

Fatigue and fracture behavior of aluminum alloys in gaseous hydrogen (PVP2023-106442)

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Hydrogen technologies include diverse range of operations



Hydrogen delivery

- **hydrogen pipelines: carbon steels**
- **Challenge: cyclic pressure**

Hydrogen storage

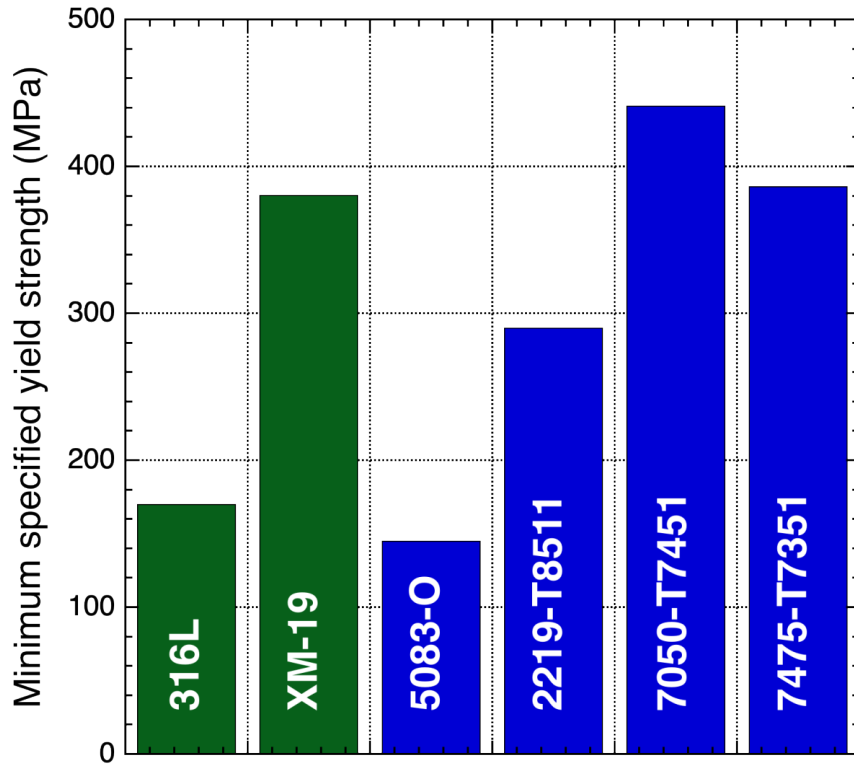
- **hydrogen forklifts: Cr-Mo ferritic steels**
- **Challenge: filling ~6/day**



Pressure manifold components

- **Valves, tubing, and other devices: austenitic stainless steels**
- **Challenges: low temperature, lower-cost alternatives (e.g., aluminum or higher strength), alloy content, welding**

Aluminum alloys can be higher strength than 316L



Solid-solution

- 5083-O: Al-Mg alloy

Precipitation-strengthened (aged)

- 2219-T8511: Al-Cu alloy

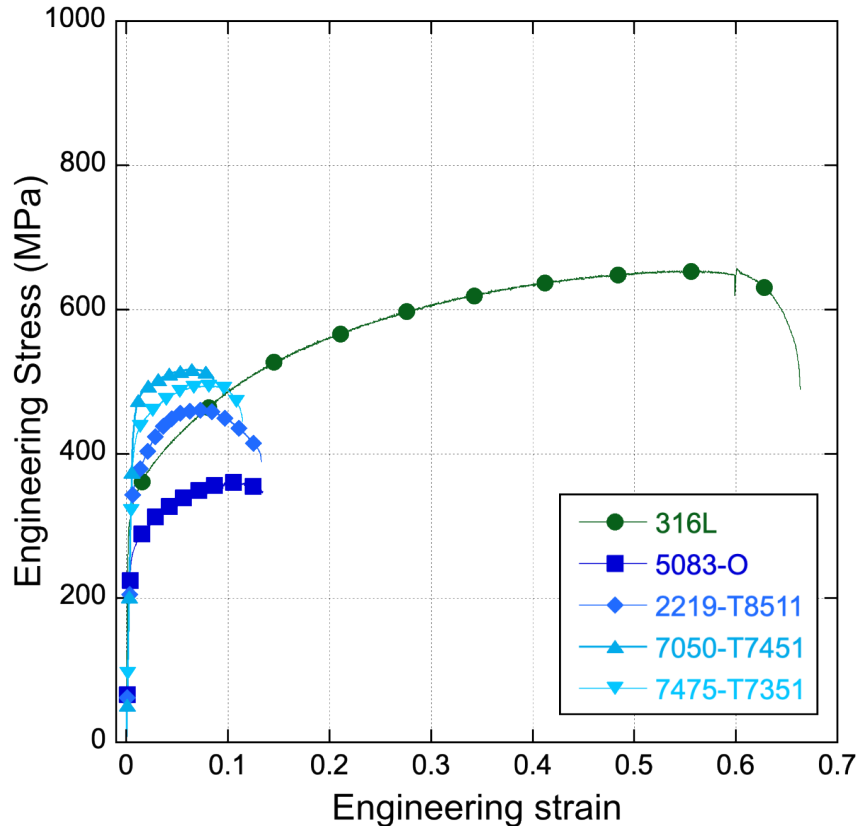
Overaged

- 7050-T7451: Al-Zn-Mg-Cu alloy
- 7475-T7351: Al-Zn-Mg-Cu alloy

Trade-off:

Aluminum alloys display significantly lower ductility, fatigue, and fracture properties than austenitic stainless steels

Tensile ductility of Al alloys can be less than 20% of 316L



- Aluminum alloys are less common than steels in pressure applications due to low fatigue and fracture resistance
- However, aluminum is commonly used in safety critical components, such as in aerospace applications
 - For example, 2219 alloy was used for liquid hydrogen tanks on the space shuttle and common in other aerospace applications
 - 7475 is an aerospace grade alloy with significantly higher fracture resistance than 7075

Standard methods used to evaluate fatigue and fracture

Instrumentation

- **Load:** internal load cell
- **Displacement:** clip gauge on specimen
- **Crack monitoring:** Direct Current Potential Difference (DCPD)

Fatigue crack growth rate (FCGR):

ASTM E647

- **Load ratio (R):** 0.1 & 0.3
- **Frequency:** 1 Hz
- **Constant load amplitude or K-control**

Fracture resistance:

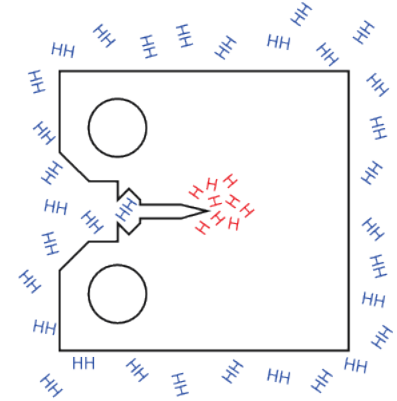
ASTM E1820 (elastic-plastic fracture)

- **Constant displacement rate:** 0.005 mm/min

Environments

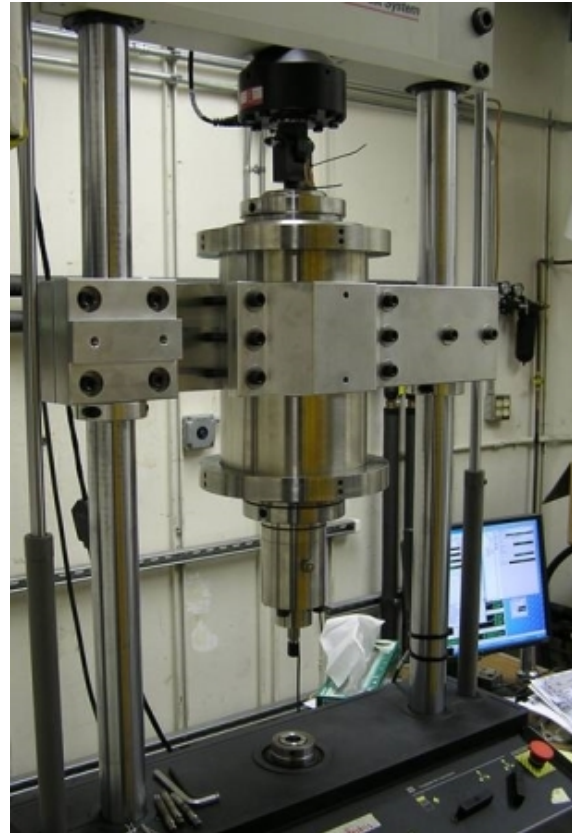
- **Laboratory air**
- **99.9999% N₂:** 100 bar
- **99.9999% H₂:** 1,000 bar (GH2)

Compact tension geometry

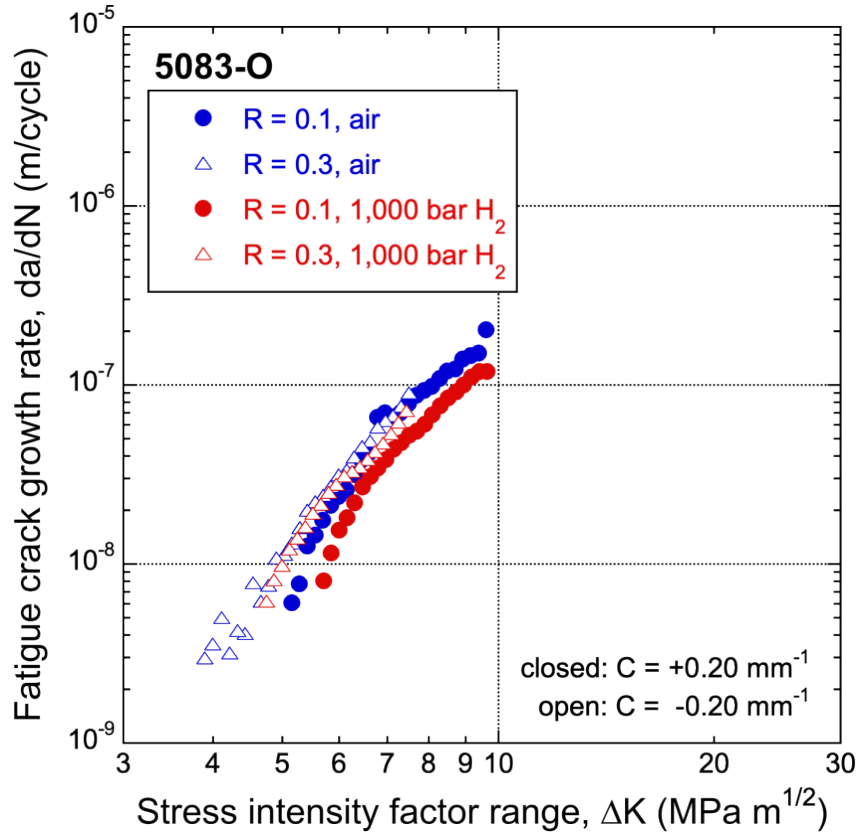


B = 12.7mm
W = 26.4 mm

**Fracture test conducted at
conclusion of fatigue test**



Aluminum alloy 5083-O

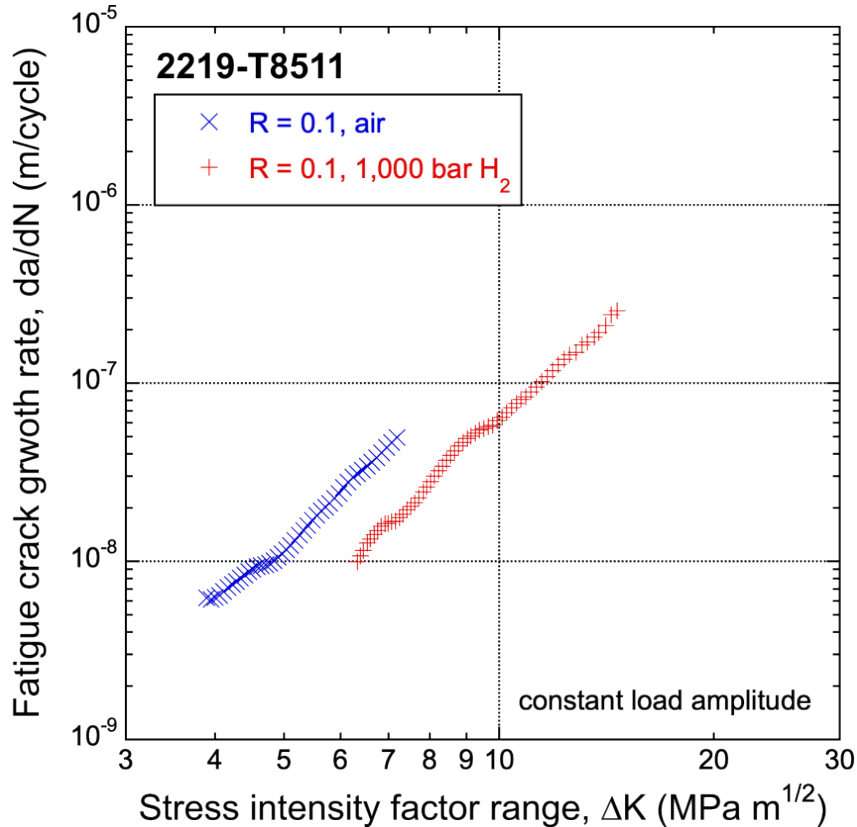


Weldable alloy common in vehicle applications with excellent environmental corrosion resistance

- No apparent degradation of fatigue crack growth due to high-pressure (1,000 bar) gaseous hydrogen**

FCGR of 5083-O is not accelerated in high-pressure GH2

Aluminum alloy 2219-T8511

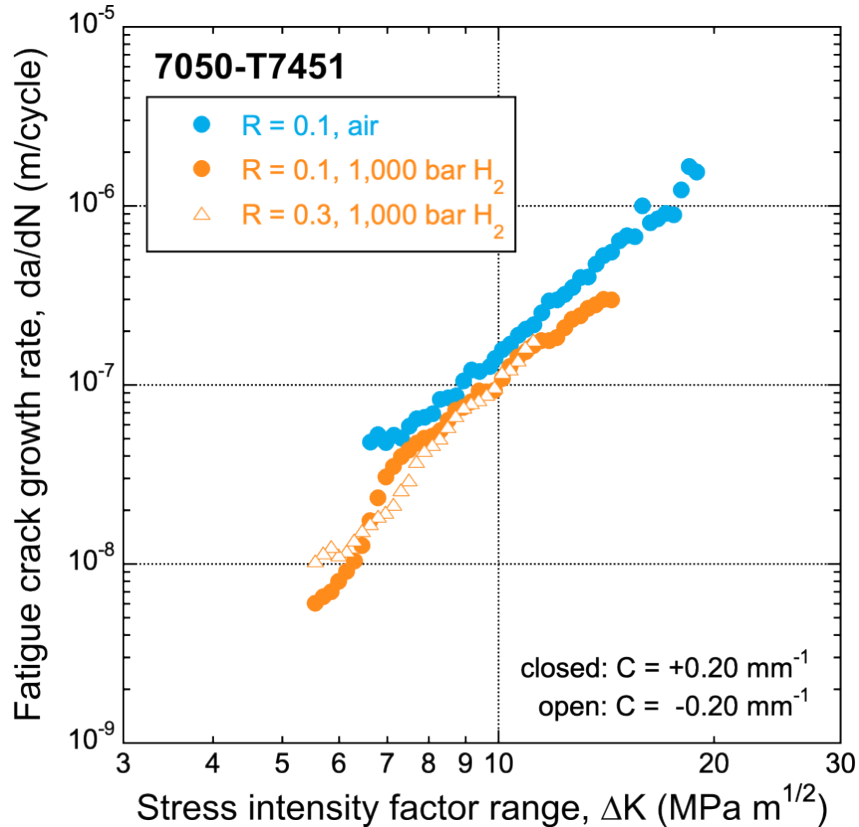


Weldable, high-strength alloy with good resistance to stress-corrosion cracking (SCC) used in aerospace applications

- **No apparent degradation of fatigue crack growth due to high-pressure (1,000 bar) gaseous hydrogen**

FCGR of 2219-T8511 is not accelerated in high-pressure GH2

Aluminum alloy 7050-T7451

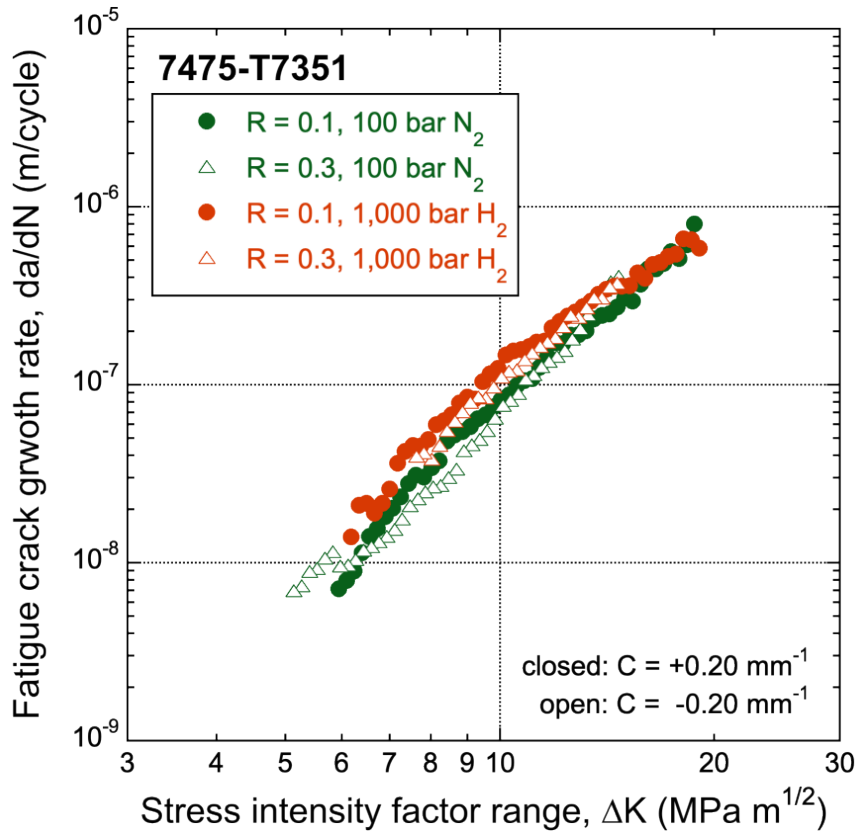


Common high-strength aluminum alloy with poor weldability and susceptibility to stress-corrosion cracking

- Overaged condition shows no apparent degradation of fatigue crack growth due to high-pressure (1,000 bar) gaseous hydrogen

FCGR of 7050-T7451 is not accelerated in high-pressure GH2

Aluminum alloy 7475-T7351

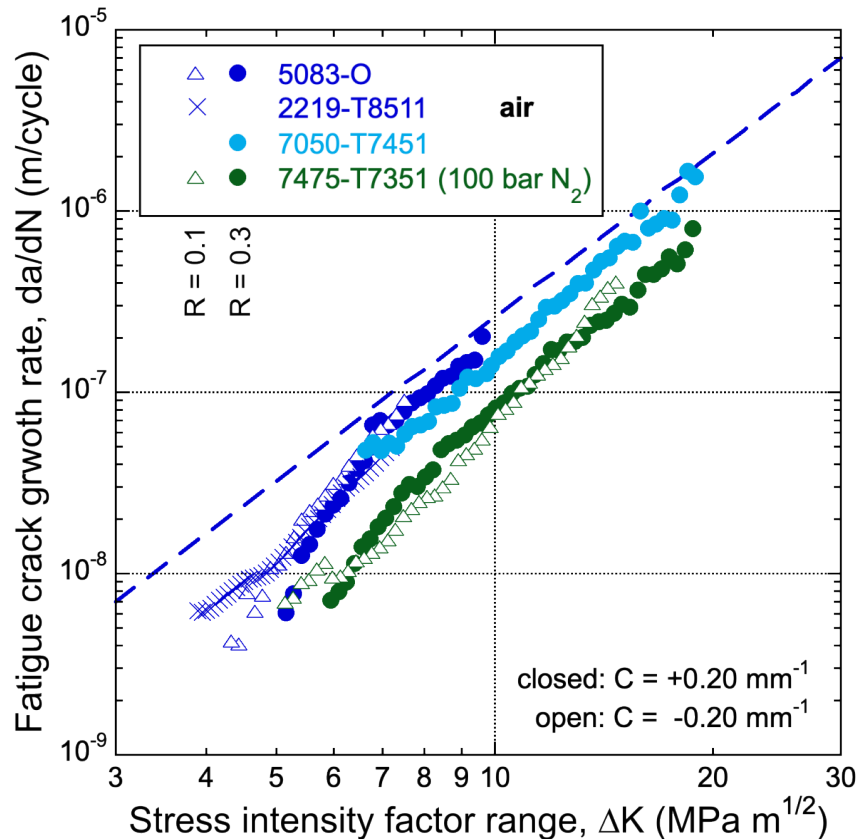


High-strength, aerospace aluminum alloy with poor weldability and susceptibility to stress-corrosion cracking

- Fatigue crack growth appears slightly lower in dry nitrogen than in GH2
- Fatigue crack growth of overaged condition in high-pressure (1,000 bar) gaseous hydrogen is similar to other alloys

FCGR of 7475-T7351 is not accelerated in high-pressure GH2

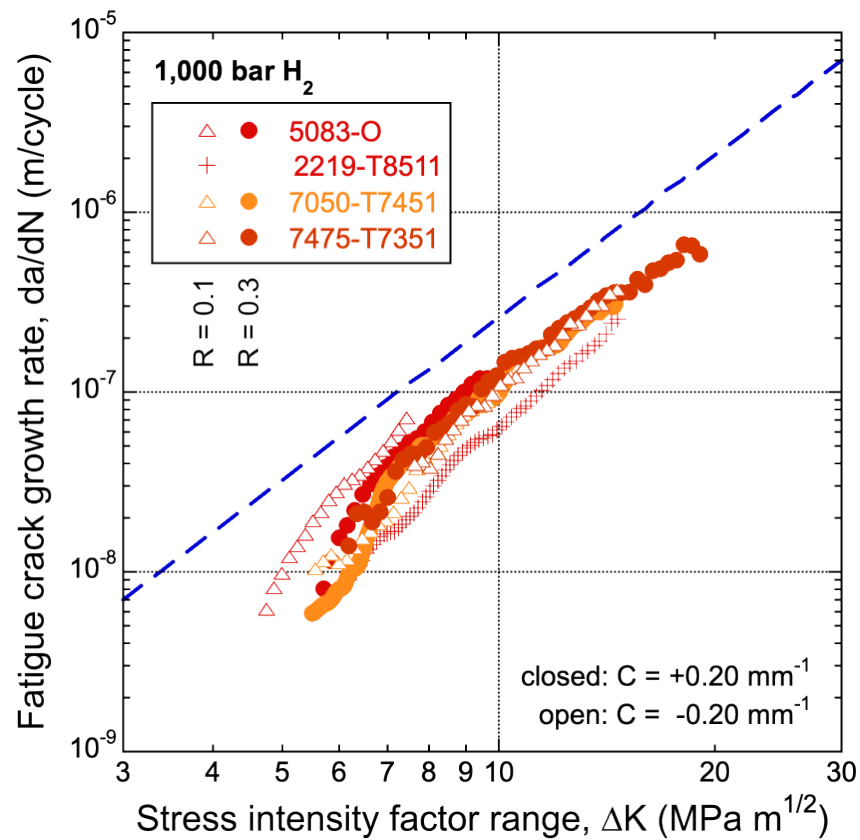
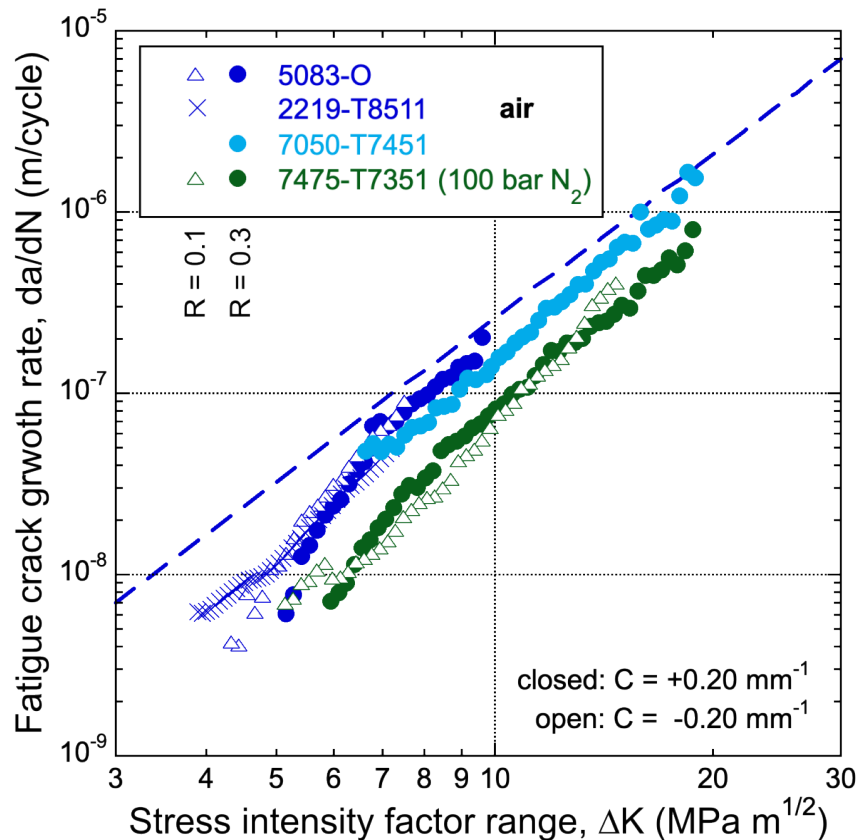
Composite data in air show consistent FCGR for all alloys



- All tested aluminum alloys show similar fatigue crack growth in air
- Fatigue crack growth rates are bounded by recommended Paris Law relationship (dashed line)
 - from International Institute of Welding
- Fatigue crack growth appears lower in dry nitrogen than in air

FCGR of aluminum alloys is sensitive to humidity

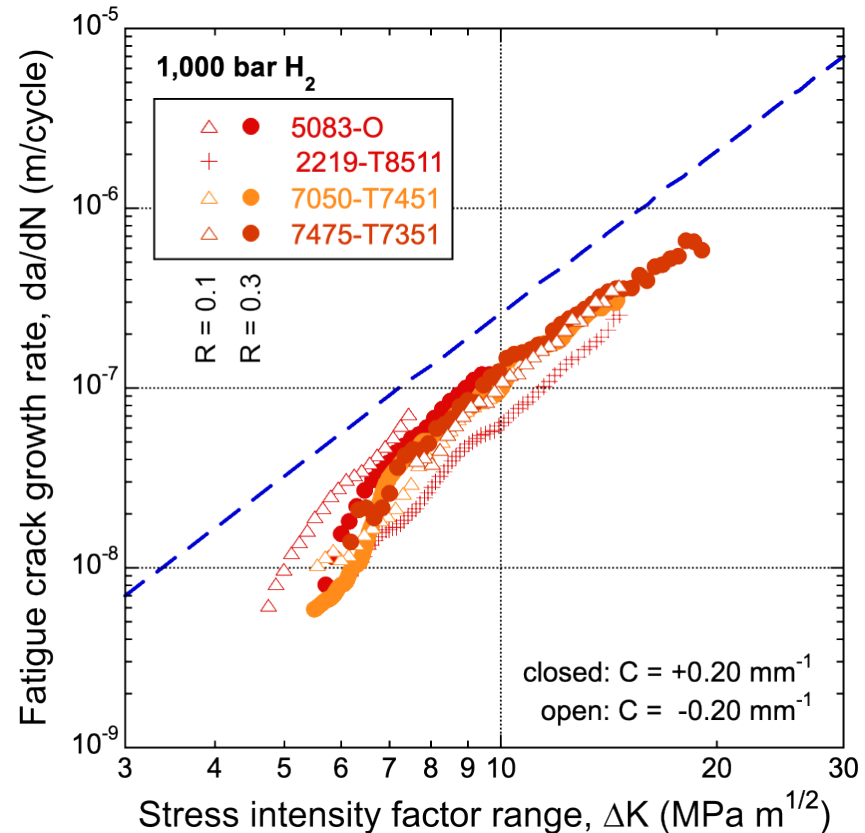
FCGR of all alloys in GH2 is slightly lower than in air



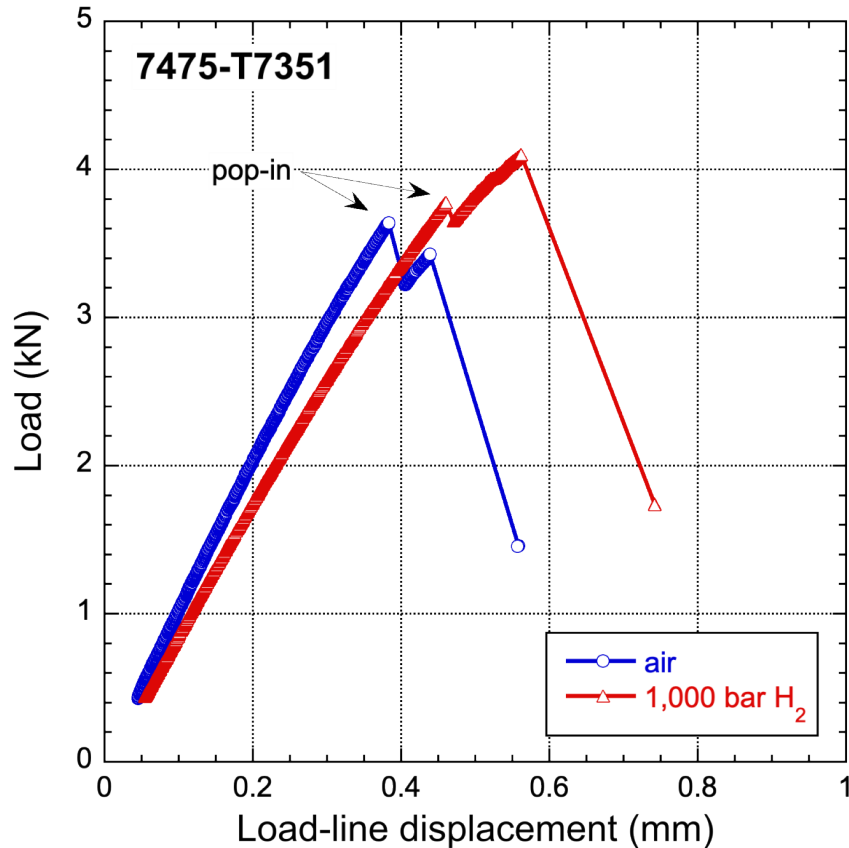
FCGR in GH2 is similar for all tested alloys

- All tested alloys display lower fatigue crack growth rates than benchmark behavior for aluminum alloys
- Some scatter in the fatigue crack growth rates
 - Perhaps related to residual moisture content in the gaseous environment (gas quality was not evaluated for all tests)

FCGR of high-strength aluminum alloys is not accelerated in dry, high-pressure GH2



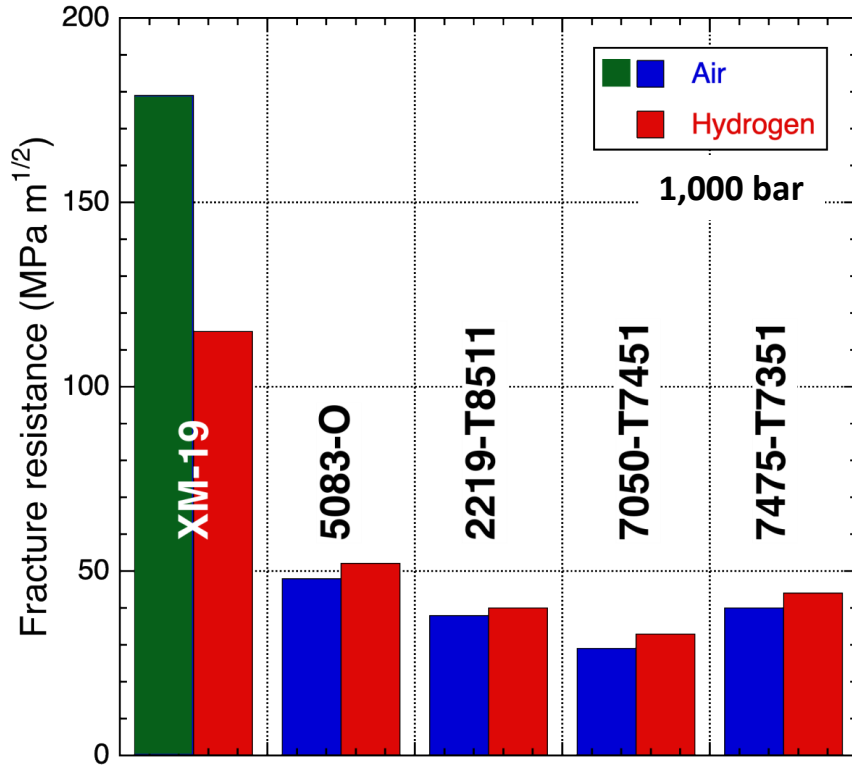
Fracture resistance is similar in GH2 and air



- High-strength aluminum alloys show 'low' fracture resistance in air, often characterized by unstable crack growth (e.g., pop-in)
- In GH2, fracture resistance is generally slightly higher than in air
 - Improved fracture resistance can be achieved in dry gas environments, including dry GH2

Dry hydrogen environments appear to be more 'inert' than laboratory air, even at high pressure

Fracture resistance of aluminum alloys in dry GH2 is better than in air



XM-19 w/ internal H from:
Metall Mater Trans **41A** (2010) 3348

- Fracture resistance of austenitic stainless steel is reduced in hydrogen environments
- In contrast, fracture resistance of aluminum alloys is not degraded in dry GH2
- However, fracture resistance of aluminum will generally be lower than common austenitic stainless steels in GH2 environments

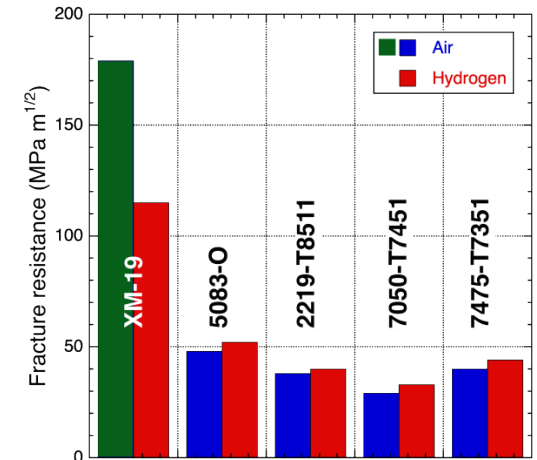
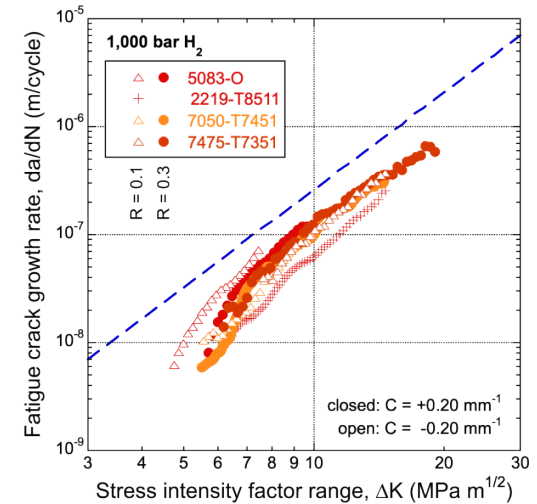
Fracture resistance of aluminum alloys is better defined in GH2 than steels, since it is unchanged (or better) relative to air

Summary

- GH2 at pressure of 1,000 bar did not degrade the fatigue crack growth behavior of the tested high-strength aluminum alloys
 - Nominally same FCGR for R = 0.1 and 0.3
- Fracture resistance also was not degraded in GH2 at pressure of 1,000 bar

Results:

- Both fatigue crack growth and fracture resistance is improved slightly in dry, high-pressure GH2 relative to tests in air
- Results likely mediated by moisture content



Thank You!

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<https://h-mat.org/>

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