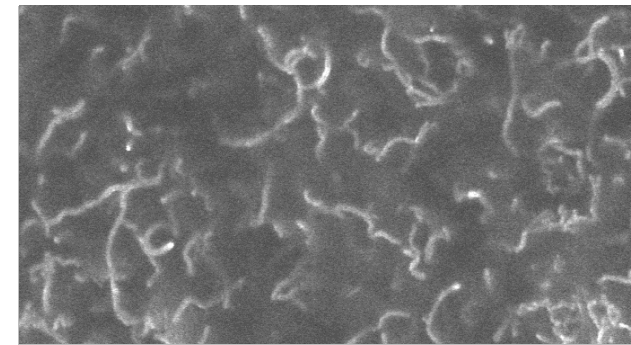
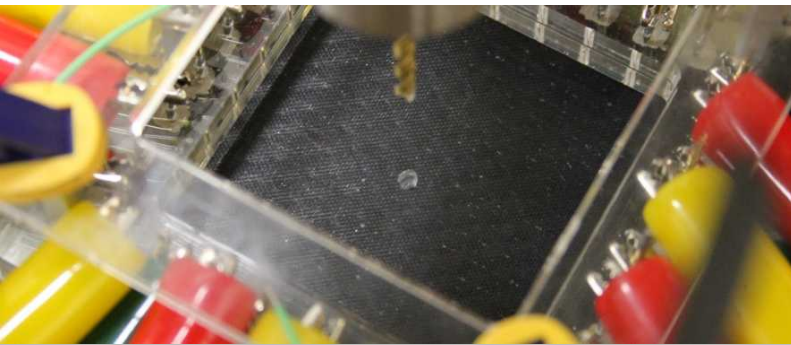


Exceptional service in the national interest



Spatially Distributed Structural Health Monitoring using Electrical Impedance Tomography

Bryan R. Loyola, Steven Paradise, and Christopher Hall

¹Sandia National Laboratories, Livermore, CA, USA



Sandia National Laboratories is a multi-program laboratory managed and operated by Sandia Corporation, a wholly owned subsidiary of Lockheed Martin Corporation, for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-AC04-94AL85000. SAND NO. 2011-XXXXP



Usage of Fiber-Reinforced Composites

- Over the past 50 years, increased usage of composite materials



Commercial aircraft systems



Future and legacy spacecraft



Military aircraft



Naval structures



Wind turbine blades



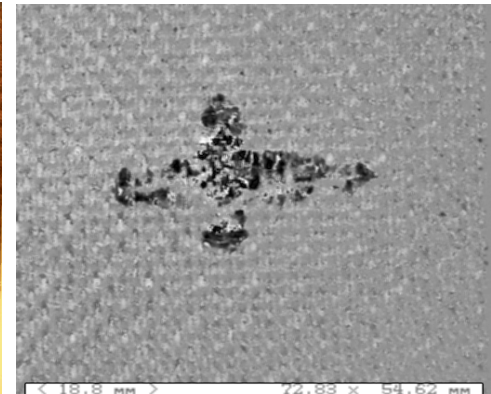
CFRP cable stay bridge

Composite Damage Modes

- Susceptible to damage due to:
 - Strain, impact, chemical penetrants, multi-axial fatigue
- Damage modes:
 - Matrix cracking
 - Fiber-breakage
 - Delamination
 - Transverse cracking
 - Fiber-matrix debonding
 - Matrix degradation
 - Blistering
- Difficult to detect
 - Internal to laminate structure
 - Nearly invisible to naked eye
 - Current methods are laborious



Visual inspection



C-SCAN ultrasound image

CFRP panel after 20 Joule impact



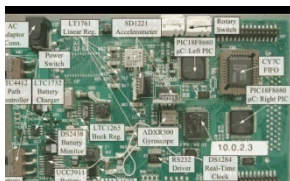
Aircraft ultrasonic inspection (Composites World)

Emerging Sensing Technologies

Wireless Sensors and Sensor Networks



WiMMS



UCI DuraNode

Wang, *et al.* (2008)

Chung, *et al.* (2005)

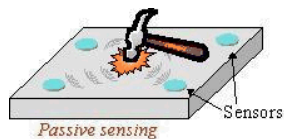
Advantages:

- Low Cost
- Dense instrumentation
- Reconfigurable

Disadvantages:

- Point sensors
- In direct damage detection
- Physics-based models

Ultrasonics and Guided-Waves



Array of piezoelectric ceramic sensors and actuators

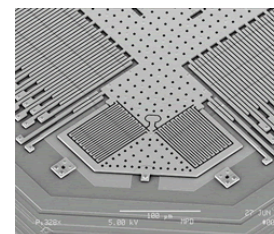
Advantages:

- Sensors and actuators
- Spatial damage detection

Disadvantages:

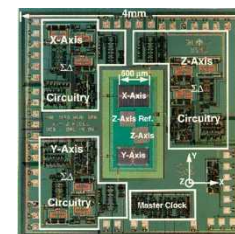
- Indirect damage detection
- Wave propagation models or pattern recognition
- Thin structures
- Expensive data acquisition

Micro-electromechanical Systems (MEMS)



AD iMEMS

Weinberg (1999)



3-axis accelerometer

Lemkin (1997)

Advantages:

- Miniaturized sensor designs
- Complex sensors/actuators

Disadvantages:

- "Top-down" design
- Expensive fabrication equipment
- High costs
- Sensor sensitivity on par with macro-scale counterpart

SHM Design Considerations

Current SHM limitations:

- Indirect sensing approaches
- Point-based sensing
- Tethered sensors
- Lack of system scalability



Boeing 787 (Boeing)

Successful SHM systems:

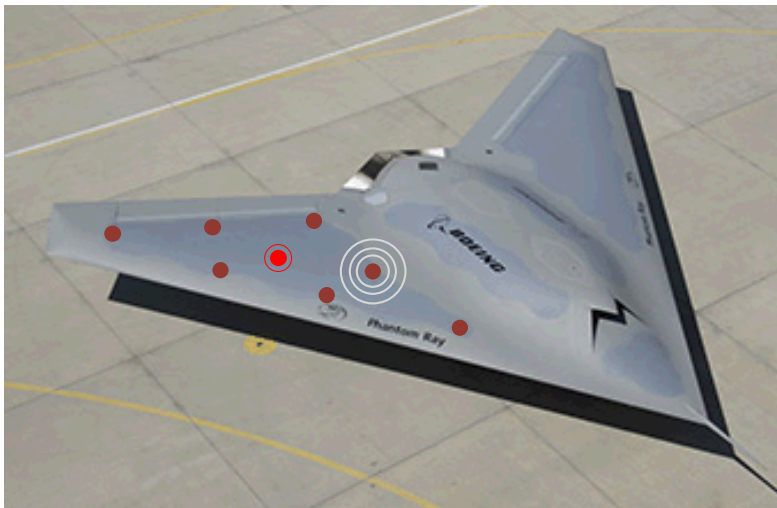
1. Directly detect and measure damage
2. Determine the damage location
3. Ascertain the size of the damage
4. Quantify the severity of the damage
5. Achieve multi-modal sensing capabilities (i.e., delamination, cracking, and chemical penetration)



Golden Gate Bridge (Wikipedia)

Spatially Distributed SHM Paradigm

- Current state-of-art in structural health monitoring:
 - Passive SHM using acoustic emissions
 - Active SHM using piezoelectric sensor/actuator pairs
- “Sensing skins” for spatial damage detection:
 - Objective is to identify the location and severity of damage
 - Monitor and detect damage over two- (or even three) dimensions
 - Direct damage detection

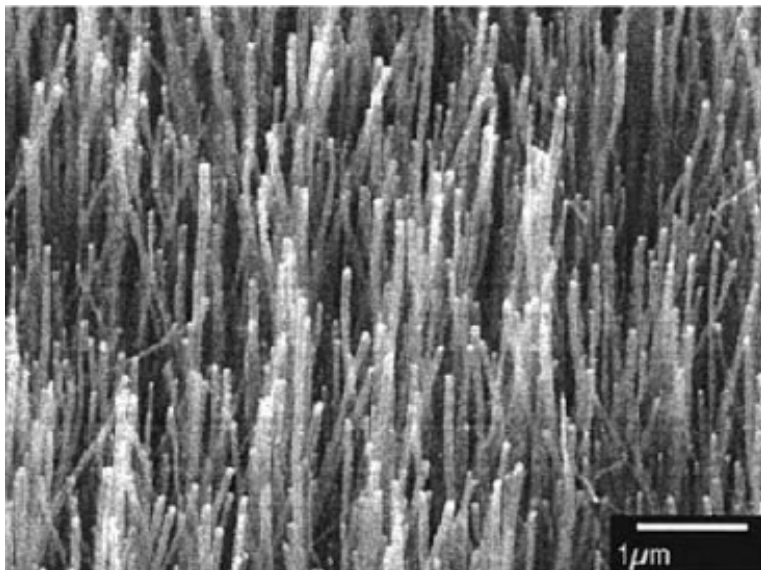


(Boeing)

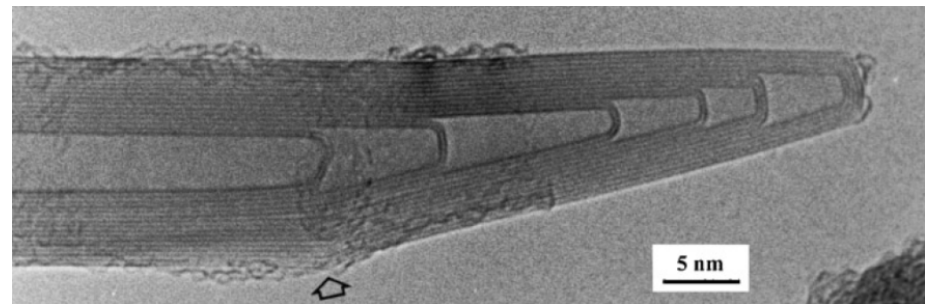
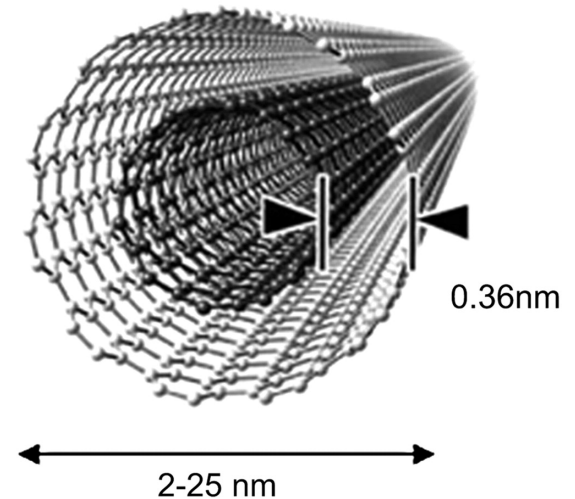


(Boeing)

- Multi-walled carbon nanotubes (MWNT):
 - Rolled concentric cylindrical structures constructed of graphene sheets
 - Diameter: 6 ~ 100 nm
 - High-aspect ratios: $\sim 10^3$ to 10^7
 - Metallic conductivity
 - Five times stiffer and ten times stronger than steel



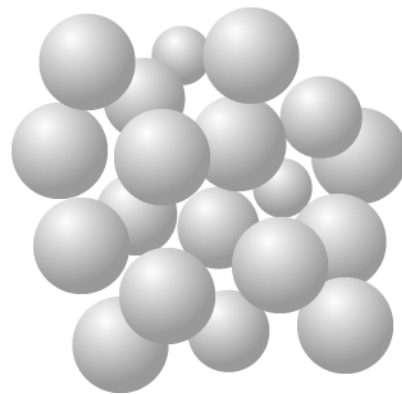
Aligned carbon nanotube forest
Thostenson, et al. (2001)



TEM imagery of an end cap of a MWNT
Harris (2004)

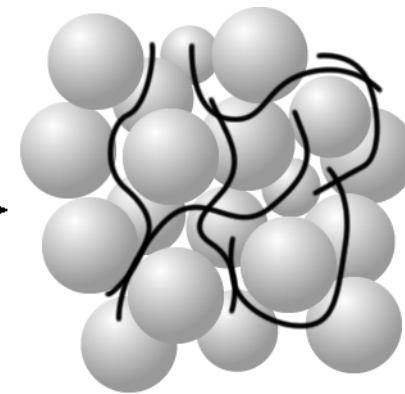
Sprayable MWNT-Latex Thin Film

- Rapid large-scale deposition
 - Required for mass deployment of methodology
- MWNT-PSS/Latex paint formulation
 - Collaborated to improve initial Sandia formulation
 - Sub-micron PVDF creates mold for MWNT organization
 - Off-the-shelf deposition method



Kynar Aquatec™ latex solution
(avg. particle size 150nm)

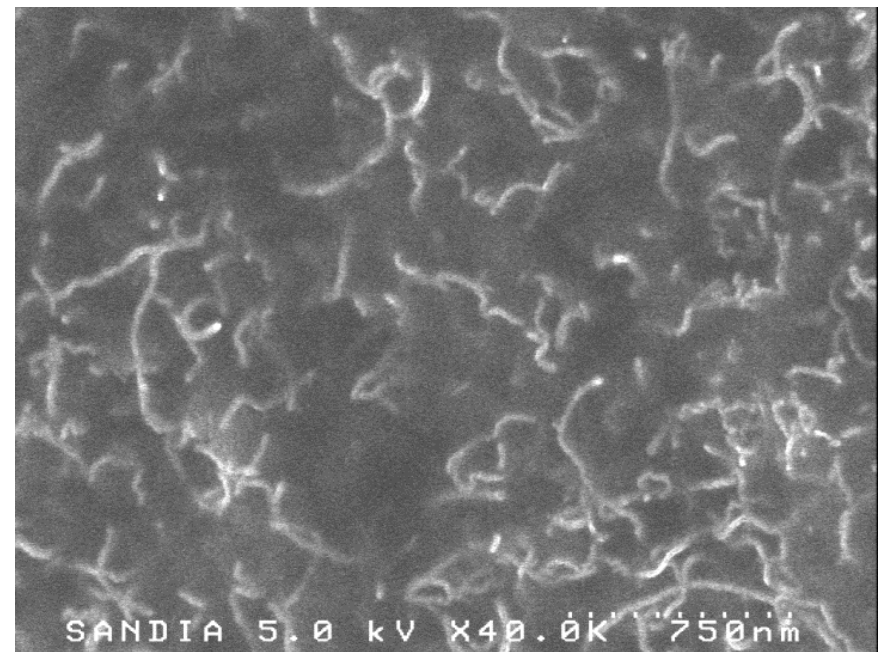
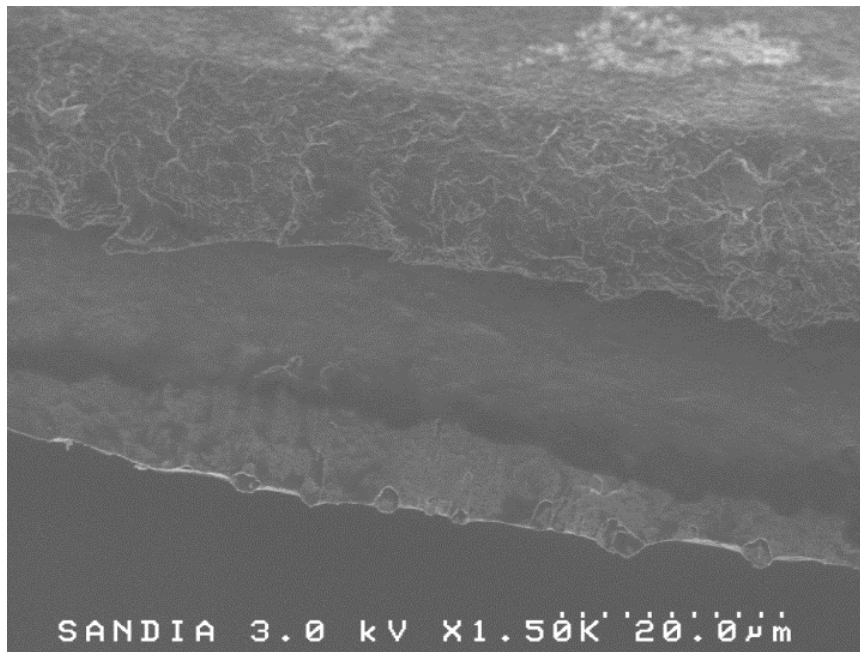
sonicate with
nanotube ink
→
(PSS wrapped MWCNT)



Forms segregated
MWCNT network

MWNT-Latex Morphology

- Creation of MWNT networks:
 - Electrical percolation above 1 wt% MWNTs
- Fiber-reinforced polymer deployment:
 - Surface applied to post-cured composites
 - Applied to fiber weaves for embedded sensing



Cross-section and MWNT network SEM images of 3wt% MWNT-Latex film

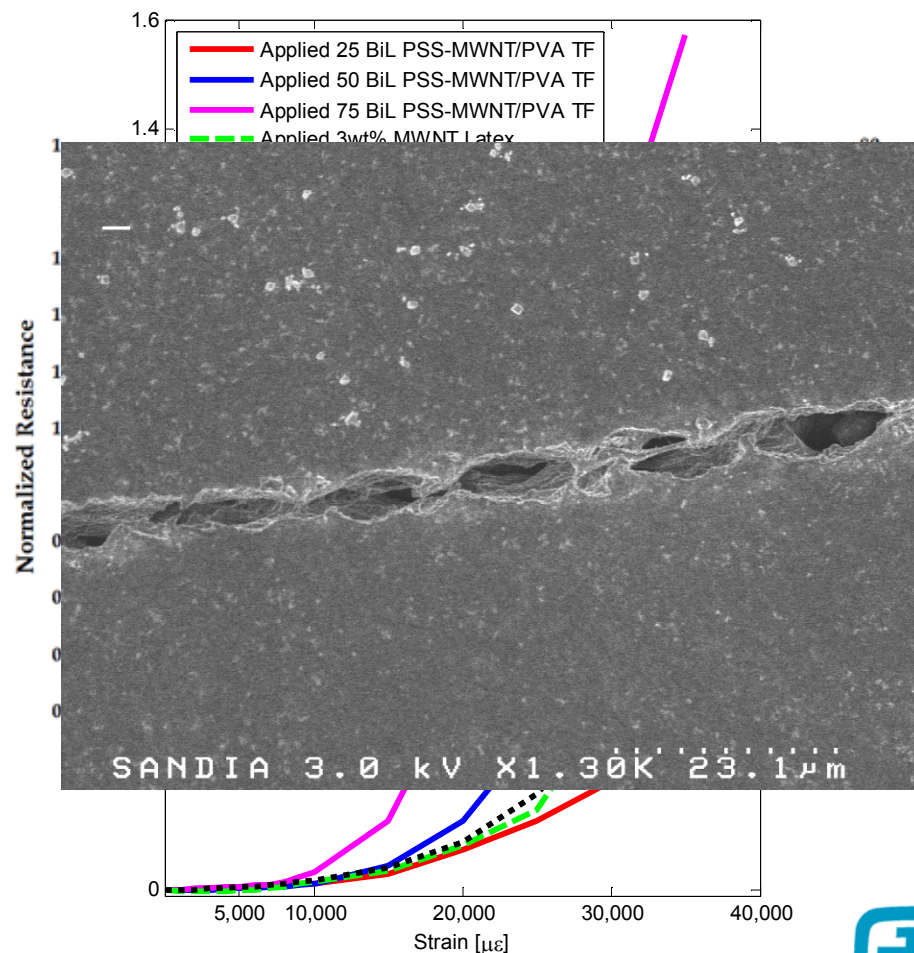
MWNT-Latex Characterization

- Electromechanical characteristics:

- Quasi-static testing
 - Nearly same sensitivity as LbL
- Bi-functional strain response
 - Linear
 - Quadratic
 - Cracking of film

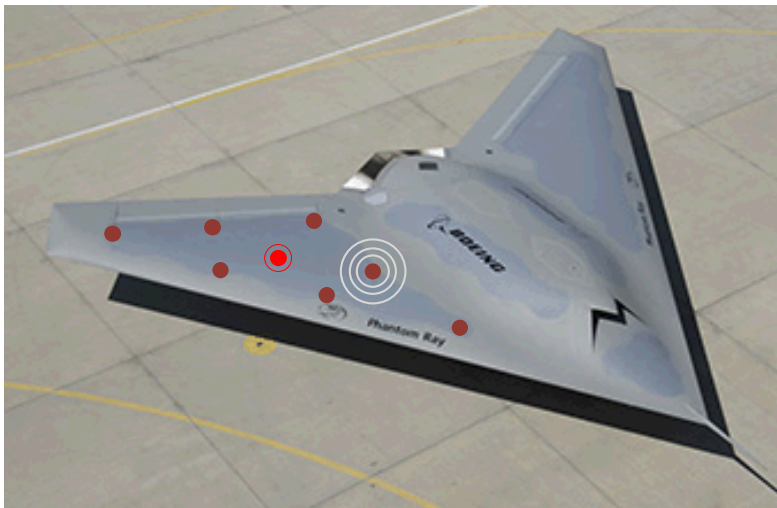
- Thermo-resistance coupling:

- -50°C to 80°C over 2 hours
- 2 hour holds
- Inversely linear relationship
- Non-linear response @ -30°C
 - $\sim T_g$ of PVDF
 - Restructuring of MWNTs



Spatially Distributed SHM Paradigm

- Current state-of-art in structural health monitoring:
 - Passive SHM using acoustic emissions
 - Active SHM using piezoelectric sensor/actuator pairs
- “Sensing skins” for spatial damage detection:
 - Objective is to identify the location and severity of damage
 - Monitor and detect damage over two- (or even three) dimensions
 - Direct damage detection



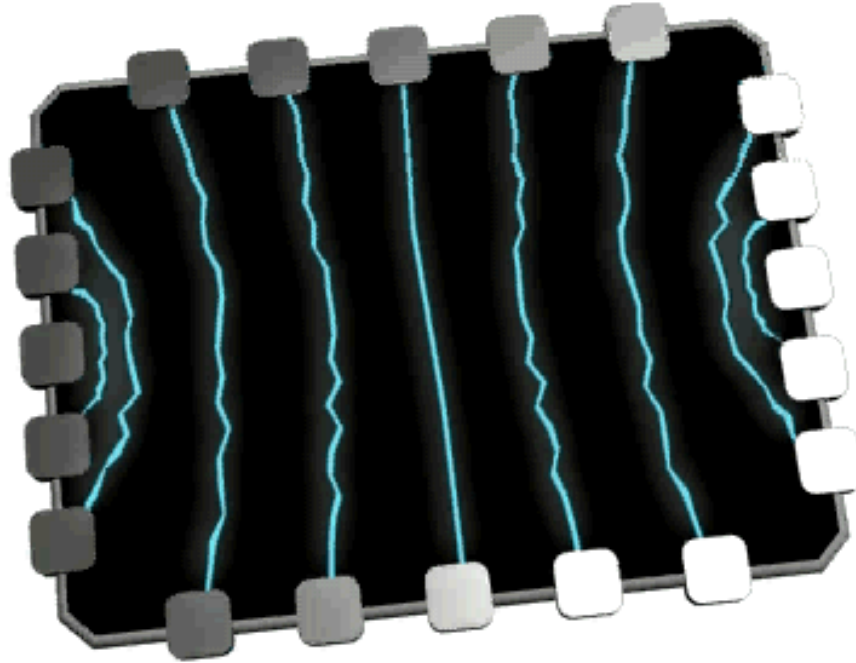
(Boeing)



(Boeing)

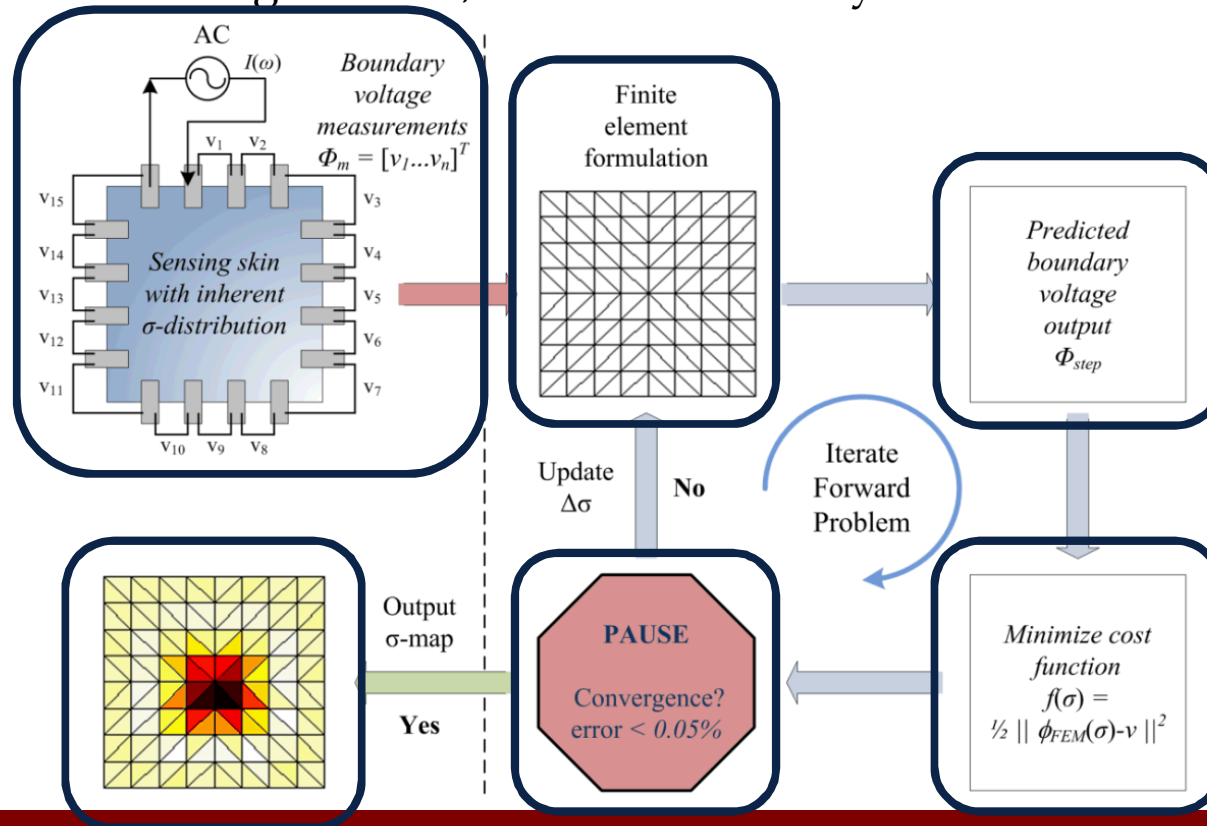
Electrical Impedance Tomography

- Overview of spatial conductivity mapping
 - Since film impedance calibrated to strain, conductivity maps can correspond to 2-D strain distribution maps



Typical EIT Reconstruction

- Laplace's equation:
 - $\nabla \cdot (\sigma \nabla \phi) = 0$, where σ can vary by orders of magnitude
 - Governs potential and conductivity relationship
- Forward problem: conductivity known, solve voltage
- Inverse problem: voltage known, solve conductivity



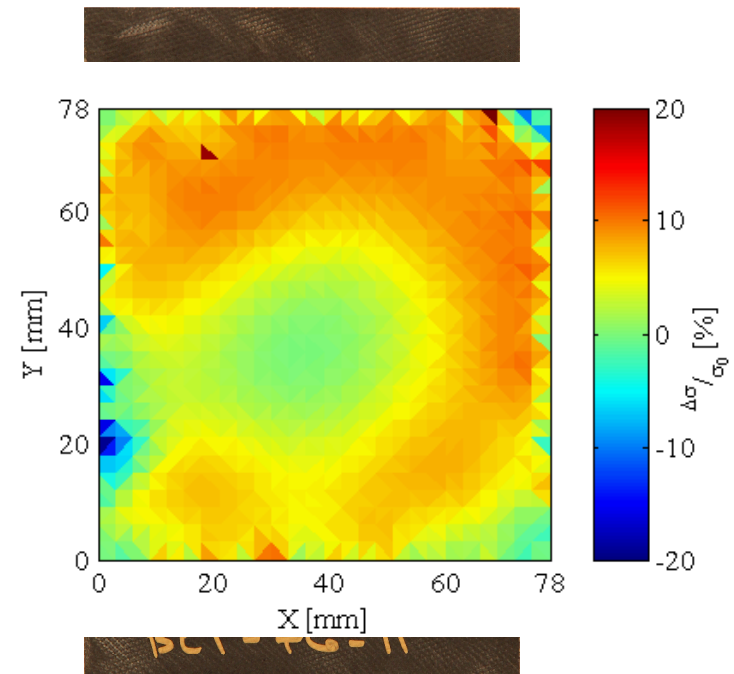
- Reconstructs small σ changes:
 - Typically difference imaging
 - $\sigma_1 - \sigma_2 \ll \sigma_2$
- Maximum a posteriori (MAP):
 - H: sensitivity matrix

$$H(\sigma_{bkgd})_{ij} = \frac{\partial V_i}{\partial \sigma_j}$$
 - Regularization hyperparameter: λ
 - Noise figure

$$NF(\lambda) = \frac{SNR_{in}}{SNR_{out}} \approx 1$$
 - Use representative σ distribution
 - W: Noise model
 - R: Regularization matrix
- Advantages:
 - Can pre-calculate H
 - Many damage modes lead to small changes in σ

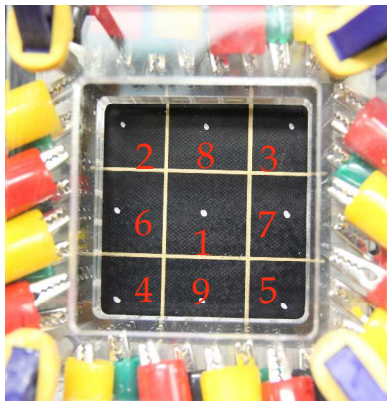
$$\frac{\Delta \sigma}{\sigma_0} = \left(\underline{H}^T \underline{W} \underline{H} + \underline{\lambda} \underline{R} \right)^{-1} \left(\underline{H}^T \underline{W} \right) \frac{\Delta V}{V_0}$$

$$\frac{\Delta \sigma}{\sigma_0} = B \Delta \frac{\Delta V}{V_0}$$



Spatially Distributed Sensitivity

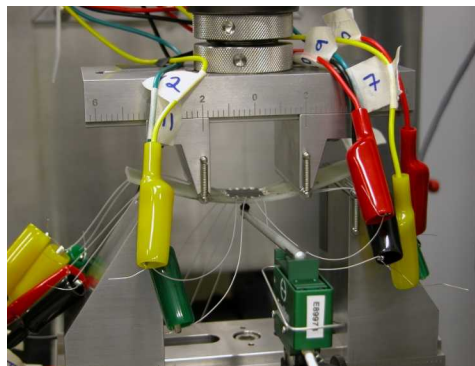
- Understand sensitivity to prescribed damage w.r.t. spatial position in sensing area
- 9 holes distributed across specimen
- 6.35 mm diameter



Spatially Distributed Sensitivity Specimen

Spatially Distributed Strain Sensitivity

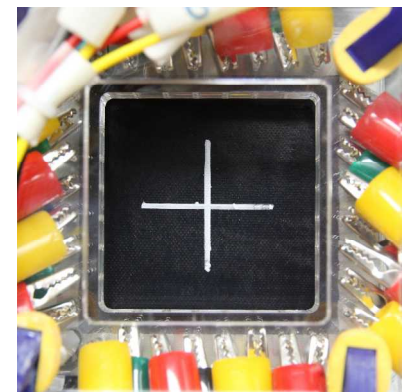
- 4-pt bending specimens
 - Homogeneous strain between inner supports
- Tensile and compressive strain values
- Small changes in conductivity



Spatially Distributed Strain Sensitivity Specimen

Damage Size Sensitivity

- Understand sensitivity to increasing damage at center of specimen
- Least sensitivity point in sensing region
- 6 progressively larger holes
 - $\frac{1}{16}''$, $\frac{1}{8}''$, $\frac{3}{16}''$, $\frac{1}{4}''$, $\frac{5}{16}''$, $\frac{3}{8}''$

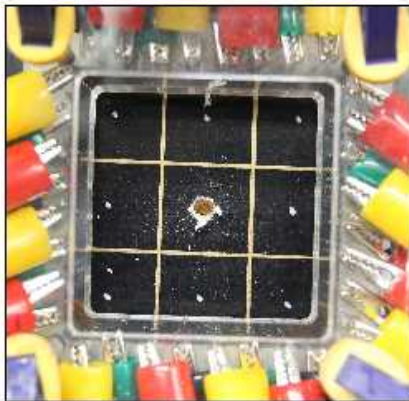


Damage Size Sensitivity Specimen

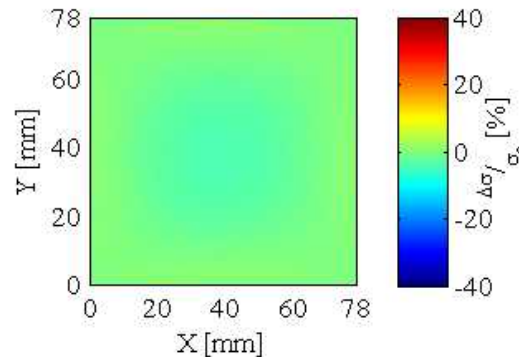
Spatially Distributed Sensitivity

- EIT Response

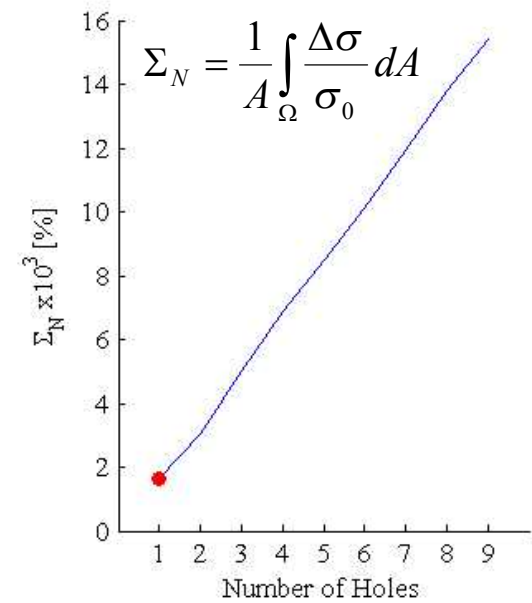
- Consistent cumulative amplitude response
 - Linear response to increasing sustained damage
- Further from center, response more disperse
 - Mean response at correct damage location



Specimen



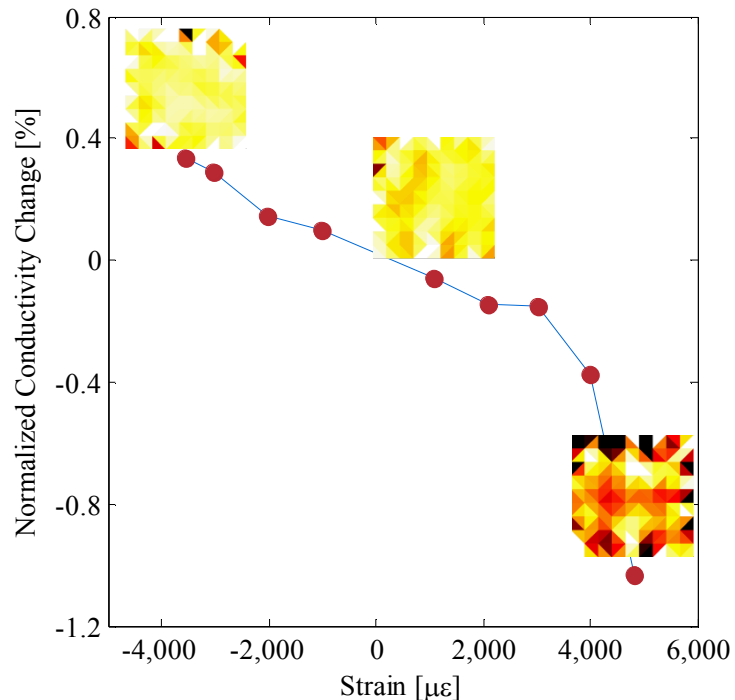
EIT Response



Damage Metric

Spatial Strain Sensing

- 4-pt bending
 - ASTM D7264
 - MWNT-Latex on GFRP
 - Stepped displacement profile
 - Tensile/compressive strain
- Strain sensitivity
 - Nearly linear

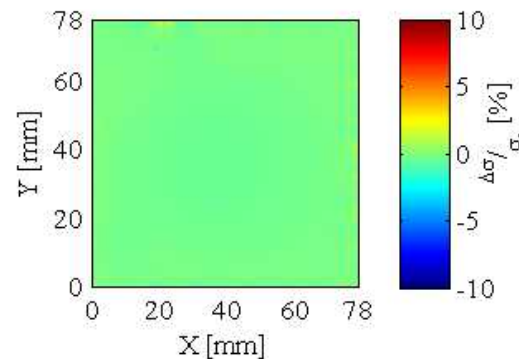


Damage Size Sensitivity

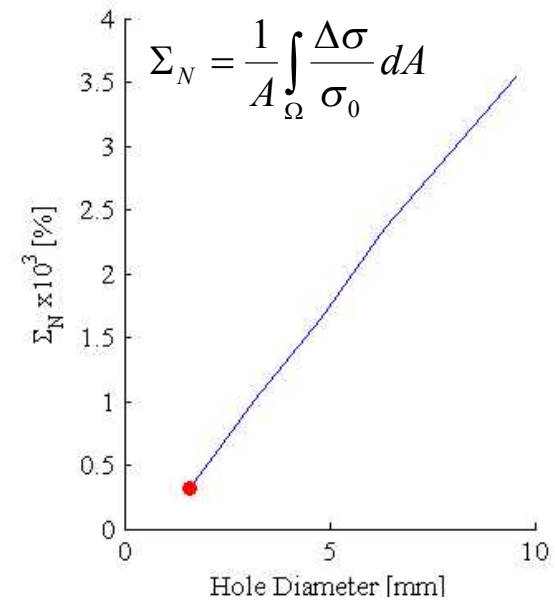
- EIT Response
 - Increasing EIT response to increasing damage size
 - Nearly linear response to size
 - EIT response at corresponding location to damage but response size is exaggerated



Specimen



EIT Response



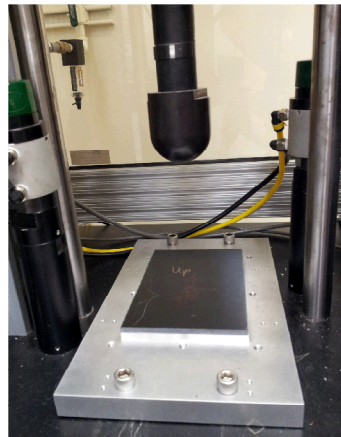
Damage Metric

Impact Damage Detection

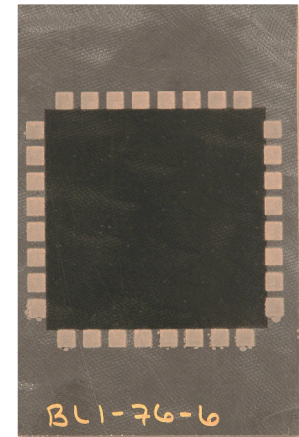
- Drop-weight impact tests
 - ASTM D7146
 - 78 mm by 78 mm sensing region
 - MWNT-latex on glass fiber weave
 - Impact energy: 20, 60, 100, 140 J
 - Before/after EIT measurements
- Verification:
 - Photographic Imaging
 - Surface damage



Drop-weight impact tester



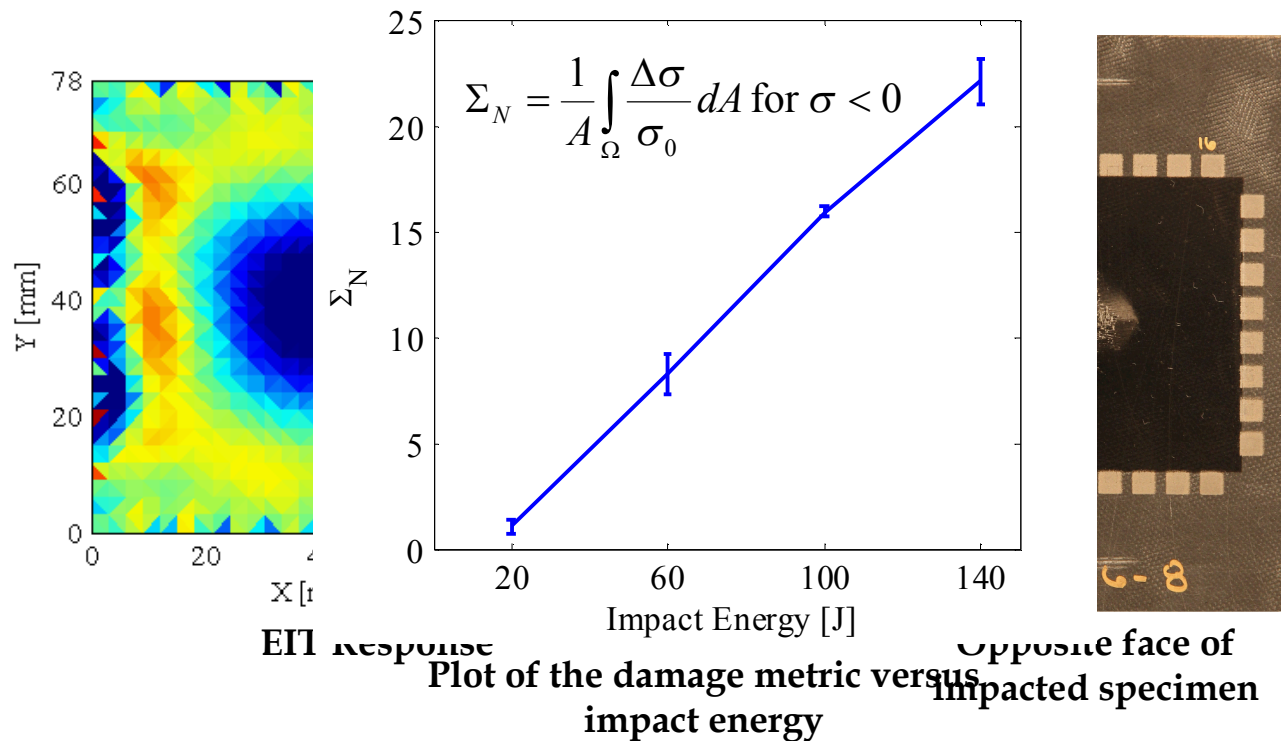
Impact setup



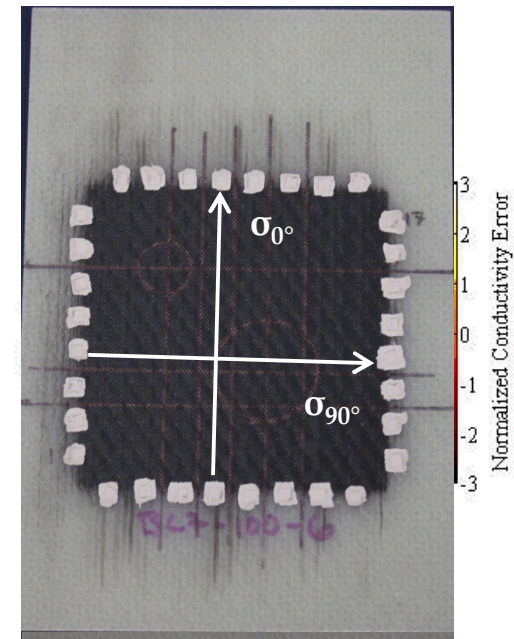
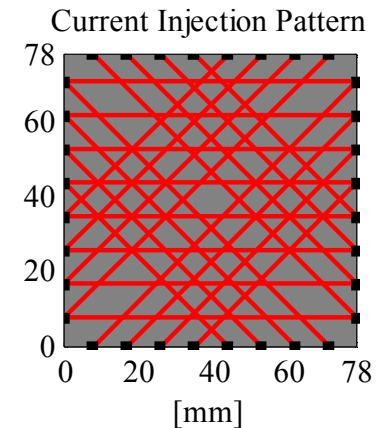
Impact specimen

Impact Damage Detection

- EIT reconstruction captures conductivity decrease in damaged region
 - Decreasing amplitude and increasing response region with increase in impact energy
- Linear response w.r.t. damage metric with good repeatability

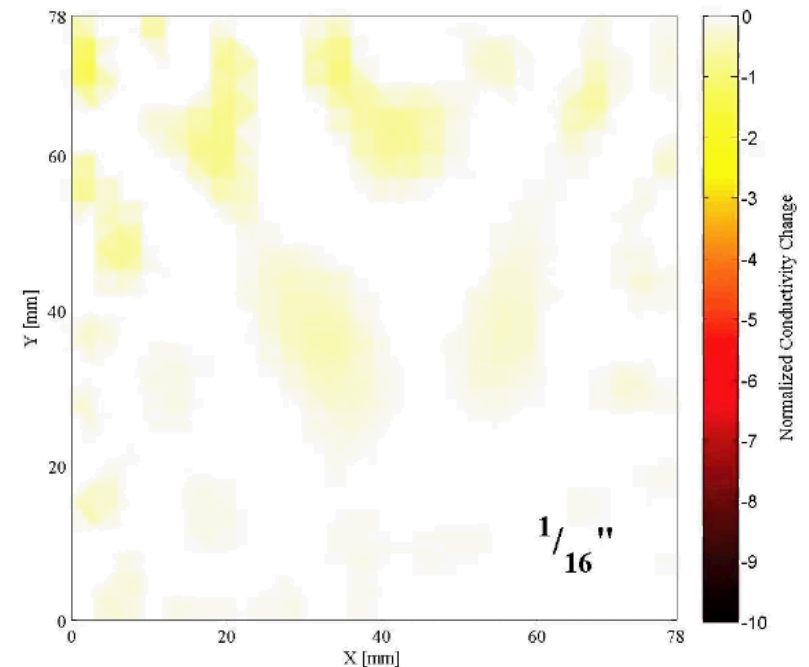
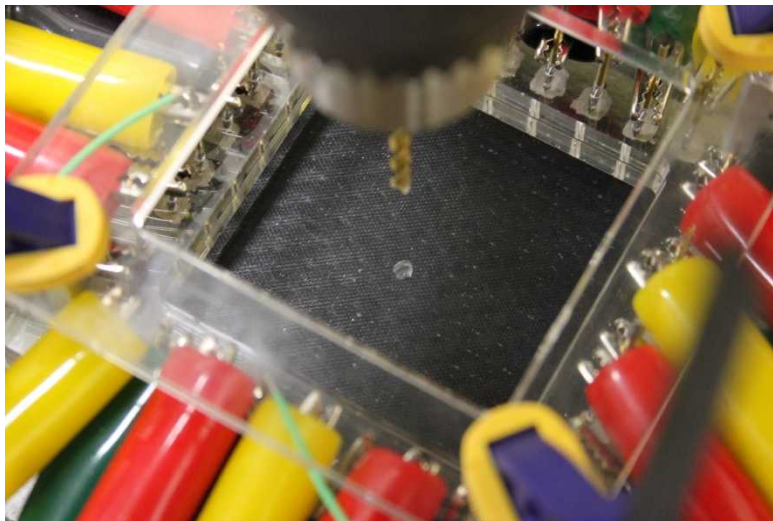


- Embedded sensing architecture
 - MWNT-Latex on GF fiber weave
 - Embedded within epoxy matrix
- Specimens
 - $[0^\circ/+45^\circ/90^\circ/-45^\circ]_{2s}$
 - Unidirectional GF
 - 150 mm x 100 mm
 - ASTM D7146 Standard
- Anisotropic EIT
 - Isotropic ► Anisotropic
 - Scalar ► Matrix: σ
 - $\sigma_{0^\circ} > \sigma_{90^\circ}$ by ~2:1
 - $\nabla \cdot (\sigma \nabla \phi) = 0$



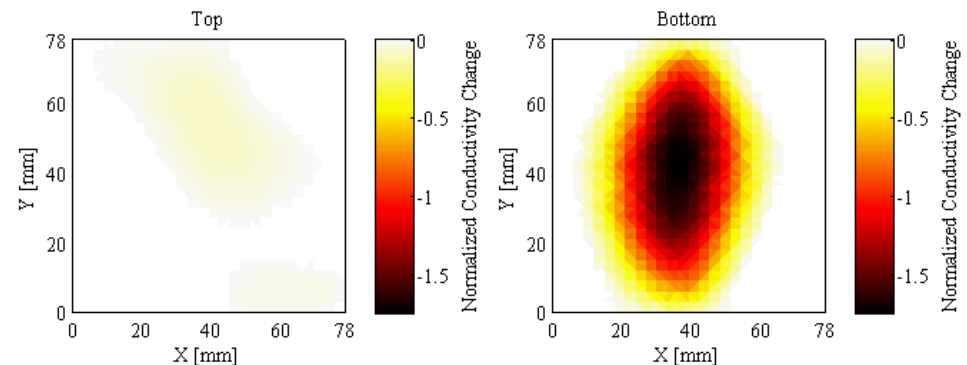
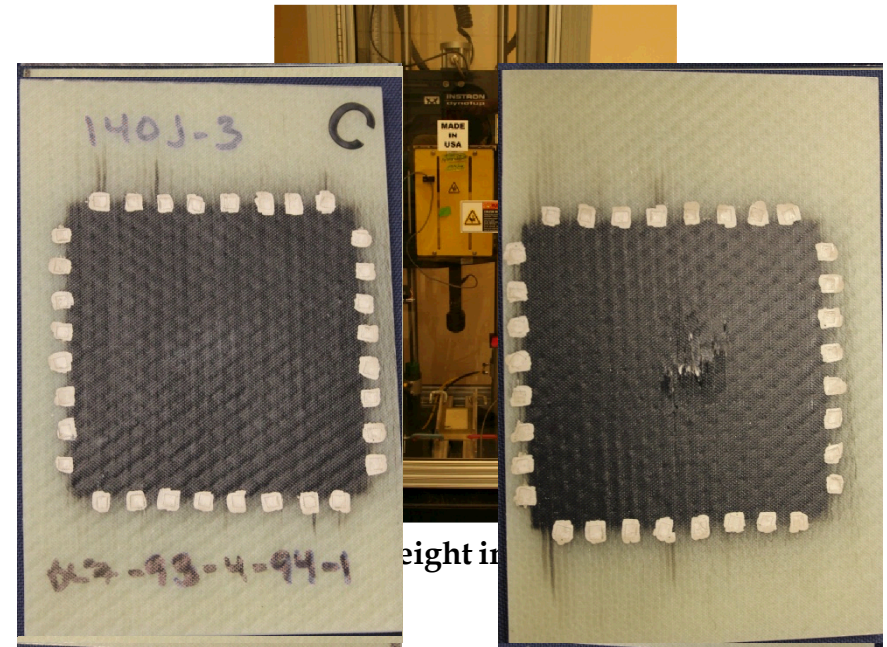
Embedded Spatial Sensitivity

- Embedded sensing validation:
 - Determine conductivity change sensitivity
 - Process:
 - Progressively larger drilled holes:
 - $1/16''$, $1/8''$, $3/16''$, $1/4''$, $5/16''$, $3/8''$, $1/2''$
 - Anisotropic EIT performed
 - Conductivity change from pristine sample



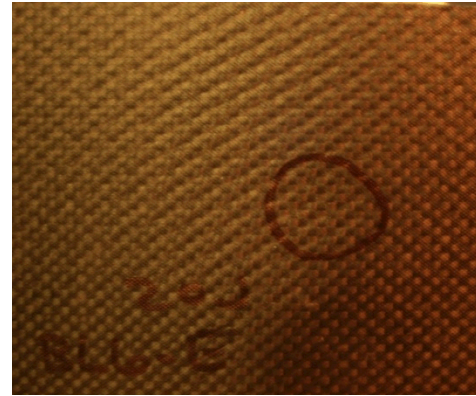
Impact Damage Detection

- Drop-weight impact tests
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 - 78 mm by 78 mm sensing region
 - MWNT-latex on glass fiber weave
 - Impact energy: 20, 60, 100, 140 J
 - Before/after EIT measurements
- Verification:
 - Thermography
 - Matrix Cracking
 - Delamination
 - Photographic Imaging
 - Surface damage

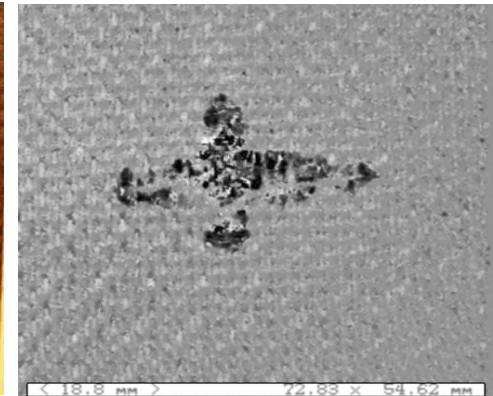


CFRP Damage Detection

- Carbon Fiber-Reinforced Polymers
 - High-strength-to-weight ratio applications
 - ~50% weight of Boeing 787
 - Primary structural material in SpaceShipTwo

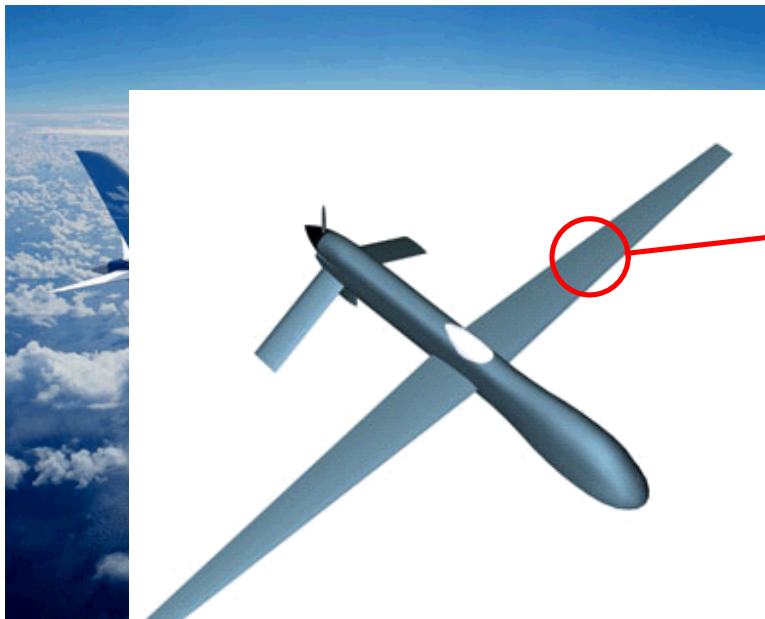


Visual inspection

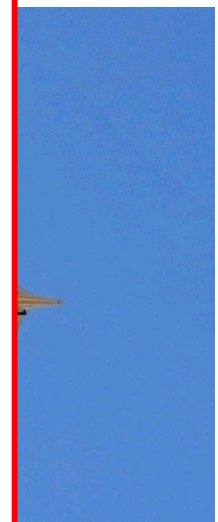


C-SCAN ultrasound image

CFRP panel after 20 Joule impact



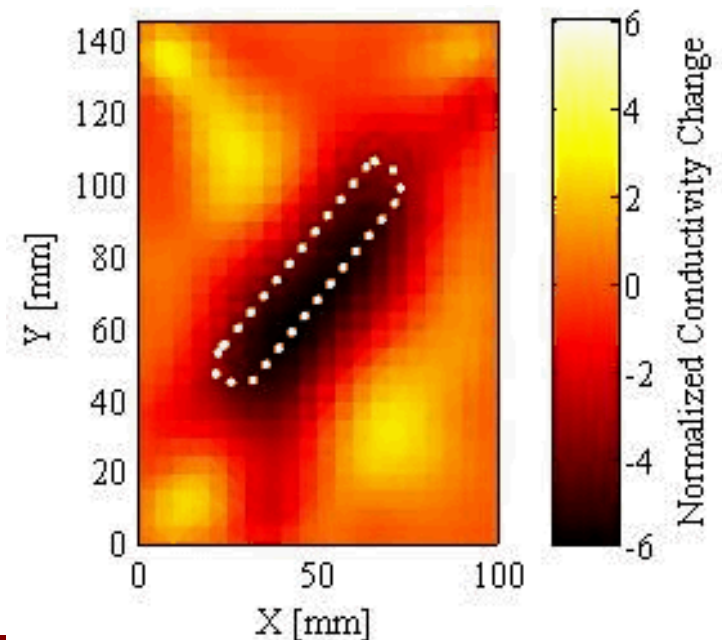
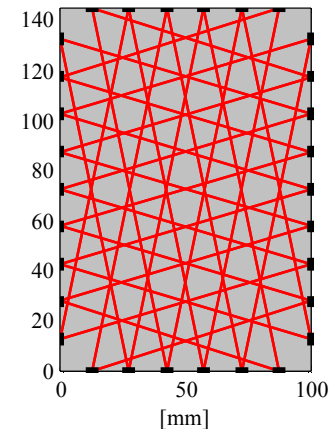
Boeing 787 (Boeing)



(actic)

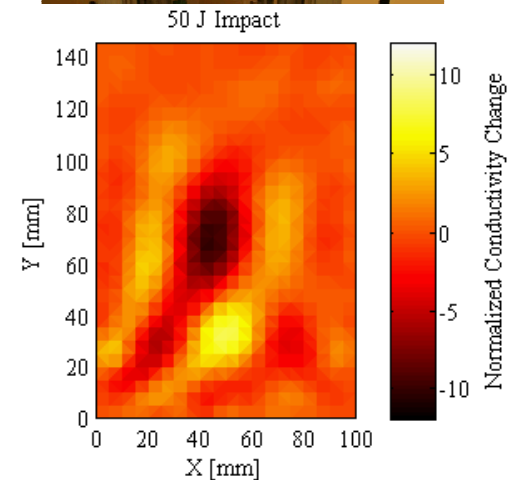
- Applied sensing measurements
 - $[\pm 45, 0/90]_{8s}$ CFRP composite
 - 100 mm x 145 mm sensing region
 - 6×9 electrodes scheme = 30 electrodes
 - 5 mm electrodes
 - 10 mm spacing
 - Anisotropic material injection scheme
- Investigate stability and efficiency:
 - Computational demand
 - ~ 1 s reconstruction time
- Accuracy characterization:
 - Conductivity:
 - Known removal of material
 - Spatial feature ID sensing resolution
 - 6.3 mm diagonal line

Current Injection Pattern



CFRP Impact Damage Detection

- Drop-weight impact tests
 - ASTM D7146
 - $[\pm 45, 0/90]_{8s}$ CFRP composite
 - 100 mm \times 145 mm specimens
 - Impact energy: 20, 35, 50 J
 - Before/after EIT measurements



- Propose a next-generation SHM system
 - Direct in situ damage detection
 - Monitor location and severity of damage
- Embedding multi-modal sensing capabilities
 - Development of MWNT-nanocomposites for SHM
 - Characterized electromechanical response to monotonic
 - Response to temperature swings
- Outline validation of EIT for damage detection
 - Applied GFRP, embedded GFRP, and CFRP specimens
 - Strain sensitivity
 - Damage sensitivity
 - Impact damage

Thank You!

Questions?

Acknowledgements:



*Exceptional
service
in the
national
interest*



