

Automated Correction of Heliostat Tracking Errors Caused by Fabrication and Installation Errors

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Overview

- **Introduction**
- **Tracking Error Sources**
- **Error Correction Theory**
- **Automated Error Correction**
- **Summary and Ongoing Work**

Introduction

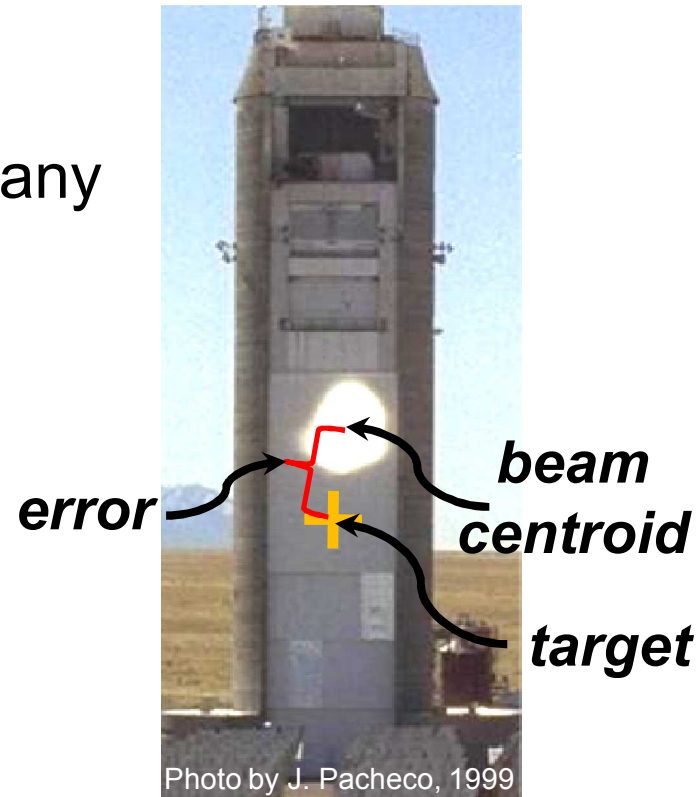
- For large heliostat fields, cost and time constraints increase likelihood of heliostat fabrication and installation errors



Photo by J. Pacheco, 1999

Introduction

- **Problem: Tracking Errors**
 - Unfeasible to identify and fix so many fabrication and installation errors
- **Solution: Adjust tracking algorithm to compensate for errors**



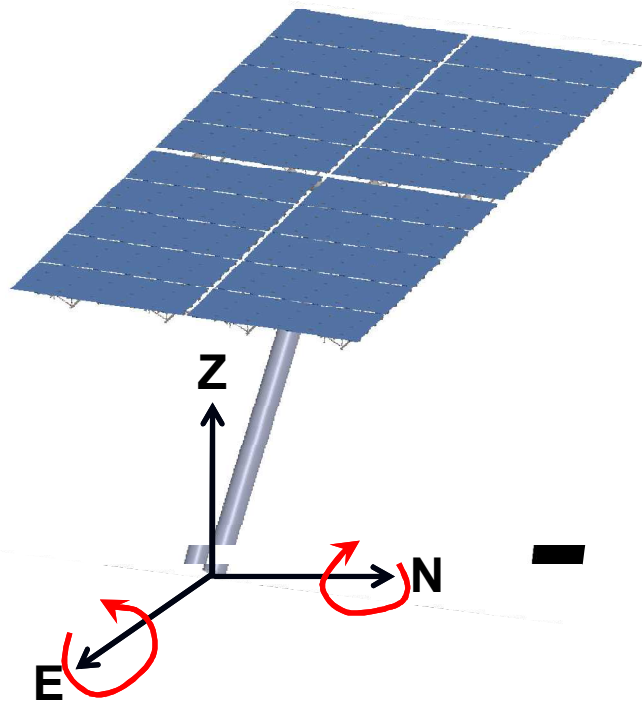


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Tracking Error Sources

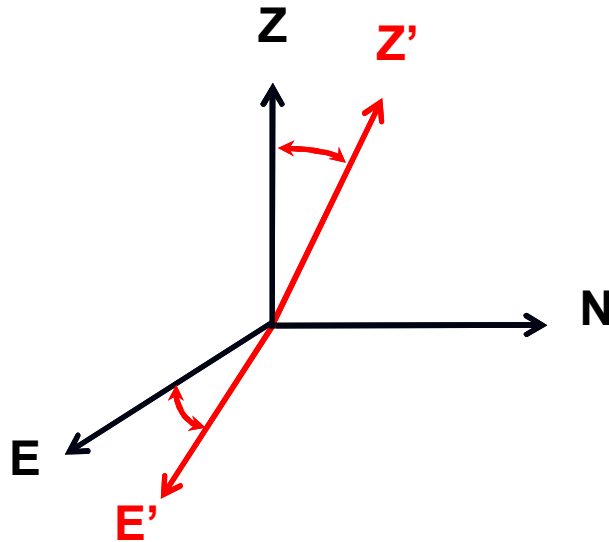
1. Pedestal Tilt



- Actual mirror coordinate system is rotated away from ideal coordinate system

Tracking Error Sources

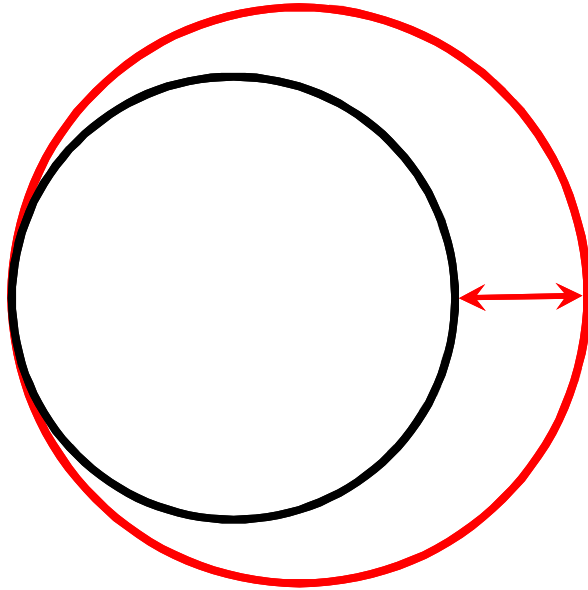
2. Reference Bias



- True Zenith and/or true East are offset from assumed Zenith and East references

Tracking Error Sources

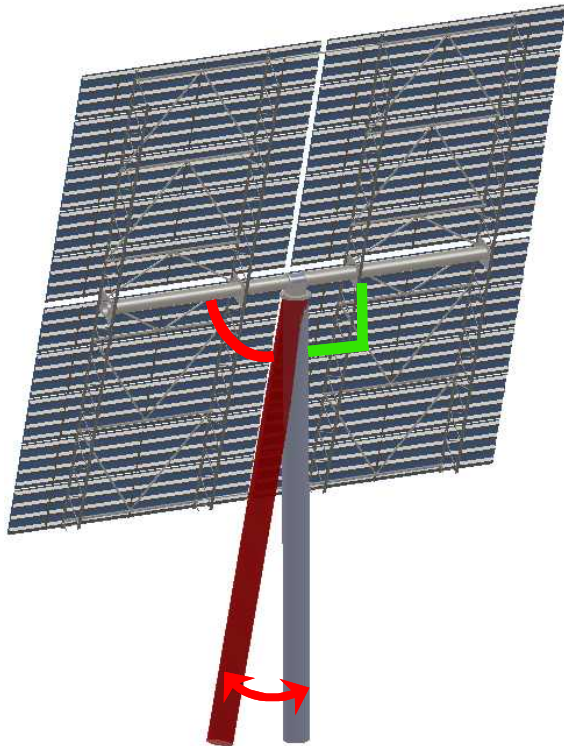
3. Drive Wheel Diameter Error



- Inaccuracy in conversion factor from angle to number of drive axis turns

Tracking Error Sources

4. Drive Axis Non-Orthogonality



- Non-square attachment between azimuthal and elevation drive axes
- An elevation-axis rotation causes an azimuthal error



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Error Correction Theory

Open-Loop Error Correction

1. Develop physical model for errors
2. Measure errors
3. Calculate error parameters
4. Adjust tracking algorithm

Error Correction Theory

Open-Loop Error Correction

1. Develop physical model for errors

a_1	←	Pedestal tilt about azimuth = 90° axis
a_2	←	Azimuth reference bias
a_3	←	Pedestal tilt about azimuth = 0 axis
a_4	←	Fractional error in azimuth wheel radius
a_5	←	Fractional error in elevation wheel radius
a_6	←	Elevation reference bias
a_7	←	Axis non-orthogonality error

$$\begin{pmatrix} \text{Azimuthal Error, } \Delta\theta \\ \text{Elevation Error, } \Delta\alpha \end{pmatrix} \approx \begin{pmatrix} \frac{\sin \theta}{\tan \alpha} & -1 & \frac{\cos \theta}{\tan \alpha} & \theta & 0 & 0 & \frac{1}{\tan \alpha} \\ -\cos \theta & 0 & \sin \theta & 0 & \alpha & -1 & 0 \end{pmatrix} \begin{pmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \\ a_5 \\ a_6 \\ a_7 \end{pmatrix}$$

Notes:

- a_1 to a_7 are assumed to be small
- θ is azimuthal angle, α is elevation angle ($\alpha=0$ at zenith)
- Method for a_1 to a_6 originally described by R. Baheti and P. Scott (1980); a_7 by C. Andraka (2000)

Error Correction Theory

Open-Loop Error Correction

2.Measure errors

θ	α	θ'	α'
Intended Azimuth (deg)	Intended Elevation (deg)	Measured Azimuth (deg)	Measured Elevation (deg)
71.01	66.32	72.69	68.17
70.84	66.12	72.51	67.96
70.66	65.93	72.33	67.76

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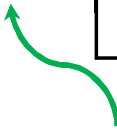
$$\begin{pmatrix} \text{Azimuthal Error, } \Delta\theta_i \\ \text{Elevation Error, } \Delta\alpha_i \end{pmatrix} \equiv \begin{pmatrix} \theta_i' - \theta_i \\ \alpha_i' - \alpha_i \end{pmatrix}$$

Error Correction Theory

Open-Loop Error Correction

3. Calculate error parameters

Least-Squares Algorithm: $\mathbf{a} = \left[\sum_i \mathbf{H}_i^T \mathbf{H}_i \right]^{-1} \left[\sum_i \mathbf{H}_i^T \mathbf{q}_i \right]$

 **Error Parameter Vector**

Definitions:

$$\mathbf{a} \equiv \begin{pmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \\ a_5 \\ a_6 \\ a_7 \end{pmatrix}$$

$$\mathbf{H}_i \equiv \begin{pmatrix} \frac{\sin \theta_i}{\tan \alpha_i} & -1 & \frac{\cos \theta_i}{\tan \alpha_i} & \theta_i & 0 & 0 & \frac{1}{\tan \alpha_i} \\ -\cos \theta_i & 0 & \sin \theta_i & 0 & \alpha_i & -1 & 0 \end{pmatrix}$$

$$\mathbf{q}_i \equiv \begin{pmatrix} \text{Azimuthal Error, } \Delta\theta \\ \text{Elevation Error, } \Delta\alpha \end{pmatrix}$$



Error Correction Theory

Open-Loop Error Correction

4. Adjust tracking algorithm

Intended Pointing Angles: $\mathbf{p}_i \equiv \begin{pmatrix} \theta_i \\ \alpha_i \end{pmatrix}$

**Adjusted Pointing Angles
Supplied to Heliostat:** $\mathbf{p}'_i = \mathbf{p}_i + \mathbf{H}_i \mathbf{a}$



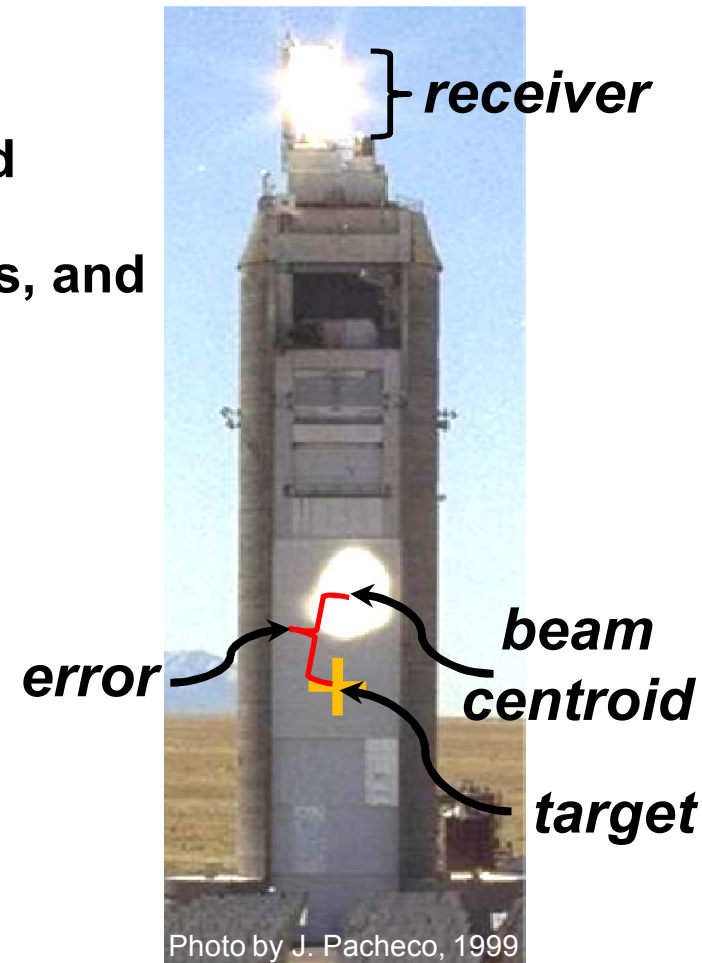
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Automated Error Correction

1. Order heliostat to point at target
2. Take photo and calculate beam centroid
3. Calculate azimuthal and elevation errors, and append to error vector
4. Calculate error parameters and adjust tracking algorithm
5. Point heliostat at receiver
6. Repeat steps 1 to 5 for next heliostat
7. After procedure is completed for all heliostats, repeat entire procedure

➤ Error vectors keep growing and error parameter estimates improve





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Summary and Ongoing Work

- **Helio­stat fab­ri­ca­tion and in­stal­la­tion er­rors are very likely for large fields**
- **Can com­pen­sa­te for er­rors using au­to­ma­ted track­ing al­go­rithm ad­just­ments**
- **Ongoing Work:**
 - Plot track­ing er­rors for a helio­stat be­fore and af­ter au­to­ma­ted er­ror cor­rec­tion