

# RAPID DESIGN OF COST-EFFECTIVE REFRACTORY HIGH ENTROPY ALLOYS STRENGTHENED BY PRECIPITATION

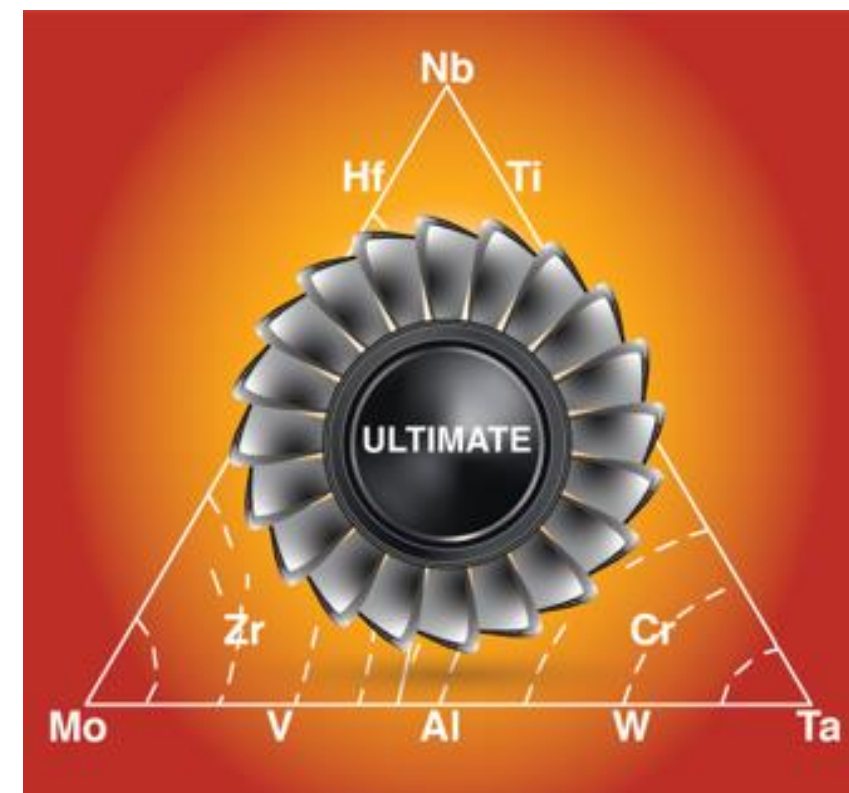
**August 20, 2024**

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# Project Overview: Target Properties

Phase I	
Properties	Project target
Creep	Creep strain <2% at 200 MPa after 100 hours
Ductility	RT elongation >1.5%
Fracture toughness	>10 MPa·m <sup>1/2</sup> at RT
Manufacturability	with <0.1mm in variation among five tensile bars manufactured specified to the appropriate ASTM E8/E8M standard.
Phase II	
Yield stress at 1300°C	>400 MPa
Solidus Temperature	≥1500°C
Density	≤9.0 g/cm <sup>3</sup>
Thermal conductivity	9-12 W/mK at RT; >24 W/mK at 1300°C;
CTE from RT to 1300°C	<2%
Thermo-mechanical fatigue	>1000 cycles at 0.45% strain and R=-1 between 100-1300°C
Creep	Coated samples retain creep strength (at 1300°C under 200MPa for 100 hours) after exposure to air at 1700°C for 100 hours
Manufacturability	A generic turbine with dimensions between 3 and 6 inches with internal cool channels with critical dimensional variation < 1% among three samples.

# Technical Approach

- Computationally interrogate key parameters
- Specify compositional **range** for alloy(s)

## Thermo-physical

- liquidus and solidus temperatures
- phase composition and mole fraction
- density
- materials cost

CALPHAD

## Intrinsic ductility

- shear instability
- VEC
- $\Delta E_B$
- Phase transformation
- Grain boundary cohesion

DFT, ML

## Yield strength

- solid solution
- precipitation
- grain refinement
- short range order
- temperature effect

MCMD, DFT,  
validated models

## Oxidation resistance

- oxygen diffusivity
- oxygen solubility
- composition gradient to promote forming  $\square$ -Al<sub>2</sub>O<sub>3</sub> scale

AIMD, DFT,  
CALPHAD, FEM

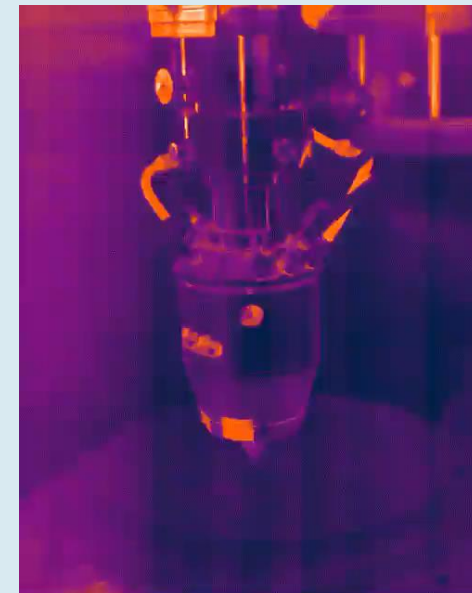
## Manufacturing ability

- minimize cracking
- reduce solidification range
- suppress ductile to brittle transition

Validated models,  
CALPHAD



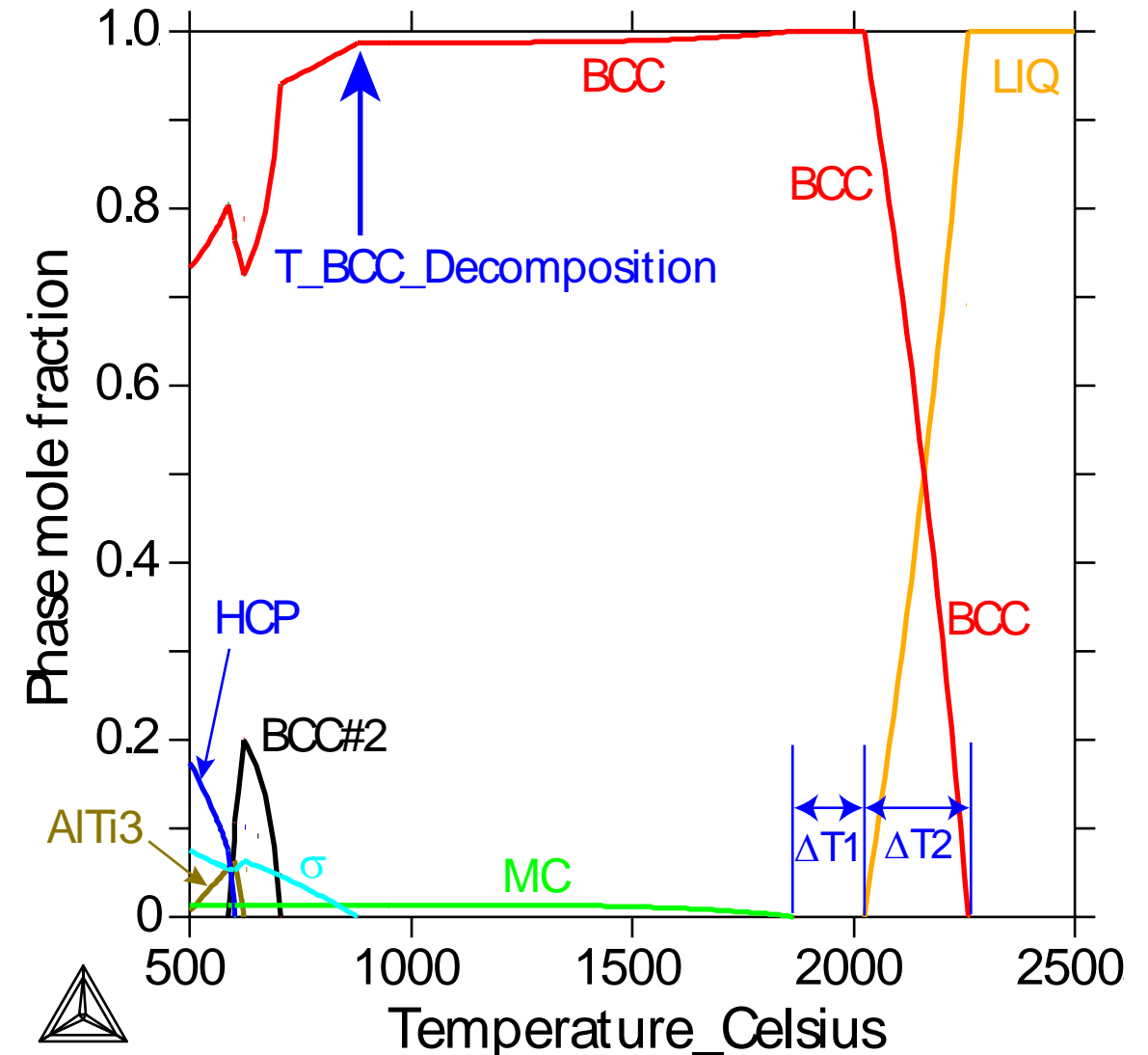
- Arc melted buttons (250g)
- Structure & property assessment
- **Simulate AM build.**  
*Single & multi-pass beam scanning of the surface to assess effect of solidification rate on cracking, segregation and precipitate formation*



- Direct Energy Deposition DED-AM
- Design of Experiments to **interrogate processing parameters** (laser power, powder flow & velocity) on the build.

# Technical Approach

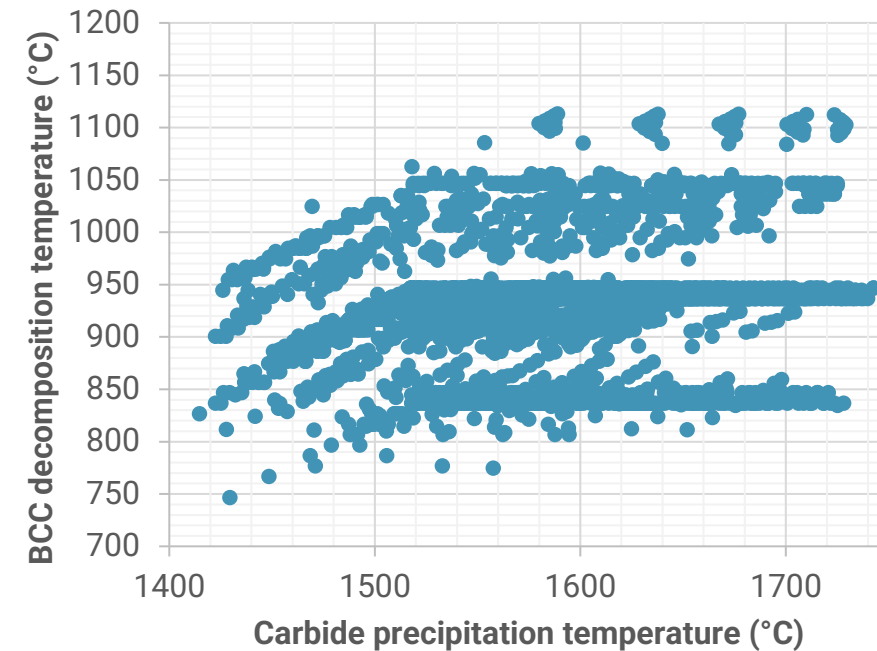
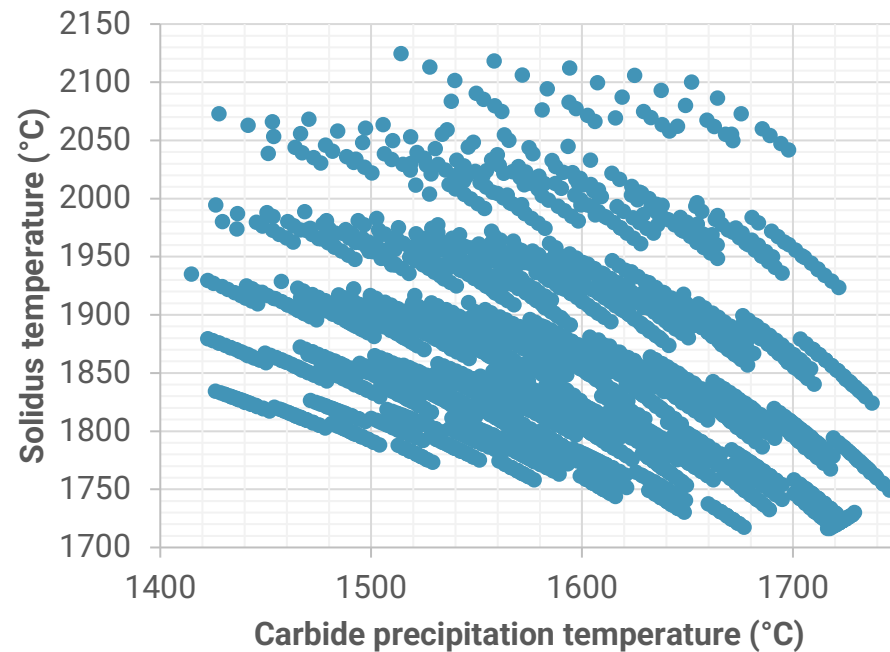
Element	Function
Nb, V, Ti	Provide for low density and high ductility
Nb, Ta, Zr, Hf, Ti, C	MC carbide formers for precipitation strengthening at elevated temperatures
Ta, Mo, V, W	Provide for solid solution strengthening. Increase melting temperature. BCC stabilizers
Ti, Zr, Hf	Increase intrinsic alloy ductility & solid solution strengthening Promote toughness through phase transformation induced ductility
Al, Cr	Increase oxidation resistance Enhance bulk alloy environmental resistance and compatibility with environmental barrier systems.





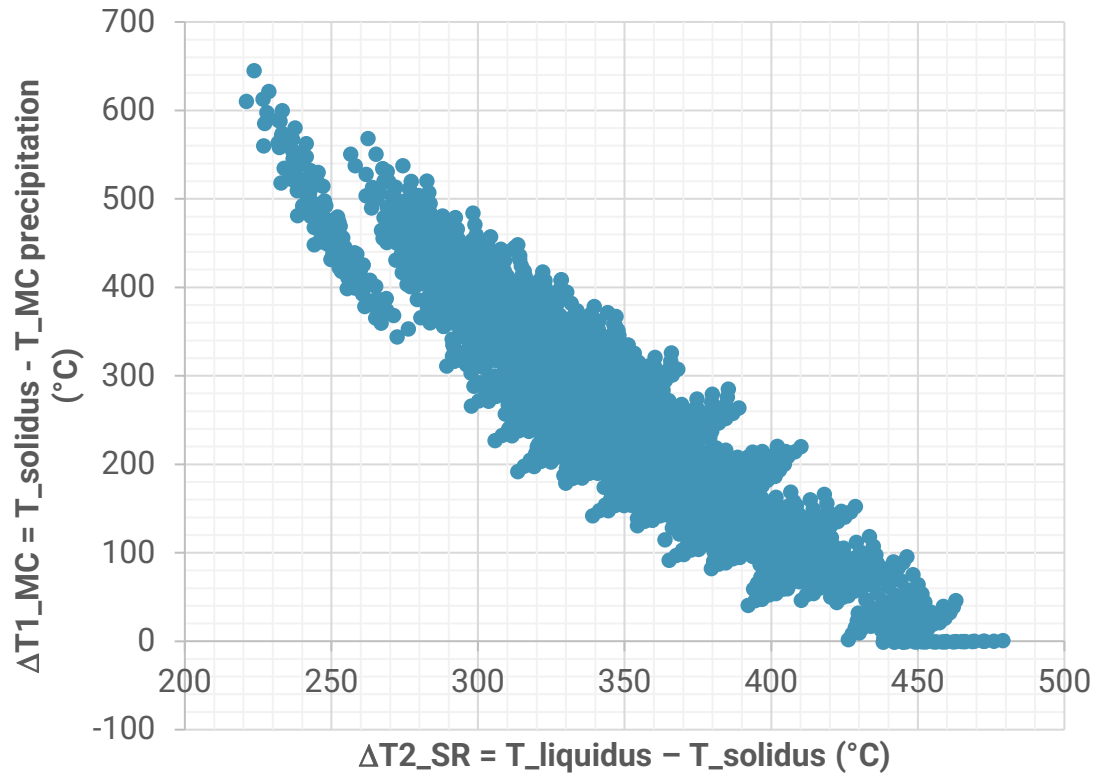
# Alloy Design: High Throughput CALPHAD

- ▶ Alloys: Nb-Ti-Ta-Mo-V-Zr-Hf-Cr-W-Al-C system
- ▶ HT CALPHAD: ThermoCalc TCNI8 database
  - Screening criteria:  $T_{\text{solidus}} \geq 1800^{\circ}\text{C}$ ,  $T_{\text{MC precipitation}} \geq 1400^{\circ}\text{C}$ ,  $T_{\text{decomposition}} \leq 1000^{\circ}\text{C}$ , density  $\leq 9.5 \text{ g/cm}^3$ , price  $\leq \$115/\text{kg}$  (C103).

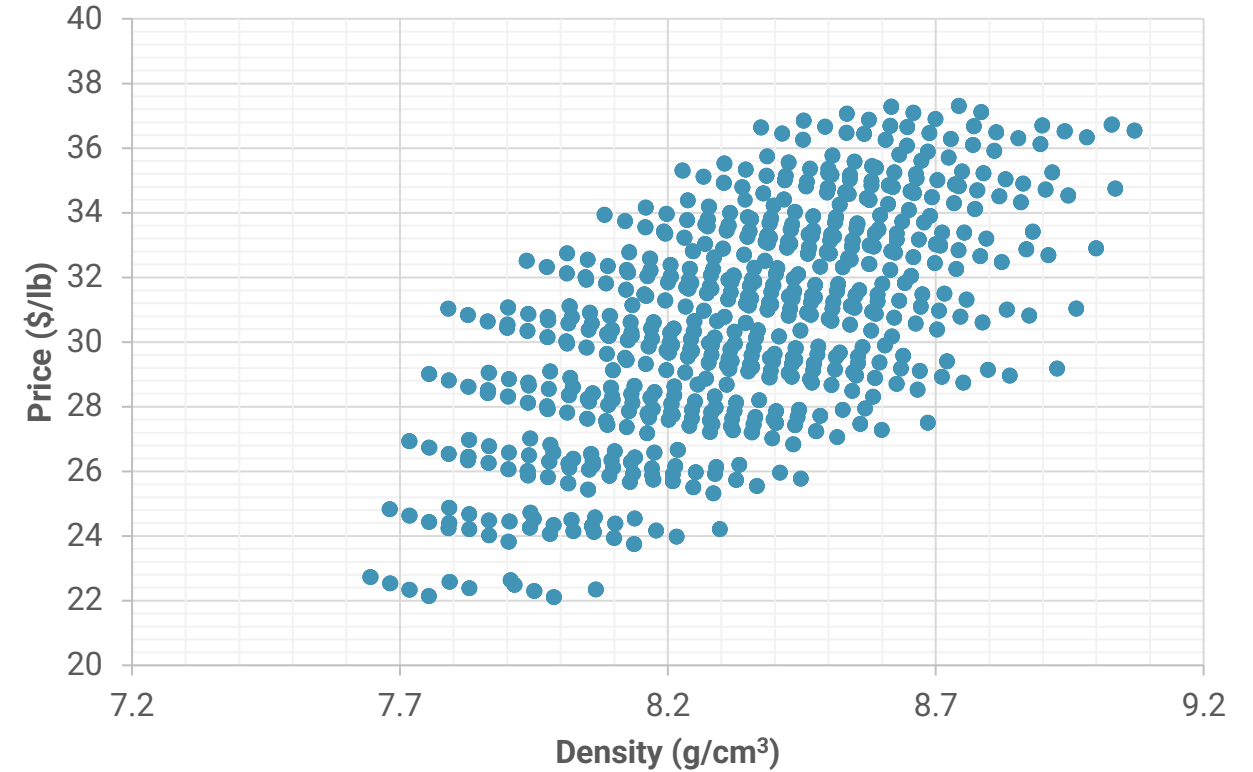


# Alloy Design Details: HT CALPHAD

- ▶  $\Delta T_2 = T_{\text{liquidus}} - T_{\text{solidus}}$ ;  $\Delta T_1 = T_{\text{solidus}} - T_{\text{MC precipitation}}$

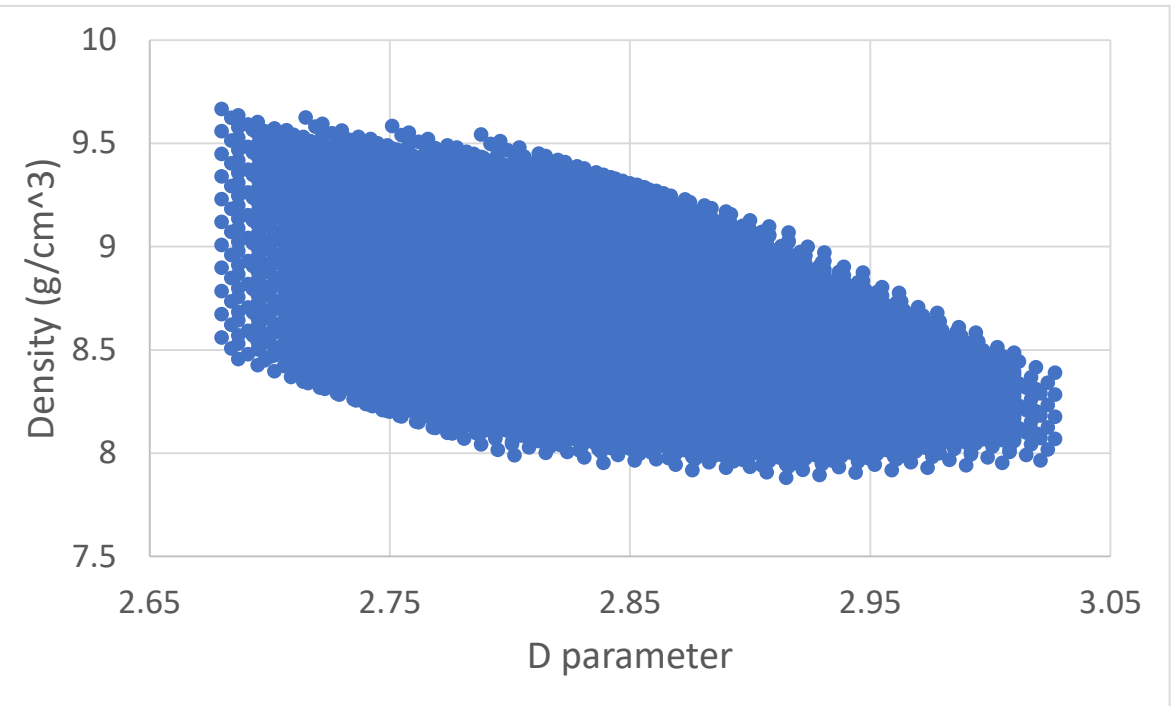
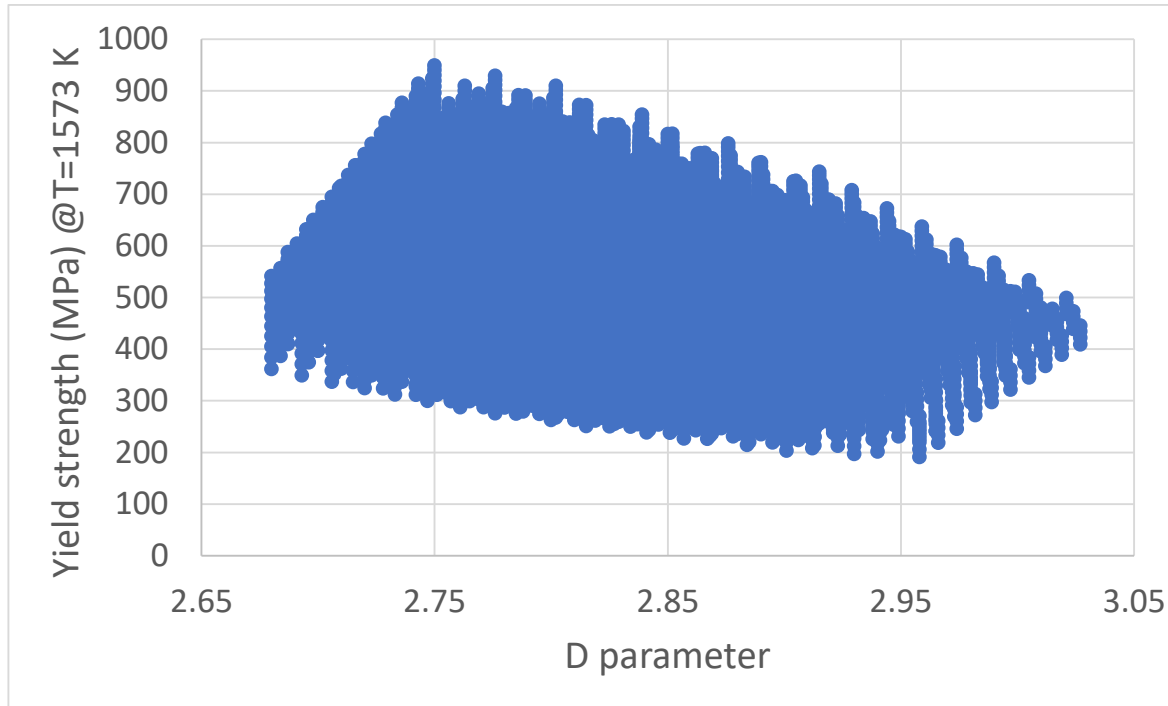


Solidification range vs BCC stability range

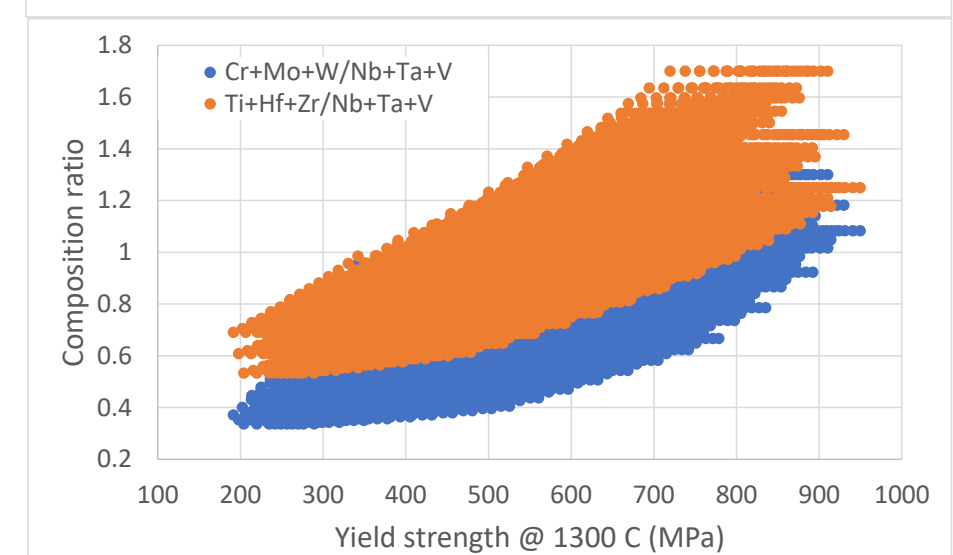
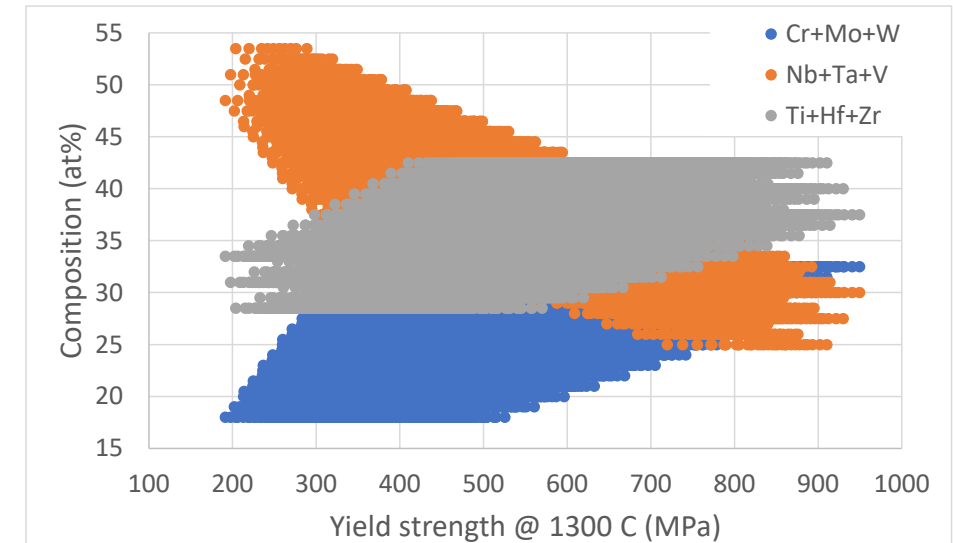
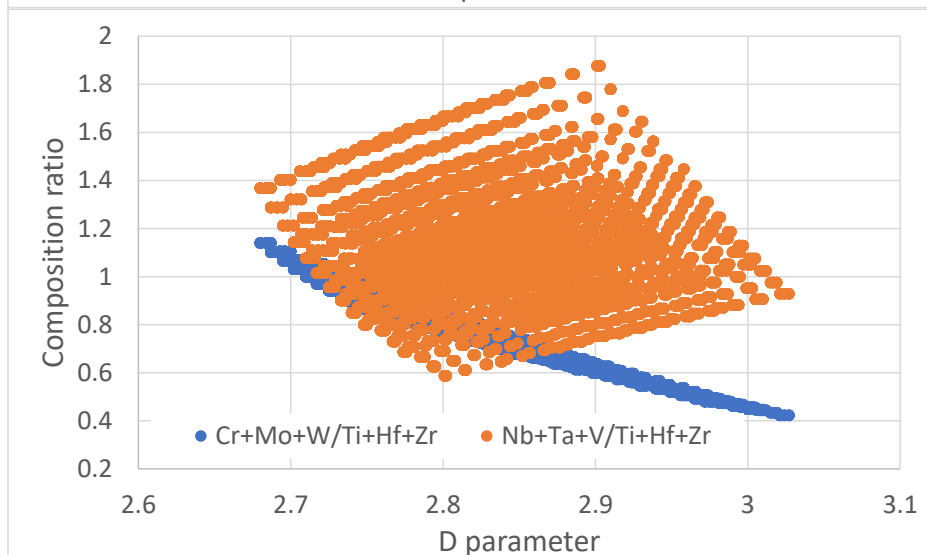
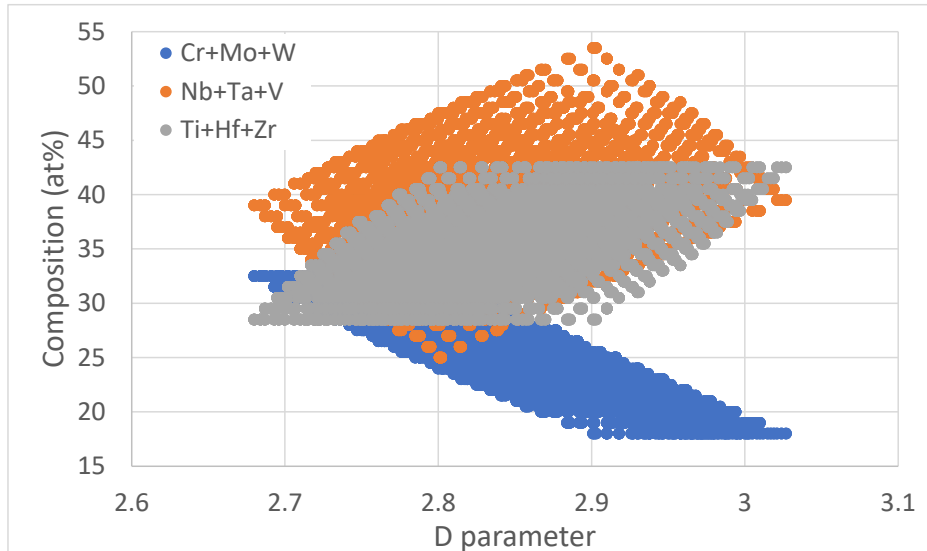


Materials cost vs density

# High Throughput Materials Design



# High Throughput Materials Design



$$F(V, T) = E_c(V) + F_{vib}(V, T) + F_{el}(V, T)$$

$$F_{vib}(V, T) = \frac{9}{8} k_B \Theta_D(V) + k_B T \left\{ 3 \ln \left[ 1 - \exp \left( -\frac{\Theta_D(V)}{T} \right) \right] - D \left( \frac{\Theta_D(V)}{T} \right) \right\}$$

- ▶  $V$  - volume
- ▶  $T$  - temperature
- ▶  $V_{eq}$  - can be obtained by solving  $V$  from  $\left( \frac{\partial F}{\partial V} \right)_T = 0$
- ▶  $E_c$  - 0 K static total energy
- ▶  $F_{vib}$  - vibrational contribution
- ▶  $F_{el}$  - thermal electronic contribution

Moruzzi et al., PRB (1988); Wang et al., IJQC 96 (2004).



# Existing Approaches to Calculate Debye Temperature from Energy-Volume Curve

- Debye temperature by Moruzzi model

$$\Theta_D = \Theta_0 \left( \frac{V_0}{V} \right)^\gamma \text{ with } \Theta_0 = 67.48 \left( \frac{r_0 B_0}{M} \right) \text{ and } V_0 = \frac{4\pi}{3} r_0^3$$

- Grüneisen parameter

$$\gamma = \frac{1}{2} (1 + B'_0) - \lambda$$

$$\lambda = \begin{cases} 2/3, & \text{Slater expression for high temperature} \\ 1, & \text{Dugdale – MacDonald for low temperature} \\ 4/3, & \text{Vashchenko – Zubarev due to free – volume theory} \end{cases}$$

Moruzzi et al., PRB (1988); Wang et al., Int. J. Quantum Chem. 96(2004);  
Slater, J. C. Introduction to Chemical Physics; McGraw-Hill: New York, 1939;  
Dugdale & MacDonald, Phys. Rev. 89(1953);  
Vashchenko & Zubarev Sov Phys. Solid. State 5(1963).

# Solution for Thermodynamic Calculations for HEAs with Arbitrary Compositions

- In this work, we propose:

- $\gamma(V) = \gamma_0 \left( \frac{V}{V_0} \right)^\delta$

- For reference pure elements/species:

- Calibrate the parameters to reproduce the experimental CTE, heat capacity, and entropy

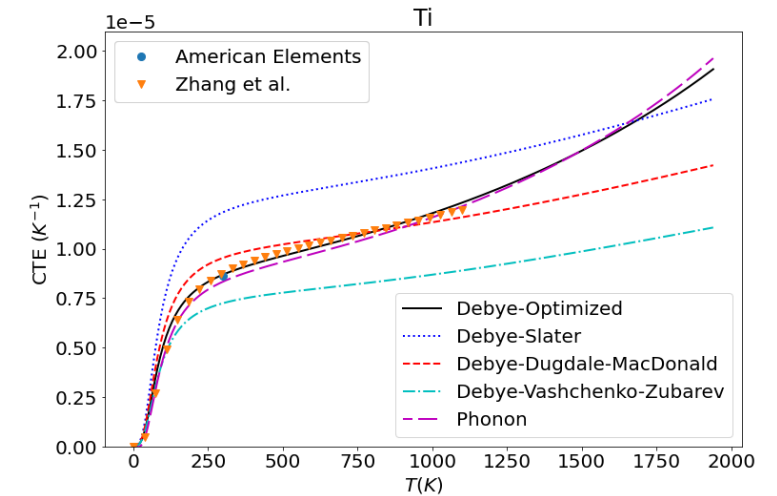
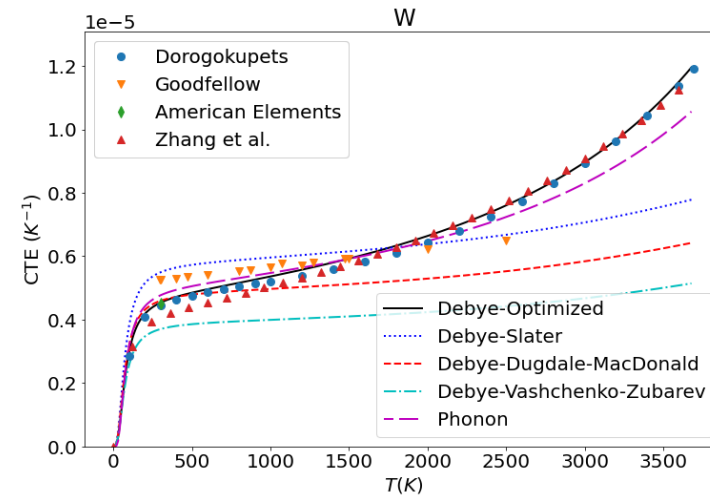
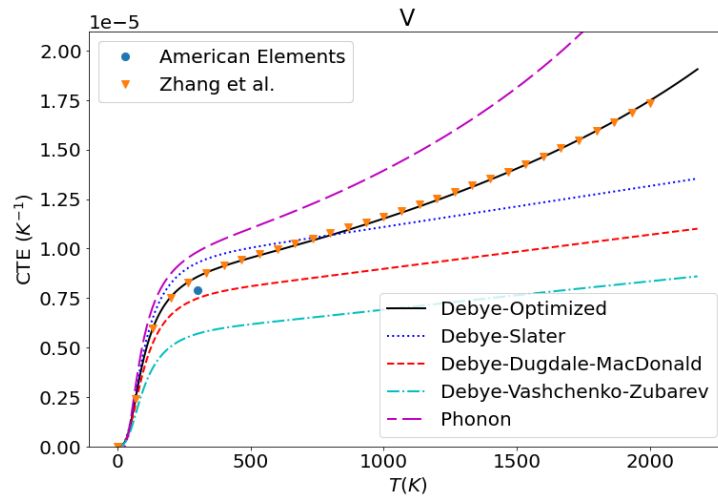
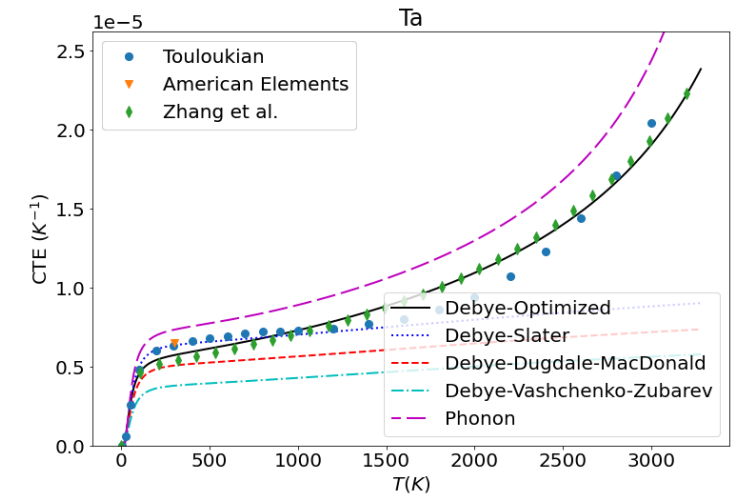
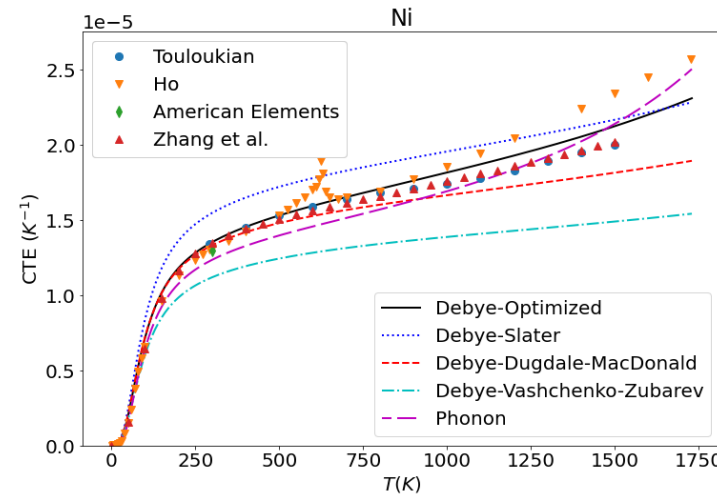
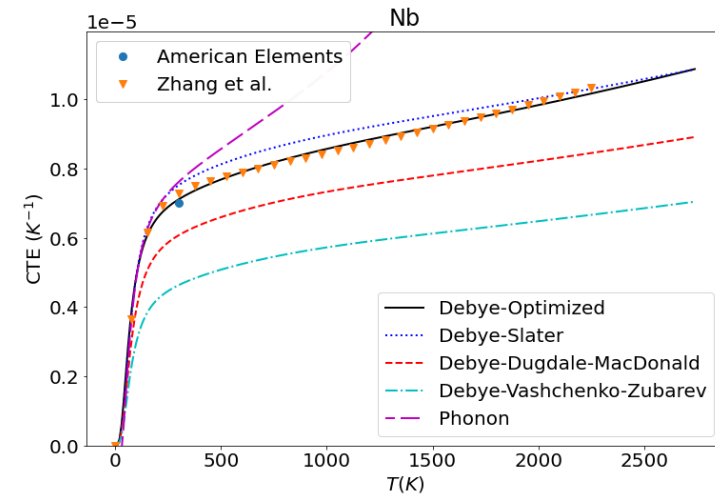
- For compounds with arbitrary composition:

- The parameters are determined by arithmetic/geometric average over pure elements/species by composition

- Implemented in DFTTK (density functional theory toolkit) package:

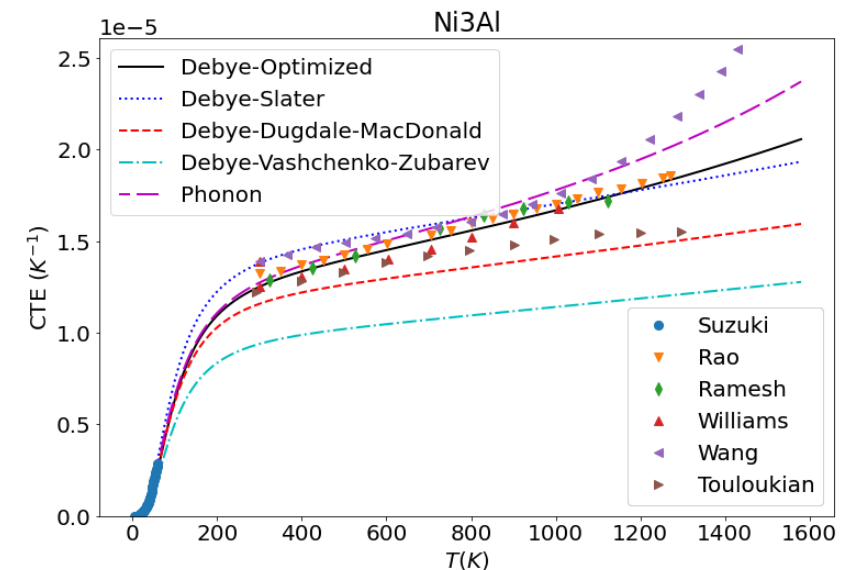
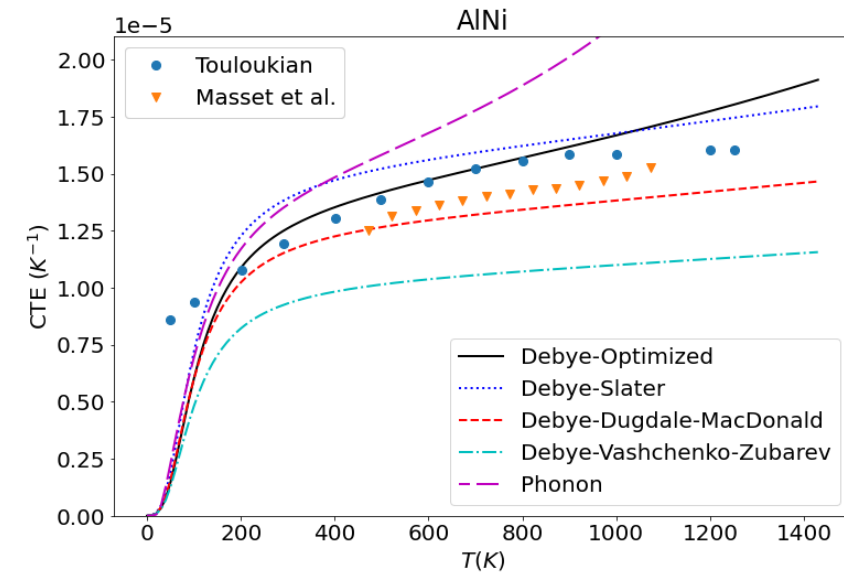
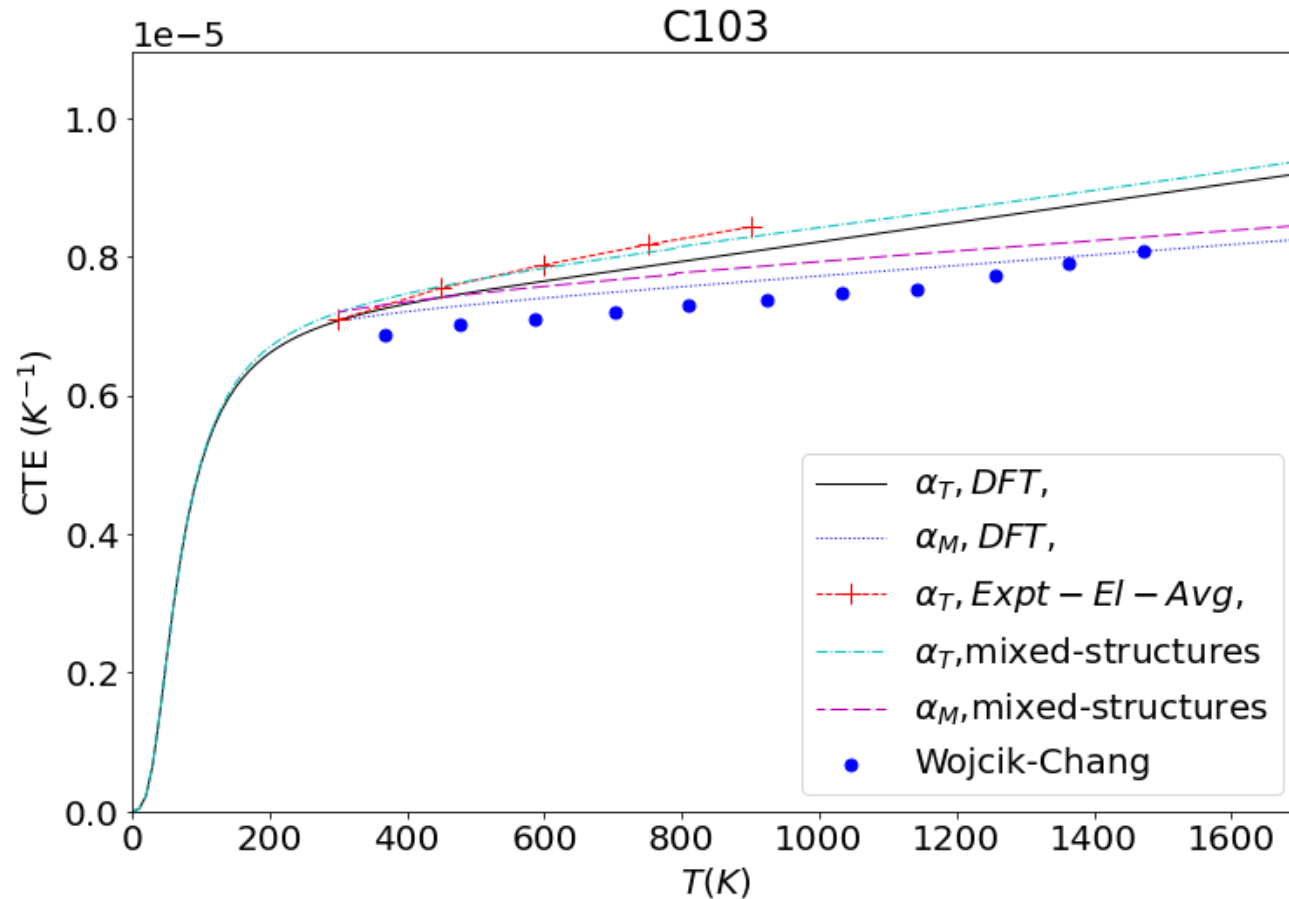
- [Welcome to dfttk's documentation! – dfttk 0.3.4 documentation](#)

# CTEs for Pure Elements



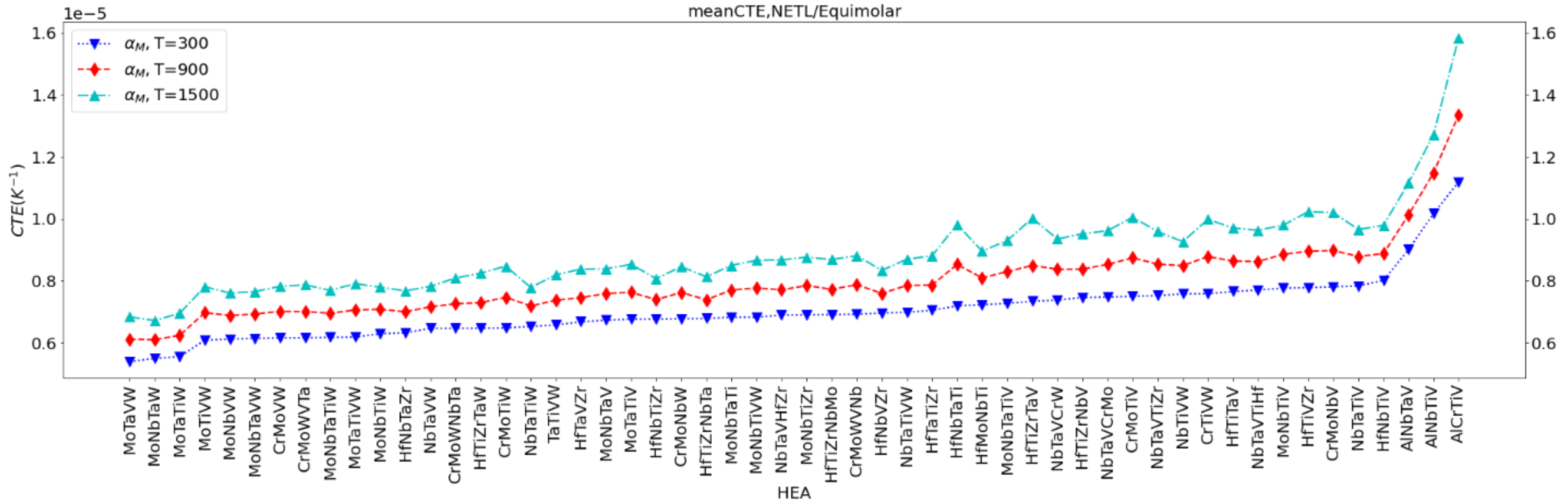
**Lines:** theoretical; **Other symbols:** experimental or critical evaluations

# Predictions for C103, NiAl and Ni<sub>3</sub>Al



**Lines:** theoretical results  
**Symbols:** experimental data

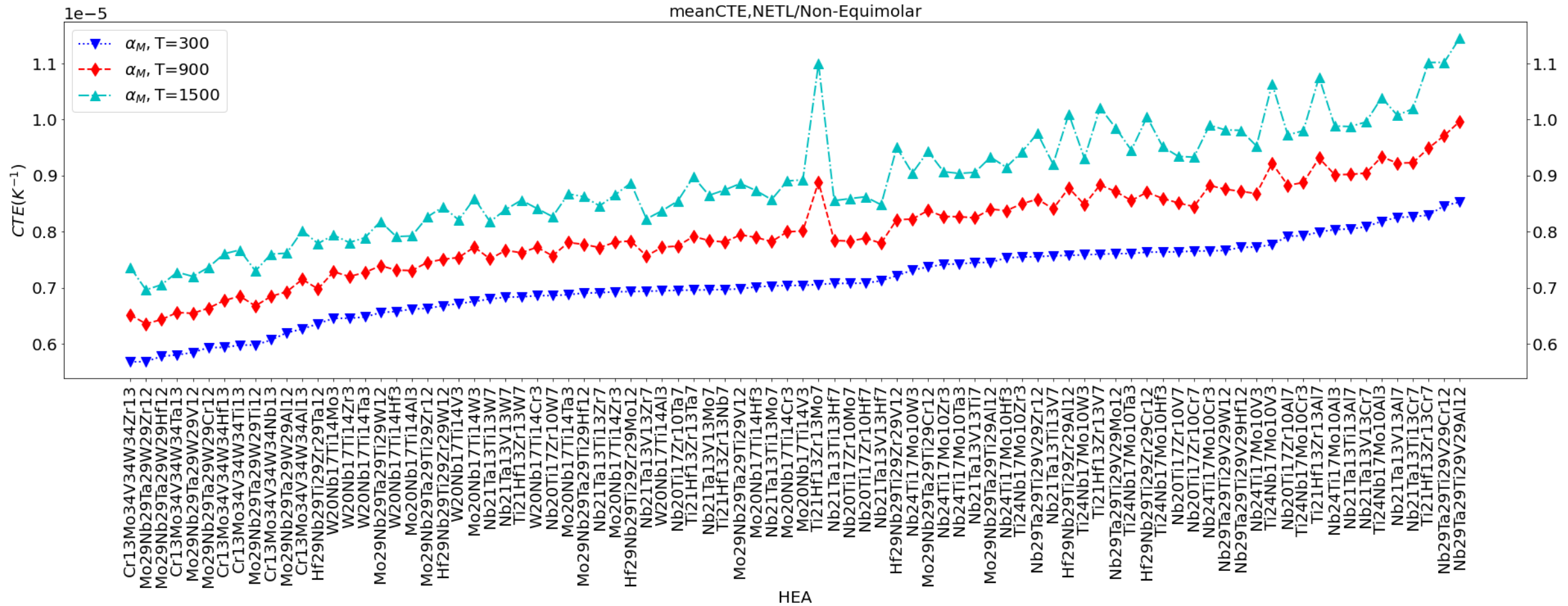
# Predictions for equimolar RHEAs



Predicted mean CTE ( $10^{-6} K^{-1}$ ) for equal molar system at the temperature 300, 900, and 1500 K.

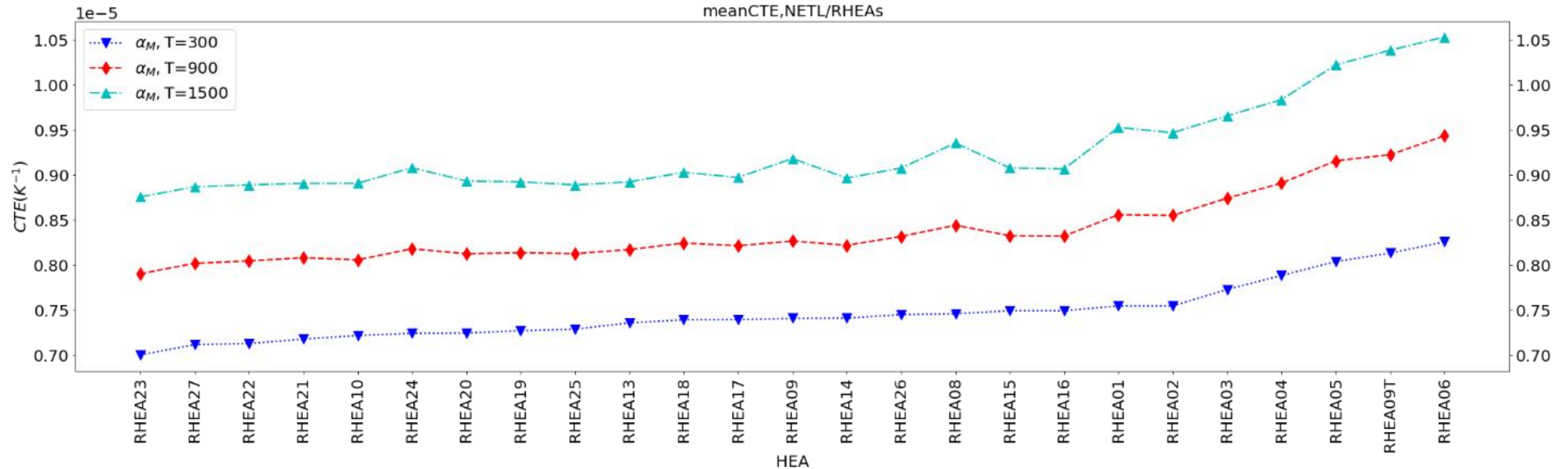


# Predictions for non-equimolar RHEAs



Predicted mean CTE ( $10^{-6} \text{ K}^{-1}$ ) for non-equimolar system at the temperature 300, 900, and 1500 K.

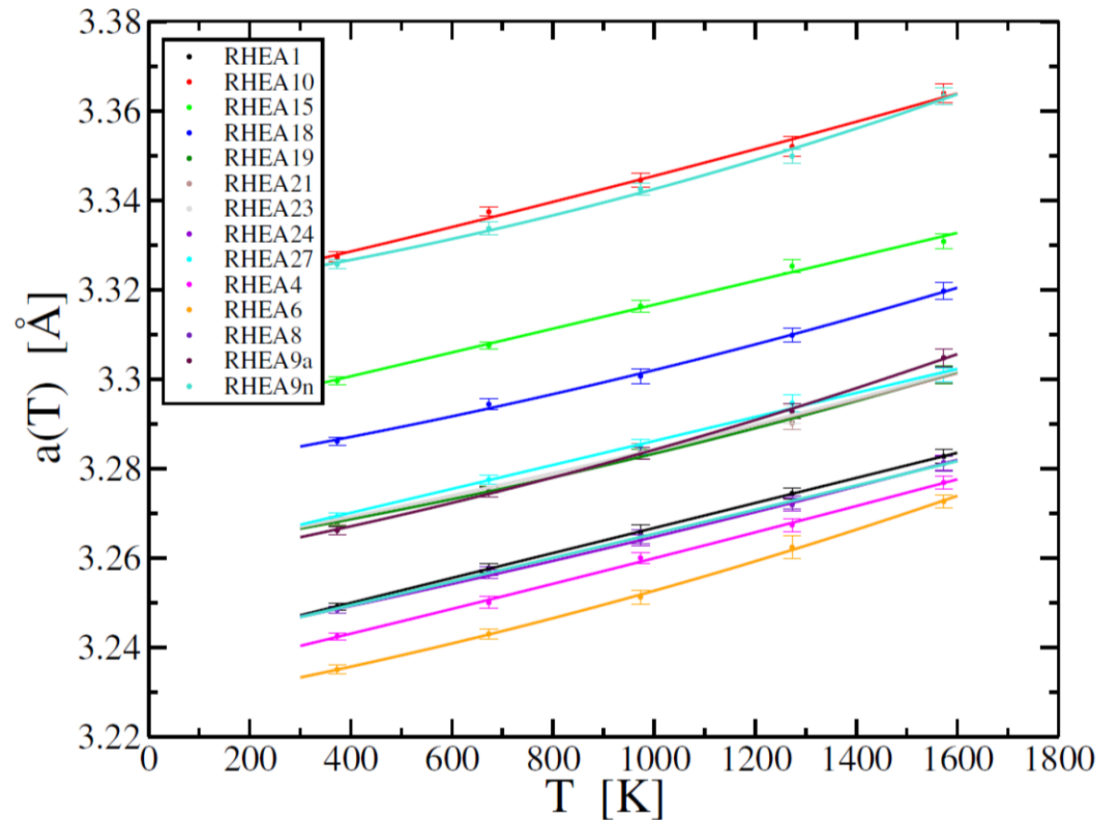
# Predictions for NETL RHEAs



Predicted mean CTE ( $10^{-6} \text{ K}^{-1}$ ) for new RHEAs system at the temperature 300, 900, and 1500 K.

# Coefficient of Thermal Expansion (CTE)

Ab initio molecular dynamics (AIMD) simulations



\*NET (%) is the net expansion from RT-1200C.

RHEA ALLOY	NET (%)*	$\alpha$ ( $10^{-6}$ )
1	1.10	8.6
4	1.12	9.1
6	1.22	11.6
8	1.05	8.3
9	1.14	11.5
10	1.12	11.6
15	1.03	9.4
18	1.05	8.1
19	1.04	9.9
23	1.12	9.0
23	1.04	9.4
24	1.06	9.1
27	1.05	8.2

- ▶ Our approach is to sum Peierls stress and solid solution strengthening for single BCC phase
  - $\sigma = \tau_p + \sigma_y (T, \dot{\epsilon})$
- ▶ As a simple approximation, the Peierls stress is calculated using the following analytical formula (Joos et al., Phys. Rev. Lett. 78(1997)2)
  - $\tau_p = G_{\text{alloy}} \frac{2\pi}{1-\nu_{\text{alloy}}} \exp\left(-\frac{2\pi}{1-\nu_{\text{alloy}}} \frac{h}{b}\right)$
- ▶ The solid solution strengthening is calculated by the theoretical model developed by Maresca and Curtin (Acta Mater. 182(2020)235-249). The main equations for the model are for the calculations of zero temperature yield stress  $\sigma_{y0}$ , the energy barrier for dislocation motion  $E_b$  and the yield stress at finite temperature  $\sigma_y(T, \dot{\epsilon})$ .

$\nu_{\text{alloy}}$ =Poisson's ration

$G_{\text{alloy}}$ =the shear modulus

$$\sigma_{y0} = 0.0915 G_{\text{alloy}} \left( \frac{1+\nu_{\text{alloy}}}{1-\nu_{\text{alloy}}} \right)^{4/3} \left[ \frac{\sum_i c_i \Delta V_i^2}{b^6} \right]^{2/3}$$

$\sigma_{y0}$  = yield stress at T=0K

$$E_b = 0.874 G_{\text{alloy}} b^3 \left( \frac{1+\nu_{\text{alloy}}}{1-\nu_{\text{alloy}}} \right)^{2/3} \left[ \frac{\sum_i c_i \Delta V_i^2}{b^6} \right]^{1/3}$$

$\Delta V$  = volume misfit

$E_b$ =the energy barrier for dislocation motion

$c_i$  = the alloy concentration for element  $i$

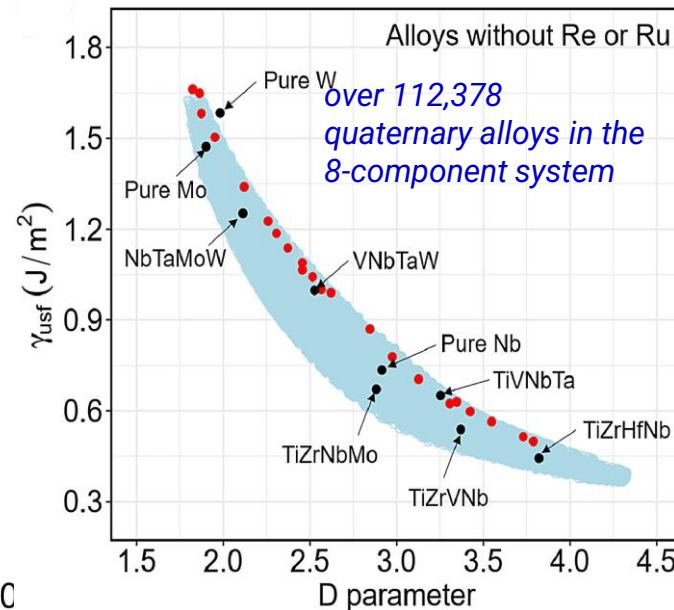
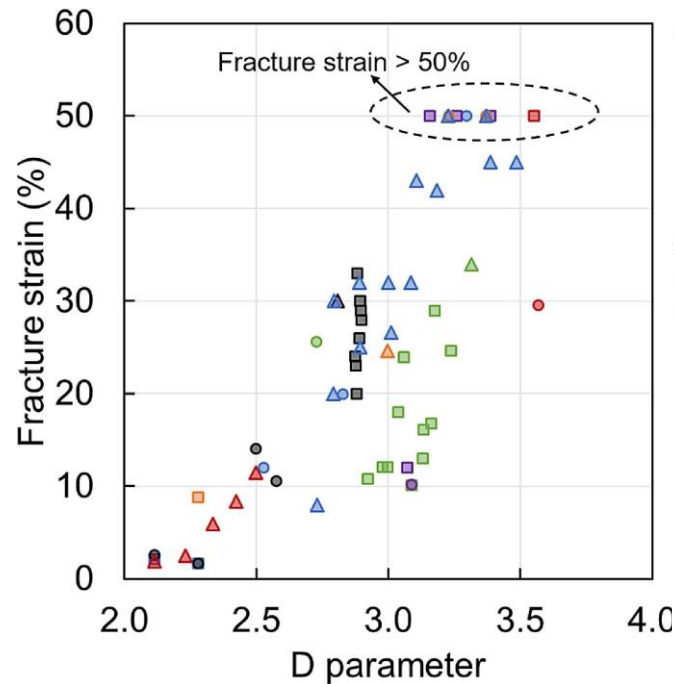
$$\sigma_y(T, \dot{\epsilon}) = \sigma_{y0} \left[ 1 - \left\{ \frac{k_B T}{\Delta E_b} \ln\left(\frac{\dot{\epsilon}_0}{\dot{\epsilon}}\right) \right\}^{2/3} \right]$$

$\sigma_y(T, \dot{\epsilon})$  =yield stress at finite temperature

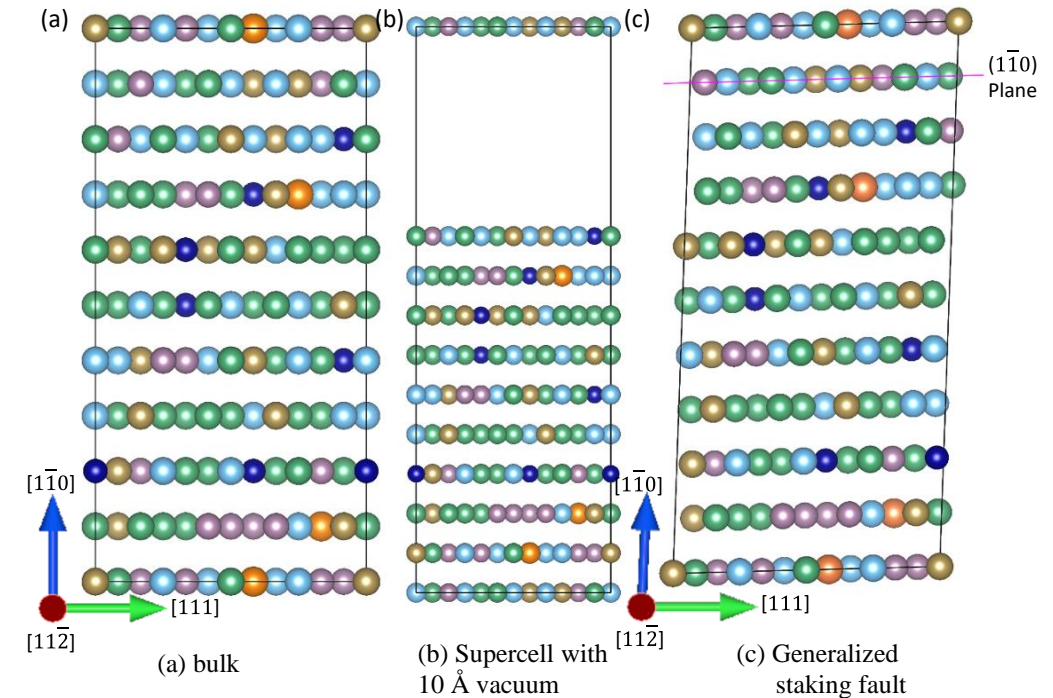


# Intrinsic Ductility: D-Parameter

$$D = \frac{\gamma_{suf}}{\gamma_{usf}} \quad \gamma_{suf} = (1-10) \text{ surface energy}; \gamma_{usf} = (1-10)[111] \text{ unstable stacking fault energy}$$



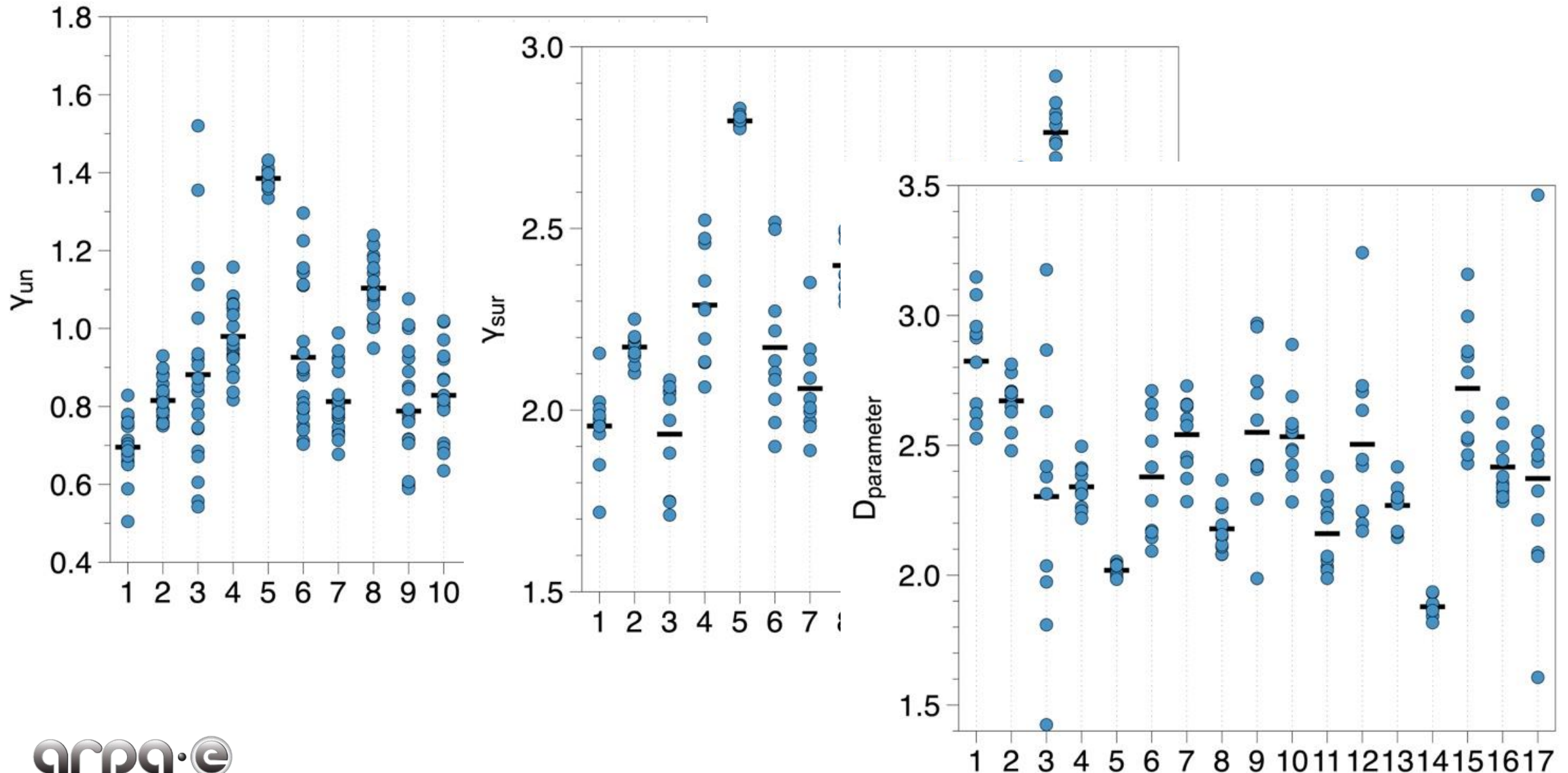
black dots: experiments are available.  
Red dots: DFT validations



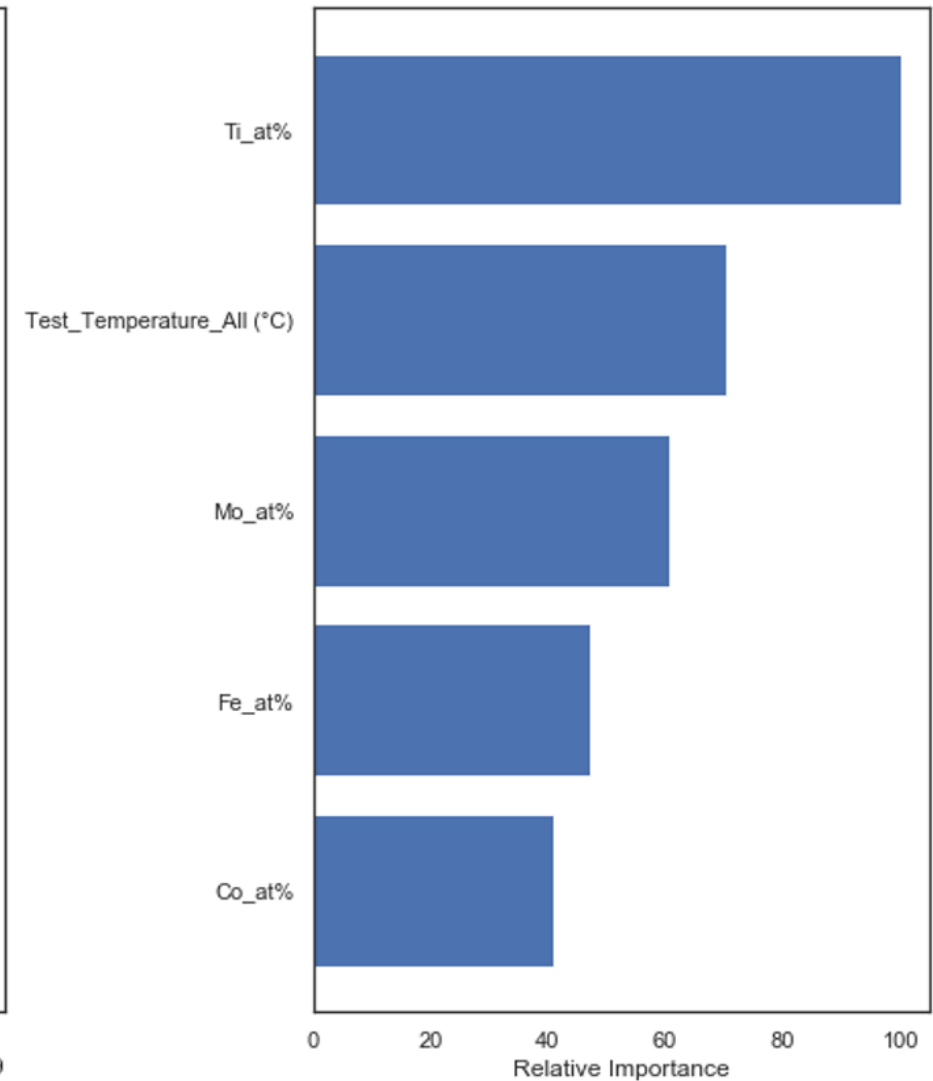
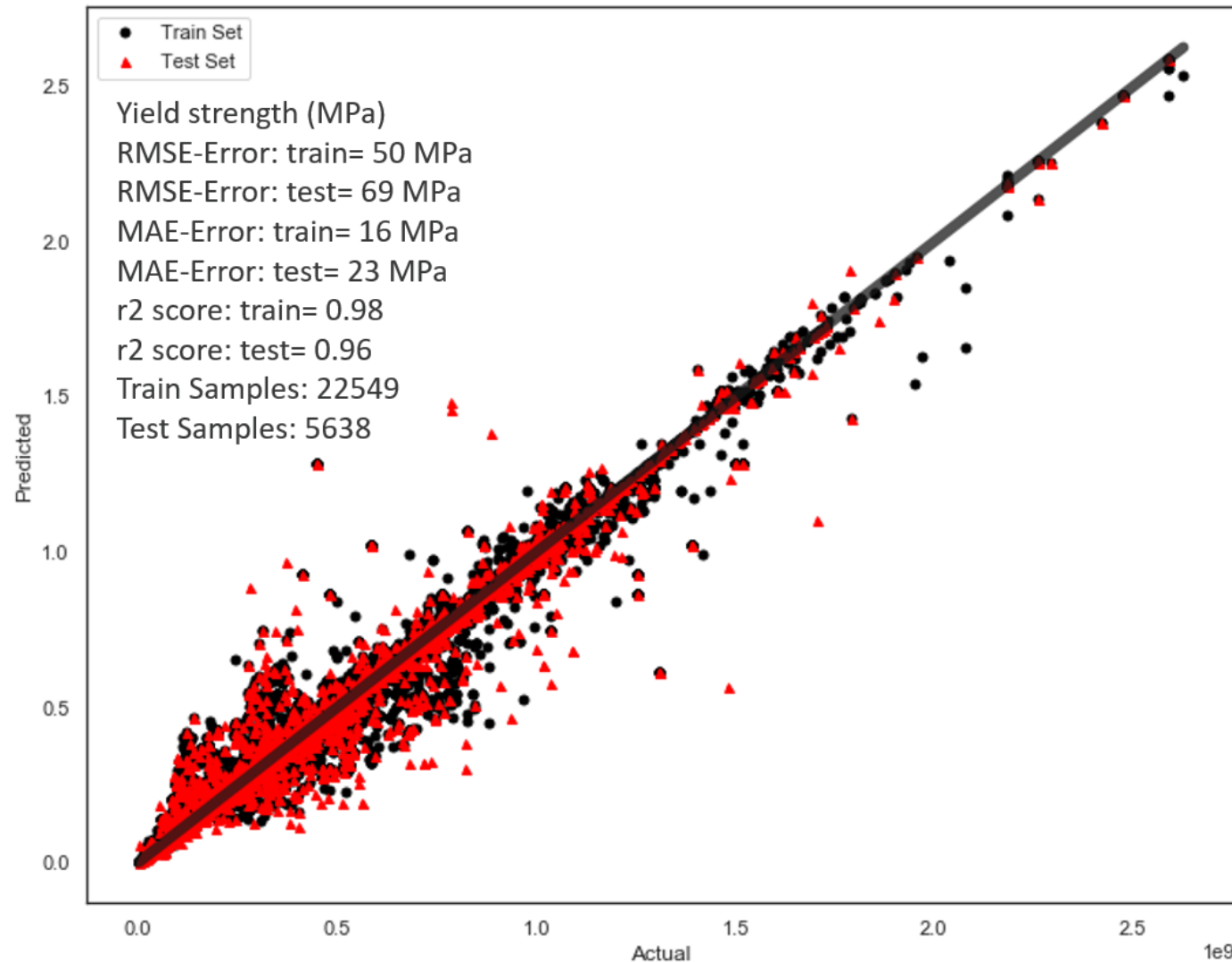
Y.J. Hu, A. Sundar, S. Ogata, L. Qi, Screening of generalized stacking fault energies, surface energies and intrinsic ductile potency of refractory multicomponent alloys, *Acta Materialia* **210** (2021) 116800



# Intrinsic Ductility: D-Parameter



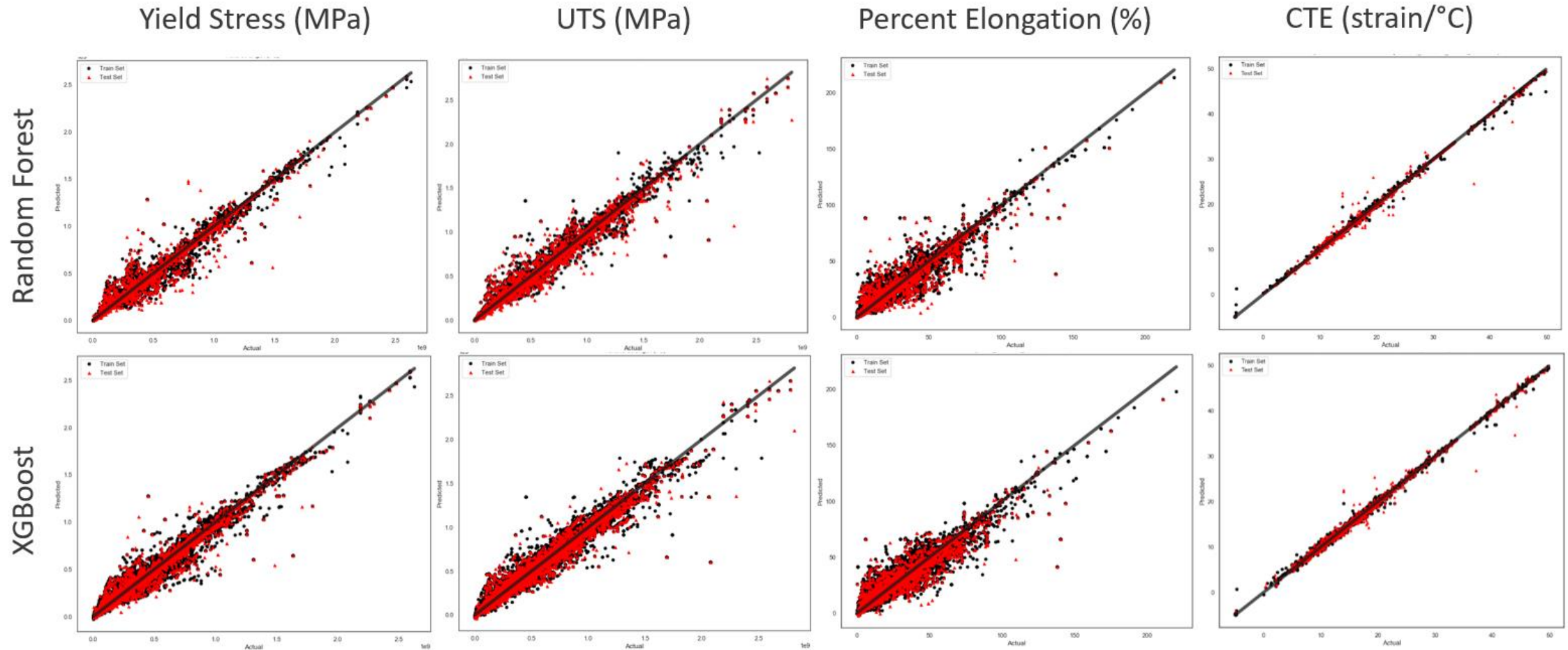
# Machine Learning Modeling



# Machine Learning Modeling

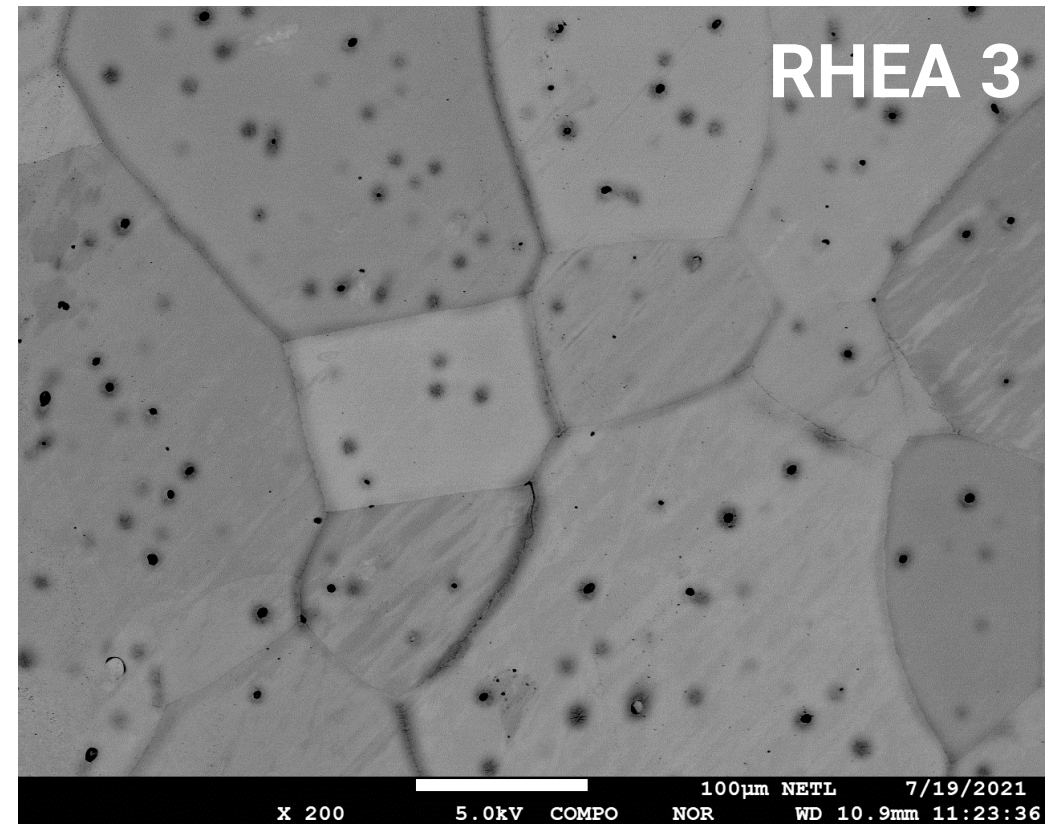
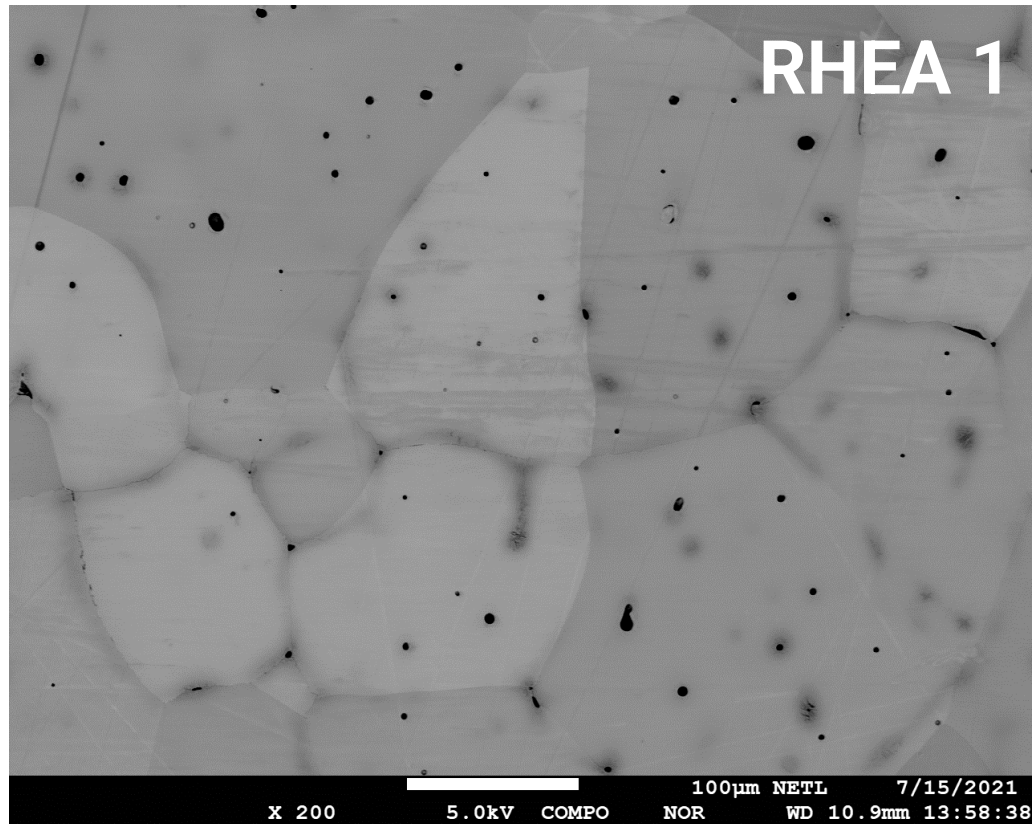
## Target Property

### Regression Model Type





# Alloy Design Results: Round #1

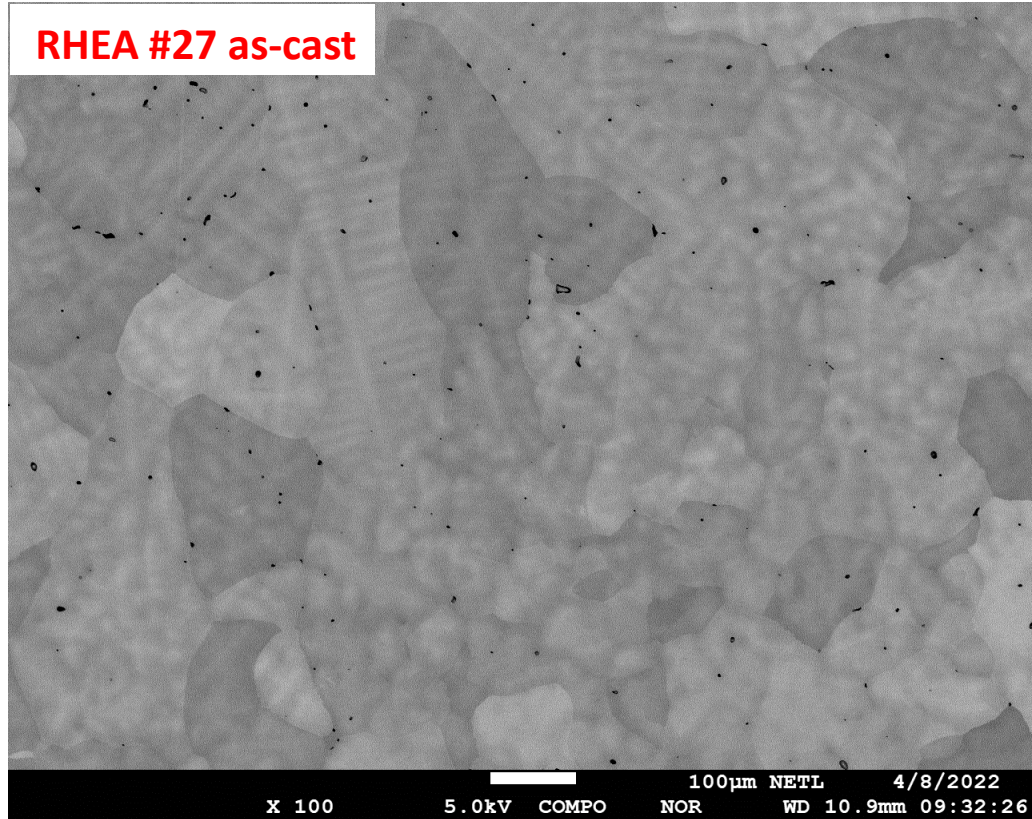


SEM backscattering electron micrographs. Those round, black features are presumably Ti-rich FCC MC carbides, as measured by EDS and WDS. The size is less than 10  $\mu\text{m}$ . They appear to be homogeneously distributed. This demonstrates the concept of MC carbide precipitation in a RHEA.

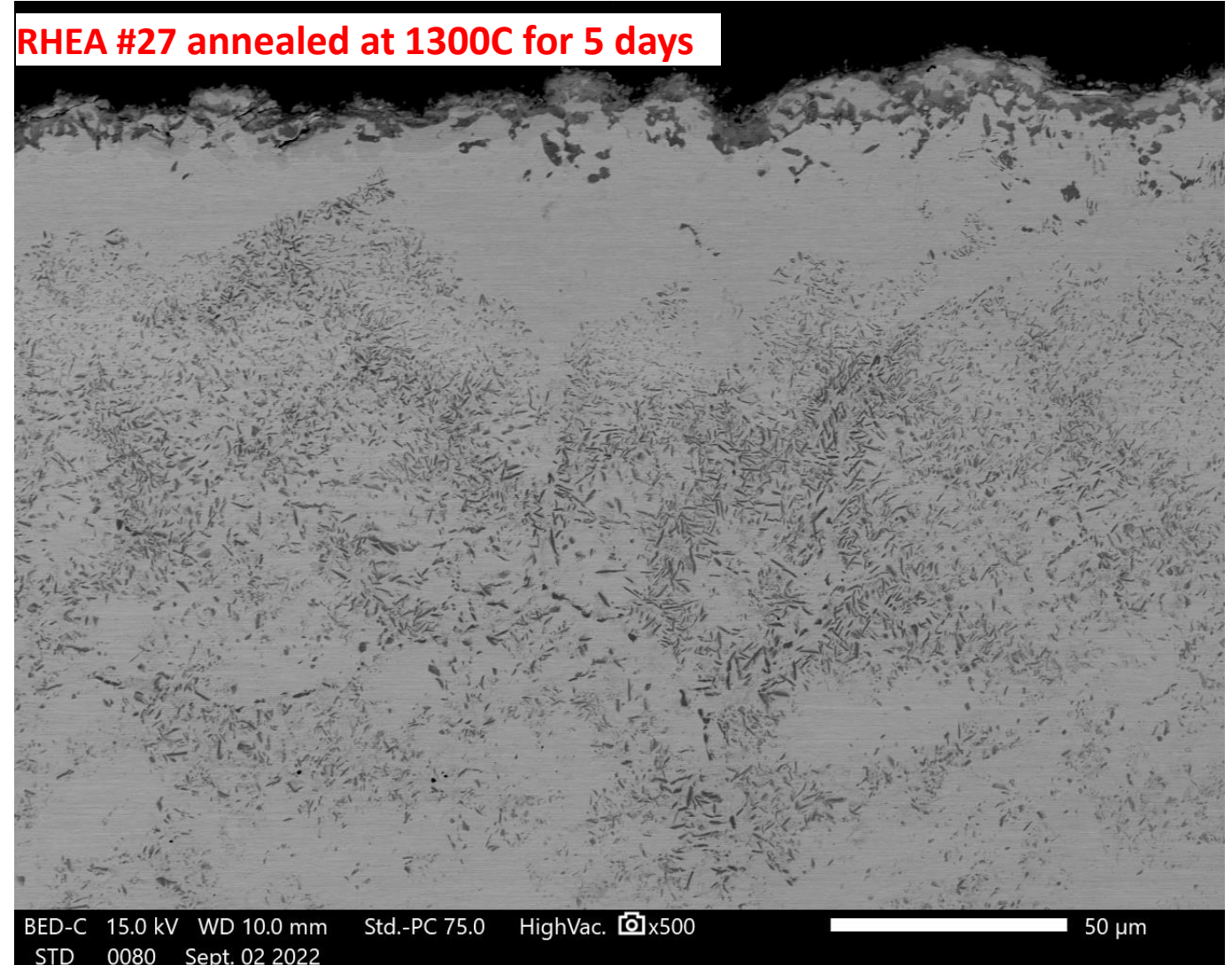


# Alloy Design Results: High Temperature

RHEA #27 as-cast

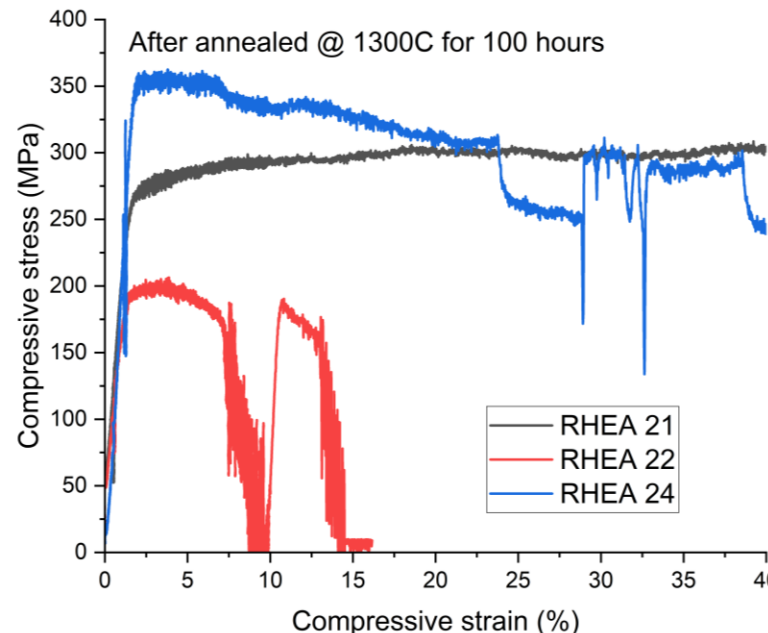
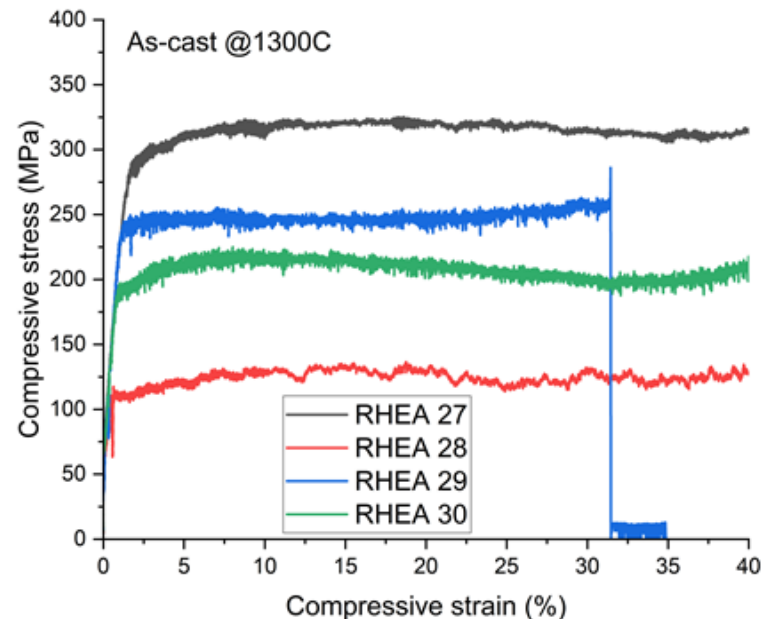
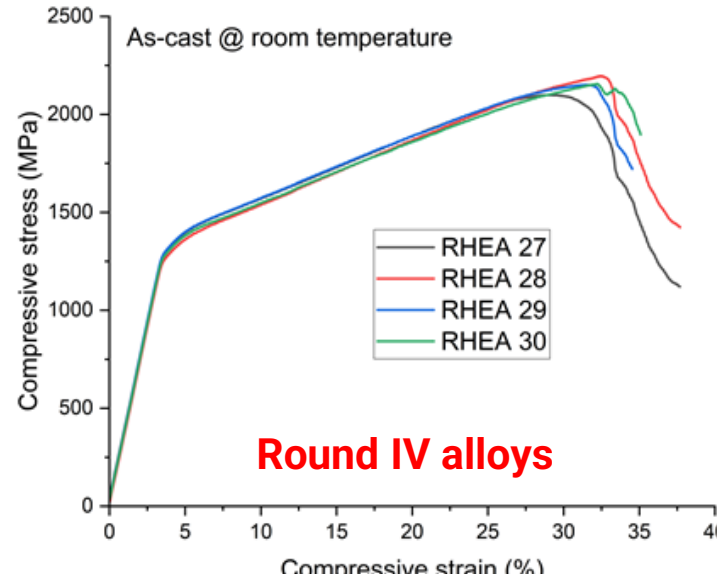
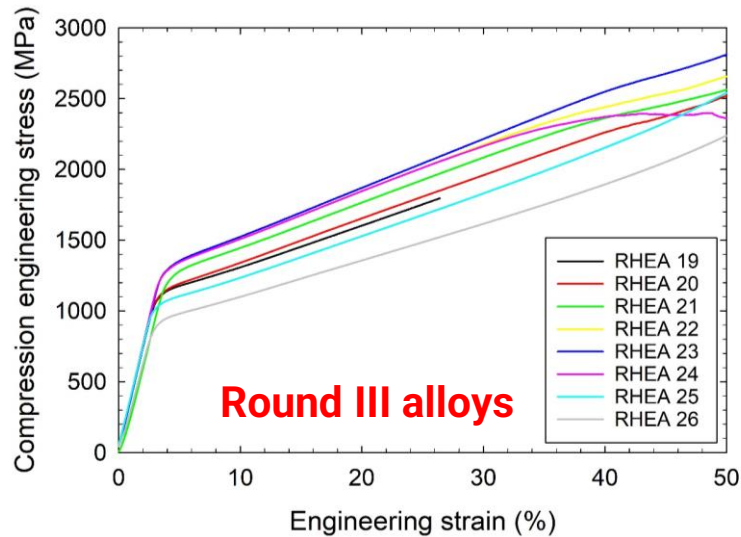


RHEA #27 annealed at 1300C for 5 days

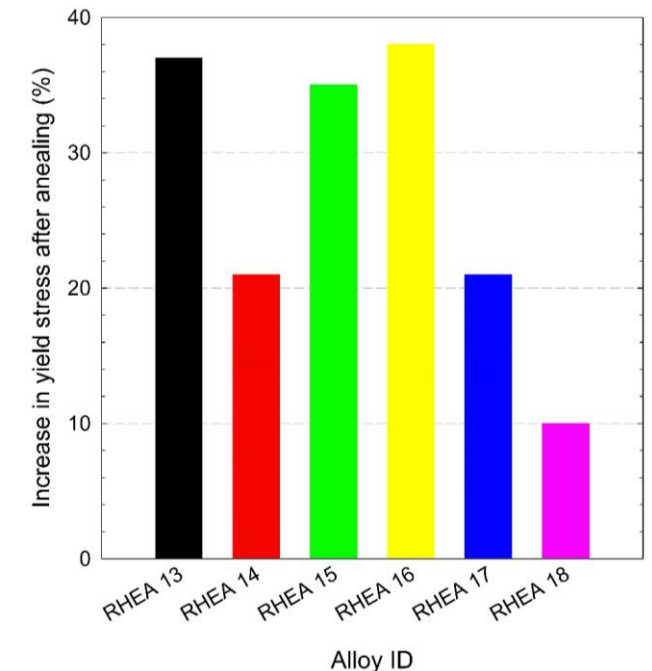




# Alloy Design Results: Mechanical Properties

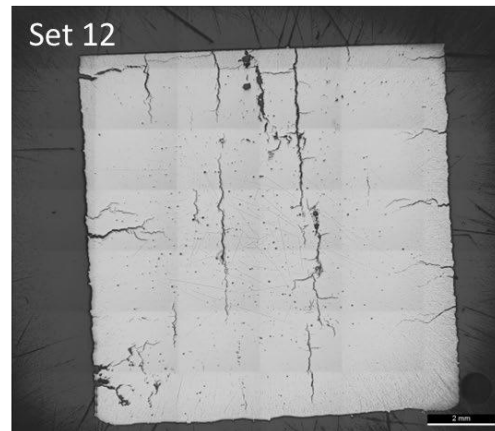
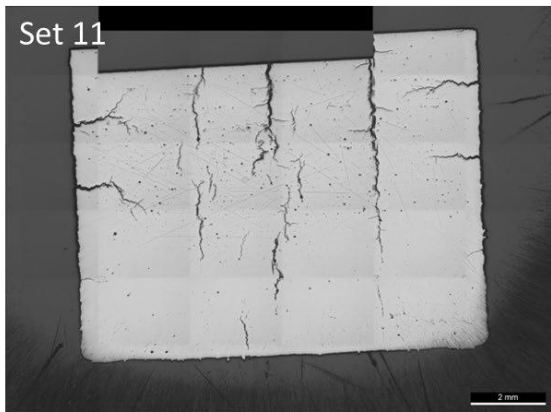
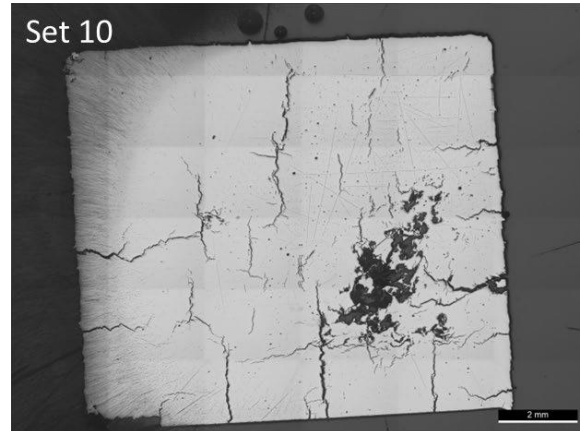
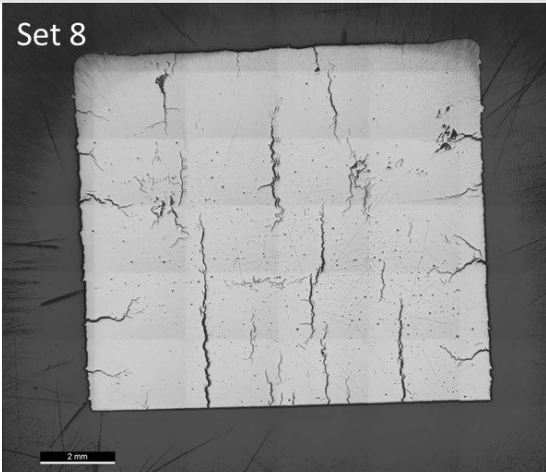


Compression tests at room temperatures before and after annealing at 1300C for 100 hours. The Y axis represents the change in yield stress due to annealing.

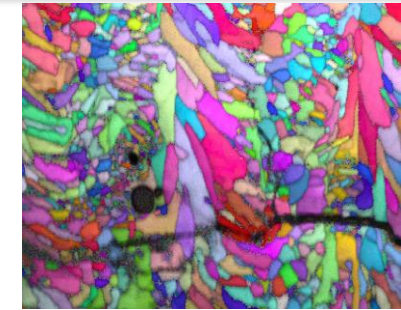


The increase in yield stress is due to precipitation of fine MC carbides and formation of fine oxides

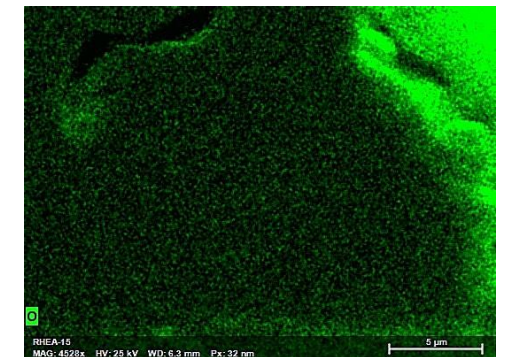
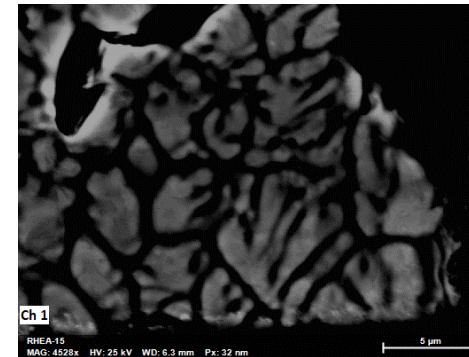
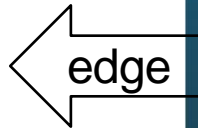
# L-DED Processing of RHEA-15



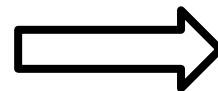
Oxygen-enrichment  
near the edge-  
emanating  
crack/free surface.



100  $\mu$ m



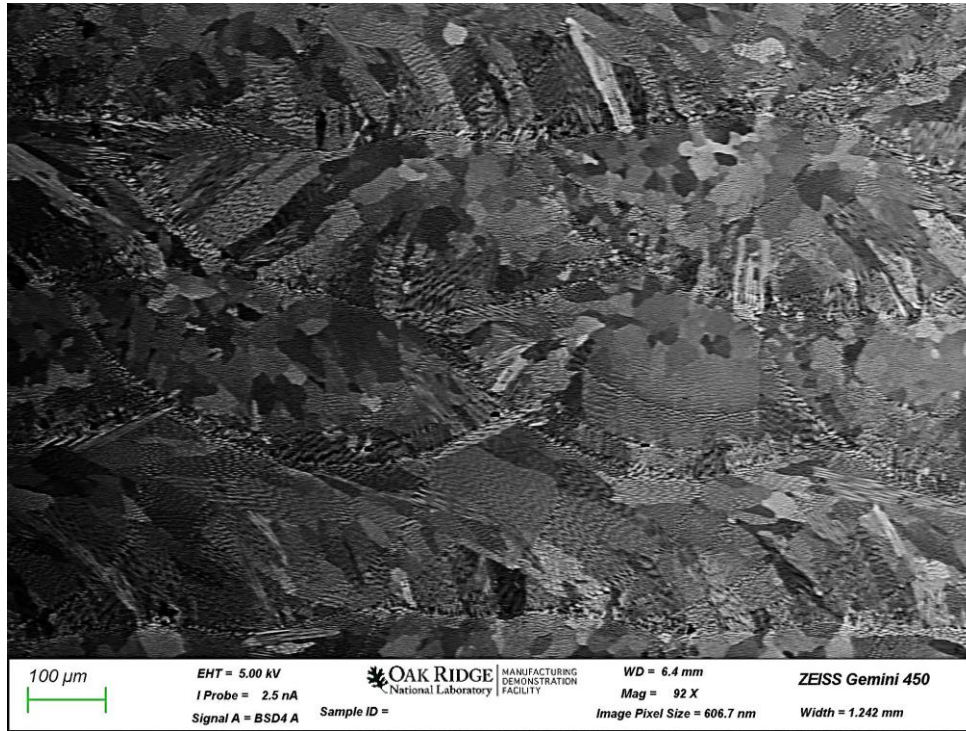
Results from  
chemical analysis



High oxygen contamination:  
**0.30 wt%**



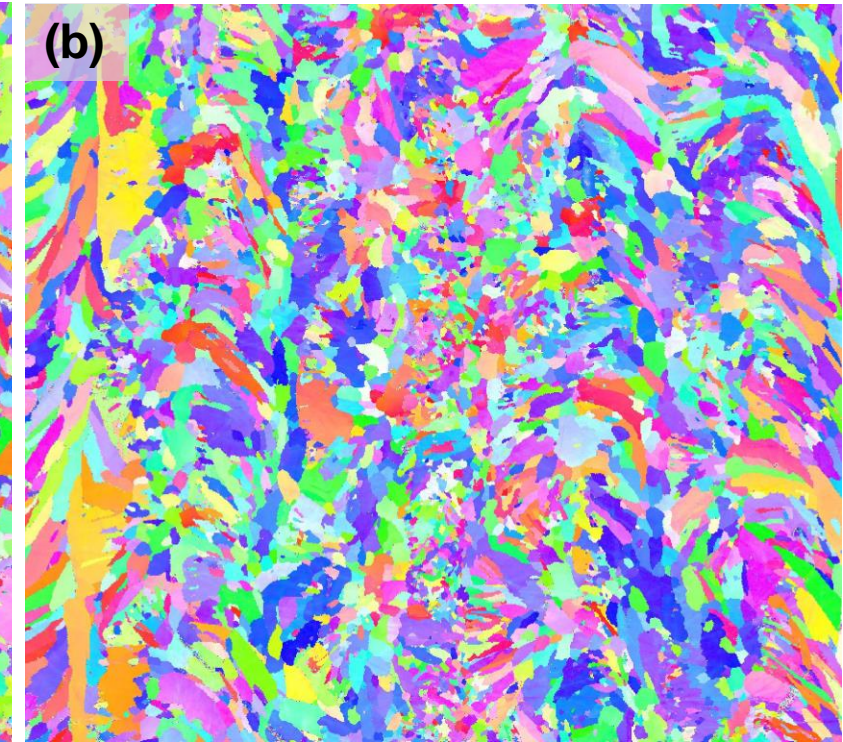
# Microstructural Evolution of L-DED Processed RHEA-15 (Round 2 Builds)



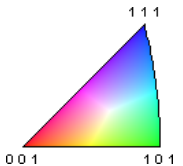
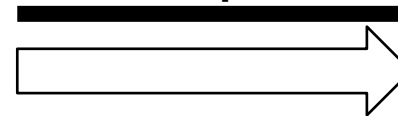
67° specimen



700 μm



700 μm



***Minimal cracking in the internal volume of Round 2 builds and no solidification cracking indicates that RHEA-15 can be processed with L-DED if O content is kept minimal.***

- High throughput, multiscale modeling are carried out to accelerate alloy design that possess balanced RT ductility & toughness, and high-temperature yield strength and creep performance, while being light weight ( $\leq 9 \text{ g/cm}^3$ ) and low cost.
- Optimized Debye-Grüneisen approach with volume dependence of the Grüneisen parameter to predict coefficient of thermal expansion (CTE)
- Arc-melt buttons are made to rapidly verify computational design
- Annealing promotes MC carbide precipitation as well as oxide dispersion, contributing about 20% increase in yield strength.
- Plasma arc melted ingots of large size are made. Tensile tests and creep tests are being planned.
- Good progress is made in mitigating thermal cracking in L-DED RHEAs.



# Acknowledgements

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