



Characterization of Heterogeneity and Mechanical Properties of SiC-SiC Composites

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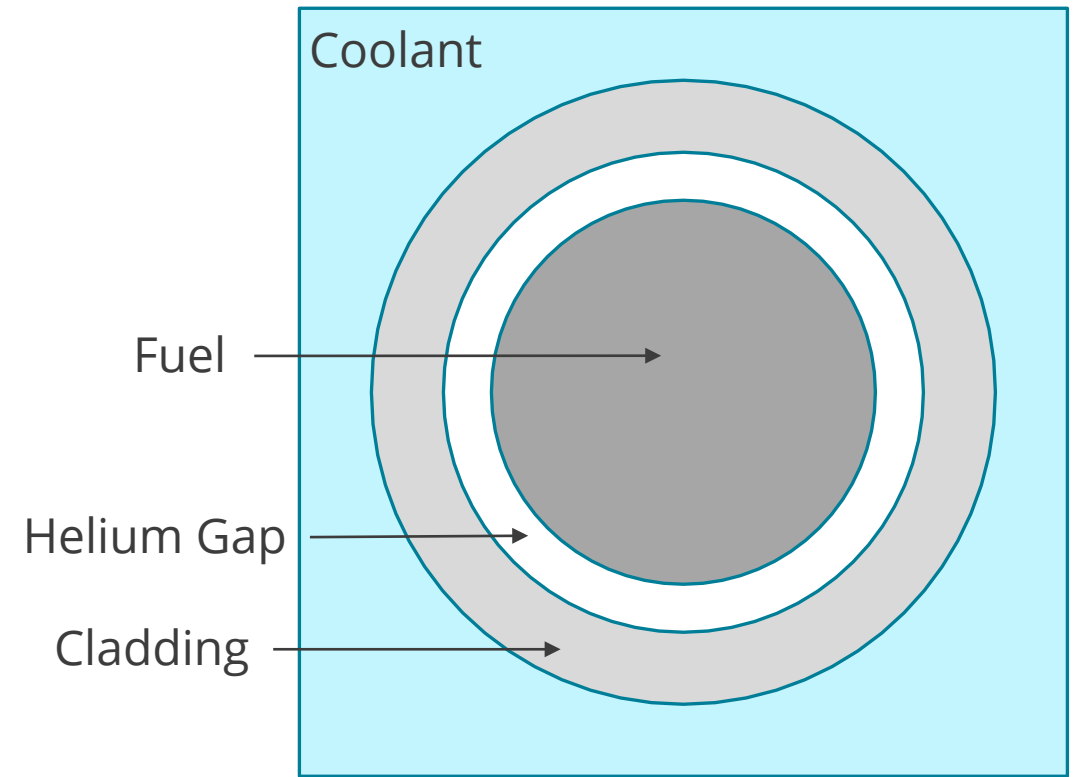


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Nuclear Fuel Cladding



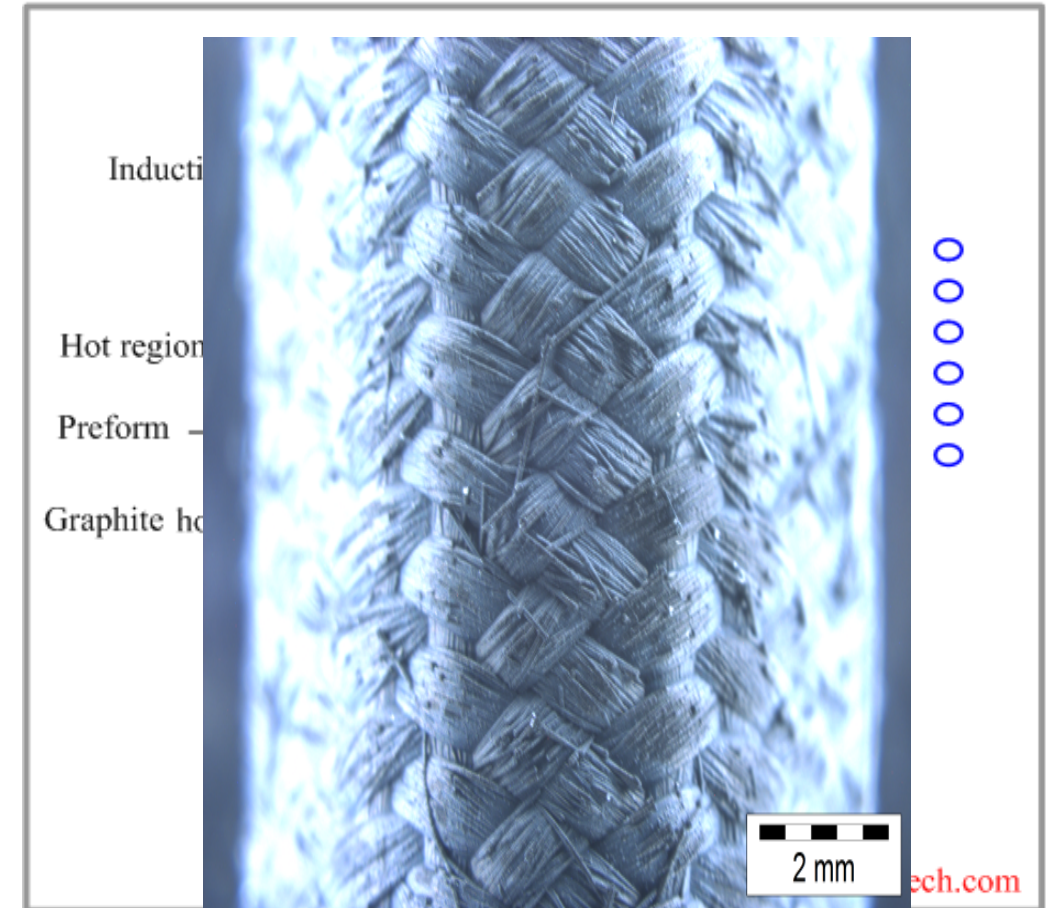
- Current nuclear fuel cladding is primarily a zirconium alloy
- Fails under very high temperatures and loses strength
 - Typically happens during Loss of Coolant Accidents (LOCA)
- Potential alternative is a SiC fiber reinforced tubular composite
 - Excellent chemical compatibility with coolant and fuel systems in high temperature gas cooled fast reactors



CVI SiC-SiC Composites



1. Hi-Nicalon Type-S SiC fibers ($\sim 10\text{ }\mu\text{m}$ diameter) are bundled and braided into tows to make a plain-weave configuration
2. 100-500nm thick PyC interphase layer is added by chemical vapor deposition
3. Matrix is densified throughout the composite via chemical vapor infiltration
4. Possible outer coating layer afterward (EBC, etc.)

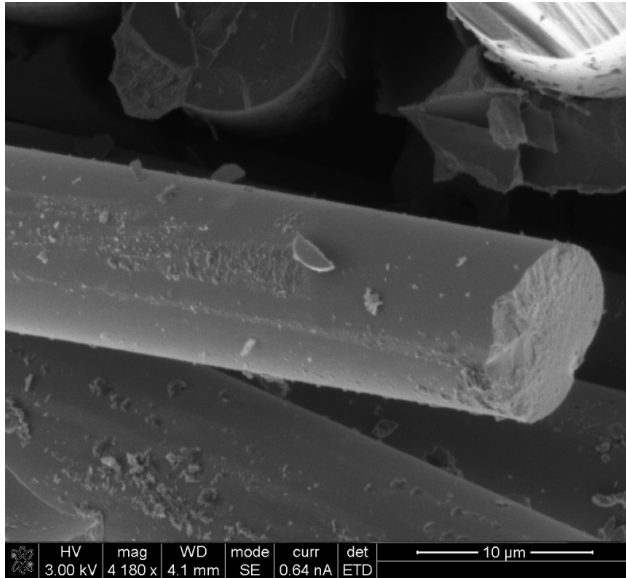


Sources of Failure in SiC-SiC Composite Tubular Cladding



- Complex stress state
 - Prefilling of tubes with helium
 - Accumulation fission gas during operation
 - Fuel pellet swelling and pellet-cladding interactions
 - LOCA failures
 - Rapid temperature rise and uneven expansion of fuel pellet
- ✓ **Contributes to scatter in strength and uncertainty in performance**
- ✓ **Incorporation of strength variability into design for reliable operation**
- Inherent microstructural heterogeneity
 - Constituents: fibers, interphase, and matrix phase
 - Geometry: fibers and tows in different orientations
 - Microstructural variabilities: porosity, local density fluctuations, tube thickness variations, weave architecture imperfections, etc

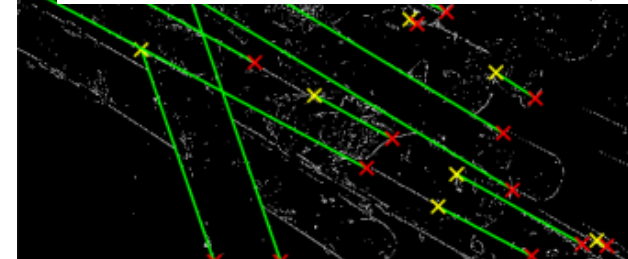
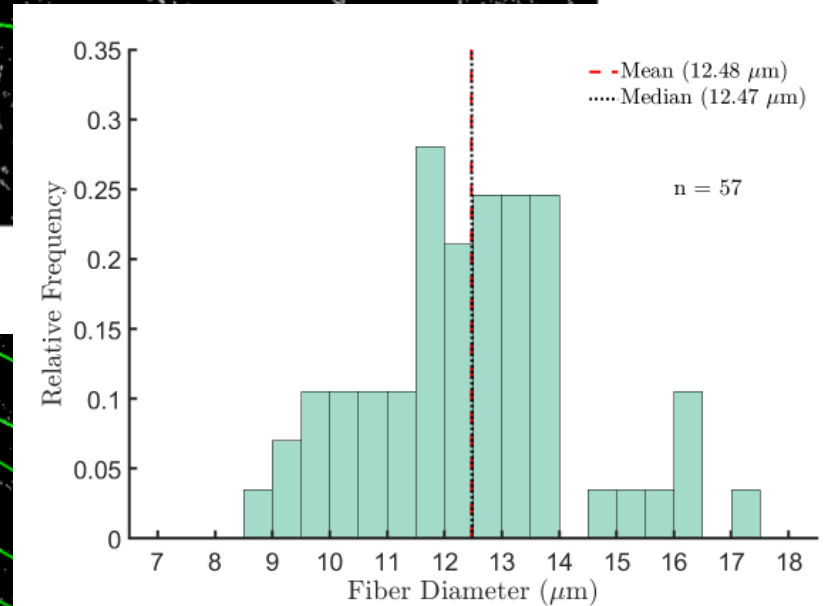
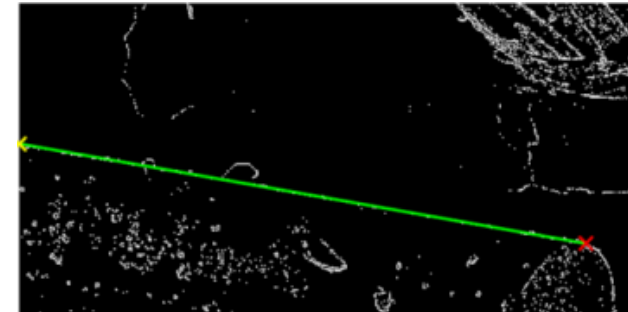
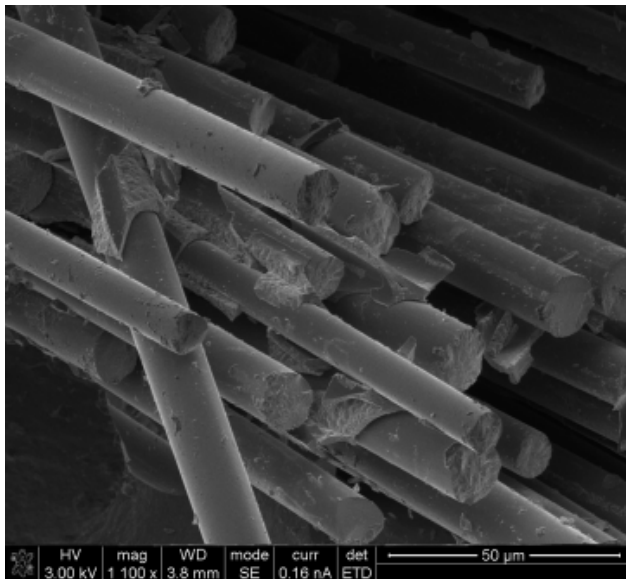
Measuring Fiber Diameter



Gaussian blur

Roberts edge
detection

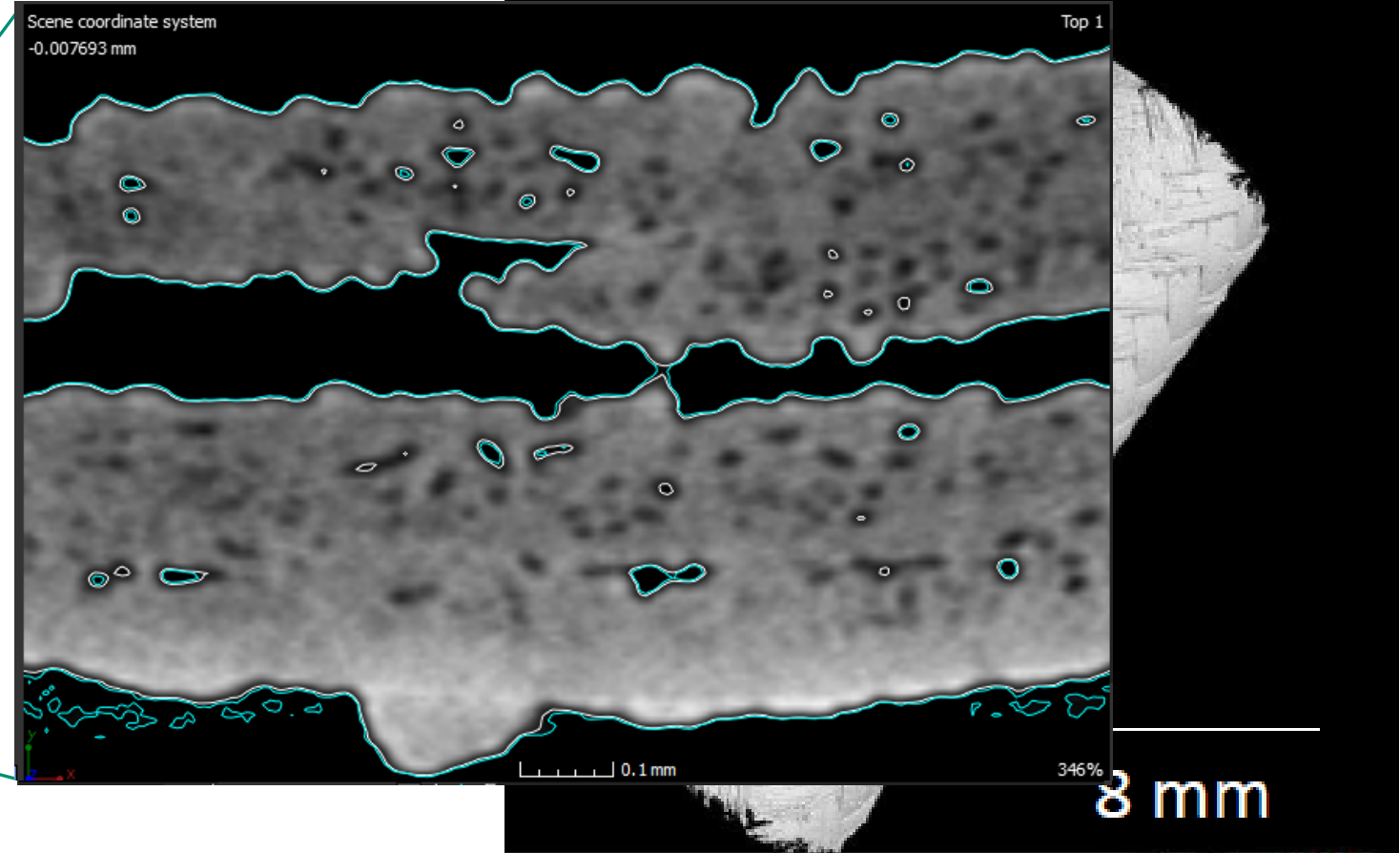
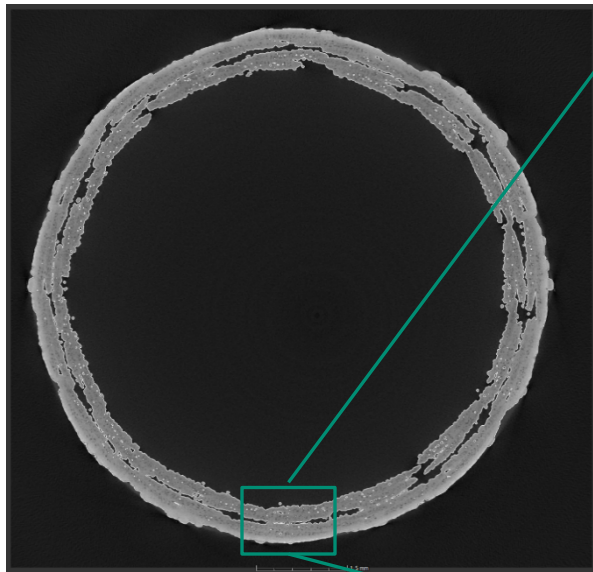
Hough line transform

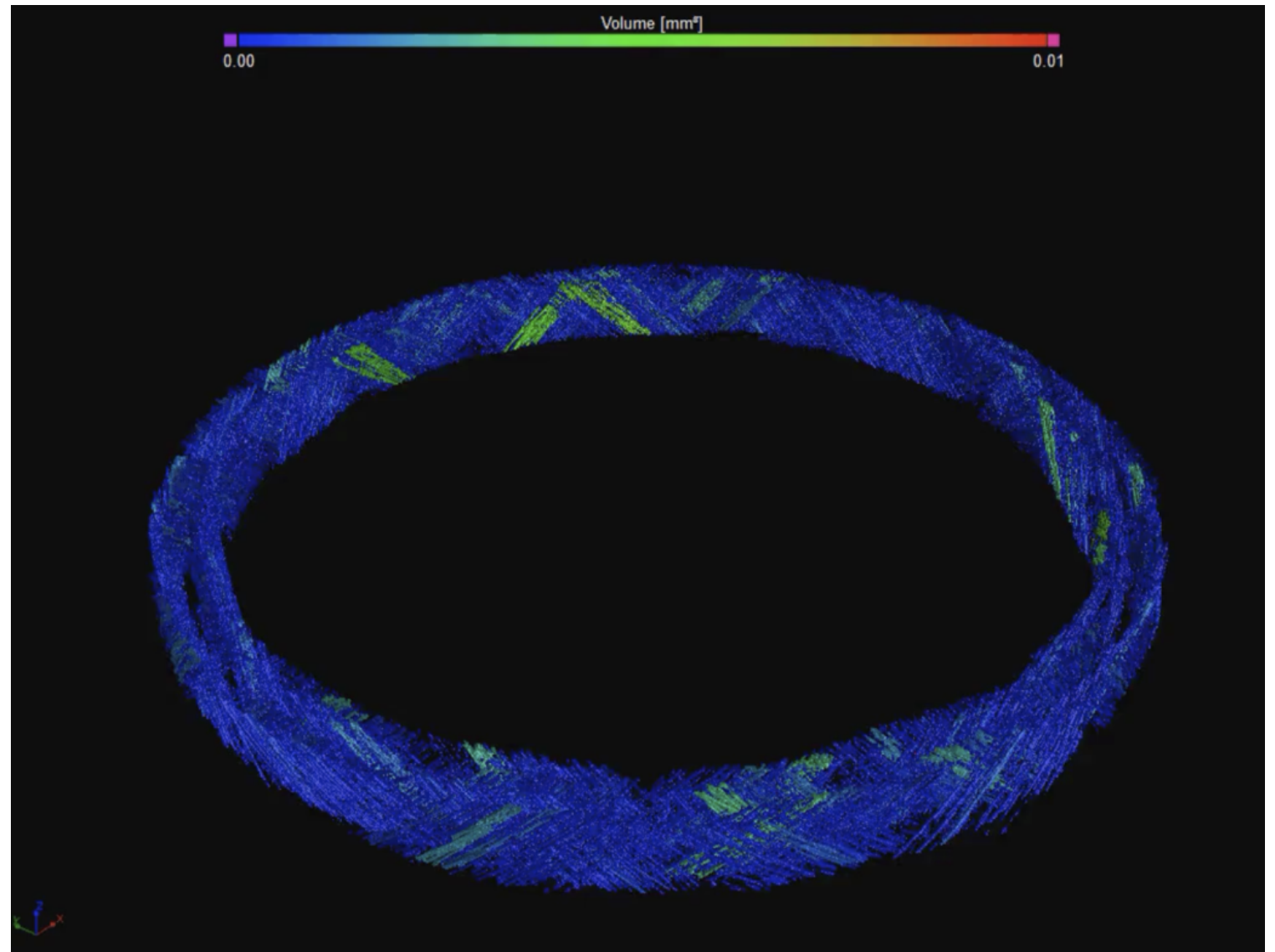
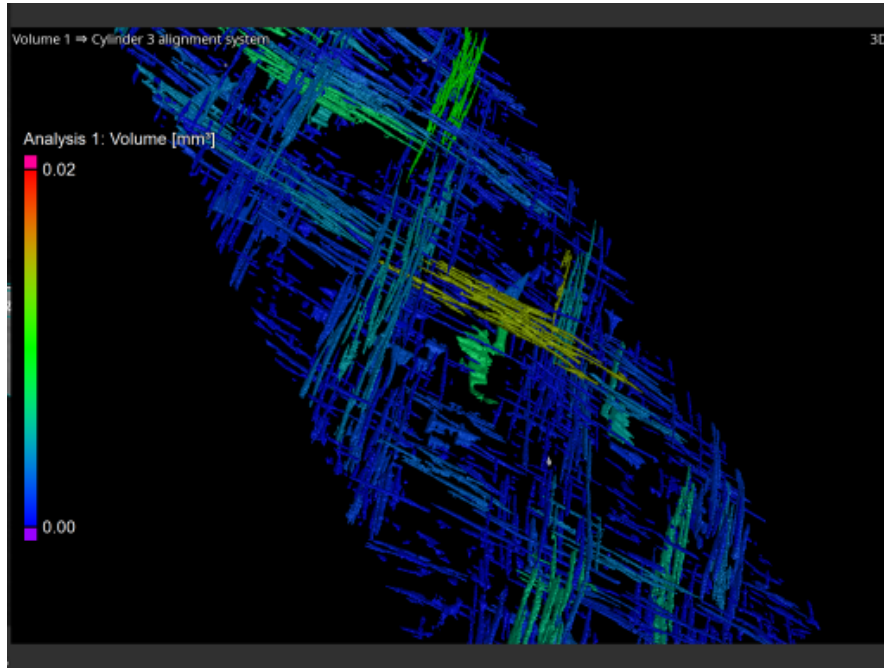


Porosity Distribution



Using X-ray computed tomography to measure the porosity shape, size, and distribution in the SiC-SiC composite





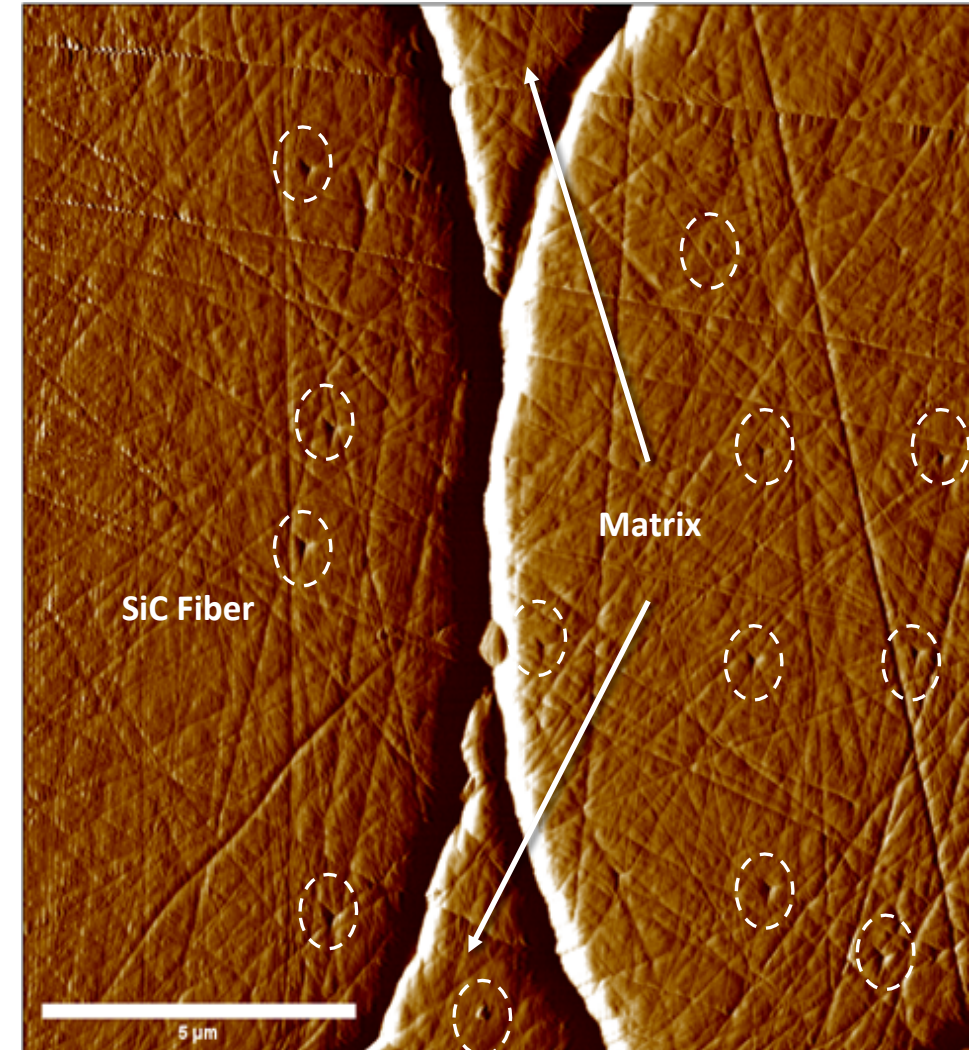
Nanoindentation



Fiber	Matrix
$366 \pm 36 \text{ GPa}$	$477 \pm 47 \text{ GPa}$

Indents were displacement-controlled
at a depth of 200 nm with room
temperature ambient conditions

At least 10 indentations per constituent

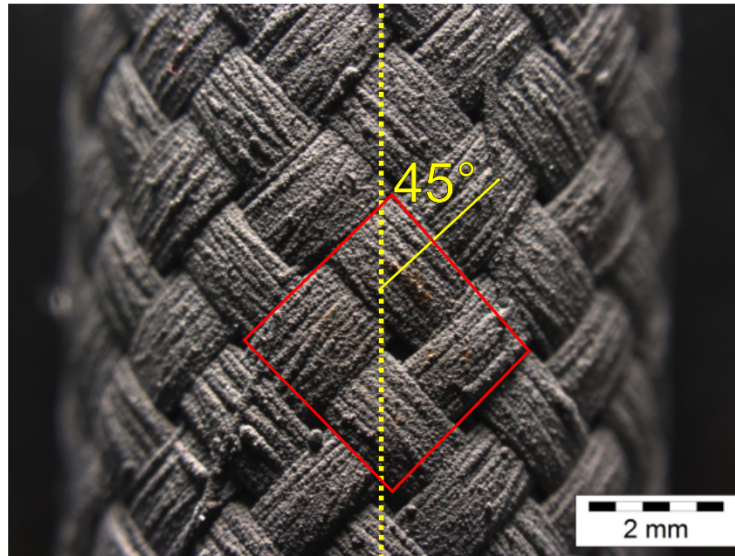


- During operation, fuel pellet swelling and pellet-cladding interactions introduces complex stress states into the cladding.
- Furthermore, LOCA can cause a rapid rise in temperature causing uneven expansion of the fuel pellets, further introducing complexity into the stress state.
- Several different loading scenarios were chosen to examine how the material behaves to help give a full range of data for failure envelope validation:
 - Axial compression
 - Tensile hoop burst
 - Flexure

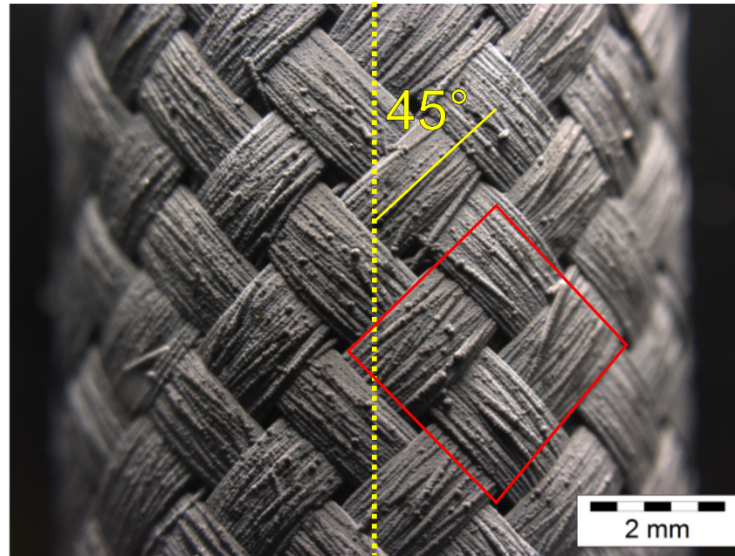
Mechanical Testing - Architectures



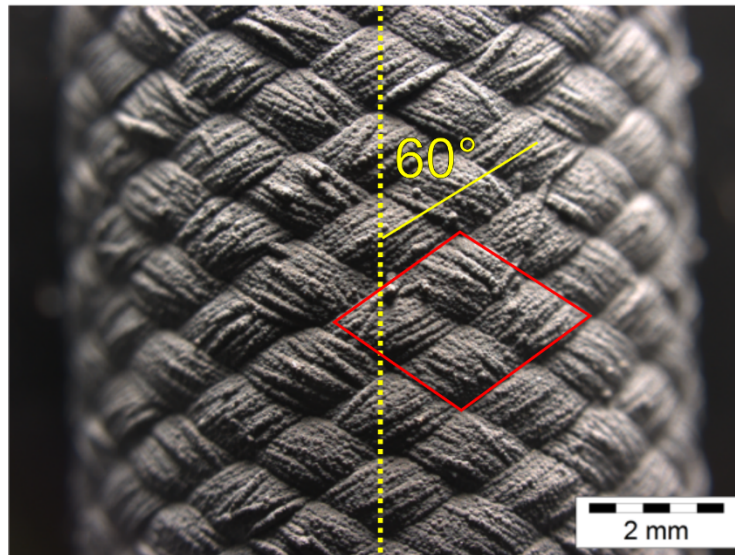
45° 2-ply
biaxial



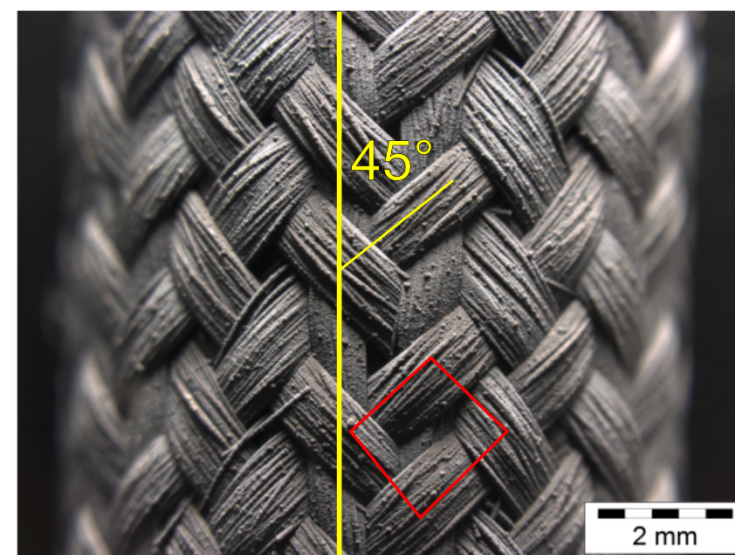
45° 3-ply
biaxial

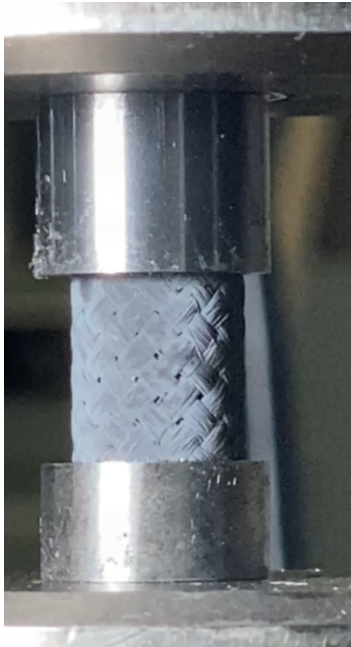


60° 2-ply
biaxial



45° 2-ply
triaxial



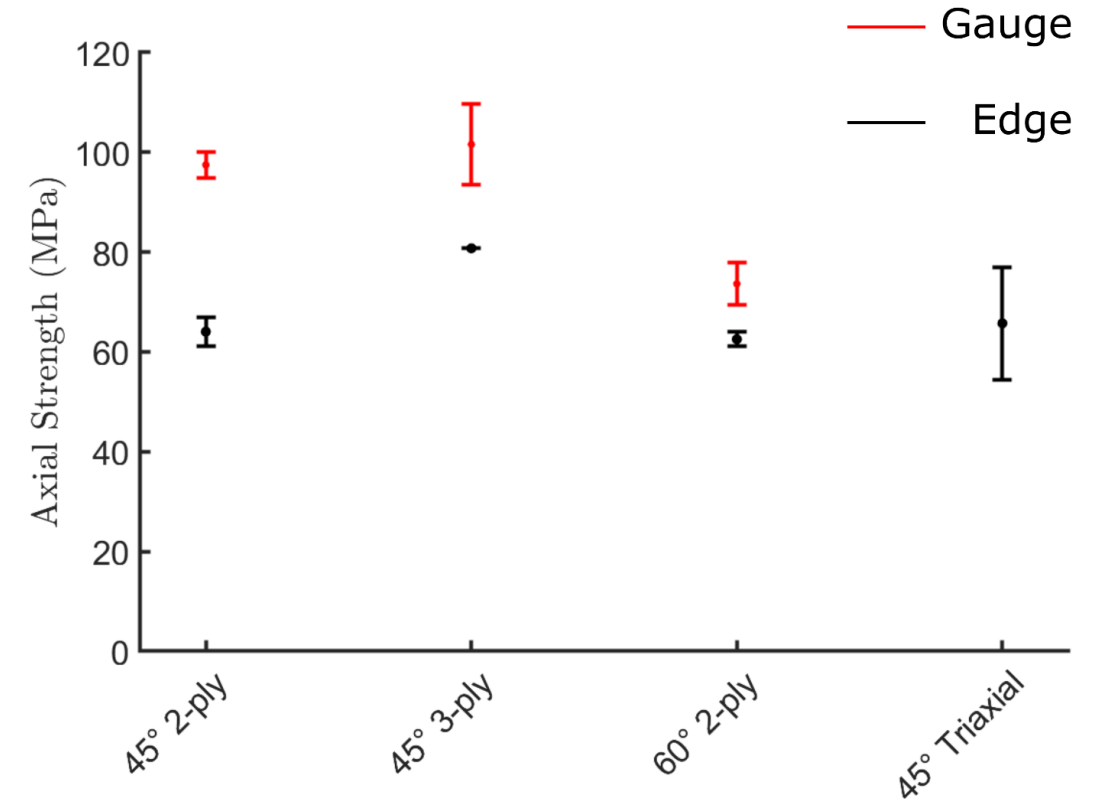
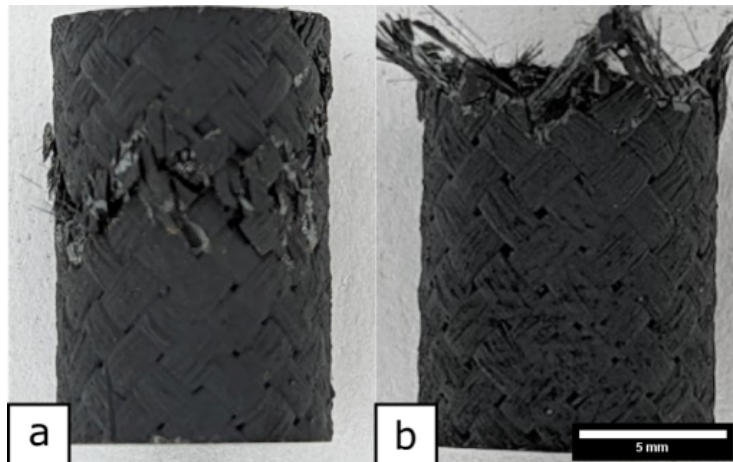


Two forms of failure identified:

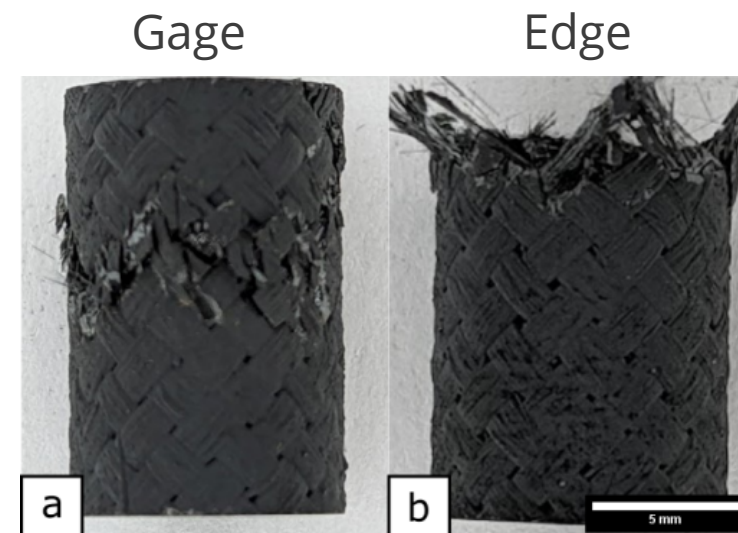
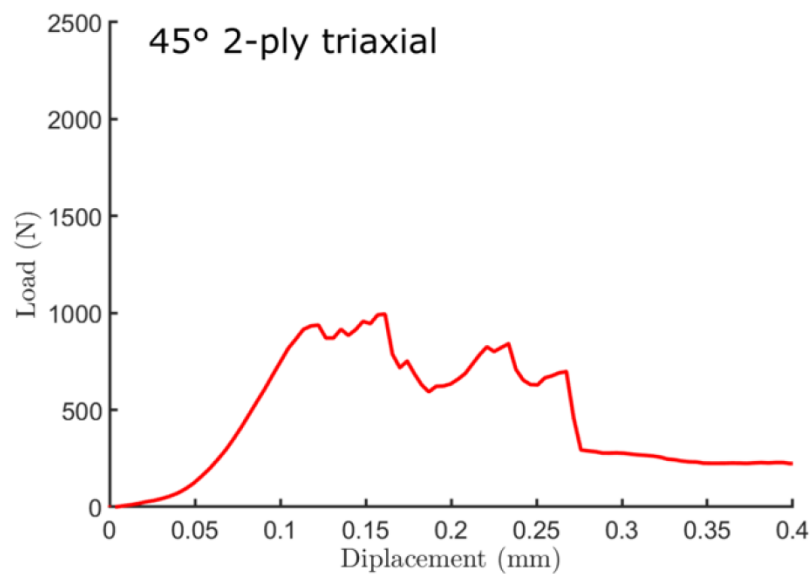
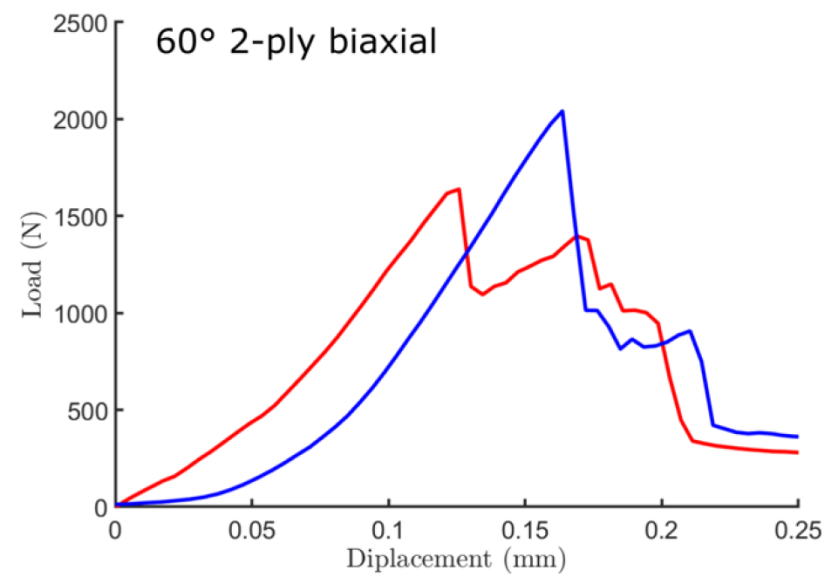
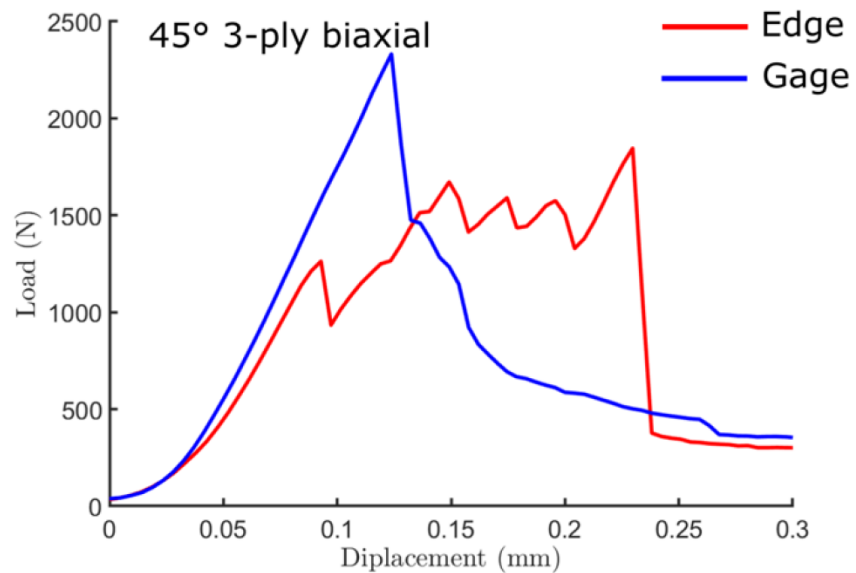
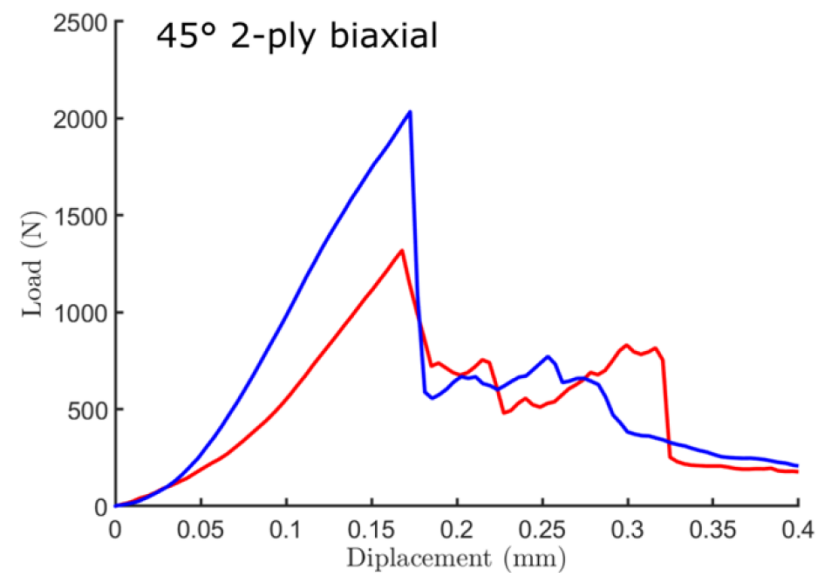
1. Macroscopic failure away from the loading surface (gauge)
2. Macroscopic failure in the vicinity of the loading surface (edge)

Gauge

Edge



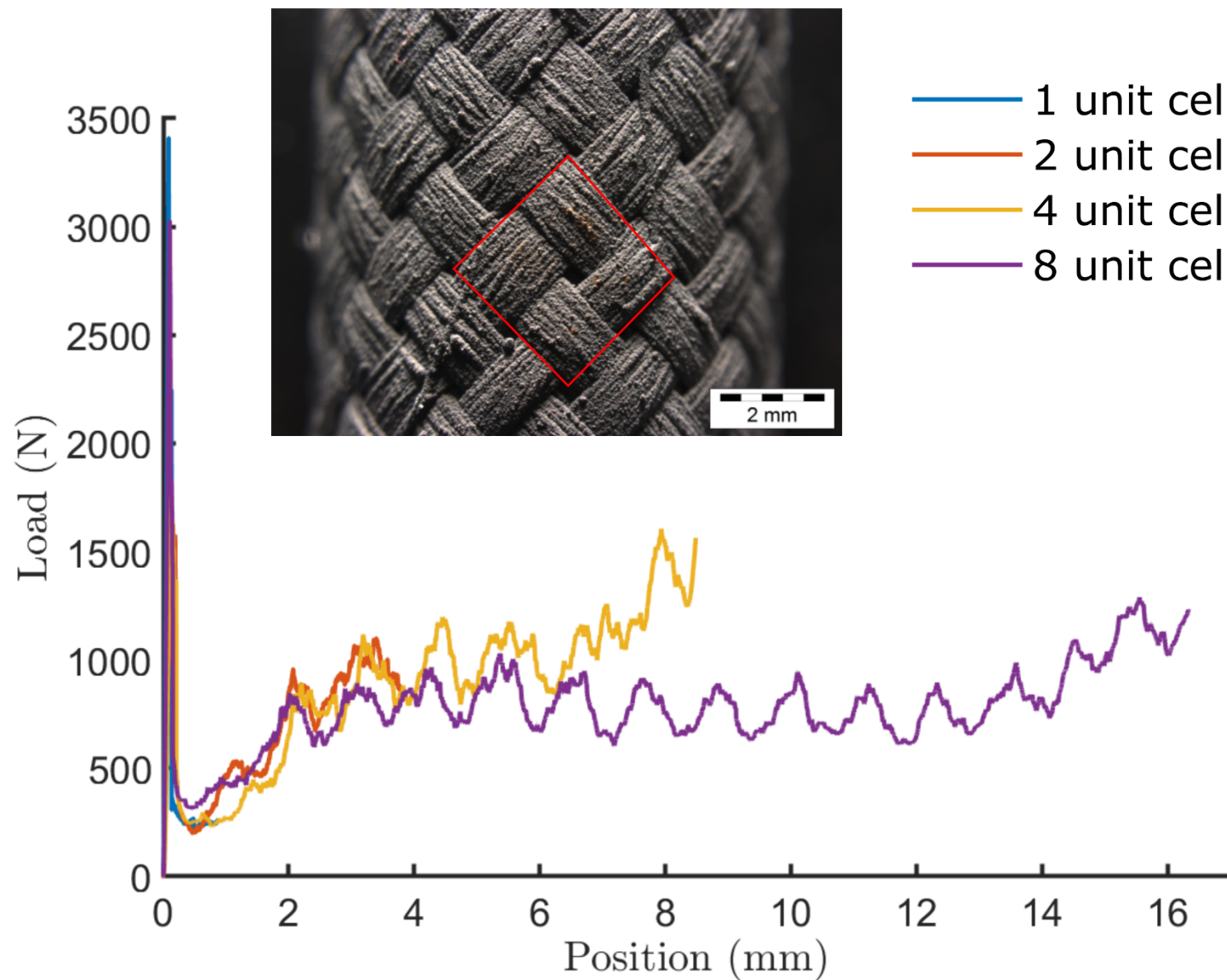
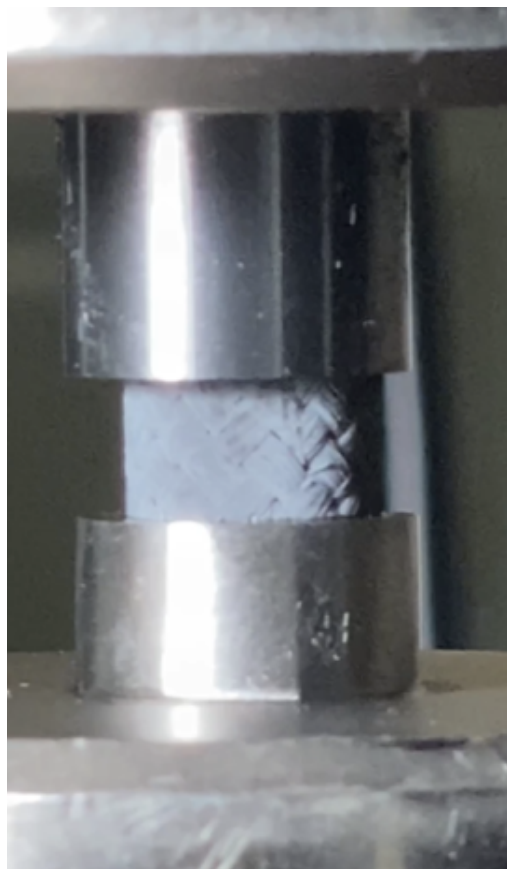
Axial Compression Load-Displacement Curves



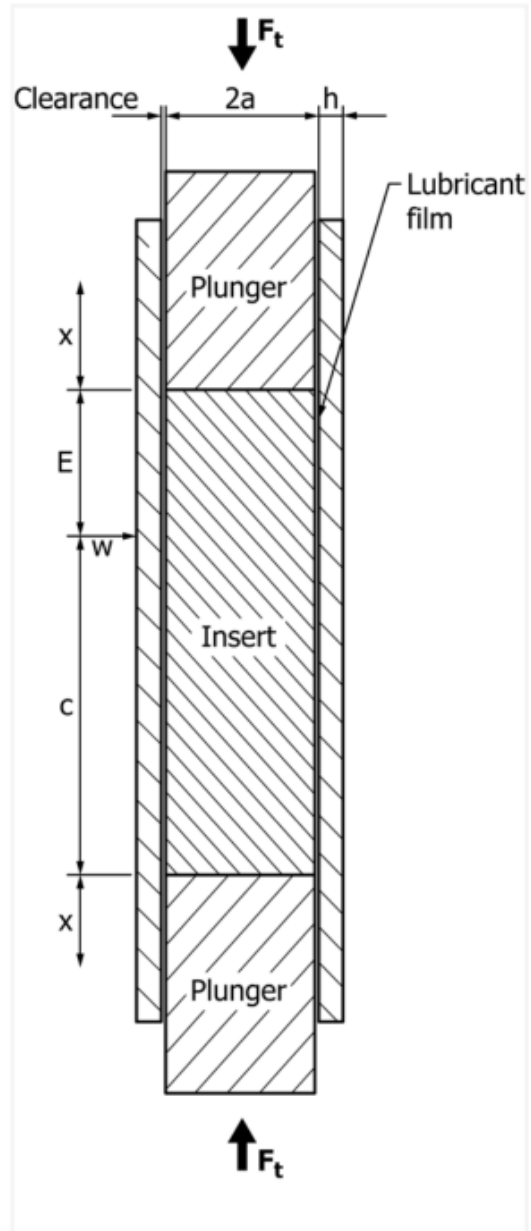
Varying Unit Cell Axial Compression



2
unit
cell



Tensile Hoop Burst



Composite is loaded via internal pressurization from radial expansion of an elastomeric insert

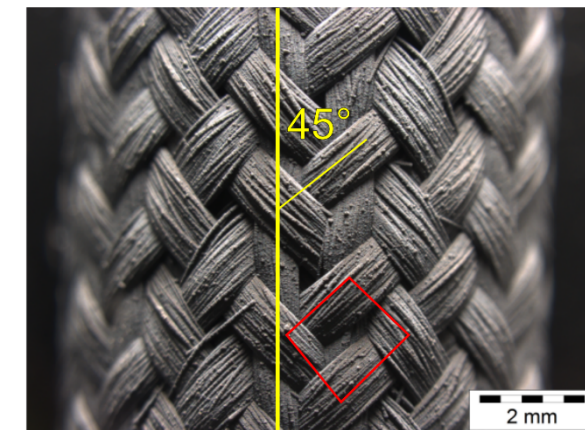
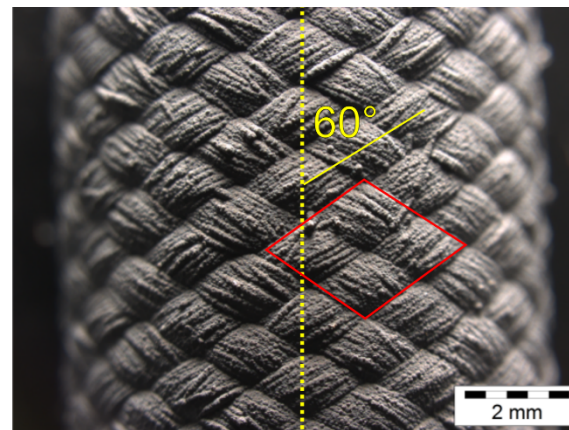
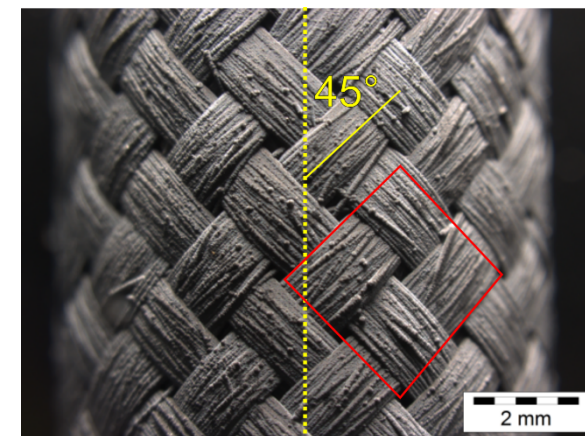
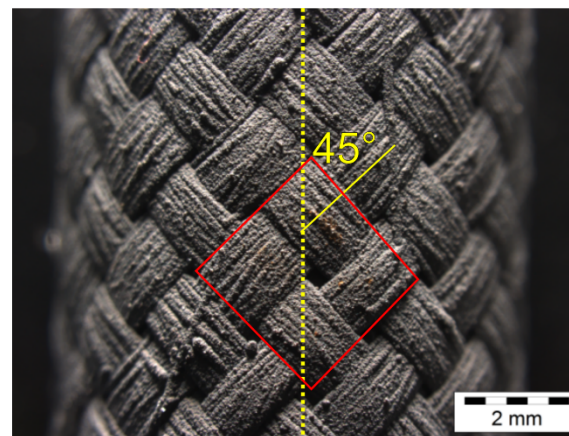
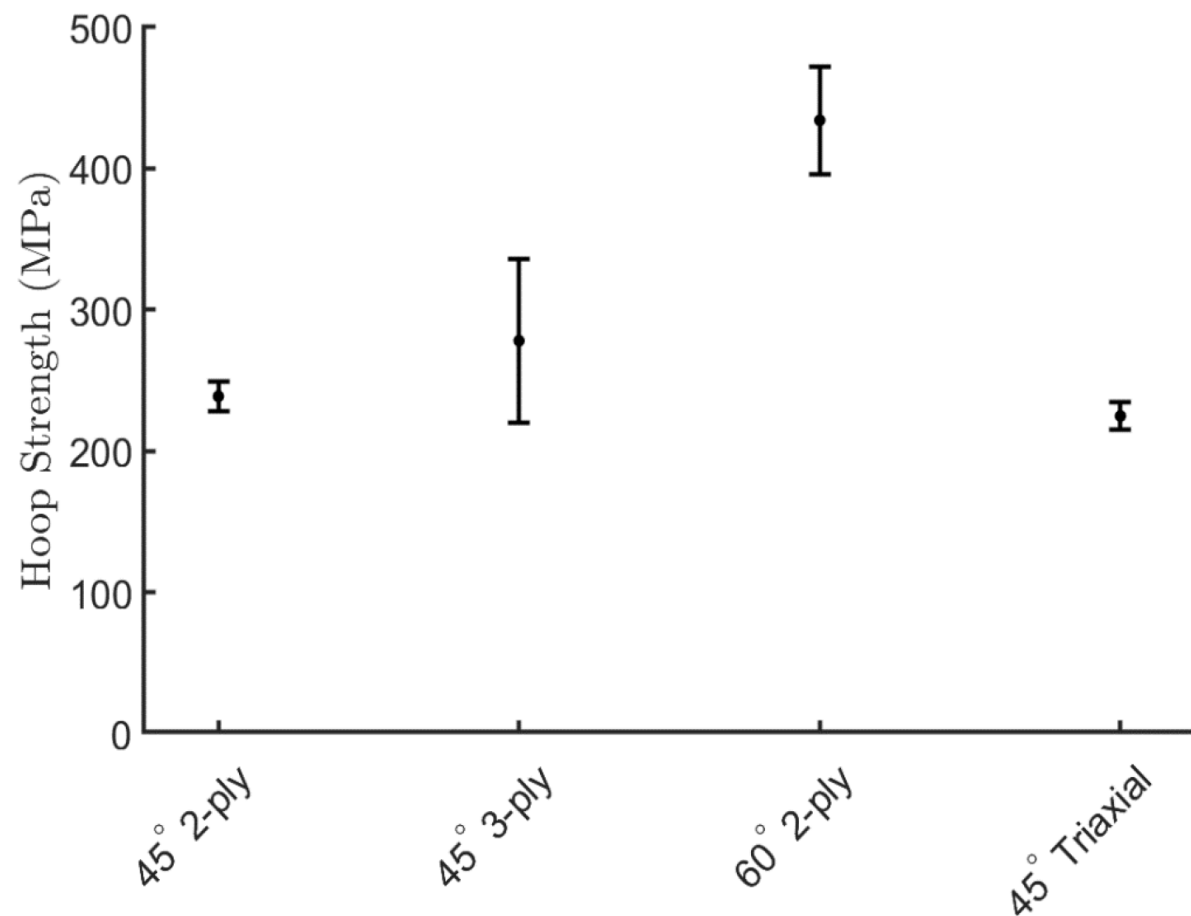
Elastomeric insert expands under uniaxial compressive loading of pushrods and exerts a uniform radial pressure on the inside of the SiC-SiC tube

Hoop tensile strength determined from the resulting maximum pressure

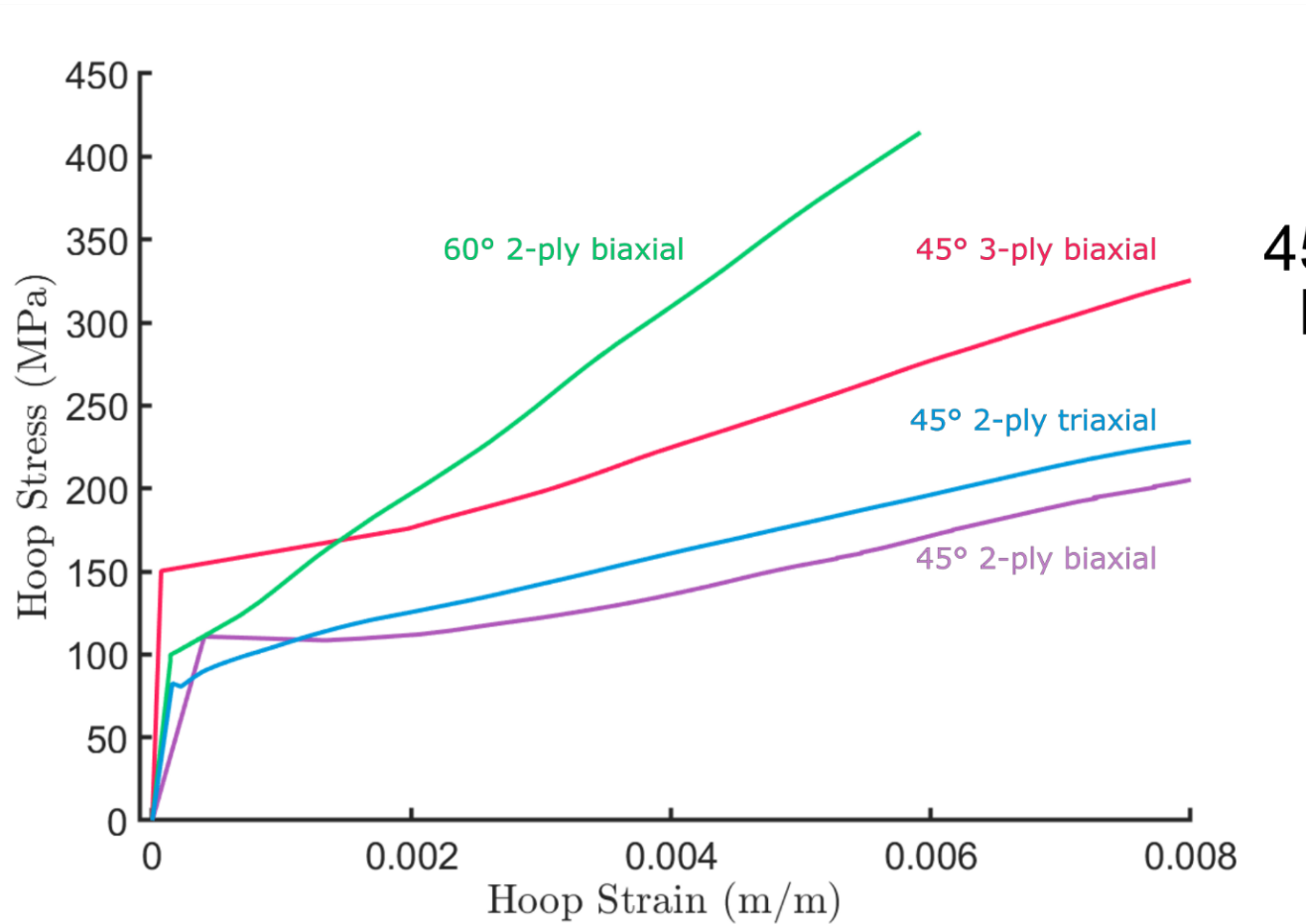
$$P = \frac{F}{\pi r_i^2}$$

$$\sigma_{hoop} = \eta_m P \frac{2r_i^2}{(r_o^2 - r_i^2)}$$

Tensile Hoop Burst – Strength Results



Tensile Hoop Burst – Stress/Strain Curves



45° 2-ply
biaxial



45° 3-ply
biaxial



60° 2-ply
biaxial



45° 2-ply
triaxial



Tensile Hoop Burst - 45° 3-ply Biaxial Failure Mode

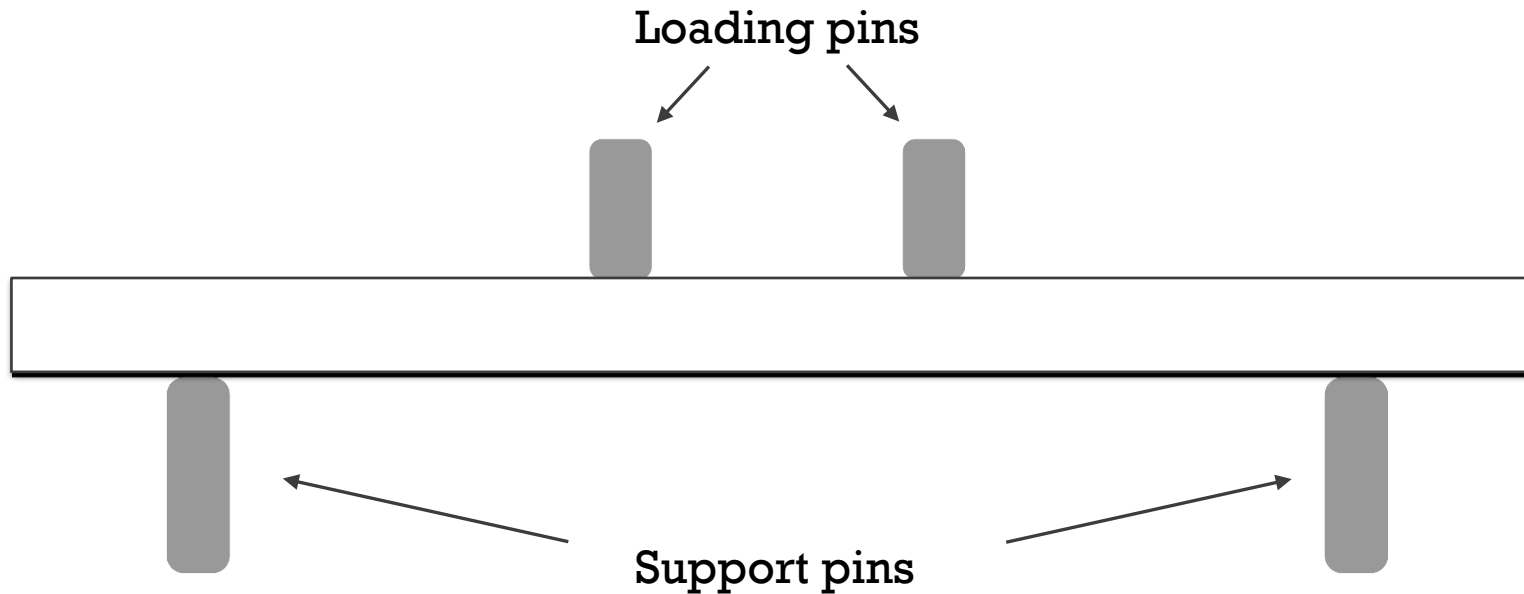


$t = 0s$

$t = 0.2s$

$t = 0.4s$

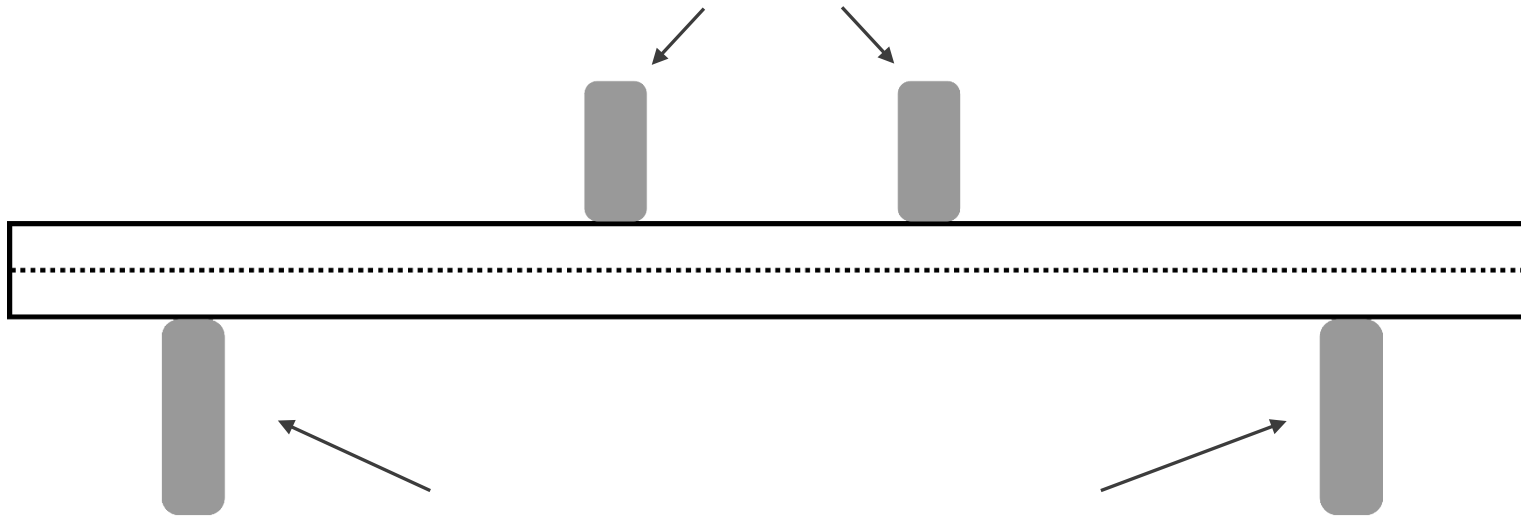




Failure of a ceramic specimen is often a function of a critical surface flaw.

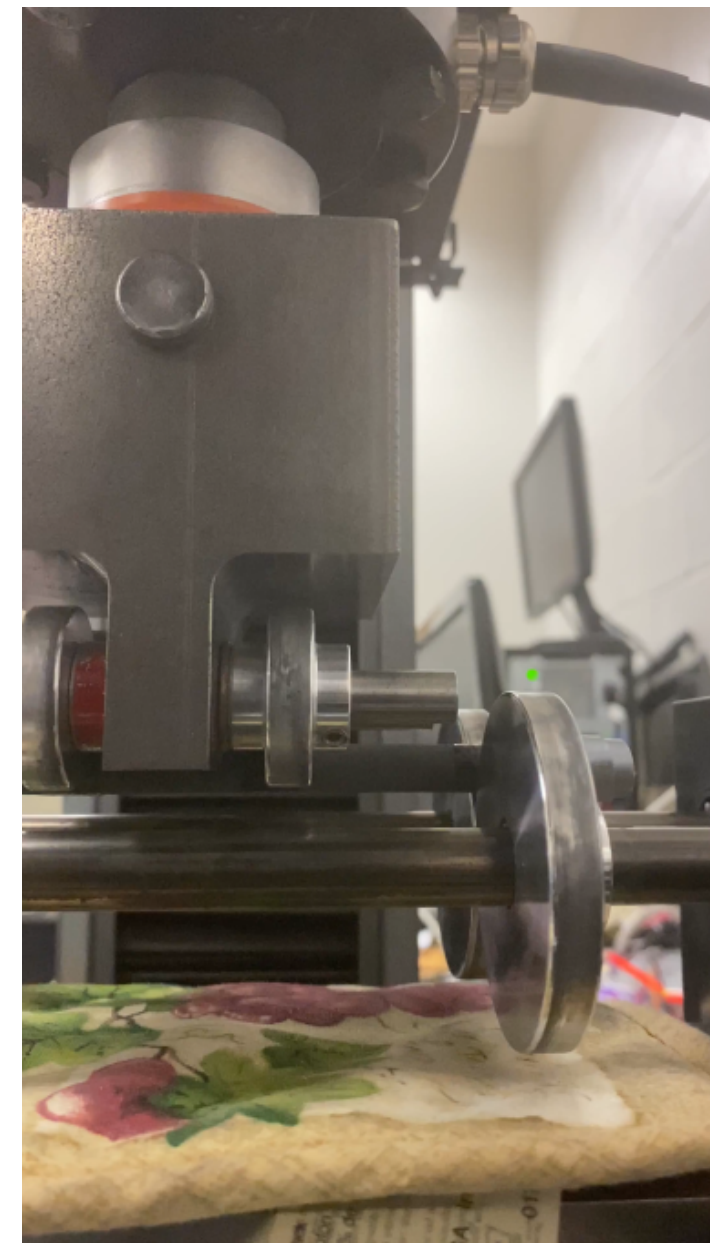
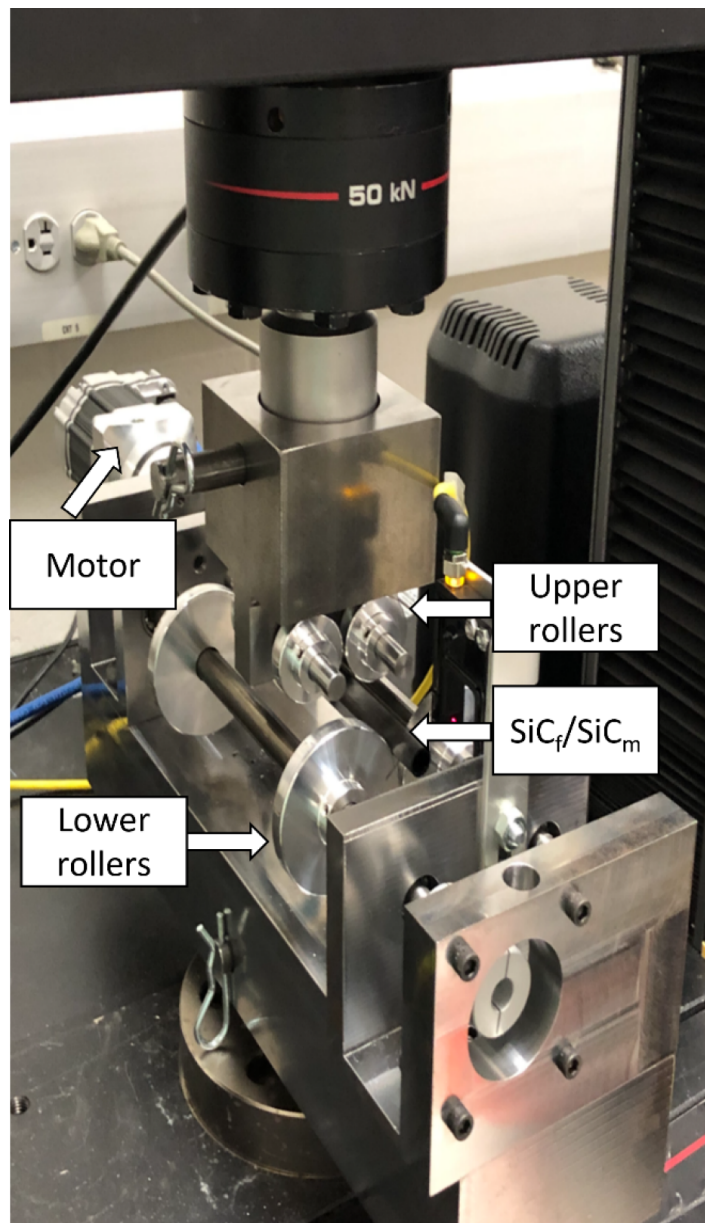
These critical flaws require a tensile stress state to grow.

The traditional 4-point bend test does not always subject the critical flaw to a tensile stress state.

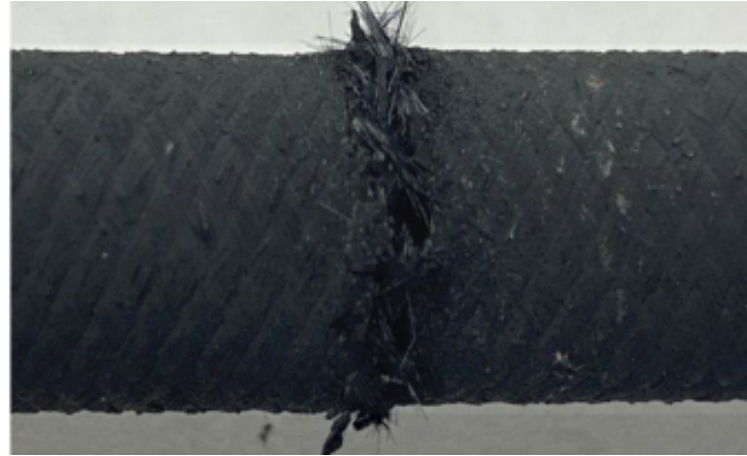


Instead, rotate the specimen as it is being loaded via rollers and it will have a higher probability to expose the critical flaw.

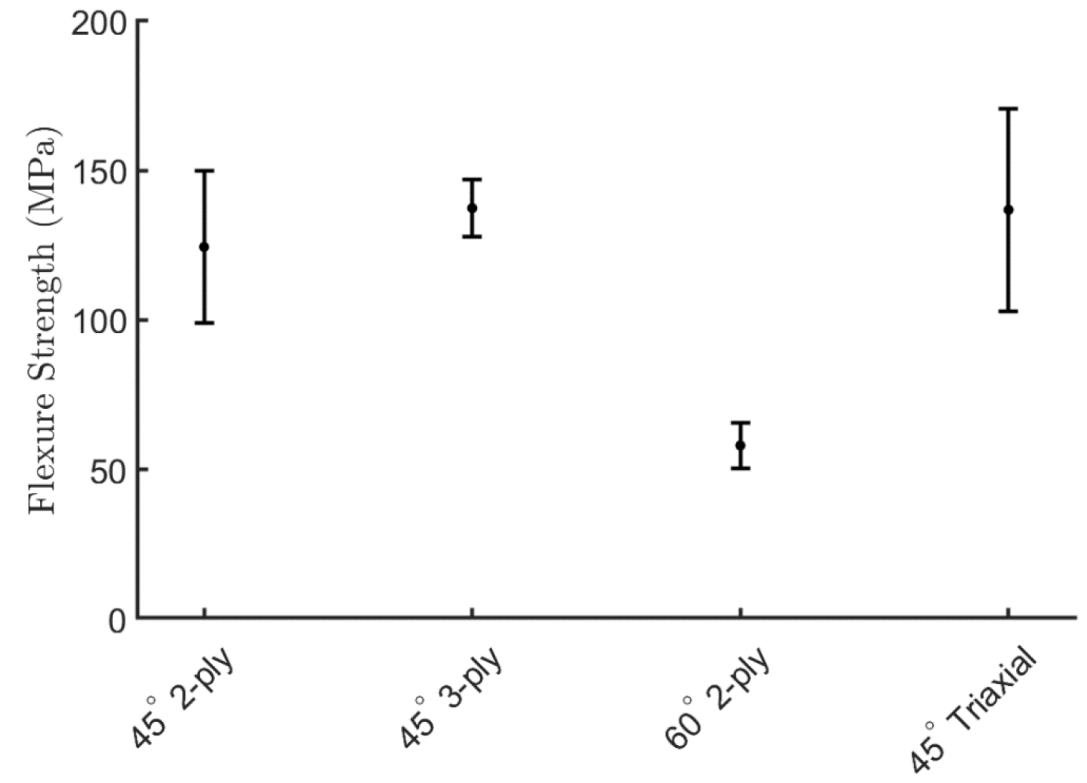
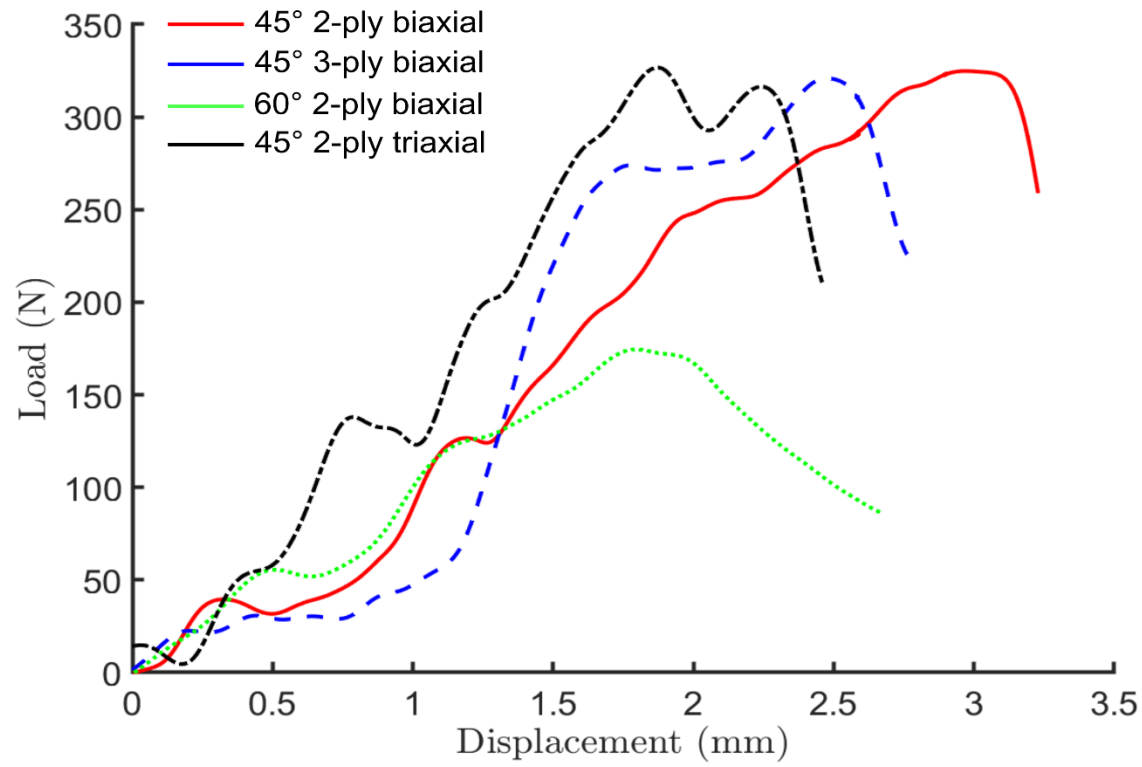
Novel Rotating Flexural Test Design



Flexural Test - Results



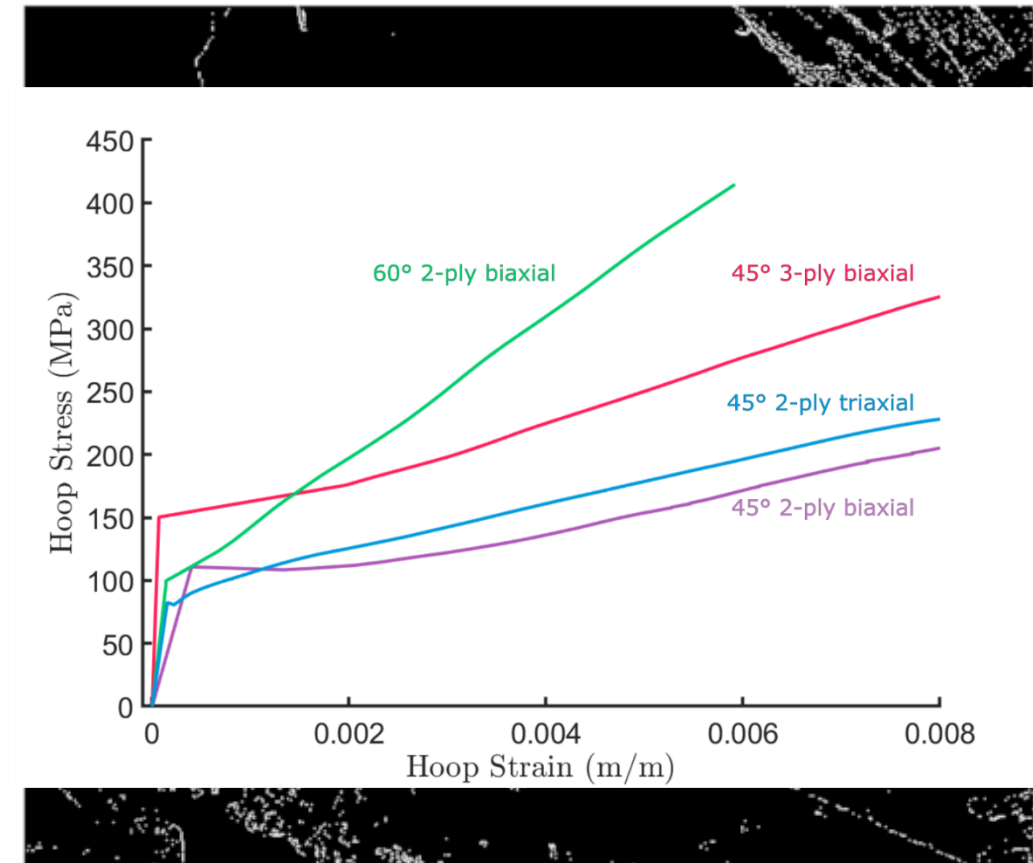
$$\sigma_f = \frac{P_{max}(L_o - L_i)r_o}{4 \left[\frac{\pi(r_o^4 - r_i^4)}{4} \right]}$$



Summary



- Heterogeneity was quantified using digital image processing algorithms and other methods
 - Fiber diameter
 - Porosity Distribution
 - Braiding angle
 - Nanoindentation
- Used a variety of loading scenarios and stress states to identify the composite's architecture influences
 - Axial compression
 - Tensile hoop burst
 - Novel rotating flexural



Acknowledgements



**Sandia
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