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TEMPERATURE DISTRIBUTION IN SPHERICAL PELLETS

by

G. M. Anderson

May 1953

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Abstract

A study is presented of the surface temperature distribution of spherical pellets in a fixed-bed, pellet-type reactor. The temperature near a point of contact is determined experimentally by means of an electrolytic tank analogue. An analytical approach involving the use of the IBM (CPC) calculator is described. Suggestions for further study are included.

Temperature Distribution in Spherical Pellets

Introduction:

This report summarizes the progress on the study of the temperature distribution in spherical pellets of a pellet type reactor. Of primary importance in this investigation is the determination of the magnitude of the peak surface temperature. Hot spots occur on the surface at the points where the pellets are in contact for in this region the coolant velocity approaches zero with a resultant reduction of heat transfer.

Two general approaches are being pursued in this program. One involves an analogue technique which utilizes an electrolytic tank current flow field to simulate the heat flow field. The other makes use of an analytical treatment with numerical results achieved by digital techniques on an IBM card programmed calculator (CPC).

The results of these two approaches are complementary in that the problems solved are somewhat different. At first it was planned to make exclusive use of the electrolytic tank because of the obvious complexity and the immense labor necessary to achieve numerical results from analytical methods. However, while awaiting the construction of the tank considerable progress was made in developing an analytical method. Of paramount importance in furthering this phase of the work was the suggestion that the CPC calculator at the Westinghouse Anacom Laboratory be employed for the numerical calculations.

The material presented is divided as follows:

- I. Mathematical Formulation of the Problem
- II. Transformation of the Problem
- III. Electrolytic Tank Analogue
- IV. Analytical Solution
- V. Conclusions and Further Work

Appendix A - Fundamental Basis of the Tank Analogue including
tank design information.

Appendix B - Mathematical Analysis of the Problem and its
solution in the form of an infinite series.

I. Mathematical Formulation of the Problem

The problem under consideration is the determination of the temperature distribution on the surface and in the interior of a spherical pellet in a fixed bed of identical pellets. The pellets have a core of fissionable material covered with a suitable cladding for the purpose of minimizing corrosion. Coolant flowing through the bed removes the heat generated by fissions in the core of the pellets.

Figure 1 shows a cross section of one possible arrangement. In this instance the matrix of spherical pellets is cubical and results in six points of contact on each sphere. Other packing arrangements may result in as many as twelve contacts.

The effects attributable to coolant variables such as velocity, viscosity and pressure are taken into account by a concept known as the film coefficient. The film coefficient relates the heat transfer between the surface and the coolant to the existing temperature differential. The effect of the coolant variables on the film coefficient is determined from experiment.

The data on the heat transfer coefficient for spherical pellets in a (1) bed is rather meager. The only available information is that given by W. H. Denton who treats the case of eight points of contact. The data shows that for a Reynolds number of 50,000, the case of interest here, the heat transfer coefficient rises rapidly from a zero value at the point of contact and attains its average value at an angle roughly five degrees removed from the contact point. Outside this region the coefficient is generally greater than the average value and at places attains a value of as much as one and a half times the average.

The impact of Denton's data on the course of this program has been considerable. The narrowness of the "dead-zone" around the point of contact implies that, as a first approximation, each point may be treated separately. The temperature decreases rapidly from a peak value at the point of contact to a value which varies slowly with the slowly varying film coefficient over the non-contact regions. The large ratio of active heat transfer area to dead-zone area ensures that the points of contact do not affect each other.

A further simplification of the problem is achieved by neglecting the variation of the film coefficient with the longitudinal angle ϕ , of Figure 2. This assumption reduces the problem to a two dimensional one with temperatures and fluxes dependent only on the r and θ variables. Actually (1) there is a variation of film coefficient with ϕ , as Denton's data shows, but it is not as important as the θ variation. The first phases of this study, in fact all of the work reported here, neglect the influence of the ϕ variable.

The final step in the reduction of the problem consists of approximating (1) the data by a step function variation of film coefficient which has zero value in a region $0 \leq \theta \leq \theta_0$ and a constant value in a region $\theta_0 \leq \theta \leq \theta_{max}$. The maximum angle, θ_{max} , should be sufficiently large so that the temperature at this point is not greatly influenced by the dead-zone angle θ_0 . In this study θ_{max} is 90° .

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For non-corroded spheres the dead-zone angle is about five degrees, as explained earlier. The effects of some corrosion in the hot-spot region may be approximately evaluated by varying θ_0 . It is presently contemplated to let θ_0 vary from five to thirty degrees. The results obtained for the larger dead-zone angles must be viewed with considerably more skepticism than the small dead-zone angles since in the former instances the interaction effects between points of contact may invalidate the assumptions employed.

The final formulation of the problem is illustrated diagrammatically in Figure 3. Heat is generated uniformly over the core region, $r \leq a$. It flows to the water from the sphere over the surface, $r = b$, which is characterized by the film coefficient variation of Figure 3B.

II. Transformation of the Problem

The problem formulated in Part I may be transformed using the principle of superposition. As this transformation is of value in the analogue approach as well as the mathematical analysis, it is explained here to unify the treatment as far as possible.

The basis of the transformation consists of the following observation. The zero film coefficient over the dead-zone, $0 < \theta < \theta_0$, implies that no thermal flux can leave the sphere in this region. This boundary condition may be achieved by adding two solutions which have equal magnitude but oppositely directed fluxes at the boundary in this region.

Consider the two cases illustrated in Figure 4. In Figure 4A the film coefficient is independent of θ . Under this condition all of the thermal flux is radial, uniform and has a surface value q_0 . The negative of the surface flux density, $-q_0$, is used in the second case, illustrated in Figure 4C, to obtain zero flux density in the dead-zone when the two solutions are added. In the latter case the heat generated in the core is taken to be zero and it is this condition which makes the transformation advantageous. In the electrolytic tank analogue it is much more convenient to apply the forcing function at the boundary than uniformly over the volume. The tank and the analytical methods are set up to solve only the second part of the problem, as the radial flow case of Figure 4A and 4B is readily calculated. The complete solution is obtained by adding the two solutions.

The justification of this transformation consists of observing that the differential equations involved are linear and that the boundary conditions are properly fulfilled.

In Figure 4D the current density for $\theta > \theta_0$ is shown dashed since it is not known in advance, but is obtained from the solution of the problem.

III. Electrolytic Tank Analogue

The fundamental basis of the current flow analogue of the heat flow field is explained in detail in Appendix A. The analogy stems from the equivalence of the differential equations governing the two phenomena.

There are two types of current flow analogues in common use. One of these makes use of a continuous conducting medium in which the flow of current and the resulting potentials are described by partial differential equations. The second type approximates the continuous medium by a mesh of lumped resistors for which the currents and potentials are described by difference equations which approximate the differential equations of the electrical field. In general the fineness of the mesh structure must be such as to adequately represent the variables in regions of rapid change.

Early experience on the present problem indicates that very steep temperature gradients are expected in the vicinity of the contact points. Consequently, in this region, at least, a fine mesh of resistors is required for a discrete type of analogue. The further consideration which admits of varying the dead-zone angle leads to a prohibitively large number of resistors for this type of representation.

The continuous type analogue is not restricted by the rapidity of the variations of the variables. However, there is another kind of difficulty in this case which arises from the necessity of representing the film coefficient. If electrolytes are used to represent the thermal conductivity of the core and clad materials it becomes a practical difficulty to connect a conductive material at the boundary to represent the film coefficient.

A compromise involving both methods is used. Electrolytes represent the conductivities of the core and clad regions. At the surface fixed resistors are connected for every degree of angle to represent the film coefficient. The one degree increments are believed adequate to handle the expected temperature gradients and the electrolytes eliminate the necessity of extending this fine mesh over the entire area represented.

The design is illustrated diagrammatically in Figure 5. The tank represents a section of the sphere of width $\Delta\phi$. It is not necessary to represent the entire sphere since the solutions are independent of ϕ under the assumptions employed. The variable thickness of the section, increasing as $r \sin \theta$, is accomplished by inclining the tank to the horizontal. A slight approximation is introduced by the fact that the sides of the tank are straight and not curved to represent the curvature of the spherical surface. This effect is minimized by keeping the angle $\Delta\phi$ small; in this instance $\Delta\phi = 0.16$ rad.

There is an insulating barrier inserted at the clad-core interface to prevent mixing of the two electrolytes which have a conductivity ratio representative of the similar thermal conductivity ratio. Connection between the two regions is achieved by using small U shaped pieces of copper separated from each other and placed at one degree intervals along the interface boundary. Similar pieces are located every degree along the outer boundary for connection to the fixed resistors which represent the surface film coefficient.

Current is fed into the tank over the dead-zone region and flows out through the film coefficient resistors. The resistors are connected to a common

ground and the measurement of potentials relative to this ground are related to the temperature difference between the corresponding point in the thermal system and the water. To this temperature must be added the temperature resulting from the uniform flow as explained in Part II.

The results of tank measurements are plotted in Figure 7 and show the functional dependence of the dimensionless ratio

$$(1) \quad \frac{K_c}{bT} \frac{T(r=b)}{q_0} = f(\theta)$$

for several values of θ_0 . The other dimensionless parameters,

$$(2) \quad \pi_1 = \left(\frac{K_c}{K_m} \right) = 2; \quad \pi_2 = \left(\frac{b}{a} \right) = 1.25$$

$$\pi_6 = \frac{hbT}{K_c} = 15.1$$

were held constant during the course of the experiment. The symbols are defined in the nomenclature at the end of the report.

An important inference to be drawn from Figure 7 is the linear dependence of hot spot surface temperature on the dead zone angle θ_0 .

IV. Analytical Solution

The heat flow problem in spherical pellets formulated in Parts I and II is a boundary value problem. There are a variety of analytical methods which have been developed to treat such problems. In general the methods involve series or iterative methods of computation which have the disadvantage of being very time consuming. The advent of modern machine methods has greatly reduced this limitation and has made these techniques available for the solution of practical problems.

Unfortunately the analytical methods deal mostly with problems which specify either the potential or the flow variable on the boundary. In the present instance the specification over part of the boundary is in the form of a film coefficient which is the ratio of heat flux to temperature differential across the sphere-water interface. A straightforward approach to this problem which would yield solutions in a reasonable time, even with machine methods, has not been found. However, a method which makes use of both series and iterative techniques has been developed that appears to offer promise of an adequate solution.

The approach is to change the problem to one of the standard types. In this case the thermal flux over the surface of the sphere is specified. The resulting temperature distribution is then determined by a series method. Comparison of the assumed thermal flux and the calculated temperature indicates the manner in which the flux assumption should be modified to achieve the constant film coefficient condition.

Figure 6 shows the assumed thermal flux distribution. The heat flow density over the region near the point of contact is uniform and directed into the sphere as required by the boundary condition. As a first approximation the flux density is assumed constant over the rest of the sphere and may be modified in successive calculations to conform with the resulting temperature distribution.

In Appendix B it is shown that the temperature distribution within the sphere may be represented by

$$(3) \quad T_c - T_o = \frac{q_0 b}{k_c} \sum_{n=1}^{\infty} \left[\left(\frac{r}{b}\right)^{2n} A_{2n} + \left(\frac{r}{b}\right)^{-(2n+1)} B_{2n} \right] P_{2n}(\theta); \quad a \leq r \leq b$$

$$(4) \quad T_m - T_o = \frac{q_0 b}{k_c} \sum_{n=1}^{\infty} \left(\frac{r}{b}\right)^{2n} C_{2n} P_{2n}(\theta); \quad 0 \leq r \leq a$$

The constants A_{2n} , B_{2n} and C_{2n} are determined from the equivalence of the temperature and thermal flux at the clad-meat interface and from the impressed thermal flux at the outer boundary. The equations are

$$(5) \quad C_{2n} = \left(\frac{a}{b}\right)^{2n} A_{2n} + \left(\frac{a}{b}\right)^{-(2n+1)} B_{2n}$$

$$(6) \quad \left(\frac{a}{b}\right)^{2n} \left[\left(\frac{k_c}{k_m}\right) - 1 \right] A_{2n} - \left(\frac{a}{b}\right)^{-(2n+1)} \left[\left(\frac{k_c}{k_m}\right) \left(1 + \frac{1}{2n}\right) + 1 \right] B_{2n} = 0$$

$$(7) \quad A_{2n} - \left(1 + \frac{1}{2n}\right) B_{2n} = - \left(1 + \frac{1}{4n}\right) \frac{2}{q_0} \int_0^1 q_s(\mu) P_{2n}(\mu) d\mu$$

$$(8) \quad \mu = \cos \theta$$

The functions $P_{2n}(\mu)$ are the Legendre polynomials of the first kind which are orthogonal over the range $0 \leq \mu \leq 1$. $q_s(\mu)$ is the impressed flux distribution.

The series of Equations (3) and (4) omit the non-varying terms corresponding to the zero values of the index n , although these terms satisfy the differential equations. The reason for this omission is that the corresponding constants A_0 , B_0 and C_0 are indeterminate for a boundary condition in terms of thermal flux density. These constants are determined by accounting for the temperature drop in the surface film.

The temperature on the surface, $r = b$ is obtained from Equation (3) by summing the series at $r = b$.

$$(9) \quad T_c - T_0 / \left(\frac{q_0 b}{K_c} \right) = \sum_{n=1}^{\infty} [A_{2n} + B_{2n}] P_{2n}(\theta)$$

The calculation of the temperature distribution for four cases of dead zone angle, θ_0 , is being performed at the Anacom Laboratory on the IBM CPC equipment. Results of this calculation will indicate the temperature for ten degree increments of θ for each of the four θ_0 's.

The next approximation to the case of constant film coefficient is obtained from these results by perturbing the heat flux assumption in the required direction. The machine program must be modified to perform an iteration since the definite integral of Equation (7) changes with the new flux distribution. For the flux distribution shown in Figure (7), the

evaluation was greatly facilitated by the fact that $q_s(\theta)$ is constant in each of the two regions. In this case the integral may be evaluated from the expression

$$(10) \quad \int P_n(\mu) d\mu = \frac{P_{n+1}(\mu) - P_{n-1}(\mu)}{2n + 1}$$

in which only the values of the Legendre polynomials at θ_0 are required.

These values may be computed with the aid of the recurrence formula,

$$(11) \quad (n+1) P_{n+1} + n P_{n-1} = (2n+1)\mu P_n$$

In the first iteration the evaluation of the definite integral may be accomplished by varying the assumed thermal flux in a series of steps so that use may be made of Equation (10). At the present time it is not planned to perform this iteration owing to the cost of the computation. Since the problem as formulated in Part I is in itself only an estimate as to what will take place in the bed, it seems inappropriate to go to great expense to represent this approximation in an exact manner. Rather these initial calculations may be viewed as complementing the electrolytic tank work. They provide additional bounds on the temperature variations which may occur.

Conclusions and Further Work

The surface temperature distribution of a spherical pellet in the vicinity of a point of contact has been determined experimentally by means of an electrolytic tank analogue. The results show the variation of surface temperature with the spherical coordinate θ for several different values of dead zone angle θ_0 . It is assumed throughout that the heat transfer coefficient is zero in the dead zone and constant outside this region.

Similar results are expected shortly from an analytical method which makes use of a series expansion technique. The necessary numerical calculations are being performed by the Westinghouse Anacom Laboratory on the IBM, CPC calculator. The results achieved analytically will differ from the electrolytic tank results and should be somewhat more pessimistic. The reason for this is that the analytical method is an iterative one and which needs several iterations to converge. At present only the first approximation is planned.

The results of this study may be expanded in several directions if it is desired. The electrolytic tank as designed can readily be used for different conductivity ratios and with relatively minor modifications for different ratios of clad to meat radii. The analytic technique is even more flexible in these regards since to change ratios one merely replaces an old ratio with a new one and repeats the computation.

The problem of modifying the film coefficient to conform more closely with the experimental data is one which may be worthy of attention. The present approach assuming the coefficient to be zero in the dead zone

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and constant outside this region is undoubtedly pessimistic. However, if the temperatures as computed thereunder are ~~too~~ large it may be necessary to explore the possibility of refining this assumption.

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Nomenclature

T_m - Temperature in the meat or core - $^{\circ}\text{F}$
 T_c - Temperature in the clad - $^{\circ}\text{F}$
 T_o - Component of temperature independent of r and θ - $^{\circ}\text{F}$
 k_m - Meat conductivity - Btu/hr sq ft $^{\circ}\text{F}/\text{ft}$.
 k_c - Clad conductivity - Btu/hr sq ft $^{\circ}\text{F}/\text{ft}$.
 h - Film coefficient - Btu/hr sq ft $^{\circ}\text{F}$
 q_v - Heat generated per unit volume - Btu/hr cu ft.
 q - Thermal flux density - Btu/hr sq ft.
 q_o - Constant surface thermal flux density corresponding to flux case of uniform film coefficient. Btu/hr sq ft.
 q_s - Surface flux density - Btu/hr sq ft.
 V - Voltage in analogue - volts
 V_a - Voltage applied to the current density source resistors.
 j - Electrical current density - Amps/sq ft.
 j_v - Volumetric production of current - amps/cu ft.
 σ_m - Electrical analogue conductivity of the meat - mhos/ft.
 σ_c - Electrical analogue conductivity of the clad - mhos/ft.
 λ - Electrical analogue conductivity of the film - mhos/sq ft.
 a_T - Radius of meat in thermal system - ft.
 b_T - Radius of clad in thermal system - ft.
 a_e - Radius of meat in electrical system - ft.
 b_e - Radius of clad in electrical system - ft.
 R_i - Resistance associated with the i^{th} connection at the surface.
(represents the film coefficient - ohms).

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R_{Si} - Current density source resistors.

r, θ, ϕ - Spherical coordinate variables.

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Appendix A

1. Basis of Current Flow Analogue

The possibility of using a current flow field to represent a thermal heat flow field stems from the equivalence of the differential equations governing the two cases.

In the thermal case

$$(12) \quad q = -k \nabla T$$

$$\nabla \cdot q = -q_v$$

In the electrical system

$$(13) \quad j = -\sigma \nabla V$$

$$\nabla \cdot j = j_v$$

The correspondance ^e between these systems of equations forms the basis of the analogy. The j_v term is not used in the analogue but is included here to show what is required if the transformation described in Part II is not employed.

The mere correspondance of the differential equations of an analogue and the system it represents is not sufficient in itself to insure that the systems are completely equivalent. It is also necessary that the particular numerical values of the dimensionless ratios characterizing the systems be made equal.

The present systems may be characterized by the following dimensionless ratios.

$$\begin{array}{ll}
 \text{Thermal} & \text{Electrical} \\
 \hline
 \pi_{1T} = (k_c/k_m) & \pi_{1e} = (\sigma_c/\sigma_m) \\
 \pi_{2T} = (b_T/a_T) & \pi_{2e} = (b_e/a_e) \\
 (14) \quad \pi_{3T} = \theta_T & \pi_{3e} = \theta_e \\
 \pi_{4T} = \theta_o & \pi_{4e} = \theta_o \\
 \pi_{5T} = (k_c/b_T)(T/q) & \pi_{5e} = (\sigma_c/b_e)(V/j) \\
 \pi_{6T} = (h b_T/k_c) & \pi_{6e} = (\lambda b_e/\sigma_c)
 \end{array}$$

For the analogue to represent the physical system it is necessary that

$$\pi_{nT} = \pi_{ne}; \quad n = 1, 2, \dots, 6.$$

Representation of the Film Coefficient

As explained earlier the film coefficient is represented by fixed resistors connected to the outer boundary of the tank at one degree intervals. The numerical value of a resistor is determined from the equivalence of the π_6 's and the size of the elemental area served by one point of connection.

The resistance is

$$(15) \quad R_i = \frac{1}{\lambda A_i} = \frac{1}{\lambda b_e^2 \sin \theta_i \Delta \theta \Delta \phi},$$

$$\theta_i = 5-1/2; 6-1/2; \dots 89-1/2.$$

and substituting for λ from Equation (14)

$$\lambda = \frac{\sigma_c}{b_e} \pi_6$$

there results

$$(16) \quad R_i = \frac{1}{\pi_6 \sigma_c b_e \sin \theta_i \Delta\theta \Delta\phi} \text{ ohms.}$$

The final selection of the resistance depends on the choice of σ_c and the value of π_6 . For the present problem

$$(17) \quad \pi_6 = 15.1; \quad b_e = 25 \text{ in.}$$

Instead of conductivity it is more common in electrical systems to use the reciprocal quantity, resistivity, and the value chosen for the clad resistivity, ρ_c , is

$$\rho_c = \frac{1}{\sigma_c} = 250 \text{ ohm-cm.}$$

The resulting expression for the film resistances is

$$(18) \quad R_i = \frac{93.6}{\sin \theta_i} \text{ ohms.}$$

The resistances for $\theta_i = 5-1/2$ to $29-1/2$ are connected to the tank through switches so they may be disconnected at times when it is desired to extend the dead-zone above 5° .

Source Resistors

The constant current density which is supplied over the dead zone angle is achieved by using a voltage source and large series resistors connected to the tank at one degree intervals. The following resistance values

$$(19) \quad R_{Si} = \frac{(200)(93.6)}{\sin \theta_i} = \frac{18,720}{\sin \theta_i} \text{ ohms; } \theta_i = 1/2, 1-1/2, \dots 29-1/2$$

reduce the source voltage by a factor of about one hundred and adequately meet the current density source condition. The switches which disconnect the film coefficient resistors at the same time connect additional source resistances when the dead zone is extended above five degrees.

In computing the value of π_5 from measured analogue voltages it is convenient to express the dead-zone current density, j_0 , in terms of the more readily measured applied voltage.

Assuming the tank voltage small compared to the applied voltage, the applied current density is

$$(20) \quad j_0 = \frac{\lambda}{200} V_a$$

where the factor of two hundred is the one used in computing the source resistors.

From Equation (14)

$$(21) \quad \lambda = \frac{\sigma_c}{b_e} \pi_6$$

and upon substituting into the expression for j_o yields

$$(22) \quad j_o = \frac{\sigma_c}{b_e} \frac{\pi_6}{200} V_a.$$

Upon this result π_5 from Equation (14) is reduced to the following form

$$\pi_5 = \frac{200}{\pi_6} \frac{V}{V_a} = 13.2 \left(\frac{V}{V_a} \right).$$

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Appendix B

Analytical Solution of the Heat Flow Problem in a Sphere when the Boundary Condition is Specified in the Form of a Thermal Flux.

The differential equation governing the steady flow of heat in a conducting region is known as Laplace's equation. In spherical coordinates the general solution of this equation, for fields which are independent of ϕ may be expressed in terms of an infinite series of the form

$$(23) \quad T = \sum_{n=0}^{\infty} \left[A_n^! r^n + B_n^! r^{-(n+1)} \right] P_n(\theta)$$

where r and θ are the spherical coordinates and $P_n(\theta)$ are the Legendre polynomials of the first kind of order n . To be perfectly general the solution should include the functions $Q_n(\theta)$, the Legendre polynomials of the second kind. However, all of these functions are infinite for $\theta = 0$ and are therefore not admissible in this problem.

The Legendre polynomials are orthogonal over the range from 0 to π . It is therefore convenient to imagine that the θ variation is extended from $\pi/2$ to π and that in this region the film coefficient is symmetrical about $\pi/2$. This in no way affects the solution in the interval from 0 to $\pi/2$ but permits the use of the orthogonal properties of the Legendre polynomials.

The symmetry about $\theta = \pi/2$ eliminates the odd-order polynomials from the solution. Separate expressions for the temperature are required for the core and clad regions because of the different conductivities. It is convenient to change the constants $A_n^!$ and $B_n^!$ slightly to make them dimensionless. Making these changes Equation (23) takes the form

$$(24) \quad T_c - T_o = \frac{q_0 b}{k_c} \sum_{n=1}^{\infty} \left[A_{2n} \left(\frac{r}{b} \right)^{2n} + B_{2n} \left(\frac{r}{b} \right)^{-(2n+1)} \right] P_{2n}(\theta)$$

$$(25) \quad T_m - T_o = \left(\frac{q_0 b}{k_c} \right) \sum_{n=1}^{\infty} C_{2n} \left(\frac{r}{a} \right)^{2n} P_{2n}(\theta)$$

where T_o is the contribution from the zero term of the series, as explained earlier. The expression for the temperature in the meat does not involve the reciprocal r terms since they are infinite at the origin.

The constants are evaluated from the boundary conditions at

$$(26) \quad r = a \quad T_m = T_c \quad k_m \frac{\partial T_m}{\partial r} = k_c \frac{\partial T_c}{\partial r}$$

and at $r = b$ $q_s = -k_c \frac{\partial T_c}{\partial r}$

$$(27) \quad = -q_0 \sum_{n=1}^{\infty} \left[2nA_{2n} - (2n+1)B_{2n} \right] P_{2n}(\theta)$$

which upon substitution from Equations (24) and (25) yield the expression given in Equations (5), (6) and (7).

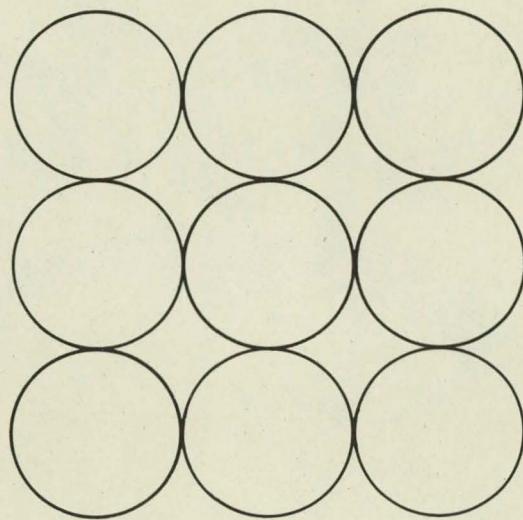
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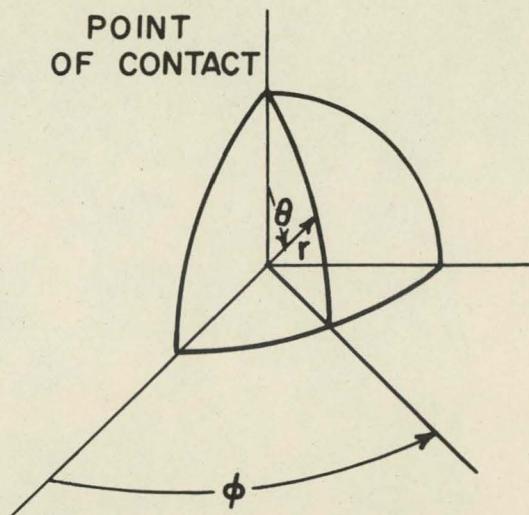
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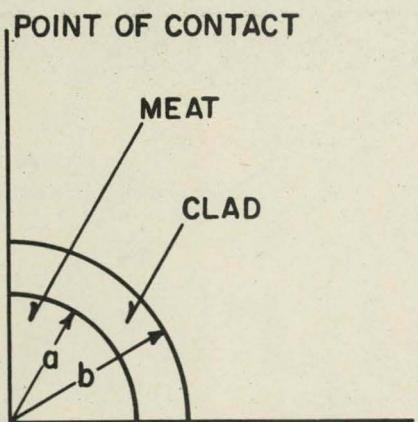
CROSS SECTION OF A PELLET BED
FIGURE 1



SECTION OF A SPHERICAL PELLET
FIGURE 2

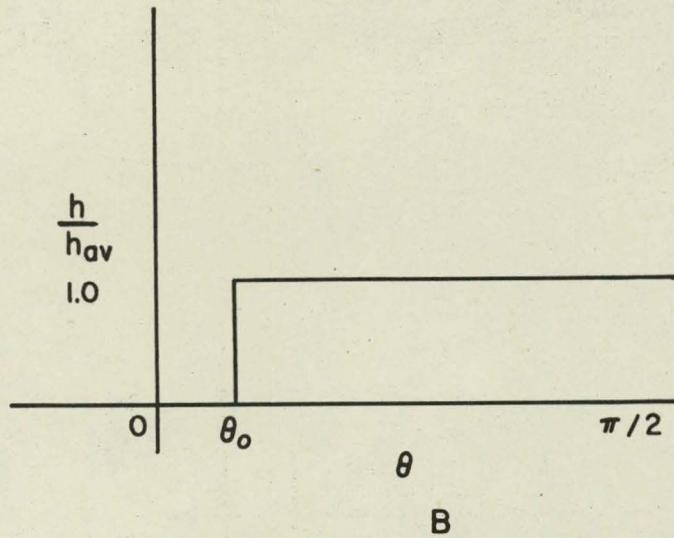
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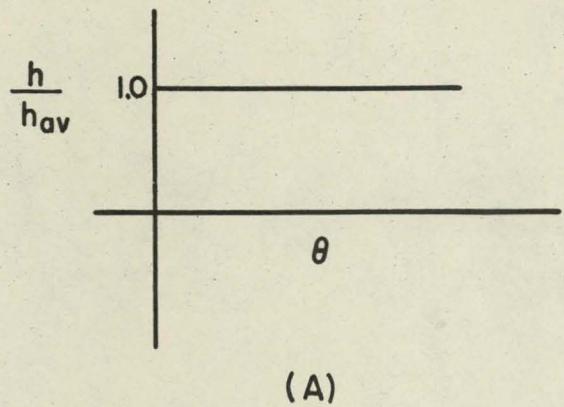
SECTION OF A SPHERICAL PELLET SHOWING
MEAT AND CLAD REGIONS



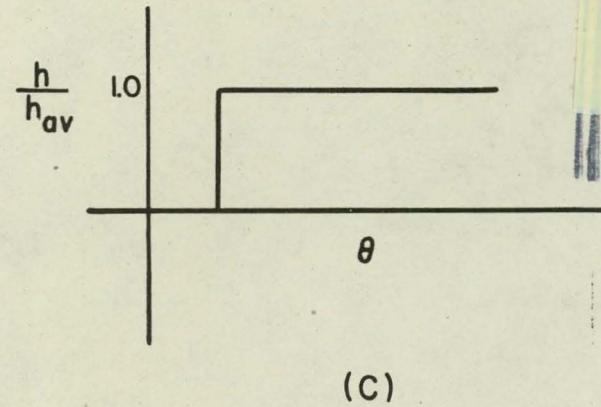
ASSUMED VARIATION OF
FILM COEFFICIENT

FIGURE 3

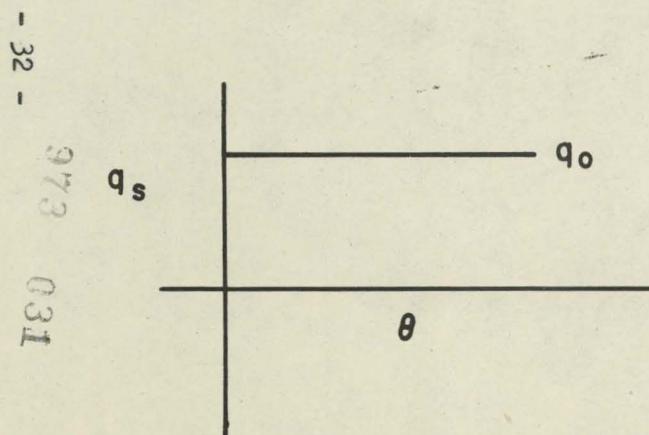
32



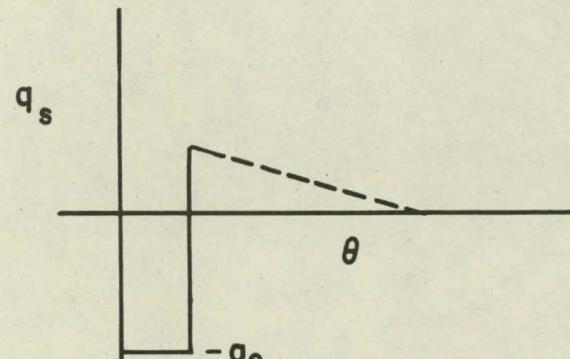
(A)



(C)



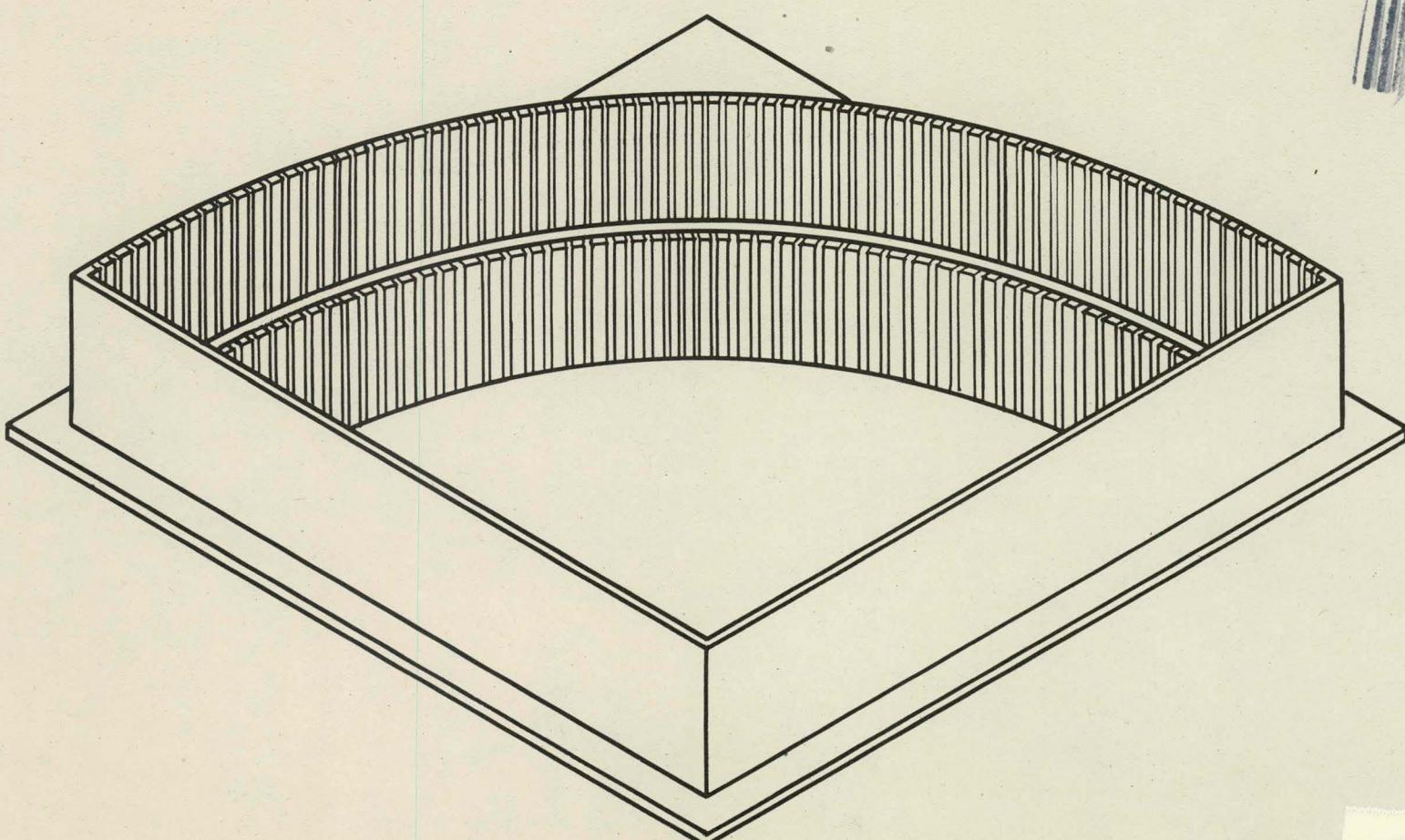
(B)



(D)

APPLICATION OF SUPERPOSITION TO
THE HEAT FLOW PROBLEM OF PART I

FIGURE 4



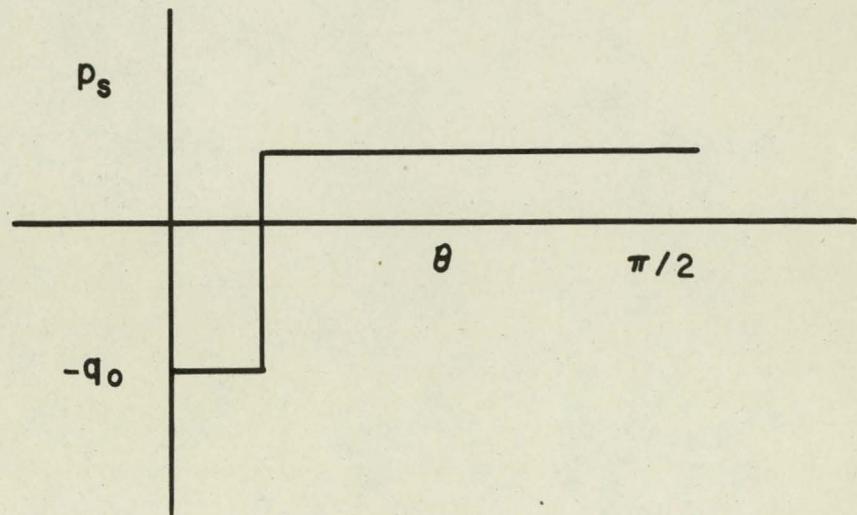
ELECTROLYTIC TANK
FIGURE 5

- 33 -

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SURFACE FLUX DENSITY ASSUMED AS A
FIRST APPROXIMATION IN THE ANALYTICAL METHOD

FIGURE 6

REF ID: A470-22-168

Series

$\Pi_5 = \frac{K_c}{bT} \frac{T(r=b)}{q_o}$ vs θ AT CONSTANT θ_o AS DETERMINED
FROM THE ELECTROLYTIC TANK

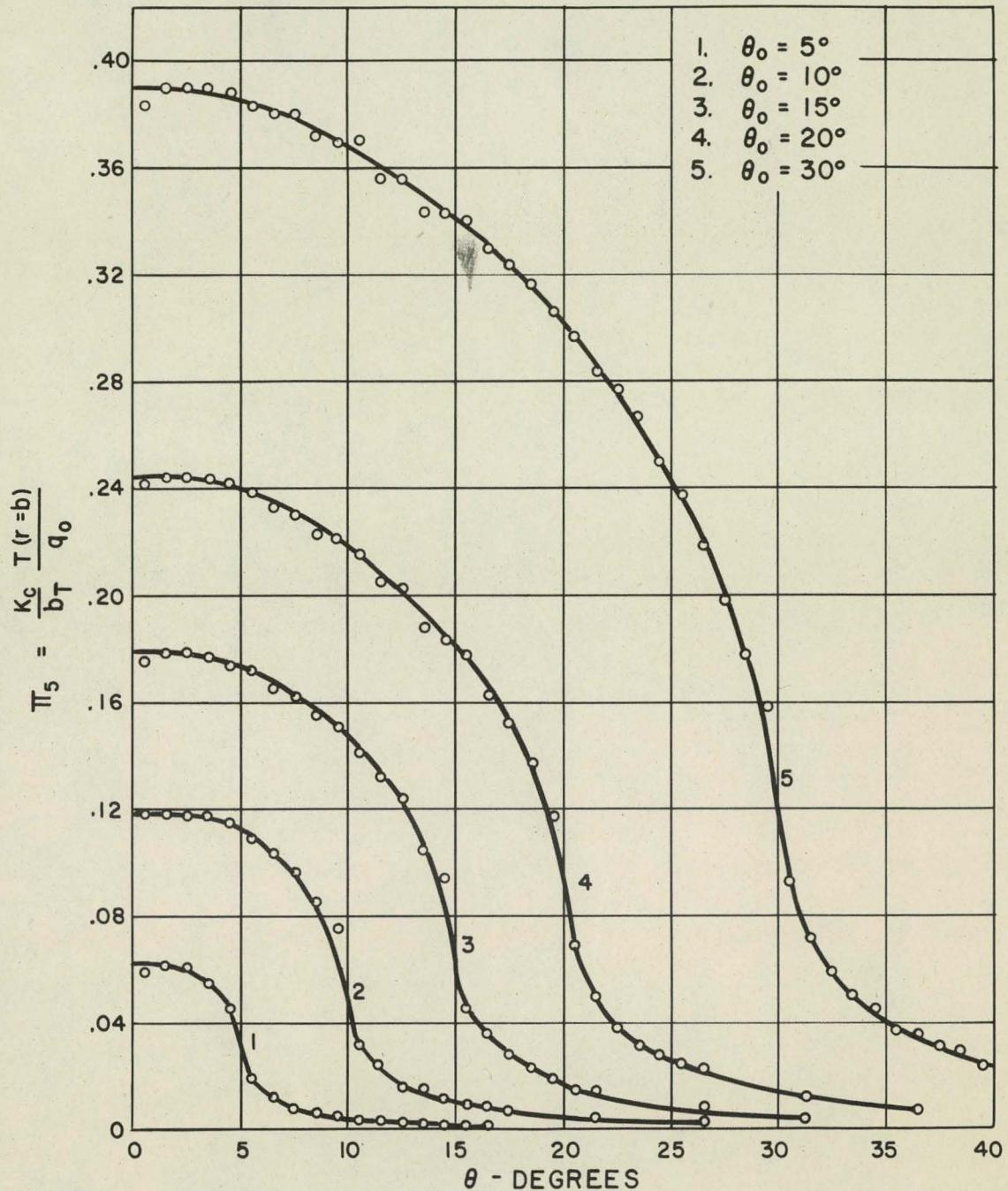


FIGURE 7