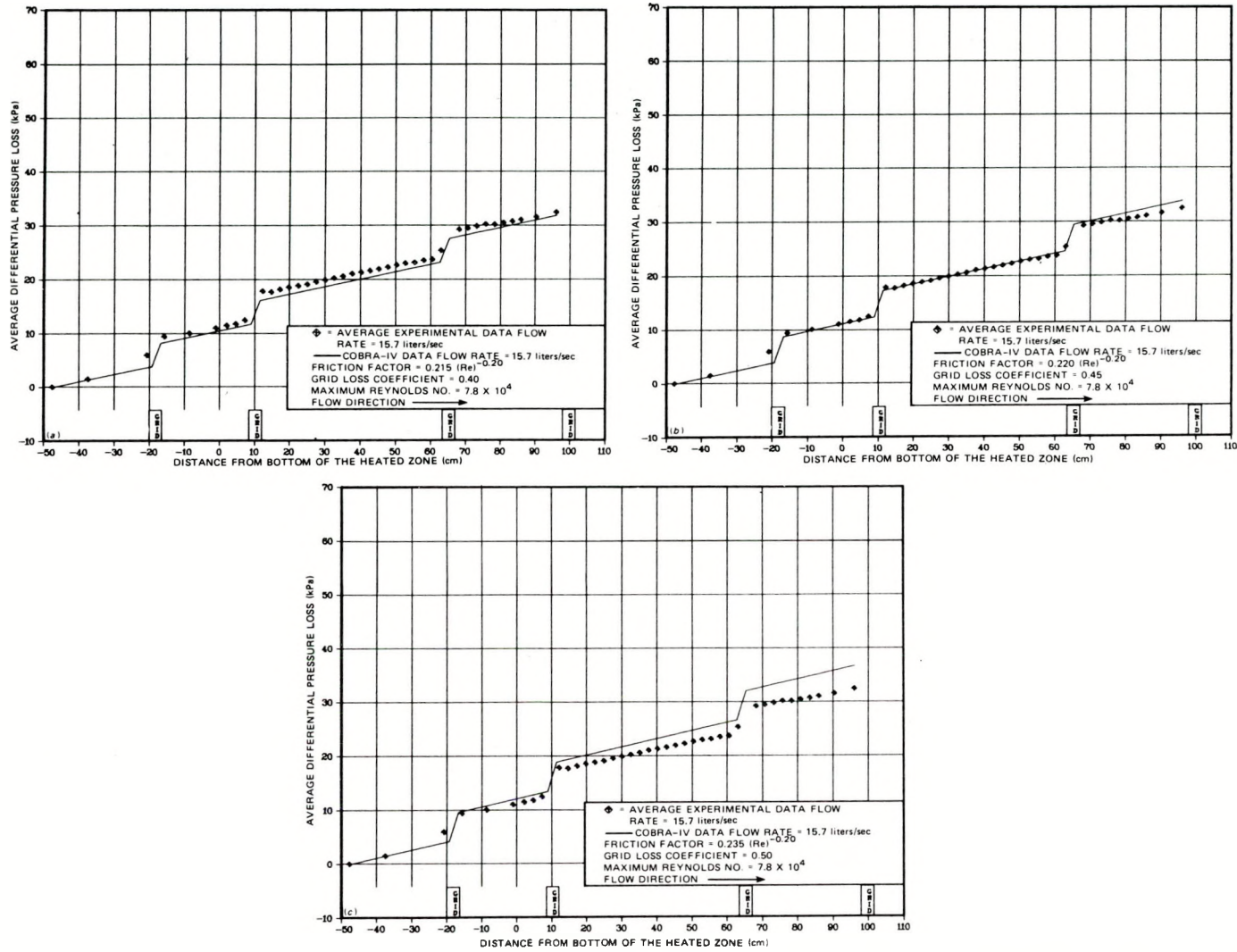


Fig. 5.38. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 15.7 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

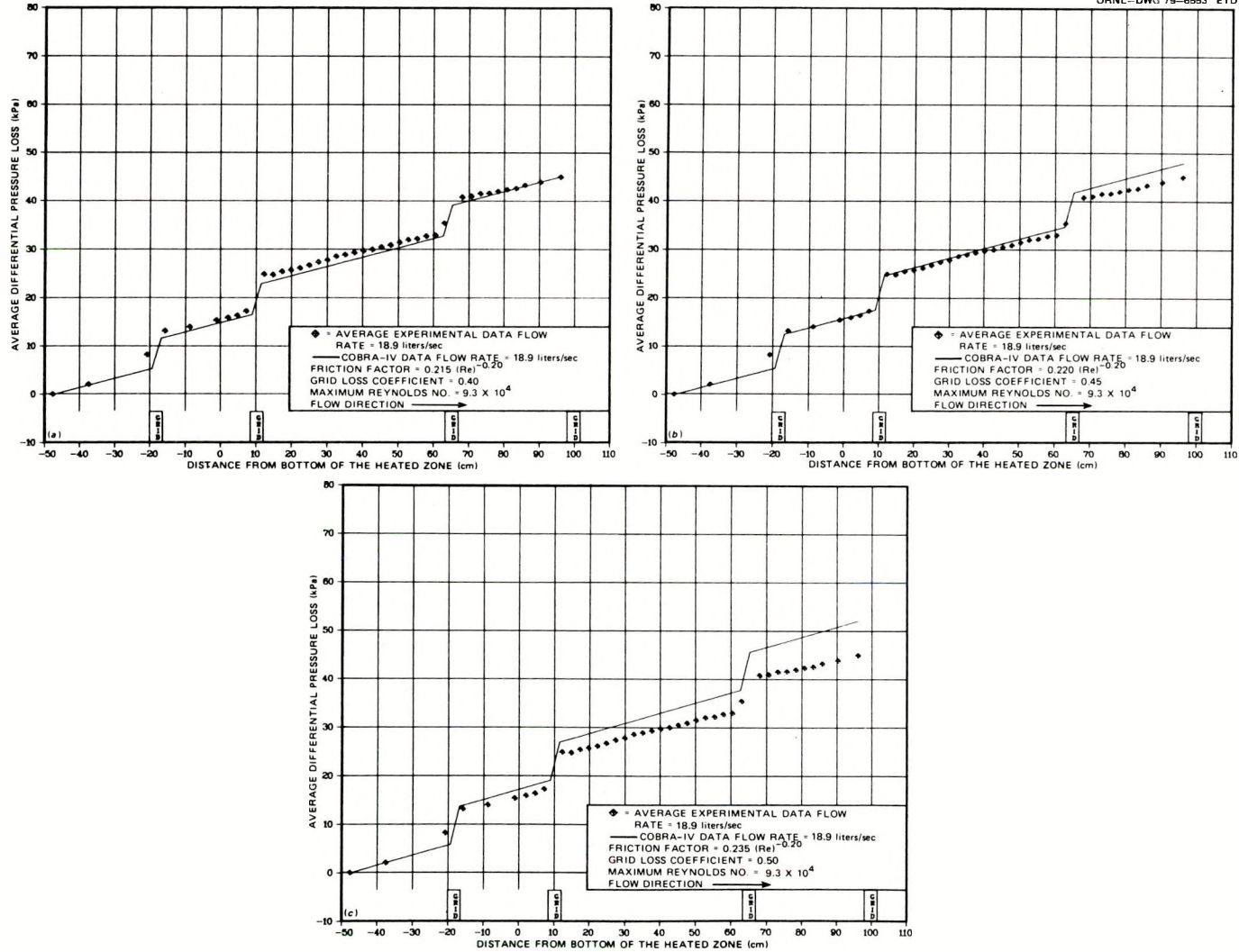


Fig. 5.39. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 18.9 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

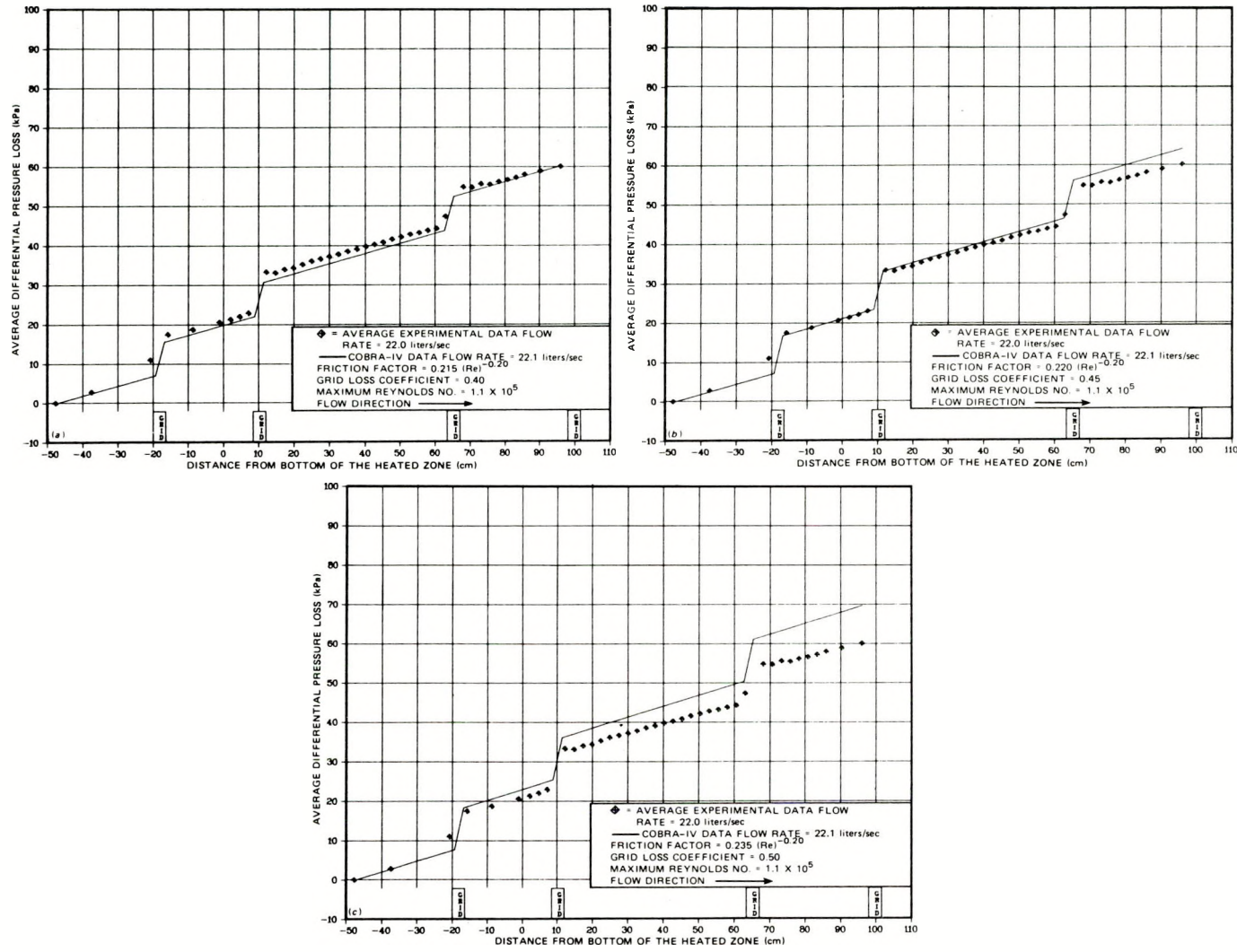


Fig. 5.40. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 22.0 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

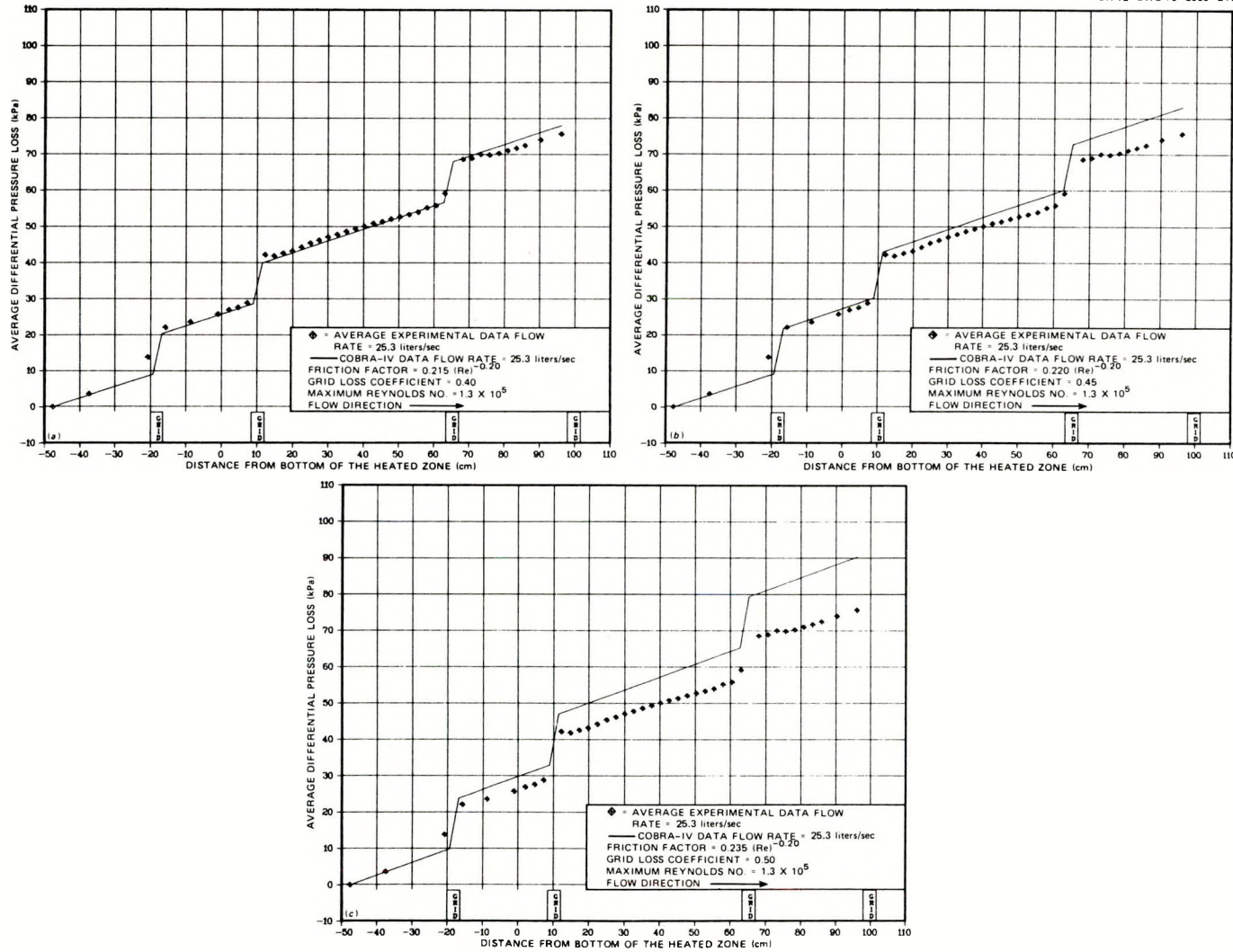


Fig. 5.41. Comparison of reference bundle/shroud 2 experimental and COBRA-IV axial pressure loss profiles; experimental flow rate = 25.3 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

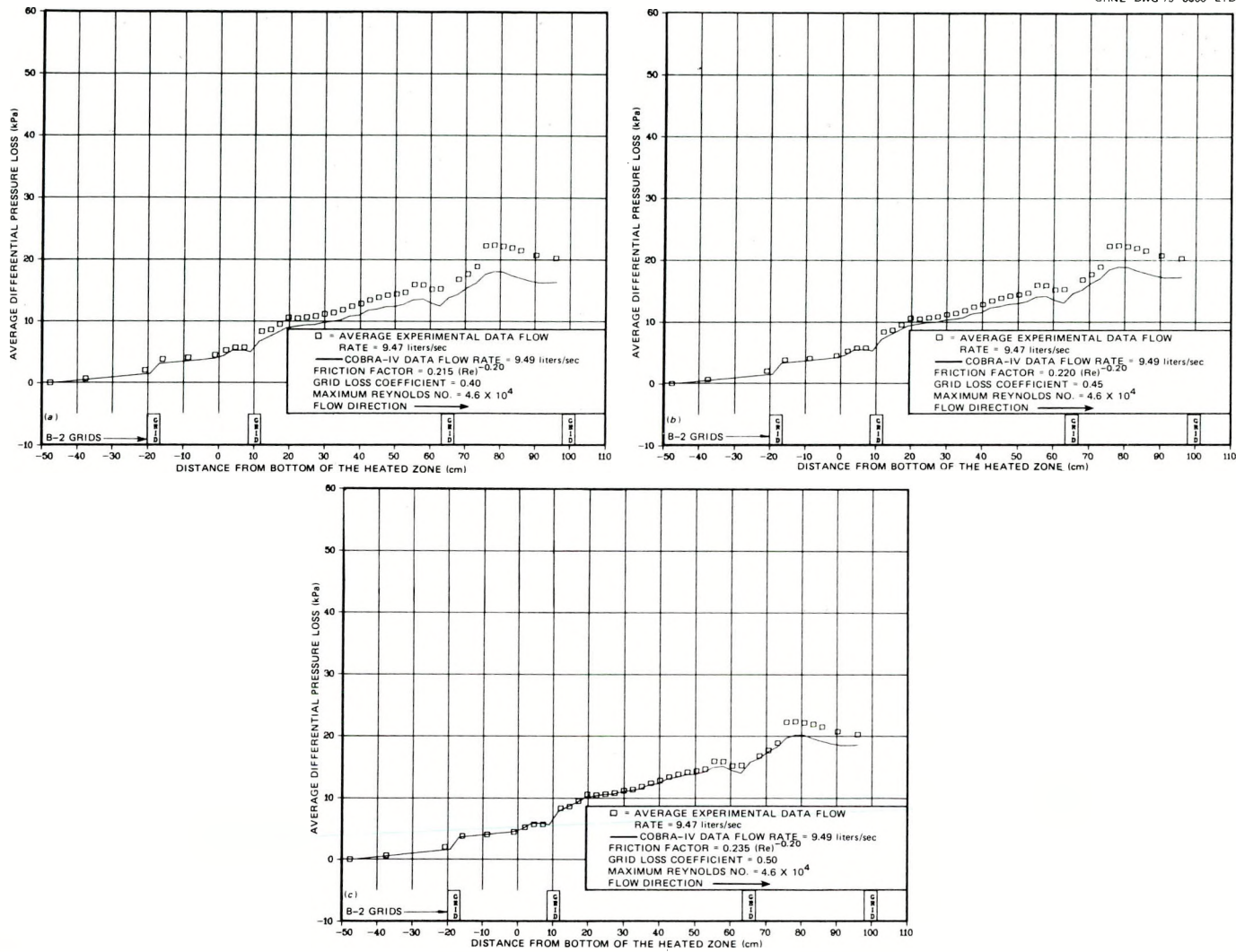


Fig. 5.42. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the minimum restriction definition; experimental flow rate = 9.47 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

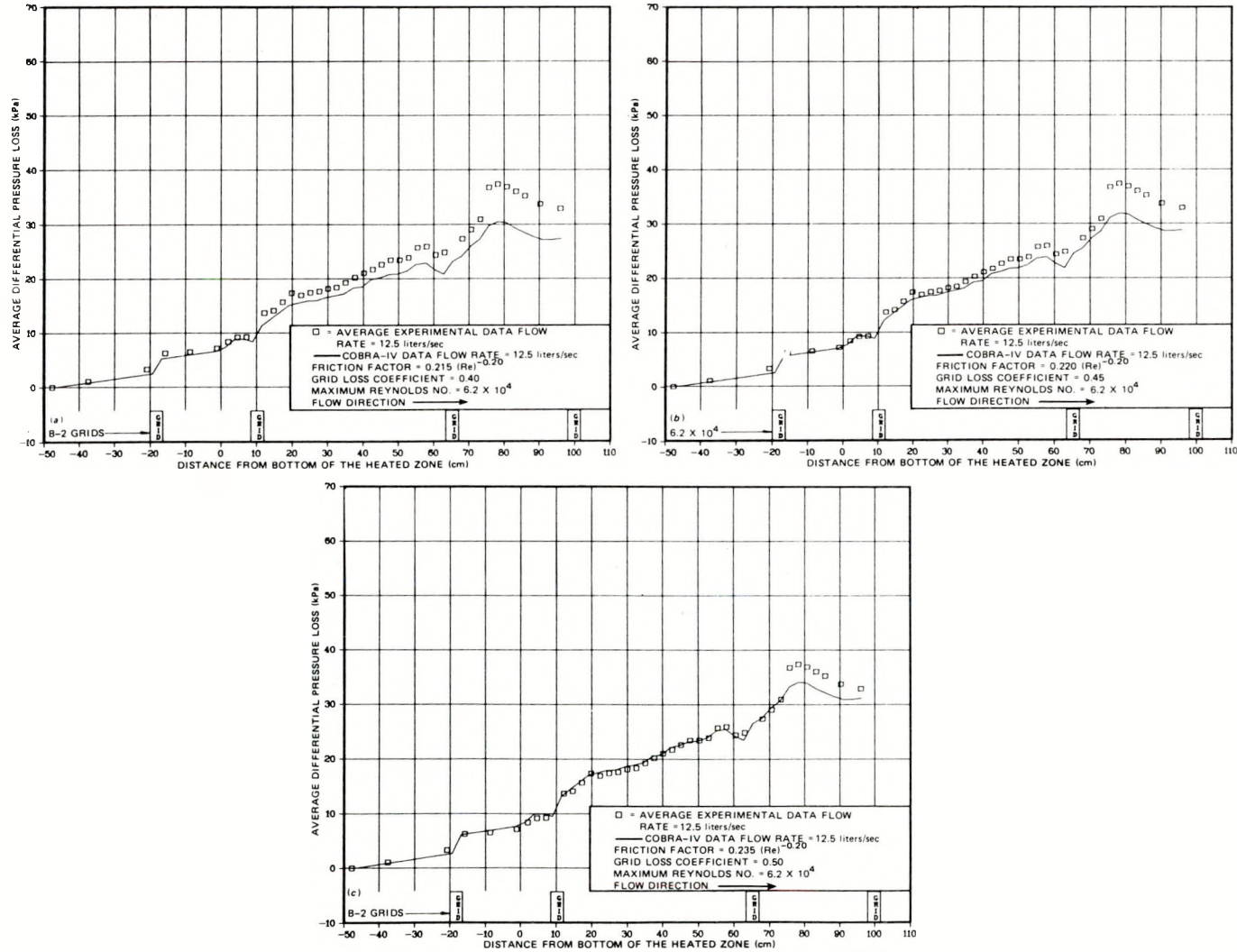


Fig. 5.43. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the minimum restriction definition; experimental flow rate = 12.5 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

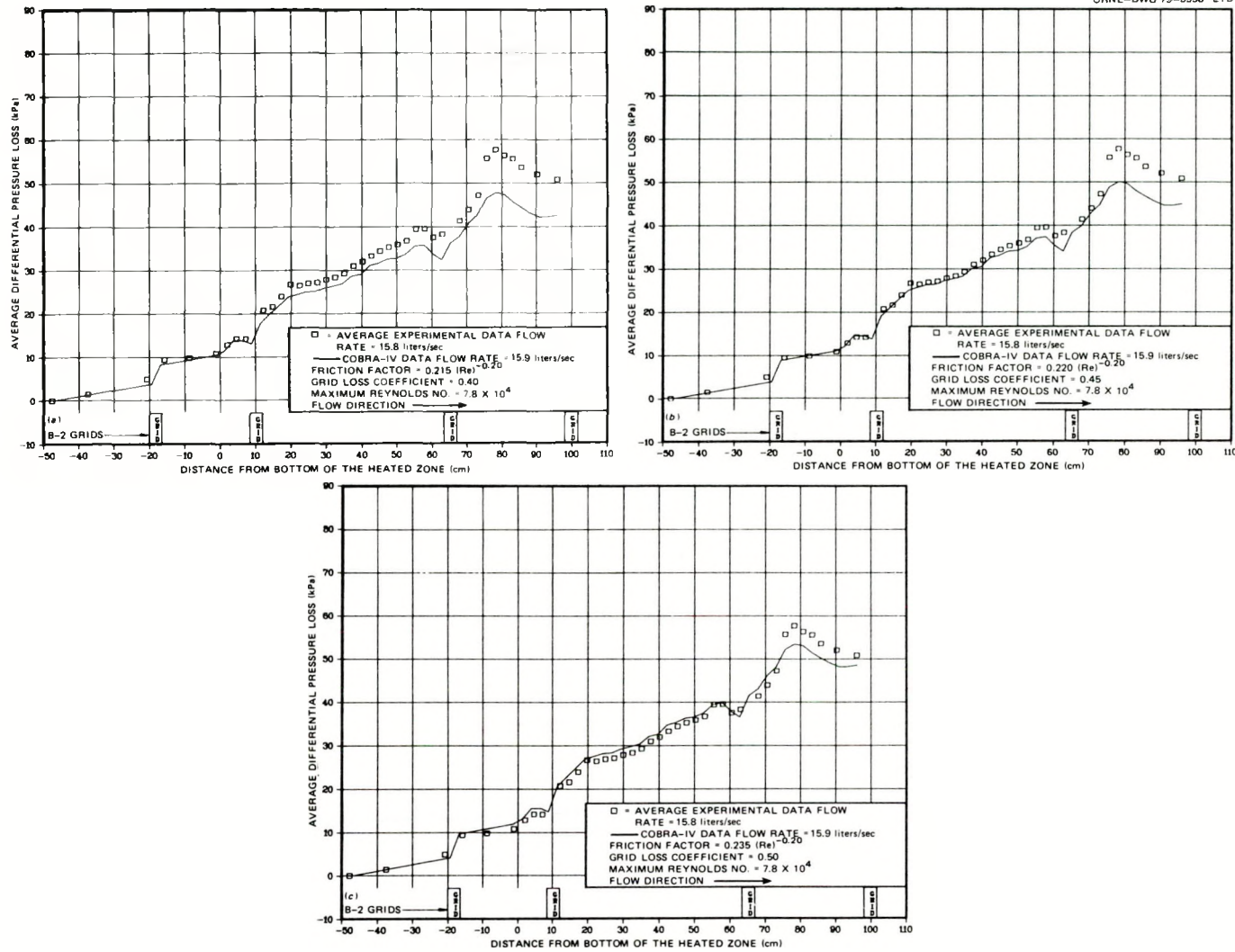


Fig. 5.44. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the minimum restriction definition; experimental flow rate = 15.8 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

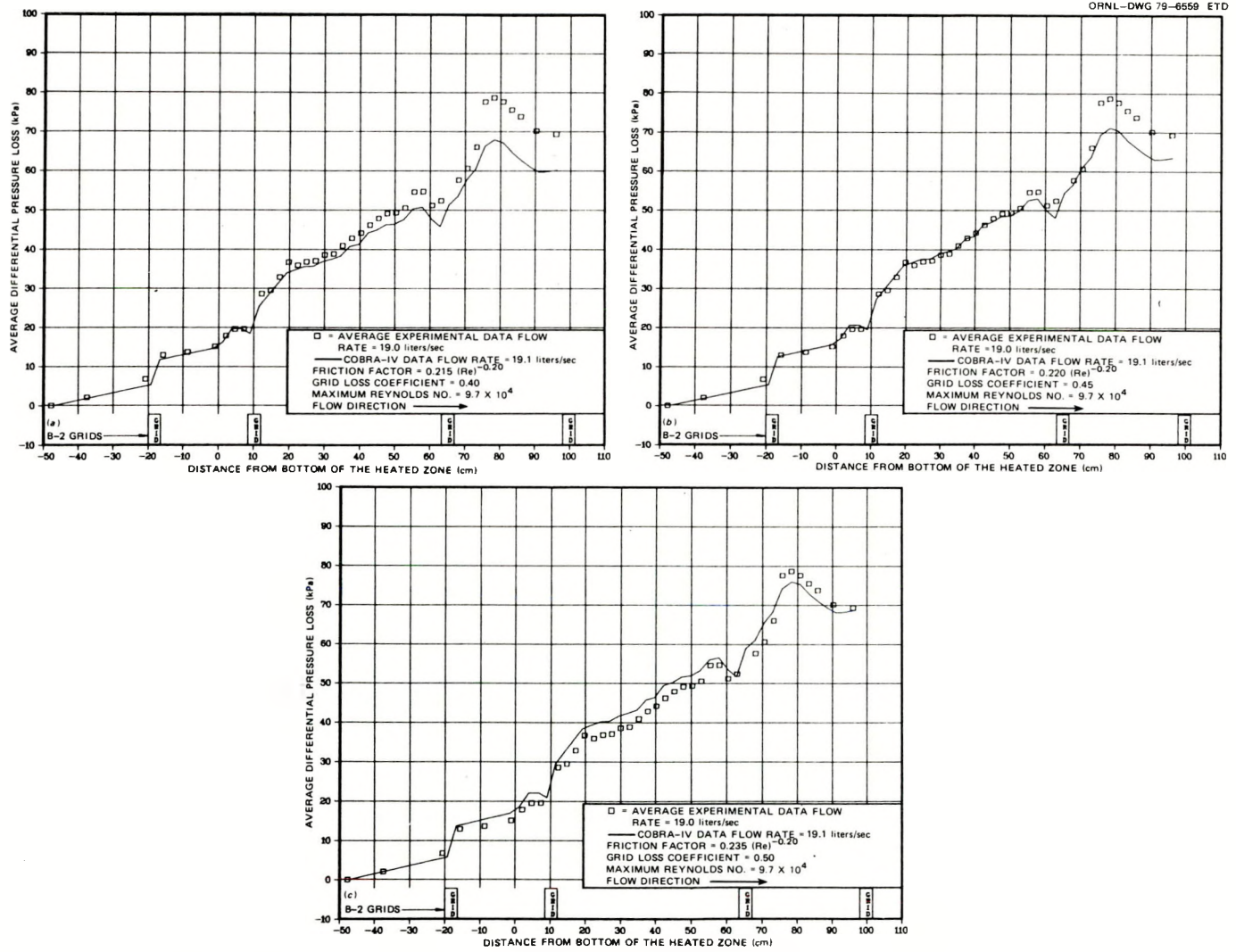


Fig. 5.45. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the minimum restriction definition; experimental flow rate = 19.0 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

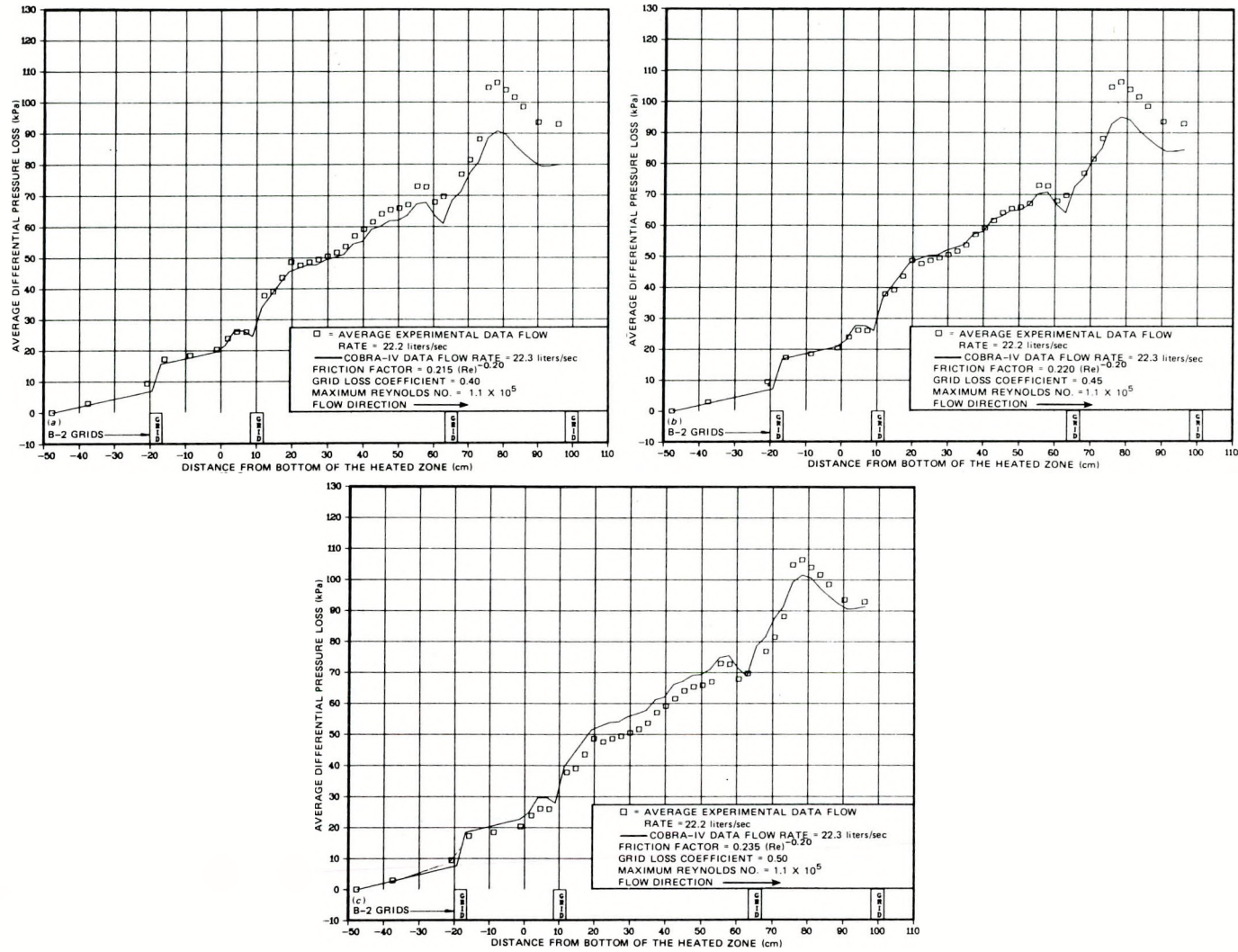


Fig. 5.46. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the minimum restriction definition; experimental flow rate = 22.2 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

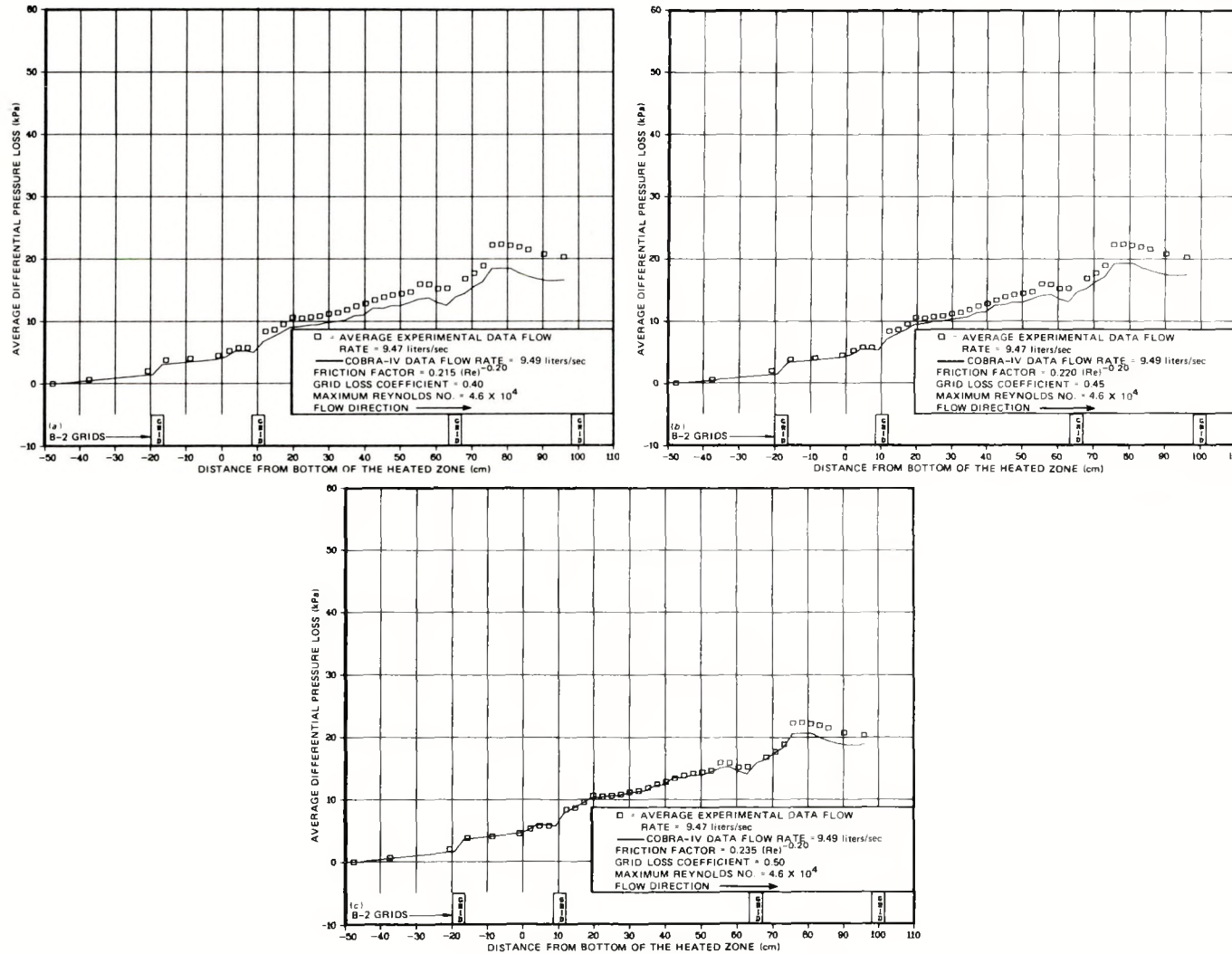


Fig. 5.47. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the maximum restriction definition; experimental flow rate = 9.47 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

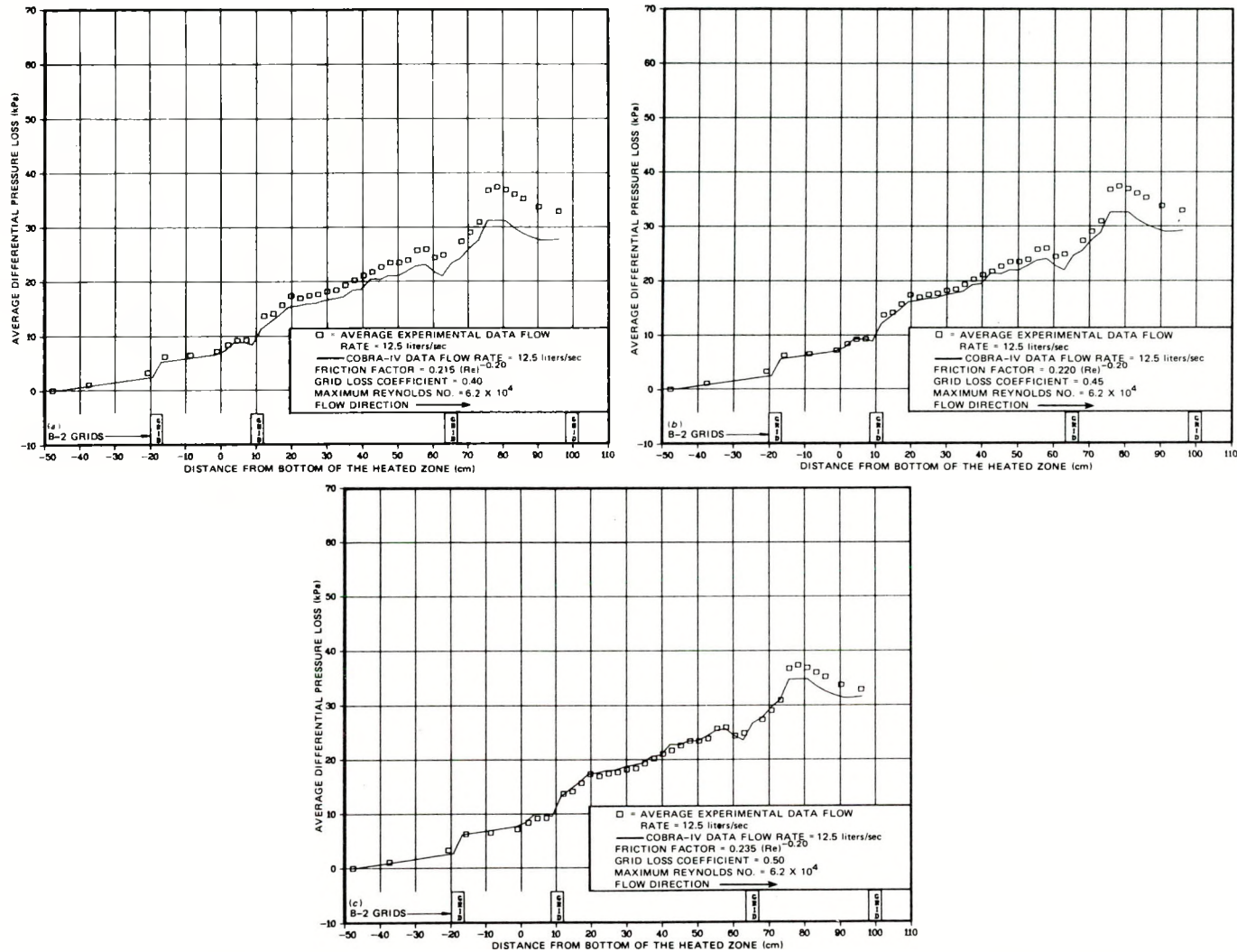


Fig. 5.48. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the maximum restriction definition; experimental flow rate = 12.5 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

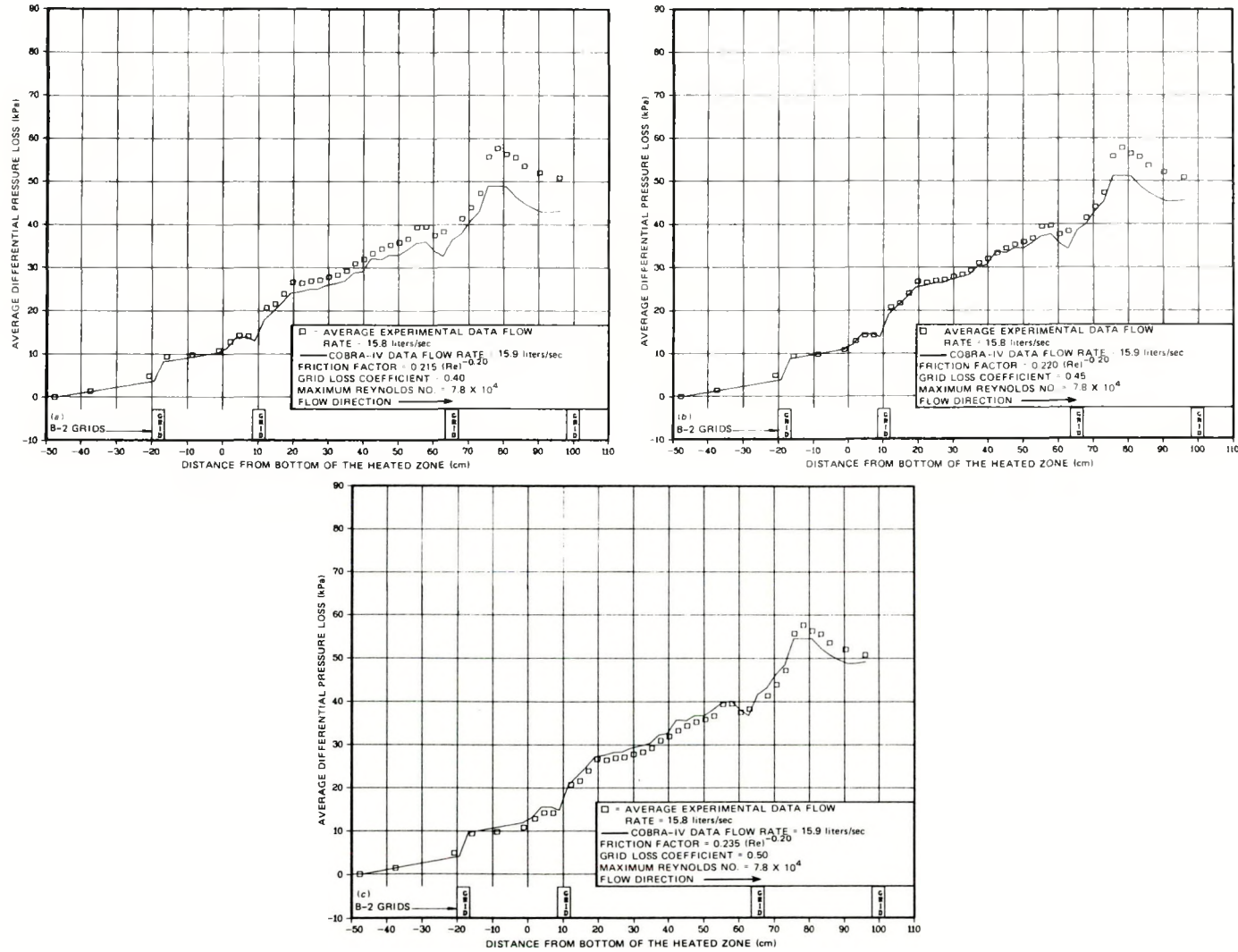


Fig. 5.49. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the maximum restriction definition; experimental flow rate = 15.8 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

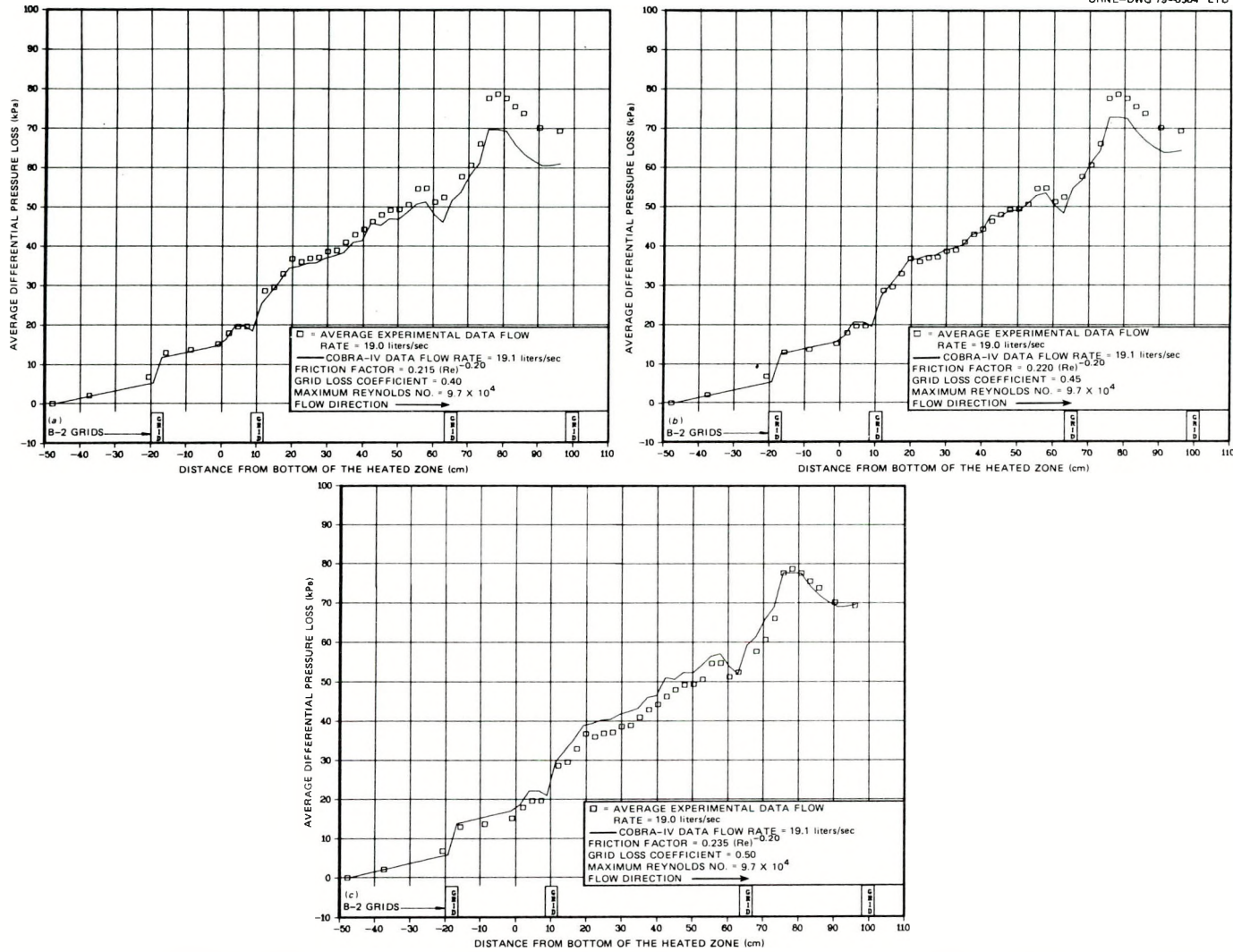


Fig. 5.50. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the maximum restriction definition; experimental flow rate = 19.0 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.

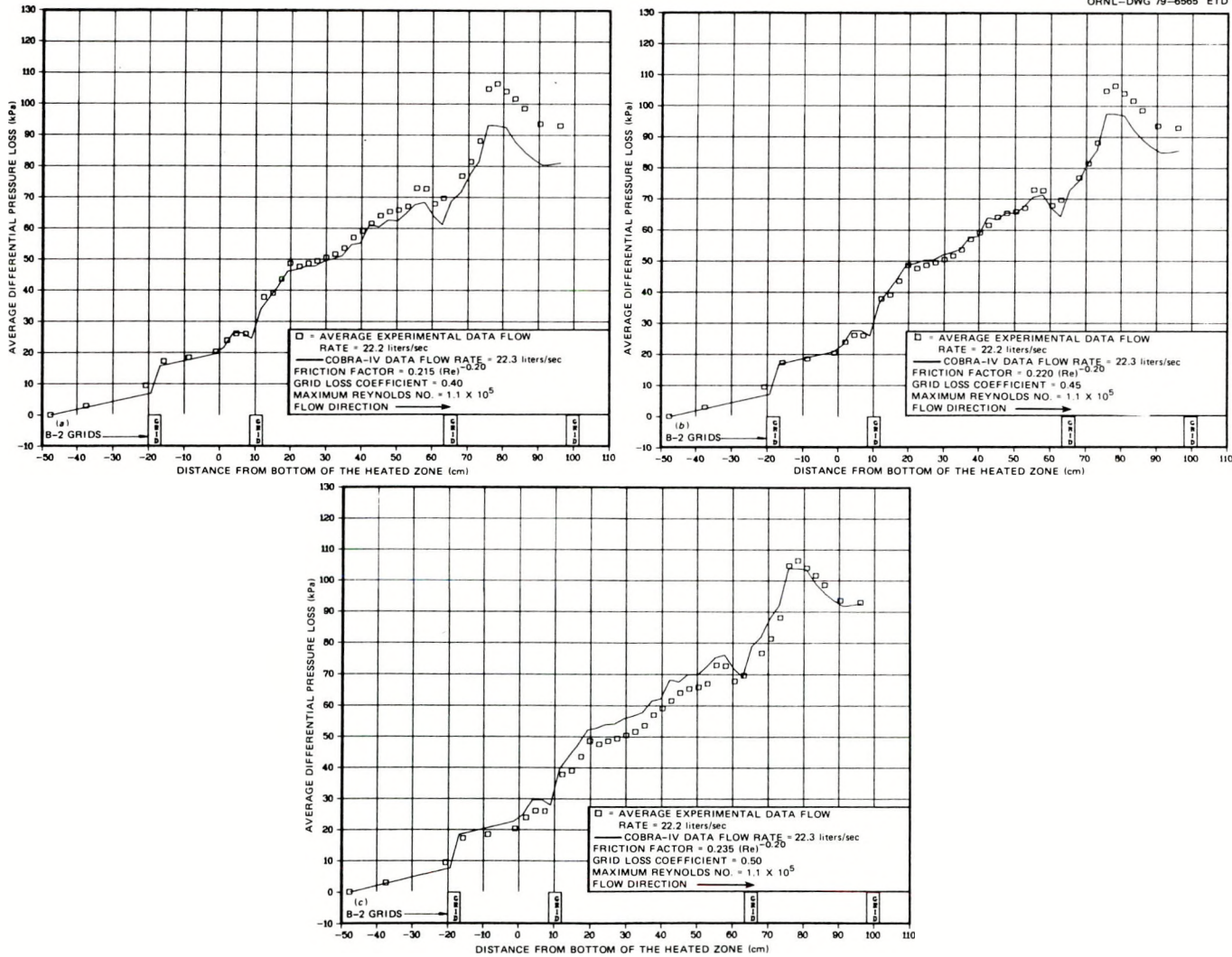
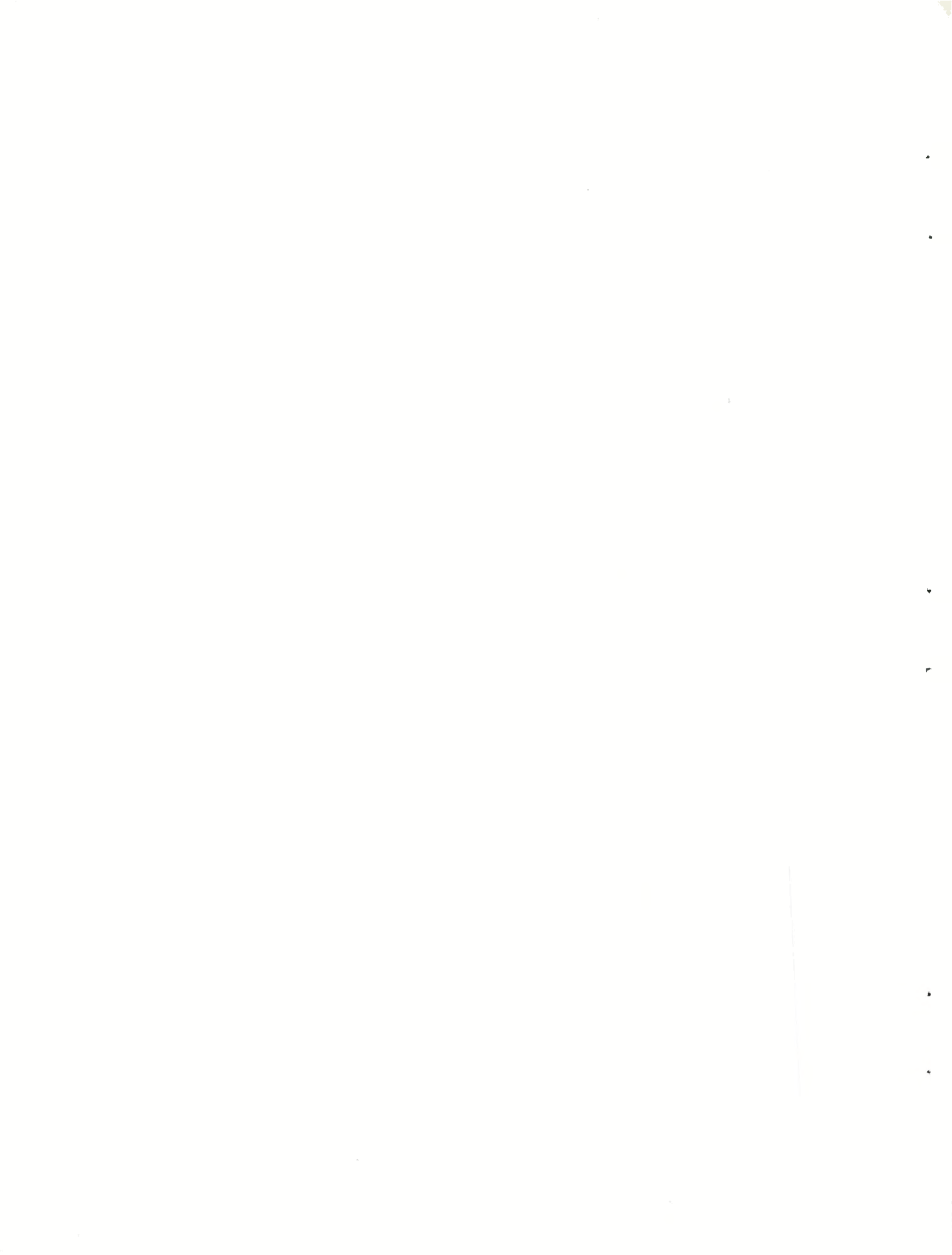


Fig. 5.51. Comparison of B-2/shroud 2 experimental and COBRA-IV axial pressure loss profiles based on the maximum restriction definition; experimental flow rate = 22.2 liters/sec. (a) Lower, (b) middle, and (c) upper correlation values.



6. CONCLUSIONS

One of the MRBT program objectives is to examine the behavior of flow resistance as a function of channel blockage. This was accomplished for B-1 and B-2 by measuring the pressure loss characteristics of those bundles as a function of flow rate (Reynolds modulus) in both the pretest and posttest conditions. These measurements were conducted in an existing flow loop, using water as the test fluid. Since these measurements did not include detailed velocity surveys (i.e., they did not provide cross-flow information about deformed regions), the resulting data do not provide more than a first-order approximation of increased flow resistance as a function of bundle-averaged flow blockage. These experimental and computation analyses in no way purport to evaluate the thermal performance of bundles with subchannel blockages. The experimental data presented are subject to significant uncertainties. Hence, these flow tests are more the proof-of-principle type of experiment rather than being of benchmark quality.

In regard to comparisons of these experimentally derived data with COBRA-IV calculated data, several observations have been made. First, a bundle-averaged approach in modeling pressure losses has been used with considerable success in determining the characteristic shapes of burst bundle pressure loss profiles. Secondly, the absolute magnitudes of these calculated pressure loss profiles were found to be very dependent upon the friction factor correlations and spacer grid pressure loss coefficients used in the calculations. The percent of variation in these correlations found necessary to closely model experimental pressure losses with COBRA-IV ($\pm 5\%$ for friction factors, $\pm 15\%$ for grid spacer pressure loss coefficients) is indicative of the precision to which reported flow areas, flow rates, and pressure measurements could be determined under the available experimental conditions.

Concluding, COBRA-IV and the bundle-averaged restriction representation constitute a viable computational technique for predicting pressure losses in this type of hydraulic system. More precise experimental data are needed, however, to determine better and more general representations

of friction factors and grid loss coefficients for the laminar, transition, and turbulent flow regimes. If more precise representations for these correlations were available, COBRA-IV could be used to predict pressure loss profiles for similar hydraulic systems that are more representative of an in-core reactor fuel element.

7. ACKNOWLEDGMENTS

The author is grateful to the many individuals who gave of their time, patience, and advice. In particular recognition is given to P. H. Hayes, E. L. Armstrong, J. L. Crowley, C. D. Griffies, and R. H. Chapman for their guidance in designing the equipment; to A. W. Longest, E. L. Biddle, C. D. Griffies, J. N. Money, C. Cross, and W. H. Glover for assisting with the flow tests; to D. M. Lister and P. A. Jallouk for their assistance in setting up COBRA-IV; and to J. E. Hardy and J. C. Conklin for reviewing this report.

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APPENDICES

Appendix A

B-1/SHROUD 1 BUNDLE-AVERAGED GEOMETRY DATA,
MINIMUM RESTRICTION DEFINITION

The data in Appendix A represent the input from card groups 5 and 6 as required by ref. 2 for the B-1/shroud 1 COBRA-IV model based on the minimum restriction definition.

Table A.1. Shroud flow area reductions for the B-1/shroud 1 COBRA-IV model based on the minimum restriction definition

AXL ^a	FRACTIONAL SHROUD FLOW AREA
-47.473	1.000
0.000	0.983
1.800	0.915
3.300	0.853
5.200	0.851
8.900	0.947
11.800	0.940
14.100	0.797
15.400	0.766
17.300	0.763
18.900	0.729
20.100	0.719
20.600	0.728
22.300	0.727
23.900	0.711
25.500	0.715
26.500	0.709
28.100	0.716
29.700	0.748
30.700	0.764
33.200	0.787
34.500	0.786
36.600	0.790
38.100	0.779
39.700	0.734
40.800	0.723
42.900	0.736
44.700	0.701
46.700	0.686
47.500	0.688
48.600	0.715
50.400	0.730
52.400	0.756
54.000	0.782
55.600	0.806
57.800	0.823
60.100	0.832
61.700	0.869
64.100	0.954
66.900	0.952
68.800	0.860
70.300	0.822
72.700	0.779
74.200	0.736
76.500	0.716
77.300	0.747
80.200	0.776
81.600	0.808
83.500	0.815
85.100	0.847
86.500	0.874
87.900	0.920
90.000	0.973
92.500	0.998
96.190	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table A.2. AFACT values for the B-1/shroud 1 COBRA-IV model based on the minimum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.473		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		0.993	0.989	0.989	0.989	0.993	0.984	0.973	0.973	0.973	0.984
1.800		0.964	0.944	0.944	0.944	0.964	0.923	0.867	0.867	0.867	0.923
3.300		0.937	0.903	0.903	0.903	0.937	0.867	0.770	0.770	0.770	0.867
5.200		0.936	0.902	0.902	0.902	0.936	0.865	0.767	0.767	0.767	0.865
8.900		0.977	0.965	0.965	0.965	0.977	0.952	0.917	0.917	0.917	0.952
11.800		0.974	0.960	0.960	0.960	0.974	0.946	0.906	0.906	0.906	0.946
14.100		0.913	0.866	0.866	0.866	0.913	0.817	0.683	0.683	0.683	0.817
15.400		0.900	0.846	0.846	0.846	0.900	0.789	0.635	0.635	0.635	0.789
17.300		0.899	0.844	0.844	0.844	0.899	0.787	0.631	0.631	0.631	0.787
18.800		0.884	0.822	0.822	0.822	0.884	0.756	0.577	0.577	0.577	0.756
20.100		0.880	0.815	0.815	0.815	0.880	0.747	0.562	0.562	0.562	0.747
20.600		0.884	0.821	0.821	0.821	0.884	0.755	0.576	0.576	0.576	0.755
22.300		0.883	0.820	0.820	0.820	0.883	0.754	0.574	0.574	0.574	0.754
23.900		0.876	0.810	0.810	0.810	0.876	0.739	0.549	0.549	0.549	0.739
25.500		0.878	0.813	0.813	0.813	0.878	0.743	0.556	0.556	0.556	0.743
26.500		0.876	0.809	0.809	0.809	0.876	0.738	0.546	0.546	0.546	0.738
28.100		0.879	0.813	0.813	0.813	0.879	0.744	0.557	0.557	0.557	0.744
29.700		0.892	0.834	0.834	0.834	0.892	0.773	0.607	0.607	0.607	0.773
30.700		0.899	0.845	0.845	0.845	0.899	0.787	0.632	0.632	0.632	0.787
33.200		0.909	0.860	0.860	0.860	0.909	0.808	0.668	0.668	0.668	0.808
34.500		0.909	0.860	0.860	0.860	0.909	0.808	0.667	0.667	0.667	0.808
36.600		0.910	0.862	0.862	0.862	0.910	0.810	0.672	0.672	0.672	0.810
38.100		0.905	0.854	0.854	0.854	0.905	0.801	0.655	0.655	0.655	0.801
39.700		0.914	0.825	0.825	0.825	0.886	0.315	0.585	0.585	0.585	0.760
40.800		0.910	0.818	0.818	0.818	0.882	0.308	0.568	0.568	0.568	0.750
42.900		0.887	0.827	0.827	0.827	0.887	0.762	0.589	0.589	0.589	0.762
44.700		0.919	0.803	0.803	0.803	0.872	0.840	0.534	0.534	0.534	0.731
46.700		0.915	0.794	0.794	0.794	0.866	0.832	0.511	0.511	0.511	0.717
47.500		0.915	0.795	0.795	0.795	0.867	0.833	0.513	0.513	0.513	0.719
48.600		0.878	0.812	0.812	0.812	0.878	0.743	0.555	0.555	0.555	0.743
50.400		0.885	0.822	0.822	0.822	0.885	0.757	0.579	0.579	0.579	0.757
52.400		0.896	0.840	0.840	0.840	0.896	0.780	0.620	0.620	0.620	0.780
54.000		0.907	0.857	0.857	0.857	0.907	0.804	0.660	0.660	0.660	0.804
55.600		0.917	0.873	0.873	0.873	0.917	0.825	0.698	0.698	0.698	0.825
57.800		0.924	0.884	0.884	0.884	0.924	0.841	0.724	0.724	0.724	0.841
60.100		0.928	0.889	0.889	0.889	0.928	0.849	0.738	0.738	0.738	0.849
61.700		0.944	0.914	0.914	0.914	0.944	0.882	0.795	0.795	0.795	0.882
64.100		0.980	0.970	0.970	0.970	0.980	0.958	0.928	0.928	0.928	0.958
66.900		0.979	0.968	0.968	0.968	0.979	0.957	0.925	0.925	0.925	0.957
68.800		0.940	0.908	0.908	0.908	0.940	0.874	0.782	0.782	0.782	0.874
70.300		0.924	0.883	0.883	0.883	0.924	0.840	0.723	0.723	0.723	0.840
72.700		0.906	0.855	0.855	0.855	0.906	0.801	0.656	0.656	0.656	0.801
74.200		0.887	0.826	0.826	0.826	0.887	0.762	0.588	0.588	0.588	0.762
76.500		0.879	0.813	0.813	0.813	0.879	0.744	0.557	0.557	0.557	0.744
77.300		0.892	0.834	0.834	0.834	0.892	0.772	0.606	0.606	0.606	0.772
80.200		0.904	0.852	0.852	0.852	0.904	0.798	0.650	0.650	0.650	0.798
81.600		0.918	0.874	0.874	0.874	0.918	0.827	0.701	0.701	0.701	0.827
83.500		0.921	0.879	0.879	0.879	0.921	0.834	0.712	0.712	0.712	0.834
85.100		0.935	0.900	0.900	0.900	0.935	0.862	0.762	0.762	0.762	0.862
86.500		0.946	0.917	0.917	0.917	0.946	0.886	0.803	0.803	0.803	0.886
87.900		0.966	0.948	0.948	0.948	0.966	0.928	0.876	0.876	0.876	0.928
90.000		0.988	0.982	0.982	0.982	0.988	0.976	0.958	0.958	0.958	0.976
92.500		0.999	0.999	0.999	0.999	0.999	0.998	0.997	0.997	0.997	0.998
96.190		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table A.2 (continued)

AXL ^a	VALUE OF I IS: 11	12	13	14	15	16	17	18	19	20
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.984	0.973	0.973	0.973	0.984	0.984	0.973	0.973	0.973	0.984
1.800	0.923	0.867	0.867	0.867	0.923	0.923	0.867	0.867	0.867	0.923
3.300	0.867	0.770	0.770	0.770	0.867	0.867	0.770	0.770	0.770	0.867
5.200	0.865	0.767	0.767	0.767	0.865	0.865	0.767	0.767	0.767	0.865
8.900	0.952	0.917	0.917	0.917	0.952	0.952	0.917	0.917	0.917	0.952
11.800	0.946	0.906	0.906	0.906	0.946	0.946	0.906	0.906	0.906	0.946
14.100	0.817	0.683	0.683	0.683	0.817	0.817	0.683	0.683	0.683	0.817
15.400	0.789	0.635	0.635	0.635	0.789	0.789	0.635	0.635	0.635	0.789
17.300	0.787	0.631	0.631	0.631	0.787	0.787	0.631	0.631	0.631	0.787
18.800	0.756	0.577	0.577	0.577	0.756	0.756	0.577	0.577	0.577	0.756
20.100	0.747	0.562	0.562	0.562	0.747	0.747	0.562	0.562	0.562	0.747
20.600	0.755	0.576	0.576	0.576	0.755	0.755	0.576	0.576	0.576	0.755
22.300	0.754	0.574	0.574	0.574	0.754	0.754	0.574	0.574	0.574	0.754
23.900	0.739	0.549	0.549	0.549	0.739	0.739	0.549	0.549	0.549	0.739
25.500	0.743	0.556	0.556	0.556	0.743	0.743	0.556	0.556	0.556	0.743
26.500	0.738	0.546	0.546	0.546	0.738	0.738	0.546	0.546	0.546	0.738
28.100	0.744	0.557	0.557	0.557	0.744	0.744	0.557	0.557	0.557	0.744
29.700	0.773	0.607	0.607	0.607	0.773	0.773	0.607	0.607	0.607	0.773
30.700	0.787	0.632	0.632	0.632	0.787	0.787	0.632	0.632	0.632	0.787
33.200	0.808	0.668	0.668	0.668	0.808	0.808	0.668	0.668	0.668	0.808
34.500	0.808	0.667	0.667	0.667	0.808	0.808	0.667	0.667	0.667	0.808
36.600	0.810	0.672	0.672	0.672	0.810	0.810	0.672	0.672	0.672	0.810
38.100	0.801	0.655	0.655	0.655	0.801	0.801	0.655	0.655	0.655	0.801
39.700	0.760	0.585	0.585	0.585	0.760	0.760	0.585	0.585	0.585	0.760
40.800	0.750	0.568	0.568	0.568	0.750	0.750	0.568	0.568	0.568	0.750
42.900	0.762	0.589	0.589	0.589	0.762	0.762	0.589	0.589	0.589	0.762
44.700	0.791	0.534	0.534	0.534	0.791	0.791	0.534	0.534	0.534	0.791
46.700	0.780	0.511	0.511	0.511	0.777	0.777	0.511	0.511	0.511	0.777
47.500	0.781	0.513	0.513	0.513	0.719	0.719	0.513	0.513	0.513	0.719
48.600	0.743	0.555	0.555	0.555	0.743	0.743	0.555	0.555	0.555	0.743
50.400	0.757	0.579	0.579	0.579	0.757	0.757	0.579	0.579	0.579	0.757
52.400	0.780	0.620	0.620	0.620	0.780	0.780	0.620	0.620	0.620	0.780
54.000	0.804	0.660	0.660	0.660	0.804	0.804	0.660	0.660	0.660	0.804
55.600	0.825	0.698	0.698	0.698	0.825	0.825	0.698	0.698	0.698	0.825
57.800	0.841	0.724	0.724	0.724	0.841	0.841	0.724	0.724	0.724	0.841
60.100	0.849	0.738	0.738	0.738	0.849	0.849	0.738	0.738	0.738	0.849
61.700	0.882	0.795	0.795	0.795	0.882	0.882	0.795	0.795	0.795	0.882
64.100	0.958	0.928	0.928	0.928	0.958	0.958	0.928	0.928	0.928	0.958
66.900	0.957	0.925	0.925	0.925	0.957	0.957	0.925	0.925	0.925	0.957
68.800	0.874	0.782	0.782	0.782	0.874	0.874	0.782	0.782	0.782	0.874
70.300	0.840	0.723	0.723	0.723	0.840	0.840	0.723	0.723	0.723	0.840
72.700	0.801	0.656	0.656	0.656	0.801	0.801	0.656	0.656	0.656	0.801
74.200	0.762	0.588	0.588	0.588	0.854	0.762	0.588	0.588	0.588	0.854
76.500	0.744	0.557	0.557	0.557	0.843	0.744	0.557	0.557	0.557	0.843
77.300	0.772	0.606	0.606	0.606	0.860	0.772	0.606	0.606	0.606	0.860
80.200	0.798	0.650	0.650	0.650	0.798	0.798	0.650	0.650	0.650	0.798
81.600	0.827	0.701	0.701	0.701	0.827	0.827	0.701	0.701	0.701	0.827
83.500	0.834	0.712	0.712	0.712	0.834	0.834	0.712	0.712	0.712	0.834
85.100	0.862	0.762	0.762	0.762	0.862	0.862	0.762	0.762	0.762	0.862
86.500	0.886	0.803	0.803	0.803	0.886	0.886	0.803	0.803	0.803	0.886
87.900	0.928	0.876	0.876	0.876	0.928	0.928	0.876	0.876	0.876	0.928
90.000	0.976	0.958	0.958	0.958	0.976	0.976	0.958	0.958	0.958	0.976
92.500	0.998	0.997	0.997	0.997	0.998	0.998	0.997	0.997	0.997	0.998
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table A.2 (continued)

AXL ^a	VALUE OF I IS: 21	22	23	24	25
-47.473	1.000	1.000	1.000	1.000	1.000
0.000	0.993	0.989	0.989	0.989	0.993
1.800	0.964	0.944	0.944	0.944	0.964
3.300	0.937	0.903	0.903	0.903	0.937
5.200	0.936	0.902	0.902	0.902	0.936
8.900	0.977	0.965	0.965	0.965	0.977
11.800	0.974	0.960	0.960	0.960	0.974
14.100	0.913	0.866	0.866	0.866	0.913
15.400	0.900	0.846	0.846	0.846	0.900
17.300	0.899	0.844	0.844	0.844	0.899
18.800	0.884	0.822	0.822	0.822	0.884
20.100	0.880	0.815	0.815	0.815	0.880
20.600	0.884	0.821	0.821	0.821	0.884
22.300	0.883	0.820	0.820	0.820	0.883
23.900	0.876	0.810	0.810	0.810	0.876
25.500	0.878	0.813	0.813	0.813	0.878
26.500	0.876	0.809	0.809	0.809	0.876
28.100	0.879	0.813	0.813	0.813	0.879
29.700	0.892	0.834	0.834	0.834	0.892
30.700	0.899	0.845	0.845	0.845	0.899
33.200	0.909	0.860	0.860	0.860	0.909
34.500	0.909	0.860	0.860	0.860	0.909
36.600	0.910	0.862	0.862	0.862	0.910
38.100	0.905	0.854	0.854	0.854	0.905
39.700	0.886	0.825	0.825	0.825	0.886
40.800	0.882	0.818	0.818	0.818	0.882
42.900	0.887	0.827	0.827	0.827	0.887
44.700	0.872	0.803	0.803	0.803	0.872
46.700	0.866	0.794	0.794	0.794	0.866
47.500	0.867	0.795	0.795	0.795	0.867
48.600	0.878	0.812	0.812	0.812	0.878
50.400	0.885	0.822	0.822	0.822	0.885
52.400	0.896	0.840	0.840	0.840	0.896
54.000	0.907	0.857	0.857	0.857	0.907
55.600	0.917	0.873	0.873	0.873	0.917
57.800	0.924	0.884	0.884	0.884	0.924
60.100	0.928	0.889	0.889	0.889	0.928
61.700	0.944	0.914	0.914	0.914	0.944
64.100	0.980	0.970	0.970	0.970	0.980
66.900	0.979	0.968	0.968	0.968	0.979
68.800	0.940	0.908	0.908	0.908	0.940
70.300	0.924	0.883	0.883	0.883	0.924
72.700	0.906	0.855	0.855	0.855	0.906
74.200	0.887	0.826	0.826	0.826	0.887
76.500	0.879	0.813	0.813	0.813	0.879
77.300	0.892	0.834	0.834	0.834	0.892
80.200	0.904	0.852	0.852	0.852	0.904
81.600	0.918	0.874	0.874	0.874	0.918
83.500	0.921	0.879	0.879	0.879	0.921
85.100	0.935	0.900	0.900	0.900	0.935
86.500	0.946	0.917	0.917	0.917	0.946
87.900	0.966	0.948	0.948	0.948	0.966
90.000	0.988	0.982	0.982	0.982	0.988
92.500	0.999	0.999	0.999	0.999	0.999
96.190	1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table A.3. GFACT values for the B-1/shroud 1 COBRA-IV model based on the minimum restriction definition

GAPXL ^a	VALUE OF K IS: 1	2	3	4	5	6	7	8	9	10
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.988	0.981	0.988	0.949	0.988	0.949	0.988	0.949	0.981	0.981
1.800	0.941	0.968	0.941	0.756	0.941	0.756	0.941	0.756	0.908	0.908
3.300	0.900	0.845	0.900	0.589	0.900	0.589	0.900	0.589	0.845	0.845
5.200	0.899	0.843	0.899	0.584	0.899	0.584	0.899	0.584	0.843	0.843
8.900	0.962	0.942	0.962	0.846	0.962	0.846	0.962	0.846	0.942	0.942
11.800	0.958	0.934	0.958	0.826	0.958	0.826	0.958	0.826	0.934	0.934
14.100	0.865	0.791	0.865	0.446	0.865	0.446	0.865	0.446	0.791	0.791
15.400	0.846	0.762	0.846	0.369	0.846	0.369	0.846	0.369	0.762	0.762
17.300	0.845	0.760	0.845	0.363	0.845	0.363	0.845	0.363	0.760	0.760
18.800	0.824	0.728	0.824	0.278	0.824	0.278	0.824	0.278	0.728	0.728
20.100	0.818	0.719	0.818	0.255	0.818	0.255	0.818	0.255	0.719	0.719
20.600	0.824	0.727	0.824	0.277	0.824	0.277	0.824	0.277	0.727	0.727
22.300	0.823	0.726	0.823	0.274	0.823	0.274	0.823	0.274	0.726	0.726
23.900	0.814	0.712	0.814	0.235	0.814	0.235	0.814	0.235	0.712	0.712
25.500	0.816	0.716	0.816	0.246	0.816	0.246	0.816	0.246	0.716	0.716
26.500	0.812	0.710	0.812	0.231	0.812	0.231	0.812	0.231	0.710	0.710
28.100	0.817	0.716	0.817	0.248	0.817	0.248	0.817	0.248	0.716	0.716
29.700	0.835	0.746	0.835	0.325	0.835	0.325	0.835	0.325	0.746	0.746
30.700	0.845	0.760	0.845	0.364	0.845	0.364	0.845	0.364	0.760	0.760
33.200	0.859	0.782	0.859	0.422	0.859	0.422	0.859	0.422	0.782	0.782
34.500	0.859	0.781	0.859	0.420	0.859	0.420	0.859	0.420	0.781	0.781
36.600	0.861	0.784	0.861	0.428	0.861	0.428	0.861	0.428	0.784	0.784
38.100	0.854	0.774	0.854	0.401	0.854	0.401	0.854	0.401	0.774	0.774
39.700	0.827	0.733	0.827	0.291	0.827	0.291	0.827	0.291	0.733	0.733
40.800	0.821	0.723	0.821	0.265	0.821	0.265	0.821	0.265	0.723	0.723
42.900	0.829	0.735	0.829	0.297	0.829	0.297	0.829	0.297	0.735	0.735
44.700	0.808	0.703	0.808	0.213	0.808	0.213	0.808	0.213	0.703	0.703
46.700	0.800	0.690	0.800	0.178	0.800	0.178	0.800	0.178	0.690	0.690
47.500	0.800	0.651	0.800	0.181	0.800	0.181	0.800	0.181	0.691	0.691
48.600	0.816	0.715	0.816	0.245	0.816	0.245	0.816	0.245	0.715	0.715
50.400	0.825	0.729	0.825	0.282	0.825	0.282	0.825	0.282	0.729	0.729
52.400	0.840	0.753	0.840	0.346	0.840	0.346	0.840	0.346	0.753	0.753
54.000	0.856	0.777	0.856	0.409	0.856	0.409	0.856	0.409	0.777	0.777
55.600	0.871	0.800	0.871	0.470	0.871	0.470	0.871	0.470	0.800	0.800
57.800	0.881	0.816	0.881	0.513	0.881	0.513	0.881	0.513	0.816	0.816
60.100	0.887	0.825	0.887	0.536	0.887	0.536	0.887	0.536	0.825	0.825
61.700	0.910	0.861	0.910	0.632	0.910	0.632	0.910	0.632	0.861	0.861
64.100	0.967	0.949	0.967	0.866	0.967	0.866	0.967	0.866	0.949	0.949
66.900	0.966	0.947	0.966	0.860	0.966	0.860	0.966	0.860	0.947	0.947
68.800	0.905	0.853	0.905	0.610	0.905	0.610	0.905	0.610	0.853	0.853
70.300	0.881	0.816	0.881	0.511	0.881	0.511	0.881	0.511	0.816	0.816
72.700	0.854	0.775	0.854	0.403	0.854	0.403	0.854	0.403	0.775	0.775
74.200	0.828	0.734	0.828	0.296	0.828	0.296	0.828	0.296	0.734	0.734
76.500	0.817	0.716	0.817	0.248	0.817	0.248	0.817	0.248	0.716	0.716
77.300	0.835	0.745	0.835	0.324	0.835	0.324	0.835	0.324	0.745	0.745
80.200	0.852	0.771	0.852	0.393	0.852	0.393	0.852	0.393	0.771	0.771
81.600	0.872	0.802	0.872	0.475	0.872	0.475	0.872	0.475	0.802	0.802
83.500	0.876	0.809	0.876	0.493	0.876	0.493	0.876	0.493	0.809	0.809
85.100	0.897	0.840	0.897	0.576	0.897	0.576	0.897	0.576	0.840	0.840
86.500	0.914	0.866	0.914	0.645	0.914	0.645	0.914	0.645	0.866	0.866
87.900	0.944	0.914	0.944	0.772	0.944	0.772	0.944	0.772	0.914	0.914
90.000	0.981	0.970	0.981	0.921	0.981	0.921	0.981	0.921	0.970	0.970
92.500	0.999	0.998	0.999	0.994	0.999	0.994	0.999	0.994	0.998	0.998
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table A.3 (continued)

GAPXL ^a	VALUE OF K IS: 11	12	13	14	15	16	17	18	19	20
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.949	0.949	0.981	0.981	0.949
1.800	0.756	0.756	0.756	0.756	0.756	0.756	0.756	0.908	0.908	0.756
3.300	0.589	0.589	0.589	0.589	0.589	0.589	0.589	0.845	0.845	0.589
5.200	0.584	0.584	0.584	0.584	0.584	0.584	0.584	0.843	0.843	0.584
8.900	0.846	0.846	0.846	0.846	0.846	0.846	0.846	0.942	0.942	0.846
11.800	0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826
14.100	0.446	0.446	0.446	0.446	0.446	0.446	0.446	0.791	0.791	0.446
15.400	0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369
17.300	0.363	0.363	0.363	0.363	0.363	0.363	0.363	0.760	0.760	0.363
18.800	0.278	0.278	0.278	0.278	0.278	0.278	0.278	0.728	0.728	0.278
20.100	0.255	0.255	0.255	0.255	0.255	0.255	0.255	0.719	0.719	0.255
20.600	0.277	0.277	0.277	0.277	0.277	0.277	0.277	0.727	0.727	0.277
22.300	0.274	0.274	0.274	0.274	0.274	0.274	0.274	0.726	0.726	0.274
23.900	0.235	0.235	0.235	0.235	0.235	0.235	0.235	0.712	0.712	0.235
25.500	0.246	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246
26.500	0.231	0.231	0.231	0.231	0.231	0.231	0.231	0.710	0.710	0.231
28.100	0.248	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248
29.700	0.325	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325
30.700	0.364	0.364	0.364	0.364	0.364	0.364	0.364	0.760	0.760	0.364
33.200	0.422	0.422	0.422	0.422	0.422	0.422	0.422	0.782	0.782	0.422
34.500	0.420	0.420	0.420	0.420	0.420	0.420	0.420	0.781	0.781	0.420
36.600	0.428	0.428	0.428	0.428	0.428	0.428	0.428	0.784	0.784	0.428
38.100	0.401	0.401	0.401	0.401	0.401	0.401	0.401	0.774	0.774	0.401
39.700	0.291	0.291	0.291	0.291	0.291	0.291	0.291	0.733	0.733	0.291
40.800	0.265	0.265	0.265	0.265	0.265	0.265	0.265	0.723	0.723	0.265
42.900	0.297	0.297	0.297	0.297	0.297	0.297	0.297	0.735	0.735	0.297
44.700	0.213	0.213	0.213	0.213	0.213	0.213	0.213	0.703	0.703	0.213
46.700	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.690	0.690	0.178
47.500	0.181	0.181	0.181	0.181	0.181	0.181	0.181	0.691	0.691	0.181
48.600	0.245	0.245	0.245	0.245	0.245	0.245	0.245	0.715	0.715	0.245
50.400	0.282	0.282	0.282	0.282	0.282	0.282	0.282	0.729	0.729	0.282
52.400	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346
54.000	0.409	0.409	0.409	0.409	0.409	0.409	0.409	0.777	0.777	0.409
55.600	0.470	0.470	0.470	0.470	0.470	0.470	0.470	0.800	0.800	0.470
57.800	0.513	0.513	0.513	0.513	0.513	0.513	0.513	0.816	0.816	0.513
60.100	0.536	0.536	0.536	0.536	0.536	0.536	0.536	0.825	0.825	0.536
61.700	0.632	0.632	0.632	0.632	0.632	0.632	0.632	0.861	0.861	0.632
64.100	0.866	0.866	0.866	0.866	0.866	0.866	0.866	0.949	0.949	0.866
66.900	0.860	0.860	0.860	0.860	0.860	0.860	0.860	0.947	0.947	0.860
68.800	0.610	0.610	0.610	0.610	0.610	0.610	0.610	0.853	0.853	0.610
70.300	0.511	0.511	0.511	0.511	0.511	0.511	0.511	0.816	0.816	0.511
72.700	0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403
74.200	0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296
76.500	0.248	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248
77.300	0.324	0.324	0.324	0.324	0.324	0.324	0.324	0.745	0.745	0.324
80.200	0.393	0.393	0.393	0.393	0.393	0.393	0.393	0.771	0.771	0.393
81.600	0.475	0.475	0.475	0.475	0.475	0.475	0.475	0.802	0.802	0.475
83.500	0.493	0.493	0.493	0.493	0.493	0.493	0.493	0.809	0.809	0.493
85.100	0.576	0.576	0.576	0.576	0.576	0.576	0.576	0.840	0.840	0.576
86.500	0.645	0.645	0.645	0.645	0.645	0.645	0.645	0.866	0.866	0.645
87.900	0.772	0.772	0.772	0.772	0.772	0.772	0.772	0.914	0.914	0.772
90.000	0.921	0.921	0.921	0.921	0.921	0.921	0.921	0.970	0.970	0.921
92.500	0.994	0.994	0.994	0.994	0.994	0.994	0.994	0.998	0.998	0.994
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table A.3 (continued)

GAPXL ^a	VALUE OF K IS: 21	22	23	24	25	26	27	28	29	30
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.949	0.981	0.981	0.949	0.949
1.800	0.756	0.756	0.756	0.756	0.756	0.756	0.908	0.908	0.756	0.756
3.300	0.589	0.589	0.589	0.589	0.589	0.589	0.845	0.845	0.589	0.589
5.200	0.584	0.584	0.584	0.584	0.584	0.584	0.843	0.843	0.584	0.584
8.900	0.846	0.846	0.846	0.846	0.846	0.846	0.942	0.942	0.846	0.846
11.800	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826	0.826
14.100	0.446	0.446	0.446	0.446	0.446	0.446	0.791	0.791	0.446	0.446
15.400	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369	0.369
17.300	0.363	0.363	0.363	0.363	0.363	0.363	0.760	0.760	0.363	0.363
18.800	0.278	0.278	0.278	0.278	0.278	0.278	0.728	0.728	0.278	0.278
20.100	0.255	0.255	0.255	0.255	0.255	0.255	0.719	0.719	0.255	0.255
20.600	0.277	0.277	0.277	0.277	0.277	0.277	0.727	0.727	0.277	0.277
22.300	0.274	0.274	0.274	0.274	0.274	0.274	0.726	0.726	0.274	0.274
23.900	0.235	0.235	0.235	0.235	0.235	0.235	0.712	0.712	0.235	0.235
25.500	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246	0.246
26.500	0.231	0.231	0.231	0.231	0.231	0.231	0.710	0.710	0.231	0.231
28.100	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248	0.248
29.700	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325	0.325
30.700	0.364	0.364	0.364	0.364	0.364	0.364	0.760	0.760	0.364	0.364
33.200	0.422	0.422	0.422	0.422	0.422	0.422	0.782	0.782	0.422	0.422
34.500	0.420	0.420	0.420	0.420	0.420	0.420	0.781	0.781	0.420	0.420
36.600	0.428	0.428	0.428	0.428	0.428	0.428	0.784	0.784	0.428	0.428
38.100	0.401	0.401	0.401	0.401	0.401	0.401	0.774	0.774	0.401	0.401
39.700	0.291	0.291	0.291	0.291	0.291	0.291	0.733	0.733	0.291	0.291
40.800	0.265	0.265	0.265	0.265	0.265	0.265	0.723	0.723	0.265	0.265
42.900	0.297	0.297	0.297	0.297	0.297	0.297	0.735	0.735	0.297	0.297
44.700	0.213	0.213	0.213	0.213	0.213	0.213	0.703	0.703	0.213	0.213
46.700	0.178	0.178	0.178	0.178	0.178	0.178	0.690	0.690	0.178	0.178
47.500	0.181	0.181	0.181	0.181	0.181	0.181	0.691	0.691	0.181	0.181
48.600	0.245	0.245	0.245	0.245	0.245	0.245	0.715	0.715	0.245	0.245
50.400	0.282	0.282	0.282	0.282	0.282	0.282	0.729	0.729	0.282	0.282
52.400	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346	0.346
54.000	0.409	0.409	0.409	0.409	0.409	0.409	0.777	0.777	0.409	0.409
55.600	0.470	0.470	0.470	0.470	0.470	0.470	0.800	0.800	0.470	0.470
57.800	0.513	0.513	0.513	0.513	0.513	0.513	0.816	0.816	0.513	0.513
60.100	0.536	0.536	0.536	0.536	0.536	0.536	0.825	0.825	0.536	0.536
61.700	0.632	0.632	0.632	0.632	0.632	0.632	0.861	0.861	0.632	0.632
64.100	0.866	0.866	0.866	0.866	0.866	0.866	0.949	0.949	0.866	0.866
66.900	0.860	0.860	0.860	0.860	0.860	0.860	0.947	0.947	0.860	0.860
68.800	0.610	0.610	0.610	0.610	0.610	0.610	0.853	0.853	0.610	0.610
70.300	0.511	0.511	0.511	0.511	0.511	0.511	0.816	0.816	0.511	0.511
72.700	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403	0.403
74.200	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296	0.296
76.500	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248	0.248
77.300	0.324	0.324	0.324	0.324	0.324	0.324	0.745	0.745	0.324	0.324
80.200	0.393	0.393	0.393	0.393	0.393	0.393	0.771	0.771	0.393	0.393
81.600	0.475	0.475	0.475	0.475	0.475	0.475	0.802	0.802	0.475	0.475
83.500	0.493	0.493	0.493	0.493	0.493	0.493	0.809	0.809	0.493	0.493
85.100	0.576	0.576	0.576	0.576	0.576	0.576	0.840	0.840	0.576	0.576
86.500	0.645	0.645	0.645	0.645	0.645	0.645	0.866	0.866	0.645	0.645
87.900	0.772	0.772	0.772	0.772	0.772	0.772	0.914	0.914	0.772	0.772
90.000	0.921	0.921	0.921	0.921	0.921	0.921	0.970	0.970	0.921	0.921
92.500	0.994	0.994	0.994	0.994	0.994	0.994	0.999	0.999	0.994	0.994
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table A.3 (continued)

GAPXL ^a	VALUE OF K IS: 31	32	33	34	35	36	37	38	39	40
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.931	0.938	0.988	0.988	0.988
1.800	0.756	0.756	0.756	0.756	0.756	0.708	0.941	0.941	0.941	0.941
3.300	0.589	0.589	0.589	0.589	0.589	0.845	0.900	0.900	0.900	0.900
5.200	0.584	0.584	0.584	0.584	0.584	0.843	0.899	0.899	0.899	0.899
8.900	0.846	0.846	0.846	0.846	0.846	0.942	0.962	0.962	0.962	0.962
11.800	0.826	0.826	0.826	0.826	0.826	0.934	0.958	0.958	0.958	0.958
14.100	0.446	0.446	0.446	0.446	0.446	0.791	0.865	0.865	0.865	0.865
15.400	0.369	0.369	0.369	0.369	0.369	0.762	0.846	0.846	0.846	0.846
17.300	0.363	0.363	0.363	0.363	0.363	0.760	0.845	0.845	0.845	0.845
18.800	0.278	0.278	0.278	0.278	0.278	0.728	0.824	0.824	0.824	0.824
20.100	0.255	0.255	0.255	0.255	0.255	0.719	0.818	0.818	0.818	0.818
20.600	0.277	0.277	0.277	0.277	0.277	0.727	0.824	0.824	0.824	0.824
22.300	0.274	0.274	0.274	0.274	0.274	0.726	0.823	0.823	0.823	0.823
23.900	0.235	0.235	0.235	0.235	0.235	0.712	0.814	0.814	0.814	0.814
25.500	0.246	0.246	0.246	0.246	0.246	0.716	0.816	0.816	0.816	0.816
26.500	0.231	0.231	0.231	0.231	0.231	0.710	0.812	0.812	0.812	0.812
28.100	0.248	0.248	0.248	0.248	0.248	0.716	0.817	0.817	0.817	0.817
29.700	0.325	0.325	0.325	0.325	0.325	0.746	0.835	0.835	0.835	0.835
30.700	0.364	0.364	0.364	0.364	0.364	0.760	0.845	0.845	0.845	0.845
33.200	0.422	0.422	0.422	0.422	0.422	0.782	0.859	0.859	0.859	0.859
34.500	0.420	0.420	0.420	0.420	0.420	0.781	0.859	0.859	0.859	0.859
36.600	0.428	0.428	0.428	0.428	0.428	0.784	0.861	0.861	0.861	0.861
38.100	0.401	0.401	0.401	0.401	0.401	0.774	0.854	0.854	0.854	0.854
39.700	0.291	0.291	0.291	0.291	0.291	0.733	0.827	0.827	0.827	0.827
40.800	0.265	0.265	0.265	0.265	0.265	0.723	0.821	0.821	0.821	0.821
42.900	0.297	0.297	0.297	0.297	0.297	0.735	0.829	0.829	0.829	0.829
44.700	0.213	0.213	0.213	0.213	0.213	0.703	0.808	0.808	0.808	0.808
46.700	0.178	0.178	0.178	0.178	0.178	0.690	0.800	0.800	0.800	0.800
47.500	0.181	0.181	0.181	0.181	0.181	0.691	0.800	0.800	0.800	0.800
48.600	0.245	0.245	0.245	0.245	0.245	0.715	0.816	0.816	0.816	0.816
50.400	0.282	0.282	0.282	0.282	0.282	0.729	0.825	0.825	0.825	0.825
52.400	0.346	0.346	0.346	0.346	0.346	0.753	0.840	0.840	0.840	0.840
54.000	0.409	0.409	0.409	0.409	0.409	0.777	0.856	0.856	0.856	0.856
55.600	0.470	0.470	0.470	0.470	0.470	0.800	0.871	0.871	0.871	0.871
57.800	0.513	0.513	0.513	0.513	0.513	0.816	0.881	0.881	0.881	0.881
60.100	0.536	0.536	0.536	0.536	0.536	0.825	0.887	0.887	0.887	0.887
61.700	0.632	0.632	0.632	0.632	0.632	0.861	0.910	0.910	0.910	0.910
64.100	0.866	0.866	0.866	0.866	0.866	0.949	0.967	0.967	0.967	0.967
66.900	0.860	0.860	0.860	0.860	0.860	0.947	0.966	0.966	0.966	0.966
68.800	0.610	0.610	0.610	0.610	0.610	0.853	0.905	0.905	0.905	0.905
70.300	0.511	0.511	0.511	0.511	0.511	0.816	0.881	0.881	0.881	0.881
72.700	0.403	0.403	0.403	0.403	0.403	0.775	0.854	0.854	0.854	0.854
74.200	0.296	0.296	0.296	0.296	0.296	0.734	0.828	0.828	0.828	0.828
76.500	0.248	0.248	0.248	0.248	0.248	0.716	0.817	0.817	0.817	0.817
77.300	0.324	0.324	0.324	0.324	0.324	0.745	0.835	0.835	0.835	0.835
80.200	0.393	0.393	0.393	0.393	0.393	0.771	0.852	0.852	0.852	0.852
81.600	0.475	0.475	0.475	0.475	0.475	0.802	0.872	0.872	0.872	0.872
83.500	0.493	0.493	0.493	0.493	0.493	0.809	0.876	0.876	0.876	0.876
85.100	0.576	0.576	0.576	0.576	0.576	0.840	0.897	0.897	0.897	0.897
86.500	0.645	0.645	0.645	0.645	0.645	0.865	0.914	0.914	0.914	0.914
87.900	0.772	0.772	0.772	0.772	0.772	0.914	0.944	0.944	0.944	0.944
90.000	0.921	0.921	0.921	0.921	0.921	0.970	0.981	0.981	0.981	0.981
92.500	0.994	0.994	0.994	0.994	0.994	0.998	0.999	0.999	0.999	0.999
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

Appendix B

B-1/SHROUD 1 BUNDLE-AVERAGED GEOMETRY DATA,
MAXIMUM RESTRICTION DEFINITION

The data in Appendix B represent the input from card groups 5 and 6 as required by ref. 2 for the B-1/shroud 1 COBRA-IV model based on the maximum restriction definition.



Table B.1. Shroud flow area reductions for the B-1/shroud 1 COBRA-IV model based on the maximum restriction definition

AXL ^a	FRACTIONAL SHROUD PLCW AREA
-47.473	1.000
0.000	0.983
1.800	0.915
3.300	0.853
5.200	0.851
8.900	0.947
11.800	0.940
14.100	0.797
15.400	0.766
17.300	0.763
18.800	0.729
20.100	0.718
20.600	0.728
22.300	0.681
23.900	0.662
25.500	0.715
26.500	0.709
28.100	0.716
29.700	0.726
30.700	0.764
33.200	0.787
34.500	0.786
36.600	0.790
38.100	0.779
39.700	0.734
40.800	0.706
42.900	0.736
44.700	0.681
46.700	0.626
47.500	0.673
49.600	0.699
50.400	0.724
52.400	0.756
54.000	0.782
55.600	0.806
57.800	0.823
60.100	0.832
61.700	0.869
64.100	0.954
66.900	0.952
68.800	0.860
70.300	0.822
72.700	0.779
74.200	0.735
76.500	0.635
77.300	0.682
80.200	0.776
81.600	0.808
83.500	0.815
85.100	0.847
86.500	0.874
87.900	0.920
90.000	0.973
92.500	0.998
96.190	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table B.2. AFACT values for the B-1/shroud 1 COBRA-IV model based on the maximum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.473		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		0.993	0.989	0.989	0.989	0.993	0.984	0.973	0.973	0.973	0.984
1.800		0.964	0.944	0.944	0.944	0.964	0.923	0.867	0.867	0.867	0.923
3.300		0.937	0.903	0.903	0.903	0.937	0.867	0.770	0.770	0.770	0.867
5.200		0.936	0.902	0.902	0.902	0.936	0.865	0.767	0.767	0.767	0.865
8.900		0.977	0.965	0.965	0.965	0.977	0.952	0.917	0.917	0.917	0.952
11.800		0.974	0.960	0.960	0.960	0.974	0.946	0.906	0.906	0.906	0.946
14.100		0.913	0.866	0.866	0.866	0.913	0.817	0.683	0.683	0.683	0.817
15.400		0.900	0.846	0.846	0.846	0.900	0.789	0.635	0.635	0.635	0.789
17.300		0.899	0.844	0.844	0.844	0.899	0.787	0.631	0.631	0.631	0.787
18.800		0.884	0.822	0.822	0.822	0.884	0.756	0.577	0.577	0.577	0.756
20.100		0.880	0.815	0.815	0.815	0.880	0.746	0.561	0.561	0.561	0.746
20.600		0.884	0.821	0.821	0.821	0.884	0.755	0.576	0.576	0.576	0.755
22.300		0.864	0.790	0.790	0.790	0.864	0.713	0.503	0.503	0.503	0.713
23.900		0.856	0.778	0.778	0.778	0.856	0.695	0.473	0.473	0.473	0.695
25.500		0.878	0.813	0.813	0.813	0.878	0.743	0.556	0.556	0.556	0.743
26.500		0.876	0.809	0.809	0.809	0.876	0.738	0.546	0.546	0.546	0.738
28.100		0.879	0.813	0.813	0.813	0.879	0.744	0.557	0.557	0.557	0.744
29.700		0.883	0.820	0.820	0.820	0.883	0.753	0.573	0.573	0.573	0.753
30.700		0.899	0.845	0.845	0.845	0.899	0.787	0.632	0.632	0.632	0.787
33.200		0.909	0.860	0.860	0.860	0.909	0.808	0.668	0.668	0.668	0.808
34.500		0.909	0.860	0.860	0.860	0.909	0.808	0.667	0.667	0.667	0.808
36.600		0.910	0.862	0.862	0.862	0.910	0.810	0.672	0.672	0.672	0.810
38.100		0.905	0.854	0.854	0.854	0.905	0.801	0.655	0.655	0.655	0.801
39.700		0.914	0.825	0.825	0.825	0.886	0.815	0.585	0.585	0.585	0.760
40.800		0.905	0.807	0.807	0.807	0.875	0.796	0.542	0.542	0.542	0.735
42.900		0.887	0.827	0.827	0.827	0.887	0.762	0.589	0.589	0.589	0.762
44.700		0.913	0.790	0.790	0.790	0.864	0.829	0.503	0.503	0.503	0.713
46.700		0.898	0.754	0.754	0.754	0.840	0.800	0.416	0.416	0.416	0.663
47.500		0.911	0.785	0.785	0.785	0.860	0.825	0.490	0.490	0.490	0.705
48.600		0.871	0.802	0.802	0.802	0.871	0.728	0.530	0.530	0.530	0.728
50.400		0.882	0.818	0.818	0.818	0.882	0.751	0.569	0.569	0.569	0.751
52.400		0.896	0.840	0.840	0.840	0.896	0.780	0.620	0.620	0.620	0.780
54.000		0.907	0.857	0.857	0.857	0.907	0.804	0.660	0.660	0.660	0.804
55.600		0.917	0.873	0.873	0.873	0.917	0.825	0.698	0.698	0.698	0.825
57.800		0.924	0.884	0.884	0.884	0.924	0.841	0.724	0.724	0.724	0.841
60.100		0.928	0.889	0.889	0.889	0.928	0.849	0.738	0.738	0.738	0.849
61.700		0.944	0.914	0.914	0.914	0.944	0.882	0.795	0.795	0.795	0.882
64.100		0.980	0.970	0.970	0.970	0.980	0.958	0.928	0.928	0.928	0.958
66.900		0.979	0.968	0.968	0.968	0.979	0.957	0.925	0.925	0.925	0.957
68.800		0.940	0.908	0.908	0.908	0.940	0.874	0.782	0.782	0.782	0.874
70.300		0.924	0.883	0.883	0.883	0.924	0.840	0.723	0.723	0.723	0.840
72.700		0.906	0.855	0.855	0.855	0.906	0.801	0.656	0.656	0.656	0.801
74.200		0.887	0.825	0.825	0.825	0.925	0.761	0.586	0.586	0.586	0.825
76.500		0.844	0.760	0.760	0.760	0.897	0.671	0.431	0.431	0.431	0.798
77.300		0.864	0.791	0.791	0.791	0.911	0.713	0.504	0.504	0.504	0.824
80.200		0.904	0.852	0.852	0.852	0.904	0.798	0.650	0.650	0.650	0.798
81.600		0.918	0.874	0.874	0.874	0.918	0.827	0.701	0.701	0.701	0.827
83.500		0.921	0.879	0.879	0.879	0.921	0.834	0.712	0.712	0.712	0.834
85.100		0.935	0.900	0.900	0.900	0.935	0.862	0.762	0.762	0.762	0.862
86.500		0.946	0.917	0.917	0.917	0.946	0.886	0.803	0.803	0.803	0.886
87.900		0.966	0.948	0.948	0.948	0.966	0.928	0.876	0.876	0.876	0.928
90.000		0.988	0.982	0.982	0.982	0.988	0.976	0.958	0.958	0.958	0.976
92.500		0.999	0.999	0.999	0.999	0.999	0.998	0.997	0.997	0.997	0.998
96.190		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table B.2 (continued)

AXL ^a	VALUE OF I IS: 11	12	13	14	15	16	17	18	19	20
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.984	0.973	0.973	0.973	0.984	0.984	0.973	0.973	0.973	0.984
1.800	0.923	0.867	0.867	0.867	0.923	0.923	0.867	0.867	0.867	0.923
3.300	0.867	0.770	0.770	0.770	0.867	0.867	0.770	0.770	0.770	0.867
5.200	0.865	0.767	0.767	0.767	0.865	0.865	0.767	0.767	0.767	0.865
8.900	0.952	0.917	0.917	0.917	0.952	0.952	0.917	0.917	0.917	0.952
11.800	0.946	0.906	0.906	0.906	0.946	0.946	0.906	0.906	0.906	0.946
14.100	0.817	0.683	0.683	0.683	0.817	0.817	0.683	0.683	0.683	0.817
15.400	0.789	0.635	0.635	0.635	0.789	0.789	0.635	0.635	0.635	0.789
17.300	0.787	0.631	0.631	0.631	0.787	0.787	0.631	0.631	0.631	0.787
18.800	0.756	0.577	0.577	0.577	0.756	0.756	0.577	0.577	0.577	0.756
20.100	0.746	0.561	0.561	0.561	0.746	0.746	0.561	0.561	0.561	0.746
20.600	0.755	0.576	0.576	0.576	0.755	0.755	0.576	0.576	0.576	0.755
22.300	0.713	0.503	0.503	0.503	0.713	0.713	0.503	0.503	0.503	0.713
23.900	0.695	0.473	0.473	0.473	0.695	0.695	0.473	0.473	0.473	0.695
25.500	0.743	0.556	0.556	0.556	0.743	0.743	0.556	0.556	0.556	0.743
26.500	0.738	0.546	0.546	0.546	0.738	0.738	0.546	0.546	0.546	0.738
28.100	0.744	0.557	0.557	0.557	0.744	0.744	0.557	0.557	0.557	0.744
29.700	0.753	0.573	0.573	0.573	0.753	0.753	0.573	0.573	0.573	0.753
30.700	0.787	0.632	0.632	0.632	0.787	0.787	0.632	0.632	0.632	0.787
33.200	0.808	0.668	0.668	0.668	0.808	0.808	0.668	0.668	0.668	0.808
34.500	0.808	0.667	0.667	0.667	0.808	0.808	0.667	0.667	0.667	0.808
36.600	0.810	0.672	0.672	0.672	0.810	0.810	0.672	0.672	0.672	0.810
38.100	0.801	0.655	0.655	0.655	0.801	0.801	0.655	0.655	0.655	0.801
39.700	0.760	0.585	0.585	0.585	0.760	0.760	0.585	0.585	0.585	0.760
40.800	0.735	0.542	0.542	0.542	0.735	0.735	0.542	0.542	0.542	0.735
42.900	0.762	0.589	0.589	0.589	0.762	0.762	0.589	0.589	0.589	0.762
44.700	0.777	0.503	0.503	0.503	0.777	0.777	0.503	0.503	0.503	0.777
46.700	0.738	0.416	0.416	0.416	0.663	0.663	0.416	0.416	0.416	0.663
47.500	0.771	0.490	0.490	0.490	0.705	0.705	0.490	0.490	0.490	0.705
48.600	0.728	0.530	0.530	0.530	0.728	0.728	0.530	0.530	0.530	0.728
50.400	0.751	0.569	0.569	0.569	0.751	0.751	0.569	0.569	0.569	0.751
52.400	0.780	0.620	0.620	0.620	0.780	0.780	0.620	0.620	0.620	0.780
54.000	0.804	0.660	0.660	0.660	0.804	0.804	0.660	0.660	0.660	0.804
55.600	0.825	0.698	0.698	0.698	0.825	0.825	0.698	0.698	0.698	0.825
57.800	0.841	0.724	0.724	0.724	0.841	0.841	0.724	0.724	0.724	0.841
60.100	0.849	0.738	0.738	0.738	0.849	0.849	0.738	0.738	0.738	0.849
61.700	0.882	0.795	0.795	0.795	0.882	0.882	0.795	0.795	0.795	0.882
64.100	0.958	0.928	0.928	0.928	0.958	0.958	0.928	0.928	0.928	0.958
66.900	0.957	0.925	0.925	0.925	0.957	0.957	0.925	0.925	0.925	0.957
68.800	0.874	0.782	0.782	0.782	0.874	0.874	0.782	0.782	0.782	0.874
70.300	0.840	0.723	0.723	0.723	0.840	0.840	0.723	0.723	0.723	0.840
72.700	0.801	0.656	0.656	0.656	0.801	0.801	0.656	0.656	0.656	0.801
74.200	0.761	0.586	0.586	0.586	0.853	0.761	0.586	0.586	0.586	0.853
76.500	0.671	0.431	0.431	0.431	0.798	0.671	0.431	0.431	0.431	0.799
77.300	0.713	0.504	0.504	0.504	0.824	0.713	0.504	0.504	0.504	0.824
80.200	0.798	0.650	0.650	0.650	0.798	0.798	0.650	0.650	0.650	0.798
81.600	0.827	0.701	0.701	0.701	0.827	0.827	0.701	0.701	0.701	0.827
83.500	0.834	0.712	0.712	0.712	0.834	0.834	0.712	0.712	0.712	0.834
85.100	0.862	0.762	0.762	0.762	0.862	0.862	0.762	0.762	0.762	0.862
86.500	0.886	0.803	0.803	0.803	0.886	0.886	0.803	0.803	0.803	0.886
87.900	0.928	0.876	0.876	0.876	0.928	0.928	0.876	0.876	0.876	0.928
90.000	0.976	0.958	0.958	0.958	0.976	0.976	0.958	0.958	0.958	0.976
92.500	0.998	0.997	0.997	0.997	0.998	0.998	0.997	0.997	0.997	0.998
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table B.2 (continued)

AXL ^a	VALUE OF I IS: 21	22	23	24	25
-47.473	1.000	1.000	1.000	1.000	1.000
0.000	0.993	0.989	0.989	0.989	0.993
1.800	0.964	0.944	0.944	0.944	0.964
3.300	0.937	0.903	0.903	0.903	0.937
5.200	0.936	0.902	0.902	0.902	0.936
8.900	0.977	0.965	0.965	0.965	0.977
11.800	0.974	0.960	0.960	0.960	0.974
14.100	0.913	0.866	0.866	0.866	0.913
15.400	0.900	0.846	0.846	0.846	0.900
17.300	0.899	0.844	0.844	0.844	0.899
18.800	0.884	0.822	0.822	0.822	0.884
20.100	0.880	0.815	0.815	0.815	0.880
20.600	0.884	0.821	0.821	0.821	0.884
22.300	0.864	0.790	0.790	0.790	0.864
23.900	0.856	0.778	0.778	0.778	0.856
25.500	0.878	0.813	0.813	0.813	0.878
26.500	0.876	0.809	0.809	0.809	0.876
28.100	0.879	0.813	0.813	0.813	0.879
29.700	0.883	0.820	0.820	0.820	0.883
30.700	0.899	0.845	0.845	0.845	0.899
33.200	0.909	0.860	0.860	0.860	0.909
34.500	0.909	0.860	0.860	0.860	0.909
36.600	0.910	0.862	0.862	0.862	0.910
38.100	0.905	0.854	0.854	0.854	0.905
39.700	0.886	0.825	0.825	0.825	0.886
40.800	0.875	0.807	0.807	0.807	0.875
42.900	0.887	0.827	0.827	0.827	0.887
44.700	0.864	0.790	0.790	0.790	0.864
46.700	0.840	0.754	0.754	0.754	0.840
47.500	0.860	0.785	0.785	0.785	0.860
48.600	0.871	0.802	0.802	0.802	0.871
50.400	0.882	0.818	0.818	0.818	0.882
52.400	0.896	0.840	0.840	0.840	0.896
54.000	0.907	0.857	0.857	0.857	0.907
55.600	0.917	0.873	0.873	0.873	0.917
57.800	0.924	0.884	0.884	0.884	0.924
60.100	0.928	0.889	0.889	0.889	0.928
61.700	0.944	0.914	0.914	0.914	0.944
64.100	0.980	0.970	0.970	0.970	0.980
66.900	0.979	0.968	0.968	0.968	0.979
68.800	0.940	0.908	0.908	0.908	0.940
70.300	0.924	0.883	0.883	0.883	0.924
72.700	0.906	0.855	0.855	0.855	0.906
74.200	0.887	0.825	0.825	0.825	0.887
76.500	0.844	0.760	0.760	0.760	0.844
77.300	0.864	0.791	0.791	0.791	0.864
80.200	0.904	0.852	0.852	0.852	0.904
81.600	0.918	0.874	0.874	0.874	0.918
83.500	0.921	0.879	0.879	0.879	0.921
85.100	0.935	0.900	0.900	0.900	0.935
86.500	0.946	0.917	0.917	0.917	0.946
87.900	0.966	0.948	0.948	0.948	0.966
90.000	0.988	0.982	0.982	0.982	0.988
92.500	0.999	0.999	0.999	0.999	0.999
96.190	1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table B.3. GFACT values for the B-1/shroud 1 COBRA-IV model based on the maximum restriction definition

GAPXL ^a	VALUE	CF	K	IS: 1	2	3	4	5	6	7	8	9	10
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.988	0.981	0.988	0.949	0.988	0.949	0.988	0.949	0.988	0.949	0.981	0.981	0.981
1.800	0.941	0.908	0.941	0.756	0.941	0.756	0.941	0.756	0.941	0.756	0.908	0.908	0.908
3.300	0.900	0.845	0.900	0.589	0.900	0.589	0.900	0.589	0.900	0.589	0.845	0.845	0.845
5.200	0.899	0.843	0.899	0.584	0.899	0.584	0.899	0.584	0.899	0.584	0.843	0.843	0.843
8.900	0.962	0.942	0.962	0.846	0.962	0.846	0.962	0.846	0.962	0.846	0.942	0.942	0.942
11.800	0.958	0.934	0.958	0.826	0.958	0.826	0.958	0.826	0.958	0.826	0.934	0.934	0.934
14.100	0.865	0.791	0.865	0.446	0.865	0.446	0.865	0.446	0.865	0.446	0.791	0.791	0.791
15.400	0.846	0.762	0.846	0.369	0.846	0.369	0.846	0.369	0.846	0.369	0.762	0.762	0.762
17.300	0.845	0.760	0.845	0.363	0.845	0.363	0.845	0.363	0.845	0.363	0.760	0.760	0.760
18.800	0.824	0.728	0.824	0.278	0.824	0.278	0.824	0.278	0.824	0.278	0.728	0.728	0.728
20.100	0.818	0.719	0.818	0.254	0.818	0.254	0.818	0.254	0.818	0.254	0.719	0.719	0.719
20.600	0.824	0.727	0.824	0.277	0.824	0.277	0.824	0.277	0.824	0.277	0.727	0.727	0.727
22.300	0.797	0.685	0.797	0.166	0.797	0.166	0.797	0.166	0.797	0.166	0.685	0.685	0.685
23.900	0.786	0.669	0.786	0.121	0.786	0.121	0.786	0.121	0.786	0.121	0.669	0.669	0.669
25.500	0.816	0.716	0.816	0.246	0.816	0.246	0.816	0.246	0.816	0.246	0.716	0.716	0.716
26.500	0.812	0.710	0.812	0.231	0.812	0.231	0.812	0.231	0.812	0.231	0.710	0.710	0.710
28.100	0.817	0.716	0.817	0.248	0.817	0.248	0.817	0.248	0.817	0.248	0.716	0.716	0.716
29.700	0.823	0.726	0.823	0.272	0.823	0.272	0.823	0.272	0.823	0.272	0.726	0.726	0.726
30.700	0.845	0.760	0.845	0.364	0.845	0.364	0.845	0.364	0.845	0.364	0.760	0.760	0.760
33.200	0.859	0.782	0.859	0.422	0.859	0.422	0.859	0.422	0.859	0.422	0.782	0.782	0.782
34.500	0.859	0.781	0.859	0.420	0.859	0.420	0.859	0.420	0.859	0.420	0.781	0.781	0.781
36.600	0.861	0.784	0.861	0.428	0.861	0.428	0.861	0.428	0.861	0.428	0.784	0.784	0.784
38.100	0.854	0.774	0.854	0.401	0.854	0.401	0.854	0.401	0.854	0.401	0.774	0.774	0.774
39.700	0.827	0.733	0.827	0.291	0.827	0.291	0.827	0.291	0.827	0.291	0.733	0.733	0.733
40.800	0.811	0.708	0.811	0.225	0.811	0.225	0.811	0.225	0.811	0.225	0.708	0.708	0.708
42.900	0.829	0.735	0.829	0.297	0.829	0.297	0.829	0.297	0.829	0.297	0.735	0.735	0.735
44.700	0.797	0.685	0.797	0.166	0.797	0.166	0.797	0.166	0.797	0.166	0.685	0.685	0.685
46.700	0.765	0.637	0.765	0.037	0.765	0.037	0.765	0.037	0.765	0.037	0.637	0.637	0.637
47.500	0.792	0.678	0.792	0.146	0.792	0.146	0.792	0.146	0.792	0.146	0.678	0.678	0.678
48.600	0.807	0.701	0.807	0.207	0.807	0.207	0.807	0.207	0.807	0.207	0.701	0.701	0.701
50.400	0.821	0.723	0.821	0.266	0.821	0.266	0.821	0.266	0.821	0.266	0.723	0.723	0.723
52.400	0.840	0.753	0.840	0.346	0.840	0.346	0.840	0.346	0.840	0.346	0.753	0.753	0.753
54.000	0.856	0.777	0.856	0.409	0.856	0.409	0.856	0.409	0.856	0.409	0.777	0.777	0.777
55.600	0.871	0.800	0.871	0.470	0.871	0.470	0.871	0.470	0.871	0.470	0.800	0.800	0.800
57.800	0.881	0.816	0.881	0.513	0.881	0.513	0.881	0.513	0.881	0.513	0.816	0.816	0.816
60.100	0.887	0.825	0.887	0.536	0.887	0.536	0.887	0.536	0.887	0.536	0.825	0.825	0.825
61.700	0.910	0.861	0.910	0.632	0.910	0.632	0.910	0.632	0.910	0.632	0.861	0.861	0.861
64.100	0.967	0.949	0.967	0.866	0.967	0.866	0.967	0.866	0.967	0.866	0.949	0.949	0.949
66.900	0.966	0.947	0.966	0.860	0.966	0.860	0.966	0.860	0.966	0.860	0.947	0.947	0.947
68.800	0.905	0.853	0.905	0.610	0.905	0.610	0.905	0.610	0.905	0.610	0.853	0.853	0.853
70.300	0.881	0.816	0.881	0.511	0.881	0.511	0.881	0.511	0.881	0.511	0.816	0.816	0.816
72.700	0.854	0.775	0.854	0.403	0.854	0.403	0.854	0.403	0.854	0.403	0.775	0.775	0.775
74.200	0.827	0.733	0.827	0.292	0.827	0.292	0.827	0.292	0.827	0.292	0.733	0.733	0.733
76.500	0.771	0.645	0.771	0.059	0.771	0.059	0.771	0.059	0.771	0.059	0.645	0.645	0.645
77.300	0.797	0.686	0.797	0.167	0.797	0.167	0.797	0.167	0.797	0.167	0.686	0.686	0.686
80.200	0.852	0.771	0.852	0.393	0.852	0.393	0.852	0.393	0.852	0.393	0.771	0.771	0.771
81.600	0.872	0.802	0.872	0.475	0.872	0.475	0.872	0.475	0.872	0.475	0.802	0.802	0.802
83.500	0.876	0.809	0.876	0.493	0.876	0.493	0.876	0.493	0.876	0.493	0.809	0.809	0.809
85.100	0.897	0.840	0.897	0.576	0.897	0.576	0.897	0.576	0.897	0.576	0.840	0.840	0.840
86.500	0.914	0.866	0.914	0.645	0.914	0.645	0.914	0.645	0.914	0.645	0.866	0.866	0.866
87.900	0.944	0.914	0.944	0.772	0.944	0.772	0.944	0.772	0.944	0.772	0.914	0.914	0.914
90.000	0.981	0.970	0.981	0.921	0.981	0.921	0.981	0.921	0.981	0.921	0.970	0.970	0.970
92.500	0.999	0.998	0.999	0.994	0.999	0.994	0.999	0.994	0.999	0.994	0.998	0.998	0.998
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table B.3 (continued)

GAPXL ^a	VALUE OF F IS:	11	12	13	14	15	16	17	18	19	20
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.949	0.949	0.949	0.981	0.981	0.949
1.800	0.756	0.756	0.756	0.756	0.756	0.756	0.756	0.756	0.908	0.908	0.756
3.300	0.589	0.589	0.589	0.589	0.589	0.589	0.589	0.589	0.845	0.845	0.589
5.200	0.584	0.584	0.584	0.584	0.584	0.584	0.584	0.584	0.843	0.843	0.584
8.900	0.846	0.846	0.846	0.846	0.846	0.846	0.846	0.846	0.942	0.942	0.846
11.800	0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826
14.100	0.446	0.446	0.446	0.446	0.446	0.446	0.446	0.446	0.791	0.791	0.446
15.400	0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369
17.300	0.363	0.363	0.363	0.363	0.363	0.363	0.363	0.363	0.760	0.760	0.363
18.800	0.278	0.278	0.278	0.278	0.278	0.278	0.278	0.278	0.728	0.728	0.278
20.100	0.254	0.254	0.254	0.254	0.254	0.254	0.254	0.254	0.719	0.719	0.254
20.600	0.277	0.277	0.277	0.277	0.277	0.277	0.277	0.277	0.727	0.727	0.277
22.300	0.166	0.166	0.166	0.166	0.166	0.166	0.166	0.166	0.685	0.685	0.166
23.900	0.121	0.121	0.121	0.121	0.121	0.121	0.121	0.121	0.669	0.669	0.121
25.500	0.246	0.246	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246
26.500	0.231	0.231	0.231	0.231	0.231	0.231	0.231	0.231	0.710	0.710	0.231
28.100	0.248	0.248	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248
29.700	0.272	0.272	0.272	0.272	0.272	0.272	0.272	0.272	0.726	0.726	0.272
30.700	0.364	0.364	0.364	0.364	0.364	0.364	0.364	0.364	0.760	0.760	0.364
33.200	0.422	0.422	0.422	0.422	0.422	0.422	0.422	0.422	0.782	0.782	0.422
34.500	0.420	0.420	0.420	0.420	0.420	0.420	0.420	0.420	0.781	0.781	0.420
36.600	0.428	0.428	0.428	0.428	0.428	0.428	0.428	0.428	0.784	0.784	0.428
38.100	0.401	0.401	0.401	0.401	0.401	0.401	0.401	0.401	0.774	0.774	0.401
39.700	0.291	0.291	0.291	0.291	0.291	0.291	0.291	0.291	0.733	0.733	0.291
40.800	0.225	0.225	0.225	0.225	0.225	0.225	0.225	0.225	0.708	0.708	0.225
42.900	0.297	0.297	0.297	0.297	0.297	0.297	0.297	0.297	0.735	0.735	0.297
44.700	0.166	0.166	0.166	0.166	0.166	0.166	0.166	0.166	0.685	0.685	0.166
46.700	0.037	0.037	0.037	0.037	0.037	0.037	0.037	0.037	0.637	0.637	0.037
47.500	0.146	0.146	0.146	0.146	0.146	0.146	0.146	0.146	0.678	0.678	0.146
48.600	0.207	0.207	0.207	0.207	0.207	0.207	0.207	0.207	0.701	0.701	0.207
50.400	0.266	0.266	0.266	0.266	0.266	0.266	0.266	0.266	0.723	0.723	0.266
52.400	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346
54.000	0.409	0.409	0.409	0.409	0.409	0.409	0.409	0.409	0.777	0.777	0.409
55.600	0.470	0.470	0.470	0.470	0.470	0.470	0.470	0.470	0.800	0.800	0.470
57.800	0.513	0.513	0.513	0.513	0.513	0.513	0.513	0.513	0.816	0.816	0.513
60.100	0.536	0.536	0.536	0.536	0.536	0.536	0.536	0.536	0.825	0.825	0.536
61.700	0.632	0.632	0.632	0.632	0.632	0.632	0.632	0.632	0.861	0.861	0.632
64.100	0.866	0.866	0.866	0.866	0.866	0.866	0.866	0.866	0.949	0.949	0.866
66.900	0.860	0.860	0.860	0.860	0.860	0.860	0.860	0.860	0.947	0.947	0.860
68.800	0.610	0.610	0.610	0.610	0.610	0.610	0.610	0.610	0.853	0.853	0.610
70.300	0.511	0.511	0.511	0.511	0.511	0.511	0.511	0.511	0.816	0.816	0.511
72.700	0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403
74.200	0.292	0.292	0.292	0.292	0.292	0.292	0.292	0.292	0.733	0.733	0.292
76.500	0.059	0.059	0.059	0.059	0.059	0.059	0.059	0.059	0.645	0.645	0.059
77.300	0.167	0.167	0.167	0.167	0.167	0.167	0.167	0.167	0.686	0.686	0.167
80.200	0.393	0.393	0.393	0.393	0.393	0.393	0.393	0.393	0.771	0.771	0.393
81.600	0.475	0.475	0.475	0.475	0.475	0.475	0.475	0.475	0.802	0.802	0.475
83.500	0.493	0.493	0.493	0.493	0.493	0.493	0.493	0.493	0.809	0.809	0.493
85.100	0.576	0.576	0.576	0.576	0.576	0.576	0.576	0.576	0.840	0.840	0.576
86.500	0.645	0.645	0.645	0.645	0.645	0.645	0.645	0.645	0.866	0.866	0.645
87.900	0.772	0.772	0.772	0.772	0.772	0.772	0.772	0.772	0.914	0.914	0.772
90.000	0.921	0.921	0.921	0.921	0.921	0.921	0.921	0.921	0.970	0.970	0.921
92.500	0.994	0.994	0.994	0.994	0.994	0.994	0.994	0.994	0.998	0.998	0.994
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table B.3 (continued)

GAPXL ^a	VALUE OF K IS: 21	22	23	24	25	26	27	28	29	30
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.949	0.981	0.981	0.949	0.949
1.800	0.756	0.756	0.756	0.756	0.756	0.756	0.908	0.908	0.756	0.756
3.300	0.589	0.589	0.589	0.589	0.589	0.589	0.845	0.845	0.589	0.589
5.200	0.584	0.584	0.584	0.584	0.584	0.584	0.843	0.843	0.584	0.584
8.900	0.846	0.846	0.846	0.846	0.846	0.846	0.942	0.942	0.846	0.846
11.800	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826	0.826
14.100	0.446	0.446	0.446	0.446	0.446	0.446	0.791	0.791	0.446	0.446
15.400	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369	0.369
17.300	0.363	0.363	0.363	0.363	0.363	0.363	0.760	0.760	0.363	0.363
18.800	0.278	0.278	0.278	0.278	0.278	0.278	0.728	0.728	0.278	0.278
20.100	0.254	0.254	0.254	0.254	0.254	0.254	0.719	0.719	0.254	0.254
20.600	0.277	0.277	0.277	0.277	0.277	0.277	0.727	0.727	0.277	0.277
22.300	0.166	0.166	0.166	0.166	0.166	0.166	0.685	0.685	0.166	0.166
23.900	0.121	0.121	0.121	0.121	0.121	0.121	0.669	0.669	0.121	0.121
25.500	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246	0.246
26.500	0.231	0.231	0.231	0.231	0.231	0.231	0.710	0.710	0.231	0.231
28.100	0.248	0.248	0.248	0.248	0.248	0.248	0.716	0.716	0.248	0.248
29.700	0.272	0.272	0.272	0.272	0.272	0.272	0.726	0.726	0.272	0.272
30.700	0.364	0.364	0.364	0.364	0.364	0.364	0.760	0.760	0.364	0.364
33.200	0.422	0.422	0.422	0.422	0.422	0.422	0.782	0.782	0.422	0.422
34.500	0.420	0.420	0.420	0.420	0.420	0.420	0.781	0.781	0.420	0.420
36.600	0.428	0.428	0.428	0.428	0.428	0.428	0.784	0.784	0.428	0.428
38.100	0.401	0.401	0.401	0.401	0.401	0.401	0.774	0.774	0.401	0.401
39.700	0.291	0.291	0.291	0.291	0.291	0.291	0.733	0.733	0.291	0.291
40.800	0.225	0.225	0.225	0.225	0.225	0.225	0.708	0.708	0.225	0.225
42.900	0.297	0.297	0.297	0.297	0.297	0.297	0.735	0.735	0.297	0.297
44.700	0.166	0.166	0.166	0.166	0.166	0.166	0.685	0.685	0.166	0.166
46.700	0.037	0.037	0.037	0.037	0.037	0.037	0.637	0.637	0.037	0.037
47.500	0.146	0.146	0.146	0.146	0.146	0.146	0.678	0.678	0.146	0.146
48.600	0.207	0.207	0.207	0.207	0.207	0.207	0.701	0.701	0.207	0.207
50.400	0.266	0.266	0.266	0.266	0.266	0.266	0.723	0.723	0.266	0.266
52.400	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346	0.346
54.000	0.409	0.409	0.409	0.409	0.409	0.409	0.777	0.777	0.409	0.409
55.600	0.470	0.470	0.470	0.470	0.470	0.470	0.800	0.800	0.470	0.470
57.800	0.513	0.513	0.513	0.513	0.513	0.513	0.816	0.816	0.513	0.513
60.100	0.536	0.536	0.536	0.536	0.536	0.536	0.825	0.825	0.536	0.536
61.700	0.632	0.632	0.632	0.632	0.632	0.632	0.861	0.861	0.632	0.632
64.100	0.866	0.866	0.866	0.866	0.866	0.866	0.949	0.949	0.866	0.866
66.900	0.860	0.860	0.860	0.860	0.860	0.860	0.947	0.947	0.860	0.860
68.800	0.610	0.610	0.610	0.610	0.610	0.610	0.853	0.853	0.610	0.610
70.300	0.511	0.511	0.511	0.511	0.511	0.511	0.816	0.816	0.511	0.511
72.700	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403	0.403
74.200	0.292	0.292	0.292	0.292	0.292	0.292	0.733	0.733	0.292	0.292
76.500	0.059	0.059	0.059	0.059	0.059	0.059	0.645	0.645	0.059	0.059
77.300	0.167	0.167	0.167	0.167	0.167	0.167	0.686	0.686	0.167	0.167
80.200	0.393	0.393	0.393	0.393	0.393	0.393	0.771	0.771	0.393	0.393
81.600	0.475	0.475	0.475	0.475	0.475	0.475	0.802	0.802	0.475	0.475
83.500	0.493	0.493	0.493	0.493	0.493	0.493	0.809	0.809	0.493	0.493
85.100	0.576	0.576	0.576	0.576	0.576	0.576	0.840	0.840	0.576	0.576
86.500	0.645	0.645	0.645	0.645	0.645	0.645	0.866	0.866	0.645	0.645
87.900	0.772	0.772	0.772	0.772	0.772	0.772	0.914	0.914	0.772	0.772
90.000	0.921	0.921	0.921	0.921	0.921	0.921	0.970	0.970	0.921	0.921
92.500	0.994	0.994	0.994	0.994	0.994	0.994	0.998	0.998	0.994	0.994
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table B.3 (continued)

GAPXL ^a	VALUE OF K IS: 31	32	33	34	35	36	37	38	39	40
-47.473	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	0.949	0.949	0.949	0.949	0.949	0.981	0.988	0.988	0.988	0.988
1.800	0.756	0.756	0.756	0.756	0.756	0.908	0.941	0.941	0.941	0.941
3.300	0.589	0.589	0.589	0.589	0.589	0.845	0.900	0.900	0.900	0.900
5.200	0.584	0.584	0.584	0.584	0.584	0.843	0.899	0.899	0.899	0.899
8.900	0.846	0.846	0.846	0.846	0.846	0.942	0.962	0.962	0.962	0.962
11.800	0.826	0.826	0.826	0.826	0.826	0.934	0.958	0.958	0.958	0.958
14.100	0.446	0.446	0.446	0.446	0.446	0.791	0.865	0.865	0.865	0.865
15.800	0.369	0.369	0.369	0.369	0.369	0.762	0.846	0.846	0.846	0.846
17.300	0.363	0.363	0.363	0.363	0.363	0.760	0.845	0.845	0.845	0.845
18.800	0.278	0.278	0.278	0.278	0.278	0.728	0.824	0.824	0.824	0.824
20.100	0.254	0.254	0.254	0.254	0.254	0.719	0.818	0.818	0.818	0.818
20.600	0.277	0.277	0.277	0.277	0.277	0.727	0.824	0.824	0.824	0.824
22.300	0.166	0.166	0.166	0.166	0.166	0.685	0.797	0.797	0.797	0.797
23.900	0.121	0.121	0.121	0.121	0.121	0.669	0.786	0.786	0.786	0.786
25.500	0.246	0.246	0.246	0.246	0.246	0.716	0.816	0.816	0.816	0.816
26.500	0.231	0.231	0.231	0.231	0.231	0.710	0.812	0.812	0.812	0.812
28.100	0.248	0.248	0.248	0.248	0.248	0.716	0.817	0.817	0.817	0.817
29.700	0.272	0.272	0.272	0.272	0.272	0.726	0.823	0.823	0.823	0.823
30.700	0.364	0.364	0.364	0.364	0.364	0.760	0.845	0.845	0.845	0.845
33.200	0.422	0.422	0.422	0.422	0.422	0.782	0.859	0.859	0.859	0.859
34.500	0.420	0.420	0.420	0.420	0.420	0.781	0.859	0.859	0.859	0.859
36.600	0.428	0.428	0.428	0.428	0.428	0.784	0.861	0.861	0.861	0.861
38.100	0.401	0.401	0.401	0.401	0.401	0.774	0.854	0.854	0.854	0.854
39.700	0.291	0.291	0.291	0.291	0.291	0.733	0.827	0.827	0.827	0.827
40.800	0.225	0.225	0.225	0.225	0.225	0.708	0.811	0.811	0.811	0.811
42.900	0.297	0.297	0.297	0.297	0.297	0.735	0.829	0.829	0.829	0.829
44.700	0.166	0.166	0.166	0.166	0.166	0.685	0.797	0.797	0.797	0.797
46.700	0.037	0.037	0.037	0.037	0.037	0.637	0.765	0.765	0.765	0.765
47.500	0.146	0.146	0.146	0.146	0.146	0.678	0.792	0.792	0.792	0.792
48.600	0.207	0.207	0.207	0.207	0.207	0.701	0.807	0.807	0.807	0.807
50.400	0.266	0.266	0.266	0.266	0.266	0.723	0.821	0.821	0.821	0.821
52.400	0.346	0.346	0.346	0.346	0.346	0.753	0.840	0.840	0.840	0.840
54.000	0.409	0.409	0.409	0.409	0.409	0.777	0.856	0.856	0.856	0.856
55.600	0.470	0.470	0.470	0.470	0.470	0.800	0.871	0.871	0.871	0.871
57.800	0.513	0.513	0.513	0.513	0.513	0.816	0.881	0.881	0.881	0.881
60.100	0.536	0.536	0.536	0.536	0.536	0.825	0.887	0.887	0.887	0.887
61.700	0.632	0.632	0.632	0.632	0.632	0.861	0.910	0.910	0.910	0.910
64.100	0.866	0.866	0.866	0.866	0.866	0.949	0.967	0.967	0.967	0.967
66.900	0.860	0.860	0.860	0.860	0.860	0.947	0.966	0.966	0.966	0.966
68.800	0.610	0.610	0.610	0.610	0.610	0.853	0.905	0.905	0.905	0.905
70.300	0.511	0.511	0.511	0.511	0.511	0.816	0.881	0.881	0.881	0.881
72.700	0.403	0.403	0.403	0.403	0.403	0.775	0.854	0.854	0.854	0.854
74.200	0.292	0.292	0.292	0.292	0.292	0.733	0.827	0.827	0.827	0.827
76.500	0.059	0.059	0.059	0.059	0.059	0.645	0.771	0.771	0.771	0.771
77.300	0.167	0.167	0.167	0.167	0.167	0.686	0.797	0.797	0.797	0.797
80.200	0.393	0.393	0.393	0.393	0.393	0.771	0.852	0.852	0.852	0.852
81.600	0.475	0.475	0.475	0.475	0.475	0.802	0.872	0.872	0.872	0.872
83.500	0.493	0.493	0.493	0.493	0.493	0.809	0.876	0.876	0.876	0.876
85.100	0.576	0.576	0.576	0.576	0.576	0.840	0.897	0.897	0.897	0.897
86.500	0.645	0.645	0.645	0.645	0.645	0.866	0.914	0.914	0.914	0.914
87.900	0.772	0.772	0.772	0.772	0.772	0.914	0.944	0.944	0.944	0.944
90.000	0.921	0.921	0.921	0.921	0.921	0.970	0.981	0.981	0.981	0.981
92.500	0.994	0.994	0.994	0.994	0.994	0.998	0.999	0.999	0.999	0.999
96.190	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

Appendix C

B-2/SHROUD 1 BUNDLE-AVERAGED GEOMETRY DATA,
MINIMUM RESTRICTION DEFINITION

The data in Appendix C represent the input from card groups 5 and 6 as required by ref. 2 for the B-2/shroud 1 COBRA-IV model based on the minimum restriction definition.



Table C.1. Shroud flow area reductions for the B-2/shroud I COBRA-IV model based on the minimum restriction definition

AXL ^a	FRACTIONAL SHROUD FLOW AREA
-47.676	1.000
0.000	1.003
1.800	0.940
3.400	0.872
5.000	0.877
6.900	0.906
8.900	0.959
11.500	0.957
13.300	0.897
15.100	0.859
16.800	0.821
18.100	0.802
19.500	0.772
21.400	0.785
23.200	0.783
25.000	0.790
26.900	0.806
28.500	0.795
30.000	0.798
32.000	0.803
34.000	0.810
35.900	0.788
37.700	0.758
39.500	0.779
41.200	0.742
43.300	0.735
44.700	0.745
46.200	0.748
47.700	0.736
49.700	0.756
51.600	0.766
53.500	0.733
54.900	0.720
56.200	0.692
57.600	0.736
59.800	0.815
61.800	0.866
63.800	0.942
66.500	0.944
68.400	0.864
70.100	0.801
71.600	0.781
73.100	0.756
74.600	0.696
76.200	0.663
78.000	0.673
79.500	0.696
81.600	0.715
83.800	0.801
86.000	0.849
88.100	0.909
89.900	0.982
91.500	0.993
95.987	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table C.2. AFACT values for the B-2/shroud 1 COBRA-IV model based on the minimum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001	1.003	1.005	1.005	1.005	1.003
1.800		0.974	0.960	0.960	0.960	0.974	0.946	0.906	0.906	0.906	0.946
3.400		0.945	0.916	0.916	0.916	0.945	0.885	0.801	0.801	0.801	0.885
5.000		0.947	0.919	0.919	0.919	0.947	0.889	0.808	0.808	0.808	0.889
6.900		0.960	0.938	0.938	0.938	0.960	0.915	0.853	0.853	0.853	0.915
8.800		0.982	0.973	0.973	0.973	0.982	0.963	0.936	0.936	0.936	0.963
11.500		0.982	0.972	0.972	0.972	0.982	0.961	0.933	0.933	0.933	0.961
13.300		0.956	0.933	0.933	0.933	0.956	0.908	0.840	0.840	0.840	0.908
15.100		0.940	0.907	0.907	0.907	0.940	0.873	0.780	0.780	0.780	0.873
16.800		0.924	0.882	0.882	0.882	0.924	0.839	0.721	0.721	0.721	0.839
18.100		0.915	0.870	0.870	0.870	0.915	0.821	0.691	0.691	0.691	0.821
19.500		0.902	0.850	0.850	0.850	0.902	0.794	0.644	0.644	0.644	0.794
21.400		0.908	0.858	0.858	0.858	0.908	0.805	0.664	0.664	0.664	0.805
23.200		0.907	0.857	0.857	0.857	0.907	0.805	0.662	0.662	0.662	0.805
25.000		0.910	0.862	0.862	0.862	0.910	0.811	0.673	0.673	0.673	0.811
26.900		0.917	0.872	0.872	0.872	0.917	0.825	0.697	0.697	0.697	0.825
28.500		0.913	0.865	0.865	0.865	0.913	0.816	0.691	0.681	0.681	0.816
30.000		0.914	0.867	0.867	0.867	0.914	0.818	0.685	0.685	0.685	0.818
32.000		0.916	0.871	0.871	0.871	0.916	0.823	0.693	0.693	0.693	0.823
34.000		0.919	0.875	0.875	0.875	0.919	0.829	0.703	0.703	0.703	0.829
35.900		0.909	0.860	0.860	0.860	0.909	0.809	0.669	0.669	0.669	0.809
37.700		0.897	0.841	0.841	0.841	0.897	0.782	0.623	0.623	0.623	0.782
39.500		0.928	0.855	0.855	0.855	0.906	0.847	0.656	0.656	0.656	0.847
41.200		0.916	0.830	0.830	0.830	0.890	0.821	0.597	0.597	0.597	0.821
43.300		0.887	0.826	0.826	0.826	0.887	0.761	0.587	0.587	0.587	0.761
44.700		0.931	0.833	0.833	0.833	0.891	0.864	0.603	0.603	0.603	0.864
46.200		0.931	0.834	0.834	0.834	0.892	0.865	0.607	0.607	0.607	0.865
47.700		0.928	0.826	0.826	0.826	0.887	0.859	0.588	0.588	0.588	0.859
49.700		0.896	0.839	0.839	0.839	0.896	0.780	0.619	0.619	0.619	0.780
51.600		0.900	0.846	0.846	0.846	0.900	0.789	0.635	0.635	0.635	0.789
53.500		0.886	0.824	0.824	0.824	0.886	0.759	0.583	0.583	0.583	0.759
54.900		0.881	0.816	0.816	0.816	0.881	0.748	0.564	0.564	0.564	0.748
56.200		0.869	0.798	0.798	0.798	0.869	0.723	0.520	0.520	0.520	0.723
57.600		0.887	0.826	0.826	0.826	0.887	0.762	0.588	0.588	0.588	0.762
59.800		0.921	0.878	0.878	0.878	0.921	0.833	0.711	0.711	0.711	0.833
61.800		0.943	0.912	0.912	0.912	0.943	0.879	0.791	0.791	0.791	0.879
63.800		0.975	0.962	0.962	0.962	0.975	0.947	0.909	0.909	0.909	0.947
66.500		0.976	0.963	0.963	0.963	0.976	0.950	0.913	0.913	0.913	0.950
68.400		0.942	0.911	0.911	0.911	0.942	0.877	0.788	0.788	0.788	0.877
70.100		0.915	0.869	0.869	0.869	0.915	0.821	0.690	0.690	0.690	0.821
71.600		0.906	0.856	0.856	0.856	0.906	0.802	0.658	0.658	0.658	0.802
73.100		0.896	0.840	0.840	0.840	0.896	0.780	0.620	0.620	0.620	0.780
74.600		0.870	0.800	0.800	0.800	0.915	0.726	0.526	0.526	0.526	0.800
76.200		0.856	0.779	0.779	0.779	0.905	0.697	0.475	0.475	0.475	0.800
78.000		0.860	0.785	0.785	0.785	0.908	0.705	0.490	0.490	0.490	0.819
79.500		0.870	0.800	0.800	0.800	0.870	0.726	0.526	0.526	0.526	0.821
81.600		0.878	0.813	0.813	0.813	0.878	0.743	0.556	0.556	0.556	0.743
83.800		0.915	0.869	0.869	0.869	0.915	0.821	0.690	0.690	0.690	0.821
86.000		0.935	0.900	0.900	0.900	0.935	0.864	0.764	0.764	0.764	0.864
88.100		0.961	0.940	0.940	0.940	0.961	0.918	0.858	0.858	0.858	0.918
89.900		0.992	0.988	0.988	0.988	0.992	0.984	0.972	0.972	0.972	0.984
91.500		0.997	0.995	0.995	0.995	0.997	0.994	0.989	0.989	0.989	0.994
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table C.2 (continued)

AXL ²	VALUE OF I IS:	11	12	13	14	15	16	17	18	19	20
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.003	1.005	1.005	1.005	1.003	1.003	1.005	1.005	1.005	1.003
1.800		0.946	0.906	0.906	0.906	0.946	0.946	0.906	0.906	0.906	0.946
3.400		0.885	0.801	0.801	0.801	0.885	0.885	0.801	0.801	0.801	0.885
5.000		0.889	0.808	0.808	0.808	0.889	0.889	0.808	0.808	0.808	0.889
6.900		0.915	0.853	0.853	0.853	0.915	0.915	0.853	0.853	0.853	0.915
8.800		0.963	0.936	0.936	0.936	0.963	0.963	0.936	0.936	0.936	0.963
11.500		0.961	0.933	0.933	0.933	0.961	0.961	0.933	0.933	0.933	0.961
13.300		0.908	0.840	0.840	0.840	0.908	0.908	0.840	0.840	0.840	0.908
15.100		0.873	0.780	0.780	0.780	0.873	0.873	0.780	0.780	0.780	0.873
16.800		0.839	0.721	0.721	0.721	0.839	0.839	0.721	0.721	0.721	0.839
18.100		0.821	0.691	0.691	0.691	0.821	0.821	0.691	0.691	0.691	0.821
19.500		0.794	0.644	0.644	0.644	0.794	0.794	0.644	0.644	0.644	0.794
21.400		0.806	0.664	0.664	0.664	0.806	0.806	0.664	0.664	0.664	0.806
23.200		0.805	0.662	0.662	0.662	0.805	0.805	0.662	0.662	0.662	0.805
25.000		0.811	0.673	0.673	0.673	0.811	0.811	0.673	0.673	0.673	0.811
26.900		0.825	0.697	0.697	0.697	0.825	0.825	0.697	0.697	0.697	0.825
28.500		0.816	0.681	0.681	0.681	0.816	0.816	0.681	0.681	0.681	0.816
30.000		0.818	0.685	0.685	0.685	0.818	0.818	0.685	0.685	0.685	0.818
32.000		0.823	0.693	0.693	0.693	0.823	0.823	0.693	0.693	0.693	0.823
34.000		0.828	0.703	0.703	0.703	0.828	0.828	0.703	0.703	0.703	0.828
35.900		0.809	0.669	0.669	0.669	0.809	0.809	0.669	0.669	0.669	0.809
37.700		0.782	0.623	0.623	0.623	0.782	0.782	0.623	0.623	0.623	0.782
39.500		0.801	0.656	0.656	0.656	0.801	0.801	0.656	0.656	0.656	0.801
41.200		0.767	0.597	0.597	0.597	0.767	0.767	0.597	0.597	0.597	0.767
43.300		0.761	0.587	0.587	0.587	0.761	0.761	0.587	0.587	0.587	0.761
44.700		0.822	0.603	0.603	0.603	0.771	0.771	0.603	0.603	0.603	0.771
46.200		0.823	0.607	0.607	0.607	0.773	0.773	0.607	0.607	0.607	0.773
47.700		0.815	0.588	0.588	0.588	0.762	0.762	0.588	0.588	0.588	0.762
49.700		0.780	0.619	0.619	0.619	0.780	0.780	0.619	0.619	0.619	0.780
51.600		0.789	0.635	0.635	0.635	0.789	0.789	0.635	0.635	0.635	0.789
53.500		0.759	0.583	0.583	0.583	0.759	0.759	0.583	0.583	0.583	0.759
54.900		0.748	0.564	0.564	0.564	0.748	0.748	0.564	0.564	0.564	0.748
56.200		0.723	0.520	0.520	0.520	0.723	0.723	0.520	0.520	0.520	0.723
57.600		0.762	0.588	0.588	0.588	0.762	0.762	0.588	0.588	0.588	0.762
59.800		0.833	0.711	0.711	0.711	0.833	0.833	0.711	0.711	0.711	0.833
61.800		0.879	0.791	0.791	0.791	0.879	0.879	0.791	0.791	0.791	0.879
63.800		0.947	0.909	0.909	0.909	0.947	0.947	0.909	0.909	0.909	0.947
66.500		0.950	0.913	0.913	0.913	0.950	0.950	0.913	0.913	0.913	0.950
68.400		0.877	0.788	0.788	0.788	0.877	0.877	0.788	0.788	0.788	0.877
70.100		0.821	0.690	0.690	0.690	0.821	0.821	0.690	0.690	0.690	0.821
71.600		0.802	0.658	0.658	0.658	0.802	0.802	0.658	0.658	0.658	0.802
73.100		0.780	0.620	0.620	0.620	0.780	0.780	0.620	0.620	0.620	0.780
74.600		0.726	0.526	0.526	0.526	0.831	0.726	0.526	0.526	0.526	0.832
76.200		0.697	0.475	0.475	0.475	0.813	0.697	0.475	0.475	0.475	0.814
78.000		0.705	0.490	0.490	0.490	0.819	0.705	0.490	0.490	0.490	0.819
79.500		0.726	0.526	0.526	0.526	0.726	0.726	0.526	0.526	0.526	0.726
81.600		0.743	0.556	0.556	0.556	0.743	0.743	0.556	0.556	0.556	0.743
83.800		0.821	0.690	0.690	0.690	0.821	0.821	0.690	0.690	0.690	0.821
86.000		0.864	0.764	0.764	0.764	0.864	0.864	0.764	0.764	0.764	0.864
88.100		0.918	0.858	0.858	0.858	0.918	0.918	0.858	0.858	0.858	0.918
89.900		0.984	0.972	0.972	0.972	0.984	0.984	0.972	0.972	0.972	0.984
91.500		0.994	0.989	0.989	0.989	0.994	0.994	0.989	0.989	0.989	0.994
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table C.2 (continued)

AXL ^a	VALUE OF I IS: 21	22	23	24	25
-47.676	1.000	1.000	1.000	1.000	1.000
0.000	1.001	1.002	1.002	1.002	1.001
1.800	0.974	0.960	0.960	0.960	0.974
3.400	0.945	0.916	0.916	0.916	0.945
5.000	0.947	0.919	0.919	0.919	0.947
6.900	0.960	0.938	0.938	0.938	0.960
8.800	0.982	0.973	0.973	0.973	0.982
11.500	0.982	0.972	0.972	0.972	0.982
13.300	0.956	0.933	0.933	0.933	0.956
15.100	0.940	0.907	0.907	0.907	0.940
16.800	0.924	0.882	0.882	0.882	0.924
18.100	0.915	0.870	0.870	0.870	0.915
19.500	0.902	0.850	0.850	0.850	0.902
21.400	0.908	0.858	0.858	0.858	0.908
23.200	0.907	0.857	0.857	0.857	0.907
25.000	0.910	0.862	0.862	0.862	0.910
26.900	0.917	0.872	0.872	0.872	0.917
28.500	0.913	0.865	0.865	0.865	0.913
30.000	0.914	0.867	0.867	0.867	0.914
32.000	0.916	0.871	0.871	0.871	0.916
34.000	0.919	0.875	0.875	0.875	0.919
35.900	0.909	0.860	0.860	0.860	0.909
37.700	0.897	0.841	0.841	0.841	0.897
39.500	0.906	0.855	0.855	0.855	0.906
41.200	0.890	0.830	0.830	0.830	0.890
43.300	0.887	0.826	0.826	0.826	0.887
44.700	0.891	0.833	0.833	0.833	0.891
46.200	0.892	0.834	0.834	0.834	0.892
47.700	0.887	0.826	0.826	0.826	0.887
49.700	0.896	0.839	0.839	0.839	0.896
51.600	0.900	0.846	0.846	0.846	0.900
53.500	0.886	0.824	0.824	0.824	0.886
54.900	0.881	0.816	0.816	0.816	0.881
56.200	0.869	0.798	0.798	0.798	0.869
57.600	0.887	0.826	0.826	0.826	0.887
59.800	0.921	0.878	0.878	0.878	0.921
61.800	0.943	0.912	0.912	0.912	0.943
63.800	0.975	0.962	0.962	0.962	0.975
66.500	0.976	0.963	0.963	0.963	0.976
68.400	0.942	0.911	0.911	0.911	0.942
70.100	0.915	0.869	0.869	0.869	0.915
71.600	0.906	0.856	0.856	0.856	0.906
73.100	0.896	0.840	0.840	0.840	0.896
74.600	0.870	0.800	0.800	0.800	0.870
76.200	0.856	0.779	0.779	0.779	0.856
78.000	0.860	0.785	0.785	0.785	0.860
79.500	0.870	0.800	0.800	0.800	0.870
81.600	0.878	0.813	0.813	0.813	0.878
83.800	0.915	0.869	0.869	0.869	0.915
86.000	0.935	0.900	0.900	0.900	0.935
88.100	0.961	0.940	0.940	0.940	0.961
89.900	0.992	0.988	0.988	0.988	0.992
91.500	0.997	0.995	0.995	0.995	0.997
95.987	1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table C.3. GFACT values for the B-2/shroud 1 COBRA-IV model based on the minimum restriction definition

GAPXL ^a	VALUE	CF	K	IS:	1	2	3	4	5	6	7	8	9	10
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.002	1.004	1.002	1.010	1.002	1.010	1.002	1.010	1.002	1.010	1.002	1.010	1.004	1.004
1.800	0.958	0.934	0.958	0.826	0.958	0.826	0.958	0.826	0.958	0.826	0.958	0.826	0.934	0.934
3.400	0.913	0.865	0.913	0.642	0.913	0.642	0.913	0.642	0.913	0.642	0.913	0.642	0.865	0.865
5.000	0.916	0.869	0.916	0.654	0.916	0.654	0.916	0.654	0.916	0.654	0.916	0.654	0.869	0.869
6.900	0.935	0.899	0.935	0.732	0.935	0.732	0.935	0.732	0.935	0.732	0.935	0.732	0.899	0.899
8.800	0.971	0.955	0.971	0.880	0.971	0.880	0.971	0.880	0.971	0.880	0.971	0.880	0.955	0.955
11.500	0.970	0.953	0.970	0.875	0.970	0.875	0.970	0.875	0.970	0.875	0.970	0.875	0.953	0.953
13.300	0.929	0.890	0.929	0.709	0.929	0.709	0.929	0.709	0.929	0.709	0.929	0.709	0.890	0.890
15.100	0.904	0.852	0.904	0.606	0.904	0.606	0.904	0.606	0.904	0.606	0.904	0.606	0.852	0.852
16.800	0.880	0.814	0.880	0.508	0.880	0.508	0.880	0.508	0.880	0.508	0.880	0.508	0.814	0.814
18.100	0.868	0.796	0.868	0.459	0.868	0.459	0.868	0.459	0.868	0.459	0.868	0.459	0.796	0.796
19.500	0.850	0.768	0.850	0.383	0.850	0.383	0.850	0.383	0.850	0.383	0.850	0.383	0.768	0.768
21.400	0.857	0.780	0.857	0.415	0.857	0.415	0.857	0.415	0.857	0.415	0.857	0.415	0.780	0.780
23.200	0.857	0.778	0.857	0.412	0.857	0.412	0.857	0.412	0.857	0.412	0.857	0.412	0.778	0.778
25.000	0.861	0.785	0.861	0.430	0.861	0.430	0.861	0.430	0.861	0.430	0.861	0.430	0.785	0.785
26.900	0.870	0.800	0.870	0.469	0.870	0.469	0.870	0.469	0.870	0.469	0.870	0.469	0.800	0.800
28.500	0.864	0.790	0.864	0.443	0.864	0.443	0.864	0.443	0.864	0.443	0.864	0.443	0.790	0.790
30.000	0.866	0.792	0.866	0.449	0.866	0.449	0.866	0.449	0.866	0.449	0.866	0.449	0.792	0.792
32.000	0.869	0.797	0.869	0.462	0.869	0.462	0.869	0.462	0.869	0.462	0.869	0.462	0.797	0.797
34.000	0.873	0.803	0.873	0.478	0.873	0.478	0.873	0.478	0.873	0.478	0.873	0.478	0.803	0.803
35.900	0.859	0.783	0.859	0.423	0.859	0.423	0.859	0.423	0.859	0.423	0.859	0.423	0.783	0.783
37.700	0.842	0.755	0.842	0.350	0.842	0.350	0.842	0.350	0.842	0.350	0.842	0.350	0.755	0.755
39.500	0.854	0.775	0.854	0.403	0.854	0.403	0.854	0.403	0.854	0.403	0.854	0.403	0.775	0.775
41.200	0.832	0.740	0.832	0.310	0.832	0.310	0.832	0.310	0.832	0.310	0.832	0.310	0.740	0.740
43.300	0.828	0.734	0.828	0.294	0.828	0.294	0.828	0.294	0.828	0.294	0.828	0.294	0.734	0.734
44.700	0.834	0.743	0.834	0.319	0.834	0.319	0.834	0.319	0.834	0.319	0.834	0.319	0.743	0.743
46.200	0.835	0.746	0.835	0.325	0.835	0.325	0.835	0.325	0.835	0.325	0.835	0.325	0.746	0.746
47.700	0.828	0.734	0.828	0.296	0.828	0.296	0.828	0.296	0.828	0.296	0.828	0.296	0.734	0.734
49.700	0.840	0.753	0.840	0.344	0.840	0.344	0.840	0.344	0.840	0.344	0.840	0.344	0.753	0.753
51.600	0.846	0.762	0.846	0.369	0.846	0.369	0.846	0.369	0.846	0.369	0.846	0.369	0.762	0.762
53.500	0.826	0.731	0.826	0.288	0.826	0.288	0.826	0.288	0.826	0.288	0.826	0.288	0.731	0.731
54.900	0.819	0.720	0.819	0.258	0.819	0.258	0.819	0.258	0.819	0.258	0.819	0.258	0.720	0.720
56.200	0.803	0.695	0.803	0.191	0.803	0.191	0.803	0.191	0.803	0.191	0.803	0.191	0.695	0.695
57.600	0.828	0.734	0.828	0.296	0.828	0.296	0.828	0.296	0.828	0.296	0.828	0.296	0.734	0.734
59.800	0.876	0.808	0.876	0.491	0.876	0.491	0.876	0.491	0.876	0.491	0.876	0.491	0.808	0.808
61.800	0.909	0.859	0.909	0.625	0.909	0.625	0.909	0.625	0.909	0.625	0.909	0.625	0.859	0.859
63.800	0.959	0.936	0.959	0.831	0.959	0.831	0.959	0.831	0.959	0.831	0.959	0.831	0.936	0.936
66.500	0.961	0.939	0.961	0.839	0.961	0.839	0.961	0.839	0.961	0.839	0.961	0.839	0.939	0.939
68.400	0.907	0.857	0.907	0.620	0.907	0.620	0.907	0.620	0.907	0.620	0.907	0.620	0.857	0.857
70.100	0.868	0.795	0.868	0.457	0.868	0.457	0.868	0.457	0.868	0.457	0.868	0.457	0.795	0.795
71.600	0.855	0.776	0.855	0.406	0.855	0.406	0.855	0.406	0.855	0.406	0.855	0.406	0.776	0.776
73.100	0.840	0.753	0.840	0.346	0.840	0.346	0.840	0.346	0.840	0.346	0.840	0.346	0.753	0.753
74.600	0.805	0.699	0.805	0.200	0.805	0.200	0.805	0.200	0.805	0.200	0.805	0.200	0.699	0.699
76.200	0.786	0.670	0.786	0.124	0.786	0.124	0.786	0.124	0.786	0.124	0.786	0.124	0.670	0.670
78.000	0.792	0.678	0.792	0.146	0.792	0.146	0.792	0.146	0.792	0.146	0.792	0.146	0.678	0.678
79.500	0.805	0.699	0.805	0.200	0.805	0.200	0.805	0.200	0.805	0.200	0.805	0.200	0.699	0.699
81.600	0.816	0.716	0.816	0.246	0.816	0.246	0.816	0.246	0.816	0.246	0.816	0.246	0.716	0.716
83.800	0.868	0.795	0.868	0.457	0.868	0.457	0.868	0.457	0.868	0.457	0.868	0.457	0.795	0.795
86.000	0.897	0.841	0.897	0.579	0.897	0.579	0.897	0.579	0.897	0.579	0.897	0.579	0.841	0.841
88.100	0.937	0.902	0.937	0.741	0.937	0.741	0.937	0.741	0.937	0.741	0.937	0.741	0.902	0.902
89.900	0.987	0.980	0.987	0.947	0.987	0.947	0.987	0.947	0.987	0.947	0.987	0.947	0.980	0.980
91.500	0.995	0.992	0.995	0.979	0.995	0.979	0.995	0.979	0.995	0.979	0.995	0.979	0.992	0.992
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table C.3 (continued)

GAPXL ^a	VALUE OF K IS: 11	12	13	14	15	16	17	18	19	20
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.010	1.010	1.004	1.004	1.000
1.800	0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826
3.400	0.642	0.642	0.642	0.642	0.642	0.642	0.642	0.865	0.865	0.642
5.000	0.654	0.654	0.654	0.654	0.654	0.654	0.654	0.869	0.869	0.654
6.900	0.732	0.732	0.732	0.732	0.732	0.732	0.732	0.899	0.899	0.732
8.800	0.880	0.880	0.880	0.880	0.880	0.880	0.880	0.955	0.955	0.880
11.500	0.875	0.875	0.875	0.875	0.875	0.875	0.875	0.953	0.953	0.875
13.300	0.709	0.709	0.709	0.709	0.709	0.709	0.709	0.850	0.850	0.709
15.100	0.606	0.606	0.606	0.606	0.606	0.606	0.606	0.852	0.852	0.606
16.800	0.508	0.508	0.508	0.508	0.508	0.508	0.508	0.814	0.814	0.508
18.100	0.459	0.459	0.459	0.459	0.459	0.459	0.459	0.796	0.796	0.459
19.500	0.383	0.383	0.383	0.383	0.383	0.383	0.383	0.768	0.768	0.383
21.400	0.415	0.415	0.415	0.415	0.415	0.415	0.415	0.780	0.780	0.415
23.200	0.412	0.412	0.412	0.412	0.412	0.412	0.412	0.778	0.778	0.412
25.000	0.430	0.430	0.430	0.430	0.430	0.430	0.430	0.785	0.785	0.430
26.900	0.469	0.469	0.469	0.469	0.469	0.469	0.469	0.800	0.800	0.469
28.500	0.443	0.443	0.443	0.443	0.443	0.443	0.443	0.790	0.790	0.443
30.000	0.449	0.449	0.449	0.449	0.449	0.449	0.449	0.792	0.792	0.449
32.000	0.462	0.462	0.462	0.462	0.462	0.462	0.462	0.797	0.797	0.462
34.000	0.478	0.478	0.478	0.478	0.478	0.478	0.478	0.803	0.803	0.478
35.900	0.423	0.423	0.423	0.423	0.423	0.423	0.423	0.783	0.783	0.423
37.700	0.350	0.350	0.350	0.350	0.350	0.350	0.350	0.755	0.755	0.350
39.500	0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403
41.200	0.310	0.310	0.310	0.310	0.310	0.310	0.310	0.740	0.740	0.310
43.300	0.294	0.294	0.294	0.294	0.294	0.294	0.294	0.734	0.734	0.294
44.700	0.319	0.319	0.319	0.319	0.319	0.319	0.319	0.743	0.743	0.319
46.200	0.325	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325
47.700	0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296
49.700	0.344	0.344	0.344	0.344	0.344	0.344	0.344	0.753	0.753	0.344
51.600	0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369
53.500	0.288	0.288	0.288	0.288	0.288	0.288	0.288	0.731	0.731	0.288
54.900	0.258	0.258	0.258	0.258	0.258	0.258	0.258	0.720	0.720	0.258
56.200	0.191	0.191	0.191	0.191	0.191	0.191	0.191	0.695	0.695	0.191
57.600	0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296
59.800	0.491	0.491	0.491	0.491	0.491	0.491	0.491	0.808	0.808	0.491
61.800	0.625	0.625	0.625	0.625	0.625	0.625	0.625	0.859	0.859	0.625
63.800	0.831	0.831	0.831	0.831	0.831	0.831	0.831	0.936	0.936	0.831
66.500	0.839	0.839	0.839	0.839	0.839	0.839	0.839	0.939	0.939	0.839
68.400	0.620	0.620	0.620	0.620	0.620	0.620	0.620	0.857	0.857	0.620
70.100	0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457
71.600	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.776	0.776	0.406
73.100	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346
74.600	0.200	0.200	0.200	0.200	0.200	0.200	0.200	0.699	0.699	0.200
76.200	0.124	0.124	0.124	0.124	0.124	0.124	0.124	0.670	0.670	0.124
78.000	0.146	0.146	0.146	0.146	0.146	0.146	0.146	0.678	0.678	0.146
79.500	0.200	0.200	0.200	0.200	0.200	0.200	0.200	0.699	0.699	0.200
81.600	0.246	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246
83.800	0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457
86.000	0.579	0.579	0.579	0.579	0.579	0.579	0.579	0.841	0.841	0.579
88.100	0.741	0.741	0.741	0.741	0.741	0.741	0.741	0.902	0.902	0.741
89.900	0.947	0.947	0.947	0.947	0.947	0.947	0.947	0.980	0.980	0.947
91.500	0.979	0.979	0.979	0.979	0.979	0.979	0.979	0.992	0.992	0.979
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table C.3 (continued)

GAPXL ^a	VALUE OF K IS: 21	22	23	24	25	26	27	28	29	30
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.010	1.004	1.004	1.010	1.010
1.800	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826	0.826
3.400	0.642	0.642	0.642	0.642	0.642	0.642	0.865	0.865	0.642	0.642
5.000	0.654	0.654	0.654	0.654	0.654	0.654	0.869	0.869	0.654	0.654
6.900	0.732	0.732	0.732	0.732	0.732	0.732	0.899	0.899	0.732	0.732
8.800	0.880	0.880	0.880	0.880	0.880	0.880	0.955	0.955	0.880	0.880
11.500	0.875	0.875	0.875	0.875	0.875	0.875	0.953	0.953	0.875	0.875
13.300	0.709	0.709	0.709	0.709	0.709	0.709	0.890	0.890	0.709	0.709
15.100	0.606	0.606	0.606	0.606	0.606	0.606	0.852	0.852	0.606	0.606
16.800	0.508	0.508	0.508	0.508	0.508	0.508	0.814	0.814	0.508	0.508
18.100	0.459	0.459	0.459	0.459	0.459	0.459	0.796	0.796	0.459	0.459
19.500	0.383	0.383	0.383	0.383	0.383	0.383	0.768	0.768	0.383	0.383
21.400	0.415	0.415	0.415	0.415	0.415	0.415	0.780	0.780	0.415	0.415
23.200	0.412	0.412	0.412	0.412	0.412	0.412	0.778	0.778	0.412	0.412
25.000	0.430	0.430	0.430	0.430	0.430	0.430	0.785	0.785	0.430	0.430
26.900	0.469	0.469	0.469	0.469	0.469	0.469	0.800	0.800	0.469	0.469
28.500	0.443	0.443	0.443	0.443	0.443	0.443	0.790	0.790	0.443	0.443
30.000	0.449	0.449	0.449	0.449	0.449	0.449	0.792	0.792	0.449	0.449
32.000	0.462	0.462	0.462	0.462	0.462	0.462	0.797	0.797	0.462	0.462
34.000	0.478	0.478	0.478	0.478	0.478	0.478	0.803	0.803	0.478	0.478
35.900	0.423	0.423	0.423	0.423	0.423	0.423	0.783	0.783	0.423	0.423
37.700	0.350	0.350	0.350	0.350	0.350	0.350	0.755	0.755	0.350	0.350
39.500	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403	0.403
41.200	0.310	0.310	0.310	0.310	0.310	0.310	0.740	0.740	0.310	0.310
43.300	0.294	0.294	0.294	0.294	0.294	0.294	0.734	0.734	0.294	0.294
44.700	0.319	0.319	0.319	0.319	0.319	0.319	0.743	0.743	0.319	0.319
46.200	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325	0.325
47.700	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296	0.296
49.700	0.344	0.344	0.344	0.344	0.344	0.344	0.753	0.753	0.344	0.344
51.600	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369	0.369
53.500	0.288	0.288	0.288	0.288	0.288	0.288	0.731	0.731	0.288	0.288
54.900	0.258	0.258	0.258	0.258	0.258	0.258	0.720	0.720	0.258	0.258
56.200	0.191	0.191	0.191	0.191	0.191	0.191	0.695	0.695	0.191	0.191
57.600	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296	0.296
59.800	0.491	0.491	0.491	0.491	0.491	0.491	0.808	0.808	0.491	0.491
61.800	0.625	0.625	0.625	0.625	0.625	0.625	0.859	0.859	0.625	0.625
63.800	0.831	0.831	0.831	0.831	0.831	0.831	0.936	0.936	0.831	0.831
66.500	0.839	0.839	0.839	0.839	0.839	0.839	0.939	0.939	0.839	0.839
68.400	0.620	0.620	0.620	0.620	0.620	0.620	0.857	0.857	0.620	0.620
70.100	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457	0.457
71.600	0.406	0.406	0.406	0.406	0.406	0.406	0.776	0.776	0.406	0.406
73.100	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346	0.346
74.600	0.200	0.200	0.200	0.200	0.200	0.200	0.699	0.699	0.200	0.200
76.200	0.124	0.124	0.124	0.124	0.124	0.124	0.670	0.670	0.124	0.124
78.000	0.146	0.146	0.146	0.146	0.146	0.146	0.678	0.678	0.146	0.146
79.500	0.200	0.200	0.200	0.200	0.200	0.200	0.699	0.699	0.200	0.200
81.600	0.246	0.246	0.246	0.246	0.246	0.246	0.716	0.716	0.246	0.246
83.800	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457	0.457
86.000	0.579	0.579	0.579	0.579	0.579	0.579	0.841	0.841	0.579	0.579
88.100	0.741	0.741	0.741	0.741	0.741	0.741	0.902	0.902	0.741	0.741
89.900	0.947	0.947	0.947	0.947	0.947	0.947	0.980	0.980	0.947	0.947
91.500	0.979	0.979	0.979	0.979	0.979	0.979	0.992	0.992	0.979	0.979
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table C.3 (continued)

GAPXL ^a	VALUE OF K IS:	31	32	33	34	35	36	37	38	39	40
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.004	1.002	1.002	1.002	1.002
1.800		0.826	0.826	0.826	0.826	0.826	0.934	0.958	0.958	0.958	0.958
3.400		0.642	0.642	0.642	0.642	0.642	0.865	0.913	0.913	0.913	0.913
5.000		0.654	0.654	0.654	0.654	0.654	0.869	0.916	0.916	0.916	0.916
6.900		0.732	0.732	0.732	0.732	0.732	0.899	0.935	0.935	0.935	0.935
8.800		0.880	0.880	0.880	0.880	0.880	0.955	0.971	0.971	0.971	0.971
11.500		0.875	0.875	0.875	0.875	0.875	0.953	0.970	0.970	0.970	0.970
13.300		0.709	0.709	0.709	0.709	0.709	0.890	0.929	0.929	0.929	0.929
15.100		0.606	0.606	0.606	0.606	0.606	0.852	0.904	0.904	0.904	0.904
16.800		0.508	0.508	0.508	0.508	0.508	0.814	0.880	0.880	0.880	0.880
18.100		0.459	0.459	0.459	0.459	0.459	0.796	0.868	0.868	0.868	0.868
19.500		0.383	0.383	0.383	0.383	0.383	0.768	0.850	0.850	0.850	0.850
21.400		0.415	0.415	0.415	0.415	0.415	0.780	0.857	0.857	0.857	0.857
23.200		0.412	0.412	0.412	0.412	0.412	0.773	0.857	0.857	0.857	0.857
25.000		0.430	0.430	0.430	0.430	0.430	0.785	0.861	0.861	0.861	0.861
26.900		0.469	0.469	0.469	0.469	0.469	0.900	0.870	0.870	0.870	0.870
28.500		0.443	0.443	0.443	0.443	0.443	0.790	0.864	0.864	0.864	0.864
30.000		0.449	0.449	0.449	0.449	0.449	0.792	0.866	0.866	0.866	0.866
32.000		0.462	0.462	0.462	0.462	0.462	0.797	0.869	0.869	0.869	0.869
34.000		0.478	0.478	0.478	0.478	0.478	0.803	0.873	0.873	0.873	0.873
35.900		0.423	0.423	0.423	0.423	0.423	0.783	0.859	0.859	0.859	0.859
37.700		0.350	0.350	0.350	0.350	0.350	0.755	0.842	0.842	0.842	0.842
39.500		0.403	0.403	0.403	0.403	0.403	0.775	0.854	0.854	0.854	0.854
41.200		0.310	0.310	0.310	0.310	0.310	0.740	0.832	0.832	0.832	0.832
43.300		0.294	0.294	0.294	0.294	0.294	0.734	0.828	0.828	0.828	0.828
44.700		0.319	0.319	0.319	0.319	0.319	0.743	0.834	0.834	0.834	0.834
46.200		0.325	0.325	0.325	0.325	0.325	0.746	0.835	0.835	0.835	0.835
47.700		0.296	0.296	0.296	0.296	0.296	0.734	0.828	0.828	0.828	0.828
49.700		0.344	0.344	0.344	0.344	0.344	0.753	0.840	0.840	0.840	0.840
51.600		0.369	0.369	0.369	0.369	0.369	0.762	0.846	0.846	0.846	0.846
53.500		0.288	0.288	0.288	0.288	0.288	0.731	0.826	0.826	0.826	0.826
54.900		0.258	0.258	0.258	0.258	0.258	0.720	0.819	0.819	0.819	0.819
56.200		0.191	0.191	0.191	0.191	0.191	0.695	0.803	0.803	0.803	0.803
57.600		0.296	0.296	0.296	0.296	0.296	0.734	0.828	0.828	0.828	0.828
59.800		0.491	0.491	0.491	0.491	0.491	0.808	0.876	0.876	0.876	0.876
61.800		0.625	0.625	0.625	0.625	0.625	0.859	0.909	0.909	0.909	0.909
63.800		0.831	0.831	0.831	0.831	0.831	0.936	0.959	0.959	0.959	0.959
66.500		0.839	0.839	0.839	0.839	0.839	0.939	0.961	0.961	0.961	0.961
68.400		0.620	0.620	0.620	0.620	0.620	0.857	0.907	0.907	0.907	0.907
70.100		0.457	0.457	0.457	0.457	0.457	0.795	0.868	0.868	0.868	0.868
71.600		0.406	0.406	0.406	0.406	0.406	0.776	0.855	0.855	0.855	0.855
73.100		0.346	0.346	0.346	0.346	0.346	0.753	0.840	0.840	0.840	0.840
74.600		0.200	0.200	0.200	0.200	0.200	0.699	0.805	0.805	0.805	0.805
76.200		0.124	0.124	0.124	0.124	0.124	0.670	0.786	0.786	0.786	0.786
78.000		0.146	0.146	0.146	0.146	0.146	0.678	0.792	0.792	0.792	0.792
79.500		0.200	0.200	0.200	0.200	0.200	0.699	0.805	0.805	0.805	0.805
81.600		0.246	0.246	0.246	0.246	0.246	0.716	0.816	0.816	0.816	0.816
83.800		0.457	0.457	0.457	0.457	0.457	0.795	0.868	0.868	0.868	0.868
86.000		0.579	0.579	0.579	0.579	0.579	0.841	0.897	0.897	0.897	0.897
88.100		0.741	0.741	0.741	0.741	0.741	0.902	0.937	0.937	0.937	0.937
89.900		0.947	0.947	0.947	0.947	0.947	0.980	0.987	0.987	0.987	0.987
91.500		0.979	0.979	0.979	0.979	0.979	0.992	0.995	0.995	0.995	0.995
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

Appendix D

B-2/SHROUD 1 BUNDLE-AVERAGED GEOMETRY DATA,
MAXIMUM RESTRICTION DEFINITION

The data in Appendix D represent the input from card groups 5 and 6 as required by ref. 2 for the B-2/shroud 1 COBRA-IV model based on the maximum restriction definition.

Table D.1. Shroud flow area reductions for the B-2/shroud 1 COBRA-IV model based on the maximum restriction definition

AXL ^a	FRACTIONAL SHROUD FLOW AREA
~47.676	1.000
0.000	1.003
1.800	0.940
3.400	0.872
5.000	0.877
6.900	0.906
8.800	0.959
11.500	0.957
13.300	0.897
15.100	0.859
16.800	0.821
18.100	0.802
19.500	0.756
21.400	0.785
23.200	0.783
25.000	0.790
26.900	0.806
28.500	0.795
30.000	0.798
32.000	0.803
34.000	0.810
35.900	0.788
37.700	0.752
39.500	0.779
41.200	0.718
43.300	0.706
44.700	0.745
46.200	0.748
47.700	0.724
49.700	0.756
51.600	0.766
53.500	0.701
54.900	0.720
56.200	0.640
57.600	0.736
59.800	0.815
61.800	0.866
63.800	0.942
66.500	0.944
68.400	0.864
70.100	0.801
71.600	0.781
73.100	0.756
74.600	0.686
76.200	0.592
78.000	0.657
79.500	0.690
81.600	0.676
83.800	0.801
86.000	0.849
88.100	0.909
89.900	0.982
91.500	0.993
95.987	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table D.2. AFACT values for the B-2/shroud 1 COBRA-IV model based on the maximum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001	1.003	1.005	1.005	1.005	1.003
1.800		0.974	0.960	0.960	0.960	0.974	0.946	0.906	0.906	0.906	0.946
3.400		0.945	0.916	0.916	0.916	0.945	0.885	0.801	0.801	0.801	0.885
5.000		0.947	0.919	0.919	0.919	0.947	0.889	0.808	0.808	0.808	0.889
6.900		0.960	0.938	0.938	0.938	0.960	0.915	0.853	0.853	0.853	0.915
8.800		0.982	0.973	0.973	0.973	0.982	0.963	0.936	0.936	0.936	0.963
11.500		0.982	0.972	0.972	0.972	0.982	0.961	0.933	0.933	0.933	0.961
13.300		0.956	0.933	0.933	0.933	0.956	0.908	0.840	0.840	0.840	0.908
15.100		0.940	0.907	0.907	0.907	0.940	0.873	0.780	0.780	0.780	0.873
16.800		0.924	0.882	0.882	0.882	0.924	0.839	0.721	0.721	0.721	0.839
18.100		0.915	0.870	0.870	0.870	0.915	0.821	0.691	0.691	0.691	0.821
19.500		0.896	0.840	0.840	0.840	0.896	0.780	0.620	0.620	0.620	0.780
21.400		0.908	0.858	0.858	0.858	0.908	0.806	0.664	0.664	0.664	0.806
23.200		0.907	0.857	0.857	0.857	0.907	0.805	0.662	0.662	0.662	0.805
25.000		0.910	0.862	0.862	0.862	0.910	0.811	0.673	0.673	0.673	0.811
26.900		0.917	0.872	0.872	0.872	0.917	0.825	0.697	0.697	0.697	0.825
28.500		0.913	0.865	0.865	0.865	0.913	0.816	0.681	0.681	0.681	0.816
30.000		0.914	0.867	0.867	0.867	0.914	0.818	0.685	0.685	0.685	0.818
32.000		0.916	0.871	0.871	0.871	0.916	0.823	0.693	0.693	0.693	0.823
34.000		0.919	0.875	0.875	0.875	0.919	0.828	0.703	0.703	0.703	0.828
35.900		0.909	0.860	0.860	0.860	0.909	0.809	0.669	0.669	0.669	0.809
37.700		0.894	0.837	0.837	0.837	0.894	0.777	0.614	0.614	0.614	0.777
39.500		0.928	0.855	0.855	0.855	0.906	0.847	0.656	0.656	0.656	0.801
41.200		0.909	0.815	0.815	0.815	0.880	0.805	0.561	0.561	0.561	0.746
43.300		0.875	0.807	0.807	0.807	0.875	0.735	0.542	0.542	0.542	0.735
44.700		0.931	0.833	0.833	0.833	0.891	0.864	0.603	0.603	0.603	0.771
46.200		0.931	0.834	0.834	0.834	0.892	0.865	0.607	0.607	0.607	0.773
47.700		0.925	0.818	0.818	0.818	0.882	0.852	0.569	0.569	0.569	0.751
49.700		0.896	0.839	0.839	0.839	0.896	0.780	0.619	0.619	0.619	0.780
51.600		0.900	0.846	0.846	0.846	0.900	0.789	0.635	0.635	0.635	0.789
53.500		0.872	0.803	0.803	0.803	0.872	0.731	0.534	0.534	0.534	0.731
54.900		0.881	0.816	0.816	0.816	0.881	0.748	0.564	0.564	0.564	0.748
56.200		0.846	0.763	0.763	0.763	0.846	0.675	0.438	0.438	0.438	0.675
57.600		0.887	0.826	0.826	0.826	0.887	0.762	0.588	0.588	0.588	0.762
59.800		0.921	0.878	0.878	0.878	0.921	0.833	0.711	0.711	0.711	0.833
61.800		0.943	0.912	0.912	0.912	0.943	0.879	0.791	0.791	0.791	0.879
63.800		0.975	0.962	0.962	0.962	0.975	0.947	0.909	0.909	0.909	0.947
66.500		0.976	0.963	0.963	0.963	0.976	0.950	0.913	0.913	0.913	0.950
68.400		0.942	0.911	0.911	0.911	0.942	0.877	0.788	0.788	0.788	0.877
70.100		0.915	0.869	0.869	0.869	0.915	0.821	0.690	0.690	0.690	0.821
71.600		0.906	0.856	0.856	0.856	0.906	0.802	0.658	0.658	0.658	0.802
73.100		0.896	0.840	0.840	0.840	0.896	0.780	0.620	0.620	0.620	0.780
74.600		0.866	0.794	0.794	0.794	0.912	0.717	0.511	0.511	0.511	0.826
76.200		0.826	0.732	0.732	0.732	0.885	0.632	0.364	0.364	0.364	0.774
78.000		0.853	0.774	0.774	0.774	0.904	0.691	0.465	0.465	0.465	0.810
79.500		0.867	0.796	0.796	0.796	0.867	0.720	0.516	0.516	0.516	0.720
81.600		0.861	0.787	0.787	0.787	0.861	0.708	0.494	0.494	0.494	0.708
83.800		0.915	0.869	0.869	0.869	0.915	0.821	0.690	0.690	0.690	0.821
86.000		0.935	0.900	0.900	0.900	0.935	0.864	0.764	0.764	0.764	0.864
88.100		0.961	0.940	0.940	0.940	0.961	0.918	0.858	0.858	0.858	0.918
89.900		0.992	0.988	0.988	0.988	0.992	0.984	0.972	0.972	0.972	0.984
91.500		0.997	0.995	0.995	0.995	0.997	0.994	0.989	0.989	0.989	0.994
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table D.2 (continued)

AXL ^a	VALUE	CF	I	IS:	11	12	13	14	15	16	17	18	19	20
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.003	1.005	1.005	1.005	1.003	1.003	1.003	1.003	1.005	1.005	1.005	1.005	1.005	1.003
1.800	0.946	0.906	0.906	0.906	0.946	0.946	0.946	0.946	0.906	0.906	0.906	0.906	0.906	0.946
3.400	0.885	0.801	0.801	0.801	0.885	0.885	0.885	0.885	0.801	0.801	0.801	0.801	0.801	0.885
5.000	0.889	0.808	0.808	0.808	0.889	0.889	0.889	0.889	0.808	0.808	0.808	0.808	0.808	0.889
6.900	0.915	0.853	0.853	0.853	0.915	0.915	0.915	0.915	0.853	0.853	0.853	0.853	0.853	0.915
8.800	0.963	0.936	0.936	0.936	0.963	0.963	0.963	0.963	0.936	0.936	0.936	0.936	0.936	0.963
11.500	0.961	0.933	0.933	0.933	0.961	0.961	0.961	0.961	0.933	0.933	0.933	0.933	0.933	0.961
13.300	0.908	0.840	0.840	0.840	0.908	0.908	0.908	0.908	0.840	0.840	0.840	0.840	0.840	0.908
15.100	0.873	0.780	0.780	0.780	0.873	0.873	0.873	0.873	0.780	0.780	0.780	0.780	0.780	0.873
16.800	0.839	0.721	0.721	0.721	0.839	0.839	0.839	0.839	0.721	0.721	0.721	0.721	0.721	0.839
18.100	0.821	0.691	0.691	0.691	0.821	0.821	0.821	0.821	0.691	0.691	0.691	0.691	0.691	0.821
19.500	0.780	0.620	0.620	0.620	0.780	0.780	0.780	0.780	0.620	0.620	0.620	0.620	0.620	0.780
21.400	0.806	0.664	0.664	0.664	0.806	0.806	0.806	0.806	0.664	0.664	0.664	0.664	0.664	0.806
23.200	0.805	0.662	0.662	0.662	0.805	0.805	0.805	0.805	0.662	0.662	0.662	0.662	0.662	0.805
25.000	0.811	0.673	0.673	0.673	0.811	0.811	0.811	0.811	0.673	0.673	0.673	0.673	0.673	0.811
26.900	0.825	0.697	0.697	0.697	0.825	0.825	0.825	0.825	0.697	0.697	0.697	0.697	0.697	0.825
28.500	0.816	0.681	0.681	0.681	0.816	0.816	0.816	0.816	0.681	0.681	0.681	0.681	0.681	0.816
30.000	0.818	0.685	0.685	0.685	0.818	0.818	0.818	0.818	0.685	0.685	0.685	0.685	0.685	0.818
32.000	0.823	0.693	0.693	0.693	0.823	0.823	0.823	0.823	0.693	0.693	0.693	0.693	0.693	0.823
34.000	0.828	0.703	0.703	0.703	0.828	0.828	0.828	0.828	0.703	0.703	0.703	0.703	0.703	0.828
35.900	0.809	0.669	0.669	0.669	0.809	0.809	0.809	0.809	0.669	0.669	0.669	0.669	0.669	0.809
37.700	0.777	0.614	0.614	0.614	0.777	0.777	0.777	0.777	0.614	0.614	0.614	0.614	0.614	0.777
39.500	0.801	0.656	0.656	0.656	0.801	0.801	0.801	0.801	0.656	0.656	0.656	0.656	0.656	0.801
41.200	0.746	0.561	0.561	0.561	0.746	0.746	0.746	0.746	0.561	0.561	0.561	0.561	0.561	0.746
43.300	0.735	0.542	0.542	0.542	0.735	0.735	0.735	0.735	0.542	0.542	0.542	0.542	0.542	0.735
44.700	0.822	0.603	0.603	0.603	0.771	0.771	0.771	0.771	0.603	0.603	0.603	0.603	0.603	0.771
46.200	0.823	0.607	0.607	0.607	0.773	0.773	0.773	0.773	0.607	0.607	0.607	0.607	0.607	0.773
47.700	0.806	0.569	0.569	0.569	0.751	0.751	0.751	0.751	0.569	0.569	0.569	0.569	0.569	0.751
49.700	0.780	0.619	0.619	0.619	0.780	0.780	0.780	0.780	0.619	0.619	0.619	0.619	0.619	0.780
51.600	0.789	0.635	0.635	0.635	0.789	0.789	0.789	0.789	0.635	0.635	0.635	0.635	0.635	0.789
53.500	0.731	0.534	0.534	0.534	0.731	0.731	0.731	0.731	0.534	0.534	0.534	0.534	0.534	0.731
54.900	0.748	0.564	0.564	0.564	0.748	0.748	0.748	0.748	0.564	0.564	0.564	0.564	0.564	0.748
56.200	0.675	0.438	0.438	0.438	0.675	0.675	0.675	0.675	0.438	0.438	0.438	0.438	0.438	0.675
57.600	0.762	0.588	0.588	0.588	0.762	0.762	0.762	0.762	0.588	0.588	0.588	0.588	0.588	0.762
59.800	0.833	0.711	0.711	0.711	0.833	0.833	0.833	0.833	0.711	0.711	0.711	0.711	0.711	0.833
61.800	0.879	0.791	0.791	0.791	0.879	0.879	0.879	0.879	0.791	0.791	0.791	0.791	0.791	0.879
63.800	0.947	0.909	0.909	0.909	0.947	0.947	0.947	0.947	0.909	0.909	0.909	0.909	0.909	0.947
66.500	0.950	0.913	0.913	0.913	0.950	0.950	0.950	0.950	0.913	0.913	0.913	0.913	0.913	0.950
68.400	0.877	0.788	0.788	0.788	0.877	0.877	0.877	0.877	0.788	0.788	0.788	0.788	0.788	0.877
70.100	0.821	0.690	0.690	0.690	0.821	0.821	0.821	0.821	0.690	0.690	0.690	0.690	0.690	0.821
71.600	0.802	0.658	0.658	0.658	0.802	0.802	0.802	0.802	0.658	0.658	0.658	0.658	0.658	0.802
73.100	0.780	0.620	0.620	0.620	0.780	0.780	0.780	0.780	0.620	0.620	0.620	0.620	0.620	0.780
74.600	0.717	0.511	0.511	0.511	0.826	0.717	0.511	0.511	0.511	0.511	0.511	0.511	0.511	0.827
76.200	0.632	0.364	0.364	0.364	0.774	0.632	0.364	0.364	0.364	0.364	0.364	0.364	0.364	0.775
78.000	0.691	0.465	0.465	0.465	0.810	0.691	0.465	0.465	0.465	0.465	0.465	0.465	0.465	0.811
79.500	0.720	0.516	0.516	0.516	0.720	0.720	0.516	0.516	0.516	0.516	0.516	0.516	0.516	0.720
81.600	0.708	0.494	0.494	0.494	0.708	0.708	0.494	0.494	0.494	0.494	0.494	0.494	0.494	0.708
83.800	0.821	0.690	0.690	0.690	0.821	0.821	0.690	0.690	0.690	0.690	0.690	0.690	0.690	0.821
86.000	0.864	0.764	0.764	0.764	0.864	0.864	0.764	0.764	0.764	0.764	0.764	0.764	0.764	0.864
88.100	0.918	0.858	0.858	0.858	0.918	0.918	0.858	0.858	0.858	0.858	0.858	0.858	0.858	0.918
89.900	0.984	0.972	0.972	0.972	0.984	0.984	0.972	0.972	0.972	0.972	0.972	0.972	0.972	0.984
91.500	0.994	0.989	0.989	0.989	0.994	0.994	0.989	0.989	0.989	0.989	0.989	0.989	0.989	0.994
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table D.2 (continued)

AXL ^a	VALUE OF I IS: 21	22	23	24	25
-47.676	1.000	1.000	1.000	1.000	1.000
0.000	1.001	1.002	1.002	1.002	1.001
1.800	0.974	0.960	0.960	0.960	0.974
3.400	0.945	0.916	0.916	0.916	0.945
5.000	0.947	0.919	0.919	0.919	0.947
6.900	0.960	0.938	0.938	0.938	0.960
8.800	0.982	0.973	0.973	0.973	0.982
11.500	0.982	0.972	0.972	0.972	0.982
13.300	0.956	0.933	0.933	0.933	0.956
15.100	0.940	0.907	0.907	0.907	0.940
16.800	0.924	0.882	0.882	0.882	0.924
18.100	0.915	0.870	0.870	0.870	0.915
19.500	0.896	0.840	0.840	0.840	0.896
21.400	0.908	0.858	0.858	0.858	0.908
23.200	0.907	0.857	0.857	0.857	0.907
25.000	0.910	0.862	0.862	0.862	0.910
26.900	0.917	0.872	0.872	0.872	0.917
28.500	0.913	0.865	0.865	0.865	0.913
30.000	0.914	0.867	0.867	0.867	0.914
32.000	0.916	0.871	0.871	0.871	0.916
34.000	0.919	0.875	0.875	0.875	0.919
35.900	0.909	0.860	0.860	0.860	0.909
37.700	0.894	0.837	0.837	0.837	0.894
39.500	0.906	0.855	0.855	0.855	0.906
41.200	0.880	0.815	0.815	0.815	0.880
43.300	0.875	0.807	0.807	0.807	0.875
44.700	0.891	0.833	0.833	0.833	0.891
46.200	0.892	0.834	0.834	0.834	0.892
47.700	0.882	0.818	0.818	0.818	0.882
49.700	0.896	0.839	0.839	0.839	0.896
51.600	0.900	0.846	0.846	0.846	0.900
53.500	0.872	0.803	0.803	0.803	0.872
54.900	0.881	0.816	0.816	0.816	0.881
56.200	0.846	0.763	0.763	0.763	0.846
57.600	0.887	0.826	0.826	0.826	0.887
59.800	0.921	0.878	0.878	0.878	0.921
61.800	0.943	0.912	0.912	0.912	0.943
63.800	0.975	0.962	0.962	0.962	0.975
66.500	0.976	0.963	0.963	0.963	0.976
68.400	0.942	0.911	0.911	0.911	0.942
70.100	0.915	0.869	0.869	0.869	0.915
71.600	0.906	0.856	0.856	0.856	0.906
73.100	0.896	0.840	0.840	0.840	0.896
74.600	0.866	0.794	0.794	0.794	0.866
76.200	0.826	0.732	0.732	0.732	0.826
78.000	0.853	0.774	0.774	0.774	0.853
79.500	0.867	0.796	0.796	0.796	0.867
81.600	0.861	0.787	0.787	0.787	0.861
83.800	0.915	0.869	0.869	0.869	0.915
86.000	0.935	0.900	0.900	0.900	0.935
88.100	0.961	0.940	0.940	0.940	0.961
89.900	0.992	0.988	0.988	0.988	0.992
91.500	0.997	0.995	0.995	0.995	0.997
95.987	1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table D.3. GFACT values for the B-2/shroud 1 COBRA-IV model based on the maximum restriction definition

GAPXL ^a	VALUE OF K IS: 1	2	3	4	5	6	7	8	9	10
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.002	1.004	1.002	1.010	1.002	1.010	1.002	1.010	1.004	1.004
1.800	0.958	0.934	0.958	0.826	0.958	0.826	0.958	0.826	0.934	0.934
3.400	0.913	0.865	0.913	0.642	0.913	0.642	0.913	0.642	0.865	0.865
5.000	0.916	0.869	0.916	0.654	0.916	0.654	0.916	0.654	0.869	0.869
6.900	0.935	0.899	0.935	0.732	0.935	0.732	0.935	0.732	0.899	0.899
8.800	0.971	0.955	0.971	0.880	0.971	0.880	0.971	0.880	0.955	0.955
11.500	0.970	0.953	0.970	0.875	0.970	0.875	0.970	0.875	0.953	0.953
13.300	0.929	0.890	0.929	0.709	0.929	0.709	0.929	0.709	0.890	0.890
15.100	0.904	0.852	0.904	0.606	0.904	0.606	0.904	0.606	0.852	0.852
16.800	0.880	0.814	0.880	0.508	0.880	0.508	0.880	0.508	0.814	0.814
18.100	0.868	0.796	0.868	0.459	0.868	0.459	0.868	0.459	0.796	0.796
19.500	0.840	0.753	0.840	0.346	0.840	0.346	0.840	0.346	0.753	0.753
21.400	0.857	0.780	0.857	0.415	0.857	0.415	0.857	0.415	0.780	0.780
23.200	0.857	0.778	0.857	0.412	0.857	0.412	0.857	0.412	0.778	0.778
25.000	0.861	0.785	0.861	0.430	0.861	0.430	0.861	0.430	0.785	0.785
26.900	0.870	0.800	0.870	0.469	0.870	0.469	0.870	0.469	0.800	0.800
28.500	0.864	0.790	0.864	0.443	0.864	0.443	0.864	0.443	0.790	0.790
30.000	0.866	0.792	0.866	0.449	0.866	0.449	0.866	0.449	0.792	0.792
32.000	0.869	0.797	0.869	0.462	0.869	0.462	0.869	0.462	0.797	0.797
34.000	0.873	0.803	0.873	0.478	0.873	0.478	0.873	0.478	0.803	0.803
35.900	0.859	0.783	0.859	0.423	0.859	0.423	0.859	0.423	0.783	0.783
37.700	0.838	0.750	0.838	0.336	0.838	0.336	0.838	0.336	0.750	0.750
39.500	0.854	0.775	0.854	0.403	0.854	0.403	0.854	0.403	0.775	0.775
41.200	0.818	0.719	0.818	0.254	0.818	0.254	0.818	0.254	0.719	0.719
43.300	0.811	0.708	0.811	0.225	0.811	0.225	0.811	0.225	0.708	0.708
44.700	0.834	0.743	0.834	0.319	0.834	0.319	0.834	0.319	0.743	0.743
46.200	0.835	0.746	0.835	0.325	0.835	0.325	0.835	0.325	0.746	0.746
47.700	0.821	0.723	0.821	0.266	0.821	0.266	0.821	0.266	0.723	0.723
49.700	0.840	0.753	0.840	0.344	0.840	0.344	0.840	0.344	0.753	0.753
51.600	0.846	0.762	0.846	0.369	0.846	0.369	0.846	0.369	0.762	0.762
53.500	0.808	0.703	0.808	0.213	0.808	0.213	0.808	0.213	0.703	0.703
54.900	0.819	0.720	0.819	0.258	0.819	0.258	0.819	0.258	0.720	0.720
56.200	0.773	0.649	0.773	0.069	0.773	0.069	0.773	0.069	0.649	0.649
57.600	0.828	0.734	0.828	0.296	0.828	0.296	0.828	0.296	0.734	0.734
59.800	0.876	0.808	0.876	0.491	0.876	0.491	0.876	0.491	0.808	0.808
61.800	0.909	0.859	0.909	0.625	0.909	0.625	0.909	0.625	0.859	0.859
63.800	0.959	0.936	0.959	0.831	0.959	0.831	0.959	0.831	0.936	0.936
66.500	0.961	0.939	0.961	0.839	0.961	0.839	0.961	0.839	0.939	0.939
68.400	0.907	0.857	0.907	0.620	0.907	0.620	0.907	0.620	0.857	0.857
70.100	0.868	0.795	0.868	0.457	0.868	0.457	0.868	0.457	0.795	0.795
71.600	0.855	0.776	0.855	0.406	0.855	0.406	0.855	0.406	0.776	0.776
73.100	0.840	0.753	0.840	0.346	0.840	0.346	0.840	0.346	0.753	0.753
74.600	0.800	0.690	0.800	0.178	0.800	0.178	0.800	0.178	0.690	0.690
76.200	0.747	0.609	0.747	0.0	0.747	0.0	0.747	0.0	0.609	0.609
78.000	0.664	0.783	0.664	0.109	0.783	0.109	0.783	0.109	0.664	0.664
79.500	0.801	0.693	0.801	0.185	0.801	0.185	0.801	0.185	0.693	0.693
81.600	0.793	0.680	0.793	0.152	0.793	0.152	0.793	0.152	0.680	0.680
83.800	0.868	0.795	0.868	0.457	0.868	0.457	0.868	0.457	0.795	0.795
86.000	0.897	0.841	0.897	0.579	0.897	0.579	0.897	0.579	0.841	0.841
88.100	0.937	0.902	0.937	0.741	0.937	0.741	0.937	0.741	0.902	0.902
89.900	0.987	0.980	0.987	0.947	0.987	0.947	0.987	0.947	0.980	0.980
91.500	0.995	0.992	0.995	0.979	0.995	0.979	0.995	0.979	0.992	0.992
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table D.3 (continued)

GAPXL ^a	VAIUE CP K IS: 11	12	13	14	15	16	17	18	19	20
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.010	1.010	1.004	1.004	1.010
1.800	0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826
3.400	0.642	0.642	0.642	0.642	0.642	0.642	0.642	0.865	0.865	0.642
5.000	0.654	0.654	0.654	0.654	0.654	0.654	0.654	0.869	0.869	0.654
6.900	0.732	0.732	0.732	0.732	0.732	0.732	0.732	0.899	0.899	0.732
8.800	0.880	0.880	0.880	0.880	0.880	0.880	0.880	0.955	0.955	0.880
11.500	0.875	0.875	0.875	0.875	0.875	0.875	0.875	0.953	0.953	0.875
13.300	0.709	0.709	0.709	0.709	0.709	0.709	0.709	0.890	0.890	0.709
15.100	0.606	0.606	0.606	0.606	0.606	0.606	0.606	0.852	0.852	0.606
16.800	0.508	0.508	0.508	0.508	0.508	0.508	0.508	0.814	0.814	0.508
18.100	0.459	0.459	0.459	0.459	0.459	0.459	0.459	0.796	0.796	0.459
19.500	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346
21.400	0.415	0.415	0.415	0.415	0.415	0.415	0.415	0.780	0.780	0.415
23.200	0.412	0.412	0.412	0.412	0.412	0.412	0.412	0.778	0.778	0.412
25.000	0.430	0.430	0.430	0.430	0.430	0.430	0.430	0.785	0.785	0.430
26.900	0.469	0.469	0.469	0.469	0.469	0.469	0.469	0.800	0.800	0.469
28.500	0.443	0.443	0.443	0.443	0.443	0.443	0.443	0.790	0.790	0.443
30.000	0.449	0.449	0.449	0.449	0.449	0.449	0.449	0.792	0.792	0.449
32.000	0.462	0.462	0.462	0.462	0.462	0.462	0.462	0.797	0.797	0.462
34.000	0.478	0.478	0.478	0.478	0.478	0.478	0.478	0.803	0.803	0.478
35.900	0.423	0.423	0.423	0.423	0.423	0.423	0.423	0.783	0.783	0.423
37.700	0.336	0.336	0.336	0.336	0.336	0.336	0.336	0.750	0.750	0.336
39.500	0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403
41.200	0.254	0.254	0.254	0.254	0.254	0.254	0.254	0.719	0.719	0.254
43.300	0.225	0.225	0.225	0.225	0.225	0.225	0.225	0.708	0.708	0.225
44.700	0.319	0.319	0.319	0.319	0.319	0.319	0.319	0.743	0.743	0.319
46.200	0.325	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325
47.700	0.266	0.266	0.266	0.266	0.266	0.266	0.266	0.723	0.723	0.266
49.700	0.344	0.344	0.344	0.344	0.344	0.344	0.344	0.753	0.753	0.344
51.600	0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369
53.500	0.213	0.213	0.213	0.213	0.213	0.213	0.213	0.703	0.703	0.213
54.900	0.258	0.258	0.258	0.258	0.258	0.258	0.258	0.720	0.720	0.258
56.200	0.069	0.069	0.069	0.069	0.069	0.069	0.069	0.649	0.649	0.069
57.600	0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296
59.800	0.491	0.491	0.491	0.491	0.491	0.491	0.491	0.808	0.808	0.491
61.800	0.625	0.625	0.625	0.625	0.625	0.625	0.625	0.859	0.859	0.625
63.800	0.831	0.831	0.831	0.831	0.831	0.831	0.831	0.936	0.936	0.831
66.500	0.839	0.839	0.839	0.839	0.839	0.839	0.839	0.939	0.939	0.839
68.400	0.620	0.620	0.620	0.620	0.620	0.620	0.620	0.857	0.857	0.620
70.100	0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457
71.600	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.776	0.776	0.406
73.100	0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346
74.600	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.690	0.690	0.178
76.200	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.609	0.609	0.0
78.000	0.109	0.109	0.109	0.109	0.109	0.109	0.109	0.664	0.664	0.109
79.500	0.185	0.185	0.185	0.185	0.185	0.185	0.185	0.693	0.693	0.185
81.600	0.152	0.152	0.152	0.152	0.152	0.152	0.152	0.680	0.680	0.152
83.800	0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457
86.000	0.579	0.579	0.579	0.579	0.579	0.579	0.579	0.841	0.841	0.579
88.100	0.741	0.741	0.741	0.741	0.741	0.741	0.741	0.902	0.902	0.741
89.900	0.947	0.947	0.947	0.947	0.947	0.947	0.947	0.980	0.980	0.947
91.500	0.979	0.979	0.979	0.979	0.979	0.979	0.979	0.992	0.992	0.979
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table D.3 (continued)

GAPXL ^a	VALUE OF K IS: 21	22	23	24	25	26	27	28	29	30
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.010	1.004	1.004	1.010	1.010
1.800	0.826	0.826	0.826	0.826	0.826	0.826	0.934	0.934	0.826	0.826
3.400	0.642	0.642	0.642	0.642	0.642	0.642	0.865	0.865	0.642	0.642
5.000	0.654	0.654	0.654	0.654	0.654	0.654	0.869	0.869	0.654	0.654
6.900	0.732	0.732	0.732	0.732	0.732	0.732	0.899	0.899	0.732	0.732
8.800	0.880	0.880	0.880	0.880	0.880	0.880	0.955	0.955	0.880	0.880
11.500	0.875	0.875	0.875	0.875	0.875	0.875	0.953	0.953	0.875	0.875
13.300	0.709	0.709	0.709	0.709	0.709	0.709	0.890	0.890	0.709	0.709
15.100	0.606	0.606	0.606	0.606	0.606	0.606	0.852	0.852	0.606	0.606
16.800	0.508	0.508	0.508	0.508	0.508	0.508	0.814	0.814	0.508	0.508
18.100	0.459	0.459	0.459	0.459	0.459	0.459	0.796	0.796	0.459	0.459
19.500	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346	0.346
21.400	0.415	0.415	0.415	0.415	0.415	0.415	0.780	0.780	0.415	0.415
23.200	0.412	0.412	0.412	0.412	0.412	0.412	0.778	0.778	0.412	0.412
25.000	0.430	0.430	0.430	0.430	0.430	0.430	0.785	0.785	0.430	0.430
26.900	0.469	0.469	0.469	0.469	0.469	0.469	0.800	0.800	0.469	0.469
28.500	0.443	0.443	0.443	0.443	0.443	0.443	0.790	0.790	0.443	0.443
30.000	0.449	0.449	0.449	0.449	0.449	0.449	0.792	0.792	0.449	0.449
32.000	0.462	0.462	0.462	0.462	0.462	0.462	0.797	0.797	0.462	0.462
34.000	0.478	0.478	0.478	0.478	0.478	0.478	0.803	0.803	0.478	0.478
35.900	0.423	0.423	0.423	0.423	0.423	0.423	0.783	0.783	0.423	0.423
37.700	0.336	0.336	0.336	0.336	0.336	0.336	0.750	0.750	0.336	0.336
39.500	0.403	0.403	0.403	0.403	0.403	0.403	0.775	0.775	0.403	0.403
41.200	0.254	0.254	0.254	0.254	0.254	0.254	0.719	0.719	0.254	0.254
43.300	0.225	0.225	0.225	0.225	0.225	0.225	0.708	0.708	0.225	0.225
44.700	0.319	0.319	0.319	0.319	0.319	0.319	0.743	0.743	0.319	0.319
46.200	0.325	0.325	0.325	0.325	0.325	0.325	0.746	0.746	0.325	0.325
47.700	0.266	0.266	0.266	0.266	0.266	0.266	0.723	0.723	0.266	0.266
49.700	0.344	0.344	0.344	0.344	0.344	0.344	0.753	0.753	0.344	0.344
51.600	0.369	0.369	0.369	0.369	0.369	0.369	0.762	0.762	0.369	0.369
53.500	0.213	0.213	0.213	0.213	0.213	0.213	0.703	0.703	0.213	0.213
54.900	0.258	0.258	0.258	0.258	0.258	0.258	0.720	0.720	0.258	0.258
56.200	0.069	0.069	0.069	0.069	0.069	0.069	0.649	0.649	0.069	0.069
57.600	0.296	0.296	0.296	0.296	0.296	0.296	0.734	0.734	0.296	0.296
59.800	0.491	0.491	0.491	0.491	0.491	0.491	0.808	0.808	0.491	0.491
61.800	0.625	0.625	0.625	0.625	0.625	0.625	0.859	0.859	0.625	0.625
63.800	0.831	0.831	0.831	0.831	0.831	0.831	0.936	0.936	0.831	0.831
66.500	0.839	0.839	0.839	0.839	0.839	0.839	0.939	0.939	0.839	0.839
68.400	0.620	0.620	0.620	0.620	0.620	0.620	0.857	0.857	0.620	0.620
70.100	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457	0.457
71.600	0.406	0.406	0.406	0.406	0.406	0.406	0.776	0.776	0.406	0.406
73.100	0.346	0.346	0.346	0.346	0.346	0.346	0.753	0.753	0.346	0.346
74.600	0.178	0.178	0.178	0.178	0.178	0.178	0.690	0.690	0.178	0.178
76.200	0.0	0.0	0.0	0.0	0.0	0.0	0.609	0.609	0.0	0.0
78.000	0.109	0.109	0.109	0.109	0.109	0.109	0.664	0.664	0.109	0.109
79.500	0.185	0.185	0.185	0.185	0.185	0.185	0.693	0.693	0.185	0.185
81.600	0.152	0.152	0.152	0.152	0.152	0.152	0.680	0.680	0.152	0.152
83.800	0.457	0.457	0.457	0.457	0.457	0.457	0.795	0.795	0.457	0.457
86.000	0.579	0.579	0.579	0.579	0.579	0.579	0.841	0.841	0.579	0.579
88.100	0.741	0.741	0.741	0.741	0.741	0.741	0.902	0.902	0.741	0.741
89.900	0.947	0.947	0.947	0.947	0.947	0.947	0.980	0.980	0.947	0.947
91.500	0.979	0.979	0.979	0.979	0.979	0.979	0.992	0.992	0.979	0.979
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table D.3 (continued)

GAPXL ^a	VAIUE CP K IS: 31	32	33	34	35	36	37	38	39	40
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.004	1.002	1.002	1.002	1.002
1.800	0.826	0.826	0.826	0.826	0.826	0.934	0.958	0.958	0.958	0.958
3.400	0.642	0.642	0.642	0.642	0.642	0.865	0.913	0.913	0.913	0.913
5.000	0.654	0.654	0.654	0.654	0.654	0.869	0.916	0.916	0.916	0.916
6.900	0.732	0.732	0.732	0.732	0.732	0.899	0.935	0.935	0.935	0.935
8.800	0.880	0.880	0.880	0.880	0.880	0.955	0.971	0.971	0.971	0.971
11.500	0.875	0.875	0.875	0.875	0.875	0.953	0.970	0.970	0.970	0.970
13.300	0.709	0.709	0.709	0.709	0.709	0.890	0.929	0.929	0.929	0.929
15.100	0.606	0.606	0.606	0.606	0.606	0.852	0.904	0.904	0.904	0.904
16.800	0.508	0.508	0.508	0.508	0.508	0.814	0.880	0.880	0.880	0.880
18.100	0.459	0.459	0.459	0.459	0.459	0.796	0.868	0.868	0.868	0.868
19.500	0.346	0.346	0.346	0.346	0.346	0.753	0.840	0.840	0.840	0.840
21.400	0.415	0.415	0.415	0.415	0.415	0.780	0.857	0.857	0.857	0.857
23.200	0.412	0.412	0.412	0.412	0.412	0.778	0.857	0.857	0.857	0.857
25.000	0.430	0.430	0.430	0.430	0.430	0.785	0.861	0.861	0.861	0.861
26.900	0.469	0.469	0.469	0.469	0.469	0.800	0.870	0.870	0.870	0.870
28.500	0.443	0.443	0.443	0.443	0.443	0.790	0.864	0.864	0.864	0.864
30.000	0.449	0.449	0.449	0.449	0.449	0.792	0.866	0.866	0.866	0.866
32.000	0.462	0.462	0.462	0.462	0.462	0.797	0.869	0.869	0.869	0.869
34.000	0.478	0.478	0.478	0.478	0.478	0.803	0.873	0.873	0.873	0.873
35.900	0.423	0.423	0.423	0.423	0.423	0.783	0.859	0.859	0.859	0.859
37.700	0.336	0.336	0.336	0.336	0.336	0.750	0.838	0.838	0.838	0.838
39.500	0.403	0.403	0.403	0.403	0.403	0.775	0.854	0.854	0.854	0.854
41.200	0.254	0.254	0.254	0.254	0.254	0.719	0.818	0.818	0.818	0.818
43.300	0.225	0.225	0.225	0.225	0.225	0.708	0.811	0.811	0.811	0.811
44.700	0.319	0.319	0.319	0.319	0.319	0.743	0.834	0.834	0.834	0.834
46.200	0.325	0.325	0.325	0.325	0.325	0.746	0.835	0.835	0.835	0.835
47.700	0.266	0.266	0.266	0.266	0.266	0.723	0.821	0.821	0.821	0.821
49.700	0.344	0.344	0.344	0.344	0.344	0.753	0.840	0.840	0.840	0.840
51.600	0.369	0.369	0.369	0.369	0.369	0.762	0.846	0.846	0.846	0.846
53.500	0.213	0.213	0.213	0.213	0.213	0.703	0.808	0.808	0.808	0.808
54.900	0.258	0.258	0.258	0.258	0.258	0.720	0.819	0.819	0.819	0.819
56.200	0.069	0.069	0.069	0.069	0.069	0.649	0.773	0.773	0.773	0.773
57.600	0.296	0.296	0.296	0.296	0.296	0.734	0.828	0.828	0.828	0.828
59.800	0.491	0.491	0.491	0.491	0.491	0.808	0.876	0.876	0.876	0.876
61.800	0.625	0.625	0.625	0.625	0.625	0.859	0.909	0.909	0.909	0.909
63.800	0.831	0.831	0.831	0.831	0.831	0.936	0.959	0.959	0.959	0.959
66.500	0.839	0.839	0.839	0.839	0.839	0.939	0.961	0.961	0.961	0.961
68.400	0.620	0.620	0.620	0.620	0.620	0.857	0.907	0.907	0.907	0.907
70.100	0.457	0.457	0.457	0.457	0.457	0.795	0.868	0.868	0.868	0.868
71.600	0.406	0.406	0.406	0.406	0.406	0.776	0.855	0.855	0.855	0.855
73.100	0.346	0.346	0.346	0.346	0.346	0.753	0.840	0.840	0.840	0.840
74.600	0.178	0.178	0.178	0.178	0.178	0.690	0.800	0.800	0.800	0.800
76.200	0.0	0.0	0.0	0.0	0.0	0.609	0.747	0.747	0.747	0.747
78.000	0.109	0.109	0.109	0.109	0.109	0.664	0.783	0.783	0.783	0.783
79.500	0.185	0.185	0.185	0.185	0.185	0.693	0.801	0.801	0.801	0.801
81.600	0.152	0.152	0.152	0.152	0.152	0.680	0.793	0.793	0.793	0.793
83.800	0.457	0.457	0.457	0.457	0.457	0.795	0.868	0.868	0.868	0.868
86.000	0.579	0.579	0.579	0.579	0.579	0.841	0.897	0.897	0.897	0.897
88.100	0.741	0.741	0.741	0.741	0.741	0.902	0.937	0.937	0.937	0.937
89.900	0.947	0.947	0.947	0.947	0.947	0.980	0.987	0.987	0.987	0.987
91.500	0.979	0.979	0.979	0.979	0.979	0.992	0.995	0.995	0.995	0.995
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

Appendix E

B-2/SHROUD 2 BUNDLE-AVERAGED GEOMETRY DATA,
MINIMUM RESTRICTION DEFINITION

The data in Appendix E represent the input from card groups 5 and 6 as required by ref. 2 for the B-2/shroud 2 COBRA-IV model based on the minimum restriction definition.

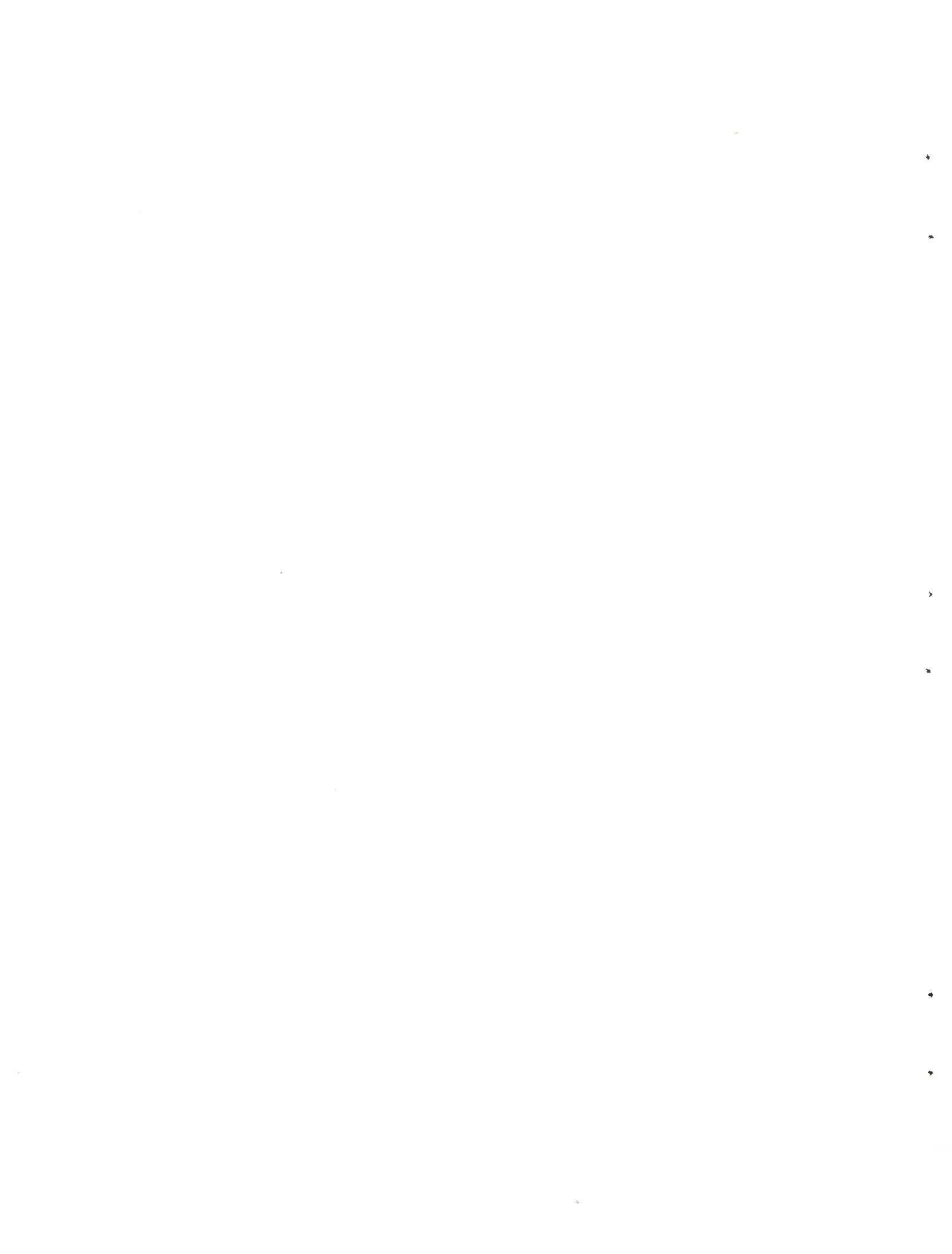


Table E.1. Shroud flow area reductions for the B-2/shroud 2 COBRA-IV model based on the minimum restriction definition

AXL ^a	FRACTIONAL SHROUD FLOW AREA
-47.676	1.000
0.000	1.003
1.800	0.950
3.400	0.894
5.000	0.898
6.900	0.922
8.800	0.966
11.500	0.964
13.300	0.915
15.100	0.883
16.900	0.851
18.100	0.835
19.500	0.810
21.400	0.821
23.200	0.820
25.000	0.826
26.900	0.838
28.500	0.830
30.000	0.832
32.000	0.836
34.000	0.842
35.900	0.823
37.700	0.799
39.500	0.817
41.200	0.785
43.300	0.780
44.700	0.788
46.200	0.790
47.700	0.780
49.700	0.797
51.600	0.805
53.500	0.778
54.900	0.767
56.200	0.744
57.600	0.780
59.800	0.846
61.800	0.889
63.800	0.951
66.500	0.954
68.400	0.887
70.100	0.835
71.600	0.818
73.100	0.797
74.600	0.747
76.200	0.720
78.000	0.728
79.500	0.747
81.600	0.763
83.800	0.835
86.000	0.874
88.100	0.924
89.900	0.985
91.500	0.994
95.987	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table E.2. AFACT values for the B-2/shroud 2 COBRA-IV model based on the minimum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001	1.002	1.005	1.005	1.005	1.002
1.800		0.983	0.964	0.964	0.964	0.983	0.963	0.906	0.906	0.906	0.963
3.400		0.964	0.924	0.924	0.924	0.964	0.921	0.801	0.801	0.801	0.921
5.000		0.965	0.927	0.927	0.927	0.965	0.924	0.808	0.808	0.808	0.924
6.900		0.973	0.944	0.944	0.944	0.973	0.942	0.853	0.853	0.853	0.942
8.800		0.988	0.976	0.976	0.976	0.988	0.975	0.936	0.936	0.936	0.975
11.500		0.988	0.974	0.974	0.974	0.988	0.974	0.933	0.933	0.933	0.974
13.300		0.971	0.939	0.939	0.939	0.971	0.937	0.840	0.840	0.840	0.937
15.100		0.960	0.916	0.916	0.916	0.960	0.913	0.780	0.780	0.780	0.913
16.800		0.950	0.894	0.894	0.894	0.950	0.890	0.721	0.721	0.721	0.890
18.100		0.944	0.882	0.882	0.882	0.944	0.878	0.691	0.691	0.691	0.878
19.500		0.936	0.864	0.864	0.864	0.936	0.859	0.644	0.644	0.644	0.859
21.400		0.939	0.872	0.872	0.872	0.939	0.867	0.664	0.664	0.664	0.867
23.200		0.939	0.871	0.871	0.871	0.939	0.866	0.662	0.662	0.662	0.866
25.000		0.941	0.876	0.876	0.876	0.941	0.871	0.673	0.673	0.673	0.871
26.900		0.945	0.885	0.885	0.885	0.945	0.880	0.697	0.697	0.697	0.880
28.500		0.942	0.879	0.879	0.879	0.942	0.874	0.681	0.681	0.681	0.874
30.000		0.943	0.880	0.880	0.880	0.943	0.876	0.685	0.685	0.685	0.876
32.000		0.944	0.883	0.883	0.883	0.944	0.879	0.693	0.693	0.693	0.879
34.000		0.946	0.887	0.887	0.887	0.946	0.883	0.703	0.703	0.703	0.883
35.900		0.940	0.874	0.874	0.874	0.940	0.869	0.669	0.669	0.669	0.869
37.700		0.932	0.856	0.856	0.856	0.932	0.851	0.623	0.623	0.623	0.851
39.500		0.938	0.869	0.869	0.869	0.938	0.864	0.656	0.656	0.656	0.864
41.200		0.927	0.847	0.847	0.847	0.927	0.841	0.597	0.597	0.597	0.841
43.300		0.925	0.843	0.843	0.843	0.925	0.837	0.587	0.587	0.587	0.837
44.700		0.928	0.849	0.849	0.849	0.928	0.843	0.603	0.603	0.603	0.843
46.200		0.929	0.850	0.850	0.850	0.929	0.845	0.607	0.607	0.607	0.845
47.700		0.925	0.843	0.843	0.843	0.925	0.837	0.588	0.588	0.588	0.837
49.700		0.931	0.855	0.855	0.855	0.931	0.849	0.619	0.619	0.619	0.849
51.600		0.934	0.861	0.861	0.861	0.934	0.856	0.635	0.635	0.635	0.856
53.500		0.925	0.841	0.841	0.841	0.925	0.835	0.583	0.583	0.583	0.835
54.900		0.921	0.834	0.834	0.834	0.921	0.828	0.564	0.564	0.564	0.828
56.200		0.913	0.817	0.817	0.817	0.913	0.810	0.520	0.520	0.520	0.810
57.600		0.925	0.843	0.843	0.843	0.925	0.837	0.588	0.588	0.588	0.837
59.800		0.948	0.890	0.890	0.890	0.948	0.886	0.711	0.711	0.711	0.886
61.800		0.962	0.920	0.920	0.920	0.962	0.917	0.791	0.791	0.791	0.917
63.800		0.984	0.965	0.965	0.965	0.984	0.964	0.909	0.909	0.909	0.964
66.500		0.984	0.967	0.967	0.967	0.984	0.966	0.913	0.913	0.913	0.966
68.400		0.962	0.919	0.919	0.919	0.962	0.916	0.788	0.788	0.788	0.916
70.100		0.944	0.882	0.882	0.882	0.944	0.878	0.690	0.690	0.690	0.878
71.600		0.938	0.870	0.870	0.870	0.938	0.865	0.658	0.658	0.658	0.865
73.100		0.931	0.855	0.855	0.855	0.931	0.850	0.620	0.620	0.620	0.850
74.600		0.914	0.820	0.820	0.820	0.914	0.813	0.526	0.526	0.526	0.813
76.200		0.905	0.800	0.800	0.800	0.905	0.793	0.475	0.475	0.475	0.793
78.000		0.908	0.806	0.806	0.806	0.908	0.798	0.490	0.490	0.490	0.798
79.500		0.914	0.820	0.820	0.820	0.914	0.813	0.526	0.526	0.526	0.813
81.600		0.920	0.831	0.831	0.831	0.920	0.825	0.556	0.556	0.556	0.825
83.800		0.944	0.882	0.882	0.882	0.944	0.878	0.690	0.690	0.690	0.878
86.000		0.957	0.910	0.910	0.910	0.957	0.907	0.764	0.764	0.764	0.907
88.100		0.974	0.946	0.946	0.946	0.974	0.944	0.858	0.858	0.858	0.944
89.900		0.995	0.989	0.989	0.989	0.995	0.989	0.972	0.972	0.972	0.989
91.500		0.998	0.996	0.996	0.996	0.998	0.996	0.989	0.989	0.989	0.996
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table E.2 (continued)

AXL ^a	VALUE OF I IS:	11	12	13	14	15	16	17	18	19	20
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.002	1.005	1.005	1.005	1.002	1.002	1.005	1.005	1.005	1.002
1.800		0.963	0.906	0.906	0.906	0.963	0.963	0.906	0.906	0.906	0.963
3.400		0.921	0.801	0.801	0.801	0.921	0.921	0.801	0.801	0.801	0.921
5.000		0.924	0.808	0.808	0.808	0.924	0.924	0.808	0.808	0.808	0.924
6.900		0.942	0.853	0.853	0.853	0.942	0.942	0.853	0.853	0.853	0.942
8.800		0.975	0.936	0.936	0.936	0.975	0.975	0.936	0.936	0.936	0.975
11.500		0.974	0.933	0.933	0.933	0.974	0.974	0.933	0.933	0.933	0.974
13.300		0.937	0.840	0.840	0.840	0.937	0.937	0.840	0.840	0.840	0.937
15.100		0.913	0.780	0.780	0.780	0.913	0.913	0.780	0.780	0.780	0.913
16.800		0.890	0.721	0.721	0.721	0.890	0.890	0.721	0.721	0.721	0.890
18.100		0.878	0.691	0.691	0.691	0.878	0.878	0.691	0.691	0.691	0.878
19.500		0.859	0.644	0.644	0.644	0.859	0.859	0.644	0.644	0.644	0.859
21.400		0.867	0.664	0.664	0.664	0.867	0.867	0.664	0.664	0.664	0.867
23.200		0.866	0.662	0.662	0.662	0.866	0.866	0.662	0.662	0.662	0.866
25.000		0.871	0.673	0.673	0.673	0.871	0.871	0.673	0.673	0.673	0.871
26.900		0.880	0.697	0.697	0.697	0.880	0.880	0.697	0.697	0.697	0.880
28.500		0.874	0.681	0.681	0.681	0.874	0.874	0.681	0.681	0.681	0.874
30.000		0.876	0.685	0.685	0.685	0.876	0.876	0.685	0.685	0.685	0.876
32.000		0.879	0.693	0.693	0.693	0.879	0.879	0.693	0.693	0.693	0.879
34.000		0.883	0.703	0.703	0.703	0.883	0.883	0.703	0.703	0.703	0.883
35.900		0.869	0.669	0.669	0.669	0.869	0.869	0.669	0.669	0.669	0.869
37.700		0.851	0.623	0.623	0.623	0.851	0.851	0.623	0.623	0.623	0.851
39.500		0.864	0.656	0.656	0.656	0.864	0.864	0.656	0.656	0.656	0.864
41.200		0.841	0.597	0.597	0.597	0.841	0.841	0.597	0.597	0.597	0.841
43.300		0.837	0.587	0.587	0.587	0.837	0.837	0.587	0.587	0.587	0.837
44.700		0.843	0.603	0.603	0.603	0.843	0.843	0.603	0.603	0.603	0.843
46.200		0.845	0.607	0.607	0.607	0.845	0.845	0.607	0.607	0.607	0.845
47.700		0.837	0.588	0.588	0.588	0.837	0.837	0.588	0.588	0.588	0.837
49.700		0.849	0.619	0.619	0.619	0.849	0.849	0.619	0.619	0.619	0.849
51.600		0.856	0.635	0.635	0.635	0.856	0.856	0.635	0.635	0.635	0.856
53.500		0.835	0.583	0.583	0.583	0.835	0.835	0.583	0.583	0.583	0.835
54.900		0.828	0.564	0.564	0.564	0.828	0.828	0.564	0.564	0.564	0.828
56.200		0.810	0.520	0.520	0.520	0.810	0.810	0.520	0.520	0.520	0.810
57.600		0.837	0.588	0.588	0.588	0.837	0.837	0.588	0.588	0.588	0.837
59.800		0.886	0.711	0.711	0.711	0.886	0.886	0.711	0.711	0.711	0.886
61.800		0.917	0.791	0.791	0.791	0.917	0.917	0.791	0.791	0.791	0.917
63.800		0.964	0.909	0.909	0.909	0.964	0.964	0.909	0.909	0.909	0.964
66.500		0.966	0.913	0.913	0.913	0.966	0.966	0.913	0.913	0.913	0.966
68.400		0.916	0.788	0.788	0.788	0.916	0.916	0.788	0.788	0.788	0.916
70.100		0.878	0.690	0.690	0.690	0.878	0.878	0.690	0.690	0.690	0.878
71.600		0.865	0.658	0.658	0.658	0.865	0.865	0.658	0.658	0.658	0.865
73.100		0.850	0.620	0.620	0.620	0.850	0.850	0.620	0.620	0.620	0.850
74.600		0.813	0.526	0.526	0.526	0.813	0.813	0.526	0.526	0.526	0.813
76.200		0.793	0.475	0.475	0.475	0.793	0.793	0.475	0.475	0.475	0.793
78.000		0.798	0.490	0.490	0.490	0.798	0.798	0.490	0.490	0.490	0.798
79.500		0.813	0.526	0.526	0.526	0.813	0.813	0.526	0.526	0.526	0.813
81.600		0.825	0.556	0.556	0.556	0.825	0.825	0.556	0.556	0.556	0.825
83.800		0.878	0.690	0.690	0.690	0.878	0.878	0.690	0.690	0.690	0.878
86.000		0.907	0.764	0.764	0.764	0.907	0.907	0.764	0.764	0.764	0.907
88.100		0.944	0.858	0.858	0.858	0.944	0.944	0.858	0.858	0.858	0.944
89.900		0.989	0.972	0.972	0.972	0.989	0.989	0.972	0.972	0.972	0.989
91.500		0.996	0.989	0.989	0.989	0.996	0.996	0.989	0.989	0.989	0.996
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table E.2 (continued)

AXL ^a	VALUE OF I IS:	21	22	23	24	25
-47.676		1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001
1.800		0.983	0.964	0.964	0.964	0.983
3.400		0.964	0.924	0.924	0.924	0.964
5.000		0.965	0.927	0.927	0.927	0.965
6.900		0.973	0.944	0.944	0.944	0.973
8.800		0.988	0.976	0.976	0.976	0.988
11.500		0.988	0.974	0.974	0.974	0.988
13.300		0.971	0.939	0.939	0.939	0.971
15.100		0.960	0.916	0.916	0.916	0.960
16.800		0.950	0.894	0.894	0.894	0.950
18.100		0.944	0.882	0.882	0.882	0.944
19.500		0.936	0.864	0.864	0.864	0.936
21.400		0.939	0.872	0.872	0.872	0.939
23.200		0.939	0.871	0.871	0.871	0.939
25.000		0.941	0.876	0.876	0.876	0.941
26.900		0.945	0.885	0.885	0.885	0.945
28.500		0.942	0.879	0.879	0.879	0.942
30.000		0.943	0.880	0.880	0.880	0.943
32.000		0.944	0.883	0.883	0.883	0.944
34.000		0.946	0.887	0.887	0.887	0.946
35.900		0.940	0.874	0.874	0.874	0.940
37.700		0.932	0.856	0.856	0.856	0.932
39.500		0.938	0.869	0.869	0.869	0.938
41.200		0.927	0.847	0.847	0.847	0.927
43.300		0.925	0.843	0.843	0.843	0.925
44.700		0.928	0.849	0.849	0.849	0.928
46.200		0.929	0.850	0.850	0.850	0.929
47.700		0.925	0.843	0.843	0.843	0.925
49.700		0.931	0.855	0.855	0.855	0.931
51.600		0.934	0.861	0.861	0.861	0.934
53.500		0.925	0.841	0.841	0.841	0.925
54.900		0.921	0.834	0.834	0.834	0.921
56.200		0.913	0.817	0.817	0.817	0.913
57.600		0.925	0.843	0.843	0.843	0.925
59.800		0.948	0.890	0.890	0.890	0.948
61.800		0.962	0.920	0.920	0.920	0.962
63.800		0.984	0.965	0.965	0.965	0.984
66.500		0.984	0.967	0.967	0.967	0.984
68.400		0.962	0.919	0.919	0.919	0.962
70.100		0.944	0.882	0.882	0.882	0.944
71.600		0.938	0.870	0.870	0.870	0.938
73.100		0.931	0.855	0.855	0.855	0.931
74.600		0.914	0.820	0.820	0.820	0.914
76.200		0.905	0.800	0.800	0.800	0.905
78.000		0.908	0.806	0.806	0.806	0.908
79.500		0.914	0.820	0.820	0.820	0.914
81.600		0.920	0.831	0.831	0.831	0.920
83.800		0.944	0.882	0.882	0.882	0.944
86.000		0.957	0.910	0.910	0.910	0.957
88.100		0.974	0.946	0.946	0.946	0.974
89.900		0.995	0.989	0.989	0.989	0.995
91.500		0.998	0.996	0.996	0.996	0.998
95.987		1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table E.3. GFACT values for the B-2/shroud 2 COBRA-IV model based on the minimum restriction definition

GAPXL ^a	VALUE OF K IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.002	1.002	1.002	1.010	1.002	1.010	1.002	1.010	1.002	1.002
1.800		0.963	0.961	0.963	0.826	0.963	0.826	0.963	0.826	0.961	0.961
3.400		0.923	0.920	0.923	0.642	0.923	0.642	0.923	0.642	0.920	0.920
5.000		0.926	0.922	0.926	0.654	0.926	0.654	0.926	0.654	0.922	0.922
6.900		0.943	0.940	0.943	0.732	0.943	0.732	0.943	0.732	0.940	0.940
8.800		0.974	0.973	0.974	0.880	0.974	0.880	0.974	0.880	0.973	0.973
11.500		0.973	0.972	0.973	0.875	0.973	0.875	0.973	0.875	0.972	0.972
13.300		0.938	0.935	0.938	0.709	0.938	0.709	0.938	0.709	0.935	0.935
15.100		0.916	0.912	0.916	0.606	0.916	0.606	0.916	0.606	0.912	0.912
16.800		0.895	0.890	0.895	0.508	0.895	0.508	0.895	0.508	0.890	0.890
18.100		0.884	0.879	0.884	0.459	0.884	0.459	0.884	0.459	0.879	0.879
19.500		0.868	0.862	0.868	0.383	0.868	0.383	0.868	0.383	0.862	0.862
21.400		0.875	0.869	0.875	0.415	0.875	0.415	0.875	0.415	0.869	0.869
23.200		0.874	0.868	0.874	0.412	0.874	0.412	0.874	0.412	0.868	0.868
25.000		0.878	0.872	0.878	0.430	0.878	0.430	0.878	0.430	0.872	0.872
26.900		0.886	0.881	0.886	0.469	0.886	0.469	0.886	0.469	0.881	0.881
28.500		0.881	0.875	0.881	0.443	0.881	0.443	0.881	0.443	0.875	0.875
30.000		0.882	0.877	0.882	0.449	0.882	0.449	0.882	0.449	0.877	0.877
32.000		0.885	0.880	0.885	0.462	0.885	0.462	0.885	0.462	0.880	0.880
34.000		0.889	0.883	0.889	0.478	0.889	0.478	0.889	0.478	0.883	0.883
35.900		0.877	0.871	0.877	0.423	0.877	0.423	0.877	0.423	0.871	0.871
37.700		0.861	0.854	0.861	0.350	0.861	0.350	0.861	0.350	0.854	0.854
39.500		0.872	0.866	0.872	0.403	0.872	0.403	0.872	0.403	0.866	0.866
41.200		0.852	0.845	0.852	0.310	0.852	0.310	0.852	0.310	0.845	0.845
43.300		0.849	0.842	0.849	0.294	0.849	0.294	0.849	0.294	0.842	0.842
44.700		0.854	0.847	0.854	0.319	0.854	0.319	0.854	0.319	0.847	0.847
46.200		0.856	0.849	0.856	0.325	0.856	0.325	0.856	0.325	0.849	0.849
47.700		0.850	0.842	0.850	0.296	0.850	0.296	0.850	0.296	0.842	0.842
49.700		0.860	0.853	0.860	0.344	0.860	0.344	0.860	0.344	0.853	0.853
51.600		0.865	0.859	0.865	0.369	0.865	0.369	0.865	0.369	0.859	0.859
53.500		0.848	0.840	0.848	0.288	0.848	0.288	0.848	0.288	0.840	0.840
54.900		0.842	0.834	0.842	0.258	0.842	0.258	0.842	0.258	0.834	0.834
56.200		0.827	0.819	0.827	0.191	0.827	0.191	0.827	0.191	0.819	0.819
57.600		0.850	0.842	0.850	0.296	0.850	0.296	0.850	0.296	0.842	0.842
59.800		0.891	0.886	0.891	0.491	0.891	0.491	0.891	0.491	0.886	0.886
61.800		0.920	0.916	0.920	0.625	0.920	0.625	0.920	0.625	0.916	0.916
63.800		0.964	0.962	0.964	0.831	0.964	0.831	0.964	0.831	0.962	0.962
66.500		0.966	0.964	0.966	0.839	0.966	0.839	0.966	0.839	0.964	0.964
68.400		0.919	0.915	0.919	0.620	0.919	0.620	0.919	0.620	0.915	0.915
70.100		0.884	0.878	0.884	0.457	0.884	0.457	0.884	0.457	0.878	0.878
71.600		0.873	0.867	0.873	0.406	0.873	0.406	0.873	0.406	0.867	0.867
73.100		0.860	0.853	0.860	0.346	0.860	0.346	0.860	0.346	0.853	0.853
74.600		0.829	0.821	0.829	0.200	0.829	0.200	0.829	0.200	0.821	0.821
76.200		0.813	0.804	0.813	0.124	0.813	0.124	0.813	0.124	0.804	0.804
78.000		0.818	0.809	0.818	0.146	0.818	0.146	0.818	0.146	0.809	0.809
79.500		0.829	0.821	0.829	0.200	0.829	0.200	0.829	0.200	0.821	0.821
81.600		0.839	0.831	0.839	0.246	0.839	0.246	0.839	0.246	0.831	0.831
83.800		0.884	0.878	0.884	0.457	0.884	0.457	0.884	0.457	0.878	0.878
86.000		0.910	0.906	0.910	0.579	0.910	0.579	0.910	0.579	0.906	0.906
88.100		0.945	0.942	0.945	0.741	0.945	0.741	0.945	0.741	0.942	0.942
89.900		0.989	0.988	0.989	0.947	0.989	0.947	0.989	0.947	0.988	0.988
91.500		0.996	0.995	0.996	0.979	0.996	0.979	0.996	0.979	0.995	0.995
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table E.3 (continued)

GAPXL ^a	VALUE OF K IS:	11	12	13	14	15	16	17	18	19	20
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.010
1.800		0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.961	0.961	0.826
3.400		0.642	0.642	0.642	0.642	0.642	0.642	0.642	0.920	0.920	0.642
5.000		0.654	0.654	0.654	0.654	0.654	0.654	0.654	0.922	0.922	0.654
6.900		0.732	0.732	0.732	0.732	0.732	0.732	0.732	0.940	0.940	0.732
8.800		0.880	0.880	0.880	0.880	0.880	0.880	0.880	0.973	0.973	0.880
11.500		0.875	0.875	0.875	0.875	0.875	0.875	0.875	0.972	0.972	0.875
13.300		0.709	0.709	0.709	0.709	0.709	0.709	0.709	0.935	0.935	0.709
15.100		0.606	0.606	0.606	0.606	0.606	0.606	0.606	0.912	0.912	0.606
16.800		0.508	0.508	0.508	0.508	0.508	0.508	0.508	0.890	0.890	0.508
18.100		0.459	0.459	0.459	0.459	0.459	0.459	0.459	0.879	0.879	0.459
19.500		0.383	0.383	0.383	0.383	0.383	0.383	0.383	0.862	0.862	0.383
21.400		0.415	0.415	0.415	0.415	0.415	0.415	0.415	0.869	0.869	0.415
23.200		0.412	0.412	0.412	0.412	0.412	0.412	0.412	0.868	0.868	0.412
25.000		0.430	0.430	0.430	0.430	0.430	0.430	0.430	0.872	0.872	0.430
26.900		0.469	0.469	0.469	0.469	0.469	0.469	0.469	0.881	0.881	0.469
28.500		0.443	0.443	0.443	0.443	0.443	0.443	0.443	0.875	0.875	0.443
30.000		0.449	0.449	0.449	0.449	0.449	0.449	0.449	0.877	0.877	0.449
32.000		0.462	0.462	0.462	0.462	0.462	0.462	0.462	0.880	0.880	0.462
34.000		0.478	0.478	0.478	0.478	0.478	0.478	0.478	0.883	0.883	0.478
35.900		0.423	0.423	0.423	0.423	0.423	0.423	0.423	0.871	0.871	0.423
37.700		0.350	0.350	0.350	0.350	0.350	0.350	0.350	0.854	0.854	0.350
39.500		0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.866	0.866	0.403
41.200		0.310	0.310	0.310	0.310	0.310	0.310	0.310	0.845	0.845	0.310
43.300		0.294	0.294	0.294	0.294	0.294	0.294	0.294	0.842	0.842	0.294
44.700		0.319	0.319	0.319	0.319	0.319	0.319	0.319	0.847	0.847	0.319
46.200		0.325	0.325	0.325	0.325	0.325	0.325	0.325	0.849	0.849	0.325
47.700		0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296
49.700		0.344	0.344	0.344	0.344	0.344	0.344	0.344	0.853	0.853	0.344
51.600		0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.859	0.859	0.369
53.500		0.288	0.288	0.288	0.288	0.288	0.288	0.288	0.840	0.840	0.288
54.900		0.258	0.258	0.258	0.258	0.258	0.258	0.258	0.834	0.834	0.258
56.200		0.191	0.191	0.191	0.191	0.191	0.191	0.191	0.819	0.819	0.191
57.600		0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296
59.800		0.491	0.491	0.491	0.491	0.491	0.491	0.491	0.886	0.886	0.491
61.800		0.625	0.625	0.625	0.625	0.625	0.625	0.625	0.916	0.916	0.625
63.800		0.831	0.831	0.831	0.831	0.831	0.831	0.831	0.962	0.962	0.831
66.500		0.839	0.839	0.839	0.839	0.839	0.839	0.839	0.964	0.964	0.839
68.400		0.620	0.620	0.620	0.620	0.620	0.620	0.620	0.915	0.915	0.620
70.100		0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457
71.600		0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.867	0.867	0.406
73.100		0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346
74.600		0.200	0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.821	0.200
76.200		0.124	0.124	0.124	0.124	0.124	0.124	0.124	0.804	0.804	0.124
78.000		0.146	0.146	0.146	0.146	0.146	0.146	0.146	0.809	0.809	0.146
79.500		0.200	0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.821	0.200
81.600		0.246	0.246	0.246	0.246	0.246	0.246	0.246	0.831	0.831	0.246
83.800		0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457
86.000		0.579	0.579	0.579	0.579	0.579	0.579	0.579	0.906	0.906	0.579
88.100		0.741	0.741	0.741	0.741	0.741	0.741	0.741	0.942	0.942	0.741
89.900		0.947	0.947	0.947	0.947	0.947	0.947	0.947	0.988	0.988	0.947
91.500		0.979	0.979	0.979	0.979	0.979	0.979	0.979	0.995	0.995	0.979
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table E.3 (continued)

GAPXL ^a	VALUE OF K IS:	21	22	23	24	25	26	27	28	29	30
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.010	1.010
1.800		0.826	0.826	0.826	0.826	0.826	0.826	0.961	0.961	0.826	0.826
3.400		0.642	0.642	0.642	0.642	0.642	0.642	0.920	0.920	0.642	0.642
5.000		0.654	0.654	0.654	0.654	0.654	0.654	0.922	0.922	0.654	0.654
6.900		0.732	0.732	0.732	0.732	0.732	0.732	0.940	0.940	0.732	0.732
8.800		0.880	0.880	0.880	0.880	0.880	0.880	0.973	0.973	0.880	0.880
11.500		0.875	0.875	0.875	0.875	0.875	0.875	0.972	0.972	0.875	0.875
13.300		0.709	0.709	0.709	0.709	0.709	0.709	0.935	0.935	0.709	0.709
15.100		0.606	0.606	0.606	0.606	0.606	0.606	0.912	0.912	0.606	0.606
16.800		0.508	0.508	0.508	0.508	0.508	0.508	0.890	0.890	0.508	0.508
18.100		0.459	0.459	0.459	0.459	0.459	0.459	0.879	0.879	0.459	0.459
19.500		0.383	0.383	0.383	0.383	0.383	0.383	0.862	0.862	0.383	0.383
21.400		0.415	0.415	0.415	0.415	0.415	0.415	0.869	0.869	0.415	0.415
23.200		0.412	0.412	0.412	0.412	0.412	0.412	0.868	0.868	0.412	0.412
25.000		0.430	0.430	0.430	0.430	0.430	0.430	0.872	0.872	0.430	0.430
26.900		0.469	0.469	0.469	0.469	0.469	0.469	0.881	0.881	0.469	0.469
28.500		0.443	0.443	0.443	0.443	0.443	0.443	0.875	0.875	0.443	0.443
30.000		0.449	0.449	0.449	0.449	0.449	0.449	0.877	0.877	0.449	0.449
32.000		0.462	0.462	0.462	0.462	0.462	0.462	0.890	0.890	0.462	0.462
34.000		0.478	0.478	0.478	0.478	0.478	0.478	0.883	0.883	0.478	0.478
35.900		0.423	0.423	0.423	0.423	0.423	0.423	0.871	0.871	0.423	0.423
37.700		0.350	0.350	0.350	0.350	0.350	0.350	0.854	0.854	0.350	0.350
39.500		0.403	0.403	0.403	0.403	0.403	0.403	0.866	0.866	0.403	0.403
41.200		0.310	0.310	0.310	0.310	0.310	0.310	0.845	0.845	0.310	0.310
43.300		0.294	0.294	0.294	0.294	0.294	0.294	0.842	0.842	0.294	0.294
44.700		0.319	0.319	0.319	0.319	0.319	0.319	0.847	0.847	0.319	0.319
46.200		0.325	0.325	0.325	0.325	0.325	0.325	0.849	0.849	0.325	0.325
47.700		0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296	0.296
49.700		0.344	0.344	0.344	0.344	0.344	0.344	0.853	0.853	0.344	0.344
51.600		0.369	0.369	0.369	0.369	0.369	0.369	0.859	0.859	0.369	0.369
53.500		0.288	0.288	0.288	0.288	0.288	0.288	0.840	0.840	0.288	0.288
54.900		0.258	0.258	0.258	0.258	0.258	0.258	0.834	0.834	0.258	0.258
56.200		0.191	0.191	0.191	0.191	0.191	0.191	0.819	0.819	0.191	0.191
57.600		0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296	0.296
59.800		0.491	0.491	0.491	0.491	0.491	0.491	0.886	0.886	0.491	0.491
61.800		0.625	0.625	0.625	0.625	0.625	0.625	0.916	0.916	0.625	0.625
63.800		0.831	0.831	0.831	0.831	0.831	0.831	0.962	0.962	0.831	0.831
66.500		0.839	0.839	0.839	0.839	0.839	0.839	0.964	0.964	0.839	0.839
68.400		0.620	0.620	0.620	0.620	0.620	0.620	0.915	0.915	0.620	0.620
70.100		0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457	0.457
71.600		0.406	0.406	0.406	0.406	0.406	0.406	0.867	0.867	0.406	0.406
73.100		0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346	0.346
74.600		0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.821	0.200	0.200
76.200		0.124	0.124	0.124	0.124	0.124	0.124	0.804	0.804	0.124	0.124
78.000		0.146	0.146	0.146	0.146	0.146	0.146	0.809	0.809	0.146	0.146
79.500		0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.821	0.200	0.200
81.600		0.246	0.246	0.246	0.246	0.246	0.246	0.831	0.831	0.246	0.246
83.800		0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457	0.457
86.000		0.579	0.579	0.579	0.579	0.579	0.579	0.906	0.906	0.579	0.579
88.100		0.741	0.741	0.741	0.741	0.741	0.741	0.942	0.942	0.741	0.741
89.900		0.947	0.947	0.947	0.947	0.947	0.947	0.988	0.988	0.947	0.947
91.500		0.979	0.979	0.979	0.979	0.979	0.979	0.995	0.995	0.979	0.979
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table E.3 (continued)

GAPXL ^a	VALUE OF K IS:	31	32	33	34	35	36	37	38	39	40
-47.676	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000	1.010	1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.002	1.002	1.002
1.800	0.826	0.826	0.826	0.826	0.826	0.826	0.961	0.963	0.963	0.963	0.963
3.400	0.642	0.642	0.642	0.642	0.642	0.642	0.920	0.923	0.923	0.923	0.923
5.000	0.654	0.654	0.654	0.654	0.654	0.654	0.922	0.926	0.926	0.926	0.926
6.900	0.732	0.732	0.732	0.732	0.732	0.732	0.940	0.943	0.943	0.943	0.943
8.800	0.880	0.880	0.880	0.880	0.880	0.880	0.973	0.974	0.974	0.974	0.974
11.500	0.875	0.875	0.875	0.875	0.875	0.875	0.972	0.973	0.973	0.973	0.973
13.300	0.709	0.709	0.709	0.709	0.709	0.709	0.935	0.938	0.938	0.938	0.938
15.100	0.606	0.606	0.606	0.606	0.606	0.606	0.912	0.916	0.916	0.916	0.916
16.800	0.508	0.508	0.508	0.508	0.508	0.508	0.890	0.895	0.895	0.895	0.895
18.100	0.459	0.459	0.459	0.459	0.459	0.459	0.879	0.884	0.884	0.884	0.884
19.500	0.383	0.383	0.383	0.383	0.383	0.383	0.862	0.868	0.868	0.868	0.868
21.400	0.415	0.415	0.415	0.415	0.415	0.415	0.869	0.875	0.875	0.875	0.875
23.200	0.412	0.412	0.412	0.412	0.412	0.412	0.868	0.874	0.874	0.874	0.874
25.000	0.430	0.430	0.430	0.430	0.430	0.430	0.872	0.878	0.878	0.878	0.878
26.900	0.469	0.469	0.469	0.469	0.469	0.469	0.881	0.886	0.886	0.886	0.886
28.500	0.443	0.443	0.443	0.443	0.443	0.443	0.875	0.881	0.881	0.881	0.881
30.000	0.449	0.449	0.449	0.449	0.449	0.449	0.877	0.882	0.882	0.882	0.882
32.000	0.462	0.462	0.462	0.462	0.462	0.462	0.880	0.885	0.885	0.885	0.885
34.000	0.478	0.478	0.478	0.478	0.478	0.478	0.881	0.889	0.889	0.889	0.889
35.900	0.423	0.423	0.423	0.423	0.423	0.423	0.871	0.877	0.877	0.877	0.877
37.700	0.350	0.350	0.350	0.350	0.350	0.350	0.854	0.861	0.861	0.861	0.861
39.500	0.403	0.403	0.403	0.403	0.403	0.403	0.866	0.872	0.872	0.872	0.872
41.200	0.310	0.310	0.310	0.310	0.310	0.310	0.845	0.852	0.852	0.852	0.852
43.300	0.294	0.294	0.294	0.294	0.294	0.294	0.842	0.849	0.849	0.849	0.849
44.700	0.319	0.319	0.319	0.319	0.319	0.319	0.847	0.854	0.854	0.854	0.854
46.200	0.325	0.325	0.325	0.325	0.325	0.325	0.849	0.856	0.856	0.856	0.856
47.700	0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.850	0.850	0.850	0.850
49.700	0.344	0.344	0.344	0.344	0.344	0.344	0.853	0.860	0.860	0.860	0.860
51.600	0.369	0.369	0.369	0.369	0.369	0.369	0.859	0.865	0.865	0.865	0.865
53.500	0.288	0.288	0.288	0.288	0.288	0.288	0.840	0.848	0.848	0.848	0.848
54.900	0.258	0.258	0.258	0.258	0.258	0.258	0.834	0.842	0.842	0.842	0.842
56.200	0.191	0.191	0.191	0.191	0.191	0.191	0.819	0.827	0.827	0.827	0.827
57.600	0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.850	0.850	0.850	0.850
59.800	0.491	0.491	0.491	0.491	0.491	0.491	0.886	0.891	0.891	0.891	0.891
61.800	0.625	0.625	0.625	0.625	0.625	0.625	0.916	0.920	0.920	0.920	0.920
63.800	0.831	0.831	0.831	0.831	0.831	0.831	0.962	0.964	0.964	0.964	0.964
66.500	0.839	0.839	0.839	0.839	0.839	0.839	0.964	0.966	0.966	0.966	0.966
68.400	0.620	0.620	0.620	0.620	0.620	0.620	0.915	0.919	0.919	0.919	0.919
70.100	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.884	0.884	0.884	0.884
71.600	0.406	0.406	0.406	0.406	0.406	0.406	0.867	0.873	0.873	0.873	0.873
73.100	0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.860	0.860	0.860	0.860
74.600	0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.829	0.829	0.829	0.829
76.200	0.124	0.124	0.124	0.124	0.124	0.124	0.804	0.813	0.813	0.813	0.813
78.000	0.146	0.146	0.146	0.146	0.146	0.146	0.809	0.818	0.818	0.818	0.818
79.500	0.200	0.200	0.200	0.200	0.200	0.200	0.821	0.829	0.829	0.829	0.829
81.600	0.246	0.246	0.246	0.246	0.246	0.246	0.831	0.839	0.839	0.839	0.839
83.800	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.884	0.884	0.884	0.884
86.000	0.579	0.579	0.579	0.579	0.579	0.579	0.906	0.910	0.910	0.910	0.910
88.100	0.741	0.741	0.741	0.741	0.741	0.741	0.942	0.945	0.945	0.945	0.945
89.900	0.947	0.947	0.947	0.947	0.947	0.947	0.988	0.989	0.989	0.989	0.989
91.500	0.979	0.979	0.979	0.979	0.979	0.979	0.995	0.996	0.996	0.996	0.996
95.987	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

Appendix F

B-2/SHROUD 2 BUNDLE-AVERAGED GEOMETRY DATA,
MAXIMUM RESTRICTION DEFINITION

The data in Appendix F represent the input from card groups 5 and 6 as required by ref. 2 for the B-2/shroud 2 COBRA-IV model based on the maximum restriction definition.



Table F.1. Shroud flow area reductions for the B-2/shroud 2 COBRA-IV model based on the maximum restriction definition

AXL ^a	FRACTIONAL SHROUD FLOW AREA
-47.676	1.000
0.000	1.003
1.800	0.950
3.400	0.894
5.000	0.898
6.900	0.922
8.800	0.966
11.500	0.964
13.300	0.915
15.100	0.883
16.800	0.851
18.100	0.835
19.500	0.797
21.400	0.821
23.200	0.820
25.000	0.826
26.900	0.838
28.500	0.830
30.000	0.832
32.000	0.836
34.000	0.842
35.900	0.823
37.700	0.794
39.500	0.817
41.200	0.766
43.300	0.756
44.700	0.788
46.200	0.790
47.700	0.770
49.700	0.797
51.600	0.805
53.500	0.751
54.900	0.767
56.200	0.700
57.600	0.780
59.800	0.846
61.800	0.889
63.800	0.951
66.500	0.954
68.400	0.887
70.100	0.835
71.600	0.818
73.100	0.797
74.600	0.739
76.200	0.661
78.000	0.715
79.500	0.742
81.600	0.730
83.800	0.835
86.000	0.874
88.100	0.924
89.900	0.985
91.500	0.994
95.987	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table F.2. AFACT values for the B-2/shroud 2 COBRA-IV model based on the maximum restriction definition

AXL ^a	VALUE OF I IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001	1.002	1.005	1.005	1.005	1.002
1.800		0.983	0.964	0.964	0.964	0.983	0.963	0.906	0.906	0.906	0.963
3.400		0.964	0.924	0.924	0.924	0.964	0.921	0.801	0.801	0.801	0.921
5.000		0.965	0.927	0.927	0.927	0.965	0.924	0.808	0.808	0.808	0.924
6.900		0.973	0.944	0.944	0.944	0.973	0.942	0.853	0.853	0.853	0.942
8.800		0.988	0.976	0.976	0.976	0.988	0.975	0.936	0.936	0.936	0.975
11.500		0.988	0.974	0.974	0.974	0.988	0.974	0.933	0.933	0.933	0.974
13.300		0.971	0.939	0.939	0.939	0.971	0.937	0.840	0.840	0.840	0.937
15.100		0.960	0.916	0.916	0.916	0.960	0.913	0.780	0.780	0.780	0.913
16.800		0.950	0.894	0.894	0.894	0.950	0.890	0.721	0.721	0.721	0.890
18.100		0.944	0.882	0.882	0.882	0.944	0.878	0.691	0.691	0.691	0.878
19.500		0.931	0.855	0.855	0.855	0.931	0.850	0.620	0.620	0.620	0.850
21.400		0.939	0.872	0.872	0.872	0.939	0.867	0.664	0.664	0.664	0.867
23.200		0.939	0.871	0.871	0.871	0.939	0.866	0.662	0.662	0.662	0.866
25.000		0.941	0.876	0.876	0.876	0.941	0.871	0.673	0.673	0.673	0.871
26.900		0.945	0.885	0.885	0.885	0.945	0.880	0.697	0.697	0.697	0.880
28.500		0.942	0.879	0.879	0.879	0.942	0.874	0.681	0.681	0.681	0.874
30.000		0.943	0.880	0.880	0.880	0.943	0.876	0.685	0.685	0.685	0.876
32.000		0.944	0.883	0.883	0.883	0.944	0.879	0.693	0.693	0.693	0.879
34.000		0.946	0.887	0.887	0.887	0.946	0.883	0.703	0.703	0.703	0.883
35.900		0.940	0.874	0.874	0.874	0.940	0.869	0.669	0.669	0.669	0.869
37.700		0.930	0.853	0.853	0.853	0.930	0.847	0.614	0.614	0.614	0.847
39.500		0.938	0.869	0.869	0.869	0.938	0.864	0.656	0.656	0.656	0.864
41.200		0.921	0.833	0.833	0.833	0.921	0.827	0.561	0.561	0.561	0.827
43.300		0.917	0.826	0.826	0.826	0.917	0.819	0.542	0.542	0.542	0.819
44.700		0.928	0.849	0.849	0.849	0.928	0.843	0.603	0.603	0.603	0.843
46.200		0.929	0.850	0.850	0.850	0.929	0.845	0.607	0.607	0.607	0.845
47.700		0.922	0.836	0.836	0.836	0.922	0.830	0.569	0.569	0.569	0.830
49.700		0.931	0.855	0.855	0.855	0.931	0.849	0.619	0.619	0.619	0.849
51.600		0.934	0.861	0.861	0.861	0.934	0.856	0.635	0.635	0.635	0.856
53.500		0.916	0.823	0.823	0.823	0.916	0.816	0.534	0.534	0.534	0.816
54.900		0.921	0.834	0.834	0.834	0.921	0.828	0.564	0.564	0.564	0.828
56.200		0.898	0.786	0.786	0.786	0.898	0.778	0.438	0.438	0.438	0.778
57.600		0.925	0.843	0.843	0.843	0.925	0.837	0.588	0.588	0.588	0.837
59.800		0.948	0.890	0.890	0.890	0.948	0.886	0.711	0.711	0.711	0.886
61.800		0.962	0.920	0.920	0.920	0.962	0.917	0.791	0.791	0.791	0.917
63.800		0.984	0.965	0.965	0.965	0.984	0.964	0.909	0.909	0.909	0.964
66.500		0.984	0.967	0.967	0.967	0.984	0.966	0.913	0.913	0.913	0.966
68.400		0.962	0.919	0.919	0.919	0.962	0.916	0.788	0.788	0.788	0.916
70.100		0.944	0.882	0.882	0.882	0.944	0.878	0.690	0.690	0.690	0.878
71.600		0.938	0.870	0.870	0.870	0.938	0.865	0.658	0.658	0.658	0.865
73.100		0.931	0.855	0.855	0.855	0.931	0.850	0.620	0.620	0.620	0.850
74.600		0.912	0.814	0.814	0.814	0.912	0.807	0.511	0.511	0.511	0.807
76.200		0.885	0.758	0.758	0.758	0.885	0.749	0.364	0.364	0.364	0.749
78.000		0.903	0.796	0.796	0.796	0.903	0.789	0.465	0.465	0.465	0.789
79.500		0.912	0.816	0.816	0.816	0.912	0.809	0.516	0.516	0.516	0.809
81.600		0.908	0.807	0.807	0.807	0.908	0.800	0.494	0.494	0.494	0.800
83.800		0.944	0.882	0.882	0.882	0.944	0.878	0.690	0.690	0.690	0.878
86.000		0.957	0.910	0.910	0.910	0.957	0.907	0.764	0.764	0.764	0.907
88.100		0.974	0.946	0.946	0.946	0.974	0.944	0.858	0.858	0.858	0.944
89.900		0.995	0.989	0.989	0.989	0.995	0.989	0.972	0.972	0.972	0.989
91.500		0.998	0.996	0.996	0.996	0.998	0.996	0.989	0.989	0.989	0.996
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table F.2 (continued)

AXL ^a	VALUE OF I IS:	11	12	13	14	15	16	17	18	19	20
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.002	1.005	1.005	1.005	1.002	1.002	1.005	1.005	1.005	1.002
1.800		0.963	0.906	0.906	0.906	0.963	0.963	0.906	0.906	0.906	0.963
3.400		0.921	0.801	0.801	0.801	0.921	0.921	0.801	0.801	0.801	0.921
5.000		0.924	0.808	0.808	0.808	0.924	0.924	0.808	0.808	0.808	0.924
6.900		0.942	0.853	0.853	0.853	0.942	0.942	0.853	0.853	0.853	0.942
8.800		0.975	0.936	0.936	0.936	0.975	0.975	0.936	0.936	0.936	0.975
11.500		0.974	0.933	0.933	0.933	0.974	0.974	0.933	0.933	0.933	0.974
13.300		0.937	0.840	0.840	0.840	0.937	0.937	0.840	0.840	0.840	0.937
15.100		0.913	0.780	0.780	0.780	0.913	0.913	0.780	0.780	0.780	0.913
16.800		0.890	0.721	0.721	0.721	0.890	0.890	0.721	0.721	0.721	0.890
18.100		0.878	0.691	0.691	0.691	0.878	0.878	0.691	0.691	0.691	0.878
19.500		0.850	0.620	0.620	0.620	0.850	0.850	0.620	0.620	0.620	0.850
21.400		0.867	0.664	0.664	0.664	0.867	0.867	0.664	0.664	0.664	0.867
23.200		0.866	0.662	0.662	0.662	0.866	0.866	0.662	0.662	0.662	0.866
25.000		0.871	0.673	0.673	0.673	0.871	0.871	0.673	0.673	0.673	0.871
26.900		0.880	0.697	0.697	0.697	0.880	0.880	0.697	0.697	0.697	0.880
28.500		0.874	0.681	0.681	0.681	0.874	0.874	0.681	0.681	0.681	0.874
30.000		0.876	0.685	0.685	0.685	0.876	0.876	0.685	0.685	0.685	0.876
32.000		0.879	0.693	0.693	0.693	0.879	0.879	0.693	0.693	0.693	0.879
34.000		0.883	0.703	0.703	0.703	0.883	0.883	0.703	0.703	0.703	0.883
35.900		0.869	0.669	0.669	0.669	0.869	0.869	0.669	0.669	0.669	0.869
37.700		0.847	0.614	0.614	0.614	0.847	0.847	0.614	0.614	0.614	0.847
39.500		0.864	0.656	0.656	0.656	0.864	0.864	0.656	0.656	0.656	0.864
41.200		0.827	0.561	0.561	0.561	0.827	0.827	0.561	0.561	0.561	0.827
43.300		0.819	0.542	0.542	0.542	0.819	0.819	0.542	0.542	0.542	0.819
44.700		0.843	0.603	0.603	0.603	0.843	0.843	0.603	0.603	0.603	0.843
46.200		0.845	0.607	0.607	0.607	0.845	0.845	0.607	0.607	0.607	0.845
47.700		0.830	0.569	0.569	0.569	0.830	0.830	0.569	0.569	0.569	0.830
49.700		0.849	0.619	0.619	0.619	0.849	0.849	0.619	0.619	0.619	0.849
51.600		0.856	0.635	0.635	0.635	0.856	0.856	0.635	0.635	0.635	0.856
53.500		0.816	0.534	0.534	0.534	0.816	0.816	0.534	0.534	0.534	0.816
54.900		0.828	0.564	0.564	0.564	0.828	0.828	0.564	0.564	0.564	0.828
56.200		0.778	0.438	0.438	0.438	0.778	0.778	0.438	0.438	0.438	0.778
57.600		0.837	0.588	0.588	0.588	0.837	0.837	0.588	0.588	0.588	0.837
59.800		0.886	0.711	0.711	0.711	0.886	0.886	0.711	0.711	0.711	0.886
61.800		0.917	0.791	0.791	0.791	0.917	0.917	0.791	0.791	0.791	0.917
63.800		0.964	0.909	0.909	0.909	0.964	0.964	0.909	0.909	0.909	0.964
66.500		0.966	0.913	0.913	0.913	0.966	0.966	0.913	0.913	0.913	0.966
68.400		0.916	0.788	0.788	0.788	0.916	0.916	0.788	0.788	0.788	0.916
70.100		0.878	0.690	0.690	0.690	0.878	0.878	0.690	0.690	0.690	0.878
71.600		0.865	0.658	0.658	0.658	0.865	0.865	0.658	0.658	0.658	0.865
73.100		0.850	0.620	0.620	0.620	0.850	0.850	0.620	0.620	0.620	0.850
74.600		0.807	0.511	0.511	0.511	0.807	0.807	0.511	0.511	0.511	0.807
76.200		0.749	0.364	0.364	0.364	0.749	0.749	0.364	0.364	0.364	0.749
78.000		0.789	0.465	0.465	0.465	0.789	0.789	0.465	0.465	0.465	0.789
79.500		0.809	0.516	0.516	0.516	0.809	0.809	0.516	0.516	0.516	0.809
81.600		0.800	0.494	0.494	0.494	0.800	0.800	0.494	0.494	0.494	0.800
83.800		0.878	0.690	0.690	0.690	0.878	0.878	0.690	0.690	0.690	0.878
86.000		0.907	0.764	0.764	0.764	0.907	0.907	0.764	0.764	0.764	0.907
88.100		0.944	0.858	0.858	0.858	0.944	0.944	0.858	0.858	0.858	0.944
89.900		0.989	0.972	0.972	0.972	0.989	0.989	0.972	0.972	0.972	0.989
91.500		0.996	0.989	0.989	0.989	0.996	0.996	0.989	0.989	0.989	0.996
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table F.2 (continued)

AXL ^a	VALUE OF I IS:	21	22	23	24	25
-47.676		1.000	1.000	1.000	1.000	1.000
0.000		1.001	1.002	1.002	1.002	1.001
1.800		0.983	0.964	0.964	0.964	0.983
3.400		0.964	0.924	0.924	0.924	0.964
5.000		0.965	0.927	0.927	0.927	0.965
6.900		0.973	0.944	0.944	0.944	0.973
8.800		0.988	0.976	0.976	0.976	0.988
11.500		0.988	0.974	0.974	0.974	0.988
13.300		0.971	0.939	0.939	0.939	0.971
15.100		0.960	0.916	0.916	0.916	0.960
16.800		0.950	0.894	0.894	0.894	0.950
18.100		0.944	0.882	0.882	0.882	0.944
19.500		0.931	0.855	0.855	0.855	0.931
21.400		0.939	0.872	0.872	0.872	0.939
23.200		0.939	0.871	0.871	0.871	0.939
25.000		0.941	0.876	0.876	0.876	0.941
26.900		0.945	0.885	0.885	0.885	0.945
28.500		0.942	0.879	0.879	0.879	0.942
30.000		0.943	0.880	0.880	0.880	0.943
32.000		0.944	0.883	0.883	0.883	0.944
34.000		0.946	0.887	0.887	0.887	0.946
35.900		0.940	0.874	0.874	0.874	0.940
37.700		0.930	0.853	0.853	0.853	0.930
39.500		0.938	0.869	0.869	0.869	0.938
41.200		0.921	0.833	0.833	0.833	0.921
43.300		0.917	0.826	0.826	0.826	0.917
44.700		0.928	0.849	0.849	0.849	0.928
46.200		0.929	0.850	0.850	0.850	0.929
47.700		0.922	0.836	0.836	0.836	0.922
49.700		0.931	0.855	0.855	0.855	0.931
51.600		0.934	0.861	0.861	0.861	0.934
53.500		0.916	0.823	0.823	0.823	0.916
54.900		0.921	0.834	0.834	0.834	0.921
56.200		0.898	0.786	0.786	0.786	0.898
57.600		0.925	0.843	0.843	0.843	0.925
59.800		0.948	0.890	0.890	0.890	0.948
61.800		0.962	0.920	0.920	0.920	0.962
63.800		0.984	0.965	0.965	0.965	0.984
66.500		0.984	0.967	0.967	0.967	0.984
68.400		0.962	0.919	0.919	0.919	0.962
70.100		0.944	0.882	0.882	0.882	0.944
71.600		0.938	0.870	0.870	0.870	0.938
73.100		0.931	0.855	0.855	0.855	0.931
74.600		0.912	0.814	0.814	0.814	0.912
76.200		0.885	0.758	0.758	0.758	0.885
78.000		0.903	0.796	0.796	0.796	0.903
79.500		0.912	0.816	0.816	0.816	0.912
81.600		0.908	0.807	0.807	0.807	0.908
83.800		0.944	0.882	0.882	0.882	0.944
86.000		0.957	0.910	0.910	0.910	0.957
88.100		0.974	0.946	0.946	0.946	0.974
89.900		0.995	0.989	0.989	0.989	0.995
91.500		0.998	0.996	0.996	0.996	0.998
95.987		1.000	1.000	1.000	1.000	1.000

^aAXL values are the centimeters from the bottom of the heated zone.

Table F.3. GFACT values for the B-2/shroud 2 COBRA-IV model based on the maximum restriction definition

GAPXL ^a	VALUE OF K IS:	1	2	3	4	5	6	7	8	9	10
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.002	1.002	1.002	1.010	1.002	1.010	1.002	1.010	1.002	1.002
1.800		0.963	0.961	0.963	0.826	0.963	0.826	0.963	0.826	0.961	0.961
3.400		0.923	0.920	0.923	0.642	0.923	0.642	0.923	0.642	0.920	0.920
5.000		0.926	0.922	0.926	0.654	0.926	0.654	0.926	0.654	0.922	0.922
6.900		0.943	0.940	0.943	0.732	0.943	0.732	0.943	0.732	0.940	0.940
8.800		0.974	0.973	0.974	0.880	0.974	0.880	0.974	0.880	0.973	0.973
11.500		0.973	0.972	0.973	0.875	0.973	0.875	0.973	0.875	0.972	0.972
13.300		0.938	0.935	0.938	0.709	0.938	0.709	0.938	0.709	0.935	0.935
15.100		0.916	0.912	0.916	0.606	0.916	0.606	0.916	0.606	0.912	0.912
16.800		0.895	0.890	0.895	0.508	0.895	0.508	0.895	0.508	0.890	0.890
18.100		0.884	0.879	0.884	0.459	0.884	0.459	0.884	0.459	0.879	0.879
19.500		0.860	0.853	0.860	0.346	0.860	0.346	0.860	0.346	0.853	0.853
21.400		0.875	0.869	0.875	0.415	0.875	0.415	0.875	0.415	0.869	0.869
23.200		0.874	0.868	0.874	0.412	0.874	0.412	0.874	0.412	0.868	0.868
25.000		0.878	0.872	0.878	0.430	0.878	0.430	0.878	0.430	0.872	0.872
26.900		0.886	0.881	0.886	0.469	0.886	0.469	0.886	0.469	0.881	0.881
28.500		0.881	0.875	0.881	0.443	0.881	0.443	0.881	0.443	0.875	0.875
30.000		0.882	0.877	0.882	0.449	0.882	0.449	0.882	0.449	0.877	0.877
32.000		0.885	0.880	0.885	0.462	0.885	0.462	0.885	0.462	0.880	0.880
34.000		0.889	0.883	0.889	0.478	0.889	0.478	0.889	0.478	0.883	0.883
35.900		0.877	0.871	0.877	0.423	0.877	0.423	0.877	0.423	0.871	0.871
37.700		0.858	0.851	0.858	0.336	0.858	0.336	0.858	0.336	0.851	0.851
39.500		0.872	0.866	0.872	0.403	0.872	0.403	0.872	0.403	0.866	0.866
41.200		0.841	0.833	0.841	0.254	0.841	0.254	0.841	0.254	0.833	0.833
43.300		0.834	0.826	0.834	0.225	0.834	0.225	0.834	0.225	0.826	0.826
44.700		0.854	0.847	0.854	0.319	0.854	0.319	0.854	0.319	0.847	0.847
46.200		0.856	0.849	0.856	0.325	0.856	0.325	0.856	0.325	0.849	0.849
47.700		0.843	0.836	0.843	0.266	0.843	0.266	0.843	0.266	0.836	0.836
49.700		0.860	0.853	0.860	0.344	0.860	0.344	0.860	0.344	0.853	0.853
51.600		0.865	0.859	0.865	0.369	0.865	0.369	0.865	0.369	0.859	0.859
53.500		0.832	0.824	0.832	0.213	0.832	0.213	0.832	0.213	0.824	0.824
54.900		0.842	0.834	0.842	0.258	0.842	0.258	0.842	0.258	0.834	0.834
56.200		0.801	0.792	0.801	0.069	0.801	0.069	0.801	0.069	0.792	0.792
57.600		0.850	0.842	0.850	0.296	0.850	0.296	0.850	0.296	0.842	0.842
59.800		0.891	0.886	0.891	0.491	0.891	0.491	0.891	0.491	0.886	0.886
61.800		0.920	0.916	0.920	0.625	0.920	0.625	0.920	0.625	0.916	0.916
63.800		0.964	0.962	0.964	0.831	0.964	0.831	0.964	0.831	0.962	0.962
66.500		0.966	0.964	0.966	0.839	0.966	0.839	0.966	0.839	0.964	0.964
68.400		0.919	0.915	0.919	0.620	0.919	0.620	0.919	0.620	0.915	0.915
70.100		0.884	0.878	0.884	0.457	0.884	0.457	0.884	0.457	0.878	0.878
71.600		0.873	0.867	0.873	0.406	0.873	0.406	0.873	0.406	0.867	0.867
73.100		0.860	0.853	0.860	0.346	0.860	0.346	0.860	0.346	0.853	0.853
74.600		0.824	0.816	0.824	0.178	0.824	0.178	0.824	0.178	0.816	0.816
76.200		0.778	0.768	0.778	0.0	0.778	0.0	0.778	0.0	0.768	0.768
78.000		0.810	0.800	0.810	0.109	0.810	0.109	0.810	0.109	0.800	0.800
79.500		0.826	0.817	0.826	0.185	0.826	0.185	0.826	0.185	0.817	0.817
81.600		0.819	0.810	0.819	0.152	0.819	0.152	0.819	0.152	0.810	0.810
83.800		0.884	0.878	0.884	0.457	0.884	0.457	0.884	0.457	0.878	0.878
86.000		0.910	0.906	0.910	0.579	0.910	0.579	0.910	0.579	0.906	0.906
88.100		0.945	0.942	0.945	0.741	0.945	0.741	0.945	0.741	0.942	0.942
89.900		0.989	0.988	0.989	0.947	0.989	0.947	0.989	0.947	0.988	0.988
91.500		0.996	0.995	0.996	0.979	0.996	0.979	0.996	0.979	0.995	0.995
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table F.3 (continued)

GAPXL ^a	VALUE OF K IS:	11	12	13	14	15	16	17	18	19	20
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.010
1.800		0.826	0.826	0.826	0.826	0.826	0.826	0.826	0.961	0.961	0.826
3.400		0.642	0.642	0.642	0.642	0.642	0.642	0.642	0.920	0.920	0.642
5.000		0.654	0.654	0.654	0.654	0.654	0.654	0.654	0.922	0.922	0.654
6.900		0.732	0.732	0.732	0.732	0.732	0.732	0.732	0.940	0.940	0.732
8.800		0.880	0.880	0.880	0.880	0.880	0.880	0.880	0.973	0.973	0.880
11.500		0.875	0.875	0.875	0.875	0.875	0.875	0.875	0.972	0.972	0.875
13.300		0.709	0.709	0.709	0.709	0.709	0.709	0.709	0.935	0.935	0.709
15.100		0.606	0.606	0.606	0.606	0.606	0.606	0.606	0.912	0.912	0.606
16.800		0.508	0.508	0.508	0.508	0.508	0.508	0.508	0.850	0.850	0.508
18.100		0.459	0.459	0.459	0.459	0.459	0.459	0.459	0.879	0.879	0.459
19.500		0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346
21.400		0.415	0.415	0.415	0.415	0.415	0.415	0.415	0.869	0.869	0.415
23.200		0.412	0.412	0.412	0.412	0.412	0.412	0.412	0.868	0.868	0.412
25.000		0.430	0.430	0.430	0.430	0.430	0.430	0.430	0.872	0.872	0.430
26.900		0.469	0.469	0.469	0.469	0.469	0.469	0.469	0.881	0.881	0.469
28.500		0.443	0.443	0.443	0.443	0.443	0.443	0.443	0.875	0.875	0.443
30.000		0.449	0.449	0.449	0.449	0.449	0.449	0.449	0.877	0.877	0.449
32.000		0.462	0.462	0.462	0.462	0.462	0.462	0.462	0.880	0.880	0.462
34.000		0.478	0.478	0.478	0.478	0.478	0.478	0.478	0.883	0.883	0.478
35.900		0.423	0.423	0.423	0.423	0.423	0.423	0.423	0.871	0.871	0.423
37.700		0.336	0.336	0.336	0.336	0.336	0.336	0.336	0.851	0.851	0.336
39.500		0.403	0.403	0.403	0.403	0.403	0.403	0.403	0.866	0.866	0.403
41.200		0.254	0.254	0.254	0.254	0.254	0.254	0.254	0.833	0.833	0.254
43.300		0.225	0.225	0.225	0.225	0.225	0.225	0.225	0.826	0.826	0.225
44.700		0.319	0.319	0.319	0.319	0.319	0.319	0.319	0.847	0.847	0.319
46.200		0.325	0.325	0.325	0.325	0.325	0.325	0.325	0.849	0.849	0.325
47.700		0.266	0.266	0.266	0.266	0.266	0.266	0.266	0.836	0.836	0.266
49.700		0.344	0.344	0.344	0.344	0.344	0.344	0.344	0.853	0.853	0.344
51.600		0.369	0.369	0.369	0.369	0.369	0.369	0.369	0.859	0.859	0.369
53.500		0.213	0.213	0.213	0.213	0.213	0.213	0.213	0.824	0.824	0.213
54.900		0.258	0.258	0.258	0.258	0.258	0.258	0.258	0.834	0.834	0.258
56.200		0.069	0.069	0.069	0.069	0.069	0.069	0.069	0.792	0.792	0.069
57.600		0.296	0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296
59.800		0.491	0.491	0.491	0.491	0.491	0.491	0.491	0.886	0.886	0.491
61.800		0.625	0.625	0.625	0.625	0.625	0.625	0.625	0.916	0.916	0.625
63.800		0.831	0.831	0.831	0.831	0.831	0.831	0.831	0.962	0.962	0.831
66.500		0.839	0.839	0.839	0.839	0.839	0.839	0.839	0.964	0.964	0.839
68.400		0.620	0.620	0.620	0.620	0.620	0.620	0.620	0.915	0.915	0.620
70.100		0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457
71.600		0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.867	0.867	0.406
73.100		0.346	0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346
74.600		0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.816	0.816	0.178
76.200		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.768	0.768	0.0
78.000		0.109	0.109	0.109	0.109	0.109	0.109	0.109	0.800	0.800	0.109
79.500		0.185	0.185	0.185	0.185	0.185	0.185	0.185	0.817	0.817	0.185
81.600		0.152	0.152	0.152	0.152	0.152	0.152	0.152	0.810	0.810	0.152
83.800		0.457	0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457
86.000		0.579	0.579	0.579	0.579	0.579	0.579	0.579	0.906	0.906	0.579
88.100		0.741	0.741	0.741	0.741	0.741	0.741	0.741	0.942	0.942	0.741
89.900		0.947	0.947	0.947	0.947	0.947	0.947	0.947	0.988	0.988	0.947
91.500		0.979	0.979	0.979	0.979	0.979	0.979	0.979	0.995	0.995	0.979
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table F.3 (continued)

GAPXL ^a	VALUE OF K IS:	21	22	23	24	25	26	27	28	29	30
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.010	1.010
1.800		0.826	0.826	0.826	0.826	0.826	0.826	0.961	0.961	0.826	0.826
3.400		0.642	0.642	0.642	0.642	0.642	0.642	0.920	0.920	0.642	0.642
5.000		0.654	0.654	0.654	0.654	0.654	0.654	0.922	0.922	0.654	0.654
6.900		0.732	0.732	0.732	0.732	0.732	0.732	0.940	0.940	0.732	0.732
8.800		0.880	0.880	0.880	0.880	0.880	0.880	0.973	0.973	0.880	0.880
11.500		0.875	0.875	0.875	0.875	0.875	0.875	0.972	0.972	0.875	0.875
13.300		0.709	0.709	0.709	0.709	0.709	0.709	0.935	0.935	0.709	0.709
15.100		0.606	0.606	0.606	0.606	0.606	0.606	0.912	0.912	0.606	0.606
16.800		0.508	0.508	0.508	0.508	0.508	0.508	0.890	0.890	0.508	0.508
18.100		0.459	0.459	0.459	0.459	0.459	0.459	0.879	0.879	0.459	0.459
19.500		0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346	0.346
21.400		0.415	0.415	0.415	0.415	0.415	0.415	0.869	0.869	0.415	0.415
23.200		0.412	0.412	0.412	0.412	0.412	0.412	0.868	0.868	0.412	0.412
25.000		0.430	0.430	0.430	0.430	0.430	0.430	0.872	0.872	0.430	0.430
26.900		0.469	0.469	0.469	0.469	0.469	0.469	0.881	0.881	0.469	0.469
28.500		0.443	0.443	0.443	0.443	0.443	0.443	0.875	0.875	0.443	0.443
30.000		0.449	0.449	0.449	0.449	0.449	0.449	0.877	0.877	0.449	0.449
32.000		0.462	0.462	0.462	0.462	0.462	0.462	0.880	0.880	0.462	0.462
34.000		0.478	0.478	0.478	0.478	0.478	0.478	0.883	0.883	0.478	0.478
35.900		0.423	0.423	0.423	0.423	0.423	0.423	0.871	0.871	0.423	0.423
37.700		0.336	0.336	0.336	0.336	0.336	0.336	0.851	0.851	0.336	0.336
39.500		0.403	0.403	0.403	0.403	0.403	0.403	0.866	0.866	0.403	0.403
41.200		0.254	0.254	0.254	0.254	0.254	0.254	0.833	0.833	0.254	0.254
43.300		0.225	0.225	0.225	0.225	0.225	0.225	0.826	0.826	0.225	0.225
44.700		0.319	0.319	0.319	0.319	0.319	0.319	0.847	0.847	0.319	0.319
46.200		0.325	0.325	0.325	0.325	0.325	0.325	0.849	0.849	0.325	0.325
47.700		0.266	0.266	0.266	0.266	0.266	0.266	0.836	0.836	0.266	0.266
49.700		0.344	0.344	0.344	0.344	0.344	0.344	0.853	0.853	0.344	0.344
51.600		0.369	0.369	0.369	0.369	0.369	0.369	0.859	0.859	0.369	0.369
53.500		0.213	0.213	0.213	0.213	0.213	0.213	0.824	0.824	0.213	0.213
54.900		0.258	0.258	0.258	0.258	0.258	0.258	0.834	0.834	0.258	0.258
56.200		0.069	0.069	0.069	0.069	0.069	0.069	0.792	0.792	0.069	0.069
57.600		0.296	0.296	0.296	0.296	0.296	0.296	0.842	0.842	0.296	0.296
59.800		0.491	0.491	0.491	0.491	0.491	0.491	0.886	0.886	0.491	0.491
61.800		0.625	0.625	0.625	0.625	0.625	0.625	0.916	0.916	0.625	0.625
63.800		0.831	0.831	0.831	0.831	0.831	0.831	0.962	0.962	0.831	0.831
66.500		0.839	0.839	0.839	0.839	0.839	0.839	0.964	0.964	0.839	0.839
68.400		0.620	0.620	0.620	0.620	0.620	0.620	0.915	0.915	0.620	0.620
70.100		0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457	0.457
71.600		0.406	0.406	0.406	0.406	0.406	0.406	0.867	0.867	0.406	0.406
73.100		0.346	0.346	0.346	0.346	0.346	0.346	0.853	0.853	0.346	0.346
74.600		0.178	0.178	0.178	0.178	0.178	0.178	0.816	0.816	0.178	0.178
76.200		0.0	0.0	0.0	0.0	0.0	0.0	0.768	0.768	0.0	0.0
78.000		0.109	0.109	0.109	0.109	0.109	0.109	0.800	0.800	0.109	0.109
79.500		0.185	0.185	0.185	0.185	0.185	0.185	0.817	0.817	0.185	0.185
81.600		0.152	0.152	0.152	0.152	0.152	0.152	0.810	0.810	0.152	0.152
83.800		0.457	0.457	0.457	0.457	0.457	0.457	0.878	0.878	0.457	0.457
86.000		0.579	0.579	0.579	0.579	0.579	0.579	0.906	0.906	0.579	0.579
88.100		0.741	0.741	0.741	0.741	0.741	0.741	0.942	0.942	0.741	0.741
89.900		0.947	0.947	0.947	0.947	0.947	0.947	0.988	0.988	0.947	0.947
91.500		0.979	0.979	0.979	0.979	0.979	0.979	0.995	0.995	0.979	0.979
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

Table F.3 (continued)

GAPXL ^a	VALUE OF K IS:	31	32	33	34	35	36	37	38	39	40
-47.676		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
0.000		1.010	1.010	1.010	1.010	1.010	1.002	1.002	1.002	1.002	1.002
1.800		0.826	0.826	0.826	0.826	0.826	0.961	0.963	0.963	0.963	0.963
3.400		0.642	0.642	0.642	0.642	0.642	0.920	0.923	0.923	0.923	0.923
5.000		0.654	0.654	0.654	0.654	0.654	0.922	0.926	0.926	0.926	0.926
6.900		0.732	0.732	0.732	0.732	0.732	0.940	0.943	0.943	0.943	0.943
8.800		0.880	0.880	0.880	0.880	0.880	0.973	0.974	0.974	0.974	0.974
11.500		0.875	0.875	0.875	0.875	0.875	0.972	0.973	0.973	0.973	0.973
13.300		0.709	0.709	0.709	0.709	0.709	0.935	0.938	0.938	0.938	0.938
15.100		0.606	0.606	0.606	0.606	0.606	0.912	0.916	0.916	0.916	0.916
16.800		0.508	0.508	0.508	0.508	0.508	0.890	0.895	0.895	0.895	0.895
18.100		0.459	0.459	0.459	0.459	0.459	0.879	0.884	0.884	0.884	0.884
19.500		0.346	0.346	0.346	0.346	0.346	0.853	0.860	0.860	0.860	0.860
21.400		0.415	0.415	0.415	0.415	0.415	0.869	0.875	0.875	0.875	0.875
23.200		0.412	0.412	0.412	0.412	0.412	0.868	0.874	0.874	0.874	0.874
25.000		0.430	0.430	0.430	0.430	0.430	0.872	0.878	0.878	0.878	0.878
26.900		0.469	0.469	0.469	0.469	0.469	0.881	0.886	0.886	0.886	0.886
28.500		0.443	0.443	0.443	0.443	0.443	0.875	0.881	0.881	0.881	0.881
30.000		0.449	0.449	0.449	0.449	0.449	0.877	0.882	0.882	0.882	0.882
32.000		0.462	0.462	0.462	0.462	0.462	0.880	0.885	0.885	0.885	0.885
34.000		0.478	0.478	0.478	0.478	0.478	0.893	0.899	0.899	0.899	0.899
35.900		0.423	0.423	0.423	0.423	0.423	0.871	0.877	0.877	0.877	0.877
37.700		0.336	0.336	0.336	0.336	0.336	0.851	0.858	0.858	0.858	0.858
39.500		0.403	0.403	0.403	0.403	0.403	0.866	0.872	0.872	0.872	0.872
41.200		0.254	0.254	0.254	0.254	0.254	0.833	0.841	0.841	0.841	0.841
43.300		0.225	0.225	0.225	0.225	0.225	0.826	0.834	0.834	0.834	0.834
44.700		0.319	0.319	0.319	0.319	0.319	0.847	0.854	0.854	0.854	0.854
46.200		0.325	0.325	0.325	0.325	0.325	0.849	0.856	0.856	0.856	0.856
47.700		0.266	0.266	0.266	0.266	0.266	0.836	0.843	0.843	0.843	0.843
49.700		0.344	0.344	0.344	0.344	0.344	0.853	0.860	0.860	0.860	0.860
51.600		0.369	0.369	0.369	0.369	0.369	0.859	0.865	0.865	0.865	0.865
53.500		0.213	0.213	0.213	0.213	0.213	0.824	0.832	0.832	0.832	0.832
54.900		0.258	0.258	0.258	0.258	0.258	0.834	0.842	0.842	0.842	0.842
56.200		0.069	0.069	0.069	0.069	0.069	0.792	0.801	0.801	0.801	0.801
57.600		0.296	0.296	0.296	0.296	0.296	0.842	0.850	0.850	0.850	0.850
59.800		0.491	0.491	0.491	0.491	0.491	0.886	0.891	0.891	0.891	0.891
61.800		0.625	0.625	0.625	0.625	0.625	0.916	0.920	0.920	0.920	0.920
63.800		0.831	0.831	0.831	0.831	0.831	0.962	0.964	0.964	0.964	0.964
66.500		0.839	0.839	0.839	0.839	0.839	0.964	0.966	0.966	0.966	0.966
68.400		0.620	0.620	0.620	0.620	0.620	0.915	0.919	0.919	0.919	0.919
70.100		0.457	0.457	0.457	0.457	0.457	0.878	0.884	0.884	0.884	0.884
71.600		0.406	0.406	0.406	0.406	0.406	0.867	0.873	0.873	0.873	0.873
73.100		0.346	0.346	0.346	0.346	0.346	0.853	0.860	0.860	0.860	0.860
74.600		0.178	0.178	0.178	0.178	0.178	0.816	0.824	0.824	0.824	0.824
76.200		0.0	0.0	0.0	0.0	0.0	0.768	0.778	0.778	0.778	0.778
78.000		0.109	0.109	0.109	0.109	0.109	0.800	0.810	0.810	0.810	0.810
79.500		0.185	0.185	0.185	0.185	0.185	0.817	0.826	0.826	0.826	0.826
81.600		0.152	0.152	0.152	0.152	0.152	0.810	0.819	0.819	0.819	0.819
83.800		0.457	0.457	0.457	0.457	0.457	0.878	0.884	0.884	0.884	0.884
86.000		0.579	0.579	0.579	0.579	0.579	0.906	0.910	0.910	0.910	0.910
88.100		0.741	0.741	0.741	0.741	0.741	0.942	0.945	0.945	0.945	0.945
89.900		0.947	0.947	0.947	0.947	0.947	0.988	0.989	0.989	0.989	0.989
91.500		0.979	0.979	0.979	0.979	0.979	0.995	0.996	0.996	0.996	0.996
95.987		1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000

^aGAPXL values are the centimeters from the bottom of the heated zone.

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