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Jeffrey G. Schreiber, Steven M. Geng,
and Gary V. Lorenz
National Aeronautics and Space Administration
Lewis Research Center

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Jeffrey G. Schreiber, Steven M. Geng,
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National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135

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RE-1000 FREE-PISTON STIRLING ENGINE

SENSITIVITY TEST RESULTS

Jeffrey G. Schreiber, Steven M. Geng, and Gary V. Lorenz
National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135

SUMMARY

The NASA Lewis Research Center has been testing a 1 kW (1.33 hp) free-piston Stirling engine at the NASA Lewis test facilities. The tests performed over the past several years have been on a single cylinder machine known as the RE-1000. The data recorded were to aid in the investigation of the dynamics and thermodynamics of the free-piston Stirling engine. The data are intended to be used primarily for computer code validation. NASA reports TM-82999, TM-83407, and TM-87126 give initial results of the engine tests.

The tests were designed to investigate the sensitivity of the engine performance to variations on the mean pressure of the working space, the working fluid used, heater and cooler temperatures, regenerator porosity, power piston mass and displacer dynamics. These tests have now been completed at NASA Lewis.

This report presents some of the detailed data collected in the sensitivity tests. In all, 781 data points were recorded. A complete description of the engine and test facility is given. Many of the data can be found in tabular form, while a microfiche containing all of the data points can be requested from NASA Lewis.

INTRODUCTION

A free-piston Stirling engine has been under test at NASA Lewis as part of the Stirling engine technology program. The engine was intended to be used for basic research of specific thermodynamic processes involved in the Stirling engine cycle. The engine, known as the model RE-1000 was designed and built by Sunpower Inc., of Athens, Ohio in 1979 for NASA Lewis. The name RE-1000 indicates that the device is a research engine producing approximately 1000 W of brake power.

The RE-1000 engine, as tested at NASA Lewis, does not represent an engine design optimized as a total system. The goal used during the initial stages of the design was to produce an engine with maximum efficiency at the design condition of 1 kW brake power output. The starting point of the design task, however, was to use an existing design of a cooler that was intended to be used on an engine with 3 kW brake power output. The size of the regenerator cavity was also taken from the existing 3 kW design. With these constraints, along with the heater tube temperature fixed at 600 °C, cooler temperature of 25 °C, mean working space pressure of 7.0 MPa, and Helium as the working fluid, the Sunpower optimization code was used to maximize the efficiency. The engine is described in references 1 and 2.

Since the design of the engine research was aimed at the need for basic Stirling cycle research, consideration was given to the ability to measure key thermodynamic characteristics. Many instrumentation ports were incorporated in the hardware during the fabrication of the engine. Also considered at the time of fabrication was the requirement for reliable operation, short teardown and rebuild times, and flexibility of configurations. Because the research to be carried out was not intended to be on the heat input system, an electric resistance heater was developed. This heater will be discussed in detail in a later section. For similar reasons, a dashpot load was installed to absorb the power generated by the engine. This load will also be discussed in a later section. Figure 1 is a cutaway view of the RE-1000 showing the internal parts with some of the key components labeled. Figure 2 shows the engine installed in the NASA Lewis test facility.

The purpose of the tests performed with the RE-1000 was to collect data throughout a test matrix and to use these data to validate computer simulations of the Stirling cycle. The test matrix developed was intended to exercise the engine over a wide range of operating conditions to find the parameters to which the engine has a great amount of sensitivity, and also to find parameters to which the engine has little sensitivity. The matrix not only varied the operating conditions of the engine, but also varied the configuration of the engine hardware. Changes in the operating condition were accomplished by varying the working fluid, the mean pressure level, the heater and cooler temperatures, and the power piston stroke. Changes in the configuration of the engine were accomplished by varying the power piston mass, the regenerator porosity, and the displacer dynamics.

At the end of the sensitivity testing, a total of 781 data points were accumulated. During some of the testing, a measurement of the dynamic pressure drops of the heat exchangers was recorded. The measurement of steady-state pressure drops for each of the engine configurations was also recorded. A discussion of the measurement of the dynamic pressure drops of the heat exchangers will be presented in a later section.

This report presents a complete description of the RE-1000 engine suitable for computer modeling of the test hardware. A description of the test methods and the instrumentation is given. The 781 data points recorded during the sensitivity tests are presented in detail along with suggestions on how some of the data should be used. Some of the data are given in the form of graphs to show the general trends of the engine performance. A sample of the detailed data readings is presented in tabular form but the entire collection of data on a microfiche card can be requested from NASA Lewis.

DESCRIPTION OF THE SENSITIVITY

Test Program

During the initial stages of the NASA Lewis free-piston Stirling test program, a need was recognized for data recorded over a wide range of operating conditions for the purpose of computer code validation. Little free-piston Stirling engine data had been published and none of it was in enough detail to be used for code validation. With the ability to change the configuration of the RE-1000 engine and the wide range of operating conditions at which the

engine could be run, a test program was designed to exercise the engine over many configurations and operating conditions.

The sensitivity tests were developed to provide this much needed data with the use of the RE-1000. The term sensitivity was used to describe these tests because the engine would be exercised throughout a matrix of operating conditions with eight parameters being varied. The data could then be studied to better understand how the engine responds not only to changes in the operating conditions, but also to changes in the configuration of the hardware. Three of the varied parameters were configurational changes of the hardware. These changes include variations in the power piston mass, regenerator porosity, and the displacer dynamics. A description of each of these configurational changes is given later in this report. The other parameters varied include the power piston stroke, the mean operating pressure in the working space, the working fluid, the heater temperature and the cooler temperature.

With the data collected in these tests, it would be possible to see which parameters strongly influence the performance of the engine and which parameters have little if any effect on the engine performance. Because these tests cover a wide range of operating conditions it would be possible to see the trends of the engine on a macro basis rather than on a small, local basis. This feature is helpful in the validation of computer simulations since a computer code should be able to predict engine performance over a wide range of conditions. Descriptions of the NASA Lewis Stirling engine simulation and the calibration of it are given in references 3, 4, 5, and 6.

Figure 3 shows a segment of the sensitivity test matrix. The parameters varied in this matrix were mean engine pressure level, heater head temperature, and cooler temperature. Inside each of the elements of this matrix the stroke of the power piston was varied. The stroke was varied in increments of 0.2 cm from 1.8 cm up to an upper limit determined by the operating conditions of the engine. In some cases, the upper limit of the stroke was determined by the presence of system instability causing the engine to operate with a low frequency beat imposed on the engine operating frequency. Other times the stroke was limited by either the condition of no load being applied to the piston or by the increased risk to the engine instrumentation. In all cases, the stroke was kept below a limit of 3.6 cm because of physical limits in the engine.

The matrix depicted in figure 4 shows the variations in the configuration of the test hardware. The configuration matrix shows that the engine was tested with all four possible combinations of the two regenerators and two displacers. In addition to these four configurations, the tests were run with the regenerator 1 and displacer 1, but the working fluid was changed to nitrogen for one series of tests and to argon for another series. A series was also run with a power piston with a mass less than the design mass. It should be noted that the matrix shown in figure 3 exists once inside of each of the elements of the matrix shown in figure 4. Table I gives a listing of the number of data points recorded for each of the elements of the configuration matrix.

The test matrix was designed to provide a method to detect the potential weakness of a Stirling engine computer simulation. When a computer simulation is calibrated against too few data points, the code may be very accurate when simulating a point near the conditions it was calibrated against, but may have significant errors when it tries to predict engine performance at an operating

point far from the calibration conditions. This may be caused by the adjustment of calibration factors that tend to counteract one another in some respect. The idea that it is desirable to calibrate a computer code over a wide range of data conditions led to the detail design of the sensitivity test program.

The data can be used to study the sensitivity of the RE-1000 to variations in any one parameter with all other parameters held constant. While this is done in most engine test programs to some degree, (i.e., the engine speed is usually varied as the pressure level is held constant in most kinematic Stirling engines as shown in ref. 7), it is possible to use data from the current sensitivity tests to look at the variation in engine performance with respect to any of the controlled parameters used as a test variable. With this in mind, the significance of some of the tests becomes more clear. If data points are used for code validation where the test parameter varied is the working fluid while all other parameters are held constant, some of the deficiencies of a code that are a function of the gas properties can be found. By using the data points recorded when the mass of the power piston was the test variable, the deficiencies of a code that are frequency dependent can be found. This can be very helpful in the calibration of a Stirling simulation since the configuration of the engine and the gas properties are held constant and the operating frequency of the engine is changed.

DESCRIPTION OF THE ENGINE

Along with the cutaway drawing of the RE-1000 free-piston Stirling engine shown in figure 1, a photograph of the engine in the test cell is shown in figure 2. The engine is relatively large and massive for its power level for several reasons. The engine as built for NASA Lewis, was intended to be a rugged test engine to be used for many hours of research. Since there was no interest in demonstrating the feasibility of commercialization or production features of the engine, no consideration was given to make the engine light or compact. A design requirement was to make the engine reliable and provide ease of maintenance. Consideration was also given to the ability to get instrumentation into key locations to enable the measurement of some thermodynamic characteristics. Generally, ports were incorporated at the connecting ducts between the heat exchangers for gas property measurements.

As can be seen in figure 1, the RE-1000 is a single cylinder free-piston Stirling engine with a posted displacer, annular regenerator and cooler, electric resistance heater head, and a dashpot load device built inside of the bounce space. The significance of the posted displacer configuration is that the displacer dynamics are coupled to the engine housing by the displacer gas spring. (The other possible configuration is to couple the displacer dynamics to a gas spring inside of the power piston.) Since the engine housing of the RE-1000 is very massive relative to any other engine component, the driving force that the displacer responds to during operation is the pressure variation of the working space. The response of the displacer generally follows the classical linear vibration theory as described in reference 8. Some nonlinearity does exist in the engine however and the linear theory does not yield a perfect representation of the dynamics.

The sliding surfaces of the engine use wear couples. The power piston is coated with a fluorocarbon material and the cylinder is coated with chrome

oxide. The displacer rods are coated with chrome oxide while the bores of the displacers are hardened stainless steel. Advances in Stirling engine technology include the use of hydrostatic and hydrodynamic gas bearings as described in references 9 and 10.

A complete list of engine parameters and dimensions is given in table II. It is intended that this list should be a complete description of the RE-1000 for its modeling on a computer. In the event of any conflict, the data presented in this table supersedes any previously published data with respect to the dimensions and configuration of the NASA Lewis RE-1000 since some alterations and more accurate measurements have occurred.

Heater. - The RE-1000 heater head is shown in figure 5. The heater was made with 34 tubes of Inconel 718. The tubes are used to form an electric resistance heater. As can be seen in figure 5, 17 of the tubes are connected at their midpoints to a semicircular electric bus bar. The other 17 tubes are connected at their midpoints to another electric bus bar. Each of the electric bus bars has two power tabs extending radially from the center of the heater. The flow of current from the one bus bar, through 17 of the tubes to the cylinder produces heat in those 17 tubes. The cylinder of the engine is at electric ground. Heat is likewise generated in the other 17 tubes as the current flows away from the head, through the tubes to the other bus bar and its power tabs, back to the electric power supply. The heat is generated by the high current, low voltage system.

The four power tabs that extend radially from the two bus bars were sized to generate heat in an attempt to minimize the heat loss from the heater tubes. With the proper amount of heat generated in each power tab, the temperature gradient in each tab would be zero at a point where the power tabs connect to the semicircular bus bars. A cross section of one power tab and its connection to the bus bar is shown in the exploded section of figure 5. The diagram shown in the exploded view illustrates how the power tab becomes neither a heat sink nor a heat source to the heater tubes. In this case, the heat conducted from the bus bars to the power tabs would be zero and the power tabs. The heat calculated in the data system as PWRIN, however, includes some of the heat generated in the power tabs that is lost to the surrounding air. This has an effect on any attempt to develop an energy balance or to calculate the engine efficiency based on the power input.

Each heater tube connects the expansion space to the hot end of the regenerator. The 34 tubes flow gas in parallel from the expansion space to a small plenum chamber at the hot end of the regenerator. As the tubes pass through the two semicircular bus bars, there is no mixing of the hot gas from one tube to another. The bus bars are brazed to the heater tubes only as a conductor for the electric current.

The heater head was fabricated with two semicircular bus bars attached to the midpoint of each of the 34 heater tubes. Since all of the heater tubes were fabricated from the same supply of tubing and therefore should be identical, every tube should be at the same temperature level during operation. The data always indicated that there was a tube to tube variation in temperature of as much as 20 °C. The variation was interpreted as misalignment of the thermocouples on the heater tubes and not a true variation of the heater tube temperatures. Since there is a voltage gradient along the length of the tube during operation, any misalignment of the thermocouple on the heater tube would cause

the thermocouple output to be the summation of the temperature induced output and the voltage gradient over the length of the thermocouple. This thought was substantiated by the fact that with each new installation of thermocouples, the highest heater tube temperature measured was on a different tube than on the previous tests.

Regenerator. - The regenerator cavity is an annular volume between the outside surface of the displacer cylinder wall and the inside surface of the gas-pressure containing wall. The regenerator matrix is a knitted 304 stainless steel material produced by Metex Corporation of Edison, New Jersey. Regenerator material is inserted into the regenerator cavity from the cold end of the heater head and becomes compressed when the cooler is attached to the heater head. The matrix structure is much like a metallic rope with a square cross section. The regenerator matrix is shown in figure 6.

The engine was tested with two different regenerators. The regenerators were similar with the exception that they had different porosities. When the engine was purchased at the beginning of the program it was built with a regenerator and a displacer that shall be referred to as regenerator 1 and displacer 1. These parts were produced as a result of the design goal of high engine efficiency within the original design constraints. At a later time in the testing of the RE-1000, NASA Lewis purchased a regenerator and displacer from Sunpower that were optimized for high power output with all other parameters of the engine held constant. These high power parts shall be referred to as regenerator 2 and displacer 2. Regenerator 1 had a porosity of 75.9 percent while regenerator 2 had a porosity of 81.2 percent. As a standard test procedure, the heat exchangers were flow tested after each rebuild. The results of the steady-state flow tests are given in figures 7 and 8.

Cooler. - The cooler of the RE-1000 is shown in figures 9 and 10. The cooler is of an annular design. The gas flow path consists of 135 rectangular passages spaced evenly around the displacer cylinder. The cooling water flows through passages in the cooler housing, parallel to the gas passages. The water flow direction during the tests was from the regenerator end of the cooler to the compression space end of the cooler. All of the parts of the cooler assembly are made of aluminum for low thermal resistance. The stainless steel displacer cylinder required a light press fit into the aluminum finned unit.

Displacer. - Tests on the RE-1000 were run with two different displacers. As discussed earlier, displacer 1 was designed for high efficiency and displacer 2 was designed for high power. The two displacers are shown in figures 11 and 12. Both of the displacers are similar in construction. The gas spring of the displacer is contained inside of the displacer body. The cold end of the displacer provides mounts for the antirotation rod and the displacer position measurement rod. The displacer position measurement rod extends into a linear voltage differential transformer (LVDT) built inside the power piston. This measures the position of the displacer relative to the power piston position since no easy method exists in the RE-1000 to measure the absolute displacer position. The antirotation rod prevents the LVDT core mounted on the displacer from becoming rotationally misaligned with respect to the LVDT windings built into the power piston.

The displacer rods are coated with chrome oxide for wear resistance. The rod is supported by the mounting spider located at the end of the rod. A small

bushing of Rulon is built into one of the legs of the spider to guide the displacer antirotation rod. The bore of the displacer, into which the displacer rod fits, is made of hardened stainless steel. On the end of the sleeve, inside of the displacer, an enclosed volume is attached to form the displacer gas spring. The volume was sized in each displacer gas spring to achieve a predetermined spring rate. Leakage of gas between the displacer gas spring and the working space is minimized by the close fit of the displacer rod in the displacer bore.

There must also be a seal to minimize leakage of the gas of the expansion and compression spaces from leaking between one another without flowing through the heat exchangers. This seal is a single ring of Teflon impregnated with molybdenum-disulfide. There was no backup ring used to provide a preload. The ring had a cross section of 3.00 mm (0.118 in.) by 2.82 mm (0.111 in.). Since little differential pressure across the displacer exists during engine operation a high drag piston ring similar to the ones used in kinematic engines is not necessary.

The two displacers and displacer rods differ in design. Displacer 1 was intended to operate with a phase angle of about 45° with respect to the power piston. Displacer 2 was designed to operate with a phase angle of approximately 80°. Displacer rod 1 was 1.663 cm (0.6548 in.) in diameter while the bore of the displacer was 1.666 cm (0.6558 in.). Displacer rod 2 was 1.808 cm (0.7120 in.) in diameter while the bore of the displacer was 1.811 cm (0.7130 in.). The combination of the gas spring stiffness, the displacer mass, and the damping to which the displacer was subject determined the dynamic response of the displacer to the pressure wave in the working space. Critical dimensions of the two displacers are given in figures 13 and 14.

One major difference between the two displacer rods was the instrumentation port provided in displacer rod 2 that could not be fit into displacer rod 1. This instrumentation port allowed the measurement of the pressure variation in the displacer gas spring as the engine was operating. Figure 12 shows the displacer rod with the instrumentation port labeled. The passage in the displacer rod caused a slight phase shift of the pressure reading but a negligible amount of attenuation. Because of this, the data reported contains a measurement of pressure amplitude but does not try to integrate the pressure-volume diagram of the gas spring. The error in the calculated loss would be too large relative to the actual value of the loss.

Power Piston. - The power piston of the RE-1000 is shown in figure 15. The main body of the piston is made of aluminum with the outer surface having a coating of a fluorocarbon anti-friction material. The larger diameter section of the piston was made of steel to increase the piston mass to the level required to make the engine resonant frequency become 30 Hz. For the tests conducted at the higher operating frequency this steel mass was replaced by a mass of the same dimensions but made of aluminum. This had the effect of making the oscillating mass of the piston lighter and therefore raised the resonant frequency to approximately 37 Hz.

A slot was machined in the side of the large mass to hold a Rulon bushing. This bushing would ride on a stationary rod during operation of the engine to prevent the piston from rotating. The bushing was made of Rulon to minimize friction. Misalignment of the power piston could cause damage to the instrumentation in the engine or the power piston itself. A new technology being

applied to free-piston Stirling engines involves the use of induced rotation to cause the displacer and the power piston to ride on hydrodynamic gas bearings. Some of the work on these bearings can be found in reference 9.

Three protrusions are on the end of the power piston toward the compression space. These three protrusions extend through the three openings of the displacer rod spider when the piston is at the inward end of its stroke. This feature was incorporated with the intent of reducing the dead volume of the engine.

The power piston uses a clearance seal to reduce gas leakage. The cylinder in which the piston rides is coated with chrome oxide. The inside diameter of the power piston cylinder is 5.722 mm (2.2528 in.), and the outside diameter of the piston is 5.718 mm (2.2514 in.).

The power piston of the RE-1000 with its original steel mass weighed 5.47 kg (12.06 lb). Other hardware attached to the power piston weighed 504 g (1.11 lb). This caused the total oscillating mass to be 5.97 kg (13.2 lb). Since this mass is nearly 15 times as large as the mass of a displacer, the dynamics of the engine are dictated by the inertia of the power piston. The operating frequency of the engine was approximately equal to the natural frequency of the piston mass rebounding against the gas spring formed by the working space. The bounce space in the large pressure vessel had a negligible effect on the frequency since its volume is about 100 times as large as the working space volume.

Centering ports. - Since a free-piston Stirling engine has no kinematic linkage to constrain the motions of the power piston or the displacer, a system is required to insure that the midpoint of the stroke of each oscillating part stays at some fixed location in the engine. This is necessary to insure that the piston and displacer do not collide with one another or with some physical limitation inside of the engine. This system is needed even if the hardware were able to be produced with perfect dimensions because the leakage past most of the seals in the engine is not symmetrical. A power piston will generally tend to drift into the working space, and in the configuration of the RE-1000, the displacer will drift toward the compression space.

Leakage in the engine is not symmetrical due to the variation of the density of the working fluid. As an example, the leakage of gas between the working space and the bounce space will be discussed. Since the bounce space undergoes no heating or cooling and the change in volume is very small compared to the total volume of the bounce space, the pressure in this space will be assumed to be constant. The pressure in the working space however undergoes a sizeable pressure variation. The equations describing the leakage between the two spaces indicate that the volumetric flow of gas will be equal in both directions. This flow rate is proportional to the square of the differential pressure across the leak path. The density of the gas flowing from the working space to the bounce space will be higher than the flow from the bounce space to the working space. There will be a net mass flow rate from the working space to the bounce space when the two volumetric flow rates are equal.

When the piston is at midstroke, a small port on the surface of the piston is aligned with a small port in the cylinder wall. The alignment of these ports allows the working space to communicate with the bounce space. This has the effect of forcing the engine to have no difference between the mean working

space pressure and the mean bounce space pressure when the engine is operating. This centering port system has been very effective and trouble free throughout the testing.

A similar system exists for the centering of the displacer stroke. In the case of the displacer, ports are aligned at midstroke that lets the small internal gas spring communicate with the large bounce space of the engine. This forces the pressure of the displacer gas spring to have its pressure equal to the mean pressure of the engine when it is at the center of its stroke. Figure 16 shows a cross section of the displacer centering port system.

When the engine is running, the centers of the piston and displacer strokes will not be at the exact same positions as when their respective centering ports are aligned. The distance between the center of the stroke and the centering port position will vary from one data point to another depending on the operating conditions. If the centering port system was made strong enough to force the center of the stroke to be at the same position as the centering port, a large loss may be incurred. This loss will be discussed later. The power piston of the RE-1000 has a maximum stroke of 4.35 cm (1.713 in.). The centering port for the power piston is aligned when the piston is 2.05 cm (0.807 in.) from the bumpers that limit its movement into the working space. Even though the piston does not touch the bumper during operation, the positions of the centering ports and of the center of the piston stroke were measured relative to these bumpers.

Displacer number 1 has a maximum stroke of 4.04 cm (1.591 in.). Its movement is limited by the end of the cylinder at the expansion space end of its stroke and by the spider that supports the displacer rod at the compression space end of its stroke. The centering port for displacer number 1 becomes aligned when the displacer is 1.90 cm (0.75 in.) from the expansion space limit of its movement. The dimensions for displacer number 2 are 5.18 cm (2.039 in.) for the maximum stroke and 2.64 cm (1.04 in.) from the end of its possible movement to the centering port position. Table III lists the volumes in the engine with both the power piston and the displacer at their respective center port positions. Locations of the different volumes are shown in figure 17.

Each of these centering port systems has some loss associated with it. This loss occurs in the form of gas flowing from the higher pressure side of the port to the low pressure side of the port each time the port opens. For the displacer centering port system this loss should not be too great. The reason for this is that the pressure in the gas spring of the displacer should be the same at both times the port opens during a cycle if there is no leakage and the spring has little hysteresis. The centering port of the displacer should only flow enough gas to correct for some slight leakage and a slight hysteresis in the spring. The power piston centering port, on the other hand, may have a great amount of loss associated with it. The reason for the loss is that the pressure in the working space at the two times during an engine cycle that the ports open is not nearly equal. One opening occurs when the power piston is in the process of compressing the working fluid and the other opening occurs when the power piston is expanding the working fluid. During the compression stroke most of the working fluid is in the relatively cool compression space while during the expansion stroke most of the working fluid is in the relatively hot expansion space. At one opening of the centering ports the working space pressure will be much greater than the bounce space

pressure, while at the other opening of the centering ports the pressure in the working space will be much less than the pressure in the bounce space. The result is that there is gas flowing from the working space to the bounce space at one opening of the ports and gas flowing in the opposite direction at the other opening of the ports. These two situations are diagramed in figure 18. For these reasons it is important to have the centering port systems of an engine no bigger than is needed to maintain proper operation.

In the RE-1000 tests were done to determine the best strength of the centering port system for the power piston. The piston originally had four centering ports. Each port was a 1.14 mm (0.045 in.) hole. Data was recorded with from one to four of the ports operational. When only one port was working, the power piston tended to operate far in toward the working space and would almost not open the centering ports at all. As the number of ports was increased, the piston ran more centered around the centering ports but the losses increased. The best combination found was with two of the four original centering ports in operation.

Leak paths. - There are several leak paths in the RE-1000 that should be included in any model to insure proper simulation. Leak paths that allow the gas in the working space to leak to and from the bounce space will be discussed. Not only are these leak paths not obvious at first look at the engine, but the number of paths that are leaking actually changes during the engine cycle.

Figure 19 shows the leak paths that involve the power piston and its cylinder. In the position that the piston is shown in figure 19(a) there are three different leak paths for the gas to use. The shortest path and therefore the path that probably allows the most leakage, is the path that connects the port in the piston to the port in the cylinder wall. This path is designated as path 1 in figure 19(a). The length of this path changes as the piston moves during operation of the engine. Path 2 connects the working space to the centering port in the cylinder wall. This port in the cylinder wall is always at the bounce space pressure level. The length of this leak path also changes during the engine cycle as the piston moves. Leak path 3 connects the centering port of the piston to the bounce space without going through the centering port in the cylinder wall. Care must be taken to realize that while the piston is on the outer half of its stroke as is shown in figure 19(a), there are three separate leak paths, but when the piston is on the inner half of its stroke there is only one leak path. Leak paths 2 and 3 do not leak because the centering port of the piston is on the working space side of the centering port of the cylinder wall. There will be no leakage between the working space and the piston centering port because they are both at the working space pressure. Similarly there will be no leakage between the bounce space and the cylinder centering port because they are both at the same pressure. A two dimensional description of the leak paths past the power piston is given in figure 20. The paths are shown for both halves of the engine cycle.

A similar situation exists for the displacer centering port system. In this case, there is leakage that involves three different gas volumes. When the displacer is on the half of its stroke toward the compression space, there are three leak paths as shown in figure 16(a). Leak path 1 connects the centering port in the displacer bore to the centering port in the displacer rod. This allows gas to leak between the displacer gas spring and the main

bounce space. Leak path 2 connects the working space to the centering port in the displacer bore. This allows leakage of gas between the working space and the displacer gas spring. Leak path 3 connects the displacer gas spring to the ports in the displacer rod. This leak path allows leakage in a parallel path to leak path 1. As was the case with the power piston, the number of leak paths in existence changes during the engine cycle. Leak path 2 changes to leak between the compression space and the main engine bounce space when the displacer has moved to the expansion space as shown in figure 16(b). Both ends of leak path 3 connect the same volume when the displacer has moved to the expansion space and therefore has no leakage.

Dashpot load. - The RE-1000 was built with a small dashpot as a load for the engine. Since the purpose of the testing was not to do research on power output conversion devices, the dashpot load system was well suited. The dashpot load device can be seen in figures 1 and 21 near the top of the engine. The dashpot was made with a carbon piston inside of a cylinder that would pump gas in the bounce space back and forth through an orifice. The orifice size was adjusted during each test to change the level of the load. The adjustment was accomplished by rotating a valve stem that had a 3° taper. The valve stem was rotated by means of an electric motor that can be call out at the top of the engine in figure 21. This load, being a damping device, produced a force resisting the piston motion that was nearly proportional to the square of the piston velocity. A water jacket was formed around the dashpot cylinder to cool the load device.

DESCRIPTION OF THE FACILITY

The RE-1000 test facility was built to supply all of the support systems needed to carry out the research program. The facility had a high pressure, 15.5 MPa, (2250 psi) gas system to supply the working fluid to the engine, a water system to cool the engine, and a dc electric power supply system to heat the engine. Along with these systems, a high speed data system was built to help in the data gathering process. Each of these systems will be described in detail. A simple schematic of the facility systems and some of the instrumentation built into the systems is shown in figure 21.

The gas system was designed with the ability to charge the engine with helium, hydrogen, nitrogen, or argon. The supply pressure to the engine was set with a remote control pressure regulator. The flow of gas into or out of the engine was controlled by motorized needle valves in the supply and vent lines respectively. To start the engine, a pair of solenoid valves alternatively connected the high pressure supply line or the low pressure vent line directly to the working space. The gas pulses would start the displacer and power piston in motion. If the motion was great enough, the engine cycle would continue to operate.

The water cooling system consisted of two separate systems. An open loop system was used to cool the dashpot load device. This system had no temperature control. A closed loop water system was used to cool the engine compression space. This water system was filled with distilled water. A heat exchanger with a feedback system controlled the temperature of this water loop. Water temperature varied during the tests from 25 to 55 °C. Temperatures and flow rates were measured in both systems to allow the calculation of heat rejection rates of the engine.

The engine heater power supply system consisted of two Sorensen electric power supplies connected in parallel. Each power supply unit had the capability of delivering 1000 A of direct current to the engine. The two power supplies were regulated by an automatic controller which used a thermocouple on one of the engine heater tubes as feedback. The heater temperature was varied from 450 to 600 °C during the engine tests. The control room for these systems is shown in figure 22.

DESCRIPTION OF THE INSTRUMENTATION

Since the purpose of these tests was to collect detailed data on the dynamics and thermodynamics of the free-piston Stirling engine, a variety of diagnostic instrumentation was installed. In all cases, the instrumentation was designed to have as little impact on the operation of the engine as possible.

All of the temperatures were measured with type K (Chromel-Alumel) thermocouples. Twelve heater tube temperatures were recorded for data use; their average was used to set the desired test conditions. Six of the heater tube temperatures were measured at the quarter-length point at the expansion space end of the heater tube, while the other six temperatures were measured at the quarter-length point of the heater tube near the regenerator end of the tube. The thermocouples attached to the heater tubes were closed ball and unsheathed. Thermocouples were added on the outer wall of the regenerator to aid in the calculation of conduction losses. The locations of the thermocouples installed on the heater head are shown in figure 23. These regenerator wall thermocouples were closed ball and sheathed.

Thermocouples used to measure the temperature rise of the water as it passed through the engine cooler or through the dashpot cooler were installed in a differential temperature configuration. This produced a signal proportional to the temperature rise of the water. The thermocouples were matched pairs to insure accuracy. These thermocouples were used in addition to the ones used to measure the water inlet temperature and the water outlet temperature because they give a more accurate reading of the differential temperature. Both the inlet and outlet temperatures were estimated to have a probable error of 1 °C compared to 0.2 °C for the temperature differential measurement.

A dynamic pressure measurement was made in the compression space. This measurement was used along with a measurement of power piston motion to calculate the indicated power of the engine. A dynamic pressure measurement was also made of the gas spring inside of displacer 2. As was mentioned earlier, the path that the gas pressure wave traveled caused a slight phase shift of the signal but a negligible amount of attenuation.

A great amount of work was done to develop techniques for accurate measurement of the dynamic pressure drop across each of the heat exchangers. Due to physical limitations in the installation of the instrumentation, measurements across the cooler, the regenerator, and across the entire heat exchanger circuit were made. Special attention was given to the response characteristics of the tubes that connected the transducers to the engine. Each tube was sized to provide the same acoustic response on both sides of the transducers. Corrections in the length of the lines were made to compensate for differences in the tube diameters and the gas temperatures. Tests

performed at NASA Lewis indicated that the transducers had insignificant sensitivity to mean pressure level but were affected by the temperature of operation. The transducer bodies were therefore water cooled so that the transducer would maintain a constant operating temperature.

The transducers installed to measure the dynamic pressure drops during engine operation were also used to measure steady-state pressure drops. Because these tests were run at steady-state conditions, the data is reported in the form of cooler pressure drop, regenerator pressure drop, and heater pressure drop. The heater pressure drop was calculated from the measurement of the total pressure drop minus the regenerator and cooler pressure drops. This same calculation was not done for the measurements of the dynamic pressure drops during operation of the engine. The reason is that while great care was taken to insure that the response was equal for both lines feeding any individual transducer, this does not guarantee that the response of the two lines feeding one transducer is the same as the response of the two lines feeding another transducer. Although any two transducers may in fact measure the pressure drop properly, there was no attempt made to synchronize one transducer with another. The reasons not to subtract one pressure drop measurement from another becomes even more apparent when it is realized that the actual pressure drops in the heat exchangers are not necessarily in phase with one another.

The displacer and power piston positions and the power piston velocity were measured. The power piston position and velocity were measured directly by a LVDT, and LVT (linear velocity transducer), respectively. The displacer position cannot be measured directly, since the displacer is completely enclosed in the working space. The displacer position was measured relative to the power piston position with the core of the LVDT attached to the displacer and the windings of the LVDT built inside of the power piston. The signal from the windings inside of the power piston was carried along wires from the moving piston that had to flex as the piston moved. These wires were supported by a piece of music wire 0.254 mm (0.010 in.) in diameter.

To obtain the absolute displacer position signal, the power piston position had to be subtracted from the displacer signal. An electric analog circuit performed the subtraction of the signals from the LVDTs. The circuit also measured the peak values of the signals to produce a measurement of the piston and displacer strokes.

A crystal-type force transducer was mounted in the link that connected the power piston to the dashpot load. Since the force transducer moves with the power piston, a wire system similar to the one used for the displacer signal had to be incorporated to relay the force transducer output signal to the data system. This dynamic measurement of the resistance force being applied to the power piston by the dashpot was used in an analog circuit along with the power piston velocity to calculate the brake power output of the engine. The equation used to calculate the power output was

$$\text{brake power} = \frac{FV \cos \theta}{2}$$

where

F amplitude of the force signal
V amplitude of the piston velocity signal
 θ phase angle between F and V signals

An error analysis method outlined in reference 11 was applied to this calculation of power. The resultant probable error was found to be 3 percent at the engine design conditions, assuming that all of the input signals maintained their individual accuracies. The indicated power output of the engine was calculated in a similar manner to that used for the brake power output. The dynamic compression space pressure was measured with a crystal type fast response pressure transducer. The equation used was

$$\text{indicated power} = \frac{PAV \cos \theta}{2}$$

where

P amplitude of the compression space pressure signal
A area of the power piston
V amplitude of the piston velocity signal
 θ phase angle between P and V

The same error analysis method was applied to this calculation and the resultant probable error was found to be 3 percent. Once again, this assumes that all of the input signals maintained their individual accuracies. Both the indicated power calculation and the brake power calculation were recorded by the data system.

Two phase angle meters were utilized in the data system. One of the meters measured the phase angle between the power piston and the displacer positions. The other phase angle meter measured the phase angle between the power piston position and the compression space pressure wave. The motions of the power piston and the displacer were not pure sine waves, and the pressure variation in the compression space also had some higher harmonic content. By strict definition, there can be no measurement of a phase angle between two signals that are not pure sine waves. By describing the measured motion of the power piston in a Fourier series it was found that approximately 97 percent of the piston motion is due to the 30 Hz fundamental frequency of the engine. The other 3 percent of the motion was contained in the higher harmonics. In the case of the displacer motion, 95 percent of the motion is in the fundamental frequency term and the remaining 5 percent is in the higher harmonics. The phase angles presented in these data represent the time interval between the positive going zero crossings of the two signals. The outputs from both of these meters were checked against other methods of measuring the phase angles and found to be accurate. The outputs from these meters were fed into the data system and appear on the data output.

A diagram showing the location of most of the instrumentation on the engine is presented in figure 24. A listing of all of the instrumentation used in the tests is given in table IV. The number located near each piece of instrumentation represented in figure 24 is an item number that can be used to identify the instrumentation listing in table IV.

Data was recorded by two different systems during the sensitivity tests. The NASA Lewis Escort system was used for the steady-state data and a separate

system was installed in the control room for recording the dynamic data. Most of the data signals were transmitted to the Escort data system. The Escort system is a microcomputer-based digital data recording and display system intended for steady-state use. The sampling rate of approximately 5000 samples per second permits the use of multiple scans, which are averaged for each data point recorded. The data system uses five scans of data recorded over a 15 sec period. Calculations are performed to indicate the statistical variation of each channel recorded over the total number of scans.

The Escort system has the ability to convert millivolt signals to engineering units with preprogrammed calibrations. The system can do any predefined calculations with the parameters and display the results on either a set of light emitting diodes (LED) or on a preprogrammed cathode ray tube (CRT) display. There were 15 LEDs available for these tests. The CRT could be programmed to have many different displays, or pages, that could be viewed during the test sessions. An important feature of the Escort system was its ability to do limit checking during the tests. The system could check for high or low limits of any of the data channels or calculated parameters, and then provide either a warning of limits violation or perform some predefined action, such as the initiation of a shutdown procedure. Both the LEDs and the two CRTs can be seen in the control room in figure 22. More information on the Escort system may be found in reference 12.

While the Escort system was able to handle the steady-state data channels, it did not fulfill the sample time requirements to record dynamic data channels. The dynamic data were handled by several different methods. Phase angles were measured with phase angle meters. The inputs to these meters were the dynamic engine parameters from the LVDTs and the compression space pressure transducer. The output from these meters was used as an input to the Escort data system. The measurements of the piston stroke, the displacer stroke, and the calculations of the indicated and brake power levels were performed by analog electronic circuits. The analog electronic circuits were checked for accuracy by comparing their measurements to measurements made by other methods and were found to be accurate. These analog circuits were recalibrated regularly to insure accuracy was maintained.

All of the dynamic signals were recorded for each data point on a Honeywell Visicorder. The parameters recorded on the Visicorder are indicated as such in table IV. Some data reduction was performed with the output from the recorder after each test run. The measurements resulting from this output include the amplitudes of the three differential pressure transducers, the amplitude of the compression space pressure wave, the center position of both the displacer and power piston strokes, and the amplitude of the pressure wave in the displacer gas spring for displacer 2. These parameters were later merged with the steady-state data and appear on the output of the steady-state data system.

Table V provides a list of the calculations performed in the data reduction process. The exact form of each calculation is presented in appendix B. This appendix also gives a calculated error for each of the calculations based on the probable error of the inputs for the calculation.

DESCRIPTION OF THE TEST PROCEDURE

The RE-1000 engine installed in the test facility is shown in figure 2. Before any test session was initiated a calibration of all pressure transducers was performed automatically by the Escort data system. The engine was purged of air by a pressure-vent cycle of the working and bounce spaces. The engine was then pressurized with the desired working fluid to some pressure level between 5.5 MPa (800 psig) and 7.0 MPa (1015 psig) depending on the test conditions to run on that particular day. Since the engine would usually start to run after fewer than five impulses from the starter system, regardless of pressure level, it was not important to be at any particular pressure level to start the engine. On most test runs the engine would become self sustaining after two impulses.

After the engine was purged of air and pressurized with the desired working fluid, the electric heat input was started and the cooling water flows were set to their proper levels. The dashpot load was set to give little or no resistance to the power piston motion. When the heater tube temperature reached 600 °C, the engine was given pressure pulses from the starter system until self sustaining operation was achieved. The dashpot needle valve orifice was then adjusted to give the proper load to the engine for the piston stroke desired.

After typically 2 min of operation, all measured temperatures had reached their steady-state values. (The shortness of the transient period is due to the very low thermal inertia, including the absence of an oil lubrication system that kinematic engines have.) Although steady-state temperature readings were obtained in about 2 min, the engine would typically be allowed to operate at the initial test conditions for 15 min or more.

When recording data, the points were recorded in order of increasing stroke. The engine would be held at the desired operating conditions, (i.e., temperature and pressure) with the power piston stroke at 1.8 cm. After this point was recorded, the dashpot setting was adjusted to allow the power piston stroke to reach 2.0 cm. The piston stroke was increased in increments of 0.2 cm and data was recorded until the stroke reached 3.6 cm or until the engine operation started to become erratic. The engine was allowed to operate at each new condition for 5 min before this new data point was recorded.

It must be noted that the test conditions were traversed in a rather orderly fashion. For example, the first series of data points recorded may have been with the conditions of 450 °C heater temperature, 25 °C cooler temperature and 5.5 MPa mean pressure level. The next three series taken would then be at the same conditions with the heater temperature set to 500, 550, and 600 °C respectively. Similar data points would then be recorded for these same operating points with the mean pressure level set at 7.0 MPa. This test procedure, although convenient, can result in misleading data. The reason is that if there were any changes in the instrumentation characteristics that were a function of time, these changes would mistakenly be interpreted as being a result of the changed engine operating conditions. If, on the other hand, the data points are traversed in a random fashion the changes in the instrumentation characteristics will show as scatter in the data. This is also true for changes in the engine as it accumulates run time if there are any parts that are susceptible to wear. This is discussed here to indicate that although it

was recognized that the data matrix should be traversed in a completely random fashion, it was felt that it was not possible within the time frame of the sensitivity tests.

SAMPLE TEST DATA POINT

Some sample data points will be discussed in this section to provide a better understanding of the format used to present the data output. Table VI contains all of the data points that have only one of the set conditions varied from the design conditions. Within this set, data points 1006 to 1012 will be examined. The mean pressure reading of the engine was set at 7000 kPa. This can be seen by the mnemonics MEANCP and MEANBP. These are the mean compression space and bounce space pressure readings respectively. The average value of the 12 heater tube temperatures was set at 600 °C which can be seen by the mnemonic TAVHTR. The cooling water inlet temperature was set at 25 °C as shown by TWINCL. The piston stroke was set at 1.8 cm on data point 1006 as indicated by the parameter PISTST and then increased by 0.2 cm on each successive data point. These parameters are specified by the sensitivity test matrix. All other values on the data sheet are a result of the engine operating at these set conditions.

In the column on the left side of the data sheet there are two sections. These sections are labeled "HEAT TO DASHPOT COOLING" and "HEAT TO COOLER". The calculation used in each of these measurements is based on the water mass flow rates and the differential temperature measurements of the respective cooling loops. The parameters TDLDP and TDLCL are the differential temperatures. Note that these measurements are not equal to the outlet temperature minus the inlet temperature in either case. This results from the locations of the thermocouples used for the inlet and outlet temperature measurements versus the locations of the thermocouples used for the differential temperature measurements. It also results from the improved accuracy of the differential temperature measurement versus the subtraction of two absolute temperature measurements. As previously stated, both the inlet and outlet temperatures were estimated to have a probable error of 1 °C while the differential temperature measure has an estimated probable error of 0.2 °C. Subtracting one temperature from another to calculate the differential temperature can lead to substantial errors when the differential temperature and the probable errors are of the same order of magnitude.

In the second column appear the measurements of the electric power input to the heater head and the calculations performed by the Escort data system. The parameter VOLTG is the voltage across the resistance of the heater head and was measured from a point on the electrically positive power tab near the semi-circular bus bar to a similar point on the electrically negative power tab. This measurement therefore was not the true voltage that existed across the heater tubes alone.

The calculated parameters Pressure Factor of the Piston (PFP), and Pressure Factor of the Displacer (PFD), are shown in graphic form in figure 25. The phasor diagram represents the operating condition at the design point. The format of the diagram follows the vector notation outlined in reference 13. The pressure wave of the engine is divided into two components, one in phase with the piston motion and one in phase with the displacer motion. In general, the component of the pressure wave in phase with the piston motion has been

caused by the piston motion. Similarly, the component of the pressure wave in phase with the displacer motion has been caused by the displacer motion. The vector PFP is the pressure variation in the working space that is the result of the piston motion, divided by the amplitude of the piston. The vector PFD is the pressure variation in the working space that is the result of the displacer motion, divided by the amplitude of the displacer. This provides an easy check on the effectiveness of the piston and displacer motions although care must be taken since it is not an exact measurement of the performance of the piston or displacer. By definition, the PFP vector is parallel to the power piston vector and the PFD vector is parallel to the displacer position vector.

A calculation of the Beale number is included in this column. The Beale number, as outlined in reference 13, gives an indication of how well the engine makes use of its swept volume and pressure level for the conversion of energy. All of the calculations are presented in appendix B.

The third column of the data sheet has several different types of measurements. The parameter PRESUP is the supply pressure to the engine. MEANBP and MEANCP are the average pressure levels of the bounce space and the working space respectively. Vibration transducers were mounted on the large base plate of the engine. The measurement represents the vibration level of the entire engine housing, measured in terms of peak velocity (cm/sec). This measurement, along with the frequency of the engine, can be used to calculate the motion of the engine housing.

The fourth column presents measurements of the mean temperatures of the gas inside the engine housing. Below these readings is a section labeled "REMOTE CALCULATIONS". The four parameters represent calculations executed by the analog circuits. All of the parameters listed under the title "DYNAMIC CALCULATIONS" were measured from the Visicorder traces. PAMPC is the amplitude of the pressure variation in the compression space. PISTCP and DISPCP give the center position of the power piston and the displacer: i.e., the distance from inward limit of the possible movement to the center of the oscillatory movement. As an example, for data point 1010 the center positions measured were 2.33 cm for the power piston and 2.37 cm for the displacer. It was stated in the "Centering Port" subsection of the section "Description of the Engine", that the centering port of the power piston opened when the power piston was 2.05 cm from the inward limit of its motion and the displacer 1 centering port opened when its position was 1.81 cm from its inward limit. This indicates that for data point 1010, the center of the center power piston stroke was $2.33 - 2.05 = 0.28$ cm from the center port location. Similarly, the displacer was operating $2.37 - 1.90 = 0.47$ cm from the center port location.

Measurements presented in this column also include the pressure amplitude inside the gas spring of displacer number 2, PDYNDB. The last three parameters listed give the amplitudes of the dynamic pressure drop measurements. PDLCLR and PDLREG are the pressure drop amplitudes across the cooler and the regenerator respectively. PDLDIS is the pressure drop across the total heat exchanger system as measured between the expansion and compression spaces.

The last column gives the individual heater tube temperature readings. From these individual readings, the average heater tube temperature was calculated and the heater temperature was set. There is some scatter in the readings of the tube temperatures. There are two effects causing this scatter.

First, there is some variation in the temperatures of the heater tubes due to the slight tube to tube variation in electrical resistance. Second, the voltage gradient along the length of the heater tube may effect the accuracy of the temperature measurement. The voltage gradient can be either added to or subtracted from the voltage signal of the thermocouple if the thermocouple is not either electrically insulated from the heater tube or known to be mounted perpendicular to the voltage gradient. Both of these effects probably contributed to the variation in the measurements, but since there were 12 thermocouples, the average of all the heater temperatures should still represent the operating temperature of the heater. Another indication that misalignment of thermocouples was responsible for some of the scatter in the heater temperatures was that a heater tube that indicated the highest temperature with one set of thermocouples was not always the same tube that indicated the highest temperature with the next set of thermocouples. Typically the thermocouples were replaced during each rebuild.

The temperatures in the lower part of the last column are the outer wall temperatures of the regenerator. The locations of these measurements are shown in figure 23. The equation used to determine the axial conduction losses of the regenerator wall from these measurements are listed in appendix B.

When any measurement is made, there is some amount of error associated with each reading. In the case of the RE-1000, being a free-piston Stirling engine (FPSE), the measurement errors are important and in some instances highly significant when calculations are performed. This must be considered when the data is being used to validate computer simulations of the engine cycle. If a code is being validated to produce good agreement in the prediction of indicated power with test data, the validation work can go no further when the code predicts a value of indicated power that is within the error band of the data. For a FPSE, the error band on the indicated power measurement is at times significant. The reason is that there are errors associated with: (1) the volume measurement, (2) the dynamic pressure measurement, and (3) matching the time scales of the two measurements. While it is possible to measure each of these parameters with acceptable accuracy, the size of the error of each signal relative to the size of the signal must be considered. When synchronizing time scales, any phase shift of the two signals relative to one another can lead to a sizeable error in the calculated value of indicated power since the phase angle between the pressure and volume signals is small. Any error in the synchronizing of signals will have a large effect when added to this relatively small phase angle.

With proper technique, measurement errors can be kept to a minimum. An example of minimizing measurement error can be seen in the calculation of heat rejected by the engine cooler. The thermocouples used measuring the temperature rise of the water as it passed through the engine cooler were connected in a differential temperature configuration. If the differential thermocouple configuration has a maximum error of 1.0 percent then the reading will indeed be very accurate. If on the other hand the differential temperatures were measured using two separate thermocouples, there will be an error associated with each of the two temperature measurements. The summation of these two errors will most likely be quite large when compared to the temperature rise intended to be measured.

Estimated probable errors associated with each of the parameters measured in this test program are listed in table IV. The error associated with each of

the calculations is presented in appendix B. In cases where the exact value of the probable error of an individual parameter was not known an estimate was made based on experience with that particular type of instrumentation.

DESCRIPTION OF THE TEST MATRIX

The test matrix was devised to generate the data necessary to characterize the sensitivity of the RE-1000 with respect to each of the parameters that define the test point, plus several configuration related parameters. The attempt was made to optimize the matrix to minimize the number of test conditions and data points. While the engine was designed to operate at temperature levels up to 650 °C, for the reasons of reliability the test matrix was limited to values of 600, 550, 500, and 450 °C. Since power and efficiency are also dependent on the cooler temperature, three levels were chosen: 25, 40, and 55 °C. The engine was designed to operate with helium as the working fluid at a mean pressure level of 7.0 MPa (1015 psig). The test matrix used pressures of 7.0 MPa (1015 psig), 5.5 MPa (798 psig), and 4.0 MPa (580 psig). Testing the engine at four heater temperatures, three cooler temperatures, and three pressure levels, would generate 36 different test conditions for each configuration. To maintain the size of the test matrix at a reasonable level, it was decided that the 14 conditions shown in dashed lines in figure 3 were eliminated. The conditions remaining in figure 3 were used for all four combinations of the two displacers and two regenerators. Figure 4 shows that only displacer 1 and regenerator 1 were used when running tests with either the light weight power piston or an alternate working fluid.

Figure 4 shows the number of data points recorded for each of the configurations. The seven versions of the test condition matrix shown in figure 26 give the data point reading numbers for each of the conditions. With this figure, the reading number for any desired condition can be found. The lower of the two numbers that appear in an element is the reading number of the data point recorded at the shortest power piston stroke tested. In all cases, this value was 1.8 cm. Each successive data reading had the power piston stroke increased by 0.2 cm. As an example, if an element has only four data readings listed in it, the engine was tested with power piston stroke settings of 1.8, 2.0, 2.2, and 2.4 cm.

All 781 data points are not needed to validate a computer code, so the test matrix was divided into two smaller subsets. These two matrixes are called the "Single Variable Data Set" (SVDS) and the "Double Variable Data Set" (DVDS). A list of the data points in the SVDS is given in table VII and a list of the data points in the DVDS is given in table VIII. The remainder of the data can be obtained on microfiche from NASA Lewis by request.

The RE-1000 was designed for high efficiency at the set conditions of 600 °C heater head temperature, 25 °C cooling water inlet temperature, 7.0 MPa mean working pressure with helium and a power piston stroke setting of 2.6 cm. The configuration used included the standard power piston, regenerator 1 and displacer 1. This test condition was recorded on data reading number 1010. Since there were eight test parameters varied during the sensitivity tests, there should be eight different curves that can be plotted showing the sensitivity of the engine to any one of the eight parameters, while the other seven parameters are kept at their respective design conditions. If these plots were generated, every one would contain data reading 1010. The collection of all of

the data points needed to produce these plots is called the SVDS. The title Single Variable Data Set indicates that only one of the eight variable parameters was allowed to deviate from the design condition. Table VII gives the data point reading numbers for all of these points, and also shows which data points are needed to generate each of the eight curves. Table VII shows that the SVDS contains only 18 data points and is therefore an easier data set to work with for computer code validation. This data set can be used to check the sensitivity of a computer code prediction against the sensitivity of the engine for any of the eight test parameters varied.

Figure 27 shows some plots generated from the SVDS. These plots indicate that performance of the RE-1000 obeys the general trends attributed to free-piston Stirling engines. As the power piston stroke is increased, the indicated power produced increased. The displacer stroke also grew as the power piston stroke increased. The displacer stroke growing can be attributed to the larger pressure wave in the working space when the piston is sweeping more volume. The larger pressure wave provides more driving force for the displacer to respond to during operation. It can also be seen in figure 27 that both the indicated power and the efficiency increase as the heater temperature increases.

In some cases the plots generated from the SVDS contain only two points. This is true for any plot of the engine performance with the power piston mass as the independent variable. The engine was tested with only two different levels of power piston mass. This is also true for the variations of the displacer dynamics, the regenerator porosity, and even with the working fluid. The reason this is true for the working fluid is that even though the engine was tested with three different working fluids, the design conditions for the other seven variables could not be reached when tests were being conducted with argon as the working fluid. In an attempt to expand the collection of data points in the SVDS, the DVDS was formed. The DVDS was generated from the SVDS by taking each data point in the SVDS and allowing the power piston stroke to be varied over its entire range. By doing this, a collection of 79 data points was formulated. The list of the data points in this collection is shown in table VIII with the lowest value number of each set being the reading number for the set point condition indicated and the power piston stroke set at 1.8 cm. Each successive reading number had the power piston stroke increased by 0.2 cm. Appendix C contains the data of the DVDS printed out in the same format as described earlier. Figure 28 shows some plots of engine performance using data from the DVDS. Unlike the SVDS, the DVDS can be used to generate plots that show families of curves. Once again it can be seen that the RE-1000 obeys the general laws of FPSE operation.

Some data not included in the SVDS or in the DVDS but that still may be of importance are presented in appendix D. These data points were recorded with both the high power displacer and the high power regenerator installed in the engine. A description of the data points can be found in appendix D.

USE OF THE TEST DATA

There are some important characteristics of the test data that must be understood when the data is being used to validate a computer simulation. These areas will be discussed to provide better insight into the interpretation

of the test data. In all cases the data will be shown to be completely valid, but it will also be shown that the data must be used in the proper manner.

As was described in the section on the sample data output, the motions of the displacer and the power piston were measured in terms of their strokes and their phase relationship to one another. The compression space pressure wave was also characterized by its amplitude and its phase angle relative to the power piston. This assumes that all three of these parameters can be characterized as sinusoidal in nature, without any higher harmonics. A study was performed to test the validity of this approach with the NASA Lewis Stirling engine simulation used in the constrained mode. More information on the details of the NASA Lewis simulation can be found in references 3, 4, 5, and 6. Some of the data points from the SVDS were chosen for this test, and the measured piston motion and displacer motion were digitized. These digitized wave forms were analyzed and their respective Fourier series were generated. Both the power piston and the displacer motion are dominated by the fundamental frequency with only a small part of the motion attributed to the harmonics. Higher harmonics of the fundamental frequency accounted for 1.4 percent of the piston motion and 3.5 percent of the displacer motion. The series solution for each of these wave forms, up to the third harmonic of the fundamental frequency were used as inputs to the NASA Lewis computer code. The results of these simulations were compared to the results of the simulations run with the inputs being in the same form as the data presented in this report. The difference in the code predictions with the two different forms of inputs was negligible. This comparison was made between the two different methods of inputting the data without looking at how well any of the code predictions compared to the test results. It was a test of how similar the two input methods were to one another, not a test of how well the simulation predicted the true engine performance. In the case of the simulations run at the design point of the engine, the results differed by less than 4 percent from one another in the prediction of indicated power. It was felt that this was smaller than the experimental error incurred in the measurement of indicated power and was therefore valid.

This style of data presentation may not be valid for other FPSE engines. The reason that it works well for the RE-1000 is due to the power piston being quite massive and therefore tending to have a sinusoidal motion with little of any higher harmonics. This is reinforced by the load being a pure damper unit which has sinusoidal load characteristics. The same would most likely not be true for a load like the hydraulic pump being tested at NASA Lewis, since the force exerted on the engine is far from sinusoidal. More about the hydraulic load can be found in reference 14. The hydraulic load can cause the engine motions to become highly nonsinusoidal in nature.

Another topic that warrants discussion is the use of the heater tube temperature measurements. The heater tubes of the RE-1000 are heated by resistance heating. The heat is generated inside the heater tube walls. An insulation blanket was placed around the heater head to help prevent heat losses from the head to the surroundings. The thermocouples used to measure the heater tube temperatures were mounted on the outside surface of the tubes indicated in figure 23 and can be seen installed on the heater head in figure 5. If the insulation around the heater head was perfect and with no loss of heat from the head, the temperature gradient through the wall of the heater tube would be flat at the outer surface of the tube. This is very different from the situation that normally exists in Stirling engines where the

heat input is from a combustion source. Since the insulation around the heater head was not perfect, there was some loss of heat from the heater tubes to the atmosphere around the engine. Therefore a temperature gradient exists at the outer surface of the tube that conducts heat toward the outer surface. This creates a point at some distance inside the tube material with a higher temperature than the measured outer wall temperature. The temperature at the inner surface of the heater tube would, as expected, be at some lower temperature because the working fluid is absorbing heat. Figure 29 shows the temperature gradients in these cases discussed.

It would increase the accuracy of a simulation if a correction method was implemented for inputting the measured outer wall temperature. The correction should account for the temperature gradient in a resistance heated tube being different than the gradient in a combustion heated tube. A comparison of the measured expansion space mean gas temperature to the calculated values may provide a means of assuring a correct simulation of the heater temperature.

It was found during these tests that even the measurement of the mean pressure level of the engine needs to be interpreted properly for best results. These measurements were made with strain gauge pressure transducers connected to the engine by means of stainless steel tubing. A small orifice was inserted into each of the stainless steel tubes used for the two mean pressure measurements. In the case of the mean working space measurement, the orifice was a micrometer needle valve with a taper of 1° on the needle. This orifice size could therefore be adjusted as the engine was running. It was found that the reading of the mean pressure could be distorted by changing the size of the orifice. If the orifice was too big, some of the pressure wave present in the engine would get to the transducer and make the output of the transducer fluctuate at the operating frequency of the engine. If the orifice size became too small, the reading of the mean pressure would become higher without the actual engine pressure being raised.

This phenomenon was caused by the change in the density of the working fluid during an engine cycle. If the mean pressure in the engine and the pressure in the transducer were at the exact same level at the initiation of a cycle, there would be the same volumetric flow of gas from the engine to the transducer during the high pressure half of the cycle as the volumetric flow from the transducer to the engine during the low pressure half of the cycle. Due to the change in the density of the gas in the working space during the cycle, the mass flow would not be the same in both directions. This causes a net mass flow of gas toward the transducer which continues until the mean pressure in the transducer is above the mean pressure in the engine to force the net mass flow to become zero.

To prove that this actually did effect the pressure readings of the engine, the output signals of the mean pressure transducers were connected to a strip chart recorder to record the pressure readings versus time. With the engine running at a steady-state condition, the orifice size of the mean compression space transducer (MEANCP), was varied. As the orifice size was reduced, the output signal of the transducer increased. As the orifice size was increased, the output signal was lowered until a point was reached where the pressure wave of the engine started to show in the output signal of the transducer. During this test the output of the mean bounce space pressure transducer (MEANBP), was observed to remain constant.

Because of these results, it may be more accurate to use the value of MEANBP as a representation of the true mean pressure level of the engine. This transducer is less susceptible to this sort of error because the pressure wave in the bounce space is extremely small in the RE-1000. The true mean pressure of the working space should be higher than that of the bounce space only by the amount needed to support the weight of the power piston. The pressure required is about 20.9 kPa (3.03 psi) for the standard power piston.

During the recording of some data points, 1020 to 1025, a failure in the thermocouples that measured the temperature differential of the water used to cool the engine resulted in erroneous calculations of the heat rejection from the engine cooler. These readings were for a series of points with the average heater tube temperature set at 500 °C. Figure 30 shows the recorded values of cooler heat rejection versus the power piston stroke. The 500 °C curve clearly does not follow the trends of the other curves. By using a curve fitting technique applied to the data points with the heater tube average temperature set at 450, 550, and 600 °C, values for the heat rejected during the 500 °C run were predicted. The predicted values of heat rejection are also plotted. In addition, figure 30 shows the measured and predicted values of the engine efficiency plotted as a function of heater temperature and piston stroke. From the closeness of the curve fitting technique used, it is felt that these analytical values are valid when coupled with the experimental data.

On some data points, the instrumentation used to measure QCOOLR proved to be correct while the trends of the internal efficiency were opposite to the expected classical Stirling engine trends. It was noted that as the cooling water inlet temperature was raised the calculation of INTEFF diverted from the calculation of EXTEFF. The only difference between the two methods used to calculate the engine brake efficiency was the representation of the heat input to the engine system. EXTEFF was based on the electric power supplied to the heater head while INTEFF was based on the summation of the power and heat flowing out of the engine system. The calculated values of efficiency agreed well with one another for data with the cooling water inlet temperature at 25 °C, however, their values diverged as the cooling water inlet temperature was raised to the values of 40 and 55 °C.

It was found that when the cooling water inlet temperature was at 25 °C, there was little heat transferred between the cooler housing and the atmosphere around the engine. At elevated cooling water temperatures some heat was lost directly from the cooler housing to the atmosphere without being detected as an increase in the cooling water temperature as it passed through the cooler. To derive a correction for this measurement would require an assumption to be made so that the heat transfer from the cooler housing to the atmosphere could be calculated. It is best to exclude the use of data points having elevated cooling water inlet temperatures when accurate energy balances are required.

CONCLUDING REMARKS

The effort to complete the sensitivity tests at the NASA Lewis test facility has been completed. A matrix with 8 parameters varied was used to determine the test conditions. The RE-1000 provided a reliable test bed for the investigation of the thermodynamics of the Stirling cycle along with the dynamics of a free-piston engine. Throughout the test program the engine

demonstrated high reliability. Most of the problems encountered involved the instrumentation used to measure the engine performance.

A complete characterization of the engine performance has been recorded and is presented in this report for the purpose of Stirling cycle computer code validation. Some of the measurements made can be of great help in understanding the losses of the Stirling cycle. For some of the data points recorded, the amplitude of the dynamic pressure drop through the heat exchangers was recorded. The measurement technique used for the dynamic pressure drop had to be perfected and therefore was not made during the initial stages of the test program. However, steady-state flow tests were performed on all of the heat exchangers used in the program. During tests utilizing one of the two displacers, the amplitude of the gas pressure inside of the displacer gas spring was recorded. Due to the nonlinearity of a gas spring, this parameter can be of great use in the computer simulation validation process.

SUMMARY OF RESULTS

The data are generally quite accurate and lend themselves well to validation purposes, especially when used as per the recommendations outlined in this report. Because there is some error associated with the measurement of any experimental parameter, care must be used when validating a computer simulation. A rough indication of the error of each measurement and calculation is given in this report.

The RE-1000 was designed to be a reliable, research engine intended to provide easy access to key areas for measurements of critical parameters. This engine was not optimized as a complete engine for high power or for high efficiency. The sensitivity test program did however obtain the following results:

1. The trends contained in the data agree well with the theoretical rules of Stirling engine performance. Power output and efficiencies measured act in accordance with previous engine tests.

2. The test program investigated many characteristics of the engine.

- a. Tests were completed over a test matrix which varied the mean operating pressure to three levels, the heater temperature to four levels, the cooler temperature to three levels, both the regenerator porosity and the displacer dynamics to two different values, the power piston mass to two values and used three different working fluids. In addition, the power piston stroke was varied over its range for each of the operating conditions.

- b. In the nearly 200 hr of operation accumulated during the sensitivity tests, the engine proved itself to be a reliable test bed for research. The engine was usually found to be more reliable than the instrumentation and supporting data system.

- c. During the course of the testing, some novel instrumentation was added to provide a better insight into the operation of the free-piston engine. These measurements included the dynamic pressure drop transducers and the dynamic pressure reading of the displacer gas spring.

d. Steady-state flow tests were performed to enable a correlation between these measurements and the dynamic pressure drop measurements to be made. Although the dynamic pressure drop transducers were working only for the tests performed with one of the two regenerators used, this should still prove to be of great benefit in the validation of computer codes.

APPENDIX A

Heat Exchanger Steady-State Flow Tests

Steady-state flow tests were performed on the RE-1000 heat exchanger assembly. These tests were done for the engine with both of the regenerators used in the sensitivity tests. The two regenerators tested were of 75.9 percent porosity and of 81.2 percent porosity. The tests were run with nitrogen at mass flow rates that gave approximately the Reynolds numbers that occur in the engine during operation.

The pressure drops were measured with the Validyne differential pressure transducers that were installed on the engine for the dynamic pressure drop measurements. The cooler pressure drop and the regenerator pressure drop were measured directly with the transducers installed. The heater pressure drop was calculated from a measurement of the total pressure drop across the heat exchanger assembly, minus the regenerator and cooler pressure drops. The tests were run from zero flow rate, up to the maximum flow desired and back down to zero flow again. If the transducers did not return to zero pressure drop at zero flow, the transducers were adjusted and the tests were rerun.

Tests were run with the mean pressure at either 1380 kPa (200 psi) or at 2070 kPa (300 psi) depending on the requirements of the flow meters being used for the particular test. The mass flow rates were measured with venturi-type flow meters. The measurements of flow rate were corrected for the variations in the temperature of the nitrogen.

The results of these flow tests are presented in the form of graphs of the pressure drops measured versus the mass flow rate. These plots are shown in figure 7 for the tests with regenerator 1 (75.9 percent porosity) and in figure 8 for regenerator 2 (81.2 percent porosity).

APPENDIX B

Data System Calculations

All of the calculations used in the data system are presented in this appendix. For each of the calculations, the parameters used as inputs are listed, the calculation is shown, and an estimate of the error of the calculation is given. The error is based on the estimated errors of the input parameters and was calculated by methods outlined in reference 11. The error of the calculation corresponds to the design point of the engine, data point 1010. A second error calculation is presented that corresponds to data point 1099. This shows the error in the calculations at a low power operating condition. The inputs to the calculations are listed in terms of their mnemonics and can be found in the list of instrumentation in table IV.

CALCULATION 1

ELECTRICAL POWER INPUT (WATTS)

power = voltage x current

PWRIN = voltg x amps

VOLTG heater head voltage (volts)

AMPS calculation 8 (amps)

$$\text{ERROR} = [(0.01)^2 + (0.02)^2]^{1/2}$$

$$\text{ERROR} = 2\% \quad (73 \text{ W})$$

$$(\text{ERROR} = 4\% \text{ @ POINT 1099})$$

CALCULATION 2

HEAT TO THE COOLING WATER BY THE COOLER (WATTS)

heat = (mass flow) (specific heat) (delta temp)

$$\text{QCOOLR} = \frac{(\text{FLOCLR}) (\text{RHOH}_2\text{O}) (\text{CPH}_2\text{O}) (\text{TDLCLR})}{60}$$

FLOCLR cooling water volumetric flow rate (l/min)

RHOH₂O water density sub-routine (gm/l)

CPH₂O water specific heat sub-routine (joules/gm °C)

TDLHTR cooling water delta temperature (°C)

$$\text{ERROR} = [(0.1/4.22)^2 + (0.2/9.34)^2]^{1/2}$$

$$\text{ERROR} = 3\% \quad (82 \text{ W})$$

$$(\text{ERROR} = 6\% \text{ @ POINT 1099})$$

CALCULATION 3

HEAT TO THE COOLING WATER BY THE DASHPOT (WATTS)

heat = (mass flow) (specific heat) (delta temp)

$$\text{QDSHPT} = \frac{(\text{FLODP}) (\text{RHOH}_2\text{O}) (\text{CPH}_2\text{O}) (\text{TDLDP})}{60}$$

FLODP dashpot water volumetric flow rate (l/min)

RHOH₂O water density sub-routine (gm/l)

CPH₂O water specific heat sub-routine (joules/gm °C)

TDLDP dashpot water delta temperature (°C)

$$\text{ERROR} = [(0.1/3.96)^2 + (0.2/2.21)^2]^{1/2}$$

$$\text{ERROR} = [(0.1/3.96)^2 + (0.2/2.21)^2]^{1/2}$$

$$\text{ERROR} = 9\% \quad (55 \text{ W})$$

$$(\text{ERROR} = 51\% @ 1099)$$

CALCULATION 4

BRAKE ENGINE EFFICIENCY, EXTERNAL (%)

$$\text{efficiency} = \frac{\text{brake power output}}{\text{electric power input}}$$

$$\text{EXTEFF} = \frac{\text{PWROUT}}{\text{PWRIN}}$$

PWROUT brake power output, analog circuit (W)
PWRIN electric heater power input (W)

$$\text{ERROR} = [(0.03)^2 + (0.02)^2]^{1/2}$$

$$\text{ERROR} = 4\%$$

$$(\text{ERROR} = 6\% @ \text{POINT } 1099)$$

CALCULATION 5

AVERAGE HEATER TUBE TEMPERATURE (°C)

$$\text{temperature} = \frac{\text{summation of 12 tube temps}}{12}$$

$$\text{TAVHTR} = \frac{\text{summation T01HTR to T12HTR}}{12}$$

TxxHTR heater tube outside wall temperature (°C)

$$\text{ERROR} = [12(10/600)^2]^{1/2}$$

$$\text{ERROR} = 6\% \quad (36 \text{ °C})$$

$$(\text{ERROR} = 8\% @ \text{POINT } 1099)$$

CALCULATION 6

BRAKE ENGINE EFFICIENCY, INTERNAL (%)

$$\text{efficiency} = \frac{\text{brake power output}}{\text{brake power} + \text{heat rejected}}$$

$$\text{INTEFF} = \frac{\text{PWROUT}}{\text{PWROUT} + \text{QCOOLR}}$$

PWROUT brake power output, analog circuit (W)
QCOOLR heat rejected by the engine cooler (W)

$$\text{ERROR} = [(0.03)^2 + (0.03)^2 + (0.03)^2]^{1/2}$$

$$\text{ERROR} = 5\%$$

$$(\text{ERROR} = 9\% \text{ @ POINT 1099})$$

CALCULATION 8

HEATER AMPERAGE (°C)

$$\text{amperage} = \text{amperage 1} + \text{amperage 2}$$

$$\text{AMPS} = \text{AMPS1} + \text{AMPS2}$$

AMPS1 amperage from power supply 1 (A)
AMPS2 amperage from power supply 2 (A)

$$\text{ERROR} = [(10/673)^2 + (10/618)^2]^{1/2}$$

$$\text{ERROR} = 2\% \quad (28 \text{ A})$$

$$(\text{ERROR} = 4\% \text{ @ POINT 1099})$$

CALCULATION 9

DISPLACER GAS CONDUCTION (WATTS)

$$\text{heat} = (\text{gas conductivity}) (\text{conduction area}) (\text{delta temp})$$

$$\text{QDISPG} = (\text{Kg}) (2.76) (\text{TGEXP-TGCOMP})$$

Kg gas conductivity (W/cm °C)
2.76 cross section of gas (cm²)
TGEXP expansion space gas temperature (°C)
TGCOMP compression space gas temperature (°C)

$$\text{ERROR} = \frac{10 + 1}{(557-56)}$$

$$\text{ERROR} = 2\% \quad (0.06 \text{ W})$$

$$(\text{ERROR} = 3\% \text{ @ POINT 1099})$$

CALCULATION 10

DISPLACER SHELL CONDUCTION (WATTS)

$$\text{heat} = (\text{shell conductivity}) (\text{conduction area}) (\text{delta temp})$$

$$QDISP = (Km) (0.0284) (TGEXP-TGCOMP)$$

Km displacer metal conductivity (W/cm °C)
 0.00284 conduction area (cm²)
 TGEXP expansion space gas temperature (°C)
 TGCOMP compression space gas temperature (°C)

$$ERROR = \frac{10 + 1}{(557-56)}$$

$$ERROR = 2\% \quad (0.28 \text{ W})$$

$$(ERROR = 3\% @ \text{POINT } 1099)$$

CALCULATION 11

REGENERATOR WALL CONDUCTION, COOL END (WATTS)

$$\text{heat} = (Km) (\text{conduction area}) (\text{delta temp})$$

$$QREG1 = 0.811 (T18REG-T19REG)$$

0.811 metal conductivity times conduction area (W cm/°C)
 T18REG regenerator wall temperature, midpoint (°C)
 T19REG regenerator wall temperature, cool end (°C)

$$ERROR = \frac{5 + 5}{(339-221)}$$

$$ERROR = 8\% \quad (7.7 \text{ W})$$

$$(ERROR = 9\% @ \text{POINT } 1099)$$

CALCULATION 12

REGENERATOR WALL CONDUCTION, COOL END (WATTS)

$$\text{heat} = (Km) (\text{conduction area}) (\text{delta temp})$$

$$QREG2 = 0.811 (T14REG-T18REG)$$

0.811 metal conductivity times conduction area (W/cm °C)
 T14REG regenerator wall temperature, hot end (°C)
 T18REG regenerator wall temperature, midpoint (°C)

$$ERROR = \frac{10 + 5}{(339-221)}$$

$$ERROR = 9\% \quad (12.5 \text{ W})$$

$$(ERROR = 13\% @ \text{POINT } 1099)$$

CALCULATION 13

INNER REGENERATOR WALL CONDUCTION (WATTS)

$$\text{heat} = (K_m) (\text{conduction area}) (\Delta \text{temp})$$

$$Q_{REG3} = 0.0774 (T_{GREGH} - T_{GREGC})$$

0.0774 metal conductivity times conduction area (W cm/°C)
 T_{GREGH} regenerator hot end gas temperature (°C)
 T_{GREGC} regenerator cold end gas temperature (°C)

$$\text{ERROR} = \frac{10 + 1}{(549 - 91)}$$

$$\text{ERROR} = 2\% \quad (0.7 \text{ W})$$

$$(\text{ERROR} = 3\% \text{ @ POINT 1099})$$

CALCULATION 14

PRESSURE FACTOR OF THE POWER PISTON (kPa/cm)

$$\text{factor} = \frac{(\text{compression space pressure amplitude}) (\text{trig term})}{(\text{piston amplitude})}$$

$$PFP = \frac{2(PAMPC) (\cos \theta_p + (\sin \theta_p / \tan \theta_d))}{(PISTST)}$$

PAMPC pressure amplitude in the compression space (kPa)
 θ_p phase angle, pressure to power piston (deg)
 PISTST power piston stroke (cm)

$$\text{ERROR} = 3\%$$

$$(\text{ERROR} = 5\% \text{ @ POINT 1099})$$

CALCULATION 15

PRESSURE FACTOR OF THE DISPLACER (kPa/cm)

$$\text{factor} = \frac{(\text{compression space pressure amplitude}) (\text{trig term})}{(\text{displacer amplitude})}$$

$$PFD = \frac{(PAMPC) (\sin \theta_p)^2}{(DISPST) (\sin \theta_d)}$$

PAMPC pressure amplitude in the compression space (kPa)
 θ_p phase angle, pressure to power piston (deg)
 DISPST displacer stroke (cm)
 θ_d phase angle, displacer to power piston (deg)

ERROR = 2%

(ERROR = 4% @ POINT 1099)

CALCULATION 16

BEALE NUMBER

$$\text{number} = \frac{\text{brake power output}}{(\text{mean pressure}) (\text{swept volume}) (\text{frequency})}$$

$$\text{NBEALE} = \frac{(100) \text{PWROUT}}{(\text{MEANBP}) (\text{PISTST}) (\text{AREA}) (\text{FREQ})}$$

PWROUT brake power output, analog circuit (W)
MEANBP mean bounce space pressure (kPa)
PISTST power piston stroke (cm)
AREA cross section area of the power piston (cm²)
FREQ operating frequency of the engine (Hz)

ERROR = 3%

(ERROR = 5% @ POINT 1099)

APPENDIX C

Double Variable Data Set, (DVDS)

Listing of the DVDS are presented in this appendix. These data points were determined by letting the stroke of the power piston be varied over its range for each one of the data points of the Single Variable Data Set (SVDS). The format of the listing is the same as used for the SVDS.

APPENDIX C. - DOUBLE VARIABLE DATA SET, (DVDS)

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 09:56:39.76 RDG 598

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING FLODP 4.17 L/MIN TWINDP 16.5 DEG.C TDLDP 1.55 DEG.C TWO DPR 18.6 DEG.C	POWER IN AMPS1 606. AMPS AMPS2 535. AMPS VOLTG 2.50 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7411. KPA MEANBP 7003. KPA MEANCP 7026. KPA	GAS TEMPERATURES TGEXP 566.0 DEG.C TGREGH 547.6 DEG.C TGREGC 96.2 DEG.C TGCOMP 45.7 DEG.C TGBOUN 28.7 DEG.C	SURFACE TEMPERATURE T01HTR 591.2 DEG.C T02HTR 617.6 DEG.C T03HTR 610.5 DEG.C T04HTR 602.9 DEG.C T05HTR 609.2 DEG.C T06HTR 586.2 DEG.C T07HTR 586.5 DEG.C T08HTR 604.9 DEG.C T09HTR 621.1 DEG.C T10HTR 552.6 DEG.C T11HTR 601.0 DEG.C T12HTR 609.9 DEG.C
HEAT TO COOLER FLOCLR 4.02 L/MIN TWINCL 24.8 DEG.C TDLCL 6.42 DEG.C TWOCLR 31.19 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 2848. WATTS 2 QCOCLR 1789. WATTS 3 QDSHPT 450. WATTS 4 EXTEFF 24.9 % 5 TAVHTR 599.5 DEG.C 6 INTEFF 28.4 % 8 AMPS 1140. AMPS 9 QDISPG 3. WATTS 10 QDISP 15. WATTS 11 QREG1 103. WATTS 12 QREG2 122. WATTS 13 QREG3 35. WATTS 14 PFP 1024.5 KPA/CM 15 PFD 385.1 KPA/CM 16 NBEALE 0.00741	VIBRATION VX1HOR 0.4 CM/S VY1VER 5.7 CM/S PHASE ANGLES PADISP 54.7 DEG. PAPRES -27.1 DEG. ENGINE SPEED FREQ 29.9 HZ	REMOTE CALCULATIONS PWROUT 709. WATTS INDPWR 739. WATTS PISTST 1.78 CM DISPST 2.18 CM DYNAMIC CALCULATIONS PAMPC 753. KPA DISPCP 1.89 CM PISTCP 2.50 CM PDYNDB***** KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 566.3 DEG.C T14REG 532.1 DEG.C T15REG 406.3 DEG.C T16REG***** DEG.C T17REG 357.7 DEG.C T18REG 381.5 DEG.C T19REG 254.3 DEG.C T03HED 221.0 DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 10:01:48.78 RDG 599

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING FLODP 4.16 L/MIN TWINDP 16.7 DEG.C TDLDP 1.83 DEG.C TWO DPR 19.1 DEG.C	POWER IN AMPS1 643. AMPS AMPS2 568. AMPS VOLTG 2.65 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7147. KPA MEANBP 6996. KPA MEANCP 7019. KPA	GAS TEMPERATURES TGEXP 563.3 DEG.C TGREGH 547.3 DEG.C TGREGC 99.6 DEG.C TGCOMP 49.4 DEG.C TGBOUN 30.1 DEG.C	SURFACE TEMPERATURE T01HTR 594.8 DEG.C T02HTR 614.9 DEG.C T03HTR 612.3 DEG.C T04HTR 601.6 DEG.C T05HTR 607.2 DEG.C T06HTR 589.3 DEG.C T07HTR 583.0 DEG.C T08HTR 604.1 DEG.C T09HTR 622.0 DEG.C T10HTR 550.5 DEG.C T11HTR 601.0 DEG.C T12HTR 607.2 DEG.C
HEAT TO COOLER FLOCLR 4.06 L/MIN TWINCL 24.9 DEG.C TDLCL 7.14 DEG.C TWOCLR 32.02 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 3214. WATTS 2 QCOCLR 2010. WATTS 3 QDSHPT 531. WATTS 4 EXTEFF 26.1 % 5 TAVHTR 599.0 DEG.C 6 INTEFF 29.4 % 8 AMPS 1212. AMPS 9 QDISPG 3. WATTS 10 QDISP 15. WATTS 11 QREG1 101. WATTS 12 QREG2 123. WATTS 13 QREG3 35. WATTS 14 PFP 1015.6 KPA/CM 15 PFD 376.6 KPA/CM 16 NBEALE 0.00783	VIBRATION VX1HOR 0.4 CM/S VY1VER 6.4 CM/S PHASE ANGLES PADISP 55.4 DEG. PAPRES -25.6 DEG. ENGINE SPEED FREQ 29.8 HZ	REMOTE CALCULATIONS PWROUT 839. WATTS INDPWR 878. WATTS PISTST 2.00 CM DISPST 2.36 CM DYNAMIC CALCULATIONS PAMPC 846. KPA DISPCP 1.99 CM PISTCP 2.58 CM PDYNDB***** KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 563.8 DEG.C T14REG 529.8 DEG.C T15REG 404.5 DEG.C T16REG***** DEG.C T17REG 364.9 DEG.C T18REG 378.5 DEG.C T19REG 254.0 DEG.C T03HED 248.6 DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 10:06:54.78 RDG 600

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING FLODP 4.13 L/MIN TWINDP 16.8 DEG.C TDLDP 2.16 DEG.C TWO DPR 19.7 DEG.C	POWER IN AMPS1 692. AMPS AMPS2 607. AMPS VOLTG 2.84 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7428. KPA MEANBP 7001. KPA MEANCP 7031. KPA	GAS TEMPERATURES TGEXP 562.2 DEG.C TGREGH 547.8 DEG.C TGREGC 103.0 DEG.C TGCOMP 53.5 DEG.C TGBOUN 32.1 DEG.C	SURFACE TEMPERATURE T01HTR 599.6 DEG.C T02HTR 617.4 DEG.C T03HTR 617.2 DEG.C T04HTR 603.1 DEG.C T05HTR 610.3 DEG.C T06HTR 592.9 DEG.C T07HTR 584.0 DEG.C T08HTR 606.1 DEG.C T09HTR 625.6 DEG.C T10HTR 550.1 DEG.C T11HTR 603.6 DEG.C T12HTR 609.4 DEG.C
HEAT TO COOLER FLOCLR 4.09 L/MIN TWINCL 25.0 DEG.C TDLCL 8.09 DEG.C TWOCLR 33.11 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 3684. WATTS 2 QCOCLR 2296. WATTS 3 QDSHPT 622. WATTS 4 EXTEFF 26.5 % 5 TAVHTR 601.6 DEG.C 6 INTEFF 29.8 % 8 AMPS 1299. AMPS 9 QDISPG 3. WATTS 10 QDISP 14. WATTS 11 QREG1 101. WATTS 12 QREG2 121. WATTS 13 QREG3 34. WATTS 14 PFP 1027.1 KPA/CM 15 PFD 372.6 KPA/CM 16 NBEALE 0.00823	VIBRATION VX1HOR 0.5 CM/S VY1VER 7.2 CM/S PHASE ANGLES PADISP 56.1 DEG. PAPRES -24.1 DEG. ENGINE SPEED FREQ 29.9 HZ	REMOTE CALCULATIONS PWROUT 976. WATTS INDPWR 1021. WATTS PISTST 2.21 CM DISPST 2.52 CM DYNAMIC CALCULATIONS PAMPC 955. KPA DISPCP 2.01 CM PISTCP 2.58 CM PDYNDB***** KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 564.7 DEG.C T14REG 530.9 DEG.C T15REG 405.9 DEG.C T16REG***** DEG.C T17REG 368.4 DEG.C T18REG 381.4 DEG.C T19REG 256.9 DEG.C T03HED 276.8 DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 10:12:06.78 RDG 601

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.12 L/MIN	AMPS1 742. AMPS	PRESUP 7134. KPA	TGEXP 557.6 DEG.C	T01HTR 598.3 DEG.C
	AMPS2 627. AMPS			T02HTR 615.7 DEG.C
TWINDP 17.1 DEG.C	VOLTG 2.98 VOLTS	MEANBP 6983. KPA	TGREGH 543.3 DEG.C	T03HTR 616.6 DEG.C
TDLDP 2.55 DEG.C		MEANCP 7021. KPA	TGREGC 101.9 DEG.C	T04HTR 599.0 DEG.C
TWODPR 20.2 DEG.C			TGCOMP 57.1 DEG.C	T05HTR 609.3 DEG.C
			TGBOUN 34.1 DEG.C	T06HTR 590.2 DEG.C
				T07HTR 580.9 DEG.C
				T08HTR 603.6 DEG.C
				T09HTR 622.6 DEG.C
				T10HTR 545.0 DEG.C
				T11HTR 601.4 DEG.C
				T12HTR 607.7 DEG.C
				T13REG 561.0 DEG.C
				T14REG 525.9 DEG.C
				T15REG 397.9 DEG.C
				T16REG***** DEG.C
				T17REG 357.2 DEG.C
				T18REG 378.2 DEG.C
				T19REG 252.1 DEG.C
				T03HED 285.9 DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
FLOCLR 4.13 L/MIN	1 PWRIN 4080. WATTS	VX1HOR 0.5 CM/S	PWROUT 1077. WATTS
TWINCL 25.1 DEG.C	2 QCOOLR 2589. WATTS	VY1VER 7.8 CM/S	INDPWR 1125. WATTS
TDLCL 9.03 DEG.C	3 QDSHPT 730. WATTS		PISTST 2.38 CM
TWOCLR 34.11 DEG.C	4 EXTEFF 26.4 %	PHASE ANGLES	DISPST 2.65 CM
	5 TAVHTR 599.2 DEG.C	PADISP 57.1 DEG.	
	6 INTEFF 29.4 %	PAPRES -22.5 DEG.	
	8 AMPS 1369. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS
	9 QDISPG 3. WATTS	FREQ 29.9 HZ	PAMPC 546. KPA
	10 QDISP 14. WATTS		DISPCP 2.05 CM
	11 QREG1 102. WATTS		PISTCP 2.56 CM
	12 QREG2 120. WATTS		PDYNDB***** KPA
	13 QREG3 34. WATTS		PDLCLR***** KPA
	14 PFP 537.5 KPA/CM		PDLREG***** KPA
	15 PFD 188.0 KPA/CM		PDLDIS***** KPA
	16 NBEALE 0.00843		

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 10:17:21.80 RDG 602

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.10 L/MIN	AMPS1 776. AMPS	PRESUP 7482. KPA	TGEXP 554.8 DEG.C	T01HTR 609.3 DEG.C
	AMPS2 651. AMPS			T02HTR 609.9 DEG.C
TWINDP 17.3 DEG.C	VOLTG 3.11 VOLTS	MEANBP 7013. KPA	TGREGH 547.0 DEG.C	T03HTR 619.7 DEG.C
TDLDP 2.94 DEG.C		MEANCP 7056. KPA	TGREGC 101.9 DEG.C	T04HTR 594.8 DEG.C
TWODPR 20.9 DEG.C			TGCOMP 61.6 DEG.C	T05HTR 604.4 DEG.C
			TGBOUN 36.9 DEG.C	T06HTR 602.3 DEG.C
				T07HTR 573.2 DEG.C
				T08HTR 605.9 DEG.C
				T09HTR 619.2 DEG.C
				T10HTR 546.2 DEG.C
				T11HTR 603.4 DEG.C
				T12HTR 603.9 DEG.C
				T13REG 553.7 DEG.C
				T14REG 514.8 DEG.C
				T15REG 376.7 DEG.C
				T16REG***** DEG.C
				T17REG 355.8 DEG.C
				T18REG 366.0 DEG.C
				T19REG 235.5 DEG.C
				T03HED 293.5 DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
FLOCLR 4.17 L/MIN	1 PWRIN 4433. WATTS	VX1HOR 0.5 CM/S	PWROUT 1216. WATTS
TWINCL 25.2 DEG.C	2 QCOOLR 2789. WATTS	VY1VER 8.5 CM/S	INDPWR 1262. WATTS
TDLCL 9.63 DEG.C	3 QDSHPT 840. WATTS		PISTST 2.58 CM
TWOCLR 34.80 DEG.C	4 EXTEFF 27.4 %	PHASE ANGLES	DISPST 2.72 CM
	5 TAVHTR 599.3 DEG.C	PADISP 57.1 DEG.	
	6 INTEFF 30.4 %	PAPRES -21.6 DEG.	
	8 AMPS 1427. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS
	9 QDISPG 3. WATTS	FREQ 29.9 HZ	PAMPC 1127. KPA
	10 QDISP 14. WATTS		DISPCP 2.11 CM
	11 QREG1 106. WATTS		PISTCP 2.64 CM
	12 QREG2 121. WATTS		PDYNDB***** KPA
	13 QREG3 34. WATTS		PDLCLR***** KPA
	14 PFP 1019.5 KPA/CM		PDLREG***** KPA
	15 PFD 362.7 KPA/CM		PDLDIS***** KPA
	16 NBEALE 0.00875		

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:35:55.03 RDG 1006

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.96 L/MIN	AMPS1 534. AMPS	PRESUP 7365. KPA	TGEXP 567.8 DEG.C	T01HTR 601.6 DEG.C
	AMPS2 503. AMPS			T02HTR 605.5 DEG.C
TWINDP 17.7 DEG.C	VOLTG 2.25 VOLTS	MEANBP 7011. KPA	TGREGH 564.5 DEG.C	T03HTR 612.6 DEG.C
TDLDP 1.15 DEG.C		MEANCP 7040. KPA	TGREGC 90.7 DEG.C	T04HTR 606.4 DEG.C
TWODPR 17.6 DEG.C			TGCOMP 39.9 DEG.C	T05HTR 614.3 DEG.C
			TGBOUN 27.9 DEG.C	T06HTR 606.3 DEG.C
				T07HTR 604.5 DEG.C
				T08HTR 584.1 DEG.C
				T09HTR 608.8 DEG.C
				T10HTR 591.8 DEG.C
				T11HTR 578.7 DEG.C
				T12HTR 580.2 DEG.C
				T13REG***** DEG.C
				T14REG 518.3 DEG.C
				T15REG 384.4 DEG.C
				T16REG 386.7 DEG.C
				T17REG 376.6 DEG.C
				T18REG 372.8 DEG.C
				T19REG 225.5 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
FLOCLR 4.22 L/MIN	1 PWRIN 2338. WATTS	VX1HOR 0.2 CM/S	PWROUT 532. WATTS
TWINCL 24.7 DEG.C	2 QCOOLR 1892. WATTS	VY1VER 5.9 CM/S	INDPWR 570. WATTS
TDLCL 6.47 DEG.C	3 QDSHPT 317. WATTS		PISTST 1.80 CM
TWOCLR 29.47 DEG.C	4 EXTEFF 22.7 %	PHASE ANGLES	DISPST 1.83 CM
	5 TAVHTR 599.6 DEG.C	PADISP 56.0 DEG.	
	6 INTEFF 21.9 %	PAPRES -19.8 DEG.	
	8 AMPS 1037. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS
	9 QDISPG 3. WATTS	FREQ 30.2 HZ	PAMPC 792. KPA
	10 QDISP 15. WATTS		DISPCP 2.13 CM
	11 QREG1 119. WATTS		PISTCP 2.31 CM
	12 QREG2 118. WATTS		PDYNDB***** KPA
	13 QREG3 37. WATTS		PDLCLR 3.22 KPA
	14 PFP 1029.1 KPA/CM		PDLREG 37.33 KPA
	15 PFD 353.7 KPA/CM		PDLDIS 65.16 KPA
	16 NBEALE 0.00544		

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:40:22.06 RDG 1007

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.94 L/MIN	AMPS1 571. AMPS	PRESUP 7469. KPA	TGEXP 564.8 DEG.C	T01HTR 602.6 DEG.C
	AMPS2 533. AMPS			T02HTR 605.0 DEG.C
TWINDP 17.8 DEG.C	VOLTG 2.40 VOLTS	MEANBP 6982. KPA	TGREGH 563.7 DEG.C	T03HTR 613.7 DEG.C
TDLDP 1.42 DEG.C		MEANCP 7013. KPA	TGREGC 92.6 DEG.C	T04HTR 605.5 DEG.C
TWODPR 19.8 DEG.C			TGCOMP 42.0 DEG.C	T05HTR 615.7 DEG.C
			TGBOUN 29.0 DEG.C	T06HTR 607.1 DEG.C
				T07HTR 605.4 DEG.C
				T08HTR 583.8 DEG.C
				T09HTR 609.7 DEG.C
				T10HTR 590.4 DEG.C
				T11HTR 577.8 DEG.C
				T12HTR 577.7 DEG.C
				T13REG***** DEG.C
				T14REG 517.9 DEG.C
				T15REG 385.7 DEG.C
				T16REG 388.3 DEG.C
				T17REG 377.9 DEG.C
				T18REG 373.3 DEG.C
				T19REG 228.0 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER
 FLOCLR 4.24 L/MIN
 TWINCL 24.7 DEG.C
 TDLCL 6.82 DEG.C
 TWOCCL 29.95 DEG.C

CALCULATED PARAMETERS
 1 PWRIN 2648. WATTS
 2 QCOOLR 2004. WATTS
 3 QDSHPT 390. WATTS
 4 EXTEFF 23.5 %
 5 TAVHTR 599.5 DEG.C
 6 INTEFF 23.7 %
 8 AMPS 1104. AMPS
 9 QDISPG 3. WATTS
 10 QDISP 15. WATTS
 11 QREG1 118. WATTS
 12 QREG2 117. WATTS
 13 QREG3 36. WATTS
 14 PFP 1018.7 KPA/CM
 15 PFD 345.8 KPA/CM
 16 NBEALE 0.00573

VIBRATION
 VX1HOR 0.3 CM/S
 VY1VER 6.6 CM/S

PHASE ANGLES
 PADISP 56.6 DEG.
 PAPRES -18.9 DEG.

ENGINE SPEED
 FREQ 30.2 HZ

REMOTE CALCULATIONS
 PWROUT 622. WATTS
 INDPWR 665. WATTS
 PISTST 2.01 CM
 DISPST 1.97 CM

DYNAMIC CALCULATIONS
 PAMPC 881. KPA
 DISPCP 2.24 CM
 PISTCP 2.34 CM
 PDYNDB***** KPA
 PDLCLR 3.73 KPA
 PDLREG 42.43 KPA
 PDLDIS 73.65 KPA

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:46:40.06 RDG 1008

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.94 L/MIN	AMPS1 604. AMPS	PRESUP 7055. KPA	TGEXP 562.6 DEG.C	T01HTR 603.8 DEG.C
	AMPS2 557. AMPS			T02HTR 604.7 DEG.C
TWINDP 17.8 DEG.C	VOLTG 2.53 VOLTS	MEANBP 6988. KPA	TGREGH 563.2 DEG.C	T03HTR 615.3 DEG.C
TDLDP 1.72 DEG.C		MEANCP 7023. KPA	TGREGC 94.9 DEG.C	T04HTR 605.1 DEG.C
TWODPR 20.2 DEG.C			TGCOMP 45.4 DEG.C	T05HTR 617.5 DEG.C
			TGBOUN 30.6 DEG.C	T06HTR 608.2 DEG.C
				T07HTR 606.3 DEG.C
				T08HTR 584.0 DEG.C
				T09HTR 610.7 DEG.C
				T10HTR 589.5 DEG.C
				T11HTR 577.5 DEG.C
				T12HTR 576.1 DEG.C
				T13REG***** DEG.C
				T14REG 518.7 DEG.C
				T15REG 388.8 DEG.C
				T16REG 389.2 DEG.C
				T17REG 380.1 DEG.C
				T18REG 376.6 DEG.C
				T19REG 232.8 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER
 FLOCLR 4.23 L/MIN
 TWINCL 24.7 DEG.C
 TDLCL 7.61 DEG.C
 TWOCCL 30.73 DEG.C

CALCULATED PARAMETERS
 1 PWRIN 2930. WATTS
 2 QCOOLR 2233. WATTS
 3 QDSHPT 470. WATTS
 4 EXTEFF 24.0 %
 5 TAVHTR 599.9 DEG.C
 6 INTEFF 23.9 %
 8 AMPS 1160. AMPS
 9 QDISPG 3. WATTS
 10 QDISP 15. WATTS
 11 QREG1 117. WATTS
 12 QREG2 115. WATTS
 13 QREG3 36. WATTS
 14 PFP 1025.2 KPA/CM
 15 PFD 339.6 KPA/CM
 16 NBEALE 0.00589

VIBRATION
 VX1HOR 0.3 CM/S
 VY1VER 7.2 CM/S

PHASE ANGLES
 PADISP 56.7 DEG.
 PAPRES -17.7 DEG.

ENGINE SPEED
 FREQ 30.2 HZ

REMOTE CALCULATIONS
 PWROUT 703. WATTS
 INDPWR 753. WATTS
 PISTST 2.20 CM
 DISPST 2.09 CM

DYNAMIC CALCULATIONS
 PAMPC 979. KPA
 DISPCP 2.27 CM
 PISTCP 2.34 CM
 PDYNDB***** KPA
 PDLCLR 4.07 KPA
 PDLREG 47.86 KPA
 PDLDIS 82.13 KPA

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:50:22.06 RDG 1009

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.97 L/MIN	AMPS1 637. AMPS	PRESUP 7339. KPA	TGEXP 561.2 DEG.C	T01HTR 602.1 DEG.C
	AMPS2 585. AMPS			T02HTR 608.6 DEG.C
TWINDP 17.8 DEG.C	VOLTG 2.66 VOLTS	MEANBP 6999. KPA	TGREGH 558.0 DEG.C	T03HTR 616.2 DEG.C
TDLDP 1.97 DEG.C		MEANCP 7036. KPA	TGREGC 94.3 DEG.C	T04HTR 604.3 DEG.C
TWODPR 20.5 DEG.C			TGCOMP 48.5 DEG.C	T05HTR 620.1 DEG.C
			TGBOUN 32.1 DEG.C	T06HTR 611.6 DEG.C
				T07HTR 610.2 DEG.C
				T08HTR 581.4 DEG.C
				T09HTR 609.4 DEG.C
				T10HTR 588.8 DEG.C
				T11HTR 573.6 DEG.C
				T12HTR 577.0 DEG.C
				T13REG***** DEG.C
				T14REG 517.8 DEG.C
				T15REG 384.9 DEG.C
				T16REG 378.4 DEG.C
				T17REG 364.0 DEG.C
				T18REG 364.0 DEG.C
				T19REG 231.2 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER
 FLOCLR 4.23 L/MIN
 TWINCL 24.7 DEG.C
 TDLCL 8.53 DEG.C
 TWOCCL 31.51 DEG.C

CALCULATED PARAMETERS
 1 PWRIN 3253. WATTS
 2 QCOOLR 2505. WATTS
 3 QDSHPT 545. WATTS
 4 EXTEFF 24.4 %
 5 TAVHTR 600.3 DEG.C
 6 INTEFF 24.1 %
 8 AMPS 1222. AMPS
 9 QDISPG 3. WATTS
 10 QDISP 15. WATTS
 11 QREG1 108. WATTS
 12 QREG2 125. WATTS
 13 QREG3 36. WATTS
 14 PFP 1012.0 KPA/CM
 15 PFD 324.2 KPA/CM
 16 NBEALE 0.00609

VIBRATION
 VX1HOR 0.3 CM/S
 VY1VER 7.9 CM/S

PHASE ANGLES
 PADISP 57.1 DEG.
 PAPRES -16.5 DEG.

ENGINE SPEED
 FREQ 30.2 HZ

REMOTE CALCULATIONS
 PWROUT 794. WATTS
 INDPWR 856. WATTS
 PISTST 2.40 CM
 DISPST 2.22 CM

DYNAMIC CALCULATIONS
 PAMPC 1063. KPA
 DISPCP 2.37 CM
 PISTCP 2.38 CM
 PDYNDB***** KPA
 PDLCLR 5.43 KPA
 PDLREG 52.95 KPA
 PDLDIS 90.28 KPA

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:55:25.06 RDG 1010

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.96 L/MIN	AMPS1	673. AMPS	PRESUP	7464. KPA	TGEXP	557.5 DEG.C	T01HTR	598.1 DEG.C
		AMPS2	619. AMPS					T02HTR	610.5 DEG.C
TWINDP	17.8 DEG.C	VOLTG	2.82 VOLTS	MEANBP	6988. KPA	TGREGH	548.5 DEG.C	T03HTR	614.9 DEG.C
TDLDP	2.21 DEG.C			MEANCP	7032. KPA	TGREGC	90.7 DEG.C	T04HTR	601.4 DEG.C
TWODPR	20.8 DEG.C					TGCOMP	55.7 DEG.C	T05HTR	621.1 DEG.C
						TGBOUN	33.4 DEG.C	T06HTR	613.0 DEG.C
								T07HTR	611.9 DEG.C
								T08HTR	576.7 DEG.C
								T09HTR	605.4 DEG.C
								T10HTR	586.4 DEG.C
								T11HTR	567.1 DEG.C
								T12HTR	576.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWRIN	3643. WATTS	VX1HOR	0.4 CM/S	PWROUT	866. WATTS	T13REG*****	DEG.C
TWINCL	24.8 DEG.C	2 QCOOLR	2736. WATTS	VY1VER	8.5 CM/S	INDPWR	939. WATTS	T14REG	510.3 DEG.C
TDLCL	9.34 DEG.C	3 QDSHPT	607. WATTS			PISTST	2.60 CM	T15REG	368.0 DEG.C
TWOCLR	32.52 DEG.C	4 EXTEFF	23.8 %	PHASE ANGLES		DISPST	2.34 CM	T16REG	350.7 DEG.C
		5 TAVHTR	598.6 DEG.C	PADISP	57.5 DEG.			T17REG	331.5 DEG.C
		6 INTEFF	24.0 %	PAPRES	-15.7 DEG.			T18REG	338.9 DEG.C
		8 AMPS	1292. AMPS					T19REG	221.0 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED				T03HED*****	DEG.C
		10 QDISP	14. WATTS	FREQ	30.1 HZ				
		11 QREG1	96. WATTS						
		12 QREG2	139. WATTS						
		13 QREG3	35. WATTS						
		14 PFP	1002.8 KPA/CM						
		15 PFD	314.1 KPA/CM						
		16 NBEALE	0.00617						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:00:55.09 RDG 1011

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.93 L/MIN	AMPS1	702. AMPS	PRESUP	7063. KPA	TGEXP	555.5 DEG.C	T01HTR	598.9 DEG.C
		AMPS2	653. AMPS					T02HTR	612.9 DEG.C
TWINDP	17.8 DEG.C	VOLTG	2.98 VOLTS	MEANBP	6987. KPA	TGREGH	545.4 DEG.C	T03HTR	617.1 DEG.C
TDLDP	2.67 DEG.C			MEANCP	7034. KPA	TGREGC	92.1 DEG.C	T04HTR	602.4 DEG.C
TWODPR	21.2 DEG.C					TGCOMP	61.1 DEG.C	T05HTR	624.1 DEG.C
						TGBOUN	35.8 DEG.C	T06HTR	616.0 DEG.C
								T07HTR	614.9 DEG.C
								T08HTR	576.4 DEG.C
								T09HTR	606.9 DEG.C
								T10HTR	586.4 DEG.C
								T11HTR	566.2 DEG.C
								T12HTR	577.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.23 L/MIN	1 PWRIN	4038. WATTS	VX1HOR	0.4 CM/S	PWROUT	955. WATTS	T13REG*****	DEG.C
TWINCL	24.9 DEG.C	2 QCOOLR	3032. WATTS	VY1VER	9.2 CM/S	INDPWR	1030. WATTS	T14REG	506.8 DEG.C
TDLCL	10.32 DEG.C	3 QDSHPT	732. WATTS			PISTST	2.80 CM	T15REG	359.0 DEG.C
TWOCLR	33.65 DEG.C	4 EXTEFF	23.7 %	PHASE ANGLES		DISPST	2.45 CM	T16REG	339.8 DEG.C
		5 TAVHTR	600.0 DEG.C	PADISP	57.5 LEG.			T17REG	317.8 DEG.C
		6 INTEFF	24.0 %	PAPRES	-15.2 DEG.			T18REG	326.8 DEG.C
		8 AMPS	1355. AMPS					T19REG	216.2 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED				T03HED*****	DEG.C
		10 QDISP	14. WATTS	FREQ	30.1 HZ				
		11 QREG1	90. WATTS						
		12 QREG2	146. WATTS						
		13 QREG3	35. WATTS						
		14 PFP	1010.0 KPA/CM						
		15 PFD	317.9 KPA/CM						
		16 NBEALE	0.00631						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:03:22.09 RDG 1012

FLUID HELIUM BAROM 14.303 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.93 L/MIN	AMPS1	742. AMPS	PRESUP	7066. KPA	TGEXP	552.5 DEG.C	T01HTR	598.1 DEG.C
		AMPS2	683. AMPS					T02HTR	613.6 DEG.C
TWINDP	17.9 DEG.C	VOLTG	3.13 VOLTS	MEANBP	7000. KPA	TGREGH	540.3 DEG.C	T03HTR	617.4 DEG.C
TDLDP	2.89 DEG.C			MEANCP	7049. KPA	TGREGC	92.6 DEG.C	T04HTR	601.2 DEG.C
TWODPR	21.4 DEG.C					TGCOMP	65.4 DEG.C	T05HTR	625.1 DEG.C
						TGBOUN	37.2 DEG.C	T06HTR	616.7 DEG.C
								T07HTR	615.6 DEG.C
								T08HTR	574.6 DEG.C
								T09HTR	606.0 DEG.C
								T10HTR	585.4 DEG.C
								T11HTR	563.2 DEG.C
								T12HTR	576.1 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.24 L/MIN	1 PWRIN	4467. WATTS	VX1HOR	0.4 CM/S	PWROUT	1026. WATTS	T13REG*****	DEG.C
TWINCL	24.9 DEG.C	2 QCOOLR	3265. WATTS	VY1VER	9.8 CM/S	INDPWR	1100. WATTS	T14REG	500.1 DEG.C
TDLCL	11.10 DEG.C	3 QDSHPT	791. WATTS			PISTST	3.00 CM	T15REG	346.2 DEG.C
TWOCLR	34.54 DEG.C	4 EXTEFF	23.0 %	PHASE ANGLES		DISPST	2.51 CM	T16REG	326.9 DEG.C
		5 TAVHTR	599.4 DEG.C	PADISP	57.6 DEG.			T17REG	303.8 DEG.C
		6 INTEFF	23.9 %	PAPRES	-14.0 DEG.			T18REG	313.6 DEG.C
		8 AMPS	1426. AMPS					T19REG	208.9 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED				T03HED*****	DEG.C
		10 QDISP	14. WATTS	FREQ	30.2 HZ				
		11 QREG1	85. WATTS						
		12 QREG2	151. WATTS						
		13 QREG3	35. WATTS						
		14 PFP	1007.4 KPA/CM						
		15 PFD	306.5 KPA/CM						
		16 NBEALE	0.00631						

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:10:31.09 RDG 1013

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.91 L/MIN	AMPS1	494. AMPS	PRESUP	7273. KPA	TGEXP	521.6 DEG.C	T01HTR	551.8 DEG.C
TWINDP	17.9 DEG.C	AMPS2	487. AMPS	MEANBP	7014. KPA	TGREGH	517.7 DEG.C	T02HTR	556.2 DEG.C
TDLDP	1.65 DEG.C	VOLTG	2.14 VOLTS	MEANCP	7048. KPA	TGREGC	88.2 DEG.C	T03HTR	562.5 DEG.C
TWODPR	20.2 DEG.C					TGCOMP	43.2 DEG.C	T04HTR	556.3 DEG.C
						TGBOUN	34.1 DEG.C	T05HTR	565.1 DEG.C
								T06HTR	557.6 DEG.C
								T07HTR	555.8 DEG.C
								T08HTR	535.2 DEG.C
								T09HTR	559.3 DEG.C
								T10HTR	543.0 DEG.C
								T11HTR	530.5 DEG.C
								T12HTR	532.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.23 L/MIN	1 PWRIN	2100. WATTS	VX1HOR	0.2 CM/S	PWROUT	479. WATTS	T13REG*****	DEG.C
TWINCL	24.8 DEG.C	2 QCOOLR	1978. WATTS	VY1VER	6.0 CM/S	INDPWR	514. WATTS	T14REG	477.8 DEG.C
TDLCL	6.73 DEG.C	3 QDSHPT	449. WATTS			PISTST	1.80 CM	T15REG	357.7 DEG.C
TWOCLR	29.88 DEG.C	4 EXTEFF	22.8 %	PHASE ANGLES		DISPST	1.81 CM	T16REG	359.1 DEG.C
		5 TAVHTR	550.5 DEG.C	PADISP	57.8 DEG.			T17REG	348.7 DEG.C
		6 INTEFF	19.5 %	PAPRES	-17.6 DEG.			T18REG	346.4 DEG.C
		8 AMPS	981. AMPS			DYNAMIC CALCULATIONS		T19REG	213.1 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	792. KPA	T03HED*****	DEG.C
		10 QDISP	14. WATTS	FREQ	30.2 HZ	DISPCP	2.16 CM		
		11 QREG1	108. WATTS			PISTCP	2.31 CM		
		12 QREG2	107. WATTS			PDYND*****	KPA		
		13 QREG3	33. WATTS			PDLCLR	3.05 KPA		
		14 PFP	1004.7 KPA/CM			PDLREG	37.33 KPA		
		15 PFD	312.6 KPA/CM			PDLDIS	68.56 KPA		
		16 NBEALE	0.00489						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:15:40.12 RDG 1014

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.94 L/MIN	AMPS1	537. AMPS	PRESUP	7176. KPA	TGEXP	518.2 DEG.C	T01HTR	553.9 DEG.C
		AMPS2	518. AMPS					T02HTR	555.2 DEG.C
TWINDP	17.9 DEG.C	VOLTG	2.30 VOLTS	MEANBP	6998. KPA	TGREGH	517.7 DEG.C	T03HTR	564.7 DEG.C
TDLDP	1.82 DEG.C			MEANCP	7034. KPA	TGREGC	89.4 DEG.C	T04HTR	556.1 DEG.C
TWODPR	20.4 DEG.C					TGCOMP	44.9 DEG.C	T05HTR	566.8 DEG.C
						TGBOUN	34.7 DEG.C	T06HTR	557.4 DEG.C
								T07HTR	555.6 DEG.C
								T08HTR	536.2 DEG.C
								T09HTR	560.4 DEG.C
								T10HTR	542.4 DEG.C
								T11HTR	530.3 DEG.C
								T12HTR	529.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWRIN	2423. WATTS	VX1HOR	0.3 CM/S	PWROUT	544. WATTS		
TWINCL	24.5 DEG.C	2 QCOOLR	2073. WATTS	VY1VER	6.6 CM/S	INDPWR	582. WATTS		
TDLCL	7.07 DEG.C	3 QDSHPT	498. WATTS			PISTST	1.99 CM		
TWOCLR	30.05 DEG.C	4 EXTEFF	22.4 %	PHASE ANGLES		DISPST	1.93 CM		
		5 TAVHTR	550.8 DEG.C	PADIS	58.1 DEG.			T13REG	***** DEG.C
		6 INTEFF	20.8 %	PAPRES	-16.7 DEG.			T14REG	476.5 DEG.C
		8 AMPS	1054. AMPS			DYNAMIC CALCULATIONS		T15REG	358.2 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	871. KPA	T16REG	359.3 DEG.C
		10 QDISP	13. WATTS	FREQ	30.1 HZ	DISPCP	2.33 CM	T17REG	351.5 DEG.C
		11 QREG1	109. WATTS			PISTCP	2.31 CM	T18REG	348.9 DEG.C
		12 QREG2	103. WATTS			PDYNDB	***** KPA	T19REG	215.1 DEG.C
		13 QREG3	33. WATTS			PDLCLR	3.56 KPA		
		14 PFP	993.7 KPA/CM			PDLREG	42.76 KPA		
		15 PFD	305.2 KPA/CM			PDLDIS	74.33 KPA	T03HED	***** DEG.C
		16 NBEALE	0.00505						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:20:22.12 RDG 1015

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.10 L/MIN	AMPS1	573. AMPS	PRESUP	7276. KPA	TGEXP	515.3 DEG.C	T01HTR	555.3 DEG.C
		AMPS2	552. AMPS					T02HTR	554.2 DEG.C
TWINDP	18.0 DEG.C	VOLTG	2.45 VOLTS	MEANBP	6994. KPA	TGREGH	516.9 DEG.C	T03HTR	566.5 DEG.C
TDLDP	1.87 DEG.C			MEANCP	7035. KPA	TGREGC	91.7 DEG.C	T04HTR	555.3 DEG.C
TWODFR	20.5 DEG.C					TGCOMP	46.8 DEG.C	T05HTR	568.3 DEG.C
						TGBOUN	34.8 DEG.C	T06HTR	557.5 DEG.C
								T07HTR	555.6 DEG.C
								T08HTR	536.4 DEG.C
								T09HTR	561.0 DEG.C
								T10HTR	541.8 DEG.C
								T11HTR	529.2 DEG.C
								T12HTR	527.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	2761. WATTS	VX1HOR	0.3 CM/S	PWROUT	627. WATTS	T13REG*****	DEG.C
TWINCL	24.4 DEG.C	2 QCOOLR	2285. WATTS	VY1VER	7.2 CM/S	INDPWR	673. WATTS	T14REG	475.3 DEG.C
TDLCL	7.82 DEG.C	3 QDSHPT	534. WATTS			PISTST	2.20 CM	T15REG	357.1 DEG.C
TWOCLR	30.61 DEG.C	4 EXTEFF	22.7 %	PHASE ANGLES		DISPST	2.06 CM	T16REG	358.8 DEG.C
		5 TAVHTR	550.7 DEG.C	PADISP	58.1 DEG.			T17REG	352.5 DEG.C
		6 INTEFF	21.5 %	PAPRES	-15.9 DEG.	DYNAMIC CALCULATIONS		T18REG	349.0 DEG.C
		8 AMPS	1125. AMPS			PAMPC	974. KPA	T19REG	215.6 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		DISPCP	2.26 CM		
		10 QDISP	13. WATTS	FREQ	30.1 HZ	PISTCP	2.31 CM		
		11 QREG1	108. WATTS			PDYNDB*****	KPA		
		12 QREG2	102. WATTS			PDLCLR	4.41 KPA		
		13 QREG3	33. WATTS			PDLREG	47.86 KPA		
		14 PFP	1003.4 KPA/CM			PDLDIS	82.13 KPA	T03HED*****	DEG.C
		15 PFD	304.9 KPA/CM						
		16 NBEALE	0.00527						

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:25:22.12 RDG 1016

FLUID HELIUM		BAROM 14.298 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.11 L/MIN	AMPS1	612. AMPS	PRESUP	7122. KPA	TGEXP	511.8 DEG.C	T01HTR	551.1 DEG.C
		AMPS2	577. AMPS					T02HTR	555.6 DEG.C
TWINDP	18.0 DEG.C	VOLTG	2.59 VOLTS	MEANBP	6992. KPA	TGREGH	510.2 DEG.C	T03HTR	564.6 DEG.C
TDLDP	2.00 DEG.C			MEANCP	7037. KPA	TGREGC	92.2 DEG.C	T04HTR	552.4 DEG.C
TWODPR	20.7 DEG.C					TGCOMP	49.6 DEG.C	T05HTR	568.3 DEG.C
						TGBOUN	35.4 DEG.C	T06HTR	558.7 DEG.C
								T07HTR	557.3 DEG.C
								T08HTR	531.3 DEG.C
								T09HTR	557.6 DEG.C
								T10HTR	538.2 DEG.C
								T11HTR	523.3 DEG.C
								T12HTR	526.2 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	3086. WATTS	VX1HOR	0.4 CM/S	PWROUT	706. WATTS		
TWINCL	24.4 DEG.C	2 QCOOLR	2470. WATTS	VY1VER	7.9 CM/S	INDPWR	764. WATTS		
TDLCL	8.46 DEG.C	3 QDSHPT	572. WATTS			PISTST	2.41 CM		
TWOCLR	31.32 DEG.C	4 EXTEFF	22.9 %	PHASE ANGLES		DISPST	2.20 CM		
		5 TAVHTR	548.7 DEG.C	PADISP	58.4 DEG.			T13REG*****	DEG.C
		6 INTEFF	22.2 %	PAPRES	-14.7 DEG.			T14REG	473.9 DEG.C
		8 AMPS	1189. AMPS			DYNAMIC CALCULATIONS		T15REG	355.5 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	1068. KPA	T16REG	351.6 DEG.C
		10 QDISP	13. WATTS	FREQ	30.1 HZ	DISPCP	2.40 CM	T17REG	340.8 DEG.C
		11 QREG1	100. WATTS			PISTCP	2.38 CM	T18REG	339.3 DEG.C
		12 QREG2	109. WATTS			PDYNDB*****	KPA	T19REG	216.2 DEG.C
		13 QREG3	32. WATTS			PDLCLR	5.60 KPA		
		14 PFP	996.6 KPA/CM			PDLREG	52.95 KPA		
		15 PFD	289.2 KPA/CM			PDLDIS	91.98 KPA	T03HED*****	DEG.C
		16 NBEALE	0.00542						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:30:25.12 RDG 1017

FLUID HELIUM		BAROM 14.298 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.08 L/MIN	AMPS1	645. AMPS	PRESUP	7187. KPA	TGEXP	510.7 DEG.C	T01HTR	550.9 DEG.C
		AMPS2	608. AMPS					T02HTR	558.5 DEG.C
TWINDP	18.0 DEG.C	VOLTG	2.73 VOLTS	MEANBP	7011. KPA	TGREGH	507.3 DEG.C	T03HTR	565.8 DEG.C
TDLDP	2.30 DEG.C			MEANCP	7062. KPA	TGREGC	92.3 DEG.C	T04HTR	553.1 DEG.C
TWODPR	21.0 DEG.C					TGCOMP	53.4 DEG.C	T05HTR	570.6 DEG.C
						TGBOUN	36.2 DEG.C	T06HTR	561.8 DEG.C
								T07HTR	560.5 DEG.C
								T08HTR	530.1 DEG.C
								T09HTR	558.1 DEG.C
								T10HTR	537.8 DEG.C
								T11HTR	521.5 DEG.C
								T12HTR	526.8 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTG CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWRIN	3424. WATTS	VX1HOR	0.4 CM/S	PWROUT	776. WATTS		
TWINCL	24.5 DEG.C	2 QCOOLR	2713. WATTS	VY1VER	0.5 CM/S	INDPWR	846. WATTS		
TDLCL	9.27 DEG.C	3 QDSHPT	651. WATTS			PISTST	2.60 CM		
TWOCLR	32.09 DEG.C	4 EXTEFF	22.7 %	PHASE ANGLES		DISPST	2.30 CM		
		5 TAVTR	549.6 DEG.C	PADISP	58.8 DEG.			T13REG*****	DEG.C
		6 INTEFF	22.2 %	PAPRES	-14.1 DEG.			T14REG	473.2 DEG.C
		8 AMPS	1253. AMPS			DYNAMIC CALCULATIONS		T15REG	352.3 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMFC	1166. KPA	T16REG	343.8 DEG.C
		10 QDISP	13. WATTS	FREQ	30.2 HZ	DISPCP	2.35 CM	T17REG	330.4 DEG.C
		11 QREG1	93. WATTS			PISTCP	2.35 CM	T18REG	330.0 DEG.C
		12 QREG2	116. WATTS			PBYNDB*****	KPA	T19REG	215.2 DEG.C
		13 QREG3	32. WATTS			PDLCLR	6.79 KPA		
		14 PFP	1003.3 KPA/CM			PDLREG	57.70 KPA	T03HED*****	DEG.C
		15 PFD	289.5 KPA/CM			PDLDIS	98.77 KPA		
		16 NBEALE	0.00549						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:35:25.12 RDG 1018

FLUID HELIUM		BAROM 14.298 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.06 L/MIN	AMPS1	695. AMPS	PRESUP	7360. KPA	TGEXP	507.8 DEG.C	T01HTR	548.6 DEG.C
		AMPS2	646. AMPS					T02HTR	561.7 DEG.C
TWINDP	18.1 DEG.C	VOLTG	2.93 VOLTS	MEANBP	7004. KPA	TGREGH	498.0 DEG.C	T03HTR	566.1 DEG.C
TDLDP	2.50 DEG.C			MEANCP	7057. KPA	TGREGC	88.8 DEG.C	T04HTR	551.9 DEG.C
TWODPR	21.3 DEG.C					TGCOMP	62.1 DEG.C	T05HTR	573.0 DEG.C
						TGBOUN	37.6 DEG.C	T06HTR	564.2 DEG.C
								T07HTR	563.3 DEG.C
								T08HTR	526.9 DEG.C
								T09HTR	555.9 DEG.C
								T10HTR	536.9 DEG.C
								T11HTR	516.2 DEG.C
								T12HTR	527.6 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	3924. WATTS	VX1HOR	0.4 CM/S	PWROUT	849. WATTS	T13REG*****	DEG.C
TWINCL	24.7 DEG.C	2 QCOOLR	3092. WATTS	VY1VER	9.2 CM/S	INDPWR	917. WATTS	T14REG	462.5 DEG.C
TDLCL	10.57 DEG.C	3 QDSHPT	707. WATTS			PISTST	2.81 CM	T15REG	327.1 DEG.C
TWOCLR	33.52 DEG.C	4 EXTEFF	21.6 %	PHASE ANGLES		DISPST	2.41 CM	T16REG	310.0 DEG.C
		5 TAVHTR	549.4 DEG.C	PADISP	58.8 DEG.			T17REG	291.2 DEG.C
		6 INTEFF	21.5 %	PAPRES	-13.4 DEG.			T18REG	299.0 DEG.C
		8 AMPS	1341. AMPS			DYNAMIC CALCULATIONS		T19REG	199.2 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMFC	1255. KPA	T03HED*****	DEG.C
		10 QDISP	13. WATTS	FREQ	30.1 HZ	DISPCP	2.37 CM		
		11 QREG1	81. WATTS			PISTCP	2.27 CM		
		12 QREG2	133. WATTS			PDYND*****	KPA		
		13 QREG3	32. WATTS			PDLCLR	9.33 KPA		
		14 PFP	996.2 KPA/CM			PDLREG	61.43 KPA		
		15 PFD	282.7 KPA/CM			PDLDIS	105.89 KPA		
		16 NBEALE	0.00559						

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:40:04.12 RDG 1019

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.05 L/MIN	AMPS1	736. AMPS	PRESUP	7090. KPA	TGEXP	503.9 DEG.C	T01HTR	547.6 DEG.C
TWINDP	18.1 DEG.C	AMPS2	683. AMPS	MEANBP	6994. KPA	TGREGH	489.9 DEG.C	T02HTR	544.3 DEG.C
TDLDP	2.77 DEG.C	VOLTG	3.10 VOLTS	MEANCP	7048. KPA	TGREGC	89.2 DEG.C	T03HTR	566.9 DEG.C
TWODPR	21.5 DEG.C					TGCOMP	68.3 DEG.C	T04HTR	551.2 DEG.C
						TGBOUN	38.9 DEG.C	T05HTR	575.0 DEG.C
								T06HTR	566.1 DEG.C
								T07HTR	565.2 DEG.C
								T08HTR	525.0 DEG.C
								T09HTR	554.6 DEG.C
								T10HTR	536.6 DEG.C
								T11HTR	512.7 DEG.C
								T12HTR	528.0 DEG.C
								T13REG*****	DEG.C
								T14REG	451.1 DEG.C
								T15REG	305.2 DEG.C
								T16REG	286.6 DEG.C
								T17REG	265.2 DEG.C
								T18REG	275.5 DEG.C
								T19REG	186.9 DEG.C
								T03HED*****	DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:37:22.48 RDG 1020

FLUID HELIUM		BAROM 14.327 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.78 L/MIN	AMPS1	522. AMPS	PRESUP	7382. KPA	TGEXP	472.1 DEG.C	T01HTR	501.1 DEG.C
TWINDP	16.6 DEG.C	AMPS2	485. AMPS	MEANBP	7010. KPA	TGREGH	468.4 DEG.C	T02HTR	506.4 DEG.C
TDLDP	0.82 DEG.C	VOLTG	2.16 VOLTS	MEANCP	7045. KPA	TGREGC	83.7 DEG.C	T03HTR	512.4 DEG.C
TWODPR	18.0 DEG.C					TGCOMP	40.0 DEG.C	T04HTR	506.2 DEG.C
						TGBOUN	27.2 DEG.C	T05HTR	514.6 DEG.C
								T06HTR	506.6 DEG.C
								T07HTR	505.1 DEG.C
								T08HTR	434.9 DEG.C
								T09HTR	507.6 DEG.C
								T10HTR	493.2 DEG.C
								T11HTR	478.8 DEG.C
								T12HTR	482.1 DEG.C
								T13REG*****	DEG.C
								T14REG	433.2 DEG.C
								T15REG	326.1 DEG.C
								T16REG	325.7 DEG.C
								T17REG	317.1 DEG.C
								T18REG	315.9 DEG.C
								T19REG	196.0 DEG.C
								T03HED*****	DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:42:25.54 RDG 1021

FLUID HELIUM		BAROM 14.327 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.53 L/MIN	AMPS1	549. AMPS	PRESUP	7585. KPA	TGEXP	470.0 DEG.C	T01HTR	503.1 DEG.C
TWINDP	16.4 DEG.C	AMPS2	515. AMPS	MEANBP	6991. KPA	TGREGH	468.6 DEG.C	T02HTR	506.0 DEG.C
TDLDP	1.07 DEG.C	VOLTG	2.28 VOLTS	MEANCP	7030. KPA	TGREGC	85.3 DEG.C	T03HTR	514.3 DEG.C
TWODPR	18.1 DEG.C					TGCOMP	42.2 DEG.C	T04HTR	506.0 DEG.C
						TGBOUN	28.3 DEG.C	T05HTR	516.5 DEG.C
								T06HTR	507.3 DEG.C
								T07HTR	506.0 DEG.C
								T08HTR	485.9 DEG.C
								T09HTR	509.0 DEG.C
								T10HTR	492.7 DEG.C
								T11HTR	478.9 DEG.C
								T12HTR	480.6 DEG.C
								T13REG*****	DEG.C
								T14REG	433.3 DEG.C
								T15REG	327.0 DEG.C
								T16REG	325.6 DEG.C
								T17REG	318.4 DEG.C
								T18REG	317.7 DEG.C
								T19REG	198.6 DEG.C
								T03HED*****	DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:47:31.54 RDG 1022

FLUID HELIUM BAROM 14.327 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.51 L/MIN	POWER IN AMPS1 581. AMPS AMPS2 547. AMPS VOLTG 2.42 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7245. KPA MEANBP 7015. KPA MEANCP 7055. KPA	GAS TEMPERATURES TGEXP 467.9 DEG.C TGREGH 468.2 DEG.C TGREGC 87.5 DEG.C TGCOMP 45.5 DEG.C TGBOUN 29.1 DEG.C	SURFACE TEMPERATURES T01HTR 504.6 DEG.C T02HTR 505.8 DEG.C T03HTR 516.3 DEG.C T04HTR 505.6 DEG.C T05HTR 518.7 DEG.C T06HTR 508.3 DEG.C T07HTR 506.9 DEG.C T08HTR 486.3 DEG.C T09HTR 510.1 DEG.C T10HTR 492.5 DEG.C T11HTR 478.4 DEG.C T12HTR 479.5 DEG.C
TWINDP 16.3 DEG.C TDLDP 1.27 DEG.C TWODPR 18.2 DEG.C				

HEAT TO COOLER FLOCLR 4.20 L/MIN TWINCL 24.8 DEG.C TDLCL ***** DEG.C TWOCLR 30.71 DEG.C	CALCULATED PARAMETERS 1 PWRIN 2730. WATTS 2 QCOOLR 2216. WATTS 3 QDSHPT 398. WATTS 4 EXTEFF 21.3 % 5 TAVHTR 501.1 DEG.C 6 INTEFF 20.8 % 8 AMPS 1128. AMPS 9 QDISPG 2. WATTS 10 QDISP 12. WATTS 11 QREG1 95. WATTS 12 QREG2 93. WATTS 13 QREG3 29. WATTS 14 PFP 1007.7 KPA/CM 15 PFD 278.8 KPA/CM 16 NBEALE 0.00488	VIBRATION VX1HOR 0.4 CM/S VY1VER 7.3 CM/S PHASE ANGLES PADISP 59.8 DEG. PAPRES -14.5 DEG. ENGINE SPEED FREQ 30.2 HZ	REMOTE CALCULATIONS PWROUT 583. WATTS INDPWR 630. WATTS PISTST 2.20 CM DISPST 2.07 CM DYNAMIC CALCULATIONS PAMPC 994. KPA DISPCP 2.24 CM PISTCP 2.29 CM PDYNDB***** KPA PDLCLR 4.58 KPA PDLREG 47.86 KPA PDLDIS 83.15 KPA	T13REG***** DEG.C T14REG 433.8 DEG.C T15REG 329.0 DEG.C T16REG 326.0 DEG.C T17REG 319.6 DEG.C T18REG 319.3 DEG.C T19REG 201.8 DEG.C T03HED***** DEG.C
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NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:52:22.54 RDG 1023

FLUID HELIUM BAROM 14.327 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.49 L/MIN	POWER IN AMPS1 618. AMPS AMPS2 576. AMPS VOLTG 2.56 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7584. KPA MEANBP 7014. KPA MEANCP 7061. KPA	GAS TEMPERATURES TGEXP 465.3 DEG.C TGREGH 466.0 DEG.C TGREGC 90.2 DEG.C TGCOMP 50.5 DEG.C TGBOUN 30.5 DEG.C	SURFACE TEMPERATURES T01HTR 504.1 DEG.C T02HTR 505.9 DEG.C T03HTR 516.7 DEG.C T04HTR 504.4 DEG.C T05HTR 519.7 DEG.C T06HTR 509.5 DEG.C T07HTR 508.0 DEG.C T08HTR 484.7 DEG.C T09HTR 509.9 DEG.C T10HTR 490.7 DEG.C T11HTR 476.3 DEG.C T12HTR 477.6 DEG.C
TWINDP 16.2 DEG.C TDLDP 1.53 DEG.C TWODPR 18.3 DEG.C				

HEAT TO COOLER FLOCLR 4.20 L/MIN TWINCL 24.8 DEG.C TDLCL ***** DEG.C TWOCLR 31.45 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3062. WATTS 2 QCOOLR 2456. WATTS 3 QDSHPT 478. WATTS 4 EXTEFF 21.5 % 5 TAVHTR 500.6 DEG.C 6 INTEFF 21.1 % 8 AMPS 1194. AMPS 9 QDISPG 2. WATTS 10 QDISP 12. WATTS 11 QREG1 93. WATTS 12 QREG2 93. WATTS 13 QREG3 29. WATTS 14 PFP 1005.0 KPA/CM 15 PFD 268.6 KPA/CM 16 NBEALE 0.00502	VIBRATION VX1HOR 0.4 CM/S VY1VER 8.0 CM/S PHASE ANGLES PADISP 60.1 DEG. PAPRES -13.6 DEG. ENGINE SPEED FREQ 30.2 HZ	REMOTE CALCULATIONS PWROUT 658. WATTS INDPWR 712. WATTS PISTST 2.41 CM DISPST 2.21 CM DYNAMIC CALCULATIONS PAMPC 1093. KPA DISPCP 2.27 CM PISTCP 2.27 CM PDYNDB***** KPA PDLCLR 5.60 KPA PDLREG 54.30 KPA PDLDIS 93.00 KPA	T13REG***** DEG.C T14REG 433.5 DEG.C T15REG 330.0 DEG.C T16REG 325.2 DEG.C T17REG 318.4 DEG.C T18REG 318.7 DEG.C T19REG 204.5 DEG.C T03HED***** DEG.C
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NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:57:22.54 RDG 1024

FLUID HELIUM BAROM 14.327 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.47 L/MIN	POWER IN AMPS1 646. AMPS AMPS2 602. AMPS VOLTG 2.68 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7572. KPA MEANBP 7000. KPA MEANCP 7048. KPA	GAS TEMPERATURES TGEXP 462.4 DEG.C TGREGH 460.8 DEG.C TGREGC 90.5 DEG.C TGCOMP 53.8 DEG.C TGBOUN 31.8 DEG.C	SURFACE TEMPERATURES T01HTR 501.0 DEG.C T02HTR 507.9 DEG.C T03HTR 515.5 DEG.C T04HTR 503.0 DEG.C T05HTR 519.9 DEG.C T06HTR 511.0 DEG.C T07HTR 510.1 DEG.C T08HTR 481.0 DEG.C T09HTR 507.7 DEG.C T10HTR 488.1 DEG.C T11HTR 471.9 DEG.C T12HTR 476.6 DEG.C
TWINDP 16.2 DEG.C TDLDP 1.71 DEG.C TWODPR 18.6 DEG.C				

HEAT TO COOLER FLOCLR 4.21 L/MIN TWINCL 24.7 DEG.C TDLCL ***** DEG.C TWOCLR 32.15 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3348. WATTS 2 QCOOLR 2681. WATTS 3 QDSHPT 531. WATTS 4 EXTEFF 21.1 % 5 TAVHTR 499.5 DEG.C 6 INTEFF 20.9 % 8 AMPS 1248. AMPS 9 QDISPG 2. WATTS 10 QDISP 12. WATTS 11 QREG1 85. WATTS 12 QREG2 100. WATTS 13 QREG3 29. WATTS 14 PFP 995.1 KPA/CM 15 PFD 254.7 KPA/CM 16 NBEALE 0.00501	VIBRATION VX1HOR 0.4 CM/S VY1VER 8.6 CM/S PHASE ANGLES PADISP 60.4 DEG. PAPRES -12.5 DEG. ENGINE SPEED FREQ 30.2 HZ	REMOTE CALCULATIONS PWROUT 707. WATTS INDPWR 765. WATTS PISTST 2.60 CM DISPST 2.30 CM DYNAMIC CALCULATIONS PAMPC 1176. KPA DISPCP 2.32 CM PISTCP 2.28 CM PDYNDB***** KPA PDLCLR 6.96 KPA PDLREG 58.04 KPA PDLDIS 99.10 KPA	T13REG***** DEG.C T14REG 431.6 DEG.C T15REG 326.4 DEG.C T16REG 317.8 DEG.C T17REG 307.7 DEG.C T18REG 308.1 DEG.C T19REG 203.0 DEG.C T03HED***** DEG.C
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APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:02:49.54 RDG 1025

FLUID HELIUM		BAROM 14.327 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE		GAS TEMPERATURES	SURFACE TEMPERATURES	
FLODP	4.46 L/MIN	AMPS1 686. AMPS	PRESUP 756. KPA		TGEXP 460.4 DEG.C	T01HTR 498.6 DEG.C	
		AMPS2 636. AMPS				T02HTR 510.9 DEG.C	
TWINDP	16.1 DEG.C	VOLTG 2.85 VOLTS	MEANBP 6987. KPA		TGREGH 453.1 DEG.C	T03HTR 515.5 DEG.C	
TDLDP	2.01 DEG.C		MEANCP 7037. KPA		TGREGC 87.7 DEG.C	T04HTR 501.4 DEG.C	
TWODPR	18.7 DEG.C				TGCOMP 59.7 DEG.C	T05HTR 522.1 DEG.C	
					TGBOUN 33.4 DEG.C	T06HTR 513.0 DEG.C	
						T07HTR 512.4 DEG.C	
						T08HTR 477.6 DEG.C	
						T09HTR 505.3 DEG.C	
						T10HTR 487.3 DEG.C	
						T11HTR 466.9 DEG.C	
						T12HTR 478.1 DEG.C	
						T13REG***** DEG.C	
						T14REG 423.8 DEG.C	
						T15REG 360.7 DEG.C	
						T16REG 290.7 DEG.C	
						T17REG 277.3 DEG.C	
						T18REG 282.5 DEG.C	
						T19REG 189.9 DEG.C	
						T03HED***** DEG.C	
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION		REMOTE CALCULATIONS		
FLOCLR 4.20 L/MIN		1 PWRIN 3764. WATTS	VX1HOR 0.4 CM/S		PWROUT 767. WATTS		
TWINCL 24.7 DEG.C		2 QCOOLR 2994. WATTS	VY1VER 9.2 CM/S		INDPWR 827. WATTS		
TDLCL ***** DEG.C		3 QDSHPT 623. WATTS			PISTST 2.80 CM		
TWOCLR 33.24 DEG.C		4 EXTEFF 20.4 %	PHASE ANGLES		DISPST 2.41 CM		
		5 TAVHTR 499.1 DEG.C	PADISP 60.8 DEG.				
		6 INTEFF 20.4 %	PAPRES -11.8 DEG.				
		8 AMPS 1322. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		
		9 QDISPG 2. WATTS	FREQ 30.2 HZ		PAMPC 1265. KPA		
		10 QDISP 11. WATTS			DISPCP 2.36 CM		
		11 QREG1 75. WATTS			PISTCP 2.26 CM		
		12 QREG2 115. WATTS			PDYNDB***** KPA		
		13 QREG3 28. WATTS			PDLCLR 9.33 KPA		
		14 PFP 987.7 KPA/CM			PDLREG 61.77 KPA		
		15 PFD 245.8 KPA/CM			PDLDIS106.23 KPA		
		16 NBEALE 0.00506					

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:13:22.54 RDG 1026

FLUID HELIUM		BAROM 14.332 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE		GAS TEMPERATURES	SURFACE TEMPERATURES	
FLODP	4.24 L/MIN	AMPS1 482. AMPS	PRESUP 7565. KPA		TGEXP 425.8 DEG.C	T01HTR 451.6 DEG.C	
		AMPS2 481. AMPS				T02HTR 457.0 DEG.C	
TWINDP	16.0 DEG.C	VOLTG 2.06 VOLTS	MEANBP 7011. KPA		TGREGH 421.4 DEG.C	T03HTR 462.9 DEG.C	
TDLDP	1.29 DEG.C		MEANCP 7050. KPA		TGREGC 79.6 DEG.C	T04HTR 456.2 DEG.C	
TWODPR	18.0 DEG.C				TGCOMP 41.8 DEG.C	T05HTR 465.8 DEG.C	
					TGBOUN 31.8 DEG.C	T06HTR 457.1 DEG.C	
						T07HTR 455.9 DEG.C	
						T08HTR 436.2 DEG.C	
						T09HTR 457.9 DEG.C	
						T10HTR 444.7 DEG.C	
						T11HTR 430.2 DEG.C	
						T12HTR 434.7 DEG.C	
						T13REG***** DEG.C	
						T14REG 391.1 DEG.C	
						T15REG 295.0 DEG.C	
						T16REG 294.0 DEG.C	
						T17REG 286.6 DEG.C	
						T18REG 286.7 DEG.C	
						T19REG 179.3 DEG.C	
						T03HED***** DEG.C	
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION		REMOTE CALCULATIONS		
FLOCLR 4.20 L/MIN		1 PWRIN 1983. WATTS	VX1HOR 0.3 CM/S		PWROUT 396. WATTS		
TWINCL 24.4 DEG.C		2 QCOOLR 2012. WATTS	VY1VER 6.0 CM/S		INDPWR 426. WATTS		
TDLCL 6.90 DEG.C		3 QDSHPT 380. WATTS			PISTST 1.82 CM		
TWOCLR 29.24 DEG.C		4 EXTEFF 20.0 %	PHASE ANGLES		DISPST 1.80 CM		
		5 TAVHTR 450.9 DEG.C	PADISP 61.3 DEG.				
		6 INTEFF 16.5 %	PAPRES -14.4 DEG.				
		8 AMPS 964. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		
		9 QDISPG 2. WATTS	FREQ 30.1 HZ		PAMPC 807. KPA		
		10 QDISP 11. WATTS			DISPCP 2.19 CM		
		11 QREG1 87. WATTS			PISTCP 2.29 CM		
		12 QREG2 85. WATTS			PDYNDB***** KPA		
		13 QREG3 26. WATTS			PDLCLR 3.56 KPA		
		14 PFP 981.6 KPA/CM			PDLREG 39.71 KPA		
		15 PFD 255.1 KPA/CM			PDLDIS 70.93 KPA		
		16 NBEALE 0.00403					

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:18:28.54 RDG 1027

FLUID HELIUM		BAROM 14.332 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE		GAS TEMPERATURES	SURFACE TEMPERATURES	
FLODP	4.16 L/MIN	AMPS1 516. AMPS	PRESUP 7195. KPA		TGEXP 423.6 DEG.C	T01HTR 452.5 DEG.C	
		AMPS2 507. AMPS				T02HTR 456.6 DEG.C	
TWINDP	16.1 DEG.C	VOLTG 2.18 VOLTS	MEANBP 7001. KPA		TGREGH 420.7 DEG.C	T03HTR 464.1 DEG.C	
TDLDP	1.34 DEG.C		MEANCP 7045. KPA		TGREGC 80.6 DEG.C	T04HTR 455.8 DEG.C	
TWODPR	18.1 DEG.C				TGCOMP 43.0 DEG.C	T05HTR 467.0 DEG.C	
					TGBOUN 31.8 DEG.C	T06HTR 457.5 DEG.C	
						T07HTR 456.3 DEG.C	
						T08HTR 436.2 DEG.C	
						T09HTR 458.4 DEG.C	
						T10HTR 444.0 DEG.C	
						T11HTR 429.2 DEG.C	
						T12HTR 433.1 DEG.C	
						T13REG***** DEG.C	
						T14REG 390.5 DEG.C	
						T15REG 295.4 DEG.C	
						T16REG 293.0 DEG.C	
						T17REG 286.8 DEG.C	
						T18REG 287.6 DEG.C	
						T19REG 181.0 DEG.C	
						T03HED***** DEG.C	
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION		REMOTE CALCULATIONS		
FLOCLR 4.12 L/MIN		1 PWRIN 2233. WATTS	VX1HOR 0.3 CM/S		PWROUT 444. WATTS		
TWINCL 24.2 DEG.C		2 QCOOLR 2064. WATTS	VY1VER 6.6 CM/S		INDPWR 474. WATTS		
TDLCL 7.10 DEG.C		3 QDSHPT 387. WATTS			PISTST 2.00 CM		
TWOCLR 29.47 DEG.C		4 EXTEFF 19.9 %	PHASE ANGLES		DISPST 1.91 CM		
		5 TAVHTR 450.9 DEG.C	PADISP 60.8 DEG.				
		6 INTEFF 17.7 %	PAPRES -13.1 DEG.				
		8 AMPS 1022. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		
		9 QDISPG 2. WATTS	FREQ 30.1 HZ		PAMPC 891. KPA		
		10 QDISP 11. WATTS			DISPCP 2.20 CM		
		11 QREG1 86. WATTS			PISTCP 2.26 CM		
		12 QREG2 83. WATTS			PDYNDB***** KPA		
		13 QREG3 26. WATTS			PDLCLR 4.07 KPA		
		14 PFP 979.5 KPA/CM			PDLREG 45.48 KPA		
		15 PFD 242.6 KPA/CM			PDLDIS 75.35 KPA		
		16 NBEALE 0.00410					

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:23:22.54 RDG 1028

FLUID HELIUM		BAROM 14.332 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.11 L/MIN	AMPS1	554. AMPS	PRESUP	7567. KPA	TGEXP	421.0 DEG.C
		AMPS2	537. AMPS				
TWINDP	16.0 DEG.C	VOLTG	2.33 VOLTS	MEANBP	6994. KPA	TGREGH	419.4 DEG.C
TDLDP	1.49 DEG.C			MEANCP	7040. KPA	TGREGC	83.4 DEG.C
TWODPR	18.2 DEG.C					TGCOMP	45.7 DEG.C
						TGBOUN	31.6 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.24 L/MIN	1 PWRIN	2544. WATTS	VX1HOR	0.3 CM/S	PWROUT	503. WATTS
TWINCL	25.4 DEG.C	2 QCOOLR	2129. WATTS	VY1VER	7.3 CM/S	INDPWR	536. WATTS
TDLCL	7.24 DEG.C	3 QDSHPT	428. WATTS			PISTST	2.20 CM
TWOCLR	31.11 DEG.C	4 EXTEFF	19.8 %	PHASE ANGLES		DISPST	2.04 CM
		5 TAVHTR	450.8 DEG.C	PADISP	61.5 DEG.		
		6 INTEFF	19.1 %	PAPRES	-12.2 DEG.		
		8 AMPS	1091. AMPS			DYNAMIC CALCULATIONS	
		9 QDISPG	2. WATTS	ENGINE SPEED		PAMPC	994. KPA
		10 QDISP	11. WATTS	FREQ	30.1 HZ	DISPCP	2.21 CM
		11 QREG1	84. WATTS			PISTCP	2.26 CM
		12 QREG2	83. WATTS			PDYNDB*****	KPA
		13 QREG3	26. WATTS			PDLCLR	4.75 KPA
		14 PFP	986.2 KPA/CM			PDLREG	47.52 KPA
		15 PFD	234.4 KPA/CM			PDLDIS	83.83 KPA
		16 NBEALE	0.00423				
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:29:31.54 RDG 1029

FLUID HELIUM		BAROM 14.332 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.05 L/MIN	AMPS1	594. AMPS	PRESUP	7241. KPA	TGEXP	418.4 DEG.C
		AMPS2	565. AMPS				
TWINDP	16.0 DEG.C	VOLTG	2.48 VOLTS	MEANBP	7010. KPA	TGREGH	418.1 DEG.C
TDLDP	1.78 DEG.C			MEANCP	7019. KPA	TGREGC	85.9 DEG.C
TWODPR	18.4 DEG.C					TGCOMP	48.6 DEG.C
						TGBOUN	32.5 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.22 L/MIN	1 PWRIN	2871. WATTS	VX1HOR	0.4 CM/S	PWROUT	566. WATTS
TWINCL	24.5 DEG.C	2 QCOOLR	2392. WATTS	VY1VER	8.0 CM/S	INDPWR	600. WATTS
TDLCL	8.17 DEG.C	3 QDSHPT	501. WATTS			PISTST	2.41 CM
TWOCLR	31.28 DEG.C	4 EXTEFF	19.7 %	PHASE ANGLES		DISPST	2.17 CM
		5 TAVHTR	450.7 DEG.C	PADISP	61.8 DEG.		
		6 INTEFF	19.1 %	PAPRES	-11.3 DEG.		
		8 AMPS	1159. AMPS			DYNAMIC CALCULATIONS	
		9 QDISPG	2. WATTS	ENGINE SPEED		PAMPC	1083. KPA
		10 QDISP	11. WATTS	FREQ	30.2 HZ	DISPCP	2.24 CM
		11 QREG1	83. WATTS			PISTCP	2.27 CM
		12 QREG2	82. WATTS			PDYNDB*****	KPA
		13 QREG3	26. WATTS			PDLCLR	5.60 KPA
		14 PFP	975.0 KPA/CM			PDLREG	53.96 KPA
		15 PFD	223.1 KPA/CM			PDLDIS	94.35 KPA
		16 NBEALE	0.00432				
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:35:10.54 RDG 1030

FLUID HELIUM		BAROM 14.332 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.11 L/MIN	AMPS1	626. AMPS	PRESUP	7562. KPA	TGEXP	415.3 DEG.C
		AMPS2	589. AMPS				
TWINDP	16.0 DEG.C	VOLTG	2.60 VOLTS	MEANBP	7007. KPA	TGREGH	415.9 DEG.C
TDLDP	1.90 DEG.C			MEANCP	7061. KPA	TGREGC	88.9 DEG.C
TWODPR	18.6 DEG.C					TGCOMP	54.6 DEG.C
						TGBOUN	33.1 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.21 L/MIN	1 PWRIN	3159. WATTS	VX1HOR	0.4 CM/S	PWROUT	605. WATTS
TWINCL	24.3 DEG.C	2 QCOOLR	2595. WATTS	VY1VER	8.6 CM/S	INDPWR	643. WATTS
TDLCL	8.88 DEG.C	3 QDSHPT	542. WATTS			PISTST	2.60 CM
TWOCLR	31.77 DEG.C	4 EXTEFF	19.2 %	PHASE ANGLES		DISPST	2.26 CM
		5 TAVHTR	449.8 DEG.C	PADISP	61.8 DEG.		
		6 INTEFF	18.9 %	PAPRES	-10.3 DEG.		
		8 AMPS	1215. AMPS			DYNAMIC CALCULATIONS	
		9 QDISPG	2. WATTS	ENGINE SPEED		PAMPC	1181. KPA
		10 QDISP	10. WATTS	FREQ	30.2 HZ	DISPCP	2.30 CM
		11 QREG1	79. WATTS			PISTCP	2.25 CM
		12 QREG2	83. WATTS			PDYNDB*****	KPA
		13 QREG3	25. WATTS			PDLCLR	6.79 KPA
		14 PFP	980.6 KPA/CM			PDLREG	58.38 KPA
		15 PFD	212.3 KPA/CM			PDLDIS	100.49 KPA
		16 NBEALE	0.00428				
							T03HED***** DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:40:31.54 RDG 1031

FLUID HELIUM		BAROM 14.337 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.11 L/MIN	AMPS1	664. AMPS	PRESUP	7552. KPA	TGEXP	413.6 DEG.C	T01HTR	449.7 DEG.C
		AMPS2	617. AMPS					T02HTR	459.5 DEG.C
TWINDP	16.1 DEG.C	VOLTG	2.74 VOLTS	MEANBP	6987. KPA	TGREGH	409.5 DEG.C	T03HTR	464.8 DEG.C
TDLLP	2.10 DEG.C			MEANCP	7045. KPA	TGREGC	86.7 DEG.C	T04HTR	452.0 DEG.C
TWODPR	18.9 DEG.C					TGCOMP	59.0 DEG.C	T05HTR	470.5 DEG.C
						TGBOUN	34.2 DEG.C	T06HTR	461.8 DEG.C
								T07HTR	461.1 DEG.C
								T08HTR	429.5 DEG.C
								T09HTR	456.1 DEG.C
								T10HTR	438.0 DEG.C
								T11HTR	419.2 DEG.C
								T12HTR	428.0 DEG.C
								T13REG*****	DEG.C
								T14REG	384.5 DEG.C
								T15REG	286.4 DEG.C
								T16REG	274.9 DEG.C
								T17REG	265.6 DEG.C
								T18REG	267.3 DEG.C
								T19REG	181.0 DEG.C
								T03HED*****	DEG.C

HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.21 L/MIN	1 PWRIN	3509. WATTS	VX1HOR	0.4 CM/S	PWROUT	670. WATTS
TWINCL	24.3 DEG.C	2 QCOOLR	2859. WATTS	VY1VER	9.2 CM/S	INDPWR	687. WATTS
TDLCL	9.79 DEG.C	3 QDSHPT	601. WATTS			PISTST	2.80 CM
TWOCLR	32.66 DEG.C	4 EXTEFF	19.1 %	PHASE ANGLES		DISPST	2.39 CM
		5 TAVHTR	449.2 DEG.C	PADISP	61.9 DEG.		
		6 INTEFF	19.0 %	PAPRES	-9.8 DEG.		
		8 AMPS	1281. AMPS				
		9 QDISPG	2. WATTS	ENGINE SPEED			
		10 QDISP	10. WATTS	FREQ	30.2 HZ		
		11 QREG1	70. WATTS				
		12 QREG2	95. WATTS				
		13 QREG3	25. WATTS				
		14 PFP	978.5 KPA/CM				
		15 PFD	205.0 KPA/CM				
		16 MBEALE	0.00441				

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:43:10.54 RDG 1032

FLUID HELIUM		BAROM 14.337 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.12 L/MIN	AMPS1	724. AMPS	PRESUP	7237. KPA	TGEXP	411.8 DEG.C	T01HTR	447.8 DEG.C
		AMPS2	663. AMPS					T02HTR	464.2 DEG.C
TWINDP	16.0 DEG.C	VOLTG	2.97 VOLTS	MEANBP	6855. KPA	TGREGH	398.5 DEG.C	T03HTR	465.5 DEG.C
TDLDP	2.20 DEG.C			MEANCP	7054. KPA	TGREGC	83.5 DEG.C	T04HTR	452.3 DEG.C
TWODPR	18.9 DEG.C					TGCOMP	67.8 DEG.C	T05HTR	473.3 DEG.C
						TGBOUN	34.6 DEG.C	T06HTR	464.8 DEG.C
								T07HTR	463.9 DEG.C
								T08HTR	426.4 DEG.C
								T09HTR	454.4 DEG.C
								T10HTR	438.6 DEG.C
								T11HTR	414.1 DEG.C
								T12HTR	430.1 DEG.C
								T13REG*****	DEG.C
								T14REG	368.1 DEG.C
								T15REG	251.0 DEG.C
								T16REG	236.0 DEG.C
								T17REG	222.4 DEG.C
								T18REG	230.0 DEG.C
								T19REG	159.6 DEG.C
								T03HED*****	DEG.C

HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.21 L/MIN	1 PWRIN	4116. WATTS	VX1HOR	0.5 CM/S	PWROUT	688. WATTS
TWINCL	24.3 DEG.C	2 QCOOLR	3252. WATTS	VY1VER	9.8 CM/S	INDPWR	692. WATTS
TDLCL	11.13 DEG.C	3 QDSHPT	632. WATTS			PISTST	2.98 CM
TWOCLR	34.14 DEG.C	4 EXTEFF	16.7 %	PHASE ANGLES		DISPST	2.40 CM
		5 TAVHTR	449.6 DEG.C	PADISP	61.6 DEG.		
		6 INTEFF	17.5 %	PAPRES	-8.4 DEG.		
		8 AMPS	1387. AMPS				
		9 QDISPG	2. WATTS	ENGINE SPEED			
		10 QDISP	10. WATTS	FREQ	30.1 HZ		
		11 QREG1	57. WATTS				
		12 QREG2	112. WATTS				
		13 QREG3	24. WATTS				
		14 PFP	961.9 KPA/CM				
		15 PFD	185.2 KPA/CM				
		16 MBEALE	0.00426				

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:38:38.30 RDG 1066

FLUID HELIUM		BAROM 14.342 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.21 L/MIN	AMPS1	512. AMPS	PRESUP	5570. KPA	TGEXP	566.1 DEG.C	T01HTR	598.8 DEG.C
		AMPS2	429. AMPS					T02HTR	607.5 DEG.C
TWINDP	22.3 DEG.C	VOLTG	2.02 VOLTS	MEANBP	5503. KPA	TGREGH	561.3 DEG.C	T03HTR	612.9 DEG.C
TDLDP	0.96 DEG.C			MEANCP	5532. KPA	TGREGC	74.3 DEG.C	T04HTR	606.0 DEG.C
TWODPR	23.9 DEG.C					TGCOMP	40.2 DEG.C	T05HTR	616.0 DEG.C
						TGBOUN	34.4 DEG.C	T06HTR	606.1 DEG.C
								T07HTR	605.4 DEG.C
								T08HTR	583.7 DEG.C
								T09HTR	606.5 DEG.C
								T10HTR	594.1 DEG.C
								T11HTR	579.1 DEG.C
								T12HTR	586.0 DEG.C
								T13REG*****	DEG.C
								T14REG	519.7 DEG.C
								T15REG	384.2 DEG.C
								T16REG	376.7 DEG.C
								T17REG	363.9 DEG.C
								T18REG	369.8 DEG.C
								T19REG	220.3 DEG.C
								T03HED*****	DEG.C

HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.41 L/MIN	1 PWRIN	1904. WATTS	VX1HOR	0.2 CM/S	PWROUT	418. WATTS
TWINCL	25.1 DEG.C	2 QCOOLR	1559. WATTS	VY1VER	4.7 CM/S	INDPWR	443. WATTS
TDLCL	5.09 DEG.C	3 QDSHPT	280. WATTS			PISTST	1.82 CM
TWOCLR	28.93 DEG.C	4 EXTEFF	21.9 %	PHASE ANGLES		DISPST	1.95 CM
		5 TAVHTR	600.2 DEG.C	PADISP	56.5 DEG.		
		6 INTEFF	21.1 %	PAPRES	-22.1 DEG.		
		8 AMPS	941. AMPS				
		9 QDISPG	3. WATTS	ENGINE SPEED			
		10 QDISP	15. WATTS	FREQ	26.7 HZ		
		11 QREG1	121. WATTS				
		12 QREG2	122. WATTS				
		13 QREG3	38. WATTS				
		14 PFP	649.2 KPA/CM				
		15 PFD	232.3 KPA/CM				
		16 MBEALE	0.00608				

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:42:44.30 RDG 1067

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.21 L/MIN	AMPS1 543. AMPS	PRESUP 5562. KPA	TGEXP 563.3 DEG.C	T01HTR 599.9 DEG.C
	AMPS2 455. AMPS			T02HTR 606.4 DEG.C
TWINDP 22.4 DEG.C	VOLTG 2.15 VOLTS	MEANBP 5482. KPA	TGREGH 561.9 DEG.C	T03HTR 612.9 DEG.C
TDLDP 1.14 DEG.C		MEANCP 5513. KPA	TGREGC 76.6 DEG.C	T04HTR 605.8 DEG.C
TWODPR 24.0 DEG.C			TGCOMP 42.0 DEG.C	T05HTR 615.8 DEG.C
			TGBOUN 34.5 DEG.C	T06HTR 606.4 DEG.C
				T07HTR 605.5 DEG.C
				T08HTR 583.5 DEG.C
				T09HTR 608.0 DEG.C
				T10HTR 591.7 DEG.C
				T11HTR 579.4 DEG.C
				T12HTR 582.5 DEG.C
				T13REG***** DEG.C
				T14REG 518.5 DEG.C
				T15REG 384.1 DEG.C
				T16REG 379.5 DEG.C
				T17REG 368.1 DEG.C
				T18REG 368.6 DEG.C
				T19REG 221.5 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.40 L/MIN	1 PWRIN 2142. WATTS	VX1HOR 0.2 CM/S	PWROUT 495. WATTS	
TWINCL 25.0 DEG.C	2 QCOOLR 1692. WATTS	VY1VER 5.2 CM/S	INDPWR 524. WATTS	
TDLCL 5.54 DEG.C	3 QDSHPT 335. WATTS		PISTST 2.01 CM	
TWOCLR 29.30 DEG.C	4 EXTEFF 23.1 %	PHASE ANGLES	DISPST 2.07 CM	
	5 TAVHTR 599.8 DEG.C	PADISP 57.8 DEG.		
	6 INTEFF 22.6 %	PAPRES -20.9 DEG.		
	8 AMPS 998. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 26.9 HZ	PAMPC 699. KPA	
	10 QDISP 15. WATTS		DISPCP 2.13 CM	
	11 QREG1 119. WATTS		PISTCP 2.45 CM	
	12 QREG2 122. WATTS		PDYNDB***** KPA	
	13 QREG3 38. WATTS		PDLCLR 2.55 KPA	
	14 PFP 805.9 KPA/CM		PDLREG 31.90 KPA	
	15 PFD 284.8 KPA/CM		PDLDIS 54.30 KPA	
	16 NBEALE 0.00652			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:47:26.35 RDG 1068

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.21 L/MIN	AMPS1 575. AMPS	PRESUP 5565. KPA	TGEXP 560.8 DEG.C	T01HTR 602.3 DEG.C
	AMPS2 478. AMPS			T02HTR 605.2 DEG.C
TWINDP 22.4 DEG.C	VOLTG 2.27 VOLTS	MEANBP 5503. KPA	TGREGH 562.5 DEG.C	T03HTR 614.6 DEG.C
TDLDP 1.31 DEG.C		MEANCP 5536. KPA	TGREGC 78.9 DEG.C	T04HTR 605.5 DEG.C
TWODPR 24.3 DEG.C			TGCOMP 43.5 DEG.C	T05HTR 616.9 DEG.C
			TGBOUN 35.0 DEG.C	T06HTR 606.9 DEG.C
				T07HTR 605.7 DEG.C
				T08HTR 584.7 DEG.C
				T09HTR 610.2 DEG.C
				T10HTR 590.8 DEG.C
				T11HTR 580.2 DEG.C
				T12HTR 579.9 DEG.C
				T13REG***** DEG.C
				T14REG 517.8 DEG.C
				T15REG 385.2 DEG.C
				T16REG 381.9 DEG.C
				T17REG 371.8 DEG.C
				T18REG 369.8 DEG.C
				T19REG 223.8 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.41 L/MIN	1 PWRIN 2387. WATTS	VX1HOR 0.2 CM/S	PWROUT 570. WATTS	
TWINCL 25.1 DEG.C	2 QCOOLR 1833. WATTS	VY1VER 5.8 CM/S	INDPWR 602. WATTS	
TDLCL 5.99 DEG.C	3 QDSHPT 383. WATTS		PISTST 2.20 CM	
TWOCLR 29.80 DEG.C	4 EXTEFF 23.9 %	PHASE ANGLES	DISPST 2.19 CM	
	5 TAVHTR 600.2 DEG.C	PADISP 58.1 DEG.		
	6 INTEFF 23.7 %	PAPRES -19.8 DEG.		
	8 AMPS 1053. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 27.0 HZ	PAMPC 768. KPA	
	10 QDISP 15. WATTS		DISPCP 2.16 CM	
	11 QREG1 118. WATTS		PISTCP 2.44 CM	
	12 QREG2 120. WATTS		PDYNDB***** KPA	
	13 QREG3 37. WATTS		PDLCLR 2.72 KPA	
	14 PFP 804.1 KPA/CM		PDLREG 35.98 KPA	
	15 PFD 279.2 KPA/CM		PDLDIS 60.75 KPA	
	16 NBEALE 0.00680			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:52:23.35 RDG 1069

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.22 L/MIN	AMPS1 607. AMPS	PRESUP 5561. KPA	TGEXP 558.0 DEG.C	T01HTR 603.6 DEG.C
	AMPS2 500. AMPS			T02HTR 604.9 DEG.C
TWINDP 22.4 DEG.C	VOLTG 2.39 VOLTS	MEANBP 5500. KPA	TGREGH 562.2 DEG.C	T03HTR 615.8 DEG.C
TDLDP 1.51 DEG.C		MEANCP 5501. KPA	TGREGC 81.1 DEG.C	T04HTR 604.8 DEG.C
TWODPR 24.5 DEG.C			TGCOMP 45.8 DEG.C	T05HTR 618.0 DEG.C
			TGBOUN 36.3 DEG.C	T06HTR 608.0 DEG.C
				T07HTR 606.6 DEG.C
				T08HTR 585.1 DEG.C
				T09HTR 611.2 DEG.C
				T10HTR 589.3 DEG.C
				T11HTR 579.9 DEG.C
				T12HTR 577.8 DEG.C
				T13REG***** DEG.C
				T14REG 517.7 DEG.C
				T15REG 386.7 DEG.C
				T16REG 383.2 DEG.C
				T17REG 373.7 DEG.C
				T18REG 370.7 DEG.C
				T19REG 226.8 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.41 L/MIN	1 PWRIN 2642. WATTS	VX1HOR 0.3 CM/S	PWROUT 646. WATTS	
TWINCL 25.1 DEG.C	2 QCOOLR 1979. WATTS	VY1VER 6.5 CM/S	INDPWR 681. WATTS	
TDLCL 6.46 DEG.C	3 QDSHPT 441. WATTS		PISTST 2.41 CM	
TWOCLR 30.43 DEG.C	4 EXTEFF 24.5 %	PHASE ANGLES	DISPST 2.32 CM	
	5 TAVHTR 600.4 DEG.C	PADISP 58.5 DEG.		
	6 INTEFF 24.6 %	PAPRES -18.6 DEG.		
	8 AMPS 1107. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 27.0 HZ	PAMPC 827. KPA	
	10 QDISP 15. WATTS		DISPCP 2.19 CM	
	11 QREG1 117. WATTS		PISTCP 2.44 CM	
	12 QREG2 119. WATTS		PDYNDB***** KPA	
	13 QREG3 37. WATTS		PDLCLR 3.56 KPA	
	14 PFP 785.9 KPA/CM		PDLREG 40.05 KPA	
	15 PFD 266.8 KPA/CM		PDLDIS 67.20 KPA	
	16 NBEALE 0.00704			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:58:23.35 RDG 1070

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.20 L/MIN	AMPS1 635. AMPS	PRESUP 5553. KPA	TGEXP 555.0 DEG.C	T01HTR 599.7 DEG.C
	AMPS2 521. AMPS			T02HTR 606.6 DEG.C
TWINDP 22.4 DEG.C	VOLTG 2.49 VOLTS	MEANBP 5494. KPA	TGREGH 554.7 DEG.C	T03HTR 614.7 DEG.C
TDLDP 1.70 DEG.C		MEANCP 5531. KPA	TGREGC 79.7 DEG.C	T04HTR 601.3 DEG.C
TWODPR 24.7 DEG.C			TGCOMP 49.4 DEG.C	T05HTR 619.1 DEG.C
			TGBOUN 37.7 DEG.C	T06HTR 609.4 DEG.C
				T07HTR 608.6 DEG.C
				T08HTR 580.8 DEG.C
				T09HTR 607.0 DEG.C
				T10HTR 586.7 DEG.C
				T11HTR 573.8 DEG.C
				T12HTR 577.7 DEG.C
				T13REG***** DEG.C
				T14REG 515.2 DEG.C
				T15REG 379.5 DEG.C
				T16REG 367.6 DEG.C
				T17REG 352.7 DEG.C
				T18REG 355.3 DEG.C
				T19REG 222.0 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.40 L/MIN	1 PWRIN 2879. WATTS	VX1HOR 0.3 CM/S	PWROUT 720. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 2137. WATTS	VY1VER 7.1 CM/S	INDPWR 759. WATTS	
TDLCL 7.00 DEG.C	3 QDSHPT 495. WATTS		PISTST 2.61 CM	
TWOCLR 30.99 DEG.C	4 EXTEFF 25.0 %	PHASE ANGLES	DISPST 2.46 CM	
	5 TAVHTR 598.8 DEG.C	PADISP 58.7 DEG.		
	6 INTEFF 25.2 %	PAPRES -18.0 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 1156. AMPS	ENGINE SPEED	PAMPC 910. KPA	
	9 QDISPG 3. WATTS	FREQ 26.9 HZ	DISPCP 2.25 CM	
	10 QDISP 14. WATTS		PISTCP 2.46 CM	
	11 QREG1 108. WATTS		PDYNDB***** KPA	
	12 QREG2 130. WATTS		PDLCLR 4.41 KPA	
	13 QREG3 37. WATTS		PDLREG 42.43 KPA	
	14 PFP 795.4 KPA/CM		PDLDIS 71.95 KPA	
	15 PFD 267.8 KPA/CM			
	16 NBEALE 0.00729			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 15:03:44.35 RDG 1071

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.18 L/MIN	AMPS1 680. AMPS	PRESUP 5546. KPA	TGEXP 550.9 DEG.C	T01HTR 596.8 DEG.C
	AMPS2 558. AMPS			T02HTR 610.6 DEG.C
TWINDP 22.4 DEG.C	VOLTG 2.68 VOLTS	MEANBP 5489. KPA	TGREGH 541.6 DEG.C	T03HTR 614.7 DEG.C
TDLDP 1.91 DEG.C		MEANCP 5530. KPA	TGREGC 74.2 DEG.C	T04HTR 598.6 DEG.C
TWODPR 24.9 DEG.C			TGCOMP 55.2 DEG.C	T05HTR 621.7 DEG.C
			TGBOUN 38.7 DEG.C	T06HTR 611.7 DEG.C
				T07HTR 611.3 DEG.C
				T08HTR 577.1 DEG.C
				T09HTR 602.4 DEG.C
				T10HTR 586.1 DEG.C
				T11HTR 567.0 DEG.C
				T12HTR 579.7 DEG.C
				T13REG***** DEG.C
				T14REG 499.2 DEG.C
				T15REG 343.6 DEG.C
				T16REG 321.0 DEG.C
				T17REG 298.4 DEG.C
				T18REG 311.2 DEG.C
				T19REG 198.0 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.41 L/MIN	1 PWRIN 3315. WATTS	VX1HOR 0.3 CM/S	PWROUT 785. WATTS	
TWINCL 25.3 DEG.C	2 QCOOLR 2441. WATTS	VY1VER 7.6 CM/S	INDPWR 833. WATTS	
TDLCL 7.97 DEG.C	3 QDSHPT 554. WATTS		PISTST 2.79 CM	
TWOCLR 32.10 DEG.C	4 EXTEFF 23.7 %	PHASE ANGLES	DISPST 2.55 CM	
	5 TAVHTR 598.1 DEG.C	PADISP 59.5 DEG.		
	6 INTEFF 24.3 %	PAPRES -17.3 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 1238. AMPS	ENGINE SPEED	PAMPC 965. KPA	
	9 QDISPG 3. WATTS	FREQ 26.8 HZ	DISPCP 2.31 CM	
	10 QDISP 14. WATTS		PISTCP 2.44 CM	
	11 QREG1 92. WATTS		PDYNDB***** KPA	
	12 QREG2 152. WATTS		PDLCLR 6.28 KPA	
	13 QREG3 36. WATTS		PDLREG 44.80 KPA	
	14 PFP 780.6 KPA/CM		PDLDIS 77.38 KPA	
	15 PFD 261.2 KPA/CM			
	16 NBEALE 0.00744			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 15:07:17.40 RDG 1072

FLUID HELIUM BAROM 14.342 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.17 L/MIN	AMPS1 696. AMPS	PRESUP 5559. KPA	TGEXP 548.3 DEG.C	T01HTR 594.4 DEG.C
	AMPS2 573. AMPS			T02HTR 613.1 DEG.C
TWINDP 22.5 DEG.C	VOLTG 2.75 VOLTS	MEANBP 5494. KPA	TGREGH 532.0 DEG.C	T03HTR 614.0 DEG.C
TDLDP 1.96 DEG.C		MEANCP 5536. KPA	TGREGC 72.3 DEG.C	T04HTR 597.7 DEG.C
TWODPR 25.0 DEG.C			TGCOMP 59.7 DEG.C	T05HTR 622.3 DEG.C
			TGBOUN 39.4 DEG.C	T06HTR 612.6 DEG.C
				T07HTR 612.1 DEG.C
				T08HTR 574.5 DEG.C
				T09HTR 599.4 DEG.C
				T10HTR 586.0 DEG.C
				T11HTR 562.9 DEG.C
				T12HTR 581.6 DEG.C
				T13REG***** DEG.C
				T14REG 487.4 DEG.C
				T15REG 321.4 DEG.C
				T16REG 296.8 DEG.C
				T17REG 270.4 DEG.C
				T18REG 288.0 DEG.C
				T19REG 184.7 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.42 L/MIN	1 PWRIN 3489. WATTS	VX1HOR 0.3 CM/S	PWROUT 756. WATTS	
TWINCL 25.5 DEG.C	2 QCOOLR 2559. WATTS	VY1VER 7.7 CM/S	INDPWR 806. WATTS	
TDLCL 8.35 DEG.C	3 QDSHPT 566. WATTS		PISTST 2.83 CM	
TWOCLR 32.86 DEG.C	4 EXTEFF 21.7 %	PHASE ANGLES	DISPST 2.44 CM	
	5 TAVHTR 597.5 DEG.C	PADISP 59.2 DEG.		
	6 INTEFF 22.8 %	PAPRES -16.2 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 1269. AMPS	ENGINE SPEED	PAMPC 999. KPA	
	9 QDISPG 3. WATTS	FREQ 26.7 HZ	DISPCP 2.34 CM	
	10 QDISP 14. WATTS		PISTCP 2.43 CM	
	11 QREG1 84. WATTS		PDYNDB***** KPA	
	12 QREG2 162. WATTS		PDLCLR***** KPA	
	13 QREG3 36. WATTS		PDLREG 26.13 KPA	
	14 PFP 794.0 KPA/CM		PDLDIS 74.67 KPA	
	15 PFD 265.6 KPA/CM			
	16 NBEALE 0.00709			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:31:24.59 RDG 1075

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.84 L/MIN	AMPS1 448. AMPS	PRESUP 4145. KPA	TGEXP 562.2 DEG.C	T01HTR 606.5 DEG.C
	AMPS2 382. AMPS			T02HTR 598.6 DEG.C
TWINDP 22.8 DEG.C	VOLTG 1.78 VOLTS	MEANBP 4011. KPA	TGREGH 562.2 DEG.C	T03HTR 613.7 DEG.C
TDLDP 0.32 DEG.C		MEANCP 4028. KPA	TGREGC 66.0 DEG.C	T04HTR 607.6 DEG.C
TWODPR 23.8 DEG.C			TGCOMP 33.8 DEG.C	T05HTR 611.9 DEG.C
			TGBOUN 28.7 DEG.C	T06HTR 596.9 DEG.C
				T07HTR 598.3 DEG.C
				T08HTR 593.9 DEG.C
				T09HTR 606.9 DEG.C
				T10HTR 598.1 DEG.C
				T11HTR 585.3 DEG.C
				T12HTR 584.7 DEG.C
				T13REG***** DEG.C
				T14REG 507.9 DEG.C
				T15REG 366.8 DEG.C
				T16REG 363.4 DEG.C
				T17REG 360.7 DEG.C
				T18REG 361.9 DEG.C
				T19REG 203.1 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.31 L/MIN	1 PWRIN 1481. WATTS	VX1HOR 0.2 CM/S	PWROUT 280. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 890. WATTS	VY1VER 4.1 CM/S	INDPWR 290. WATTS	
TDLCL 2.97 DEG.C	3 QDSHPT 86. WATTS		PISTST 1.82 CM	
TWOCLR 27.79 DEG.C	4 EXTEFF 18.9 %	PHASE ANGLES	DISPST 1.97 CM	
	5 TAVHTR 600.2 DEG.C	PADISP 55.9 DEG.		
	6 INTEFF 23.9 %	PAPRES -23.8 DEG.		
	8 AMPS 830. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 22.8 HZ	PAMPC 438. KPA	
	10 QDISP 15. WATTS		DISPCP 1.76 CM	
	11 QREG1 129. WATTS		PISTCP 2.34 CM	
	12 QREG2 118. WATTS		PDYNDB***** KPA	
	13 QREG3 38. WATTS		PDLCLR 1.19 KPA	
	14 PFP 573.1 KPA/CM		PDLREG 17.65 KPA	
	15 PFD 216.8 KPA/CM		PDLDIS 31.90 KPA	
	16 NBEALE 0.00656			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:37:54.59 RDG 1076

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.85 L/MIN	AMPS1 464. AMPS	PRESUP 4155. KPA	TGEXP 560.4 DEG.C	T01HTR 605.6 DEG.C
	AMPS2 396. AMPS			T02HTR 600.3 DEG.C
TWINDP 22.8 DEG.C	VOLTG 1.85 VOLTS	MEANBP 4008. KPA	TGREGH 562.3 DEG.C	T03HTR 614.9 DEG.C
TDLDP 0.45 DEG.C		MEANCP 4025. KPA	TGREGC 67.6 DEG.C	T04HTR 605.9 DEG.C
TWODPR 23.9 DEG.C			TGCOMP 36.1 DEG.C	T05HTR 614.6 DEG.C
			TGBOUN 29.0 DEG.C	T06HTR 597.9 DEG.C
				T07HTR 599.1 DEG.C
				T08HTR 592.2 DEG.C
				T09HTR 606.3 DEG.C
				T10HTR 597.4 DEG.C
				T11HTR 584.1 DEG.C
				T12HTR 586.1 DEG.C
				T13REG***** DEG.C
				T14REG 511.9 DEG.C
				T15REG 373.5 DEG.C
				T16REG 365.9 DEG.C
				T17REG 362.8 DEG.C
				T18REG 366.2 DEG.C
				T19REG 209.5 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.31 L/MIN	1 PWRIN 1589. WATTS	VX1HOR 0.2 CM/S	PWROUT 327. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 968. WATTS	VY1VER 4.4 CM/S	INDPWR 345. WATTS	
TDLCL 3.24 DEG.C	3 QDSHPT 119. WATTS		PISTST 1.99 CM	
TWOCLR 28.07 DEG.C	4 EXTEFF 20.6 %	PHASE ANGLES	DISPST 2.13 CM	
	5 TAVHTR 600.4 DEG.C	PADISP 55.7 DEG.		
	6 INTEFF 25.3 %	PAPRES -22.8 DEG.		
	8 AMPS 860. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 22.8 HZ	PAMPC 472. KPA	
	10 QDISP 15. WATTS		DISPCP 2.05 CM	
	11 QREG1 127. WATTS		PISTCP 2.36 CM	
	12 QREG2 118. WATTS		PDYNDB***** KPA	
	13 QREG3 38. WATTS		PDLCLR 1.36 KPA	
	14 PFP 562.5 KPA/CM		PDLREG 21.72 KPA	
	15 PFD 208.3 KPA/CM		PDLDIS 36.66 KPA	
	16 NBEALE 0.00700			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:42:21.59 RDG 1077

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.83 L/MIN	AMPS1 501. AMPS	PRESUP 4150. KPA	TGEXP 556.6 DEG.C	T01HTR 603.9 DEG.C
	AMPS2 420. AMPS			T02HTR 599.1 DEG.C
TWINDP 22.9 DEG.C	VOLTG 1.98 VOLTS	MEANBP 4000. KPA	TGREGH 560.5 DEG.C	T03HTR 613.3 DEG.C
TDLDP 0.57 DEG.C		MEANCP 4020. KPA	TGREGC 69.3 DEG.C	T04HTR 605.2 DEG.C
TWODPR 24.1 DEG.C			TGCOMP 38.1 DEG.C	T05HTR 613.6 DEG.C
			TGBOUN 29.9 DEG.C	T06HTR 599.1 DEG.C
				T07HTR 599.9 DEG.C
				T08HTR 589.4 DEG.C
				T09HTR 606.7 DEG.C
				T10HTR 593.7 DEG.C
				T11HTR 582.2 DEG.C
				T12HTR 583.3 DEG.C
				T13REG***** DEG.C
				T14REG 512.1 DEG.C
				T15REG 376.4 DEG.C
				T16REG 369.4 DEG.C
				T17REG 363.7 DEG.C
				T18REG 364.3 DEG.C
				T19REG 212.9 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.29 L/MIN	1 PWRIN 1824. WATTS	VX1HOR 0.2 CM/S	PWROUT 387. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 1071. WATTS	VY1VER 4.9 CM/S	INDPWR 407. WATTS	
TDLCL 3.59 DEG.C	3 QDSHPT 151. WATTS		PISTST 2.20 CM	
TWOCLR 28.42 DEG.C	4 EXTEFF 21.2 %	PHASE ANGLES	DISPST 2.27 CM	
	5 TAVHTR 599.1 DEG.C	PADISP 57.1 DEG.		
	6 INTEFF 26.5 %	PAPRES -21.7 DEG.		
	8 AMPS 921. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 23.0 HZ	PAMPC 541. KPA	
	10 QDISP 15. WATTS		DISPCP 2.06 CM	
	11 QREG1 123. WATTS		PISTCP 2.39 CM	
	12 QREG2 120. WATTS		PDYNDB***** KPA	
	13 QREG3 38. WATTS		PDLCLR 1.87 KPA	
	14 PFP 574.3 KPA/CM		PDLREG 24.44 KPA	
	15 PFD 209.8 KPA/CM		PDLDIS 40.73 KPA	
	16 NBEALE 0.00746			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:47:21.59 RDG 1078

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 3.82 L/MIN	AMPS1 516. AMPS	PRESUP 4146. KPA	TGEXP 555.0 DEG.C	T01HTR 605.0 DEG.C
	AMPS2 434. AMPS			T02HTR 600.1 DEG.C
TWINDP 22.9 DEG.C	VOLTG 2.04 VOLTS	MEANBP 3995. KPA	TGREGH 561.3 DEG.C	T03HTR 613.7 DEG.C
TDLDP 0.80 DEG.C		MEANCP 4018. KPA	TGREGC 71.0 DEG.C	T04HTR 604.9 DEG.C
TWODPR 24.3 DEG.C			TGCOMP 40.2 DEG.C	T05HTR 614.0 DEG.C
			TGBOUN 30.9 DEG.C	T06HTR 600.5 DEG.C
				T07HTR 600.9 DEG.C
				T08HTR 589.6 DEG.C
				T09HTR 607.9 DEG.C
				T10HTR 592.2 DEG.C
				T11HTR 582.9 DEG.C
				T12HTR 582.0 DEG.C
				T13REG***** DEG.C
				T14REG 512.8 DEG.C
				T15REG 378.0 DEG.C
				T16REG 371.9 DEG.C
				T17REG 367.3 DEG.C
				T18REG 365.8 DEG.C
				T19REG 215.0 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.30 L/MIN	1 PWRIN 1939. WATTS	VX1HOR 0.2 CM/S	PWROUT 446. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 1176. WATTS	VY1VER 5.4 CM/S	INDPWR 466. WATTS	
TDLCL 3.94 DEG.C	3 QDSHPT 212. WATTS		PISTST 2.41 CM	
TWOCLR 28.82 DEG.C	4 EXTEFF 23.0 %	PHASE ANGLES	DISPST 2.39 CM	
	5 TAVHTR 599.5 DEG.C	PADISP 58.1 DEG.		
	6 INTEFF 27.5 %	PAPRES -20.3 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 950. AMPS	ENGINE SPEED	PAMPC 576. KPA	
	9 QDISPG 3. WATTS	FREQ 23.0 HZ	DISPCP 2.13 CM	
	10 QDISP 15. WATTS		PISTCP 2.37 CM	
	11 QREG1 122. WATTS		PDYNDB***** KPA	
	12 QREG2 119. WATTS		PDLCLR 2.04 KPA	
	13 QREG3 38. WATTS		PDLREG 27.15 KPA	
	14 PFP 551.3 KPA/CM		PDLDIS 45.48 KPA	
	15 PFD 197.3 KPA/CM			
	16 NBEALE 0.00784			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:52:24.59 RDG 1079

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.08 L/MIN	AMPS1 539. AMPS	PRESUP 4143. KPA	TGEXP 553.1 DEG.C	T01HTR 605.6 DEG.C
	AMPS2 451. AMPS			T02HTR 599.1 DEG.C
TWINDP 23.0 DEG.C	VOLTG 2.13 VOLTS	MEANBP 3988. KPA	TGREGH 560.1 DEG.C	T03HTR 614.7 DEG.C
TDLDP 0.81 DEG.C		MEANCP 4012. KPA	TGREGC 70.5 DEG.C	T04HTR 603.7 DEG.C
TWODPR 24.4 DEG.C			TGCOMP 41.6 DEG.C	T05HTR 615.3 DEG.C
			TGBOUN 32.0 DEG.C	T06HTR 600.7 DEG.C
				T07HTR 601.4 DEG.C
				T08HTR 589.7 DEG.C
				T09HTR 607.8 DEG.C
				T10HTR 591.6 DEG.C
				T11HTR 581.8 DEG.C
				T12HTR 581.3 DEG.C
				T13REG***** DEG.C
				T14REG 512.2 DEG.C
				T15REG 376.1 DEG.C
				T16REG 367.6 DEG.C
				T17REG 363.5 DEG.C
				T18REG 363.0 DEG.C
				T19REG 213.7 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.31 L/MIN	1 PWRIN 2109. WATTS	VX1HOR 0.2 CM/S	PWROUT 500. WATTS	
TWINCL 25.2 DEG.C	2 QCOOLR 1242. WATTS	VY1VER 5.8 CM/S	INDPWR 520. WATTS	
TDLCL 4.15 DEG.C	3 QDSHPT 228. WATTS		PISTST 2.61 CM	
TWOCLR 29.17 DEG.C	4 EXTEFF 23.7 %	PHASE ANGLES	DISPST 2.55 CM	
	5 TAVHTR 599.4 DEG.C	PADISP 58.4 DEG.		
	6 INTEFF 28.7 %	PAPRES -19.7 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 990. AMPS	ENGINE SPEED	PAMPC 645. KPA	
	9 QDISPG 3. WATTS	FREQ 23.0 HZ	DISPCP 2.18 CM	
	10 QDISP 15. WATTS		PISTCP 2.40 CM	
	11 QREG1 121. WATTS		PDYNDB***** KPA	
	12 QREG2 121. WATTS		PDLCLR 2.72 KPA	
	13 QREG3 38. WATTS		PDLREG 29.53 KPA	
	14 PFP 568.8 KPA/CM		PDLDIS 49.21 KPA	
	15 PFD 200.5 KPA/CM			
	16 NBEALE 0.00815			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:58:30.59 RDG 1080

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.11 L/MIN	AMPS1 568. AMPS	PRESUP 4141. KPA	TGEXP 551.4 DEG.C	T01HTR 606.1 DEG.C
	AMPS2 475. AMPS			T02HTR 602.6 DEG.C
TWINDP 23.1 DEG.C	VOLTG 2.25 VOLTS	MEANBP 3986. KPA	TGREGH 552.6 DEG.C	T03HTR 618.9 DEG.C
TDLDP 0.93 DEG.C		MEANCP 4010. KPA	TGREGC 65.5 DEG.C	T04HTR 600.0 DEG.C
TWODPR 24.7 DEG.C			TGCOMP 45.0 DEG.C	T05HTR 621.4 DEG.C
			TGBOUN 33.3 DEG.C	T06HTR 602.5 DEG.C
				T07HTR 604.1 DEG.C
				T08HTR 590.2 DEG.C
				T09HTR 603.7 DEG.C
				T10HTR 594.6 DEG.C
				T11HTR 577.3 DEG.C
				T12HTR 585.2 DEG.C
				T13REG***** DEG.C
				T14REG 506.8 DEG.C
				T15REG 358.9 DEG.C
				T16REG 335.6 DEG.C
				T17REG 331.6 DEG.C
				T18REG 342.7 DEG.C
				T19REG 201.7 DEG.C
				T03HED***** DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.33 L/MIN	1 PWRIN 2342. WATTS	VX1HOR 0.2 CM/S	PWROUT 567. WATTS	
TWINCL 25.3 DEG.C	2 QCOOLR 1402. WATTS	VY1VER 6.3 CM/S	INDPWR 592. WATTS	
TDLCL 4.67 DEG.C	3 QDSHPT 266. WATTS		PISTST 2.82 CM	
TWOCLR 29.62 DEG.C	4 EXTEFF 24.2 %	PHASE ANGLES	DISPST 2.71 CM	
	5 TAVHTR 600.6 DEG.C	PADISP 58.6 DEG.		
	6 INTEFF 28.8 %	PAPRES -19.5 DEG.	DYNAMIC CALCULATIONS	
	8 AMPS 1043. AMPS	ENGINE SPEED	PAMPC 699. KPA	
	9 QDISPG 3. WATTS	FREQ 22.8 HZ	DISPCP 2.21 CM	
	10 QDISP 14. WATTS		PISTCP 2.50 CM	
	11 QREG1 114. WATTS		PDYNDB***** KPA	
	12 QREG2 133. WATTS		PDLCLR 3.53 KPA	
	13 QREG3 38. WATTS		PDLREG 31.22 KPA	
	14 PFP 568.0 KPA/CM		PDLDIS 51.93 KPA	
	15 PFD 202.2 KPA/CM			
	16 NBEALE 0.00860			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 15:02:18.59 RDG 1081

FLUID HELIUM BAROM 14.288 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.10 L/MIN	AMPS1 588. AMPS	PRESUP 4139. KPA	TGEXP 548.9 DEG.C	T01HTR 605.2 DEG.C
	AMPS2 491. AMPS			T02HTR 603.1 DEG.C
TWINDP 23.1 DEG.C	VOLTG 2.32 VOLTS	MEANBP 3983. KPA	TGREGH 549.8 DEG.C	T03HTR 619.5 DEG.C
TDLDP 1.06 DEG.C		MEANCP 4011. KPA	TGREGC 65.1 DEG.C	T04HTR 597.7 DEG.C
TWODPR 24.8 DEG.C			TGCCOMP 47.0 DEG.C	T05HTR 622.8 DEG.C
			TGBOUN 33.9 DEG.C	T06HTR 603.2 DEG.C
				T07HTR 604.6 DEG.C
				T08HTR 588.7 DEG.C
				T09HTR 602.6 DEG.C
				T10HTR 593.1 DEG.C
				T11HTR 575.1 DEG.C
				T12HTR 584.3 DEG.C
				T13REG***** DEG.C
				T14REG 505.5 DEG.C
				T15REG 354.8 DEG.C
				T16REG 328.7 DEG.C
				T17REG 324.6 DEG.C
				T18REG 338.1 DEG.C
				T19REG 198.9 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.34 L/MIN	1 PWRIN 2507. WATTS	VX1HOR 0.3 CM/S	PWROUT 622. WATTS	
TWINCL 25.3 DEG.C	2 QCOOLR 1500. WATTS	VY1VER 6.7 CM/S	INDPWR 647. WATTS	
TDLCL 4.99 DEG.C	3 QDSHPT 302. WATTS		PISTST 3.01 CM	
TWOCLR 29.90 DEG.C	4 EXTEFF 24.8 %	PHASE ANGLES	DISPST 2.86 CM	
	5 TAVHTR 600.0 DEG.C	PADISP 58.9 DEG.		
	6 INTEFF 29.3 %	PAPRES -19.3 DEG.		
	8 AMPS 1079. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 22.7 HZ	PAMPC 733. KPA	
	10 QDISP 14. WATTS		DISPCP 2.11 CM	
	11 QREG1 113. WATTS		PISTCP 2.56 CM	
	12 QREG2 136. WATTS		PDYNDB***** KPA	
	13 QREG3 38. WATTS		PDLCLR 3.39 KPA	
	14 PFP 557.0 KPA/CM		PDLREG 33.26 KPA	
	15 PFD 197.2 KPA/CM		PDLDIS 56.68 KPA	
	16 MBEALE 0.00890			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:16:33.40 RDG 1117

FLUID HELIUM BAROM 14.239 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.40 L/MIN	AMPS1 560. AMPS	PRESUP 7088. KPA	TGEXP 568.8 DEG.C	T01HTR 600.0 DEG.C
	AMPS2 479. AMPS			T02HTR 605.4 DEG.C
TWINDP 25.0 DEG.C	VOLTG 2.22 VOLTS	MEANBP 7007. KPA	TGREGH 563.8 DEG.C	T03HTR 611.6 DEG.C
TDLDP 1.19 DEG.C		MEANCP 7047. KPA	TGREGC 100.3 DEG.C	T04HTR 606.7 DEG.C
TWODPR 26.7 DEG.C			TGCCOMP 52.1 DEG.C	T05HTR 615.9 DEG.C
			TGBOUN 39.5 DEG.C	T06HTR 605.5 DEG.C
				T07HTR 604.2 DEG.C
				T08HTR 582.6 DEG.C
				T09HTR 609.2 DEG.C
				T10HTR 591.2 DEG.C
				T11HTR 579.1 DEG.C
				T12HTR 582.9 DEG.C
				T13REG***** DEG.C
				T14REG 521.2 DEG.C
				T15REG 389.1 DEG.C
				T16REG 392.1 DEG.C
				T17REG 384.9 DEG.C
				T18REG 378.9 DEG.C
				T19REG 232.9 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.36 L/MIN	1 PWRIN 2303. WATTS	VX1HOR 0.3 CM/S	PWROUT 503. WATTS	
TWINCL 39.6 DEG.C	2 QCOOLR 1710. WATTS	VY1VER 6.0 CM/S	INDPWR 544. WATTS	
TDLCL 5.69 DEG.C	3 QDSHPT 363. WATTS		PISTST 1.81 CM	
TWOCLR 44.13 DEG.C	4 EXTEFF 21.9 %	PHASE ANGLES	DISPST 1.86 CM	
	5 TAVHTR 599.5 DEG.C	PADISP 57.9 DEG.		
	6 INTEFF 22.7 %	PAPRES -18.7 DEG.		
	8 AMPS 1039. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 30.1 HZ	PAMPC 797. KPA	
	10 QDISP 15. WATTS		DISPCP 2.02 CM	
	11 QREG1 118. WATTS		PISTCP 2.36 CM	
	12 QREG2 115. WATTS		PDYNDB***** KPA	
	13 QREG3 36. WATTS		PDLCLR 3.05 KPA	
	14 PFP 1009.7 KPA/CM		PDLREG 35.30 KPA	
	15 PFD 324.6 KPA/CM		PDLDIS 65.16 KPA	
	16 MBEALE 0.00512			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:21:24.40 RDG 1118

FLUID HELIUM BAROM 14.239 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.39 L/MIN	AMPS1 597. AMPS	PRESUP 7085. KPA	TGEXP 565.3 DEG.C	T01HTR 601.1 DEG.C
	AMPS2 508. AMPS			T02HTR 604.4 DEG.C
TWINDP 25.1 DEG.C	VOLTG 2.36 VOLTS	MEANBP 7005. KPA	TGREGH 563.9 DEG.C	T03HTR 612.7 DEG.C
TDLDP 1.45 DEG.C		MEANCP 7047. KPA	TGREGC 103.2 DEG.C	T04HTR 605.5 DEG.C
TWODPR 27.0 DEG.C			TGCCOMP 54.8 DEG.C	T05HTR 616.6 DEG.C
			TGBOUN 40.7 DEG.C	T06HTR 606.3 DEG.C
				T07HTR 604.7 DEG.C
				T08HTR 582.5 DEG.C
				T09HTR 610.0 DEG.C
				T10HTR 588.9 DEG.C
				T11HTR 578.6 DEG.C
				T12HTR 579.5 DEG.C
				T13REG***** DEG.C
				T14REG 521.3 DEG.C
				T15REG 392.8 DEG.C
				T16REG 396.2 DEG.C
				T17REG 389.8 DEG.C
				T18REG 382.9 DEG.C
				T19REG 237.7 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.37 L/MIN	1 PWRIN 2608. WATTS	VX1HOR 0.3 CM/S	PWROUT 587. WATTS	
TWINCL 39.7 DEG.C	2 QCOOLR 1842. WATTS	VY1VER 6.7 CM/S	INDPWR 631. WATTS	
TDLCL 6.10 DEG.C	3 QDSHPT 442. WATTS		PISTST 2.01 CM	
TWOCLR 44.66 DEG.C	4 EXTEFF 22.5 %	PHASE ANGLES	DISPST 1.97 CM	
	5 TAVHTR 599.2 DEG.C	PADISP 57.9 DEG.		
	6 INTEFF 24.2 %	PAPRES -18.0 DEG.		
	8 AMPS 1105. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 30.2 HZ	PAMPC 886. KPA	
	10 QDISP 14. WATTS		DISPCP 2.23 CM	
	11 QREG1 118. WATTS		PISTCP 2.37 CM	
	12 QREG2 112. WATTS		PDYNDB***** KPA	
	13 QREG3 36. WATTS		PDLCLR 3.73 KPA	
	14 PFP 1009.5 KPA/CM		PDLREG 41.07 KPA	
	15 PFD 327.9 KPA/CM		PDLDIS 80.10 KPA	
	16 MBEALE 0.00538			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:26:36.40 RDG 1119

FLUID HELIUM BAROM 14.234 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.38 L/MIN	AMPS1 629. AMPS	PRESUP 7086. KPA	TGEXP 562.9 DEG.C	T01HTR 602.1 DEG.C
TWINDP 25.2 DEG.C	AMPS2 536. AMPS	MEANBP 7009. KPA	TGREGH 563.0 DEG.C	T02HTR 604.2 DEG.C
TDLDP 1.63 DEG.C	VOLTG 2.49 VOLTS	MEANCP 7053. KPA	TGREGC 105.7 DEG.C	T03HTR 614.5 DEG.C
TWODPR 27.4 DEG.C			TGCOMP 57.4 DEG.C	T04HTR 605.0 DEG.C
			TGBOUN 41.6 DEG.C	T05HTR 618.6 DEG.C
				T06HTR 607.2 DEG.C
				T07HTR 605.7 DEG.C
				T08HTR 582.5 DEG.C
				T09HTR 611.1 DEG.C
				T10HTR 588.5 DEG.C
				T11HTR 578.0 DEG.C
				T12HTR 578.3 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.36 L/MIN	1 PWRIN 2900. WATTS	VX1HOR 0.4 CM/S	PWROUT 667. WATTS	T13REG***** DEG.C
TWINCL 39.7 DEG.C	2 QCOOLR 2073. WATTS	VY1VER 7.3 CM/S	INDPWR 718. WATTS	T14REG 521.9 DEG.C
TDLCL 6.88 DEG.C	3 QDSHPT 496. WATTS		PISTST 2.19 CM	T15REG 395.2 DEG.C
TWOCLR 45.37 DEG.C	4 EXTEFF 23.0 %	PHASE ANGLES	DISPST 2.09 CM	T16REG 397.0 DEG.C
	5 TAVHTR 599.6 DEG.C	PADISP 58.4 DEG.		T17REG 390.9 DEG.C
	6 INTEFF 24.3 %	PAPRES -16.8 DEG.	DYNAMIC CALCULATIONS	T18REG 384.7 DEG.C
	8 AMPS 1165. AMPS	ENGINE SPEED	PAMPC 970. KPA	T19REG 241.6 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.2 HZ	DISPCP 2.26 CM	
	10 QDISP 14. WATTS		PISTCP 2.39 CM	
	11 QREG1 116. WATTS		PDYNDB***** KPA	
	12 QREG2 111. WATTS		PDLCLR 4.24 KPA	
	13 QREG3 35. WATTS		PDLREG 45.48 KPA	
	14 PFP 1003.3 KPA/CM		PDLDIS 80.78 KPA	
	15 PFD 314.8 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00558			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:31:27.40 RDG 1120

FLUID HELIUM BAROM 14.234 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.38 L/MIN	AMPS1 667. AMPS	PRESUP 7089. KPA	TGEXP 559.6 DEG.C	T01HTR 602.7 DEG.C
TWINDP 25.3 DEG.C	AMPS2 568. AMPS	MEANBP 7011. KPA	TGREGH 561.7 DEG.C	T02HTR 604.0 DEG.C
TDLDP 1.97 DEG.C	VOLTG 2.64 VOLTS	MEANCP 7061. KPA	TGREGC 108.4 DEG.C	T03HTR 616.3 DEG.C
TWODPR 27.7 DEG.C			TGCOMP 60.5 DEG.C	T04HTR 603.9 DEG.C
			TGBOUN 42.9 DEG.C	T05HTR 620.6 DEG.C
				T06HTR 608.3 DEG.C
				T07HTR 606.7 DEG.C
				T08HTR 582.1 DEG.C
				T09HTR 611.5 DEG.C
				T10HTR 587.4 DEG.C
				T11HTR 576.6 DEG.C
				T12HTR 576.3 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.39 L/MIN	1 PWRIN 3258. WATTS	VX1HOR 0.4 CM/S	PWROUT 757. WATTS	T13REG***** DEG.C
TWINCL 39.8 DEG.C	2 QCOOLR 2345. WATTS	VY1VER 8.0 CM/S	INDPWR 814. WATTS	T14REG 522.0 DEG.C
TDLCL 7.74 DEG.C	3 QDSHPT 598. WATTS		PISTST 2.42 CM	T15REG 397.2 DEG.C
TWOCLR 46.23 DEG.C	4 EXTEFF 23.2 %	PHASE ANGLES	DISPST 2.22 CM	T16REG 397.8 DEG.C
	5 TAVHTR 599.7 DEG.C	PADISP 58.4 DEG.		T17REG 391.7 DEG.C
	6 INTEFF 24.4 %	PAPRES -15.7 DEG.	DYNAMIC CALCULATIONS	T18REG 386.3 DEG.C
	8 AMPS 1234. AMPS	ENGINE SPEED	PAMPC 1083. KPA	T19REG 245.2 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.2 HZ	DISPCP 2.27 CM	
	10 QDISP 14. WATTS		PISTCP 2.38 CM	
	11 QREG1 114. WATTS		PDYNDB***** KPA	
	12 QREG2 110. WATTS		PDLCLR 5.09 KPA	
	13 QREG3 35. WATTS		PDLREG 51.25 KPA	
	14 PFP 1010.9 KPA/CM		PDLDIS 89.26 KPA	
	15 PFD 309.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00575			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:36:21.40 RDG 1121

FLUID HELIUM BAROM 14.234 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON				
HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.37 L/MIN	AMPS1 702. AMPS	PRESUP 7093. KPA	TGEXP 557.0 DEG.C	T01HTR 602.6 DEG.C
TWINDP 25.4 DEG.C	AMPS2 595. AMPS	MEANBP 7012. KPA	TGREGH 559.4 DEG.C	T02HTR 604.7 DEG.C
TDLDP 2.18 DEG.C	VOLTG 2.78 VOLTS	MEANCP 7066. KPA	TGREGC 109.0 DEG.C	T03HTR 616.7 DEG.C
TWODPR 28.0 DEG.C			TGCOMP 64.0 DEG.C	T04HTR 603.0 DEG.C
			TGBOUN 44.0 DEG.C	T05HTR 621.4 DEG.C
				T06HTR 609.5 DEG.C
				T07HTR 608.0 DEG.C
				T08HTR 580.9 DEG.C
				T09HTR 611.3 DEG.C
				T10HTR 586.0 DEG.C
				T11HTR 574.7 DEG.C
				T12HTR 574.8 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.36 L/MIN	1 PWRIN 3599. WATTS	VX1HOR 0.4 CM/S	PWROUT 845. WATTS	T13REG***** DEG.C
TWINCL 39.9 DEG.C	2 QCOOLR 2532. WATTS	VY1VER 8.6 CM/S	INDPWR 907. WATTS	T14REG 519.6 DEG.C
TDLCL 8.42 DEG.C	3 QDSHPT 662. WATTS		PISTST 2.61 CM	T15REG 391.9 DEG.C
TWOCLR 46.99 DEG.C	4 EXTEFF 23.5 %	PHASE ANGLES	DISPST 2.35 CM	T16REG 391.7 DEG.C
	5 TAVHTR 599.5 DEG.C	PADISP 58.5 DEG.		T17REG 386.0 DEG.C
	6 INTEFF 25.0 %	PAPRES -14.9 DEG.	DYNAMIC CALCULATIONS	T18REG 378.7 DEG.C
	8 AMPS 1296. AMPS	ENGINE SPEED	PAMPC 1171. KPA	T19REG 241.7 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.3 HZ	DISPCP 2.29 CM	
	10 QDISP 14. WATTS		PISTCP 2.34 CM	
	11 QREG1 111. WATTS		PDYNDB***** KPA	
	12 QREG2 114. WATTS		PDLCLR 6.11 KPA	
	13 QREG3 35. WATTS		PDLREG 56.34 KPA	
	14 PFP 1007.1 KPA/CM		PDLDIS 96.73 KPA	
	15 PFD 300.9 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00593			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:41:21.40 RDG 1122

FLUID HELIUM BAROM 14.234 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.37 L/MIN	AMPS1 744. AMPS	PRESUP 7093. KPA	TGEXP 553.0 DEG.C	T01HTR 600.3 DEG.C
	AMPS2 627. AMPS			T02HTR 606.8 DEG.C
TWINDP 25.6 DEG.C	VOLTG 2.91 VOLTS	MEANBP 7014. KPA	TGREGH 550.7 DEG.C	T03HTR 614.7 DEG.C
TDLDP 2.52 DEG.C		MEANCP 7072. KPA	TGREGC 104.8 DEG.C	T04HTR 601.2 DEG.C
TWODPR 28.5 DEG.C			TGCOMP 70.5 DEG.C	T05HTR 621.1 DEG.C
			TGBOUN 45.8 DEG.C	T06HTR 610.5 DEG.C
				T07HTR 608.9 DEG.C
				T08HTR 577.5 DEG.C
				T09HTR 608.8 DEG.C
				T10HTR 583.8 DEG.C
				T11HTR 569.9 DEG.C
				T12HTR 574.2 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.35 L/MIN	1 PWRIN 3994. WATTS	VX1HOR 0.4 CM/S	PWROUT 912. WATTS	T13REG***** DEG.C
TWINCL 39.9 DEG.C	2 QCOOLR 2889. WATTS	VY1VER 9.1 CM/S	INDPWR 976. WATTS	T14REG 505.1 DEG.C
TELCL 9.62 DEG.C	3 QDSHPT 765. WATTS		PISTST 2.77 CM	T15REG 358.4 DEG.C
TWOCLR 48.16 DEG.C	4 EXTEFF 22.8 %	PHASE ANGLES	DISPST 2.44 CM	T16REG 355.7 DEG.C
	5 TAVHTR 598.1 DEG.C	PADISP 58.5 DEG.		T17REG 348.0 DEG.C
	6 INTEFF 24.0 %	PAPRES -14.4 DEG.		T18REG 338.4 DEG.C
	8 AMPS 1371. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 216.9 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.1 HZ	PAMPC 1196. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.39 CM	
	11 QREG1 99. WATTS		PISTCP 2.38 CM	
	12 QREG2 135. WATTS		PDYNDB***** KPA	
	13 QREG3 35. WATTS		PDLCLR 6.28 KPA	
	14 PFP 967.1 KPA/CM		PDLREG 55.66 KPA	
	15 PFD 285.6 KPA/CM		PDLDIS 97.07 KPA	
	16 NBEALE 0.00606			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 13:46:04.27 RDG 1196

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.00 L/MIN	AMPS1 557. AMPS	PRESUP 7217. KPA	TGEXP 570.3 DEG.C	T01HTR 599.5 DEG.C
	AMPS2 461. AMPS			T02HTR 607.4 DEG.C
TWINDP 18.2 DEG.C	VOLTG 2.22 VOLTS	MEANBP 6995. KPA	TGREGH 563.9 DEG.C	T03HTR 611.0 DEG.C
TDLDP 1.58 DEG.C		MEANCP 7025. KPA	TGREGC 113.1 DEG.C	T04HTR 608.2 DEG.C
TWODPR 20.5 DEG.C			TGCOMP 61.3 DEG.C	T05HTR 615.2 DEG.C
			TGBOUN 37.7 DEG.C	T06HTR 607.0 DEG.C
				T07HTR 605.8 DEG.C
				T08HTR 582.1 DEG.C
				T09HTR 610.2 DEG.C
				T10HTR 591.5 DEG.C
				T11HTR 579.7 DEG.C
				T12HTR 584.1 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.15 L/MIN	1 PWRIN 2263. WATTS	VX1HOR 0.4 CM/S	PWROUT 473. WATTS	T13REG***** DEG.C
TWINCL 53.9 DEG.C	2 QCOOLR 1278. WATTS	VY1VER 5.8 CM/S	INDPWR 512. WATTS	T14REG 522.9 DEG.C
TDLCL 4.48 DEG.C	3 QDSHPT 439. WATTS		PISTST 1.80 CM	T15REG 394.2 DEG.C
TWOCLR 57.49 DEG.C	4 EXTEFF 20.9 %	PHASE ANGLES	DISPST 1.82 CM	T16REG 396.6 DEG.C
	5 TAVHTR 600.1 DEG.C	PADISP 58.3 DEG.		T17REG 385.4 DEG.C
	6 INTEFF 27.0 %	PAPRES -17.8 DEG.		T18REG 380.3 DEG.C
	8 AMPS 1018. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 241.3 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.1 HZ	PAMPC 787. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.06 CM	
	11 QREG1 113. WATTS		PISTCP 2.36 CM	
	12 QREG2 116. WATTS		PDYNDB***** KPA	
	13 QREG3 35. WATTS		PDLCLR 3.05 KPA	
	14 PFP 997.0 KPA/CM		PDLREG 34.28 KPA	
	15 PFD 311.0 KPA/CM		PDLDIS 61.43 KPA	
	16 NBEALE 0.00486			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 13:50:19.27 RDG 1197

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.02 L/MIN	AMPS1 598. AMPS	PRESUP 7260. KPA	TGEXP 567.2 DEG.C	T01HTR 600.1 DEG.C
	AMPS2 490. AMPS			T02HTR 606.6 DEG.C
TWINDP 18.3 DEG.C	VOLTG 2.37 VOLTS	MEANBP 6995. KPA	TGREGH 562.6 DEG.C	T03HTR 612.9 DEG.C
TDLDP 1.80 DEG.C		MEANCP 7027. KPA	TGREGC 116.1 DEG.C	T04HTR 607.5 DEG.C
TWODPR 20.8 DEG.C			TGCOMP 63.2 DEG.C	T05HTR 617.2 DEG.C
			TGBOUN 38.8 DEG.C	T06HTR 607.8 DEG.C
				T07HTR 606.6 DEG.C
				T08HTR 581.6 DEG.C
				T09HTR 611.3 DEG.C
				T10HTR 590.3 DEG.C
				T11HTR 578.8 DEG.C
				T12HTR 582.1 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.15 L/MIN	1 PWRIN 2582. WATTS	VX1HOR 0.5 CM/S	PWROUT 557. WATTS	T13REG***** DEG.C
TWINCL 54.0 DEG.C	2 QCOOLR 1444. WATTS	VY1VER 6.5 CM/S	INDPWR 602. WATTS	T14REG 523.7 DEG.C
TDLCL 5.06 DEG.C	3 QDSHPT 504. WATTS		PISTST 2.01 CM	T15REG 397.9 DEG.C
TWOCLR 58.16 DEG.C	4 EXTEFF 21.6 %	PHASE ANGLES	DISPST 1.96 CM	T16REG 399.7 DEG.C
	5 TAVHTR 600.2 DEG.C	PADISP 58.7 DEG.		T17REG 387.7 DEG.C
	6 INTEFF 27.8 %	PAPRES -16.7 DEG.		T18REG 382.7 DEG.C
	8 AMPS 1088. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 246.2 DEG.C
	9 QDISPG 3. WATTS	FREQ 30.1 HZ	PAMPC 886. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.15 CM	
	11 QREG1 111. WATTS		PISTCP 2.37 CM	
	12 QREG2 114. WATTS		PDYNDB***** KPA	
	13 QREG3 35. WATTS		PDLCLR 3.90 KPA	
	14 PFP 998.3 KPA/CM		PDLREG 39.37 KPA	
	15 PFD 303.3 KPA/CM		PDLDIS 71.29 KPA	
	16 NBEALE 0.00512			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 13:55:22.27 RDG 1198

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.01 L/MIN	POWER IN AMPS1 633. AMPS AMPS2 516. AMPS VOLTG 2.51 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7273. KPA MEANBP 6991. KPA MEANCP 7027. KPA	GAS TEMPERATURES TGEXP 564.3 DEG.C TGREGH 561.6 DEG.C TGREGC 118.7 DEG.C TGCOMP 66.2 DEG.C TGBOUN 39.9 DEG.C	SURFACE TEMPERATURES T01HTR 600.3 DEG.C T02HTR 607.0 DEG.C T03HTR 613.9 DEG.C T04HTR 607.0 DEG.C T05HTR 618.5 DEG.C T06HTR 609.5 DEG.C T07HTR 608.2 DEG.C T08HTR 580.8 DEG.C T09HTR 612.3 DEG.C T10HTR 588.8 DEG.C T11HTR 577.9 DEG.C T12HTR 580.4 DEG.C
TWINDP 18.5 DEG.C TDLDP 2.06 DEG.C TWODPR 21.2 DEG.C				
HEAT TO COOLER FLOCLR 4.14 L/MIN TWINCL 53.8 DEG.C TDLCL 5.79 DEG.C TWOCLE 58.75 DEG.C	CALCULATED PARAMETERS 1 PWRIN 2882. WATTS 2 QCOOLR 1646. WATTS 3 QDSHPT 575. WATTS 4 EXTEFF 22.0 % 5 TAVHTR 600.4 DEG.C 6 INTEFF 27.9 % 8 AMPS 1149. AMPS 9 QDISPG 3. WATTS 10 QDISP 14. WATTS 11 QREG1 108. WATTS 12 QREG2 115. WATTS 13 QREG3 34. WATTS 14 PFP 993.0 KPA/CM 15 PFD 297.3 KPA/CM 16 NBEALE 0.00531	VIBRATION VX1HOR 0.5 CM/S VY1VER 7.2 CM/S PHASE ANGLES PADISP 58.8 DEG. PAPRES -15.8 DEG. ENGINE SPEED FREQ 30.1 HZ	REMOTE CALCULATIONS PWROUT 635. WATTS INDPWR 687. WATTS PISTST 2.21 CM DISPST 2.08 CM DYNAMIC CALCULATIONS PAMPC 974. KPA DISPCP 2.15 CM PISTCP 2.39 CM PDYNDB***** KPA PDLCLR 4.41 KPA PDLREG 44.12 KPA PDLDIS 78.06 KPA	T13REG***** DEG.C T14REG 524.2 DEG.C T15REG 400.3 DEG.C T16REG 401.8 DEG.C T17REG 389.2 DEG.C T18REG 382.9 DEG.C T19REG 249.6 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 14:00:22.27 RDG 1199

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.01 L/MIN	POWER IN AMPS1 663. AMPS AMPS2 545. AMPS VOLTG 2.64 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7294. KPA MEANBP 6992. KPA MEANCP 7033. KPA	GAS TEMPERATURES TGEXP 561.2 DEG.C TGREGH 561.0 DEG.C TGREGC 121.9 DEG.C TGCOMP 69.7 DEG.C TGBOUN 41.7 DEG.C	SURFACE TEMPERATURES T01HTR 601.4 DEG.C T02HTR 607.2 DEG.C T03HTR 614.5 DEG.C T04HTR 606.8 DEG.C T05HTR 619.1 DEG.C T06HTR 610.8 DEG.C T07HTR 609.4 DEG.C T08HTR 580.6 DEG.C T09HTR 613.5 DEG.C T10HTR 586.9 DEG.C T11HTR 577.6 DEG.C T12HTR 577.9 DEG.C
TWINDP 18.6 DEG.C TDLDP 2.39 DEG.C TWODPR 21.7 DEG.C				
HEAT TO COOLER FLOCLR 4.15 L/MIN TWINCL 53.9 DEG.C TDLCL 6.70 DEG.C TWOCLE 59.60 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3188. WATTS 2 QCOOLR 1911. WATTS 3 QDSHPT 667. WATTS 4 EXTEFF 22.0 % 5 TAVHTR 600.5 DEG.C 6 INTEFF 26.9 % 8 AMPS 1208. AMPS 9 QDISPG 3. WATTS 10 QDISP 14. WATTS 11 QREG1 106. WATTS 12 QREG2 114. WATTS 13 QREG3 34. WATTS 14 PFP 1001.2 KPA/CM 15 PFD 291.6 KPA/CM 16 NBEALE 0.00540	VIBRATION VX1HOR 0.5 CM/S VY1VER 7.8 CM/S PHASE ANGLES PADISP 58.9 DEG. PAPRES -14.6 DEG. ENGINE SPEED FREQ 30.2 HZ	REMOTE CALCULATIONS PWROUT 702. WATTS INDPWR 759. WATTS PISTST 2.40 CM DISPST 2.17 CM DYNAMIC CALCULATIONS PAMPC 1073. KPA DISPCP 2.22 CM PISTCP 2.38 CM PDYNDB***** KPA PDLCLR 5.09 KPA PDLREG 48.53 KPA PDLDIS 84.51 KPA	T13REG***** DEG.C T14REG 523.7 DEG.C T15REG 402.1 DEG.C T16REG 405.0 DEG.C T17REG 392.3 DEG.C T18REG 383.3 DEG.C T19REG 252.8 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 14:05:40.27 RDG 1200

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.01 L/MIN	POWER IN AMPS1 694. AMPS AMPS2 569. AMPS VOLTG 2.77 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7311. KPA MEANBP 6986. KPA MEANCP 7031. KPA	GAS TEMPERATURES TGEXP 558.0 DEG.C TGREGH 558.6 DEG.C TGREGC 123.0 DEG.C TGCOMP 74.7 DEG.C TGBOUN 43.1 DEG.C	SURFACE TEMPERATURES T01HTR 600.3 DEG.C T02HTR 607.1 DEG.C T03HTR 614.6 DEG.C T04HTR 605.0 DEG.C T05HTR 619.5 DEG.C T06HTR 611.7 DEG.C T07HTR 610.3 DEG.C T08HTR 578.4 DEG.C T09HTR 612.6 DEG.C T10HTR 584.7 DEG.C T11HTR 575.1 DEG.C T12HTR 575.6 DEG.C
TWINDP 18.7 DEG.C TDLDP 2.63 DEG.C TWODPR 22.1 DEG.C				
HEAT TO COOLER FLOCLR 4.15 L/MIN TWINCL 54.0 DEG.C TDLCL 7.39 DEG.C TWOCLE 60.48 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3493. WATTS 2 QCOOLR 2108. WATTS 3 QDSHPT 734. WATTS 4 EXTEFF 22.2 % 5 TAVHTR 599.6 DEG.C 6 INTEFF 26.9 % 8 AMPS 1263. AMPS 9 QDISPG 3. WATTS 10 QDISP 14. WATTS 11 QREG1 103. WATTS 12 QREG2 114. WATTS 13 QREG3 34. WATTS 14 PFP 994.2 KPA/CM 15 PFD 280.7 KPA/CM 16 NBEALE 0.00551	VIBRATION VX1HOR 0.6 CM/S VY1VER 8.4 CM/S PHASE ANGLES PADISP 59.1 DEG. PAPRES -13.7 DEG. ENGINE SPEED FREQ 30.2 HZ	REMOTE CALCULATIONS PWROUT 777. WATTS INDPWR 837. WATTS PISTST 2.60 CM DISPST 2.29 CM DYNAMIC CALCULATIONS PAMPC 1161. KPA DISPCP 2.25 CM PISTCP 2.34 CM PDYNDB***** KPA PDLCLR 6.11 KPA PDLREG 53.29 KPA PDLDIS 91.98 KPA	T13REG***** DEG.C T14REG 522.7 DEG.C T15REG 402.6 DEG.C T16REG 402.4 DEG.C T17REG 388.9 DEG.C T18REG 382.4 DEG.C T19REG 255.2 DEG.C T03HED***** DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 14:10:19.27 RDG 1201

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.01 L/MIN	AMPS1 723. AMPS	PRESUP 7305. KPA	TGEXP 556.4 DEG.C	T01HTR 602.6 DEG.C
	AMPS2 598. AMPS			T02HTR 608.4 DEG.C
WINDP 18.9 DEG.C	VOLTG 2.89 VOLTS	MEANBP 6991. KPA	TGREGH 557.1 DEG.C	T03HTR 617.3 DEG.C
OLDP 2.99 DEG.C		MEANCP 7039. KPA	TGREGC 120.6 DEG.C	T04HTR 604.0 DEG.C
MODPR 22.5 DEG.C			TGCOMP 76.3 DEG.C	T05HTR 623.0 DEG.C
			TGBOUN 44.5 DEG.C	T06HTR 612.7 DEG.C
				T07HTR 611.5 DEG.C
				T08HTR 579.8 DEG.C
				T09HTR 612.4 DEG.C
				T10HTR 585.8 DEG.C
				T11HTR 574.0 DEG.C
				T12HTR 576.0 DEG.C
				T13REG***** DEG.C
				T14REG 518.9 DEG.C
				T15REG 389.9 DEG.C
				T16REG 385.9 DEG.C
				T17REG 377.6 DEG.C
				T18REG 372.5 DEG.C
				T19REG 245.5 DEG.C
				T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 14:15:34.27 RDG 1202

FLUID HELIUM BAROM 14.327 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.02 L/MIN	AMPS1 783. AMPS	PRESUP 7280. KPA	TGEXP 552.0 DEG.C	T01HTR 602.5 DEG.C
	AMPS2 647. AMPS			T02HTR 612.2 DEG.C
WINDP 19.0 DEG.C	VOLTG 3.13 VOLTS	MEANBP 6995. KPA	TGREGH 546.6 DEG.C	T03HTR 617.6 DEG.C
OLDP 3.29 DEG.C		MEANCP 7050. KPA	TGREGC 115.6 DEG.C	T04HTR 603.5 DEG.C
MODPR 22.9 DEG.C			TGCOMP 88.7 DEG.C	T05HTR 624.6 DEG.C
			TGBOUN 46.0 DEG.C	T06HTR 614.3 DEG.C
				T07HTR 612.8 DEG.C
				T08HTR 578.2 DEG.C
				T09HTR 610.6 DEG.C
				T10HTR 586.3 DEG.C
				T11HTR 569.6 DEG.C
				T12HTR 577.0 DEG.C
				T13REG***** DEG.C
				T14REG 496.1 DEG.C
				T15REG 341.2 DEG.C
				T16REG 338.3 DEG.C
				T17REG 331.2 DEG.C
				T18REG 322.7 DEG.C
				T19REG 211.5 DEG.C
				T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:34:29.52 RDG 1353

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.44 L/MIN	AMPS1 553. AMPS	PRESUP 7180. KPA	TGEXP 570.0 DEG.C	T01HTR 595.9 DEG.C
	AMPS2 485. AMPS			T02HTR 614.9 DEG.C
WINDP 18.9 DEG.C	VOLTG 2.24 VOLTS	MEANBP 7016. KPA	TGREGH 554.2 DEG.C	T03HTR 605.8 DEG.C
OLDP 1.14 DEG.C		MEANCP 7044. KPA	TGREGC 90.3 DEG.C	T04HTR 611.3 DEG.C
MODPR 20.7 DEG.C			TGCOMP 41.5 DEG.C	T05HTR 611.4 DEG.C
			TGBOUN 33.6 DEG.C	T06HTR 607.4 DEG.C
				T07HTR 606.7 DEG.C
				T08HTR 577.0 DEG.C
				T09HTR 607.6 DEG.C
				T10HTR 592.4 DEG.C
				T11HTR 576.1 DEG.C
				T12HTR 588.5 DEG.C
				T13REG***** DEG.C
				T14REG 512.8 DEG.C
				T15REG 379.1 DEG.C
				T16REG 384.1 DEG.C
				T17REG 369.9 DEG.C
				T18REG 354.5 DEG.C
				T19REG 221.1 DEG.C
				T03HED***** DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:39:20.52 RDG 1354

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP	4.42 L/MIN	AMPS1 589. AMPS	PRESUP 7176. KPA	TGEXP 567.5 DEG.C	T01HTR 599.0 DEG.C
		AMPS2 517. AMPS			T02HTR 614.7 DEG.C
TWINDP	18.8 DEG.C	VOLTG 2.39 VOLTS	MEANBP 7013. KPA	TGREGH 555.3 DEG.C	T03HTR 608.7 DEG.C
TDLDP	1.29 DEG.C		MEANCP 7046. KPA	TGREGC 92.9 DEG.C	T04HTR 610.9 DEG.C
TWODPR	20.9 DEG.C			TGCOMP 44.3 DEG.C	T05HTR 613.6 DEG.C
				TGBOUN 33.8 DEG.C	T06HTR 608.5 DEG.C
					T07HTR 607.7 DEG.C
					T08HTR 578.9 DEG.C
					T09HTR 609.5 DEG.C
					T10HTR 592.5 DEG.C
					T11HTR 576.9 DEG.C
					T12HTR 586.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR	4.19 L/MIN	1 PWRIN 2649. WATTS	VX1HOR 0.5 CM/S	PWROUT 570. WATTS	
TWINCL	25.0 DEG.C	2 QCOOLR 1999. WATTS	VY1VER 7.0 CM/S	INDPWR 590. WATTS	
TDLCL	6.88 DEG.C	3 QDSHPT 396. WATTS		PISTST 2.00 CM	
TWOCLR	30.79 DEG.C	4 EXTEFF 21.5 %	PHASE ANGLES	DISPST 1.58 CM	
		5 TAVHTR 600.6 DEG.C	PADISP 76.9 DEG.		T13REG***** DEG.C
		6 INTEFF 22.2 %	PAPRES -13.8 DEG.		T14REG 513.0 DEG.C
		8 AMPS 1106. AMPS	ENGINE SPEED	PAMPC 994. KPA	T15REG 381.2 DEG.C
		9 QDISPG 3. WATTS	FREQ 31.5 HZ	DISPCP 2.56 CM	T16REG 385.9 DEG.C
		10 QDISP 15. WATTS		PISTCP 2.29 CM	T17REG 375.8 DEG.C
		11 QREG1 109. WATTS		PDYNDB 938. KPA	T18REG 359.7 DEG.C
		12 QREG2 124. WATTS		PDICLR 5.09 KPA	T19REG 224.8 DEG.C
		13 QREG3 36. WATTS		PDLREG 39.03 KPA	
		14 PFP 1020.2 KPA/CM		PDLDIS 70.26 KPA	T03HED***** DEG.C
		15 PFD 309.4 KPA/CM			
		16 NBEALE 0.00503			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:44:41.52 RDG 1355

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP	4.42 L/MIN	AMPS1 630. AMPS	PRESUP 7179. KPA	TGEXP 565.0 DEG.C	T01HTR 599.9 DEG.C
		AMPS2 546. AMPS			T02HTR 614.3 DEG.C
TWINDP	18.8 DEG.C	VOLTG 2.54 VOLTS	MEANBP 7002. KPA	TGREGH 554.0 DEG.C	T03HTR 610.2 DEG.C
TDLDP	1.56 DEG.C		MEANCP 7038. KPA	TGREGC 94.7 DEG.C	T04HTR 609.3 DEG.C
TWODPR	21.1 DEG.C			TGCOMP 46.4 DEG.C	T05HTR 615.5 DEG.C
				TGBOUN 34.4 DEG.C	T06HTR 609.3 DEG.C
					T07HTR 608.5 DEG.C
					T08HTR 578.5 DEG.C
					T09HTR 609.6 DEG.C
					T10HTR 591.2 DEG.C
					T11HTR 575.1 DEG.C
					T12HTR 584.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR	4.21 L/MIN	1 PWRIN 2988. WATTS	VX1HOR 0.6 CM/S	PWROUT 665. WATTS	
TWINCL	25.0 DEG.C	2 QCOOLR 2231. WATTS	VY1VER 7.6 CM/S	INDPWR 688. WATTS	
TDLCL	7.64 DEG.C	3 QDSHPT 478. WATTS		PISTST 2.20 CM	
TWOCLR	31.41 DEG.C	4 EXTEFF 22.2 %	PHASE ANGLES	DISPST 1.70 CM	
		5 TAVHTR 600.5 DEG.C	PADISP 78.3 DEG.		T13REG***** DEG.C
		6 INTEFF 23.0 %	PAPRES -13.3 DEG.		T14REG 514.0 DEG.C
		8 AMPS 1175. AMPS	ENGINE SPEED	PAMPC 1107. KPA	T15REG 385.7 DEG.C
		9 QDISPG 3. WATTS	FREQ 31.5 HZ	DISPCP 2.66 CM	T16REG 386.1 DEG.C
		10 QDISP 15. WATTS		PISTCP 2.31 CM	T17REG 377.2 DEG.C
		11 QREG1 108. WATTS		PDYNDB 1013. KPA	T18REG 362.9 DEG.C
		12 QREG2 123. WATTS		PDICLR 6.11 KPA	T19REG 230.2 DEG.C
		13 QREG3 36. WATTS		PDLREG 47.52 KPA	
		14 PFP 1026.0 KPA/CM		PDLDIS 84.17 KPA	T03HED***** DEG.C
		15 PFD 306.8 KPA/CM			
		16 NBEALE 0.00532			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:49:20.52 RDG 1356

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING		POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP	4.41 L/MIN	AMPS1 651. AMPS	PRESUP 7174. KPA	TGEXP 562.0 DEG.C	T01HTR 600.8 DEG.C
		AMPS2 578. AMPS			T02HTR 613.9 DEG.C
TWINDP	18.8 DEG.C	VOLTG 2.66 VOLTS	MEANBP 6993. KPA	TGREGH 552.7 DEG.C	T03HTR 611.7 DEG.C
TDLDP	1.83 DEG.C		MEANCP 7034. KPA	TGREGC 96.7 DEG.C	T04HTR 608.2 DEG.C
TWODPR	21.2 DEG.C			TGCOMP 49.0 DEG.C	T05HTR 617.1 DEG.C
				TGBOUN 35.1 DEG.C	T06HTR 610.3 DEG.C
					T07HTR 609.5 DEG.C
					T08HTR 578.6 DEG.C
					T09HTR 610.0 DEG.C
					T10HTR 590.1 DEG.C
					T11HTR 574.1 DEG.C
					T12HTR 583.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR	4.20 L/MIN	1 PWRIN 3269. WATTS	VX1HOR 0.6 CM/S	PWROUT 737. WATTS	
TWINCL	25.1 DEG.C	2 QCOOLR 2347. WATTS	VY1VER 8.3 CM/S	INDPWR 758. WATTS	
TDLCL	8.05 DEG.C	3 QDSHPT 562. WATTS		PISTST 2.41 CM	
TWOCLR	32.04 DEG.C	4 EXTEFF 22.5 %	PHASE ANGLES	DISPST 1.79 CM	
		5 TAVHTR 600.6 DEG.C	PADISP 77.6 DEG.		T13REG***** DEG.C
		6 INTEFF 23.9 %	PAPRES -12.3 DEG.		T14REG 513.8 DEG.C
		8 AMPS 1229. AMPS	ENGINE SPEED	PAMPC 1191. KPA	T15REG 386.6 DEG.C
		9 QDISPG 3. WATTS	FREQ 31.3 HZ	DISPCP 2.70 CM	T16REG 386.8 DEG.C
		10 QDISP 15. WATTS		PISTCP 2.32 CM	T17REG 378.8 DEG.C
		11 QREG1 108. WATTS		PDYNDB 1113. KPA	T18REG 364.8 DEG.C
		12 QREG2 121. WATTS		PDICLR 6.96 KPA	T19REG 232.2 DEG.C
		13 QREG3 35. WATTS		PDLREG 50.57 KPA	
		14 PFP 1013.0 KPA/CM		PDLDIS 88.92 KPA	T03HED***** DEG.C
		15 PFD 289.6 KPA/CM			
		16 NBEALE 0.00544			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:54:20.52 RDG 1357

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.38 L/MIN	AMPS1 680. AMPS	PRESUP 7163. KPA	TGEXP 559.5 DEG.C	T01HTR 601.7 DEG.C
	AMPS2 605. AMPS			T02HTR 614.3 DEG.C
TWINDP 18.8 DEG.C	VOLTG 2.78 VOLTS	MEANBP 6992. KPA	TGREGH 551.9 DEG.C	T03HTR 612.7 DEG.C
TDLDP 1.99 DEG.C		MEANCNP 7033. KPA	TGREGC 98.6 DEG.C	T04HTR 608.1 DEG.C
TWODPR 21.4 DEG.C			TGCOMP 51.6 DEG.C	T05HTR 618.4 DEG.C
			TGBOUN 35.8 DEG.C	T06HTR 611.3 DEG.C
				T07HTR 610.5 DEG.C
				T08HTR 578.3 DEG.C
				T09HTR 610.9 DEG.C
				T10HTR 589.1 DEG.C
				T11HTR 573.5 DEG.C
				T12HTR 582.0 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.22 L/MIN	1 PWRIN 3575. WATTS	VX1HOR 0.7 CM/S	PWROUT 798. WATTS	T13REG***** DEG.C
TWINCL 25.0 DEG.C	2 QCOOLR 2563. WATTS	VY1VER 8.9 CM/S	INDPWR 818. WATTS	T14REG 514.2 DEG.C
TDLCL 8.75 DEG.C	3 QDSHPT 605. WATTS		PISTST 2.61 CM	T15REG 390.1 DEG.C
TWOCLR 32.75 DEG.C	4 EXTEFF 22.3 %	PHASE ANGLES	DISPST 1.88 CM	T16REG 386.3 DEG.C
	5 TAVHTR 600.9 DEG.C	PADISP 77.7 DEG.		T17REG 378.5 DEG.C
	6 INTEFF 23.7 %	PAPRES -11.3 DEG.		T18REG 364.0 DEG.C
	8 AMPS 1285. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 237.4 DEG.C
	9 QDISPG 3. WATTS	FREQ 31.3 HZ	PAMPC 1299. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.65 CM	
	11 QREG1 103. WATTS		PISTCP 2.36 CM	
	12 QREG2 122. WATTS		PDYNDB 1163. KPA	
	13 QREG3 35. WATTS		PDLCLR 7.81 KPA	
	14 PFP 1020.3 KPA/CM		PDLREG 57.36 KPA	
	15 PFD 277.8 KPA/CM		PDLDIS 99.10 KPA	
	16 NBEALE 0.00545			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:59:38.52 RDG 1358

FLUID HELIUM BAROM 14.474 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.36 L/MIN	AMPS1 711. AMPS	PRESUP 7175. KPA	TGEXP 556.8 DEG.C	T01HTR 601.8 DEG.C
	AMPS2 630. AMPS			T02HTR 614.1 DEG.C
TWINDP 18.7 DEG.C	VOLTG 2.90 VOLTS	MEANBP 7019. KPA	TGREGH 550.4 DEG.C	T03HTR 613.3 DEG.C
TDLDP 2.25 DEG.C		MEANCNP 7064. KPA	TGREGC 101.1 DEG.C	T04HTR 607.2 DEG.C
TWODPR 21.6 DEG.C			TGCOMP 55.2 DEG.C	T05HTR 619.1 DEG.C
			TGBOUN 37.0 DEG.C	T06HTR 612.3 DEG.C
				T07HTR 611.6 DEG.C
				T08HTR 577.5 DEG.C
				T09HTR 611.3 DEG.C
				T10HTR 587.4 DEG.C
				T11HTR 572.3 DEG.C
				T12HTR 579.8 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.22 L/MIN	1 PWRIN 3891. WATTS	VX1HOR 0.7 CM/S	PWROUT 870. WATTS	T13REG***** DEG.C
TWINCL 25.0 DEG.C	2 QCOOLR 2795. WATTS	VY1VER 9.5 CM/S	INDPWR 887. WATTS	T14REG 513.4 DEG.C
TDLCL 9.55 DEG.C	3 QDSHPT 684. WATTS		PISTST 2.79 CM	T15REG 390.1 DEG.C
TWOCLR 33.62 DEG.C	4 EXTEFF 22.4 %	PHASE ANGLES	DISPST 1.96 CM	T16REG 387.2 DEG.C
	5 TAVHTR 600.6 DEG.C	PADISP 78.1 DEG.		T17REG 378.5 DEG.C
	6 INTEFF 23.7 %	PAPRES -10.9 DEG.		T18REG 363.7 DEG.C
	8 AMPS 1341. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 239.1 DEG.C
	9 QDISPG 3. WATTS	FREQ 31.3 HZ	PAMPC 1383. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.74 CM	
	11 QREG1 101. WATTS		PISTCP 2.33 CM	
	12 QREG2 121. WATTS		PDYNDB 1250. KPA	
	13 QREG3 35. WATTS		PDLCLR 8.65 KPA	
	14 PFP 1011.9 KPA/CM		PDLREG 61.43 KPA	
	15 PFD 271.3 KPA/CM		PDLDIS 104.87 KPA	
	16 NBEALE 0.00553			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 13:52:29.52 RDG 1359

FLUID HELIUM BAROM 14.489 PSI REGENERATOR 1 DISPLACER 2 STANDARD PISTON

T TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.04 L/MIN	AMPS1 765. AMPS	PRESUP 7172. KPA	TGEXP 552.7 DEG.C	T01HTR 602.9 DEG.C
	AMPS2 648. AMPS			T02HTR 612.8 DEG.C
WINDP 18.8 DEG.C	VOLTG 3.05 VOLTS	MEANBP 7019. KPA	TGREGH 548.5 DEG.C	T03HTR 615.7 DEG.C
DLDP 2.33 DEG.C		MEANCNP 7072. KPA	TGREGC 98.1 DEG.C	T04HTR 604.7 DEG.C
WODPR 21.7 DEG.C			TGCOMP 56.8 DEG.C	T05HTR 621.4 DEG.C
			TGBOUN 34.0 DEG.C	T06HTR 611.1 DEG.C
				T07HTR 610.6 DEG.C
				T08HTR 577.6 DEG.C
				T09HTR 609.8 DEG.C
				T10HTR 587.0 DEG.C
				T11HTR 570.0 DEG.C
				T12HTR 577.7 DEG.C
T TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
LOCLR 4.22 L/MIN	1 PWRIN 4315. WATTS	VX1HOR 0.8 CM/S	PWROUT 928. WATTS	T13REG***** DEG.C
WINCL 24.8 DEG.C	2 QCOOLR 3040. WATTS	VY1VER 10.3 CM/S	INDPWR 945. WATTS	T14REG 511.7 DEG.C
DLCL 10.38 DEG.C	3 QDSHPT 656. WATTS		PISTST 3.00 CM	T15REG 389.4 DEG.C
WOCLR 34.04 DEG.C	4 EXTEFF 21.5 %	PHASE ANGLES	DISPST 2.05 CM	T16REG 383.5 DEG.C
	5 TAVHTR 600.1 DEG.C	PADISP 78.0 DEG.		T17REG 380.1 DEG.C
	6 INTEFF 23.4 %	PAPRES -10.0 DEG.		T18REG 371.1 DEG.C
	8 AMPS 1413. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	T19REG 240.3 DEG.C
	9 QDISPG 3. WATTS	FREQ 31.2 HZ	PAMPC 1506. KPA	T03HED***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.82 CM	
	11 QREG1 106. WATTS		PISTCP 2.34 CM	
	12 QREG2 114. WATTS		PDYNDB 1313. KPA	
	13 QREG3 35. WATTS		PDLCLR 11.20 KPA	
	14 PFP 1026.1 KPA/CM		PDLREG 69.58 KPA	
	15 PFD 260.8 KPA/CM		PDLDIS 117.77 KPA	
	16 NBEALE 0.00550			

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 13:58:08.52 RDG 1360

FLUID HELIUM		BAROM 14.489 PSI	REGENERATOR 1	DISPLACER 2	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.06 L/MIN	AMPS1	785. AMPS	PRESUP	7033. KPA	TGEXP	551.0 DEG.C
		AMPS2	675. AMPS				
TWINDP	18.8 DEG.C	VOLTG	3.16 VOLTS	MEANBP	6982. KPA	TGREGH	547.3 DEG.C
TDLDP	2.74 DEG.C			MEANCP	7038. KPA	TGREGC	100.3 DEG.C
TWODPR	22.0 DEG.C					TGCOMP	60.3 DEG.C
						TGBOUN	36.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.22 L/MIN	1 PWRIN	4609. WATTS	VXIHOR	0.8 CM/S	PWROUT	987. WATTS
TWINCL	24.7 DEG.C	2 QCOOLR	3285. WATTS	VYIVER	11.0 CM/S	INDPNR	1000. WATTS
TDLCL	11.22 DEG.C	3 QDSHPT	774. WATTS			PISTST	3.21 CM
TWOCLR	34.76 DEG.C	4 EXTEFF	21.4 %	PHASE ANGLES		DISPST	2.13 CM
		5 TAVHTR	600.0 DEG.C	PADISP	77.9 DEG.		
		6 INTEFF	23.1 %	PAPRES	-9.2 DEG.	DYNAMIC CALCULATIONS	
		8 AMPS	1460. AMPS	ENGINE SPEED		PAMPC	1609. KPA
		9 QDISPG	3. WATTS	FREQ	31.1 HZ	DISPCP	2.91 CM
		10 QDISP	14. WATTS			PISTCP	2.36 CM
		11 QREG1	104. WATTS			PDYNDB	1350. KPA
		12 QREG2	114. WATTS			PDLCLR	11.71 KPA
		13 QREG3	35. WATTS			PDLREG	77.04 KPA
		14 PFP	1024.3 KPA/CM			PDLDIS	128.29 KPA
		15 PFD	247.1 KPA/CM				
		16 NBEALE	0.00552				
							T13REG***** DEG.C
							T14REG 511.8 DEG.C
							T15REG 390.4 DEG.C
							T16REG 383.8 DEG.C
							T17REG 380.3 DEG.C
							T18REG 371.0 DEG.C
							T19REG 242.4 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:36:18.08 RDG 1548

FLUID NITROGEN		BAROM 14.362 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.01 L/MIN	AMPS1	477. AMPS	PRESUP	7146. KPA	TGEXP	569.6 DEG.C
		AMPS2	373. AMPS				
TWINDP	19.7 DEG.C	VOLTG	1.88 VOLTS	MEANBP	7003. KPA	TGREGH	552.4 DEG.C
TDLDP	0.73 DEG.C			MEANCP	7041. KPA	TGREGC	103.0 DEG.C
TWODPR	20.6 DEG.C					TGCOMP	35.5 DEG.C
						TGBOUN	25.2 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.32 L/MIN	1 PWRIN	1600. WATTS	VXIHOR	0.4 CM/S	PWROUT	180. WATTS
TWINCL	25.0 DEG.C	2 QCOOLR	1426. WATTS	VYIVER	4.8 CM/S	INDPNR	194. WATTS
TDLCL	4.76 DEG.C	3 QDSHPT	203. WATTS			PISTST	1.78 CM
TWOCLR	28.40 DEG.C	4 EXTEFF	11.3 %	PHASE ANGLES		DISPST	1.19 CM
		5 TAVHTR	600.3 DEG.C	PADISP	39.3 DEG.		
		6 INTEFF	11.2 %	PAPRES	-7.0 DEG.	DYNAMIC CALCULATIONS	
		8 AMPS	850. AMPS	ENGINE SPEED		PAMPC	753. KPA
		9 QDISPG	0. WATTS	FREQ	27.9 HZ	DISPCP	2.18 CM
		10 QDISP	15. WATTS			PISTCP	2.28 CM
		11 QREG1	119. WATTS			PDYNDB*****	KPA
		12 QREG2	97. WATTS			PDLCLR	8.49 KPA
		13 QREG3	35. WATTS			PDLREG	40.73 KPA
		14 PFP	965.8 KPA/CM			PDLDIS	109.29 KPA
		15 PFD	245.8 KPA/CM				
		16 NBEALE	0.00202				
							T13REG***** DEG.C
							T14REG 489.5 DEG.C
							T15REG 372.9 DEG.C
							T16REG 384.9 DEG.C
							T17REG 385.7 DEG.C
							T18REG 369.5 DEG.C
							T19REG 223.2 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:41:21.08 RDG 1549

FLUID NITROGEN		BAROM 14.357 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.01 L/MIN	AMPS1	502. AMPS	PRESUP	7143. KPA	TGEXP	570.0 DEG.C
		AMPS2	389. AMPS				
TWINDP	19.5 DEG.C	VOLTG	1.96 VOLTS	MEANBP	7008. KPA	TGREGH	552.4 DEG.C
TDLDP	0.83 DEG.C			MEANCP	7047. KPA	TGREGC	104.2 DEG.C
TWODPR	20.5 DEG.C					TGCOMP	36.7 DEG.C
						TGBOUN	25.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.31 L/MIN	1 PWRIN	1749. WATTS	VXIHOR	0.5 CM/S	PWROUT	212. WATTS
TWINCL	25.0 DEG.C	2 QCOOLR	1545. WATTS	VYIVER	5.4 CM/S	INDPNR	228. WATTS
TDLCL	5.16 DEG.C	3 QDSHPT	232. WATTS			PISTST	1.99 CM
TWOCLR	28.78 DEG.C	4 EXTEFF	12.1 %	PHASE ANGLES		DISPST	1.29 CM
		5 TAVHTR	599.8 DEG.C	PADISP	38.6 DEG.		
		6 INTEFF	12.1 %	PAPRES	-6.6 DEG.	DYNAMIC CALCULATIONS	
		8 AMPS	891. AMPS	ENGINE SPEED		PAMPC	837. KPA
		9 QDISPG	0. WATTS	FREQ	27.9 HZ	DISPCP	2.22 CM
		10 QDISP	15. WATTS			PISTCP	2.32 CM
		11 QREG1	116. WATTS			PDYNDB*****	KPA
		12 QREG2	99. WATTS			PDLCLR	9.84 KPA
		13 QREG3	35. WATTS			PDLREG	46.16 KPA
		14 PFP	958.7 KPA/CM			PDLDIS	121.51 KPA
		15 PFD	238.3 KPA/CM				
		16 NBEALE	0.00213				
							T13REG***** DEG.C
							T14REG 492.6 DEG.C
							T15REG 376.7 DEG.C
							T16REG 386.5 DEG.C
							T17REG 386.5 DEG.C
							T18REG 370.2 DEG.C
							T19REG 226.9 DEG.C
							T03HED***** DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:46:18.08 RDG 1550

FLUID NITROGEN BAROM 14.357 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.03 L/MIN	AMPS1 528. AMPS	PRESUP 7157. KPA	TGEXP 569.4 DEG.C	T01HTR 601.3 DEG.C
	AMPS2 406. AMPS			T02HTR 615.8 DEG.C
TWINDP 19.3 DEG.C	VOLTG 2.05 VOLTS	MEANBP 7012. KPA	TGREGH 553.1 DEG.C	T03HTR 601.5 DEG.C
TDLDP 0.91 DEG.C		MEANCP 7052. KPA	TGREGC 106.5 DEG.C	T04HTR 618.7 DEG.C
TWODPR 20.4 DEG.C			TGCOMP 38.2 DEG.C	T05HTR 600.1 DEG.C
			TGBOUN 26.1 DEG.C	T06HTR 603.8 DEG.C
				T07HTR 603.2 DEG.C
				T08HTR 584.4 DEG.C
				T09HTR 607.5 DEG.C
				T10HTR 597.6 DEG.C
				T11HTR 581.0 DEG.C
				T12HTR 585.4 DEG.C

AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
FLOCLR 4.33 L/MIN	1 PWRIN 1916. WATTS	VXIHOR 0.5 CM/S	PWROUT 240. WATTS
TWINCL 25.0 DEG.C	2 QCOOLR 1714. WATTS	VYIVER 6.1 CM/S	INDPWR 258. WATTS
TDLCL 5.70 DEG.C	3 QDSHPT 256. WATTS		PISTST 2.18 CM
TWOCLR 29.17 DEG.C	4 EXTEFF 12.5 %	PHASE ANGLES	DISPST 1.39 CM
	5 TAVHTR 600.0 DEG.C	PADISP 37.9 DEG.	
	6 INTEFF 12.3 %	PAPRES -6.3 DEG.	
	8 AMPS 934. AMPS		DYNAMIC CALCULATIONS
	9 QDISPG 0. WATTS	ENGINE SPEED	PAMPC 925. KPA
	10 QDISP 15. WATTS	FREQ 27.9 HZ	DISPCP 2.31 CM
	11 QREG1 115. WATTS		PISTCP 2.40 CM
	12 QREG2 100. WATTS		PDYNDB***** KPA
	13 QREG3 35. WATTS		PDLCLR 11.54 KPA
	14 PFP 962.8 KPA/CM		PDLREG 52.27 KPA
	15 PFD 239.0 KPA/CM		PDLDIS134.06 KPA
	16 NBEALE 0.00219		T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:51:18.08 RDG 1551

FLUID NITROGEN BAROM 14.357 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.01 L/MIN	AMPS1 546. AMPS	PRESUP 7160. KPA	TGEXP 568.4 DEG.C	T01HTR 601.8 DEG.C
	AMPS2 422. AMPS			T02HTR 615.0 DEG.C
TWINDP 19.2 DEG.C	VOLTG 2.13 VOLTS	MEANBP 7010. KPA	TGREGH 552.9 DEG.C	T03HTR 602.6 DEG.C
TDLDP 0.99 DEG.C		MEANCP 7049. KPA	TGREGC 108.3 DEG.C	T04HTR 617.5 DEG.C
TWODPR 20.4 DEG.C			TGCOMP 39.6 DEG.C	T05HTR 601.1 DEG.C
			TGBOUN 26.6 DEG.C	T06HTR 603.6 DEG.C
				T07HTR 603.0 DEG.C
				T08HTR 584.4 DEG.C
				T09HTR 607.4 DEG.C
				T10HTR 597.0 DEG.C
				T11HTR 580.0 DEG.C
				T12HTR 584.0 DEG.C

AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
FLOCLR 4.34 L/MIN	1 PWRIN 2061. WATTS	VXIHOR 0.6 CM/S	PWROUT 268. WATTS
TWINCL 25.1 DEG.C	2 QCOOLR 1781. WATTS	VYIVER 6.8 CM/S	INDPWR 289. WATTS
TDLCL 5.92 DEG.C	3 QDSHPT 277. WATTS		PISTST 2.38 CM
TWOCLR 29.56 DEG.C	4 EXTEFF 13.0 %	PHASE ANGLES	DISPST 1.48 CM
	5 TAVHTR 599.8 DEG.C	PADISP 37.4 DEG.	
	6 INTEFF 13.1 %	PAPRES -6.0 DEG.	
	8 AMPS 968. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS
	9 QDISPG 0. WATTS	FREQ 28.0 HZ	PAMPC 1009. KPA
	10 QDISP 15. WATTS		DISPCP 2.37 CM
	11 QREG1 114. WATTS		PISTCP 2.42 CM
	12 QREG2 99. WATTS		PDYNDB***** KPA
	13 QREG3 34. WATTS		PDLCLR 12.90 KPA
	14 PFP 957.7 KPA/CM		PDLREG 57.02 KPA
	15 PFD 232.4 KPA/CM		PDLDIS140.51 KPA
	16 NBEALE 0.00224		T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:56:18.08 RDG 1552

FLUID NITROGEN BAROM 14.357 PSI REGENERATOR 1 DISPLACER 1 STANDARD PISTON

T TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.00 L/MIN	AMPS1 565. AMPS	PRESUP 7160. KPA	TGEXP 566.9 DEG.C	T01HTR 602.2 DEG.C
	AMPS2 437. AMPS			T02HTR 615.1 DEG.C
WINDP 19.2 DEG.C	VOLTG 2.20 VOLTS	MEANBP 7010. KPA	TGREGH 552.1 DEG.C	T03HTR 604.1 DEG.C
LDLP 1.10 DEG.C		MEANCP 7058. KPA	TGREGC 109.5 DEG.C	T04HTR 616.8 DEG.C
WODPR 20.4 DEG.C			TGCOMP 40.9 DEG.C	T05HTR 602.7 DEG.C
			TGBOUN 27.3 DEG.C	T06HTR 604.2 DEG.C
				T07HTR 603.7 DEG.C
				T08HTR 584.2 DEG.C
				T09HTR 607.4 DEG.C
				T10HTR 596.9 DEG.C
				T11HTR 579.0 DEG.C
				T12HTR 583.9 DEG.C

T TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS
LOCLR 4.35 L/MIN	1 PWRIN 2210. WATTS	VXIHOR 0.6 CM/S	PWROUT 292. WATTS
WINCL 25.2 DEG.C	2 QCOOLR 1948. WATTS	VYIVER 7.5 CM/S	INDPWR 314. WATTS
DLCL 6.46 DEG.C	3 QDSHPT 306. WATTS		PISTST 2.56 CM
WOCLR 29.99 DEG.C	4 EXTEFF 13.2 %	PHASE ANGLES	DISPST 1.56 CM
	5 TAVHTR 600.0 DEG.C	PADISP 36.5 DEG.	
	6 INTEFF 13.0 %	PAPRES -5.5 DEG.	
	8 AMPS 1002. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS
	9 QDISPG 0. WATTS	FREQ 28.0 HZ	PAMPC 1102. KPA
	10 QDISP 15. WATTS		DISPCP 2.47 CM
	11 QREG1 112. WATTS		PISTCP 2.41 CM
	12 QREG2 99. WATTS		PDYNDB***** KPA
	13 QREG3 34. WATTS		PDLCLR 14.93 KPA
	14 PFP 967.5 KPA/CM		PDLREG 61.09 KPA
	15 PFD 226.5 KPA/CM		PDLDIS154.77 KPA
	16 NBEALE 0.00226		T03HED***** DEG.C

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 10:54:18.64 RDG 1604

FLUID HELIUM		BAROM 14.386 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.58 L/MIN	AMPS1	531. AMPS	PRESUP	7243. KPA	TGEXP	568.2 DEG.C	T01HTR	599.2 DEG.C		
TWINDP	20.6 DEG.C	AMPS2	399. AMPS	MEANBP	6994. KPA	TGREGH	565.8 DEG.C	T02HTR	609.4 DEG.C		
TDLDP	1.22 DEG.C	VOLTG	2.05 VOLTS	MEANCP	7024. KPA	TGREGC	92.5 DEG.C	T03HTR	606.0 DEG.C		
TWODPR	21.7 DEG.C					TGCOMP	36.2 DEG.C	T04HTR	608.2 DEG.C		
						TGBOUN	27.3 DEG.C	T05HTR	609.2 DEG.C		
								T06HTR	608.3 DEG.C		
								T07HTR	607.5 DEG.C		
								T08HTR	581.7 DEG.C		
								T09HTR	609.0 DEG.C		
								T10HTR	586.6 DEG.C		
								T11HTR	582.4 DEG.C		
								T12HTR	581.5 DEG.C		
								T13REG*****	DEG.C		
								T14REG	519.1 DEG.C		
								T15REG	387.9 DEG.C		
								T16REG	394.2 DEG.C		
								T17REG	391.1 DEG.C		
								T18REG	375.7 DEG.C		
								T19REG	227.8 DEG.C		
								T03HED*****	DEG.C		

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 11:00:27.64 RDG 1605

FLUID HELIUM		BAROM 14.386 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.54 L/MIN	AMPS1	574. AMPS	PRESUP	7312. KPA	TGEXP	565.5 DEG.C	T01HTR	600.4 DEG.C		
TWINDP	20.1 DEG.C	AMPS2	423. AMPS	MEANBP	7009. KPA	TGREGH	565.7 DEG.C	T02HTR	608.3 DEG.C		
TDLDP	1.38 DEG.C	VOLTG	2.20 VOLTS	MEANCP	7045. KPA	TGREGC	96.4 DEG.C	T03HTR	607.8 DEG.C		
TWODPR	21.4 DEG.C					TGCOMP	38.4 DEG.C	T04HTR	608.0 DEG.C		
						TGBOUN	28.3 DEG.C	T05HTR	611.3 DEG.C		
								T06HTR	609.0 DEG.C		
								T07HTR	608.3 DEG.C		
								T08HTR	581.6 DEG.C		
								T09HTR	610.9 DEG.C		
								T10HTR	584.8 DEG.C		
								T11HTR	581.9 DEG.C		
								T12HTR	579.0 DEG.C		
								T13REG*****	DEG.C		
								T14REG	520.7 DEG.C		
								T15REG	393.0 DEG.C		
								T16REG	399.2 DEG.C		
								T17REG	395.0 DEG.C		
								T18REG	380.8 DEG.C		
								T19REG	234.1 DEG.C		
								T03HED*****	DEG.C		

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 11:04:30.64 RDG 1606

FLUID HELIUM		BAROM 14.381 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.50 L/MIN	AMPS1	599. AMPS	PRESUP	7318. KPA	TGEXP	563.8 DEG.C	T01HTR	604.0 DEG.C		
TWINDP	19.7 DEG.C	AMPS2	448. AMPS	MEANBP	6998. KPA	TGREGH	567.7 DEG.C	T02HTR	607.5 DEG.C		
TDLDP	1.57 DEG.C	VOLTG	2.30 VOLTS	MEANCP	7035. KPA	TGREGC	98.2 DEG.C	T03HTR	610.3 DEG.C		
TWODPR	21.2 DEG.C					TGCOMP	40.2 DEG.C	T04HTR	608.3 DEG.C		
						TGBOUN	29.1 DEG.C	T05HTR	613.3 DEG.C		
								T06HTR	609.6 DEG.C		
								T07HTR	608.9 DEG.C		
								T08HTR	584.1 DEG.C		
								T09HTR	613.5 DEG.C		
								T10HTR	584.4 DEG.C		
								T11HTR	583.4 DEG.C		
								T12HTR	576.9 DEG.C		
								T13REG*****	DEG.C		
								T14REG	521.0 DEG.C		
								T15REG	394.3 DEG.C		
								T16REG	401.6 DEG.C		
								T17REG	400.4 DEG.C		
								T18REG	386.0 DEG.C		
								T19REG	236.3 DEG.C		
								T03HED*****	DEG.C		

APPENDIX C. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 11:09:27.64 RDG 1607

FLUID HELIUM		BAROM 14.381 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
EAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.33 L/MIN	AMPS1	632. AMPS	PRESUP	7309. KPA	TGEXP	561.1 DEG.C	T01HTR	605.5 DEG.C		
		AMPS2	480. AMPS					T02HTR	606.9 DEG.C		
TWINDP	19.4 DEG.C	VOLTG	2.45 VOLTS	MEANBP	6991. KPA	TGREGH	567.0 DEG.C	T03HTR	612.8 DEG.C		
TDLDP	1.81 DEG.C			MEANCP	7031. KPA	TGREGC	100.5 DEG.C	T04HTR	607.3 DEG.C		
TWODPR	21.2 DEG.C					TGCOMP	42.8 DEG.C	T05HTR	615.8 DEG.C		
						TGBOUN	29.7 DEG.C	T06HTR	610.6 DEG.C		
								T07HTR	609.8 DEG.C		
								T08HTR	584.3 DEG.C		
								T09HTR	614.4 DEG.C		
								T10HTR	583.7 DEG.C		
								T11HTR	582.5 DEG.C		
								T12HTR	575.3 DEG.C		
EAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS					
FLOCLR	4.26 L/MIN	1 PWRIN	2721. WATTS	VX1HOR	0.4 CM/S	PWROUT	498. WATTS	T13REG*****	DEG.C		
TWINCL	25.4 DEG.C	2 QCOOLR	2338. WATTS	VY1VER	6.0 CM/S	INDPWR	504. WATTS	T14REG	521.6 DEG.C		
TDLCL	7.91 DEG.C	3 QDSHPT	546. WATTS			PISTST	2.39 CM	T15REG	395.8 DEG.C		
TWOCRL	31.12 DEG.C	4 EXTEFF	18.3 %	PHASE ANGLES		DISPST	1.78 CM	T16REG	401.9 DEG.C		
		5 TAVHTR	600.7 DEG.C	PADISP	40.5 DEG.	PDYNDB*****	KPA	T17REG	401.2 DEG.C		
		6 INTEFF	17.5 %	PAPRES	-8.7 DEG.	PDLCLR	3.90 KPA	T18REG	388.8 DEG.C		
		8 AMPS	1112. AMPS	ENGINE SPEED		PDLREG	36.66 KPA	T19REG	239.4 DEG.C		
		9 QDISPG	3. WATTS	FREQ	37.1 HZ	PDLDIS	71.27 KPA	T03HED*****	DEG.C		
		10 QDISP	15. WATTS								
		11 QREG1	121. WATTS								
		12 QREG2	108. WATTS								
		13 QREG3	36. WATTS								
		14 PFP	997.3 KPA/CM								
		15 PFD	267.4 KPA/CM								
		16 NBEALE	0.00312								

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 14:42:20.19 RDG 1608

FLUID HELIUM		BAROM 14.322 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
EAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.12 L/MIN	AMPS1	653. AMPS	PRESUP	7070. KPA	TGEXP	558.4 DEG.C	T01HTR	606.1 DEG.C		
		AMPS2	517. AMPS					T02HTR	606.3 DEG.C		
TWINDP	18.0 DEG.C	VOLTG	2.58 VOLTS	MEANBP	7016. KPA	TGREGH	566.1 DEG.C	T03HTR	614.3 DEG.C		
TDLDP	2.44 DEG.C			MEANCP	7063. KPA	TGREGC	102.3 DEG.C	T04HTR	606.5 DEG.C		
TWODPR	20.5 DEG.C					TGCOMP	46.1 DEG.C	T05HTR	617.5 DEG.C		
						TGBOUN	32.6 DEG.C	T06HTR	611.5 DEG.C		
								T07HTR	610.5 DEG.C		
								T08HTR	584.0 DEG.C		
								T09HTR	615.3 DEG.C		
								T10HTR	582.2 DEG.C		
								T11HTR	581.7 DEG.C		
								T12HTR	573.3 DEG.C		
EAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS					
FLOCLR	4.26 L/MIN	1 PWRIN	3018. WATTS	VX1HOR	0.4 CM/S	PWROUT	572. WATTS	T13REG***** DEG.C			
TWINCL	25.3 DEG.C	2 QCOOLR	2618. WATTS	VY1VER	6.0 CM/S	INDPWR	571. WATTS	T14REG 522.1 DEG.C			
TDLCL	8.86 DEG.C	3 QDSHPT	701. WATTS			PISTST	2.56 CM	T15REG 398.0 DEG.C			
TWOCRL	31.88 DEG.C	4 EXTEFF	19.0 %	PHASE ANGLES		DISPST	1.90 CM	T16REG 402.5 DEG.C			
		5 TAVHTR	600.8 DEG.C	PADISP	41.6 DEG.	PDYNDB*****	KPA	T17REG 402.6 DEG.C			
		6 INTEFF	17.9 %	PAPRES	-8.1 DEG.	PDLCLR	4.41 KPA	T18REG 390.7 DEG.C			
		8 AMPS	1170. AMPS	ENGINE SPEED		PDLREG	40.05 KPA	T19REG 243.5 DEG.C			
		9 QDISPG	3. WATTS	FREQ	37.4 HZ	PDLDIS	75.69 KPA	T03HED***** DEG.C			
		10 QDISP	15. WATTS								
		11 QREG1	119. WATTS								
		12 QREG2	107. WATTS								
		13 QREG3	36. WATTS								
		14 PFP	987.2 KPA/CM								
		15 PFD	245.5 KPA/CM								
		16 NBEALE	0.00331								

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 14:47:50.19 RDG 1609

FLUID HELIUM		BAROM 14.317 PSI		REGENERATOR 1		DISPLACER 1		LIGHT		PISTON	
EAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES			
FLODP	4.11 L/MIN	AMPS1	685. AMPS	PRESUP	7072. KPA	TGEXP	555.2 DEG.C	T01HTR	606.3 DEG.C		
		AMPS2	546. AMPS					T02HTR	606.0 DEG.C		
TWINDP	18.0 DEG.C	VOLTG	2.71 VOLTS	MEANBP	7017. KPA	TGREGH	563.8 DEG.C	T03HTR	615.3 DEG.C		
TDLDP	2.62 DEG.C			MEANCP	7068. KPA	TGREGC	104.2 DEG.C	T04HTR	605.2 DEG.C		
TWODPR	20.6 DEG.C					TGCOMP	48.4 DEG.C	T05HTR	619.0 DEG.C		
						TGBOUN	33.2 DEG.C	T06HTR	612.0 DEG.C		
								T07HTR	611.2 DEG.C		
								T08HTR	583.1 DEG.C		
								T09HTR	615.3 DEG.C		
								T10HTR	580.5 DEG.C		
								T11HTR	579.8 DEG.C		
								T12HTR	571.8 DEG.C		
EAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS					
FLOCLR	4.26 L/MIN	1 PWRIN	3342. WATTS	VX1HOR	0.4 CM/S	PWROUT	639. WATTS				
TWINCL	25.3 DEG.C	2 QCOOLR	2826. WATTS	VY1VER	6.5 CM/S	INDPWR	633. WATTS				
TDLCL	9.57 DEG.C	3 QDSHPT	749. WATTS			PISTST	2.77 CM				
TWOCRL	32.66 DEG.C	4 EXTEFF	19.1 %	PHASE ANGLES		DISPST	2.00 CM				
		5 TAVHTR	600.5 DEG.C	PADISP	42.0 DEG.						
		6 INTEFF	18.4 %	PAPRES	-7.6 DEG.	DYNAMIC CALCULATIONS					
		8 AMPS	1231. AMPS	ENGINE SPEED		PAMPC	1201. KPA	T13REG*****	DEG.C		
		9 QDISPG	3. WATTS	FREQ	37.6 HZ	DISPCP	2.34 CM	T14REG	521.9 DEG.C		
		10 QDISP	14. WATTS			PISTCP	2.37 CM	T15REG	399.0 DEG.C		
		11 QREG1	118. WATTS			PDYNDB*****	KPA	T16REG	401.8 DEG.C		
		12 QREG2	106. WATTS			PDLCLR	5.26 KPA	T17REG	402.7 DEG.C		
		13 QREG3	36. WATTS			PDLREG	44.12 KPA	T18REG	391.1 DEG.C		
		14 PFP	987.4 KPA/CM			PDLDIS	82.47 KPA	T19REG	245.7 DEG.C		
		15 PFD	238.4 KPA/CM					T03HED*****	DEG.C		
		16 NBEALE	0.00341								

APPENDIX C. - CONCLUDED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 14:52:20.19 RDG 1610

FLUID HELIUM BAROM 14.317 PSI REGENERATOR 1 DISPLACER 1 LIGHT PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.10 L/MIN	AMPS1 714. AMPS	PRESUP 7077. KPA	TGEXP 552.3 DEG.C	T01HTR 607.1 DEG.C
	AMPS2 574. AMPS			T02HTR 605.6 DEG.C
TWINDP 18.0 DEG.C	VOLTG 2.84 VOLTS	MEANBP 7018. KPA	TGREGH 561.9 DEG.C	T03HTR 617.6 DEG.C
TDLDP 2.88 DEG.C		MEANCP 7075. KPA	TGREGC 105.4 DEG.C	T04HTR 603.8 DEG.C
TWODPR 20.8 DEG.C			TGCOMP 51.3 DEG.C	T05HTR 621.5 DEG.C
			TGBOUN 33.9 DEG.C	T06HTR 612.5 DEG.C
				T07HTR 611.9 DEG.C
				T08HTR 582.9 DEG.C
				T09HTR 615.1 DEG.C
				T10HTR 580.3 DEG.C
				T11HTR 578.2 DEG.C
				T12HTR 570.8 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.27 L/MIN	1 PWRIN 3664. WATTS	VX1HOR 0.4 CM/S	PWROUT 705. WATTS	T13REG***** DEG.C
TWINCL 25.3 DEG.C	2 QCOOLR 3090. WATTS	VY1VER 7.0 CM/S	INDPWR 691. WATTS	T14REG 521.6 DEG.C
TDLCL 10.44 DEG.C	3 QDSHPT 823. WATTS		PISTST 2.99 CM	T15REG 398.3 DEG.C
TWOCLR 33.54 DEG.C	4 EXTEFF 19.2 %	PHASE ANGLES	DISPST 2.10 CM	T16REG 398.9 DEG.C
	5 TAVHTR 600.6 DEG.C	PADISP 42.4 DEG.		T17REG 401.3 DEG.C
	6 INTEFF 18.6 %	PAPRES -7.2 DEG.	DYNAMIC CALCULATIONS	T18REG 391.4 DEG.C
	8 AMPS 1289. AMPS		PAMPC 1314. KPA	T19REG 246.8 DEG.C
	9 QDISPG 3. WATTS	ENGINE SPEED	DISPCP 2.33 CM	
	10 QDISP 14. WATTS	FREQ 37.7 HZ	PISTCP 2.34 CM	
	11 QREG1 117. WATTS		PDYNDB***** KPA	
	12 QREG2 106. WATTS		PDLCLR 6.45 KPA	
	13 QREG3 35. WATTS		PDLREG 48.19 KPA	
	14 PFP 992.5 KPA/CM		PDLDIS 89.60 KPA	
	15 PFD 230.9 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00347			

APPENDIX D

Other Data Points of Interest

By the definition of the SVDS and the DVDS, data points with both the high power regenerator and the high power displacer were eliminated. Since these data points may be of great interest for computer code validation, they were presented in this appendix. Although all of the data points are available on a microfiche card that may be obtained from NASA Lewis, it was felt that it would be easier to access the data from an appendix.

The 69 data points included in this appendix are those that have any combination of two of the following three set conditions: 7.0 MPa pressure 25 °C cooler temperature, and 600 °C heater temperature. If viewed from the format of figure 27, these include the row of 25 °C and 7.0 MPa, 25 °C, and 600 °C, and also the pair of 600 °C and 7.0 MPa.

APPENDIX D. - OTHER DATA POINTS OF INTEREST.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 15:42:22.48 RDG 774

FLUID HELIUM		BAROM 14.411 PSI	REGENERATOR 2	DISPLACER 2	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.02 L/MIN	AMPS1	634. AMPS	PRESUP	7345. KPA	TGEXP	563.9 DEG.C
		AMPS2	499. AMPS				
TWINDP	17.0 DEG.C	VOLTG	2.50 VOLTS	MEANBP	6998. KPA	TGREGH	542.4 DEG.C
TDLDP	2.06 DEG.C			MEANCP	7049. KPA	TGREGC	115.9 DEG.C
TWODPR	19.9 DEG.C					TGCOMP	68.4 DEG.C
						TGBOUN	44.1 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	3.94 L/MIN	1 PWRIN	2831. WATTS	VXIHOR	0.2 CM/S	PWROUT	606. WATTS
TWINCL	53.7 DEG.C	2 QCOOLR	1783. WATTS	VYIVER	6.5 CM/S	INDPWR	644. WATTS
TDLCL	6.59 DEG.C	3 QDSHPT	577. WATTS			PISTST	1.81 CM
TWOCLR	58.57 DEG.C	4 EXTEFF	21.4 %	PHASE ANGLES		DISPST	1.80 CM
		5 TAVHTR	599.6 DEG.C	PADISP	84.4 DEG.		
		6 INTEFF	25.4 %	PAPRES	-17.5 DEG.		
		8 AMPS	1133. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	32.4 HZ	PAMPC	915. KPA
		10 QDISP	14. WATTS			DISPCP	2.54 CM
		11 QREG1	101. WATTS			PISTCP	2.38 CM
		12 QREG2	113. WATTS			PDYNDB	1138. KPA
		13 QREG3	33. WATTS			PDLCLR*****	KPA
		14 PFP	995.8 KPA/CM			PDLREG*****	KPA
		15 PFD	306.5 KPA/CM			PDLDIS*****	KPA
		16 NBEALE	0.00576				
							T01HTR 599.0 DEG.C
							T02HTR 609.0 DEG.C
							T03HTR 606.6 DEG.C
							T04HTR 597.5 DEG.C
							T05HTR 596.7 DEG.C
							T06HTR 600.7 DEG.C
							T07HTR 572.7 DEG.C
							T08HTR 605.0 DEG.C
							T09HTR 610.4 DEG.C
							T10HTR 597.6 DEG.C
							T11HTR 596.0 DEG.C
							T12HTR 604.5 DEG.C
							T13REG 548.3 DEG.C
							T14REG 510.8 DEG.C
							T15REG 385.7 DEG.C
							T16REG***** DEG.C
							T17REG 369.5 DEG.C
							T18REG 371.4 DEG.C
							T19REG 247.4 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 15:48:10.48 RDG 775

FLUID HELIUM		BAROM 14.416 PSI	REGENERATOR 2	DISPLACER 2	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.00 L/MIN	AMPS1	668. AMPS	PRESUP	7249. KPA	TGEXP	561.9 DEG.C
		AMPS2	527. AMPS				
TWINDP	17.1 DEG.C	VOLTG	2.63 VOLTS	MEANBP	6996. KPA	TGREGH	540.6 DEG.C
TDLDP	2.18 DEG.C			MEANCP	7052. KPA	TGREGC	118.4 DEG.C
TWODPR	19.9 DEG.C					TGCOMP	71.4 DEG.C
						TGBOUN	43.5 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	3.99 L/MIN	1 PWRIN	3148. WATTS	VXIHOR	0.2 CM/S	PWROUT	684. WATTS
TWINCL	54.5 DEG.C	2 QCOOLR	1781. WATTS	VYIVER	7.1 CM/S	INDPWR	729. WATTS
TDLCL	6.49 DEG.C	3 QDSHPT	608. WATTS			PISTST	2.00 CM
TWOCLR	59.87 DEG.C	4 EXTEFF	21.7 %	PHASE ANGLES		DISPST	1.93 CM
		5 TAVHTR	599.7 DEG.C	PADISP	84.5 DEG.		
		6 INTEFF	27.7 %	PAPRES	-16.5 DEG.		
		8 AMPS	1196. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	32.2 HZ	PAMPC	1004. KPA
		10 QDISP	14. WATTS			DISPCP	2.51 CM
		11 QREG1	100. WATTS			PISTCP	2.42 CM
		12 QREG2	112. WATTS			PDYNDB	1188. KPA
		13 QREG3	33. WATTS			PDLCLR*****	KPA
		14 PFP	991.8 KPA/CM			PDLREG*****	KPA
		15 PFD	296.7 KPA/CM			PDLDIS*****	KPA
		16 NBEALE	0.00592				
							T01HTR 599.7 DEG.C
							T02HTR 609.0 DEG.C
							T03HTR 607.8 DEG.C
							T04HTR 596.9 DEG.C
							T05HTR 596.9 DEG.C
							T06HTR 600.5 DEG.C
							T07HTR 571.9 DEG.C
							T08HTR 604.3 DEG.C
							T09HTR 611.4 DEG.C
							T10HTR 598.4 DEG.C
							T11HTR 595.8 DEG.C
							T12HTR 604.2 DEG.C
							T13REG 547.6 DEG.C
							T14REG 510.0 DEG.C
							T15REG 385.6 DEG.C
							T16REG***** DEG.C
							T17REG 369.9 DEG.C
							T18REG 372.0 DEG.C
							T19REG 249.1 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 15:52:31.48 RDG 776

FLUID HELIUM		BAROM 14.416 PSI	REGENERATOR 2	DISPLACER 2	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	4.00 L/MIN	AMPS1	707. AMPS	PRESUP	7356. KPA	TGEXP	558.7 DEG.C
		AMPS2	556. AMPS				
TWINDP	17.0 DEG.C	VOLTG	2.78 VOLTS	MEANBP	7008. KPA	TGREGH	537.8 DEG.C
TDLDP	2.40 DEG.C			MEANCP	7066. KPA	TGREGC	121.3 DEG.C
TWODPR	20.2 DEG.C					TGCOMP	74.3 DEG.C
						TGBOUN	43.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.03 L/MIN	1 PWRIN	3513. WATTS	VXIHOR	0.3 CM/S	PWROUT	772. WATTS
TWINCL	54.6 DEG.C	2 QCOOLR	2021. WATTS	VYIVER	7.8 CM/S	INDPWR	816. WATTS
TDLCL	7.29 DEG.C	3 QDSHPT	669. WATTS			PISTST	2.20 CM
TWOCLR	60.81 DEG.C	4 EXTEFF	22.0 %	PHASE ANGLES		DISPST	2.05 CM
		5 TAVHTR	599.2 DEG.C	PADISP	85.0 DEG.		
		6 INTEFF	27.6 %	PAPRES	-15.5 DEG.		
		8 AMPS	1263. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	32.1 HZ	PAMPC	1083. KPA
		10 QDISP	14. WATTS			DISPCP	2.52 CM
		11 QREG1	97. WATTS			PISTCP	2.43 CM
		12 QREG2	112. WATTS			PDYNDB	1263. KPA
		13 QREG3	32. WATTS			PDLCLR*****	KPA
		14 PFP	973.2 KPA/CM			PDLREG*****	KPA
		15 PFD	283.3 KPA/CM			PDLDIS*****	KPA
		16 NBEALE	0.00608				
							T01HTR 599.5 DEG.C
							T02HTR 608.7 DEG.C
							T03HTR 608.2 DEG.C
							T04HTR 595.7 DEG.C
							T05HTR 596.3 DEG.C
							T06HTR 599.4 DEG.C
							T07HTR 570.2 DEG.C
							T08HTR 603.3 DEG.C
							T09HTR 611.7 DEG.C
							T10HTR 598.4 DEG.C
							T11HTR 595.0 DEG.C
							T12HTR 603.5 DEG.C
							T13REG 546.0 DEG.C
							T14REG 508.9 DEG.C
							T15REG 386.0 DEG.C
							T16REG***** DEG.C
							T17REG 369.2 DEG.C
							T18REG 371.2 DEG.C
							T19REG 251.4 DEG.C
							T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 15:57:19.48 RDG 777

FLUID HELIUM BAROM 14.416 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 3.98 L/MIN	POWER IN AMPS1 748. AMPS AMPS2 592. AMPS VOLTG 2.95 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7351. KPA MEANBP 6998. KPA MEANCP 7062. KPA	GAS TEMPERATURES TGEXP 555.7 DEG.C TGREGH 534.7 DEG.C TGREGC 124.6 DEG.C TGCOMP 78.3 DEG.C TGBOUN 44.0 DEG.C	SURFACE TEMPERATURES T01HTR 599.5 DEG.C T02HTR 609.0 DEG.C T03HTR 609.3 DEG.C T04HTR 595.4 DEG.C T05HTR 597.1 DEG.C T06HTR 598.4 DEG.C T07HTR 569.4 DEG.C T08HTR 602.2 DEG.C T09HTR 613.3 DEG.C T10HTR 598.6 DEG.C T11HTR 594.7 DEG.C T12HTR 603.3 DEG.C
HEAT TO COOLER FLOCLR 4.05 L/MIN TWINCL 54.6 DEG.C TDLCL 8.13 DEG.C TWOCLR 61.75 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3951. WATTS 2 QCOOLR 2261. WATTS 3 QDSHPT 771. WATTS 4 EXTEFF 22.4 % 5 TAVHTR 599.2 DEG.C 6 INTEFF 28.1 % 8 AMPS 1340. AMPS 9 QDISPG 3. WATTS 10 QDISP 14. WATTS 11 QREG1 95. WATTS 12 QREG2 111. WATTS 13 QREG3 32. WATTS 14 PFP 986.6 KPA/CM 15 PFD 277.1 KPA/CM 16 NBEALE 0.00641	VIBRATION VX1HOR 0.3 CM/S VY1VER 8.5 CM/S PHASE ANGLES PADISP 84.9 DEG. PAPRES -14.7 DEG. ENGINE SPEED FREQ 32.0 HZ	REMOTE CALCULATIONS PWROUT 886. WATTS INDPWR 911. WATTS PISTST 2.40 CM DISPST 2.20 CM DYNAMIC CALCULATIONS PAMPC 1196. KPA DISPCP 2.55 CM PISTCP 2.38 CM PDYNDB 1350. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 545.2 DEG.C T14REG 508.7 DEG.C T15REG 387.7 DEG.C T16REG***** DEG.C T17REG 368.9 DEG.C T18REG 372.3 DEG.C T19REG 254.6 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 16:02:16.48 RDG 778

FLUID HELIUM BAROM 14.411 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 3.97 L/MIN	POWER IN AMPS1 792. AMPS AMPS2 619. AMPS VOLTG 3.11 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7349. KPA MEANBP 6998. KPA MEANCP 7066. KPA	GAS TEMPERATURES TGEXP 553.7 DEG.C TGREGH 532.1 DEG.C TGREGC 127.1 DEG.C TGCOMP 82.4 DEG.C TGBOUN 44.7 DEG.C	SURFACE TEMPERATURES T01HTR 600.1 DEG.C T02HTR 611.1 DEG.C T03HTR 611.9 DEG.C T04HTR 596.2 DEG.C T05HTR 599.5 DEG.C T06HTR 597.9 DEG.C T07HTR 570.3 DEG.C T08HTR 602.1 DEG.C T09HTR 616.0 DEG.C T10HTR 599.2 DEG.C T11HTR 595.2 DEG.C T12HTR 605.1 DEG.C
HEAT TO COOLER FLOCLR 4.07 L/MIN TWINCL 54.6 DEG.C TDLCL 8.69 DEG.C TWOCLR 62.74 DEG.C	CALCULATED PARAMETERS 1 PWRIN 4387. WATTS 2 QCOOLR 2428. WATTS 3 QDSHPT 865. WATTS 4 EXTEFF 22.8 % 5 TAVHTR 600.4 DEG.C 6 INTEFF 29.2 % 8 AMPS 1410. AMPS 9 QDISPG 3. WATTS 10 QDISP 13. WATTS 11 QREG1 93. WATTS 12 QREG2 111. WATTS 13 QREG3 31. WATTS 14 PFP 989.0 KPA/CM 15 PFD 267.2 KPA/CM 16 NBEALE 0.00666	VIBRATION VX1HOR 0.4 CM/S VY1VER 9.1 CM/S PHASE ANGLES PADISP 83.6 DEG. PAPRES -13.7 DEG. ENGINE SPEED FREQ 32.0 HZ	REMOTE CALCULATIONS PWROUT 1000. WATTS INDPWR 1005. WATTS PISTST 2.61 CM DISPST 2.30 CM DYNAMIC CALCULATIONS PAMPC 1294. KPA DISPCP 2.77 CM PISTCP 2.51 CM PDYNDB 1450. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 545.5 DEG.C T14REG 509.1 DEG.C T15REG 389.4 DEG.C T16REG***** DEG.C T17REG 363.3 DEG.C T18REG 372.4 DEG.C T19REG 258.0 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 16:06:19.48 RDG 779

FLUID HELIUM BAROM 14.411 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 3.97 L/MIN	POWER IN AMPS1 834. AMPS AMPS2 644. AMPS VOLTG 3.26 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7347. KPA MEANBP 6998. KPA MEANCP 7070. KPA	GAS TEMPERATURES TGEXP 549.9 DEG.C TGREGH 529.2 DEG.C TGREGC 129.9 DEG.C TGCOMP 86.5 DEG.C TGBOUN 45.1 DEG.C	SURFACE TEMPERATURES T01HTR 599.5 DEG.C T02HTR 610.0 DEG.C T03HTR 612.1 DEG.C T04HTR 594.4 DEG.C T05HTR 599.4 DEG.C T06HTR 596.6 DEG.C T07HTR 568.1 DEG.C T08HTR 600.7 DEG.C T09HTR 616.0 DEG.C T10HTR 599.0 DEG.C T11HTR 593.9 DEG.C T12HTR 603.8 DEG.C
HEAT TO COOLER FLOCLR 4.08 L/MIN TWINCL 54.7 DEG.C TDLCL 9.17 DEG.C TWOCLR 63.87 DEG.C	CALCULATED PARAMETERS 1 PWRIN 4815. WATTS 2 QCOOLR 2570. WATTS 3 QDSHPT 932. WATTS 4 EXTEFF 22.3 % 5 TAVHTR 599.5 DEG.C 6 INTEFF 29.5 % 8 AMPS 1477. AMPS 9 QDISPG 3. WATTS 10 QDISP 13. WATTS 11 QREG1 90. WATTS 12 QREG2 110. WATTS 13 QREG3 31. WATTS 14 PFP 993.6 KPA/CM 15 PFD 263.2 KPA/CM 16 NBEALE 0.00669	VIBRATION VX1HOR 0.4 CM/S VY1VER 9.7 CM/S PHASE ANGLES PADISP 83.8 DEG. PAPRES -13.1 DEG. ENGINE SPEED FREQ 31.9 HZ	REMOTE CALCULATIONS PWROUT 1075. WATTS INDPWR 1093. WATTS PISTST 2.80 CM DISPST 2.41 CM DYNAMIC CALCULATIONS PAMPC 1393. KPA DISPCP 2.82 CM PISTCP 2.48 CM PDYNDB 1525. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG 543.8 DEG.C T14REG 508.4 DEG.C T15REG 391.2 DEG.C T16REG***** DEG.C T17REG 362.5 DEG.C T18REG 372.8 DEG.C T19REG 262.0 DEG.C T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/21/83 16:10:19.48 RDG 780

FLUID HELIUM		BAROM 14.411 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	3.99 L/MIN	AMPS1	876. AMPS	PRESUP	7350. KPA	TGEXP	547.1 DEG.C	T01HTR	601.4 DEG.
		AMPS2	681. AMPS			TGREGH	528.7 DEG.C	T02HTR	609.4 DEG.
TWINDP	17.0 DEG.C	VOLTG	3.44 VOLTS	MEANBP	7008. KPA	TGREGC	132.8 DEG.C	T03HTR	613.8 DEG.
TDLDP	3.67 DEG.C			MEANCP	7087. KPA	TGCOMP	90.3 DEG.C	T04HTR	593.1 DEG.
TWODPR	21.6 DEG.C					TGBOUN	46.2 DEG.C	T05HTR	599.5 DEG.
								T06HTR	597.6 DEG.
								T07HTR	565.5 DEG.
								T08HTR	600.9 DEG.
								T09HTR	616.2 DEG.
								T10HTR	600.7 DEG.
								T11HTR	594.6 DEG.
								T12HTR	603.6 DEG.
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.05 L/MIN	1 PWRIN	5355. WATTS	VX1HOR	0.5 CM/S	PWROUT	1175. WATTS	T13REG	542.0 DEG.
TWINCL	54.0 DEG.C	2 QCOOLR	2919. WATTS	VY1VER	10.4 CM/S	INDPWR	1186. WATTS	T14REG	507.0 DEG.
TDLCL	10.48 DEG.C	3 QDSHPT	1018. WATTS			PISTST	3.00 CM	T15REG	391.8 DEG.
TWOCLR	64.54 DEG.C	4 EXTEFF	21.9 %	PHASE ANGLES		DISPST	2.39 CM	T16REG	***** DEG.
		5 TAVHTR	599.7 DEG.C	PADISP	84.3 DEG.			T17REG	366.9 DEG.
		6 INTEFF	28.7 %	PAPRES	-12.0 DEG.			T18REG	376.1 DEG.
		8 AMPS	1557. AMPS	ENGINE SPEED				T19REG	265.2 DEG.
		9 QDISPG	3. WATTS	FREQ	32.1 HZ	PAMPC	1516. KPA	T03HED	***** DEG.
		10 QDISP	13. WATTS			DISPCP	2.89 CM		
		11 QREG1	90. WATTS			PISTCP	2.35 CM		
		12 QREG2	106. WATTS			PDYNDB	1625. KPA		
		13 QREG3	31. WATTS			PDLCLR	***** KPA		
		14 PFP	1008.2 KPA/CM			PDLREG	***** KPA		
		15 PFD	264.5 KPA/CM			PDLDIS	***** KPA		
		16 NBEALE	0.00678						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:00:19.43 RDG 804

FLUID HELIUM		BAROM 14.136 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.12 L/MIN	AMPS1	623. AMPS	PRESUP	7076. KPA	TGEXP	562.6 DEG.C	T01HTR	599.6 DEG.C
		AMPS2	515. AMPS			TGREGH	540.1 DEG.C	T02HTR	610.4 DEG.C
TWINDP	20.3 DEG.C	VOLTG	2.50 VOLTS	MEANBP	7010. KPA	TGREGC	102.3 DEG.C	T03HTR	609.8 DEG.C
TDLDP	1.54 DEG.C			MEANCP	7051. KPA	TGCOMP	56.2 DEG.C	T04HTR	595.3 DEG.C
TWODPR	22.4 DEG.C					TGBOUN	36.6 DEG.C	T05HTR	600.9 DEG.C
								T06HTR	601.1 DEG.C
								T07HTR	574.0 DEG.C
								T08HTR	602.8 DEG.C
								T09HTR	607.4 DEG.C
								T10HTR	598.8 DEG.C
								T11HTR	593.9 DEG.C
								T12HTR	607.6 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.37 L/MIN	1 PWRIN	2844. WATTS	VX1HOR	0.2 CM/S	PWROUT	638. WATTS	T13REG	552.8 DEG.C
TWINCL	39.4 DEG.C	2 QCOOLR	1629. WATTS	VY1VER	6.5 CM/S	INDPWR	669. WATTS	T14REG	515.4 DEG.C
TDLCL	5.40 DEG.C	3 QDSHPT	441. WATTS			PISTST	1.80 CM	T15REG	389.0 DEG.C
TWOCLR	44.66 DEG.C	4 EXTEFF	22.4 %	PHASE ANGLES		DISPST	1.78 CM	T16REG	***** DEG.C
		5 TAVHTR	600.1 DEG.C	PADISP	83.6 DEG.			T17REG	364.8 DEG.C
		6 INTEFF	28.1 %	PAPRES	-18.4 DEG.			T18REG	376.4 DEG.C
		8 AMPS	1139. AMPS	ENGINE SPEED				T19REG	246.1 DEG.C
		9 QDISPG	3. WATTS	FREQ	32.4 HZ	PAMPC	930. KPA	T03HED	***** DEG.C
		10 QDISP	14. WATTS			DISPCP	2.52 CM		
		11 QREG1	106. WATTS			PISTCP	2.44 CM		
		12 QREG2	113. WATTS			PDYNDB	1075. KPA		
		13 QREG3	34. WATTS			PDLCLR	***** KPA		
		14 PFP	1019.6 KPA/CM			PDLREG	***** KPA		
		15 PFD	332.1 KPA/CM			PDLDIS	***** KPA		
		16 NBEALE	0.00608						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:06:16.43 RDG 805

FLUID HELIUM		BAROM 14.140 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.13 L/MIN	AMPS1	661. AMPS	PRESUP	7084. KPA	TGEXP	560.4 DEG.C	T01HTR	599.2 DEG.C
		AMPS2	541. AMPS			TGREGH	537.7 DEG.C	T02HTR	610.0 DEG.C
TWINDP	20.3 DEG.C	VOLTG	2.64 VOLTS	MEANBP	6997. KPA	TGREGC	105.7 DEG.C	T03HTR	609.8 DEG.C
TDLDP	1.80 DEG.C			MEANCP	7043. KPA	TGCOMP	59.6 DEG.C	T04HTR	595.5 DEG.C
TWODPR	22.7 DEG.C					TGBOUN	38.0 DEG.C	T05HTR	600.5 DEG.C
								T06HTR	599.8 DEG.C
								T07HTR	572.6 DEG.C
								T08HTR	602.4 DEG.C
								T09HTR	609.4 DEG.C
								T10HTR	598.6 DEG.C
								T11HTR	594.0 DEG.C
								T12HTR	606.5 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.37 L/MIN	1 PWRIN	3173. WATTS	VX1HOR	0.2 CM/S	PWROUT	727. WATTS	T13REG	550.8 DEG.C
TWINCL	39.5 DEG.C	2 QCOOLR	1886. WATTS	VY1VER	7.1 CM/S	INDPWR	765. WATTS	T14REG	513.6 DEG.C
TDLCL	6.25 DEG.C	3 QDSHPT	516. WATTS			PISTST	1.98 CM	T15REG	388.1 DEG.C
TWOCLR	45.58 DEG.C	4 EXTEFF	22.9 %	PHASE ANGLES		DISPST	1.91 CM	T16REG	***** DEG.C
		5 TAVHTR	599.9 DEG.C	PADISP	84.2 DEG.			T17REG	363.4 DEG.C
		6 INTEFF	27.8 %	PAPRES	-17.5 DEG.			T18REG	376.2 DEG.C
		8 AMPS	1202. AMPS	ENGINE SPEED				T19REG	247.4 DEG.C
		9 QDISPG	3. WATTS	FREQ	32.3 HZ	PAMPC	1019. KPA	T03HED	***** DEG.C
		10 QDISP	14. WATTS			DISPCP	2.54 CM		
		11 QREG1	105. WATTS			PISTCP	2.45 CM		
		12 QREG2	111. WATTS			PDYNDB	1175. KPA		
		13 QREG3	33. WATTS			PDLCLR	***** KPA		
		14 PFP	1012.1 KPA/CM			PDLREG	***** KPA		
		15 PFD	323.8 KPA/CM			PDLDIS	***** KPA		
		16 NBEALE	0.00632						

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:10:34.48 RDG 806

FLUID HELIUM		BAROM 14.140 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
EAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.12 L/MIN	AMPS1	702. AMPS	PRESUP	7058. KPA	TGEXP	556.9 DEG.C	T01HTR	599.3 DEG.C
		AMPS2	574. AMPS					T02HTR	609.0 DEG.C
TWINDP	20.2 DEG.C	VOLTG	2.80 VOLTS	MEANBP	6989. KPA	TGREGH	535.3 DEG.C	T03HTR	610.3 DEG.C
TDLDP	2.13 DEG.C			MEANCP	7036. KPA	TGREGC	108.4 DEG.C	T04HTR	594.1 DEG.C
TWODPR	23.0 DEG.C					TGCOMP	62.1 DEG.C	T05HTR	599.8 DEG.C
						TGBOUN	38.8 DEG.C	T06HTR	598.9 DEG.C
								T07HTR	570.4 DEG.C
								T08HTR	601.5 DEG.C
								T09HTR	609.8 DEG.C
EAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCIR	4.37 L/MIN	1 PWRIN	3568. WATTS	VX1HOR	0.3 CM/S	PWROUT	827. WATTS	T10HTR	598.7 DEG.C
TWINCL	39.6 DEG.C	2 QCOOLR	2145. WATTS	VY1VER	7.8 CM/S	INDPWR	868. WATTS	T11HTR	593.4 DEG.C
TDLCL	7.12 DEG.C	3 QDSHPT	612. WATTS			PISTST	2.19 CM	T12HTR	605.2 DEG.C
TWOCLR	46.49 DEG.C	4 EXTEFF	23.2 %	PHASE ANGLES		DISPST	2.04 CM		
		5 TAVHTR	599.2 DEG.C	PADISP	84.5 DEG.			T13REG	548.0 DEG.C
		6 INTEFF	27.8 %	PAPRES	-16.6 DEG.			T14REG	511.3 DEG.C
		8 AMPS	1276. AMPS			DYNAMIC CALCULATIONS		T15REG	386.5 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	1127. KPA	T16REG*****	DEG.C
		10 QDISP	14. WATTS	FREQ	32.2 HZ	DISPCP	2.58 CM	T17REG	363.4 DEG.C
		11 QREG1	104. WATTS			PISTCP	2.46 CM	T18REG	376.1 DEG.C
		12 QREG2	110. WATTS			PDYNDB	1263. KPA	T19REG	248.0 DEG.C
		13 QREG3	33. WATTS			PDLCLR*****	KPA		
		14 PFP	1012.9 KPA/CM			PDLREG*****	KPA	T03HED*****	DEG.C
		15 PFD	315.8 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00652						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:15:19.48 RDG 807

FLUID HELIUM		BAROM 14.140 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.12 L/MIN	AMPS1	745. AMPS	PRESUP	7098. KPA	TGEXP	554.4 DEG.C	T01HTR	600.7 DEG.C
		AMPS2	602. AMPS					T02HTR	607.8 DEG.C
TWINDP	20.3 DEG.C	VOLTG	2.95 VOLTS	MEANBP	7009. KPA	TGREGH	534.3 DEG.C	T03HTR	611.5 DEG.C
TDLDP	2.47 DEG.C			MEANCP	7065. KPA	TGREGC	113.3 DEG.C	T04HTR	593.6 DEG.C
TWODPR	23.3 DEG.C					TGCOMP	65.8 DEG.C	T05HTR	598.8 DEG.C
						TGBOUN	40.1 DEG.C	T06HTR	599.5 DEG.C
								T07HTR	567.7 DEG.C
								T08HTR	601.9 DEG.C
								T09HTR	611.4 DEG.C
								T10HTR	600.1 DEG.C
								T11HTR	594.3 DEG.C
								T12HTR	603.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.38 L/MIN	1 PWRIN	3977. WATTS	VX1HOR	0.3 CM/S	PWROUT	930. WATTS	T13REG	545.5 DEG.C
TWINCL	39.7 DEG.C	2 QCOOLR	2443. WATTS	VY1VER	8.4 CM/S	INDPWR	973. WATTS	T14REG	509.0 DEG.C
TDLCL	8.08 DEG.C	3 QDSHPT	708. WATTS			PISTST	2.40 CM	T15REG	384.4 DEG.C
TWOCLR	47.47 DEG.C	4 EXTEFF	23.4 %	PHASE ANGLES		DISPST	2.18 CM	T16REG*****	DEG.C
		5 TAVHTR	599.3 DEG.C	PADISP	84.6 DEG.			T17REG	366.0 DEG.C
		6 INTEFF	27.6 %	PAPRES	-15.7 DEG.			T18REG	376.0 DEG.C
		8 AMPS	1347. AMPS			DYNAMIC CALCULATIONS		T19REG	247.9 DEG.C
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	1235. KPA	T03HED*****	DEG.C
		10 QDISP	14. WATTS	FREQ	32.2 HZ	DISPCP	2.62 CM		
		11 QREG1	104. WATTS			PISTCP	2.46 CM		
		12 QREG2	108. WATTS			PDYNDB	1363. KPA		
		13 QREG3	33. WATTS			PDLCLR*****	KPA		
		14 PFP	1018.6 KPA/CM			PDLREG*****	KPA		
		15 PFD	308.2 KPA/CM			PDLDIS*****	KPA		
		16 NBEAL	0.00670						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:21:04.48 RDG 808

FLUID HELIUM		BAROM 14.140 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.12 L/MIN	AMPS1	772. AMPS	PRESUP	7092. KPA	TGEXP	551.3 DEG.C	T01HTR	602.3 DEG.C
		AMPS2	627. AMPS					T02HTR	606.0 DEG.C
TWINDP	20.3 DEG.C	VOLTG	3.08 VOLTS	MEANBP	6987. KPA	TGREGH	533.7 DEG.C	T03HTR	612.4 DEG.C
TDLDP	2.83 DEG.C			MEANCP	7051. KPA	TGREGC	116.6 DEG.C	T04HTR	592.6 DEG.C
TWODPR	23.6 DEG.C					TGCOMP	69.6 DEG.C	T05HTR	596.3 DEG.C
						TGBOUN	41.5 DEG.C	T06HTR	599.9 DEG.C
								T07HTR	564.9 DEG.C
								T08HTR	602.1 DEG.C
								T09HTR	612.0 DEG.C
								T10HTR	601.6 DEG.C
								T11HTR	595.1 DEG.C
								T12HTR	602.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.37 L/MIN	1 PWRIN	4307. WATTS	VX1HOR	0.4 CM/S	PWROUT	1006. WATTS		
TWINCL	39.7 DEG.C	2 QCOOLR	2686. WATTS	VY1VER	9.0 CM/S	INDPWR	1051. WATTS		
TDLCL	8.90 DEG.C	3 QDSHPT	811. WATTS			PISTST	2.59 CM		
TWOCLR	48.29 DEG.C	4 EXTEFF	23.4 %	PHASE ANGLES		DISPST	2.28 CM		
		5 TAVHTR	599.0 DEG.C	PADISP	84.4 DEG.			T13REG	542.2 DEG.C
		6 INTEFF	27.2 %	PAPRES	-14.7 DEG.	DYNAMIC CALCULATIONS		T14REG	504.7 DEG.C
		8 AMPS	1399. AMPS	ENGINE SPEED		PAMPC	1324. KPA	T15REG	379.1 DEG.C
		9 QDISPG	3. WATTS	FREQ	32.0 HZ	DISPCP	2.64 CM	T16REG	***** DEG.C
		10 QDISP	14. WATTS			PISTCP	2.44 CM	T17REG	368.6 DEG.C
		11 QREG1	107. WATTS			PDYNDB	1438. KPA	T18REG	376.3 DEG.C
		12 QREG2	104. WATTS			PDLCLR	***** KPA	T19REG	244.8 DEG.C
		13 QREG3	32. WATTS			PDLREG	***** KPA		
		14 PFP	1012.5 KPA/CM			PDLDIS	***** KPA	T03HED	***** DEG.C
		15 PFD	296.8 KPA/CM						
		16 NBEALE	0.00676						

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APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/06/84 13:25:19.48 RDG 809

FLUID HELIUM		BAROM 14.140 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.11 L/MIN	AMPS1	807. AMPS	PRESUP	7071. KPA	TGEXP	548.0 DEG.C	T01HTR	602.0 DEG.C
TWINDP	20.3 DEG.C	AMPS2	657. AMPS	MEANBP	6995. KPA	TGREGH	531.8 DEG.C	T02HTR	605.5 DEG.C
TDLDP	3.06 DEG.C	VOLTG	3.23 VOLTS	MEANCNP	7063. KPA	TGREGC	119.2 DEG.C	T03HTR	612.7 DEG.C
TWODPR	23.9 DEG.C					TGCOMP	73.6 DEG.C	T04HTR	591.3 DEG.C
						TGBOUN	43.2 DEG.C	T05HTR	596.1 DEG.C
								T06HTR	599.0 DEG.C
								T07HTR	563.2 DEG.C
								T08HTR	601.3 DEG.C
								T09HTR	612.4 DEG.C
								T10HTR	601.5 DEG.C
								T11HTR	594.8 DEG.C
								T12HTR	601.4 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.39 L/MIN	1 PWRIN	4736. WATTS	VX1HOR	0.4 CM/S	PWROUT	1089. WATTS	T13REG	540.6 DEG.C
TWINCL	39.7 DEG.C	2 QCOOLR	3031. WATTS	VY1VER	9.6 CM/S	INDPWR	1138. WATTS	T14REG	503.7 DEG.C
TDLCL	10.00 DEG.C	3 QDSHPT	874. WATTS			PISTST	2.79 CM	T15REG	380.7 DEG.C
TWOCLR	49.39 DEG.C	4 EXTEFF	23.0 %	PHASE ANGLES		DISPST	2.40 CM	T16REG*****	DEG.C
		5 TAVHTR	598.4 DEG.C	PADISP	84.9 DEG.			T17REG	368.5 DEG.C
		6 INTEFF	26.4 %	PAPRES	-13.8 DEG.			T18REG	376.1 DEG.C
		8 AMPS	1464. AMPS	ENGINE SPEED				T19REG	248.8 DEG.C
		9 QDISPG	3. WATTS	FREQ	31.9 HZ			T03HED*****	DEG.C
		10 QDISP	13. WATTS						
		11 QREG1	103. WATTS						
		12 QREG2	104. WATTS						
		13 QREG3	32. WATTS						
		14 PFP	1014.3 KPA/CM						
		15 PFD	284.2 KPA/CM						
		16 NBEALE	0.00680						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/11/84 13:59:30.48 RDG 810

FLUID HELIUM		BAROM 14.489 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.04 L/MIN	AMPS1	838. AMPS	PRESUP	7493. KPA	TGEXP	545.2 DEG.C	T01HTR	604.3 DEG.C
TWINDP	17.7 DEG.C	AMPS2	709. AMPS	MEANBP	6998. KPA	TGREGH	529.9 DEG.C	T02HTR	606.4 DEG.C
TDLDP	3.08 DEG.C	VOLTG	3.40 VOLTS	MEANCNP	7065. KPA	TGREGC	116.5 DEG.C	T03HTR	614.7 DEG.C
TWODPR	21.4 DEG.C					TGCOMP	75.0 DEG.C	T04HTR	591.2 DEG.C
						TGBOUN	38.1 DEG.C	T05HTR	595.7 DEG.C
								T06HTR	600.3 DEG.C
								T07HTR	561.9 DEG.C
								T08HTR	601.4 DEG.C
								T09HTR	613.5 DEG.C
								T10HTR	603.2 DEG.C
								T11HTR	595.0 DEG.C
								T12HTR	601.5 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.59 L/MIN	1 PWRIN	5261. WATTS	VX1HOR	0.6 CM/S	PWROUT	1219. WATTS	T13REG*****	DEG.C
TWINCL	39.3 DEG.C	2 QCOOLR	3181. WATTS	VY1VER	10.2 CM/S	INDPWR	1240. WATTS	T14REG	501.8 DEG.C
TDLCL	10.03 DEG.C	3 QDSHPT	866. WATTS			PISTST	2.98 CM	T15REG	380.7 DEG.C
TWOCLR	48.65 DEG.C	4 EXTEFF	23.2 %	PHASE ANGLES		DISPST	2.48 CM	T16REG*****	DEG.C
		5 TAVHTR	599.1 DEG.C	PADISP	84.7 DEG.			T17REG	369.1 DEG.C
		6 INTEFF	27.7 %	PAPRES	-12.8 DEG.			T18REG	378.4 DEG.C
		8 AMPS	1547. AMPS	ENGINE SPEED				T19REG	250.2 DEG.C
		9 QDISPG	3. WATTS	FREQ	32.1 HZ			T03HED*****	DEG.C
		10 QDISP	13. WATTS						
		11 QREG1	104. WATTS						
		12 QREG2	100. WATTS						
		13 QREG3	32. WATTS						
		14 PFP	1025.8 KPA/CM						
		15 PFD	275.7 KPA/CM						
		16 NBEALE	0.00710						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/11/84 14:04:09.48 RDG 811

FLUID HELIUM		BAROM 14.489 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.03 L/MIN	AMPS1	868. AMPS	PRESUP	7503. KPA	TGEXP	542.6 DEG.C	T01HTR	606.1 DEG.C
TWINDP	17.7 DEG.C	AMPS2	734. AMPS	MEANBP	6988. KPA	TGREGH	530.1 DEG.C	T02HTR	605.2 DEG.C
TDLDP	3.45 DEG.C	VOLTG	3.54 VOLTS	MEANCNP	7061. KPA	TGREGC	122.5 DEG.C	T03HTR	616.0 DEG.C
TWODPR	21.8 DEG.C					TGCOMP	79.2 DEG.C	T04HTR	590.4 DEG.C
						TGBOUN	41.0 DEG.C	T05HTR	596.1 DEG.C
								T06HTR	601.2 DEG.C
								T07HTR	559.9 DEG.C
								T08HTR	602.7 DEG.C
								T09HTR	614.0 DEG.C
								T10HTR	605.1 DEG.C
								T11HTR	596.9 DEG.C
								T12HTR	601.2 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.51 L/MIN	1 PWRIN	5666. WATTS	VX1HOR	0.6 CM/S	PWROUT	1290. WATTS	T13REG*****	DEG.C
TWINCL	39.8 DEG.C	2 QCOOLR	3468. WATTS	VY1VER	10.9 CM/S	INDPWR	1299. WATTS	T14REG	499.9 DEG.C
TDLCL	11.15 DEG.C	3 QDSHPT	969. WATTS			PISTST	3.20 CM	T15REG	377.8 DEG.C
TWOCLR	50.71 DEG.C	4 EXTEFF	22.8 %	PHASE ANGLES		DISPST	2.59 CM	T16REG*****	DEG.C
		5 TAVHTR	599.6 DEG.C	PADISP	84.2 DEG.			T17REG	371.7 DEG.C
		6 INTEFF	27.1 %	PAPRES	-12.0 DEG.			T18REG	378.2 DEG.C
		8 AMPS	1601. AMPS	ENGINE SPEED				T19REG	250.0 DEG.C
		9 QDISPG	3. WATTS	FREQ	32.0 HZ			T03HED*****	DEG.C
		10 QDISP	13. WATTS						
		11 QREG1	104. WATTS						
		12 QREG2	99. WATTS						
		13 QREG3	32. WATTS						
		14 PFP	1025.1 KPA/CM						
		15 PFD	265.2 KPA/CM						
		16 NBEALE	0.00703						

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/11/84 14:07:18.48 RDG 812

FLUID HELIUM BAROM 14.494 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.03 L/MIN	AMPS1 895. AMPS	PRESUP 7427. KPA	TGEXP 541.0 DEG.C	T01HTR 607.2 DEG.C
	AMPS2 791. AMPS			T02HTR 606.2 DEG.C
TWINDP 17.7 DEG.C	VOLTG 3.70 VOLTS	MEANBP 7014. KPA	TGREGH 530.6 DEG.C	T03HTR 617.1 DEG.C
TDLDP 3.69 DEG.C		MEANCP 7097. KPA	TGREGC 127.5 DEG.C	T04HTR 591.2 DEG.C
TWODPR 22.2 DEG.C			TGCOMP 83.6 DEG.C	T05HTR 596.3 DEG.C
			TGBOUN 42.8 DEG.C	T06HTR 601.3 DEG.C
				T07HTR 559.4 DEG.C
				T08HTR 604.2 DEG.C
				T09HTR 616.8 DEG.C
				T10HTR 606.1 DEG.C
				T11HTR 598.7 DEG.C
				T12HTR 601.1 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.51 L/MIN	1 PWRIN 6242. WATTS	VX1HOR 0.7 CM/S	PWROUT 1377. WATTS	
TWINCL 39.9 DEG.C	2 QCOOLR 3840. WATTS	VY1VER 11.6 CM/S	INDPWR 1378. WATTS	
TDLCL 12.34 DEG.C	3 QDSHPT 1035. WATTS		PISTST 3.39 CM	
TWOCLR 52.00 DEG.C	4 EXTEFF 22.1 %	PHASE ANGLES	DISPST 2.70 CM	
	5 TAVHTR 600.5 DEG.C	PADISP 83.9 DEG.		T13REG***** DEG.C
	6 INTEFF 26.4 %	PAPRES -11.0 DEG.		T14REG 499.6 DEG.C
	8 AMPS 1686. AMPS		DYNAMIC CALCULATIONS	T15REG 379.5 DEG.C
	9 QDISPG 3. WATTS	ENGINE SPEED	PAMPC 1762. KPA	T16REG***** DEG.C
	10 QDISP 13. WATTS	FREQ 32.1 HZ	DISPCP 2.84 CM	T17REG 597.5 DEG.C
	11 QREG1 101. WATTS		PISTCP 2.04 CM	T18REG 378.3 DEG.C
	12 QREG2 98. WATTS		PDYNDB 1750. KPA	T19REG 253.4 DEG.C
	13 QREG3 31. WATTS		PDLCLR***** KPA	
	14 PFP 1040.7 KPA/CM		PDLREG***** KPA	
	15 PFD 251.4 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00702			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 13:57:19.44 RDG 862

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.32 L/MIN	AMPS1 401. AMPS	PRESUP 4030. KPA	TGEXP 557.6 DEG.C	T01HTR 599.7 DEG.C
	AMPS2 468. AMPS			T02HTR 606.4 DEG.C
TWINDP 17.9 DEG.C	VOLTG 1.91 VOLTS	MEANBP 3998. KPA	TGREGH 545.1 DEG.C	T03HTR 603.2 DEG.C
TDLDP 0.85 DEG.C		MEANCP 4026. KPA	TGREGC 79.1 DEG.C	T04HTR 599.0 DEG.C
TWODPR 19.4 DEG.C			TGCOMP 36.8 DEG.C	T05HTR 597.5 DEG.C
			TGBOUN 33.8 DEG.C	T06HTR 602.2 DEG.C
				T07HTR 578.8 DEG.C
				T08HTR 606.1 DEG.C
				T09HTR 607.3 DEG.C
				T10HTR 598.6 DEG.C
				T11HTR 598.4 DEG.C
				T12HTR 602.8 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.17 L/MIN	1 PWRIN 1663. WATTS	VX1HOR 0.2 CM/S	PWROUT 355. WATTS	
TWINCL 25.0 DEG.C	2 QCOOLR 915. WATTS	VY1VER 4.4 CM/S	INDPWR 369. WATTS	
TDLCL 3.16 DEG.C	3 QDSHPT 256. WATTS		PISTST 1.82 CM	
TWOCLR 28.66 DEG.C	4 EXTEFF 21.4 %	PHASE ANGLES	DISPST 1.89 CM	
	5 TAVHTR 600.0 DEG.C	PADISP 80.0 DEG.		T13REG***** DEG.C
	6 INTEFF 28.0 %	PAPRES -21.7 DEG.		T14REG 508.6 DEG.C
	8 AMPS 869. AMPS		DYNAMIC CALCULATIONS	T15REG 375.0 DEG.C
	9 QDISPG 3. WATTS	ENGINE SPEED	PAMPC 541. KPA	T16REG***** DEG.C
	10 QDISP 15. WATTS	FREQ 24.9 HZ	DISPCP 2.63 CM	T17REG 598.6 DEG.C
	11 QREG1 108. WATTS		PISTCP 2.50 CM	T18REG 352.8 DEG.C
	12 QREG2 126. WATTS		PDYNDB 625. KPA	T19REG 219.7 DEG.C
	13 QREG3 36. WATTS		PDLCLR***** KPA	
	14 PFP 591.7 KPA/CM		PDLREG***** KPA	
	15 PFD 214.5 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00764			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:02:19.44 RDG 863

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.34 L/MIN	AMPS1 424. AMPS	PRESUP 4033. KPA	TGEXP 554.9 DEG.C	T01HTR 599.6 DEG.C
	AMPS2 486. AMPS			T02HTR 605.6 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.00 VOLTS	MEANBP 4004. KPA	TGREGH 543.1 DEG.C	T03HTR 602.9 DEG.C
TDLDP 0.90 DEG.C		MEANCP 4036. KPA	TGREGC 79.7 DEG.C	T04HTR 597.2 DEG.C
TWODPR 19.4 DEG.C			TGCOMP 37.6 DEG.C	T05HTR 597.4 DEG.C
			TGBOUN 33.4 DEG.C	T06HTR 601.3 DEG.C
				T07HTR 577.5 DEG.C
				T08HTR 604.4 DEG.C
				T09HTR 606.8 DEG.C
				T10HTR 598.5 DEG.C
				T11HTR 597.4 DEG.C
				T12HTR 601.8 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.16 L/MIN	1 PWRIN 1821. WATTS	VX1HOR 0.2 CM/S	PWROUT 405. WATTS	
TWINCL 24.8 DEG.C	2 QCOOLR 987. WATTS	VY1VER 4.7 CM/S	INDPWR 424. WATTS	
TDLCL 3.42 DEG.C	3 QDSHPT 272. WATTS		PISTST 1.99 CM	
TWOCLR 28.77 DEG.C	4 EXTEFF 22.3 %	PHASE ANGLES	DISPST 2.01 CM	
	5 TAVHTR 599.2 DEG.C	PADISP 80.8 DEG.		T13REG***** DEG.C
	6 INTEFF 29.1 %	PAPRES -20.9 DEG.		T14REG 509.0 DEG.C
	8 AMPS 910. AMPS		DYNAMIC CALCULATIONS	T15REG 377.1 DEG.C
	9 QDISPG 3. WATTS	ENGINE SPEED	PAMPC 600. KPA	T16REG***** DEG.C
	10 QDISP 15. WATTS	FREQ 24.8 HZ	DISPCP 2.64 CM	T17REG 357.9 DEG.C
	11 QREG1 107. WATTS		PISTCP 2.56 CM	T18REG 352.6 DEG.C
	12 QREG2 127. WATTS		PDYNDB 664. KPA	T19REG 220.9 DEG.C
	13 QREG3 36. WATTS		PDLCLR***** KPA	
	14 PFP 599.7 KPA/CM		PDLREG***** KPA	
	15 PFD 215.8 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00800			T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:07:19.44 RDG 864

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURE
FLODP 4.33 L/MIN	AMPS1 448. AMPS	PRESUP 4036. KPA	TGEXP 551.9 DEG.C	T01HTR 600.5 DEG.C
	AMPS2 504. AMPS			T02HTR 605.3 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.10 VOLTS	MEANBP 4009. KPA	TGREGH 541.5 DEG.C	T03HTR 604.0 DEG.C
TDLDP 0.99 DEG.C		MEANCP 4044. KPA	TGREGC 81.1 DEG.C	T04HTR 596.5 DEG.C
TWODPR 19.6 DEG.C			TGCOMP 39.2 DEG.C	T05HTR 597.8 DEG.C
			TGBOUN 33.5 DEG.C	T06HTR 601.2 DEG.C
				T07HTR 576.8 DEG.C
				T08HTR 603.4 DEG.C
				T09HTR 607.4 DEG.C
				T10HTR 599.3 DEG.C
				T11HTR 597.1 DEG.C
				T12HTR 601.3 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.17 L/MIN	1 PWRIN 1999. WATTS	VX1HOR 0.2 CM/S	PWROUT 449. WATTS	T13REG***** DEG.C
TWINCL 24.8 DEG.C	2 QCOOLR 1111. WATTS	VY1VER 5.1 CM/S	INDPWR 471. WATTS	T14REG 508.2 DEG.C
TDLCL 3.84 DEG.C	3 QDSHPT 299. WATTS		PISTST 2.15 CM	T15REG 376.7 DEG.C
TWOCLR 29.15 DEG.C	4 EXTEFF 22.5 %	PHASE ANGLES	DISPST 2.12 CM	T16REG***** DEG.C
	5 TAVHTR 599.2 DEG.C	PADISP 81.0 DEG.		T17REG 358.4 DEG.C
	6 INTEFF 28.8 %	PAPRES -20.0 DEG.	DYNAMIC CALCULATIONS	T18REG 353.4 DEG.C
	8 AMPS 953. AMPS	ENGINE SPEED	PAMPC 640. KPA	T19REG 221.9 DEG.C
	9 QDISPG 3. WATTS	FREQ 24.7 HZ	DISPCP 2.67 CM	
	10 QDISP 15. WATTS		PISTCP 2.58 CM	
	11 QREG1 107. WATTS		PDYNDB 738. KPA	
	12 QREG2 126. WATTS		PDLCLR***** KPA	
	13 QREG3 36. WATTS		PDLREG***** KPA	
	14 PFP 592.3 KPA/CM		PDLDIS***** KPA	
	15 PFD 209.2 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00821			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:12:19.44 RDG 865

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURE
FLODP 4.34 L/MIN	AMPS1 479. AMPS	PRESUP 4040. KPA	TGEXP 550.7 DEG.C	T01HTR 601.4 DEG.C
	AMPS2 526. AMPS			T02HTR 605.3 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.21 VOLTS	MEANBP 4014. KPA	TGREGH 540.3 DEG.C	T03HTR 604.8 DEG.C
TDLDP 1.20 DEG.C		MEANCP 4053. KPA	TGREGC 82.5 DEG.C	T04HTR 596.0 DEG.C
TWODPR 19.7 DEG.C			TGCOMP 41.1 DEG.C	T05HTR 598.4 DEG.C
			TGBOUN 33.7 DEG.C	T06HTR 600.9 DEG.C
				T07HTR 576.5 DEG.C
				T08HTR 602.3 DEG.C
				T09HTR 608.6 DEG.C
				T10HTR 600.2 DEG.C
				T11HTR 597.2 DEG.C
				T12HTR 600.7 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.16 L/MIN	1 PWRIN 2225. WATTS	VX1HOR 0.3 CM/S	PWROUT 516. WATTS	T13REG***** DEG.C
TWINCL 24.7 DEG.C	2 QCOOLR 1228. WATTS	VY1VER 5.6 CM/S	INDPWR 547. WATTS	T14REG 508.3 DEG.C
TDLCL 4.25 DEG.C	3 QDSHPT 363. WATTS		PISTST 2.39 CM	T15REG 376.6 DEG.C
TWOCLR 29.61 DEG.C	4 EXTEFF 23.2 %	PHASE ANGLES	DISPST 2.30 CM	T16REG***** DEG.C
	5 TAVHTR 599.4 DEG.C	PADISP 81.9 DEG.		T17REG 357.3 DEG.C
	6 INTEFF 29.6 %	PAPRES -18.8 DEG.	DYNAMIC CALCULATIONS	T18REG 351.2 DEG.C
	8 AMPS 1005. AMPS	ENGINE SPEED	PAMPC 704. KPA	T19REG 222.8 DEG.C
	9 QDISPG 3. WATTS	FREQ 24.6 HZ	DISPCP 2.63 CM	
	10 QDISP 14. WATTS		PISTCP 2.61 CM	
	11 QREG1 104. WATTS		PDYNDB 807. KPA	
	12 QREG2 127. WATTS		PDLCLR***** KPA	
	13 QREG3 35. WATTS		PDLREG***** KPA	
	14 PFP 583.6 KPA/CM		PDLDIS***** KPA	
	15 PFD 199.7 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00849			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:17:19.44 RDG 866

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURE
FLODP 4.33 L/MIN	AMPS1 500. AMPS	PRESUP 4055. KPA	TGEXP 549.3 DEG.C	T01HTR 602.6 DEG.C
	AMPS2 545. AMPS			T02HTR 606.2 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.30 VOLTS	MEANBP 4008. KPA	TGREGH 539.7 DEG.C	T03HTR 606.7 DEG.C
TDLDP 1.27 DEG.C		MEANCP 4049. KPA	TGREGC 84.1 DEG.C	T04HTR 596.4 DEG.C
TWODPR 19.8 DEG.C			TGCOMP 42.9 DEG.C	T05HTR 600.2 DEG.C
			TGBOUN 34.1 DEG.C	T06HTR 601.0 DEG.C
				T07HTR 577.2 DEG.C
				T08HTR 602.1 DEG.C
				T09HTR 610.3 DEG.C
				T10HTR 601.5 DEG.C
				T11HTR 598.0 DEG.C
				T12HTR 601.3 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.18 L/MIN	1 PWRIN 2408. WATTS	VX1HOR 0.3 CM/S	PWROUT 572. WATTS	T13REG***** DEG.C
TWINCL 24.9 DEG.C	2 QCOOLR 1362. WATTS	VY1VER 6.0 CM/S	INDPWR 607. WATTS	T14REG 508.9 DEG.C
TDLCL 4.70 DEG.C	3 QDSHPT 383. WATTS		PISTST 2.59 CM	T15REG 377.5 DEG.C
TWOCLR 30.07 DEG.C	4 EXTEFF 23.8 %	PHASE ANGLES	DISPST 2.43 CM	T16REG***** DEG.C
	5 TAVHTR 600.3 DEG.C	PADISP 81.4 DEG.		T17REG 357.0 DEG.C
	6 INTEFF 29.6 %	PAPRES -18.3 DEG.	DYNAMIC CALCULATIONS	T18REG 350.7 DEG.C
	8 AMPS 1045. AMPS	ENGINE SPEED	PAMPC 753. KPA	T19REG 224.4 DEG.C
	9 QDISPG 3. WATTS	FREQ 24.4 HZ	DISPCP 2.64 CM	
	10 QDISP 14. WATTS		PISTCP 2.63 CM	
	11 QREG1 102. WATTS		PDYNDB 850. KPA	
	12 QREG2 128. WATTS		PDLCLR***** KPA	
	13 QREG3 35. WATTS		PDLREG***** KPA	
	14 PFP 579.0 KPA/CM		PDLDIS***** KPA	
	15 PFD 196.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00877			

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:22:19.44 RDG 867

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.34 L/MIN	AMPS1 526. AMPS	PRESUP 4050. KPA	TGEXP 545.6 DEG.C	T01HTR 602.7 DEG.C
	AMPS2 563. AMPS			T02HTR 604.9 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.40 VOLTS	MEANBP 3997. KPA	TGREGH 537.7 DEG.C	T03HTR 606.9 DEG.C
TDLDP 1.39 DEG.C		MEANCP 4045. KPA	TGREGC 85.7 DEG.C	T04HTR 595.0 DEG.C
TWODPR 20.0 DEG.C			TGCOMP 45.2 DEG.C	T05HTR 599.7 DEG.C
			TGBOUN 34.8 DEG.C	T06HTR 599.9 DEG.C
				T07HTR 575.8 DEG.C
				T08HTR 600.1 DEG.C
				T09HTR 610.4 DEG.C
				T10HTR 601.5 DEG.C
				T11HTR 596.9 DEG.C
				T12HTR 599.7 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.17 L/MIN	1 PWRIN 2617. WATTS	VX1HOR 0.3 CM/S	PWROUT 629. WATTS	
TWINCL 24.9 DEG.C	2 QCOOLR 1507. WATTS	VY1VER 6.5 CM/S	INDPWR 668. WATTS	
TDLCL 5.21 DEG.C	3 QDSHPT 420. WATTS		PISTST 2.80 CM	
TWOCLR 30.65 DEG.C	4 EXTEFF 24.0 %	PHASE ANGLES	DISPST 2.57 CM	
	5 TAVHTR 599.5 DEG.C	PADISP 81.5 DEG.		
	6 INTEFF 29.4 %	PAPRES -17.4 DEG.		
	8 AMPS 1090. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 24.3 HZ	PAMPC 817. KPA	T13REG***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.67 CM	T14REG 507.4 DEG.C
	11 QREG1 102. WATTS		PISTCP 2.68 CM	T15REG 376.9 DEG.C
	12 QREG2 127. WATTS		PDYNDB 888. KPA	T16REG***** DEG.C
	13 QREG3 35. WATTS		PDLCLR***** KPA	T17REG 356.9 DEG.C
	14 PFP 583.2 KPA/CM		PDLREG***** KPA	T18REG 350.4 DEG.C
	15 PFD 192.6 KPA/CM		PDLDIS***** KPA	T19REG 225.2 DEG.C
	16 NBEALE 0.00900			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:27:19.44 RDG 868

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.34 L/MIN	AMPS1 552. AMPS	PRESUP 4050. KPA	TGEXP 542.4 DEG.C	T01HTR 602.9 DEG.C
	AMPS2 582. AMPS			T02HTR 603.4 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.50 VOLTS	MEANBP 3993. KPA	TGREGH 536.1 DEG.C	T03HTR 607.4 DEG.C
TDLDP 1.59 DEG.C		MEANCP 4043. KPA	TGREGC 87.5 DEG.C	T04HTR 593.6 DEG.C
TWODPR 20.2 DEG.C			TGCOMP 47.4 DEG.C	T05HTR 598.9 DEG.C
			TGBOUN 35.6 DEG.C	T06HTR 599.2 DEG.C
				T07HTR 574.0 DEG.C
				T08HTR 598.7 DEG.C
				T09HTR 610.4 DEG.C
				T10HTR 601.8 DEG.C
				T11HTR 596.2 DEG.C
				T12HTR 597.9 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.17 L/MIN	1 PWRIN 2836. WATTS	VX1HOR 0.3 CM/S	PWROUT 683. WATTS	
TWINCL 25.0 DEG.C	2 QCOOLR 1631. WATTS	VY1VER 6.9 CM/S	INDPWR 727. WATTS	
TDLCL 5.64 DEG.C	3 QDSHPT 480. WATTS		PISTST 2.99 CM	
TWOCLR 31.16 DEG.C	4 EXTEFF 24.1 %	PHASE ANGLES	DISPST 2.70 CM	
	5 TAVHTR 598.7 DEG.C	PADISP 82.0 DEG.		
	6 INTEFF 29.5 %	PAPRES -16.6 DEG.		
	8 AMPS 1134. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 24.2 HZ	PAMPC 866. KPA	T13REG***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.73 CM	T14REG 504.7 DEG.C
	11 QREG1 101. WATTS		PISTCP 2.69 CM	T15REG 375.3 DEG.C
	12 QREG2 125. WATTS		PDYNDB 938. KPA	T16REG***** DEG.C
	13 QREG3 35. WATTS		PDLCLR***** KPA	T17REG 356.8 DEG.C
	14 PFP 577.9 KPA/CM		PDLREG***** KPA	T18REG 350.9 DEG.C
	15 PFD 184.9 KPA/CM		PDLDIS***** KPA	T19REG 225.8 DEG.C
	16 NBEALE 0.00918			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:32:19.46 RDG 869

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.33 L/MIN	AMPS1 580. AMPS	PRESUP 4043. KPA	TGEXP 539.8 DEG.C	T01HTR 604.4 DEG.C
	AMPS2 603. AMPS			T02HTR 601.7 DEG.C
TWINDP 17.9 DEG.C	VOLTG 2.61 VOLTS	MEANBP 4004. KPA	TGREGH 535.1 DEG.C	T03HTR 608.6 DEG.C
TDLDP 1.73 DEG.C		MEANCP 4058. KPA	TGREGC 89.0 DEG.C	T04HTR 591.6 DEG.C
TWODPR 20.3 DEG.C			TGCOMP 50.1 DEG.C	T05HTR 597.9 DEG.C
			TGBOUN 36.7 DEG.C	T06HTR 599.1 DEG.C
				T07HTR 571.7 DEG.C
				T08HTR 597.1 DEG.C
				T09HTR 610.2 DEG.C
				T10HTR 603.4 DEG.C
				T11HTR 595.8 DEG.C
				T12HTR 595.9 DEG.C
HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.18 L/MIN	1 PWRIN 3088. WATTS	VX1HOR 0.3 CM/S	PWROUT 749. WATTS	
TWINCL 25.1 DEG.C	2 QCOOLR 1814. WATTS	VY1VER 7.3 CM/S	INDPWR 797. WATTS	
TDLCL 6.25 DEG.C	3 QDSHPT 522. WATTS		PISTST 3.17 CM	
TWOCLR 31.89 DEG.C	4 EXTEFF 24.3 %	PHASE ANGLES	DISPST 2.74 CM	
	5 TAVHTR 598.1 DEG.C	PADISP 83.1 DEG.		
	6 INTEFF 29.2 %	PAPRES -15.9 DEG.		
	8 AMPS 1183. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	FREQ 24.3 HZ	PAMPC 930. KPA	T13REG***** DEG.C
	10 QDISP 14. WATTS		DISPCP 2.80 CM	T14REG 502.7 DEG.C
	11 QREG1 104. WATTS		PISTCP 2.75 CM	T15REG 374.4 DEG.C
	12 QREG2 120. WATTS		PDYNDB 1013. KPA	T16REG***** DEG.C
	13 QREG3 35. WATTS		PDLCLR***** KPA	T17REG 359.6 DEG.C
	14 PFP 583.7 KPA/CM		PDLREG***** KPA	T18REG 355.3 DEG.C
	15 PFD 187.4 KPA/CM		PDLDIS***** KPA	T19REG 226.8 DEG.C
	16 NBEALE 0.00947			T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:42:19.46 RDG 870

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.31 L/MIN	AMPS1	470. AMPS	PRESUP	5524. KPA	TGEXP	560.7 DEG.C	T01HTR	598.8 DEG
		AMPS2	519. AMPS					T02HTR	607.4 DEG
TWINDP	17.9 DEG.C	VOLTG	2.18 VOLTS	MEANBP	5492. KPA	TGREGH	542.5 DEG.C	T03HTR	603.2 DEG
TDLDP	1.23 DEG.C			MEANCP	5529. KPA	TGREGC	88.0 DEG.C	T04HTR	597.9 DEG
TWODPR	19.8 DEG.C					TGCOMP	40.3 DEG.C	T05HTR	597.0 DEG
						TGBOUN	35.2 DEG.C	T06HTR	600.3 DEG
								T07HTR	576.4 DEG
								T08HTR	604.5 DEG
								T09HTR	608.9 DEG
								T10HTR	597.6 DEG
								T11HTR	596.7 DEG
								T12HTR	602.5 DEG
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.18 L/MIN	1 PWRIN	2152. WATTS	VX1HOR	0.2 CM/S	PWROUT	479. WATTS		
TWINCL	25.0 DEG.C	2 QCOOLR	1268. WATTS	VY1VER	5.2 CM/S	INDPWR	513. WATTS		
TDLCL	4.37 DEG.C	3 QDSHPT	368. WATTS			PISTST	1.78 CM		
TWOCLR	29.94 DEG.C	4 EXTEFF	22.3 %	PHASE ANGLES		DISPST	1.77 CM		
		5 TAVHTR	599.3 DEG.C	PADISP	80.8 DEG.			T13REG*****	DEG
		6 INTEFF	27.4 %	PAPRES	-19.9 DEG.			T14REG	508.9 DEG
		8 AMPS	989. AMPS			DYNAMIC CALCULATIONS		T15REG	376.9 DEG
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	733. KPA	T16REG*****	DEG
		10 QDISP	15. WATTS	FREQ	28.8 HZ	DISPCP	2.60 CM	T17REG	357.0 DEG
		11 QREG1	103. WATTS			PISTCP	2.48 CM	T18REG	353.7 DEG
		12 QREG2	126. WATTS			PDYNDP	825. KPA	T19REG	226.2 DEG
		13 QREG3	35. WATTS			PDLCLR*****	KPA		
		14 PFP	819.1 KPA/CM			PDLREG*****	KPA		
		15 PFD	286.5 KPA/CM			PDLDIS*****	KPA	T03HED*****	DEG
		16 NBEALE	0.00662						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:47:25.51 RDG 871

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.30 L/MIN	AMPS1	508. AMPS	PRESUP	5533. KPA	TGEXP	557.4 DEG.C	T01HTR	599.0 DEG
		AMPS2	548. AMPS					T02HTR	606.3 DEG
TWINDP	18.0 DEG.C	VOLTG	2.33 VOLTS	MEANBP	5503. KPA	TGREGH	540.7 DEG.C	T03HTR	603.6 DEG
TDLLP	1.42 DEG.C			MEANCP	5543. KPA	TGREGC	90.0 DEG.C	T04HTR	596.4 DEG
TWODPR	20.0 DEG.C					TGCOMP	42.5 DEG.C	T05HTR	596.6 DEG
						TGBOUN	35.2 DEG.C	T06HTR	599.5 DEG
								T07HTR	574.2 DEG
								T08HTR	603.6 DEG
								T09HTR	609.2 DEG
								T10HTR	597.7 DEG
								T11HTR	596.6 DEG
								T12HTR	600.9 DEG
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.17 L/MIN	1 PWRIN	2455. WATTS	VX1HOR	0.2 CM/S	PWROUT	566. WATTS		
TWINCL	24.9 DEG.C	2 QCOOLR	1427. WATTS	VY1VER	5.9 CM/S	INDPWR	604. WATTS		
TDLCL	4.93 DEG.C	3 QDSHPT	425. WATTS			PISTST	1.99 CM		
TWOCLR	30.43 DEG.C	4 EXTEFF	23.0 %	PHASE ANGLES		DISPST	1.91 CM		
		5 TAVHTR	598.6 DEG.C	PADISP	82.0 DEG.			T13REG	***** DEG
		6 INTEFF	28.4 %	PAPRES	-18.9 DEG.			T14REG	506.5 DEG
		8 AMPS	1056. AMPS			DYNAMIC CALCULATIONS		T15REG	375.4 DEG
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	807. KPA	T16REG	***** DEG
		10 QDISP	15. WATTS	FREQ	28.8 HZ	DISPCP	2.61 CM	T17REG	357.0 DEG
		11 QREG1	102. WATTS			PISTCP	2.51 CM	T18REG	352.1 DEG
		12 QREG2	125. WATTS			PDYNDP	913. KPA	T19REG	226.3 DEG
		13 QREG3	35. WATTS			PDLCLR	***** KPA		
		14 PFP	805.7 KPA/CM			PDLREG	***** KPA	T03HED	***** DEG
		15 PFD	276.0 KPA/CM			PDLDIS	***** KPA		
		16 NBEALE	0.00699						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:52:19.51 RDG 872

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.33 L/MIN	AMPS1	544. AMPS	PRESUP	5918. KPA	TGEXP	555.5 DEG.C	T01HTR	600.6 DEG.
TWINDP	18.0 DEG.C	AMPS2	576. AMPS	MEANBP	5502. KPA	TGREGH	539.7 DEG.C	T02HTR	606.9 DEG.
TDLDP	1.54 DEG.C	VOLTG	2.47 VOLTS	MEANCP	5549. KPA	TGREGC	91.9 DEG.C	T03HTR	606.2 DEG.
TWODPR	20.2 DEG.C					TGCOMP	45.1 DEG.C	T04HTR	596.3 DEG.
						TGBOUN	35.5 DEG.C	T05HTR	597.6 DEG.
								T06HTR	599.9 DEG.
								T07HTR	573.6 DEG.
								T08HTR	603.6 DEG.
								T09HTR	610.7 DEG.
								T10HTR	599.5 DEG.
								T11HTR	597.3 DEG.
								T12HTR	601.3 DEG.
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.16 L/MIN	1 PWRIN	2767. WATTS	VX1HOR	0.3 CM/S	PWROUT	654. WATTS	T13REG	***** DEG.
TWINCL	24.9 DEG.C	2 QCOOLR	1625. WATTS	VY1VER	6.6 CM/S	INDPWR	694. WATTS	T14REG	506.2 DEG.
TDLCL	5.63 DEG.C	3 QDSHPT	465. WATTS			PISTST	2.19 CM	T15REG	375.3 DEG.
TWOCLR	31.12 DEG.C	4 EXTEFF	23.6 %	PHASE ANGLES		DISPST	2.05 CM	T16REG	***** DEG.
		5 TAVHTR	599.4 DEG.C	PADISP	83.0 DEG.			T17REG	358.2 DEG.
		6 INTEFF	28.7 %	PAPRES	-17.9 DEG.			T18REG	354.7 DEG.
		8 AMPS	1120. AMPS			DYNAMIC CALCULATIONS		T19REG	228.1 DEG.
		9 QDISPG	3. WATTS	ENGINE SPEED		PAMPC	901. KPA	T03HED	***** DEG.
		10 QDISP	14. WATTS	FREQ	28.8 HZ	DISPCP	2.65 CM		
		11 QREG1	103. WATTS			PISTCP	2.53 CM		
		12 QREG2	123. WATTS			PDYNDB	1000. KPA		
		13 QREG3	35. WATTS			PDLCLR	***** KPA		
		14 PFP	813.2 KPA/CM			PDLREG	***** KPA		
		15 PFD	272.7 KPA/CM			PDLDIS	***** KPA		
		16 NBEALE	0.00733						

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 14:57:19.51 RDG 873

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.32 L/MIN	AMPS1 579. AMPS	PRESUP 5700. KPA	TGEXP 553.0 DEG.C	T01HTR 601.3 DEG.C
	AMPS2 601. AMPS			T02HTR 606.6 DEG.C
WINDP 18.0 DEG.C	VOLTG 2.60 VOLTS	MEANBP 5504. KPA	TGREGH 538.3 DEG.C	T03HTR 607.7 DEG.C
DLDP 1.74 DEG.C		MEANCP 5553. KPA	TGREGC 93.4 DEG.C	T04HTR 595.5 DEG.C
WODPR 20.4 DEG.C			TGCOMP 47.8 DEG.C	T05HTR 597.8 DEG.C
			TGBOUN 36.2 DEG.C	T06HTR 599.6 DEG.C
				T07HTR 572.5 DEG.C
				T08HTR 603.0 DEG.C
				T09HTR 611.2 DEG.C
				T10HTR 600.4 DEG.C
				T11HTR 597.3 DEG.C
				T12HTR 601.1 DEG.C
TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
LOCLR 4.20 L/MIN	1 PWRIN 3068. WATTS	VX1HOR 0.3 CM/S	PWROUT 735. WATTS	T13REG***** DEG.C
WINCL 25.1 DEG.C	2 QCCOOLR 1834. WATTS	VY1VER 7.2 CM/S	INDPWR 780. WATTS	T14REG 505.8 DEG.C
DLCL 6.30 DEG.C	3 QDSHPT 522. WATTS		PISTST 2.39 CM	T15REG 373.5 DEG.C
WOCCLR 31.83 DEG.C	4 EXTEFF 24.0 %	PHASE ANGLES	DISPST 2.17 CM	T16REG***** DEG.C
	5 TAVHTR 599.5 DEG.C	PADISP 83.7 DEG.		T17REG 359.0 DEG.C
	6 INTEFF 28.6 %	PAPRES -16.9 DEG.	DYNAMIC CALCULATIONS	T18REG 356.5 DEG.C
	8 AMPS 1180. AMPS	ENGINE SPEED	PAMPC 974. KPA	T19REG 229.0 DEG.C
	9 QDISPG 3. WATTS	FREQ 28.7 HZ	DISPCP 2.69 CM	
	10 QDISP 14. WATTS		PISTCP 2.57 CM	
	11 QREG1 103. WATTS		PDYNDB 1088. KPA	
	12 QREG2 121. WATTS		PDLCLR***** KPA	
	13 QREG3 34. WATTS		PDLREG***** KPA	
	14 PFP 805.5 KPA/CM		PDLDIS***** KPA	
	15 PFD 262.4 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00756			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 15:02:19.51 RDG 874

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.29 L/MIN	AMPS1 615. AMPS	PRESUP 5643. KPA	TGEXP 549.9 DEG.C	T01HTR 601.5 DEG.C
	AMPS2 629. AMPS			T02HTR 605.9 DEG.C
WINDP 18.0 DEG.C	VOLTG 2.74 VOLTS	MEANBP 5508. KPA	TGREGH 535.7 DEG.C	T03HTR 608.8 DEG.C
DLDP 1.97 DEG.C		MEANCP 5565. KPA	TGREGC 96.1 DEG.C	T04HTR 594.5 DEG.C
WODPR 20.6 DEG.C			TGCOMP 50.8 DEG.C	T05HTR 598.1 DEG.C
			TGBOUN 37.1 DEG.C	T06HTR 598.8 DEG.C
				T07HTR 571.0 DEG.C
				T08HTR 601.4 DEG.C
				T09HTR 611.9 DEG.C
				T10HTR 600.8 DEG.C
				T11HTR 596.3 DEG.C
				T12HTR 600.1 DEG.C
TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
LOCLR 4.20 L/MIN	1 PWRIN 3407. WATTS	VX1HOR 0.3 CM/S	PWROUT 817. WATTS	T13REG***** DEG.C
WINCL 25.1 DEG.C	2 QCCOOLR 2041. WATTS	VY1VER 7.9 CM/S	INDPWR 868. WATTS	T14REG 505.0 DEG.C
DLCL 7.01 DEG.C	3 QDSHPT 588. WATTS		PISTST 2.61 CM	T15REG 374.9 DEG.C
WOCCLR 32.72 DEG.C	4 EXTEFF 24.0 %	PHASE ANGLES	DISPST 2.31 CM	T16REG***** DEG.C
	5 TAVHTR 599.1 DEG.C	PADISP 84.2 DEG.		T17REG 359.3 DEG.C
	6 INTEFF 28.6 %	PAPRES -15.9 DEG.	DYNAMIC CALCULATIONS	T18REG 357.7 DEG.C
	8 AMPS 1243. AMPS	ENGINE SPEED	PAMPC 1063. KPA	T19REG 232.3 DEG.C
	9 QDISPG 3. WATTS	FREQ 28.7 HZ	DISPCP 2.71 CM	
	10 QDISP 14. WATTS		PISTCP 2.58 CM	
	11 QREG1 102. WATTS		PDYNDB 1138. KPA	
	12 QREG2 119. WATTS		PDLCLR***** KPA	
	13 QREG3 34. WATTS		PDLREG***** KPA	
	14 PFP 807.0 KPA/CM		PDLDIS***** KPA	
	15 PFD 253.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00773			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 15:07:19.52 RDG 875

FLUID HELIUM BAROM 14.332 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
LODP 4.30 L/MIN	AMPS1 649. AMPS	PRESUP 5623. KPA	TGEXP 547.9 DEG.C	T01HTR 602.4 DEG.C
	AMPS2 651. AMPS			T02HTR 606.0 DEG.C
WINDP 18.0 DEG.C	VOLTG 2.86 VOLTS	MEANBP 5500. KPA	TGREGH 535.2 DEG.C	T03HTR 610.2 DEG.C
DLDP 2.18 DEG.C		MEANCP 5562. KPA	TGREGC 98.1 DEG.C	T04HTR 594.5 DEG.C
WODPR 20.8 DEG.C			TGCOMP 54.0 DEG.C	T05HTR 599.1 DEG.C
			TGBOUN 38.5 DEG.C	T06HTR 599.1 DEG.C
				T07HTR 570.5 DEG.C
				T08HTR 601.9 DEG.C
				T09HTR 613.4 DEG.C
				T10HTR 601.8 DEG.C
				T11HTR 597.3 DEG.C
				T12HTR 600.4 DEG.C
TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
LOCLR 4.19 L/MIN	1 PWRIN 3721. WATTS	VX1HOR 0.3 CM/S	PWROUT 893. WATTS	T13REG***** DEG.C
WINCL 25.2 DEG.C	2 QCCOOLR 2244. WATTS	VY1VER 8.5 CM/S	INDPWR 947. WATTS	T14REG 505.3 DEG.C
DLCL 7.72 DEG.C	3 QDSHPT 652. WATTS		PISTST 2.79 CM	T15REG 376.4 DEG.C
WOCCLR 33.35 DEG.C	4 EXTEFF 24.0 %	PHASE ANGLES	DISPST 2.43 CM	T16REG***** DEG.C
	5 TAVHTR 599.7 DEG.C	PADISP 84.1 DEG.		T17REG 359.4 DEG.C
	6 INTEFF 28.5 %	PAPRES -15.3 DEG.	DYNAMIC CALCULATIONS	T18REG 357.9 DEG.C
	8 AMPS 1301. AMPS	ENGINE SPEED	PAMPC 1142. KPA	T19REG 234.3 DEG.C
	9 QDISPG 3. WATTS	FREQ 28.6 HZ	DISPCP 2.74 CM	
	10 QDISP 14. WATTS		PISTCP 2.59 CM	
	11 QREG1 100. WATTS		PDYNDB 1225. KPA	
	12 QREG2 120. WATTS		PDLCLR***** KPA	
	13 QREG3 34. WATTS		PDLREG***** KPA	
	14 PFP 811.8 KPA/CM		PDLDIS***** KPA	
	15 PFD 249.2 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00793			

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 15:12:19.57 RDG 876

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.29 L/MIN	AMPS1	678. AMPS	PRESUP	5655. KPA	TGEXP	544.3 DEG.C	T01HTR	603.1 DEG.C
		AMPS2	673. AMPS					T02HTR	604.9 DEG.C
TWINDP	18.0 DEG.C	VOLTG	2.97 VOLTS	MEANBP	5515. KPA	TGREGH	534.2 DEG.C	T03HTR	610.5 DEG.C
TDLDP	2.44 DEG.C			MEANCP	5587. KPA	TGREGC	101.1 DEG.C	T04HTR	593.5 DEG.C
TWODPR	21.1 DEG.C					TGCOMP	57.5 DEG.C	T05HTR	598.7 DEG.C
						TGBOUN	39.8 DEG.C	T06HTR	598.3 DEG.C
								T07HTR	569.2 DEG.C
								T08HTR	601.1 DEG.C
								T09HTR	614.2 DEG.C
								T10HTR	602.4 DEG.C
								T11HTR	597.6 DEG.C
								T12HTR	598.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.20 L/MIN	1 PWRIN	4017. WATTS	VX1HOR	0.4 CM/S	PWROUT	965. WATTS	T13REG*****	DEG.C
TWINCL	25.2 DEG.C	2 QCOOLR	2473. WATTS	VY1VER	9.0 CM/S	INDPWR	1019. WATTS	T14REG	503.6 DEG.C
TDLCL	8.49 DEG.C	3 QDSHPT	729. WATTS			PISTST	2.99 CM	T15REG	375.5 DEG.C
TWOCLR	34.20 DEG.C	4 EXTEFF	24.0 %	PHASE ANGLES		DISPST	2.55 CM	T16REG*****	DEG.C
		5 TAVHTR	599.3 DEG.C	PADISP	82.7 DEG.			T17REG	363.4 DEG.C
		6 INTEFF	28.1 %	PAPRES	-14.6 DEG.			T18REG	359.0 DEG.C
		8 AMPS	1351. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	235.8 DEG.C
		9 QDISPG	3. WATTS	FREQ	28.4 HZ	PAMPC	1211. KPA	T03HED*****	DEG.C
		10 QDISP	14. WATTS			DISPCP	2.79 CM		
		11 QREG1	100. WATTS			PISTCP	2.56 CM		
		12 QREG2	117. WATTS			PDYNDB	1263. KPA		
		13 QREG3	34. WATTS			PDLCLR*****	KPA		
		14 PFP	809.8 KPA/CM			PDLREG*****	KPA		
		15 PFD	240.9 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00802						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 15:17:19.57 RDG 877

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.27 L/MIN	AMPS1	710. AMPS	PRESUP	5643. KPA	TGEXP	541.6 DEG.C	T01HTR	605.2 DEG.C
		AMPS2	698. AMPS					T02HTR	602.6 DEG.C
TWINDP	18.0 DEG.C	VOLTG	3.10 VOLTS	MEANBP	5502. KPA	TGREGH	534.2 DEG.C	T03HTR	611.6 DEG.C
TDLDP	2.67 DEG.C			MEANCP	5583. KPA	TGREGC	104.1 DEG.C	T04HTR	591.6 DEG.C
TWODPR	21.4 DEG.C					TGCOMP	60.3 DEG.C	T05HTR	596.5 DEG.C
						TGBOUN	41.3 DEG.C	T06HTR	599.2 DEG.C
								T07HTR	565.8 DEG.C
								T08HTR	601.7 DEG.C
								T09HTR	613.7 DEG.C
								T10HTR	604.1 DEG.C
								T11HTR	598.4 DEG.C
								T12HTR	596.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	4369. WATTS	VX1HOR	0.4 CM/S	PWROUT	1049. WATTS	T13REG*****	DEG.C
TWINCL	25.3 DEG.C	2 QCOOLR	2740. WATTS	VY1VER	9.7 CM/S	INDPWR	1101. WATTS	T14REG	499.6 DEG.C
TDLCL	9.38 DEG.C	3 QDSHPT	792. WATTS			PISTST	3.20 CM	T15REG	371.9 DEG.C
TWOCLR	35.17 DEG.C	4 EXTEFF	24.0 %	PHASE ANGLES		DISPST	2.58 CM	T16REG*****	DEG.C
		5 TAVHTR	598.9 DEG.C	PADISP	84.0 DEG.			T17REG	367.7 DEG.C
		6 INTEFF	27.7 %	PAPRES	-13.7 DEG.			T18REG	363.3 DEG.C
		8 AMPS	1408. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	235.4 DEG.C
		9 QDISPG	3. WATTS	FREQ	28.4 HZ	PAMPC	1304. KPA	T03HED*****	DEG.C
		10 QDISP	14. WATTS			DISPCP	2.83 CM		
		11 QREG1	104. WATTS			PISTCP	2.53 CM		
		12 QREG2	111. WATTS			PDYNDB	1338. KPA		
		13 QREG3	33. WATTS			PDLCLR*****	KPA		
		14 PFP	811.9 KPA/CM			PDLREG*****	KPA		
		15 PFD	240.9 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00817						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/16/84 15:22:19.57 RDG 878

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.26 L/MIN	AMPS1	739. AMPS	PRESUP	5619. KPA	TGEXP	540.1 DEG.C	T01HTR	606.6 DEG.C
		AMPS2	720. AMPS					T02HTR	603.3 DEG.C
TWINDP	17.9 DEG.C	VOLTG	3.22 VOLTS	MEANBP	5511. KPA	TGREGH	533.9 DEG.C	T03HTR	613.5 DEG.C
TDLDP	2.93 DEG.C			MEANCP	5595. KPA	TGREGC	106.2 DEG.C	T04HTR	591.7 DEG.C
TWODPR	21.6 DEG.C					TGCOMP	63.7 DEG.C	T05HTR	598.5 DEG.C
						TGBOUN	42.5 DEG.C	T06HTR	599.9 DEG.C
								T07HTR	565.4 DEG.C
								T08HTR	602.5 DEG.C
								T09HTR	615.0 DEG.C
								T10HTR	605.5 DEG.C
								T11HTR	599.4 DEG.C
								T12HTR	597.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.23 L/MIN	1 PWRIN	4694. WATTS	VX1HOR	0.3 CM/S	PWROUT	1112. WATTS	T13REG*****	DEG.C
TWINCL	25.2 DEG.C	2 QCOOLR	3002. WATTS	VY1VER	10.2 CM/S	INDPWR	1165. WATTS	T14REG	501.2 DEG.C
TDLCL	10.22 DEG.C	3 QDSHPT	869. WATTS			PISTST	3.39 CM	T15REG	374.4 DEG.C
TWOCLR	35.97 DEG.C	4 EXTEFF	23.7 %	PHASE ANGLES		DISPST	2.62 CM	T16REG*****	DEG.C
		5 TAVHTR	599.9 DEG.C	PADISP	84.6 DEG.			T17REG	368.3 DEG.C
		6 INTEFF	27.0 %	PAPRES	-13.0 DEG.			T18REG	364.4 DEG.C
		8 AMPS	1458. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	238.5 DEG.C
		9 QDISPG	3. WATTS	FREQ	28.3 HZ	PAMPC	1393. KPA	T03HED*****	DEG.C
		10 QDISP	14. WATTS			DISPCP	2.81 CM		
		11 QREG1	102. WATTS			PISTCP	2.58 CM		
		12 QREG2	111. WATTS			PDYNDB	1425. KPA		
		13 QREG3	33. WATTS			PDLCLR*****	KPA		
		14 PFP	817.9 KPA/CM			PDLREG*****	KPA		
		15 PFD	240.5 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00817						

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 12:35:29.83 RDG 909

FLUID HELIUM BAROM 14.450 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

EAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.11 L/MIN	AMPS1 520. AMPS	PRESUP 7093. KPA	TGEXP 420.9 DEG.C	T01HTR 450.9 DEG.C
	AMPS2 521. AMPS			T02HTR 456.7 DEG.C
TWINDP 17.4 DEG.C	VOLTG 2.25 VOLTS	MEANBP 7011. KPA	TGREGH 402.3 DEG.C	T03HTR 458.1 DEG.C
TDLDP 1.22 DEG.C		MEANCP 7039. KPA	TGREGC 80.3 DEG.C	T04HTR 448.1 DEG.C
TWODPR 19.2 DEG.C			TGCOMP 43.3 DEG.C	T05HTR 445.7 DEG.C
			TGBOUN 32.5 DEG.C	T06HTR 452.8 DEG.C
				T07HTR 423.7 DEG.C
				T08HTR 455.8 DEG.C
				T09HTR 459.0 DEG.C
				T10HTR 450.0 DEG.C
				T11HTR 447.1 DEG.C
				T12HTR 454.2 DEG.C
EAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.14 L/MIN	1 PWRIN 2338. WATTS	VX1HOR 0.3 CM/S	PWROUT 463. WATTS	T13REG***** DEG.C
TWINCL 25.1 DEG.C	2 QCOOLR 1518. WATTS	VY1VER 6.2 CM/S	INDPWR 461. WATTS	T14REG 377.6 DEG.C
TDLCL 5.28 DEG.C	3 QDSHPT 350. WATTS		PISTST 1.80 CM	T15REG 281.7 DEG.C
TWOCRL 30.71 DEG.C	4 EXTEFF 19.8 %	PHASE ANGLES	DISPST 1.68 CM	T16REG***** DEG.C
	5 TAVHTR 450.2 DEG.C	PADISP 87.6 DEG.		T17REG 273.9 DEG.C
	6 INTEFF 23.4 %	PAPRES -13.1 DEG.	DYNAMIC CALCULATIONS	T18REG 283.1 DEG.C
	8 AMPS 1041. AMPS		PAMPC 910. KPA	T19REG 179.1 DEG.C
	9 QDISPG 2. WATTS	ENGINE SPEED	DISPCP 2.60 CM	
	10 QDISP 11. WATTS	FREQ 31.6 HZ	PISTCP 2.36 CM	
	11 QREG1 84. WATTS		PDYNDB 1063. KPA	
	12 QREG2 77. WATTS		PDLCLR***** KPA	
	13 QREG3 25. WATTS		PDLREG***** KPA	
	14 PFP 993.5 KPA/CM		PDLDIS***** KPA	
	15 PFD 245.7 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00452			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 12:40:20.83 RDG 910

FLUID HELIUM BAROM 14.445 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

EAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.10 L/MIN	AMPS1 556. AMPS	PRESUP 7096. KPA	TGEXP 418.9 DEG.C	T01HTR 450.2 DEG.C
	AMPS2 547. AMPS			T02HTR 456.6 DEG.C
TWINDP 17.3 DEG.C	VOLTG 2.38 VOLTS	MEANBP 6998. KPA	TGREGH 399.9 DEG.C	T03HTR 457.9 DEG.C
TDLDP 1.41 DEG.C		MEANCP 7032. KPA	TGREGC 82.9 DEG.C	T04HTR 447.5 DEG.C
TWODPR 19.4 DEG.C			TGCOMP 45.9 DEG.C	T05HTR 445.7 DEG.C
			TGBOUN 32.9 DEG.C	T06HTR 451.7 DEG.C
				T07HTR 422.6 DEG.C
				T08HTR 454.9 DEG.C
				T09HTR 459.7 DEG.C
				T10HTR 449.3 DEG.C
				T11HTR 446.4 DEG.C
				T12HTR 453.6 DEG.C
EAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.16 L/MIN	1 PWRIN 2625. WATTS	VX1HOR 0.3 CM/S	PWROUT 520. WATTS	T13REG***** DEG.C
TWINCL 25.2 DEG.C	2 QCOOLR 1723. WATTS	VY1VER 6.9 CM/S	INDPWR 517. WATTS	T14REG 377.0 DEG.C
TDLCL 5.97 DEG.C	3 QDSHPT 402. WATTS		PISTST 2.00 CM	T15REG 282.7 DEG.C
TWOCRL 31.38 DEG.C	4 EXTEFF 19.8 %	PHASE ANGLES	DISPST 1.78 CM	T16REG***** DEG.C
	5 TAVHTR 449.7 DEG.C	PADISP 88.3 DEG.		T17REG 273.5 DEG.C
	6 INTEFF 23.2 %	PAPRES -12.0 DEG.	DYNAMIC CALCULATIONS	T18REG 282.3 DEG.C
	8 AMPS 1103. AMPS		PAMPC 1009. KPA	T19REG 181.6 DEG.C
	9 QDISPG 2. WATTS	ENGINE SPEED	DISPCP 2.60 CM	
	10 QDISP 11. WATTS	FREQ 31.5 HZ	PISTCP 2.38 CM	
	11 QREG1 82. WATTS		PDYNDB 1125. KPA	
	12 QREG2 77. WATTS		PDLCLR***** KPA	
	13 QREG3 25. WATTS		PDLREG***** KPA	
	14 PFP 993.8 KPA/CM		PDLDIS***** KPA	
	15 PFD 235.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00460			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 12:45:23.83 RDG 911

FLUID HELIUM BAROM 14.445 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

EAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.09 L/MIN	AMPS1 591. AMPS	PRESUP 7094. KPA	TGEXP 417.1 DEG.C	T01HTR 450.9 DEG.C
	AMPS2 574. AMPS			T02HTR 457.2 DEG.C
TWINDP 17.0 DEG.C	VOLTG 2.51 VOLTS	MEANBP 6999. KPA	TGREGH 398.5 DEG.C	T03HTR 458.6 DEG.C
TDLDP 1.59 DEG.C		MEANCP 7031. KPA	TGREGC 85.7 DEG.C	T04HTR 447.3 DEG.C
TWODPR 19.2 DEG.C			TGCOMP 49.0 DEG.C	T05HTR 447.2 DEG.C
			TGBOUN 33.5 DEG.C	T06HTR 451.7 DEG.C
				T07HTR 421.8 DEG.C
				T08HTR 455.0 DEG.C
				T09HTR 461.0 DEG.C
				T10HTR 449.9 DEG.C
				T11HTR 446.7 DEG.C
				T12HTR 453.8 DEG.C
EAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.16 L/MIN	1 PWRIN 2928. WATTS	VX1HOR 0.3 CM/S	PWROUT 573. WATTS	T13REG***** DEG.C
TWINCL 25.2 DEG.C	2 QCOOLR 1933. WATTS	VY1VER 7.6 CM/S	INDPWR 567. WATTS	T14REG 377.4 DEG.C
TDLCL 6.70 DEG.C	3 QDSHPT 452. WATTS		PISTST 2.20 CM	T15REG 284.3 DEG.C
TWOCRL 32.22 DEG.C	4 EXTEFF 19.6 %	PHASE ANGLES	DISPST 1.89 CM	T16REG***** DEG.C
	5 TAVHTR 450.1 DEG.C	PADISP 88.6 DEG.		T17REG 274.7 DEG.C
	6 INTEFF 22.9 %	PAPRES -11.1 DEG.	DYNAMIC CALCULATIONS	T18REG 281.6 DEG.C
	8 AMPS 1164. AMPS		PAMPC 1102. KPA	T19REG 184.2 DEG.C
	9 QDISPG 2. WATTS	ENGINE SPEED	DISPCP 2.60 CM	
	10 QDISP 10. WATTS	FREQ 31.4 HZ	PISTCP 2.38 CM	
	11 QREG1 79. WATTS		PDYNDB 1213. KPA	
	12 QREG2 78. WATTS		PDLCLR***** KPA	
	13 QREG3 24. WATTS		PDLREG***** KPA	
	14 PFP 990.0 KPA/CM		PDLDIS***** KPA	
	15 PFD 224.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00462			

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 12:50:20.83 RDG 912

FLUID HELIUM		BAROM 14.445 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.08 L/MIN	AMPS1	625. AMPS	PRESUP	7102. KPA	TGEXP	414.5 DEG.C	T01HTR	450.9 DEG.C
		AMPS2	603. AMPS					T02HTR	456.8 DEG.C
TWINDP	16.9 DEG.C	VOLTG	2.65 VOLTS	MEANBP	7012. KPA	TGREGH	396.7 DEG.C	T03HTR	458.8 DEG.C
TDLDP	1.73 DEG.C			MEANCP	7050. KPA	TGREGC	88.6 DEG.C	T04HTR	446.5 DEG.C
TWODPR	19.4 DEG.C					TGCOMP	52.4 DEG.C	T05HTR	446.9 DEG.C
						TGBOUN	34.1 DEG.C	T06HTR	451.1 DEG.C
								T07HTR	420.2 DEG.C
								T08HTR	454.5 DEG.C
								T09HTR	461.6 DEG.C
								T10HTR	450.1 DEG.C
								T11HTR	446.4 DEG.C
								T12HTR	453.1 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.15 L/MIN	1 PWRIN	3251. WATTS	VX1HOR	0.4 CM/S	PWROUT	618. WATTS	T13REG*****	DEG.C
TWINCL	25.4 DEG.C	2 QCOOLR	2181. WATTS	VY1VER	8.2 CM/S	INDPWR	619. WATTS	T14REG	376.0 DEG.C
TDLCL	7.57 DEG.C	3 QDSHPT	491. WATTS			PISTST	2.40 CM	T15REG	284.5 DEG.C
TWOCLR	33.08 DEG.C	4 EXTEFF	19.0 %	PHASE ANGLES		DISPST	2.00 CM	T16REG*****	DEG.C
		5 TAVHTR	449.7 DEG.C	PADISP	88.7 DEG.			T17REG	273.5 DEG.C
		6 INTEFF	22.1 %	PAPRES	-10.3 DEG.			T18REG	281.9 DEG.C
		8 AMPS	1227. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	185.8 DEG.C
		9 QDISPG	2. WATTS	FREQ	31.3 HZ	PAMPC	1153. KPA	T03HED*****	DEG.C
		10 QDISP	10. WATTS			DISPCP	2.61 CM		
		11 QREG1	78. WATTS			PISTCP	2.39 CM		
		12 QREG2	76. WATTS			PDYNDB	1300. KPA		
		13 QREG3	24. WATTS			PDLCLR*****	KPA		
		14 PFP	949.7 KPA/CM			PDLREG*****	KPA		
		15 PFD	206.1 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00457						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 12:55:26.94 RDG 913

FLUID HELIUM		BAROM 14.445 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.09 L/MIN	AMPS1	658. AMPS	PRESUP	7106. KPA	TGEXP	411.3 DEG.C	T01HTR	451.7 DEG.C
		AMPS2	630. AMPS					T02HTR	455.0 DEG.C
TWINDP	16.9 DEG.C	VOLTG	2.78 VOLTS	MEANBP	6994. KPA	TGREGH	396.4 DEG.C	T03HTR	458.7 DEG.C
TDLDP	1.92 DEG.C			MEANCP	7034. KPA	TGREGC	92.0 DEG.C	T04HTR	445.0 DEG.C
TWODPR	19.5 DEG.C					TGCOMP	56.7 DEG.C	T05HTR	445.5 DEG.C
						TGBOUN	35.0 DEG.C	T06HTR	451.1 DEG.C
								T07HTR	417.4 DEG.C
								T08HTR	454.4 DEG.C
								T09HTR	461.7 DEG.C
								T10HTR	450.6 DEG.C
								T11HTR	447.0 DEG.C
								T12HTR	451.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.18 L/MIN	1 PWRIN	3583. WATTS	VX1HOR	0.4 CM/S	PWROUT	663. WATTS	T13REG*****	DEG.C
TWINCL	25.4 DEG.C	2 QCOOLR	2427. WATTS	VY1VER	8.8 CM/S	INDPWR	662. WATTS	T14REG	373.1 DEG.C
TDLCL	8.37 DEG.C	3 QDSHPT	546. WATTS			PISTST	2.60 CM	T15REG	281.1 DEG.C
TWOCLR	34.05 DEG.C	4 EXTEFF	18.5 %	PHASE ANGLES		DISPST	2.11 CM	T16REG*****	DEG.C
		5 TAVHTR	449.1 DEG.C	PADISP	87.4 DEG.			T17REG	276.3 DEG.C
		6 INTEFF	21.5 %	PAPRES	-9.6 DEG.			T18REG	280.5 DEG.C
		8 AMPS	1288. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	184.5 DEG.C
		9 QDISPG	2. WATTS	FREQ	31.1 HZ	PAMPC	1206. KPA	T03HED*****	DEG.C
		10 QDISP	10. WATTS			DISPCP	2.62 CM		
		11 QREG1	78. WATTS			PISTCP	2.39 CM		
		12 QREG2	75. WATTS			PDYNDB	1363. KPA		
		13 QREG3	24. WATTS			PDLCLR*****	KPA		
		14 PFP	922.8 KPA/CM			PDLREG*****	KPA		
		15 PFD	191.0 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00456						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 13:00:20.94 RDG 914

FLUID HELIUM		BAROM 14.445 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.09 L/MIN	AMPS1	690. AMPS	PRESUP	7103. KPA	TGEXP	409.9 DEG.C	T01HTR	452.2 DEG.C
		AMPS2	659. AMPS					T02HTR	455.1 DEG.C
TWINDP	17.0 DEG.C	VOLTG	2.92 VOLTS	MEANBP	7012. KPA	TGREGH	395.8 DEG.C	T03HTR	459.6 DEG.C
TDLDP	2.15 DEG.C			MEANCP	7054. KPA	TGREGC	95.1 DEG.C	T04HTR	444.5 DEG.C
TWODPR	19.8 DEG.C					TGCOMP	60.7 DEG.C	T05HTR	446.0 DEG.C
						TGBOUN	35.9 DEG.C	T06HTR	451.0 DEG.C
								T07HTR	416.4 DEG.C
								T08HTR	454.3 DEG.C
								T09HTR	462.6 DEG.C
								T10HTR	451.2 DEG.C
								T11HTR	447.2 DEG.C
								T12HTR	451.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.18 L/MIN	1 PWRIN	3934. WATTS	VX1HOR	0.4 CM/S	PWROUT	706. WATTS	T13REG*****	DEG.C
TWINCL	25.5 DEG.C	2 QCOOLR	2718. WATTS	VY1VER	9.4 CM/S	INDPWR	702. WATTS	T14REG	373.5 DEG.C
TDLCL	9.38 DEG.C	3 QDSHPT	612. WATTS			PISTST	2.80 CM	T15REG	282.5 DEG.C
TWOCLR	35.03 DEG.C	4 EXTEFF	18.0 %	PHASE ANGLES		DISPST	2.22 CM	T16REG*****	DEG.C
		5 TAVHTR	449.3 DEG.C	PADISP	87.5 DEG.			T17REG	278.3 DEG.C
		6 INTEFF	20.6 %	PAPRES	-8.9 DEG.			T18REG	281.6 DEG.C
		8 AMPS	1349. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS		T19REG	187.5 DEG.C
		9 QDISPG	2. WATTS	FREQ	31.1 HZ	PAMPC	1285. KPA	T03HED*****	DEG.C
		10 QDISP	10. WATTS			DISPCP	2.69 CM		
		11 QREG1	76. WATTS			PISTCP	2.38 CM		
		12 QREG2	75. WATTS			PDYNDB	1488. KPA		
		13 QREG3	23. WATTS			PDLCLR*****	KPA		
		14 PFP	914.1 KPA/CM			PDLREG*****	KPA		
		15 PFD	179.3 KPA/CM			PDLDIS*****	KPA		
		16 NBEALE	0.00451						

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 13:05:21.01 RDG 915

FLUID HELIUM BAROM 14.445 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.06 L/MIN	POWER IN AMPS1 737. AMPS AMPS2 695. AMPS VOLTG 3.11 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7106. KPA MEANBP 6998. KPA MEANCP 7043. KPA	GAS TEMPERATURES TGEXP 405.7 DEG.C TGREGH 394.6 DEG.C TGREGC 98.1 DEG.C TGCOMP 64.5 DEG.C TGBOUN 36.9 DEG.C	SURFACE TEMPERATURES T01HTR 454.4 DEG.C T02HTR 456.2 DEG.C T03HTR 461.7 DEG.C T04HTR 445.3 DEG.C T05HTR 446.3 DEG.C T06HTR 452.6 DEG.C T07HTR 415.5 DEG.C T08HTR 455.7 DEG.C T09HTR 465.3 DEG.C T10HTR 453.1 DEG.C T11HTR 448.9 DEG.C T12HTR 452.0 DEG.C
AT TO COOLER FLOCLR 4.20 L/MIN TWINCL 25.5 DEG.C TDLCL 10.29 DEG.C TWOCLR 35.91 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 4448. WATTS 2 QCOOLR 2998. WATTS 3 QDSHPT 651. WATTS 4 EXTEFF 16.7 % 5 TAVHTR 450.6 DEG.C 6 INTEFF 19.9 % 8 AMPS 1432. AMPS 9 QDISPG 2. WATTS 10 QDISP 10. WATTS 11 QREG1 74. WATTS 12 QREG2 74. WATTS 13 QREG3 23. WATTS 14 PFP 923.9 KPA/CM 15 PFD 168.2 KPA/CM 16 NBEALE 0.00445	VIBRATION VXIHOR 0.5 CM/S VYIVER 10.0 CM/S PHASE ANGLES PADISP 87.4 DEG. PAPRES -8.1 DEG. ENGINE SPEED FREQ 31.0 HZ	REMOTE CALCULATIONS PWROUT 743. WATTS INDPWR 732. WATTS PISTST 2.99 CM DISPST 2.32 CM DYNAMIC CALCULATIONS PAMPC 1388. KPA DISPCP 2.71 CM PISTCP 2.35 CM PDYNDB 1538. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 370.1 DEG.C T15REG 280.8 DEG.C T16REG***** DEG.C T17REG 278.1 DEG.C T18REG 279.1 DEG.C T19REG 188.4 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/19/84 13:17:24.01 RDG 916

FLUID HELIUM BAROM 14.445 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.03 L/MIN	POWER IN AMPS1 742. AMPS AMPS2 717. AMPS VOLTG 3.19 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7110. KPA MEANBP 7016. KPA MEANCP 7059. KPA	GAS TEMPERATURES TGEXP 406.1 DEG.C TGREGH 395.8 DEG.C TGREGC 102.3 DEG.C TGCOMP 69.4 DEG.C TGBOUN 39.1 DEG.C	SURFACE TEMPERATURES T01HTR 455.6 DEG.C T02HTR 456.5 DEG.C T03HTR 462.7 DEG.C T04HTR 445.1 DEG.C T05HTR 447.9 DEG.C T06HTR 453.1 DEG.C T07HTR 415.3 DEG.C T08HTR 456.4 DEG.C T09HTR 466.4 DEG.C T10HTR 454.2 DEG.C T11HTR 449.9 DEG.C T12HTR 452.4 DEG.C
AT TO COOLER FLOCLR 4.20 L/MIN TWINCL 25.6 DEG.C TDLCL 11.58 DEG.C TWOCLR 37.42 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 4657. WATTS 2 QCOOLR 3372. WATTS 3 QDSHPT 730. WATTS 4 EXTEFF 16.4 % 5 TAVHTR 451.3 DEG.C 6 INTEFF 18.5 % 8 AMPS 1459. AMPS 9 QDISPG 2. WATTS 10 QDISP 10. WATTS 11 QREG1 72. WATTS 12 QREG2 75. WATTS 13 QREG3 23. WATTS 14 PFP 990.6 KPA/CM 15 PFD 171.3 KPA/CM 16 NBEALE 0.00428	VIBRATION VXIHOR 0.4 CM/S VYIVER 10.5 CM/S PHASE ANGLES PADISP 87.0 DEG. PAPRES -7.5 DEG. ENGINE SPEED FREQ 31.0 HZ	REMOTE CALCULATIONS PWROUT 764. WATTS INDPWR 754. WATTS PISTST 3.19 CM DISPST 2.43 CM DYNAMIC CALCULATIONS PAMPC 1585. KPA DISPCP 2.72 CM PISTCP 2.23 CM PDYNDB 1600. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 374.1 DEG.C T15REG 284.6 DEG.C T16REG***** DEG.C T17REG 282.5 DEG.C T18REG 281.8 DEG.C T19REG 193.1 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:25:18.80 RDG 921

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.17 L/MIN	POWER IN AMPS1 519. AMPS AMPS2 549. AMPS VOLTG 2.35 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7379. KPA MEANBP 7012. KPA MEANCP 7027. KPA	GAS TEMPERATURES TGEXP 466.3 DEG.C TGREGH 449.3 DEG.C TGREGC 85.6 DEG.C TGCOMP 43.1 DEG.C TGBOUN 30.7 DEG.C	SURFACE TEMPERATURES T01HTR 501.8 DEG.C T02HTR 506.5 DEG.C T03HTR 508.2 DEG.C T04HTR 493.0 DEG.C T05HTR 496.5 DEG.C T06HTR 504.1 DEG.C T07HTR 471.9 DEG.C T08HTR 506.7 DEG.C T09HTR 509.8 DEG.C T10HTR 501.2 DEG.C T11HTR 497.9 DEG.C T12HTR 504.1 DEG.C
AT TO COOLER FLOCLR 4.23 L/MIN TWINCL 25.0 DEG.C TDLCL 5.45 DEG.C TWOCLR 30.63 DEG.C	CALCULATED PARAMETERS 1 PWIRIN 2513. WATTS 2 QCOOLR 1598. WATTS 3 QDSHPT 385. WATTS 4 EXTEFF 21.1 % 5 TAVHTR 500.6 DEG.C 6 INTEFF 25.0 % 8 AMPS 1069. AMPS 9 QDISPG 2. WATTS 10 QDISP 12. WATTS 11 QREG1 95. WATTS 12 QREG2 89. WATTS 13 QREG3 28. WATTS 14 PFP 1025.4 KPA/CM 15 PFD 288.3 KPA/CM 16 NBEALE 0.00515	VIBRATION VXIHOR 0.3 CM/S VYIVER 6.5 CM/S PHASE ANGLES PADISP 85.9 DEG. PAPRES -15.1 DEG. ENGINE SPEED FREQ 31.9 HZ	REMOTE CALCULATIONS PWROUT 531. WATTS INDPWR 567. WATTS PISTST 1.80 CM DISPST 1.69 CM DYNAMIC CALCULATIONS PAMPC 935. KPA DISPCP 2.59 CM PISTCP 2.34 CM PDYNDB 1075. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 420.1 DEG.C T15REG 310.7 DEG.C T16REG***** DEG.C T17REG 304.2 DEG.C T18REG 310.4 DEG.C T19REG 193.8 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:30:18.80 RDG 922

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:35:30.80 RDG 923

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:40:18.80 RDG 924

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APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:45:18.80 RDG 925

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.17 L/MIN	POWER IN AMPS1 680. AMPS AMPS2 653. AMPS VOLTG 2.92 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7177. KPA MEANBP 7005. KPA MEANCP 7024. KPA	GAS TEMPERATURES TGEXP 456.9 DEG.C TGREGH 442.7 DEG.C TGREGC 96.4 DEG.C TGCOMP 57.0 DEG.C TGBOUN 34.6 DEG.C	SURFACE TEMPERATURES T01HTR 502.9 DEG.C T02HTR 505.9 DEG.C T03HTR 510.5 DEG.C T04HTR 495.3 DEG.C T05HTR 495.7 DEG.C T06HTR 501.7 DEG.C T07HTR 467.0 DEG.C T08HTR 505.0 DEG.C T09HTR 513.5 DEG.C T10HTR 501.9 DEG.C T11HTR 497.7 DEG.C T12HTR 501.4 DEG.C
TWINDP 16.6 DEG.C TDLDP 2.22 DEG.C TWODPR 19.4 DEG.C				
AT TO COOLER FLOCLR 4.25 L/MIN TWINCL 25.2 DEG.C TDLCL 8.69 DEG.C TWOCLR 33.98 DEG.C	CALCULATED PARAMETERS 1 PWRIN 3888. WATTS 2 QCOOLR 2566. WATTS 3 QDSHPT 644. WATTS 4 EXTEFF 20.4 % 5 TAVHTR 499.9 DEG.C 6 INTEFF 23.6 % 8 AMPS 1333. AMPS 9 QDISPG 2. WATTS 10 QDISP 11. WATTS 11 QREG1 89. WATTS 12 QREG2 85. WATTS 13 QREG3 27. WATTS 14 PFP 1012.7 KPA/CM 15 PFD 242.5 KPA/CM 16 NBEALE 0.00542	VIBRATION VXIHOR 0.4 CM/S VYIVER 9.0 CM/S PHASE ANGLES PADISP 85.4 DEG. PAPRES -11.4 DEG. ENGINE SPEED FREQ 31.4 HZ	REMOTE CALCULATIONS PWROUT 794. WATTS INDPWR 841. WATTS PISTST 2.60 CM DISPST 2.15 CM DYNAMIC CALCULATIONS PAMPC 1319. KPA DISPCP 2.71 CM PISTCP 2.35 CM PDYNDB 1450. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 416.5 DEG.C T15REG 311.5 DEG.C T16REG***** DEG.C T17REG 305.9 DEG.C T18REG 311.5 DEG.C T19REG 201.5 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:50:21.80 RDG 926

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.15 L/MIN	POWER IN AMPS1 715. AMPS AMPS2 684. AMPS VOLTG 3.07 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7167. KPA MEANBP 6996. KPA MEANCP 7021. KPA	GAS TEMPERATURES TGEXP 455.9 DEG.C TGREGH 442.1 DEG.C TGREGC 99.9 DEG.C TGCOMP 61.3 DEG.C TGBOUN 36.1 DEG.C	SURFACE TEMPERATURES T01HTR 504.7 DEG.C T02HTR 506.3 DEG.C T03HTR 512.2 DEG.C T04HTR 495.1 DEG.C T05HTR 496.8 DEG.C T06HTR 502.3 DEG.C T07HTR 466.1 DEG.C T08HTR 505.6 DEG.C T09HTR 515.0 DEG.C T10HTR 503.6 DEG.C T11HTR 498.8 DEG.C T12HTR 501.8 DEG.C
TWINDP 16.6 DEG.C TDLDP 2.53 DEG.C TWODPR 19.7 DEG.C				
AT TO COOLER FLOCLR 4.25 L/MIN TWINCL 25.2 DEG.C TDLCL 9.82 DEG.C TWOCLR 35.08 DEG.C	CALCULATED PARAMETERS 1 PWRIN 4293. WATTS 2 QCOOLR 2895. WATTS 3 QDSHPT 730. WATTS 4 EXTEFF 19.9 % 5 TAVHTR 500.7 DEG.C 6 INTEFF 22.8 % 8 AMPS 1398. AMPS 9 QDISPG 2. WATTS 10 QDISP 11. WATTS 11 QREG1 87. WATTS 12 QREG2 86. WATTS 13 QREG3 26. WATTS 14 PFP 985.0 KPA/CM 15 PFD 224.7 KPA/CM 16 NBEALE 0.00541	VIBRATION VXIHOR 0.4 CM/S VYIVER 9.7 CM/S PHASE ANGLES PADISP 85.4 DEG. PAPRES -10.6 DEG. ENGINE SPEED FREQ 31.3 HZ	REMOTE CALCULATIONS PWROUT 853. WATTS INDPWR 901. WATTS PISTST 2.80 CM DISPST 2.27 CM DYNAMIC CALCULATIONS PAMPC 1383. KPA DISPCP 2.71 CM PISTCP 2.31 CM PDYNDB 1525. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 415.8 DEG.C T15REG 311.0 DEG.C T16REG***** DEG.C T17REG 305.9 DEG.C T18REG 310.3 DEG.C T19REG 202.5 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 14:55:18.80 RDG 927

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING FLODP 4.14 L/MIN	POWER IN AMPS1 747. AMPS AMPS2 710. AMPS VOLTG 3.20 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7398. KPA MEANBP 7009. KPA MEANCP 7037. KPA	GAS TEMPERATURES TGEXP 455.3 DEG.C TGREGH 440.8 DEG.C TGREGC 103.2 DEG.C TGCOMP 65.1 DEG.C TGBOUN 37.6 DEG.C	SURFACE TEMPERATURES T01HTR 504.7 DEG.C T02HTR 505.4 DEG.C T03HTR 512.5 DEG.C T04HTR 494.0 DEG.C T05HTR 496.1 DEG.C T06HTR 501.8 DEG.C T07HTR 464.1 DEG.C T08HTR 505.1 DEG.C T09HTR 515.2 DEG.C T10HTR 503.9 DEG.C T11HTR 498.7 DEG.C T12HTR 500.9 DEG.C
TWINDP 16.7 DEG.C TDLDP 2.77 DEG.C TWODPR 19.9 DEG.C				
AT TO COOLER FLOCLR 4.27 L/MIN TWINCL 25.2 DEG.C TDLCL 10.76 DEG.C TWOCLR 36.07 DEG.C	CALCULATED PARAMETERS 1 PWRIN 4664. WATTS 2 QCOOLR 3188. WATTS 3 QDSHPT 798. WATTS 4 EXTEFF 19.2 % 5 TAVHTR 500.2 DEG.C 6 INTEFF 22.0 % 8 AMPS 1457. AMPS 9 QDISPG 2. WATTS 10 QDISP 11. WATTS 11 QREG1 85. WATTS 12 QREG2 85. WATTS 13 QREG3 26. WATTS 14 PFP 1014.4 KPA/CM 15 PFD 218.5 KPA/CM 16 NBEALE 0.00533	VIBRATION VXIHOR 0.4 CM/S VYIVER 10.3 CM/S PHASE ANGLES PADISP 85.2 DEG. PAPRES -9.8 DEG. ENGINE SPEED FREQ 31.3 HZ	REMOTE CALCULATIONS PWROUT 897. WATTS INDPWR 945. WATTS PISTST 2.99 CM DISPST 2.36 CM DYNAMIC CALCULATIONS PAMPC 1516. KPA DISPCP 2.72 CM PISTCP 2.25 CM PDYNDB 1588. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 414.4 DEG.C T15REG 310.7 DEG.C T16REG***** DEG.C T17REG 308.2 DEG.C T18REG 309.9 DEG.C T19REG 204.6 DEG.C T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:00:18.80 RDG 928

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.14 L/MIN	AMPS1 775. AMPS	PRESUP 7387. KPA	TGEXP 454.0 DEG.C	T01HTR 505.2 DEG.C
	AMPS2 738. AMPS			T02HTR 504.5 DEG.C
TWINDP 16.6 DEG.C	VOLTG 3.34 VOLTS	MEANBP 7007. KPA	TGREGH 439.9 DEG.C	T03HTR 513.0 DEG.C
TDLDP 3.02 DEG.C		MEANCP 7044. KPA	TGREGC 107.4 DEG.C	T04HTR 493.0 DEG.C
TWODPR 20.2 DEG.C			TGCOMP 70.1 DEG.C	T05HTR 495.4 DEG.C
			TGBOUN 39.0 DEG.C	T06HTR 501.6 DEG.C
				T07HTR 462.0 DEG.C
				T08HTR 505.0 DEG.C
				T09HTR 515.9 DEG.C
				T10HTR 504.0 DEG.C
				T11HTR 499.0 DEG.C
				T12HTR 500.0 DEG.C
				T13REG***** DEG.C
				T14REG 412.3 DEG.C
				T15REG 309.5 DEG.C
				T16REG***** DEG.C
				T17REG 310.4 DEG.C
				T18REG 309.5 DEG.C
				T19REG 206.6 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.28 L/MIN	1 PWRIN 5062. WATTS	VX1HOR 0.4 CM/S	PWROUT 956. WATTS	
TWINCL 25.3 DEG.C	2 QCOOLR 3545. WATTS	VY1VER10.9 CM/S	INDPWR 985. WATTS	
TDLCL 11.95 DEG.C	3 QDSHPT 869. WATTS		PISTST 3.20 CM	
TWOCLR 37.41 DEG.C	4 EXTEFF 18.9 %	PHASE ANGLES	DISPST 2.47 CM	
	5 TAVHTR 499.9 DEG.C	PADISP 84.9 DEG.		
	6 INTEFF 21.2 %	PAPRES -9.1 DEG.		
	8 AMPS 1514. AMPS		DYNAMIC CALCULATIONS	
	9 QDISPG 2. WATTS	ENGINE SPEED	PAMPC 1624. KPA	
	10 QDISP 11. WATTS	FREQ 31.2 HZ	DISPCP 2.74 CM	
	11 QREG1 83. WATTS		PISTCP 2.19 CM	
	12 QREG2 83. WATTS		PDYND 1638. KPA	
	13 QREG3 26. WATTS		PDLCLR***** KPA	
	14 PFP 1018.2 KPA/CM		PDLREG***** KPA	
	15 PFD 207.8 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00533			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:02:57.83 RDG 929

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.14 L/MIN	AMPS1 811. AMPS	PRESUP 7383. KPA	TGEXP 452.2 DEG.C	T01HTR 504.4 DEG.C
	AMPS2 767. AMPS			T02HTR 503.5 DEG.C
TWINDP 16.6 DEG.C	VOLTG 3.49 VOLTS	MEANBP 7005. KPA	TGREGH 437.7 DEG.C	T03HTR 512.8 DEG.C
TDLDP 3.14 DEG.C		MEANCP 7040. KPA	TGREGC 111.0 DEG.C	T04HTR 491.5 DEG.C
TWODPR 20.3 DEG.C			TGCOMP 74.4 DEG.C	T05HTR 495.2 DEG.C
			TGBOUN 39.8 DEG.C	T06HTR 500.3 DEG.C
				T07HTR 459.6 DEG.C
				T08HTR 504.0 DEG.C
				T09HTR 515.9 DEG.C
				T10HTR 503.5 DEG.C
				T11HTR 498.1 DEG.C
				T12HTR 499.0 DEG.C
				T13REG***** DEG.C
				T14REG 410.6 DEG.C
				T15REG 310.2 DEG.C
				T16REG***** DEG.C
				T17REG 311.1 DEG.C
				T18REG 310.3 DEG.C
				T19REG 209.5 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.28 L/MIN	1 PWRIN 5505. WATTS	VX1HOR 0.4 CM/S	PWROUT 955. WATTS	
TWINCL 25.3 DEG.C	2 QCOOLR 3855. WATTS	VY1VER11.5 CM/S	INDPWR 998. WATTS	
TDLCL 12.99 DEG.C	3 QDSHPT 903. WATTS		PISTST 3.39 CM	
TWOCLR 38.37 DEG.C	4 EXTEFF 17.3 %	PHASE ANGLES	DISPST 2.61 CM	
	5 TAVHTR 499.0 DEG.C	PADISP 84.7 DEG.		
	6 INTEFF 19.8 %	PAPRES -8.2 DEG.		
	8 AMPS 1578. AMPS		DYNAMIC CALCULATIONS	
	9 QDISPG 2. WATTS	ENGINE SPEED	PAMPC 1732. KPA	
	10 QDISP 11. WATTS	FREQ 31.2 HZ	DISPCP 2.75 CM	
	11 QREG1 82. WATTS		PISTCP 2.07 CM	
	12 QREG2 81. WATTS		PDYND 1813. KPA	
	13 QREG3 25. WATTS		PDLCLR***** KPA	
	14 PFP 1023.6 KPA/CM		PDLREG***** KPA	
	15 PFD 189.9 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00501			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:13:21.83 RDG 930

FLUID HELIUM BAROM 14.268 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.16 L/MIN	AMPS1 543. AMPS	PRESUP 7173. KPA	TGEXP 654.3 DEG.C	T01HTR 551.4 DEG.C
	AMPS2 549. AMPS			T02HTR 555.6 DEG.C
TWINDP 16.6 DEG.C	VOLTG 2.42 VOLTS	MEANBP 7011. KPA	TGREGH 495.7 DEG.C	T03HTR 558.4 DEG.C
TDLDP 1.96 DEG.C		MEANCP 7026. KPA	TGREGC 89.6 DEG.C	T04HTR 546.3 DEG.C
TWODPR 19.0 DEG.C			TGCOMP 45.3 DEG.C	T05HTR 545.8 DEG.C
			TGBOUN 37.5 DEG.C	T06HTR 553.3 DEG.C
				T07HTR 520.2 DEG.C
				T08HTR 556.0 DEG.C
				T09HTR 558.5 DEG.C
				T10HTR 550.9 DEG.C
				T11HTR 547.3 DEG.C
				T12HTR 553.1 DEG.C
				T13REG***** DEG.C
				T14REG 462.4 DEG.C
				T15REG 340.6 DEG.C
				T16REG***** DEG.C
				T17REG 332.7 DEG.C
				T18REG 339.2 DEG.C
				T19REG 210.2 DEG.C
				T03HED***** DEG.C

HEAT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.26 L/MIN	1 PWRIN 2637. WATTS	VX1HOR 0.3 CM/S	PWROUT 590. WATTS	
TWINCL 24.9 DEG.C	2 QCOOLR 1742. WATTS	VY1VER 6.5 CM/S	INDPWR 630. WATTS	
TDLCL 5.89 DEG.C	3 QDSHPT 566. WATTS		PISTST 1.79 CM	
TWOCLR 31.05 DEG.C	4 EXTEFF 22.4 %	PHASE ANGLES	DISPST 1.70 CM	
	5 TAVHTR 549.7 DEG.C	PADISP 83.9 DEG.		
	6 INTEFF 25.3 %	PAPRES -16.7 DEG.		
	8 AMPS 1092. AMPS		DYNAMIC CALCULATIONS	
	9 QDISPG 3. WATTS	ENGINE SPEED	PAMPC 1058. KPA	
	10 QDISP 17. WATTS	FREQ 32.2 HZ	DISPCP 2.68 CM	
	11 QREG1 105. WATTS		PISTCP 2.40 CM	
	12 QREG2 100. WATTS		PDYND 1075. KPA	
	13 QREG3 31. WATTS		PDLCLR***** KPA	
	14 PFP 1171.7 KPA/CM		PDLREG***** KPA	
	15 PFD 357.9 KPA/CM		PDLDIS***** KPA	
	16 NBEALE 0.00571			

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:18:18.83 RDG 931

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.17 L/MIN	AMPS1 588. AMPS	PRESUP 7404. KPA	TGEXP 695.6 DEG.C	T01HTR 551.1 DEG.C
	AMPS2 587. AMPS			T02HTR 557.4 DEG.C
TWINDP 16.6 DEG.C	VOLTG 2.60 VOLTS	MEANBP 7005. KPA	TGREGH 492.7 DEG.C	T03HTR 559.3 DEG.C
TDLDP 2.11 DEG.C		MEANCP 7032. KPA	TGREGC 92.4 DEG.C	T04HTR 546.8 DEG.C
TWODPR 19.3 DEG.C			TGCOMP 47.9 DEG.C	T05HTR 547.3 DEG.C
			TGBOUN 37.2 DEG.C	T06HTR 551.9 DEG.C
				T07HTR 521.0 DEG.C
				T08HTR 555.5 DEG.C
				T09HTR 561.3 DEG.C
				T10HTR 550.7 DEG.C
				T11HTR 547.2 DEG.C
				T12HTR 553.6 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.24 L/MIN	1 PWRIN 3053. WATTS	VX1HOR 0.3 CM/S	PWROUT 690. WATTS	T13REG***** DEG.C
TWINCL 24.8 DEG.C	2 QCOOLR 1981. WATTS	VY1VER 7.2 CM/S	INDPWR 734. WATTS	T14REG 463.5 DEG.C
TDLCL 6.73 DEG.C	3 QDSHPT 612. WATTS		PISTST 2.01 CM	T15REG 344.0 DEG.C
TWOCLR 31.75 DEG.C	4 EXTEFF 22.6 %	PHASE ANGLES	DISPST 1.86 CM	T16REG***** DEG.C
	5 TAVHTR 550.3 DEG.C	PADISP 84.5 DEG.		T17REG 331.1 DEG.C
	6 INTEFF 25.8 %	PAPRES -15.7 DEG.	DYNAMIC CALCULATIONS	T18REG 339.3 DEG.C
	8 AMPS 1174. AMPS	ENGINE SPEED	PAMPC 1043. KPA	T19REG 215.5 DEG.C
	9 QDISPG 4. WATTS	FREQ 32.0 HZ	DISPCP 2.66 CM	
	10 QDISP 18. WATTS		PISTCP 2.41 CM	
	11 QREG1 100. WATTS		PDYNDB 1200. KPA	
	12 QREG2 101. WATTS		PDLCLR***** KPA	
	13 QREG3 31. WATTS		PDLREG***** KPA	
	14 PFP 1028.5 KPA/CM		PDLDIS***** KPA	
	15 PFD 305.5 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00597			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:23:18.86 RDG 932

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.18 L/MIN	AMPS1 623. AMPS	PRESUP 7308. KPA	TGEXP 713.1 DEG.C	T01HTR 551.6 DEG.C
	AMPS2 613. AMPS			T02HTR 556.6 DEG.C
TWINDP 16.6 DEG.C	VOLTG 2.74 VOLTS	MEANBP 7019. KPA	TGREGH 490.5 DEG.C	T03HTR 559.8 DEG.C
TDLDP 2.28 DEG.C		MEANCP 7059. KPA	TGREGC 95.4 DEG.C	T04HTR 545.7 DEG.C
TWODPR 19.5 DEG.C			TGCOMP 51.2 DEG.C	T05HTR 546.2 DEG.C
			TGBOUN 37.7 DEG.C	T06HTR 551.0 DEG.C
				T07HTR 519.0 DEG.C
				T08HTR 554.7 DEG.C
				T09HTR 561.9 DEG.C
				T10HTR 550.6 DEG.C
				T11HTR 546.7 DEG.C
				T12HTR 552.3 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.23 L/MIN	1 PWRIN 3388. WATTS	VX1HOR 0.4 CM/S	PWROUT 764. WATTS	T13REG***** DEG.C
TWINCL 24.9 DEG.C	2 QCOOLR 2200. WATTS	VY1VER 7.8 CM/S	INDPWR 813. WATTS	T14REG 461.0 DEG.C
TDLCL 7.49 DEG.C	3 QDSHPT 663. WATTS		PISTST 2.20 CM	T15REG 342.3 DEG.C
TWOCLR 32.54 DEG.C	4 EXTEFF 22.5 %	PHASE ANGLES	DISPST 1.97 CM	T16REG***** DEG.C
	5 TAVHTR 549.7 DEG.C	PADISP 84.5 DEG.		T17REG 331.0 DEG.C
	6 INTEFF 25.8 %	PAPRES -14.8 DEG.	DYNAMIC CALCULATIONS	T18REG 340.8 DEG.C
	8 AMPS 1236. AMPS	ENGINE SPEED	PAMPC 1142. KPA	T19REG 216.0 DEG.C
	9 QDISPG 4. WATTS	FREQ 31.9 HZ	DISPCP 2.62 CM	
	10 QDISP 19. WATTS		PISTCP 2.38 CM	
	11 QREG1 101. WATTS		PDYNDB 1263. KPA	
	12 QREG2 98. WATTS		PDLCLR***** KPA	
	13 QREG3 31. WATTS		PDLREG***** KPA	
	14 PFP 1030.8 KPA/CM		PDLDIS***** KPA	
	15 PFD 297.0 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00604			

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:28:21.86 RDG 933

FLUID HELIUM BAROM 14.263 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.19 L/MIN	AMPS1 654. AMPS	PRESUP 7129. KPA	TGEXP 722.9 DEG.C	T01HTR 551.2 DEG.C
	AMPS2 639. AMPS			T02HTR 556.2 DEG.C
TWINDP 16.7 DEG.C	VOLTG 2.87 VOLTS	MEANBP 6996. KPA	TGREGH 487.8 DEG.C	T03HTR 560.1 DEG.C
TDLDP 2.53 DEG.C		MEANCP 7028. KPA	TGREGC 97.4 DEG.C	T04HTR 544.6 DEG.C
TWODPR 19.7 DEG.C			TGCOMP 54.1 DEG.C	T05HTR 546.6 DEG.C
			TGBOUN 38.2 DEG.C	T06HTR 550.1 DEG.C
				T07HTR 517.5 DEG.C
				T08HTR 553.5 DEG.C
				T09HTR 561.9 DEG.C
				T10HTR 550.3 DEG.C
				T11HTR 545.7 DEG.C
				T12HTR 551.9 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.23 L/MIN	1 PWRIN 3706. WATTS	VX1HOR 0.4 CM/S	PWROUT 831. WATTS	T13REG***** DEG.C
TWINCL 24.9 DEG.C	2 QCOOLR 2440. WATTS	VY1VER 8.4 CM/S	INDPWR 883. WATTS	T14REG 460.6 DEG.C
TDLCL 8.30 DEG.C	3 QDSHPT 737. WATTS		PISTST 2.39 CM	T15REG 343.1 DEG.C
TWOCLR 33.38 DEG.C	4 EXTEFF 22.4 %	PHASE ANGLES	DISPST 2.06 CM	T16REG***** DEG.C
	5 TAVHTR 549.1 DEG.C	PADISP 85.3 DEG.		T17REG 329.5 DEG.C
	6 INTEFF 25.4 %	PAPRES -13.5 DEG.	DYNAMIC CALCULATIONS	T18REG 339.7 DEG.C
	8 AMPS 1293. AMPS	ENGINE SPEED	PAMPC 1240. KPA	T19REG 218.5 DEG.C
	9 QDISPG 4. WATTS	FREQ 31.8 HZ	DISPCP 2.66 CM	
	10 QDISP 19. WATTS		PISTCP 2.39 CM	
	11 QREG1 98. WATTS		PDYNDB 1300. KPA	
	12 QREG2 98. WATTS		PDLCLR***** KPA	
	13 QREG3 30. WATTS		PDLREG***** KPA	
	14 PFP 1027.4 KPA/CM		PDLDIS***** KPA	
	15 PFD 280.9 KPA/CM			T03HED***** DEG.C
	16 NBEALE 0.00607			

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:33:18.88 RDG 934

FLUID HELIUM		BAROM 14.263 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP 4.20 L/MIN		AMPS1 692. AMPS		PRESUP 7454. KPA		TGEXP 733.3 DEG.C		T01HTR 553.2 DEG	
		AMPS2 669. AMPS						T02HTR 555.6 DEG	
TWINDP 16.7 DEG.C		VOLTG 3.02 VOLTS		MEANBP 7000. KPA		TGREGH 488.0 DEG.C		T03HTR 561.1 DEG	
TDLDP 2.81 DEG.C				MEANCP 7034. KPA		TGREGC 101.2 DEG.C		T04HTR 543.9 DEG	
TWODPR 20.0 DEG.C						TGCOMP 59.0 DEG.C		T05HTR 545.9 DEG	
						TGBOUN 38.9 DEG.C		T06HTR 551.0 DEG	
								T07HTR 515.5 DEG	
								T08HTR 554.1 DEG	
								T09HTR 563.0 DEG	
								T10HTR 552.3 DEG	
								T11HTR 547.2 DEG	
								T12HTR 550.8 DEG	
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR 4.23 L/MIN		1 PWRIN 4107. WATTS		VXIHOR 0.4 CM/S		PWROUT 919. WATTS			
TWINCL 25.1 DEG.C		2 QCOOLR 2715. WATTS		VYIVER 9.1 CM/S		INDPWR 977. WATTS			
TDLCL 9.24 DEG.C		3 QDSHPT 822. WATTS				PISTST 2.60 CM			
TWOCLR 34.47 DEG.C		4 EXTEFF 22.4 %		PHASE ANGLES		DISPST 2.19 CM			
		5 TAVHTR 549.5 DEG.C		PADISP 83.9 DEG.					
		6 INTEFF 25.3 %		PAPRES -13.0 DEG.					
		8 AMPS 1361. AMPS		ENGINE SPEED		DYNAMIC CALCULATIONS			
		9 QDISPG 4. WATTS		FREQ 31.7 HZ		PAMPC 1339. KPA		T13REG***** DEG	
		10 QDISP 19. WATTS				DISPCP 2.70 CM		T14REG 458.6 DEG	
		11 QREG1 99. WATTS				PISTCP 2.36 CM		T15REG 340.1 DEG	
		12 QREG2 97. WATTS				PDYNDB 1425. KPA		T16REG***** DEG	
		13 QREG3 30. WATTS				PDLCLR***** KPA		T17REG 332.7 DEG	
		14 PFP 1028.0 KPA/CM				PDLREG***** KPA		T18REG 338.4 DEG	
		15 PFD 276.9 KPA/CM				PDLDIS***** KPA		T19REG 216.6 DEG	
		16 NBEALE 0.00621						T03HED***** DEG	

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:38:18.88 RDG 935

FLUID HELIUM		BAROM 14.263 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP 4.17 L/MIN		AMPS1 724. AMPS		PRESUP 7132. KPA		TGEXP 746.2 DEG.C		T01HTR 554.5 DEG	
		AMPS2 694. AMPS						T02HTR 555.4 DEG	
TWINDP 16.7 DEG.C		VOLTG 3.15 VOLTS		MEANBP 6990. KPA		TGREGH 487.6 DEG.C		T03HTR 562.4 DEG	
TDLDP 2.99 DEG.C				MEANCP 7027. KPA		TGREGC 104.2 DEG.C		T04HTR 543.8 DEG	
TWODPR 20.3 DEG.C						TGCOMP 62.8 DEG.C		T05HTR 546.5 DEG	
						TGBOUN 40.0 DEG.C		T06HTR 551.5 DEG	
								T07HTR 514.7 DEG	
								T08HTR 554.7 DEG	
								T09HTR 564.6 DEG	
								T10HTR 553.5 DEG	
								T11HTR 548.3 DEG	
								T12HTR 550.7 DEG	
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR 4.24 L/MIN		1 PWRIN 4467. WATTS		VXIHOR 0.4 CM/S		PWROUT 981. WATTS			
TWINCL 25.2 DEG.C		2 QCOOLR 2967. WATTS		VYIVER 9.7 CM/S		INDPWR 1042. WATTS			
TDLCL 10.08 DEG.C		3 QDSHPT 869. WATTS				PISTST 2.79 CM			
TWOCLR 35.43 DEG.C		4 EXTEFF 22.0 %		PHASE ANGLES		DISPST 2.30 CM			
		5 TAVHTR 550.1 DEG.C		PADISP 83.7 DEG.					
		6 INTEFF 24.9 %		PAPRES -12.2 DEG.					
		8 AMPS 1418. AMPS		ENGINE SPEED		DYNAMIC CALCULATIONS			
		9 QDISPG 4. WATTS		FREQ 31.5 HZ		PAMPC 1422. KPA		T13REG***** DEG	
		10 QDISP 19. WATTS				DISPCP 2.71 CM		T14REG 457.8 DEG	
		11 QREG1 98. WATTS				PISTCP 2.35 CM		T15REG 339.7 DEG	
		12 QREG2 97. WATTS				PDYNDB 1519. KPA		T16REG***** DEG	
		13 QREG3 30. WATTS				PDLCLR***** KPA		T17REG 335.1 DEG	
		14 PFP 1020.0 KPA/CM				PDLREG***** KPA		T18REG 338.7 DEG	
		15 PFD 263.5 KPA/CM				PDLDIS***** KPA		T19REG 218.3 DEG	
		16 NBEALE 0.00621						T03HED***** DEG	

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:43:18.91 RDG 936

FLUID HELIUM		BAROM 14.263 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP 4.17 L/MIN		AMPS1 762. AMPS		PRESUP 7470. KPA		TGEXP 766.9 DEG.C		T01HTR 555.6 DEG	
		AMPS2 723. AMPS						T02HTR 555.1 DEG	
TWINDP 16.7 DEG.C		VOLTG 3.30 VOLTS		MEANBP 7007. KPA		TGREGH 486.6 DEG.C		T03HTR 563.9 DEG	
TDLDP 3.27 DEG.C				MEANCP 7036. KPA		TGREGC 107.3 DEG.C		T04HTR 542.8 DEG	
TWODPR 20.5 DEG.C						TGCOMP 67.0 DEG.C		T05HTR 546.7 DEG	
						TGBOUN 41.2 DEG.C		T06HTR 551.8 DEG	
								T07HTR 513.0 DEG	
								T08HTR 554.7 DEG	
								T09HTR 565.1 DEG	
								T10HTR 554.8 DEG	
								T11HTR 548.9 DEG	
								T12HTR 550.4 DEG	
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR 4.29 L/MIN		1 PWRIN 4898. WATTS		VXIHOR 0.4 CM/S		PWROUT 1051. WATTS			
TWINCL 25.2 DEG.C		2 QCOOLR 3294. WATTS		VYIVER 10.3 CM/S		INDPWR 1113. WATTS			
TDLCL 11.07 DEG.C		3 QDSHPT 949. WATTS				PISTST 2.99 CM			
TWOCLR 36.46 DEG.C		4 EXTEFF 21.5 %		PHASE ANGLES		DISPST 2.40 CM			
		5 TAVHTR 550.2 DEG.C		PADISP 83.6 DEG.					
		6 INTEFF 24.2 %		PAPRES -11.5 DEG.					
		8 AMPS 1485. AMPS		ENGINE SPEED		DYNAMIC CALCULATIONS			
		9 QDISPG 4. WATTS		FREQ 31.5 HZ		PAMPC 1531. KPA		T13REG***** DEG	
		10 QDISP 20. WATTS				DISPCP 2.73 CM		T14REG 457.2 DEG	
		11 QREG1 95. WATTS				PISTCP 2.32 CM		T15REG 339.7 DEG	
		12 QREG2 97. WATTS				PDYNDB 1613. KPA		T16REG***** DEG	
		13 QREG3 29. WATTS				PDLCLR***** KPA		T17REG 337.7 DEG	
		14 PFP 1027.2 KPA/CM				PDLREG***** KPA		T18REG 338.1 DEG	
		15 PFD 256.0 KPA/CM				PDLDIS***** KPA		T19REG 220.3 DEG	
		16 NBEALE 0.00621						T03HED***** DEG	

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:48:18.91 RDG 937

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.16 L/MIN	AMPS1 796. AMPS	PRESUP 7263. KPA	TGEXP 794.8 DEG.C	T01HTR 555.3 DEG.C
	AMPS2 745. AMPS			T02HTR 553.4 DEG.C
WINDP 16.8 DEG.C	VOLTG 3.43 VOLTS	MEANBP 6991. KPA	TGREGH 484.9 DEG.C	T03HTR 563.6 DEG.C
DLDP 3.41 DEG.C		MEANCP 7052. KPA	TGREGC 109.6 DEG.C	T04HTR 540.9 DEG.C
WODPR 20.9 DEG.C			TGCOMP 70.9 DEG.C	T05HTR 545.2 DEG.C
			TGBOUN 42.3 DEG.C	T06HTR 550.7 DEG.C
				T07HTR 510.3 DEG.C
				T08HTR 553.8 DEG.C
				T09HTR 564.8 DEG.C
				T10HTR 554.3 DEG.C
				T11HTR 548.3 DEG.C
				T12HTR 548.8 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.29 L/MIN	1 PWRIN 5287. WATTS	VX1HOR 0.4 CM/S	PWROUT 1119. WATTS	
WINCL 25.3 DEG.C	2 QCOOLR 3592. WATTS	VY1VER10.9 CM/S	INDPWR 1168. WATTS	
DLCL 12.07 DEG.C	3 QDSHPT 987. WATTS		PISTST 3.19 CM	
WOCLR 37.61 DEG.C	4 EXTEFF 21.2 %	PHASE ANGLES	DISPST 2.50 CM	
	5 TAVHTR 549.1 DEG.C	PADISP 83.3 DEG.		
	6 INTEFF 23.7 %	PAPRES -10.6 DEG.		
	8 AMPS 1541. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 4. WATTS	FREQ 31.4 HZ	PAMPC 1629. KPA	T13REG***** DEG.C
	10 QDISP 21. WATTS		DISPCP 2.78 CM	T14REG 453.6 DEG.C
	11 QREG1 94. WATTS		PISTCP 2.26 CM	T15REG 337.9 DEG.C
	12 QREG2 94. WATTS		PDYNDB 1675. KPA	T16REG***** DEG.C
	13 QREG3 29. WATTS		PDLCLR***** KPA	T17REG 338.8 DEG.C
	14 PFP 1024.9 KPA/CM		PDLREG***** KPA	T18REG 337.9 DEG.C
	15 PFD 240.9 KPA/CM		PDLDIS***** KPA	T19REG 222.1 DEG.C
	16 NBEALE 0.00622			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 15:51:21.91 RDG 938

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

T TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.16 L/MIN	AMPS1 836. AMPS	PRESUP 7132. KPA	TGEXP 811.7 DEG.C	T01HTR 556.4 DEG.C
	AMPS2 783. AMPS			T02HTR 554.2 DEG.C
WINDP 16.9 DEG.C	VOLTG 3.60 VOLTS	MEANBP 7008. KPA	TGREGH 484.0 DEG.C	T03HTR 565.6 DEG.C
DLDP 3.60 DEG.C		MEANCP 7069. KPA	TGREGC 113.0 DEG.C	T04HTR 540.9 DEG.C
WODPR 21.1 DEG.C			TGCOMP 75.4 DEG.C	T05HTR 546.1 DEG.C
			TGBOUN 43.1 DEG.C	T06HTR 551.1 DEG.C
				T07HTR 509.1 DEG.C
				T08HTR 555.1 DEG.C
				T09HTR 566.5 DEG.C
				T10HTR 555.6 DEG.C
				T11HTR 549.8 DEG.C
				T12HTR 549.4 DEG.C
T TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.27 L/MIN	1 PWRIN 5831. WATTS	VX1HOR 0.5 CM/S	PWROUT 1144. WATTS	
WINCL 25.4 DEG.C	2 QCOOLR 3944. WATTS	VY1VER11.6 CM/S	INDPWR 1191. WATTS	
DLCL 13.30 DEG.C	3 QDSHPT 1042. WATTS		PISTST 3.39 CM	
WOCLR 38.73 DEG.C	4 EXTEFF 19.6 %	PHASE ANGLES	DISPST 2.53 CM	
	5 TAVHTR 550.0 DEG.C	PADISP 84.5 DEG.		
	6 INTEFF 22.5 %	PAPRES -9.7 DEG.		
	8 AMPS 1618. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG 4. WATTS	FREQ 31.4 HZ	PAMPC 1747. KPA	T13REG***** DEG.C
	10 QDISP 21. WATTS		DISPCP 2.82 CM	T14REG 453.3 DEG.C
	11 QREG1 92. WATTS		PISTCP 2.19 CM	T15REG 339.9 DEG.C
	12 QREG2 92. WATTS		PDYNDB 1763. KPA	T16REG***** DEG.C
	13 QREG3 29. WATTS		PDLCLR***** KPA	T17REG 340.2 DEG.C
	14 PFP 1032.2 KPA/CM		PDLREG***** KPA	T18REG 339.7 DEG.C
	15 PFD 234.0 KPA/CM		PDLDIS***** KPA	T19REG 226.3 DEG.C
	16 NBEALE 0.00597			T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:00:18.91 RDG 939

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

AT TO DASHPOT COOLING	POWER IN	ENGINE CHARGE PRESSURE	GAS TEMPERATURES	SURFACE TEMPERATURES
FLODP 4.20 L/MIN	AMPS1 564. AMPS	PRESUP 7105. KPA	TGEXP***** DEG.C	T01HTR 601.8 DEG.C
	AMPS2 560. AMPS			T02HTR 606.0 DEG.C
WINDP 16.9 DEG.C	VOLTG 2.50 VOLTS	MEANBP 6989. KPA	TGREGH 542.8 DEG.C	T03HTR 609.0 DEG.C
DLDP 2.18 DEG.C		MEANCP 7025. KPA	TGREGC 93.8 DEG.C	T04HTR 595.9 DEG.C
WODPR 19.5 DEG.C			TGCOMP 46.4 DEG.C	T05HTR 596.1 DEG.C
			TGBOUN 39.9 DEG.C	T06HTR 603.7 DEG.C
				T07HTR 569.5 DEG.C
				T08HTR 606.4 DEG.C
				T09HTR 609.0 DEG.C
				T10HTR 601.4 DEG.C
				T11HTR 597.6 DEG.C
				T12HTR 603.2 DEG.C
AT TO COOLER	CALCULATED PARAMETERS	VIBRATION	REMOTE CALCULATIONS	
FLOCLR 4.24 L/MIN	1 PWRIN 2812. WATTS	VX1HOR 0.3 CM/S	PWROUT 640. WATTS	
WINCL 24.9 DEG.C	2 QCOOLR 1842. WATTS	VY1VER 6.6 CM/S	INDPWR 695. WATTS	
DLCL 6.26 DEG.C	3 QDSHPT 637. WATTS		PISTST 1.79 CM	
WOCLR 31.39 DEG.C	4 EXTEFF 22.8 %	PHASE ANGLES	DISPST 1.73 CM	
	5 TAVHTR 600.0 DEG.C	PADISP 82.5 DEG.		
	6 INTEFF 25.8 %	PAPRES -18.2 DEG.		
	8 AMPS 1124. AMPS	ENGINE SPEED	DYNAMIC CALCULATIONS	
	9 QDISPG***** WATTS	FREQ 32.3 HZ	PAMPC 945. KPA	T13REG***** DEG.C
	10 QDISP ***** WATTS		DISPCP 2.60 CM	T14REG 505.8 DEG.C
	11 QREG1 115. WATTS		PISTCP 2.38 CM	T15REG 371.1 DEG.C
	12 QREG2 111. WATTS		PDYNDB 1088. KPA	T16REG***** DEG.C
	13 QREG3 35. WATTS		PDLCLR***** KPA	T17REG 363.5 DEG.C
	14 PFP 1044.0 KPA/CM		PDLREG***** KPA	T18REG 369.2 DEG.C
	15 PFD 343.2 KPA/CM		PDLDIS***** KPA	T19REG 227.3 DEG.C
	16 NBEALE 0.00615			T03HED***** DEG.C

APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:05:21.91 RDG 940

FLUID HELIUM		BAROM 14.258 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.25 L/MIN	AMPS1	619. AMPS	PRESUP	7476. KPA	TGEXPI	1030.6 DEG.C	T01HTR	599.3 DEG.C
		AMPS2	577. AMPS					T02HTR	604.9 DEG.C
TWINDP	16.9 DEG.C	VOLTG	2.67 VOLTS	MEANBP	7007. KPA	TGREGH	537.5 DEG.C	T03HTR	607.6 DEG.C
TDLDP	2.15 DEG.C			MEANCP	7052. KPA	TGREGC	95.6 DEG.C	T04HTR	593.1 DEG.C
TWODPR	19.7 DEG.C					TGCOMP	48.3 DEG.C	T05HTR	594.6 DEG.C
						TGBOUN	39.3 DEG.C	T06HTR	599.8 DEG.C
								T07HTR	567.2 DEG.C
								T08HTR	603.1 DEG.C
								T09HTR	607.8 DEG.C
								T10HTR	599.0 DEG.C
								T11HTR	594.7 DEG.C
								T12HTR	600.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.24 L/MIN	1 PWRIN	3189. WATTS	VX1HOR	0.3 CM/S	PWROUT	738. WATTS		
TWINCL	24.8 DEG.C	2 QCOOLR	1976. WATTS	VY1VER	7.3 CM/S	INDPWR	802. WATTS		
TDLCL	6.71 DEG.C	3 QDSHPT	635. WATTS			PISTST	2.00 CM		
TWOCLR	31.85 DEG.C	4 EXTEFF	23.1 %	PHASE ANGLES		DISPST	1.86 CM		
		5 TAVHTR	597.7 DEG.C	PADISP	83.1 DEG.			T13REG*****	DEG.C
		6 INTEFF	27.2 %	PAPRES	-17.1 DEG.			T14REG	503.9 DEG.C
		8 AMPS	1196. AMPS			DYNAMIC CALCULATIONS		T15REG	372.4 DEG.C
		9 QDISPG	6. WATTS	ENGINE SPEED		PAMPC	1043. KPA	T16REG*****	DEG.C
		10 QDISP	28. WATTS	FREQ	32.2 HZ	DISPCP	2.62 CM	T17REG	360.5 DEG.C
		11 QREG1	111. WATTS			PISTCP	2.43 CM	T18REG	367.8 DEG.C
		12 QREG2	110. WATTS			PDYNDB	1163. KPA	T19REG	230.4 DEG.C
		13 QREG3	34. WATTS			PDLCLR*****	KPA		
		14 PFP	1036.2 KPA/CM			PDLREG*****	KPA		
		15 PFD	332.4 KPA/CM			PDLDIS*****	KPA	T03HED*****	DEG.C
		16 NBEALE	0.00638						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:10:18.91 RDG 941

FLUID HELIUM		BAROM 14.258 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.35 L/MIN	AMPS1	660. AMPS	PRESUP	7124. KPA	TGEXP	986.8 DEG.C	T01HTR	601.7 DEG.C
		AMPS2	601. AMPS					T02HTR	606.8 DEG.C
TWINDP	16.9 DEG.C	VOLTG	2.81 VOLTS	MEANBP	6999. KPA	TGREGH	536.7 DEG.C	T03HTR	610.6 DEG.C
TDLDP	2.28 DEG.C			MEANCP	7045. KPA	TGREGC	97.4 DEG.C	T04HTR	594.1 DEG.C
TWODPR	19.8 DEG.C					TGCOMP	51.3 DEG.C	T05HTR	597.2 DEG.C
						TGBOUN	39.0 DEG.C	T06HTR	601.5 DEG.C
								T07HTR	567.6 DEG.C
								T08HTR	604.3 DEG.C
								T09HTR	610.3 DEG.C
								T10HTR	601.3 DEG.C
								T11HTR	596.1 DEG.C
								T12HTR	602.9 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWRIN	3547. WATTS	VX1HOR	0.4 CM/S	PWROUT	822. WATTS		
TWINCL	24.8 DEG.C	2 QCOOLR	2199. WATTS	VY1VER	7.9 CM/S	INDPWR	892. WATTS		
TDLCL	7.51 DEG.C	3 QDSHPT	690. WATTS			PISTST	2.19 CM		
TWOCLR	32.55 DEG.C	4 EXTEFF	23.2 %	PHASE ANGLES		DISPST	1.97 CM		
		5 TAVHTR	599.5 DEG.C	PADISP	83.9 DEG.			T13REG*****	DEG.C
		6 INTEFF	27.2 %	PAPRES	-15.8 DEG.			T14REG	505.4 DEG.C
		8 AMPS	1261. AMPS			DYNAMIC CALCULATIONS		T15REG	374.0 DEG.C
		9 QDISPG	5. WATTS	ENGINE SPEED		PAMPC	1132. KPA	T16REG*****	DEG.C
		10 QDISP	27. WATTS	FREQ	32.2 HZ	DISPCP	2.64 CM	T17REG	360.6 DEG.C
		11 QREG1	110. WATTS			PISTCP	2.44 CM	T18REG	369.3 DEG.C
		12 QREG2	110. WATTS			PDYNDB	1288. KPA	T19REG	233.3 DEG.C
		13 QREG3	34. WATTS			PDLCLR*****	KPA		
		14 PFP	1025.3 KPA/CM			PDLREG*****	KPA		
		15 PFD	315.8 KPA/CM			PDLDIS*****	KPA	T03HED*****	DEG.C
		16 NBEALE	0.00649						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:15:18.94 RDG 942

FLUID HELIUM		BAROM 14.258 PSI		REGENERATOR 2		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.34 L/MIN	AMPS1	696. AMPS	PRESUP	7458. KPA	TGEXP	951.3 DEG.C	T01HTR	602.5 DEG.C
		AMPS2	631. AMPS					T02HTR	607.0 DEG.C
TWINDP	16.9 DEG.C	VOLTG	2.96 VOLTS	MEANBP	6996. KPA	TGREGH	535.0 DEG.C	T03HTR	612.0 DEG.C
TDLDP	2.48 DEG.C			MEANCP	7047. KPA	TGREGC	100.2 DEG.C	T04HTR	593.9 DEG.C
TWODPR	20.0 DEG.C					TGCOMP	54.6 DEG.C	T05HTR	597.7 DEG.C
						TGBOUN	39.4 DEG.C	T06HTR	601.2 DEG.C
								T07HTR	566.7 DEG.C
								T08HTR	604.3 DEG.C
								T09HTR	611.7 DEG.C
								T10HTR	601.9 DEG.C
								T11HTR	596.4 DEG.C
								T12HTR	602.7 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.24 L/MIN	1 PWRIN	3929. WATTS	VX1HOR	0.4 CM/S	PWROUT	908. WATTS	T13REG*****	DEG.C
TWINCL	24.9 DEG.C	2 QCOOLR	2471. WATTS	VY1VER	8.6 CM/S	INDPWR	988. WATTS	T14REG	504.3 DEG.C
TDLCL	8.39 DEG.C	3 QDSHPT	750. WATTS			PISTST	2.40 CM	T15REG	373.6 DEG.C
TWOCLR	33.45 DEG.C	4 EXTEFF	23.1 %	PHASE ANGLES		DISPST	2.09 CM	T16REG*****	DEG.C
		5 TAVHTR	599.8 DEG.C	PADISP	83.8 DEG.	DYNAMIC CALCULATIONS		T17REG	360.1 DEG.C
		6 INTEFF	26.9 %	PAPRES	-14.9 DEG.	PAMPC	1245. KPA	T18REG	370.5 DEG.C
		7 AMPS	1327. AMPS	ENGINE SPEED		DISPCP	2.61 CM	T19REG	234.8 DEG.C
		8 QDISPG	5. WATTS	FREQ	32.0 HZ	PISTCP	2.45 CM	T03HED*****	DEG.C
		10 QDISP	25. WATTS			PDYNDP	1375. KPA		
		11 QREG1	110. WATTS			PDLCLR*****	KPA		
		12 QREG2	108. WATTS			PDLREG*****	KPA		
		13 QREG3	34. WATTS			PDLDIS*****	KPA		
		14 PFP	1033.5 KPA/CM						
		15 PFD	308.4 KPA/CM						
		16 NBEALE	0.00659						

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APPENDIX D. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:24:31.06 RDG 943

FLUID HELIUM				BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON							
EAT TO DASHPOT COOLING				POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.34 L/MIN			AMPS1	730. AMPS	PRESUP	7459. KPA	TGEXP	900.7 DEG.C	T01HTR	602.5 DEG.C
				AMPS2	660. AMPS					T02HTR	605.8 DEG.C
TWINDP	16.9 DEG.C			VOLTG	3.10 VOLTS	MEANBP	7007. KPA	TGREGH	532.2 DEG.C	T03HTR	612.3 DEG.C
TDLDP	2.75 DEG.C					MEANCP	7059. KPA	TGREGC	103.1 DEG.C	T04HTR	592.2 DEG.C
TWODPR	20.3 DEG.C							TGCOMP	58.6 DEG.C	T05HTR	597.3 DEG.C
								TGBOUN	40.2 DEG.C	T06HTR	600.1 DEG.C
										T07HTR	564.5 DEG.C
										T08HTR	602.7 DEG.C
										T09HTR	611.6 DEG.C
										T10HTR	601.8 DEG.C
										T11HTR	595.6 DEG.C
										T12HTR	601.5 DEG.C
EAT TO COOLER				CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.26 L/MIN			1 PWRIN	4314. WATTS	VX1HOR	0.5 CM/S	PWROUT	1001. WATTS	T13REG*****	DEG.C
TWINCL	25.0 DEG.C			2 QCOOLR	2723. WATTS	VY1VER	9.2 CM/S	INDPWR	1078. WATTS	T14REG	502.8 DEG.C
TDLCL	9.20 DEG.C			3 QDSHPT	831. WATTS			PISTST	2.59 CM	T15REG	373.7 DEG.C
TWOCLR	34.45 DEG.C			4 EXTEFF	23.2 %	PHASE ANGLES		DISPST	2.20 CM	T16REG*****	DEG.C
				5 TAVHTR	599.0 DEG.C	PADISP	83.7 DEG.			T17REG	359.9 DEG.C
				6 INTEFF	26.9 %	PAPRES	-14.3 DEG.			T18REG	370.5 DEG.C
				8 AMPS	1390. AMPS	ENGINE SPEED				T19REG	237.4 DEG.C
				9 QDISPG	5. WATTS	FREQ	31.9 HZ			T03HED*****	DEG.C
				10 QDISP	24. WATTS						
				11 QREG1	108. WATTS						
				12 QREG2	107. WATTS						
				13 QREG3	33. WATTS						
				14 PFP	1029.1 KPA/CM						
				15 PFD	301.7 KPA/CM						
				16 NBEALE	0.00674						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:28:19.08 RDG 944

FLUID HELIUM				BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON							
HEAT TO DASHPOT COOLING				POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURE	
FLODP	4.33 L/MIN			AMPS1	765. AMPS	PRESUP	7439. KPA	TGEXP	898.6 DEG.C	T01HTR	603.5 DEG.C
				AMPS2	689. AMPS					T02HTR	604.9 DEG.C
TWINLP	16.9 DEG.C			VOLTG	3.25 VOLTS	MEANBP	6982. KPA	TGREGH	532.0 DEG.C	T03HTR	612.9 DEG.C
TDLDP	2.97 DEG.C					MEANCP	7031. KPA	TGREGC	106.5 DEG.C	T04HTR	591.7 DEG.C
TWODPR	20.5 DEG.C							TGCOMP	62.7 DEG.C	T05HTR	596.8 DEG.C
								TGBOUN	40.7 DEG.C	T06HTR	599.9 DEG.C
										T07HTR	563.0 DEG.C
										T08HTR	603.2 DEG.C
										T09HTR	613.2 DEG.C
										T10HTR	602.7 DEG.C
										T11HTR	597.1 DEG.C
										T12HTR	600.1 DEG.C
HEAT TO COOLER				CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.27 L/MIN			1 PWRIN	4725. WATTS	VX1HOR	0.5 CM/S	PWROUT	1095. WATTS	T13REG*****	DEG.C
TWINCL	25.1 DEG.C			2 QCOOLR	2991. WATTS	VY1VER	9.9 CM/S	INDPWR	1171. WATTS	T14REG	501.0 DEG.C
TDLCL	10.10 DEG.C			3 QDSHPT	895. WATTS			PISTST	2.81 CM	T15REG	371.8 DEG.C
TWOCLR	35.40 DEG.C			4 EXTEFF	23.2 %	PHASE ANGLES		DISPST	2.32 CM	T16REG*****	DEG.C
				5 TAVHTR	599.1 DEG.C	PADISP	82.5 DEG.			T17REG	362.9 DEG.C
				6 INTEFF	26.8 %	PAPRES	-13.6 DEG.			T18REG	367.9 DEG.C
				8 AMPS	1455. AMPS	ENGINE SPEED				T19REG	237.5 DEG.C
				9 QDISPG	5. WATTS	FREQ	31.6 HZ			T03HED*****	DEG.C
				10 QDISP	24. WATTS						
				11 QREG1	106. WATTS						
				12 QREG2	108. WATTS						
				13 QREG3	33. WATTS						
				14 PFP	1026.6 KPA/CM						
				15 PFD	294.4 KPA/CM						
				16 NBEALE	0.00688						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:32:22.08 RDG 945

FLUID HELIUM				BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON							
HEAT TO DASHPOT COOLING				POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.32 L/MIN			AMPS1	803. AMPS	PRESUP	7141. KPA	TGEXP	921.7 DEG.C	T01HTR	604.3 DEG.C
				AMPS2	721. AMPS					T02HTR	604.0 DEG.C
TWINDP	17.0 DEG.C			VOLTG	3.40 VOLTS	MEANBP	7036. KPA	TGREGH	531.0 DEG.C	T03HTR	613.6 DEG.C
TDLDP	3.24 DEG.C					MEANCP	7105. KPA	TGREGC	110.4 DEG.C	T04HTR	590.8 DEG.C
TWODPR	20.8 DEG.C							TGCOMP	67.4 DEG.C	T05HTR	596.2 DEG.C
								TGBOUN	42.1 DEG.C	T06HTR	599.8 DEG.C
										T07HTR	560.6 DEG.C
										T08HTR	603.5 DEG.C
										T09HTR	614.0 DEG.C
										T10HTR	603.4 DEG.C
										T11HTR	597.8 DEG.C
										T12HTR	599.2 DEG.C
HEAT TO COOLER				CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.25 L/MIN			1 PWRIN	5185. WATTS	VX1HOR	0.5 CM/S	PWROUT	1187. WATTS	T13REG*****	DEG.C
TWINCL	25.1 DEG.C			2 QCOOLR	3303. WATTS	VY1VER	10.5 CM/S	INDPWR	1260. WATTS	T14REG	499.3 DEG.C
TDLCL	11.19 DEG.C			3 QDSHPT	975. WATTS			PISTST	2.99 CM	T15REG	370.6 DEG.C
TWOCLR	36.56 DEG.C			4 EXTEFF	22.9 %	PHASE ANGLES		DISPST	2.42 CM	T16REG*****	DEG.C
				5 TAVHTR	598.9 DEG.C	PADISP	82.9 DEG.			T17REG	365.1 DEG.C
				6 INTEFF	26.4 %	PAPRES	-12.8 DEG.			T18REG	367.9 DEG.C
				8 AMPS	1524. AMPS	ENGINE SPEED				T19REG	238.3 DEG.C
				9 QDISPG	5. WATTS	FREQ	31.7 HZ			T03HED*****	DEG.C
				10 QDISP	24. WATTS						
				11 QREG1	105. WATTS						
				12 QREG2	107. WATTS						
				13 QREG3	33. WATTS						
				14 PFP	1032.1 KPA/CM						
				15 PFD	284.2 KPA/CM						
				16 NBEALE	0.00693						

APPENDIX D. - CONCLUDED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:35:28.08 RDG 946

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.30 L/MIN	POWER IN AMPS1 829. AMPS AMPS2 758. AMPS VOLTG 3.54 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7448. KPA MEANBP 7005. KPA MEANCP 7072. KPA	GAS TEMPERATURES TGEXP 969.6 DEG.C TGREGH 531.8 DEG.C TGREGC 113.5 DEG.C TGCOMP 70.9 DEG.C TGBOUN 42.9 DEG.C	SURFACE TEMPERATURE T01HTR 607.8 DEG.C T02HTR 603.5 DEG.C T03HTR 616.3 DEG.C T04HTR 590.5 DEG.C T05HTR 596.4 DEG.C T06HTR 602.2 DEG.C T07HTR 558.9 DEG.C T08HTR 604.7 DEG.C T09HTR 615.1 DEG.C T10HTR 606.8 DEG.C T11HTR 599.6 DEG.C T12HTR 599.0 DEG.C
TWINDP 17.1 DEG.C TDLDP 3.45 DEG.C TWODPR 21.1 DEG.C				
HEAT TO COOLER FLOCLR 4.27 L/MIN TWINCL 25.2 DEG.C TDLCL 12.10 DEG.C TWOCLE 37.42 DEG.C	CALCULATED PARAMETERS 1 PWRIN 5619. WATTS 2 QCOOLR 3587. WATTS 3 QDSHPT 1034. WATTS 4 EXTEFF 22.7 % 5 TAVHTR 600.1 DEG.C 6 INTEFF 26.2 % 8 AMPS 1587. AMPS 9 QDISPG 5. WATTS 10 QDISP 26. WATTS 11 QREG1 107. WATTS 12 QREG2 104. WATTS 13 QREG3 32. WATTS 14 PFP 1036.2 KPA/CM 15 PFD 278.1 KPA/CM 16 NBEALE 0.00702	VIBRATION VX1HOR 0.6 CM/S VY1VER11.2 CM/S PHASE ANGLES PADISP 82.1 DEG. PAPRES -12.2 DEG. ENGINE SPEED FREQ 31.6 HZ	REMOTE CALCULATIONS PWROUT 1274. WATTS INDPWR 1348. WATTS PISTST 3.19 CM DISPST 2.53 CM DYNAMIC CALCULATIONS PAMPC 1644. KPA DISPCP 2.80 CM PISTCP 2.30 CM PDYNDB 1700. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 497.7 DEG.C T15REG 368.7 DEG.C T16REG***** DEG.C T17REG 370.2 DEG.C T18REG 369.9 DEG.C T19REG 238.2 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:37:28.08 RDG 947

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.33 L/MIN	POWER IN AMPS1 844. AMPS AMPS2 805. AMPS VOLTG 3.68 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7445. KPA MEANBP 6998. KPA MEANCP 7074. KPA	GAS TEMPERATURES TGEXP1005.3 DEG.C TGREGH 531.1 DEG.C TGREGC 117.2 DEG.C TGCOMP 75.3 DEG.C TGBOUN 43.5 DEG.C	SURFACE TEMPERATURE T01HTR 608.8 DEG.C T02HTR 603.4 DEG.C T03HTR 617.5 DEG.C T04HTR 589.7 DEG.C T05HTR 596.9 DEG.C T06HTR 602.3 DEG.C T07HTR 557.6 DEG.C T08HTR 604.9 DEG.C T09HTR 616.2 DEG.C T10HTR 607.7 DEG.C T11HTR 600.4 DEG.C T12HTR 599.0 DEG.C
TWINDP 17.1 DEG.C TDLDP 3.67 DEG.C TWODPR 21.4 DEG.C				
HEAT TO COOLER FLOCLR 4.27 L/MIN TWINCL 25.2 DEG.C TDLCL 13.21 DEG.C TWOCLE 38.48 DEG.C	CALCULATED PARAMETERS 1 PWRIN 6067. WATTS 2 QCOOLR 3910. WATTS 3 QDSHPT 1106. WATTS 4 EXTEFF 22.1 % 5 TAVHTR 600.4 DEG.C 6 INTEFF 25.5 % 8 AMPS 1649. AMPS 9 QDISPG 5. WATTS 10 QDISP 26. WATTS 11 QREG1 105. WATTS 12 QREG2 103. WATTS 13 QREG3 32. WATTS 14 PFP 1046.3 KPA/CM 15 PFD 272.4 KPA/CM 16 NBEALE 0.00697	VIBRATION VX1HOR 0.6 CM/S VY1VER11.8 CM/S PHASE ANGLES PADISP 82.4 DEG. PAPRES -11.4 DEG. ENGINE SPEED FREQ 31.6 HZ	REMOTE CALCULATIONS PWROUT 1340. WATTS INDPWR 1410. WATTS PISTST 3.38 CM DISPST 2.57 CM DYNAMIC CALCULATIONS PAMPC 1757. KPA DISPCP 2.82 CM PISTCP 2.19 CM PDYNDB 1763. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 498.2 DEG.C T15REG 371.0 DEG.C T16REG***** DEG.C T17REG 372.2 DEG.C T18REG 371.5 DEG.C T19REG 242.1 DEG.C T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 01/24/84 16:40:31.08 RDG 948

FLUID HELIUM BAROM 14.258 PSI REGENERATOR 2 DISPLACER 2 STANDARD PISTON

HEAT TO DASHPOT COOLING FLODP 4.31 L/MIN	POWER IN AMPS1 859. AMPS AMPS2 865. AMPS VOLTG 3.86 VOLTS	ENGINE CHARGE PRESSURE PRESUP 7112. KPA MEANBP 6998. KPA MEANCP 7094. KPA	GAS TEMPERATURES TGEXP 605.1 DEG.C TGREGH 532.2 DEG.C TGREGC 121.7 DEG.C TGCOMP 80.3 DEG.C TGBOUN 44.7 DEG.C	SURFACE TEMPERATURE T01HTR 612.9 DEG.C T02HTR 607.0 DEG.C T03HTR 622.4 DEG.C T04HTR 592.7 DEG.C T05HTR 601.0 DEG.C T06HTR 605.5 DEG.C T07HTR 559.8 DEG.C T08HTR 608.5 DEG.C T09HTR 621.0 DEG.C T10HTR 612.1 DEG.C T11HTR 604.4 DEG.C T12HTR 602.3 DEG.C
TWINDP 17.2 DEG.C TDLDP 3.92 DEG.C TWODPR 21.7 DEG.C				
HEAT TO COOLER FLOCLR 4.28 L/MIN TWINCL 25.2 DEG.C TDLCL 14.69 DEG.C TWOCLE 39.99 DEG.C	CALCULATED PARAMETERS 1 PWRIN 6649. WATTS 2 QCOOLR 4363. WATTS 3 QDSHPT 1176. WATTS 4 EXTEFF 21.1 % 5 TAVHTR 604.1 DEG.C 6 INTEFF 24.3 % 8 AMPS 1724. AMPS 9 QDISPG 3. WATTS 10 QDISP 15. WATTS 11 QREG1 102. WATTS 12 QREG2 103. WATTS 13 QREG3 32. WATTS 14 PFP 1077.5 KPA/CM 15 PFD 279.5 KPA/CM 16 NBEALE 0.00683	VIBRATION VX1HOR 0.6 CM/S VY1VER12.5 CM/S PHASE ANGLES PADISP 81.6 DEG. PAPRES -10.7 DEG. ENGINE SPEED FREQ 31.7 HZ	REMOTE CALCULATIONS PWROUT 1400. WATTS INDPWR 1495. WATTS PISTST 3.60 CM DISPST 2.58 CM DYNAMIC CALCULATIONS PAMPC 1919. KPA DISPCP 2.89 CM PISTCP 2.04 CM PDYNDB 1888. KPA PDLCLR***** KPA PDLREG***** KPA PDLDIS***** KPA	T13REG***** DEG.C T14REG 499.2 DEG.C T15REG 372.6 DEG.C T16REG***** DEG.C T17REG 373.7 DEG.C T18REG 372.0 DEG.C T19REG 246.0 DEG.C T03HED***** DEG.C

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TABLE I. - NUMBER OF DATA POINTS PER CONFIGURATION

[Regenerator and displacer 1 were designed for high efficiency. Regenerator and displacer 2 were designed for high power.]

Regenerator	Displacer	Power piston	Working fluid	Number of data points
No. 1	No. 1	Std	Helium	150
No. 1	No. 2	↓	↓	174
No. 2	No. 2	↓	↓	189
No. 2	No. 1	↓	↓	108
No. 1	↓	↓	Argon	12
No. 1	↓	↓	Nitrogen	64
No. 1	↓	Light	Helium	84

TABLE II. - DESCRIPTION OF THE GEOMETRY OF THE RE-1000

Number of cylinders	1
Housing weight, Kg (lb)	416 (917)
Type	free-piston with dashpot
Design working fluid	helium
Design frequency, Hz	30
Design pressure, Mpa	7.0
Bounce space volume, cm ³ (in. ³)	20,500 (1250)
Design power, W	1000
Design displacer phase angle	45
Cylinder bore, cm (in.)	5.722 (2.2527)
Maximum displacer 1 stroke, cm (in.)	4.01 (1.579)
Displacer 1 length, cm (in.)	15.21 (5.99)
Maximum displacer 2 stroke, cm (in.)	5.18 (2.039)
Displacer 2 length, cm (in.)	14.05 (5.53)
Maximum power piston stroke, cm (in.)	4.35 (1.713)
Cooler	
Description	135 rectangular gas passages
Passage width, cm (in.)	0.0508 (0.020)
Passage depth, cm (in.)	0.376 (0.148)
Length, cm (in.)	7.92 (3.118)
Flow area, cm ² (in. ²)	2.58 (0.400)
Wetted perimeter, cm (in.)	115.2 (45.354)
Volume, cm ³ (in. ³)	20.42 (1.246)
Heater	
Description	tubular
Tube material	Inconel 718
Number of tubes	34
Tube length, cm (in.)	18.34 (7.220)
Tube inside diameter, mm (in.)	2.362 (0.093)
Tube outside diameter, mm (in.)	3.175 (0.125)
Design maximum wall temperature, °C (°F)	650 (1202)
Regenerator	
Length containing wire mesh, cm (in.)	6.446 (2.538)
Outside diameter, cm (in.)	7.18 (2.825)
Inside diameter, cm (in.)	6.07 (2.391)
Inner wall thickness, cm (in.)	0.13 (0.05)
Matrix material	304SS Metex
Wire diameter, mm (in.)	0.0889 (0.0035)
Porosity, regenerator 1, percent	75.9
Porosity, regenerator 2, percent	81.2
Weight of matrix, regenerator 1, g (lb)	139 (0.31)
Weight of matrix, regenerator 2, g (lb)	108 (0.24)
Pistons	
Standard power piston oscillating mass, kg (lb)	5.97 (13.17)
Light power piston oscillating mass, kg (lb)	3.48 (7.67)
Power piston diameter, cm (in.)	5.718 (2.2514)

TABLE II. - Continued

Displacer 1 mass, kg (lb)	0.426 (0.939)
Displacer 1 diameter, cm (in.)	5.67 (2.232)
Displacer 1 rod diameter, cm (in.)	1.663 (0.6548)
Displacer 1 bore, cm (in.)	1.666 (0.6560)
Displacer 2 mass, kg (lb)	0.381 (0.840)
Displacer 2 diameter, cm (in.)	5.67 (2.232)
Displacer 2 rod diameter, cm (in.)	1.806 (0.7110)
Displacer 2 bore, cm (in.)	1.808 (0.7118)
Dead volumes	
Displacer 1/cylinder annular gap, cm ³ (in. ³)	9.83 (0.60)
Displacer 2/cylinder annular gap, cm ³ (in. ³)	8.36 (0.51)
Expansion space instrumentation ports, cm ³ (in. ³)	1.64 (0.10)
Expansion space to heater tube junction, cm ³ (in. ³)	3.80 (0.23)
Heater tubes, cm ³ (in. ³)	27.4 (1.67)
Heater tube to regenerator plenum junction, cm ³ (in. ³)	5.90 (0.36)
Regenerator hot end plenum, cm ³ (in. ³)	4.10 (0.25)
Regenerator plenum ring, cm ³ (in. ³)	0.83 (0.05)
Instrumentation ports (heater/regenerator), cm ³ (in. ³)	1.64 (0.10)
Regenerator 1, cm ³ (in. ³)	56.1 (3.42)
Regenerator 2, cm ³ (in. ³)	69.1 (3.67)
Regenerator cold end plenum, cm ³ (in. ³)	4.23 (0.26)
Instrumentation ports (regenerator/cooler), cm ³ (in. ³)	3.41 (0.21)
Cooler, cm ³ (in. ³)	20.42 (1.25)
Cooler plenum at the compression space, cm ³ (in. ³)	7.15 (0.44)
Compression space instrumentation ports, cm ³ (in. ³)	3.15 (0.19)
Cylinder ports, cm ³ (in. ³)	1.21 (0.07)
Piston/spider clearance, cm ³ (in. ³)	18.4 (1.21)
Annular ring around the spider, cm ³ (in. ³)	3.82 (0.23)
Displacer LVDT core, cm ³ (in. ³)	2.95 (0.18)
Power piston center port, cm ³ (in. ³)	5.90 (0.36)
Materials	
Heater head	
Regenerator outer cylinder	316SS
Expansion space dome	316SS
Regenerator inner cylinder wall	316SS
Displacer	321SS
Cooler	6061-T6 Al
Cylinder	
Power piston	6061-T6 Al with chrome oxide coating
Displacer	304SS with chrome oxide coating
Standard power piston	6061-T6 Al body with Xylan Coating and mild steel mass
Light power piston	6061-T6 Al body with Xylan coating and Al mass
Diametral clearances	
Displacer 1 rod/rod cylinder, mm (in.)	0.030 (0.0012)
Displacer 1 body/displacer cylinder, mm (in.)	0.381 (0.015)
Displacer 2 rod/rod cylinder, mm (in.)	0.020 (0.0008)
Displacer 2 body/displacer cylinder, mm (in.)	0.381 (0.015)
Power piston/piston cylinder, mm (in.)	0.033 (0.001)

TABLE II. - Concluded

Displacer gas spring	
No. 1 design mean volume, cm ³ (in. ³)	31.79 (1.94)
No. 1 rod diameter, cm (in.)	1.663 (0.655)
No. 2 design mean volume, cm ³ (in. ³)	18.8 (1.147)
No. 2 rod diameter, cm (in.)	1.806 (0.711)
Center ports	
Power piston port location, distance from inward limit to center port opening position, cm (in.)	2.05 (0.81)
Displacer 1, distance from expansion space limit to position where the center port opens, cm (in.)	1.90 (0.75)
Displacer 2, distance from expansion space limit to position where the center port opens, cm (in.)	2.64 (1.04)
Center port diameter, power piston, mm (in.)	1.1 (0.042)
Center port diameter, displacers, mm (in.)	1.0 (0.040)

TABLE III. - RE-1000 ENGINE VOLUMES WHEN THE PISTON
AND DISPLACER AT THEIR CENTER PORT POSITIONS

Expansion space	cm ³ (in. ³)
Clearance around displacer 1	9.83 (0.60)
Clearance around displacer 2	8.36 (0.51)
Between the displacer and the end of the cylinder	50.1 (3.06)
Instrumentation ports	1.64 (0.10)
Total with D1	61.6 (3.76)
Expansion-heater connecting duct	
Expansion to heater tube junction	3.80 (0.232)
Heater assembly	
Heated portion of the tubes	27.4 (1.67)
Heater-regenerator duct	
Heater tube to regenerator plenum junction	5.90 (0.360)
Regenerator hot end plenum	4.10 (0.250)
Regenerator plenum ring	0.83 (0.051)
Instrumentation ports	1.64 (0.10)
Total	12.47 (0.761)
Regenerator	
Regenerator dead volume R1	56.1 (3.42)
Regenerator dead volume R2	60.1 (3.67)
Regenerator-cooler duct	
Regenerator cold end plenum ring	0.84 (0.051)
Regenerator cold end plenum	4.23 (0.258)
Instrumentation ports	0.93 (0.057)
Heater flange fittings	3.41 (0.208)
Total	9.41 (0.574)
Cooler	
Cooler passages	20.4 (1.25)
Cooler-compression space duct	
Cooler plenum at compression space	7.14 (0.436)
Cylinder ports	1.21 (0.074)
Total	8.36 (0.510)
Compression space	
Between displacer and spider with D1	49.0 (2.99)
Between displacer and spider with D2	58.5 (3.57)
From piston to spider passages	16.06 (0.98)
Annular space around spider	3.82 (0.23)
Between power piston and spider	18.4 (1.12)
Power piston centering port passage	5.90 (0.36)
Displacer LVDT inside of piston	2.95 (0.18)
Anti rotation rod clearance holes	5.16 (0.32)
Instrumentation ports	3.15 (0.19)
Total with D1	104.4 (6.37)

TABLE IV. - RE-1000 INSTRUMENTATION

Mnemonic	Parameter	Range	Type	SS	V	E±
1 MEANCP	Mean comp space pressure, kPa	0-13800	Strain gauge	X		20
2 MEANBP	Mean bounce pressure, kPa	0-1380	Strain gauge			20
3 PRESUP	Gas supply pressure, kPa	0-13800	Strain gauge			20
4 T01HTR	Heater tube metal temp, °C	400-825	Thermocouple			10
5 T02HTR						
6 T03HTR						
7 T04HTR						
8 T05HTR						
9 T06HTR						
10 T07HTR						
11 T08HTR						
12 T09HTR						
13 T10HTR						
14 T11HTR						
15 T12HTR						
16 T03HED	Head metal temp, °C					
17 T13REG	Regen. vert. profile, °C					
18 T14REG	Regen. vert. profile, °C					
19 T15REG	Regen. circular profile, °C	250-825				5
20 T16REG						
21 T17REG						
22 T18REG						
23 T19REG	Regen. vert. profile, °C	20-250				1
24 TGBOUN	Regen. vert. profile, °C	20-80				
25 TGCOMP	Comp. space gas temp, °C	20-250				
26 TGREGC	Regen.-cooler gas temp, °C	20-250				
27 TGREGH	Regen.-heater gas temp, °C	400-825				10
28 TGEXP	Expansion space gas temp, °C	400-825				10
29 TWINDP	Dashpot water inlet temp, °C	10-70				1
30 TDLDP	Dashpot water delta temp, °C	0-20				0.2
31 TWODP	Dashpot water outlet temp, °C	10-70				1
32 TWINCL	Cooler water inlet temp, °C	10-70				1
33 IDLCL	Cooler water delta temp, °C	0-20				0.2
34 TWOCL	Cooler outlet temp, °C	10-70				0.2
35 AMPS1	Heater amps, supply 1, A	0-1000	Ammeter			10
36 AMPS2	Heater amps, supply 2, A	0-1000	Ammeter			10
37 VOLTG	Heater voltage, V	0-20	Voltmeter			0.01
38 FLODP	Dashpot water flow, l/min	0-10	Turbine meter			0.1
39 FLOCLR	Cooler water flow, l/min	0-10	Turbine meter			
40 VX1HOR	Horiz. vibration, cm/sec	0-3.8	Accelerometer			
41 VY1VER	Vert. vibration, cm/sec	0-3.8	Accelerometer			
42 PISTST	Piston stroke, cm	0-4	Stroke meter			0.02
43 DISPST	Displacer stroke, cm	0-4	Stroke meter			0.02
44 INDPWR	Indicated power, W	0-3000	Analog circ.			40
45 PWROUT	Brake power, kW	0-3.0	Analog circ.			40
46 FPIST	Load force on piston, N	0-1600	F transducer	X		20
47 XPIST	Piston position, cm	±2.0	LVDT	X		0.02

TABLE IV. - Concluded

Mnemonic	Parameter	Range	Type	SS	V	E±
48 XDOTP	Piston velocity, m/sec	+8.0	LVT		X	0.1
49 XDISP	Displacer position, cm	+2.0	LVDT		↓	0.04
50 PDYNC	Comp. space pressure, kPa	+2000	Crystal			10
51 PDLCLR	Cooler delta pressure, kPa	68.9	Diff. P xducer		↓	2.5
52 PDLREG	Regen. delta pressure, kPa	138	Diff. P xducer	X		5
53 PDLDIS	Displacer delta pressure, kPa	138	Diff. P xducer		↓	5
54 PDYNDB	Disp. gas spring press., MPa	10.0	Strain gauge		↓	0.1
55 PAPRES	Phase angle of pressure, deg	0-360	Phase meter	X		1
56 PADISP	Phase angle of displacer, deg	0-360	Phase meter	X		1
57 FREQ	Engine frequency, Hz	0-50	Freq. to dc	X		0.05
58 PAMPC	Comp. pressure ampl., kPa	0-2000	Crystal		X	20

TABLE V. - RE-1000 CALCULATIONS

Mnemonic	Description of the calculation
PWRIN	Electric power input to the heater head
QCOOLR	Heat rejected by the engine cooler
QDSHPT	Heat rejected by the dashpot
EXTEFF	Engine efficiency based on the brake power output and the heater power input
TAVHTR	Average heater tube outside temperature
INTEFF	Engine efficiency based on the brake power output and brake power plus QCOOLR used as the input
AMPS	Total amperage to the heater head
QDISPG	Heat conduction through the gas in the displacer
QDISP	Heat conduction through the displacer body
QREG1	Outer regenerator wall conduction
QREG2	Outer regenerator wall conduction
QREG3	Inner regenerator wall conduction
PWROUT	Brake power output, analog calculation
INDPWR	Indicated power output, analog calculation
PISTST	Power piston stroke
DISPST	Displacer stroke

TABLE VI. - RE-1000 SAMPLE DATA POINTS SINGLE VARIABLE DATA SET

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/17/83 10:17:21.80 RDG 602

FLUID HELIUM		BAROM 14.268 PSI		REGENERATOR 2		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.10 L/MIN	AMPS1	776. AMPS	PRESUP	7482. KPA	TGEXP	554.8 DEG.C	T01HTR	609.3 DEG.C
		AMPS2	651. AMPS					T02HTR	609.9 DEG.C
TWINDP	17.3 DEG.C	VOLTG	3.11 VOLTS	MEANBP	7013. KPA	TGREGH	547.0 DEG.C	T03HTR	619.7 DEG.C
TDLDP	2.94 DEG.C			MEANCP	7056. KPA	TGREGC	101.9 DEG.C	T04HTR	594.8 DEG.C
TWODPR	20.9 DEG.C					TGCOMP	61.6 DEG.C	T05HTR	604.4 DEG.C
						TGBOUN	36.9 DEG.C	T06HTR	602.3 DEG.C
								T07HTR	573.2 DEG.C
								T08HTR	605.9 DEG.C
								T09HTR	619.2 DEG.C
								T10HTR	546.2 DEG.C
								T11HTR	603.4 DEG.C
								T12HTR	603.9 DEG.C
								T13REG	553.7 DEG.C
								T14REG	514.8 DEG.C
								T15REG	376.7 DEG.C
								T16REG*****	DEG.C
								T17REG	355.8 DEG.C
								T18REG	366.0 DEG.C
								T19REG	235.5 DEG.C
								T03HED	293.5 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.17 L/MIN	1 PWRIN	4433. WATTS	VX1HOR	0.5 CM/S	PWROUT	1216. WATTS		
TWINCL	25.2 DEG.C	2 QCOOLR	2789. WATTS	VY1VER	8.5 CM/S	INDPWR	1262. WATTS		
TDLCL	9.63 DEG.C	3 QDSHPT	840. WATTS	PHASE ANGLES		PISTST	2.58 CM		
TWOCLR	34.80 DEG.C	4 EXTEFF	27.4 %	PADISP	57.1 DEG.	DISPST	2.72 CM		
		5 TAVHTR	599.3 DEG.C	PAPRES	-21.6 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	30.4 %	ENGINE SPEED		PAMPC	1127. KPA		
		8 AMPS	1427. AMPS	FREQ	29.9 HZ	DISPCP	2.11 CM		
		9 QDISPG	3. WATTS			PISTCP	2.64 CM		
		10 QDISP	14. WATTS			PDYNDB*****	KPA		
		11 QREG1	106. WATTS			PDLCLR*****	KPA		
		12 QREG2	121. WATTS			PDLREG*****	KPA		
		13 QREG3	34. WATTS			PDLDIS*****	KPA		
		14 PFP	1019.5 KPA/CM						
		15 PFD	362.7 KPA/CM						
		16 NBEALE	0.00875						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:35:55.03 RDG 1006

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.96 L/MIN	AMPS1	534. AMPS	PRESUP	7365. KPA	TGEXP	567.8 DEG.C	T01HTR	601.6 DEG.C
		AMPS2	503. AMPS					T02HTR	605.5 DEG.C
TWINDP	17.7 DEG.C	VOLTG	2.25 VOLTS	MEANBP	7011. KPA	TGREGH	564.5 DEG.C	T03HTR	612.6 DEG.C
TDLDP	1.15 DEG.C			MEANCP	7040. KPA	TGREGC	90.7 DEG.C	T04HTR	606.4 DEG.C
TWODPR	19.6 DEG.C					TGCOMP	39.9 DEG.C	T05HTR	61.3 DEG.C
						TGBOUN	27.9 DEG.C	T06HTR	600.3 DEG.C
								T07HTR	604.5 DEG.C
								T08HTR	584.1 DEG.C
								T09HTR	608.8 DEG.C
								T10HTR	591.8 DEG.C
								T11HTR	578.7 DEG.C
								T12HTR	580.2 DEG.C
								T13REG*****	DEG.C
								T14REG	518.3 DEG.C
								T15REG	384.4 DEG.C
								T16REG	386.7 DEG.C
								T17REG	376.6 DEG.C
								T18REG	372.8 DEG.C
								T19REG	225.5 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWRIN	2338. WATTS	VX1HOR	0.2 CM/S	PWROUT	532. WATTS		
TWINCL	24.7 DEG.C	2 QCOOLR	1892. WATTS	VY1VER	5.9 CM/S	INDPWR	570. WATTS		
TDLCL	6.47 DEG.C	3 QDSHPT	317. WATTS	PHASE ANGLES		PISTST	1.80 CM		
TWOCLR	29.47 DEG.C	4 EXTEFF	22.7 %	PADISP	56.0 DEG.	DISPST	1.83 CM		
		5 TAVHTR	599.6 DEG.C	PAPRES	-19.8 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	21.9 %	ENGINE SPEED		PAMPC	792. KPA		
		8 AMPS	1037. AMPS	FREQ	30.2 HZ	DISPCP	2.13 CM		
		9 QDISPG	3. WATTS			PISTCP	2.31 CM		
		10 QDISP	15. WATTS			PDYNDB*****	KPA		
		11 QREG1	119. WATTS			PDLCLR	3.22 KPA		
		12 QREG2	118. WATTS			PDLREG	37.33 KPA		
		13 QREG3	37. WATTS			PDLDIS	65.16 KPA		
		14 PFP	1029.1 KPA/CM						
		15 PFD	353.7 KPA/CM						
		16 NBEALE	0.00544						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:40:22.06 RDG 1007

FLUID HELIUM		BAROM 14.303 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	3.94 L/MIN	AMPS1	571. AMPS	PRESUP	7469. KPA	TGEXP	564.8 DEG.C	T01HTR	602.6 DEG.C
		AMPS2	533. AMPS					T02HTR	605.0 DEG.C
TWINDP	17.8 DEG.C	VOLTG	2.40 VOLTS	MEANBP	6982. KPA	TGREGH	563.7 DEG.C	T03HTR	613.7 DEG.C
TDLDP	1.42 DEG.C			MEANCP	7013. KPA	TGREGC	92.6 DEG.C	T04HTR	605.5 DEG.C
TWODPR	19.8 DEG.C					TGCOMP	42.0 DEG.C	T05HTR	615.7 DEG.C
						TGBOUN	29.0 DEG.C	T06HTR	607.1 DEG.C
								T07HTR	605.4 DEG.C
								T08HTR	583.8 DEG.C
								T09HTR	609.7 DEG.C
								T10HTR	590.4 DEG.C
								T11HTR	577.8 DEG.C
								T12HTR	577.7 DEG.C
								T13REG*****	DEG.C
								T14REG	517.9 DEG.C
								T15REG	385.7 DEG.C
								T16REG	388.3 DEG.C
								T17REG	377.9 DEG.C
								T18REG	373.3 DEG.C
								T19REG	228.0 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.24 L/MIN	1 PWRIN	2648. WATTS	VX1HOR	0.3 CM/S	PWROUT	622. WATTS		
TWINCL	24.7 DEG.C	2 QCOOLR	2004. WATTS	VY1VER	6.6 CM/S	INDPWR	665. WATTS		
TDLCL	6.82 DEG.C	3 QDSHPT	390. WATTS	PHASE ANGLES		PISTST	2.01 CM		
TWOCLR	29.95 DEG.C	4 EXTEFF	23.5 %	PADISP	56.6 DEG.	DISPST	1.97 CM		
		5 TAVHTR	599.5 DEG.C	PAPRES	-18.9 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	23.7 %	ENGINE SPEED		PAMPC	881. KPA		
		8 AMPS	1104. AMPS	FREQ	30.2 HZ	DISPCP	2.24 CM		
		9 QDISPG	3. WATTS			PISTCP	2.34 CM		
		10 QDISP	15. WATTS			PDYNDB*****	KPA		
		11 QREG1	118. WATTS			PDLCLR	3.73 KPA		
		12 QREG2	117. WATTS			PDLREG	42.43 KPA		
		13 QREG3	36. WATTS			PDLDIS	73.65 KPA		
		14 PFP	1018.7 KPA/CM						
		15 PFD	345.8 KPA/CM						
		16 NBEALE	0.00573						

TABLE VI. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:46:40.06 RDG 1008

FLUID HELIUM		BAROM 14.303 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	3.94 L/MIN	AMPS1	604. AMPS	PRESUP	7055. KPA	TGEXP	562.6 DEG.C
		AMPS2	357. AMPS				
TWINDP	17.8 DEG.C	VOLTG	2.53 VOLTS	MEANBP	6988. KPA	TGREGH	563.2 DEG.C
TDLDP	1.72 DEG.C			MEANCP	7023. KPA	TGREGC	94.9 DEG.C
TWODPR	20.2 DEG.C					TGCOMP	45.4 DEG.C
						TGBOUN	30.6 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.23 L/MIN	1 PWRIN	2930. WATTS	VX1HOR	0.3 CM/S	PWROUT	703. WATTS
TWINCL	24.7 DEG.C	2 QCOOLR	2233. WATTS	VY1VER	7.2 CM/S	INDPWR	753. WATTS
TDLCL	7.61 DEG.C	3 QDSHPT	470. WATTS			PISTST	2.20 CM
TWOCLR	30.73 DEG.C	4 EXTEFF	24.0 %	PHASE ANGLES		DISPST	2.09 CM
		5 TAVHTR	599.9 DEG.C	PADISP	56.7 DEG.		
		6 INTEFF	23.9 %	PAPRES	-17.7 DEG.		
		8 AMPS	1160. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	30.2 HZ	PAMPC	979. KPA
		10 QDISP	15. WATTS			DISPCP	2.27 CM
		11 QREG1	117. WATTS			PISTCP	2.34 CM
		12 QREG2	115. WATTS			PDYNDB*****	KPA
		13 QREG3	36. WATTS			PDLCLR	4.07 KPA
		14 PFP	1025.2 KPA/CM			PDLREG	47.86 KPA
		15 PFD	339.6 KPA/CM			PDLDIS	82.13 KPA
		16 NBEALE	6.00589				
							T13REG***** DEG.C
							T14REG 518.7 DEG.C
							T15REG 388.8 DEG.C
							T16REG 389.2 DEG.C
							T17REG 380.1 DEG.C
							T18REG 376.6 DEG.C
							T19REG 232.8 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:50:22.06 RDG 1009

FLUID HELIUM		BAROM 14.303 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	3.97 L/MIN	AMPS1	637. AMPS	PRESUP	7339. KPA	TGEXP	561.2 DEG.C
		AMPS2	585. AMPS				
TWINDP	17.8 DEG.C	VOLTG	2.66 VOLTS	MEANBP	6999. KPA	TGREGH	558.7 DEG.C
TDLDP	1.97 DEG.C			MEANCP	7036. KPA	TGREGC	94.3 DEG.C
TWODPR	20.5 DEG.C					TGCOMP	48.5 DEG.C
						TGBOUN	32.1 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.23 L/MIN	1 PWRIN	3253. WATTS	VX1HOR	0.3 CM/S	PWROUT	794. WATTS
TWINCL	24.7 DEG.C	2 QCOOLR	2505. WATTS	VY1VER	7.9 CM/S	INDPWR	856. WATTS
TDLCL	8.53 DEG.C	3 QDSHPT	545. WATTS			PISTST	2.40 CM
TWOCLR	31.51 DEG.C	4 EXTEFF	24.4 %	PHASE ANGLES		DISPST	2.22 CM
		5 TAVHTR	600.3 DEG.C	PADISP	57.1 DEG.		
		6 INTEFF	24.1 %	PAPRES	-16.5 DEG.		
		8 AMPS	1222. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	30.2 HZ	PAMPC	1063. KPA
		10 QDISP	15. WATTS			DISPCP	2.37 CM
		11 QREG1	108. WATTS			PISTCP	2.38 CM
		12 QREG2	125. WATTS			PDYNDB*****	KPA
		13 QREG3	36. WATTS			PDLCLR	5.43 KPA
		14 PFP	1012.0 KPA/CM			PDLREG	52.95 KPA
		15 PFD	324.2 KPA/CM			PDLDIS	90.28 KPA
		16 NBEALE	0.00609				
							T13REG***** DEG.C
							T14REG 517.8 DEG.C
							T15REG 384.9 DEG.C
							T16REG 378.4 DEG.C
							T17REG 364.0 DEG.C
							T18REG 364.0 DEG.C
							T19REG 231.2 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 15:55:25.06 RDG 1010

FLUID HELIUM		BAROM 14.303 PSI	REGENERATOR 1	DISPLACER 1	STANDARD PISTON		
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES	
FLODP	3.96 L/MIN	AMPS1	673. AMPS	PRESUP	7464. KPA	TGEXP	557.5 DEG.C
		AMPS2	619. AMPS				
TWINDP	17.8 DEG.C	VOLTG	2.82 VOLTS	MEANBP	6988. KPA	TGREGH	548.5 DEG.C
TDLDP	2.21 DEG.C			MEANCP	7032. KPA	TGREGC	90.7 DEG.C
TWODPR	20.8 DEG.C					TGCOMP	55.7 DEG.C
						TGBOUN	33.4 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS	
FLOCLR	4.22 L/MIN	1 PWRIN	3643. WATTS	VX1HOR	0.4 CM/S	PWROUT	866. WATTS
TWINCL	24.8 DEG.C	2 QCOOLR	2736. WATTS	VY1VER	8.5 CM/S	INDPWR	939. WATTS
TDLCL	9.34 DEG.C	3 QDSHPT	607. WATTS			PISTST	2.60 CM
TWOCLR	32.52 DEG.C	4 EXTEFF	25.8 %	PHASE ANGLES		DISPST	2.34 CM
		5 TAVHTR	598.6 DEG.C	PADISP	57.5 DEG.		
		6 INTEFF	24.0 %	PAPRES	-15.7 DEG.		
		8 AMPS	1292. AMPS	ENGINE SPEED		DYNAMIC CALCULATIONS	
		9 QDISPG	3. WATTS	FREQ	30.1 HZ	PAMPC	1147. KPA
		10 QDISP	14. WATTS			DISPCP	2.37 CM
		11 QREG1	96. WATTS			PISTCP	2.33 CM
		12 QREG2	139. WATTS			PDYNDB*****	KPA
		13 QREG3	35. WATTS			PDLCLR	6.79 KPA
		14 PFP	1002.8 KPA/CM			PDLREG	55.66 KPA
		15 PFD	314.1 KPA/CM			PDLDIS	95.71 KPA
		16 NBEALE	0.00617				
							T13REG***** DEG.C
							T14REG 510.3 DEG.C
							T15REG 368.0 DEG.C
							T16REG 350.7 DEG.C
							T17REG 331.5 DEG.C
							T18REG 338.9 DEG.C
							T19REG 221.0 DEG.C
							T03HED***** DEG.C

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:00:55.09 RDG 1011

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:03:22.09 RDG 1012

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/26/84 16:30:25.12 RDG 1017

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TABLE VI. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 10:57:22.54 RDG 1024

FLUID HELIUM		BAROM 14.327 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.47 L/MIN	AMPS1	646. AMPS	PRESUP	7572. KPA	TGEXP	462.4 DEG.C	T01HTR	501.0 DEG.C
TWINDP	16.2 DEG.C	AMPS2	602. AMPS	MEANBP	7000. KPA	TGREGH	460.8 DEG.C	T02HTR	507.9 DEG.C
TDLDP	1.71 DEG.C	VOLTG	2.68 VOLTS	MEANCP	7048. KPA	TGREGC	90.5 DEG.C	T03HTR	515.5 DEG.C
TWODPR	18.6 DEG.C					TGCOMP	53.8 DEG.C	T04HTR	503.0 DEG.C
						TGBOUN	31.8 DEG.C	T05HTR	519.9 DEG.C
								T06HTR	511.0 DEG.C
								T07HTR	510.1 DEG.C
								T08HTR	481.0 DEG.C
								T09HTR	507.7 DEG.C
								T10HTR	488.1 DEG.C
								T11HTR	471.9 DEG.C
								T12HTR	476.6 DEG.C
								T13REG*****	DEG.C
								T14REG	431.6 DEG.C
								T15REG	326.4 DEG.C
								T16REG	317.8 DEG.C
								T17REG	307.7 DEG.C
								T18REG	308.1 DEG.C
								T19REG	203.0 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	3348. WATTS	VX1HOR	0.4 CM/S	PWROUT	707. WATTS		
TWINCL	24.7 DEG.C	2 QC00LR	2681. WATTS	VY1VER	8.6 CM/S	INDPWR	765. WATTS		
TDLCL	***** DEG.C	3 QDSHPT	531. WATTS	PHASE ANGLES		PISTST	2.60 CM		
TWOCLR	32.15 DEG.C	4 EXTEFF	21.1 %	PADISP	60.4 DEG.	DISPST	2.30 CM		
		5 TAVHTR	499.5 DEG.C	PAPRES	-12.5 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	20.9 %	ENGINE SPEED		PAMPC	1176. KPA		
		8 AMPS	1248. AMPS	FREQ	30.2 HZ	DISPCP	2.32 CM		
		9 QDISPG	2. WATTS			PISTCP	2.28 CM		
		10 QDISP	12. WATTS			PDYNDB*****	KPA		
		11 QREG1	85. WATTS			PDLCLR	6.96 KPA		
		12 QREG2	100. WATTS			PDLREG	58.04 KPA		
		13 QREG3	29. WATTS			PDLDIS	99.10 KPA		
		14 PFP	995.1 KPA/CM						
		15 PFD	254.7 KPA/CM						
		16 NBEALE	0.00501						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 07/27/84 11:35:10.54 RDG 1030

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.11 L/MIN	AMPS1	626. AMPS	PRESUP	7562. KPA	TGEXP	415.3 DEG.C	T01HTR	452.3 DEG.C
TWINDP	16.0 DEG.C	AMPS2	589. AMPS	MEANBP	7007. KPA	TGREGH	415.9 DEG.C	T02HTR	456.8 DEG.C
TDLDP	1.90 DEG.C	VOLTG	2.60 VOLTS	MEANCP	7061. KPA	TGREGC	88.9 DEG.C	T03HTR	465.5 DEG.C
TWODPR	18.6 DEG.C					TGCOMP	54.6 DEG.C	T04HTR	453.6 DEG.C
						TGBOUN	33.1 DEG.C	T05HTR	469.0 DEG.C
								T06HTR	459.9 DEG.C
								T07HTR	458.8 DEG.C
								T08HTR	432.9 DEG.C
								T09HTR	458.6 DEG.C
								T10HTR	439.4 DEG.C
								T11HTR	424.0 DEG.C
								T12HTR	426.7 DEG.C
								T13REG*****	DEG.C
								T14REG	389.2 DEG.C
								T15REG	299.6 DEG.C
								T16REG	294.9 DEG.C
								T17REG	288.2 DEG.C
								T18REG	286.8 DEG.C
								T19REG	190.0 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.21 L/MIN	1 PWRIN	3159. WATTS	VX1HOR	0.4 CM/S	PWROUT	605. WATTS		
TWINCL	24.3 DEG.C	2 QC00LR	2595. WATTS	VY1VER	8.6 CM/S	INDPWR	643. WATTS		
TDLCL	8.88 DEG.C	3 QDSHPT	542. WATTS	PHASE ANGLES		PISTST	2.60 CM		
TWOCLR	31.77 DEG.C	4 EXTEFF	19.2 %	PADISP	61.8 DEG.	DISPST	2.26 CM		
		5 TAVHTR	449.8 DEG.C	PAPRES	-10.3 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	18.9 %	ENGINE SPEED		PAMPC	1181. KPA		
		8 AMPS	1215. AMPS	FREQ	30.2 HZ	DISPCP	2.30 CM		
		9 QDISPG	2. WATTS			PISTCP	2.25 CM		
		10 QDISP	10. WATTS			PDYNDB*****	KPA		
		11 QREG1	79. WATTS			PDLCLR	6.79 KPA		
		12 QREG2	83. WATTS			PDLREG	58.38 KPA		
		13 QREG3	25. WATTS			PDLDIS	100.49 KPA		
		14 PFP	980.6 KPA/CM						
		15 PFD	212.3 KPA/CM						
		16 NBEALE	0.00428						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/03/84 14:58:23.35 RDG 1070

FLUID HELIUM		BAROM 14.342 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.20 L/MIN	AMPS1	635. AMPS	PRESUP	5553. KPA	TGEXP	555.0 DEG.C	T01HTR	599.7 DEG.C
TWINDP	22.4 DEG.C	AMPS2	521. AMPS	MEANBP	5494. KPA	TGREGH	554.7 DEG.C	T02HTR	606.6 DEG.C
TDLDP	1.70 DEG.C	VOLTG	2.49 VOLTS	MEANCP	5531. KPA	TGREGC	79.7 DEG.C	T03HTR	614.7 DEG.C
TWODPR	24.7 DEG.C					TGCOMP	49.4 DEG.C	T04HTR	601.3 DEG.C
						TGBOUN	37.7 DEG.C	T05HTR	619.1 DEG.C
								T06HTR	609.4 DEG.C
								T07HTR	608.6 DEG.C
								T08HTR	580.8 DEG.C
								T09HTR	607.0 DEG.C
								T10HTR	586.7 DEG.C
								T11HTR	573.8 DEG.C
								T12HTR	577.7 DEG.C
								T13REG*****	DEG.C
								T14REG	515.2 DEG.C
								T15REG	379.5 DEG.C
								T16REG	367.6 DEG.C
								T17REG	352.7 DEG.C
								T18REG	355.3 DEG.C
								T19REG	222.0 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.40 L/MIN	1 PWRIN	2879. WATTS	VX1HOR	0.3 CM/S	PWROUT	720. WATTS		
TWINCL	25.2 DEG.C	2 QC00LR	2137. WATTS	VY1VER	7.1 CM/S	INDPWR	759. WATTS		
TDLCL	7.00 DEG.C	3 QDSHPT	495. WATTS	PHASE ANGLES		PISTST	2.61 CM		
TWOCLR	30.99 DEG.C	4 EXTEFF	25.0 %	PADISP	58.7 DEG.	DISPST	2.46 CM		
		5 TAVHTR	598.8 DEG.C	PAPRES	-18.0 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	25.2 %	ENGINE SPEED		PAMPC	910. KPA		
		8 AMPS	1156. AMPS	FREQ	26.9 HZ	DISPCP	2.25 CM		
		9 QDISPG	3. WATTS			PISTCP	2.46 CM		
		10 QDISP	14. WATTS			PDYNDB*****	KPA		
		11 QREG1	108. WATTS			PDLCLR	4.41 KPA		
		12 QREG2	130. WATTS			PDLREG	42.43 KPA		
		13 QREG3	37. WATTS			PDLDIS	71.95 KPA		
		14 PFP	795.4 KPA/CM						
		15 PFD	267.8 KPA/CM						
		16 NBEALE	0.00729						

TABLE VI. - CONTINUED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/06/84 14:52:24.59 RDG 1079

FLUID HELIUM		BAROM 14.288 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.08 L/MIN	AMPS1	539. AMPS	PRESUP	4143. KPA	TGEXP	553.1 DEG.C	T01HTR	605.6 DEG.C
TWINDP	23.0 DEG.C	AMPS2	451. AMPS	MEANBP	3988. KPA	TGREGH	560.1 DEG.C	T02HTR	599.1 DEG.C
TDLDP	0.81 DEG.C	VOLTG	2.13 VOLTS	MEANCP	4012. KPA	TGREGC	70.5 DEG.C	T03HTR	614.7 DEG.C
TWODPR	24.4 DEG.C					TGCOMP	41.6 DEG.C	T04HTR	603.7 DEG.C
						TGBOUN	32.0 DEG.C	T05HTR	615.3 DEG.C
								T06HTR	600.7 DEG.C
								T07HTR	601.4 DEG.C
								T08HTR	589.7 DEG.C
								T09HTR	607.8 DEG.C
								T10HTR	591.6 DEG.C
								T11HTR	581.8 DEG.C
								T12HTR	581.3 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.31 L/MIN	1 PWRIN	2109. WATTS	VX1HOR	0.2 CM/S	PWROUT	500. WATTS	T13REG*****	DEG.C
TWINCL	25.2 DEG.C	2 QCOOLR	1242. WATTS	VY1VER	5.8 CM/S	INDPWR	520. WATTS	T14REG	512.2 DEG.C
TDLCL	4.15 DEG.C	3 QDSHPT	228. WATTS			PISTST	2.61 CM	T15REG	376.1 DEG.C
TWOCLR	29.17 DEG.C	4 EXTEFF	23.7 %	PHASE ANGLES		DISPST	2.55 CM	T16REG	367.6 DEG.C
		5 TAVHTR	599.4 DEG.C	PADISP	58.4 DEG.			T17REG	363.5 DEG.C
		6 INTEFF	28.7 %	PAPRES	-19.7 DEG.			T18REG	363.0 DEG.C
		8 AMPS	990. AMPS	ENGINE SPEED				T19REG	353.7 DEG.C
		9 QDISPG	3. WATTS	FREQ	23.0 HZ				
		10 QDISP	15. WATTS						
		11 QREG1	121. WATTS						
		12 QREG2	121. WATTS						
		13 QREG3	38. WATTS						
		14 PFP	568.8 KPA/CM						
		15 PFD	200.5 KPA/CM						
		16 NBEALE	0.00815						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/08/84 13:36:21.40 RDG 1121

FLUID HELIUM		BAROM 14.234 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.37 L/MIN	AMPS1	702. AMPS	PRESUP	7093. KPA	TGEXP	557.0 DEG.C	T01HTR	602.6 DEG.C
TWINDP	25.4 DEG.C	AMPS2	595. AMPS	MEANBP	7012. KPA	TGREGH	559.4 DEG.C	T02HTR	604.7 DEG.C
TDLDP	2.18 DEG.C	VOLTG	2.78 VOLTS	MEANCP	7066. KPA	TGREGC	109.0 DEG.C	T03HTR	616.7 DEG.C
TWODPR	28.0 DEG.C					TGCOMP	64.0 DEG.C	T04HTR	603.0 DEG.C
						TGBOUN	44.0 DEG.C	T05HTR	621.4 DEG.C
								T06HTR	609.5 DEG.C
								T07HTR	608.0 DEG.C
								T08HTR	580.9 DEG.C
								T09HTR	611.3 DEG.C
								T10HTR	586.0 DEG.C
								T11HTR	574.7 DEG.C
								T12HTR	574.8 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.36 L/MIN	1 PWRIN	3599. WATTS	VX1HOR	0.4 CM/S	PWROUT	845. WATTS	T13REG*****	DEG.C
TWINCL	39.9 DEG.C	2 QCOOLR	2532. WATTS	VY1VER	8.6 CM/S	INDPWR	907. WATTS	T14REG	519.6 DEG.C
TDLCL	8.42 DEG.C	3 QDSHPT	662. WATTS			PISTST	2.61 CM	T15REG	391.9 DEG.C
TWOCLR	46.99 DEG.C	4 EXTEFF	23.5 %	PHASE ANGLES		DISPST	2.35 CM	T16REG	391.7 DEG.C
		5 TAVHTR	599.5 DEG.C	PADISP	58.5 DEG.			T17REG	386.0 DEG.C
		6 INTEFF	25.0 %	PAPRES	-14.9 DEG.			T18REG	378.7 DEG.C
		8 AMPS	1296. AMPS	ENGINE SPEED				T19REG	241.7 DEG.C
		9 QDISPG	3. WATTS	FREQ	30.3 HZ				
		10 QDISP	14. WATTS						
		11 QREG1	111. WATTS						
		12 QREG2	114. WATTS						
		13 QREG3	35. WATTS						
		14 PFP	1007.1 KPA/CM						
		15 PFD	300.9 KPA/CM						
		16 NBEALE	0.00593						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 08/27/84 14:05:40.27 RDG 1200

FLUID HELIUM		BAROM 14.332 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.01 L/MIN	AMPS1	694. AMPS	PRESUP	7311. KPA	TGEXP	558.0 DEG.C	T01HTR	600.3 DEG.C
TWINDP	18.7 DEG.C	AMPS2	569. AMPS	MEANBP	6986. KPA	TGREGH	558.6 DEG.C	T02HTR	607.1 DEG.C
TDLDP	2.63 DEG.C	VOLTG	2.77 VOLTS	MEANCP	7031. KPA	TGREGC	123.0 DEG.C	T03HTR	614.6 DEG.C
TWODPR	22.1 DEG.C					TGCOMP	74.7 DEG.C	T04HTR	605.0 DEG.C
						TGBOUN	43.1 DEG.C	T05HTR	619.5 DEG.C
								T06HTR	611.7 DEG.C
								T07HTR	610.3 DEG.C
								T08HTR	578.4 DEG.C
								T09HTR	612.6 DEG.C
								T10HTR	584.7 DEG.C
								T11HTR	575.1 DEG.C
								T12HTR	575.6 DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.15 L/MIN	1 PWRIN	3493. WATTS	VX1HOR	0.6 CM/S	PWROUT	777. WATTS	T13REG*****	DEG.C
TWINCL	54.0 DEG.C	2 QCOOLR	2108. WATTS	VY1VER	8.4 CM/S	INDPWR	837. WATTS	T14REG	522.7 DEG.C
TDLCL	7.39 DEG.C	3 QDSHPT	734. WATTS			PISTST	2.60 CM	T15REG	402.6 DEG.C
TWOCLR	60.48 DEG.C	4 EXTEFF	22.2 %	PHASE ANGLES		DISPST	2.29 CM	T16REG	402.4 DEG.C
		5 TAVHTR	599.6 DEG.C	PADISP	59.1 DEG.			T17REG	388.9 DEG.C
		6 INTEFF	26.9 %	PAPRES	-13.7 DEG.			T18REG	382.4 DEG.C
		8 AMPS	1263. AMPS	ENGINE SPEED				T19REG	255.2 DEG.C
		9 QDISPG	3. WATTS	FREQ	30.2 HZ				
		10 QDISP	14. WATTS						
		11 QREG1	103. WATTS						
		12 QREG2	114. WATTS						
		13 QREG3	34. WATTS						
		14 PFP	994.2 KPA/CM						
		15 PFD	280.7 KPA/CM						
		16 NBEALE	0.00551						

TABLE VI. - CONCLUDED.

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 09/26/84 10:54:20.52 RDG 1357

FLUID HELIUM		BAROM 14.474 PSI		REGENERATOR 1		DISPLACER 2		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.38 L/MIN	AMPS1	680. AMPS	PRESUP	7163. KPA	TGEXP	559.5 DEG.C	T01HTR	601.7 DEG.C
TWINDP	18.8 DEG.C	AMPS2	605. AMPS	MEANBP	6992. KPA	TGREGH	551.9 DEG.C	T02HTR	614.3 DEG.C
TDLDP	1.99 DEG.C	VOLTG	2.78 VOLTS	MEANCP	7033. KPA	TGREGC	98.6 DEG.C	T03HTR	612.7 DEG.C
TWODPR	21.4 DEG.C					TGCOMP	51.6 DEG.C	T04HTR	608.1 DEG.C
						TGBOUN	35.8 DEG.C	T05HTR	618.4 DEG.C
								T06HTR	611.3 DEG.C
								T07HTR	610.5 DEG.C
								T08HTR	578.3 DEG.C
								T09HTR	610.9 DEG.C
								T10HTR	589.1 DEG.C
								T11HTR	573.5 DEG.C
								T12HTR	582.0 DEG.C
								T13REG*****	DEG.C
								T14REG	514.2 DEG.C
								T15REG	390.1 DEG.C
								T16REG	386.3 DEG.C
								T17REG	378.5 DEG.C
								T18REG	364.0 DEG.C
								T19REG	237.4 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.22 L/MIN	1 PWIRIN	3575. WATTS	VX1HOR	0.7 CM/S	PWROUT	798. WATTS		
TWINCL	25.0 DEG.C	2 QCOOLR	2563. WATTS	VY1VER	8.9 CM/S	INDPWR	818. WATTS		
TDLCL	8.75 DEG.C	3 QDSHPT	605. WATTS	PHASE ANGLES		PISTST	2.61 CM		
TWOCLR	32.75 DEG.C	4 EXTEFF	22.3 %	PADISP	77.7 DEG.	DISPST	1.88 CM		
		5 TAVHTR	600.9 DEG.C	PAPRES	-11.3 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	23.7 %	ENGINE SPEED		PAMPC	1299. KPA		
		8 AMPS	1285. AMPS	FREQ	31.3 HZ	DISPCP	2.65 CM		
		9 QDISPG	3. WATTS			PISTCP	2.36 CM		
		10 QDISP	14. WATTS			PBYNDB	1163. KPA		
		11 QREG1	103. WATTS			PDLCLR	7.81 KPA		
		12 QREG2	122. WATTS			PDLREG	57.36 KPA		
		13 QREG3	35. WATTS			PDLDIS	99.10 KPA		
		14 PFP	1020.3 KPA/CM						
		15 PFD	277.8 KPA/CM						
		16 NBEALE	0.00545						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 11/27/84 10:56:18.08 RDG 1552

FLUID NITROGEN		BAROM 14.357 PSI		REGENERATOR 1		DISPLACER 1		STANDARD PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.00 L/MIN	AMPS1	565. AMPS	PRESUP	7160. KPA	TGEXP	566.9 DEG.C	T01HTR	602.2 DEG.C
TWINDP	19.2 DEG.C	AMPS2	437. AMPS	MEANBP	7010. KPA	TGREGH	552.1 DEG.C	T02HTR	615.1 DEG.C
TDLDP	1.10 DEG.C	VOLTG	2.20 VOLTS	MEANCP	7058. KPA	TGREGC	109.5 DEG.C	T03HTR	604.1 DEG.C
TWODPR	20.4 DEG.C					TGCOMP	40.9 DEG.C	T04HTR	616.8 DEG.C
						TGBOUN	27.3 DEG.C	T05HTR	602.7 DEG.C
								T06HTR	604.2 DEG.C
								T07HTR	603.7 DEG.C
								T08HTR	584.2 DEG.C
								T09HTR	607.4 DEG.C
								T10HTR	596.9 DEG.C
								T11HTR	579.0 DEG.C
								T12HTR	583.9 DEG.C
								T13REG*****	DEG.C
								T14REG	497.4 DEG.C
								T15REG	385.4 DEG.C
								T16REG	391.0 DEG.C
								T17REG	391.1 DEG.C
								T18REG	375.3 DEG.C
								T19REG	236.7 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.35 L/MIN	1 PWIRIN	2210. WATTS	VX1HOR	0.6 CM/S	PWROUT	292. WATTS		
TWINCL	25.2 DEG.C	2 QCOOLR	1948. WATTS	VY1VER	7.5 CM/S	INDPWR	314. WATTS		
TDLCL	6.46 DEG.C	3 QDSHPT	306. WATTS	PHASE ANGLES		PISTST	2.56 CM		
TWOCLR	29.99 DEG.C	4 EXTEFF	13.2 %	PADISP	36.5 DEG.	DISPST	1.56 CM		
		5 TAVHTR	600.0 DEG.C	PAPRES	-5.5 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	13.0 %	ENGINE SPEED		PAMPC	1102. KPA		
		8 AMPS	1002. AMPS	FREQ	28.0 HZ	DISPCP	2.47 CM		
		9 QDISPG	0. WATTS			PISTCP	2.41 CM		
		10 QDISP	15. WATTS			PBYNDB*****	KPA		
		11 QREG1	112. WATTS			PDLCLR	14.93 KPA		
		12 QREG2	99. WATTS			PDLREG	61.09 KPA		
		13 QREG3	34. WATTS			PDLDIS	154.77 KPA		
		14 PFP	967.5 KPA/CM						
		15 PFD	226.5 KPA/CM						
		16 NBEALE	0.00226						

NASA LEWIS SENSITIVITY TEST DATA RE 1000 FREE PISTON STIRLING ENGINE TEST D003, REC 12/07/84 14:42:20.19 RDG 1608

FLUID HELIUM		BAROM 14.322 PSI		REGENERATOR 1		DISPLACER 1		LIGHT PISTON	
HEAT TO DASHPOT COOLING		POWER IN		ENGINE CHARGE PRESSURE		GAS TEMPERATURES		SURFACE TEMPERATURES	
FLODP	4.12 L/MIN	AMPS1	653. AMPS	PRESUP	7070. KPA	TGEXP	558.4 DEG.C	T01HTR	606.1 DEG.C
TWINDP	18.0 DEG.C	AMPS2	517. AMPS	MEANBP	7016. KPA	TGREGH	566.1 DEG.C	T02HTR	606.3 DEG.C
TDLDP	2.44 DEG.C	VOLTG	2.58 VOLTS	MEANCP	7063. KPA	TGREGC	102.3 DEG.C	T03HTR	614.3 DEG.C
TWODPR	20.5 DEG.C					TGCOMP	46.1 DEG.C	T04HTR	606.5 DEG.C
						TGBOUN	32.6 DEG.C	T05HTR	617.5 DEG.C
								T06HTR	611.5 DEG.C
								T07HTR	610.5 DEG.C
								T08HTR	584.0 DEG.C
								T09HTR	615.3 DEG.C
								T10HTR	582.2 DEG.C
								T11HTR	581.7 DEG.C
								T12HTR	573.3 DEG.C
								T13REG*****	DEG.C
								T14REG	522.1 DEG.C
								T15REG	398.0 DEG.C
								T16REG	402.5 DEG.C
								T17REG	402.6 DEG.C
								T18REG	390.7 DEG.C
								T19REG	243.5 DEG.C
								T03HED*****	DEG.C
HEAT TO COOLER		CALCULATED PARAMETERS		VIBRATION		REMOTE CALCULATIONS			
FLOCLR	4.26 L/MIN	1 PWIRIN	3018. WATTS	VX1HOR	0.4 CM/S	PWROUT	572. WATTS		
TWINCL	25.3 DEG.C	2 QCOOLR	2618. WATTS	VY1VER	6.0 CM/S	INDPWR	571. WATTS		
TDLCL	8.86 DEG.C	3 QDSHPT	701. WATTS	PHASE ANGLES		PISTST	2.56 CM		
TWOCLR	31.88 DEG.C	4 EXTEFF	19.0 %	PADISP	41.6 DEG.	DISPST	1.90 CM		
		5 TAVHTR	600.8 DEG.C	PAPRES	-8.1 DEG.	DYNAMIC CALCULATIONS			
		6 INTEFF	17.9 %	ENGINE SPEED		PAMPC	1102. KPA		
		8 AMPS	1170. AMPS	FREQ	37.4 HZ	DISPCP	2.28 CM		
		9 QDISPG	3. WATTS			PISTCP	2.37 CM		
		10 QDISP	15. WATTS			PBYNDB*****	KPA		
		11 QREG1	119. WATTS			PDLCLR	4.41 KPA		
		12 QREG2	107. WATTS			PDLREG	40.05 KPA		
		13 QREG3	36. WATTS			PDLDIS	75.69 KPA		
		14 PFP	987.2 KPA/CM						
		15 PFD	245.5 KPA/CM						
		16 NBEALE	0.00331						

TABLE VII. - LIST OF SINGLE VARIABLE DATA POINTS

[Regenerator and displacer 1 were designed for high efficiency. Regenerator and displacer 2 were designed for high power.]

Nominal Values								
Reading Number	Gas	Heater, °C	Cooler, °C	Pressure, MPa	Stroke, cm	Disp.	Regen.	Pist.
1006-1012	He	600	25	7.0	Var 1.8, 2.0, ...	1	1	Std
1010, 1070, 1079	He	600	25	Var 7.0, 5.5, 4.0	2.6	1	1	Std
1010, 1017, 1024, 1030	He	Var 600, 550 500, 450	25	7.0	2.6	1	1	Std
1010, 1121, 1200	He	600	Var 25, 40, 55	7.0	2.6	1	1	Std
1010, 1357	He	600	25	7.0	2.6	Var 1, 2	1	Std
1010, 602	He	600	25	7.0	2.6	1	Var 1, 2	Std
1010, 1608	He	600	25	7.0	2.6	1	1	Var Std and Light
1010, 1552	Var He, N ₂	600	25	7.0	2.6	1	1	Std

TABLE VIII. - LIST OF DOUBLE VARIABLE DATA POINTS

SVDS reading numbers	DVDS reading numbers (stroke varied starting at 1.8 cm)	
602	598-602	(1.8-2.6 cm)
1006	1006-1012	(1.8-3.0 cm)
1007		
1008		
1009		
1010		
1011		
1012		
1017	1013-1019	
1024	1020-1025	(1.8-2.8 cm)
1030	1026-1032	(1.8-3.0 cm)
1070	1066-1072	
1079	1075-1081	
1121	1117-1122	(1.8-2.8 cm)
1200	1196-1202	(1.8-3.0 cm)
1357	1353-1360	(1.8-3.2 cm)
1552	1548-1552	(1.8-2.6 cm)
1608	1604-1610	(1.8-3.0 cm)

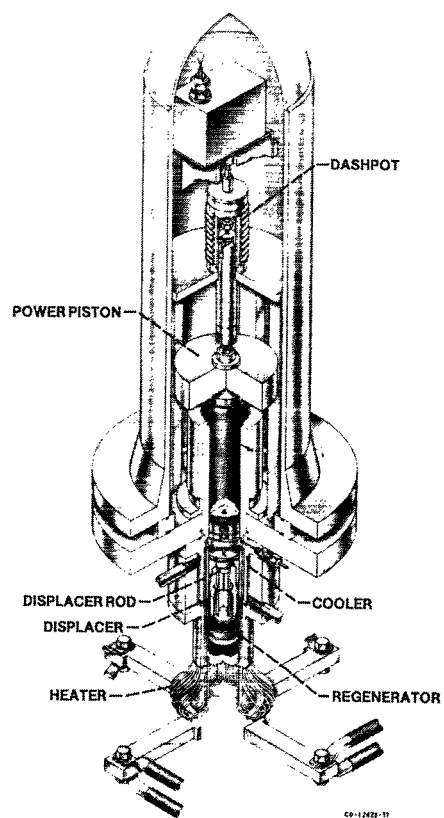


FIGURE 1. - COMPONENT LAYOUT OF RE-1000 FREE PISTON STIRLING ENGINE.

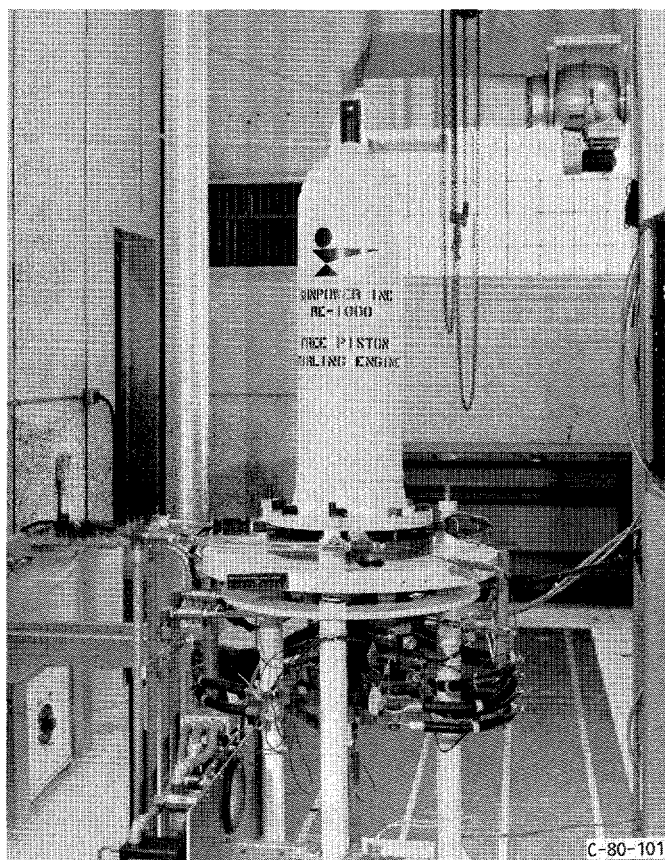


FIGURE 2. - RE-1000 FREE-PISTON STIRLING ENGINE IN TEST CELL.

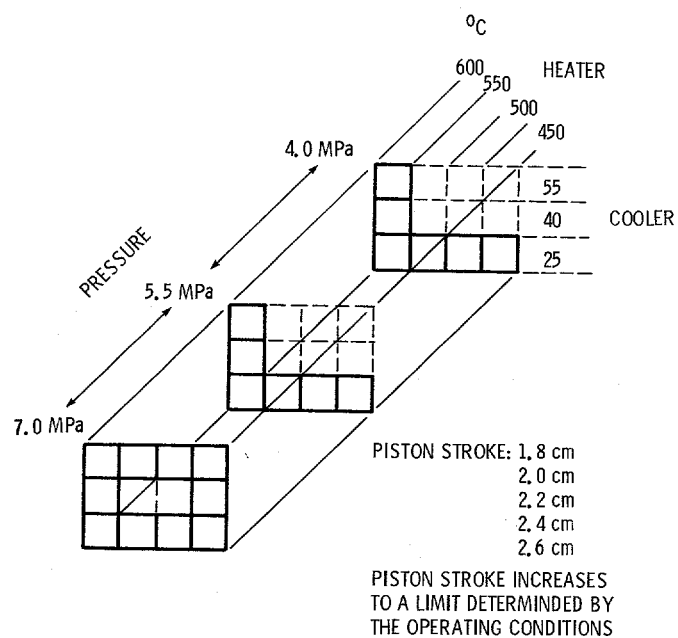


FIGURE 3.- RE-1000 TEST MATRIX.

		DISPLACER		
		2	1	
STANDARD PISTON		189	108	REGENERATOR 2
		174	150	REGENERATOR 1
NITROGEN			64	
ARGON			12	
LIGHT POWER PISTON			84	

FIGURE 4.- CONFIGURATION MATRIX WITH NUMBER OF READINGS.

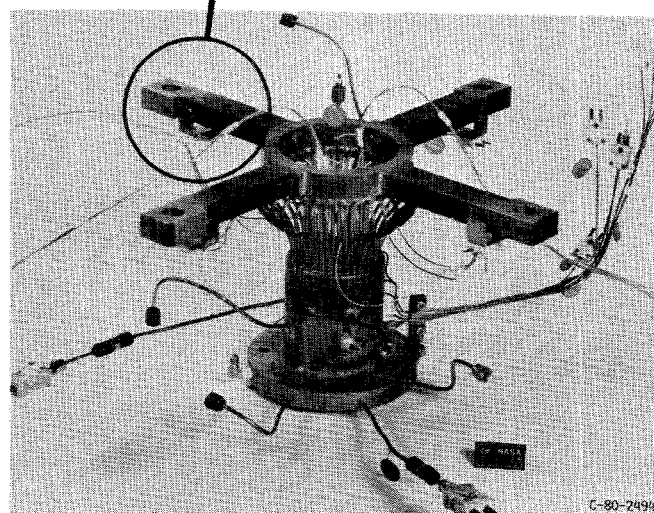
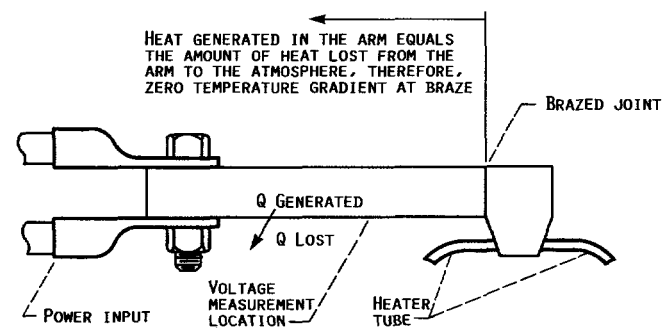


FIGURE 5.- RE-1000 HEATER HEAD.

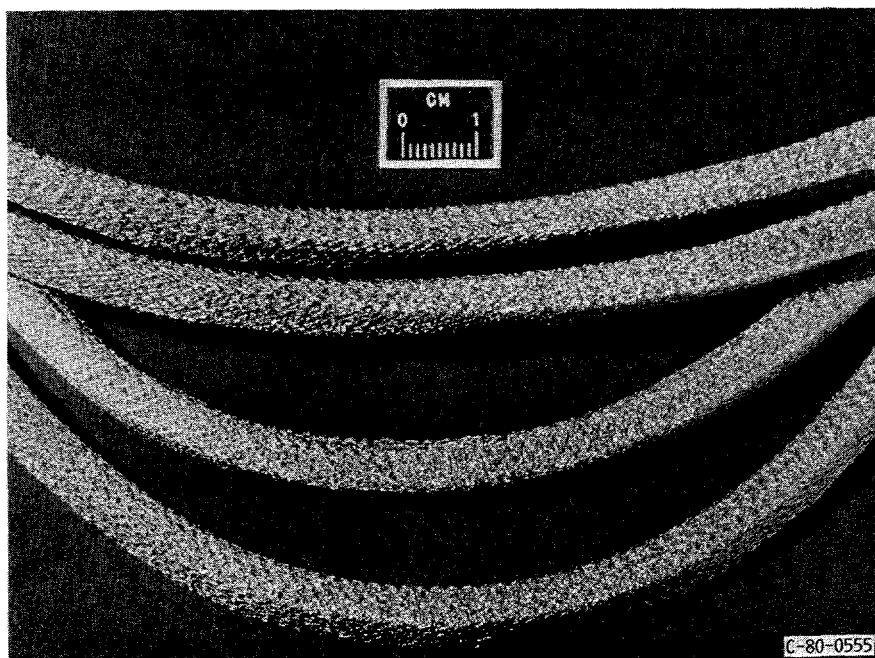


FIGURE 6. - METEX KNITTED REGENERATOR MATRIX.

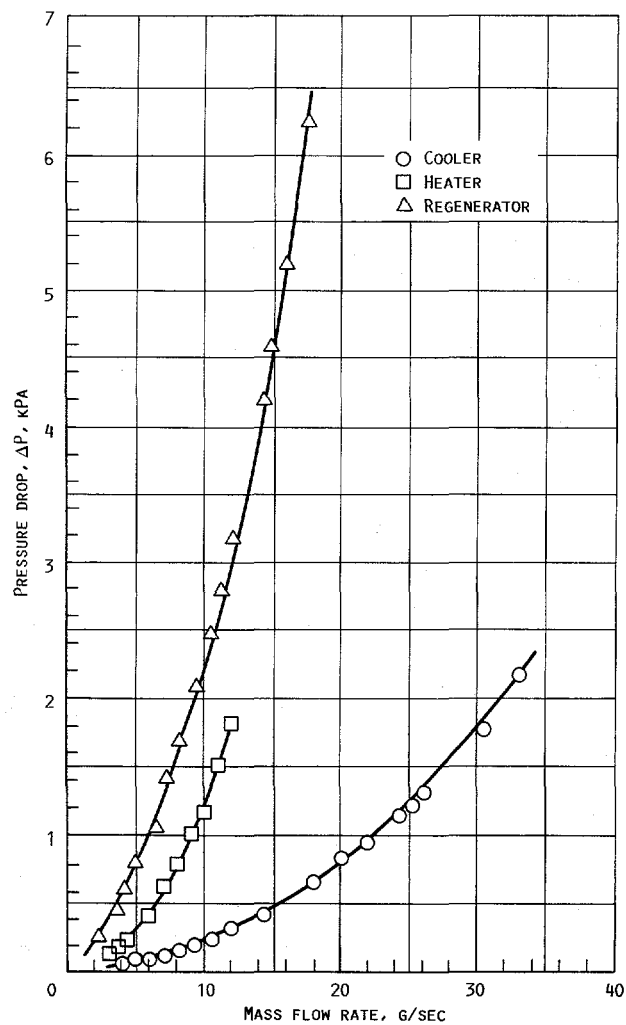


FIGURE 7.- PRESSURE DROP AS A FUNCTION OF MASS FLOW RATE WITH REGENERATOR 1. TESTED WITH NITROGEN AT 2070 KPA FOR THE COOLER AND REGENERATOR, AND 1380 KPA FOR THE HEATER. NITROGEN TEMPERATURE WAS 15 °C.

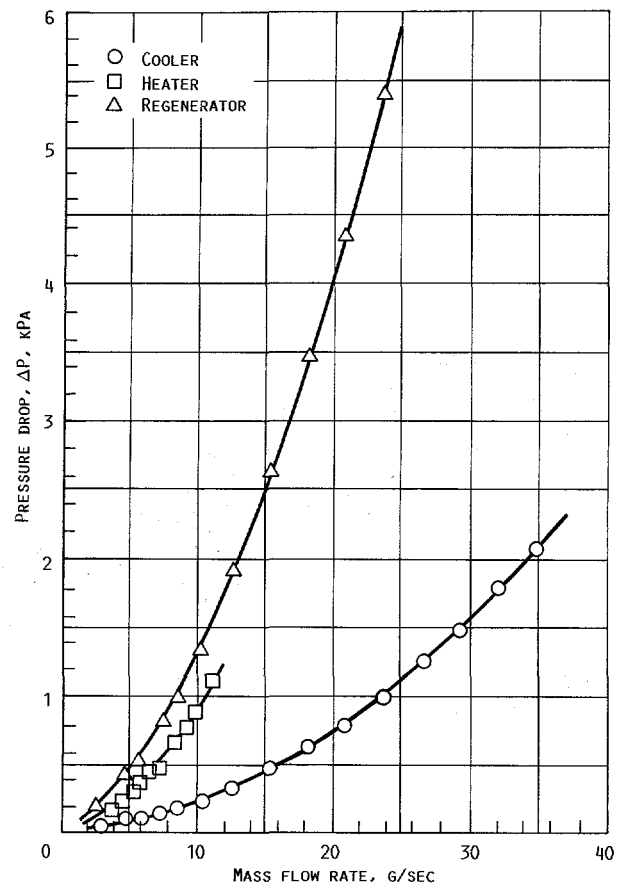
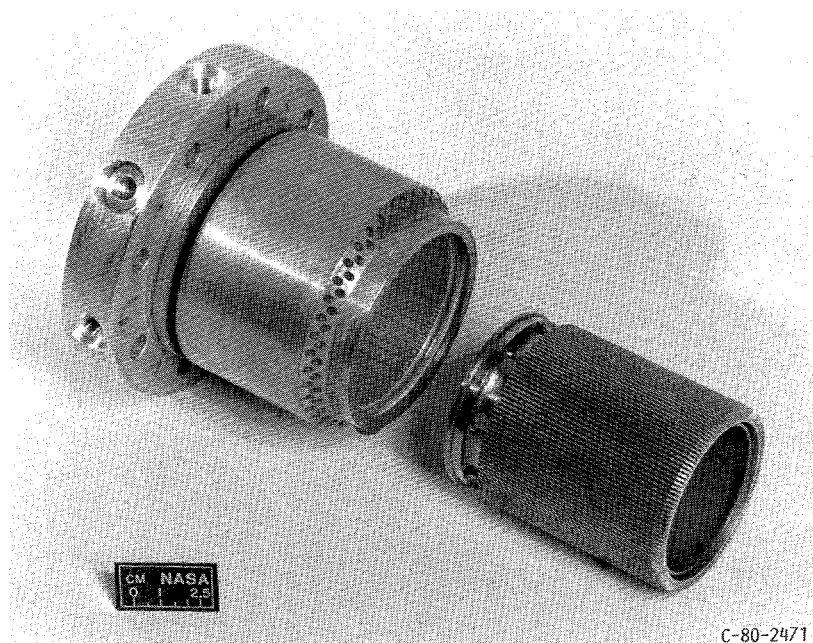


FIGURE 8.- PRESSURE DROP AS A FUNCTION OF MASS FLOW RATE WITH REGENERATOR 2. TESTED WITH NITROGEN AT 2070 kPa FOR THE COOLER AND REGENERATOR, AND 1380 kPa FOR THE HEATER. NITROGEN TEMPERATURE WAS 10 °C.



C-80-2471

FIGURE 9. - RE-1000 COOLER AND HOUSING.

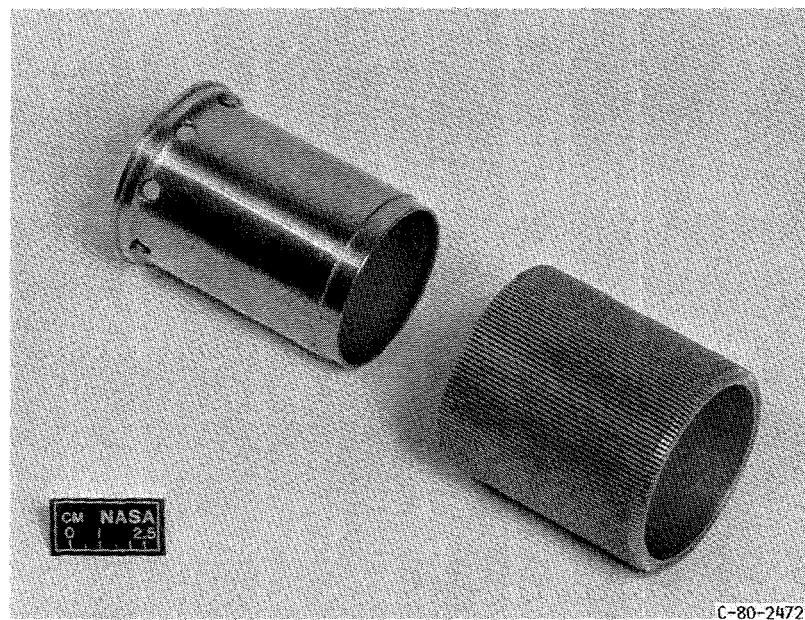


FIGURE 10. - DISPLACER CYLINDER AND COOLER GAS PASSAGES.

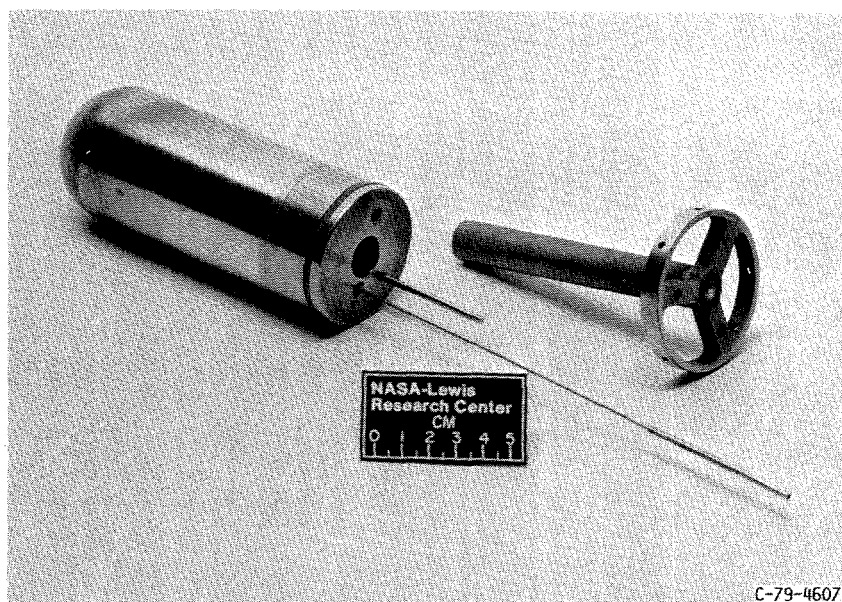


FIGURE 11. - DISPLACER AND DISPLACER ROD 1.

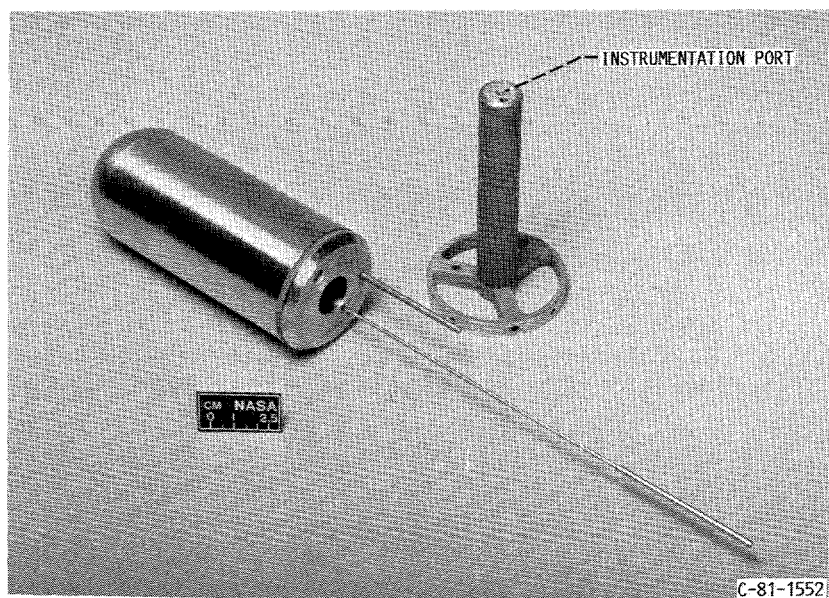


FIGURE 12. - DISPLACER AND DISPLACER ROD 2.

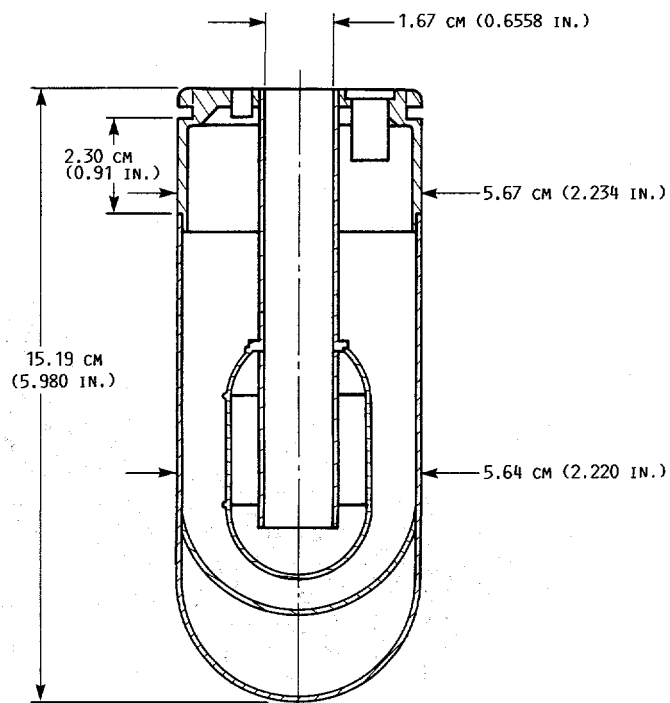


FIGURE 13.- DISPLACER 1 CROSS-SECTION. DISPLACER WEIGHT, 426 G (0.94 LB); GAS SPRING MEAN VOLUME, 31.79 cm^3 (1.94 in.^3).

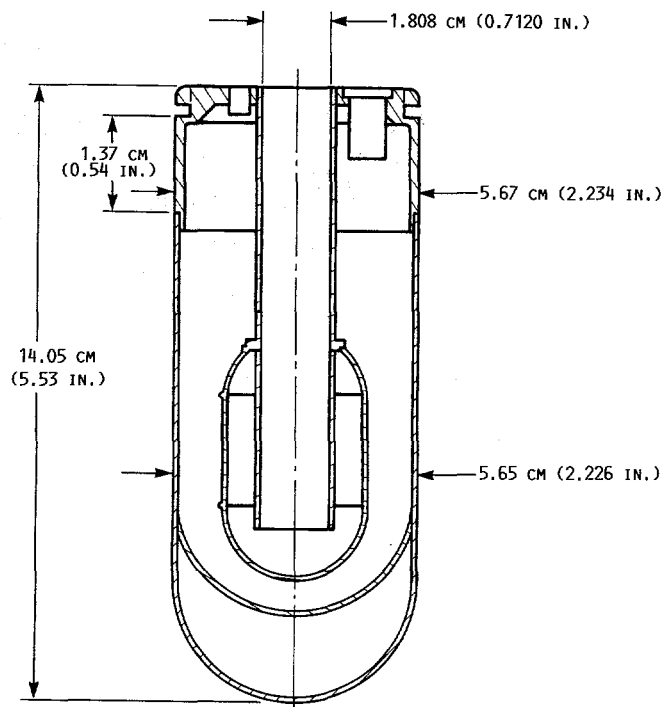


FIGURE 14.- DISPLACER 2 CROSS-SECTION. DISPLACER WEIGHT 381.36 (0.84 LB), GAS SPRING MEAN VOLUME, 18.80 CM³ (1.147 IN.³).

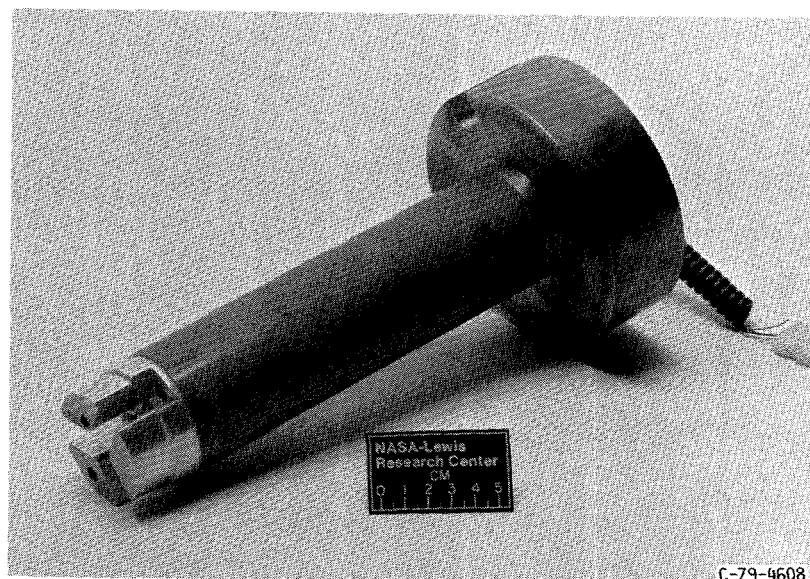
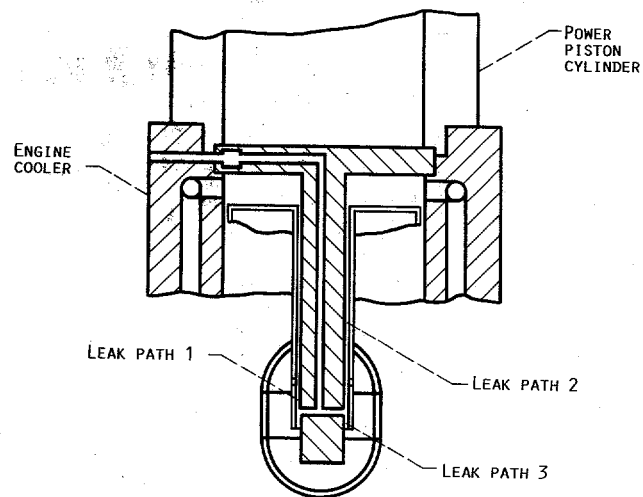
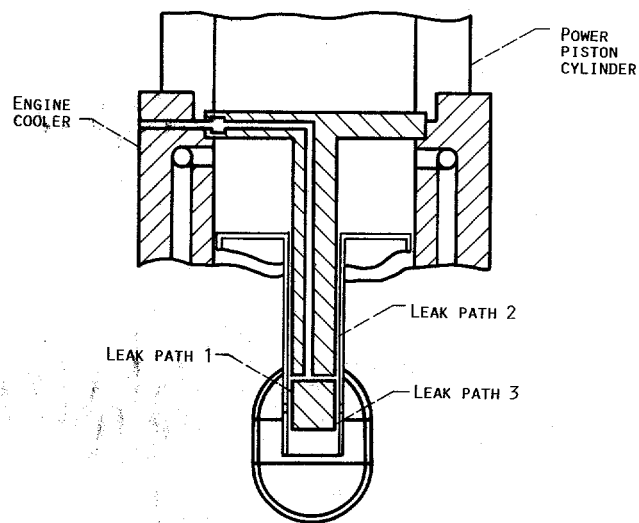


FIGURE 15. - RE-1000 FREE-PISTON STIRLING ENGINE POWER PISTON.



(A) WHEN DISPLACER IS AT COMPRESSION SPACE.



(B) WHEN DISPLACER IS AT EXPANSION SPACE.

FIGURE 16. - DISPLACER ROD MOUNTING WITH CENTER PORTS AND LEAK PATHS SHOW.

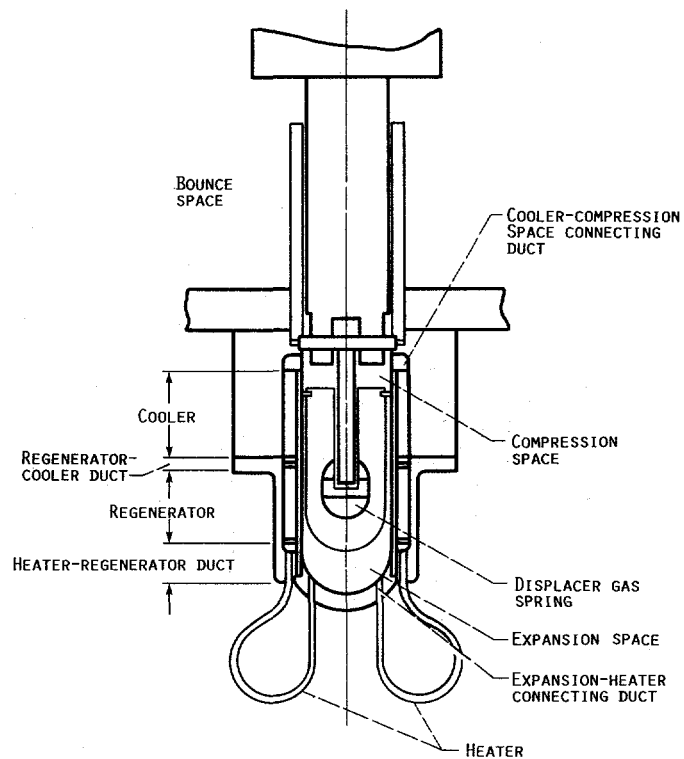


FIGURE 17.- DEFINITIONS OF VOLUMES IN THE ENGINE.

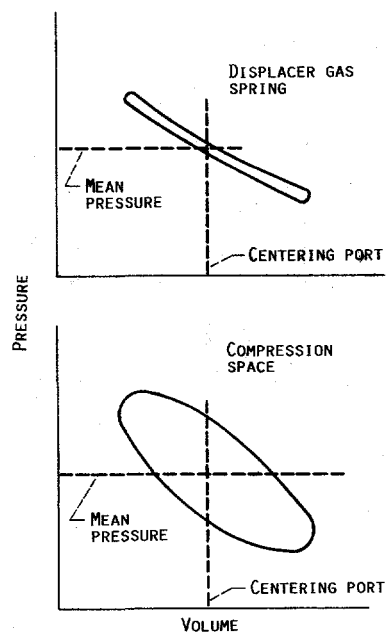


FIGURE 18.- RE-1000 ENGINE CENTERING PORT PRESSURE DROP FOR DISPLACER GAS SPRING AND COMPRESSION SPACE.

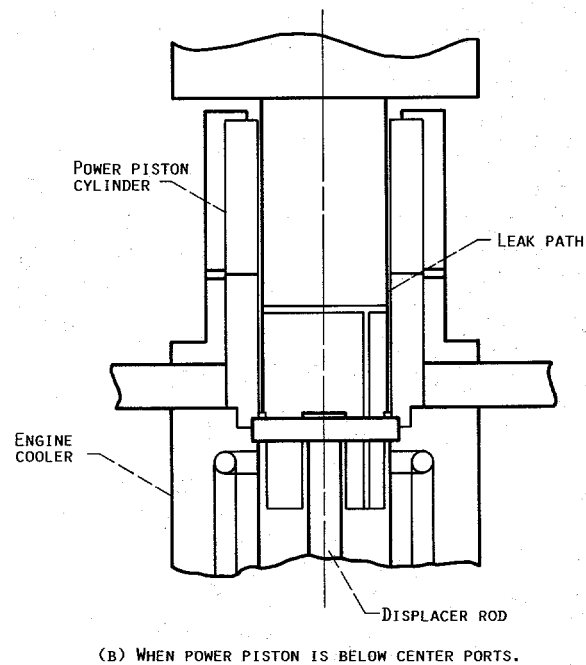
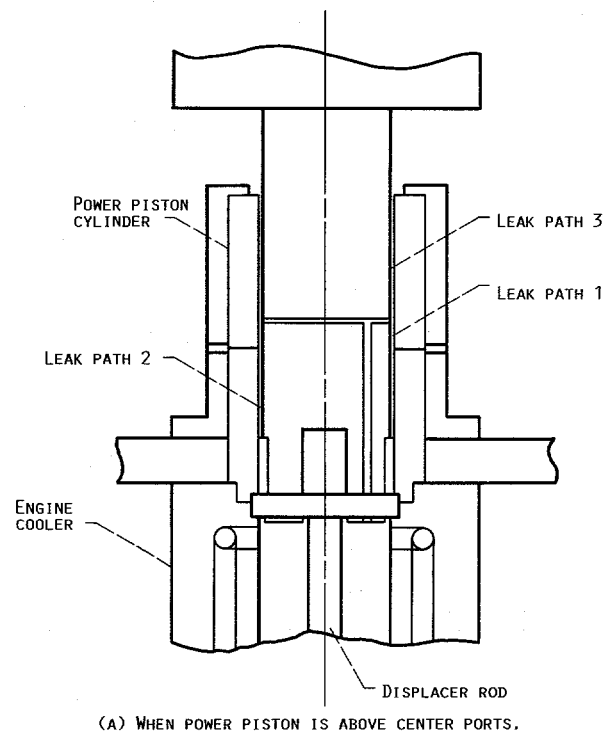
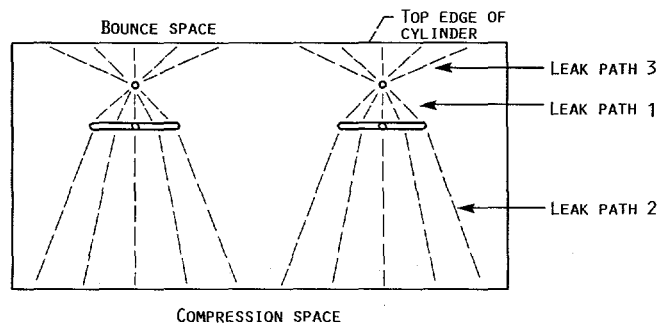
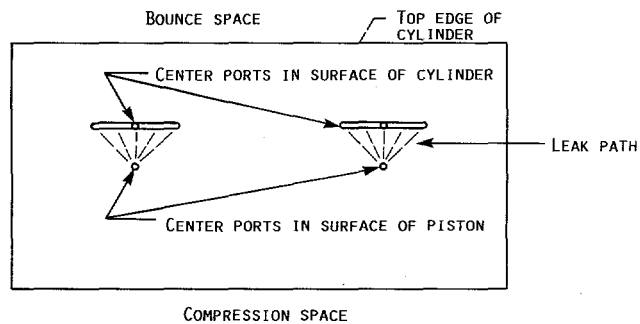


FIGURE 19.- SCHEMATIC OF LEAK PATHS PAST POWER PISTON.



(A) WHEN POWER PISTON IS ABOVE CENTER PORTS.



(B) WHEN POWER PISTON IS BELOW CENTER PORTS.

FIGURE 20. - GEOMETRY OF LEAK PATHS PAST POWER PISTON. THE CENTER PORT IN THE POWER PISTON CYLINDER IS A 1.1-MM-DIAMETER HOLE WITHIN A GROOVE 30 MM LONG, 1.25 MM WIDE, AND 0.75 MM DEEP.

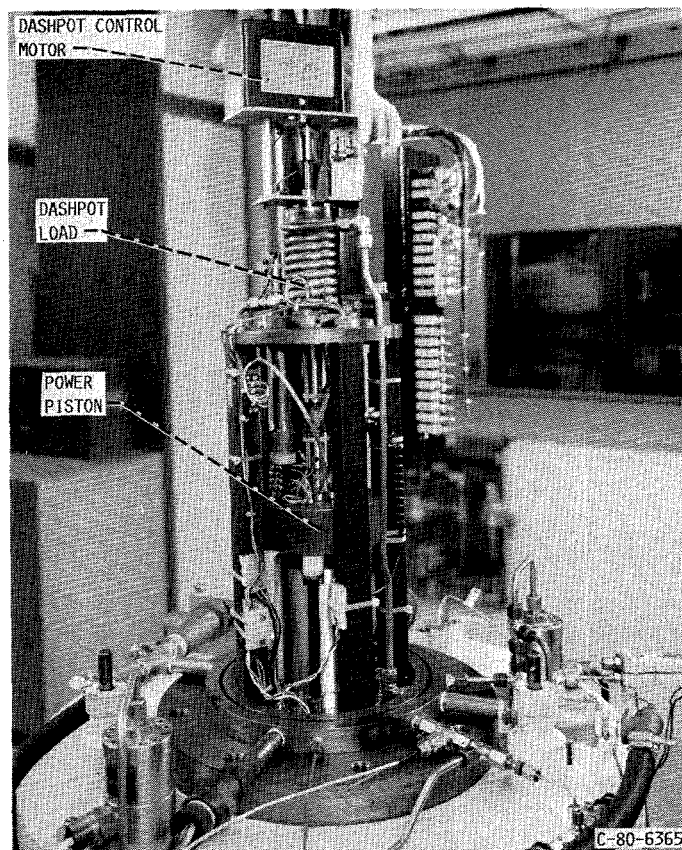


FIGURE 21. - RE-1000 IN TEST CELL WITH PRESSURE VESSEL REMOVED.

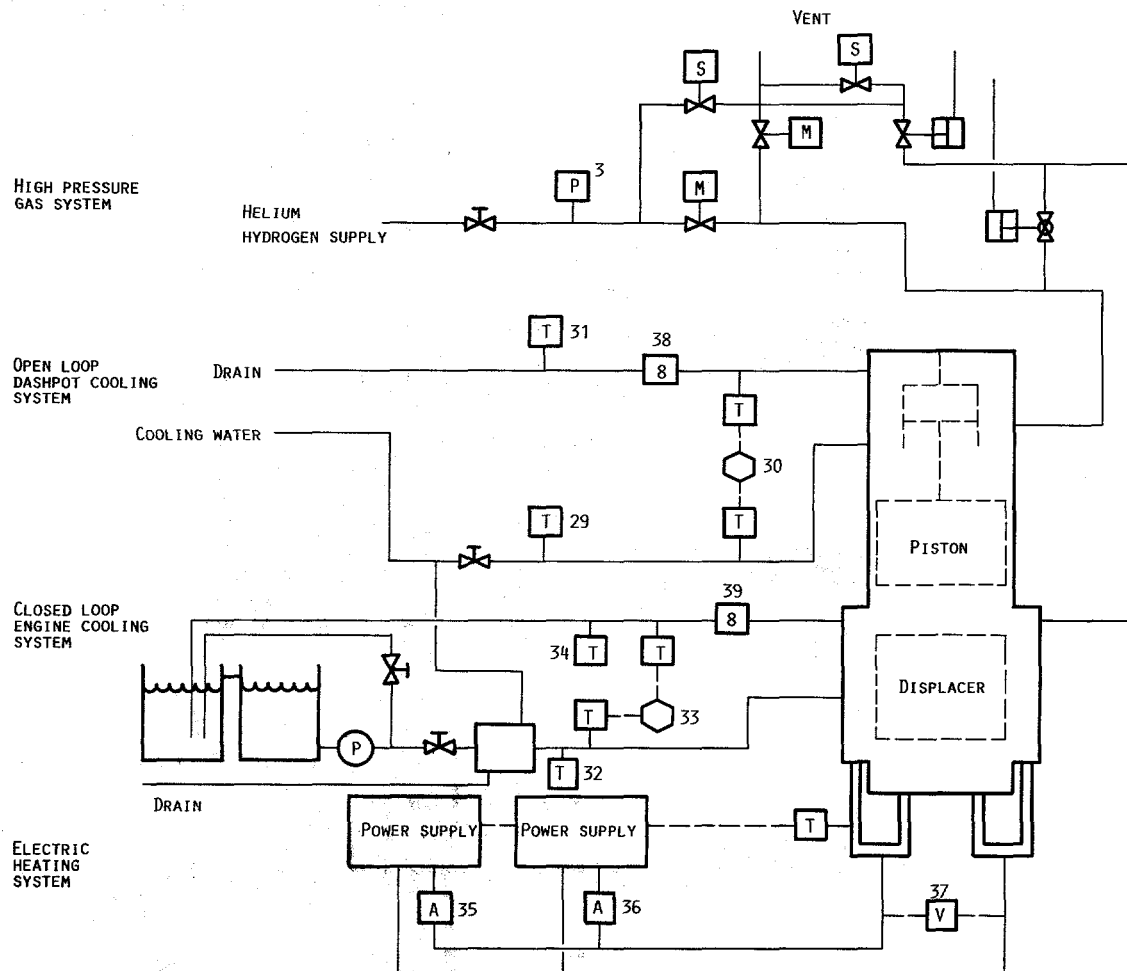


FIGURE 22.- RE-1000 TEST SCHEMATIC.

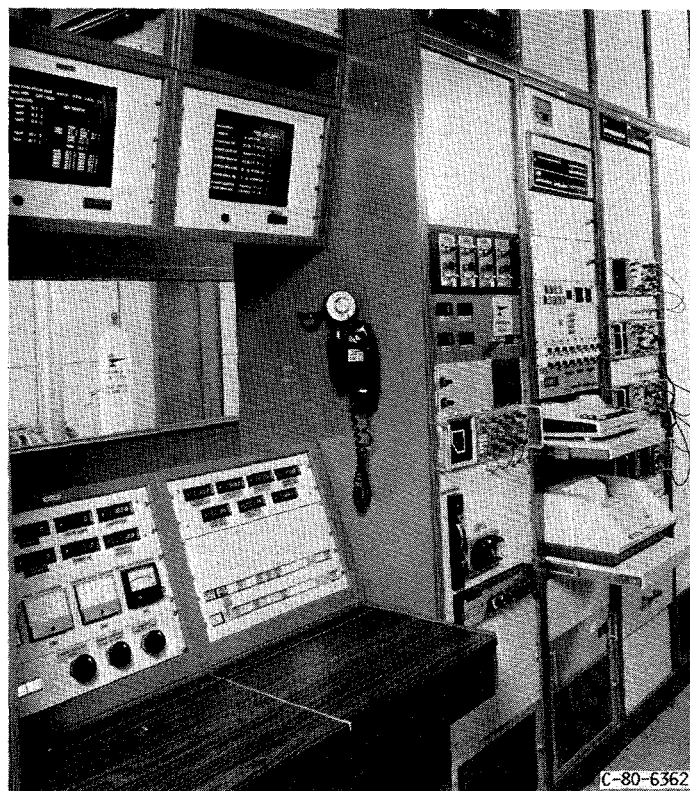


FIGURE 23. - CONTROL ROOM.

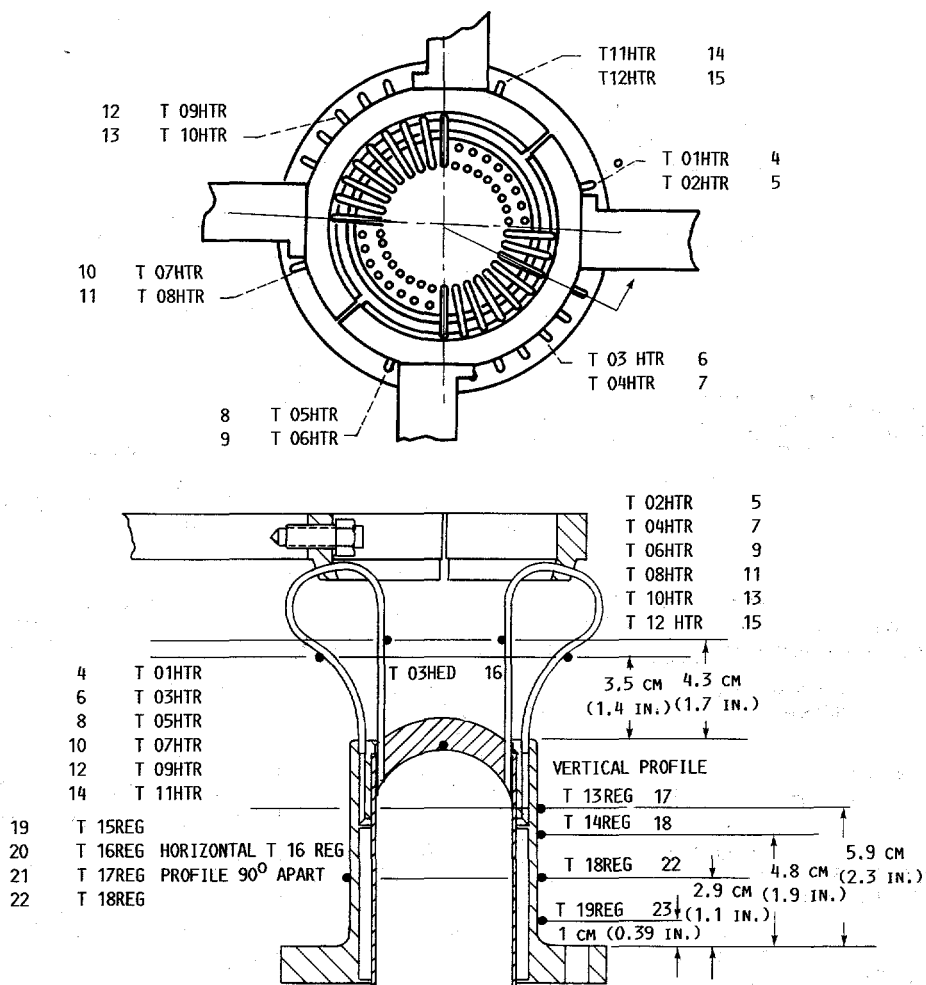


FIGURE 24.- RE-1000 HEATER HEAD THERMOCOUPLE LOCATIONS.

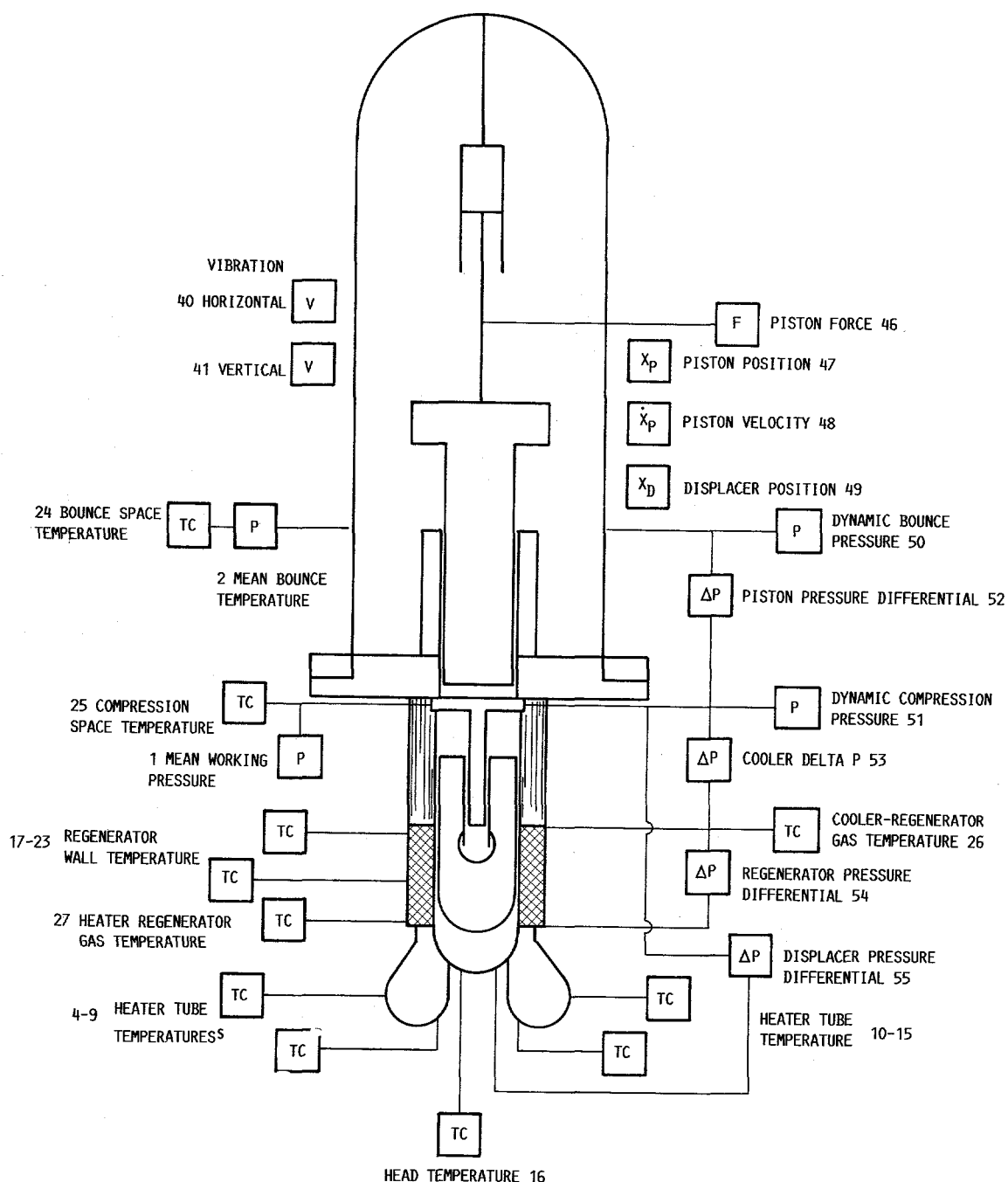


FIGURE 25.- INSTRUMENTATION LAYOUT OF RE-1000.

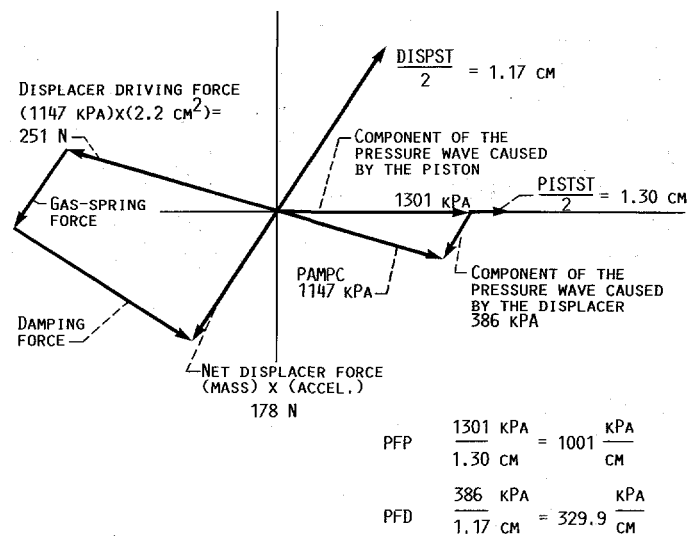


FIGURE 26.- PHASOR DIAGRAM OF THE RE-1000 AT DATA POINT 1010, DESIGN CONDITION.

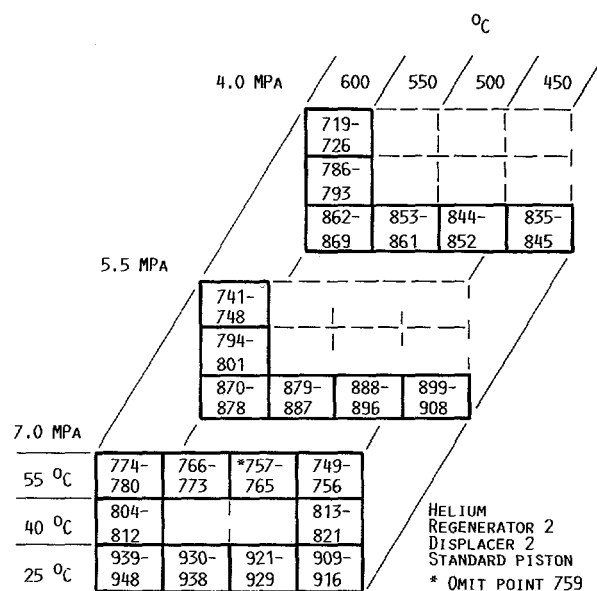
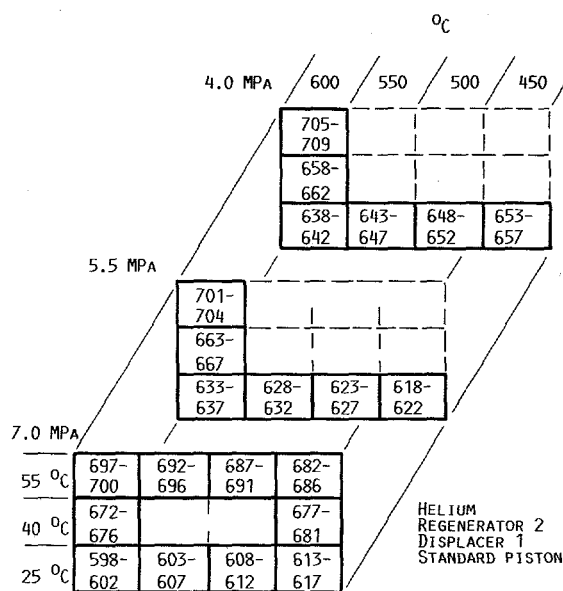
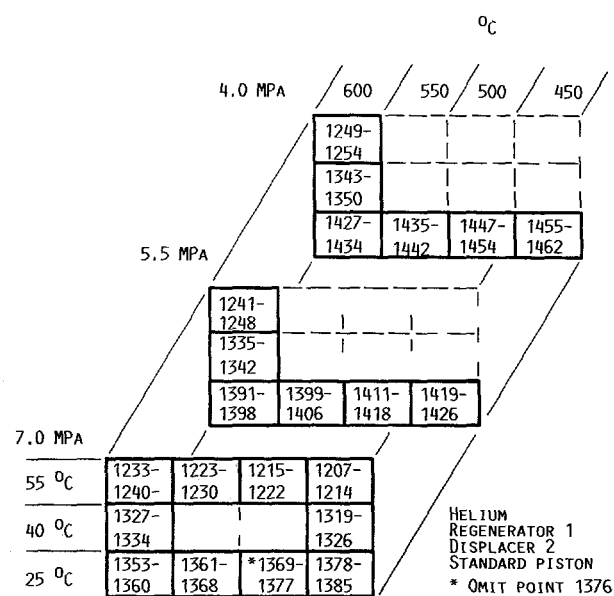
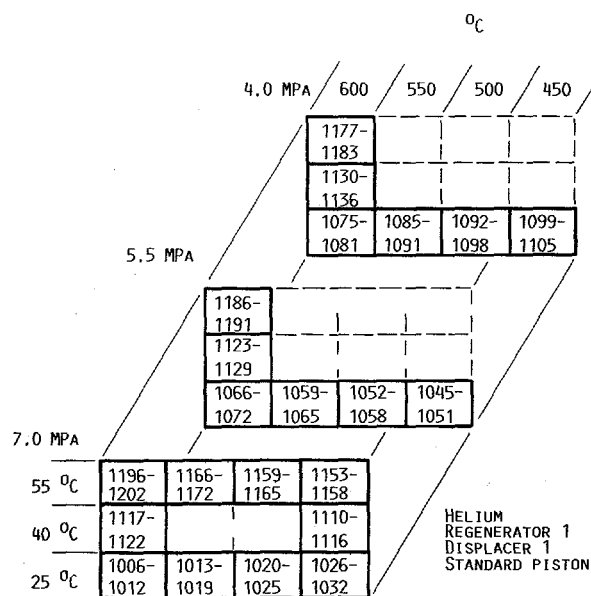


FIGURE 27.- TEST MATRIX WITH READING NUMBERS.

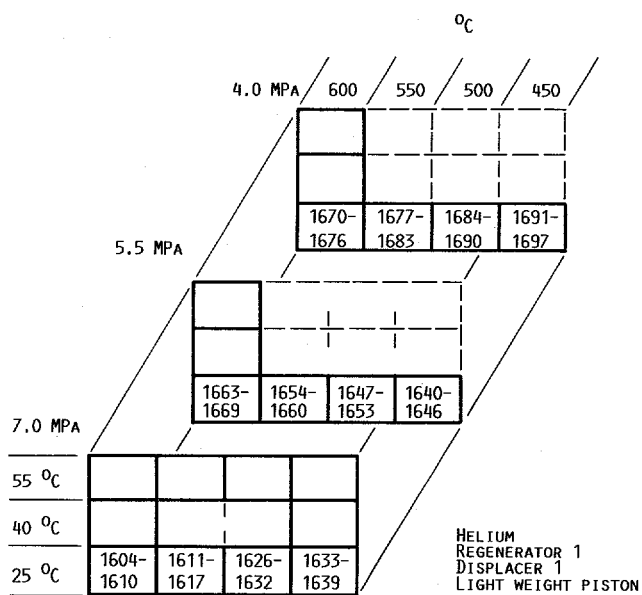
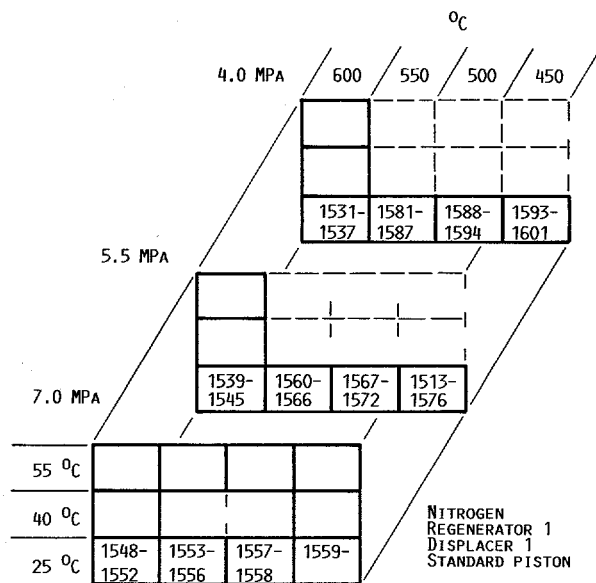
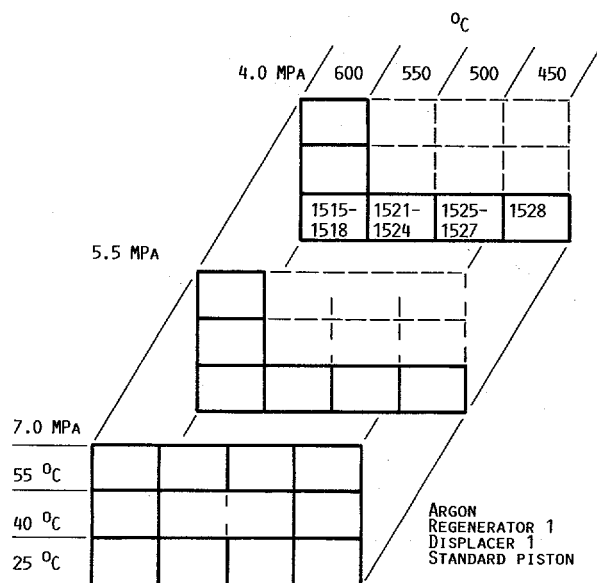


FIGURE 27.- CONCLUDED.

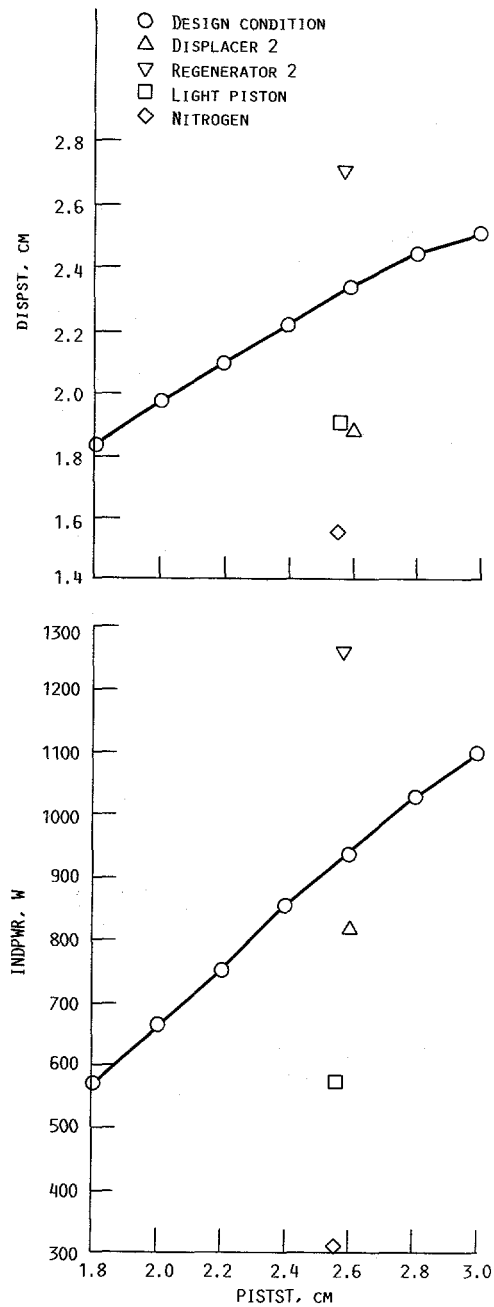


FIGURE 28.- DATA PLOTTED FROM THE SINGLE VARIABLE DATA SET (SVDS).

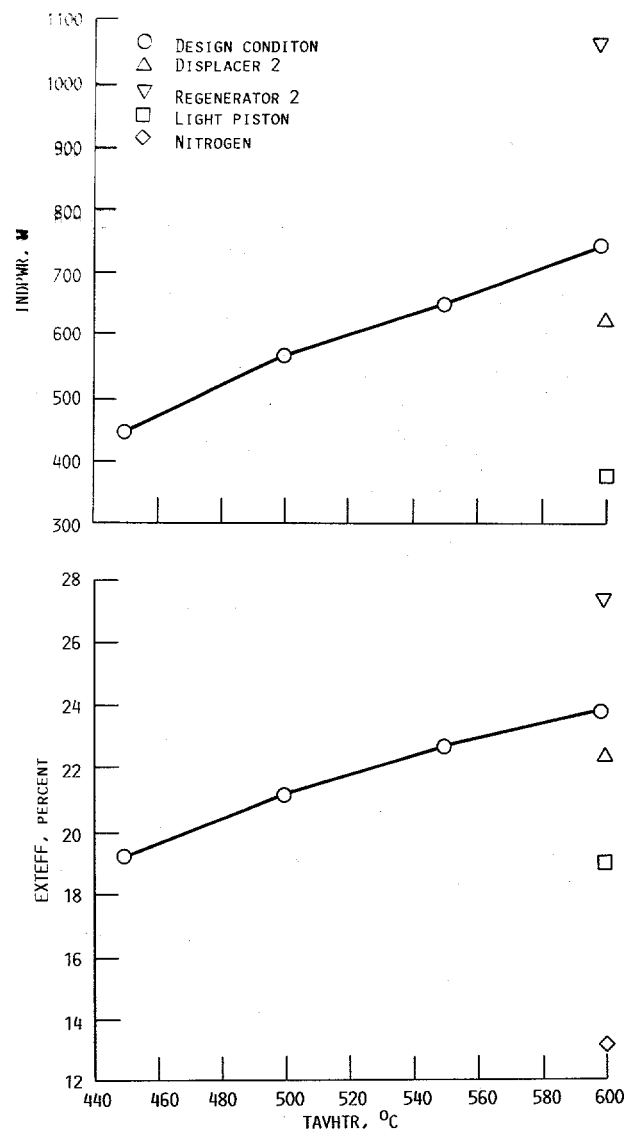


FIGURE 28.- CONCLUDED.

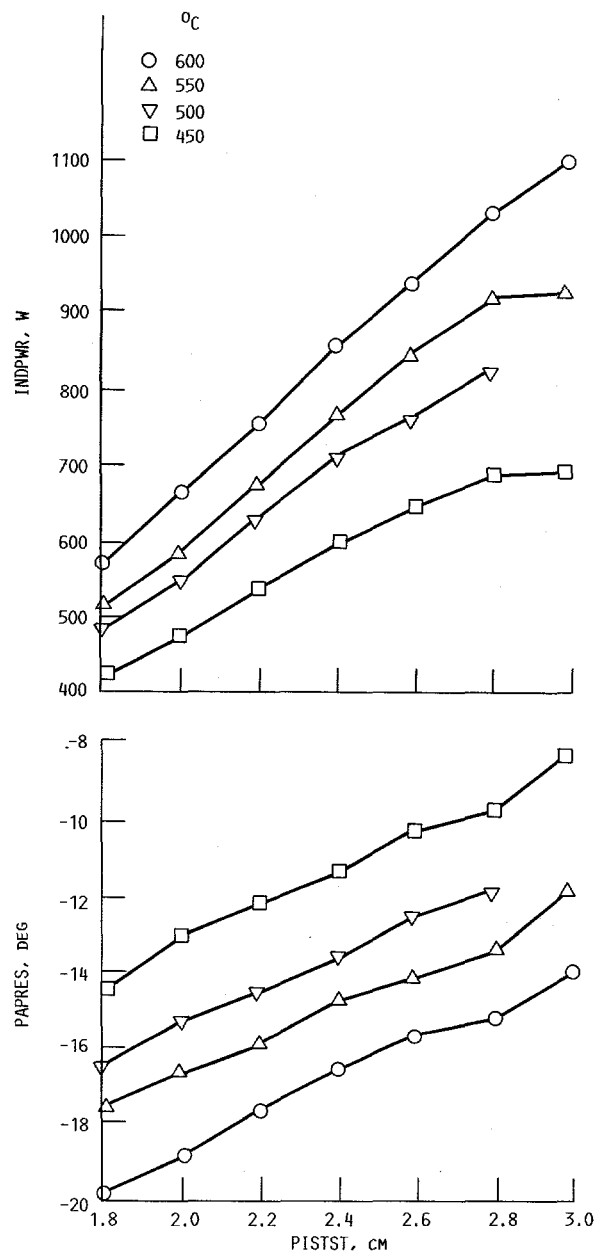


FIGURE 29.- DATA PLOTTED FROM THE DOUBLE VARIABLE DATA SET (DVDS).

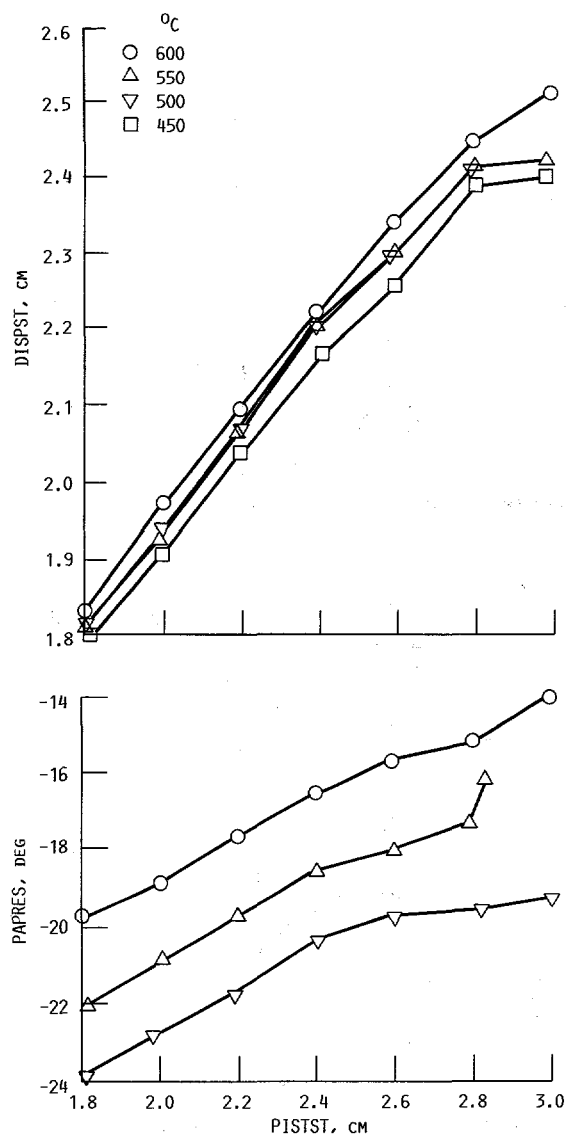


FIGURE 29.- CONCLUDED.

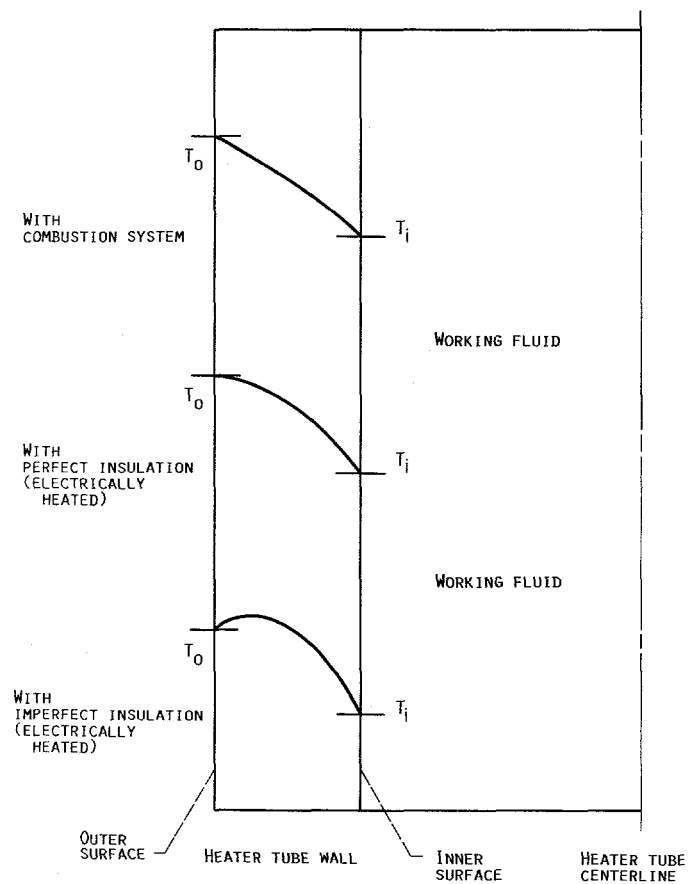


FIGURE 30.- HEATER TUBE TEMPERATURE PROFILES.

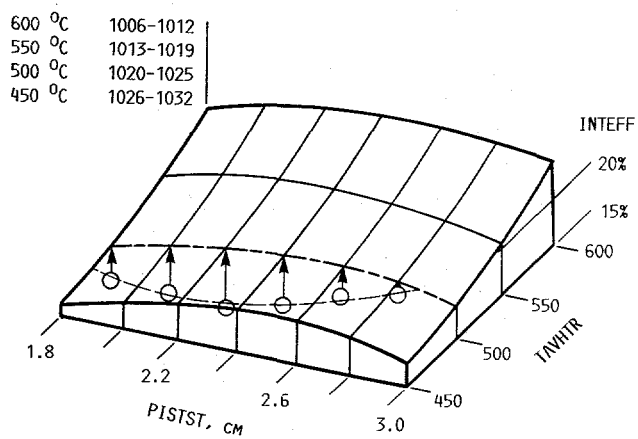
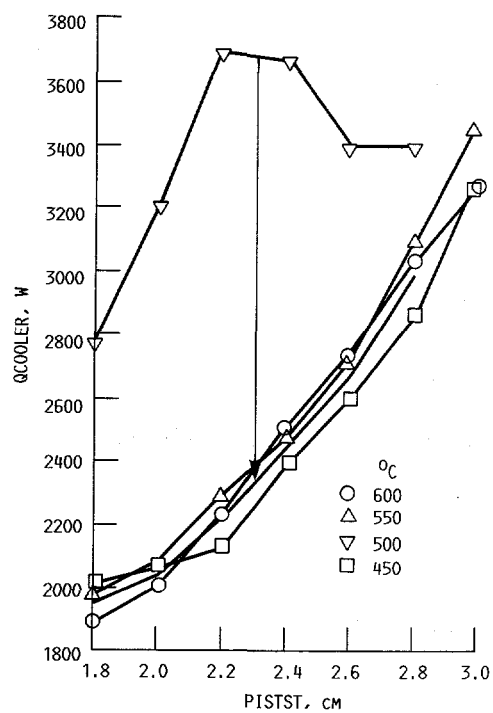


FIGURE 31.- CORRECTION OF FAULTY QCOOLER READING FOR DATA POINTS 1020-1025.

1. Report No. NASA TM-88846 Corrected Copy		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle RE-1000 Free-Piston Stirling Engine Sensitivity Test Results				5. Report Date	
				6. Performing Organization Code 506-55-6E	
7. Author(s) Jeffrey G. Schreiber, Steven M. Geng, and Gary V. Lorenz				8. Performing Organization Report No. E-3277	
				10. Work Unit No.	
9. Performing Organization Name and Address National Aeronautics and Space Administration Lewis Research Center Cleveland, Ohio 44135				11. Contract or Grant No.	
				13. Type of Report and Period Covered Technical Memorandum	
12. Sponsoring Agency Name and Address U.S. Department of Energy Division of Building and Community Systems Washington, DC 20545				14. Sponsoring Agency Code Report No. DOE/NASA/1005-11	
15. Supplementary Notes Final Report. Prepared under Interagency Agreement DE-AI05-820R1005.					
16. Abstract The NASA Lewis Research Center has been testing a 1 kW (1.33 hp) free-piston Stirling engine at the NASA Lewis test facilities. The tests performed over the past several years have been on a single cylinder machine known as the RE-1000. The data recorded were to aid in the investigation of the dynamics and thermodynamics of the free-piston Stirling engine. The data are intended to be used primarily for computer code validation. NASA reports TM-82999, TM-83407, and TM-87126 give initial results of the engine tests. The tests were designed to investigate the sensitivity of the engine performance to variations on the mean pressure of the working space, the working fluid used, heater and cooler temperatures, regenerator porosity, power piston mass and displacer dynamics. These tests have now been completed at NASA Lewis. This report presents some of the detailed data collected in the sensitivity tests. In all, 781 data points were recorded. A complete description of the engine and test facility is given. Many of the data can be found in tabular form, while a microfiche containing all of the data points can be requested from the NASA Lewis.					
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