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ANALYTIC MODELING OF A DEEP SHIELDING PROBLEM

By

TYLER J. REMEDES

A DISSERTATION PRESENTED TO THE GRADUATE SCHOOL
OF THE UNIVERSITY OF FLORIDA IN PARTIAL FULFILLMENT
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Previous generations of scientists would make tremendous efforts to simplify non-tractable problems and generate simpler models that preserved the fundamental physics. This process involved applying assumptions and simplifications to reduce the complexity of the problem until it reached a solvable form. Each assumption and simplification was chosen and applied with the intent to preserve the essential physics of the problem, since, if the core physics of the problem were eliminated, the simplified model served no purpose. Moreover, if done correctly, solutions to the reduced model would serve as useful approximations to the original problem. In a sense, solving the simple models laid the ground-work for and provided insight into the more complex problem. Today, however, the affordability of high performance computing has essentially replaced the process for analyzing complex problems. Rather than “building up” a problem by understanding smaller, simpler models, a user generally relies on powerful computational tools to directly arrive at solutions to complex problems. As computational resources grow, users continue trying to simulate new, more complex, or more detailed problems, resulting in continual stress on both the code and computational resources. When these resources are limited, the user will have to make concessions by simplifying the problem while trying to preserve important details. In the context of MCNP, simplifications typically come as reductions in geometry, or by using variance reduction techniques. Both approaches can influence the physics of the problem, leading to potentially inaccurate

251 or non-physical results. Errors can also be introduced as a result of faulty input into a
252 computational tool: something as simple as transposing numbers in a tally input can
253 result in incorrect answers.

254 In this paradigm, reduced complexity computational and analytical models still have
255 an important purpose. The explicit form of an analytic solution is arguably the best way
256 to understand the qualitative properties of simple models [2]. In contrast to “building
257 up” a complex problem through understanding simpler problems, results from detailed
258 computational scenarios can be better explained by “building down” the complex model
259 through simple models rooted in the fundamental or essential phenomenology. Simplified
260 analytic and computational models can be used to 1) increase a user’s confidence in the
261 computational solution of a complex model, 2) confirm there are no user input errors, and
262 3) ensure essential assumptions of the simulation tool are preserved.

263 This process of using analytic models to develop a more valuable analysis of
264 simulation results is named the results analysis methodology. The utility of the results
265 assessment methodology and a complimentary sensitivity analysis is exemplified through
266 the analysis of the neutron flux in a dry used fuel storage cask. This application was
267 chosen due to current scientific interest in used nuclear fuel storage.

CHAPTER 1 INTRODUCTION AND MOTIVATION

1.1 Motivation

America is the largest producer of nuclear power in the world, with 98 reactors producing approximately 805 billion kilowatt-hours of power in 2017 [3]. Despite being the largest producer of nuclear energy, the United states has not established a permanent used nuclear fuel storage facility. Instead, nuclear power plants store used fuel on site, many using storage casks or canisters. A Savannah River National Laboratory report states nearly 100,000 fuel assemblies are stored in more than 2,000 casks at 75 storage sites [4].

Fuel casks are designed to store and protect spent nuclear fuel while shielding power plant workers and others from harmful radiation generated by unstable radioisotopes created through the fission process. There does not exist a singular design of a spent fuel cask due to multiple companies designing fuel casks and various types of spent nuclear fuel which need to be stored. While each design is varied, there exist certain components which are found across many spent fuel cask designs. Spent fuel casks typically have a right cylindrical shape with layers of high atomic number and low atomic number materials, such as steel alloys and concrete respectively. Layering materials with different atomic numbers provides radiation shielding for both gamma rays and neutrons, which are the two most penetrating types of radiation emitted by radioisotopes present in the fuel (e.q., O-17, Cm-242, and Sr-90). High atomic number materials are used to mainly shield gamma rays, whereas low atomic number materials are used to mainly shield neutrons. For this reason, most spent fuel casks have an inner region where spent fuel is stored, an outer region made of low atomic number materials (i.e. concrete) and high atomic number materials (i.e. steel alloys). Layered materials are also utilized in baseplates and lids.

Materials in a spent fuel cask are specifically chosen to be multi-functional. Spent fuel casks must conduct heat away from spent fuel rods, protect fuel from damage, prevent proliferation of radioactive materials, as well as shield radiation. Heat conduction is

achieved by using materials with high thermal conductivities to draw heat away from spent fuel rods to the environment. Thick layers of materials, such as steel and concrete, protect the cask contents from environmental or other sources of damage (i.e. a hurricane or a cask being dropped during transportation). Casks are also designed to prevent proliferation by, for example, featuring welded lids or the addition of security tags to discourage unauthorized access to spent fuel. Finally, spent fuel casks are designed to shield employees and the public from the harmful radiation produced by decaying radioisotopes created in the fuel during the power making process.

If a cask inadequately performs any of the above functions, it may become necessary to open the cask for a visual inspection. This is a costly and time consuming endeavor. Greulich et. al. state the cost to re-open a cask could be in the millions of dollars and require man-months of time [5]. The process of opening a cask to visually inspect the contents also carries an increased risk of exposing workers to radiation. The high costs associated with opening a cask would certainly make visual inspection an unappealing option. Simulation based and experimental research has been motivated by the desire to develop a non-destructive assay technique to verify cask contents.

Analyzing the capabilities of technology to ensure the contents of a spent fuel cask has motivated many scientific investigations, with a large reliance on computational simulations [5–7]. Simulation results can then be correlated to experimental observations in order to identify promising techniques to inspect the interior of a cask without opening the cask. Neutron flux and dose are common measurable quantities sought after in the simulation and experimental works surrounding radiation shielding investigations of spent fuel casks. In reality, these two quantities are the same with the latter being a scalar multiple of the former. These works tend to be concerned with the neutron flux at or beyond the surface of the cask, since the radiation environment exterior to the spent fuel cask is potentially harmful to worker safety. Understanding the interior neutron flux is useful in any simulation studying the exterior neutron flux. The neutron flux at the

surface of the spent fuel cask is directly dependent on the physics occurring interior to the spent fuel cask. Ultimately, the behavior of the interior neutron flux is controlled by the configuration and choice of materials inside the fuel cask. The relationship between interior structure and exterior neutron flux has prompted many simulation investigations using radiation transport codes. Further, simulation tools are not only used to design non-destructive assay techniques, but are also used to validate radiation transport codes as applied to spent fuel casks.

Ideally, simulation results should be compared to a series of identical or similar experiments *and* numerous results from other computational and numerical tools, and analogous analytical models. Computational, numerical and analytical tools act complimentary to experiments, in that the former tend not to be limited by physical restraints such as, but not limited to, detector placement, experimental design challenges, personnel safety, and costs. Nonetheless, experimental data is highly sought after since analytical models only provide exact solutions for the most simplistic non-physical problems and computational and numerical tools only approximate solutions, albeit these approximations can be quite accurate. Unfortunately, limited amounts of experimental data result in an increased reliance on computational and numerical tools. To further exacerbate the issue, it is of utmost importance that conclusions can be confidently drawn from simulation results. In the case of spent fuel casks, human lives and livelihood depend on the correctness of simulation results. The results assessment methodology provides a way to ensure the appropriateness and inerrancy of computational and numerical tools.

The results assessment methodology formulates analogs which are designed to share phenomenological physics with its more detailed counterpart. A discussion motivating the use of analogs is, therefore, useful. Fickett describes analogs as a qualitative representation of the original, constructed, not derived, in order to maximize simplicity while minimizing loss important properties [8]. Further, analogs have the following benefits 1) exact solutions are simpler to find and more likely to exist, 2) mathematical rigor in determining

analytical solutions is reduced, and 3) salient physics is more readily observable after the removal of extraneous features. The simplified computational and analytical models used in this work are developed as analogs. Before further description of the analytical models acting as analogs in this work, it is important to discuss the processes of validation, verification, uncertainty quantification and sensitivity analysis as applied to general computational tools and to simulations of used fuel casks.

1.2 Practices for Code Reliability, Confidence, and Predictive Capability

The behavior of physical systems is commonly described using complex mathematical expressions, typically consisting of differential equations. Exact solutions of these equations (also variously known as analytical or closed-form solutions) tend to be limited to only the simplest scenarios. Indeed, the cost of exactly solving these equations often involves the extensive use of simplifying assumptions to reduce the complexity of an equation to a form where an analytical solution is possible. Approximating a differential equation as a series of coupled linear equations became an alternative to finding direct analytical solutions. Unfortunately, discretization introduces a degree of error into the solution proportional to the fidelity to which a problem was discretized. Further, discretization requires a high degree of computational rigor and, therefore, was not a realistic technique for solving differential equations until adequate advancements in computation had occurred. However, the modern-day advancement of computational power has motivated the development of tools which approximate the solutions of complex differential equations in broad sets of circumstances via approximation techniques, as opposed to simplifying assumption techniques that may yield closed-form solutions only in special cases.

These simulation tools, or simulation codes, often rely on algebraic calculations to approximate solutions of the complex differential equations which describe real-world physics. The processes of verification and validation generate evidence “that computer [codes] have adequate accuracy and level of detail for their intended use” [9]. Verification

assesses “the numerical accuracy of the solution to a computational model,” and validation “addresses the physical modeling accuracy of a computational simulation by comparing the computational results with experimental data” [10]. Stated another way, verification studies if a code solves equations correctly, and validation investigates the utility of a code through comparison with experimental data. Beyond verification and validation, uncertainty quantification has been added to quantify the accuracy with which simulation codes predict outcomes. Sensitivity analysis can be considered a type of uncertainty quantification which stratifies input parameters based on degree of impact to the error of simulation results. A short description of verification, validation, and sensitivity analysis will now be discussed.

Verifying a simulation tool requires demonstrating that the code is approximately solving the underlying equations as intended by the code developer [11]. Two examples of verification methods are benchmarking and comparison to analytical models. In benchmarking, results from a simulation tool are compared to known solutions from experiments, numerical tools, or other verified simulation tools.

Validation is ensuring a simulation tool approximately solves a representative set of equations consistent with the applications of the code. Validation relies on comparing experimental, analytical or numerical results against simulation results and validation is conducted on an application specific situations. Simulation tools are validated for different applications on a case-by-case basis. Validation commonly requires experimental data for a given application. However, sometimes experimental data is limited or non-existent since experiments can be financially burdensome, potentially risky to public and worker health, or difficult to conduct due to proprietary reasons. Difficulties in obtaining experimental data have led to alternative methods for validation, namely comparison between simulation codes. This methods attempts to validate a code by comparing results with a previously validated simulation code for an application.

Verification and validation determine the accuracy of a simulation tool for a specific application, however, neither certifies the usage of a particular code in solving a problem. Rather, it is the responsibility of the analyst to ensure that a particular simulation is relevant to the solution of a problem - ensuring the simulation is made in accordance with the manner which the code is verified and validated. Uncertainty analysis and sensitivity analysis aid in this process.

Simulation tools require parameters, or data provided by the user, such as physical properties measured through experiments (e.g., cross section data, viscosity, or thermal conductivity). These values have associated error; measurement error is an example. Uncertainty quantification is concerned with quantifying the error on the final result due to the error of input parameters, as well as, error introduced by the simulation tool itself. Further, uncertainty quantification is not concerned with the truth of a model, rather how parameters effect the solution [12]. Sensitivity analysis improves on uncertainty quantification by identifying which parameters most influence the result.

The typical approach to computational sensitivity analysis requires performing many simulations where a change is made in each computation - an approach called the direct method [13]. This process requires excessive computational resources. Applying sensitivity analysis techniques to analytical models allows for the identification of sensitivities without requiring as much computational resources, a strength of analytical sensitivity analysis. Further, if an equation yields an analytical solution, the sensitivities of an equation to its parameters can be found with minimal computational resources and requires solving sensitivity equations only once.

The history of sensitivity analysis as applied to differential equations is broad and extensive, therefore, only previous research that pertains to this work will be discussed. The first methodology for sensitivity analysis was developed on linear electrical circuits by Bode in 1945 [14]. At that time, sensitivity analysis motivated the use of feedback in circuit design. From its origins in circuit control, sensitivity analysis permeated

many others fields of science, including nuclear engineering, and many methods were developed. McKay provides an introduction into basic definitions and concepts related to sensitivity analysis [15]. Cacuci unified and generalized the direct method and the perturbation methods of sensitivity analysis in 1980 based on Frechet-derivatives [13]. A year later, Cacuci further generalized his methodology to analyze systems of response along arbitrary directions using the Gâteaux-derivative (G-derivative). This non-linear operator determines system responses to multiple perturbations in input parameters simultaneously. In doing so, Cacuci developed the Forward Sensitivity Analysis Procedure (FSAP) and Adjoint Sensitivity Analysis Procedure. The FSAP will be used to find sensitivities of the linear differential equations in this work.

The overall purpose of performing verification, validation, uncertainty quantification, and sensitivity analysis procedures is to identify the accuracy of a particular code for given scenarios. Ultimately, a code user must decide if a code adequately simulates the problem and if the user can have confidence that the simulated results are an accurate portrayal of the real-world problem. While the processes of verification, validation, and uncertainty quantification have been and continue to be extensively developed, there exists a limitation - how can a code be validated if there is no experimental data for comparison? The purpose of this document is to introduce a methodology aimed at answering this question.

1.3 State of Current Used Fuel Cask Research

Interest in experimental and simulation work stems from the need to ensure the safety and security of spent fuel casks; since there is currently no long term, national storage plan. Before discussing how simulations have been used to study spent fuel casks, it is important to take an aside and discuss the verification and validation of a commonly used radiation transport simulation code, the Monte Carlo N-Particle (MCNP) simulation code.

MCNP has been extensively verified and includes a series of benchmark problems. Further, Mosteller compiled a list of documents which discuss verification efforts on

MCNP [16]. analytical models have also been used in validation efforts [17, 18]. analytical models provide an exact solutions against which simulation tools can be compared. However, exact analytical solutions are often only available for heavily simplified problems which do not represent physical systems. Nonetheless, excellent agreement has been achieved between simple MCNP models and analytical solutions. Verification is considered an activity in mathematics where a successful test demonstrates that the governing equations of a simulation tool were solved correctly [19]. Validation of a simulation code is undertaken after verification.

MCNP has also undergone general validation in multiple disciplines within nuclear engineering; including but not limited to radiation shielding [20], criticality [21], and intermediate and high-energy physics [22] where MCNP results were compared to simple experiments. In order to validate computational tools as applied to spent fuel casks, scientist have turned to a comparative method where results from other radiation transport codes are compared with MCNP [23, 24]. However, discrepancies between results from different simulation tools are attributed to different physics being included in each tool. While this may be the driving factor leading to the apparent disagreement, this conclusion would benefit from identifying the physics seen in one simulation tool and neglected in the other. Comparison with experiments and other simulation tools is a valid, imperative, and important technique for validating simulation results, but more analysis should be done in order to increase confidence that simulation results can be trusted.

There exists a limited body of experimental work which measured the radiation dose at or near the surface of various used fuel casks. Unfortunately, none of the experiments were conducted on HI-STORM 100 spent fuel casks. Hence, discussion of past experiments will include radiation measurements performed on any spent fuel cask, including but not limited to experiments compared to any radiation transport code. Thiele et. al. even include a comparison between experimental results and the results from two radiation transport simulation tools (comparing Monaco/MARVIC with SAS4/MORSE) [25].

Both simulation tools are developed as part of the Standardized Computer Analysis for Licensing Evaluation (SCALE) packages by Oak Ridge National Laboratory. Since these radiation transport codes are not used in this work, no further explanation of the codes will be given. The author's report concludes that simulation tools can be applied for the assessment of dry storage casks. While experimental validation of simulation results is arguably the best way to corroborate simulation results, it is still important to not treat experimental data as sacrosanct [19]. Experiments still include measurement and procedural errors, and without the validation of multiple experiments of the same cask, the result of a lone experiment should not be considered to validate or invalidate simulation results. There also exists a large number of various dry fuel storage casks and experimental data may not exist to validate simulation results against. Computational and numerical tools have proven useful as an alternative to experimental data. Zioc et al. measure the thermal neutron and gamma ray radiation signatures from six different spent fuel cask designs; the HI-STORM 100 was not one of the six. The authors posit the radiation signature can be used as an identifier for individual casks. Their experiments proved inconclusive resulting from limitations of the imaging devices used. Wharton et al. used MCNP to determine the fraction of gamma rays which would be detected by a high purity germanium detector placed at the top surface of two spent fuel cask designs [26]. These simulations were used to determine the feasibility of a system designed to used passive gamma radiation to determine if a fuel bundle was present or absent from a spent fuel cask. The authors concluded that the thick shielding of the spent fuel casks measured sufficiently scattered radiation and the system was not capable of resolving discrete gamma ray peaks. This resulted in the measurements being stopped without fully testing the capabilities of the system. It should be noted, the MCNP results suggested the system was capable of performing the measurements and distinguishing between empty and filled fuel storage positions. This work serves as an example for the importance of corroborating simulation results with further investigations.

Simulation studies of the HI-STORM 100 spent fuel cask using MCNP are more numerous than experimental studies. Priest conducted an in-depth investigation of neutron and gamma flux and dose rates interior to a HI-STORM 100 spent fuel cask with the purpose of identifying an imaging system capable of withstanding the harsh environment inside the multi-purpose canister (MPC) [27]. The author performed simulations using multiple MPC configurations with used nuclear fuel from both pressurized water and boiling water reactors.

Harkness et. al. used MCNP to investigate the validity using helium-4 fast neutron detectors to determine if fuel had been removed from a HI-STORM 100 spent fuel cask [7]. This work describes a methodology to generate a source definition for MCNP based on data provided in the Next Generation Safeguards Initiative. This fuel rod composition data was aged using ORIGEN-S, a material irradiation and decay calculation code, to create an MCNP compatible source definition. A further description of this process will be provided later in this work. The neutron flux and energy spectrum at the surface of the cask were tallied as part of this investigation. From the results of MCNP simulations, the authors concluded that neutron spectroscopy was feasible using helium-4 detectors, however, confidently determining if all fuel was present in a sealed spent fuel cask required further investigation.

Miller et. al. determined the feasibility of using a mono-energetic photon source to verify the contents of a sealed HI-STORM 100 spent fuel cask [28]. The authors simulated photon transport through the spent fuel cask and found a 1000-fold reduction in the transmitted flux when a fuel assembly is present as compared to a reduction of two in the transmitted flux when there is no assembly present. The authors further corroborated their work using analytical calculations to predict the scale of the uncollided flux for when a fuel assembly is present and when there is no fuel assembly. The results from their analytical modeling agreed with corresponding MCNP simulations.

Kelly et. al. performed an uncertainty analysis in radiation dose exterior to a HI-STORM 100S (a variant of the HI-STORM 100 cask) spent fuel cask based on variabilities in concrete composition and density [29]. The authors state that density variations in the concrete have the largest effect on radiation shielding capabilities. Varying concrete composition mostly affected neutron and associated capture gamma ray dose rates.

Because of the interest in modeling radiation transport in spent fuel casks, research has not been limited to using MCNP as a simulation tool nor has it been limited to a single cask design. Gao et. al. use the radiation transport code MAVRIC (a radiation transport code developed by Oakridge National Laboratory and distributed in with the SCALE code package) to simulate neutron and gamma transport through a TN-32 spent fuel cask [30]. In this work, the authors explore the effect of two geometries and two sets of cross section data on the neutron and gamma fluxes at the surface of the cask. The authors used a detailed model which included details of individual fuel rods and a homogenous model which calculated a homogenous fuel definition that simplified the geometry in each fuel cell. The authors also used two sets of cross section data. The first set were continuous energy cross section data and the second were multigroup cross sections. The authors concluded that changes to the geometry of the problem had a larger effect on the result than changing how the cross section data was handled.

Interest in verifying cask contents has led to simulations investigating methods for tomographic imaging. These investigations rely on simulation tools as a proof of concept and to aid experimental design. Liao and Yang have used cosmic-ray muon simulations to aid in experimental design choices for a spent fuel cask tomography system [31, 32]. The authors used Geant4 (another radiation transport code) and MCNP to simulate cosmic-ray muon transport through a spent fuel cask as well as through a test setup to guide experimental design. The authors then conducted experiments using the prototype

muon imaging systems. The authors concluded they were able to detect a quarter of a missing fuel bundle located anywhere in the cask.

Greulich et. al. also investigated the possibility of tomographic imaging techniques in verifying the contents of a spent fuel cask [5]. The authors simulated neutron transport through a TN-32 spent fuel cask using MCNP. Using a beam source of neutrons incident at the surface of the cask, the uncollided flux of neutrons leaving the cask provides information which can be used to reconstruct an image of the interior of the cask.

The previously described works were all interested in either radiation dose or radiation flux values at or exterior to the surface of the cask. Since dose is directly proportional to flux, and since the exterior neutron flux is a direct result of how interior cask structure affects the interior flux, the aim of this work is to investigate the interior neutron flux so as to have the most general relevance to existing work. The neutron flux was chosen over other types of radiation as the governing equation for neutrons is linear and provides a basis and proof-of-concept for the results assessment methodology.

The body of work focusing on simulations of spent fuel cask is quite large, which demonstrates scientific interest in simulating spent fuel casks. However, experimental data to validate simulated results is limited. Further, the final safety analysis report delivered by Holtec when licensing the HI-STROM 100 spent fuel canister system did not include any experimental data pertaining to the radiation shielding capabilities of this design [1]. Instead, MCNP was used to demonstrate the cask design was capable of attenuating radiation to an adequate level. Maintaining a safe environment for power plant workers and members of the public is of utmost importance and an alternate method for validating the accuracy of simulation results is needed if simulation results are to be relied upon in the absence of experimental data. The discrepancy between the amount of simulated results and experimental data identifies the need to validate or otherwise reinforce confidence in simulation results without relying on experimental data.

1.4 General Description of the Work

The results assessment and sensitivity analysis methods presented in this work act complimentary to existing techniques - verification, validation, uncertainty analysis, and sensitivity analysis- in order to develop a more valuable analysis. This work includes high-fidelity MCNP simulations of the interior neutron flux from a Holtec HI-STORM 100 spent fuel cask, and the attendant analytical analysis of the simulation results in the absence of significant experimental validation data. A detailed model of the HI-STORM 100 spent fuel cask is simulated in MCNP to investigate the neutron flux interior to the fuel cask. Owing to a lack of validation data against which to compare these simulation results, an analytical analysis framework called "simulation results assessment" (or, henceforth, "results assessment") is developed and applied to provide an alternative (but not replacement) means for enhancing confidence in the computational model. The accuracy of the model is assessed by first developing simplified analytical and MCNP computational models. The design of these analogous models is made to retain essential physics while reducing geometric complexities. Since the essential physics is preserved, the neutron flux found using the analogous models will approximate the neutron flux interior to the cask of the detailed model. Developing analogous models is an iterative process where the initial simplified models were overly simplified and lost essential physics. Essential physics was identified from locations where disagreements between the results of the detailed model and the analogous models occur. More detailed analogs are developed in order to rectify differences observed between the two sets of results until a final set of analogous models are found. This process identified physical details that must be preserved in the detailed model in order for the detailed model to accurately simulate reality. A sensitivity analysis is also conducted on the final analogous model in each material region as well as on the detailed model in order to further validate the accuracy of the detailed model through the comparison of sensitivity structures between the models. This is also an iterative process which involves further refinement of analogous models

and input parameters in order to achieve comparable sensitivity structures between the detailed and analogous models. Finally confidence that the detailed model accurately simulates the interior neutron flux of a HI-STORM 100 spent fuel cask is increased after reaching comparable results and sensitivity structures between all models.

1.4.1 Results Assessment

A detailed model of this cask is developed in the MCNP code to predict the neutron flux in its interior. In an attempt to isolate essential physics, 1) five other MCNP simulations are developed to model various analogous problems, and 2) analytical models are developed to explain key characteristics of the flux seen in these analogous problems. The results of the simplified calculations are then used to reveal the fundamental physics controlling the shape and other characteristics of the flux distribution resulting from the complex model. This procedure is phenomenological in nature, and is thus intended to capture elemental physical processes that are occurring within sub-regions of the full-scale system. Therefore, while no single analytical solution is expected to be available for the full-scale system, any understanding gained in the sub-regions reinforces confidence that the integrated scales are being simulated in accordance with physical intuition. This outcome is valuable in cases where experimental data is sparse or nonexistent. A complimentary investigation of sensitivity structures produces a quantitative basis for comparison of analytical and computational models.

1.4.2 Sensitivity Analysis

The procedure of quantifying comparisons between analytical models, reduced geometry computational models, and the full model is demonstrated through the inclusion of sensitivity analysis procedures. Forward modeling of sensitivity structures is conceptually simple but computationally expensive for large problems, as it involves sampling a space of possible parameter values and executing a new simulation for each value. For analytical models the procedure is the opposite: using a generalized notion of the directional derivative, sensitivity structures can be computed in closed-form. The

comparison of these two methods forms the final component of this work. In addition to basic physics phenomenology, the sensitivity structure arising from analytical models can be compared to that found from forward sensitivity modeling of full-scale simulations. When these structures compare favorably, confidence in the full-scale simulations is once again reinforced.

1.5 General Overview of Chapters

This document discusses the rigorous analysis of a HI-STORM 100 used fuel cask using the results assessment methodology and a sensitivity analysis procedure. The results assessment methodology is discussed in chapters 2, 3, and 4, and chapters 5, 6 and 7 describe the process of adding a complimentary sensitivity analysis.

The second chapter of this document introduces the detailed MCNP model of the HI-STORM 100 used fuel cask. This model is used to demonstrate the results analysis methodology. The results of the simulated interior neutron flux are shown and features are identified in this chapter. A feature is defined in more depth in chapter 2.

Chapter 3 introduces the various analytical models used in this work. The neutron transport equation is derived and then reduced through application of assumptions and simplifications. From a reduced form of the neutron transport equation, the multigroup discrete ordinates equation and diffusion approximation are developed. The two equations form the basis of the analytical modeling used in this work.

The results assessment methodology is demonstrated in chapter 4. This chapter discusses why each analytical model is chosen as well as how each reduced complexity computational model is developed. After describing how the models are determined, each previously identified feature of the interior neutron flux is analyzed using the results assessment methodology.

Chapter 5 provides an overview into sensitivity analysis using the FSAP. A record of previous sensitivity analysis techniques is also provided. This chapter also introduces the sensitivity analysis process which will be used for the computational models in MCNP.

669 Chapter 6 provides foundational theory of the FSAP.

670 Comparisons between the FSAP analysis on analytical models and MCNP results are
671 discussed in Chapter 7.

672 The last chapter includes final thoughts and conclusion regarding the work.

673 Recommendations for future work are also provided in chapter 8.

CHAPTER 2

DISCUSSION OF MAIN PROBLEM

Dry storage casks provide protection, shielding, security, and cooling for used nuclear fuel which has spent at least one year in a spent fuel pool [33]. Shielding is especially important as used nuclear fuel is highly radioactive after being removed from a reactor and shielding is required to protect civilians, radiation plant workers, and the environment. The storage of used nuclear fuel has become a challenge in the United States since there is no long-term storage location. Instead, used nuclear fuel is stored in dry storage casks at the facility where it was generated. These casks are designed to 1) shield harmful radiation generated by the used nuclear fuel, 2) conduct decay heat away from fuel rods to prevent damage to the fuel and cladding, 3) protect spent nuclear fuel from environmental damage and other hazards, and 4) prevent proliferation of nuclear materials. Large efforts have been made in studying and designing casks to accomplish these challenges. While each function is imperative in analyzing the efficacy of a spent fuel cask, this work is only concerned with the radiation shielding capabilities of a Holtec International HI-STORM 100 spent fuel canister system.

Figure 2-1 is a diagram of the HI-STORM 100 spent fuel canister system partially loaded into an overpack of the same name. These two components together, the canister and overpack, will be referred to as a spent fuel cask. The HI-STORM 100 canister system is chosen as it is the most common used fuel storage system in the United States (750 canisters have been loaded before 2017) [4]. The overpack consists of two parts: a cylindrical dual material structure welded to a baseplate and a dual material removable lid. Both parts of the overpack use a combination of concrete and carbon steel to shield radiation, protect fuel, and prevent proliferation of nuclear material. Four vents are located at both the top and bottom of the overpack. These vents allow air to circulate between the overpack and MPC, removing heat caused by decaying isotopes in the spent fuel. Spent fuel rods are stored in the MPC, the central cylinder in Fig. 2-1. Figure 2-2 is

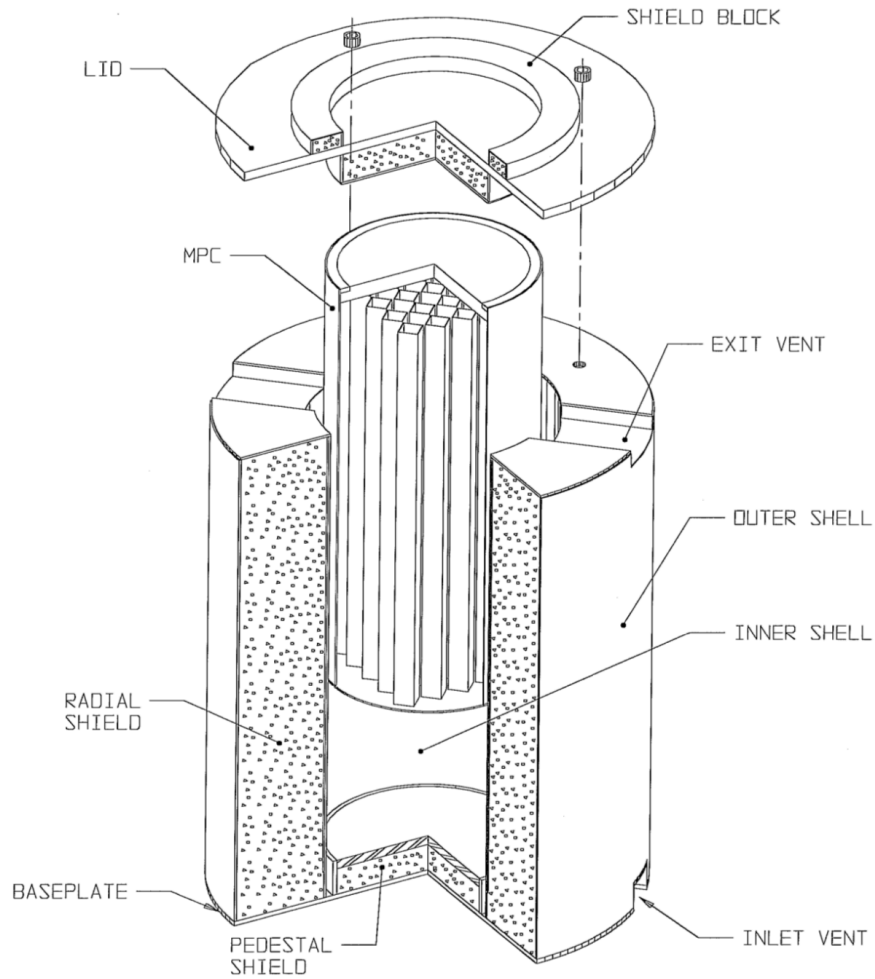


Figure 2-1. The Holtec HI-STORM 100 spent fuel cask system is designed to protect fuel, transfer decay heat to the environment, prevent proliferation of nuclear material, and attenuate radiation [1]. The MPC is seen partially inserted into the steel and concrete overpack. Current designs of the HI-STORM 100 do not use the inner shell and, therefore, the inner shield is not modeled in MCNP.

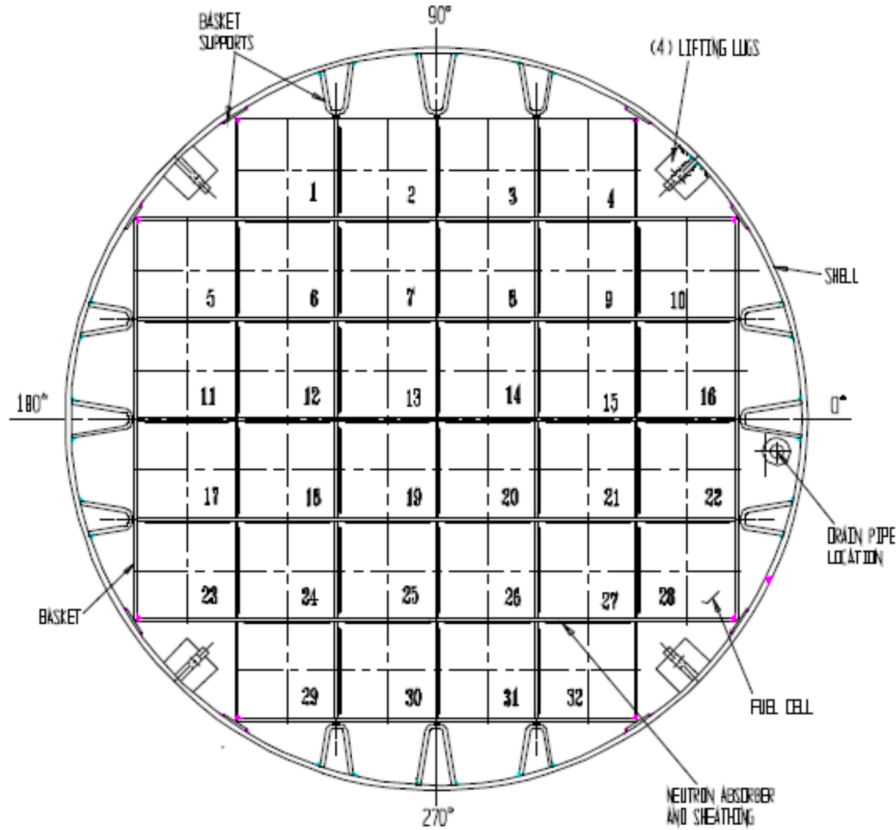


Figure 2-2. A cross section view of the multi-purpose canister. While there are multiple designs which accommodate different amounts of fuel, the MPC-32 is chosen for this work [1]. The MPC-32 is capable of holding 32 fuel bundles, one bundle in each square lattice element. The fuel basket and cylindrical wall of the MPC are made using stainless steel 304 and the canister is sealed by welding a baseplate to the bottom and a lid and closure ring to the top of the cylinder respectively.

the top-down cross section view of the MPC. Each cell in the honeycomb structure houses a single fuel bundle.

Power plant workers must be protected from the radiation produced by spent nuclear fuel rods, hence opening a sealed MPC is an expensive and potentially dangerous task. Therefore, alternative methods are being explored to ensure the content and integrity of fuel components which do not require opening a cask. A sample of these techniques includes neutron spectroscopy, deduction of interior structure based on exterior dose rates, and neutron based computer tomography which were previously discussed in detail

in Section 1.3. Each of these techniques relies on simulations using various radiation source definitions, virtual detectors, and simulated cask designs to determine specific quantities related to the neutron flux within the spent fuel cask. The key metric of this work is the interior neutron flux spatial distribution of the HI-STORM 100 spent fuel cask, as this quantity is shared among research in spent fuel casks. Clearly, simulation tools have become an important part of investigating the efficacy of a nondestructive evaluation technique, and ensuring the accuracy of these results is even more important since experimental data associated with the techniques is limited.

2.1 Description of Detailed Model

The MPC and overpack are modeled using the MCNP simulation code to determine the simulated interior neutron flux spatial distribution averaged over the height of the cask as a function of radial distance from the centerline. Figures 2-3 and 2-4 show, respectively, a side view and cross section of the cask geometry simulated in MCNP. This model is called the “detailed model” throughout this work and models the geometry of the cask down to the individual fuel rod level. Each fuel rod acts as a source term for neutrons produced from spontaneous fission and (α, n) reactions.

Figure 2-5 shows a single fuel cell cross section from the detailed model. The fuel cell contains two neutron absorbing pads composed of boron-carbide and aluminum, 264 fuel rods with zircalloy cladding and 25 water rods representing instrumentation. Fuel rod composition is determined using data from the Next Generation Safeguards Initiative which analyzed the composition of Westinghouse 17x17 fuel bundles with various degrees of initial ^{235}U enrichment and burn-up values [34]. This work investigates fuel with an initial enrichment of 3% ^{235}U and a burn-up value of 30 GWd/MTU. The composition of each individual fuel rod is unique, since fission fragment distribution is probabilistic, which introduces variance in the local neutron flux. These variations in fuel rod composition could influence the flux and potentially hide salient physics. Identifying and explaining salient physics is a goal of this work. Therefore, an average fuel rod composition is

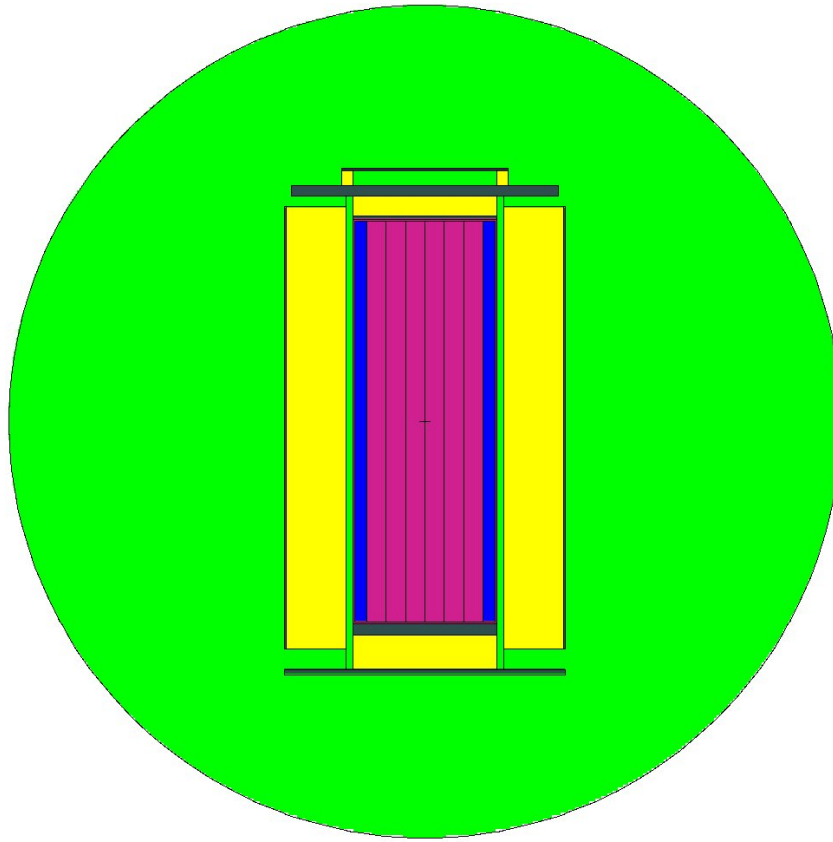


Figure 2-3. The side view of the HI-STORM 100 spent fuel cask (canister and overpack) modeled in MCNP. This is referred to as the detailed model.

determined based on the mass of each isotope present in a single spent fuel bundle in order to more clearly investigate the effects of geometry, detail, and non-fuel materials without influence from loading patterns of specific fuel rods.

The associated intrinsic neutron source is included via an MCNP neutron source definition. This definition is found using the ORIGEN-S 0-dimensional irradiation and decay code supplied with the SCALE package from Oak Ridge National Laboratory [35]. The neutron energy spectrum associated with the intrinsic source is shown in Fig. 2-6. The source spectrum results from spontaneous fission of isotopes in the fuel (such as ^{252}Cf) and (α, n) reactions occurring in the irradiated fuel.

Fig. 2-7 depicts the height-averaged scalar neutron flux as a function of radial position within the HI-STORM 100 spent fuel cask predicted using MCNP. The color

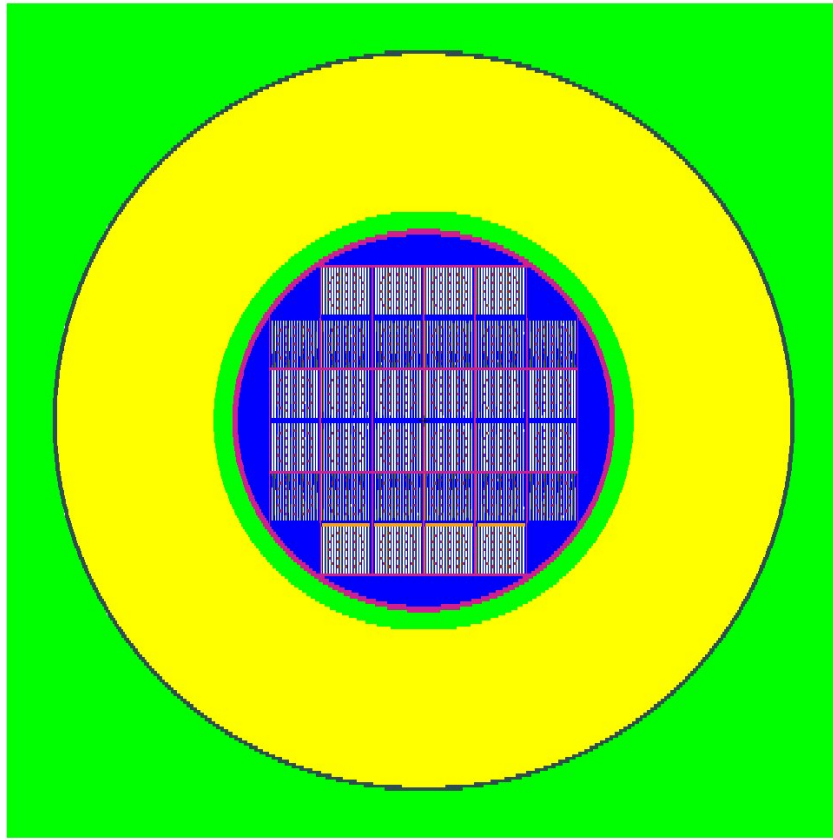


Figure 2-4. The top view of the HI-STORM 100 spent fuel cask modeled in MCNP. This view shows the fuel arrangement of the detailed model. This image shows the extent of geometric details which range from millimeters to meters.

of the line is related to the material through which the neutron flux is being simulated:
 fuel is green (the entire area interior to the MPC is considered the fuel region), MPC is
 blue, air is yellow, concrete is red, and carbon steel is black. The vertical lines designate
 interfaces between material boundaries; green is the interface between the fuel region and
 MPC, blue is the interface between the MPC and dry air, yellow is the interface between
 air and the concrete annulus, red is the interface between concrete and carbon steel,
 and black is exterior face of the cast. Figure 2-7 shows about half (54%) of the neutron
 flux is attenuated in the fuel region, and the concrete further reduces the flux by 39%.
 This result is intuitively sensible: the fuel region is comparatively dense and contains
 neutron-absorbing materials (e.g., boron), while the thick concrete overpack region is

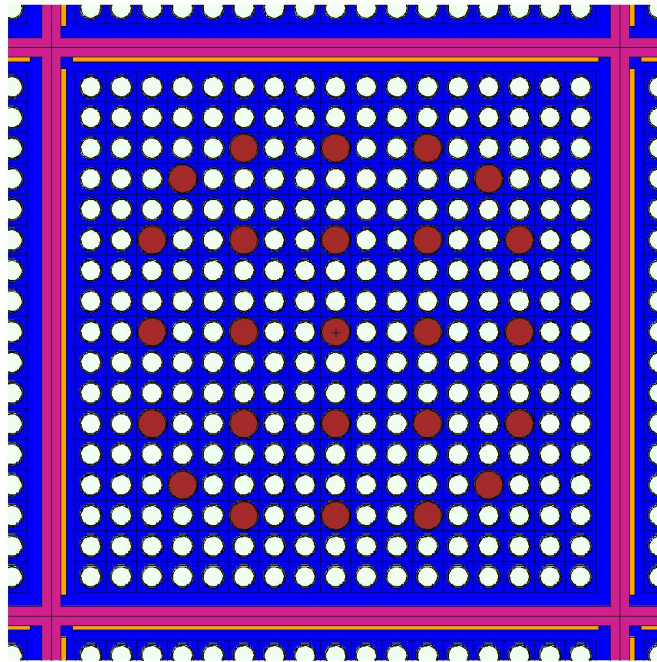


Figure 2-5. The zoomed in image of a single fuel cell cross section in the detailed model. There are neutron absorbing pads (orange rectangles) placed along the interior left and upper faces of the fuel basket (pink regions). Fuel rods (small white circles) include a fuel region, helium gap, and cladding, the helium gap and cladding are not visible in the figure. The larger red circles are the cross sectional view of water cylinders which represent instruments used for monitoring the safety of the HI-STORM 100 spent fuel cask system.

composed principally of highly thermalizing isotopes (e.g., hydrogen). Together, these processes are indicative of the observed dramatic reduction in neutron flux throughout the cask. However, advancing beyond intuition requires definitive answers to a variety of additional questions, namely:

- Are the results correct?
- Could a mistake have been made in the simulation input?
- Was an assumption made that neglected important physics?
- Does the problem include physics or exist in a physical regime outside the viability of the simulated tool used?

While corroborating simulation results with intuition is qualitatively valuable, quantitative or semi-quantitative assessments and their associated effects on confidence in simulation

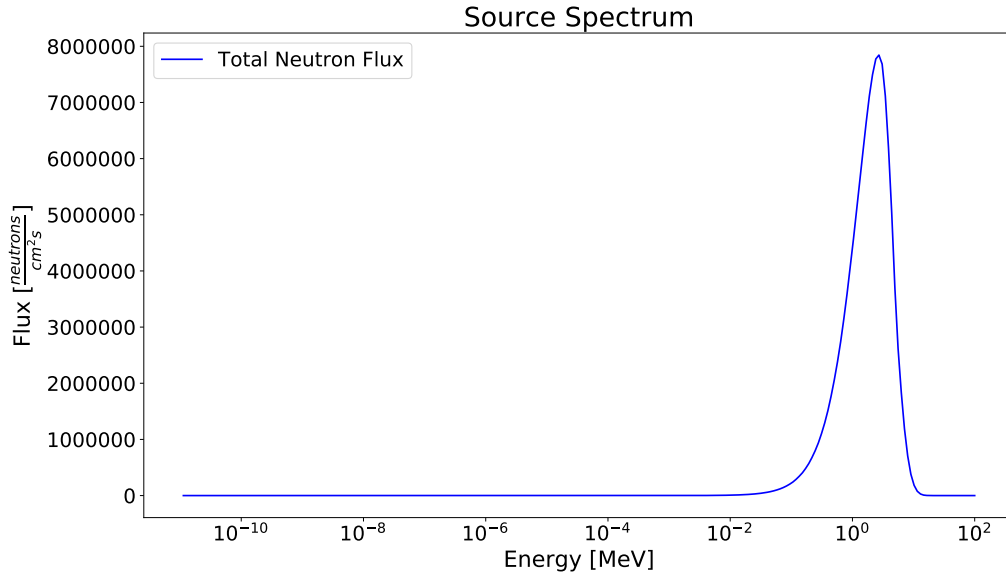


Figure 2-6. The source spectrum used in MCNP simulations. The spectrum is a result of spontaneous fission and (α, n) reactions.

results demands that the preceding questions be comprehensively addressed. The purpose of this work is to answer these questions by 1) identifying key features of the neutron flux spatial distribution as simulated in the detailed model, 2) developing simple physical models to determine the cause of each feature, and 3) gain confidence in the accuracy of the solution and inerrancy of the simulation process. In order to identify features in the neutron flux, each material region in the spent fuel cask is analyzed briefly.

2.2 Mathematical Model Choice

In the interest of constructing a complementary analytic representation of the neutron population behavior within the cask fuel region, inspection of the various features appearing in Figs. 2-1 and suggests several modeling simplifications.

- Aside from various isotope production and depletion processes featuring characteristic time scales spanning weeks to years, the spent fuel cask is essentially a static object. It is therefore assumed that the analytical representation of the cask is entirely time-independent (hereafter referred to as “static”).
- The neutron energy spectrum within the fuel region is essentially “fast”; that is, it principally exists at fission neutron energies (i.e., 1-2 MeV) with minimal

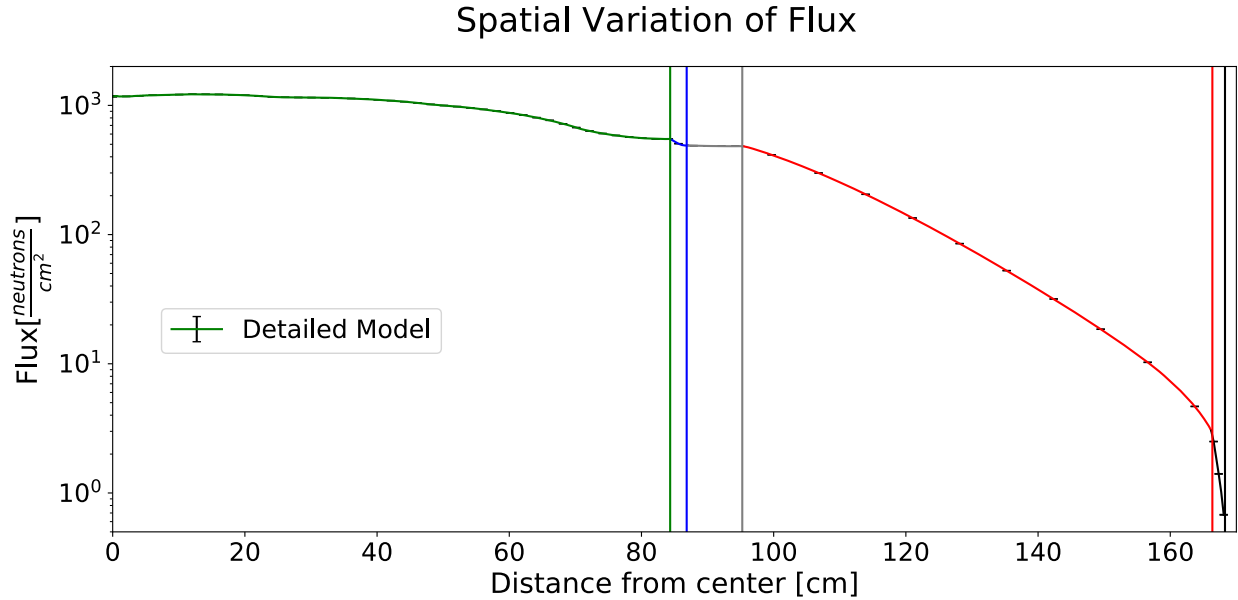


Figure 2-7. The interior neutron flux spatial distribution of the simulated HI-STORM 100 spent fuel cask. The vertical lines represent interfaces between material regions.

thermalization. As such, the analytical model used to characterize the cask fuel region is taken be approximately monoenergetic.

- Owing to the high hydrogen content in the concrete annulus, the energy spectrum can be represented with two energy groups (one thermal and one fast). An analytic model characterizing the neutron transport in concrete is assumed to require two energy groups.

As a consequence of these observations and associated simplifications, a static, monoenergetic balance law model is used to characterize the neutron population information within the cask fuel region.

2.2.1 Fuel Region

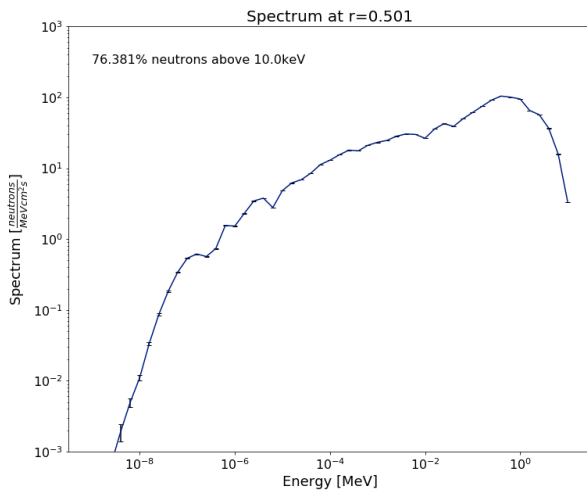
The fuel region of the HI-STORM 100 spent fuel cask features various materials including spent UO_2 nuclear fuel, a stainless steel basket, boron-containing neutron absorbing pads, and helium backfill. The geometric configuration of these materials is highly complex, as depicted in Fig. 2-1. Unfortunately, a single mathematical model capable of describing the neutron flux in the fuel region would not be tractable. Therefore, a simplified model must be developed using assumptions and approximations derived from

physics occurring in the model. In order to identify appropriate simplification, the energy spectrum and angular distribution of the neutron flux and cross section data of various materials are analyzed at various locations in the fuel region.

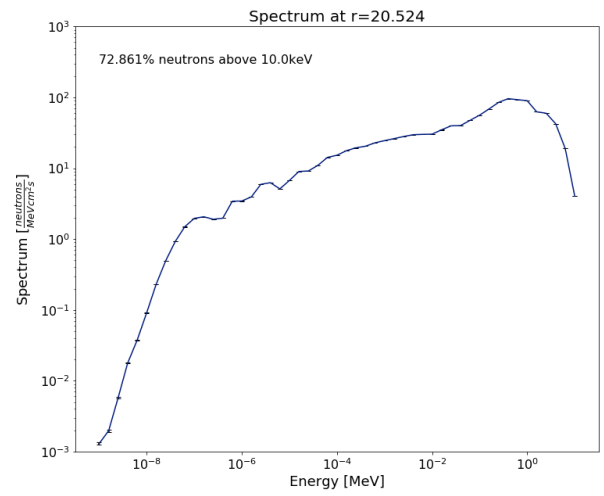
Figure 2-8 shows the energy spectrum of the neutron flux throughout the spent fuel cask. These plots show the neutron flux has little variation throughout the fuel region. This is a result of evenly distributing fuel rods through the fuel region. Further, the lack of thermalizing materials in the fuel cask means there is little change in the neutron spectrum. Therefore, it can be assumed that energy dependence of the neutrons can be handled uniformly throughout the fuel region. This is a very helpful assumption that allows for uniform treatment of material properties throughout the fuel region with respect to energy. Unfortunately, there have been no assumptions concerning how to handle neutron energy-dependence at this point, (e.g., is a monoenergetic method appropriate, or will a different model be require?).

Analysis of the energy spectrum will determine how to best handle energy-dependence. The percent of neutrons above 10keV varies between $\sim 78\%$ at inner radius values to $\sim 71\%$ at the edge of the fuel region as shown in Figs. 2-8a-2-8h. A monoenergetic handling of the energy-dependence can be assumed since the majority of neutrons have energies between 10keV and 10MeV, using an appropriate group weighting spectrum described by Bell and Glasstone [36]. After choosing a method for handling energy-dependence, it becomes necessary to determine a method for handling directional-dependence of the neutron flux.

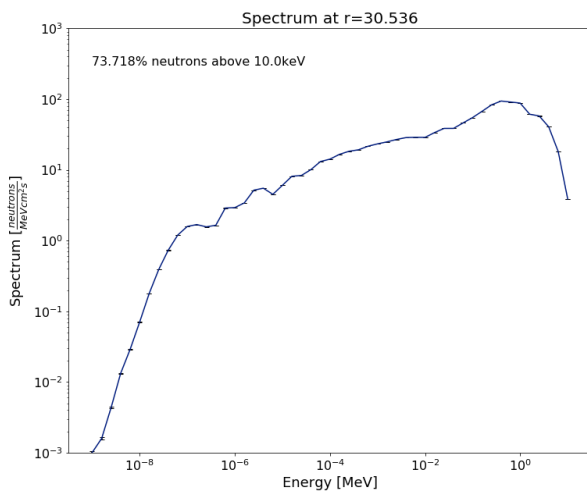
Figure 2-9 shows the angular distribution of the flux 0.5cm from the centerline (Fig. 2-9a) and at the edge of the fuel region (Fig. 2-9b). The angular distribution was tallied at these locations to capture the two extents of the angular flux. A perfectly isotropic flux would be a horizontal line with zero slope. If half of the neutron population is traveling in either direction (inward and outward), then the neutron flux can be approximated as isotropic with the understanding that deviations from isotropy will lead to errors in



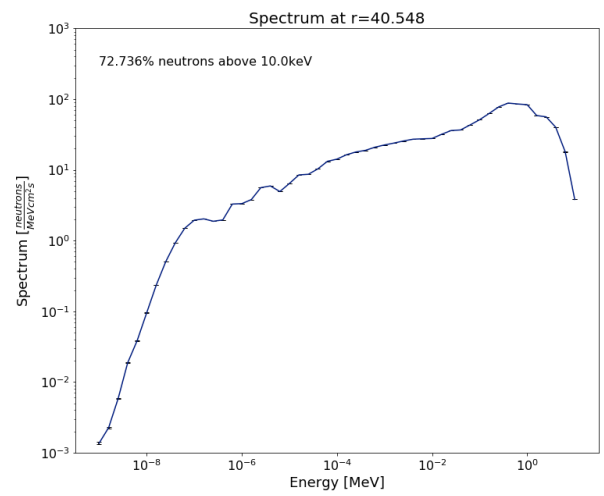
(a)



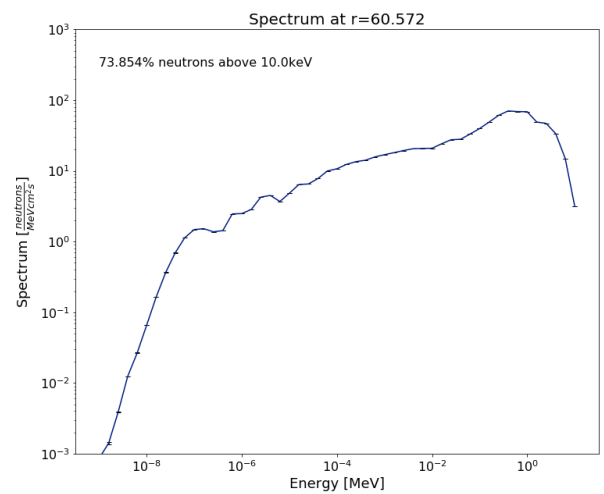
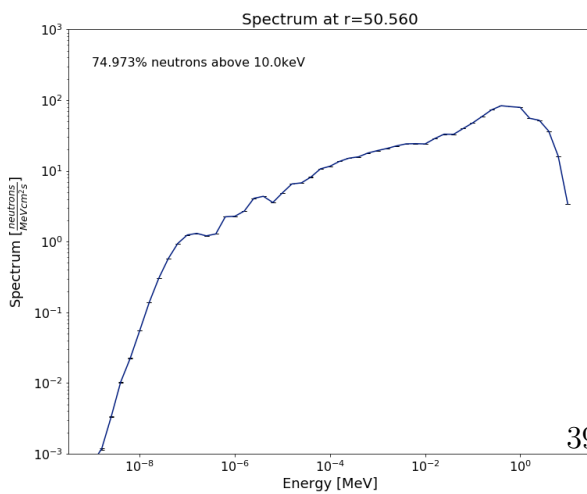
(b)

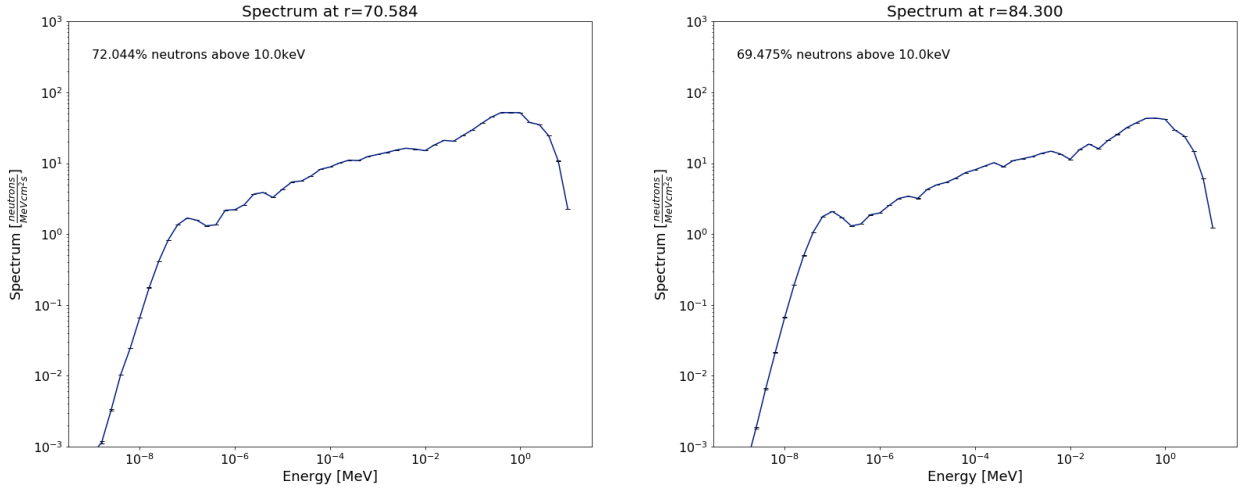


(c)



(d)



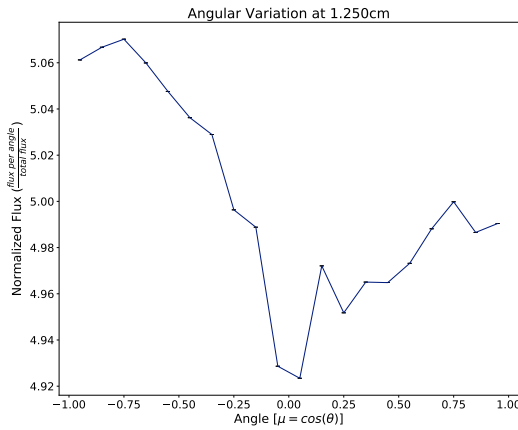


(g)

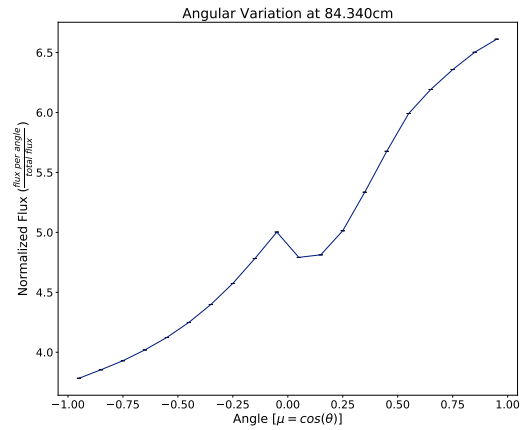
(h)

Figure 2-8. Energy spectrum of the neutron flux at various locations in the MPC where fuel rods are stored.

the results. Figure 2-9a shows the neutron flux is slightly inward-peaked 0.5cm from the centerline with 50.278% of all neutrons traveling toward the centerline. This indicates the flux can be approximated as isotropic near the centerline, a perfectly isotropic flux would have 50% of neutrons scattering toward the center of the fuel region. The flux at the outer edge has an outward peaked flux as shown in Fig. 2-9b. This is because the neutron population density is high in the fuel region, since the source of neutrons is in the fuel region, and neutrons are diffusing, or leaking, out of the fuel region where the neutron population density is lower. The percent of neutrons traveling outward from the fuel region 57.290% at the surface of the fuel region. While, the angular distribution in Fig. 2-9b shows the neutrons are slightly forward peaked, the angular distribution of the neutron flux deviates from isotropic by only $\sim 7\%$, thus, can be approximated as isotropic with the understanding that this approximation may lead to some disagreement between analytic and computational results.



(a)



(b)

Figure 2-9. Angular distribution of the neutron flux at 2-9a 0.50 cm and at 2-9b the inner surface of the MPC (84.34 cm) from the centerline of the fuel cask.

Figure 2-10 shows the mean-free-path (MFP) of each of the materials in the fuel region. The MFP is the average distance between neutron interactions in a material. Figure 2-10 shows the MFP in the fuel (blue), cladding (orange), helium (green), stainless steel (red), and neutron absorbing material (purple). The source flux is also shown in grey to identify which energy regions are most important (i.e., energy regions where the source flux is higher are more important). Assessing the MFP of each material helps to identify other assumptions and approximations that aid in determine the appropriate mathematical model to represent the neutron flux. The MFP of helium is about 1km where the source flux is most intense near 1MeV. The thickest region of helium occurs between the fuel cells and edge of the fuel region and is on the order of 10cm thick. The MFP is approximately two orders of magnitude larger, meaning there will be a negligible number of neutrons interacting in helium. The first material assumption is that helium outside of the fuel cells can be neglected. The remaining materials have a MFP of approximately 1cm at 1MeV. These materials show up in the fuel region on the same order, therefore, the remaining materials cannot be neglected. However, since these materials are evenly distributed (i.e., the materials exist throughout the fuel region and

not just at a single location) and since the remaining materials have similar MFP's, a homogenization technique can be used to approximate the geometry in the fuel region.

A cylinder shaped homogenous fuel material is made based on the weight ratio of each material in the fuel region. The volume of the homogenous cylinder of fuel material is determined to preserve the volume from the 32 original fuel cells, and the radius of the cylinder is approximately 75cm. The volume around the cylinder of homogenous fuel is treated as a vacuum in the mathematical model. The radius of the homogenized fuel is about two orders of magnitude greater than the MFP of the materials used in the fuel region (e.g., $\sim 100\text{cm}$ radius of fuel $\gg \sim 1\text{cm}$ MFP). Hence, the diffusion equation is an appropriate model since the fuel material is much thicker than the neutron's MFP. Therefore, the monoenergetic diffusion equation is an appropriate mathematical model to represent the neutron flux in the fuel region, given the previous identified assumptions and approximations derived from physical properties of materials in the fuel region.

A monoenergetic diffusion approximation is an appropriate choice of an analytic model for the fuel region, however, that may not be the case for other materials in the cask. It is important to identify how the flux behaves in the remaining materials of the fuel cask and to identify appropriate models.

2.2.2 Stainless Steel MPC

The MPC encompasses the fuel area in a 2.5cm thick stainless steel 304 cylindrical container. Figure 2-11 shows the MFP in stainless steel 304. The most important thing to notice from the figure is that the MFP is on a similar order of magnitude as the thickness of the MPC. The diffusion equation is not an appropriate model when a material's thickness is fewer than a couple MFP's thick. Therefore, the diffusion approximation is unlikely to be an appropriate mathematical model. Instead, the multigroup discrete ordinates equation is a better approximation in this situation.

The number of energy groups and angles required to adequately model neutron transport in the stainless steel is still needed. Analyzing the energy spectrum at the

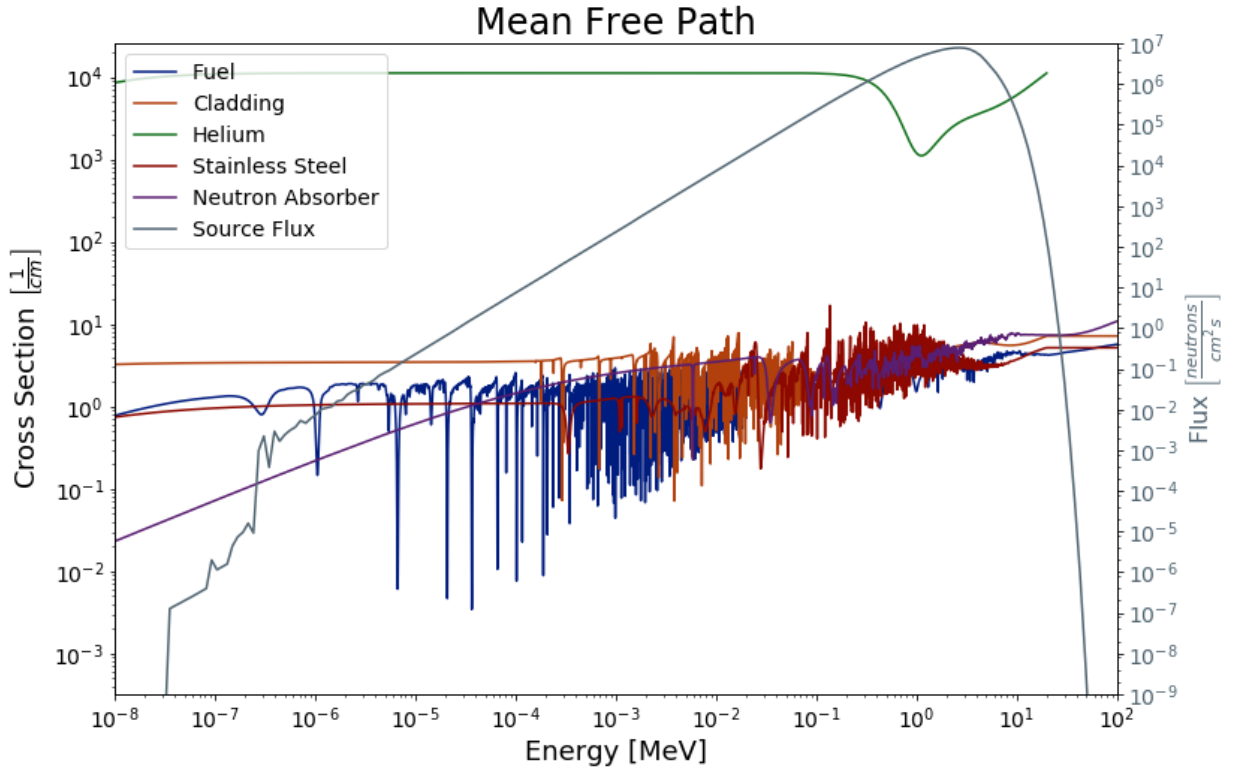


Figure 2-10. The mean-free-path, or distance between interactions, of the materials in the fuel region. The source flux is provided in order to identify energy ranges of greater importance.

interior and exterior surface of the MPC aid in finding an appropriate number of energy groups. Figure 2-12 shows the neutron energy spectrum at the interior surface (Fig. 2-12a) and exterior surface (Fig. 2-12b) of the MPC. At the interior surface of the MPC, the neutron flux is $\sim 70\%$ above 10keV and a single energy model would be appropriate. This would be preferable since the group structure in the MPC would match the energy group boundaries in the fuel region. However, the number of slow neutrons increases though the thickness of the MPC, and Fig. 2-12b shows that $\sim 59\%$ of neutrons are above 10keV. Hence, a two group analytic model is preferable.

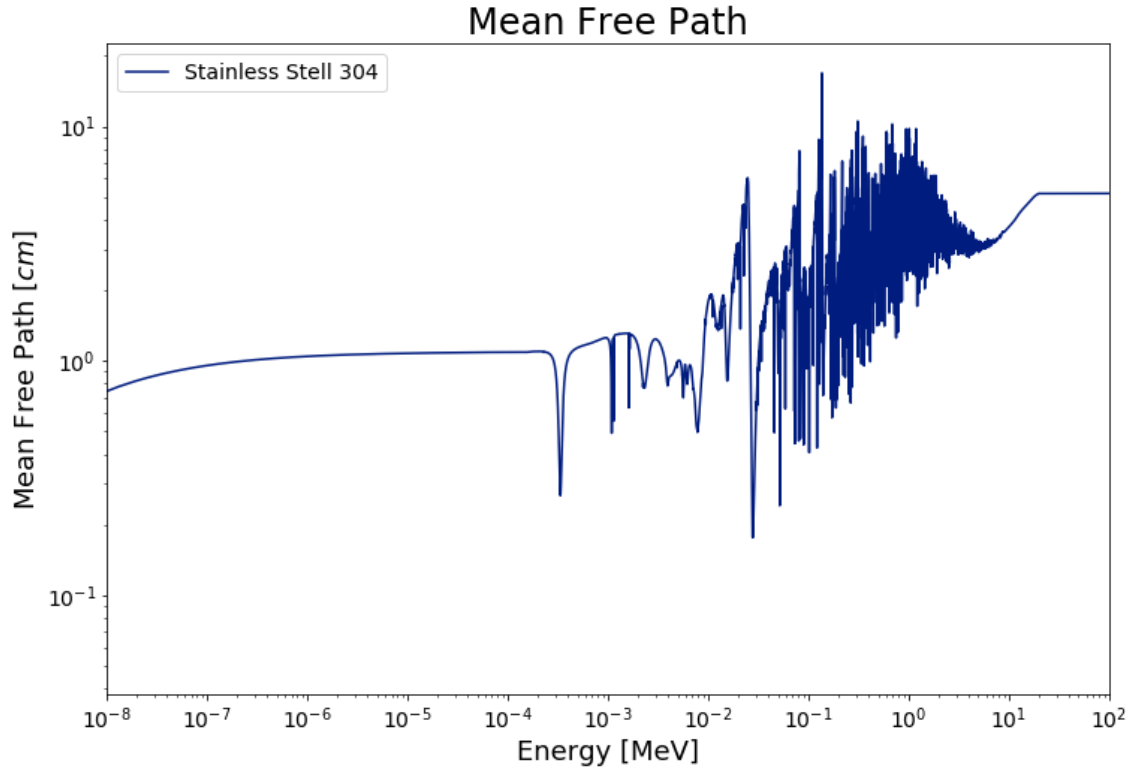
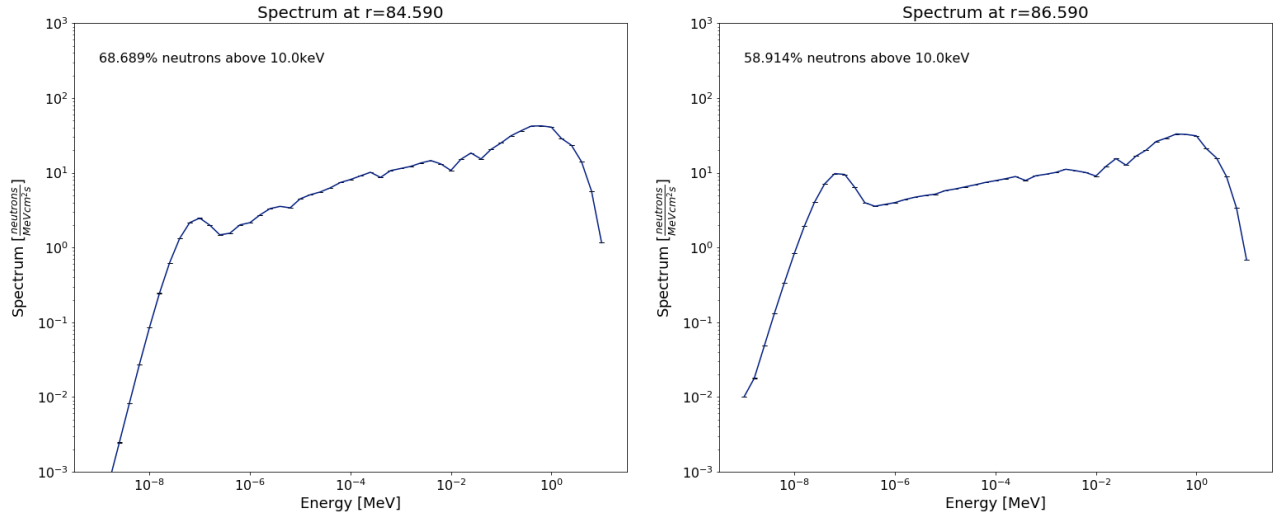


Figure 2-11. The mean-free-path of neutrons in stainless steel 304.

Analysis of the angular distribution (Fig. 2-13) helps to determine the number of angles to use in the multigroup discrete ordinates approximation. Figure 2-13a is the angular distribution of the flux at the interior surface of the MPC. Approximately 57% of the neutrons are forward scattering at this point in the MPC and Fig. 2-13b shows that the number of forward scattered neutrons is relatively similar ($\sim 56\%$). In the fuel region, the flux is considered isotropic even though over half of the neutrons are traveling away from the centerline near the outer surface of the cask. This is an acceptable approximation since there exists location in the fuel region that are closer to isotropic. However, the neutron flux in the stainless steel is only forward-pointed, which alludes to using two angles to approximate the neutron flux. Finally, the multigroup discrete



(a)

(b)

Figure 2-12. The neutron energy spectrum at a) 84.6cm and b) 86.6cm in the stainless steel MPC.

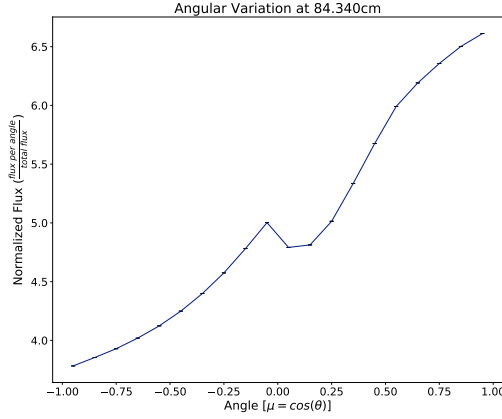
ordinates approximation with two energy groups and two angles is chosen to model neutron transport in the stainless steel.

2.2.3 Dry Air Gap

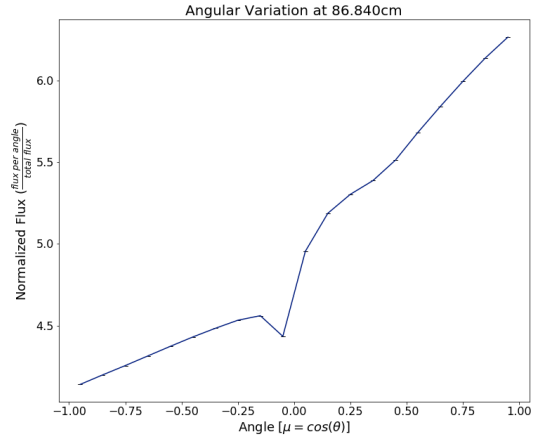
Surrounding the MPC is a gap of dry air for heat removal from the fuel. Figure 2-14 shows the mean free path of neutrons in dry air. The MFP is two orders of magnitude larger than the thickness of the air gap (~ 10 cm). Meaning, the air gap can be treated as a vacuum and there is no need for a mathematical model in this region.

2.2.4 Concrete Annulus

The 71.12cm thick concrete annulus provides nearly half the neutron shielding capabilities in the spent fuel cask due to scattering on hydrogen. Following a similar method as before, the MFP of neutrons in concrete are investigated. Concrete, being a thermalizing material, is expected to change the neutron energy spectrum through down-scattering neutrons, so both fast and thermal energies need to be taken into account when analyzing Fig. 2-15. At higher energies, 1MeV, the concrete is about 7 MFP's thick.



(a)



(b)

Figure 2-13. The neutron angular distribution at the a) inner surface and b) outer surface of the MPC.

Therefore, diffusion may not be an appropriate model for these energies of neutrons. However, at lower energies, 1eV, the concrete is about 35 MFP's thick. At lower energies, the diffusion approximation is an appropriate model. Overall, analysis of Fig. 2-15 would indicate that a multigroup discrete ordinates approximation would be better suited as an analytic model in the entire concrete. Further investigation of the neutron energy spectrum and angular distribution will aid in solidifying a model choice.

The energy spectrum does change significantly over the thickness of the concrete annulus. Figure 2-16a shows the neutron energy spectrum at the inside surface of the concrete annulus. The neutron flux is $\sim 53\%$ above 10keV at the innermost surface of the concrete. The neutron flux is quickly thermalized and less than a third of the neutron flux is above 10keV after the neutrons have traveled ten centimeters into the concrete (Fig. 2-16b). At the exiting surface, less than 6% of the neutrons remain above 10keV as shown in Fig. 2-16h. The large change in neutron energies means more than one energy will be required to model transport in concrete. The shape of the flux shows the presence of two local maxima in the neutron spectrum that occur throughout the concrete region, one near

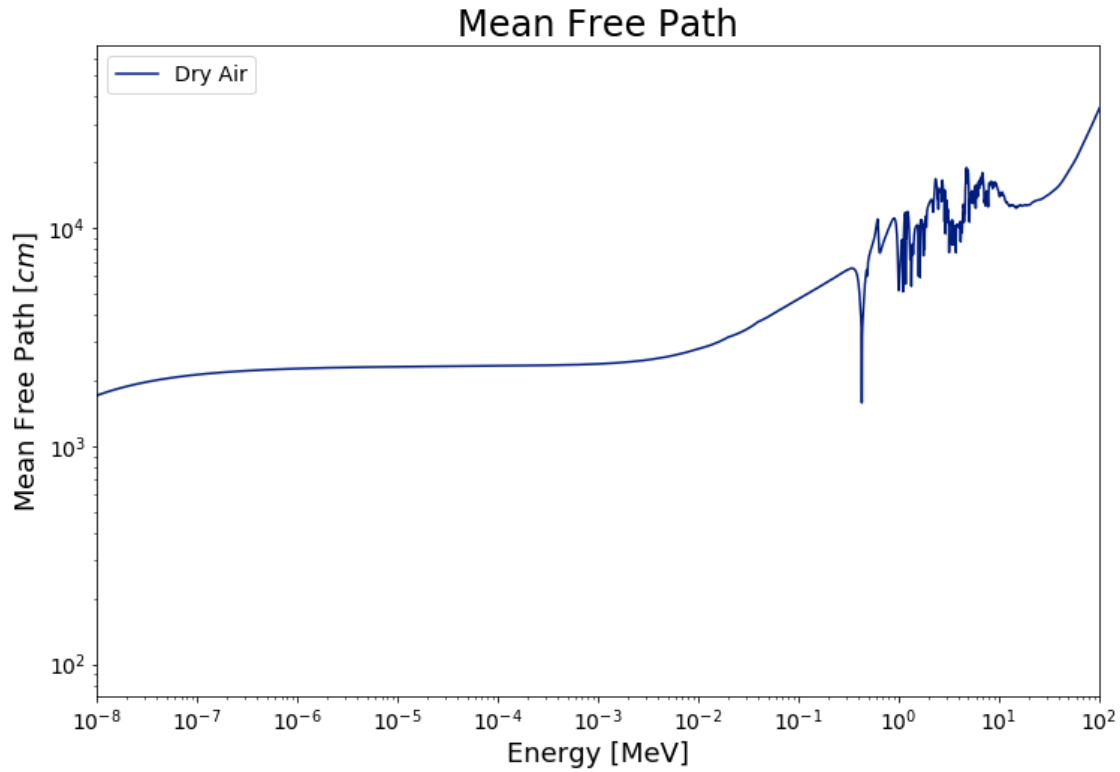


Figure 2-14. The mean-free-path of neutrons in the air gap. The low density of gaseous air lead to a high MFP. The air gap can be treated as a streaming region since the MFP is much larger than the thickness of the air gap.

1MeV and the other near 0.1eV. Therefore, a two energy group model is expected to be adequate. Analysis of the angular distribution will indicate the number of angles necessary for the multigroup discrete ordinates model.

Figure 2-17 show the angular distribution at the entering and exiting surfaces of the concrete annulus. Analysis of the angular distribution shows the neutron flux is forward-peaked with $\sim 55\%$ of the neutrons traveling outward at the inner surface of the concrete annulus. At the exiting surface, $\sim 68\%$ of the neutrons are traveling outward.

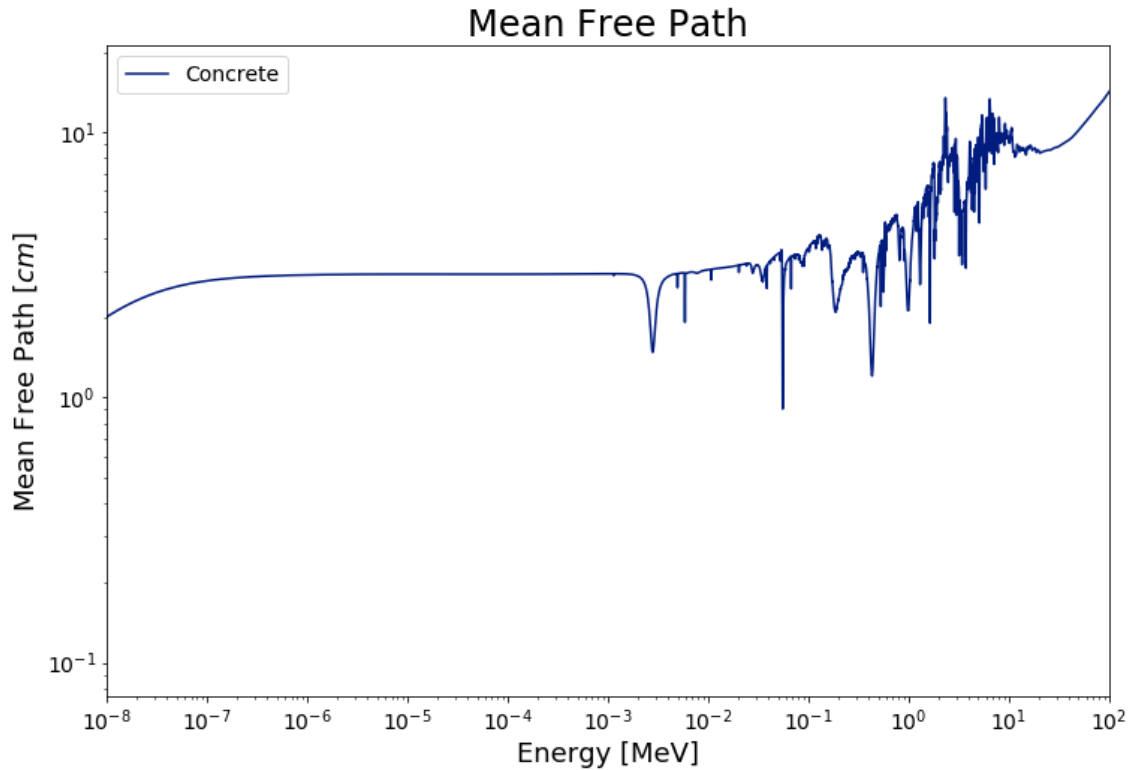


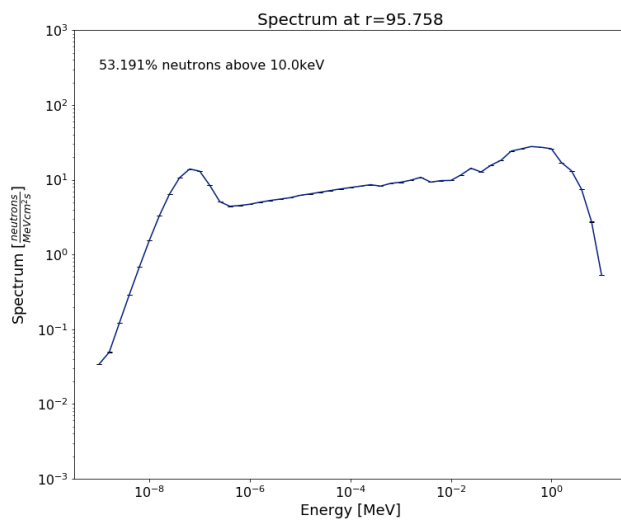
Figure 2-15. The mean-free-path of neutrons in the concrete annulus.

This results confirms the model choice of the multigroup discrete ordinates approximation.

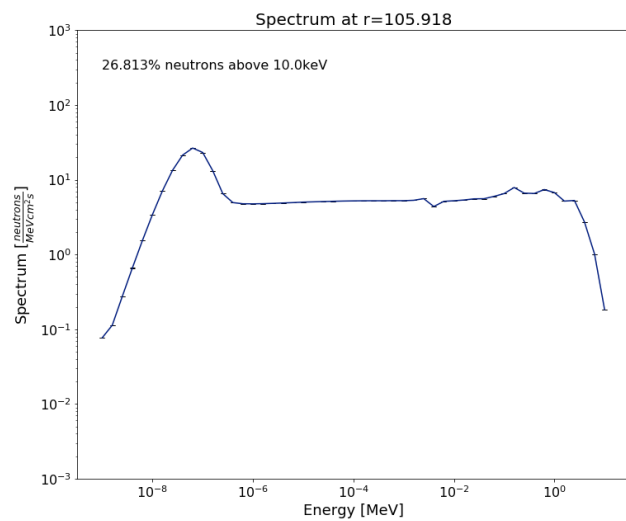
Further, two angles are adequate to describe the flux to a first-order-approximation.

2.2.5 Carbon Steel Outer Shell

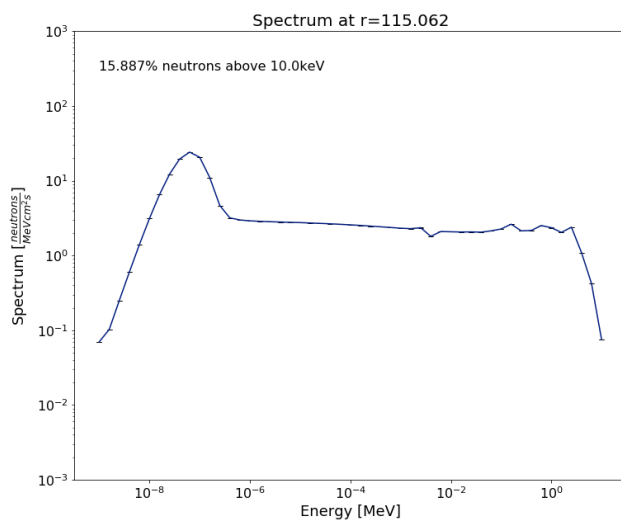
The 1.905cm thick carbon steel shell is the final material being analyzed in the spent fuel cask. Using a similar analysis as with previous materials, the MFP is compared to the thickness of the steel shell to aid in determining a mathematical model. Figure 2-18 shows the MFP of neutrons in carbon steel. The most probable energy of neutrons leaving entering the carbon steel shell is about 0.1MeV, shown in Fig. 2-16h. Using this information, the most probable MFP of neutrons in the carbon steel shell is ~ 1 cm. This is on the order of the magnitude of the carbon steel shell thickness. Therefore, the diffusion



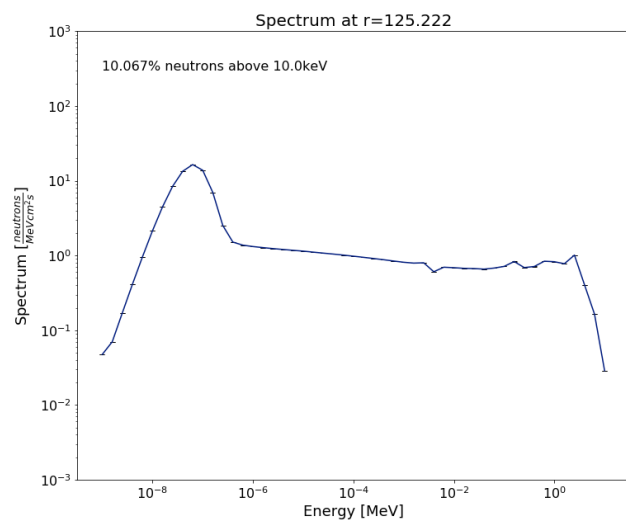
(a)



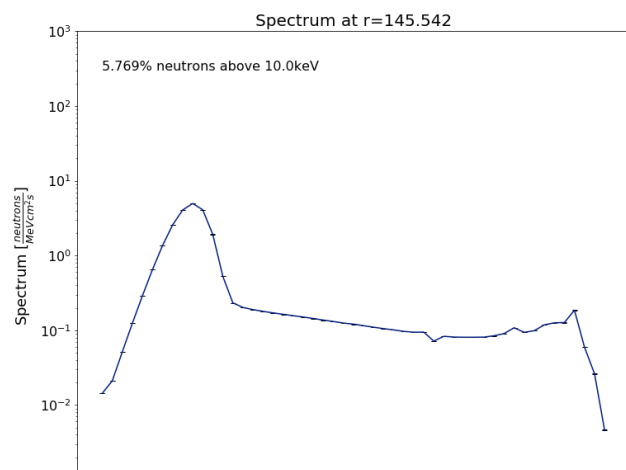
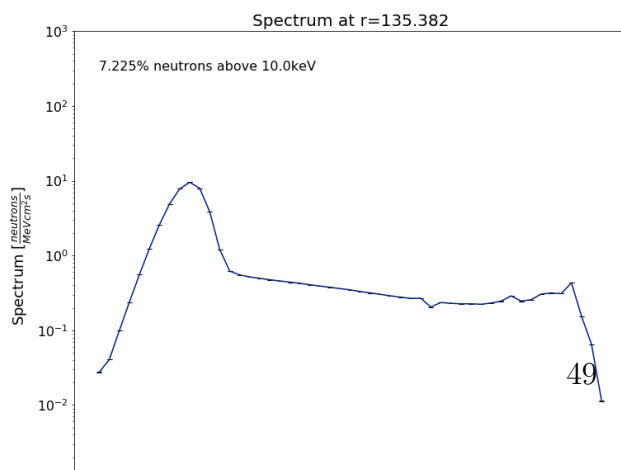
(b)

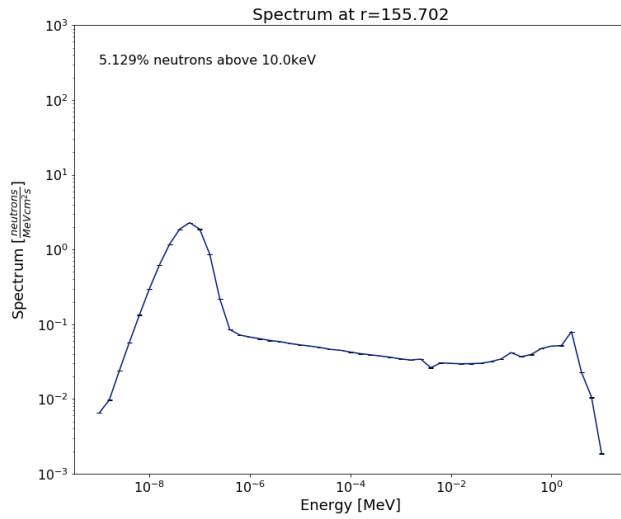


(c)

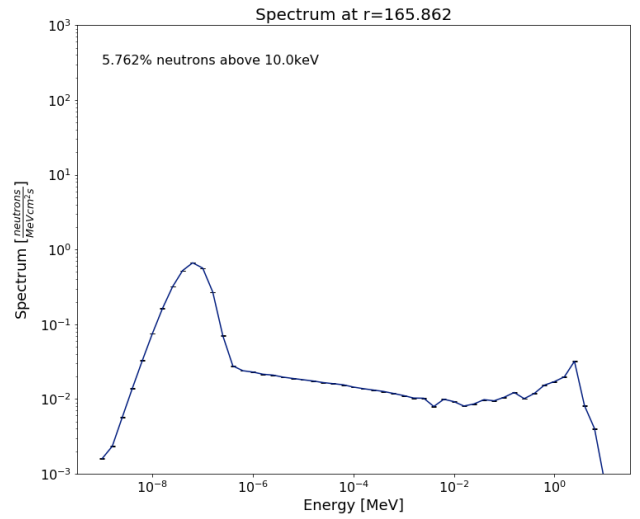


(d)



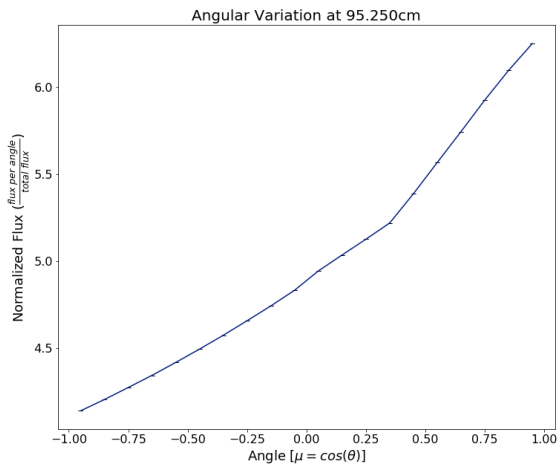


(g)

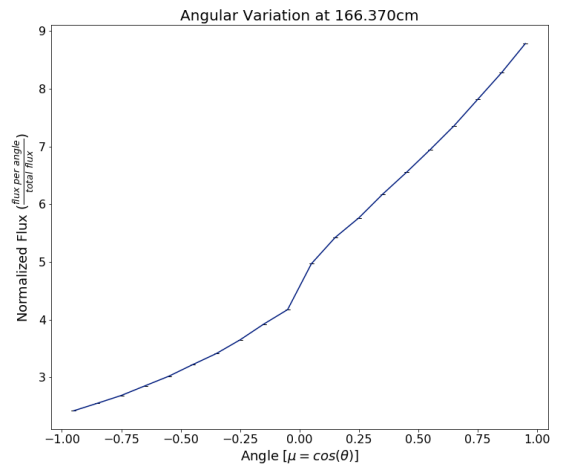


(h)

Figure 2-16. Energy spectrum of neutrons throughout the concrete annulus.



(a)



(b)

Figure 2-17. The angular distribution of the neutron flux at a) the inner surface (95.25cm) and the b) outer surface (166.37cm) of the concrete annulus.

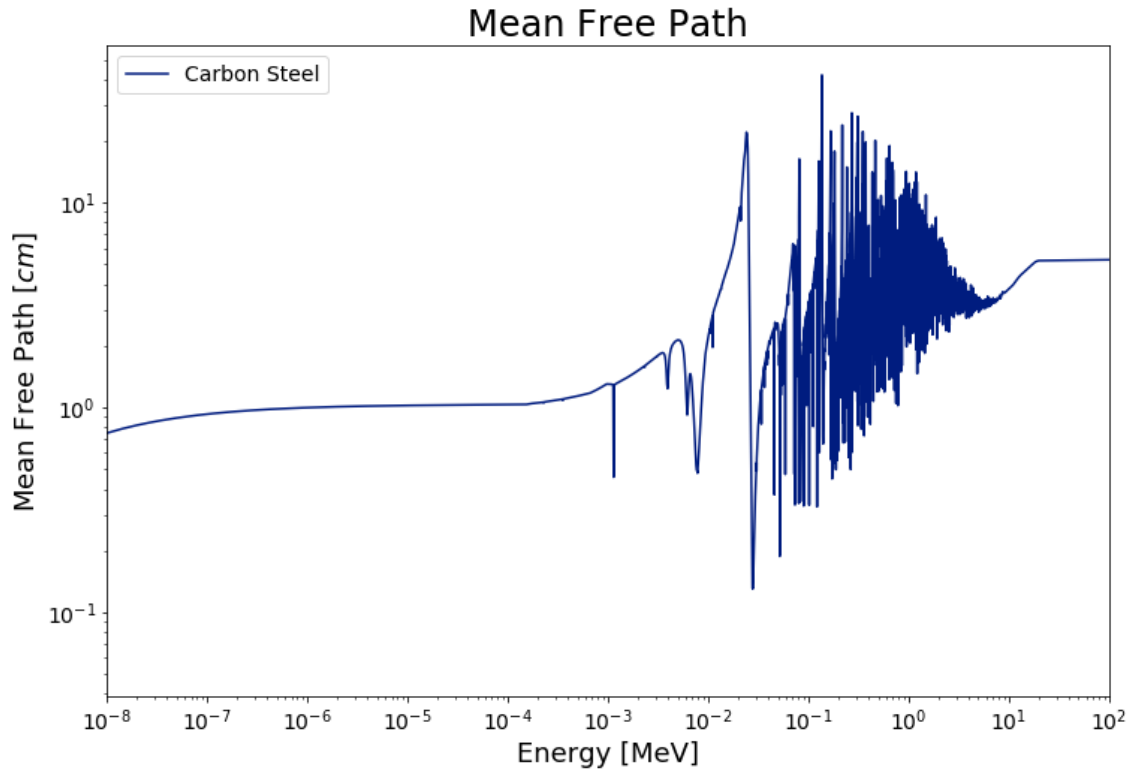
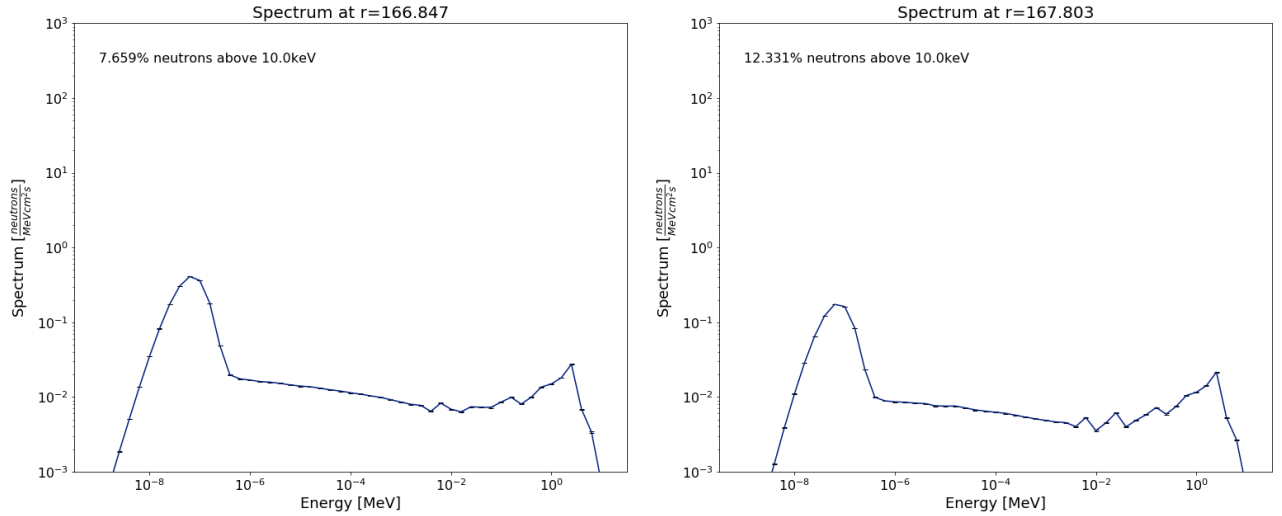


Figure 2-18. The mean-free-path of neutrons in carbon steel.

equation is likely a poor choice of mathematical model and the multigroup discrete ordinates equation is likely a better choice.

Figure 2-19 shows the energy spectrum at the inner surface (Fig. 2-19a) and outer surface (Fig. 2-19b). A small number of neutrons are bred in carbon steel resulting in the growing number of neutrons above 10keV. For this reason, two energy groups should be used to model the neutron flux in the carbon steel.

Finally, the angular distribution graphs of the neutron flux entering the carbon steel (Fig. 2-20a) and leaving the carbon steel shell (Fig. 2-20b) show the neutron flux is forward peaked. In fact, at the inner carbon steel surface $\sim 68\%$ of the flux is traveling outward and that fraction increases to $\sim 97\%$ of neutrons traveling outward at the exiting

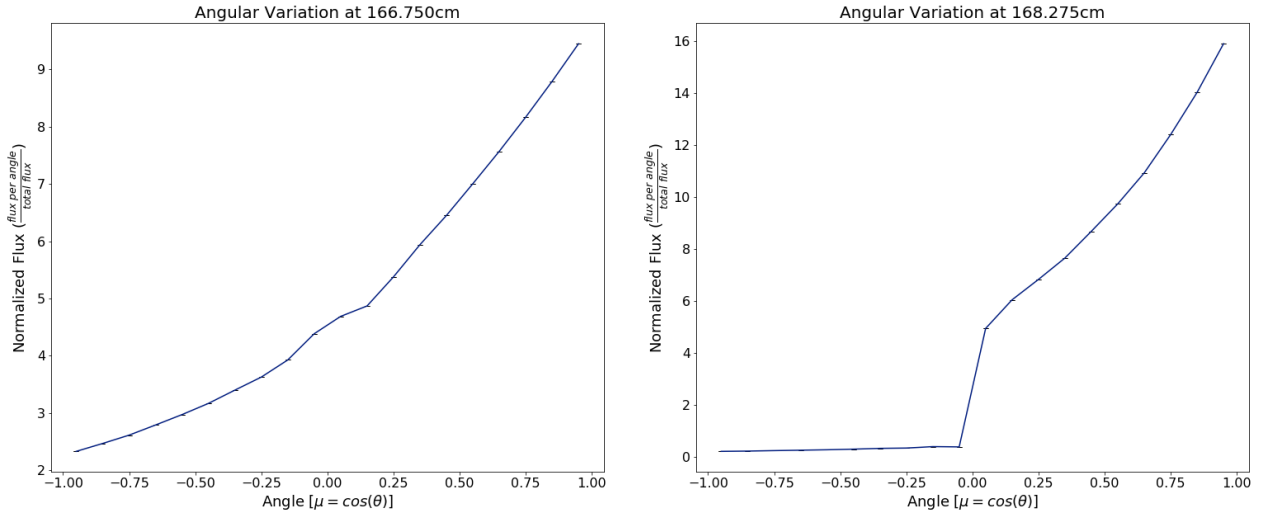


(a) (b)

Figure 2-19. The neutron energy spectrum near the a) inner surface (166.847cm) and the b) outer surface (167.803cm) of the carbon steel shell. The two peaks in each figure allude to a two energy group model.

surface of the cask. Nearly all of the neutrons are traveling away from the cask because the cask is placed in dry air. As shown previously, the MFP of neutrons in dry air is large, greater than 1km, resulting in a small number of neutrons returning to the cask after leaving. The small number of returning neutrons provides boundary condition information for the final model. Therefore, the outermost boundary of the spent fuel cask can be treated as non-reentrant. Further, two angular groups are capable of approximating the flux since the distribution is strongly preferential in a forward direction.

By no means are the previous choices in analytic models meant to be the most exhaustive means of describing the neutron flux in each material. Rather, choices were made in order to keep the models as simplistic as possible while capturing the physics of the spent fuel cask in an attempt to highlight inherent phenomenon in the problem. As will be seen during the sensitivity analysis portion of the work, even these simplistic models yield complex sensitivity results. Therefore, identifying any physical meaning using



(a) (b)

Figure 2-20. The angular distribution of the neutron flux at the a) inner surface and b) outer surface of the carbon steel shell. Since the flux is heavily forward-pointed, two directions can be used to model the flux.

the analytic models becomes challenging, if possible, even when using very simple models. While the mathematical models have been chosen, the geometry which each will be solved in has yet to be determined, which will be discussed in Chp. 3.

2.3 Identification of Features

“Features” are locations in the simulated neutron flux spatial distribution shown in Fig. 2-7 which appear to be the result of a physical process. Using a reduced complexity analytic or computational model to reproduce a feature yields two benefits: 1) the physical process that generates the feature in question is identified and, 2) confidence is gained in the accuracy of the simulation result. Confidence in the simulation result is gained when a feature is determined to be a result of an understood physical process. That is, the feature should exist in the problem, is being modeled correctly in the code, and is not a computational artifact. Ensuring agreement between simplified and complex models also corroborates the accuracy of the simulation input itself. Something as simple as inputting

an incorrect area or volume would not result in a fatal error message in MCNP, but would lead to incorrect neutron flux results. The process of reproducing features using simplified analytic and computational models provides an opportunity to identify errors in the simulation input and addressing these errors leads to increased confidence in the accuracy of a simulation.

There are five features discussed in this paper which are identified as:

1. The “flat” flux region (highlighted in Fig. 2-21): The flux in this region smoothly decreases by approximately 36% even though intuition suggests the flux should increase in the fuel pins and decrease in the space between fuel pins.
2. The abrupt level-off region (highlighted in Fig. 2-22): The flux only decreases $\sim 3\%$ over the region $65\text{ cm} \leq r \leq 84.1\text{ cm}$ from the cask centerline.
3. Periodic depressions (highlighted in Fig. 2-23): There is a $\sim 2\%$ reduction in the flux near 25 cm, 50 cm, and 75 cm from the cask centerline.
4. The asymmetric flux: Figure 2-24 is a density plot of the neutron flux when looking at a center slice of the cask from above. Figure 2-25 is a contour plot to better illustrate the neutron flux asymmetry present in Fig. 2-24. The neutron flux in the upper left section (above the diagonal line) of the plot is less than the neutron flux in the lower right section (below the diagonal line) of the image. This asymmetry is most obvious at the outer edge of the fuel region.
5. The concrete flux (Fig. 2-26): The concrete region provides the second-most significant reduction in the neutron flux within the cask. Identifying the processes which attenuate radiation in this region provides evidence the overpack was modeled correctly.

The remaining chapters will discuss how the results assessment methodology is used to identify the salient physics in each of the previously identified features, as well as, how confidence is gained in the simulation results of the detailed model through sensitivity analysis. However, the next chapter will provide an in-depth background on neutron

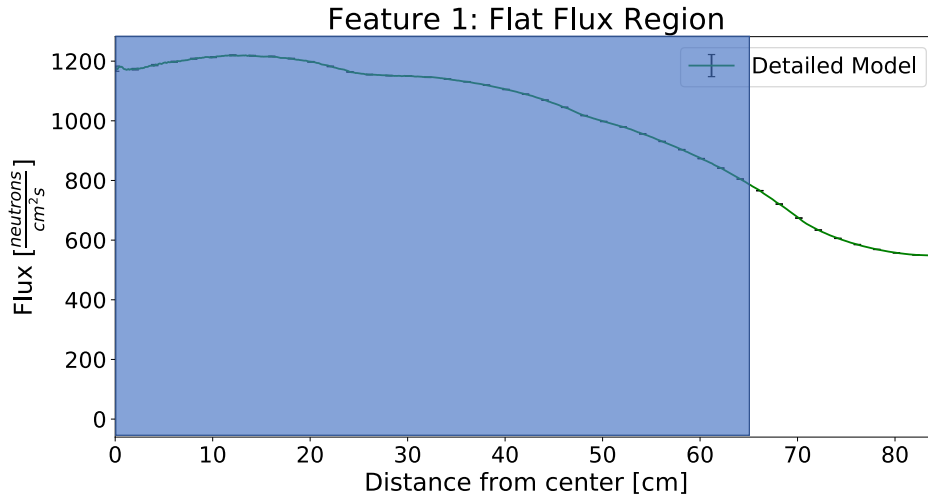


Figure 2-21. The neutron flux spatial distribution between the cask centerline and inner face of the MPC. The highlighted region is considered the flat flux region. This neutron flux is relatively flat and does not vary on the same order as the physical dimensions of materials in this region.

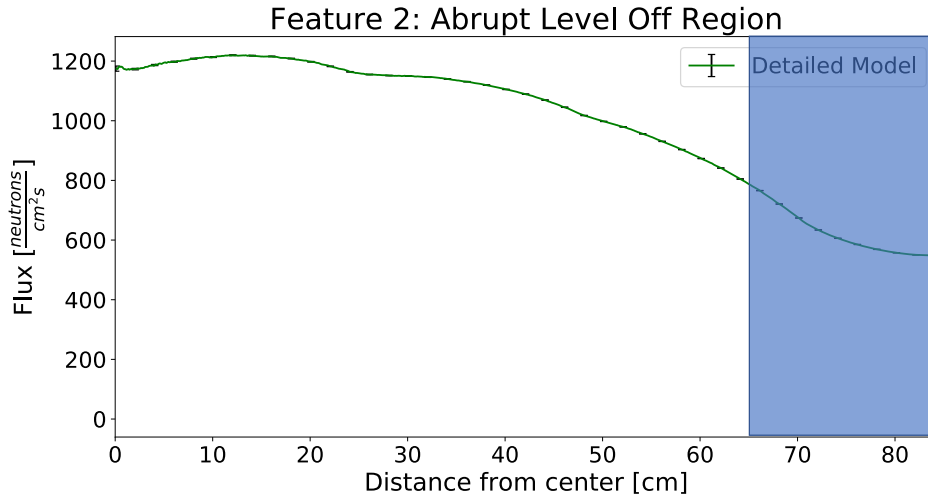


Figure 2-22. The flux stops decreasing and instead levels-off in the abrupt level-off region. The flux decreases less than 3% over the last ten centimeters before the interface between the fuel region and MPC.

1010 transport theory and the development of the analytic models which will be used in the
 1011 analysis before we can discuss the features further.

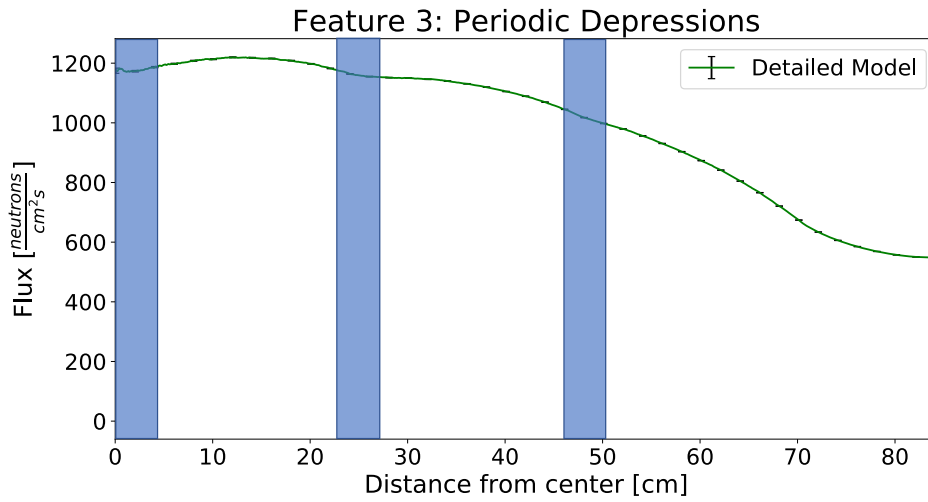


Figure 2-23. There are three depressions in the neutron flux spatial distribution located approximately 22 cm apart. The flux decreases about 2% at each depression.

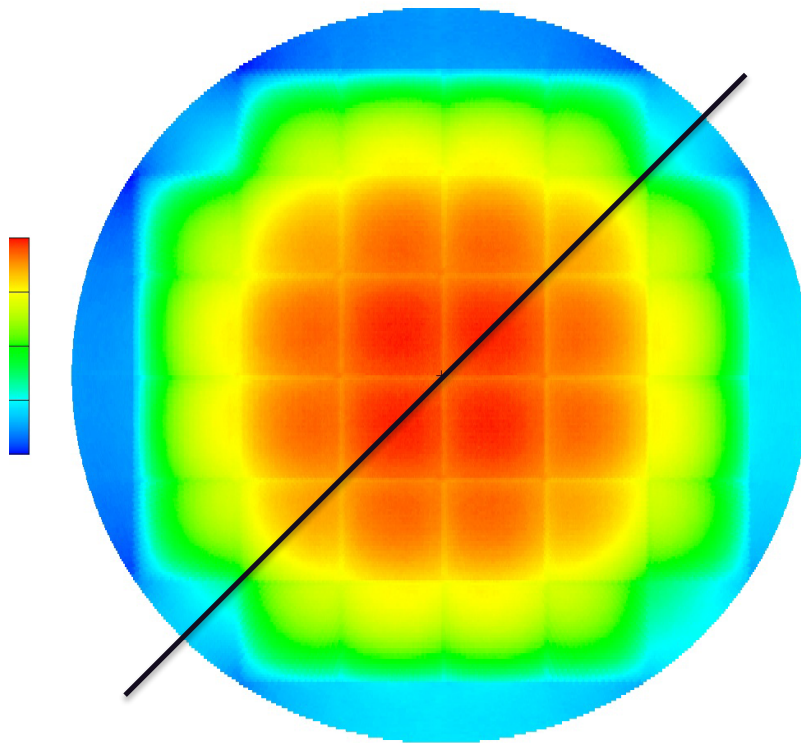


Figure 2-24. A density plot of the neutron flux at a “central slice” of the fuel cask as viewed from above. This plot shows the neutron flux is less in the upper left section than in the lower right section. The asymmetry is most evident in the blue and light blue sections at the outer radius of the figure.

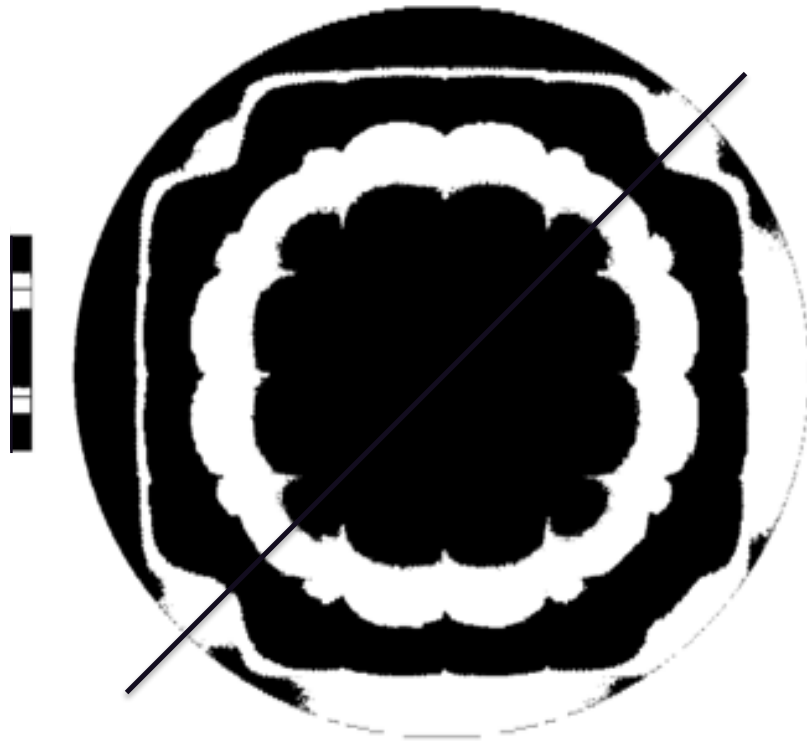


Figure 2-25. A contrast plot emphasizing the asymmetry of the flux values.

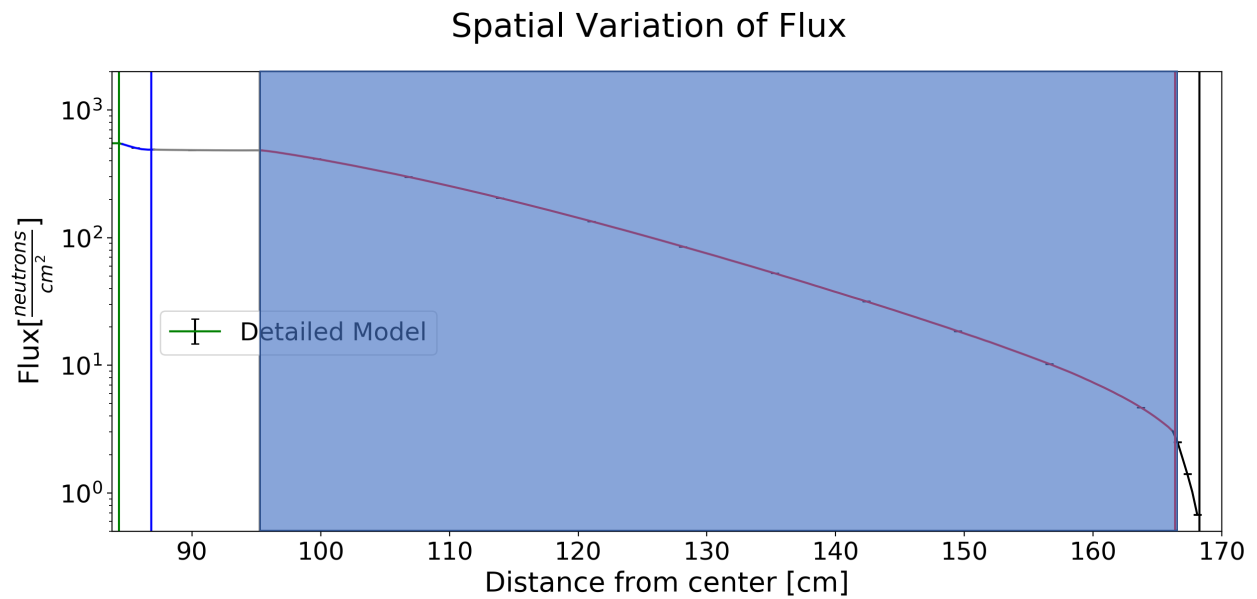


Figure 2-26. The overpack accounts for about half of the reduction to the neutron flux. The purpose of investigating this region is to determine which physical processes are responsible for the attenuation.

CHAPTER 3 THEORY

The behavior of any nuclear system is governed by the distribution of radioactive particles within the system. In the case of a system containing neutrons, the distribution of neutrons can be found by solving the neutron transport equation (NTE). The NTE is also referred to as the Boltzmann transport equation because of its similarity to Boltzmann's equation governing the kinetic theory of gas. Finding an analytic solution of the NTE for even the simplest geometries is a challenging task. Not only because the equation is an integro-differential equation defined over a seven variable phase space, but also because the solution of the NTE depends on parameters which are heavily dependent on the solution to the original equation. The NTE can be solved for the flux by applying assumptions and approximations to reduce the complexity of the equation. The NTE is derived before applying assumptions and simplifications to reduce the NTE into two, distinct tractable approximations; known as (1) the diffusion approximation and (2) the multigroup discrete ordinates equations.

Before deriving the NTE, it is important to define terms which will be used. The neutron angular density,

$$N(\mathbf{r}, \hat{\Omega}, E, t),$$

describes the expected number of neutrons in the region of phase space defined by a neutron's position vector \mathbf{r} , direction of travel $\hat{\Omega}$, and kinetic energy E at time t . It follows that the expected number of neutrons at time t in a volume element dV having energies in dE about E and directions within a narrow beam $d\hat{\Omega}$ about $\hat{\Omega}$ can be described by

$$N(\mathbf{r}, \hat{\Omega}, E, t) dV d\hat{\Omega} dE.$$

The angular flux is defined as the product of speed v and the number of neutrons,

$$\varphi(\mathbf{r}, \hat{\Omega}, E, t) = vN(\mathbf{r}, \hat{\Omega}, E, t).$$

Using the angular flux, the reaction rate is defined as

$$R_x(\mathbf{r}, \hat{\Omega}, E, t) = \Sigma_x(E) \varphi(\mathbf{r}, \hat{\Omega}, E, t)$$

where $R_x(\mathbf{r}, \hat{\Omega}, E, t)$ is the frequency of interactions between neutrons and surrounding materials. The parameter $\Sigma_x(E)$ called the macroscopic cross section for reaction “x” (e.q., total reaction cross section, absorption cross section, scattering cross section). The macroscopic cross section describes the probability of an interaction occurring per unit length as a function of incoming neutron energy.

3.1 Derivation of the Boltzmann Transport Equation for Neutrons by Derivatives

The NTE can be derived by “following” a group of neutrons, referred to as a packet, through a material and describe how neutrons are gained or lost in time. Neutrons with energy E are lost from the packet as a result of a collision over the distance $v\Delta t$, whereas neutrons that do not interact over the distance $v\Delta t$ remain in the packet. The probability of a neutron being removed from the packet over the distance $v\Delta t$ can then be written as

$$\begin{aligned} \text{Probability of a neutron} \\ \text{being removed from the packet} \end{aligned} \equiv \Sigma_t(\mathbf{r}, E) v \Delta t, \quad (3-1)$$

and the probability of a neutron remaining in the packet over the distance $v\Delta t$ is defined as

$$\begin{aligned} \text{Probability of a neutron} \\ \text{remaining in the packet} \end{aligned} \equiv 1 - \Sigma_t(\mathbf{r}, E) v \Delta t. \quad (3-2)$$

Using 3-2, the number of neutrons remaining in the packet after traveling a small distance of $v\Delta t$ is

$$\begin{aligned} \text{Number of neutrons} \\ \text{remaining in packet} \end{aligned} \equiv N(\mathbf{r}, \hat{\Omega}, E, t) [1 - \Sigma_t(\mathbf{r}, E) v \Delta t] dV d\hat{\Omega} dE. \quad (3-3)$$

Eqn. 3-3 adjusts the neutron population accounting for neutrons which left the packet through interactions, however, neutrons can enter the packet through two mechanisms: 1) internal neutron source or 2) by scattering from one packet into another. The number of

1046 neutrons which enter the packet from an internal neutron source is given by

$$\begin{aligned} &\text{Number of neutrons entering} \\ &\text{packet from internal sources} \end{aligned} \equiv s(\mathbf{r}, \hat{\Omega}, E, t) dV d\hat{\Omega} dE \Delta t. \quad (3-4)$$

Neutrons can also enter the packet through scattering interactions, called inscattering. An inscattering reaction occurs when a neutron belonging to the packet described by a volume element dV with energies in dE' about E' and directions within $d\hat{\Omega}'$ about $\hat{\Omega}'$ undergoes a scattering event leaving the neutron traveling in $d\hat{\Omega}$ about $\hat{\Omega}$ with energy in dE about E , adding this neutron to the packet $(\mathbf{r}, \hat{\Omega}, E, t)$. The probability of neutrons with energy E' and direction $\hat{\Omega}'$ which scatter into the energy $E + dE$ with direction in $\hat{\Omega} + d\hat{\Omega}$ can be written as:

$$\begin{aligned} &\text{Probability of neutrons entering} \\ &\text{packet due to inscattering} \end{aligned} \equiv \Sigma_s(\mathbf{r}, \hat{\Omega}' \rightarrow \hat{\Omega}, E' \rightarrow E, t) v N(\mathbf{r}, \hat{\Omega}', E', t). \quad (3-5)$$

Integrating definition 3-5 over all initial energies dE' and initial directions $d\hat{\Omega}'$ yields the number of neutrons that enter the packet due to inscattering,

$$\begin{aligned} &\text{Number of neutrons entering} \\ &\text{packet due to inscattering} \end{aligned} \equiv \left[\int_{4\pi} d\hat{\Omega}' \int_0^\infty dE' \Sigma_s(\mathbf{r}, \hat{\Omega}' \rightarrow \hat{\Omega}, E' \rightarrow E, t) v N(\mathbf{r}, \hat{\Omega}', E', t) \right] dV d\hat{\Omega} dE \Delta t. \quad (3-6)$$

The neutron density at $\mathbf{r} + \hat{\Omega}v\Delta t$ at time $t + \Delta t$ is found by adding 3-3, 3-4, and 3-6 and dividing that sum by $dV d\hat{\Omega} dE$:

$$\begin{aligned} N(\mathbf{r} + \hat{\Omega}v\Delta t, \hat{\Omega}, E, t + \Delta t) = & \\ & N(\mathbf{r}, \hat{\Omega}, E, t)(1 - \Sigma_t v \Delta t) \\ & + \left[\int_{4\pi} d\hat{\Omega}' \int_0^\infty dE' \Sigma_s(\mathbf{r}, \hat{\Omega}' \rightarrow \hat{\Omega}, E' \rightarrow E, t) N(\mathbf{r}, \hat{\Omega}', E', t) \right] \Delta t + S \Delta t. \end{aligned} \quad (3-7)$$

Dividing Eqn. 3-7 and taking the limit as $\Delta t \rightarrow 0$ yields the result, after rearranging terms,

$$\begin{aligned} \lim_{\Delta t \rightarrow 0} \left[\frac{N(\mathbf{r} + \hat{\Omega}v\Delta t, \hat{\Omega}, E, t + \Delta t) - N(\mathbf{r}, \hat{\Omega}, E, t)}{\Delta t} \right] + \Sigma_t v N(\mathbf{r}, \hat{\Omega}, E, t) \\ = \int_{4\pi} d\hat{\Omega}' \int_0^\infty dE' \Sigma_s(\mathbf{r}, \hat{\Omega}' \rightarrow \hat{\Omega}, E' \rightarrow E, t) N(\mathbf{r}, \hat{\Omega}', E', t) + S. \end{aligned} \quad (3-8)$$

1047 Simplifying the first term requires adding and subtracting $N(\mathbf{r}, \hat{\Omega}, E, t + \Delta t)$ to the
 1048 second term in the numerator of the fraction in Eqn. 3-8 and simplifying the expressions
 1049 individually. Adding $N(\mathbf{r}, \hat{\Omega}, E, t + \Delta t)$ to the second term in the numerator gives

$$\lim_{\Delta t \rightarrow 0} \left[\frac{N(\mathbf{r}, \hat{\Omega}, E, t + \Delta t) - N(\mathbf{r}, \hat{\Omega}, E, t)}{\Delta t} \right] = \frac{\partial N}{\partial t}. \quad (3-9)$$

Subtracting $N(\mathbf{r}, \hat{\Omega}, E, t + \Delta t)$ from the first term in the numerator leads to a less trivial expression, but it is more readily derived when decomposed in Cartesian coordinates as

$$\begin{aligned} \lim_{\Delta t \rightarrow 0} \frac{N(\mathbf{r} + \hat{\Omega}v\Delta t, \hat{\Omega}, E, t) - N(\mathbf{r}, \hat{\Omega}, E, t + \Delta t)}{\Delta t} = \\ \lim_{\Delta t \rightarrow 0} \frac{N(x + \Omega_x v\Delta t, y + \Omega_y v\Delta t, z + \Omega_z v\Delta t, \hat{\Omega}, E, t) - N(x, y, z, \hat{\Omega}, E, t + \Delta t)}{\Delta t}, \end{aligned} \quad (3-10)$$

where \mathbf{r} and $\hat{\Omega}$ have components x , y , z and Ω_x , Ω_y , Ω_z respectively. The infinitesimal $\hat{\Omega}v\Delta t$ is equivalent to Δx . Equation 3-10 is then solved using the chain rule.

$$\begin{aligned} \lim_{\Delta t \rightarrow 0} \frac{N(x + \Delta x, y + \Delta y, z + \Delta z) - N(x, y, z)}{\Delta t} = \\ \frac{\Delta N}{\Delta x} \frac{\Delta x}{\Delta t} + \frac{\Delta N}{\Delta y} \frac{\Delta y}{\Delta t} + \frac{\Delta N}{\Delta z} \frac{\Delta z}{\Delta t} = \\ v\Omega_x \frac{\partial N}{\partial x} + v\Omega_y \frac{\partial N}{\partial y} + v\Omega_z \frac{\partial N}{\partial z} = v\hat{\Omega} \cdot \nabla N \end{aligned} \quad (3-11)$$

1050 Inserting the results of Eqn. 3-9 and Eqn. 3-11 into Eqn. 3-8, and using the definition

1051

$$\varphi(\mathbf{r}, \hat{\Omega}, E, t) \equiv N(\mathbf{r}, \hat{\Omega}, E, t)v \quad (3-12)$$

yields the NTE,

$$\frac{1}{v} \frac{\partial \varphi}{\partial t} + \hat{\Omega} \cdot \nabla \varphi + \Sigma_t \varphi(\mathbf{r}, \hat{\Omega}, E, t) = \int_{4\pi} d\hat{\Omega}' \int_0^\infty dE' \Sigma_s(\mathbf{r}, \hat{\Omega}', E', t) \varphi(\mathbf{r}, \hat{\Omega}', E', t) + S(\mathbf{r}, \hat{\Omega}, E, t). \quad (3-13)$$

3.1.1 External Neutron Sources

Discussion of external sources is precluded in the previous section since external source can be handled has boundary conditions when solving for the neutron flux. However, it is important to take a moment to discuss internal source as many texts simply introduce internal source, but provide little further discussion.

A brief dimensional analysis can provide insight into how source terms are defined in the NTE. Since the neutron transport equation describes the number of neutrons in a volume at a point in time, then the units must be $\frac{\text{neutrons}}{\text{Length}^3 \text{Time}}$, or in SI units $\frac{\text{neutrons}}{\text{cm}^3 \text{s}}$. This is easily confirmed by checking the units of one term in Eqn. 3-13. Analyzing the dimensions of the interaction term,

$$\Sigma_t \varphi(\mathbf{r}, \hat{\Omega}, E, t) \equiv \left[\frac{1}{\text{Length}} \right] \left[\frac{\text{neutrons}}{\text{Length}^2 \text{Time}} \right] = \left[\frac{\text{neutrons}}{\text{Length}^3 \text{Time}} \right], \quad (3-14)$$

which confirms the previous statement. Therefore, any source term must have these same units.

Neutron sources can be categorized into two types: 1) flux-driven sources and 2) decay reactions. The distinction is made by how the source strength, or the number of neutrons per volume per unit time, varies with the radiation flux. In flux-driven sources, the source strength changes proportionally with the radiation flux. That is because these sources produce neutrons through reactions that occur when radiation interacts with the target nucleus and produces neutrons. Decay reaction sources do not depend on the neutron flux. Instead, these decay reactions, or simply decays, occur when a nucleus is left in an unstable energy state, typically resulting from some other nuclear reaction. These nuclei need to release energy in order to arrive at a stable energy state. Occasionally,

nuclei get rid of excess energy by ejecting one or more neutrons. Flux-driven sources are handled as boundary conditions when solving the mathematical models. Decay reaction sources show up as the source term, S , in the diffusion approximation.

3.2 Reduction of NTE

Upon inspection of Eqn. 3-13, there are four derivatives on the left-hand side of the equation (one in time and three spatial derivatives) and three integrals on the right-hand side of the equation (one in energy and two in direction). Equations containing both integrals and derivatives are called integro-differential equation and are among the hardest forms of problems to solve. Further, the NTE is a function of seven variables; three spatial, two direction, one energy, and one time. In its current form, the NTE has no complete analytic solution. Therefore, assumptions and approximations are applied to reduce Eqn. 3-13 into a tractable form. The following sections will discuss how the multigroup discrete ordinates equation and the 1-D planar diffusion approximation are derived from the NTE.

3.2.1 Treatment of Time Dependence

The time dependence is contained in the first term in Eqn. 3-13. Assuming the neutron flux is unchanging or slowly changing in time will simplify the time-derivative to zero. This is a fair assumption since the time between neutron interactions is much smaller than the time over which the neutron flux is evolving [37]. In this assumption φ is taken to be independent of time, and

$$\frac{\partial \varphi}{\partial t} = 0. \quad (3-15)$$

Then Eqn. 3-13 becomes the steady-state neutron transport equation,

$$\hat{\Omega} \cdot \nabla \varphi(\mathbf{r}, \hat{\Omega}, E) + \Sigma_t(\mathbf{r}, E) \varphi(\mathbf{r}, \hat{\Omega}, E) = \int_{4\pi} d\hat{\Omega}' \int_0^\infty dE' \Sigma_s(E' \rightarrow E, \hat{\Omega}' \rightarrow \hat{\Omega}) \varphi'(\mathbf{r}, \hat{\Omega}', E') + s(\mathbf{r}, E, \hat{\Omega}). \quad (3-16)$$

Even after eliminating the partial derivative in time, Eqn. 3-13 is still not tractable due to the three spatial derivatives and three integrals. Therefore, further reduction is necessary.

3.2.2 Reduction to 1-D Planar

Reducing the problem from three spatial dimensions to one spatial dimension eliminates two of the three spatial derivatives and one of the two direction derivatives. As an aside, the components of the direction vector $\hat{\Omega}$ are ϕ and θ components. ϕ is the azimuthal angle and θ is the polar angle. It is common to define the variable μ in terms of θ as

$$\mu \equiv \cos \theta,$$

where μ is defined over the range $[-1, 1]$ and ϕ is defined over the range $[0, 2\pi]$. Integrating Eqn. 3-16 over y , z , and ϕ reduces the dimensionality of the problem as

$$\int_{-\infty}^{\infty} dy \int_{-\infty}^{\infty} dz \int_0^{2\pi} d\phi \hat{\Omega} \cdot \nabla \varphi(\mathbf{r}, \hat{\Omega}, E) + \Sigma_t(\mathbf{r}, E) \varphi(\mathbf{r}, \hat{\Omega}, E) - \int_{4\pi} d\hat{\Omega}' \int_0^{\infty} dE' \Sigma_s(E' \rightarrow E, \hat{\Omega}' \rightarrow \hat{\Omega}) \varphi'(\mathbf{r}, \hat{\Omega}', E') + s(\mathbf{r}, E, \hat{\Omega}).$$

Solving the integrals yields:

$$\mu \frac{\partial}{\partial x} \varphi(x, E, \mu) + \Sigma_t(x, E) \varphi(x, E, \mu) = 2\pi \int_0^{\infty} \int_{-1}^1 \Sigma_s(x, E', \mu' \rightarrow E, \mu) \varphi(x, E', \mu') d\mu' dE' + S(x, E, \mu). \quad (3-17)$$

Eqn. 3-17 is the steady-state 1-D planar form of the NTE. While this equation appears much simpler to solve, the derivative on the left-hand side and two integrals on the right-hand side indicate the equation is still an integro-differential equation and further simplification is required to arrive at a tractable form. There are two common reductions to Eqn. 3-17, 1) the multigroup discrete ordinates approximation and 2) the diffusion approximation. The following sections apply each of these approximations to

the NTE in order to arrive at two tractable forms of the NTE which will be used in the remainder of this work.

3.3 Multigroup Discrete Ordinates Approximation

The multigroup discrete ordinates equations handle the two integrals on the right-hand side of Eqn. 3-17 by treating the integral over energy as integrals over energy ranges and approximating the integral over μ by evaluating the neutron flux at discrete angles within the full range of $[-1, 1]$. The final result is a set of coupled, first-order ordinary differential equations that are analytically tractable.

3.3.1 Treatment of Energy Dependence

The first step in developing multigroup equations is to divide the neutron energy range of interest into a finite number of energy groups, E_g , where $g = 1, 2, \dots, G$. The order of the energy group number is such that energy decreases as the group number increases, (e.g., $E_g > E_{g+1}$) [36]. Energy groups are typically chosen such that the cross section shows little variation within a group. This is done in order for the group averaged cross section to best represent the energy-dependent cross section values of that group.

The 1-D planar time-independent neutron transport equation, Eqn. 3-17, is reproduced below for the readers convenience.

$$\mu \frac{\partial}{\partial x} \varphi(x, E, \mu) + \Sigma_t(x, E) \varphi(x, E, \mu) = 2\pi \int_0^\infty \int_{-1}^1 \Sigma_s(x; E', \mu' \rightarrow E, \mu) \varphi(x, E', \mu') d\mu' dE' + S(x, E, \mu). \quad (3-17)$$

Integrating Eqn. 3-17 over g yields

$$\underbrace{\int_g \mu \frac{\partial}{\partial x} \varphi(x, E, \mu) dE}_{(1)} + \underbrace{\int_g \Sigma_t(x, E) \varphi(x, E, \mu) dE}_{(2)} = \underbrace{\int_g 2\pi \int_0^\infty \int_{-1}^1 \Sigma_s(x; E', \mu' \rightarrow E, \mu) \varphi(x, E', \mu') d\mu' dE' dE}_{(3)} + \underbrace{\int_g S(x, E, \mu) dE}_{(4)}, \quad (3-18)$$

where each term will be discussed individually. Before continuing, it is important to define the the group flux and group cross sections as:

$$\varphi_g(x, \mu) \equiv \int_{E_g}^{E_{g-1}} \varphi(x, E, \mu) dE = \int_g \varphi(x, E, \mu) dE, \quad (3-19)$$

$$\Sigma_{t,g}(x, \mu) \equiv \frac{\int_g \Sigma_t(x, E, \mu) \varphi(x, E, \mu) dE}{\varphi_g(x, E, \mu)}, \quad (3-20)$$

$$\Sigma_{s,g' \rightarrow g}(x, \mu) \equiv \frac{\int_{g'} \varphi(x, E', \mu) \int_g \int_{-1}^1 \Sigma_s(x; E', \mu' \rightarrow E, \mu) d\mu' dE dE'}{\varphi_{g'}(x, E', \mu)}. \quad (3-21)$$

1118 $\varphi_g(x, \mu)$ is the group averaged flux, $\Sigma_{t,g}(x, \mu)$ is the group averaged cross section, and

1119 $\Sigma_{s,g' \rightarrow g}(x, \mu)$ is the group to group, or transfer, cross section.

1120 Definitions 3-19 - 3-21 are used to rewrite Eqn. 3-18 term by term. The first term of

1121 Eqn. 3-18 is rewritten in terms of the group flux, 3-19 as

$$\int_g \mu \frac{\partial}{\partial x} \varphi(x, E, \mu) dE = \mu \frac{\partial}{\partial x} \varphi_g(x, \mu). \quad (3-22)$$

Rewriting the second term in Eqn. 3-18 using the total group cross section, Eqn. 3-20, yields

$$\int_g \Sigma_t(x, E, \mu) \varphi(x, E, \mu) dE = \Sigma_{t,g}(x, \mu) \varphi_g(x, \mu). \quad (3-23)$$

The third term in Eqn. 3-18 requires a bit more work. If the integral of dE' is taken over each individual energy group rather than over 0 to ∞ , then

$$\int_0^\infty dE' = \sum_{g'=1}^G \int_{E_{g'}}^{E_{g'-1}} dE' = \sum_{g'=1}^G \int_{g'} dE',$$

and the third term can be expressed using group constants, Eqn. 3-24.

$$\int_{g'} \varphi(x E', \mu) \int_g \Sigma(x; E', \mu' \rightarrow E, \mu) dE dE' = \sum_{g'=1}^G \Sigma_{g' \rightarrow g}(x, \mu) \varphi_{g'}(x, \mu) \quad (3-24)$$

1122 Finally, the fourth term is the group source term, Eqn. 3-25. The group source term
 1123 describes an arbitrary internal source of neutrons with energy in group g .

$$\int_g S(x, E, \mu) dE \equiv S_g(x, \mu). \quad (3-25)$$

1124 Using the redefined terms, Eqns. 3-22 - 3-25, Eqn. 3-18 becomes a set of equations
 1125 characterizing the flux in each energy group:

$$\mu \frac{\partial \varphi_g}{\partial x} + \Sigma_{t,g} \varphi_g = 2\pi \sum_{g'=1}^G \int_{-1}^1 \Sigma_{s,g' \rightarrow g} \varphi_{g'} + S_g, \quad g = 1, 2, \dots, G. \quad (3-26)$$

1126 3.3.2 Treatment of Directional Dependence

Equation 3-26 is a set of monoenergetic neutron transport equations where each equation defines the flux for the energy group g . Therefore, if a method for handling the directional dependence can be found for a single equation in the set of equations, the same method can be extended to all equations in Eqn. 3-26. The discrete ordinates method can be used to handle the integral over μ . By first assuming isotropic scattering, the in-scattering term reduces to

$$2\pi \sum_{g'=1}^G \int_{-1}^1 \Sigma_{s,g' \rightarrow g} \varphi_{g'} d\mu' = \frac{1}{2} \sum_{g'=1}^G \Sigma_{s,g' \rightarrow g} \int_{-1}^1 \varphi_{g'} d\mu',$$

1127 and Eqn. 3-26 reduces to

$$\mu \frac{\partial \varphi_g}{\partial x} + \Sigma_{t,g} \varphi_g = \frac{1}{2} \sum_{g'=1}^G \Sigma_{s,g' \rightarrow g} \int_{-1}^1 \varphi_{g'} + S_g, \quad g = 1, 2, \dots, G. \quad (3-27)$$

1128 Discrete ordinates treats directional dependence by evaluating the integral over μ at
 1129 a unique set of directions, $\{\mu_i\}$. Evaluating the integral in Eqn. 3-27 at each value of μ_i
 1130 leads to a weighted sum of neutron fluxes, Eqn. 3-28.

$$\int_{-1}^1 \varphi_{g'} = \sum_{j=1}^N \omega_j \phi_{g'}(x, \mu_j) \quad (3-28)$$

1131 Evaluating Eqn. 3-27 along the set of direction vectors $\{\mu_i\}$, using Eqn. 3-28, results in
 1132 the multigroup discrete ordinates equations:

$$\mu_i \frac{d\phi_i^g}{dx} + \Sigma_t^g \phi_i^g = \frac{1}{2} \sum_{j=1}^N \omega_j \sum_{g'=1}^G \Sigma_{s,g' \rightarrow g} \phi_j^{g'} + S_i^g, \quad g = 1, 2, \dots, G; i = 1, 2, \dots, N. \quad (3-29)$$

For the purpose of this work, a set of equations are derived from Eqn. 3-29 using two energy groups ($g=1,2$) and two directions ($i=1,2$). Iterating over both indices one at a time leads to the following set of equations:

$$g = 1, i = 1$$

$$\mu_1 \frac{d\phi_1^1}{dx} + \Sigma_t^1 \phi_1^1 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 1} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 1} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 1} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 1} \omega_2 \phi_2^2 \right) + S_1^1; \quad (3-30)$$

$$g = 1, i = 2$$

$$\mu_2 \frac{d\phi_2^1}{dx} + \Sigma_t^1 \phi_2^1 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 1} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 1} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 1} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 1} \omega_2 \phi_2^2 \right) + S_2^1; \quad (3-31)$$

$$g = 2, i = 1$$

$$\mu_1 \frac{d\phi_1^2}{dx} + \Sigma_t^2 \phi_1^2 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 2} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 2} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 2} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 2} \omega_2 \phi_2^2 \right) + S_1^2; \quad (3-32)$$

$$g = 2, i = 2$$

$$\mu_2 \frac{d\phi_2^2}{dx} + \Sigma_t^2 \phi_2^2 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 2} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 2} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 2} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 2} \omega_2 \phi_2^2 \right) + S_2^2. \quad (3-33)$$

In a material where scattering from lower energy groups to higher energy groups does not occur and there is no internal neutron source, these equations reduce to:

$$g = 1, i = 1$$

$$\mu_1 \frac{d\phi_1^1}{dx} + \Sigma_t^1 \phi_1^1 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 1} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 1} \omega_2 \phi_2^1 \right); \quad (3-34)$$

$$g = 1, i = 2$$

$$\mu_2 \frac{d\phi_2^1}{dx} + \Sigma_t^1 \phi_2^1 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 1} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 1} \omega_2 \phi_2^1 \right); \quad (3-35)$$

$$g = 2, i = 1$$

$$\mu_1 \frac{d\phi_1^2}{dx} + \Sigma_t^2 \phi_1^2 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 2} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 2} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 2} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 2} \omega_2 \phi_2^2 \right); \quad (3-36)$$

$$g = 2, i = 2$$

$$\mu_2 \frac{d\phi_2^2}{dx} + \Sigma_t^2 \phi_2^2 = \frac{1}{2} \left(\Sigma_s^{1 \rightarrow 2} \omega_1 \phi_1^1 + \Sigma_s^{1 \rightarrow 2} \omega_2 \phi_2^1 + \Sigma_s^{2 \rightarrow 2} \omega_1 \phi_1^2 + \Sigma_s^{2 \rightarrow 2} \omega_2 \phi_2^2 \right). \quad (3-37)$$

1133

3.4 Reduction to Diffusion Approximation

The diffusion approximation is an alternative reduction of the NTE. There are several methods for deriving the diffusion approximation, however, this derivation uses Legendre polynomial expansions to account for angular dependence in the equation. The neutron transport equation can be simplified through the use of spherical harmonics, which in 1-D, reduce to Legendre polynomials to expand the angular flux and source terms while assuming an isotropic angular differential cross section. The 1-D planar, monoenergetic, NTE with isotropic scattering is

$$\mu \frac{\partial}{\partial x} \varphi(x, \mu) + \Sigma_t(x) \varphi(x, \mu) = \frac{1}{2} \int_{-1}^1 \Sigma_s(x, \mu' \rightarrow \mu) \varphi(x, \mu') d\mu' + S(x, \mu) \quad (3-38)$$

1134

Expanding the angular flux with Legendre polynomials separates the directional and

1135

spatial components of the angular flux. Legendre polynomials exhibit an orthogonality

1136

property, Eqn. 3-39, and a "3-term recursion" relationship, Eqn. 3-40, which are used in

1137 deriving the diffusion approximation.

$$\int_{-1}^1 d\mu P_l(\mu) P_m(\mu) = \frac{2}{2l+1} \delta_{lm} \quad (3-39)$$

1138

$$(2l+1)\mu P_l(\mu) = (l+1)P_{l+1}(\mu) + (l)P_{l-1}(\mu) \quad (3-40)$$

Expanding the angular flux in Eqn. 3-38 yields:

$$\begin{aligned} \mu \frac{\partial}{\partial x} \left[\sum_l \frac{2l+1}{2} \phi_l(x) P_l(\mu) \right] + \Sigma_t \sum_l \frac{2l+1}{2} \phi_l(x) P_l(\mu) = \\ \frac{1}{2} \int_{-1}^1 d\mu' \Sigma_s(x, \mu_0) \sum_l \frac{2l+1}{2} \phi_l(x) P_l(\mu') + S(x, \mu). \end{aligned} \quad (3-41)$$

Requiring the projections of Eqn. 3-41 against Legendre polynomials of degree m (e.q., P_m) to be equal to 0 leads to

$$\begin{aligned} \int_{-1}^1 d\mu \mu \frac{\partial}{\partial x} \left[\sum_{l=0}^1 \frac{2l+1}{2} \phi_l(x) P_l(\mu) P_m(\mu) \right] + \int_{-1}^1 d\mu \Sigma_t \sum_{l=0}^1 \frac{2l+1}{2} \phi_l(x) P_l(\mu) P_m(\mu) = \\ \frac{1}{2} \int_{-1}^1 d\mu P_m(\mu) \int_{-1}^1 d\mu' \Sigma_s(x, \mu_0) \sum_{l=0}^1 \frac{2l+1}{2} \phi_l(x) P_l(\mu') + \int_{-1}^1 d\mu S(x, \mu) P_m(\mu). \end{aligned} \quad (3-42)$$

1139 The summation is truncated at $l = 1$ since the first two terms are all that is necessary for
1140 finding the diffusion approximation.

Using the recurrence relationship, Eqn. 3-40, in the first term of Eqn. 3-42 yields

$$\sum_{l=0}^1 \frac{\partial \phi_l(x)}{\partial x} \left[\int_{-1}^1 d\mu \frac{l+1}{2} P_{l+1}(\mu) P_m(\mu) + \int_{-1}^1 d\mu \frac{l}{2} P_{l-1}(\mu) P_m(\mu) \right].$$

Applying the orthogonality gives,

$$\frac{(m-1)+1}{2} \frac{2}{2m+1} \frac{\partial \phi_{m-1}(x)}{\partial x} + \frac{m+1}{2} \frac{2}{2m+1} \frac{\partial \phi_{m+1}(x)}{\partial x},$$

or,

$$\frac{m}{2m+1} \frac{\partial \phi_{m-1}(x)}{\partial x} + \frac{m+1}{2m+1} \frac{\partial \phi_{m+1}(x)}{\partial x}.$$

The second term in Eqn. 3-42 is also solved using the orthogonality property as,

$$\Sigma_t \sum_{l=0}^1 \frac{2l+1}{2} \phi_l(x) \int_{-1}^1 d\mu P_l(\mu) P_m(\mu)$$

such that,

$$\Sigma_t \sum_{l=0}^1 \frac{2l+1}{2} \phi_m(x) \frac{2}{2m+1},$$

or,

$$\Sigma_t \phi_m(x).$$

Solving the third term of Eqn. 3-42 involves calculating the values for $P_{l/m}(\mu)$ for $l, m = 0, 1$, which are $P_0(\mu) = 1$ and $P_1(\mu) = \mu$. Note each integral evaluates to 0 when either l or m is odd. Alternatively, the scattering term evaluates to $2\phi_m$ when l and m are 0.

$$\frac{1}{2} \Sigma_s(x, \mu_0) \sum_{l=0}^1 \frac{2l+1}{2} \int_{-1}^1 d\mu' \phi_l(x) P_l(\mu') \int_{-1}^1 d\mu P_m(\mu) = \begin{cases} 2\phi_0; & l \text{ and } m = 0 \\ 0; & \text{else.} \end{cases}$$

And the final term in Eqn. 3-42 is simply redefined as:

$$S_m \equiv \int_{-1}^1 d\mu S(x, \hat{\Omega}) P_m(\mu).$$

1141 For an isotropic source, $S_m = 0$ for $m > 0$.

1142 Combining the terms leads to the final set of P_1 equations, Eqns. 3-43 and 3-44.

$$\frac{\partial \phi_1}{\partial x} + \Sigma_t \phi_0 = \Sigma_s \phi_0 + S_0 \quad (3-43)$$

1143

$$\frac{1}{3} \frac{\partial \phi_0}{\partial x} + \frac{2}{5} \frac{\partial \phi_2}{\partial x} + \Sigma_t \phi_1 = S_1. \quad (3-44)$$

1144 If this set of equations were solved for ϕ_0 , the result would be the diffusion approximation.

1145 Unfortunately, there are three unknowns (ϕ_0 , ϕ_1 , and ϕ_2) and two equations. In fact, this

1146 set of equations will always have more unknown variables than equations. Therefore, a

1147 closure condition is needed to truncate the set of equations by setting $\phi_n = 0$ for $n \geq 2$.

Eqn. 3-44 then becomes

$$\phi_1 = \frac{-1}{3\Sigma_t} \frac{\partial \phi_0}{\partial x} \quad (3-45)$$

which is Fick's Law.

Substituting Fick's Law in Eqn. 3-43 for ϕ_1

$$\frac{\partial}{\partial x} \left[\frac{-1}{3\Sigma_t} \frac{\partial \phi_0}{\partial x} \right] + \Sigma_t \phi_0 = \Sigma_s \phi_0 + S_0,$$

which simplifies to the 1-D, mono-energetic, steady state diffusion approximation:

$$-D \frac{\partial^2 \phi_0}{\partial x^2} + \Sigma_a \phi_0 = S_0, \quad (3-46)$$

where D , the diffusion coefficient is defined as

$$D \equiv \frac{-1}{3\Sigma_t}, \quad (3-47)$$

when D is independent of x . The second derivative, $\frac{\partial^2}{\partial x^2}$, results from expressing the Laplacian operator in a planar coordinates systems where the coordinate-independent diffusion approximation is

$$-D \nabla^2 \phi_0 + \Sigma_a \phi_0 = S_0, \quad (3-48)$$

from Duderstadt & Hamilton [37]. Given the cylindrical geometry of the cask, the diffusion equation is expected to be applied in a cylindrical coordinate system. Equation 3-49 is the 1-D cylindrical, steady-state monoenergetic diffusion equation where the Laplacian has been expressed in cylindrical coordinates.

$$-D \frac{1}{r} \frac{d}{dr} \left(r \frac{d\phi_0}{dr} \right) + \Sigma_a \phi_0 = S_0 \quad (3-49)$$

3.5 Cylindrical to Polar Coordinate Shift

The cylindrical shape of the spent fuel cask immediately lends to a cylindrical geometry for the mathematical models. However, given the large radius of the cask, it is expected that there exists a point along the radius of the cask where polar geometry can

1163 be relaxed to a planar geometry with negligible effect to the neutron flux. This point can
 1164 be found through a dimensional analysis by developing non-dimensional forms for both the
 1165 radial and planar diffusion equations.

1166 Non-dimensional analysis is a process where an equation is rewritten in a manner such
 1167 that there are no units in the problem (i.e., all parameters and variables in an expression
 1168 are redefined using ratios rather than dimensional quantities). A comparison can be
 1169 made between the non-dimensional forms of the 1-D polar diffusion approximation and
 1170 1-D planar diffusion approximation to determine the location where planar geometry is
 1171 appropriate.

1172 Starting with the geometry-independent diffusion equation,

$$-D\nabla^2\phi + \Sigma_a\phi = S. \quad (3-46)$$

1173 where the second derivative has been written using the gradient, D is the diffusion
 1174 coefficient, ϕ is the scalar flux, Σ_a is macroscopic absorption cross section, and S is the
 1175 source term. The monoenergetic, steady-state, 1-D planar diffusion approximation:

$$-D\frac{d^2\phi}{dx^2} + \Sigma_a\phi = S. \quad (3-46)$$

1176 Dividing the equation by $-D$ and defining $L^{-2} \equiv \frac{\Sigma_a}{D}$,

$$\frac{d^2\phi}{dx^2} - \frac{1}{L^2}\phi + \frac{S}{D} = 0. \quad (3-50)$$

1177 Non-dimensionalizing x ,

$$\tilde{x} = \frac{x}{L}, \quad (3-51)$$

1178 where \tilde{x} is the non-dimensionalized form of x . The first derivative becomes

$$dx = L d\tilde{x} \quad (3-52)$$

1179 in non-dimensional form.

1180 The second order differential of x , dx^2 , becomes

$$dx^2 = L^2 d\tilde{x}. \quad (3-53)$$

Eqn. 3-46 then becomes

$$\frac{1}{L^2} \frac{d^2\phi}{dx^2} - \frac{1}{L^2} \phi + \frac{S}{D} = 0,$$

1181 or,

$$\frac{d^2\phi}{d\tilde{x}^2} - \phi + \frac{L^2 S}{D} = 0. \quad (3-54)$$

1182 Note: $\frac{L^2 S}{D}$ has units of $Length^{-2} Time^{-1}$, which are the same units as ϕ . So,

$$\tilde{\phi} = \frac{\phi}{L^2 S/D}, \quad (3-55)$$

1183 or,

$$\phi = \tilde{\phi} \frac{L^2 S}{D}, \quad (3-56)$$

1184 where $\tilde{\phi}$ is the non-dimensionalized form of ϕ . The second differential of ϕ becomes

$$d^2\phi = \frac{L^2 S}{D} d^2\tilde{\phi}. \quad (3-57)$$

1185 Using $\tilde{\phi}$, Eqn. 3-54 is written as

$$\frac{L^2 S}{D} \frac{d^2\tilde{\phi}}{d\tilde{x}^2} - \frac{L^2 S}{D} \tilde{\phi} + \frac{L^2 S}{D} = 0, \quad (3-58)$$

1186 or,

$$\frac{d^2\tilde{\phi}}{d\tilde{x}^2} - \tilde{\phi} + 1 = 0. \quad (3-59)$$

1187 The 1-D planar diffusion approximation is now expressed in a non-dimensional form.

1188 Expressing the gradient in Eqn. 3-46 in 1-D polar coordinates yields

$$\frac{1}{r} \frac{d}{dr} \left(r \frac{d\phi}{dr} \right) - \frac{1}{L^2} \phi + \frac{S}{D} = 0, \quad (3-60)$$

1189 or,

$$\frac{d^2\phi}{dr^2} + \frac{1}{r} \frac{d\phi}{dr} - \frac{1}{L^2} \phi + \frac{S}{D} = 0. \quad (3-61)$$

1190 Let

$$\tilde{r} = \frac{r}{L}, \quad (3-62)$$

1191 and,

$$\tilde{\phi} = \frac{\phi D}{L^2 S} \quad (3-63)$$

1192 Using the non-dimensionalized variables defined in Eqns. 3-62 and 3-63, Eqn. 3-61 can
1193 be rewritten as

$$\frac{d^2 \tilde{\phi}}{d\tilde{r}^2} + \frac{1}{\tilde{r}} \frac{d\tilde{\phi}}{d\tilde{r}} - \tilde{\phi} + 1 = 0. \quad (3-64)$$

1194 Then, the curvilinear form of the diffusion equation is

$$\frac{d^2 \tilde{\phi}}{d\tilde{r}^2} + \frac{k}{\tilde{r}} \frac{d\tilde{\phi}}{d\tilde{r}} - \tilde{\phi} + 1 = 0, \quad (3-65)$$

1195 where $k = 0$ for planar geometries and $k = 1$ for cylindrical geometries. Further, plotting
1196 the variable $\frac{k}{\tilde{r}}$ for $k = 1$ will show the location where accounting for polar geometries
1197 becomes negligible. Figure 3-1 shows the result from the previous dimensional analysis
1198 using material properties of the fuel materials. Near 1cm into the fuel material, results
1199 calculated using a cylindrical and planar geometries agree within 10% (denoted by the
1200 vertical black line in Fig. 3-1). After 1cm materials can be approximated using planar
1201 equations, however, the flux in the fuel region will need to be approximated using a polar
1202 diffusion equation.

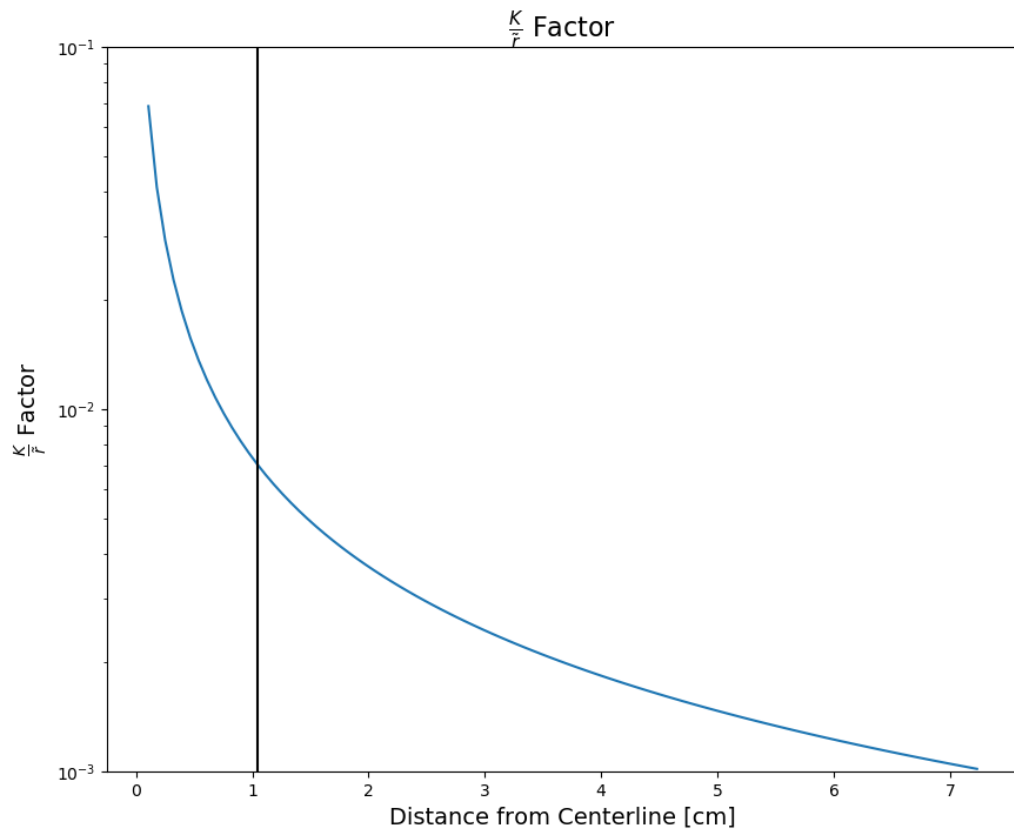


Figure 3-1. As the factor $\frac{k}{r}$ decreases, the planar solutions better approximate polar solutions in the homogenous fuel material. The location of the black vertical line shows the point where the factor $\frac{k}{r}$ is 10% of its initial value.

CHAPTER 4

ANALYSIS OF SUB-PROBLEMS

This chapter analyzing each sub-problem in depth according to the results assessment methodology using the mathematical models identified in Sec. 2 and derived in Sec. 3. Through the explanation of the causes of each feature, confidence is gained in the correctness of the detailed MCNP simulation.

4.1 Identification of Mathematical Models in Each Region

Chapter 2 identified the analytic model choice for each material region. However, differential equations only yield unique solutions when coupled with boundary conditions. Therefore, a discussion identifying appropriate boundary conditions in each material is provided. The fuel region has a unique geometry-induced feature at the center of the cylindrical fuel region where the radius is 0. The geometry at the center of the cask suggests the central symmetry boundary condition which limits the solution to a finite value at the centerline of the cask, where $r = 0$, as

$$\lim_{r \rightarrow 0} \phi(r) < \infty. \quad (4-1)$$

Further, at the exiting surface of the fuel region, an approximate non-reentrant boundary condition associated with Eqn. 3-49 is

$$\phi(\mathbf{r}_b + d) = 0, \quad (4-2)$$

where \mathbf{r}_b is the vector of positions comprising the outer surface S of V , and d is an “extrapolation distance” [suggested via notation to be uniformly added to each element of \mathbf{r}_b in Eqn. 4-2] given by Eqn. 4-2] given by

$$d = 2.13D. \quad (4-3)$$

Equation 4-2 is intended to qualitatively reproduce the neutron flux behavior at the outer surface of a non-reentrant convex body, as otherwise observed from more general neutron transport scenarios (see, for example, Lamarsh and Baratta).

The remaining materials share the same choice of analytic model, Eqn. 3-29. Therefore, a single boundary condition can be used in each model, specifically, choosing to have a continuous flux boundary condition for each partial flux is satisfactory between each material, Eqn. 4-4. In Eqn. 4-4, a and b represent the two materials on either side of the boundary surface, i is the direction and g is the energy group number. The multigroup discrete ordinates equation is a first-order set of coupled differential equations and a single boundary condition for each flux is adequate.

$$\phi_{a,i}^g = \phi_{b,i}^g \quad (4-4)$$

Finally, the analytic model changes at the boundary between the fuel region and the MPC. That is, the flux in the fuel is analytically predicted using the diffusion approximation where the flux in the remaining materials is predicted using the multigroup discrete ordinates equation. The flux in the fuel is given as a single flux, treated at a single energy group and with no angle dependence. However, the multigroup discrete ordinates equations treats the flux in two energy groups and two angles. In order to “stitch” the flux together at the interface between the fuel region and MPC, energy and angular distribution data is approximated at the interior surface of the MPC. Figure 2-8h shows the percentage of neutrons above 10keV as 73.854%. Therefore, 73.854% of the neutrons calculated with the diffusion equation are considered “fast” neutrons and this is called the fast energy group or group 1 flux. The remaining neutrons are placed in the “thermal” energy or group 2 flux. A similar approach is used to treat the angular dependence at the interface. From the analysis of the detailed cask, 57.283% of the neutrons are traveling forward, or outward from the fuel region. Therefore, these neutrons are considered the right moving flux and remaining neutrons are considered left moving. This rough analysis

of the flux at the interface provides a suitable boundary condition for the flux at the inner surface of the MPC, where Eqns. 4-5-4-8 show the boundary conditions for each partial neutron flux.

$$\phi_1^{MPC,1}(x = 84.34) = (0.73854)(0.57383)\phi_{fuel} \quad (4-5)$$

$$\phi_2^{MPC,1}(x = 84.34) = (0.73854)(0.42617)\phi_{fuel} \quad (4-6)$$

$$\phi_1^{MPC,2}(x = 84.34) = (0.26146)(0.57383)\phi_{fuel} \quad (4-7)$$

$$\phi_2^{MPC,2}(x = 84.34) = (0.26146)(0.42617)\phi_{fuel} \quad (4-8)$$

4.2 Discussion of Sub-problems

4.2.1 Flat Region

Initially, the flatness of the first feature suggests that a reduction in fine structure detail can be used to adequately represent a substantial portion the fuel region. Each fuel pin is approximately 1cm in diameter, yet the neutron flux spatial distribution does not show variations at the centimeter level. Fluctuations in the neutron flux spatial distribution at the centimeter level would require any simplified models to also preserve geometric structures at the centimeter level, but the absence of these fluctuations implies that geometric reductions are possible. Therefore, an MCNP model is developed with a homogenized fuel in the MPC similar to the analytic model.

For the purpose of clarity, this fuel composition is called “fully homogenized” since it incorporates all the materials inside the MPC. The fully homogenized fuel composition is determined by calculating the mass fractions of each material in the MPC (the stainless steel basket, the neutron absorbing pads, the helium backfill, and the fuel rods). Finally, the density of the fully homogenized fuel is corrected to account for the various densities of each material in the MPC ($10.44 \frac{g}{cm^3}$ for a single fuel rod vs. $2.31 \frac{g}{cm^3}$ for the fully homogenized fuel). The entire interior volume of the MPC is filled with the fully homogenized fuel material. Figure 4-1 is a cross section view of the corresponding

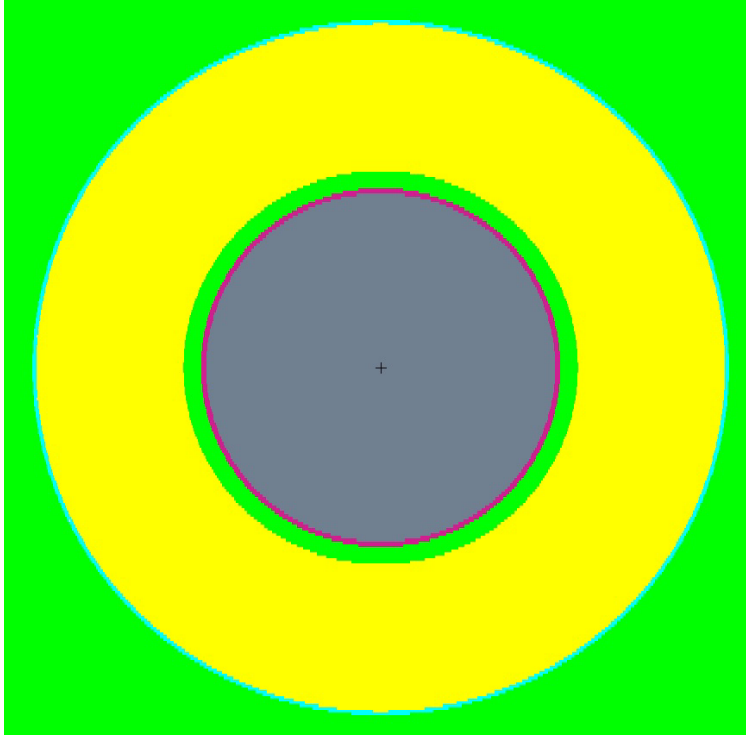


Figure 4-1. The homogeneous model. The gray circle is the fully homogenized fuel which fills the entire volume interior to the MPC.

MCNP model using the fully homogenized fuel material. This model is referred to as the “homogenous model”.

Figure 4-2 shows the homogeneous model neutron flux spatial distribution through the fuel region of the MPC, together with the complementary result from the detailed model. The homogenous model over predicts the neutron flux spatial distribution by 20-25% through the fuel region. Even though the reduced model overpredicts the detailed flux, the shape of the neutron flux spatial distribution predicted in both models shows a steady decrease across the inner 65 cm. The relative flatness of the two fluxes is evidence that geometric attenuation is less important than the material properties within the MPC. To further corroborate this notion, Fig. 4-2 also includes results from an analytic model: the dotted line appearing in this figure is a result from monoenergetic, 1D cylindrical diffusion theory, Eqn. 3-46. In this analytic setting, the monoenergetic scalar neutron flux

across a 1D cylindrical region with constant material properties is given by

$$\phi(r) = \frac{\alpha}{DB^2} \left(1 - \frac{I_0(Br)}{I_0(B\tilde{r})} \right); \quad B \equiv \sqrt{\frac{\Sigma_a - \bar{\nu}\Sigma_f}{D}}. \quad (4-9)$$

where α is the intrinsic neutron source, B is the material buckling as indicated in terms of the macroscopic total absorption cross section Σ_a , macroscopic fission cross section Σ_f , and mean number of neutrons per fission $\bar{\nu}$, and diffusion coefficient D , I_0 is the modified Bessel function of the first kind, and \tilde{r} is the extrapolated radius of the fuel region. The spatial curvature of the scalar flux appearing in Eq. 4-9 is controlled principally by the material buckling B ; as the value of B increases (resulting when absorption physics is dominant over scattering physics) the neutron flux spatial distribution calculated in Eq. 4-9 produces a flat distribution in r - as in fuel region of both computational models. This result is discussed further in conjunction with the sensitivity discussion corresponding to the diffusion approximation. The flatness of the diffusion model is proof that the flatness seen in the MCNP models is due to material properties being dominant over geometry. While the diffusion model captures the essential physics giving rise to the flat flux region, it does not adequately capture the abrupt level off within the fuel region for $r > 65\text{cm}$.

4.2.2 Abrupt Level-off Region

In order to better capture the physics which describes the second feature, a second model is developed. The purpose of this model is to capture the physics associated with the neutron flux spatial distribution suddenly flattening before exiting the MPC. Intuitively, since geometric attenuation is minimal and the mfp for neutrons (70,000 cm at 1 MeV) is much greater than the thickness of the region between the fuel basket and MPC wall (10 cm), a free streaming (i.e., constant flux) approximation is likely to be valid there. To corroborate this notion, the homogeneous model is further modified to add an annulus of helium around a fuel region which is reduced in radius in a manner which preserves the volume of the original 32 fuel cells. This model is referred to as the “helium model”. Fig. 4-3 shows the difference between the homogenous and helium models. The

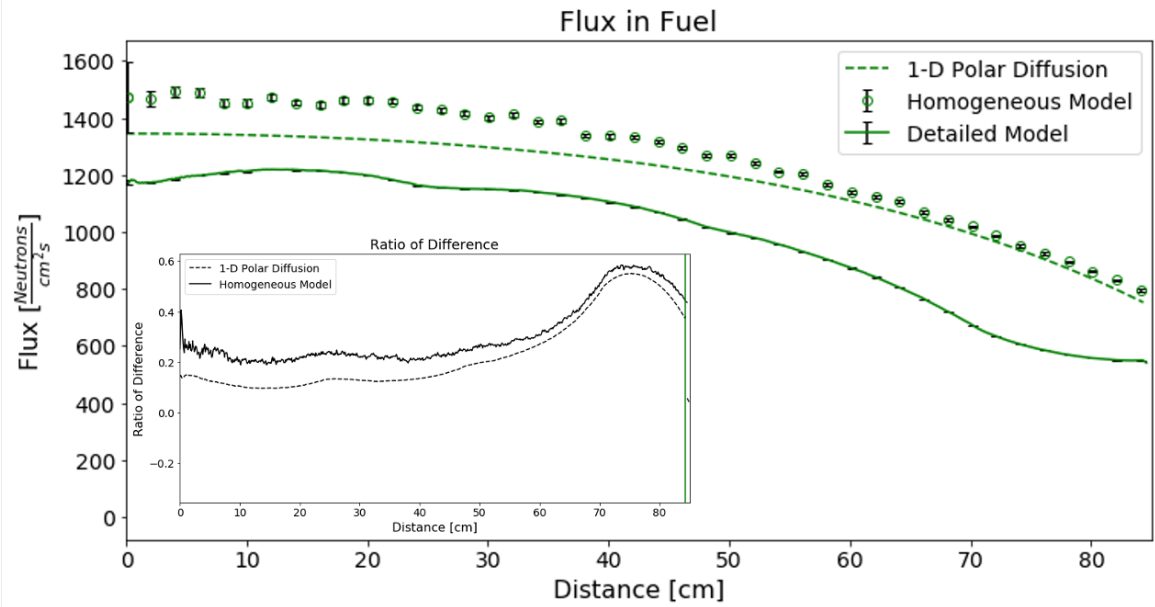


Figure 4-2. The results of the simulated neutron flux spatial distribution from the homogenous model (circles) is similarly flat to the neutron flux spatial distribution of the detailed model (solid line). The flux calculated using the diffusion approximation (dotted line) is also plotted against the two MCNP models. The diffusion approximation also shows the flatness of the neutron flux spatial distribution.

1287 composition of the fuel region is changed to account for the helium now present in the
 1288 annulus. The new homogenized fuel composition, called the partially homogenized fuel
 1289 composition, is made using the mass fractions of materials in the 32 fuel cells (the stainless
 1290 steel fuel basket, the neutron absorbing pads, the helium interior to the fuel cells, the fuel
 1291 rods) and the density of the material is adjusted to account for the reduced amount of
 1292 helium ($2.95 \frac{g}{cm^3}$).

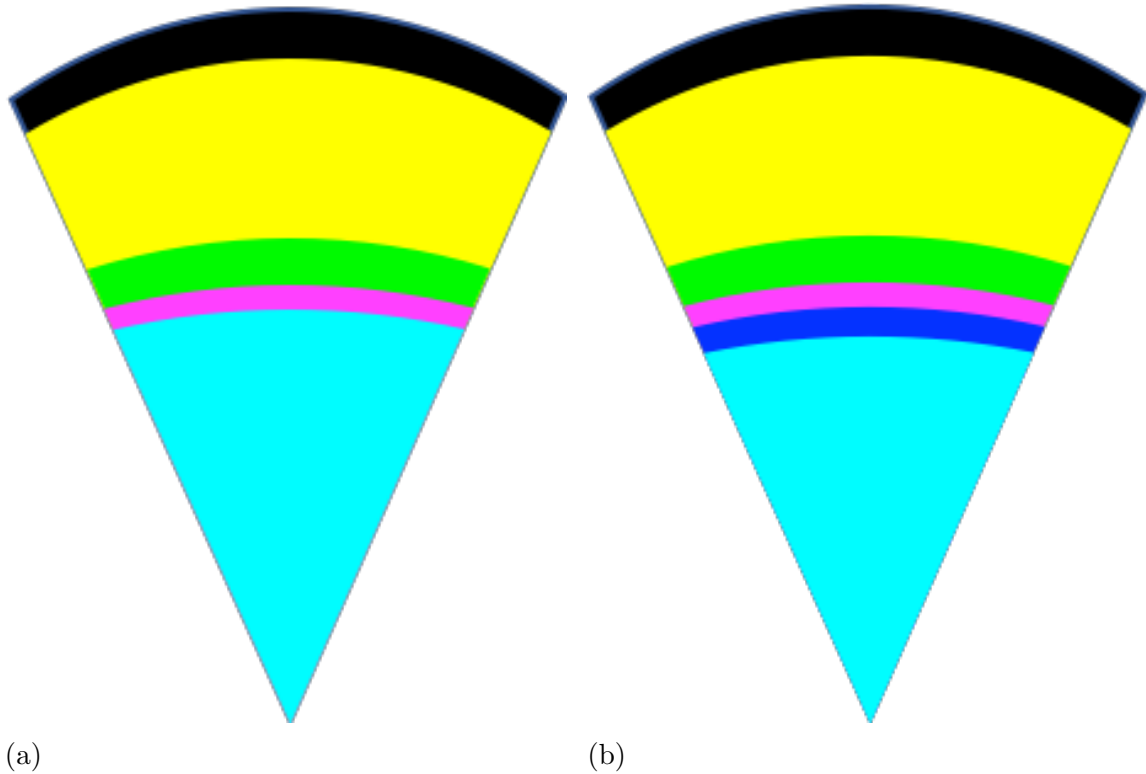


Figure 4-3. A) Section views of the homogeneous model, B) Helium model. The helium model includes an annulus of helium gas, ~ 10 cm thick, added around the homogenized fuel to allow streaming at the edge of the fuel region. Not to scale.

1293 Figure 4-4 shows the results of the simulated flux in the helium model as compared
 1294 to the detailed model. The fuel region, comprised of partially homogenized fuel material,
 1295 has a smaller radius and the analytic solution is held constant for $r > \tilde{r}$. The increased
 1296 density of the fuel in the helium model increases the total neutron absorption and thus
 1297 lowers the amplitude of the neutron flux spatial distribution. The flux flattens out over

the last 20 cm, which is a result of adding the non-interacting helium annulus. The helium model better demonstrates that the flatness of the two MCNP models and the analytic model match, with the exception of the three depressions present in the detailed models. These results do show neutrons streaming through the helium region exterior to the fuel cells before exiting into the MPC even though the helium model and the analytic model do not capture the small depressions.

4.2.3 Inter-bundle Depressions

To this point, the simulation results assessment has shown that explanation of causes for the first two features does not necessitate simulation of geometric details at the individual fuel pin level. However, the physics associated with the three small depressions in the detailed model (seen in Fig. 2-23) has not been explained. Intuition suggests it seems necessary that some level of geometric detail needs to be added back into the reduced complexity simulations to identify the cause of the final two features.

The scalar flux depressions depicted in Fig. 2-23 represent the third feature and are presumed to be caused by the neutron absorbing pads that are present between fuel bundles, located at $-71.62 < x < -71.41$ cm, $-47.61 < x < -47.40$ cm, $-23.61 < x < -23.40$ cm, $0.40 < x < 0.61$ cm, $24.40 < x < 24.61$ cm, $48.41 < x < 48.62$ cm. These pads contain ^{10}B , which has a propensity of absorbing thermal neutrons. To corroborate this notion, reintegrating the stainless steel basket structure and neutron absorbing pads is expected to capture the depressions not found in the previous models. Again, comparing the mfp of neutrons in stainless steel 304, the neutron absorbing pads, and fuel rods in Fig. 4-5 shows the mfp is dominated by the absorbing component at a level of approximately 10cm (or less, depending on the energy of the incident neutrons). These mfp's are similar to the physical thickness of the stainless steel, neutron absorbing pads, and fuel in the MPC. Therefore, the neutrons will undergo an appreciable number of

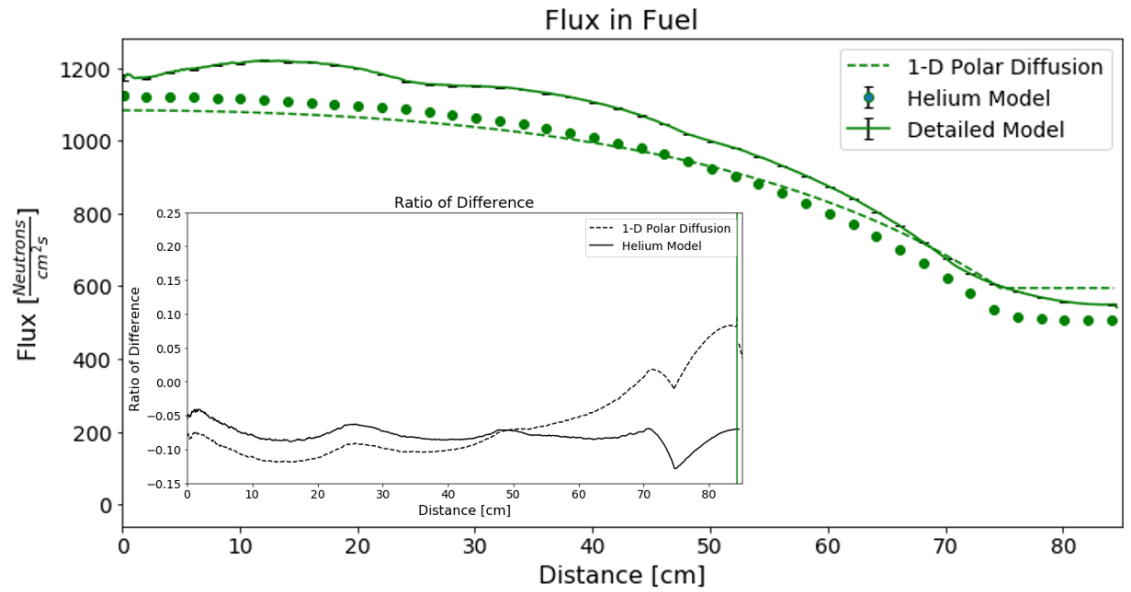


Figure 4-4. The neutron flux spatial distribution simulated by the helium model (circles) captures the neutron flux spatial distribution flattening out in the detailed model (solid line) over the 20 cm region before exiting the fuel region. The diffusion approximation (dotted line) also captures the flux flattening near 65 cm from the cask centerline after adding a helium annulus for neutron streaming.

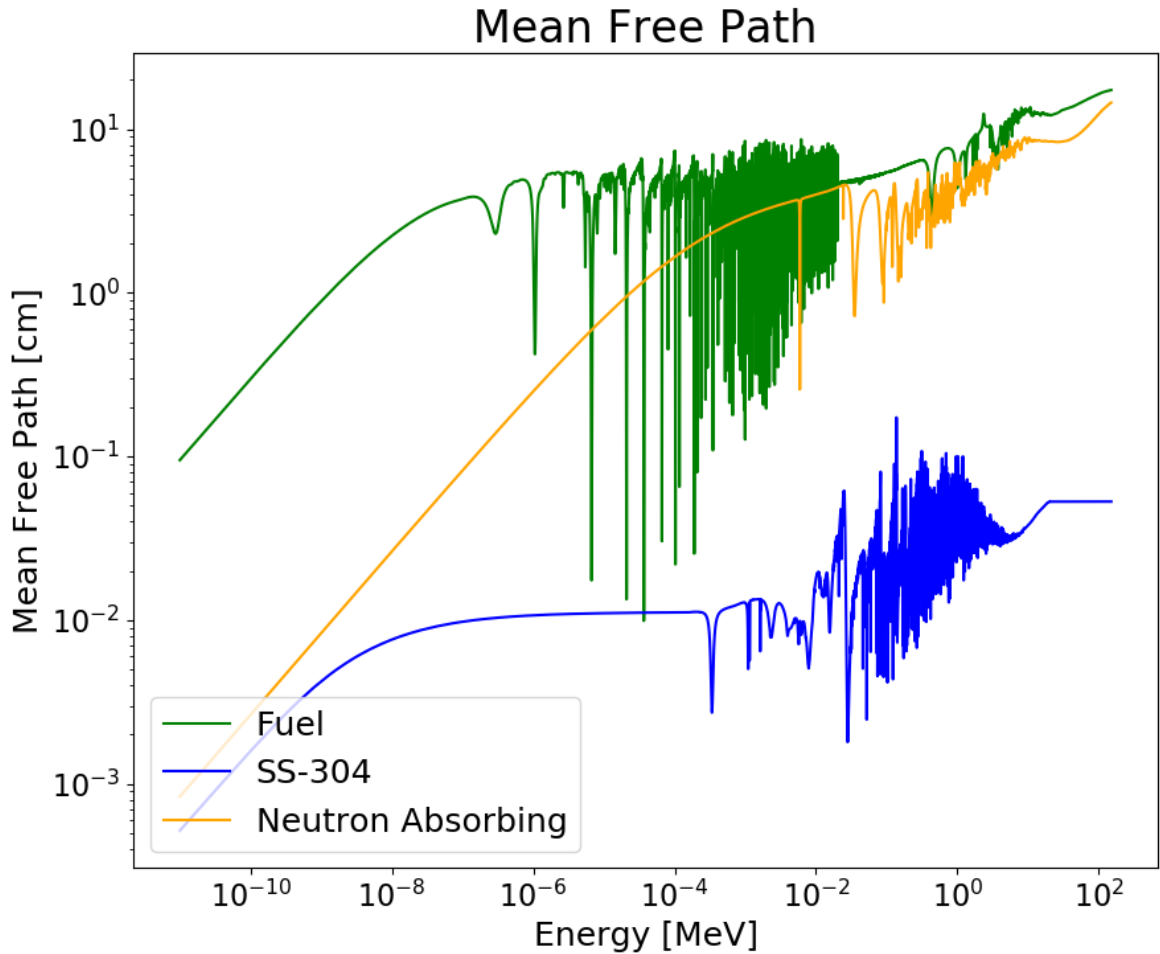


Figure 4-5. The mean free paths for stainless steel 304 (blue), neutron absorbing pad material (orange), and fuel pin material (green). These three mean free paths are similar to the physical thicknesses of each material implying that the steel and neutron absorbing pads need to be included in MCNP simulations as discrete materials instead of being incorporated into the homogenized fuel.

interactions in the stainless steel and neutron absorbing materials. However, unlike in the fuel, no neutrons are being generated in the steel and neutron absorbing materials, and so the flux is expected to decrease therein.

Another MCNP model is developed to describe the cause of the depressions, Fig. 4-6.

This multi-layered model is called the “1-D basket model” and represents a single row

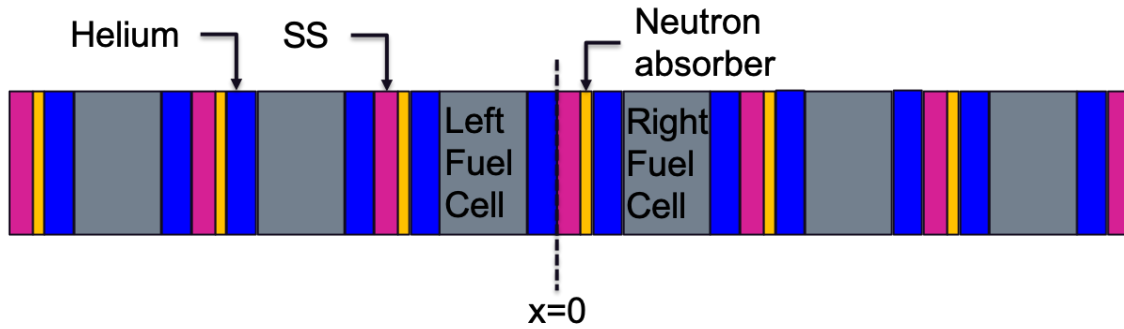


Figure 4-6. The 1-D basket model used to identify the cause of the small depressions. The model is repeating layers of stainless steel (pink), neutron absorbing pads (orange), helium (blue), and cell homogenized fuel (gray).

of fuel cells from the detailed model with one difference: the volume attributed to fuel materials. In this model, the interior volume of each fuel cell contains a cell homogenized fuel composition with helium on both sides and neutron absorbing pad to the left. The cell homogenized fuel composition is determined using the mass fraction of materials which comprise the 264 fuel rods and helium between the fuel rods in each cell. The volume of the cell homogenized fuel material is defined to be equal to the volume of a single fuel bundle.

The simulated neutron flux spatial distribution through the 1-D basket model is shown in Fig. 4-7. The simplified basket model has six small depressions present in the flux around ± 25 cm, ± 50 cm, ± 75 cm. These depressions correspond to a 1-2 % local reduction in the flux, which is similar in location and magnitude to the depressions present in the simulated neutron flux spatial distribution in the detailed model. The depressions in the neutron flux spatial distribution occur within the stainless steel and neutron absorbing pad materials. The flux increases in the fuel as neutrons are born from spontaneous fission decays and (α, n) reactions. The combination of the absorption events in the neutron absorbing pads and source events in the fuel cause the depressions observed in the neutron flux spatial distribution.

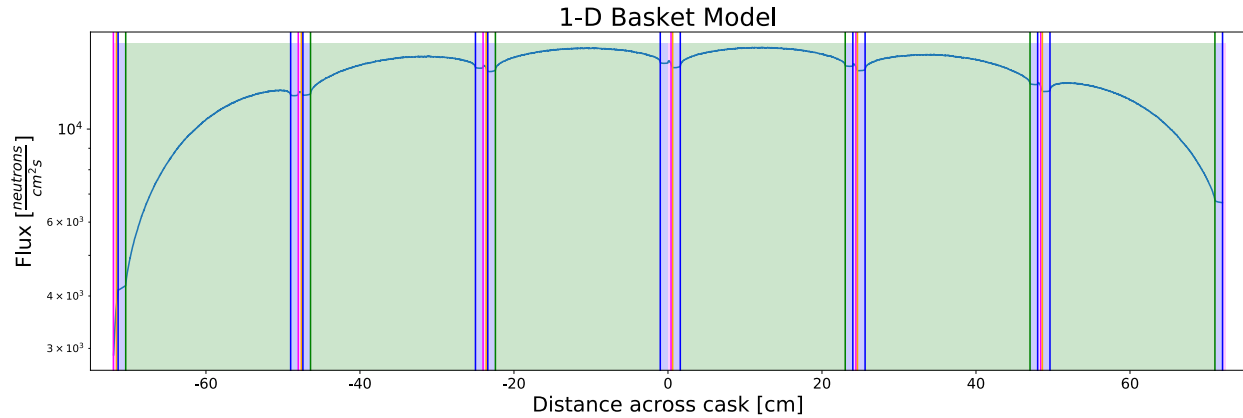


Figure 4-7. The neutron flux spatial distribution simulated from the 1-D basket model. The colors are representative of each material: stainless steel 304 (pink), neutron absorbing pad (orange), helium (blue), and cell homogenized fuel (green). There are depressions present in the flux which occur within the stainless steel and neutron absorbing pads.

4.2.4 Azimuthally Asymmetric Flux

The final feature, the flux asymmetry (seen in Fig. 2-24), is also explained using the 1-D basket model. The detailed model shows a higher flux leaving the bottom right of section of the cask as compared to the top left section of the cask. This discrepancy is seen at the leftmost and rightmost exiting surfaces in Fig. 4-7. The leftmost face has a lower exiting flux value than the value observed at the rightmost face. Figure 4-6 shows the reason for the asymmetry: a neutron born in the left fuel cell and traveling left will pass through three neutron absorbing pads before exiting the left face, which is the same number of neutron absorbing pads that same neutron would have to pass through if it were traveling right. Conversely, if a neutron is born in the right fuel cell and traveling to the left, it passes through four neutron absorbing pads. However, if that same neutron were to travel right, it only potentially encounters two neutron absorbing pads. The number of neutron absorbing pads a neutron potentially encounters is not the same based on the the location of neutron generation and direction of travel because of the placement of neutron absorbing pads in the MPC. The asymmetric loading of these pads directly affects the neutron flux spatial distribution exiting the spent fuel cask.

To further corroborate this notion, the detailed model was adjusted, replacing the stainless steel structure and neutron absorbing pads with vacuum. Figure 4-8 compares the ratio of the neutron flux spatial distribution averaged over the top left section and the flux averaged over the bottom right section from the detailed model where one simulation replaced neutron absorbing pads with vacuum and the original detailed model. The maximum deviation of the ratios of neutron flux spatial densities is 0.1% as a result of replacing non-fuel structure in the MPC with vacuum, confirming the results from the basket model. In contrast, the maximum deviation of these same ratios in the original detailed model is nearly 10%.

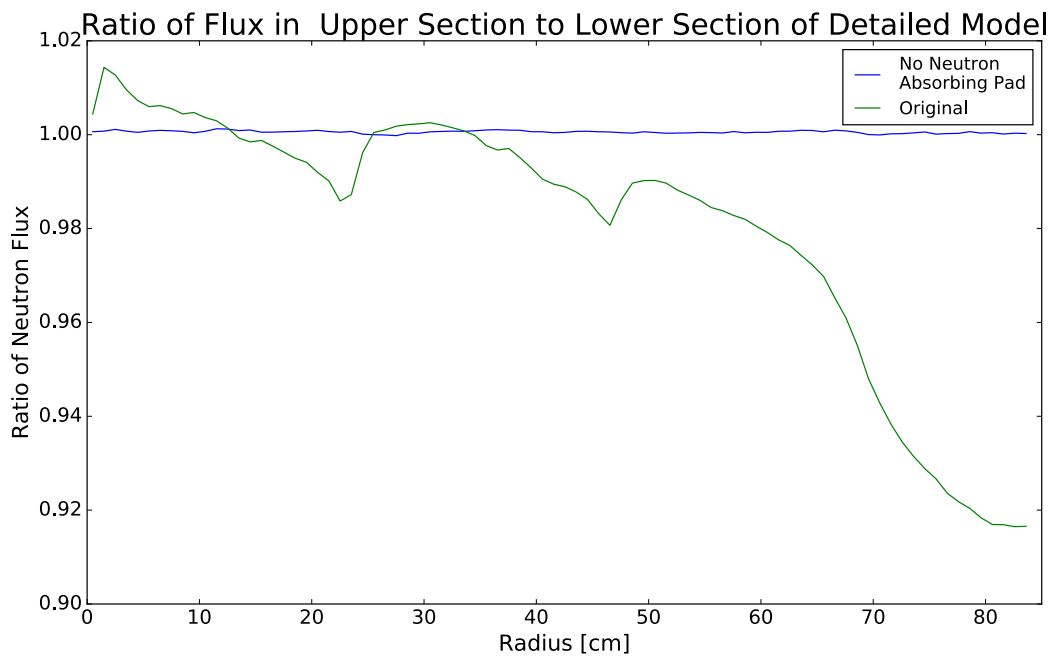


Figure 4-8. The ratio of the neutron flux spatial distribution in the upper left section of the fuel region to the neutron flux spatial distribution in the lower right section of the fuel region. This ratio is nearly 1 over the entirety of the fuel region, confirming the assumption that removing the neutron absorbing pads removes the previously identified depressions.

Previous findings have shown that geometric structures finer than the stainless steel baskets, neutron absorbing pads, and helium annulus are unnecessary for characterization of the spatial flux distribution arising from the detailed model. A final model, the

“cruciform model”, is developed to ensure no important physics are neglected in the reduced-order modeling and analysis process. This model uses the cell homogenized fuel definition in each of the 32 original fuel cells. In doing so, the stainless steel fuel basket and neutron absorbing pads are retained and discrete from the homogenized fuel. The helium surrounding the 32 fuel cells is also retained.

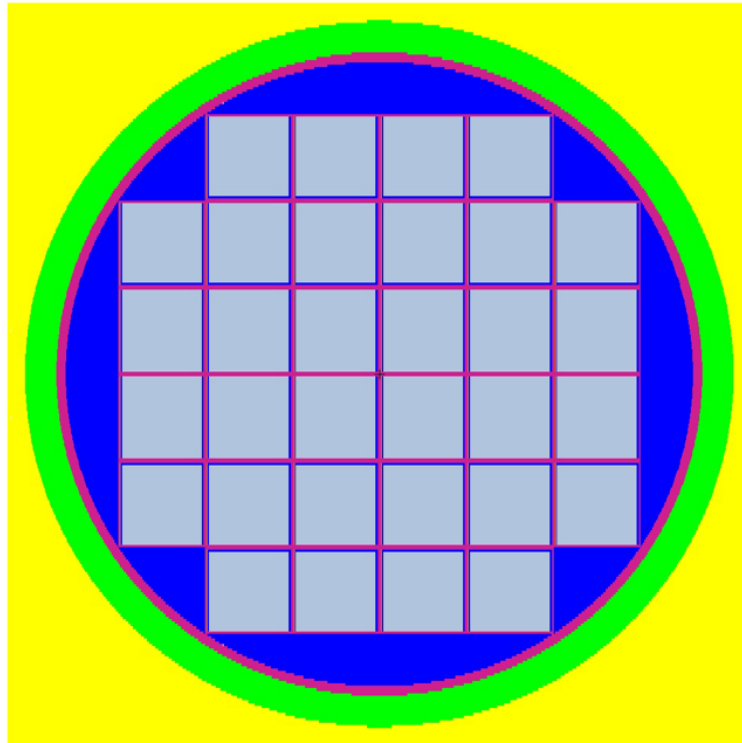


Figure 4-9. The cruciform model. The gray squares are cell homogenized fuel, the stainless steel fuel basket and MPC are pink, the helium annulus is blue, the air exterior to the MPC is green, and concrete is yellow. The neutron absorbing pads (orange) are present in this diagram, but are too thin to be seen here.

The neutron spatial flux distribution simulated by the cruciform model is shown in Fig. 4-10. These results underpredicts the flux from the detailed model by 5-7% through the entire fuel region, including in the helium annulus. Moreover, these results can also be interpreted as the cruciform model accounting for the physics relevant to the detailed model’s spatial neutron flux distribution at a level greater than 90%. That is, further fine detail additions to the cruciform model will “close the gap” with respect to the detailed model at a sub-10% level.

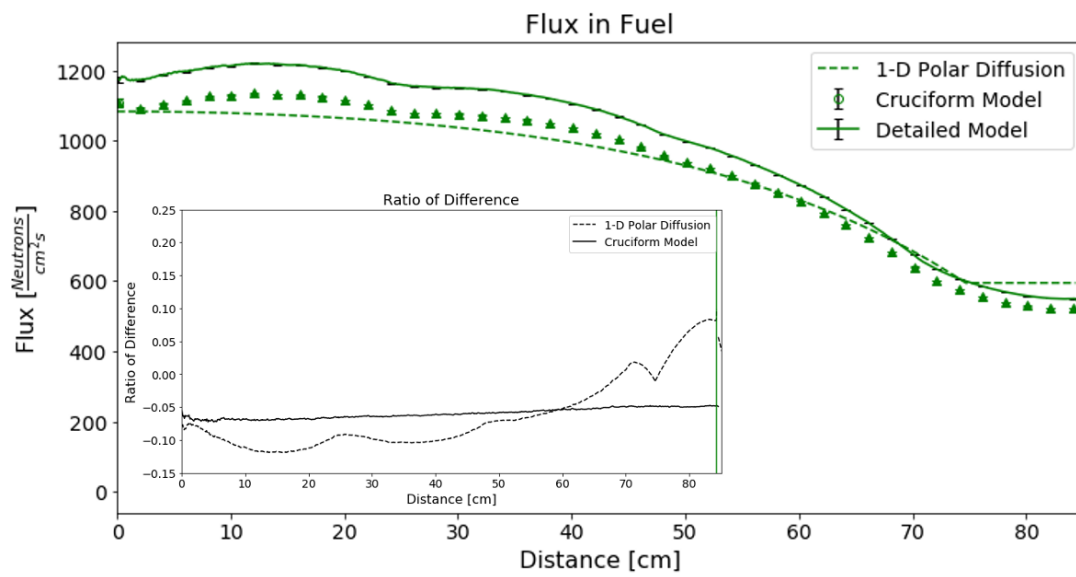


Figure 4-10. The neutron flux spatial distribution of the cruciform model (triangles) capture the flatness of, the leveling off of, and the depressions in the neutron flux spatial distribution seen in the detailed model (solid).

4.2.5 Non-exponential Decay in Concrete

The hydrogen content in concrete is responsible for thermalizing the neutron flux an attenuating neutrons. Figure 4-11 compares the neutron flux from the detailed model (solid line), the E_2S_2 analytic solution (red dashed), the fast energy group E_2S_2 solution (blue dotted), the thermal energy group E_2S_2 solution (brown dotted), and the MCNP helium model (dotted). In concrete, the neutron flux experiences a shift in energies as a result of downscattering occurring on hydrogen atoms. The analytic solutions confirm the observed shift in energies. The fast flux (the blue dotted line) decreases exponentially through the concrete regions. Intuitively, the exponential decrease is behaves similarly to an uncollided flux calculation, where the uncollided neutron flux decreases exponentially with thickness as neutrons undergo interactions in a material. In the case of concrete these interactions are mainly scattering since the scattering ratio ($\frac{\Sigma_s}{\Sigma_t}$) in the fast region for concrete is 99.5%. A high scattering ratio at fast neutron energies breeds thermal neutrons, a conclusion consistent with the initial increase in the thermal neutron flux in Fig. 4-11. As the fast neutron population decreases, the rate at which neutrons are thermalized decreases as well, which when combined with loss terms, causes the populations of both the fast and thermal neutron fluxes to decrease as a function of thickness. Both the analog MCNP model and the analytic model capture the physics of the detailed model within 10%, with the exception of the last 6cm of the analytic model.

The reason the analytic model shows higher disagreement with the detailed model in the outer 6cm is a result of the boundary conditions. The E_2S_2 equations are solved using a continuous flux boundary condition at both surfaces of the model. While considering the neutron flux as continuous is a physically consistent boundary condition, higher order effects (e.q., continuity of derivatives) are not being considered. Further, the outermost boundary condition assumptions that no neutrons will re-enter the cask after leaving.

While this assumption is nearly physically constituent, it will still act as source of error to materials within the cask.

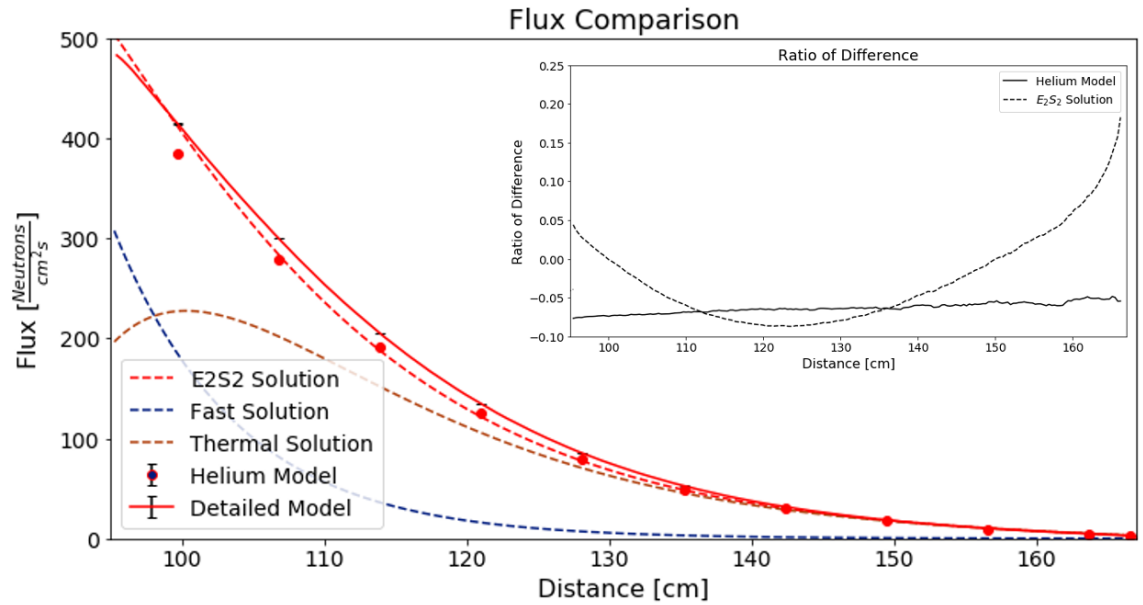


Figure 4-11. The neutron flux spatial distribution of the analytic E_2S_2 model (dashed line), helium model (circles), and detailed model (solid lines). The fast and thermal portions of the E_2S_2 solutions are shown in the blue and brown dotted lines respectively. The inset graphs shows the error between the analog models and detailed model.

4.2.6 Flux in MPC and Carbon Steel Shell

The MPC and carbon steel shell are the final material regions left to discuss. The thinness of these materials leads to a relatively simple discussion. Figure 4-12 compares

1415 the neutron flux from the detailed model (solid blue), the E₂S₂ model solution (dotted
1416 blue line), and the analog helium model (circles). The fast and thermal components of the
1417 E₂S₂ solution are displayed as the dark blue and brown lines respectively. Even though the
1418 thickness of the stainless steel is a similar to the MFP, some of the fast neutrons undergo
1419 scattering interactions and thermalize which results in an increase in the thermal flux.
1420 The error between the analog models and the detailed model is less than 10%. In fact, the
1421 analytic model agrees with the detailed model within 5%, which is better than the helium
1422 model.

1423 Figure 4-13 shows the neutron flux in the carbon steel shell. The flux in the carbon
1424 steel shell is almost entirely thermal since the concrete has already thermalized the
1425 neutron flux. The analytic model captures this behavior, unfortunately, the analytic model
1426 does not capture an increase in source source neutrons in the carbon steel which was
1427 observed in the detailed model. This is a result of assuming the number of neutrons bred
1428 through interactions is negligible and not including these fast neutrons in the E₂S₂ model.
1429 However, the analytic model agrees within 10-40% over the thickness of the carbon steel.
1430 Overall, this level of agreement is acceptable since the neutron flux is so small, in fact, the
1431 flux at the exiting surface of the cask is $0.68 \frac{1}{cm^2s}$ as predicted by the detailed model and
1432 $0.91 \frac{1}{cm^2s}$ as predicted with the E₂S₂ solution. The scale of the neutron flux is low making
1433 the larger error values acceptable. The error between the detailed and analytic models
1434 increases through the carbon steel shell as a result of the boundary conditions. At the
1435 exiting surface of the cask, the analytic solution is assumed to have a vacuum boundary
1436 condition. Meaning, none of the neutrons which leave the cask will return. While this
1437 assumption is appropriate (since the cask in the detailed model is surrounded by air), it
1438 does not exactly replicate the conditions in the detailed model. The result is an increase in
1439 error values near the outer surface of the cask.

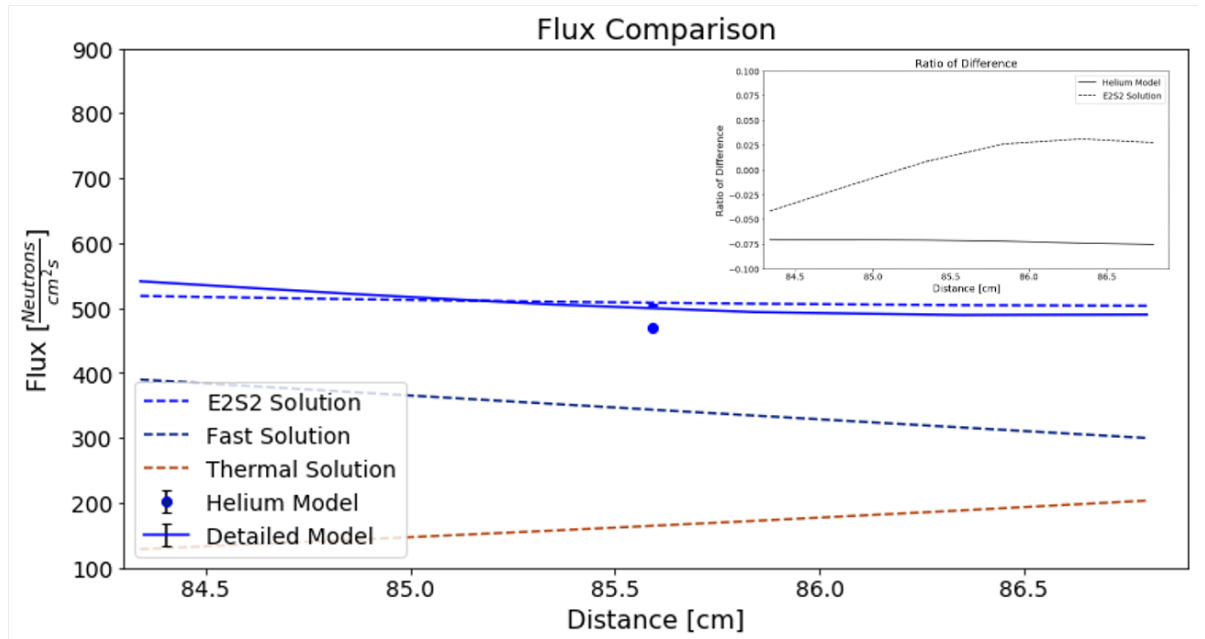


Figure 4-12. The neutron flux spatial distribution of the analytic E₂S₂ model (dashed line), helium model (circles), and detailed model (solid lines). The fast and thermal portions of the E₂S₂ solutions are shown in the blue and brown dotted lines respectively. The inset graphs shows the error between the analog models and detailed model.

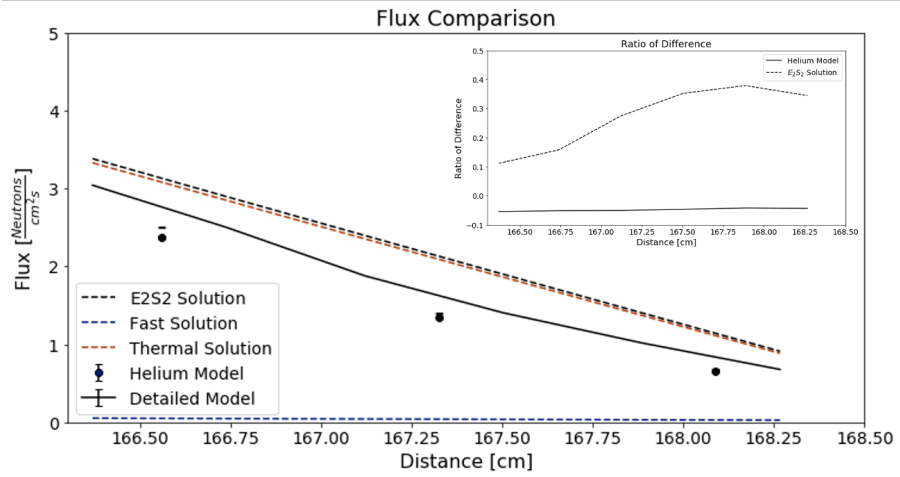


Figure 4-13. The neutron flux spatial distribution of the analytic E₂S₂ model (dashed line), helium model (circles), and detailed model (solid lines). The fast and thermal portions of the E₂S₂ solutions are shown in the blue and brown dotted lines respectively. The inset graphs shows the error between the analog models and detailed model.

4.3 Summary

Using reduced complexity analytic and computational models to analyze the simulation results of a high-fidelity computational model allows for the quantification of effects of any assumptions invoked when developing the latter model. Ensuring important physics are preserved in the course of conducting simulations increases the likelihood of correct results. This work exemplified this notion through a process referred to as "simulation results assessment." As a demonstration, this work included post-simulation analysis of a detailed MCNP model of a HI STORM 100 spent nuclear fuel cask. A series of reduced analytic and computational models were developed and used to identify the physics which causes features in the neutron flux spatial distribution as calculated by the detailed model. In the HI-STORM 100 model, the stainless steel basket, neutron absorbing pads, and helium annulus around the fuel cells are important physical components that need to be preserved in modeling. Retaining the individual fuel pin structure was found to be less important than broadly capturing the lumped material properties inside the individual fuel cells. These results were corroborated using the cruciform model, which appears to capture the physics relevant to the neutron flux spatial distribution in the detailed model beyond the 90% level. The major features of the neutron flux spatial distribution simulated by the detailed model are expected to be

1458 correct since the this model preserves material fuel properties and the geometric structure
1459 of the neutron absorbing pads and helium annulus.

CHAPTER 5

SENSITIVITY ANALYSIS

From a practical standpoint, known functional forms (as opposed to numerical tables) are optimal for both exhaustive insight into essential phenomenology and broader application, especially if those forms are differentiable or integrable. To this point, as noted by Barenblatt,

“... for a long time ... [these] solutions were treated by most researchers as though they were merely isolated ‘exact’ solutions to special problems: elegant, sometimes useful, but extremely limited in significance. It was only gradually realized that these solutions were actually of much broader significance ...”

and moreover, by Polyanin and Zaitsev,

“... exact solutions of differential equations play an important role in the proper understanding of qualitative features of many phenomena and processes in various areas of natural science ...”

and finally, by Sachdev,

“... the search for exact solutions is now motivated by the desire to understand the mathematical structure of the solutions, and hence, a deeper understanding of the physical phenomena described by them. Analysis, computation, and not insignificantly, intuition all pave the way to their discovery ...”

As such, even in an age dominated by computational studies there remains a distinct role for the development and implementation of analytical treatments. As summarized by Sachdev,

“... understanding the validity and place of exact/approximate analytical solution[s] in the general context can be greatly enhanced by numerical simulation. In short, there must be a continuous interplay of analysis and computation if a ... problem is to be successfully tackled.”

Now-ubiquitous quantitative code verification efforts may thus be viewed as a core element within a broader program of study; chapter 4 features a complementary analytical study to aid in the understanding of essential phenomenologies underpinning increasingly complicated computational science simulations. The remainder of chapter 5 motivates sensitivity analysis in the broader context of the results assessment methodology as a validation technique and introduces fundamental theory.

5.1 Motivation of Sensitivity Analysis

As characterized by Saltelli et al.,

“... sensitivity analysis is aimed ... at priority setting, to determine what factor most needs better determination, and to identify the weak links of the assessment chain (those that propagate most variance in the output) ...”

Sensitivity analysis is a prime example of a field that has in recent years been almost wholly subsumed by purely computational endeavors; the literature is voluminous surrounding site-specific efforts and implementations within physics, engineering, biology, earth science, population dynamics, economics, and many other areas. This state of affairs is also reflected in the expositions of both the canonical primers and various critiques surrounding the subject.

The evolution of sensitivity analysis from its historical and largely analytical roots to modern-day computational programs of study has largely proceeded in tandem with parallel developments in large-scale computational science. The reasons for this outcome include but are not necessarily limited to:

1. Sensitivity analysis has emerged as one of the cornerstone processes through which generalized mathematical models or codes may be assessed through an integrated program of verification, validation, and uncertainty quantification. Modern-day sensitivity studies are therefore most commonly encountered in the context of code evaluations.

2. Fully coupled, global sensitivity analyses (as opposed to their “one factor at a time” or purely local or derivative-based counterparts) are increasingly viewed as necessary components of fully rigorous code assessment strategies. These techniques typically demand repeated code execution under coupled sampling spanning the entire space of possible parameter realizations.
3. Tremendous advances in the power and widespread availability of high-performance computing resources has made conceptually and computationally simple “brute-force method” sensitivity analysis approaches more viable than ever before. In these studies, even large or otherwise complicated codes can be rapidly, repeatedly executed under conjoined parameter sampling.

In these often-necessary realizations of sensitivity analysis practices, complementary analytical studies assume the same historical relevance and underpinning basis, limitations, and advantages as for the attendant mathematical models.

A classical example of this phenomenology with broad relevance to the nuclear engineering community is adjoint analysis, as detailed among many others by Keepin, Henry, and Lewins. As noted by Lewins,

“For sensitivity and uncertainty analysis, one frequently is interested in a formulation relating the change in a given performance parameter to many different alterations in the system input or design variables.[\[38\]](#)”

Further, Stacey, Greenspan, and Lewins and Becker discuss the appropriateness of analytical adjoint based approaches in nuclear engineering, since adjoint sensitivity calculations are best suited for applications when one is interested in the change in response to many input parameters [\[38–40\]](#). Nuclear engineering is not unique in that a desired response (e.g., the neutron flux) depends on many input parameters (e.g., nuclear data). The extent of these problems have made for a desirable application of many modern-day computational approaches for executing adjoint-based sensitivity analysis studies. However, if an analytic model has a closed-form solution, then solving the system

from the forward direction is more appropriate since this process is considerably more straightforward and does not require knowledge of the adjoint solution.

Closely related to analytical adjoint methods is Cacuci’s “forward sensitivity analysis procedure” (FSAP) based on the concept of the Gâteaux generalized directional derivative. This formalism (if not its practical implementation) is entirely analytical, and often features minimal computational overhead; in turn, however, it only provides decoupled sensitivity information on the local or first-order level. As such, when implemented in the context of analytical mathematical models, Cacuci’s FSAP shares the same general drawbacks and benefits associated with analytical modeling in general.

In short, the potentially narrow scope of analytically computed sensitivity information is counterbalanced by a more complete and lucid functional representation that may prove informative of more general developments. This important idea forms the motivation of this work, to be initiated in a specific context with relevance to the nuclear engineering community.

The notions previously set forth are entirely general, and may be applied in any number of contexts. Of particular relevance to this study is the implementation of the aforementioned techniques in the context of modeling and simulation of containers intended for the long-term storage of spent nuclear reactor fuel.

Complementary analytical sensitivity analysis modeling of spent fuel cask scenarios is intended to serve two essential purposes:

1. These studies can serve as guides for targeting future, entirely computational sensitivity analysis studies pertaining to spent fuel cask scenarios, thus potentially achieving computational cost savings where necessary.
2. Analytical studies may also serve as guides for interpreting, understanding, and rigorizing certain results of existing and future computational studies pertaining to spent fuel casks.

The primary objective of this work is therefore to execute an analytical sensitivity analysis study along the analytic models identified in Chpt. 4. In support of this objective, Sec. 5.2 provides a discussion of the the general concepts of sensitivity analysis as well as a detailed description of the methods used in this work. Chapter 6 applies appropriate sensitivity analysis methods on the analytic and computational models. Chapter 7 compares the sensitivity analysis results between the analytic and computational models and discusses implications of those comparisons.

5.2 Local Sensitivity Analysis Primer

Saltelli, Chan, and Scott define sensitivity analysis as the study of “relationships between information flowing in and out of a model[41].” That is, sensitivity analysis investigates how perturbations in input parameter values influence a system’s response, where input parameters are data values passed by the user or calculated by a model and are used in the calculation of output variables. The most common input parameters appearing in nuclear engineering models are cross sections, which are derived from material properties supplied by a model or code user. In order to better understand the general process of sensitivity analysis, Oblow and Pin provide a short description of the procedure [42]. To begin, consider the set of linear equations

$$\mathbf{R} = \mathbf{F}(\mathbf{y}, \boldsymbol{\alpha}), \quad (5-1)$$

where

\mathbf{R} is a vector of the system responses,

\mathbf{F} is a vector of the model equations (e.g., vector containing the diffusion equation),

\mathbf{y} is the state vector (e.g., vector of ϕ values),

$\boldsymbol{\alpha}$ is the vector of the system parameters,

where the vector \mathbf{F} can also represent nonlinear model equations, however, the following discussion is limited to linear equations for the purpose of this work.

Local sensitivity information describes first-order sensitivities, meaning, the sensitivity information is related to the first derivative of \mathbf{R} , (e.g. , $\frac{\partial \mathbf{R}}{\partial \boldsymbol{\alpha}}$). Further, the first derivative describes the ratio of change in a system's response caused by changing the value of a parameter [43]. Hence, taking the derivative of Eqn. 5-1 over each parameter, $\boldsymbol{\alpha}_i$, independently yields

$$\frac{dR}{d\alpha_i} = \frac{\partial \mathbf{F}}{\partial \mathbf{y}} \frac{d\mathbf{y}}{d\alpha_i}. \quad (5-2)$$

Since \mathbf{F} contains the analytic models described by the user, the value $\frac{\partial \mathbf{F}}{\partial \mathbf{y}}$ can be calculated directly. Simplifying the final derivative in Eqn. 5-2 requires using the chain rule on Eqn. 5-1 to arrive at

$$\frac{d\mathbf{y}}{d\alpha_i} = \frac{\partial \mathbf{F}}{\partial \mathbf{y}} \frac{d\mathbf{y}}{d\alpha_i} + \frac{\partial \mathbf{F}}{\partial \boldsymbol{\alpha}} \frac{d\boldsymbol{\alpha}}{d\alpha_i}. \quad (5-3)$$

Re-expressing Eqn. 5-2, using Eqn. 5-3, yields the sought after sensitivity information $\frac{d\mathbf{R}}{d\boldsymbol{\alpha}}$. However, this approach can be algebraically involved since it requires solving the set of equations \mathbf{F} for each parameter variation.

In response to this problem, Cacuci developed a method for determining sensitivity information for all parameters simultaneously, given the function \mathbf{F} has a solution [13]. Cacuci utilizes the Gâteaux differential (G-differential), a form of the directional derivative, to find the differential value corresponding to each parameter simultaneously.

In this paradigm, the unperturbed response value (the value of the response function where all parameters are unperturbed) is defined as

$$\mathbf{R}(\mathbf{e}^0), \quad (5-4)$$

where $\mathbf{e}^0 = (\mathbf{y}^0, \boldsymbol{\alpha}^0)$ and the superscript 0 denotes the nominal, or unperturbed, value. If, moreover, the vector \mathbf{h}_α contains the perturbation values for M number of parameters as

$$\mathbf{h}_\alpha \equiv (\delta\alpha_1, \delta\alpha_2, \dots, \delta\alpha_M). \quad (5-5)$$

1610 Sensitivity information of the response function caused by the variations \mathbf{h}_α is found
 1611 by taking the G-differential, $\delta \mathbf{R}(\mathbf{e}^0; \mathbf{h})$, of the response function, where \mathbf{h} is the
 1612 concatenation of the perturbed parameter values and the perturbed state values;

$$\mathbf{h} \equiv (\mathbf{h}_y, \mathbf{h}_\alpha). \quad (5-6)$$

1613 Taking the G-differential of the response thus yields

$$\delta \mathbf{R}(\mathbf{e}^0; \mathbf{h}) \equiv \left. \frac{d}{d\epsilon} [\mathbf{R}(\mathbf{e}^0 + \epsilon \mathbf{h})] \right|_{\epsilon=0} = \lim_{\epsilon \rightarrow 0} \frac{\mathbf{R}(\mathbf{e}^0 + \epsilon \mathbf{h}) - \mathbf{R}(\mathbf{e}^0)}{\epsilon}, \quad (5-7)$$

1614 where ϵ is interpreted as an infinitesimal deviation from the nominal value of a given
 1615 parameter, and the rightmost expression is the definition of the G-derivative. In general,
 1616 the evaluated result of Eqn. 5-7 can be written as

$$\delta R(\mathbf{e}^0; \mathbf{h}) = \sum_i^M \eta_i \delta \alpha_i, \quad (5-8)$$

1617 where η_i contains sensitivity information for the parameter α_i . The values of η_i are used
 1618 to calculate the sought after sensitivity coefficients, which provide a relative comparison
 1619 between parameters. The sensitivity coefficients are thus calculated using $\delta \mathbf{R}$ as

$$S_{\alpha_i} = \frac{\delta R}{\delta \alpha_i} \frac{\alpha_i}{R(\mathbf{e}^0)} = \eta_i \frac{\alpha_i}{R(\mathbf{e}^0)}, \quad (5-9)$$

1620 where S_{α_i} is the sensitivity coefficient for parameter α_i [44].

1621 The sensitivity coefficients are used to determine the the “importance” of each
 1622 parameter. Parameters with larger sensitivity coefficients have a larger impact on the
 1623 the system response. The signs of sensitivity coefficients is also important, as the signs
 1624 indicate the direction of change in the response given a change in a parameter. Meaning,
 1625 if the sensitivity coefficient has a negative value for a given parameter, increasing the value
 1626 of that parameter will cause the value of the response to decrease. On the other hand, if
 1627 the sensitivity coefficient has a positive value, increasing the associated parameter value
 1628 will cause an increase in the response value.

Chapter 6 investigates the effects of perturbing parameters relating to nuclear data in
the solution to Eqn. 4-9, the solution to the 1D cylindrical diffusion equation.

CHAPTER 6
SENSITIVITY THEORY OF REDUCED PHYSICS MODELS

6.1 Local Sensitivity Analysis of Representative Spent Fuel Cask Model

6.1.1 Fuel Region

Section 3 introduced the diffusion approximation which uses experimental data in the form of cross sections to predict the neutron flux through the fuel region of the cask. Taking, the solution to Eqn. 3-46 with the boundary conditions given in Eqns. 4-1 and 4-2 is

$$\phi^0(r) = \frac{S^0}{D^0(B^0)^2} \left(1 - \frac{I_0(B^0 r)}{I_0(B^0 \tilde{r}^0)} \right); \quad B^0 \equiv \sqrt{\frac{\Sigma_a^0 - \bar{\nu}^0 \Sigma_f^0}{D^0}}. \quad (3-49)$$

where S^0 is the intrinsic neutron source, I_0 is the modified Bessel function of the first kind, and \tilde{r}^0 is the extrapolated radius of the fuel region equivalent to $r_b^0 + d^0$. The superscript 0 denotes the nominal value of each input parameter or response function.

Identifying the unperturbed input parameters from Eqn. 3-49 as

$$\boldsymbol{\alpha}^0 \equiv (S^0, D^0, B^0, \tilde{r}^0), \quad (6-1)$$

and the perturbation vector, \mathbf{h}_α as

$$\mathbf{h}_\alpha \equiv (\delta S, \delta D, \delta B, \delta \tilde{r}), \quad (6-2)$$

the vector \mathbf{h}_u becomes

$$\mathbf{h}_u \equiv (\delta \phi). \quad (6-3)$$

Then, the vector of nominal input parameters and response functions is defined as

$$\mathbf{e}^0 \equiv (\phi^0(r), \boldsymbol{\alpha}_0), \quad (6-4)$$

where the response function is

$$R(\mathbf{e}^0) = \phi^0(r). \quad (6-5)$$

1647 Finally, determining the sensitivities for each input parameter using Eqns. 6-1-6-5 in Eqn.
1648 5-7 is equivalent to replacing each input parameter in Eqn. 3-49 with

$$\alpha_i^0 \rightarrow (\alpha_i^0 + \epsilon \delta \alpha_i). \quad (6-6)$$

1649 Using Eqn. 6-6 to expand the input parameters in Eqn. 3-49 gives

$$\delta R(\mathbf{e}^0; \mathbf{h}) = \frac{d}{d\epsilon} \left[\frac{(S^0 + \epsilon \delta S)}{(D^0 + \epsilon \delta D)((B^0 + \epsilon \delta B))^2} \left(1 - \frac{I_0((B^0 + \epsilon \delta B)r)}{I_0((B^0 + \epsilon \delta B)(\tilde{r}^0 + \epsilon \delta \tilde{r}))} \right) \right] \Big|_{\epsilon=0}. \quad (6-7)$$

1650 Evaluating Eqn. 6-7 yields

$$\delta R(\mathbf{e}^0; \mathbf{h}) = \eta_1(r) \delta S + \eta_2(r) \delta D + \eta_3(r) \delta B + \eta_4(r) \delta \tilde{r}, \quad (6-8)$$

where the r -dependent functions appearing in Eqn. 6-8 are defined by

$$\eta_1 \equiv \frac{1 - \frac{I_0(B^0 r)}{I_0(B^0 \tilde{r}^0)}}{(B^0)^2 D^0}, \quad (6-9)$$

$$\eta_2 \equiv \frac{-S^0}{(B^0)^2 (D^0)^2} \left(1 - \frac{I_0(B^0 r)}{I_0(B^0 \tilde{r}^0)} \right), \quad (6-10)$$

$$\eta_3 \equiv \frac{-2S^0 \left(1 - \frac{I_0(B^0 r)}{I_0(B^0 \tilde{r}^0)} \right)}{(B^0)^3 D^0} - \frac{S^0 r I_1(B^0 r)}{(B^0)^2 D^0 I_0(B^0 \tilde{r}^0)} + \frac{S^0 \tilde{r}^0 I_0(B^0 r) I_1(B^0 \tilde{r}^0)}{(B^0)^2 D^0 (I_0(B^0 \tilde{r}^0))^2}, \quad (6-11)$$

$$\eta_4 \equiv \frac{S^0 I_0(B^0 r) I_1(B^0 \tilde{r}^0)}{B^0 D^0 (I_0(B^0 \tilde{r}^0))^2}, \quad (6-12)$$

$$(6-13)$$

and the associated sensitivity coefficients are summarized as

$$S_{c,S} = \eta_1 \frac{S^0}{\phi(r)}, \quad (6-14)$$

$$S_{c,D} = \eta_2 \frac{D^0}{\phi(r)}, \quad (6-15)$$

$$S_{c,B} = \eta_3 \frac{B^0}{\phi(r)}, \quad (6-16)$$

$$S_{c,\tilde{r}} = \eta_4 \frac{\tilde{r}^0}{\phi(r)}. \quad (6-17)$$

Equation 3-49 indicates that some of the input parameters appearing within may be defined in terms of other, more fundamental input parameters, such as how Σ_a^0 appears in the definition of B^0 as well as in D^0 . In practice, the values for D^0 , B^0 , and \tilde{r}^0 are calculated from experimental data or geometry (in the case of \tilde{r}^0). Therefore, it is necessary to express each of the above input parameters according to their individual definitions using Eqns. 6-18 - 6-20:

$$D^0 \equiv \frac{1}{3(\Sigma_s^0 + \Sigma_c^0 + \Sigma_f^0)}, \quad (6-18)$$

$$B^0 \equiv \sqrt{\frac{\Sigma_a^0 - \bar{\nu}^0 \Sigma_f^0}{D^0}} = \sqrt{\frac{\Sigma_c^0 + \Sigma_f^0(1 - \bar{\nu}^0)}{\frac{1}{3(\Sigma_s^0 + \Sigma_c^0 + \Sigma_f^0)}}}, \quad (6-19)$$

$$\tilde{r}^0 \equiv r_b^0 + \frac{0.710}{(\Sigma_s^0 + \Sigma_c^0 + \Sigma_f^0)}, \quad (6-20)$$

1651 where Σ_c^0 is the nominal capture cross section and r_b^0 is the nominal cask fuel region outer
 1652 radius, and the nominal total absorption cross section is redefined using $\Sigma_a^0 \equiv \Sigma_c^0 + \Sigma_f^0$.
 1653 Fundamental sensitivity coefficient results written in terms of the parameters Σ_s , Σ_c , Σ_f ,
 1654 and r_b are then determined by applying the G-derivative to each of Eqns. 6-18-6-20 and
 1655 substituting the results into their respective places in Eqn. 6-9 - 6-12.

1656 Redefining the sensitivity coefficients for B , D , and \tilde{r} in terms of those for Σ_c , Σ_s , $\bar{\nu}$,
 1657 Σ_f , and r_b is a straightforward process similar to how the coefficients were found for B , D ,
 1658 and \tilde{r} above. Taking the G-derivative of each of Eqns. 6-18-6-20, each equation is redefined
 1659 to be expressible in the terms $\delta\Sigma_c$, $\delta\Sigma_s$, $\delta\bar{\nu}$, $\delta\Sigma_f$, and δr_b . These definitions are then used
 1660 in the sensitivity coefficients summarized in Eqn. 6-14 - 6-17 to yield the final expressions.

Applying Eqn. 5-7 to Eqns. 6-18-6-20 using the following definitions for \mathbf{e}^0 and \mathbf{h} ,

$$\mathbf{e}^0 \equiv (\phi^0, \Sigma_c^0, \Sigma_s^0, \bar{\nu}^0, \Sigma_f^0, r_b^0) \quad (6-21)$$

$$\mathbf{h} \equiv (\delta\phi, \delta\Sigma_c, \delta\Sigma_s, \delta\bar{\nu}, \delta\Sigma_f, \delta r_b) \quad (6-22)$$

yields

$$\delta D(\mathbf{e}^0; \mathbf{h}) = \frac{d}{d\epsilon} \left[\frac{1}{3((\Sigma_s^0 + \epsilon\delta\Sigma_s) + (\Sigma_c^0 + \epsilon\delta\Sigma_c) + (\Sigma_f^0 + \epsilon\delta\Sigma_f))} \right] \Big|_{\epsilon=0}, \quad (6-23)$$

$$\delta B(\mathbf{e}^0; \mathbf{h}) = \frac{d}{d\epsilon} \left[\sqrt{\frac{(\Sigma_c^0 + \epsilon\delta\Sigma_c) + (\Sigma_f^0 + \epsilon\delta\Sigma_f)(1 - (\bar{\nu}^0 + \epsilon\delta\bar{\nu}))}{\frac{1}{3((\Sigma_s^0 + \epsilon\delta\Sigma_s) + (\Sigma_c^0 + \epsilon\delta\Sigma_c) + (\Sigma_f^0 + \epsilon\delta\Sigma_f))}}} \right] \Big|_{\epsilon=0}, \quad (6-24)$$

$$\delta\tilde{r}(\mathbf{e}^0; \mathbf{h}) = \frac{d}{d\epsilon} \left[(r_b^0 + \epsilon\delta r_b) + \frac{0.7104}{(\Sigma_s^0 + \epsilon\delta\Sigma_s) + (\Sigma_c^0 + \epsilon\delta\Sigma_c) + (\Sigma_f^0 + \epsilon\delta\Sigma_f)} \right] \Big|_{\epsilon=0}. \quad (6-25)$$

Evaluating Eqns. 6-23-6-25 determines the variations δB , δD , and $\delta\tilde{r}$ as

$$\begin{aligned} \delta B = & \frac{\delta\Sigma_s\sqrt{3}(-\bar{\nu}^0\Sigma_f^0 + \Sigma_c^0 + \Sigma_f^0)}{2\sqrt{(-\bar{\nu}^0\Sigma_f^0 + \Sigma_c^0 + \Sigma_f^0)(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)}} - \\ & \frac{\delta\bar{\nu}\sqrt{3}\Sigma_f^0(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)}{2\sqrt{(-\bar{\nu}^0\Sigma_f^0 + \Sigma_c^0 + \Sigma_f^0)(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)}} + \frac{\delta\Sigma_c\sqrt{3}(-\bar{\nu}^0\Sigma_f^0 + 2\Sigma_c^0 + 2\Sigma_f^0 + \Sigma_s^0)}{2\sqrt{(-\bar{\nu}^0\Sigma_f^0 + \Sigma_c^0 + \Sigma_f^0)(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)}} + \\ & \frac{\delta\Sigma_f\sqrt{3}(-\bar{\nu}^0(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0) - \bar{\nu}^0\Sigma_f^0 + 2\Sigma_c^0 + 2\Sigma_f^0 + \Sigma_s^0)}{2\sqrt{(-\bar{\nu}^0\Sigma_f^0 + \Sigma_c^0 + \Sigma_f^0)(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)}}, \end{aligned} \quad (6-26)$$

$$\delta D = \frac{\delta\Sigma_c + \delta\Sigma_s + \delta\Sigma_f}{3(\Sigma_c^0 + \Sigma_s^0 + \Sigma_f^0)^2} \quad (6-27)$$

$$\delta\tilde{r} = \frac{0.7104\delta r_b}{\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0} - \frac{0.7104r_b^0(\delta\Sigma_c + \delta\Sigma_f + \delta\Sigma_s)}{(\Sigma_c^0 + \Sigma_f^0 + \Sigma_s^0)^2}. \quad (6-28)$$

These values are then substituted into Eqn. 6-8 in order to determine the sensitivity coefficients. Chapter 7 discusses the results of the sensitivity analysis.

6.2 Monte Carlo Based Sensitivity Analysis

CHAPTER 7 DISCUSSION OF SENSITIVITY ANALYSIS

Chpt. 6. derived the sensitivity coefficients for each analytic model used in the problem. This chapter will explain the trends in the sensitivity coefficients. Further, the results from the sensitivity coefficients are compared between the analytic and MCNP results.

7.1 Comparison of Results

The representative homogeneous fuel composition employed in the helium model may therefore be used to determine an associated set of nominal input parameters S^0 , Σ_c^0 , Σ_s^0 , $\bar{\nu}^0$, and Σ_f^0 for use with the analytical results appearing in Sec. 6.1.1, featuring an associated quantification of their relevance to the detailed model. Given fuel composition of the helium model, these input parameters are evaluated using the nuclear data processing code NJOY code [45], where the necessary calculations proceed by weighting the cross section values against the neutron source energy spectrum. Otherwise, the nominal input parameter r_b^0 is the radius of the homogenized fuel material, 74.68 cm. Table 7-1 provides a summary of parameter values calculated for the homogeneous fuel associated with the helium model.

Figure 7-1 depicts the sensitivity coefficients $S_{c,i}$ associated with the elemental parameters $i = S$, Σ_c , Σ_s , $\bar{\nu}$, Σ_f , and r_b appearing within the analytical model given by Eqn. 4-9, as calculated using Eqs. 6-9-6-12, 6-14-6-17, and 6-26-6-28 and the data appearing in Table 7-1. Several trends are immediately evident from Fig. 7-1:

Table 7-1. Summary of cross section data in the homogenized fuel.

Parameter	Values
S^0	20.1430 $\frac{\text{neutrons}}{\text{cm}^3\text{s}}$
Σ_c^0	0.0607976 $\frac{1}{\text{cm}}$
Σ_f^0	9.002E-3 $\frac{1}{\text{cm}}$
Σ_s^0	0.1032 $\frac{1}{\text{cm}}$
$\bar{\nu}^0$	2.6475 <i>neutrons</i>
r_b	74.68 <i>cm</i>

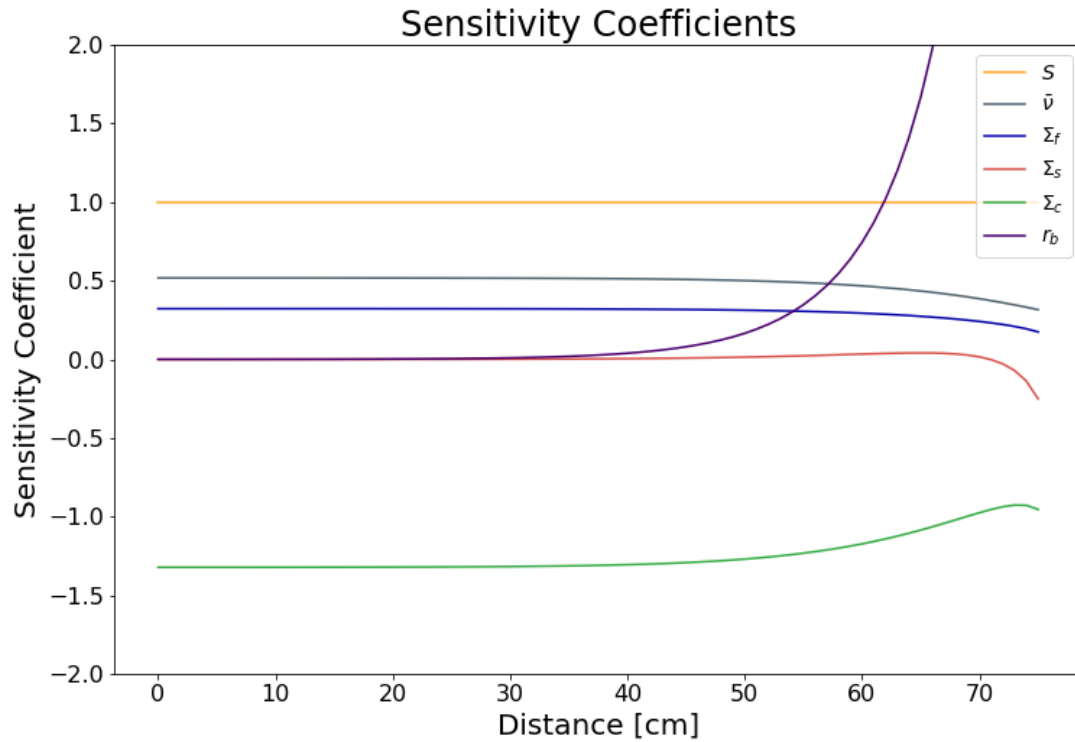


Figure 7-1. Analytical sensitivity coefficients as a function of the cylindrical radius in the homogenized fuel region.

- The sensitivity coefficient associated with the intrinsic neutron source term S is identically one since the source term itself appears simply as a scalar multiplier within Eqn. 4-9.
- The sensitivity coefficient associated with the capture cross section Σ_c is negative throughout the entire homogenous fuel region. This phenomenon indicates that as the capture cross section increases, the neutron flux decreases. This behavior is physically plausible since capture is a pure loss mechanism (i.e., as more neutrons are lost to capture, the value of the neutron flux becomes smaller). S_{c,Σ_c} has an inflection point and increases in value near 73 cm from the centerline, since loss terms are forcing the flux to meet to the boundary condition in Eqn. 4-2.
- The sensitivity coefficient of r_b exhibits the most dramatic change across the radius of the cask. In fact, the value increases to 13.238 at 74.68 cm. Perturbing r_b is effectually perturbing the location of the boundary condition, Eqn. 4-2. For this reason, S_{c,r_b} increases drastically from 40 cm to 74.78 cm since boundary conditions are imperative in constructing unique solutions. This also explains why the value is less than 0.04 for the first 40 cm, as the flux at these values is less affected by the boundary condition at 74.68 cm and more affected by the boundary condition at the

centerline, Eqn. 4-1. Finally, the values are positive since increasing the radius value would force the flux to remain at higher values through the radius of the fuel.

- Figure 7-1 shows that positive perturbations in $\bar{\nu}$ cause uniformly positive perturbations in the neutron flux. This trend is physically plausible since increasing the number of neutrons generated through fission events will increase the flux value throughout a multiplying material. Along these same lines, the sensitivity coefficient for the fission cross section Σ_f is also uniformly positive since increasing the likelihood of fission will in turn increase the number of neutrons in the homogeneous fuel material (i.e., as the number of neutrons available for transport increases, the flux increases). Moreover, while there appears to be a strong correlation between $S_{c,\bar{\nu}}$ and S_{c,Σ_f} as appearing in Fig. 7-1, the two coefficients are not identical since Σ_f appears decoupled $\bar{\nu}$ as part of its inclusion in the definition of D given by Eqn. 4-9.
- Otherwise, the sensitivity coefficients associated with Σ_f , $\bar{\nu}$, Σ_s , and Σ_c all have a similar shape: they are nearly flat for a majority of the cask's radial extent, before trending toward zero near the outer surface of the cask. This phenomenon is a consequence of all these terms appearing within the definition of B as given by Eqn. 4-9, which in turn controls the shape of the analytical neutron flux. The relationship between these input parameters demonstrates how the structure of the neutron flux is related to the structure of the sensitivity coefficients, since the G-derivative is a linear operator.
- The sensitivity coefficient associated with the scattering cross section Σ_s exhibits the most non-trivial behavior; it is positive and increasing for $r < 66.84$ cm, positive and decreasing for $66.84 \text{ cm} < r < 70.93$ cm, and negative for $r > 70.93$ cm to the cask outer radius. In turn, these features are indicative of the relative importance of a variety of gain and loss mechanisms occurring within Eqn. 4-9. In particular, for $r < 70.93$ cm neutron scattering serves a gain mechanism: it acts to spatially redistribute but otherwise preserve the neutron flux within the monoenergetic diffusion model (i.e., in the absence of thermalization). For $r > 70.93$ cm, neutron scattering is a loss mechanism: scattering in proximity to the outer boundary of the fuel region serves to increase leakage processes. The inflection point occurring at $r = 66.84$ cm is then indicative of the spatial location where the role of neutron scattering begins to transition: its presence owes to the approximate non-reentrant boundary condition given by Eqn. 4-2, which is intended to include leakage mechanics within the analytical diffusion model. That is, if the neutron flux was instead terminated at the physical extent of the fuel region, the analytical model would predict no neutron leakage and rather a zero neutron flux there. In this case, S_{c,Σ_s} would then be uniformly positive, which is clearly a non-physical result in the neighborhood of the cask outer boundary.

To further understand and better rank the importance of the various competing physical phenomenologies included in Eqn. 4-9, Fig. 7-2 depicts the absolute value of

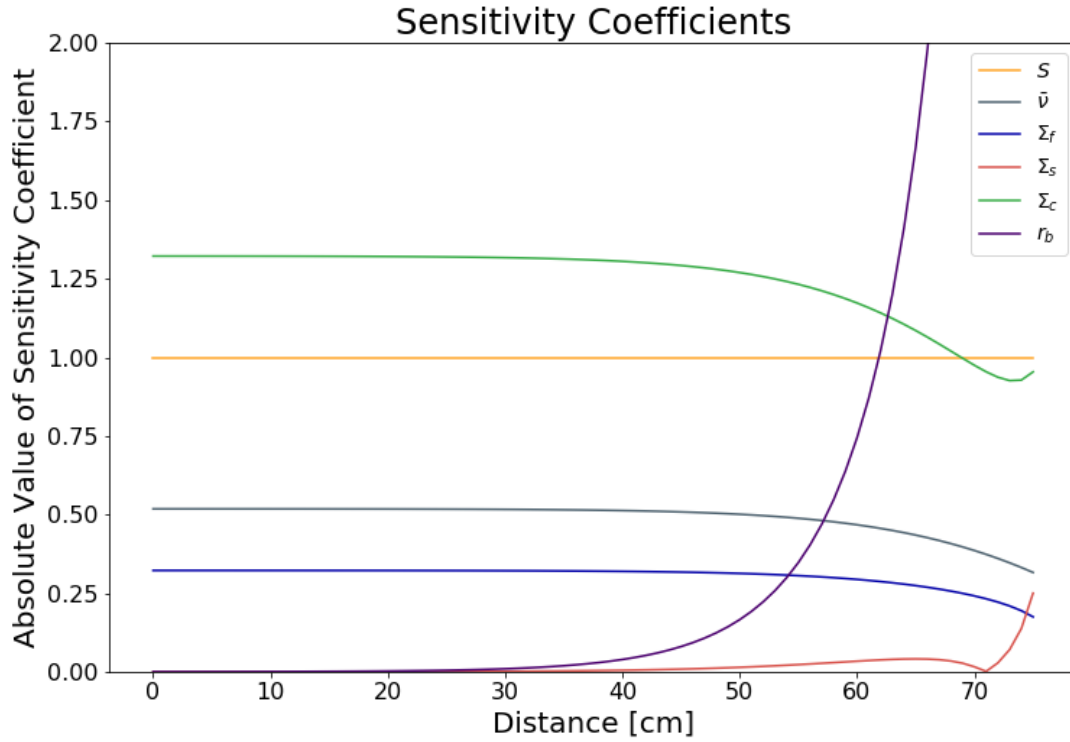


Figure 7-2. The absolute values of the sensitivity coefficients depicted in Fig. 7-1.

each sensitivity coefficient plotted in Fig. 7-1. Several additional trends are immediately evident from Fig. 7-2:

- For a majority of the cask radius, Σ_c is the most important input parameter; however, its importance drops near the cask outer radius as a result of the increase in S_{c,Σ_s} caused by leakage.
- For a majority of the cask radius, S is the second most important input parameter; however, near 60 cm, S_{c,r_b} quickly becomes the most important parameter and $S_{c,S}$ is briefly the third most important parameter before becoming the second the important parameter near 69 cm.
- For a majority of the cask radius, $\bar{\nu}$ and Σ_f are the third and fourth most sensitive parameters, respectively. However, the sharp increase in S_{c,r_b} relegates $\bar{\nu}$ and Σ_f to the fourth and fifth most important parameters near 55 cm.
- For the majority of the cask, r_b is the fifth most important parameter until approximately 35 cm where S, c, r_b increase and overtakes all the other parameters to become the most important parameter in the system.

- For a majority of the cask radius, Σ_s is the least important input parameter; however, it becomes the fourth most important parameter near the cask outer radius.

These importance trends manifest in Figs. 7-1 and 7-2 due principally to the r -dependent interplay between the capture and leakage loss mechanisms present in Eqn. 4-9. For example, capture is the dominant loss mechanism near the cask centerline, as shown in Fig. 7-1 a neutron initially located there is most likely to undergo many interactions before escaping from the cask outer surface. Conversely, leakage becomes an increasingly important loss mechanism near the cask outer radius, the importance of which is observed to eventually exceed that of capture. This physical interplay noticeably manifests in the behavior of S_{c,Σ_s} and S_{c,Σ_c} as depicted in Figs. 7-1 and 7-2: for example, at the point where S_{c,Σ_s} changes sign, S_{c,Σ_c} changes slope.

7.2 Summary

CHAPTER 8 CONCLUSIONS

Analytical models are useful tools for enhancing traditional analysis from the extensive computational modeling used in nuclear engineering. Performing sensitivity analysis reveals the underlying mathematical structure inherent to a scenario, leading to a deeper understanding of the salient physics. Incorporating a study of appropriate analytical models acts as part of a broader program of study which underpins the results from increasingly complicated computational science simulations. Further, the addition of analytically computed sensitivity information proves informative as a guide in interpreting, understanding, and rigorizing results of existing and future computational studies.

In the spirit of established analytical and computational model comparison techniques and outcomes the various analytical results, examples, and commentary provided in Chps.. 4, 6, and 7 represent an example of how an incorporated analytic and analytical sensitivity analysis studies can be used to set up, precondition, and eventually inform or compare against a complementary computational sensitivity analysis study. Within this conceptual strategy, and against the backdrop of the detailed MNCP computational model of a HI-STORM 100 spent nuclear fuel storage cask, the results appearing herein exemplify a more general recipe justifying the development and execution of local sensitivity analysis formalisms within the context of surrogate analytical models:

1. Establish a high-fidelity computational model, and extract key features of the simulation output.
2. Based on these key features, establish a reduced-fidelity model of the same underlying scenario; preferably this model is amenable to analytical or semi-analytical solution.
3. Execute a sensitivity analysis study on the reduced-fidelity model; again, preferably this study will be amenable to analytical or semi-analytical evaluation.

1796 4. Scenario dependent evaluation of the analytical or semi-analytical sensitivity
1797 structure requires nominal input parameters; these must also be consistent with the
1798 key features extracted from the high-fidelity computational model.

1799 5. Establish scenario-dependent sensitivity trends and input parameter importance
1800 ranking to precondition additional high-fidelity computational sensitivity analysis
1801 studies.

1802 Chapters 4, 6, and 7 exemplify this process in its application to the HI-STORM 100
1803 spent fuel cask and complimentary analytic models. For example, in the case of the fuel
1804 region, the parasitic capture cross section was found to be the parameter causing the most
1805 uncertainty in the neutron flux.

1806 More broadly, results of this type are capable of guiding future research to reduce
1807 uncertainty in the most impactful input parameters inherent to a given scenario of
1808 interest. Further, by identifying the most impactful parameters a code user can identify
1809 if any simplifications were made when developing an input which would affect the results.
1810 From these conclusions, a user could either change the input to address any insufficiencies
1811 or explain the insufficiencies and identify pathways for improvement. Either decision
1812 results in a more thorough examination of the problem, which is ultimately the goal of any
1813 scientific study.

1814 Further, the analytical results provided in this work are intended to be informative
1815 of complementary studies performed using computational tools. A process exemplified
1816 in Chp. 6, perhaps the most meaningful application of this work is the performance of a
1817 purely computational, local sensitivity analysis study in the context of both the detailed
1818 and helium models, using MNCP. In such an activity, the results of this work serve two
1819 principal purposes:

1820 1. The analytical results are used to guide more expensive (in terms of time or
1821 resources) computational studies, by identifying input parameters that are either

particularly important or rapidly variable at some physical location within a fuel cask geometry or physics model, or somehow otherwise impactful.

2. The analytical results are directly compared to computationally derived, local sensitivity coefficient information, thus further illuminating not only the possible sufficiency and limitations of various analytical models, but also the most important physics occurring within neutron transport simulation of spent fuel cask scenarios.

8.0.1 Recommendations for Future Work

In addition to this necessary program of study, there appears to be a nearly limitless sequence of higher-fidelity analytical fuel cask models in which the G-derivative formalism may be brought to bear. Candidate analytical models along these lines include but are not necessarily limited to multi-group neutron diffusion models, multi-group Pn or Sn neutron transport models, and multi-group integral or integro-differential neutron transport models. Depending on the physical processes of interest, each of these models may be formulated as static or time-dependent, in various representative geometries, and featuring any number of multi-material regions. Again, the ultimate intent of analytical sensitivity analysis studies within any of these formalisms is to enable comparison to complementary computational results.

Finally, and as indicated in Chp. ??, programs of sensitivity analysis as applied to computational models of spent nuclear fuel casks appears to be an area ripe for further advancement in research and development. This being the case, and in tandem with the aforementioned potential for new, analogous analytical treatments, there also appears to be ample opportunity for the computational evaluation of not only local sensitivity information as pertaining to spent fuel casks, but also the more complete global metrics as described by Saltelli et al., and many other authors.

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BIOGRAPHICAL SKETCH

1947

1948

Tyler Joseph Remedés began his academic career at Colorado School of Mines.

1949

He always enjoyed a challenge and chose to pursue a Bachelors of Science degree in

1950

Engineering Physics. While at Mines, he was an undergraduate researcher in Dr. Uwe

1951

Griefe's research group where he helped develop, prepare, and test organic neutron

1952

scintillation detectors. It was through this experience that he decided to continue his

1953

education at the University of Florida. Tyler's time at UF saw research in many areas of

1954

nuclear engineering as he explored various realms of nuclear engineering, including nuclear

1955

fuels, cosmic radiation shielding, nuclear imaging, and finally neutronics. Tyler spent

1956

his first summer at Los Alamos National Laboratory in 2016 where he worked on signal

1957

processing for ultra-fast radiation detection. In 2018, Tyler returned to Los Alamos, this

1958

time to stay, and worked with Dr. Scott Ramsey and Mr. Joe Schmidt learning about the

1959

utility of analytics as applied to neutronics. His time working with Dr. Ramsey and Mr.

1960

Schmidt has been a transformative period for Tyler.