

Exceptional service in the national interest



Sandia Cooler Program Overview

Terry Johnson, DMTS

Presented to Whirlpool

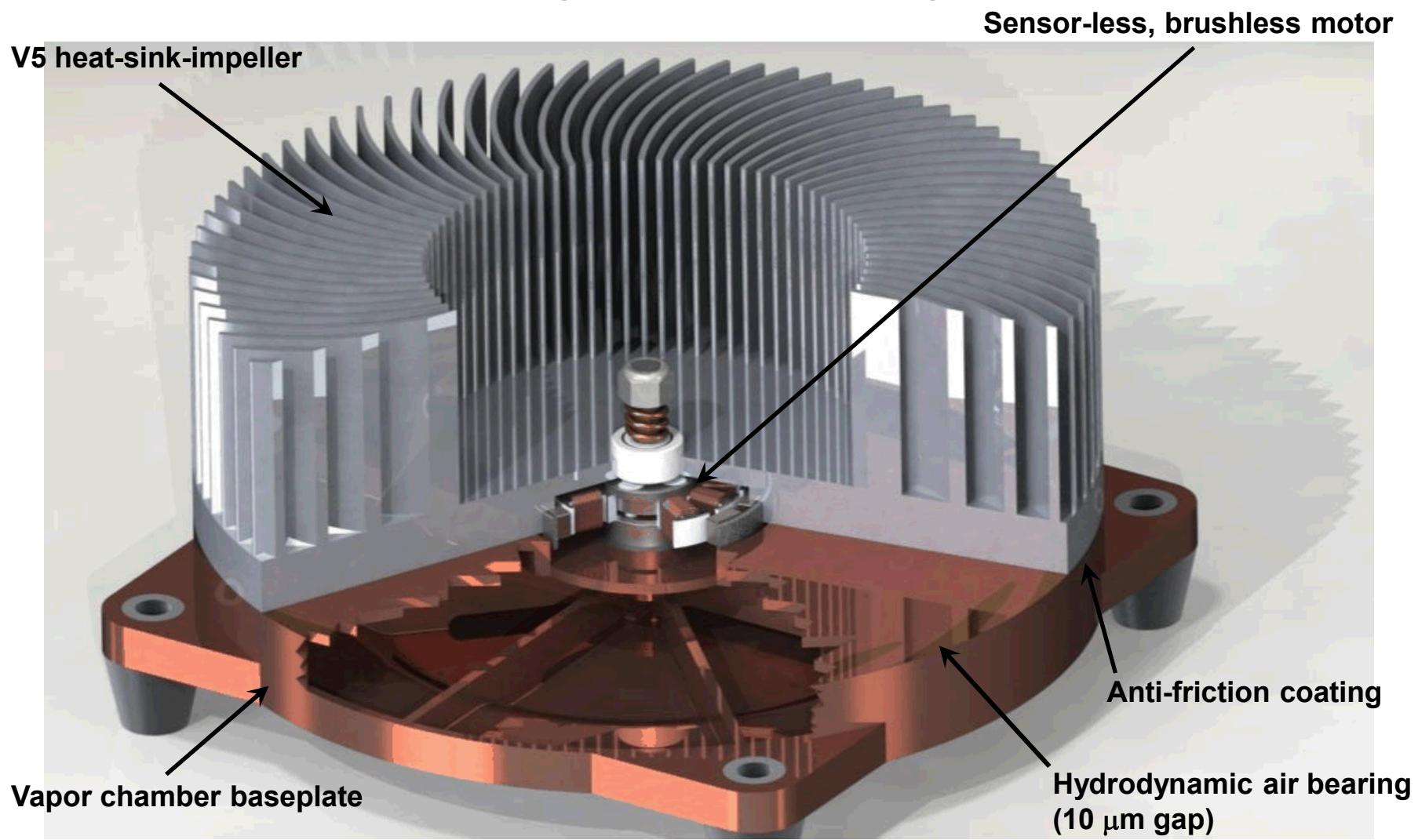
November 19, 2013



Sandia National Laboratories is a multi-program laboratory managed and operated by Sandia Corporation, a wholly owned subsidiary of Lockheed Martin Corporation, for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-AC04-94AL85000.



Latest CPU cooler design represents the culmination of several years of development



V5 objective: fully matured radial air bearing heat exchanger technology, tech transfer ready

V5 performance goals: $R = 0.1 \text{ C/W}$ at 3000 rpm, very low noise, 5 W power consumption

The team includes 14 scientists, engineers and technologists



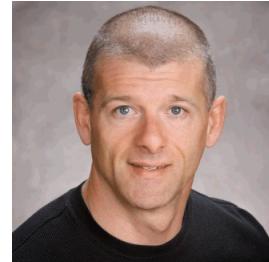
Mark Zimmerman
BSEE: Motor control development



Mike Leick
MSME: Motor control and anti-friction coatings



Jeff Koplow
PhD Chem: Inventor, technical advisor, axial flow R&D lead



Terry Johnson
MSME: Radial flow project lead and system engineer



Imane Khalil
PhD ME: Project Manager



Ryan Gorman
EE: Motor control development



Nathan Spencer
MSME: Structural dynamics



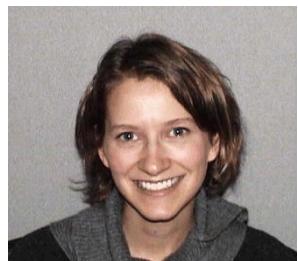
Justin Vanness
MSME: Motor control development



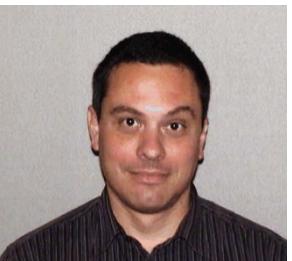
Wayne Staats
PhD ME: CFD for radial and axial flow impeller design



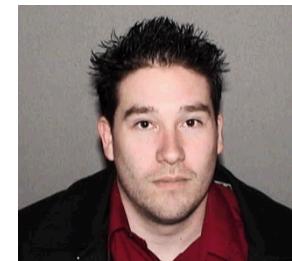
Kent Smith
Mech. Tech.: Fabrication and mechanical design



Patricia Gharagholoo
PhD ME: CFD/Heat Transfer



Marco Arienti
PhD ME: CFD/Heat Transfer



Daniel Matthew
BSME: Impeller fabrication and mechanical design



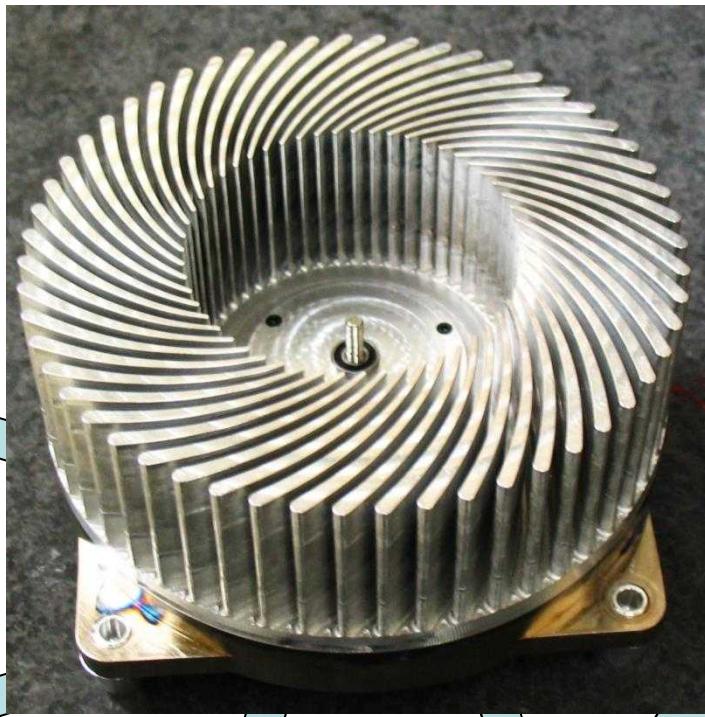
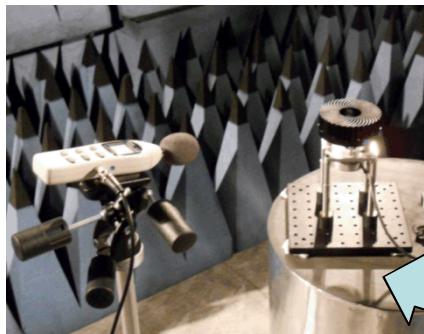
Ethan Hecht
PhD ChE: Impeller performance characterization



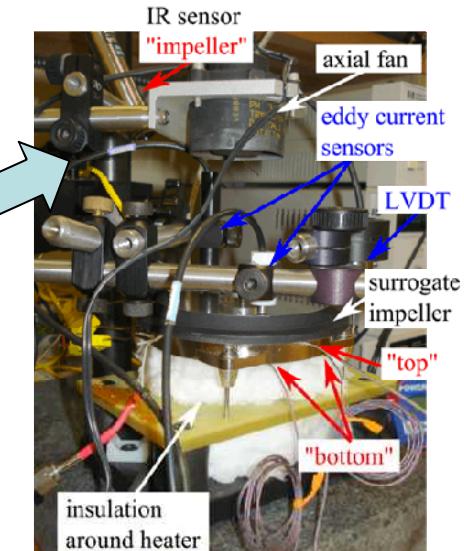
Arthur Kariya
PhD ME: Heat pipe design

Test stands have been developed to evaluate all aspects of the Sandia Cooler

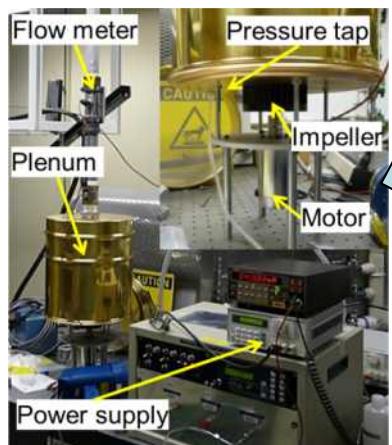
Acoustic



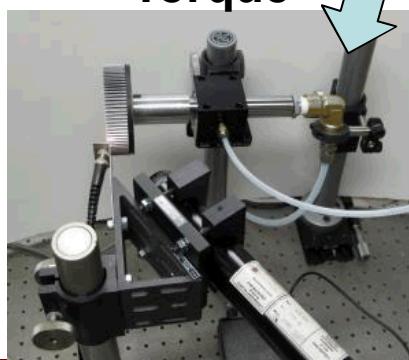
Thermal Resistance



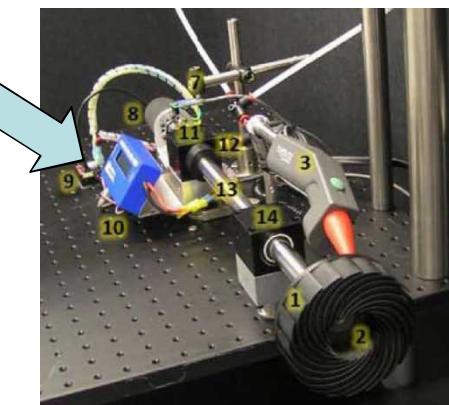
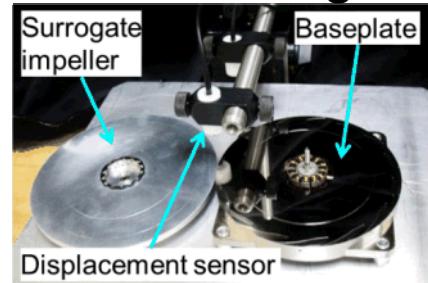
Pressure-Flow



Torque

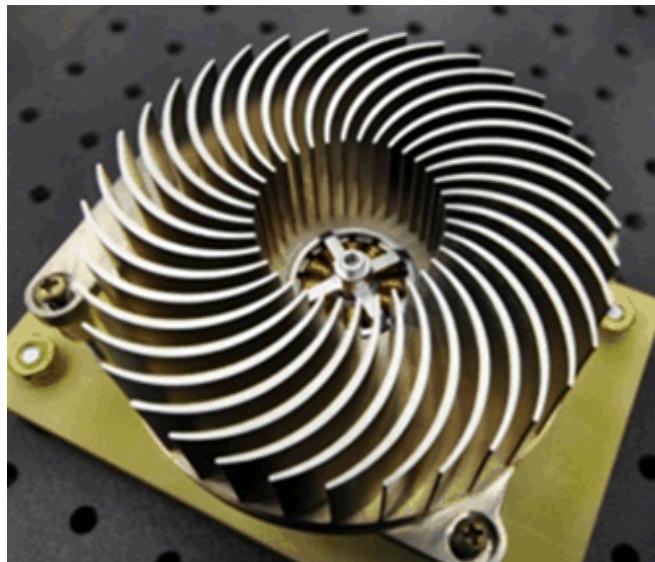


Air Bearing



Three different impeller geometries have been extensively characterized

V4



| | |
|------------|-------------------|
| OD | 4.0" |
| ID | 1.5" |
| Fin Height | 1.0" |
| # Fins | 36 |
| Shape | Intersecting arcs |

V5



| | |
|------------|-------|
| OD | 4.0" |
| ID | 2.0" |
| Fin Height | 0.95" |
| # Fins | 80 |
| Shape | Arcs |

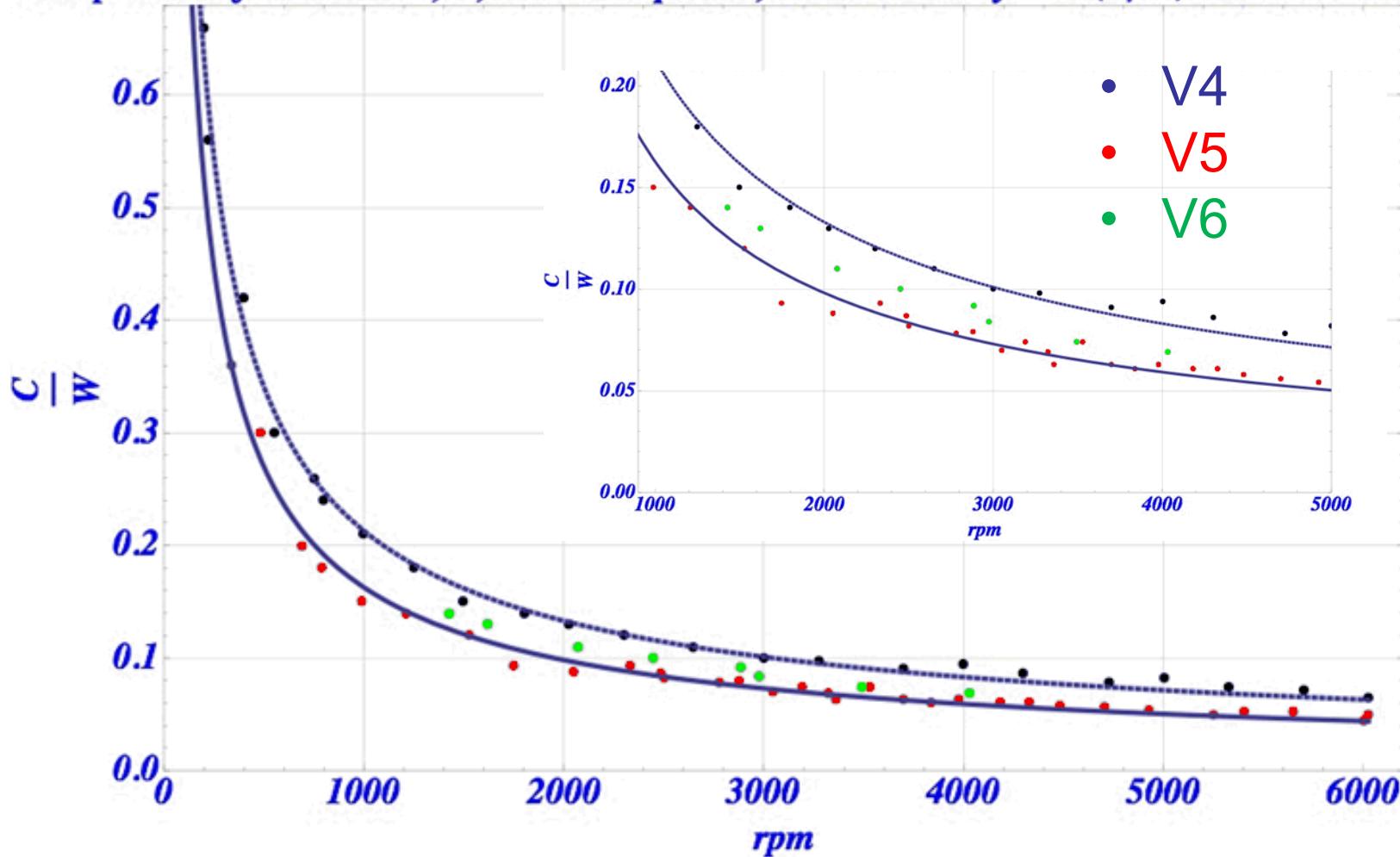
V6



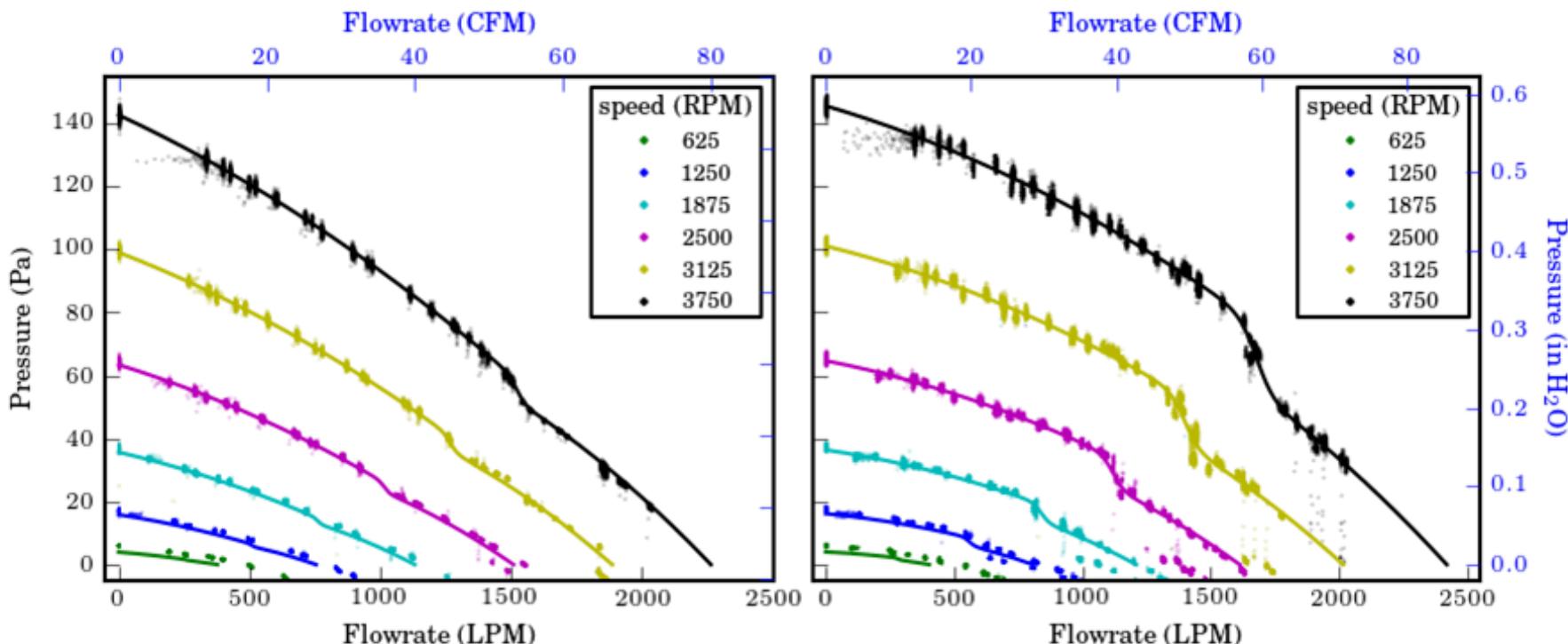
| | |
|------------|------------|
| OD | 4" |
| ID | 2.0" |
| Fin Height | 1.18" |
| # Fins | 55 |
| Shape | Log spiral |

V5 impeller has the lowest thermal resistance tested

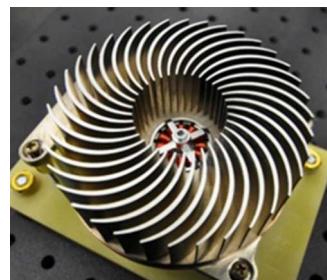
Comparison of Version IV, V, and VI Impellers, Thermal Decay – R (C/W) As a Function of Rpm



Pressure-flow curves were measured for several 4" impellers; V5 performed best



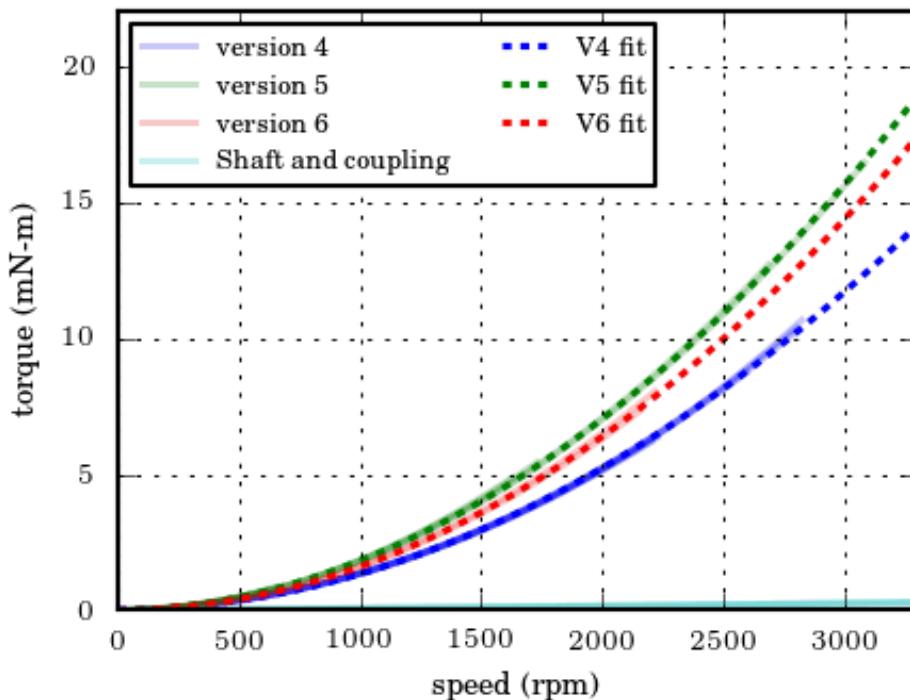
V4



V5



Impeller torque measured vs. speed; power consumption includes impeller and air gap torque



Impeller power:

$$P = \tau \times \omega$$

@2500 rpm V5 $P = 3W$

Air gap power:

$$\tau = \frac{\pi * \mu * \omega (r_o^4 - r_i^4)}{2 * h}$$

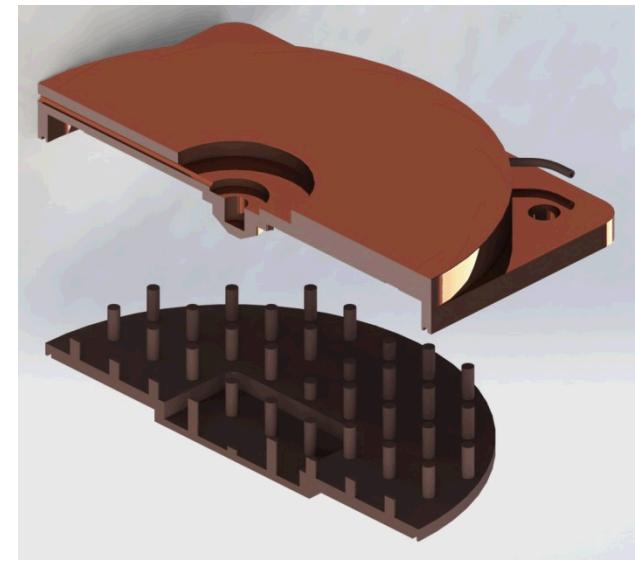
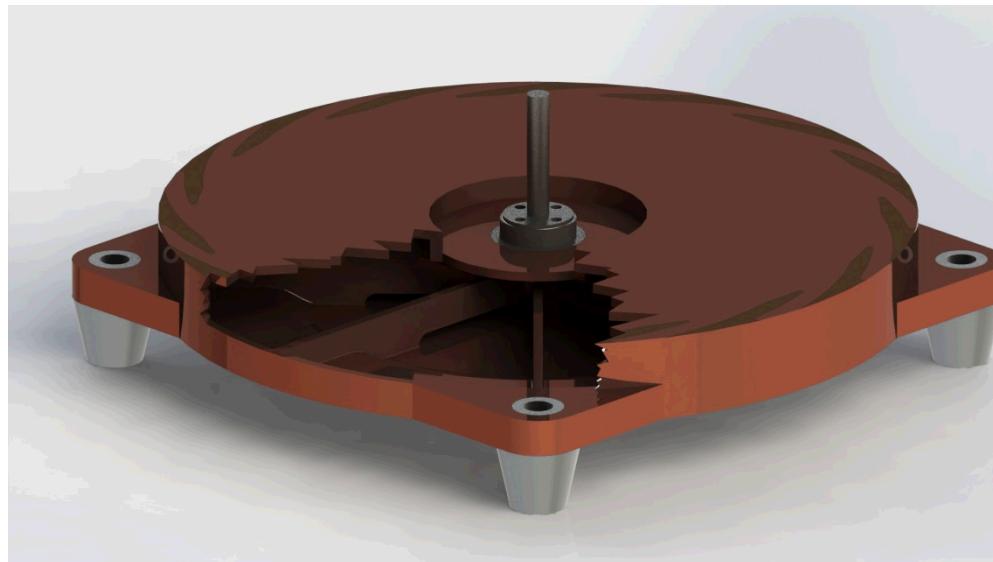
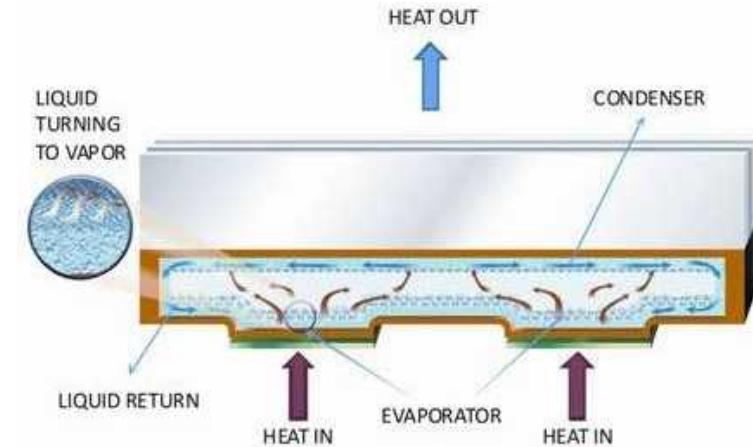
For a 10 micron air gap $P = 1.3W$

Total power:

$$P_{\text{mech}} = 3W + 1.3W = 4.3W$$

Baseplate: Vapor Chamber Incorporation

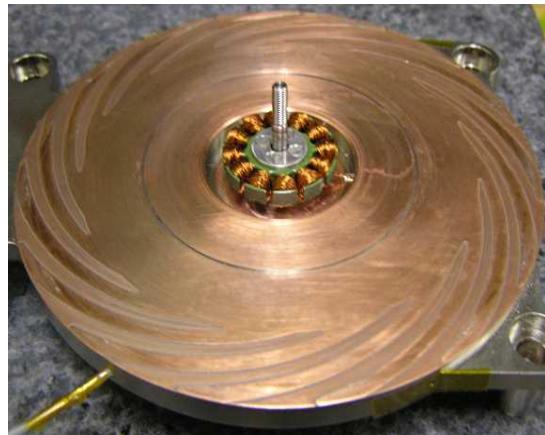
- Spreading resistance of solid baseplate was unacceptably high
- Vapor chamber solution from Thermacore



Air bearing design was improved through experiment and analysis

Original Design

Greater lift than needed
Significant pre-load for 10 μm gap
Groove area and depth larger than required



V5 Design

Good stiffness with less thermal resistance
Less sensitivity to impeller speed
Groove area still larger than required



Final Design

Maximum stiffness at a 10 μm gap
Minimal pre-load
Minimum thermal resistance

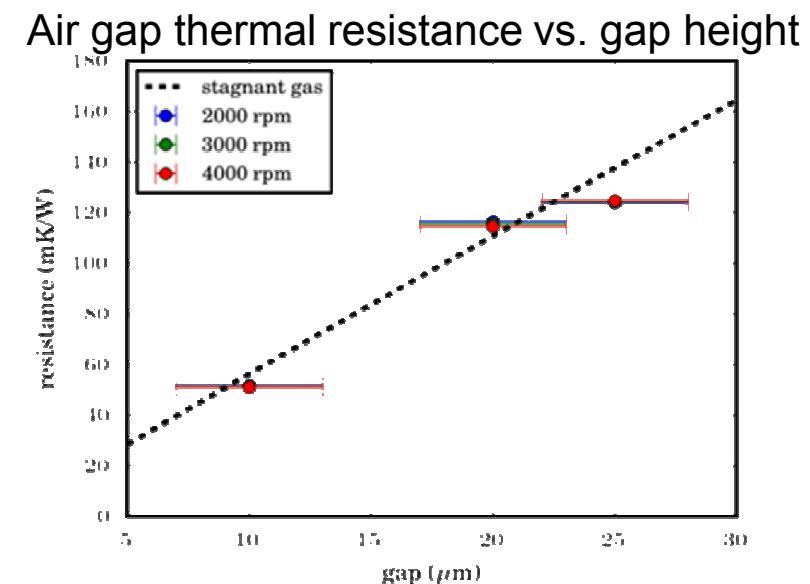
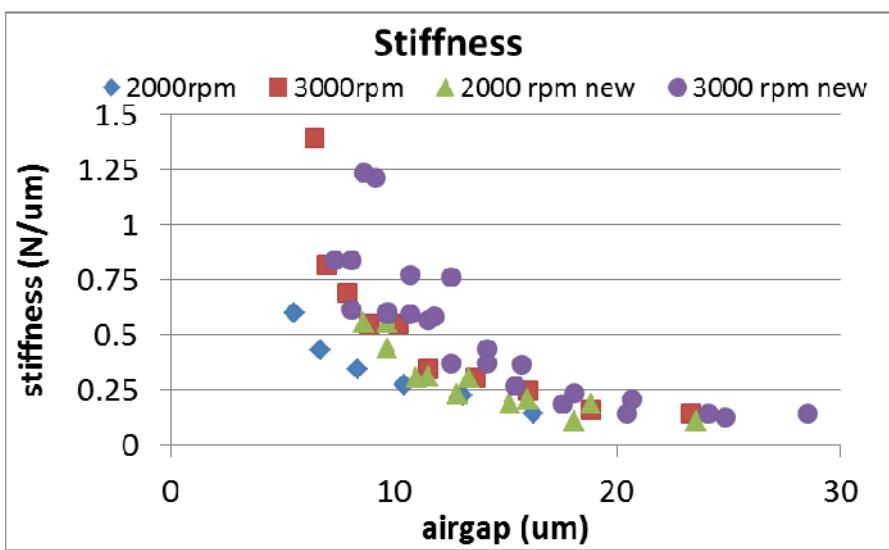
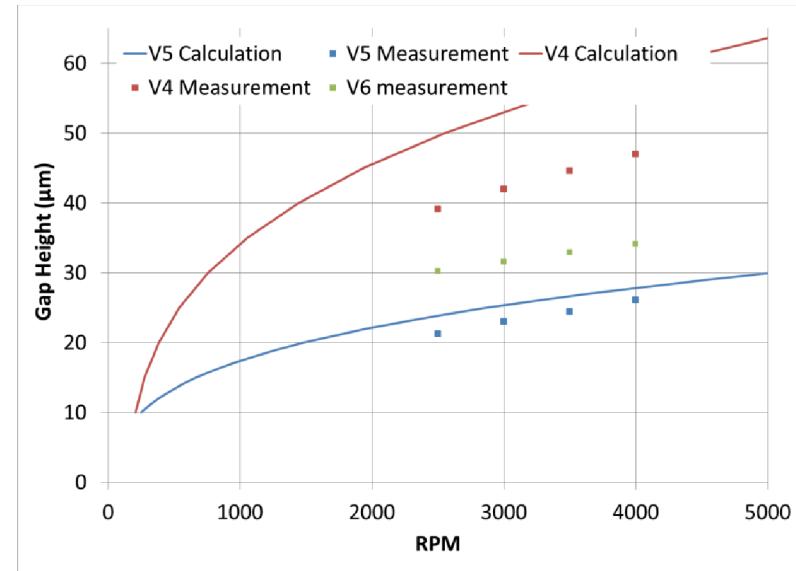
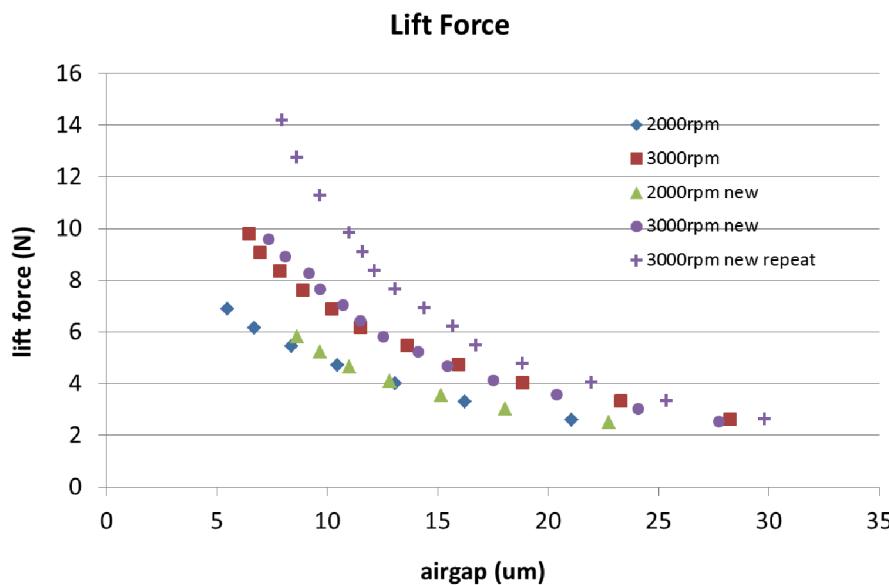


| Parameters | |
|---|------------------|
| \emptyset _Impeller | 101.6 mm |
| Groove Depth | 81 μm |
| λ , $r_{\text{Inner}}/r_{\text{Outer}}$ | 0.75 |
| α , Groove Angle | 15° |
| k, # of Grooves | 15 |
| g, ridge width/groove width | 1.0 |

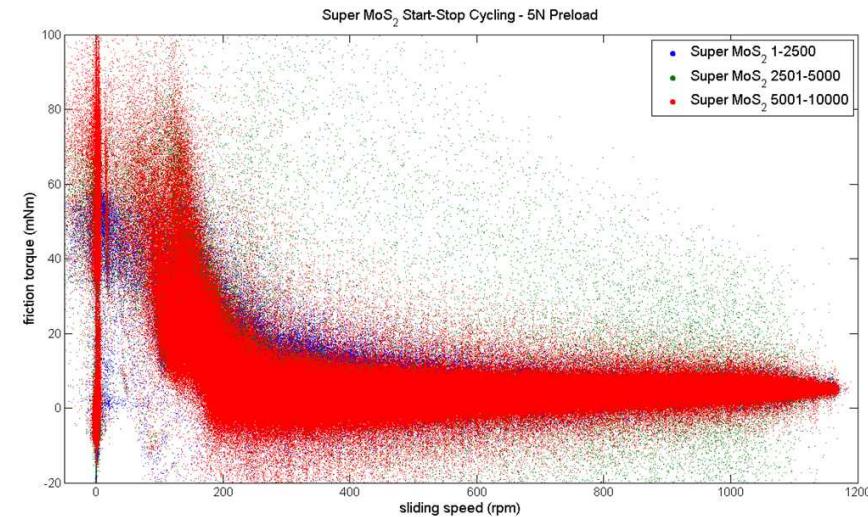
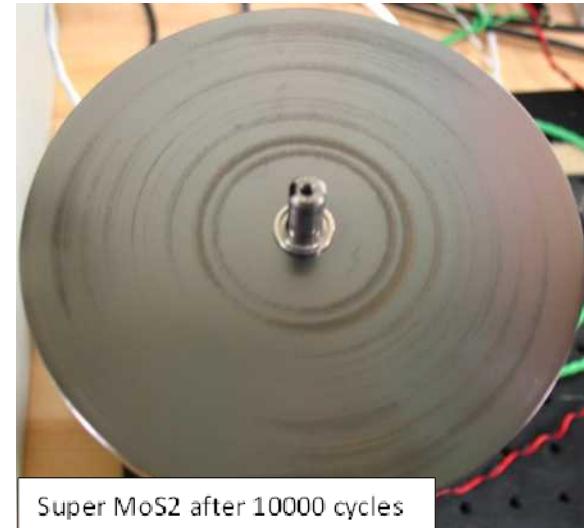
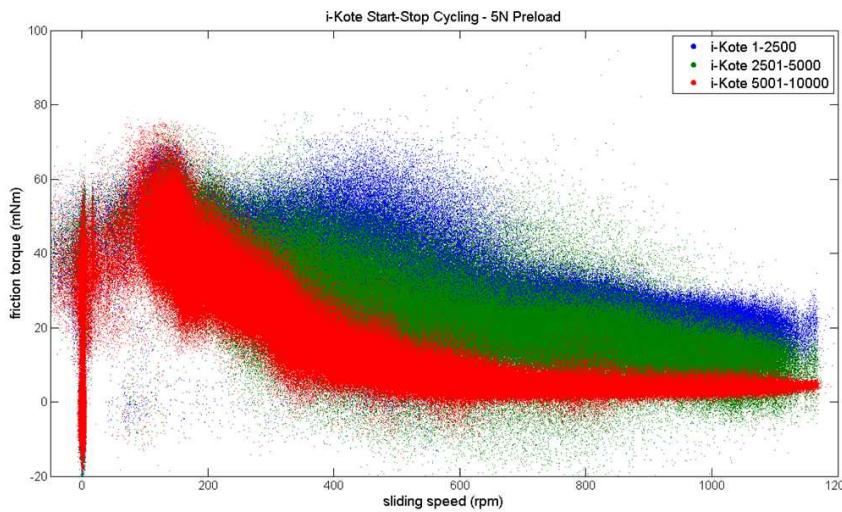
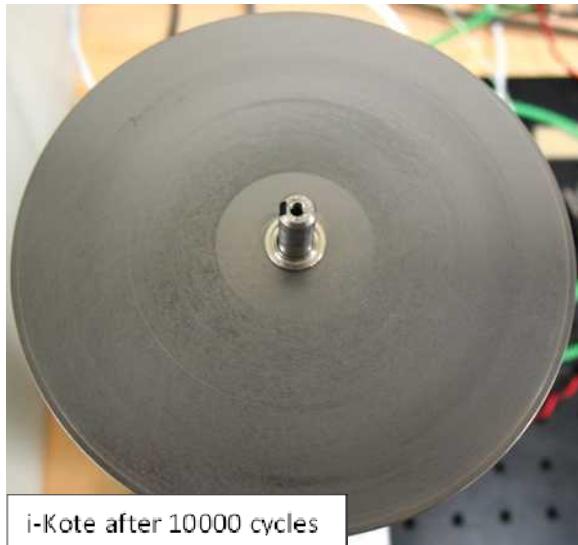
| Parameters | |
|---|------------------|
| \emptyset _Impeller | 101.6 mm |
| Groove Depth | 25 μm |
| λ , $r_{\text{Inner}}/r_{\text{Outer}}$ | 0.9 |
| α , Groove Angle | 15° |
| k, # of Grooves | 15 |
| g, ridge width/groove width | 1.0 |

| Parameters | |
|---|------------------|
| \emptyset _Impeller | 101.6 mm |
| Groove Depth | 35 μm |
| λ , $r_{\text{Inner}}/r_{\text{Outer}}$ | 0.9 |
| α , Groove Angle | 12° |
| k, # of Grooves | 15 |
| g, ridge width/groove width | 1.4 |

Air bearing provides stiff, low friction interface but thermal resistance is significant

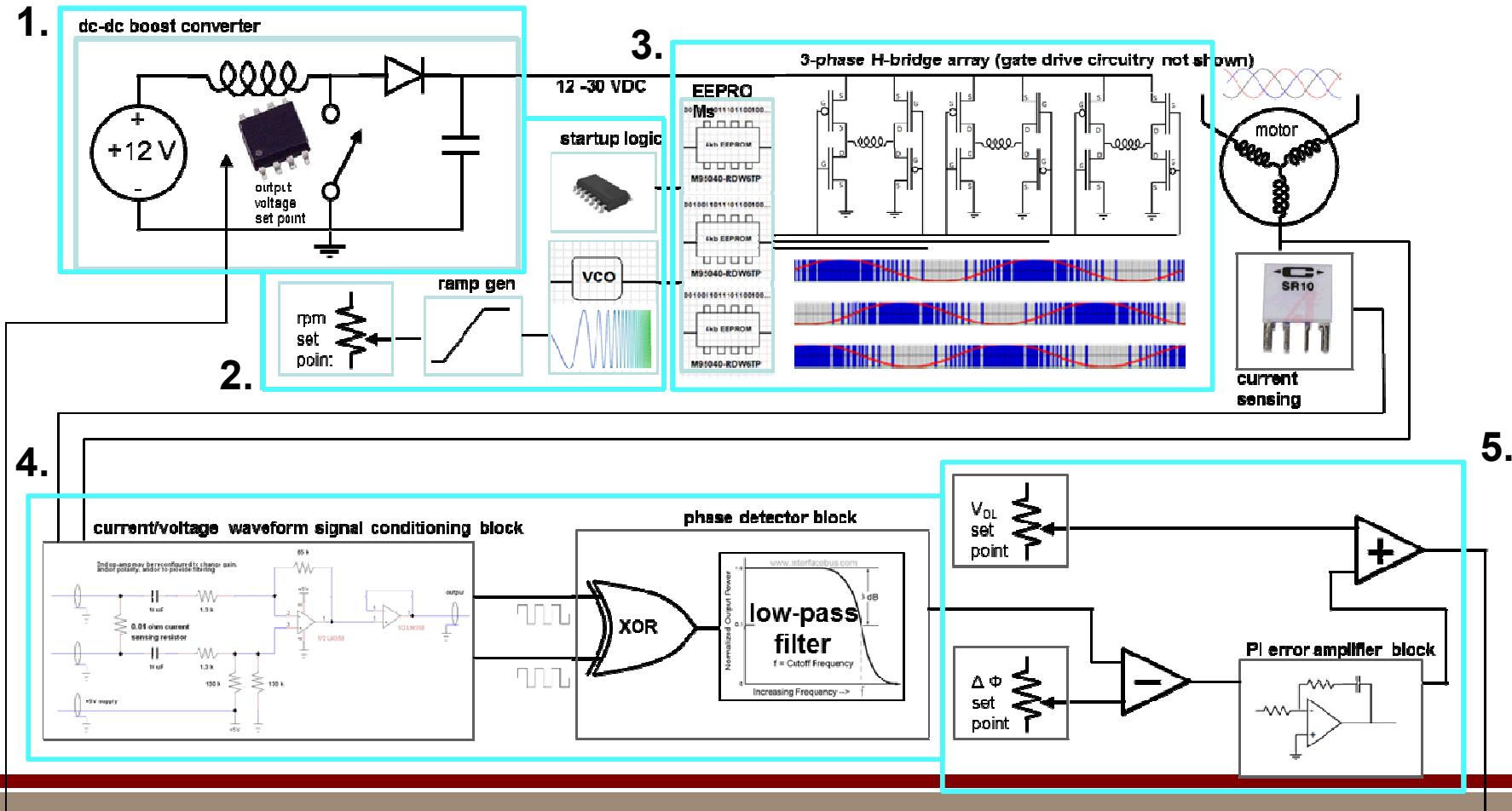


Two anti-friction coatings perform well out to 15,000 start/stop cycles



Custom motor controller in final stages of development

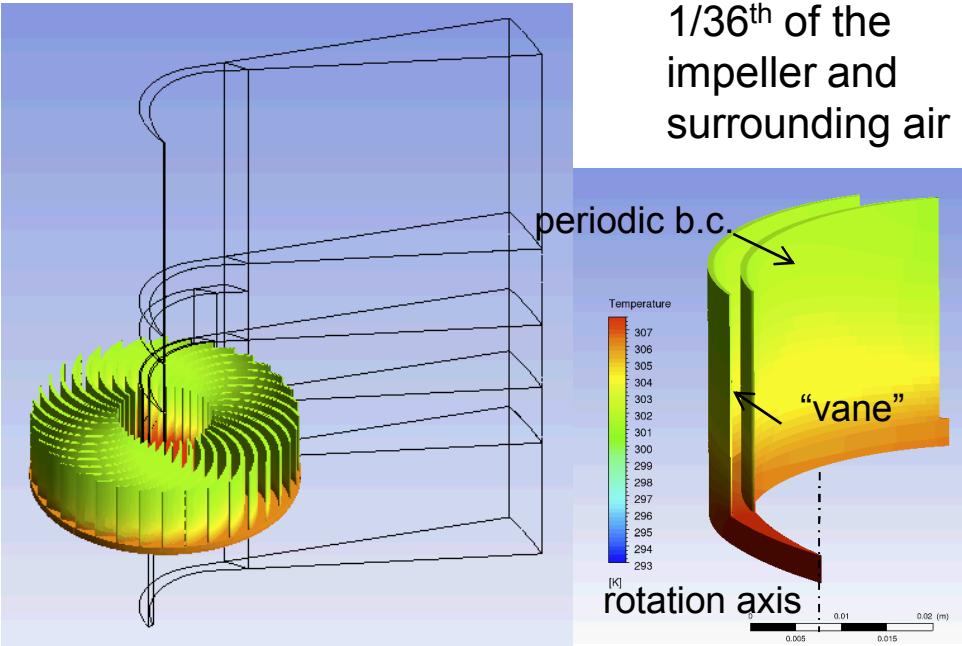
Five primary blocks: 1. Boost converter – prototyping complete, 2. VCO – final tuning for dynamic range complete, 3. H-Bridge – final tuning for efficiency and min heat loss complete, 4. Phase Detect – prototyping complete, 5. PI Loop – ongoing development



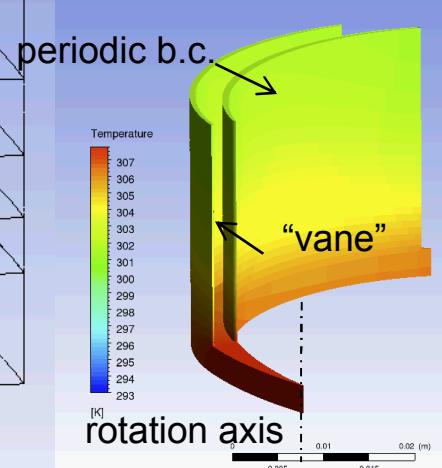
MODELING AND ANALYSIS

Computational fluid dynamics (CFD) models tell us a lot about the cooler performance

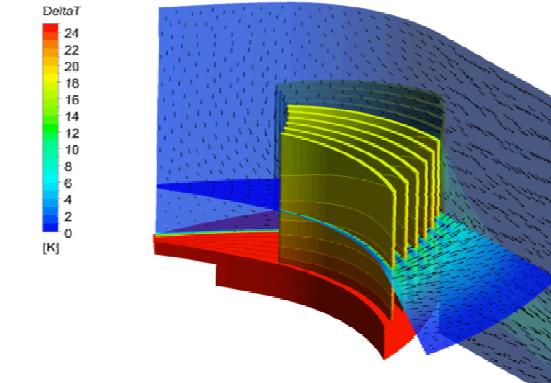
Example: V4 with 36 blades



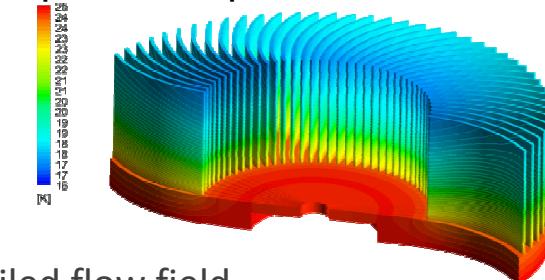
1/36th of the impeller and surrounding air



Flow field and air temperature



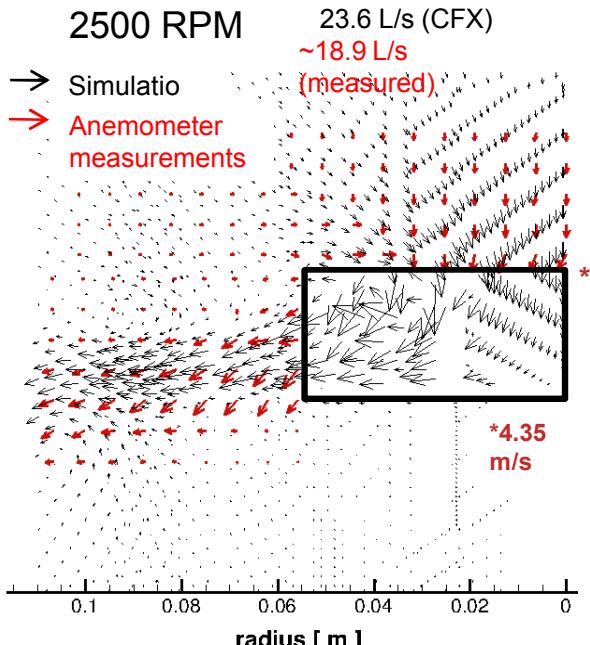
Impeller temperature and heat flux



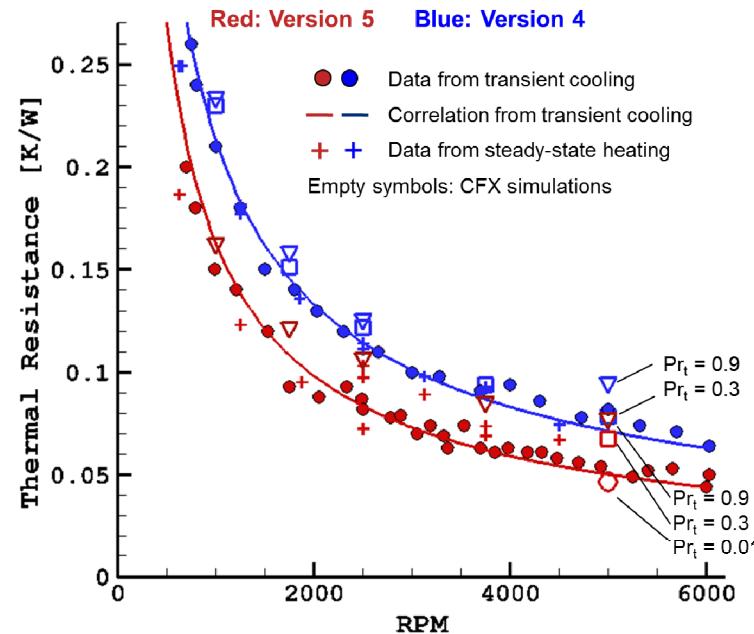
- ANSYS CFX V14.0
- Conjugate heat transfer (solid and fluid computation)
- Rotational reference frame for impeller
- Periodic boundary conditions take advantage of symmetry
- Reynolds-Averaged Navier Stokes (RANS) equations for flow field
 - Shear Stress Transport model
- Detailed flow field
- Temperature distribution in air
- Torque and power consumption
- Heat transfer coefficient
- Temperature distribution within solid regions
- Fin efficiency
- Where solid material is efficiently being used

CFD models have been experimentally validated

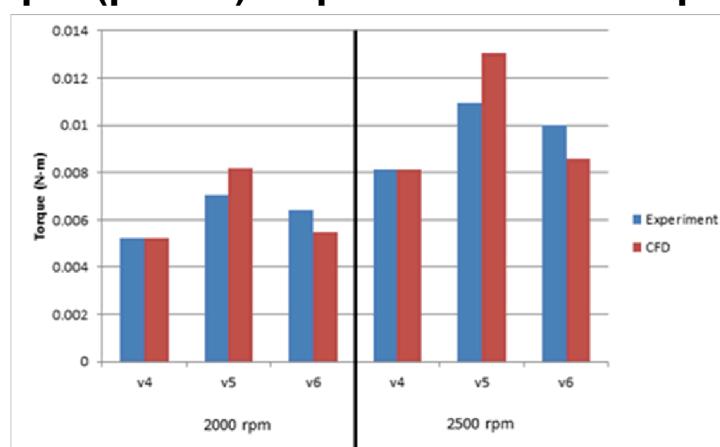
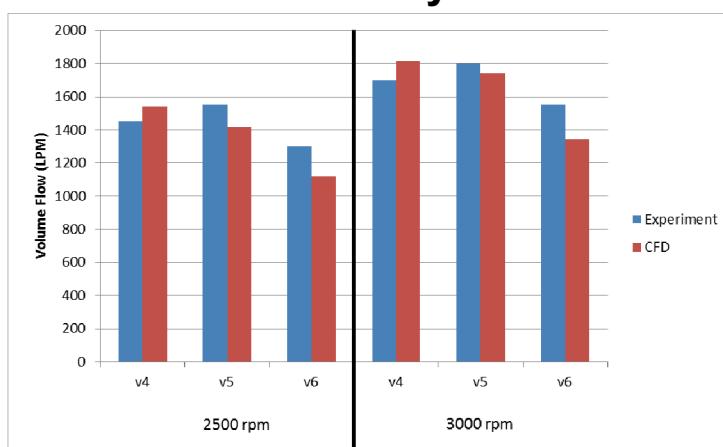
Flow Field



Thermal Resistance



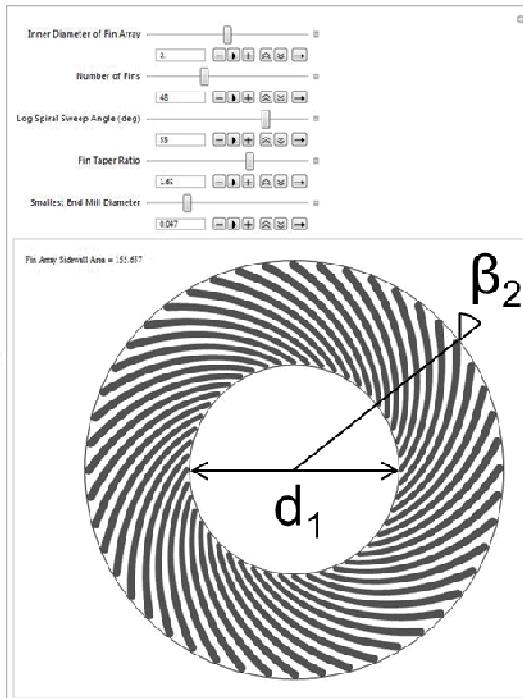
Torque (power) required to rotate impellers



CFD and design models have been used to carry out impeller parameter and scaling studies

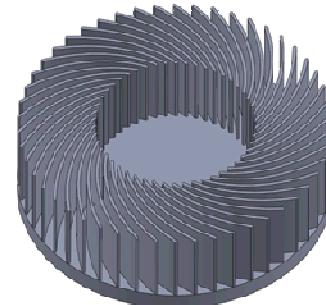
1. Generate Equations

Preliminary information (e.g. surface area) determined from Mathematica model



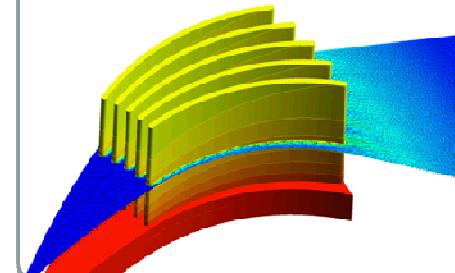
2. Create Geometry

3D geometry generated from vector equations in SolidWorks



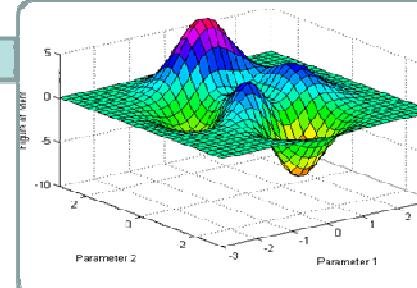
3. Simulate

Flow field and conjugate heat transfer simulated in ANSYS CFX



4. Evaluate

Determine sensitivity of thermal resistance to input parameters



1. Inner diameter (d_1)
2. Blade angle (β_2)
3. Number of fins (n)
4. Minimum endmill diameter (d_e)
5. Fin Taper Rate (power law dependence of blade width on radius)

Initial scaling study shows thermal resistance vs. motor power tradeoffs

CFD results for scale-up of V6 impeller

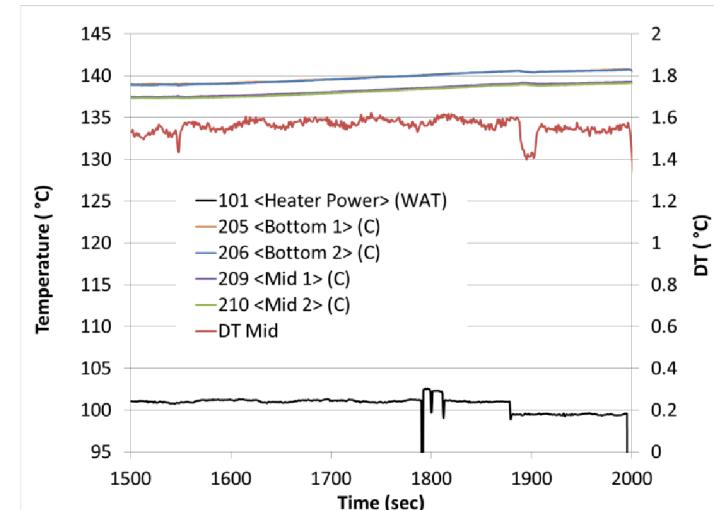
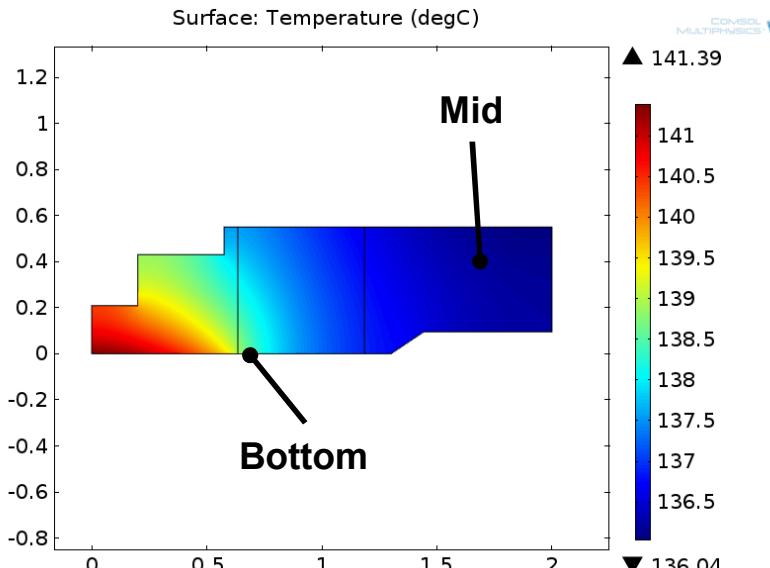
| Height (cm) | Diameter (cm) | Speed (rpm) | R (K/W) | Torque (J) | Mass Flow (kg/s) | Power (W) |
|-------------|---------------|-------------|---------|------------|------------------|-----------|
| 3 | 10 | 2500 | 0.097 | 0.0092 | 0.026 | 2.4 |
| 3 | 10 | 2500 | 0.118 | 0.0085 | 0.023 | 2.2 |
| 1.5 | 15 | 1666 | 0.119 | 0.012 | 0.021 | 2.1 |
| 1.5 | 15 | 2500 | 0.082 | 0.028 | 0.034 | 7.4 |
| 1.5 | 15 | 3000 | 0.071 | 0.041 | 0.041 | 12.9 |
| 1.5 | 15 | 5000 | 0.047 | 0.099 | 0.087 | 52.0 |
| 3 | 15 | 1666 | 0.079 | 0.021 | 0.038 | 3.7 |
| 3 | 15 | 1666 | 0.082 | 0.022 | 0.039 | 3.8 |
| 3 | 15 | 2500 | 0.061 | 0.050 | 0.061 | 13.0 |
| 3 | 15 | 2500 | 0.058 | 0.054 | 0.060 | 14.1 |
| 3 | 15 | 3000 | 0.054 | 0.073 | 0.074 | 22.8 |
| 3 | 15 | 5000 | 0.030 | 0.223 | 0.13 | 117.0 |
| 4.5 | 15 | 1666 | 0.051 | 0.033 | 0.059 | 5.7 |
| 3 | 20 | 1250 | 0.053 | 0.038 | 0.061 | 5.0 |
| 3 | 20 | 2500 | 0.030 | 0.143 | 0.15 | 37.5 |
| 4.5 | 20 | 2500 | 0.028 | 0.243 | 0.17 | 63.5 |
| 6 | 20 | 1250 | 0.031 | 0.079 | 0.11 | 10.4 |
| 6 | 20 | 2500 | 0.022 | 0.331 | 0.23 | 86.6 |
| 6 | 20 | 5000 | 0.017 | 1.353 | 0.48 | 708.5 |

- V6 geometry: 55 fins, 45° 1" inner radius, 3 cm height, 1.5 power law
- Uniform in-plane scaling; 1.5X and 2X
- Independent vertical scaling for some cases; 0.5X, 1X, 1.5X, and 2X
- Speed scaled inversely with diameter based on V6 @2500rpm for some cases

BACKUP

Vapor chamber baseplate improvement over solid Cu less than expected

Modeling and experiment used to determine Cu baseplate R_{th}



| | Solid Cu | Vapor Chamber |
|-----------------------|----------|---------------|
| ΔT @ 100W | 1.9 °C | 1.2 °C |
| R _{th} (C/W) | 0.04 | 0.02? |

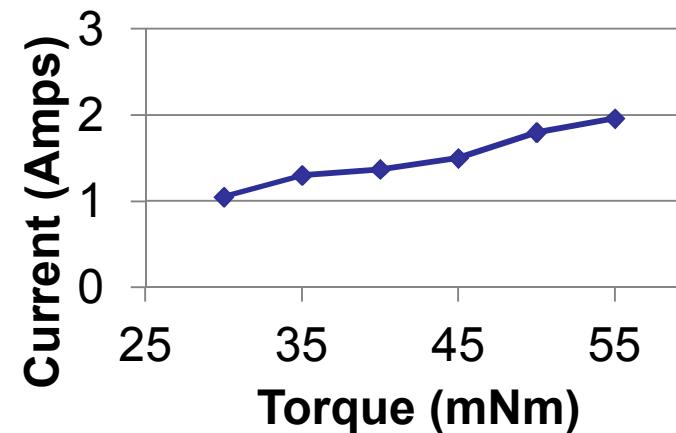
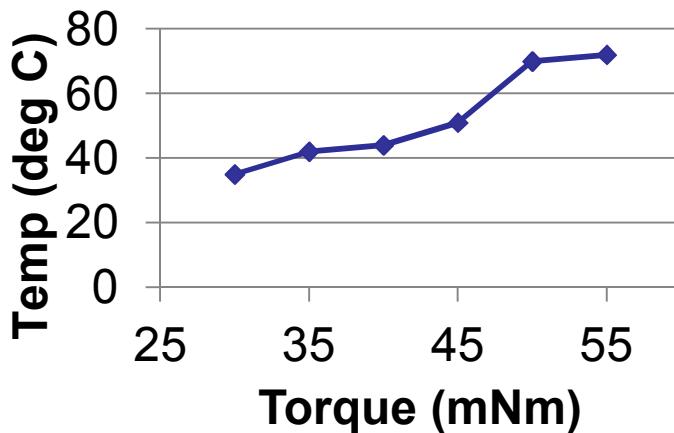
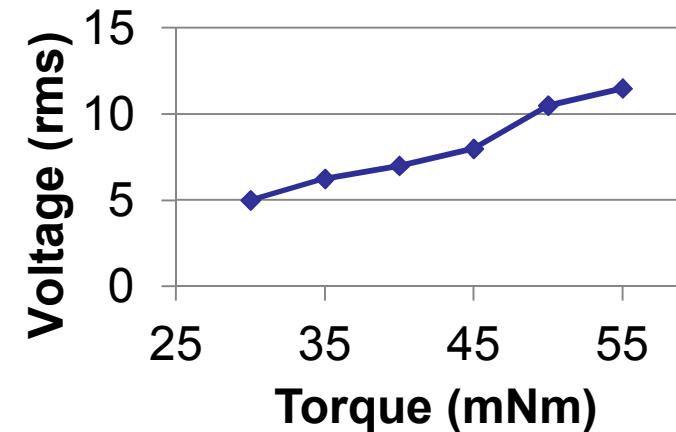
V5 device compared to standard and after-market CPU coolers



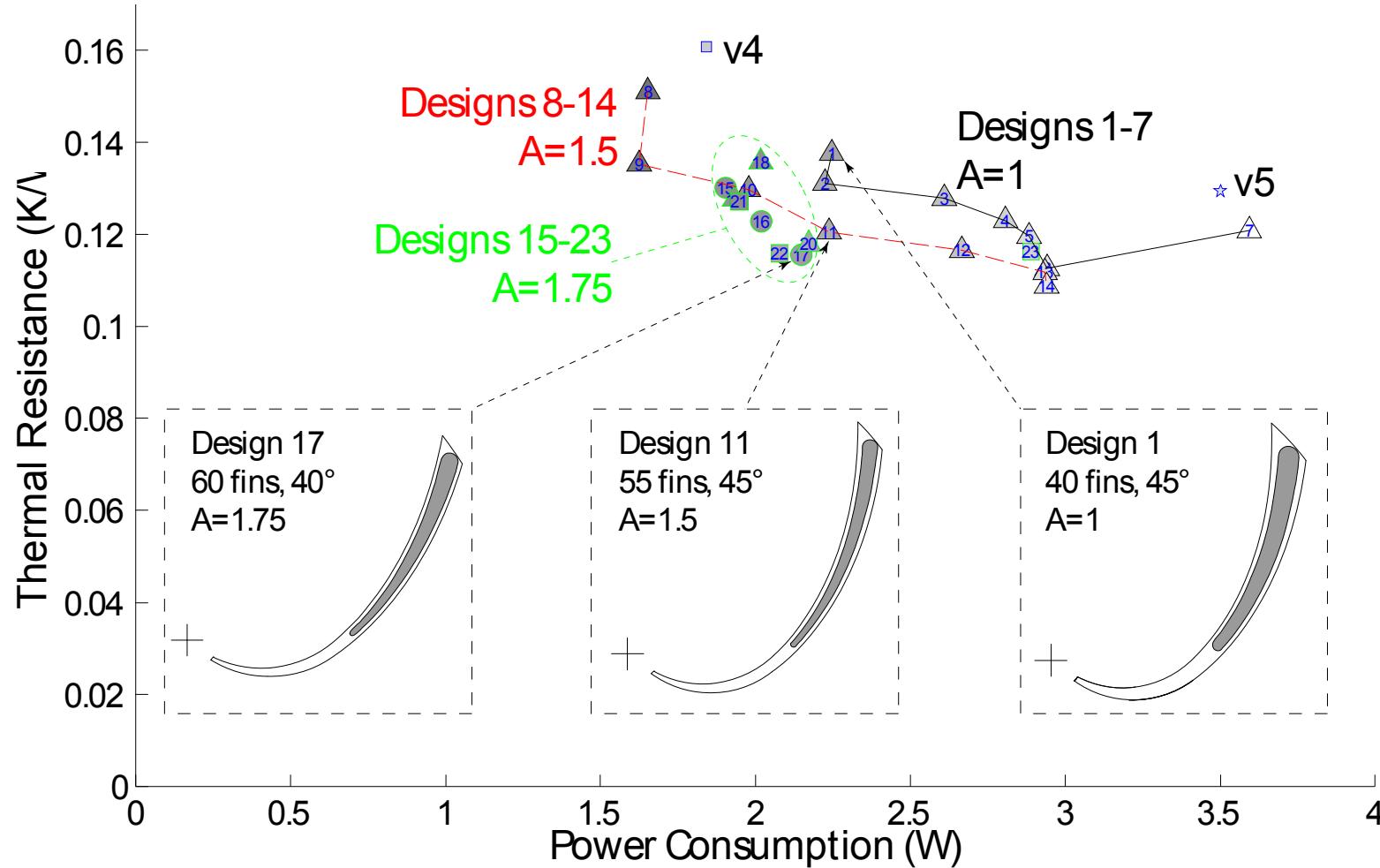
| | V5 (2500 rpm, 10 µm gap, Vap. Ch.) | V5 (3000 rpm, 5 µm gap, Vap. Ch.) | Intel i7 OEM cooler (Nidec F10T12MS1Z7) | Noctua NH-D14 |
|---|------------------------------------|-----------------------------------|---|---------------|
| R_{th} (C/W) | 0.156 | 0.12 | 0.252 | 0.10 |
| Impeller | 0.084 | 0.073 | | |
| Air gap | 0.052 | 0.027 | | |
| Baseplate | 0.02 | 0.02 | | |
| Power consumption (W) | ~5 | ~10 | 3.6 | 2.3 |
| Noise (dBA) | 48 | 49 | 28.5 | 30.5 |
| Dimensions (mm) | 100X100X46 | 100X100X46 | 100X100X62 | 140X158X160 |
| Volume (cc) | 460 | 460 | 620 | 3540 |
| Specific cooling capacity (W/C/cc) | 0.014 | 0.018 | 0.0064 | 0.0028 |
| Price (\$) | ? | ? | \$12 | \$80 |

Motor can overcome start-up torque with reduced contact area

- Motor can produce up to 55 mNm
 - potential higher but experienced voltage saturation from amplifiers
- 3-phase motor with 34 gauge windings ramped from 0 to 300 rpm in 1.5 seconds



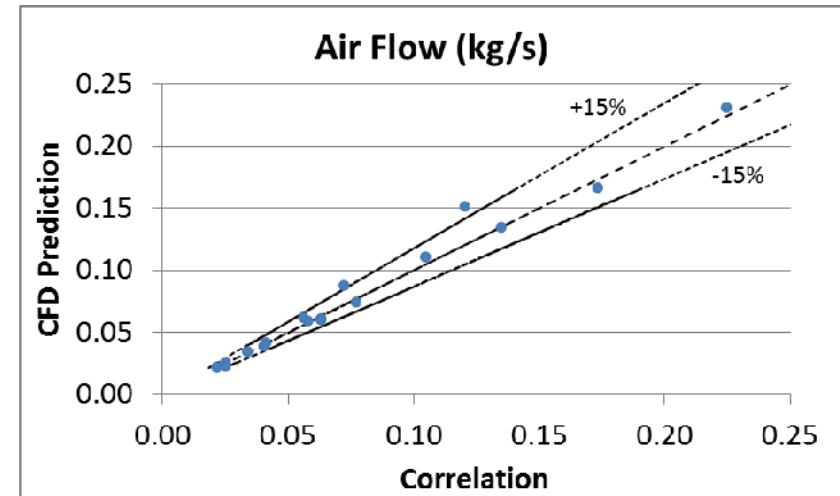
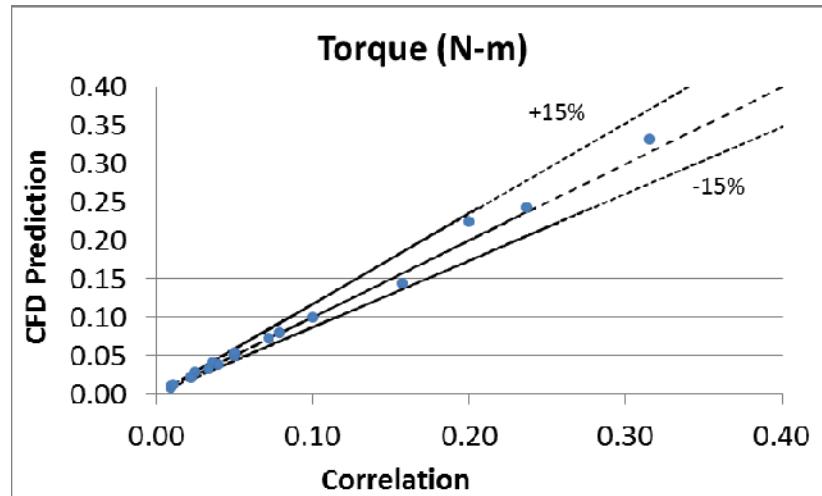
40 different permutations of the impeller geometry
were modeled to find an improved design



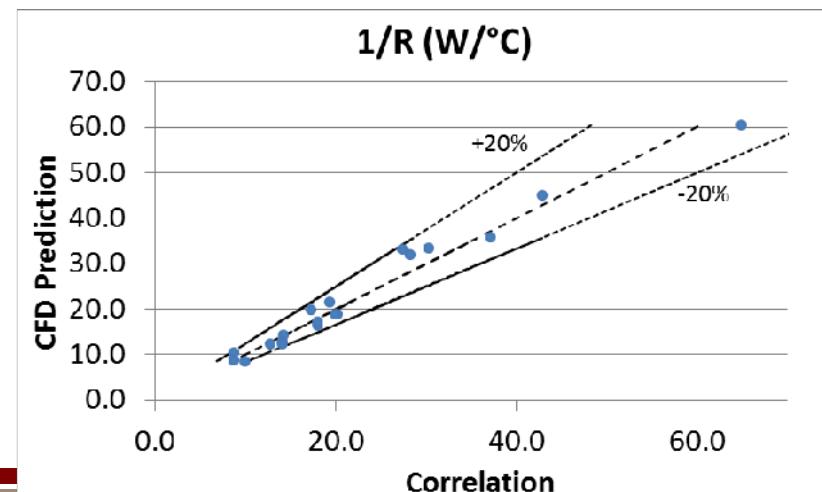
Correlations based on CFD studies predict impeller performance to within $\pm 20\%$

$$\tau = 4.8 \times 10^{-12} h \omega^2 d^4$$

$$Q = 1.16 \times 10^{-7} h^{0.9} \omega^{1.1} d^{2.25}$$



$$\frac{1}{R} = 2.82 \times 10^{-3} h^{0.5} \omega^{0.6} d^{1.8}$$



Note, since $Power = \tau \times \omega$:

$$P = 4.8 \times 10^{-12} h \omega^3 d^4$$

ω in rad/s
h in cm
d in cm

Ten Demonstration Units will be completed by January 2014

Most components are complete and ready to assemble

Impellers:

- 5 are complete
 - machined, coated, motor rotor installed
- 6 more have been machined, not coated



Vapor chamber baseplates:

- 9 are complete, 1 more in progress
 - Machined and coated



Shafts:

- 10 are complete



Motor Stators:

- 3 wound and ready

