



# Rate-dependent dynamic cylindrical cavity expansion equations for conical- and ogival-nosed projectiles

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Zherui Guo

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**Zherui Guo**

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**Idaho National Laboratory  
Idaho Falls, Idaho 83415**

**<http://www.inl.gov>**

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9 **Rate-dependent dynamic cylindrical cavity expansion**  
10 **equations for conical- and ogival-nosed projectiles**

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13 **Zherui Guo<sup>\*1</sup>**  
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17 \*Corresponding author  
18 Zherui Guo  
19 Tel.: +1 (208) 526-3607.  
20 E-mail address: [Zherui.Guo@inl.gov](mailto:Zherui.Guo@inl.gov)  
21 ORCiD: 0000-0002-8832-8435

22  
23  
24  
25 <sup>1</sup>Idaho National Laboratory, Idaho Falls, ID, USA  
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1  
2 **Abstract**  
3

4 The dynamic cylindrical cavity expansion model for a rate-dependent target material was  
5 previously derived by Warren to examine the effects of strain-rate sensitivity on the radial stress  
6 acting on a perforating projectile. However, the equations presented were largely analytical and  
7 were not further applied to predict the ballistic performance of ductile target plates. The current  
8 work expands on Warren's derivation to model the dynamics of conical and ogival geometries,  
9 and the rate-dependent model is compared to prior experimental results of 7.62-mm APM2  
10 rounds impacting 6061-T6511 aluminum alloy plates. Results show that including rate effects  
11 improves the ballistic performance prediction, even for a marginally strain-rate sensitive material  
12 such as Al6061-T651. However, existing semi-empirical variations of the cavity expansion  
13 model can provide the same degree of accuracy if target material rate-sensitivity parameters are  
14 not readily available.

15  
16 **Keywords:** Ballistic impact. Cavity expansion. Strain-rate sensitivity. Plate perforation.  
17

18 **Introduction**  
19

20 In an early series of works [1–4], Forrestal and colleagues derived cylindrical cavity expansion  
21 (CCE) equations to model the dynamic response of rate-independent, strain-hardening target  
22 plates under ballistic impact. In the CCE model, the target plate is idealized as infinitesimally  
23 thin layers (i.e. plane strain conditions) perpendicular to the direction of projectile perforation. A  
24 set of closed-form equations were then derived <sup>[5]</sup> to predict the ballistic performance of  
25 aluminum target plates against ogival- and conical-nosed long rod projectiles at normal incidence  
26 impact using the rate-independent CCE model. Subsequent works [6–12] used a semi-empirical  
27 variation of the CCE model to successfully predict the ballistic impact performance aluminum  
28 and steel plates impacted by armor-piercing rounds.

29  
30 In a parallel series of works, Forrestal et al. derived closed-form spherical cavity-expansion  
31 (SCE) equations for long rods penetrating deep metallic targets [13–15]. In contrast to plate  
32 perforation dynamics, which can be accurately described by CCE models, deep target penetration  
33 dynamics are more accurately represented by SCE models. Earlier SCE formulations assumed  
34 rate-insensitivity of the target materials, and the depth of penetration results were well-predicted  
35 for aluminum targets. Warren & Forrestal [16] later developed closed-form SCE models for the  
36 deep target penetration of rate-dependent materials (rSCE).

37  
38 Following the rSCE model, Warren [17] then derived cylindrical cavity expansion equations for  
39 rate-dependent materials (rCCE). While the radial expansion stress was shown to increase as a  
40 result of rate effects, the rCCE model itself was not applied to predict the results of ballistic  
41 perforation experiments.

42  
43 In this work, equations for the rCCE model are derived and applied to the ballistic perforation of  
44 ductile target plates. The rate-independent CCE equations are presented in the first section to  
45 form a basis for analysis, before rate effects (as derived by Warren) are included in the  
46 subsequent section. Equations are then derived for the target resistive forces acting on generic  
47 conical- and ogival-nosed projectiles, before model predictions are presented for 7.62-mm  
48 APM2 rounds impacting 6061-T6511 aluminum alloy plates. Different variations of the

1 cylindrical cavity expansion model are compared to ballistic experimental results to explain the  
2 success of semi-empirical model predictions as demonstrated in previous works.  
3  
4

5 **Derivation of model equations**  
6

7 *Rate-independent cylindrical cavity expansion (CCE)*  
8

9 The stress-strain law for a typical strain-hardening material is given by the modified Ludwik  
10 equation  
11

12 (1)  
13

14 In Equation 1,  $\sigma$  is the true stress,  $E$  is the target elastic modulus,  $\varepsilon$  is true strain,  $Y$  is the quasi-  
15 static target yield stress, and  $n$  is a strain-hardening exponent obtained from fitting Equation 1 to  
16 quasi-static uniaxial compression test data to high strains greater than 1.0 [18]. The exact  
17 solution for the radial stress  $\sigma_r$  at the cavity surface when expanding radially with velocity  $V$  is  
18 the sum of two loading responses: quasi-static strain hardening and target inertia.  
19  
20

21 (2)  
22  
23 (3)  
24  
25 (4)  
26

27 In Equations 2 to 4,  $\rho_t$  is the target density,  $\nu$  is the Poisson's ratio, and  $B$  is a target radial inertia  
28 given by  
29  
30

31 (5)  
32  
33  
34 (6)  
35

36 Since  $B$  is velocity-dependent, an empirical parameter  $B_0$  is usually curve-fitted so that a closed-  
37 form solution may be obtained as [3]  
38  
39

40 (7)  
41

42 Assuming a frictionless perforation process [5], the perforation dynamics can be modeled with  
43 the solution of the differential equation  
44  
45

46 (8)  
47

48 where  $z$  is the penetration depth,  $V_z$  the axial velocity, and  $m_p$  is the projectile mass given as

1  
2  
3 (9a)  
4  
5  
6  
7

(b)

8 with  $a$  being the projectile radius,  $\rho_p$  the projectile density,  $L$  the projectile shank length,  $k_l$  the  
9 nose shape factor, and the projectile nose length. From Ref. [5], the axial force  $F_z$  acting on the  
10 projectile nose is given as

11  
12  
13 (10)  
14

15  $N$  is a dimensionless nose shape factor dependent on the target inertia, the derivation of which  
16 will be expanded upon in subsequent sections. Solution of Equations 8 to 10 gives

17  
18  
19 (11)  
20

21 The ballistic limit velocity  $V_{bl}$  is obtained by integrating the perforation depth from to the plate  
22 thickness , and the perforation velocity  $V_z$  from to .

23  
24  
25 (12a)  
26  
27

(b)

31 (c)  
32  
33 For a striking velocity  $V_s$  above the ballistic limit  $V_{bl}$ , the residual velocity  $V_r$  can be calculated  
34 by integrating  $V_z$  from to in Equation 12 to give

35  
36  
37 (13)  
38  
39

(14)

42 Multiplying Equation 12c by and substituting into Equation 14, the residual velocity can be  
43 explicitly expressed as a function of  $V_s$  and  $V_{bl}$

44  
45  
46 (15)  
47

48 From Equation 12a,  $C$  is a dimensionless term related to the inertia of the target material via  $B_0$   
49 in Equation 7. A three-term series approximation for the exponential terms in Equations 12c and

1 15 gives  
2  
3  
4  
5  
6  
7  
8

(16)

(17)

9 These series approximations allow for the inertial effects via the  $C$  term to be easily excluded for  
10 certain materials where target inertia is negligible during the perforation process. Prior studies  
11 often excluded the effects of target inertia in Equation 16 [5,19], but instead used an empirical  
12 constant  $\kappa$  for ballistic performance prediction such that

13  
14  
15  
16  
17 *Rate-dependent cylindrical cavity expansion (rCCE)*  
18

(18)

19 Warren & Forrestal [16,17] subsequently derived a strain-rate sensitive perforation model,  
20 starting with a rate-decoupled stress-strain law [20,21]:  
21  
22  
23  
24  
25

(19)

(20)

26 In Equations 19 and 20,  $Y_d$  is the dynamic yield stress of the target material. The rate-sensitivity  
27 modification to the post-yield stress consists of a curve-fitted rate parameter  $\alpha$ , reference strain-  
28 rate  $\dot{\epsilon}_0$ , and strain-rate sensitivity exponent  $\beta$ . When  $\alpha = 0$ , the rate-independent modified Ludwik  
29 equation (Equation 1) is recovered. The exact solution for the rate-dependent radial stress is now  
30 the sum of three loading responses: quasi-static strain hardening, target inertia, and strain-rate  
31 sensitivity. The equations are given as  
32  
33  
34

(21)

(22)

(23)

42 where  $R_c$  is the cavity radius, usually taken as the projectile shank radius for the cylindrical  
43 cavity expansion model. The subscript  $R$  indicates rate-dependent forms of the variable. Warren  
44 showed that strain rate effects have a negligible influence on the strain-hardening portion of the  
45 stress-strain curve [17]. Therefore, Equation 22 can be approximated with Equation 3 without  
46 significant deviation, essentially decoupling the strain-hardening effects from the strain rate  
47 effects to give  
48

1  
2  
3  
4  
5 (24)

## Derivation of resistive force equations

7 For a rate-dependent target material, the axial force acting on the projectile nose after a change  
8 of variable is

9  
10  
11 (25)  
12

13 Equation 25 requires the following set of integrals to be solved

14  
15  
16 (26)  
17  
18  
19 (27)  
20  
21  
22 (28)  
23  
24  
25 (29)  
26

27 By inspection, a general form of the set of integrals can be expressed as

28  
29  
30 (30)  
31  
32

33 *Conical-nosed projectiles*

34

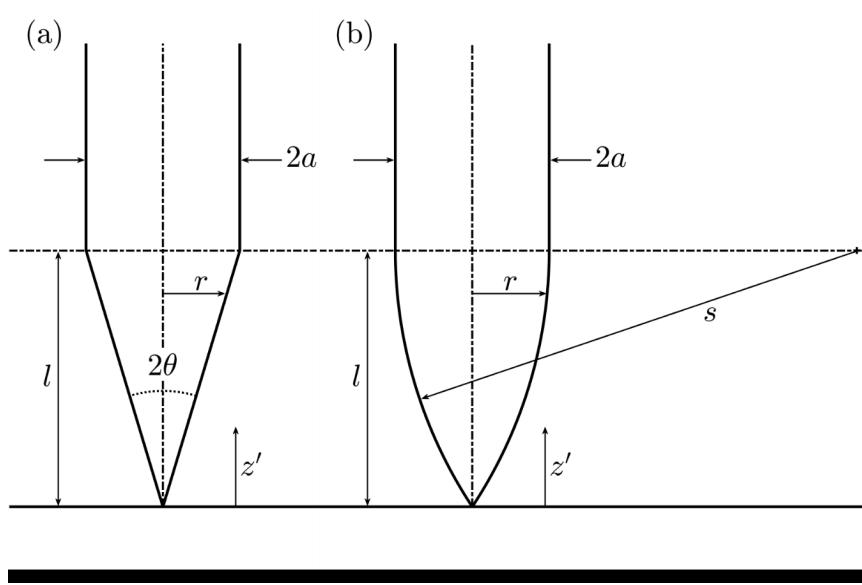


Figure 1: Dimensions of (a) conical-nosed and (b) ogival-nosed projectiles.

From Figure 1a, for a conical-nosed projectile with apex angle  $2\theta$ . Solving the integrals in Equations 27 to 29, we get

(31)

Factoring out  $I_0$  from Equation 26 and defining gives

(32)

(33)

(34)

### Ogival-nosed projectile

From Figure 1b, for an ogival-nosed projectile

(35)

(36)

The variables  $r$ ,  $s$ , and  $m$ , where  $m$  is the caliber-radius-head (CRH) of the projectile [5]. The special case of  $m = 2$  for Equation 30 was solved by Forrestal & Warren [5] to give

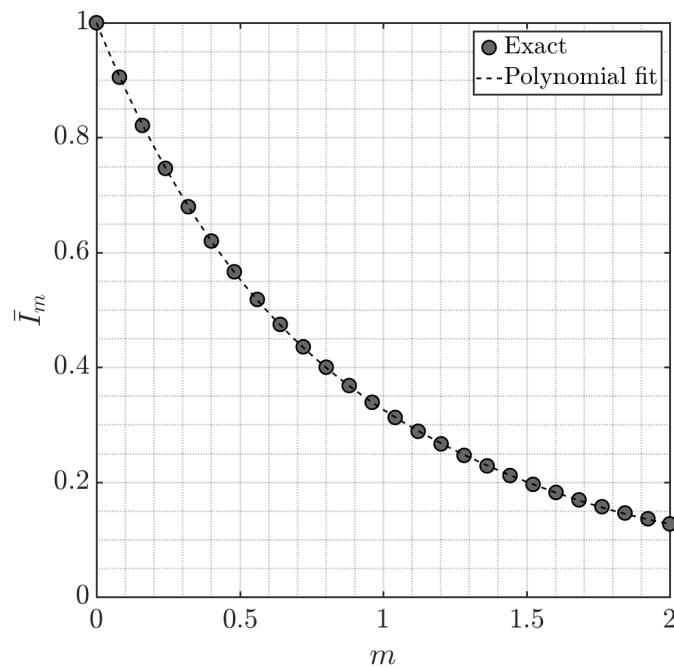
1  
2  
3 (37a)  
4  
5  
6  
7

(b)

8 Unlike the conical-nosed projectile, the solution for  $I_\beta$  in this case requires numerical integration,  
9 since there is no generalized closed-form integral solution. Using exact solutions at  $= 1$  and  $= N$   
10 as fixed end constraints, a near-exact polynomial fit for  $\bar{I}_m$  can be obtained for numerical solutions  
11 to Equation 30 (Figure 2).  $\bar{I}_m$  can be evaluated with

12  
13  
14 (38)  
15

16 The fit coefficients  $C_1$  to  $C_4$  for different caliber-radius-head values are given in Table I.  
17



18  
19  
20 **Figure 2:** Polynomial fit of numerical solutions to integral for  $= 3$ .  
21  
22

23 **Table I:** Coefficients for different CRH values.  
24

CRH	$C_4$	$C_3$	$C_2$	$C_1$
1	0.0171	-0.113	0.340	-0.590
2	0.0321	-0.211	0.599	-1.008
3	0.0453	-0.293	0.798	-1.225
4	0.0568	-0.361	0.953	-1.371
5	0.0668	-0.420	1.080	-1.481
6	0.0758	-0.471	1.187	-1.568

7	0.0838	-0.516	1.280	-1.641
8	0.0912	-0.557	1.362	-1.703
9	0.0979	-0.594	1.435	-1.757
10	0.1041	-0.628	1.501	-1.805

1  
2  
3 Solution with the rate-dependent force (Equation 32) for either nose geometry gives  
4  
5  
6  
7  
8  
9 Analytical solutions for Equation 39 again require numerical integration to calculate  $V_{bl}$ , since  $\beta$   
10 is a real, non-integer value.  
11  
12

(39)

### 13 Numerical results

14 Numerical results for 6061-T6511 aluminum alloy are presented and compared to prior ballistic  
15 plate perforation data by Ryan et al. [10]. Warren and Forrestal [16] gave material properties for  
16 alloy 6061-T6511 as  $E = 68.9$  GPa,  $\nu = 0.33$ ,  $Y = 276$  MPa,  $\rho_t = 2710$  kg/m<sup>3</sup>,  $n = 0.072$ . The rate-  
17 sensitivity constants were obtained from prior studies across a broad range of strain rates [16],  
18 from quasi-static compression ( $10^{-3}$  s<sup>-1</sup>) to pressure-shear experiments ( $10^5$  s<sup>-1</sup>). These values are  
19 reported as  $\alpha = 32.0$  MPa,  $\beta = 0.348$  (given as  $m$  in reference), and  $\gamma = 1000$  s<sup>-1</sup>. The projectile is a  
20 7.62-mm APM2 round with radius  $a = 3.93$  mm and mass  $m_p = 10.8$  g. The AP round has a CRH  
21 of approximately 3.0, giving  $N = 0.1272$  and  $\gamma = 0.6589$ . Five variations of the CCE model are  
22 plotted in Figure 3 for comparison, and their respective sums of squared errors (SSE) are  
23 reported as follows:  
24

- 25 I) Rate-independent ( $\alpha = 0$ ), no target inertia ( $C = 0$ ), SSE =  $1.59 \times 10^6$ ;
- 26 II) Rate-independent ( $\alpha = 0$ ), with target inertia, SSE =  $4.04 \times 10^5$ ;
- 27 III) Rate-dependent, no target inertia ( $C = 0$ ), SSE =  $8.08 \times 10^5$ ;
- 28 IV) Rate-dependent, with target inertia, SSE =  $1.67 \times 10^5$ ; and
- 29 V) Rate-independent ( $\alpha = 0$ ), no target inertia ( $C = 0$ ), empirical fit ( $\kappa = 1.211$ ), SSE =  
30  $1.33 \times 10^5$ .

31

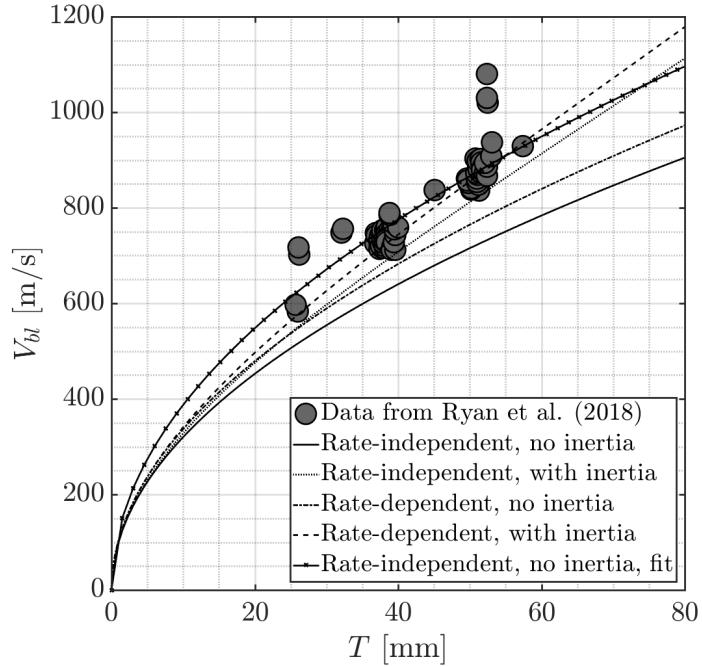


Figure 3: Variations of CCE model predictions for 7.62-mm APM2 round impacting 6061-T6511 aluminum alloy plates of different thicknesses.

The largest difference in ballistic performance appears to stem from target inertia effects, especially for much thicker targets, since  $C$  is a quadratic function of the target thickness. The rate-dependent model with inertia (Variation IV) provides the best prediction of the experimental data without empirically determined constants. It appears that ballistic performance predictions may be improved even though aluminum is often considered a relatively rate-insensitive material – this improvement in prediction may be true across the board for other more rate-sensitive materials.

Naturally, the empirically scaled rate-independent model (Equation 18, Variation V) in Figure 3 best fits the experimental data, as is the purpose of a least-squares fit. Such an empirical scaling has been previously used to model the plate perforation of AP rounds on an extensive range of aluminum alloys to great degrees of accuracy [6–11,22]. However, a comparison of Variations IV and V shows that the inclusion of rate effects results in a nonlinear response that cannot be linearly scaled, and such effects may be more pronounced for other materials that are considerably more rate-sensitive than aluminum.

Nonetheless, the frequent lack of experimental data across a broad range of strain rates to obtain the constants  $\alpha$  and  $\beta$  for any target material has to be a consideration when including rate-sensitivity effects for ballistic performance prediction. Further work examining more rate-sensitive targets would provide further insight into the rate effects on the perforation dynamics in order to improve ballistic performance predictions.

1      **Conclusions**

2  
3      Ballistic perforation equations for the rate-dependent cylindrical cavity expansion model were  
4      solved for conical- and ogival-nosed projectile geometries using material rate-sensitivity  
5      constants presented by Warren & Forrestal. Numerical results were presented for 7.62-mm  
6      APM2 rounds impacting relatively rate-insensitive 6061-T6511 aluminum alloy plates and  
7      compared with experimental ballistic impact data. The rate-dependent cavity expansion model  
8      was in good agreement with experimental data without the need for empirically determined  
9      constants, but simplified semi-empirical cavity expansion models derived with quasi-static  
10     properties provide an accurate performance prediction. This latter approach can be advantageous,  
11     as fewer material parameters are required for first order approximations.

12  
13      **Declaration of Competing Interest**

14  
15      The author(s) declare no known competing financial interests or personal relationships in the  
16     publication of this paper.

17  
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23      expansion model.

24  
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