

ULTRAHIGH TEMPERATURE INTERMETALLIC ALLOYS

M.P. Brady, J.H. Zhu, C.T. Liu, P.F. Tortorelli, J.L. Wright, and C.A. Carmichael
Metals and Ceramics Division
Oak Ridge National Laboratory
Oak Ridge, TN 37831-6115 USA

ORNL/CP-99045

ABSTRACT

CONF-980561-

A new family of Cr-Cr₂Ta intermetallic alloys based on Cr-(6-10)Ta (at.%) is under development for structural use in oxidizing environments in the 1000-1300°C (1832-2372°F) temperature range. Developmental objectives relate to high temperature strength and oxidation resistance and room temperature fracture toughness. The 1200°C (2192°F) strength goals have been met: yield and fracture strengths of 275 MPa (40 ksi) and 345 MPa (50 ksi), respectively, were achieved. Progress in attaining reasonable fracture toughness of Cr-Cr₂Ta alloys has been made; current alloys exhibit room-temperature values of about 10-12 MPa \sqrt{m} (1.1 MPa \sqrt{m} = 1 ksi $\sqrt{in.}$). Oxidation rates of these alloys at 950°C (1742°F) in air are in the range of those reported for chromia-forming alloys. At 1100°C (2012°F) in air, chromia volatility was significant but, nevertheless, no scale spallation and positive weight gains of 1-5 mg/cm² have been observed during 120-h, 6-cycle oxidation screening tests. These mechanical and oxidative properties represent substantial improvement over Cr-Cr₂Nb and Cr-Cr₂Zr alloys previously developed.

INTRODUCTION

The objective of this work is to develop high-strength, oxidation- and corrosion-resistant intermetallic alloys for use as hot components in advanced fossil energy conversion and combustion systems that meet the 65% efficiency goal of the Vision 21 Concept. Successful development of these alloys is expected to allow improvement in thermal efficiency primarily through decreased cooling requirements in advanced gas turbines. Initially targeted applications include components such as vanes, seals, and nozzles. However, these materials may also find other uses as wear-resistant parts in coal handling systems (for example nozzles), drill bits for oil/gas wells, valve guides in diesel engines, and interconnects in solid oxide fuel cells.

Development efforts have focused on two-phase, in-situ composite alloys of the type Cr-Cr₂X, where Cr₂X is a Cr-based refractory Laves phase compound (Cr₂Nb, Cr₂Zr, or Cr₂Ta). Toughness and oxidation resistance are provided by a soft Cr_{ss} (solid solution) matrix. The Cr₂X Laves phase acts as a reinforcement to the Cr_{ss} matrix in order to provide high temperature strength. Recently, we identified a promising new class of alloys based on the Cr-Cr₂Ta system^{1,2} (Fig. 1), with melting points in the 1600-1700°C (2912-3092°F) range³. Preliminary evaluation suggested that the Cr-Cr₂Ta based alloys possess superior mechanical properties and oxidation resistance to the Cr-Cr₂Nb and Cr-Cr₂Zr alloys previously developed in this program³. This report presents an overview of current Cr-Cr₂Ta alloy development objectives and performance.

DISCUSSION OF CURRENT ACTIVITIES

Three properties are key to the initial development of the two phase Cr-Cr₂Ta based alloys: room temperature fracture toughness, high temperature strength, and high temperature oxidation resistance. The target mechanical properties are room temperature fracture toughness $\geq 15-20$ MPa \sqrt{m} (1.1 MPa \sqrt{m} = 1 ksi $\sqrt{in.}$) and 345 MPa (50 ksi) tensile fracture strength at 1200°C (2192°F). Oxidation kinetics in the

MASTER

The submitted manuscript has been authored
by a contractor of the U.S. Government under
contract No. DE-AC05-96OR22464.
Accordingly, the U.S. Government retains a
nonexclusive, royalty-free license to publish
or reproduce the published form of this
contribution, or allow others to do so, for
U.S. Government purposes.

RECEIVED

AUG 13 1998

OSTI

DISCLAIMER

This report was prepared as an account of work sponsored by an agency of the United States Government. Neither the United States Government nor any agency thereof, nor any of their employees, makes any warranty, express or implied, or assumes any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represents that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States Government or any agency thereof. The views and opinions of authors expressed herein do not necessarily state or reflect those of the United States Government or any agency thereof.

DISCLAIMER

**Portions of this document may be illegible
in electronic image products. Images are
produced from the best available original
document.**

range of commercial chromia (Cr_2O_3) forming alloys is the goal for temperatures up to about 900-1000°C (1652-1832°F). At higher temperatures, where chromia scale volatility is significant⁴, the goal is noncatastrophic oxidation behavior of the bare substrate alloy such that use of a protective coating is a reasonable protection approach (ie. local coating failure would not result in immediate component loss).

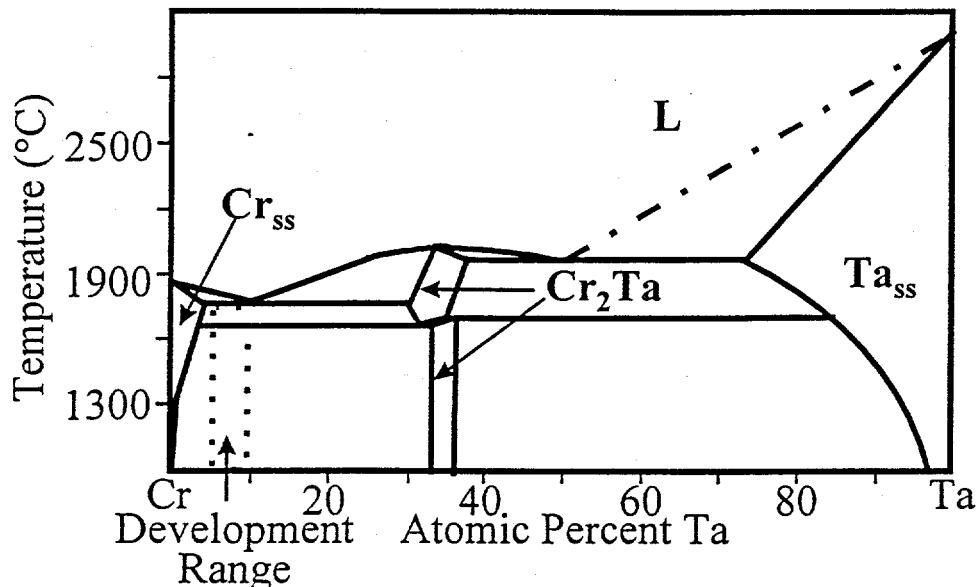


Fig. 1- Schematic binary Cr-Ta phase diagram^{1,2}.

ALLOY PREPARATION AND PROCESSING

Alloys based on the Cr-Cr₂Ta system weighing 400-500 g were prepared by arc melting and drop casting in a copper mold (2.5 cm diam x 7.6 cm long) preheated to 200°C (392°F). Small 40-60 g castings were also prepared by arc melting and drop casting in a chilled copper mold. High-purity chromium and other metal chips were used as charge materials. The alloys were easily cast and showed no evidence of cracking. After heat treatment, the larger 400-500 g alloy ingots were clad inside Mo billets and hot extruded at 1480°C (2696°F) and an extrusion ratio of 4:1. The hot-extruded material was then heat treated and used for tensile specimens. The small 40-60 g castings were used as-cast and/or after heat treatment to provide specimens for oxidation exposures and room temperature fracture toughness evaluation. The development range is based on Cr-(6-10)Ta atomic percent, at.-% (Fig. 1). Detailed alloy compositions are not reported due to patent considerations, and are denoted in the report by the letter "X" followed by a number.

ROOM TEMPERATURE FRACTURE TOUGHNESS

Chevron notched three-point bend samples approximately 3 mm x 4 mm x (20-25) mm were used to evaluate fracture toughness at room temperature. A modulus of 250 GPa (36 ksi) was estimated for the Cr-Cr₂Ta based alloys. A modulus of 300 GPa (43 ksi) was used for the Cr-(0-0.1X10) alloys. Details of this technique are provided in ref. 5.

Current Cr-Cr₂Ta based alloys have a room temperature fracture toughness in the 10-12 MPa \sqrt{m} range, with one alloy test sample exhibiting a toughness of 12.8 MPa \sqrt{m} (Table 1). This represents significant progress beyond the 7-8 MPa \sqrt{m} room temperature fracture toughness exhibited by the best Cr-Cr₂Nb based alloys previously developed under this initiative⁶. However, it is still short of the goal levels of 15-20 MPa \sqrt{m} .

Room temperature fracture toughness values of several binary Cr-Ta and microalloyed Cr alloys are shown in Table 2. A comparison of Table 1 with Table 2 reveals that the alloy additions to the Cr-(8-10)Ta base have not significantly increased room temperature fracture toughness. The binary Cr-(8-10)Ta alloys exhibit room temperature fracture toughness values in the 9-10 MPa \sqrt{m} range, only slightly below the 10-12 MPa \sqrt{m} range achieved with further alloying. (It should be noted the alloying strategy adopted thus far has proven effective in improving high temperature strength and oxidation resistance).

Table 1- Room temperature fracture toughness of selected Cr-Cr₂Ta based alloys.

Alloy No.	Composition (At.%)	Microstructure	Condition and Heat Treatment	Fracture Toughness (MPa \sqrt{m})	
				Best	Average (# of tests)
CN144	Cr-8Ta-5X1-0.5X3-0.011X8	Hypoeutectic	extruded, 1200°C	11.0	11.0 (1)
			cast, 1000°C	10.2	8.3 (2)
			cast, 1200°C	8.3	7.9 (2)
CN145	Cr-10Ta-5X1-0.5X3-0.011X8	Hypereutectic	cast, 1000°C	8.7	6.0 (2)
			cast, 1200°C	12.8	11.1 (2)
CN147	Cr-9.5Ta-5X1-0.5X3-0.011X8	Eutectic	cast, 1200°C	11.7	10.7 (4)

Table 2- Room temperature fracture toughness of selected binary Cr-Ta and Cr alloys.

Composition (At.%)	Microstructure	Condition and Heat Treatment	Fracture Toughness (MPa \sqrt{m})	
			Best	Average (# of tests)
Binary Cr-Ta				
Cr-8Ta	Hypoeutectic	cast, 1000°C	10.4	9.8 (3)
Cr-11Ta	Hypereutectic	cast, 1000°C	9.8	9.7 (2)
Cr-9.8Ta	≈ Eutectic	cast, 1000°C	9.0	9.4 (2)
Cr Matrix				
Pure Cr	Cr	cast, 1000°C	16.1	15.2 (2)
Cr-0.1X10		cast, 1000°C	34	22.3 (3)

Alloying for further improvement in room temperature fracture toughness will focus on two key areas: the Cr_{ss} phase and the Cr_{ss}/Laves phase interface. Pure Cr containing 200 O and 25 N weight parts per million, processed under similar conditions to the Cr-Cr₂Ta based alloys, exhibits a room temperature fracture toughness of 15-16 MPa \sqrt{m} (Table 2). Microalloying additions to improve the toughness of Cr are currently under evaluation. For example, the addition of 0.1 at.% of X10 was recently found to increase the room temperature fracture toughness to level of 34 MPa \sqrt{m} in one test sample (Table 2).

Manipulation of the Cr_{ss} /Laves phase interface is the second route under consideration to further improve toughness. Understanding of processing/composition/microstructure relationships in the Cr-Ta system, including phase equilibria studies to determine the eutectic composition, has been achieved. Microstructures consisting of a fine dispersion of Laves particles in the Cr_{ss} matrix or lamellar Cr_{ss} /Laves eutectic structures can now be produced (Fig. 2). Very preliminary results suggest that the lamellar eutectic structures (Fig. 2a) may exhibit greater room temperature fracture toughness than the fine dispersed structures (Fig. 2b). Future work will concentrate on confirming and optimizing this effect through adjustments of microstructural parameters such as lamellar spacing and lamellar colony grain size.

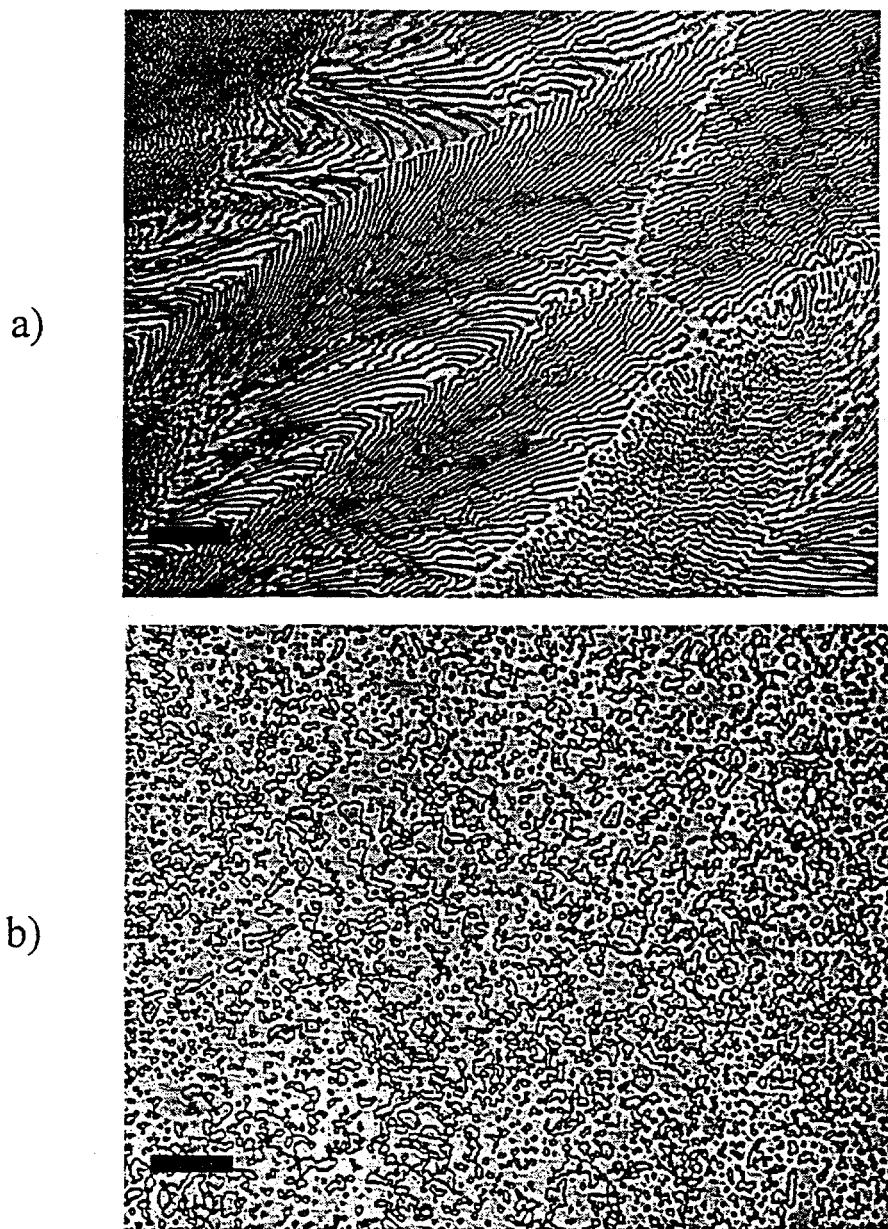


Fig. 2- Possible microstructures for Cr-Cr₂Ta based alloys (light microscopy). a) lamellar eutectic
b) Cr₂Ta Laves particles dispersed in a Cr_{ss} matrix.

TENSILE PROPERTIES

Button-type tensile specimens with gage dimensions 0.31 cm diam x 0.97-1.3 cm long were cut from hot-extruded material by electro-discharge machining, followed by grinding and polishing with "0" Emery paper. The specimens were tensile tested at room temperature and 800°C (1472°F) in air, and at 1200°C (2192°F) in vacuum (crosshead speed of approximately 0.25 cm/minute). Tensile properties are summarized in Tables 3-5.

At room temperature, no macroscopic yielding prior to fracture was observed. However, tensile fracture strengths in excess of 700 MPa (100 ksi) were achieved (Table 3). At 800°C in air, tensile elongations on the order of 1% were observed in several alloys, with yield strengths of around 900 MPa (130 ksi) (Table 4). This suggests that the ductile to brittle transition temperature (DBTT) was just under 800°C (1472°F). At 1200°C (2192°F), elongations were generally in the 20-30% range, with several alloys exhibiting tensile fracture strengths in excess of 345 MPa (50 ksi) (Table 5). Thus, the Cr-Cr₂Ta based alloys meet the high temperature strength goals of the program.

Table 3- Room temperature tensile properties.

Alloy No.	Composition Cr-8Ta-0.5X3 Base (At.%)	Yield Strength ksi (MPa)	Fracture Strength ksi (MPa)	Elongation %
CN130	5X1	>104 (717)	104 (717)	*
CN132	5X1-3X2-0.05X6	>71 (489)	71 (489)	*
CN133	2.5X1-3X2	>101 (696)	101 (696)	*
CN136	6.5X1-0.5X6	>87 (600)	87 (600)	*
CN137	6.5X1-0.05X6-0.022X8	>92 (634)	92 (634)	*
CN144	5X1-0.011X8	>104 (717)	104 (717)	*
CN146	8X7-0.011X8	>80 (552)	80 (552)	*

*Fracture Prior to Macroscopic Yielding

Table 4- 800°C (1472°C) tensile properties.

Alloy No.	Composition Cr-8Ta-0.5X3 Base (At.%)	Yield Strength ksi (MPa)	Fracture Strength ksi (MPa)	Elongation %
CN130	5X1	120 (827)	120 (827)	0.8
CN132	5X1-3X2-0.05X6	134 (924)	137 (944)	1.2
CN133	2.5X1-3X2	>69 (476)	69 (476)	*
CN136	6.5X1-0.5X6	>118 (813)	118 (813)	*
CN137	6.5X1-0.05X6-0.022X8	128 (882)	131 (903)	0.9
CN144	5X1-0.011X8	130 (896)	133 (917)	1.0
CN146	8X7-0.011X8	>116 (800)	116 (800)	*

*Fracture Prior to Macroscopic Yielding

Table 5- 1200°C (2192°F) tensile properties.

Alloy No.	Composition Cr-8Ta-0.5X3 Base (At.%)	Yield Strength ksi (MPa)	Fracture Strength ksi (MPa)	Elongation %
CN130	5X1	37 (255)	45 (310)	16.6
CN132	5X1-3X2-.05X6	42 (290)	51 (352)	18.5
CN133	2.5X1-3X2	30 (207)	36 (248)	22.0
CN136	6.5X1-0.5X6	46 (317)	56 (386)	14.3
CN137	6.5X1-0.05X6-0.022X8	37 (255)	45 (310)	28.1
CN144	5X1-0.011X8	37 (255)	46 (317)	39.0
CN146	8X7-0.011X8	30 (206)	36 (248)	25.8

HIGH TEMPERATURE OXIDATION

The development effort focused on optimization of chromia scale growth kinetics of Cr-Cr₂Ta based alloys through microalloying, with an emphasis on reactive element (RE) additions. A series of alloys based on Cr-8Ta-5X1 with microalloying additions in the 0.05 to 1 at.% range was selected for evaluation. Disk shaped oxidation specimens of 8-13 mm in diameter and 1 mm thickness were sectioned from as-cast and/or heat treated material and polished to a 600 grit finish using SiC paper. Evaluation of oxidation was performed by thermogravimetric analysis (TGA) at 950°C (1742°F) in dry air for 168 h and short term cyclic oxidation screening at 1100°C (2012°F). Screening tests at 1100°C (2012°F) were performed in a tube furnace open to room air. The samples were placed in a covered alumina crucible and inserted/removed from the furnace at temperature after intervals of 1, 4, 10, 30, 48, and 120 h of cumulative exposure. Significant chromia volatility occurred as evidenced by green stains on the underside of the alumina crucible lid.

At 950°C (1742°F) in air, RE modified Cr-8Ta alloys exhibited a 2-7 fold decrease in the rate of oxidation relative to binary Cr-8Ta (Fig. 3). No isothermal scale cracking was observed for the modified alloys during a 168 h isothermal exposure and the scale remained adherent on cooling. This behavior is in contrast to binary Cr-8Ta, which exhibited a relatively low rate of oxidation but suffered from isothermal scale cracking and scale spallation on cooling to room temperature. The best RE modified Cr-8Ta based alloy oxidized at a rate within the range defined at the high end by Waspalloy, a commercial chromia-forming superalloy, and at the low end by MA754, a Y₂O₃ dispersed Ni-Cr alloy (1000°C data) which is among the most oxidation resistant chromia-forming alloys (Fig. 3). It should be noted that the oxidation kinetics of the RE modified Cr-8Ta based alloys were not parabolic, although the rate of oxidation did decrease with time.

Microalloying additions were also successful at reducing the rate of oxidation at 1100°C (2012°F). The alloys generally exhibited a 3-5 fold decrease in the rate of oxidation and exhibited no scale spallation during the short term 1100°C (2012°F) cyclic oxidation screening test. The most recent RE modified Cr-8Ta based alloy exhibited a 50 fold decrease in the rate of oxidation as compared with Cr-8Ta-5X1 (Fig. 4), although it should be noted that this result has not yet been repeated.

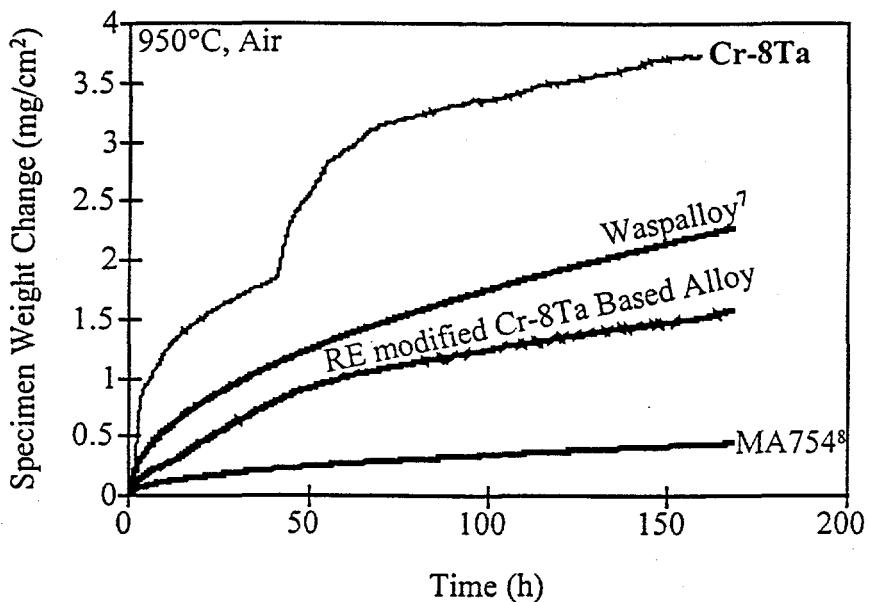


Fig. 3-TGA data at 950°C in dry air.

Overall, microalloying of the Cr-Cr₂Ta based alloys was successful in reducing the rate of oxidation and in improving scale adherence, although further compositional optimization is needed. Short term (< 200 h) behavior was in the target goal range of chromia-forming oxidation kinetics below 900-1000°C (1652-1832°F) and noncatastrophic oxidation at 1100°C (2012°F). Although the behavior thus far is promising, longer term exposures (>1000 h) are needed (and planned) before an assessment can be made of the oxidation performance of the microalloyed Cr-Cr₂Ta based alloys. Of particular concern is possible embrittlement from oxygen/nitrogen. Preliminary results suggest that the chromia scale formed on the Cr-Cr₂Ta based alloys can provide some protection from this type of embrittlement, but only if the scale remains adherent.

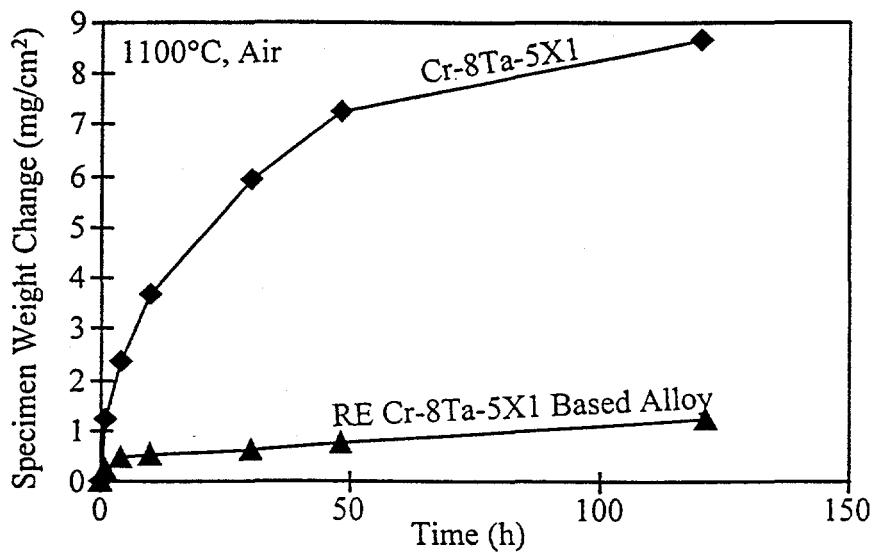


Fig. 4- Cyclic oxidation data at 1100°C in air.

SUMMARY AND FUTURE WORK

The room temperature fracture toughness, high temperature strength, and oxidation resistance of a series of developmental Cr-Cr₂Ta based alloys were evaluated. The developmental goal of 345 MPa (50 ksi) for tensile fracture strength at 1200°C (2192°F) was met. Significant improvement in room temperature fracture toughness over previously developed Cr-Cr₂Nb based alloys was achieved, with the Cr-Cr₂Ta based alloys exhibiting toughness in the 10-12 MPa \sqrt{m} range as compared with 7-8 MPa \sqrt{m} for Cr-Cr₂Nb. Although short of the goal of 15-20 MPa \sqrt{m} , further improvements are thought possible through microalloying of the Cr_{ss} phase and manipulation of the Cr_{ss}/Laves phase interface. Microalloying with reactive elements successfully reduced the rate of chromia scale formation during short term 950°C (1742°F) and 1100°C (2012°F) exposures in air. Long-term oxidation studies and optimization of microalloying levels are planned.

REFERENCES

1. M. Venkatraman and J.P. Neumann, in Binary Alloy Phase Diagrams, Vol. 1, T.B. Massalaski, J.L. Murray, L.H. Bennett, and H. Baker eds., ASM, p. 867 (1986).
2. J.H. Zhu and C.T. Liu, unpublished research (1997).
3. M.P. Brady, J.H. Zhu, C.T. Liu, P.F. Tortorelli, J.L. Wright, C.A. Carmichael, and L. R. Walker, in *Proc. of the Eleventh Annual Conference on Fossil Energy Materials*, R.R. Judkins (comp.), U. S. Department of Energy, p. 195 (December 1997).
4. P. Kofstad, High-Temperature Corrosion, Elsevier, London, 1988.
5. J.H. Schneibel, C.A. Carmichael, E.D. Specht, and R. Subramanian, *Intermetallics*, 5, p. 61 (1997).
6. C. T. Liu, P. F. Tortorelli, J. A. Horton, D. S. Easton, J. H. Schneibel, L. Heatherly, C. A. Carmichael, M. Howell, and J. L. Wright, in *Proc. Ninth Annual Conf. Fossil Energy Materials*, N. C. Cole and R. R. Judkins (comp.), U. S. Department of Energy, p. 415 (August 1995).
7. J.H. Chen, P.M. Rogers, and J.A. Little, *Oxidation of Metals*, Nos. 5/6, p. 381 (1997).
8. B.A. Pint and L. Hobbs, unpublished research (1990).

ACKNOWLEDGEMENTS

The authors thank Dewey Easton for the hot extrusions, and Jim Distefano, Lee Pike, and Ian for their reviews of the manuscript. This research was sponsored by the Fossil Energy Advanced Research and Technology Development (AR&TD) Materials Program, U S. Department of Energy, under contract DE-AC05-96OR22464 with Lockheed Martin Energy Research Corporation.