

Polyhedral finite elements and multiscale error estimation of material models

Joe Bishop

Engineering Sciences Center
Sandia National Laboratories
Albuquerque, NM

Texas Tech University, February 17, 2020

My background

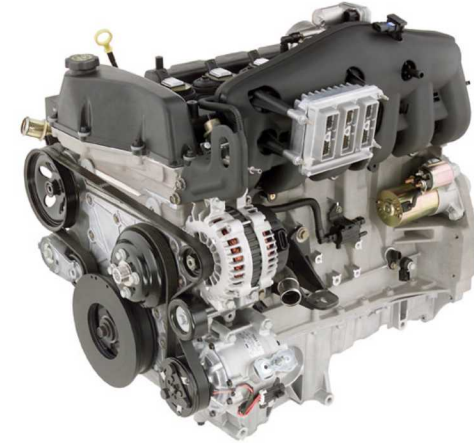
1. Ph.D. Texas A&M Univ., Aerospace Engineering, 1995

2. General Motors, Powertrain Division, 1996 - 2004

- FEA, thermal-structural analysis of engines
- "production environment (rapid design to analysis)
- fatigue (low and high cycle)
- Group Leader of head and block group, 2002 - 2004

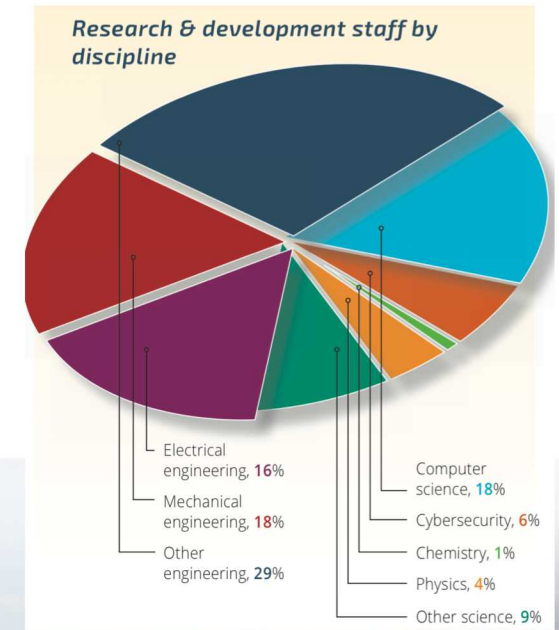
3. Sandia National Laboratories, 2004 - present

- support of product-development engineering
- impact/penetration
- geologic storage of CO₂
- pervasive fracture modeling
- polyhedral finite elements
- meshfree discretizations (rapid design to analysis)
- multiscale modeling, error estimation
- metal additive manufacturing
- department manager, 2018 - present



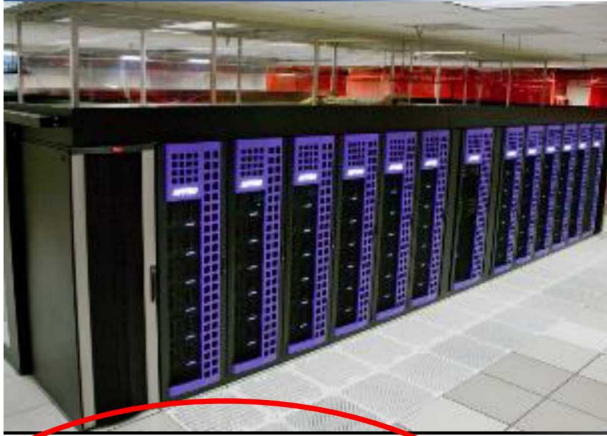
Sandia National Laboratories

- Established in 1945 as Los Alamos Z division
- Sandia Laboratories established in 1949
- Government owned, contractor operated, for DOE's National Nuclear Security Administration (NNSA)
- Responsible for design of non-nuclear components for nuclear weapons
- Sandia teams with the two "physics" laboratories: LANL, LLNL
- First managed by AT&T Bell Labs, 1949-1993
- FFRDC, Federally Funded R&D Center (may perform work for industry)
- \$3.8 billion budget
- 14,000 employees: 2600 B.S, 4600 M.S., 2100 Ph.D.
- 950 students; 250 post-docs

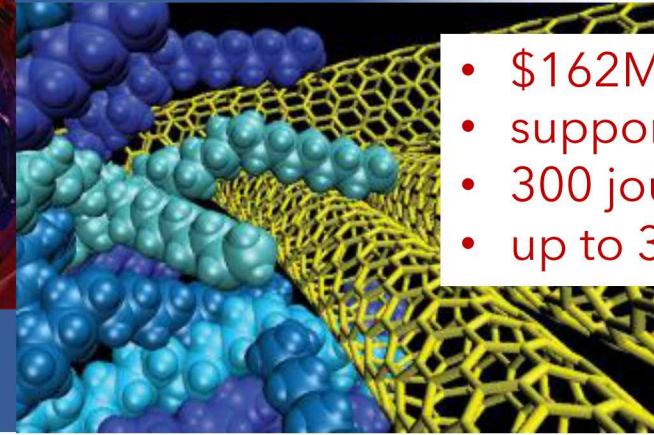


Research Foundations (Lab directed R&D)

Computing &
Information Sciences

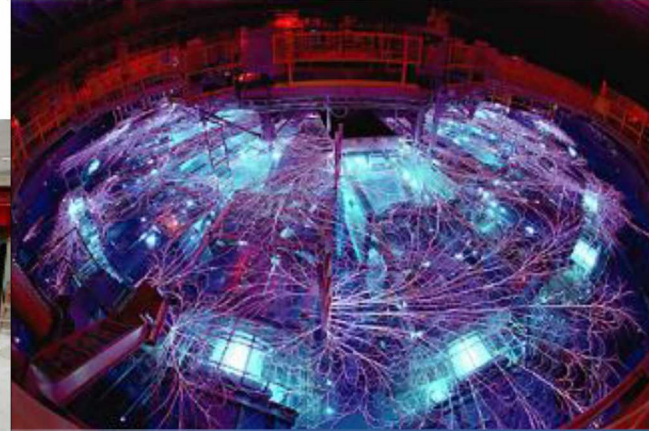


Materials Sciences

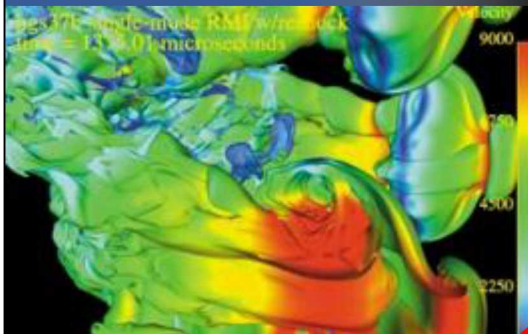


- \$162M in FY19
- supported 130 post docs
- 300 journal pubs
- up to 30% univ. collab.

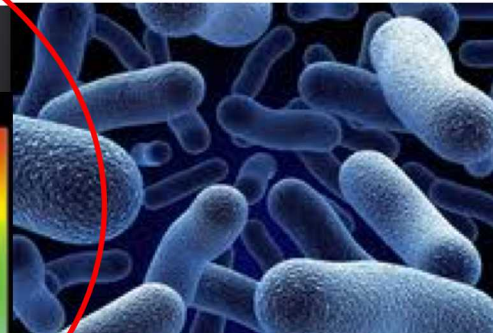
Radiation Effects &
High Energy Density Science



Engineering Sciences



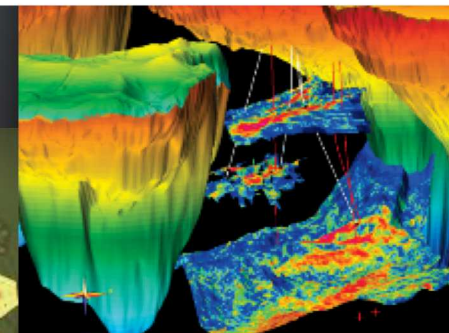
Bioscience



Nanodevices &
Microsystems

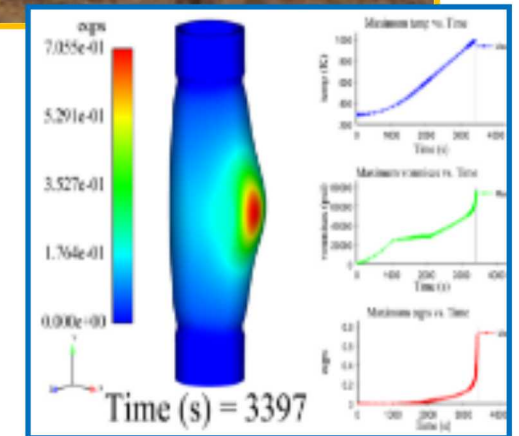


Geoscience



Engineering Sciences Center

- Mission: provide validated, science-based engineering expertise and solutions across the life cycle of products to inform engineering decisions.
- Integrate theory, computational simulation and experimental discovery/validation across length and time scales to develop the technical basis for complex systems.
- 600 staff members
- Core technical areas:
 - shock physics and energetics
 - solid mechanics
 - structural dynamics
 - thermal and fire sciences
 - aerosciences (hypersonics)
 - fluid mechanics



Structural Mechanics and Research Group

- 6 departments:
 - Structural Dynamics (linear)
 - Structural Mechanics (nonlinear)
 - Component Science and Mechanics (engineering mechanics)
 - Computational Material Mechanics (material models)
 - Environments Engineering
 - Computational Shock Physics
- 100 staff members
- Two summer intern institutes:
 - RAMS: Research & Applications of Mechanics of Structures (12 students)
 - NOMAD: Nonlinear Mechanics and Dynamics (18 students)

www.sandia.gov/careers/students_postdocs/

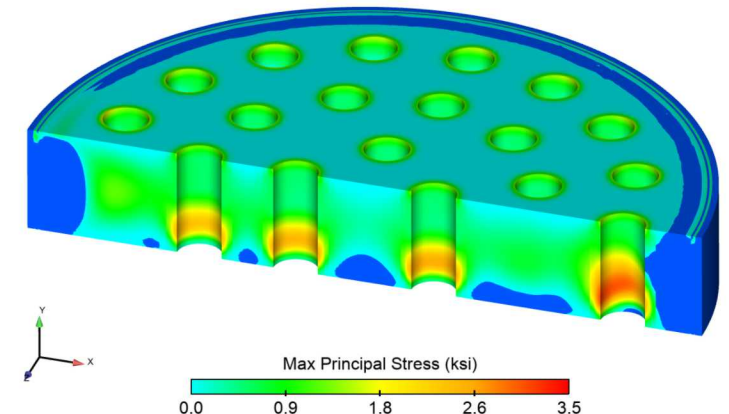
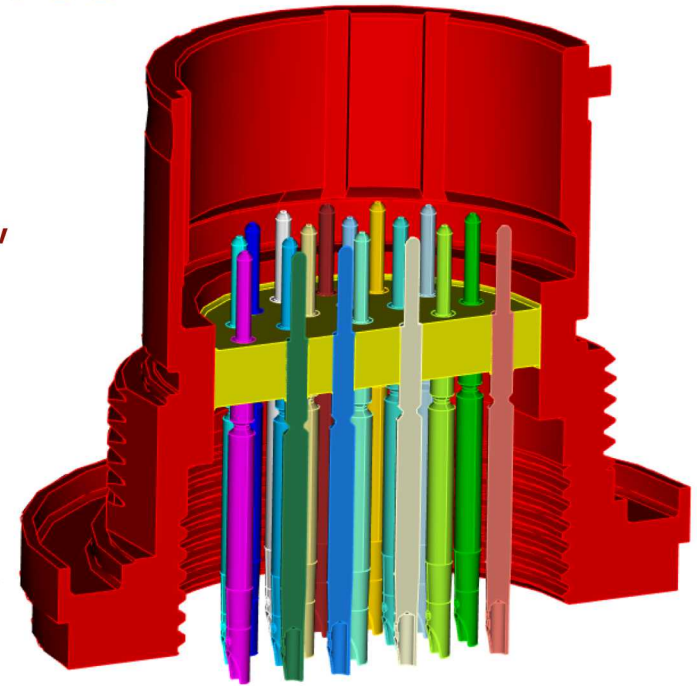
Component Science and Mechanics

- About 15 staff members: 2/3 M.S., 1/3 Ph.D.
- Tightly integrated with "product realization teams"
- FEA analysis, engineering mechanics

Research areas:

- contact mechanics
- fracture mechanics
- advanced stress analysis, error estimation
- parametric reduced-order models (multifidelity)
- object-oriented modeling (e.g. Modelica)
- rapid design to analysis methodologies

hermitic
connector



Outline

1. Multiscale modeling and model-form error estimation
2. Polyhedral finite element formulation

Solid mechanics: multiple scales

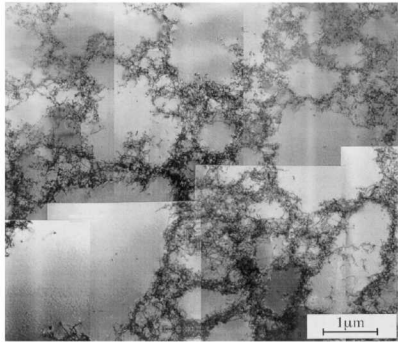


dislocation dynamics

Peach-Koehler relation

$$\mathbf{F} = (\mathbf{b} \cdot \boldsymbol{\sigma}) \times \hat{\boldsymbol{\xi}} \sim O(1/r)$$

dislocation cell structure



(Zaiser, M. and P. Hähner, 1999)

crystal plasticity

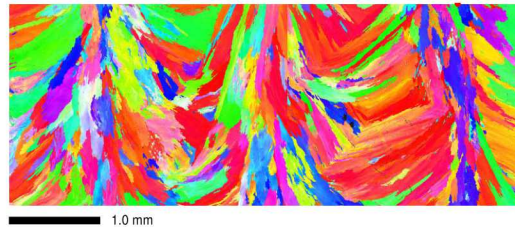
conservation of linear momentum $\frac{\partial \mathbf{P}}{\partial \mathbf{X}} : \mathbf{I} + \rho_o \mathbf{b} = \rho_o \ddot{\mathbf{u}}$ in Ω

plastic velocity gradient:

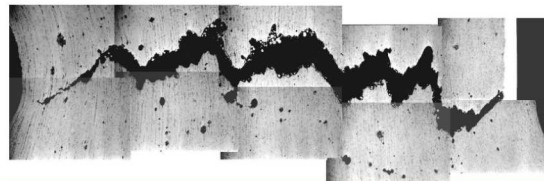
$$L^p = \sum_{\alpha=1}^N \dot{\gamma}^\alpha P^\alpha$$

$$P^\alpha = m^\alpha \otimes n^\alpha$$

AM 304L microstructure (LENS)



ductile void growth & coalescence



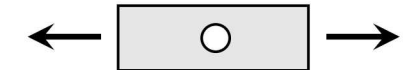
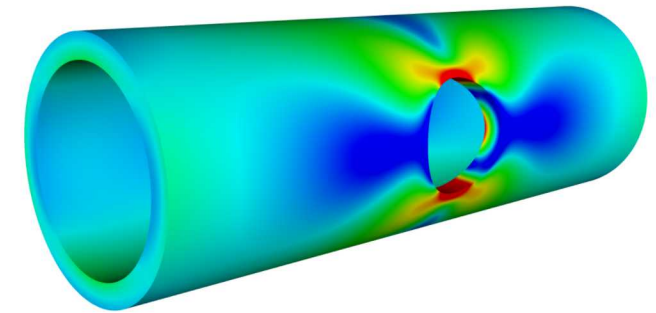
Benzerga, 1999)

macroscale plasticity

conservation of linear momentum $\frac{\partial \mathbf{P}}{\partial \mathbf{X}} : \mathbf{I} + \rho_o \mathbf{b} = \rho_o \ddot{\mathbf{u}}$ in Ω

yield surface $f(\boldsymbol{\sigma}, \bar{\boldsymbol{\epsilon}}^p) = \phi(\boldsymbol{\sigma}) - \sigma_y(\bar{\boldsymbol{\epsilon}}^p) = 0$

$$\phi(\boldsymbol{\sigma}) = \left\{ \frac{1}{2} (|\sigma_1 - \sigma_2|^a + |\sigma_1 - \sigma_3|^a + |\sigma_2 - \sigma_3|^a) \right\}^{1/a}$$



10^{-6}

10^{-4}

10^{-2}

Length Scale (m), time scale (s)

Multiscale modeling challenges for solid mechanics

1. First principles prediction of strain-hardening behavior in ductile materials
2. Stress-strain behavior is path, rate and T dependent (infinite dimensional).
3. Strain localization, fracture
4. Limits of homogenization theory (lack of scale separation)
5. How to include material variability at macroscale? (homogenization filters fine scale)
6. Concurrent multiscale modeling?
7. error estimation (epistemic uncertainty)

Types of uncertainty (UQ)

Types of aleatory uncertainty

- microstructure (material variability)
- geometric
- loading

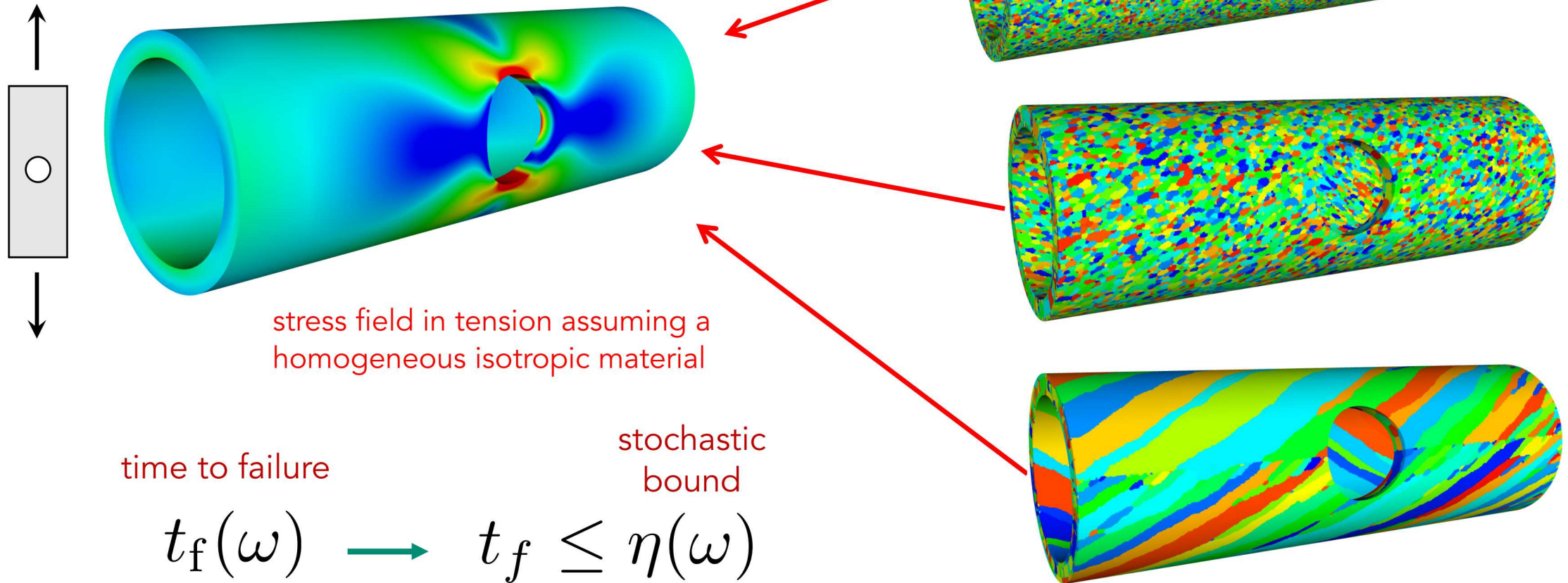
Types of errors (epistemic uncertainty)

- model-form error
- homogenization error
 - lack of scale separation
 - surface effects
 - lack of a representative volume element

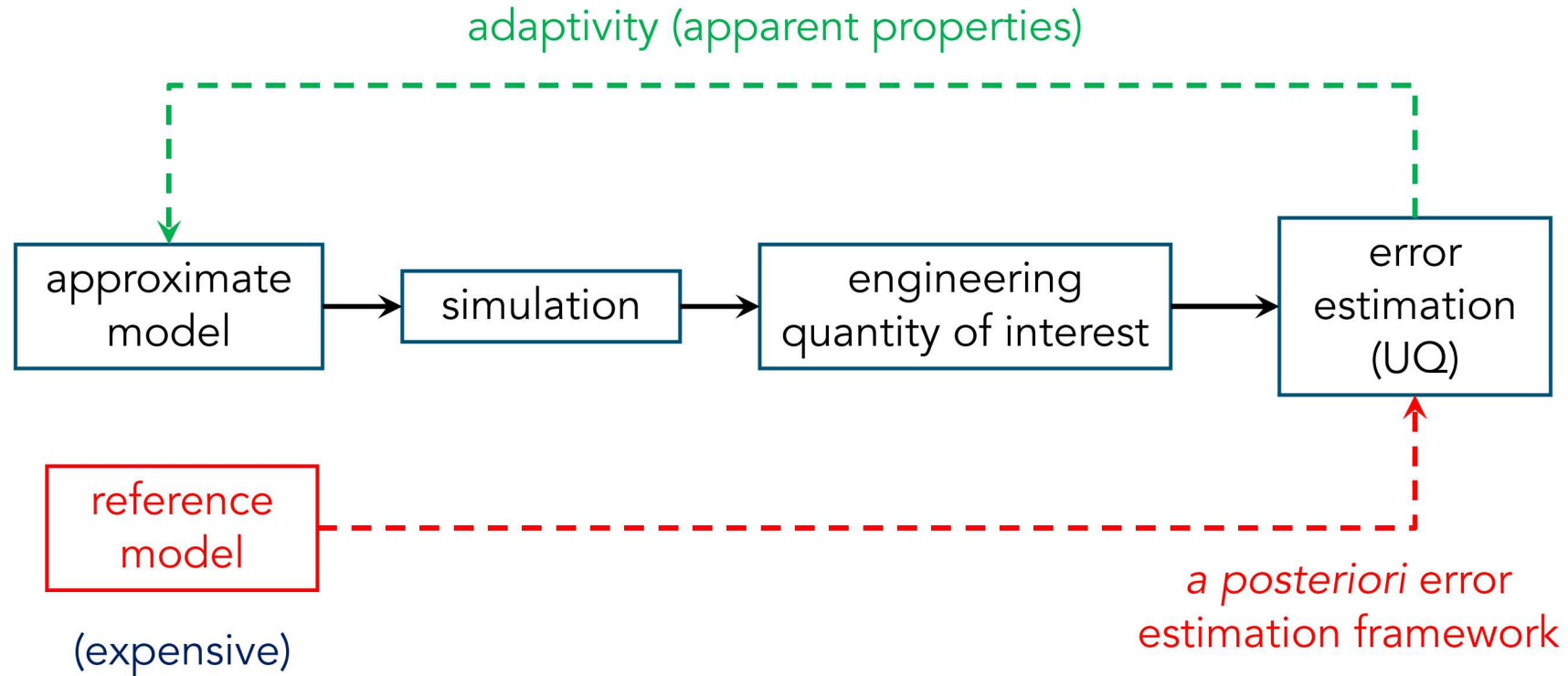
Error estimation as UQ (epistemic)



- simple macroscale model
- deterministic (**simulate once**)
- assess error *a posteriori*



Error estimation and adaptivity



time to failure

$$t_f(\omega) \longrightarrow t_f \leq \overset{\text{stochastic bound}}{\eta(\omega)}$$

Error estimation as UQ



single scale

- Zohdi, T.I., J.T. Oden, and G.J. Rodin, Hierarchical modeling of heterogeneous bodies. *Computer Methods in Applied Mechanics and Engineering*, 1996, 138:273-298.

- Oden, J.T. and T.I. Zohdi, Analysis and adaptive modeling of highly heterogeneous elastic structures. *Computer Methods in Applied Mechanics and Engineering*, 1997, 148: 367-391.

multi-scale

- Oden, J.T. and K. Vemaganti, Adaptive hierarchical modeling of heterogeneous structures. *Physica D: Nonlinear Phenomena*, 1999, 133:404-415.

goal oriented

- Oden, J.T. and K.S. Vemaganti, Estimation of Local Modeling Error and Goal-Oriented Adaptive Modeling of Heterogeneous Materials: I. Error Estimates and Adaptive Algorithms. *Journal of Computational Physics*, 2000, 164:22-47.

- Oden, J.T. and S. Prudhomme, *Goal-oriented error estimation and adaptivity for the finite element method*. *Computers & Mathematics with Applications*, 2001, 41:735-756.

nonlinear

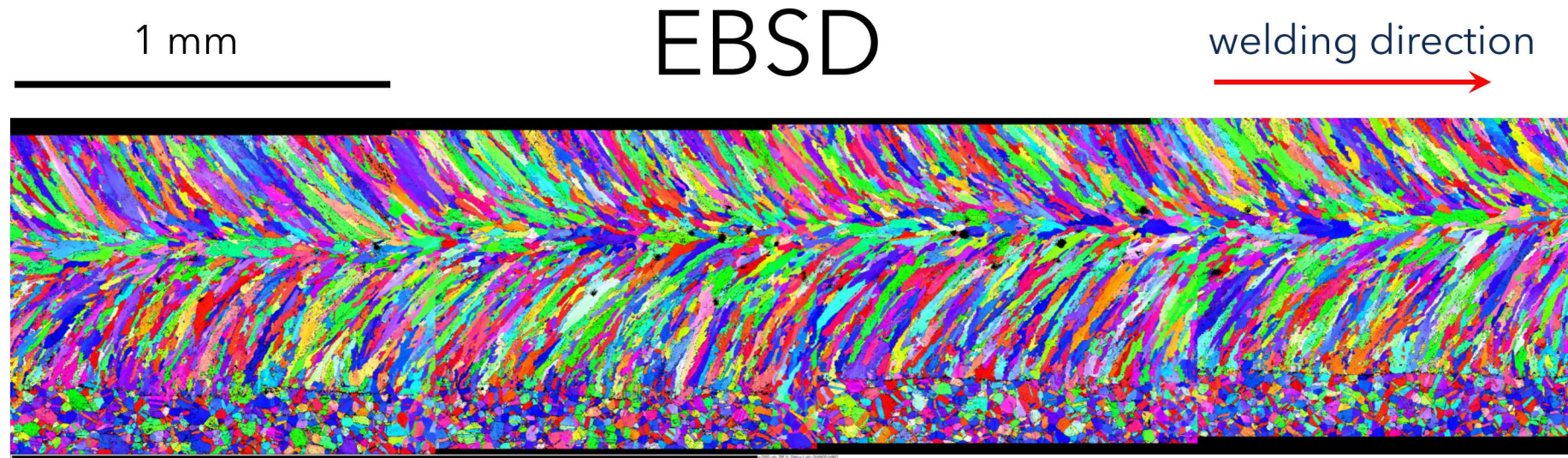
- Oden, J.T., et al., *Modeling error and adaptivity in nonlinear continuum mechanics*. *Computer Methods in Applied Mechanics and Engineering*, 2001, 190:6663-6684.

UQ

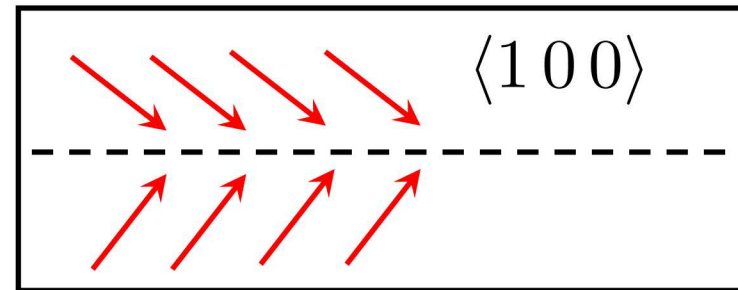
- Romkes, A. and J.T. Oden, Adaptive modeling of wave propagation in heterogeneous elastic solids. *Computer Methods in Applied Mechanics and Engineering*, 2004, 193:539-559.

- need a “reference model” (or advanced experiments)
- adapt macroscale model to reduce error or uncertainty

Motivation: Laser weld, 304L



How to homogenize?



Stainless steel 304L single crystal elasticity constants

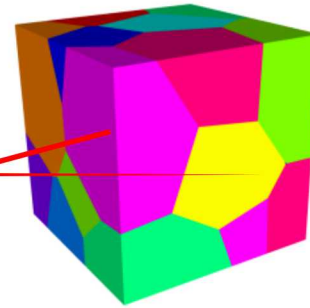


austenite FCC (cubic symmetry)

$$C_{11} = 204.6 \text{ GPa}$$

$$C_{12} = 137.7 \text{ GPa}$$

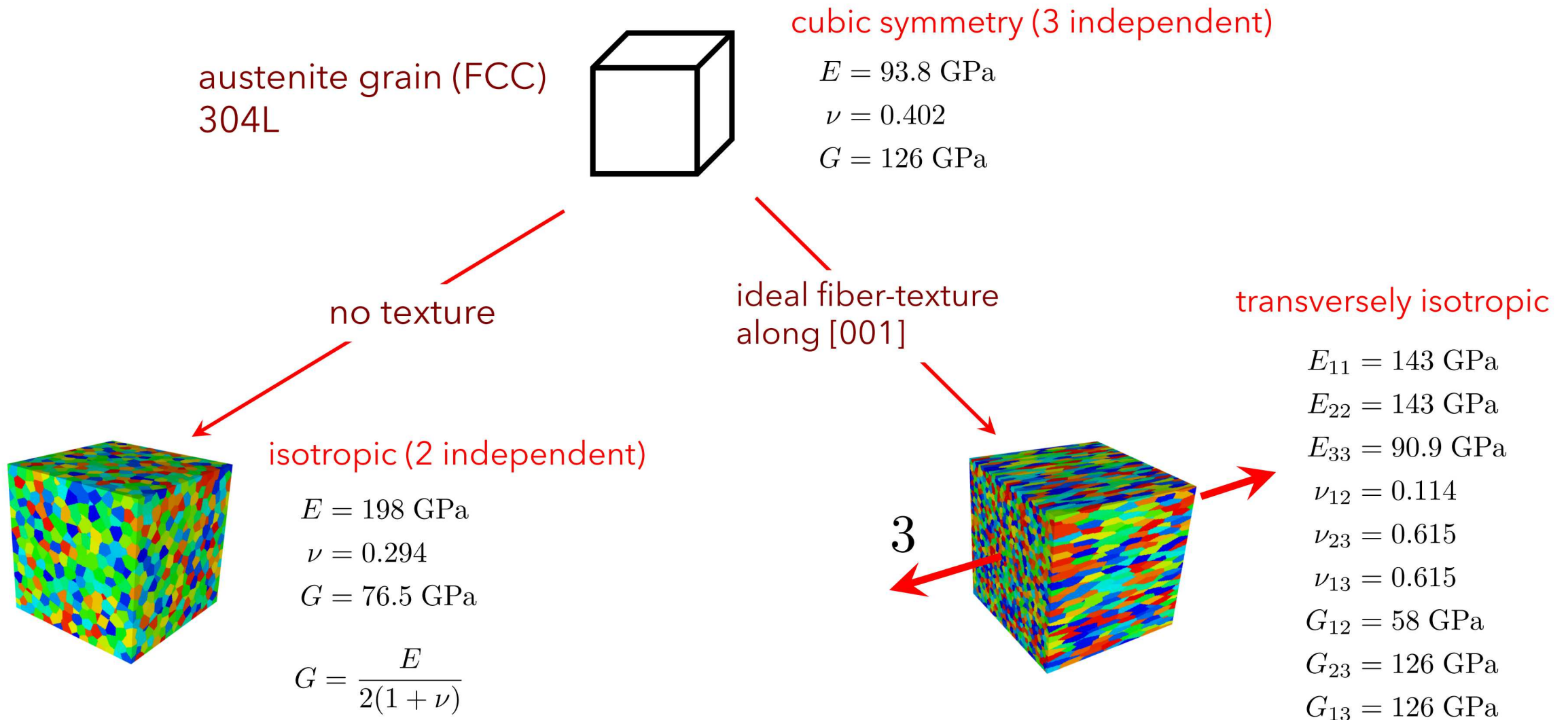
$$C_{44} = 126.2 \text{ GPa}$$



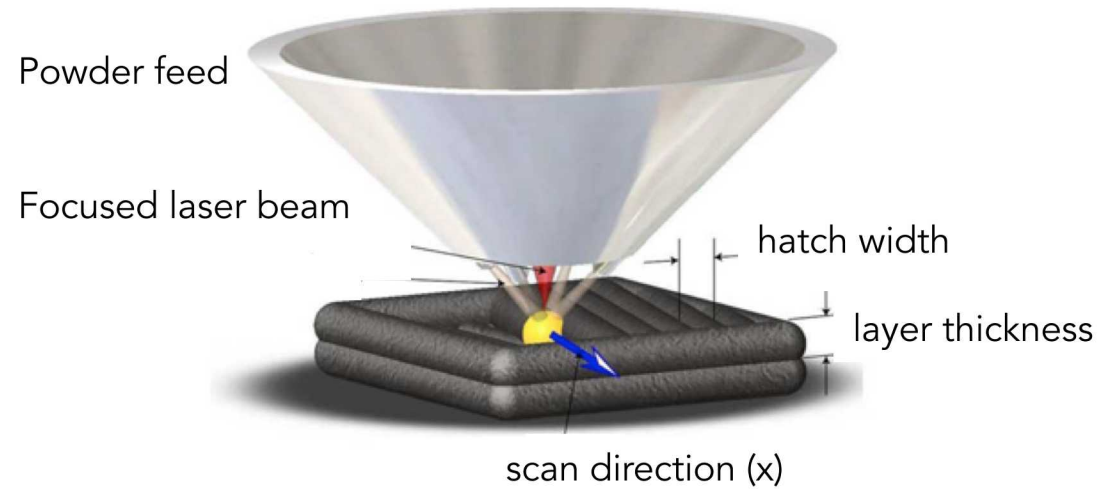
anisotropy ratio $A = \frac{2C_{44}}{C_{11} - C_{12}} = 3.8$

(Ledbetter, 1984)

Effect of texture on homogenized elastic properties

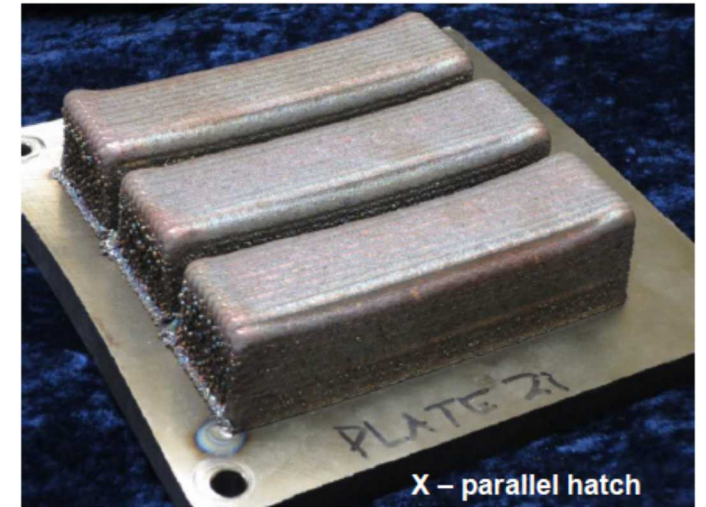


LENS, Laser Engineered Net Shaping



LENS deposition

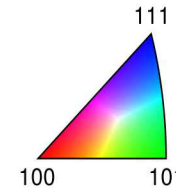
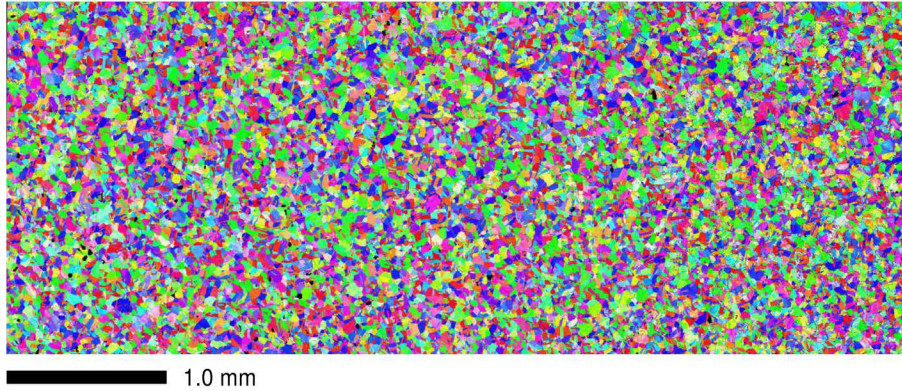
(T. Palmer, PSU)



Motivation: Microstructure comparison, LENS 304L



Wrought



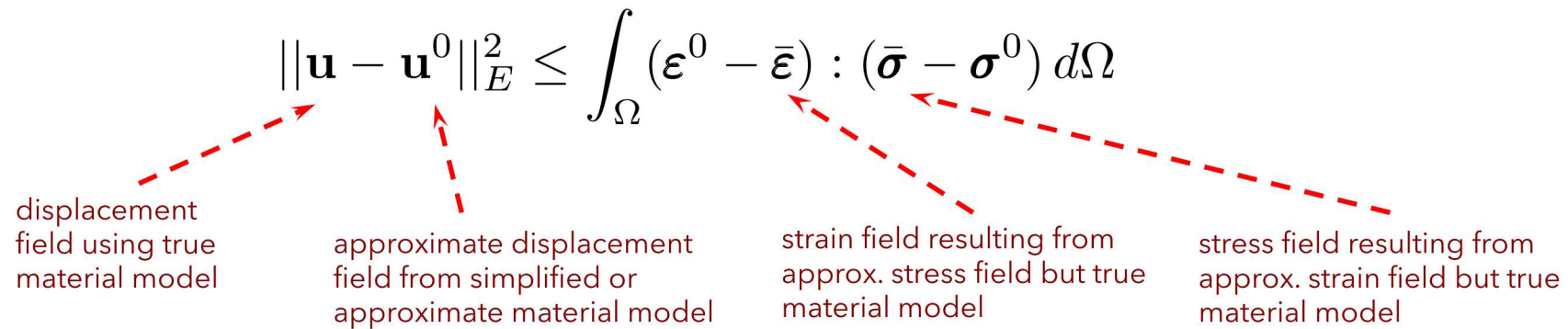
3.8 kW LENS



- How to homogenize?
- Material or structure?

Error estimation: material-model error

(Zohdi, Oden, Rodin, 1996, CMAME)

$$\|\mathbf{u} - \mathbf{u}^0\|_E^2 \leq \int_{\Omega} (\boldsymbol{\epsilon}^0 - \bar{\boldsymbol{\epsilon}}) : (\bar{\boldsymbol{\sigma}} - \boldsymbol{\sigma}^0) d\Omega$$


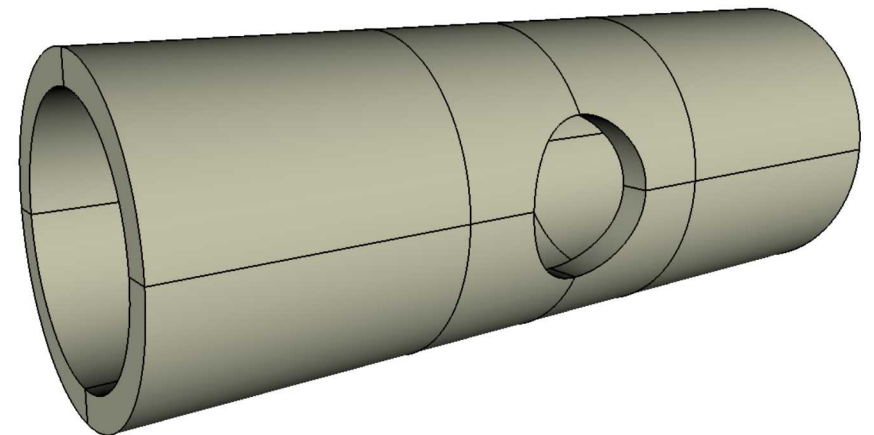
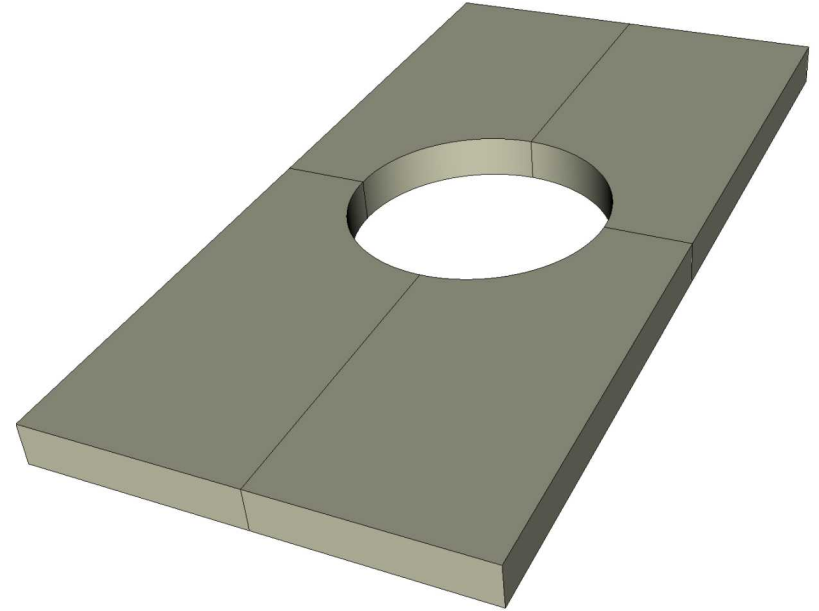
The diagram illustrates the components of the error estimation formula. Red dashed arrows point from the following text labels to the corresponding terms in the equation above:

- displacement field using true material model (points to $\mathbf{u}$)
- approximate displacement field from simplified or approximate material model (points to $\mathbf{u}^0$)
- strain field resulting from approx. stress field but true material model (points to $\boldsymbol{\epsilon}^0$)
- stress field resulting from approx. strain field but true material model (points to $\bar{\boldsymbol{\sigma}}$)

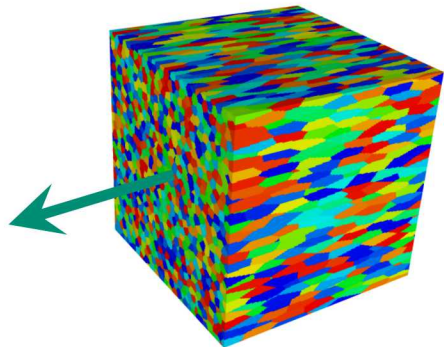
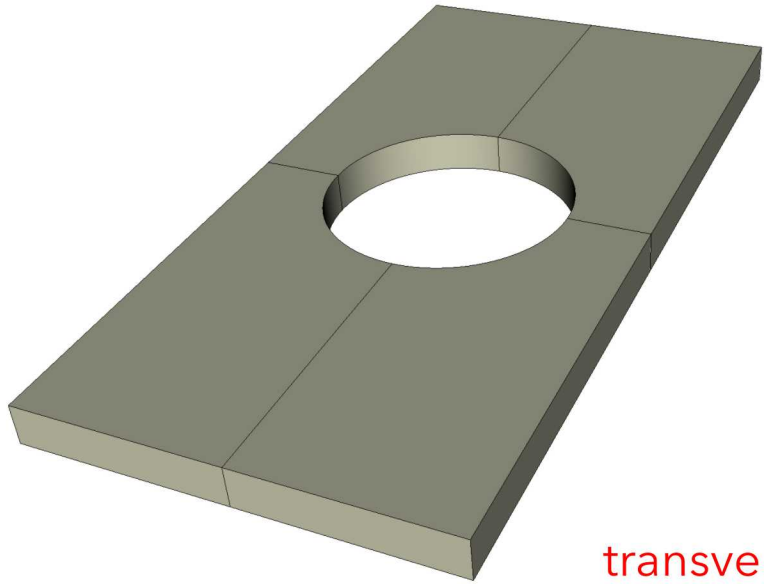
Key point: Can bound error using only known quantities from the approximate simulation. Don't need to run "true" model simulation.

Examples

1. anisotropic plate with hole (single scale)
2. tube with side hole: metal additive manufacturing (multiscale)



Example: anisotropic plate with hole



transversely isotropic

$$E_{11} = 143 \text{ GPa}$$

$$E_{22} = 143 \text{ GPa}$$

$$E_{33} = 91.8 \text{ GPa}$$

$$\nu_{12} = 0.115$$

$$\nu_{23} = 0.610$$

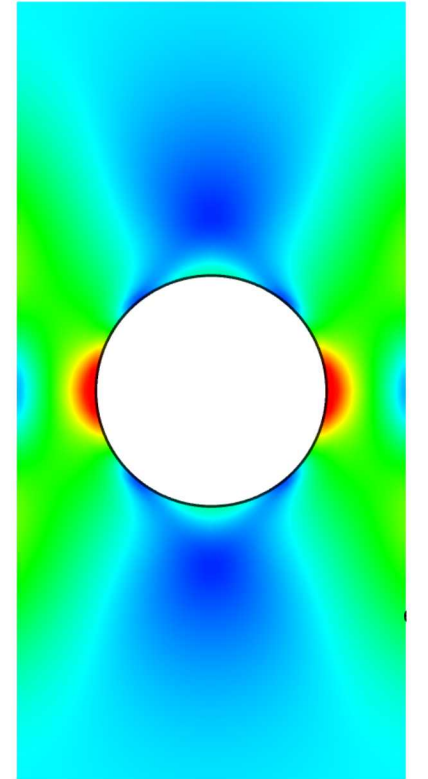
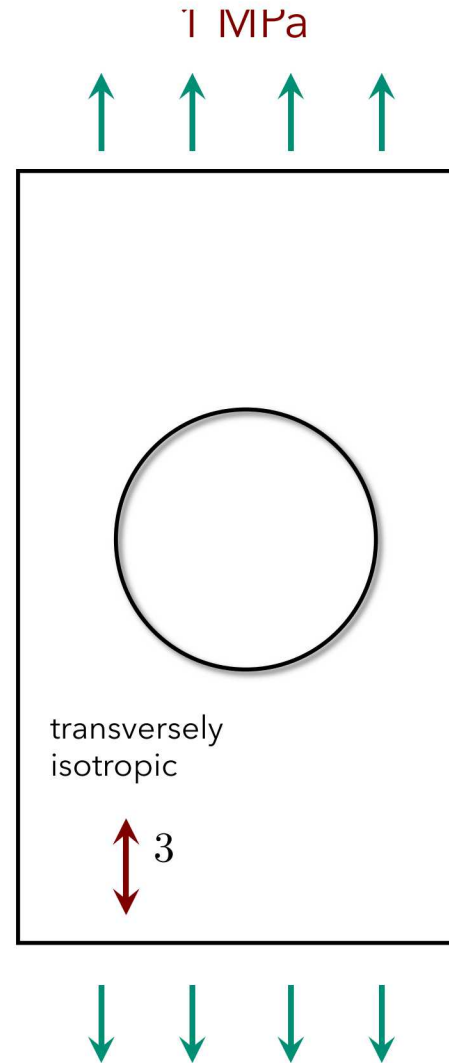
$$\nu_{13} = 0.610$$

$$G_{12} = 64 \text{ GPa}$$

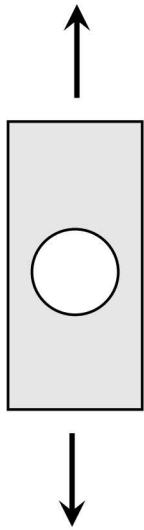
$$G_{23} = 126 \text{ GPa}$$

$$G_{13} = 126 \text{ GPa}$$

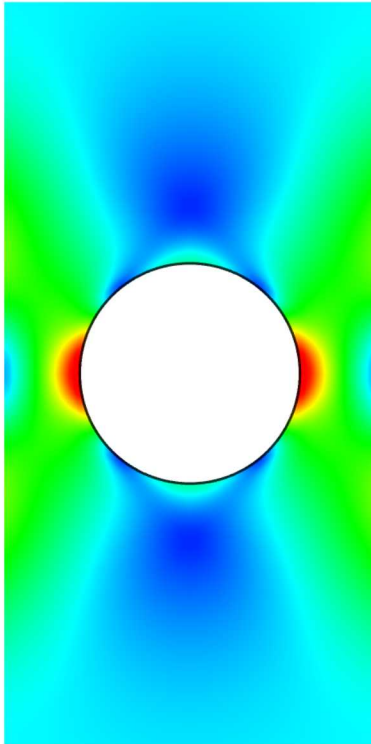
3



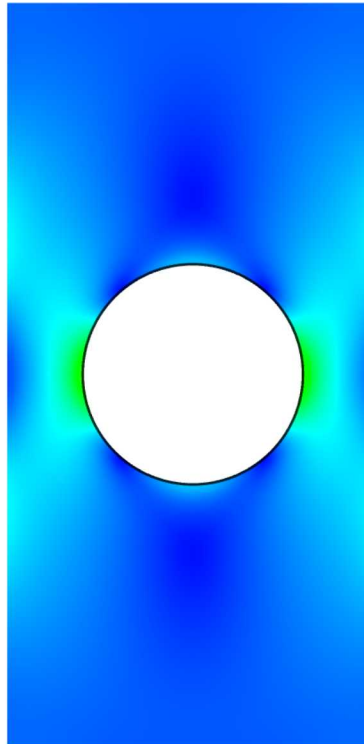
Strain, magnitude



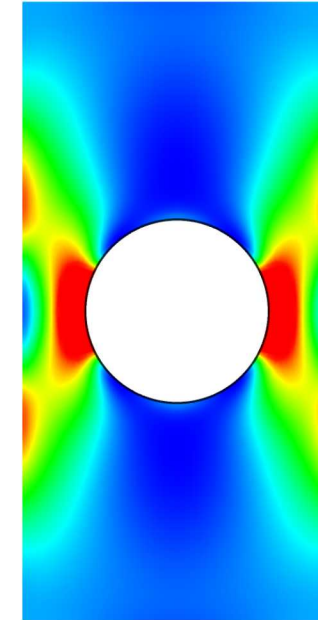
exact



$E, \mathbf{v}$
homogeneous

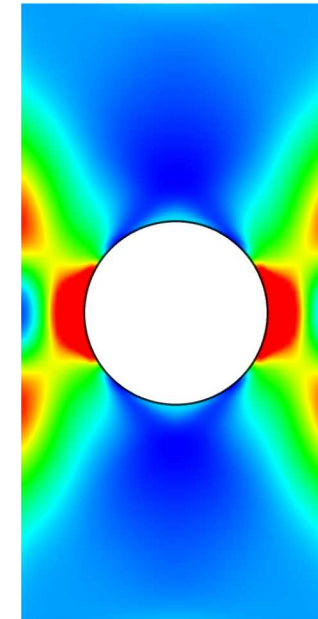


estimated
error



9.234×10^{-2}
 8.963×10^{-2}

exact error



Model adaptivity

$$\int_{\Omega} (\mathbf{e}^0 - \bar{\mathbf{e}}) : (\bar{\boldsymbol{\sigma}} - \boldsymbol{\sigma}^0) d\Omega \approx \sum_{i=1}^N V_e \eta_i^2$$

error indicator $\eta_i^2 \approx (\boldsymbol{\epsilon}^0 - \bar{\boldsymbol{\epsilon}}) : (\bar{\boldsymbol{\sigma}} - \boldsymbol{\sigma}^0) = (\mathbb{I} - \mathbb{C}^{-1} \mathbb{C}^0) \boldsymbol{\epsilon}^0 : (\mathbb{C} - \mathbb{C}^0) \boldsymbol{\epsilon}^0$

consider fixed

How can we decrease this error, but still use an isotropic material model?

Consider a heterogeneous isotropic material model?

$$\mathbb{C}^0 = \mathbb{C}^{\text{iso}} = a \mathbb{J} + b \mathbb{K} = 3K \mathbb{J} + 2\mu \mathbb{K}$$

$$\mathbb{J} = \frac{1}{3} \delta_{ij} \delta_{kl} \quad \text{hydrostatic operator}$$

$$\mathbb{K} = \mathbb{I} - \mathbb{J} \quad \text{deviatoric operator}$$

Let bulk modulus K and shear modulus μ vary with position.

Model adaptivity

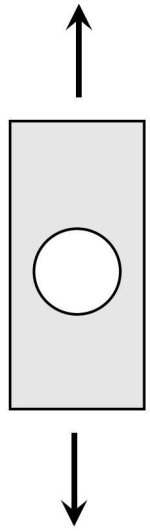
necessary conditions

$$\frac{\partial(\eta_i^2)}{\partial a} = 0 \quad \text{and} \quad \frac{\partial(\eta_i^2)}{\partial b} = 0$$

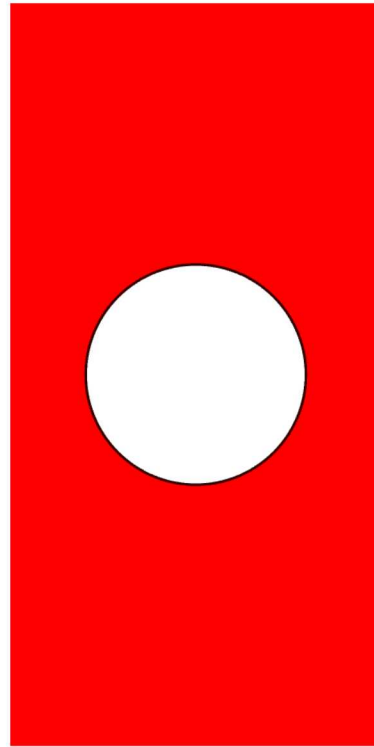
Can derive this 2 by 2 system of equations for the updated bulk and shear moduli.

$$\begin{bmatrix} 2(\mathbb{C}^{-1} \mathbb{J} \boldsymbol{\epsilon}^0) : (\mathbb{J} \boldsymbol{\epsilon}^0) & (\mathbb{C}^{-1} \mathbb{K} \boldsymbol{\epsilon}^0) : (\mathbb{J} \boldsymbol{\epsilon}^0) + (\mathbb{C}^{-1} \mathbb{J} \boldsymbol{\epsilon}^0) : (\mathbb{K} \boldsymbol{\epsilon}^0) \\ \text{(sym)} & 2(\mathbb{C}^{-1} \mathbb{K} \boldsymbol{\epsilon}^0) : (\mathbb{K} \boldsymbol{\epsilon}^0) \end{bmatrix} \begin{Bmatrix} 3K_{\text{new}} \\ 2\mu_{\text{new}} \end{Bmatrix} = \begin{Bmatrix} ||\mathbb{J} \boldsymbol{\epsilon}^0||_2^2 + (\mathbb{C}^{-1} \mathbb{J} \boldsymbol{\epsilon}^0) : (\mathbb{C} \boldsymbol{\epsilon}^0) \\ ||\mathbb{K} \boldsymbol{\epsilon}^0||_2^2 + (\mathbb{C}^{-1} \mathbb{K} \boldsymbol{\epsilon}^0) : (\mathbb{C} \boldsymbol{\epsilon}^0) \end{Bmatrix}$$

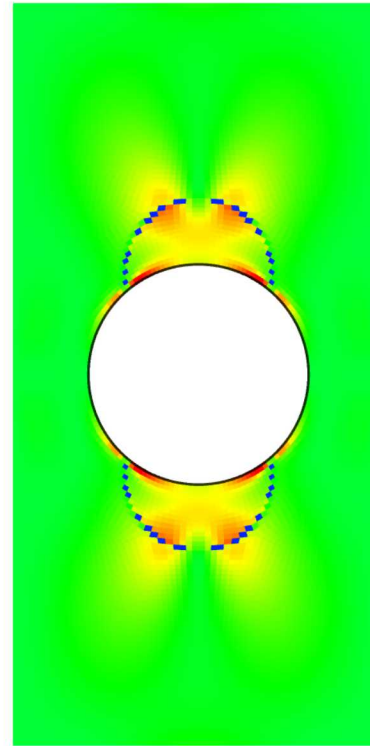
Young's modulus



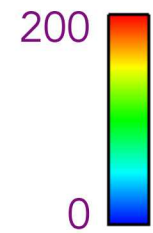
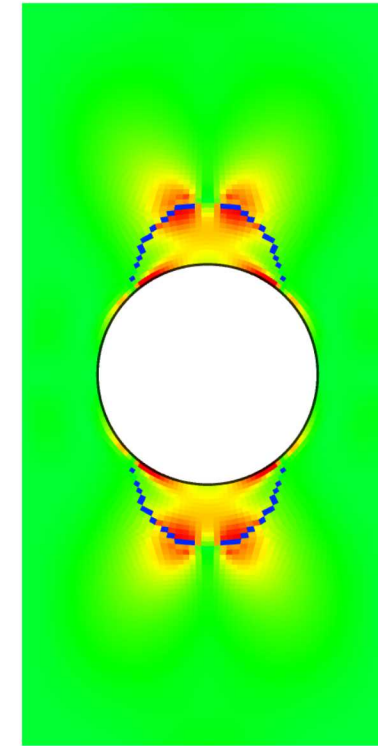
homogeneous



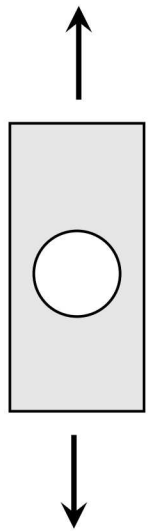
iteration 1



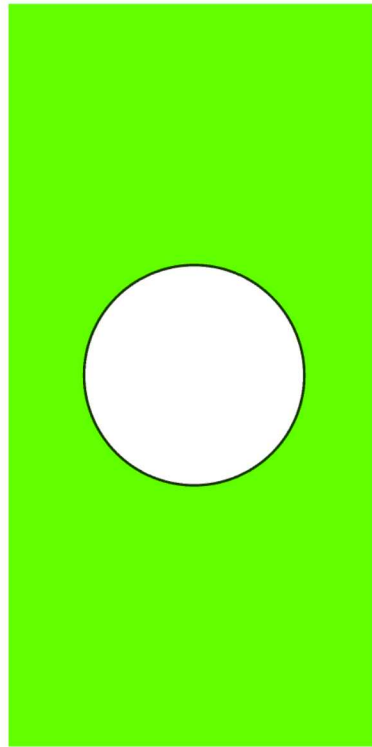
iteration 2



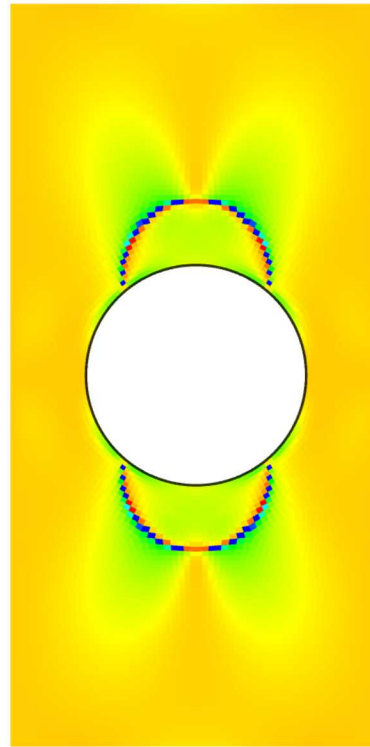
Poisson's ratio



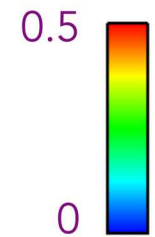
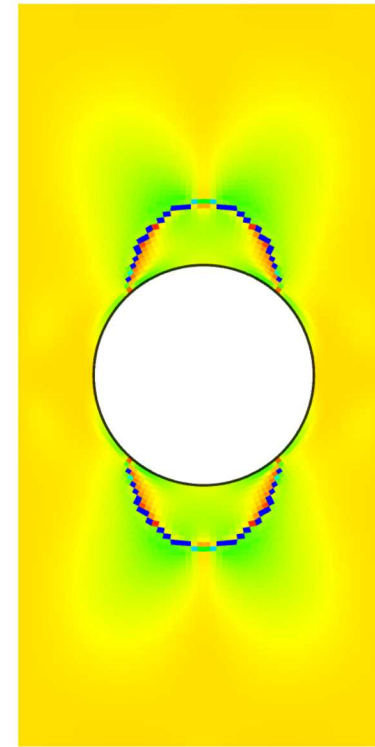
homogeneous



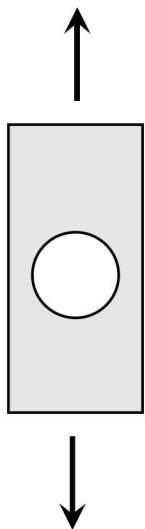
iteration 1



iteration 2

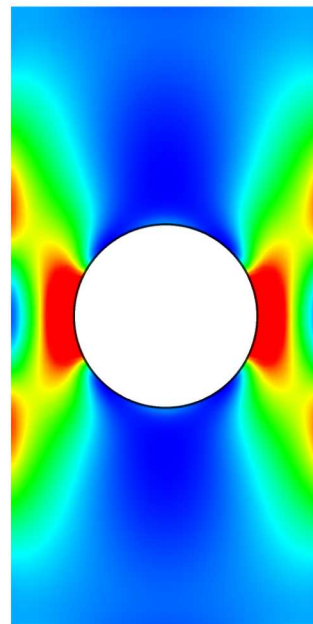


Error bound



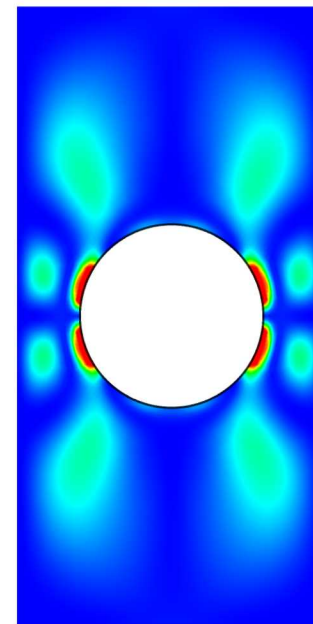
estimated error

$E, \mathbf{v}$, homogeneous



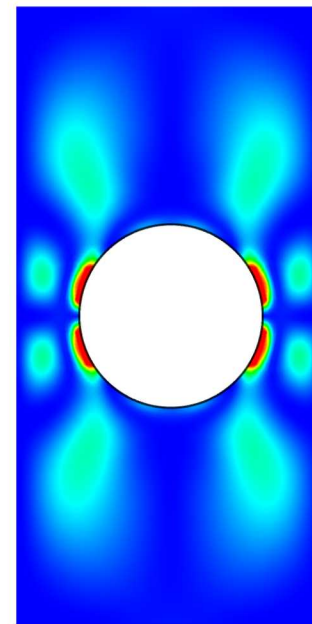
9.234×10^{-2}
 8.963×10^{-2}

iteration 1



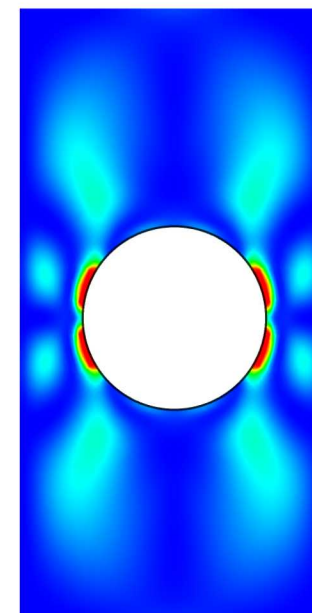
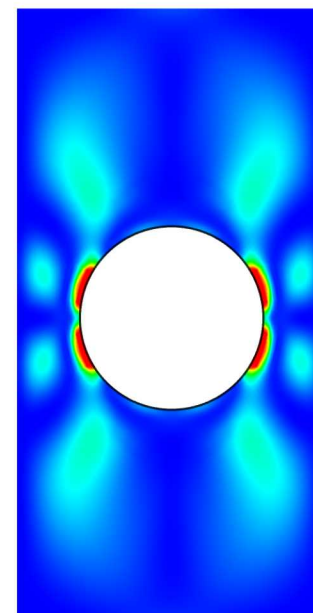
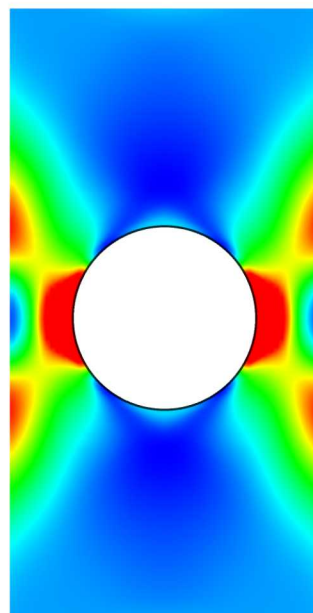
5.554×10^{-2}
 5.447×10^{-2}

iteration 2

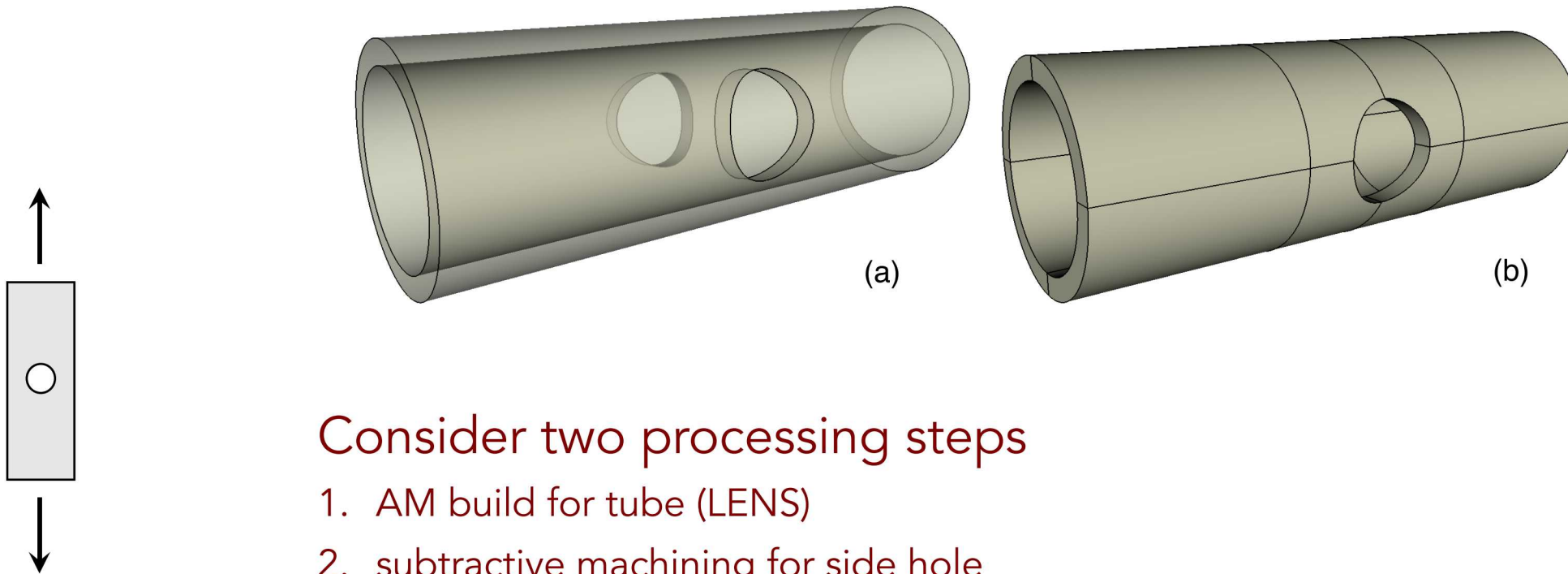


5.482×10^{-2}
 5.378×10^{-2}

exact error



Example (multiscale)



Consider two processing steps

1. AM build for tube (LENS)
2. subtractive machining for side hole

Consider two synthetic microstructures:

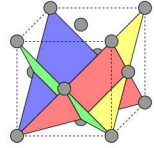
1. equiaxed (wrought)
2. additive (LENS using Kinetic Monte Carlo)

FCC crystal plasticity model



plastic velocity
gradient:

$$L^p = \sum_{\alpha=1}^N \dot{\gamma}^{\alpha} P^{\alpha}$$



Schmid tensor:

$$P^{\alpha} = m^{\alpha} \otimes n^{\alpha}$$

slip system slip
rates:

$$\dot{\gamma}^{\alpha} = \dot{\gamma}_0 \left(\frac{\tau^{\alpha}}{g} \right)^{1/m} \cdot \text{sign}(\tau^{\alpha})$$

slip system
hardening:

$$g = g_o + (g_{so} - g_o) \left[1 - \exp \left(- \frac{G_o}{g_{so} - g_o} \gamma \right) \right]$$

$$\gamma = \sum_{s=1}^N |\gamma^s| \quad (\text{sum over slip systems})$$

Single crystal elastic constants (austenite)

$$C_{11} = 204.6 \text{ GPa}$$

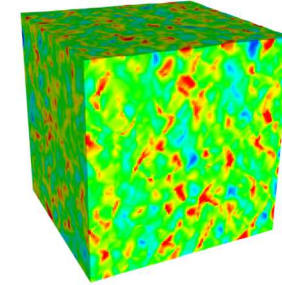
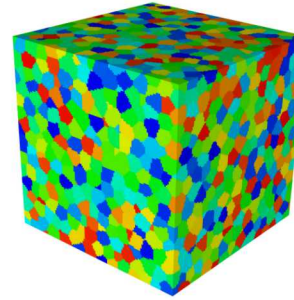
$$C_{12} = 137.7 \text{ GPa}$$

$$C_{44} = 126.2 \text{ GPa}$$

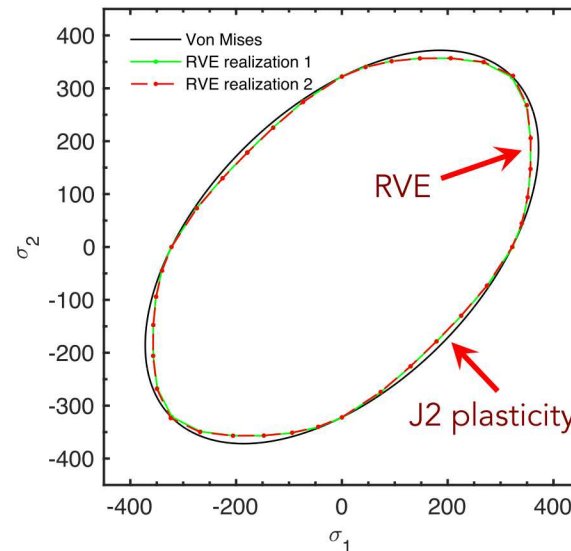
Not considering:

- grain boundary effects (Hall-Petch effect)
- twinning
- dislocation substructures
- latent hardening

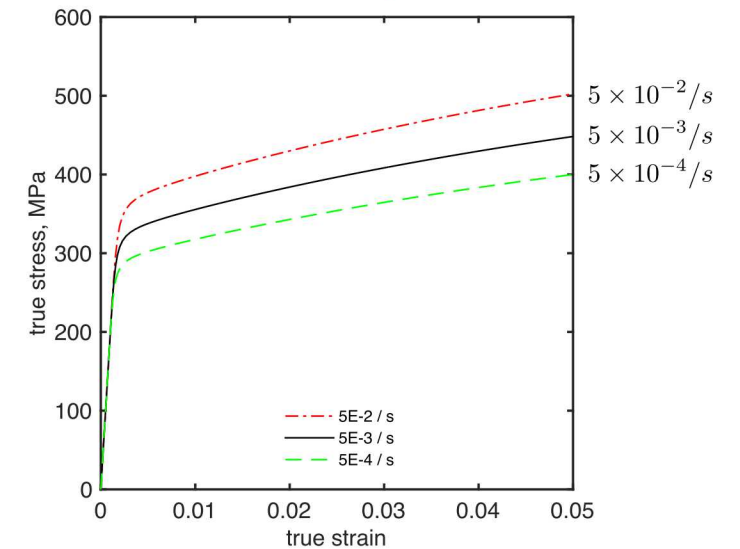
Representative Volume Element (RVE) response



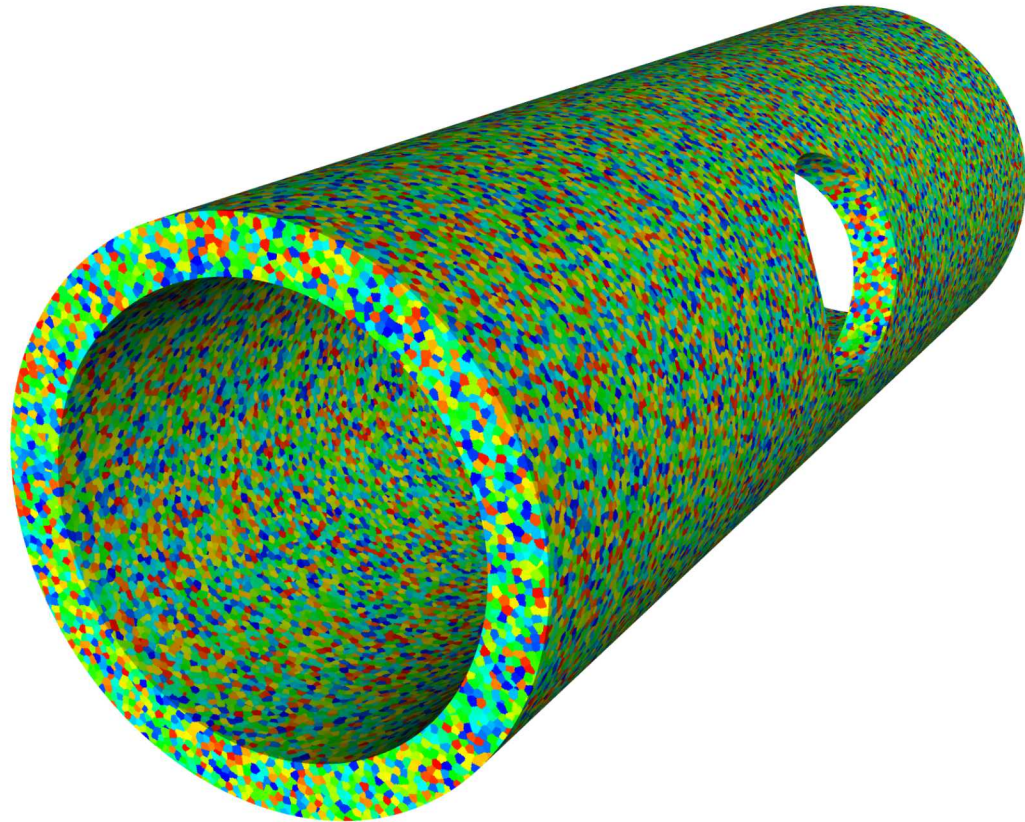
yield surface



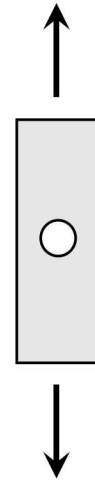
strain-rate dependence



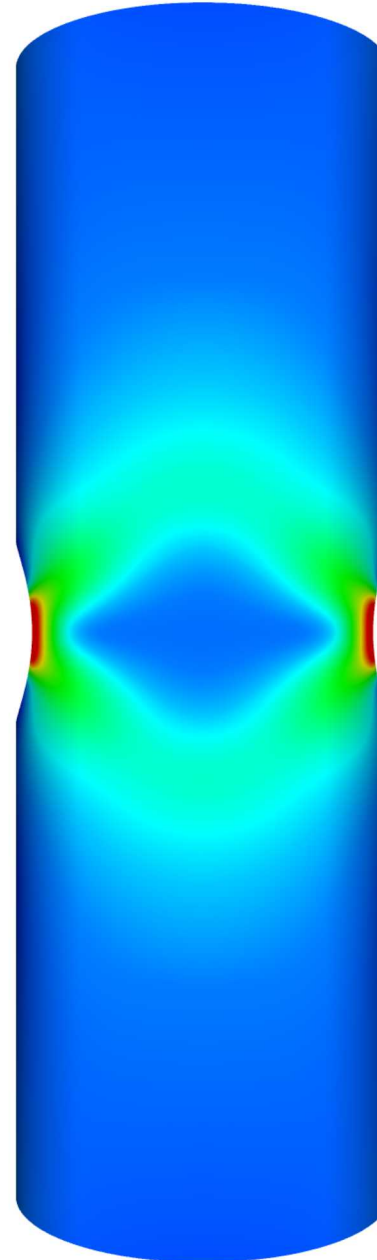
Equiaxed microstructure



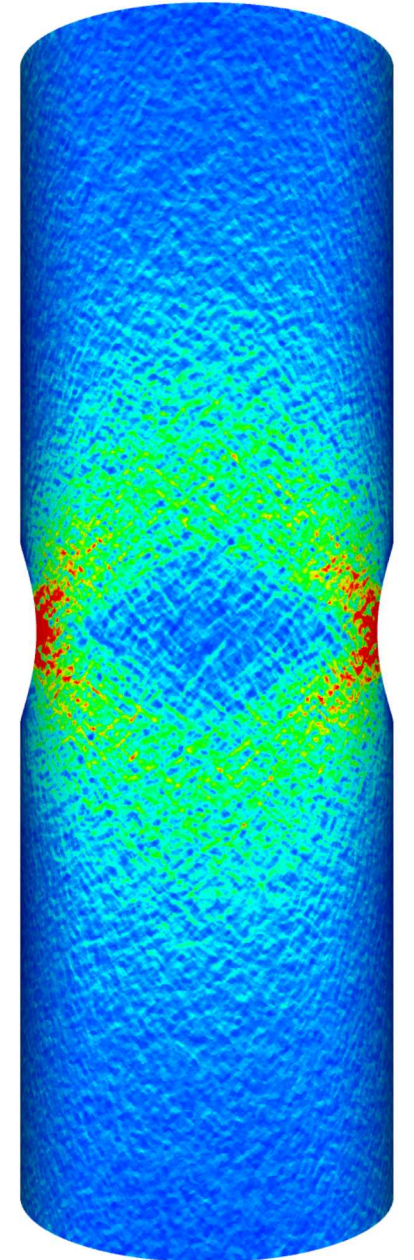
- 358,000 grains
- voxelated with 28M finite elements
- crystal plasticity
- assume no preferred grain orientation (no texture)
- homogenized response is isotropic



homogenized



direct simulation

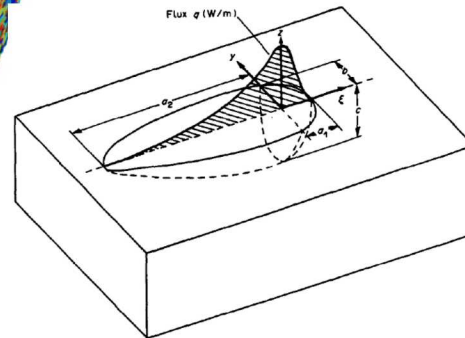
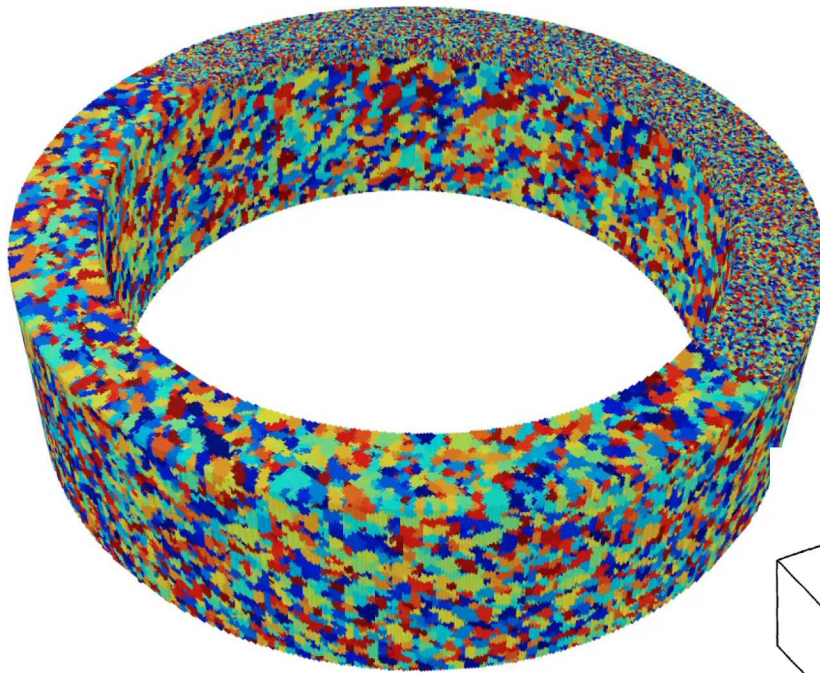


Synthetic additive microstructure using Kinetic Monte Carlo (KMC)

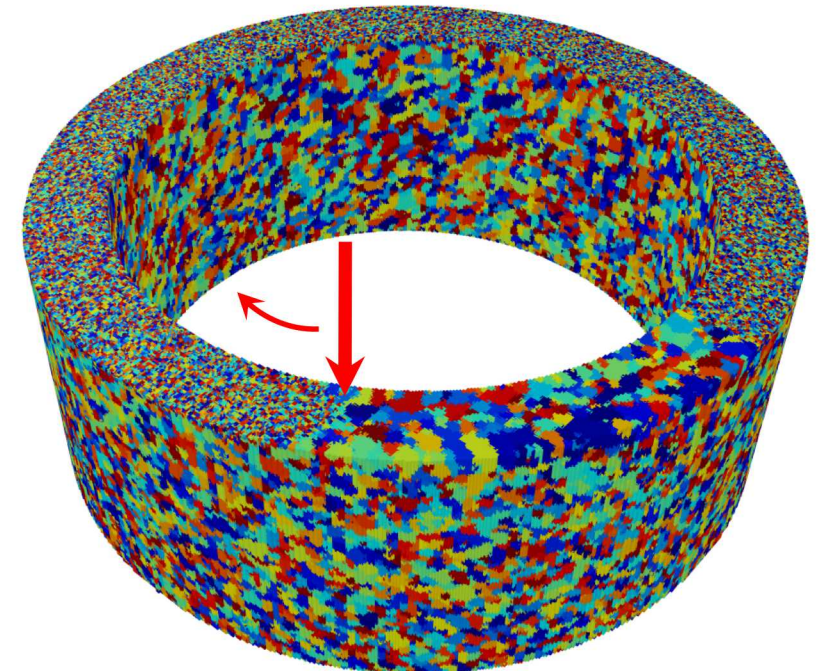
(Theron Rodgers, SNL)



- KMC additive simulation
- single laser pass per layer
- 100 layers
- overlap with previous layer



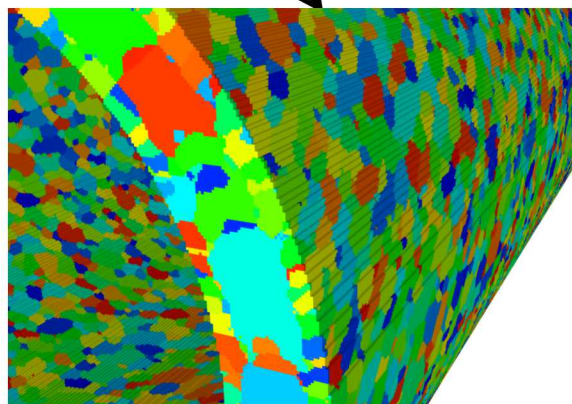
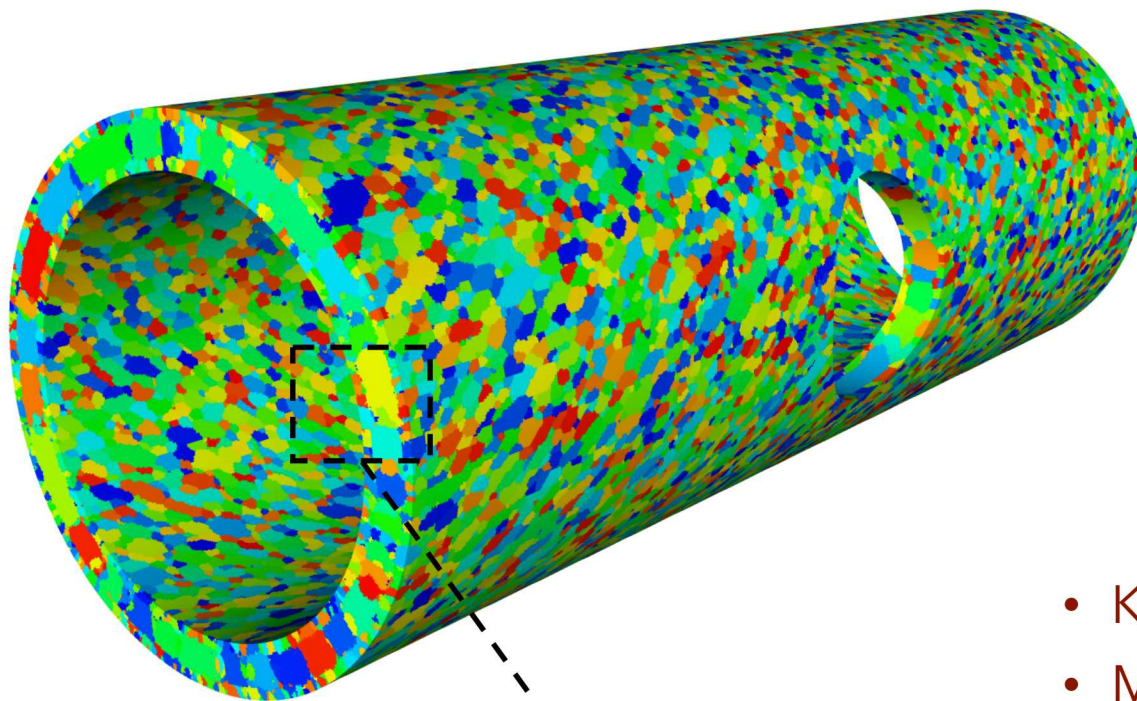
double ellipsoid heat-source model



Synthetic additive microstructure using KMC

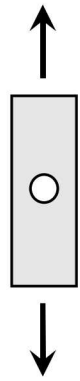


(Theron Rodgers, Jon Madison, SNL)

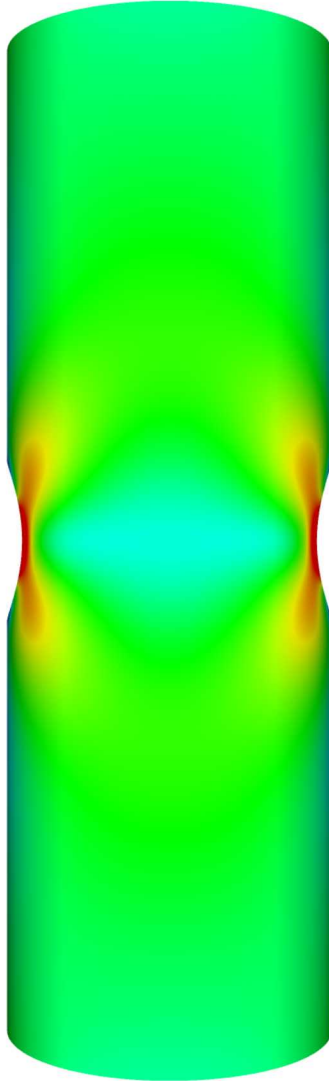


- KMC (SPPARKS) voxelated geometry, 55 million
- Map to a conformal finite-element mesh of 22M elements

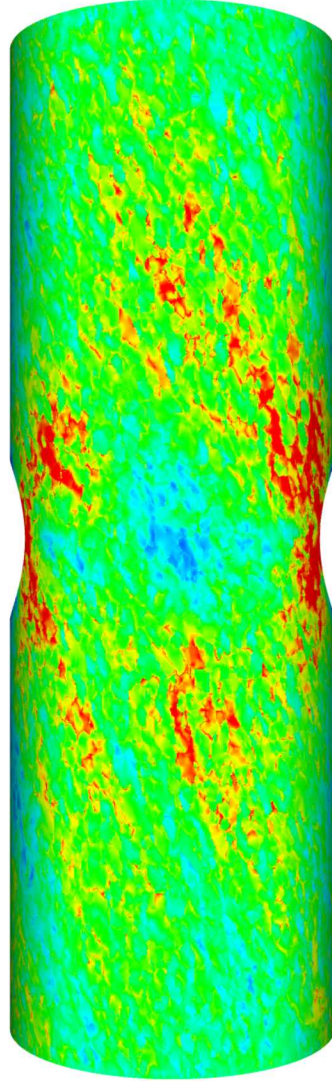
AM stress results (KMC)



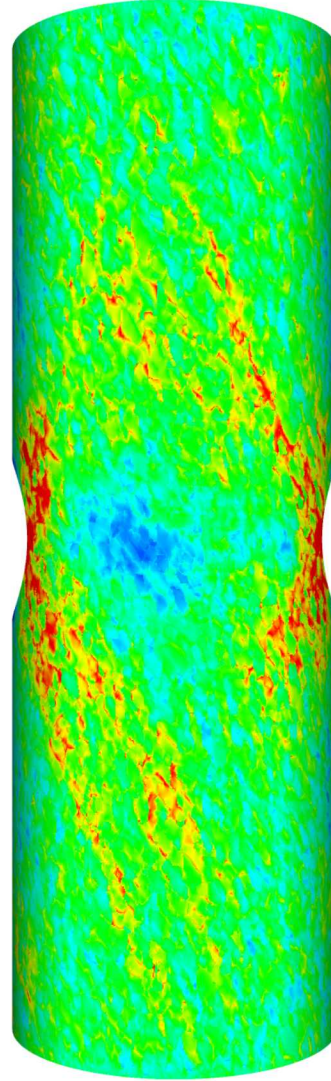
homogeneous,
isotropic



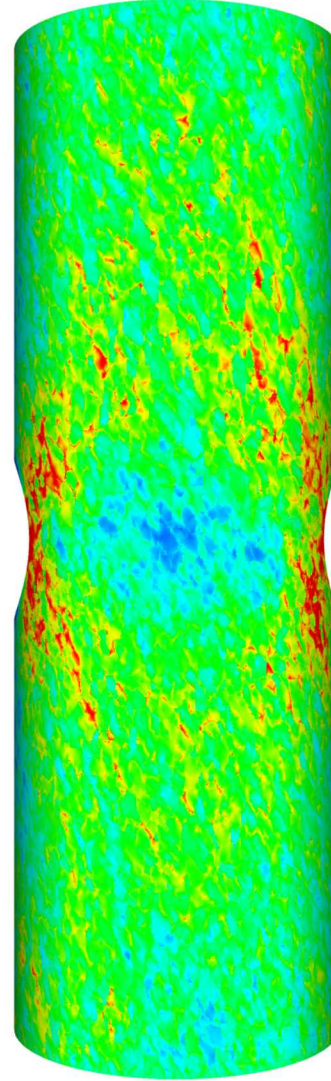
realization 1



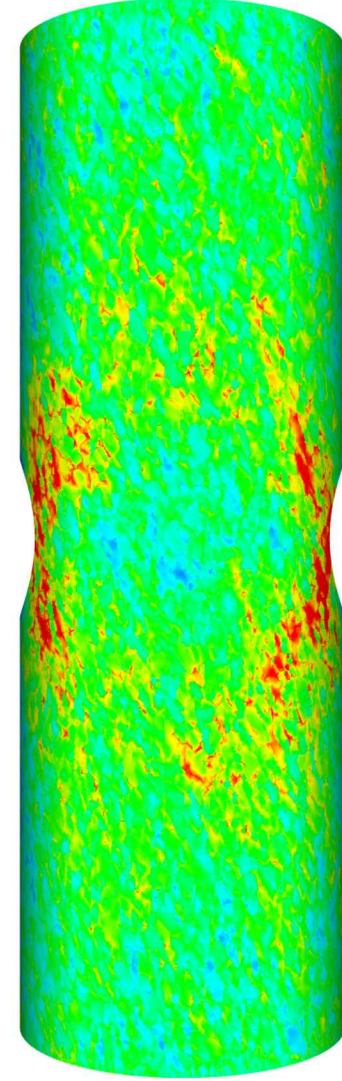
2



3

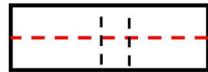


4

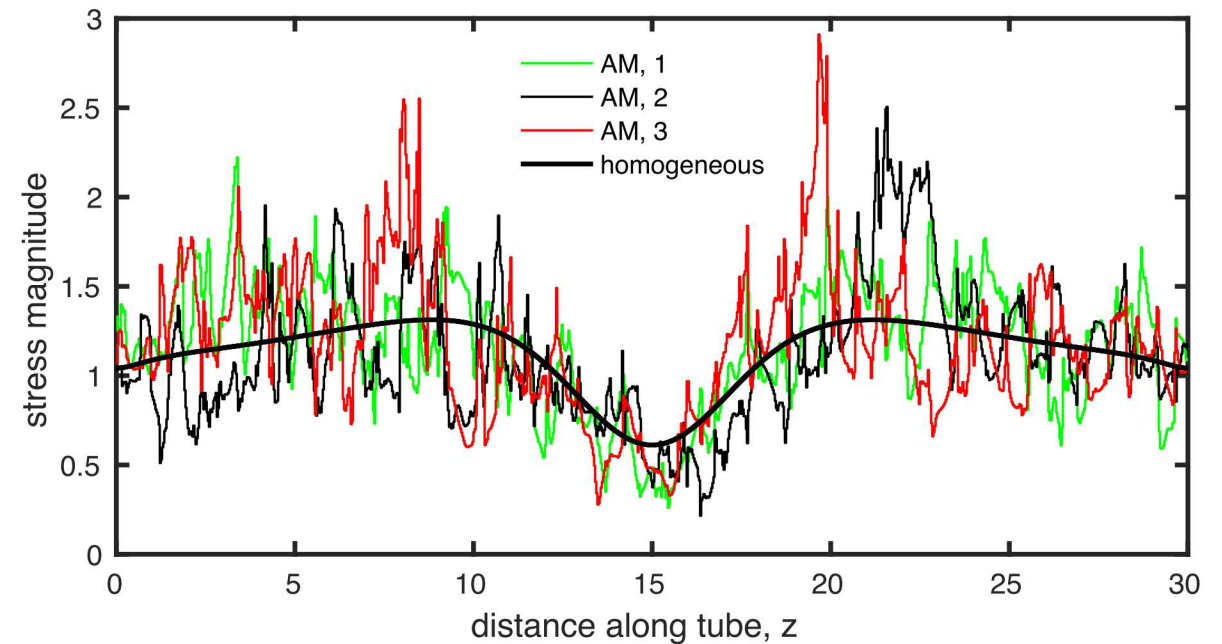
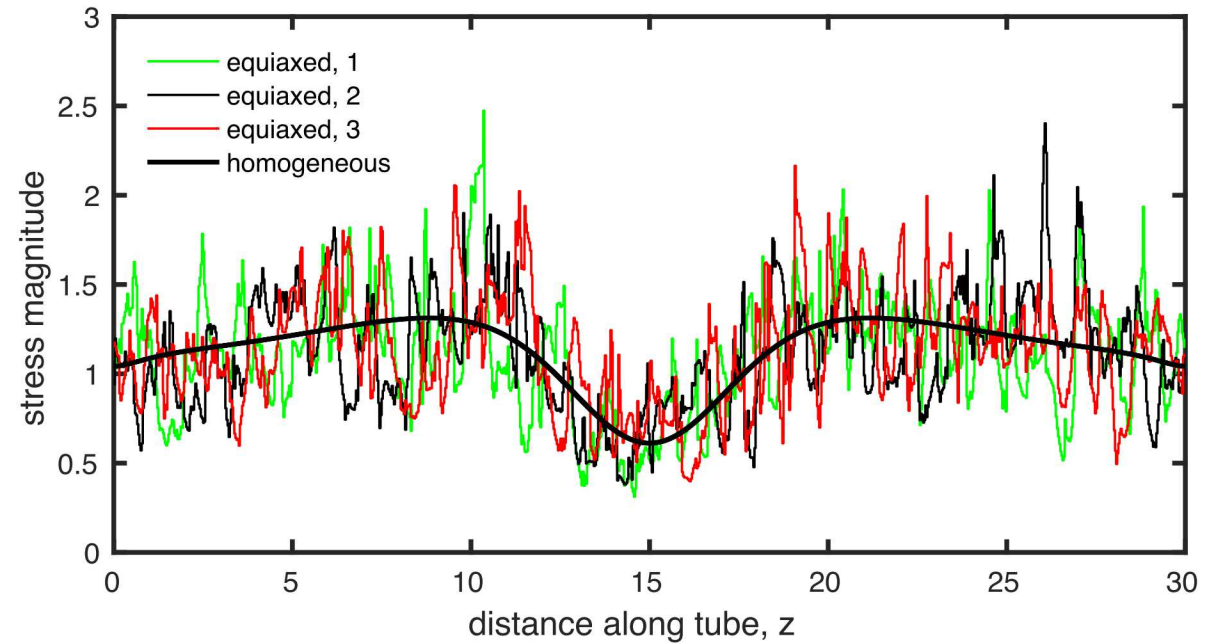


Stress trace

Equiaxed stress
result



AM (KMC) stress
result



Error estimation: multi-scale



Previous error estimate will perform poorly if applied here.

- similar to Voigt assumption in composites theory (uniform strain)
- need to use a modified error estimate in energy norm.

(Oden, Zohdi, 1997)

$$\|\mathbf{w} - \mathbf{u}\|_E^2 = 2[J(\mathbf{w}) - J(\mathbf{u}^0)] + \underbrace{\|\mathbf{u} - \mathbf{u}^0\|_E^2}_{\text{previous estimate}}$$

Diagram illustrating the error estimation formula and its components:

- $\mathbf{w}$: approximate displacement field from submodeling
- $\mathbf{u}$: displacement field using true material model
- $J(\mathbf{w})$: potential energy functional
- $J(\mathbf{u}^0)$: potential energy functional
- $\mathbf{u}^0$: previous estimate

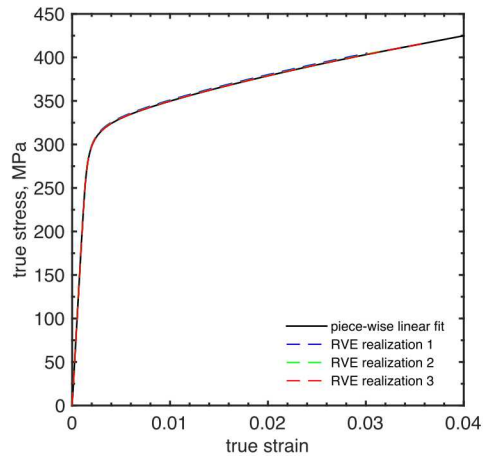
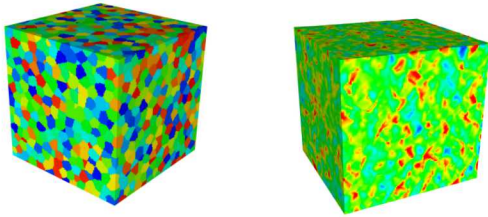
$$\|\mathbf{u} - \mathbf{u}^0\|_E^2 \leq \sum_{i=1}^N V_e \eta_i^2$$

How to get approximate microscale displacement field, $\mathbf{w}$?
Use a Dirichlet projection (submodeling)

Homogenization-localization duality



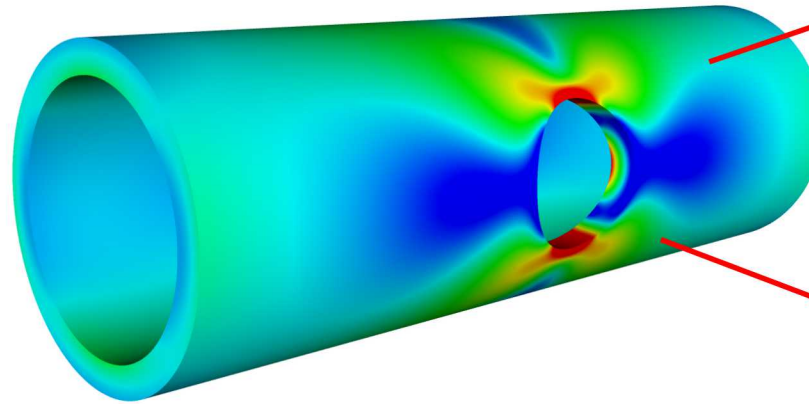
Homogenize
(filter fine scale)



Macroscale simulation

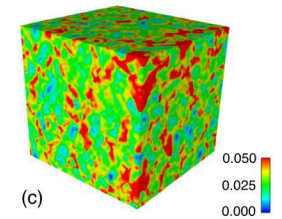


$$\frac{\partial \mathbf{P}}{\partial \mathbf{X}} : \mathbf{I} + \rho_o \mathbf{b} = \rho_o \ddot{\mathbf{u}} \quad \text{in } \Omega$$

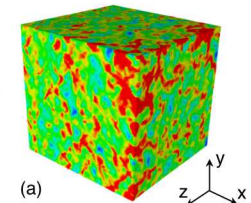


Localization
(recover fine scale)

$\sigma_{ij} \epsilon_{ij}$



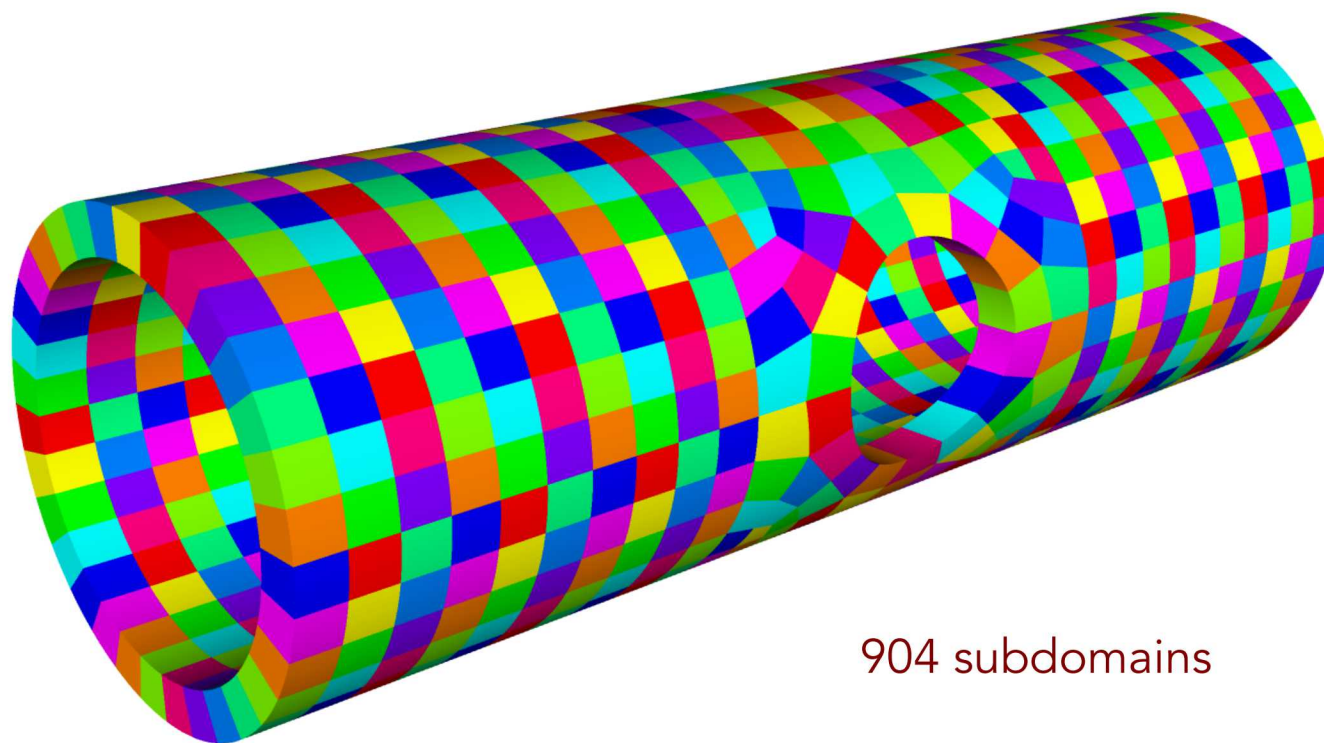
$\sigma_{ij} \epsilon_{ij}$



Assumptions:

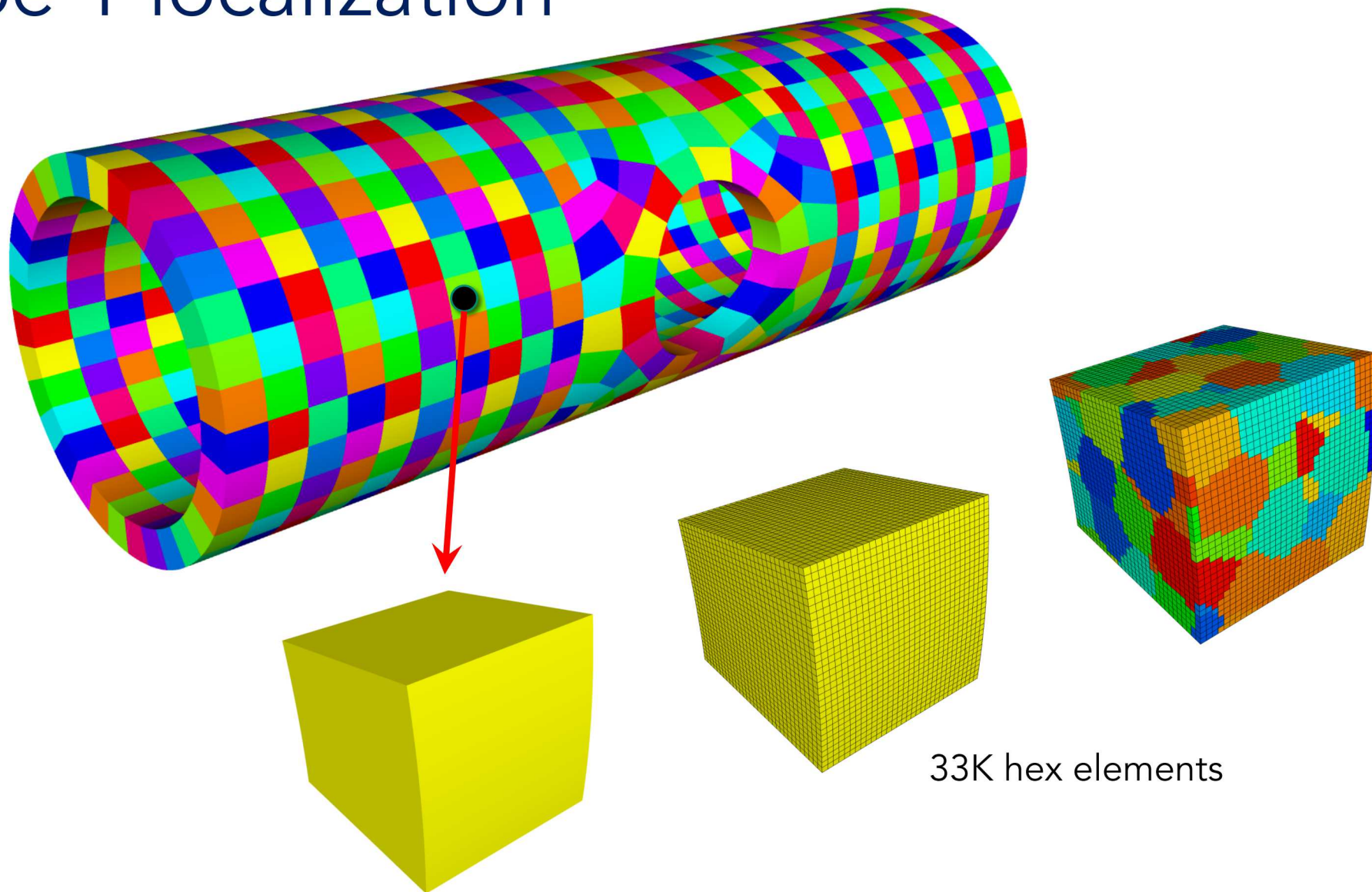
- scale separation
- RVE well defined
- no surface effects

First, partition structure into non-overlapping subdomains.

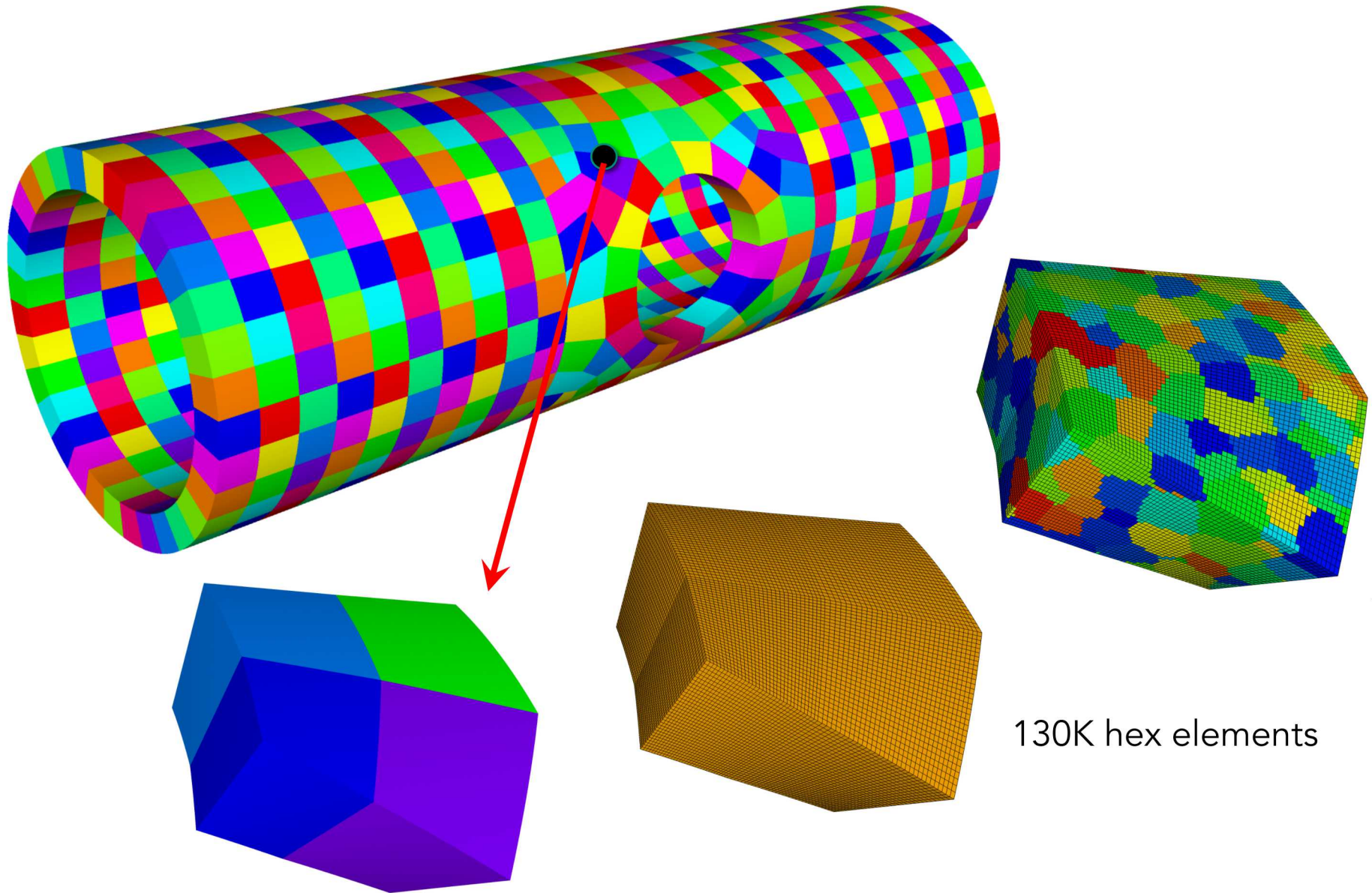


904 subdomains

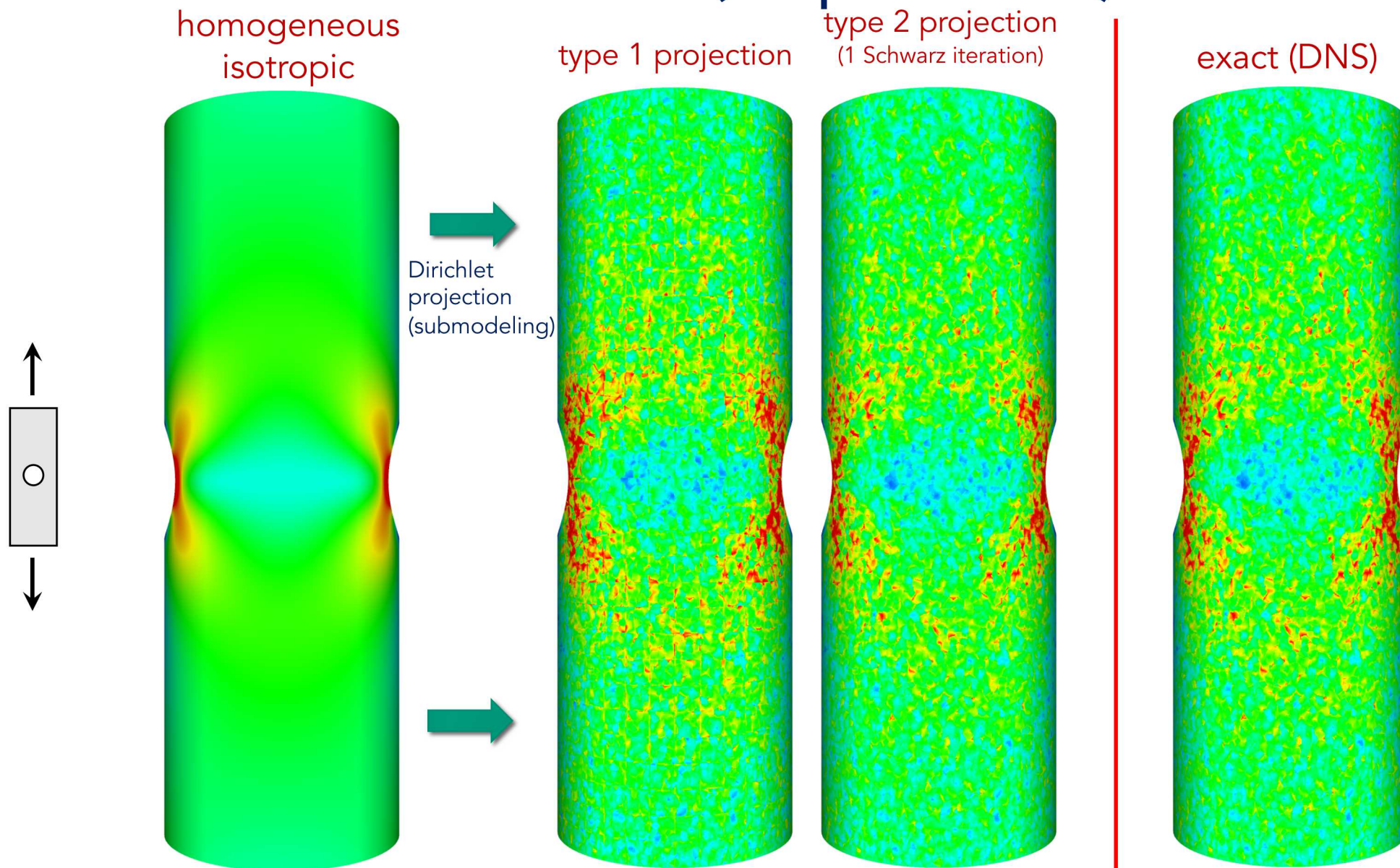
Type 1 localization



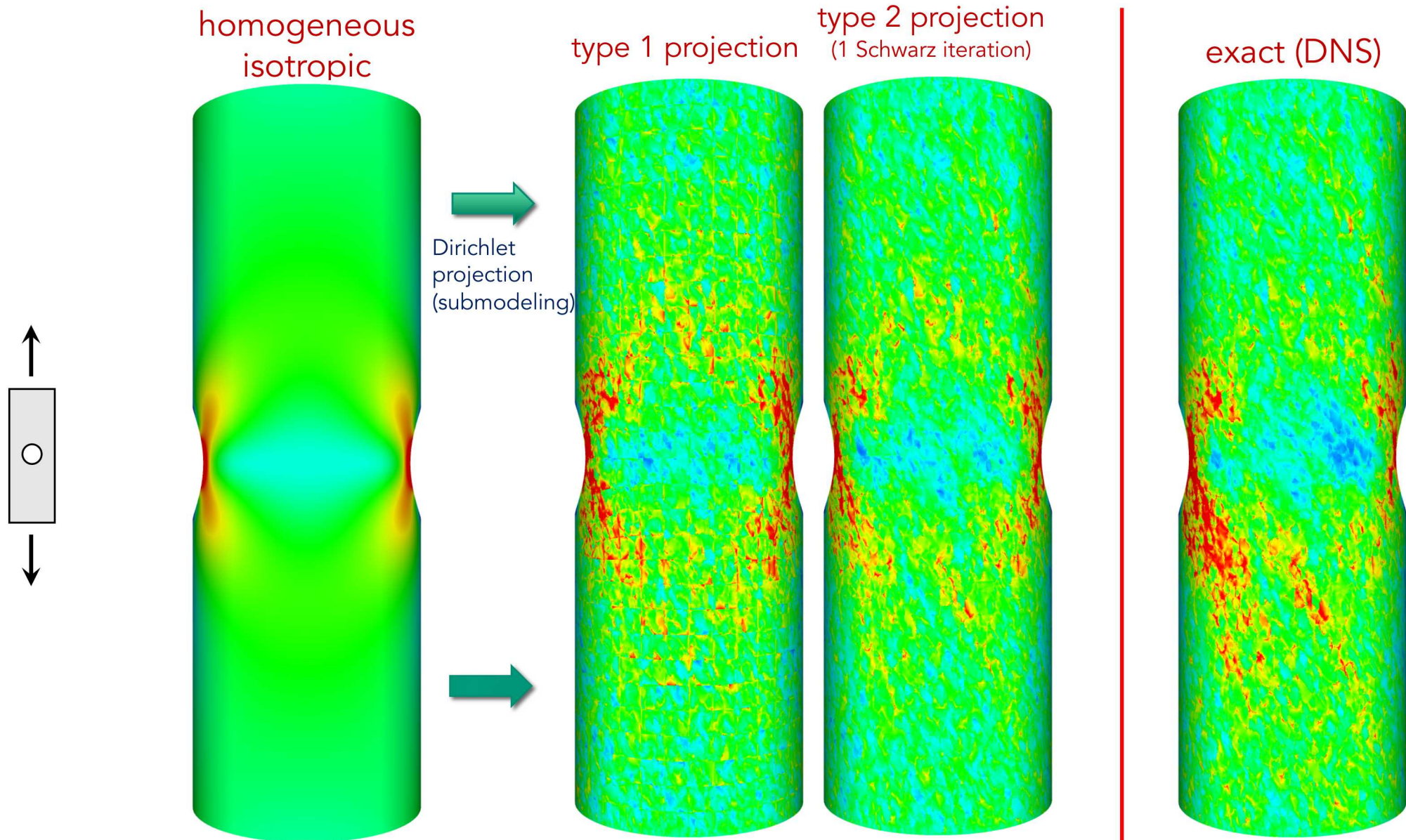
Type 2 localization (overlapping)



Localization results (equiaxed)



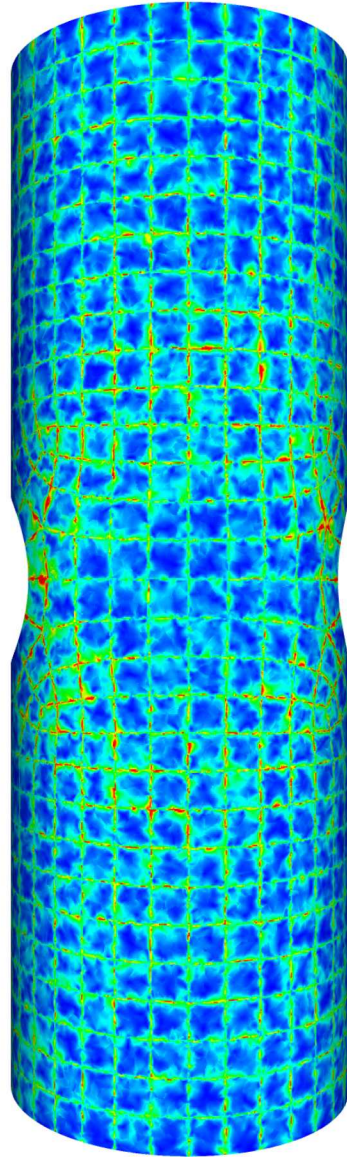
Localization stress results (AM)



Stress error

Equiaxed microstructure

Type 1

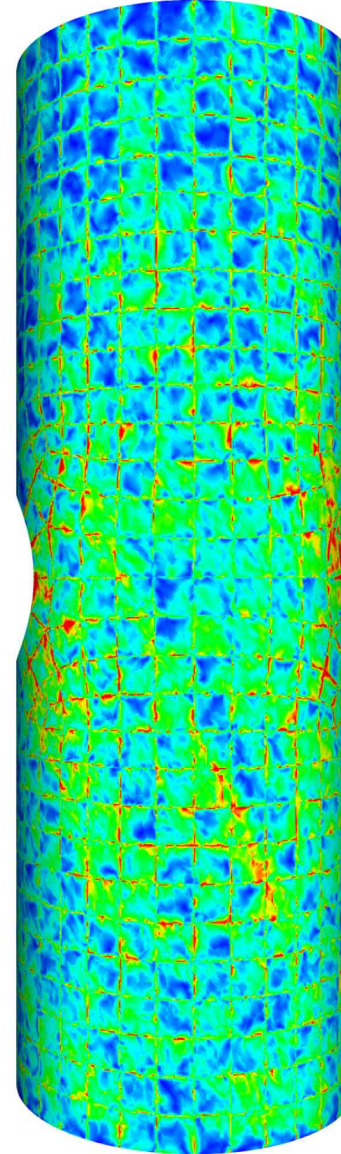


Type 2

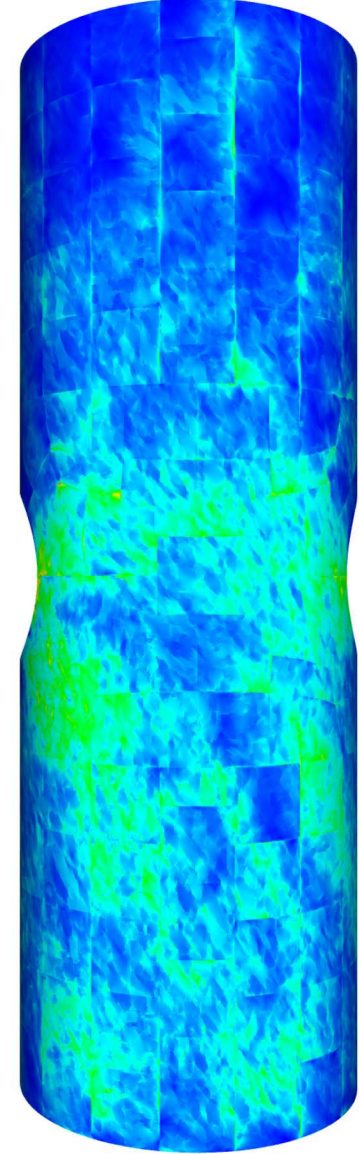


AM microstructure

Type 1



Type 2



How to adapt material properties to reduce error?

How can we decrease this error, but still use an isotropic material model?

Consider a **heterogeneous** isotropic material model?

$$\mathbb{C}^0 = \mathbb{C}^{\text{iso}} = a \mathbb{J} + b \mathbb{K} = 3K \mathbb{J} + 2\mu \mathbb{K}$$

$$\mathbb{J} = \frac{1}{3} \delta_{ij} \delta_{kl} \quad \text{hydrostatic operator}$$

$$\mathbb{K} = \mathbb{I} - \mathbb{J} \quad \text{deviatoric operator}$$

Let bulk modulus K and shear modulus μ vary with position.

How to determine? Minimize error mean constitutive response.

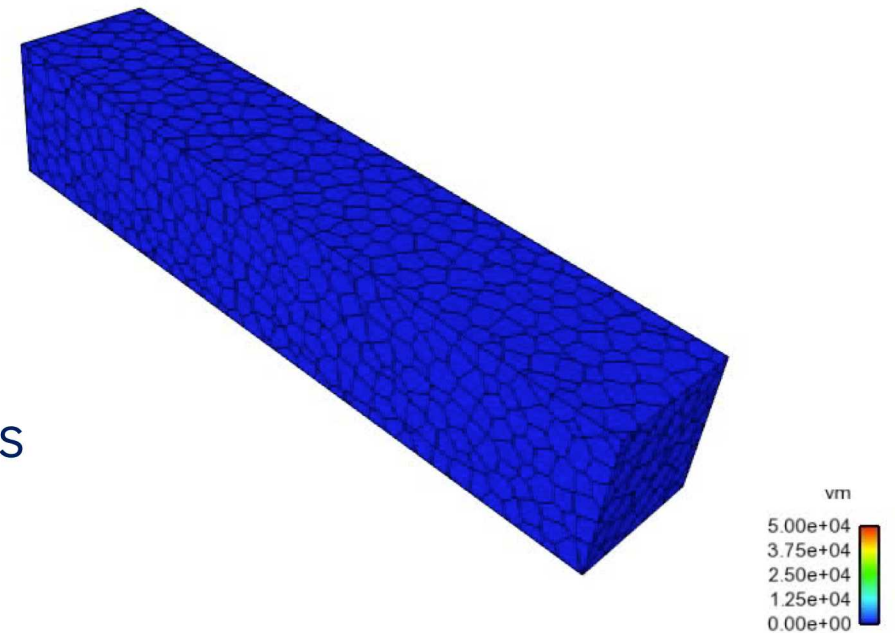
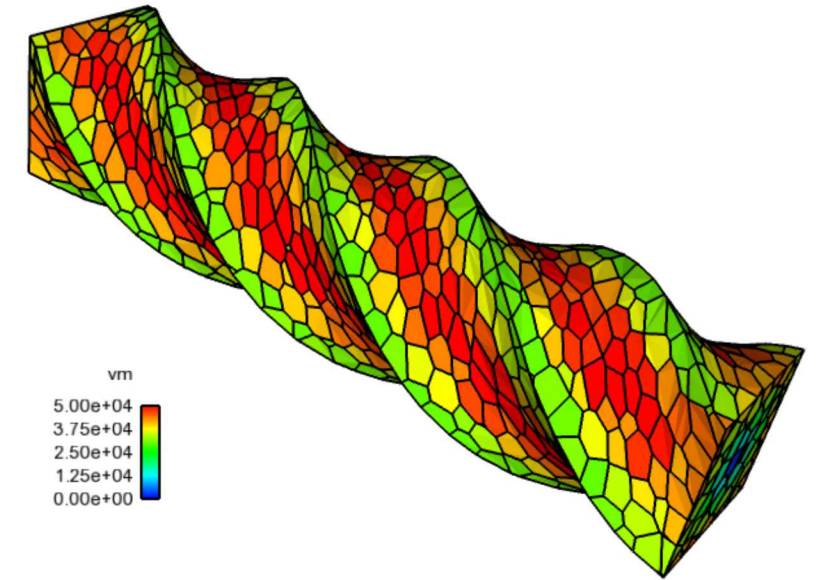
$$\left. \begin{aligned} e_{\sigma}^2 &:= ||\langle \boldsymbol{\sigma} \rangle - \mathbb{C} \langle \boldsymbol{\epsilon} \rangle||^2 \\ e_{\epsilon}^2 &:= ||\langle \boldsymbol{\epsilon} \rangle - \mathbb{S} \langle \boldsymbol{\sigma} \rangle||^2 \\ e^2 &:= [\langle \boldsymbol{\sigma} \rangle - \mathbb{C} \langle \boldsymbol{\epsilon} \rangle] : [\langle \boldsymbol{\epsilon} \rangle - \mathbb{S} \langle \boldsymbol{\sigma} \rangle] \end{aligned} \right\} \begin{aligned} \frac{\partial e^2}{\partial K} &= 0 \\ \frac{\partial e^2}{\partial \mu} &= 0 \end{aligned} \left\{ \begin{aligned} 3K &= \frac{\langle \sigma_{kk} \rangle}{\langle \epsilon_{kk} \rangle} \\ 2\mu &= \frac{\text{dev} \langle \boldsymbol{\sigma} \rangle : \text{dev} \langle \boldsymbol{\epsilon} \rangle}{\text{dev} \langle \boldsymbol{\epsilon} \rangle : \text{dev} \langle \boldsymbol{\epsilon} \rangle} \\ 2\mu &= \frac{\text{dev} \langle \boldsymbol{\sigma} \rangle : \text{dev} \langle \boldsymbol{\sigma} \rangle}{\text{dev} \langle \boldsymbol{\epsilon} \rangle : \text{dev} \langle \boldsymbol{\sigma} \rangle} \\ 2\mu &= \frac{\text{dev} \langle \boldsymbol{\sigma} \rangle : \text{dev} \langle \boldsymbol{\sigma} \rangle}{\text{dev} \langle \boldsymbol{\epsilon} \rangle : \text{dev} \langle \boldsymbol{\epsilon} \rangle} \end{aligned} \right.$$

Why polyhedra?

- Increased flexibility in finite element discretizations (tetrahedra and hexahedra are special cases)
- Enables hybrid meshing, e.g. hexahedral-dominant using frame fields
- Enables cut-cell approaches
- Voronoi meshing

How to generate meshes?

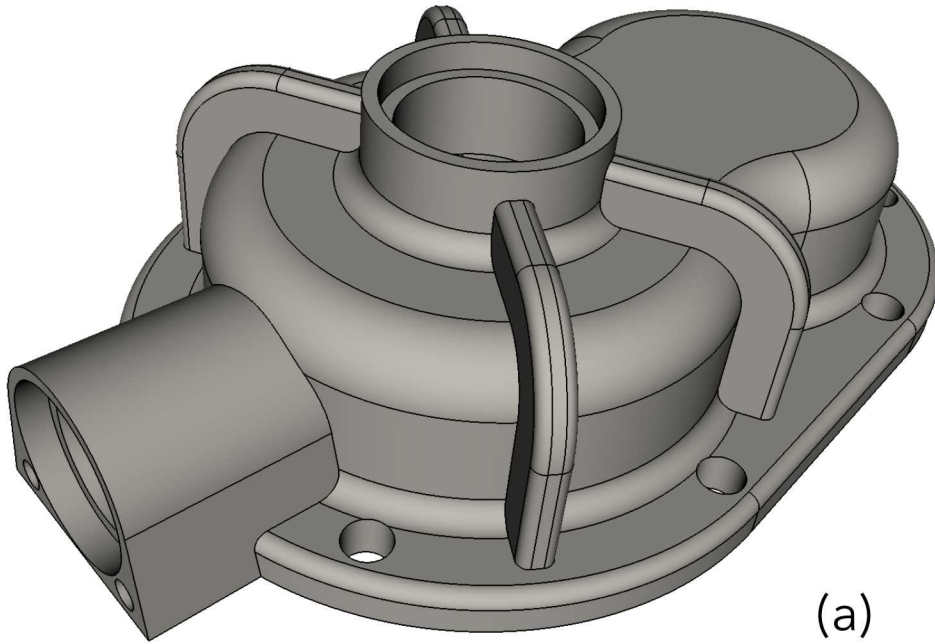
Voronoi is challenging on general shapes



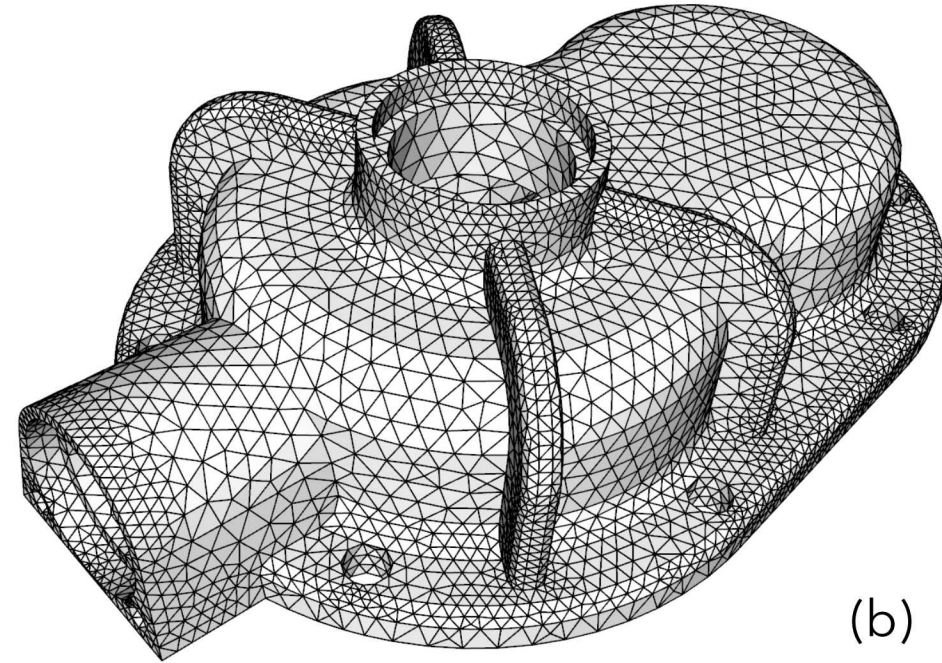
Another approach to forming polyhedral elements



Start with a tetrahedral mesh and then form dual polyhedral cells.

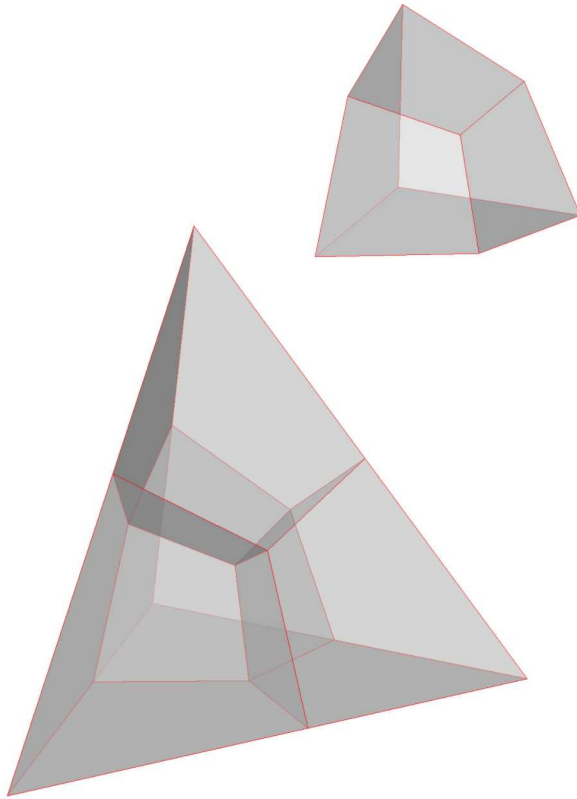


(a)

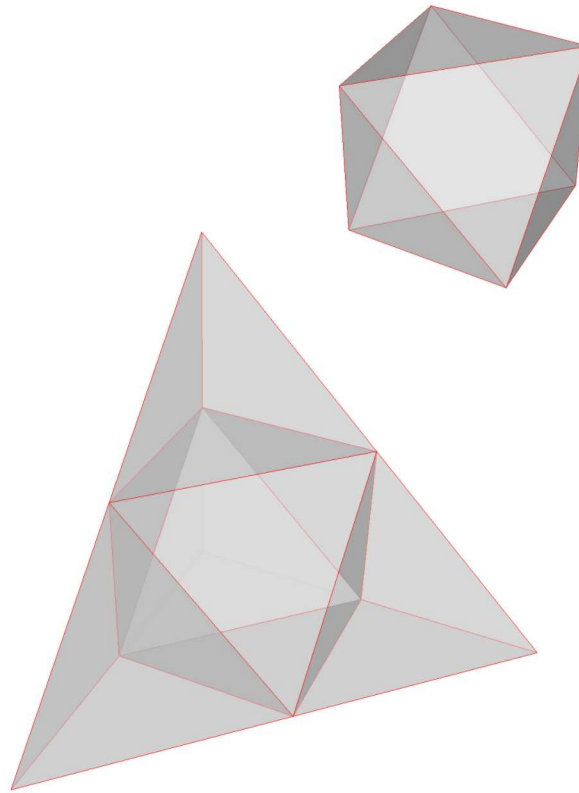


(b)

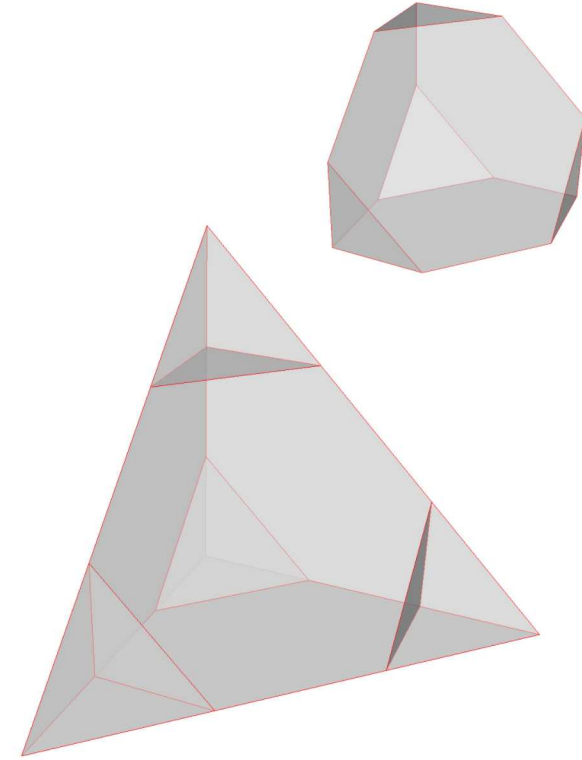
Three types of tetrahedral subdivision



barycentric

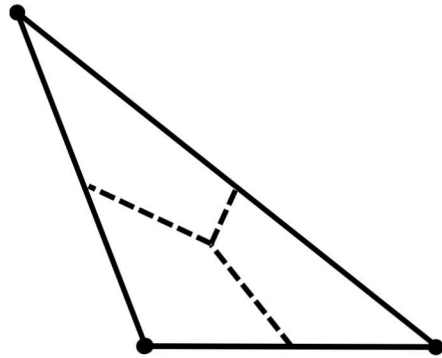


full truncation

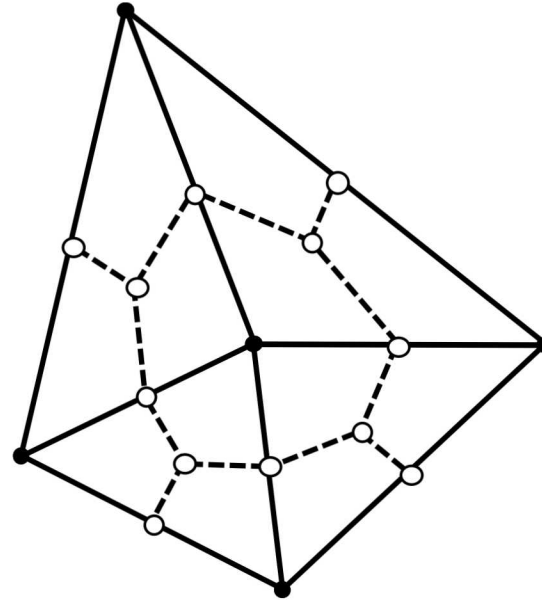


partial truncation

Barycentric subdivision and aggregation

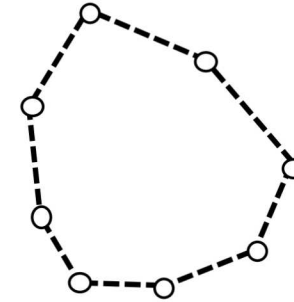


(a)



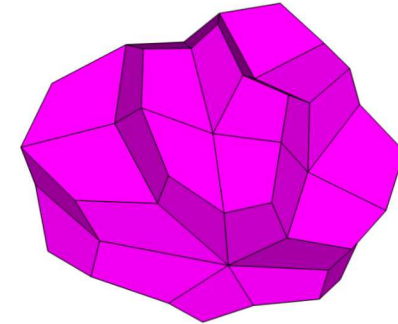
(b)

aggregate of
quadrilaterals in 2D



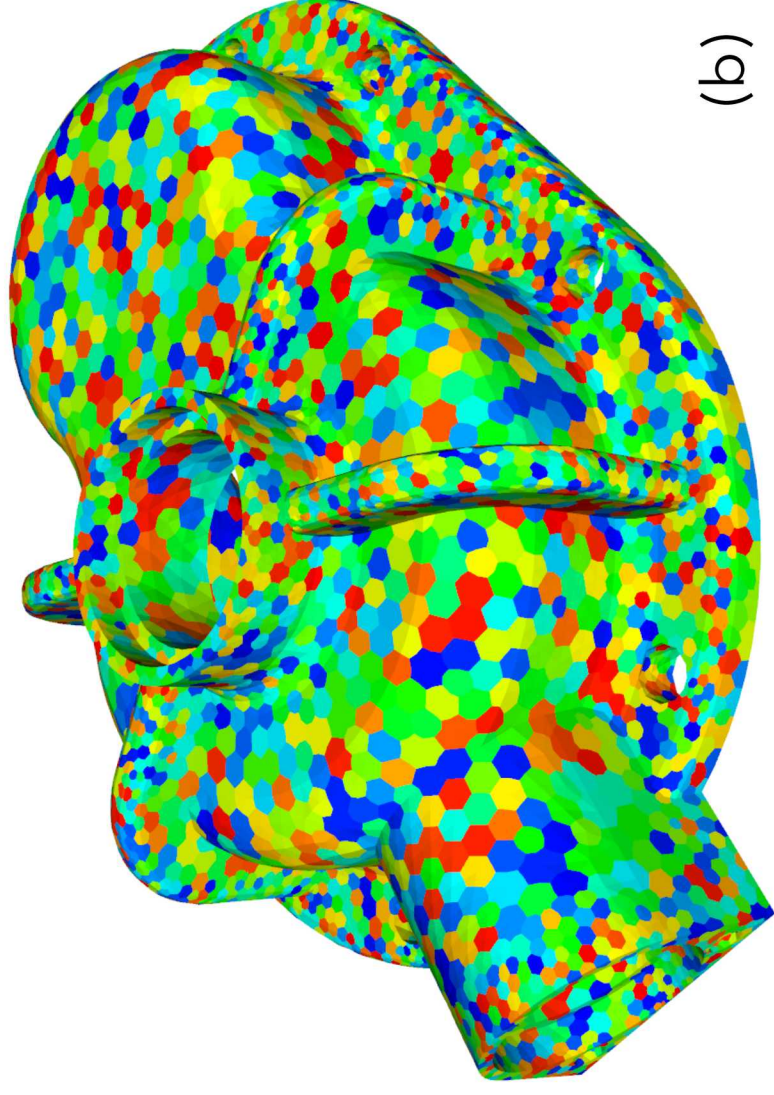
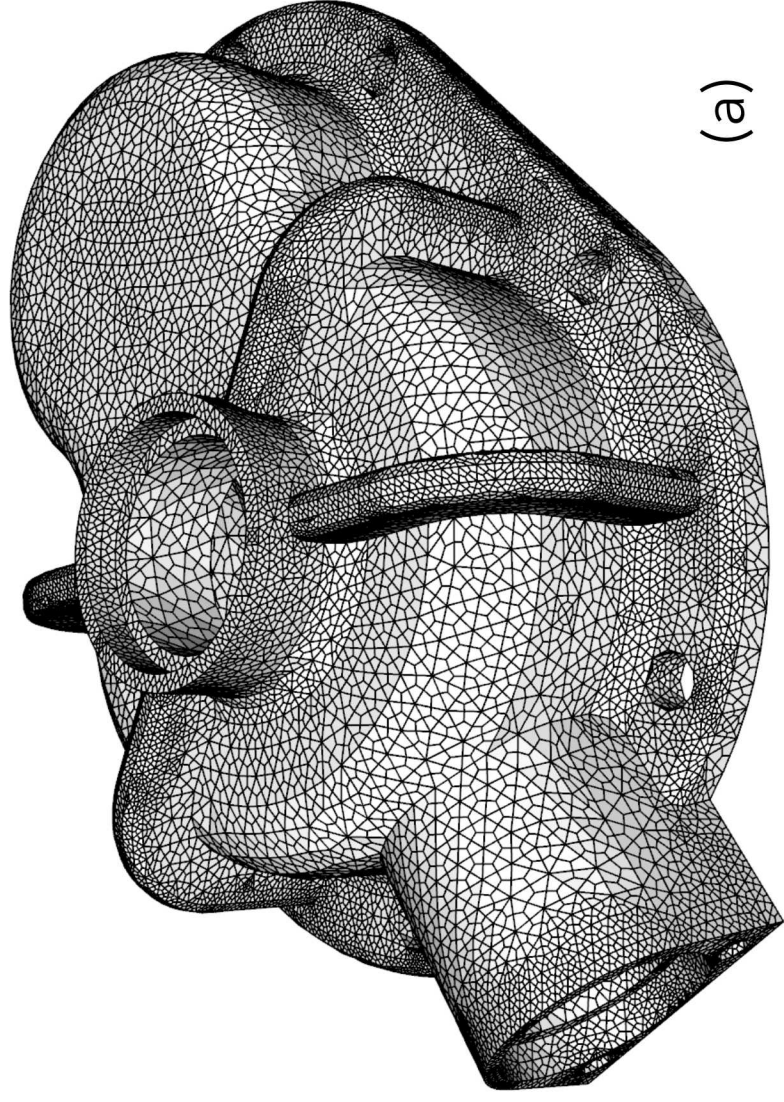
(c)

aggregate of
hexahedra in 3D

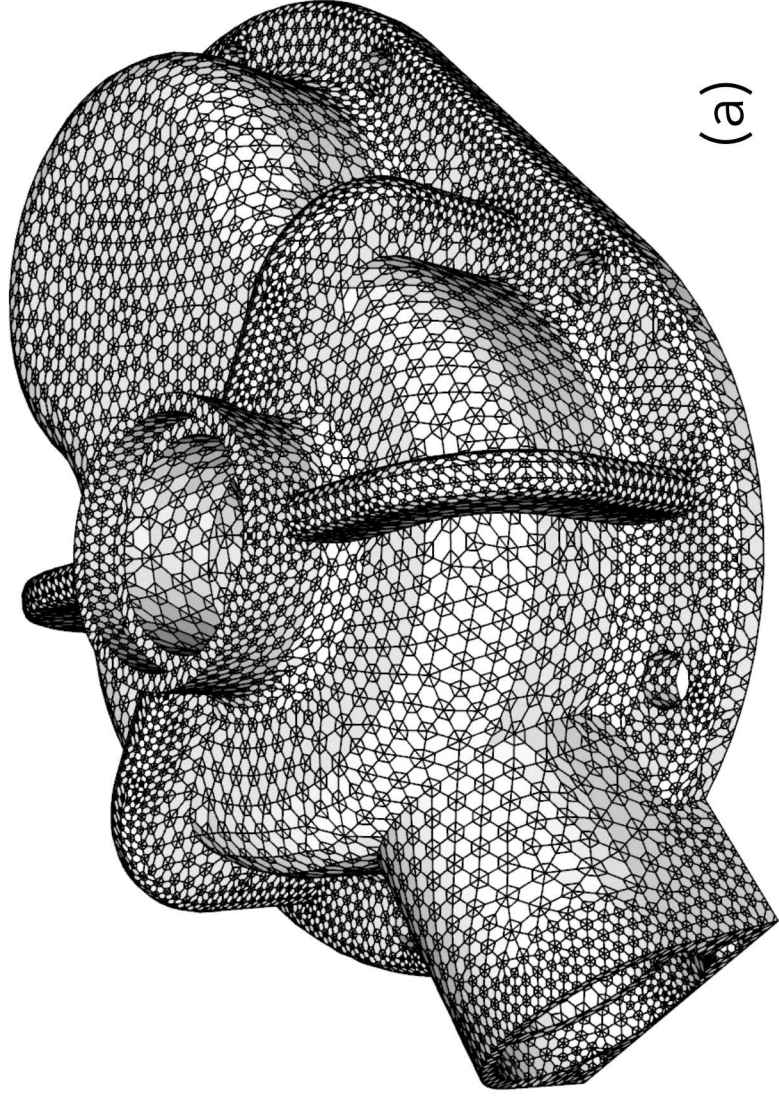


(d)

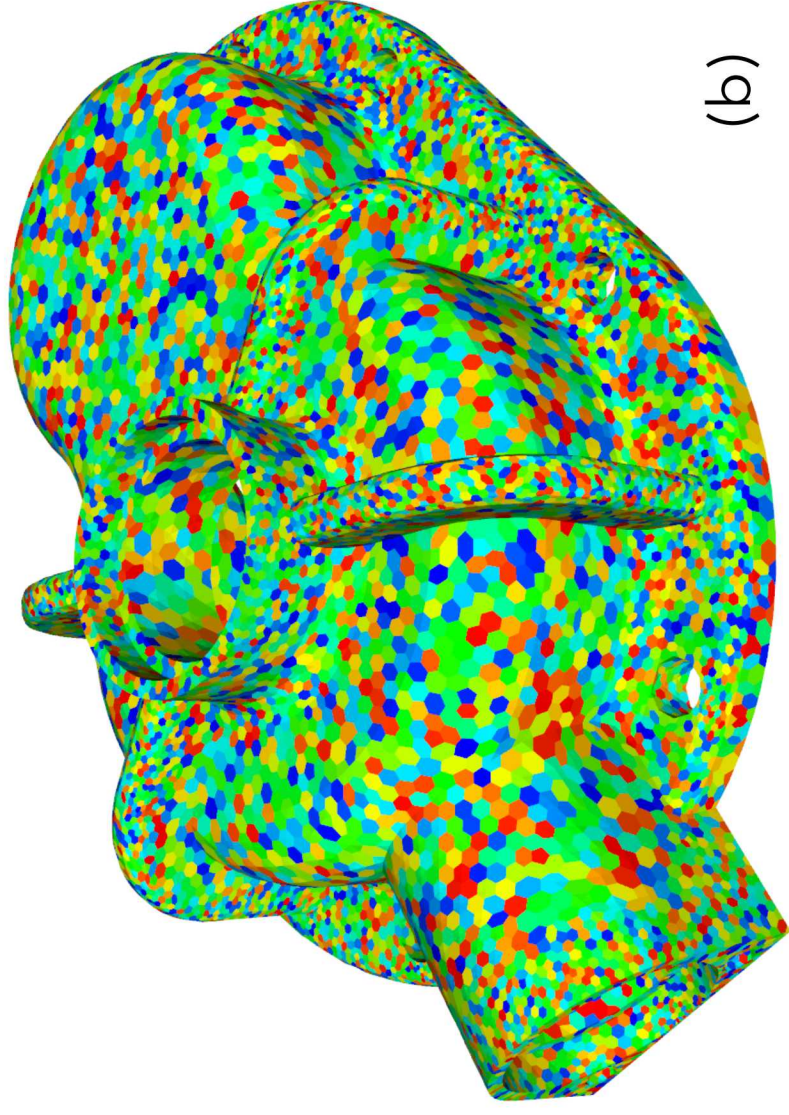
Barycentric subdivision and aggregation



Partial truncation and aggregation



(a)



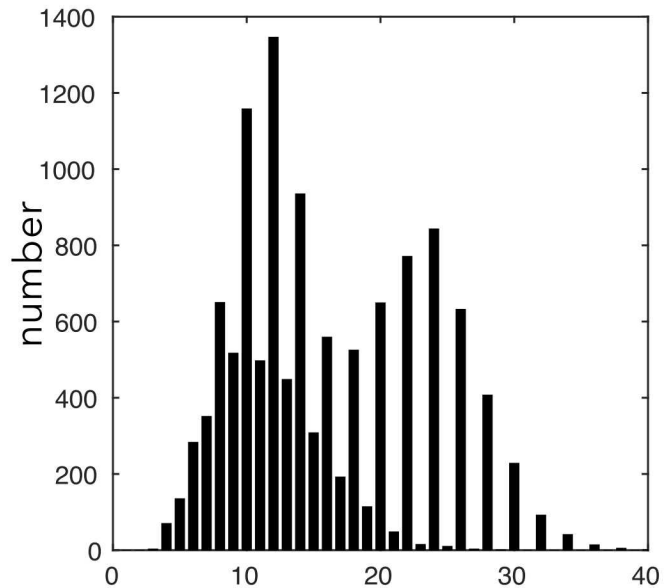
(b)

Element geometry



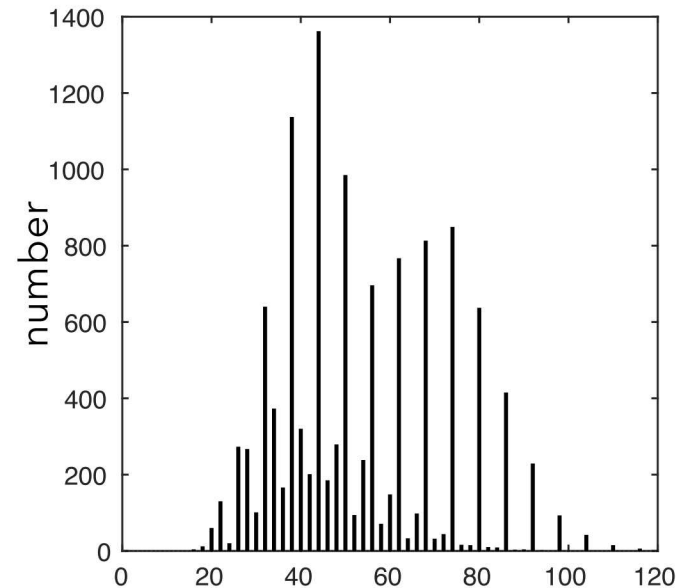
number of poly element vertices

number of attached
tetrahedra



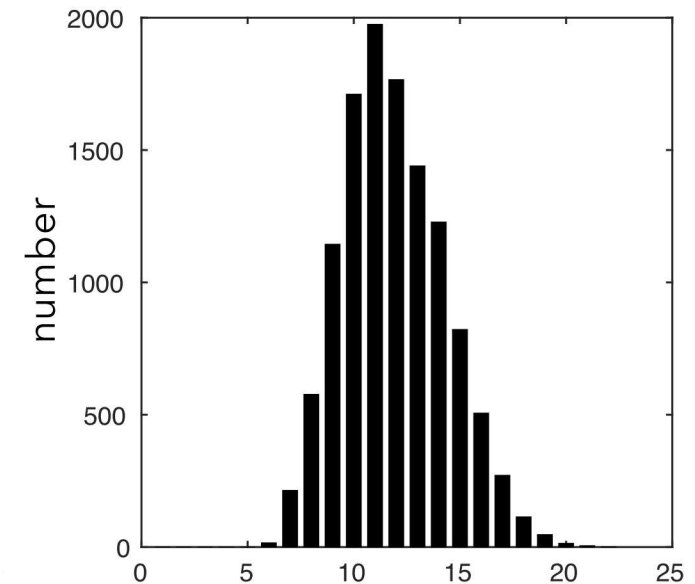
(a) number of attached tetrahedra

barycentric subdivision



(b) number of element vertices

truncated subdivision



(c) number of element vertices

Governing equations (total-Lagrangian formulation)

strong form

$$\frac{\partial \mathbf{P}}{\partial \mathbf{X}} : \mathbf{I} = \rho_0 \ddot{\mathbf{u}}$$

$$\mathbf{u} = \bar{\mathbf{u}} \quad \text{on} \quad \Gamma_0^u \quad \text{and} \quad \mathbf{P} \cdot \mathbf{N} = \mathbf{t}_0 \quad \text{on} \quad \Gamma_0^t$$

weak form

find the trial functions $\mathbf{u} \in \mathbf{H}^1(\Omega_0)$ such that

$$\int_{\Gamma_0^t} \mathbf{t}_0 \cdot \mathbf{v} \, dS - \int_{\Omega_0} \mathbf{P} : (\partial \mathbf{v} / \partial \mathbf{X}) \, d\mathbf{X} = \int_{\Omega_0} \rho_0 \ddot{\mathbf{u}} \cdot \mathbf{v} \, d\mathbf{X}$$

for all test functions $\mathbf{v} \in \mathbf{H}_0^1(\Omega_0)$

Finite element formulation

- Bubnov-Galerkin formulation (trial and test functions use same space)
 - Total-Lagrangian formulation (integrate weak form on original configuration).
 - Minimize number of integration points while avoiding artificial stabilization.
 - Typically, number of quadrature points $\sim$ number of vertices (first order).
 - Mean-dilation ($\bar{F}$) formulation for nearly-incompressible materials
 - Compatible with standard trilinear hexahedron
-
- Going to explore the use of two types of shape functions:
 1. harmonic,
 2. maximum entropy (max-ent)

Harmonic shape functions

Harmonic functions minimize the Dirichlet energy given by the following functional:

$$J(\psi) := \frac{1}{2} \int_{\Omega_e} \nabla \psi \cdot \nabla \psi \, d\mathbf{X} \quad \text{with } \psi \in H^1(\Omega_e)$$

The minimizer of this functional satisfies the following variational problem:

find $\psi \in H^1(\Omega_e)$ with $\psi = \bar{\psi}$ on Γ_e such that

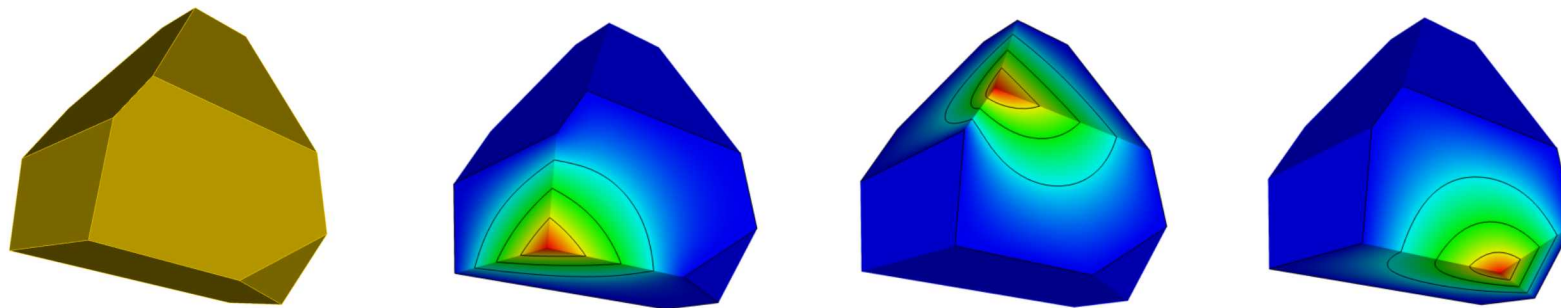
$$\int_{\Omega_e} \nabla \psi \cdot \nabla v \, d\mathbf{X} = 0$$

for all test functions $v \in H_0^1(\Omega_e)$

The strong form of this variational problem is given by:

$$\nabla^2 \psi = 0 \quad \text{in } \Omega_e \quad \text{with } \psi = \bar{\psi} \quad \text{on } \Gamma_e$$

Harmonic shape functions



$$\sum_{a=1}^{N_v} \psi_a(\mathbf{X}) = 1$$

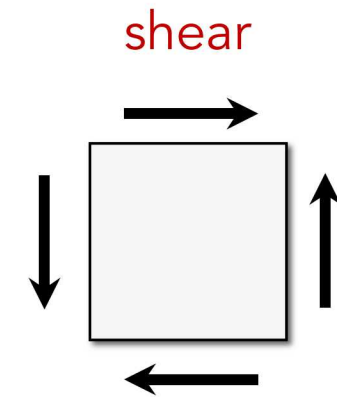
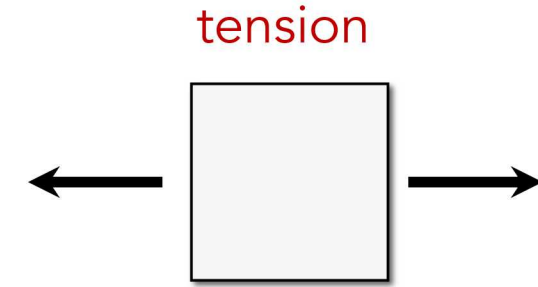
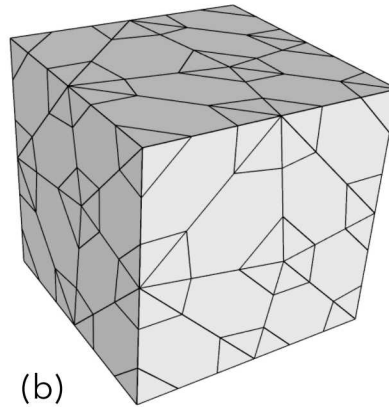
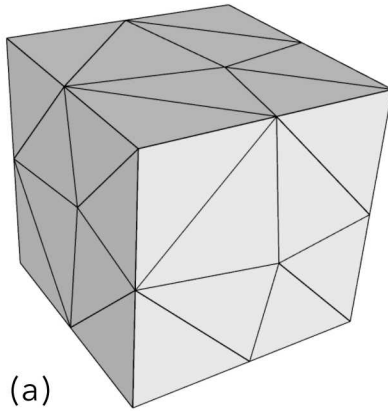
partition of unity

$$\sum_{a=1}^{N_v} \psi_a(\mathbf{X}) \mathbf{X}_a = \mathbf{X}$$

linear reproducibility

Patch test

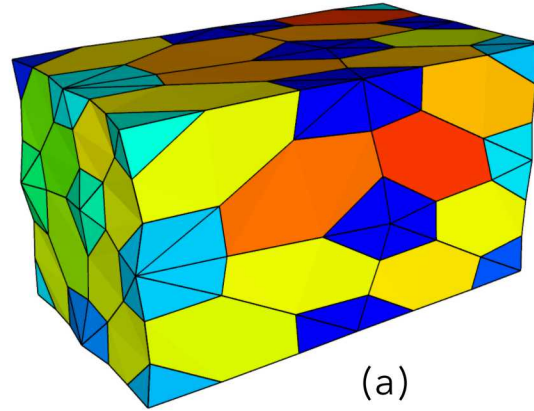
- a test for element consistency
- Uniform tractions should result in uniform stress/strain field.



Patch test

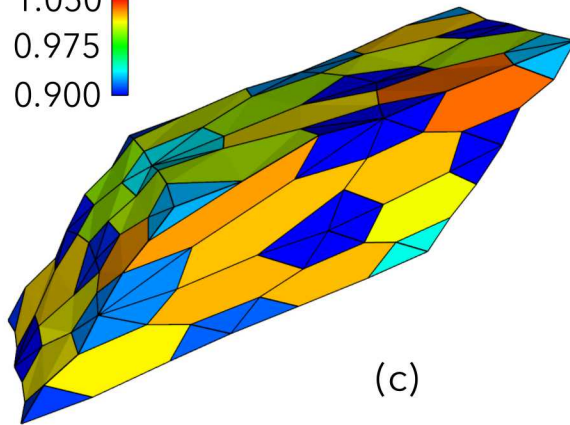


tension



1.050
0.975
0.900

shear



- Failed patch test!
- Can be traced to use of low-order integration scheme.
- Due to lack of "integration consistency" or violation of the discrete divergence theorem.

Integration consistency



Divergence theorem states that:

$$\int_{\Omega_e} \nabla \psi_a \, d\mathbf{X} = \int_{\Gamma_e} \psi_a \, \mathbf{N} \, dS$$

where $\mathbf{N}$ is the outward unit normal vector on Γ_e

In discrete form:

$$\sum_{k=1}^{N_Q} w_k \nabla \psi_{ak} = \sum_{l=1}^{N_Q^\Gamma} w_l^\Gamma \psi_{al} \mathbf{N}_l, \quad (a = 1, \dots, N_v)$$

where $\psi_{ak} := \psi_a(\mathbf{X}_k)$, and $\mathbf{X}_k$ is the position of the k -th quadrature point

- For non-polynomial shape functions, this will not be satisfied in general.
- This will result in a lack of consistency (failure of the engineering patch test).

Shape function derivative correction

- “Tweak” the shape function derivatives to satisfy the integration consistency condition.
- Maintain the reproducing properties of the derivatives.
- Minimize the least-squares difference between the new derivatives and the old.
- Only performed once during simulation (pre-processing step).

$$\min_{\boldsymbol{\xi}_k \in \mathbf{R}^3} \sum_{k=1}^{N_Q} w_k \|\boldsymbol{\xi}_k - \nabla \psi_{ak}\|^2 \quad \text{subject to the constraints} \quad \sum_{k=1}^{N_Q} w_k \boldsymbol{\xi}_k - \sum_{l=1}^{N_Q^\Gamma} w_l^\Gamma \psi_{al} \mathbf{N}_l = 0$$

This constrained optimization problem can be solved using the method of Lagrange multipliers:

$$\mathcal{L}(\boldsymbol{\xi}_k, \boldsymbol{\lambda}) := \sum_{k=1}^{N_Q} w_k \|\boldsymbol{\xi}_k - \nabla \psi_{ak}\|^2 + \boldsymbol{\lambda} \cdot \left(\sum_{k=1}^{N_Q} w_k \boldsymbol{\xi}_k - \sum_{l=1}^{N_Q^\Gamma} w_l^\Gamma \psi_{al} \mathbf{N}_l \right)$$

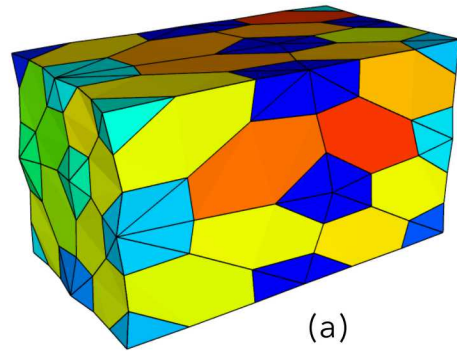
Patch test



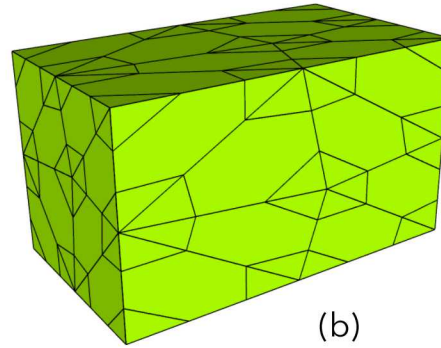
failed patch test

successful

tension

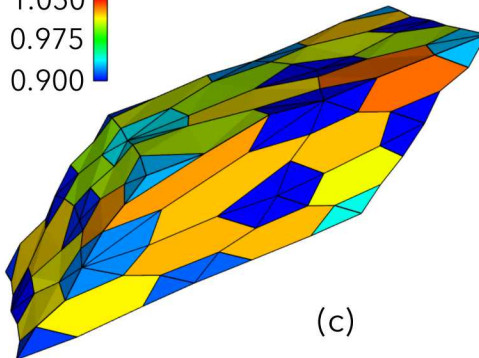


(a)

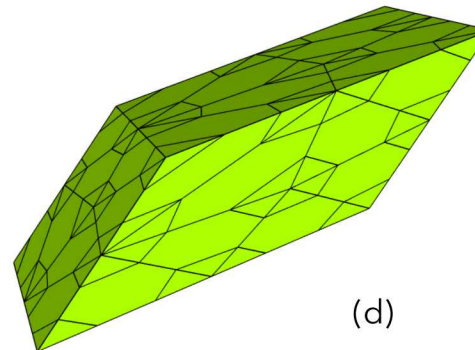


(b)

1.050
0.975
0.900



(c)



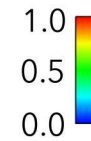
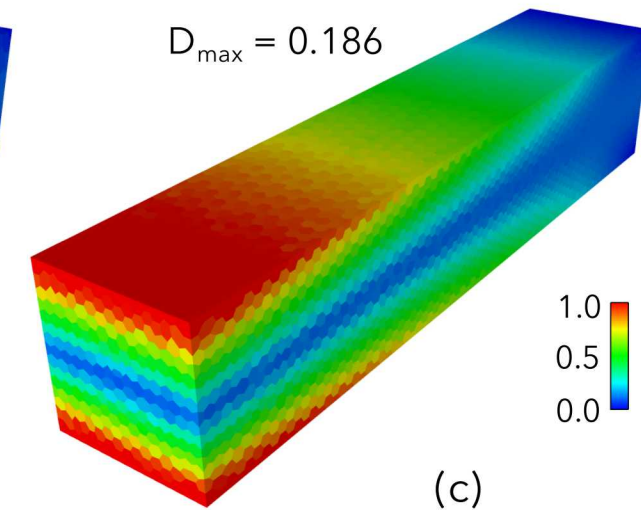
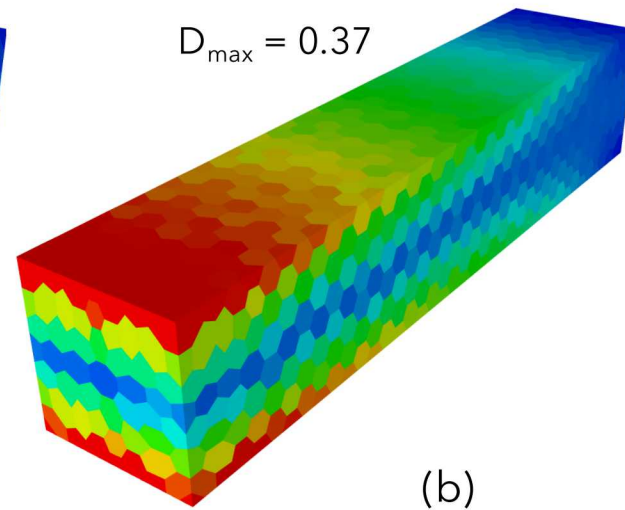
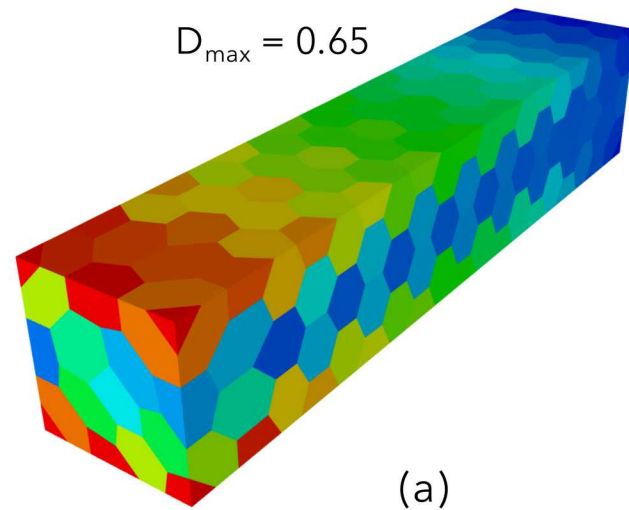
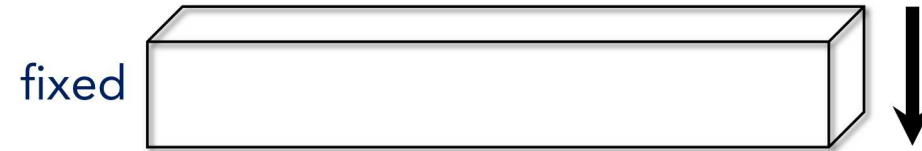
(d)

shear

Error in stress with and without correction

applied traction state	without correction	with correction
tension	0.18	$9.6 \cdot 10^{-13}$
shear	0.13	$2.7 \cdot 10^{-12}$

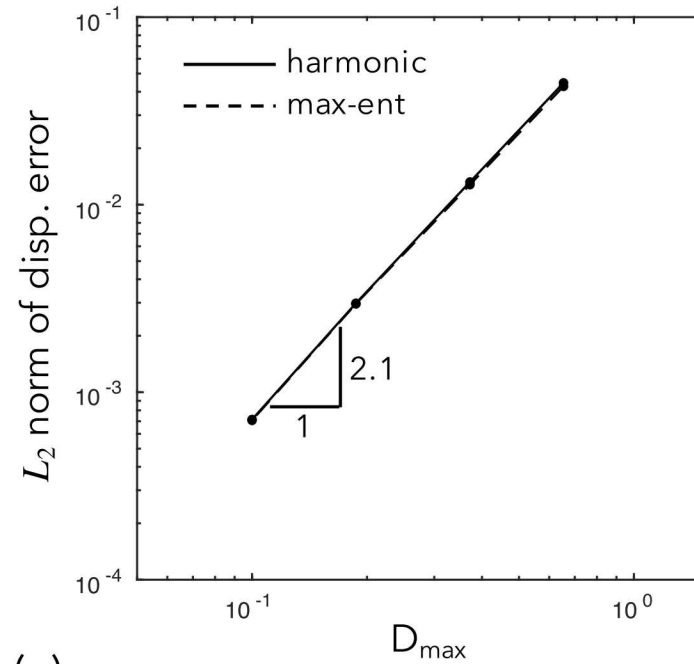
Beam bending with shear load



Convergence, harmonic vs. max-ent

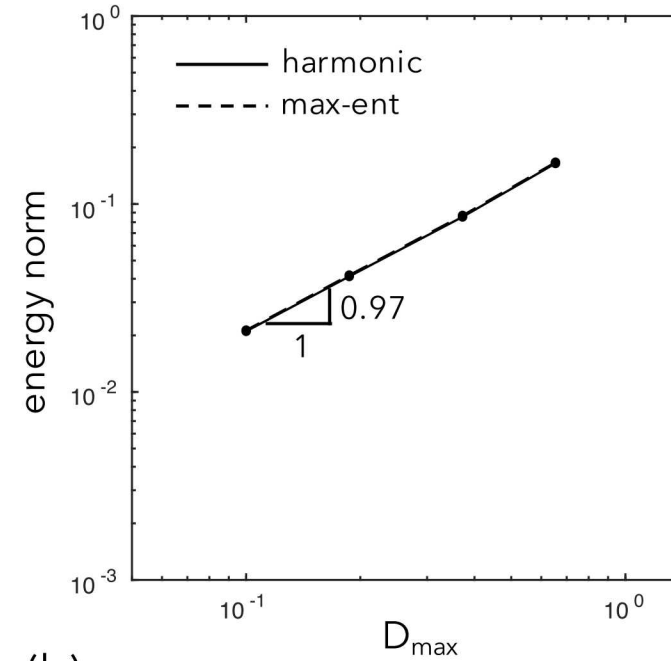


L_2 norm



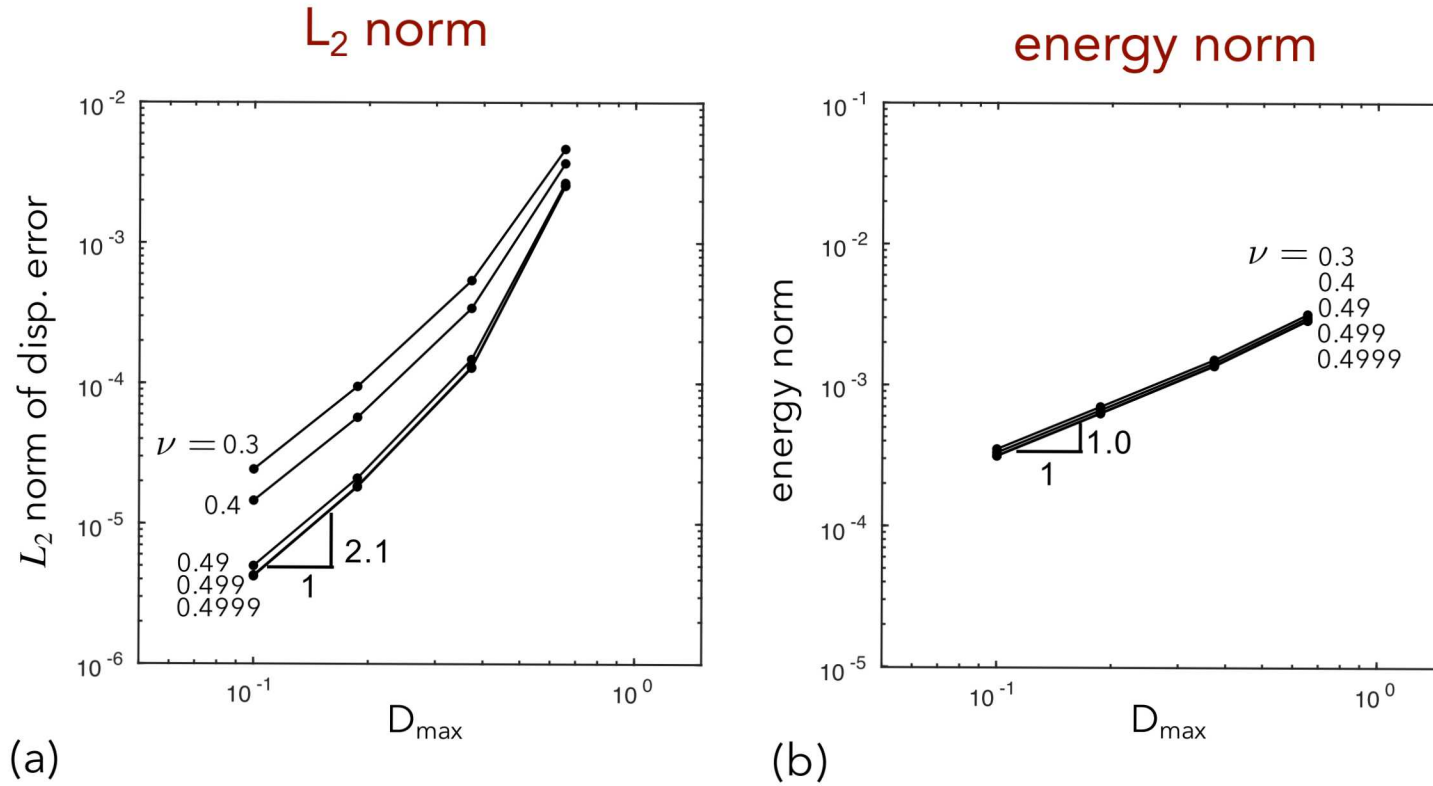
(a)

energy norm

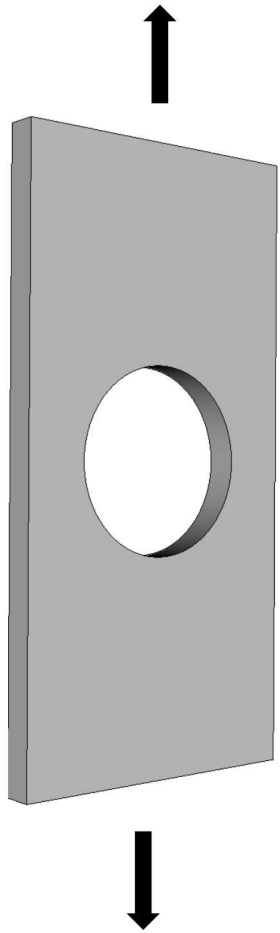


(b)

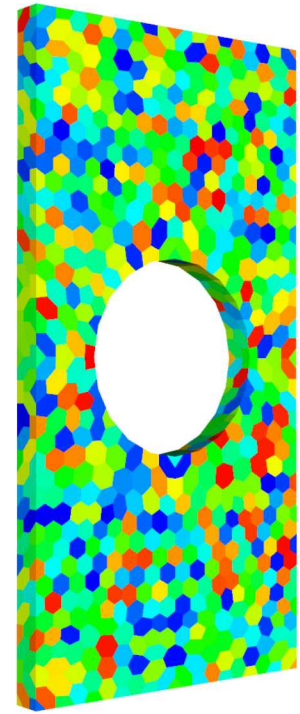
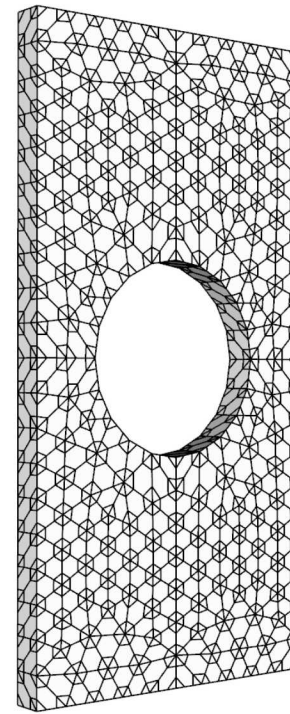
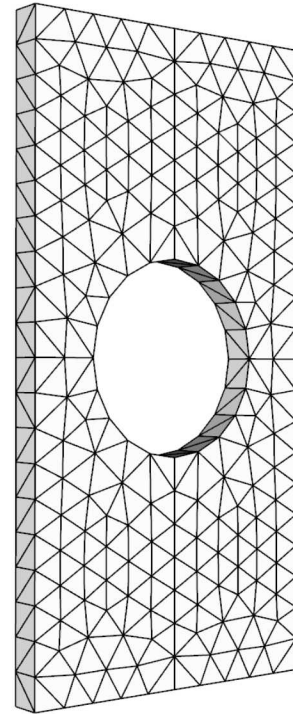
Convergence, harmonic vs. max-ent



Example: elastic-plastic plate



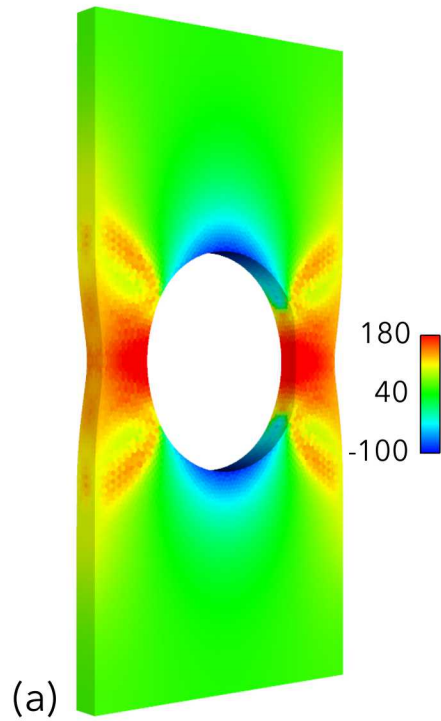
- uniaxial extension
- elastic-plastic constitutive model (J2)



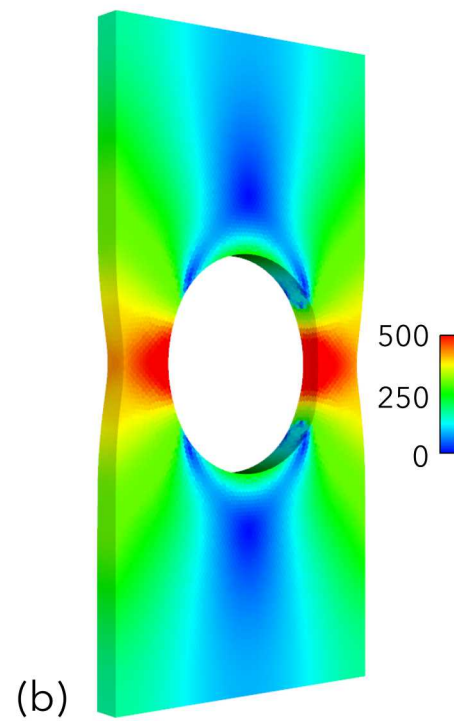
Simulation results



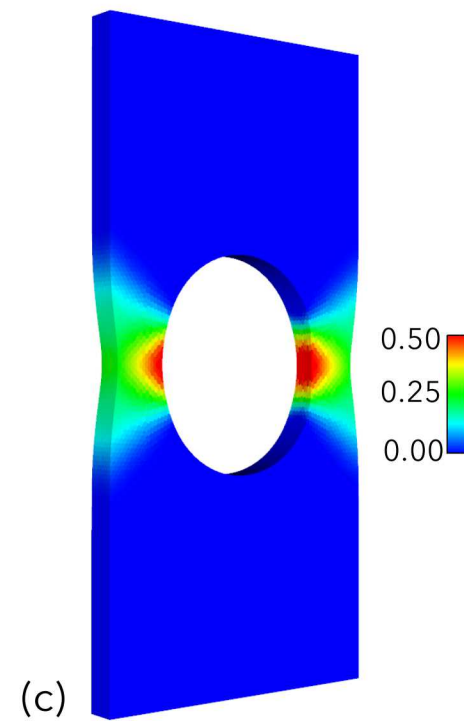
hydrostatic stress



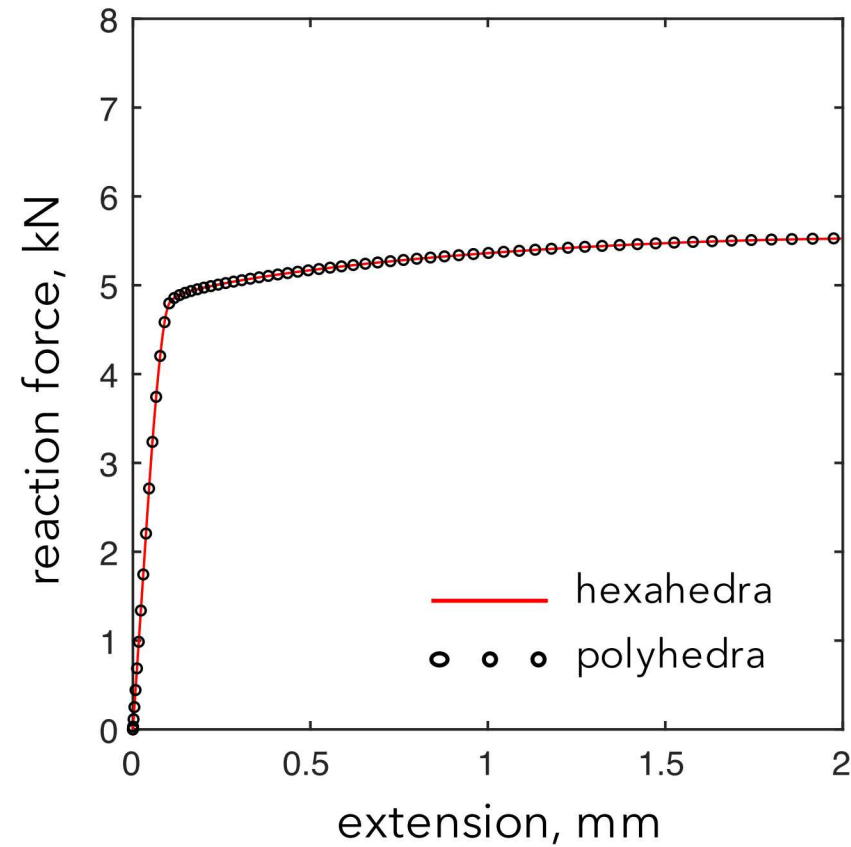
von Mises stress



equivalent plastic strain



Comparison of reaction forces





Model-form error estimation, DNS

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Polyhedral FEA

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- Bishop, J., (2014) "A displacement-based finite element formulation for general polyhedra using harmonic shape functions," *Int. J. for Numerical Methods in Engineering*, 97:1-31.
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