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Puncture of Thin Aluminum 7075-T651 Plates

Experiments and Simulations

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Introduction

Dynamic metallic plate puncture problems are of interest in many applications

They are demanding on the elastic-plastic and ductile failure models in FEAs.

- Temperature and rate dependencies may need to be addressed
- Tensile-dominated and shear-dominated failure modes can be activated
- Possible sensitivity to element size



Objectives

- Identify if calibrations of thermal-mechanical material models made from one lot can accurately predict puncture of plates from a different lot.

t , mm	σ_o , MPa	σ_u , MPa	ε_f (%)
1	490 (-11)	564(-9)	12 (1)
1.6	517 (-6)	588 (-5)	12.7 (5)
2	518 (-6)	583 (-6)	11.4 (-6)
3.2	501 (-9)	571 (-8)	12.9 (7)
4.8	495 (-10)	571 (-8)	12.3 (2)
12.7	551	618	12.1 ← Stock used for calibrations

- Compare FE predictions of plate puncture experiments to experimentally identified:
 - Minimum punch velocity to cause puncture
 - Force-time histories
 - Failure mode

The graphic features a central dark blue diamond with the text "Material Model Calibration" in white. This diamond is surrounded by a white border and is flanked by two diagonal lines of colored segments (cyan, orange, green, red, purple) extending from the corners towards the center. The background is white with faint, light blue geometric patterns.

Material Model Calibration



Plasticity Model Calibration

J₂ Yield Function with Isotropic Hardening:

$$f(\sigma_{ij}, \bar{\varepsilon}^p, T, \dot{\varepsilon}^p) = \phi(\sigma_{ij}) - \sigma_y(\bar{\varepsilon}^p, T, \dot{\varepsilon}^p) \quad \phi(\sigma_{ij}) = \sqrt{\frac{3}{2} s_{ij} s_{ij}}$$

Associated Flow Rule:

$$\dot{\varepsilon}_{ij}^p = \lambda \frac{\partial \phi}{\partial \varepsilon_{ij}^p}$$

Hardening Function:

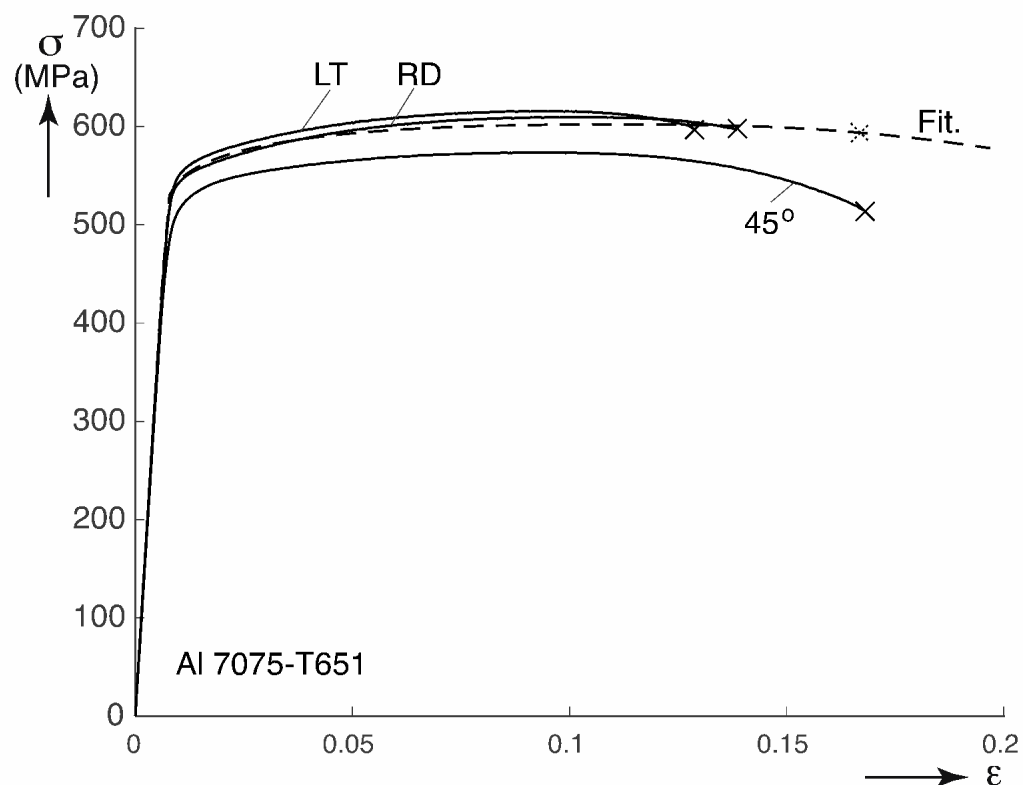
$$\sigma_y(\bar{\varepsilon}^p, T, \dot{\varepsilon}^p) = \left[\sigma_y^o(T) + A(T) (\bar{\varepsilon}^p)^{n(T)} \right] \left[1 + C(T) \ln \left(\frac{\dot{\varepsilon}^p}{\dot{\varepsilon}_o} \right) \right]$$

Adiabatic Heating:

$$T(t) = T(0) + \frac{\beta^{TQ}}{\rho c_p} \int_0^t \phi \frac{\partial \bar{\varepsilon}^p}{\partial \tau} d\tau$$



Plasticity Calibration: Effect of Temperature



Values at 20 °C

ρ , kg/m ³	E , GPa	ν	c_p , J/kg-K	β^{TQ}
2810	71.7	0.33	960	0.7

σ_y , MPa	A , MPa	n	C ,	$\dot{\epsilon}_o$, 1/s
517	405	0.41	0.008	1.6×10^{-4}

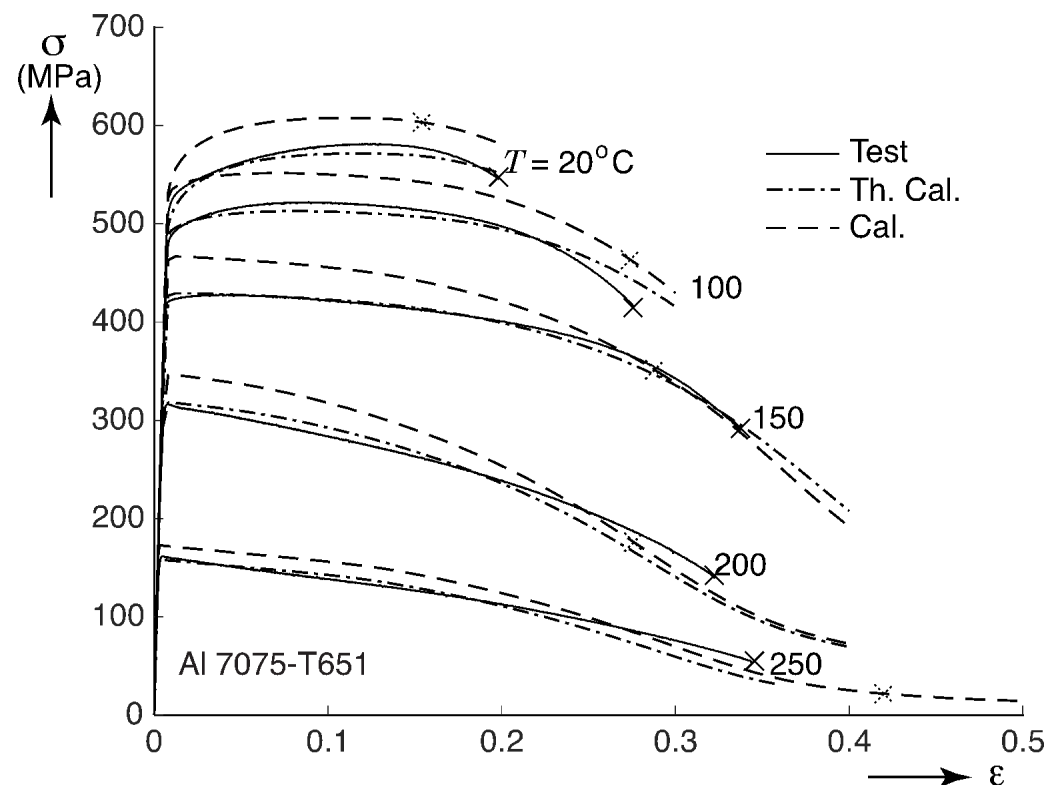
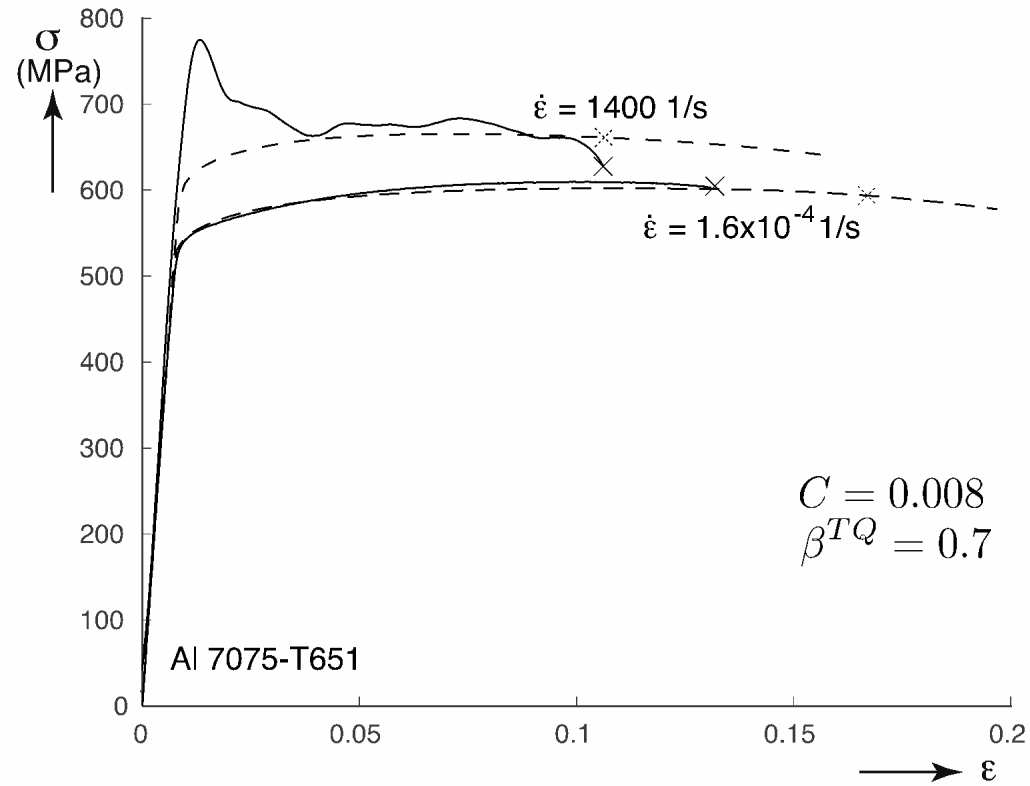


Table of
Function
Values

T , °C	$\frac{\sigma_y^0(T)}{\sigma_y^0(20)}$	$\frac{A(T)}{A(20)}$	$\frac{n(T)}{n(20)}$	$\frac{C(T)}{C(20)}$
20	1.0	1.0	1.0	1.0
100	1.01	0.8	1.5	1.0
150	0.89	0.75	2.2	1.0
200	0.66	0.3	2.44	1.33
250	0.32	0.0	2.44	4.0
750	0.0	0.0	2.44	13.3



Plasticity Calibration: Effect of Strain Rate





Ductile Failure Model (One-way coupling to plasticity)

Wilkins
Model

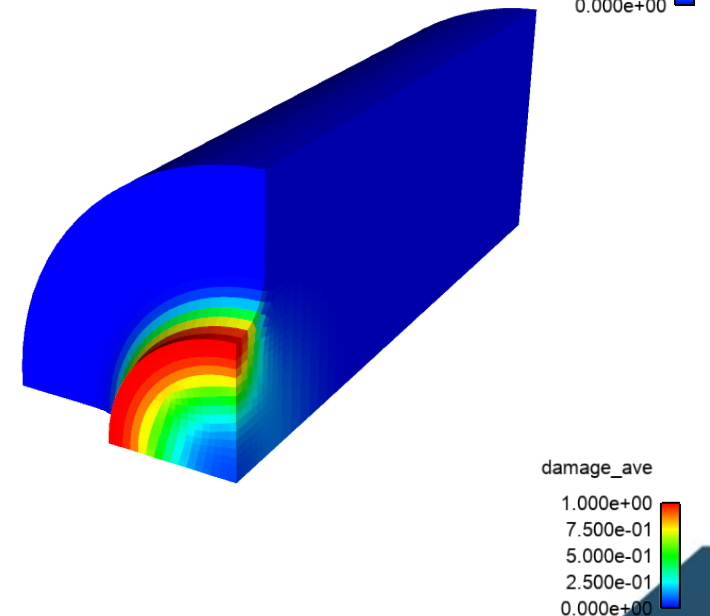
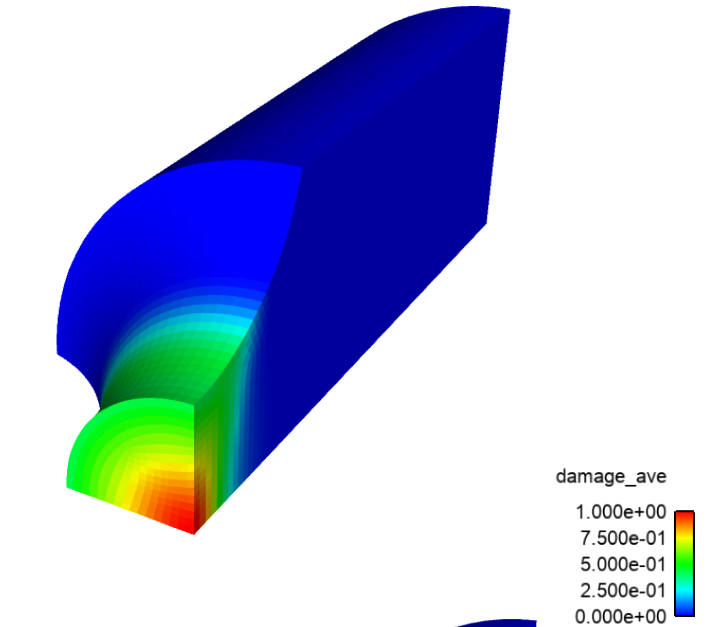
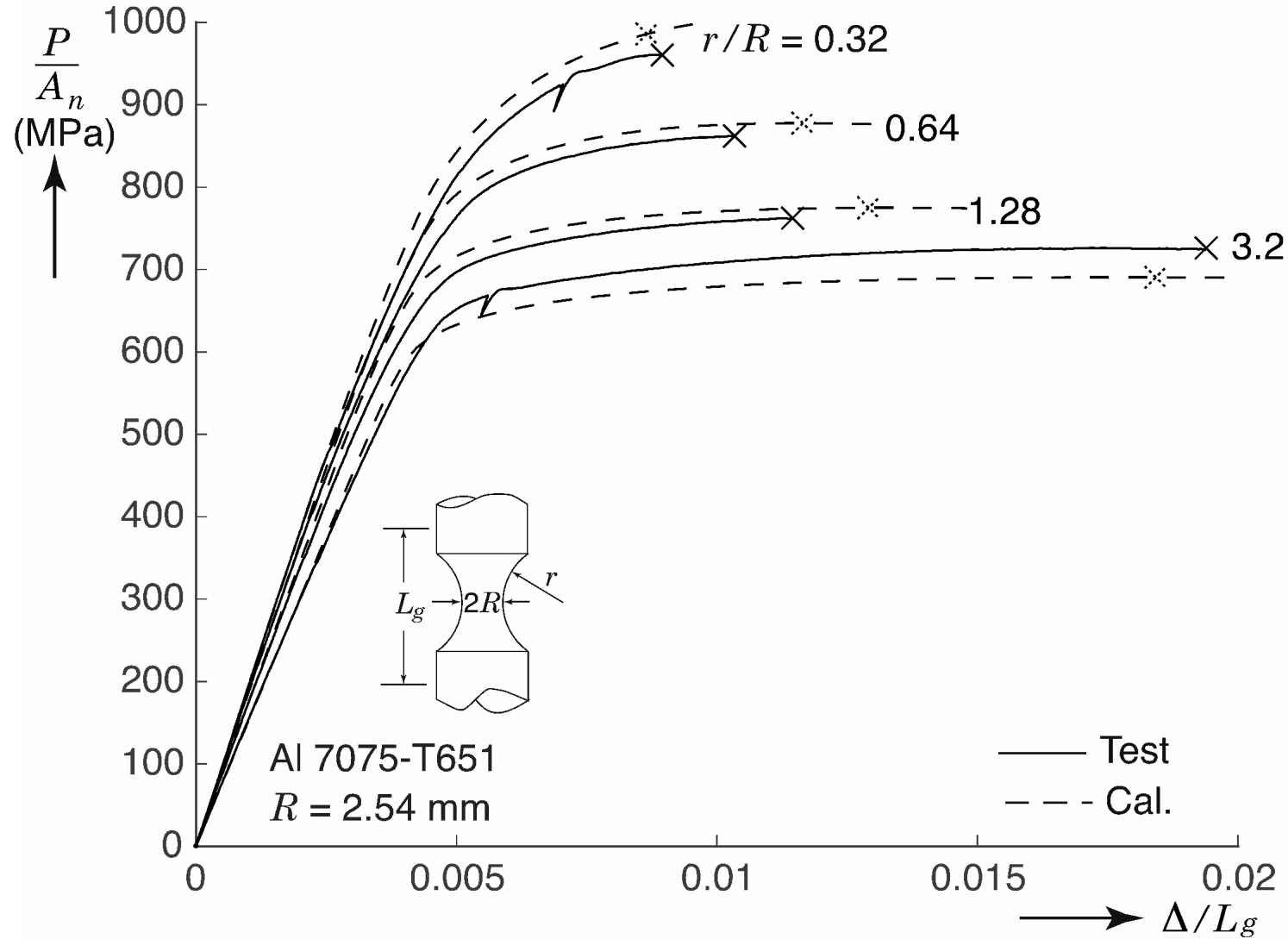
$$D = \frac{1}{D^{\text{cr}}} \int_0^{\bar{\varepsilon}^p} w_1(\sigma_m) w_2(\theta) w_3(\dot{\bar{\varepsilon}}^p) w_4(T) d\hat{\varepsilon}^p$$

$$w_1 = \left(\frac{1}{1 - \frac{\sigma_m}{B}} \right)^\alpha \quad w_2 = (2 - \mathcal{A})^\beta, \quad \mathcal{A} = \max \left(\frac{s_2}{s_3}, \frac{s_2}{s_1} \right)$$

$$w_3(\dot{\bar{\varepsilon}}^p) = \frac{1}{1 + D_4 \ln \frac{\dot{\bar{\varepsilon}}^p}{\dot{\varepsilon}_0}} \quad w_4(T) = \frac{1}{1 + D_5 \frac{T - T_{\text{ref}}}{T_{\text{melt}} - T_{\text{ref}}}}$$

B , GPa	α	β	D^{cr}	D_4	D_5	T_{melt} , ° C	T_{ref} , ° C
2.07	4.1	0.6	0.3	-0.039	22.6	750	20

Ductile Failure Model Calibration: QS Notched Tension Tests





Ductile Failure Model Calibration: QS Shear Dominated Tests

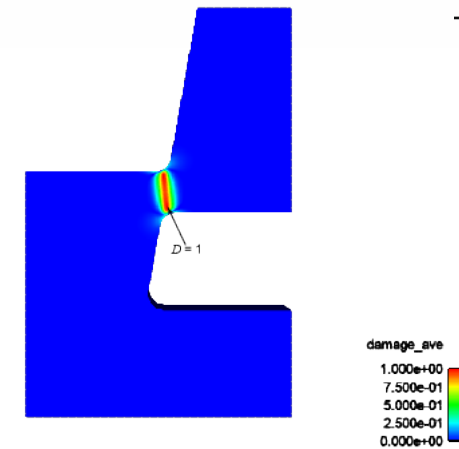
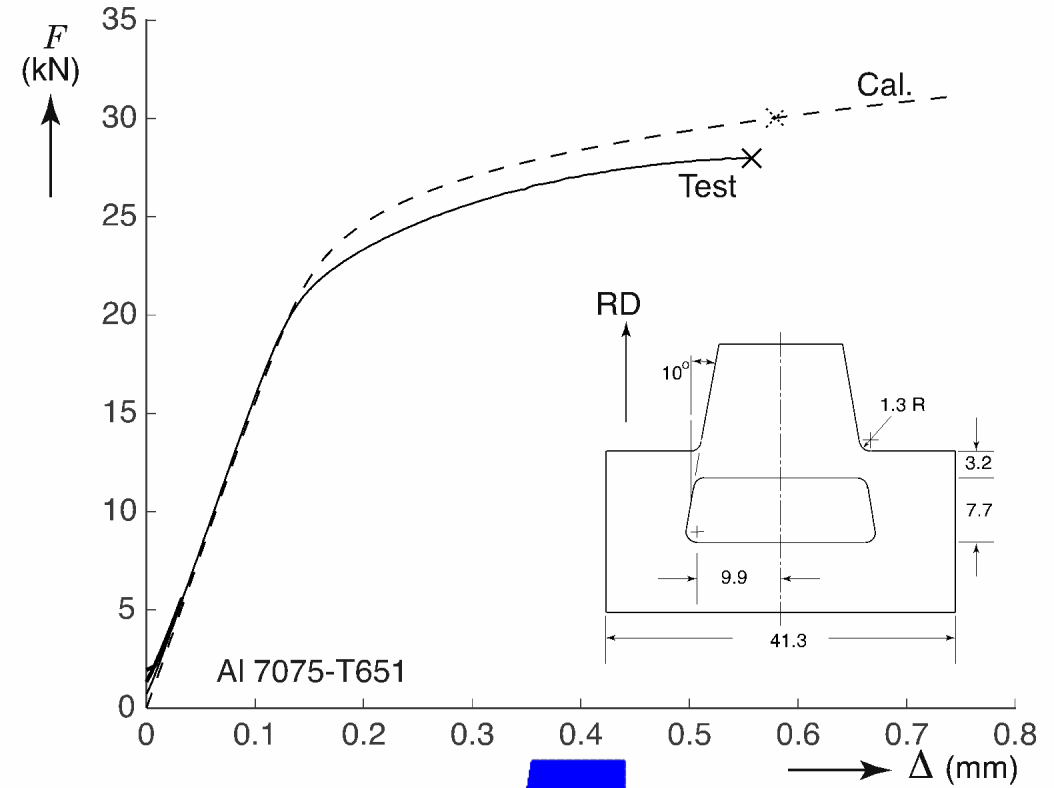
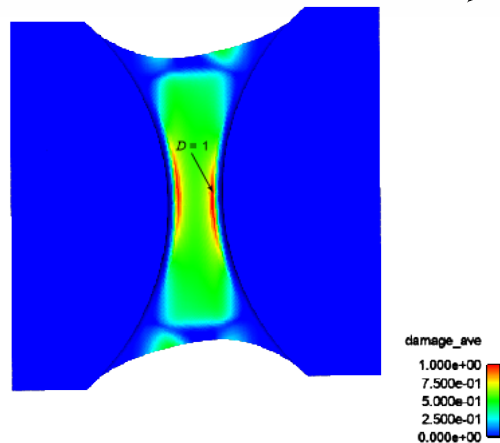
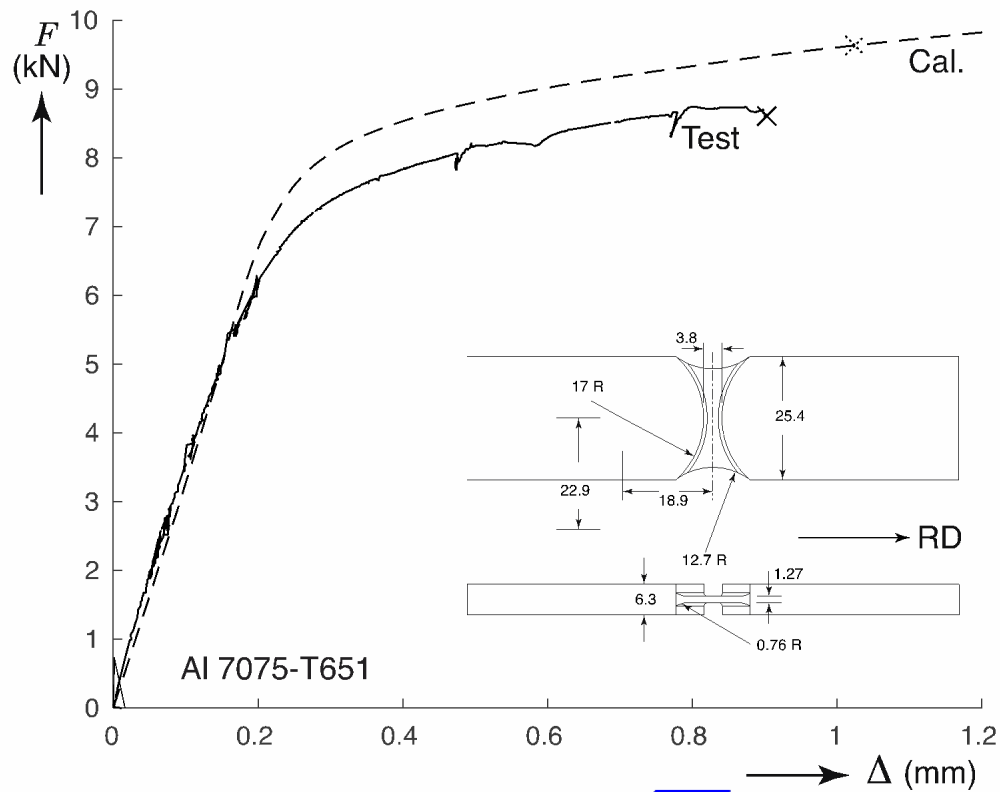
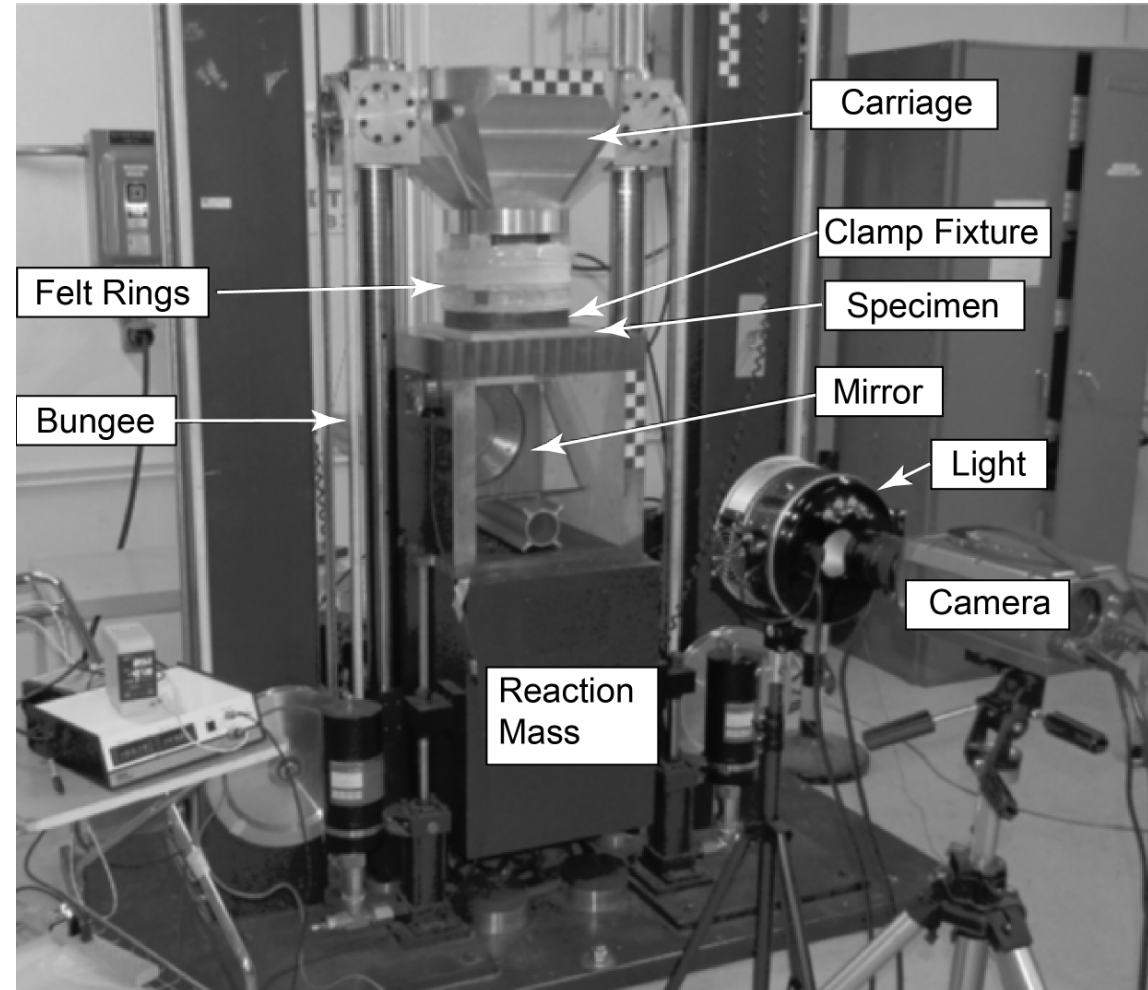
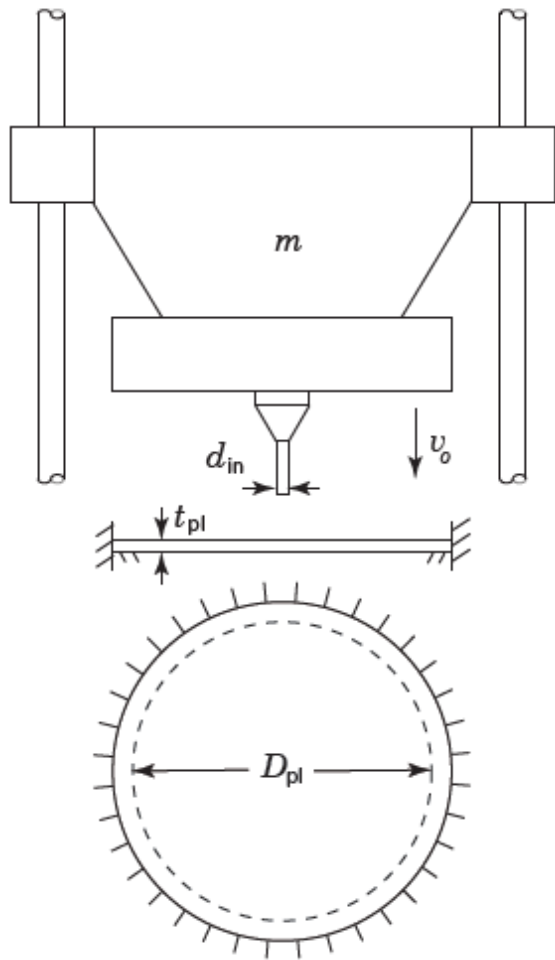




Plate Puncture



Experimental Setup



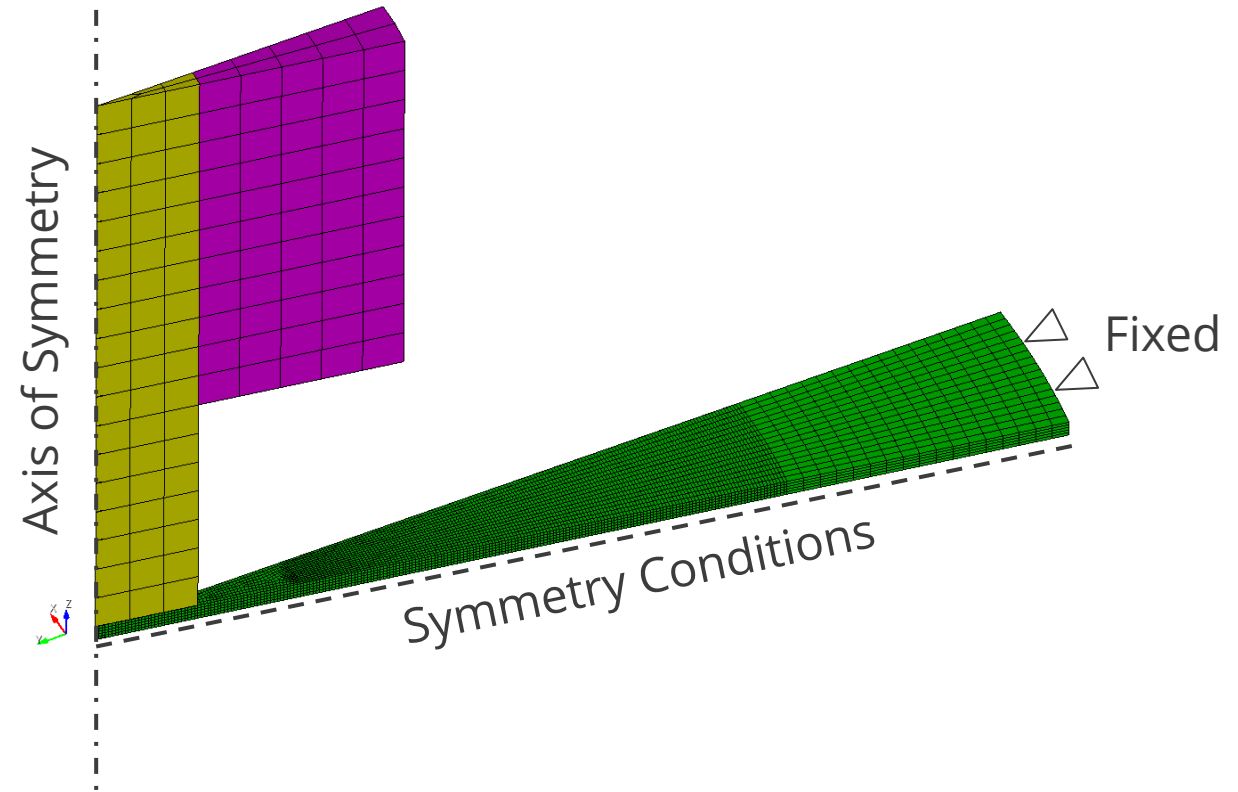
$$D_{pl} =$$
$$m = 1$$

$$t_{pl} = 1.1.6.2.3.2.4.8 \text{ mm. } d_{in} = 25.4 \text{ mm}$$



Finite Element Numerical Modeling

- Explicit, transient-dynamic finite element model.
- Reduced-integration 8-node hexahedral elements
- Frictional hard contact between punch and plate
- Punch is rigid, plate is elastic-plastic.
- Failure propagation modeled via element deletion.
- Simulations conducted with 3, 5 and 10 elements through the plate thickness.
- 180° wedge for 4.8, 3.2 and 2.0 mm thick
- 10° wedge for 2.0, 1.7, 1.0 mm thick

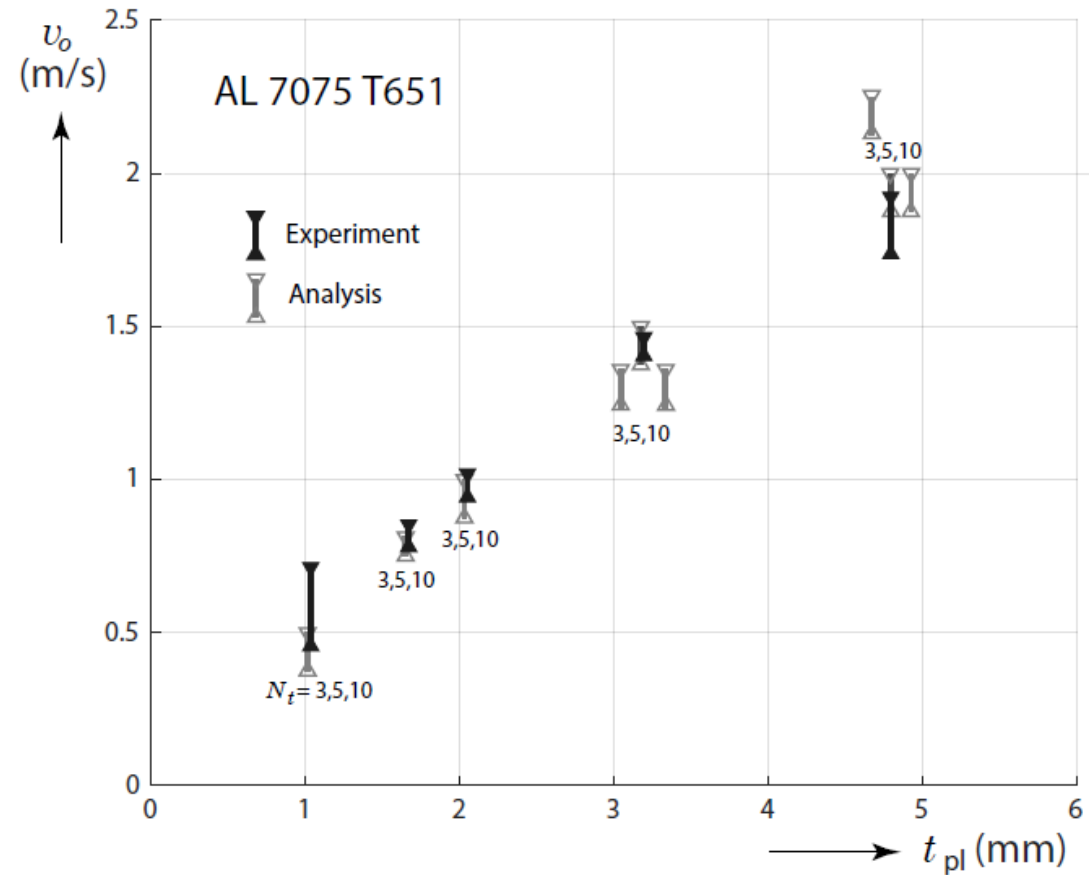




Comparison of Measured and Predicted Puncture Speed Brackets

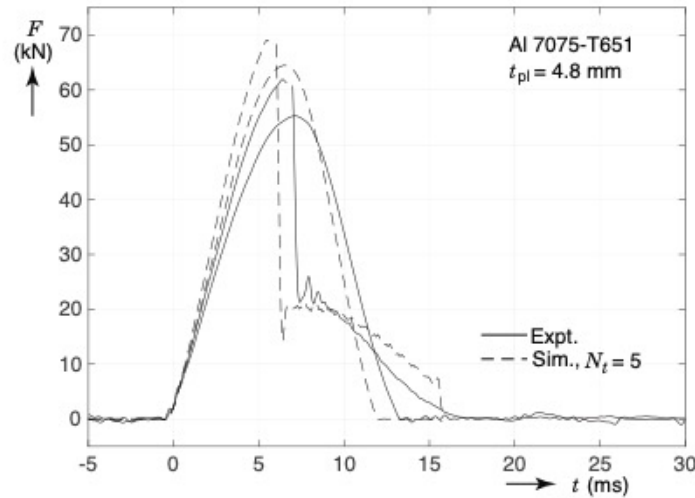
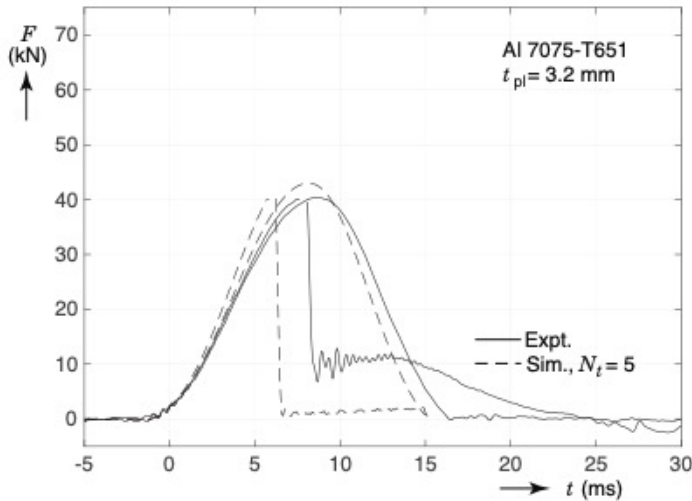
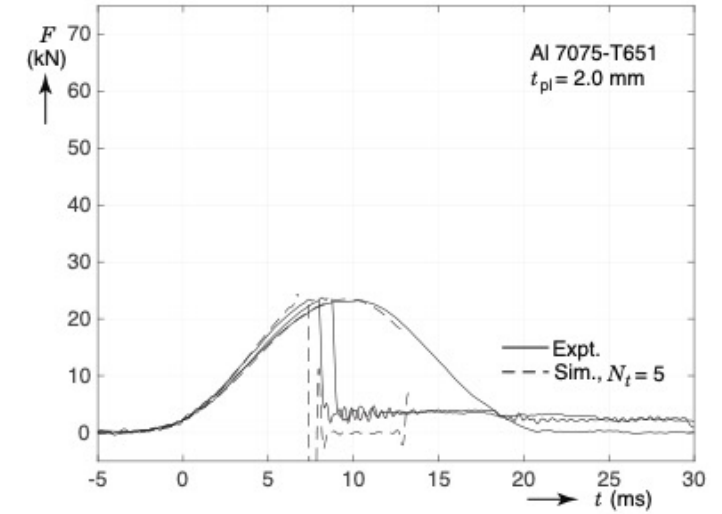
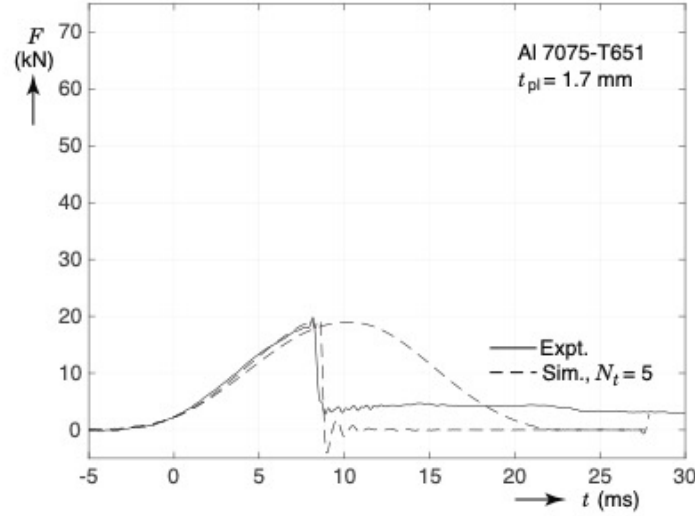
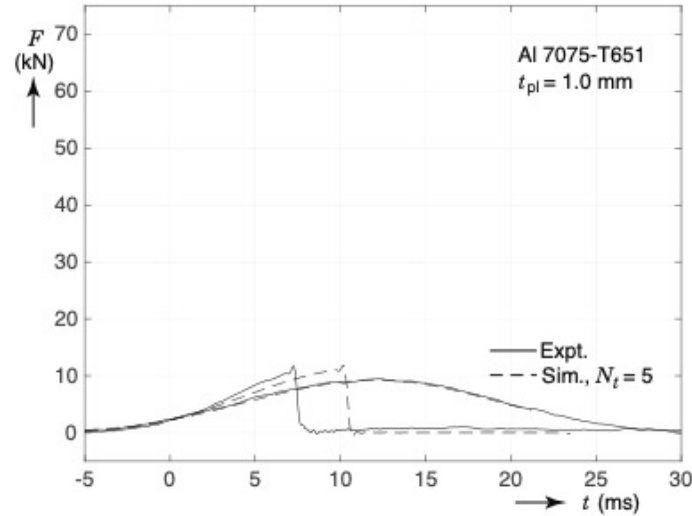
Basic Plastic Properties (% difference wrt 12.7 mm)

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Force-Time History Comparisons



Basic Material Properties

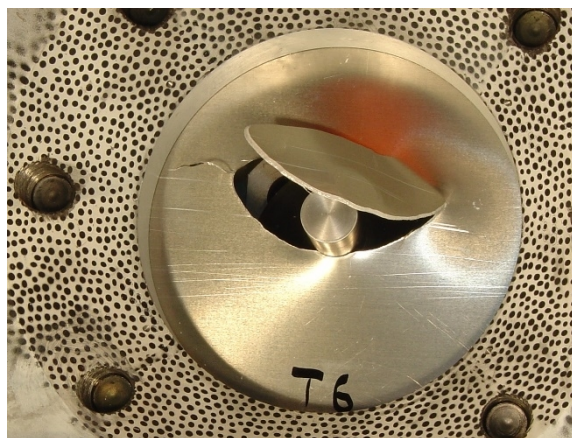
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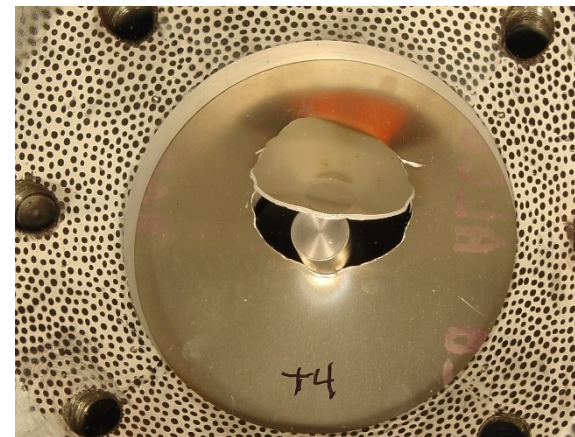
Experimental Failure Configurations



1.0 mm



1.7 mm



2.0 mm



3.2 mm

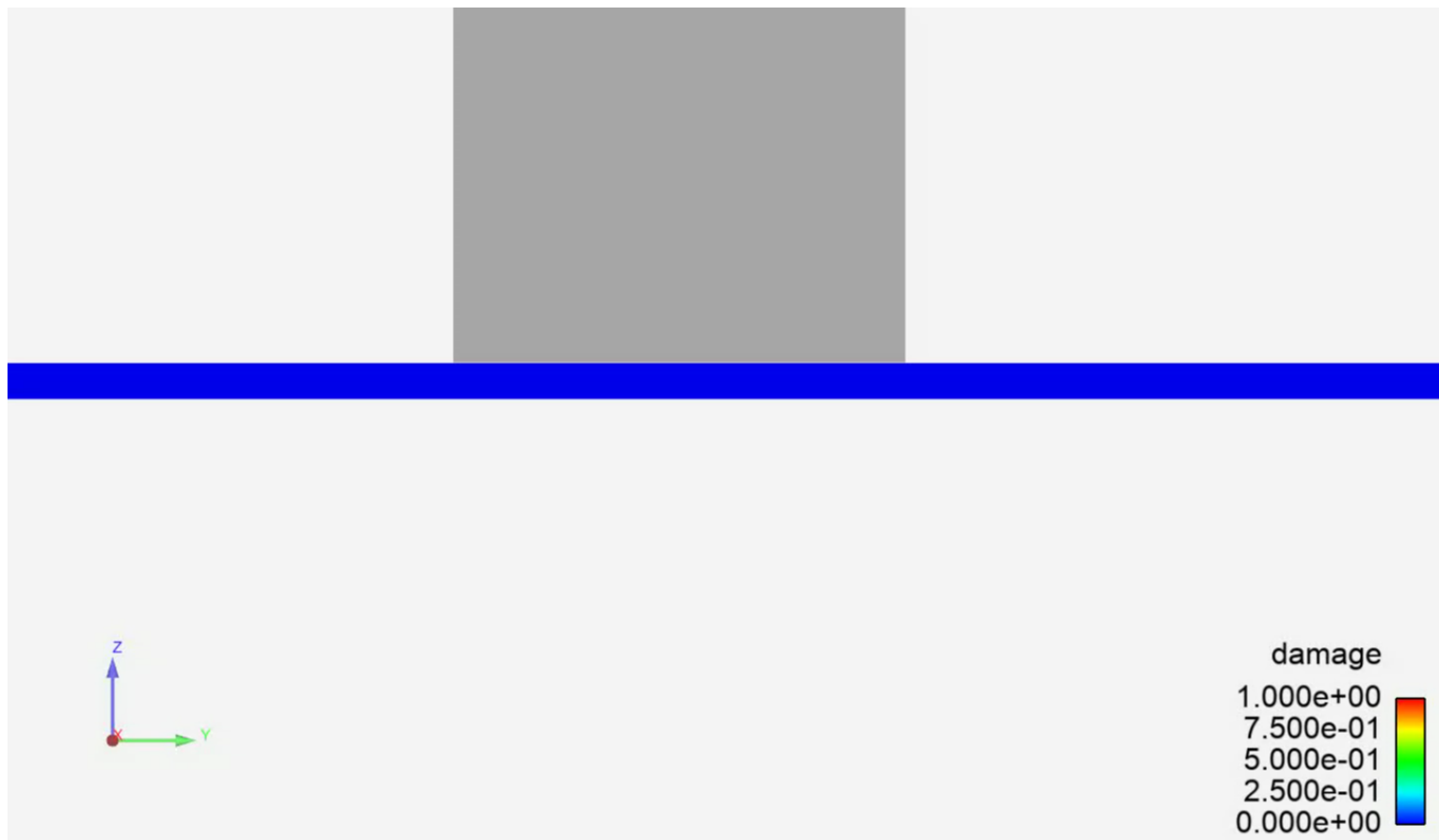


4.8 mm

Failure initiates under edge of punch and propagates outwards.

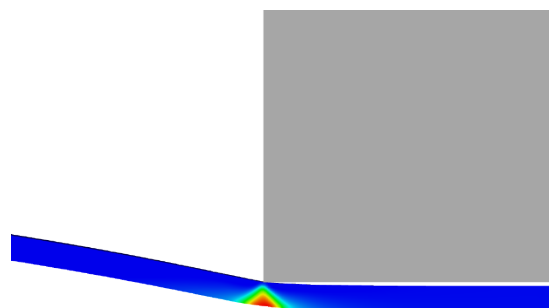


Progression of Damage $t = 2 \text{ mm}$



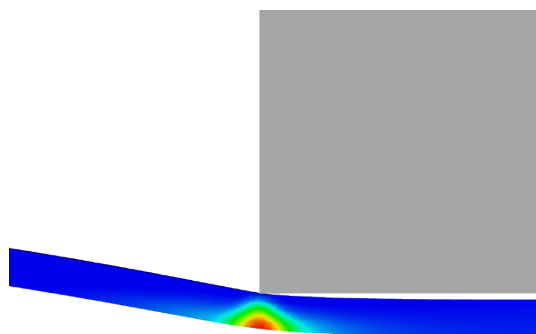


Damage at Failure ($N_t = 5$)



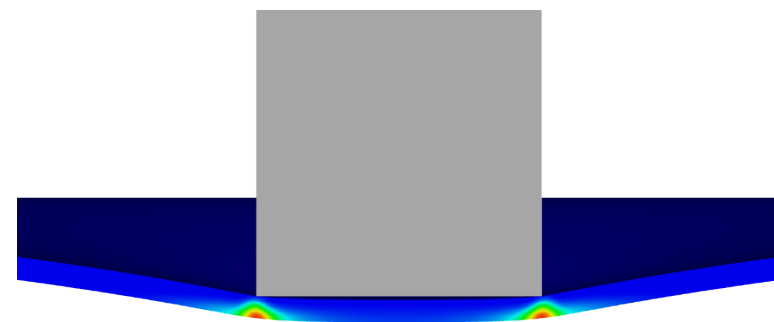
damage
1.000e+00
7.500e-01
5.000e-01
2.500e-01
0.000e+00

1.0 mm



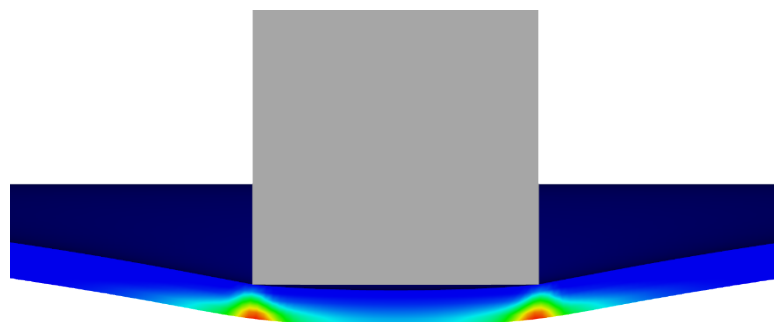
damage
1.000e+00
7.500e-01
5.000e-01
2.500e-01
0.000e+00

1.7 mm



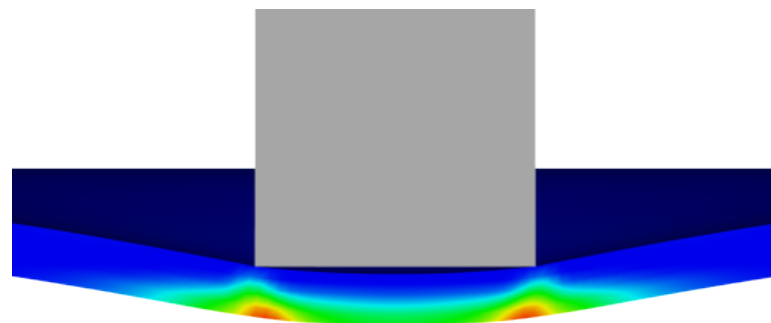
damage
1.000e+00
7.500e-01
5.000e-01
2.500e-01
0.000e+00

2.0 mm



damage
1.000e+00
7.500e-01
5.000e-01
2.500e-01
0.000e+00

3.2 mm



damage
1.00
0.75
0.50
0.25
0.00

$t_{pl} = 4.8$ mm



Conclusions

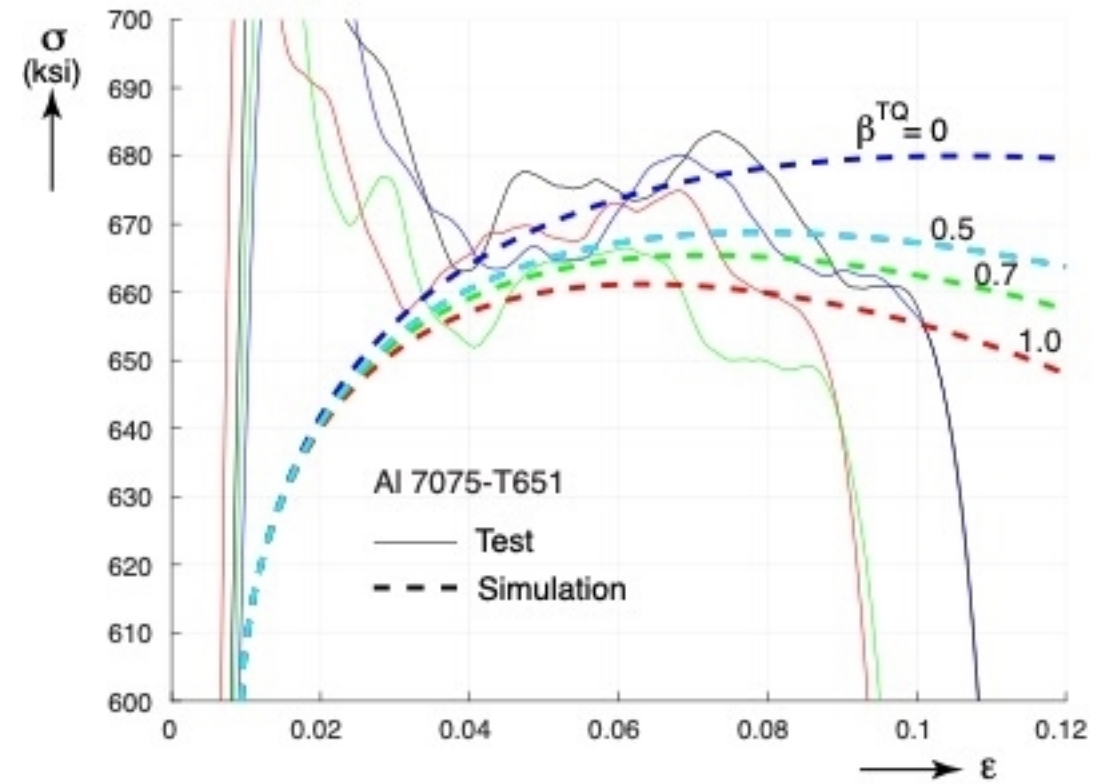
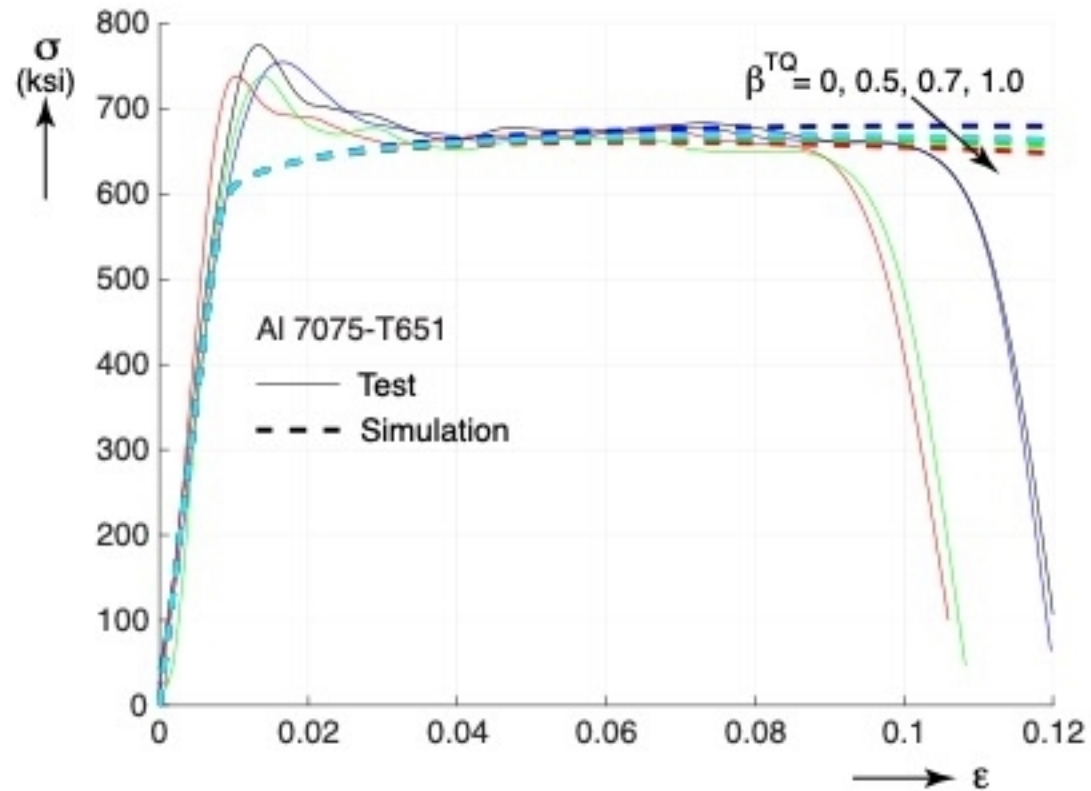
- Thermal-mechanical plasticity and ductile failure model calibrations were made from one lot of 7075 for finite element simulation of Al 7075-T651 plate puncture tests of different lots.
- Predictions for the threshold puncture punch velocity and punch Force-time histories agree reasonably well with experiments
- Not very strong element size sensitivity within the ranges studied
- Calibrated plasticity and ductile failure models were appropriate for predictions on puncture velocity, and Force-time histories but not for predicting final flap shape.

Thank you for your Attention
Questions?





Calibration of β^{TQ}





Failure Model Calibration procedure

INPUT

Experiment

$d_i, i = 1, 6$

Simulations

$\{\bar{\varepsilon}^p, \sigma_1, \sigma_2, \sigma_3\}_i, i = 1, 6$

Calibration Parameters

$\{e_1, \dots, e_N\}_i, i = 1, 6$

B
 $\{\alpha_{\min}, \alpha_{\max}, \Delta\alpha\}$

$\{\beta_{\min}, \beta_{\max}, \Delta\beta\}$

$\{D_{\min}^{cr}, D_{\max}^{cr}, \Delta D\}$

PREREQUISITE CALCULATIONS

$$\sigma_m = \frac{1}{3} \sum_{j=1}^3 \sigma_j$$

$$s_j = \sigma_j - \sigma_m, j = 1, 2, 3$$

$$A = \max\left(\frac{s_2}{s_3}, \frac{s_2}{s_1}\right)$$

TABULATION

for $\alpha = \alpha_{\min}$ to α_{\max} increment $\Delta\alpha$ do:

for $\beta = \beta_{\min}$ to β_{\max} increment $\Delta\beta$ do:

for $D^{cr} = D_{\min}^{cr}$ to D_{\max}^{cr} increment ΔD^{cr} do:

for $i = 1$ to 6 increment 1 do:

for $k = 1$ to N increment 1 do:

$$D_k = \frac{1}{D^{cr}} \int_0^{\bar{\varepsilon}^p} w_1 w_2 d\bar{\varepsilon}^p$$

Find e_i with lowest \bar{d}_k when $D_k = 1$

$$E_i = \bar{d}_i - d_i$$

$$E = \sum_{i=1}^6 E_i^2$$

Tabulate $\{E, \alpha, \beta, D^{cr}, \text{other diagnostics}\}$



Histories at First Failure Location

