

Hydrogen-Assisted Fatigue of Steel Pressure Vessels

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with

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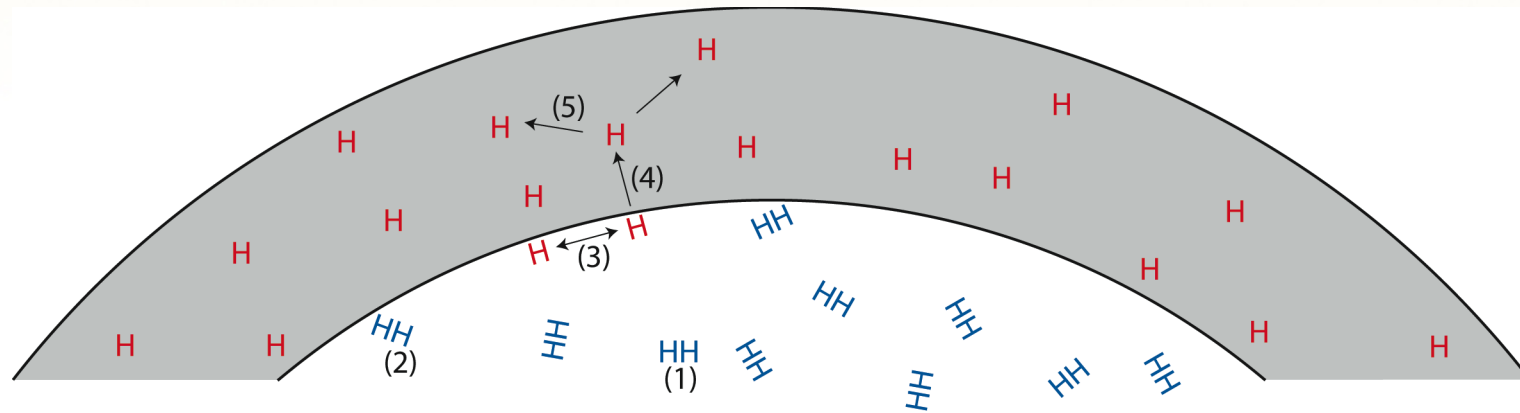
Kevin Nibur, Hy-Performance Materials Testing

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Hydrogen transport is essential to hydrogen embrittlement



- (1) Hydrogen gas
- (2) Physisorption
- (3) Dissociation
- (4) Dissolution
- (5) Diffusion

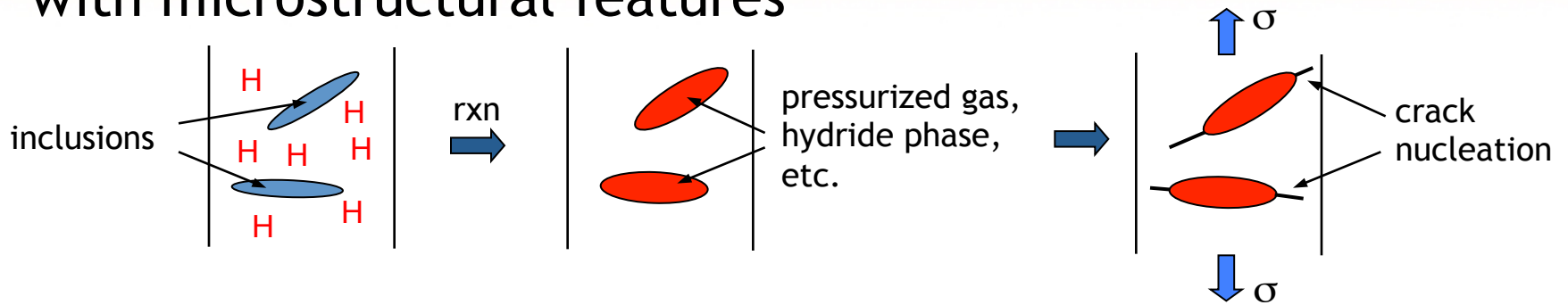
Internal Hydrogen



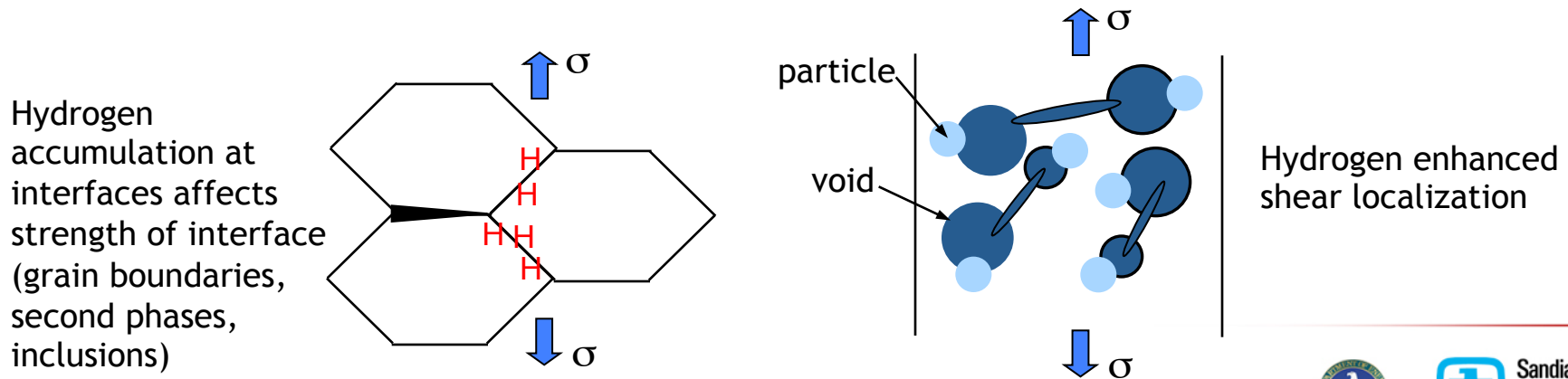
**Hydrogen
Embrittlement**

A variety of mechanisms can produce hydrogen-assisted fracture in metals

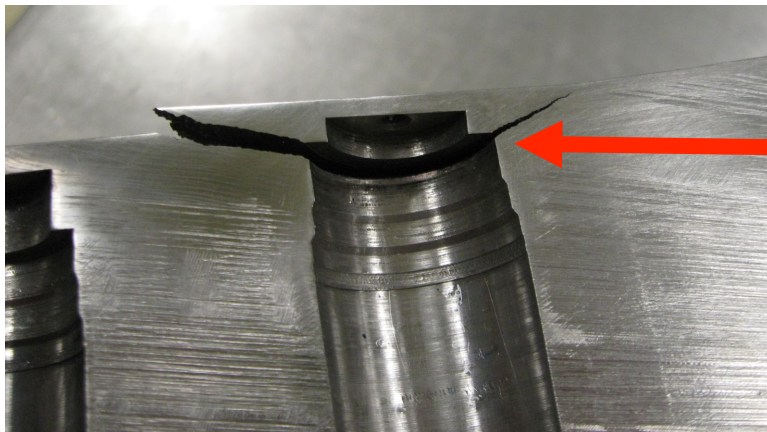
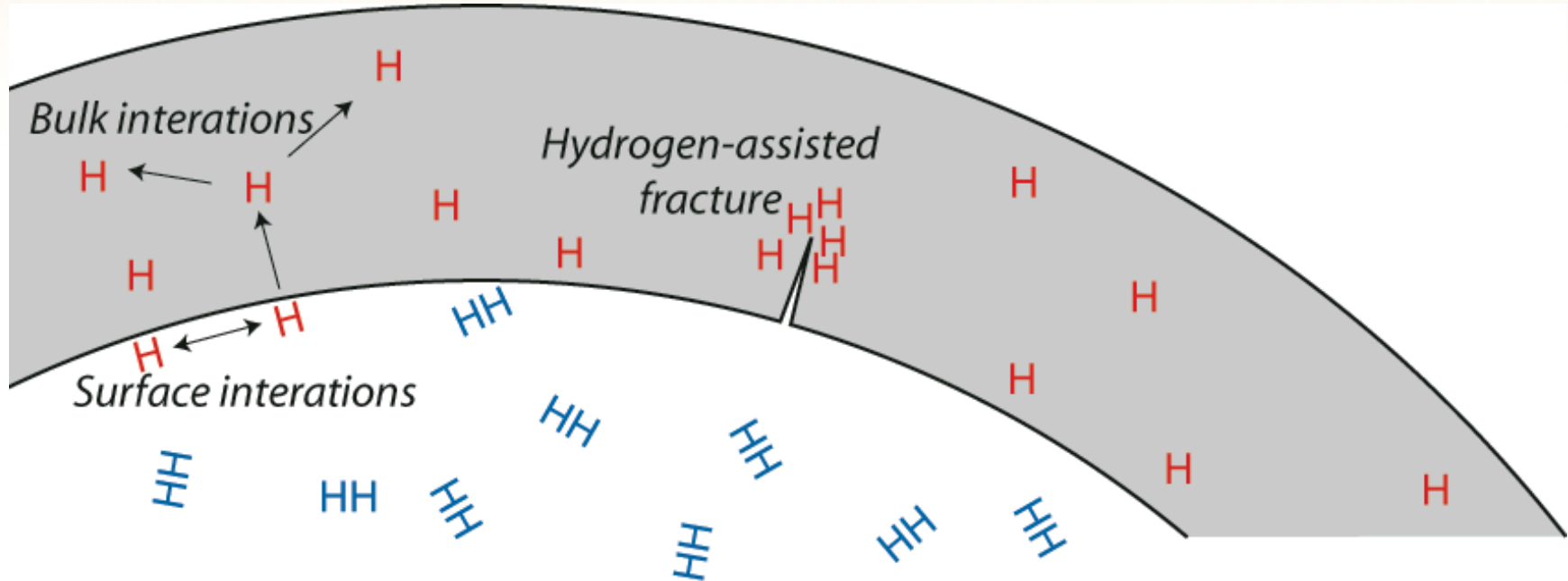
Hydrogen attack: chemical reaction of atomic hydrogen with microstructural features



Hydrogen solute effects: solute hydrogen enhanced failure of interfaces and deformation mechanisms



Hydrogen-defect interactions are critical to hydrogen-assisted fracture



Hydrogen-induced failure in high-pressure compressor initiated at stress concentration

Outline and Objectives

- Demonstrate performance test method for evaluating compatibility of pressure vessel with gaseous hydrogen
- Determine failure characteristics of commercial pressure vessels
 - Do the pressure vessels leak-before-burst when cycled with gaseous hydrogen?
- Compare full-scale testing for steel pressure vessels for gaseous hydrogen with engineering design methods
 - Fracture mechanics-based design
 - Stress-life design
- Describe method proposed in CSA standard

Fracture and fatigue resistance of steels is degraded by exposure to hydrogen



Hydrogen-induced failure of transport cylinder

Motivation:

innovative applications are expanding design space beyond engineering experience



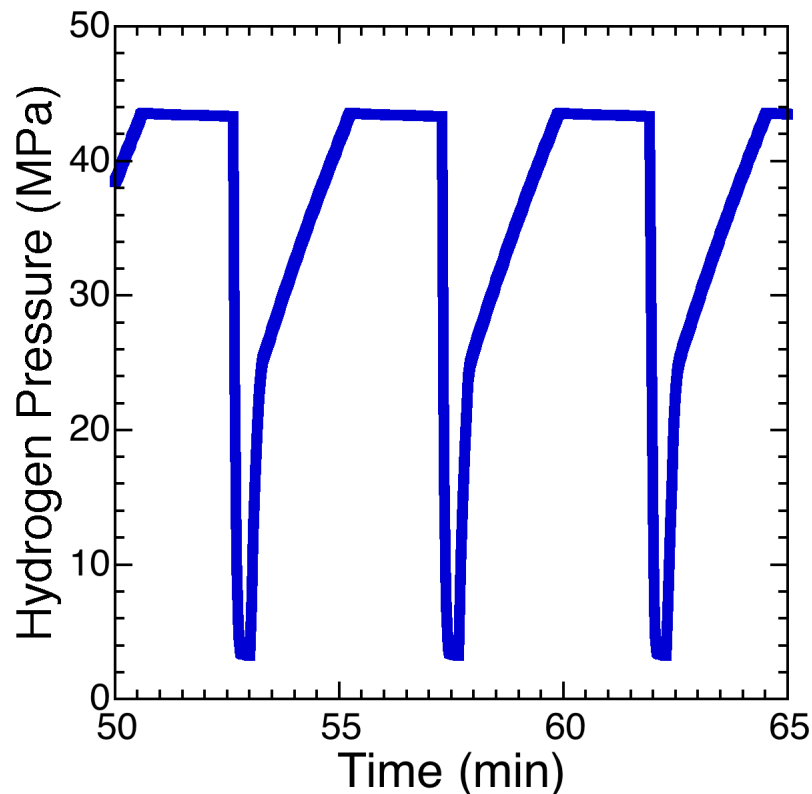
>10,000 refueling cycles are anticipated for hydrogen-powered industrial trucks

Pressure cycle designed for accelerated testing



Consider 35 MPa gaseous hydrogen fuel system

- Nominal pressure of 35 MPa
- Allow 25% over-pressure during rapid filling
- Minimum system pressure of ~3 MPa

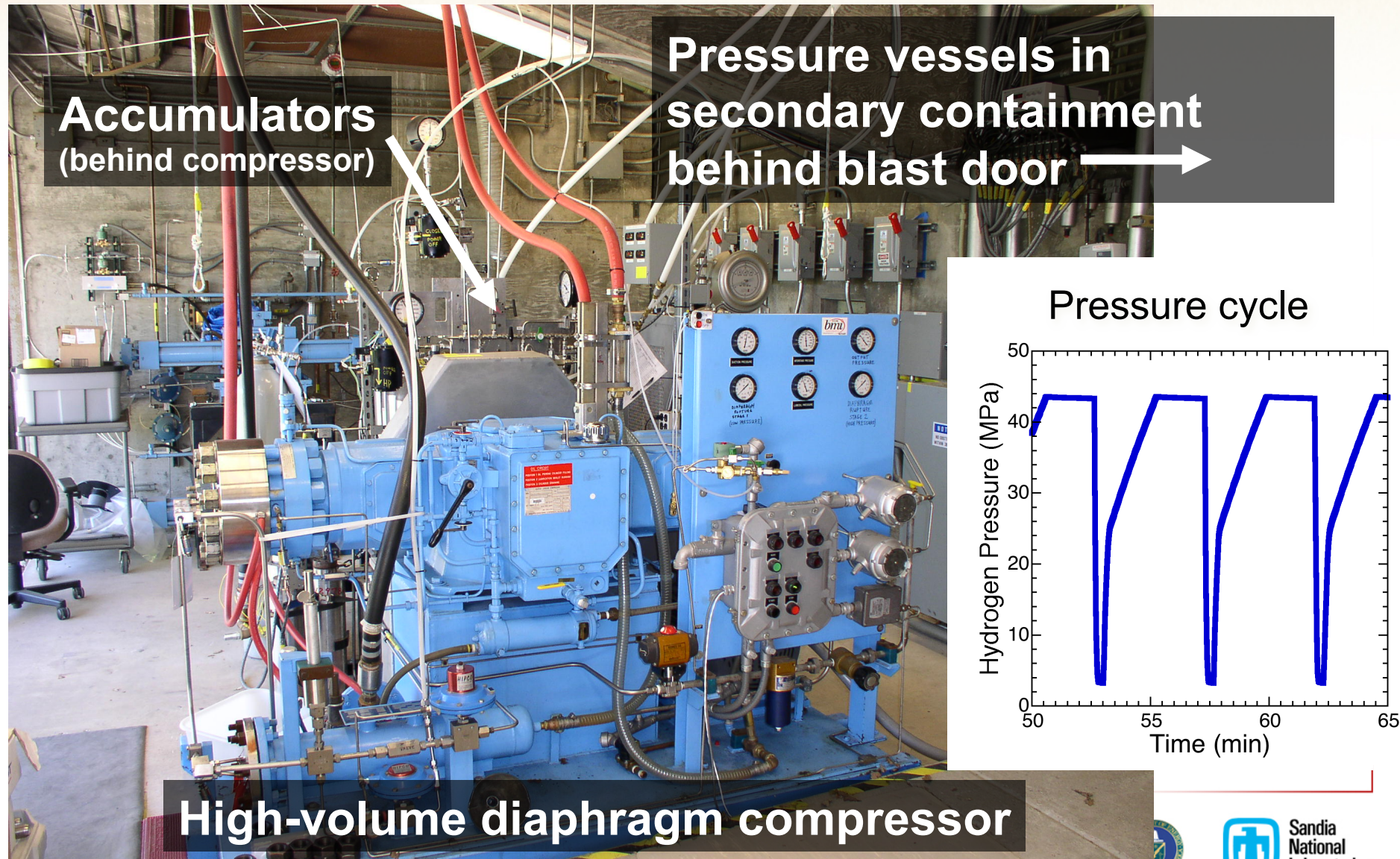


Pressure cycle for testing

- maximum $P = 43.5$ MPa
- 2-minute hold at maximum P
- rapid depressurization to 3 MPa
- 30-second hold at minimum P
- pressurization time ~ 2 min

*4 to 5 minute cycle time
(~300 cycles per day)*

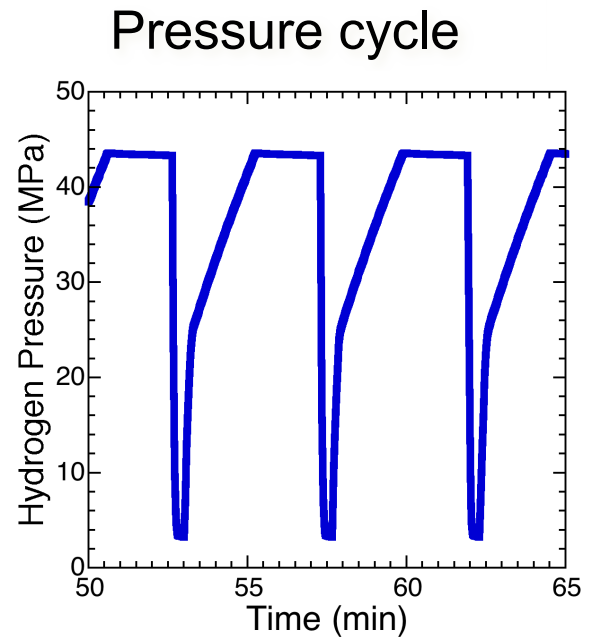
Closed-loop gas-handling system capable of simultaneously pressurizing 10 pressure vessels



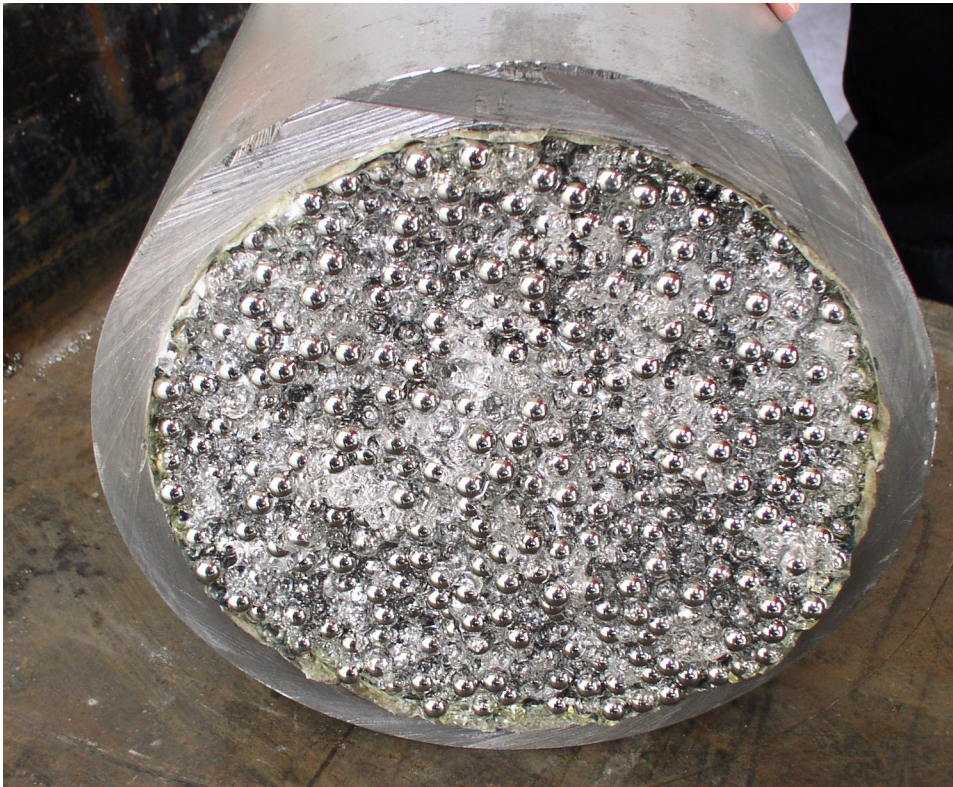
Accumulators
(behind compressor)

Pressure vessels in
secondary containment
behind blast door

High-volume diaphragm compressor



Free volume within vessels reduced to facilitate pressure cycling



- Bladder used to isolate PV surface from filler material
- Epoxy and steel used as filler
- Volume reduction 90-95%
- Gas quality inspected periodically
 - typical analysis
 - oxygen <2 ppm
 - hydrocarbons <5 ppm
 - water <5 ppm

Sectioned pressure vessel showing vessel, bladder, steel ball bearings and epoxy

Pressure vessels consistent with design rules for transportable gas cylinders

- Two pressure vessel designs from different manufacturers
 - Nominal hoop stress at $P = 43.5$ MPa
 - T1 design: ~ 340 MPa
 - T2 design: ~ 305 MPa



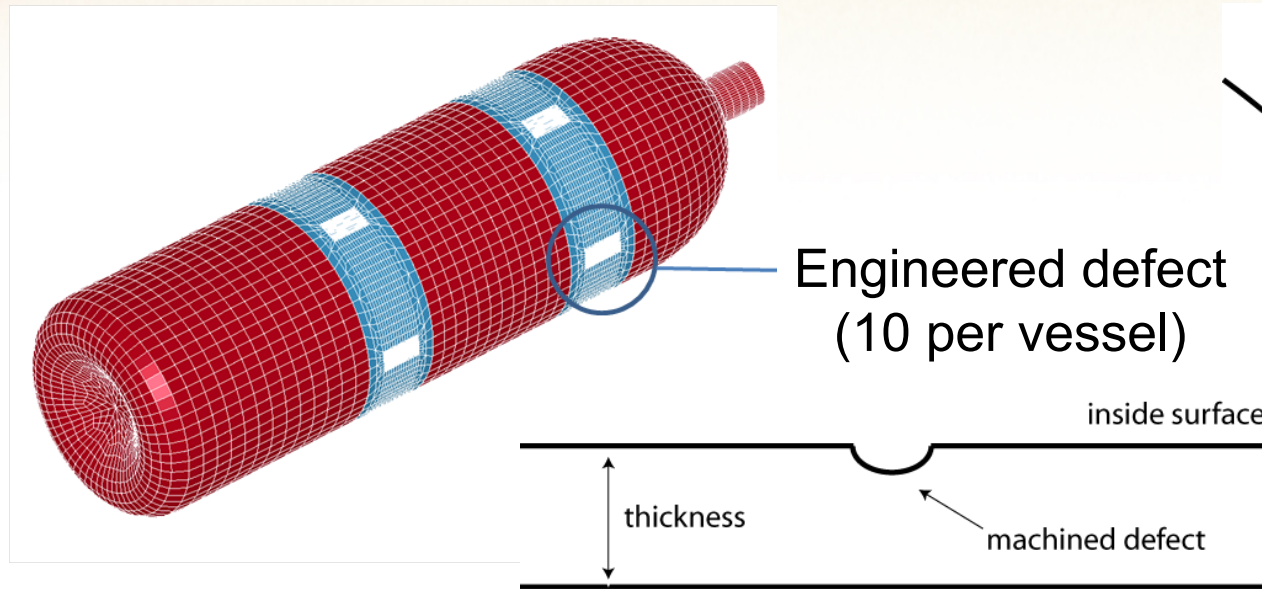
- Steel for both pressure vessels designs: 4130X
 - Quench and tempered, 1 wt% Cr - 0.25 wt% Mo
 - UTS for transport applications: 700 to 900 MPa
 - T1 design: ~ 750 MPa
 - T2 design: ~ 850 MPa

Typical design rule: maximum wall stress $< 40\%$ of UTS

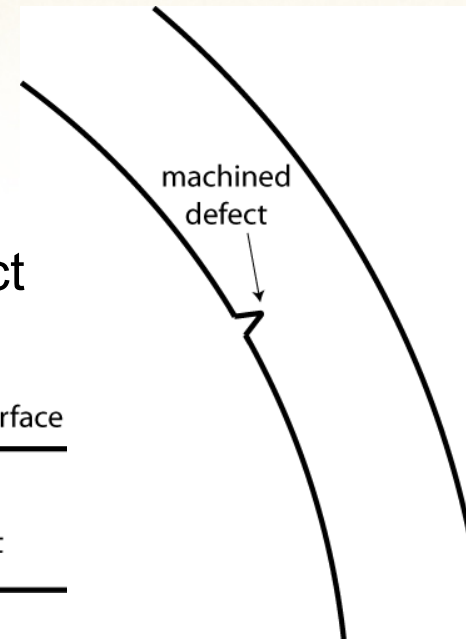
T1 design: 300 MPa

T2 design: 340 MPa

Engineered defects used to initiate failures



Elliptical engineered defect
Aspect ratio = 1/3 (depth/length)



V-notch in profile
Nominal root
radius 0.05mm
(actual ~0.12mm)

Depth of engineered defects

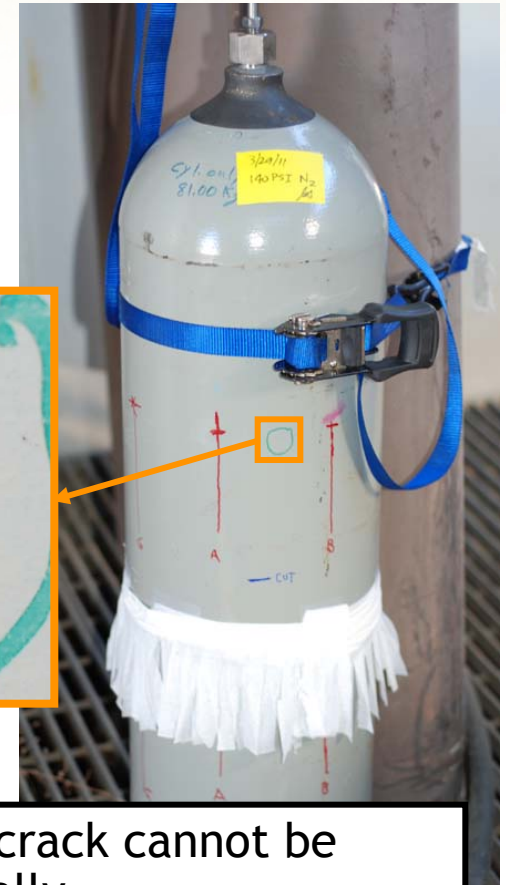
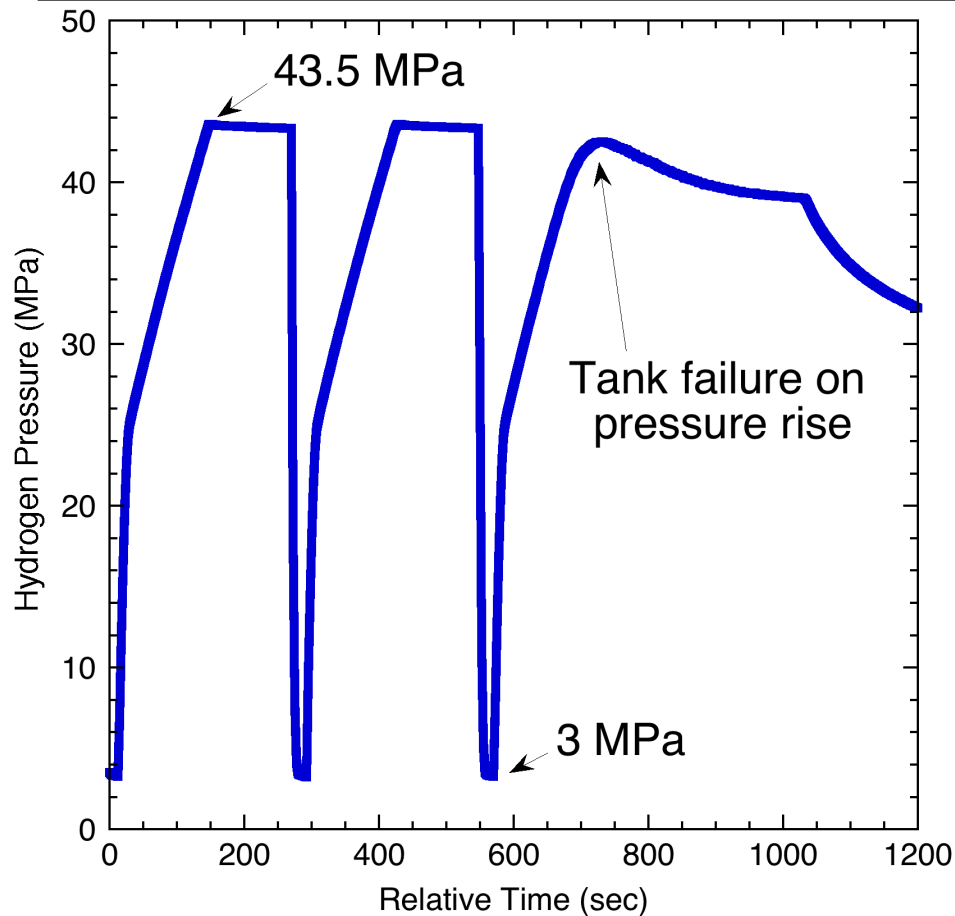
- Typically all 10 defects similar for a given vessel
- Smallest defects ~2% of wall thickness
- Largest defects ~10% of wall thickness
- For one vessel, aspect ratios were 1/2 and 1/12

Outline and Objectives

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- Compare full-scale testing for steel pressure vessels for gaseous hydrogen with engineering design methods
 - Fracture mechanics-based design
 - Stress-life design
- Describe method proposed in CSA standard

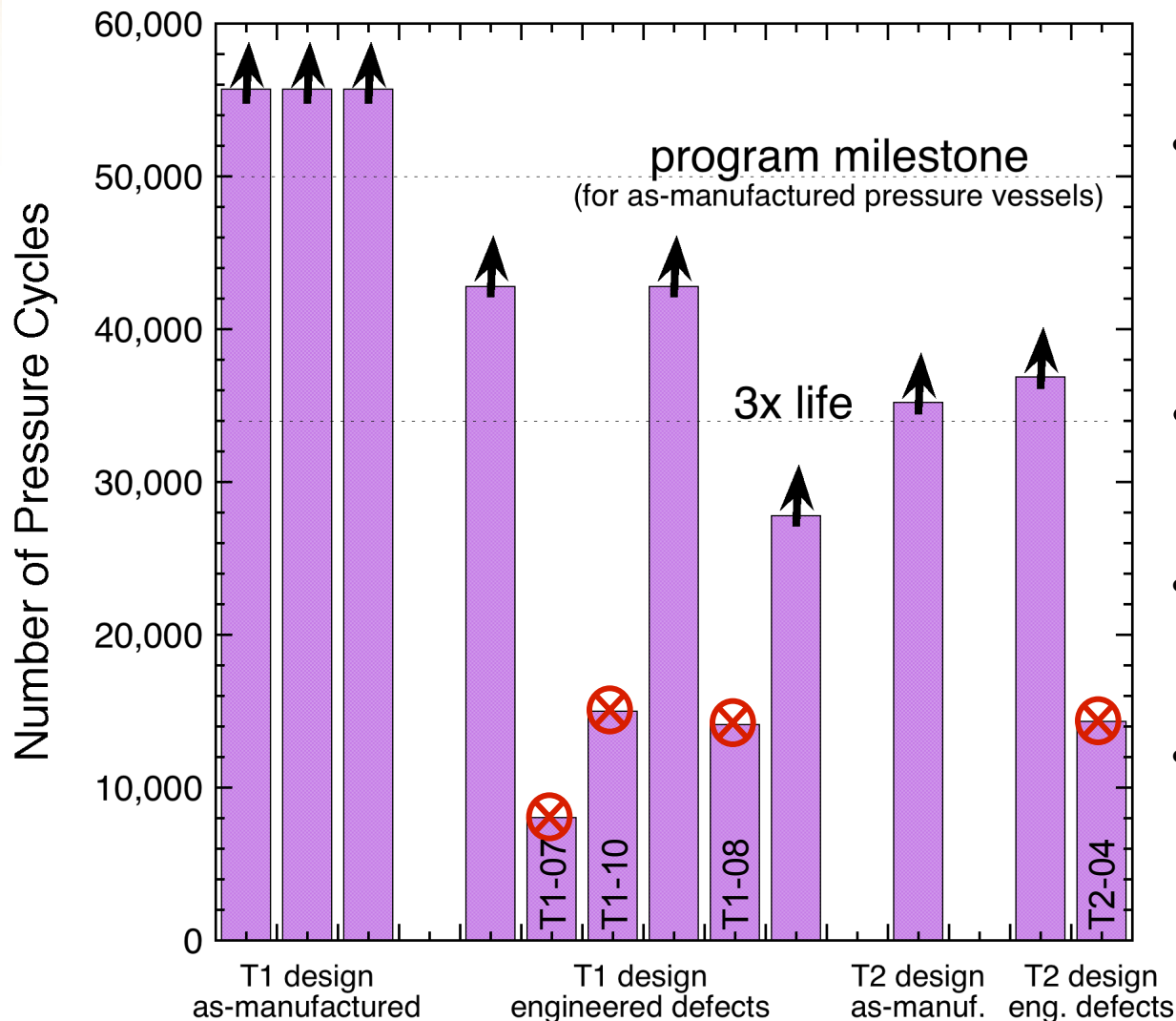
All observed failures are *leak-before-burst*

- At failure, pressure vessel “slowly” leaks gas into secondary containment



- Through-wall crack cannot be detected visually

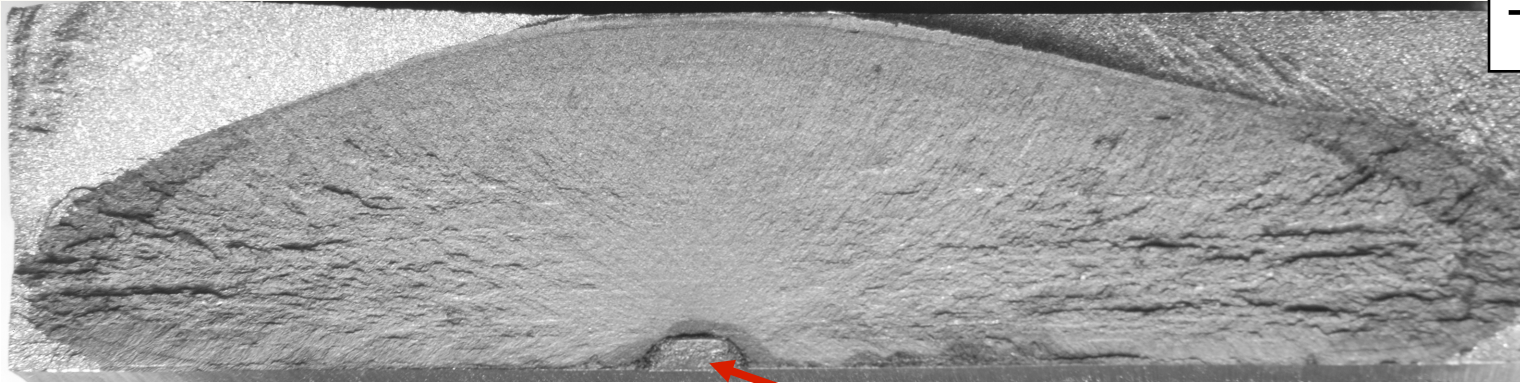
Commercial pressure vessels exceed lifetime target of 11,250 cycles by >3x



- Each pressure vessel with engineered defects contains 10 nominally equivalent defects
- Arrows indicate pressure vessels that did not fail
- In failed vessels, all defects initiate a crack
- All four failures were leak before burst

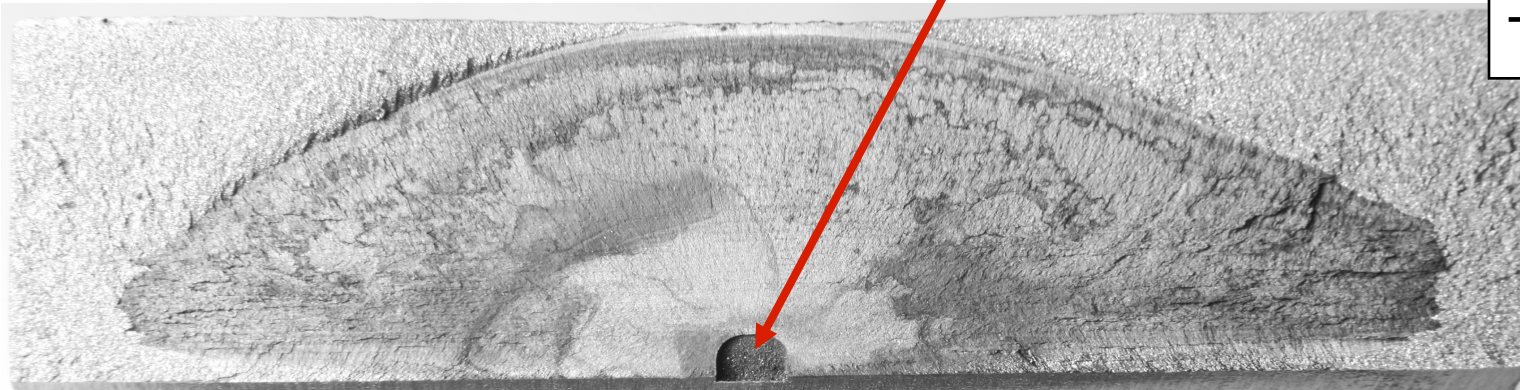
Through-wall cracks extend from “critical” engineered defect

wall thickness

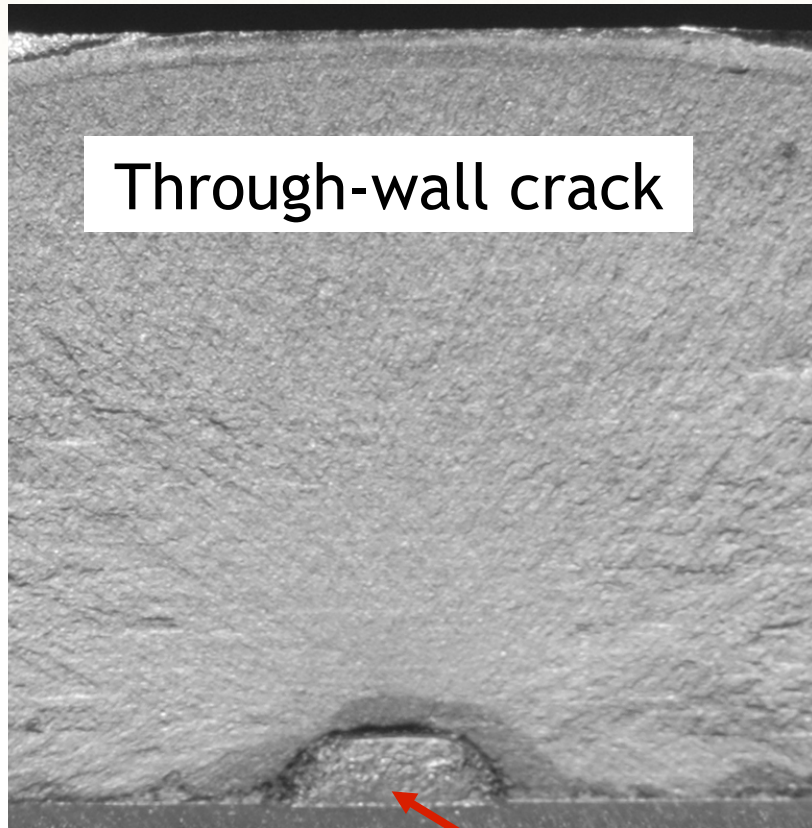


engineered defect

wall thickness

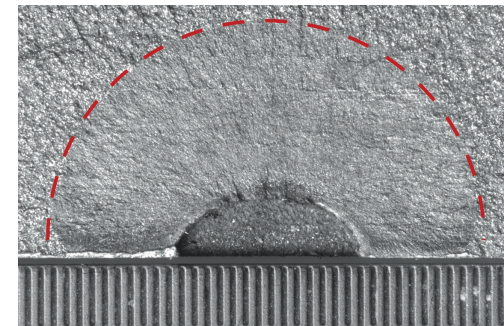
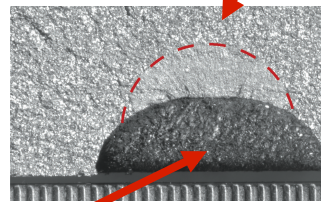


Cracks extend from all engineered defects



Non-through-wall (growing) cracks have semicircular profile

- Smaller engineered defect
- Greater number of cycles = more crack extension

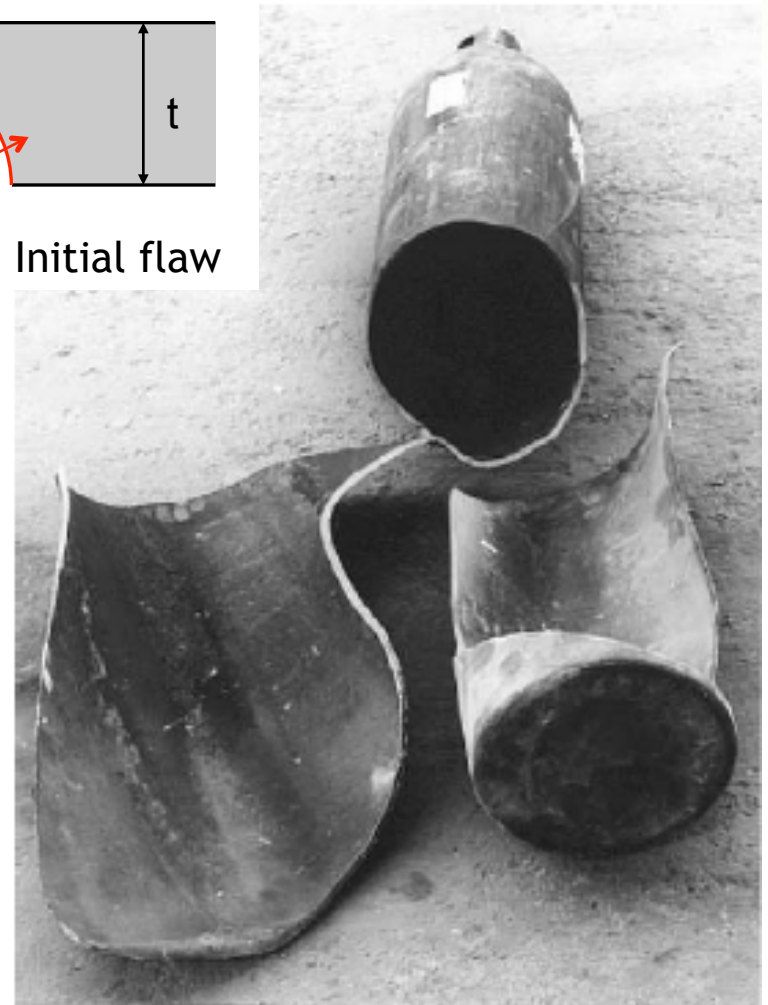
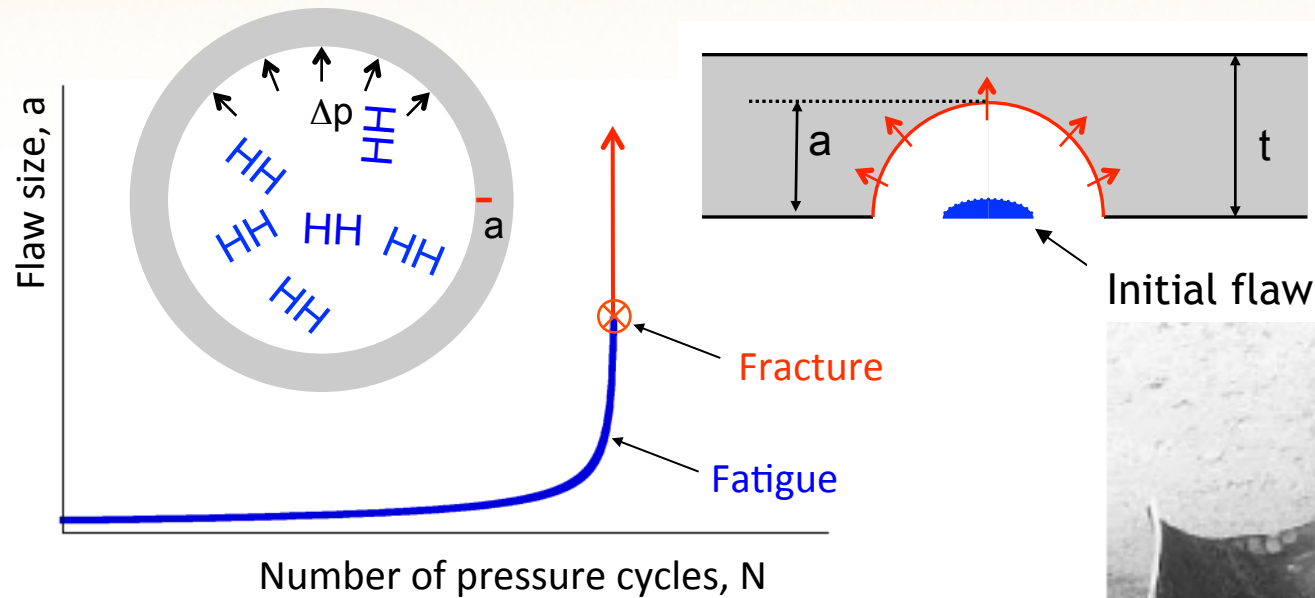


Same size engineered defect
(same vessel)

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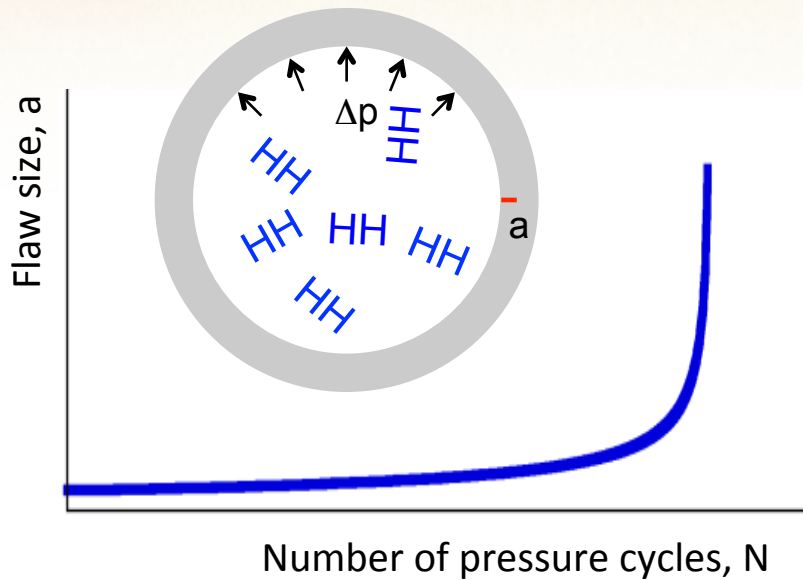
Fracture process consists of two components: fatigue and fracture



Fracture process can be modeled with materials properties measured in gaseous hydrogen:

- **Fatigue** crack growth under cyclic stress
- **Fracture** resistance

Fracture mechanics provides a methodology for quantifying crack extension



Fracture mechanics implies that single parameter (K) uniquely characterizes the cracking response

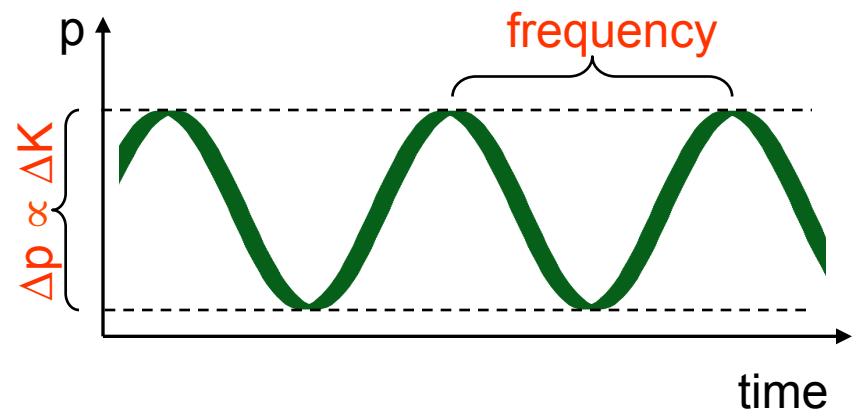
- Stress intensity factor (K) depends on pressure/load (p), size of the crack (a) and geometry
- Fatigue crack growth (da/dN) depends on ΔK

In the laboratory, one can measure the material's response to fatigue

$$da/dN = f(\Delta K)$$

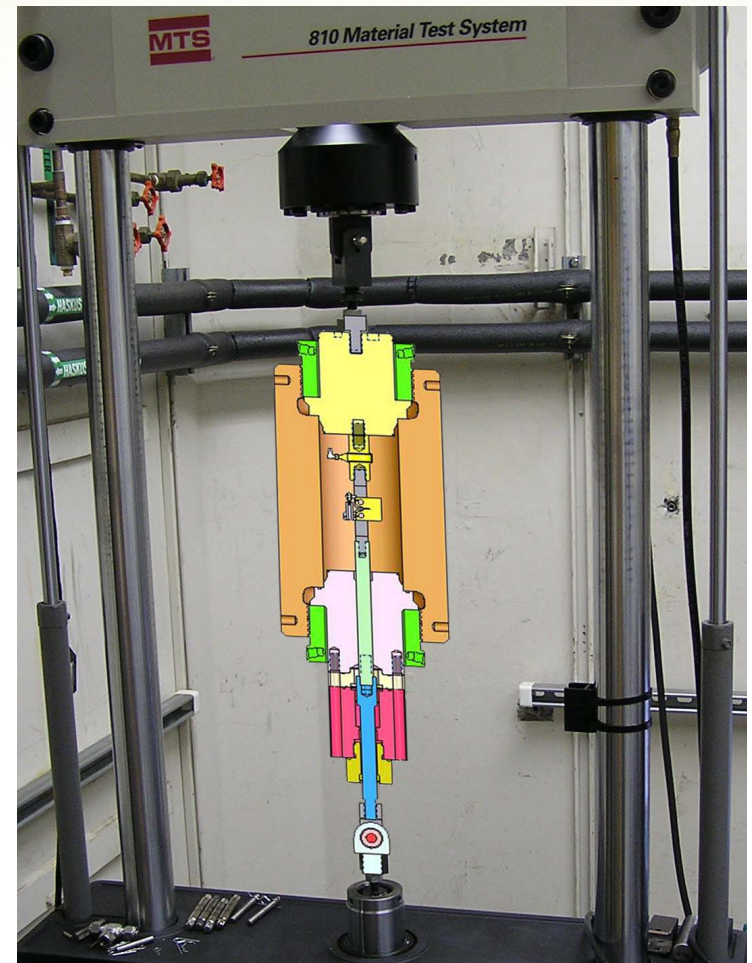
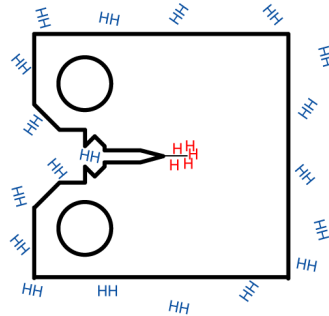
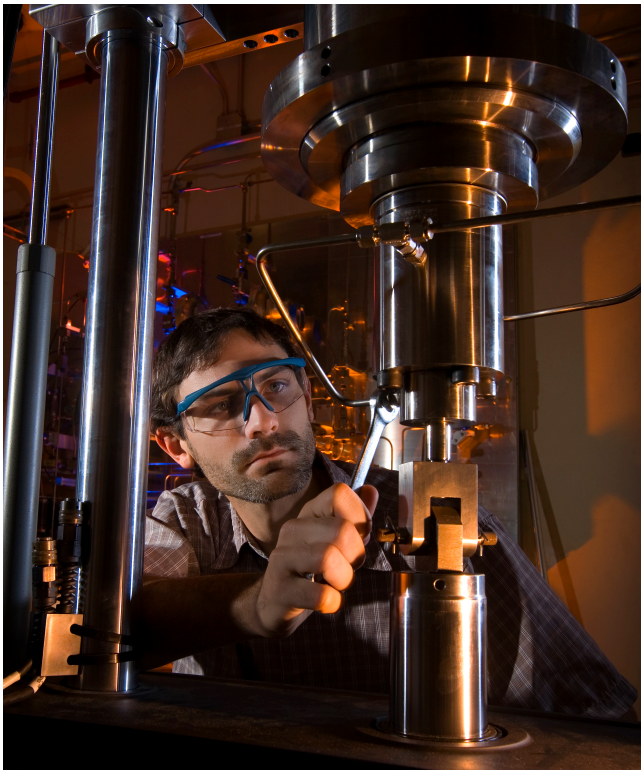
And integrate for a known geometry

$$a_{i+1} = a_i + (da/dN)_i \Delta N$$



Fracture mechanics tests can be conducted in gaseous hydrogen at high pressure

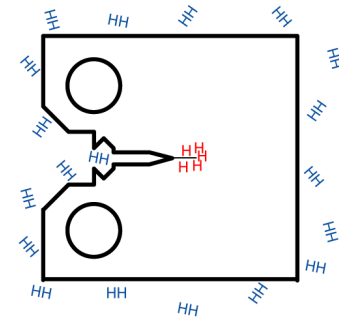
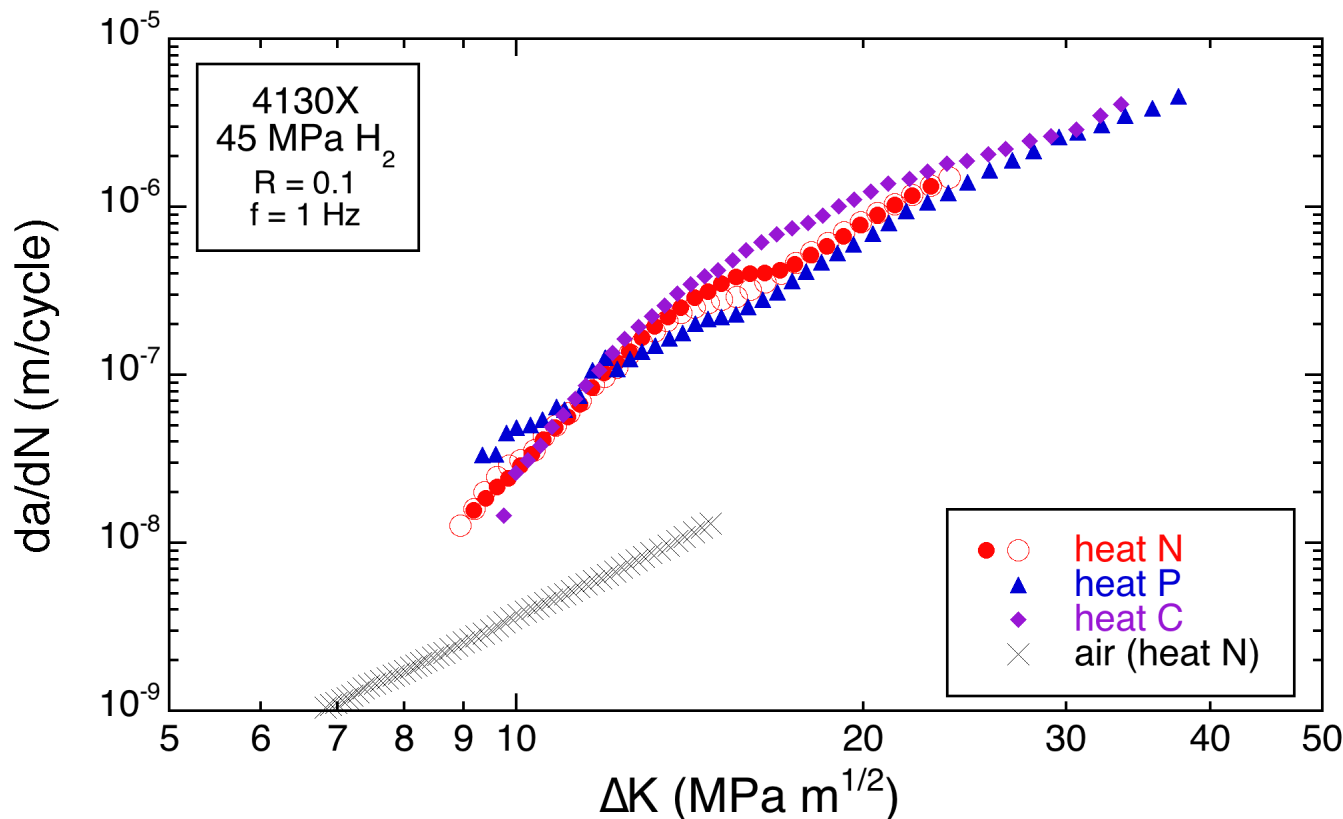
- Fatigue crack growth: da/dN
 - ASTM E647, constant load amplitude
- Fracture resistance: K_{IH}
 - ASTM E1820, elastic-plastic analysis using J-R curve determination



Fatigue crack in gaseous hydrogen is an order of magnitude greater than in air

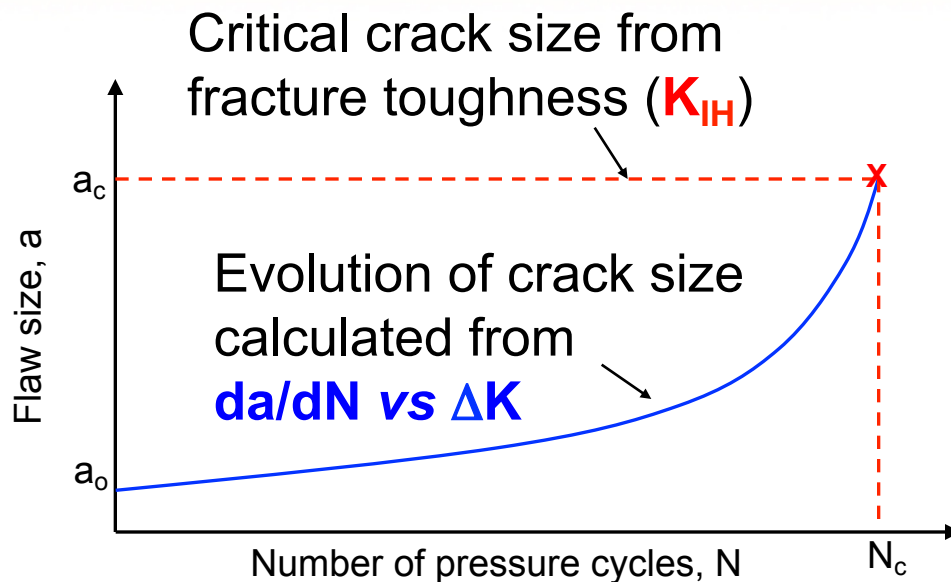
Fatigue crack growth rates measured in gaseous hydrogen at pressure of 45 MPa

- 3 heats of 4130X steel from pressure vessels
- (unlike fracture resistance, fatigue crack growth in ferritic steels appears to be relatively insensitive to hydrogen pressure)



Fatigue life qualification by fracture mechanics (crack growth methodology)

ASME BPVC VIII.3 KD-10 (KD-4)



Stress intensity
at $a/t = 0.8$

T1: 55 MPa m^{1/2}
T2: 64 MPa m^{1/2}

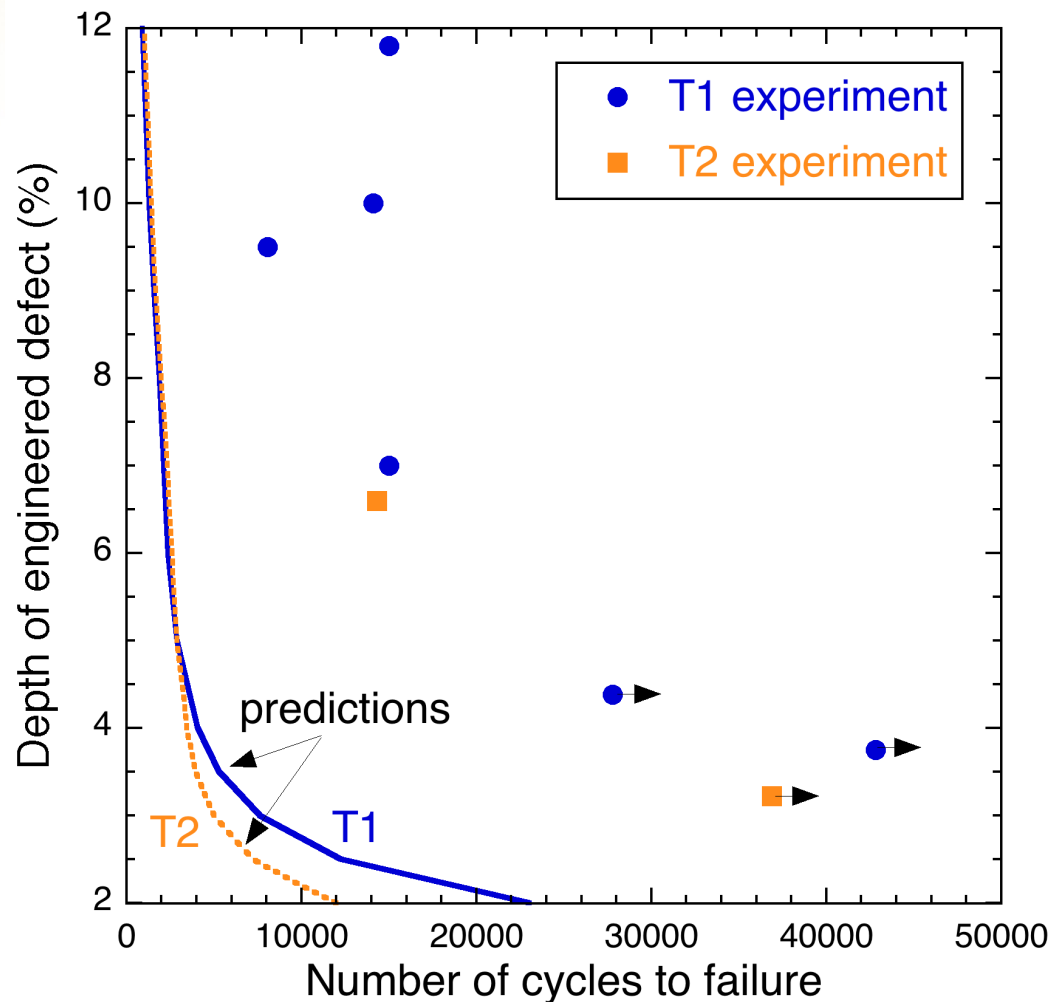
$${}^\dagger K_{JIH} = 59 \text{ MPa m}^{1/2}$$

Assumptions:

- a_c = thickness
- semicircular propagating cracks
- use data[†] for $R = 0.1$ and $f = 0.1$ Hz

[†] 4130X steel measured in gaseous H₂ at pressure of 45 MPa
Nibur et al. (PVP2010-25827)

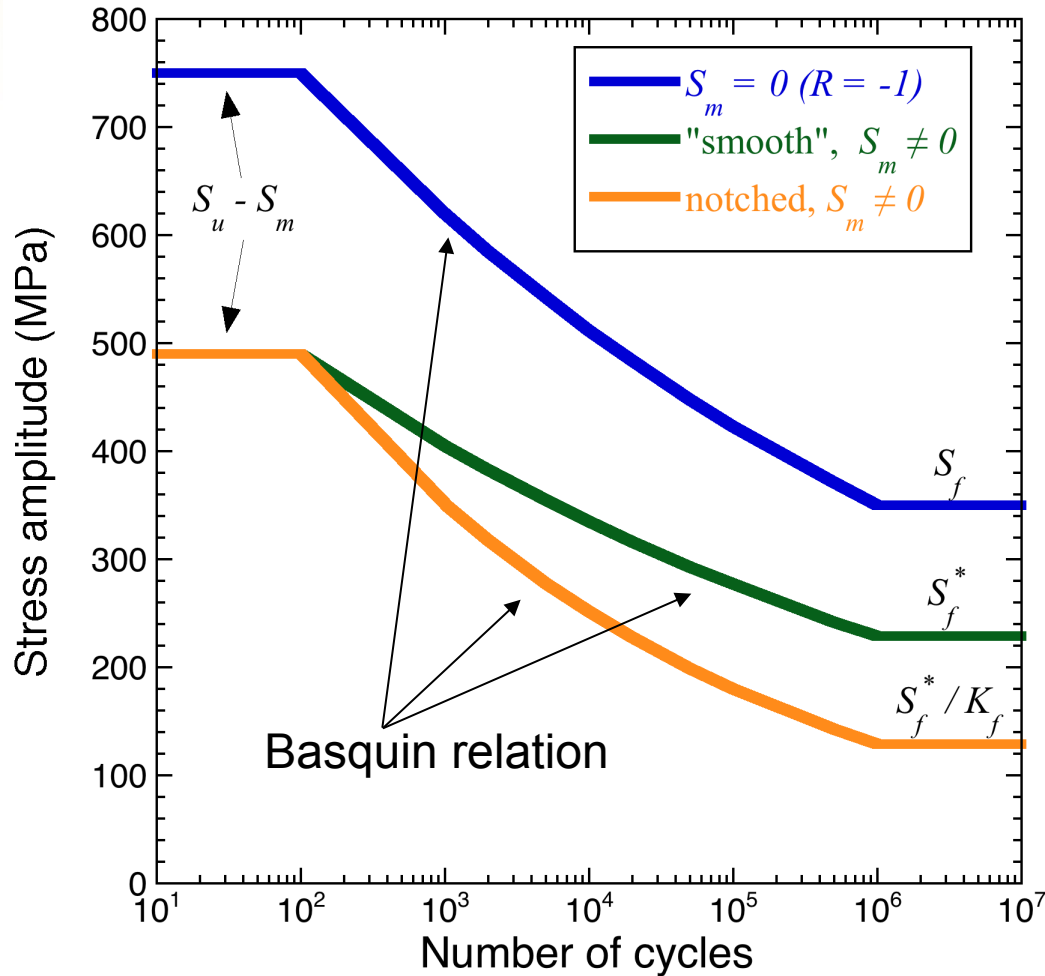
Fracture mechanics is overly conservative when defects are not initially growing



- Curves are predictions based on *crack growth* only (of semicircular flaw)
- Arrows indicate vessels that did not fail
- Failures use measured dimensions (others assume nominal dimensions)

- Fatigue life calculation is conservative by factor of 4 or more
- For small initial defects, effective safety factor approaches 10

Fatigue life methods offer framework for incorporating crack initiation



- Idealized S-N curves based on
 - Materials properties: S_u (UTS) and S_f
 - Geometry and loading: K_f and S_m

Effect of mean stress:

$$S_f^* = S_f \left[1 - \frac{S_m}{S_u} \right]$$

Effect of notch:

$$S_f^N = \frac{S_f}{K_f}$$

Effect of hydrogen on S-N curve and fatigue limit is unknown

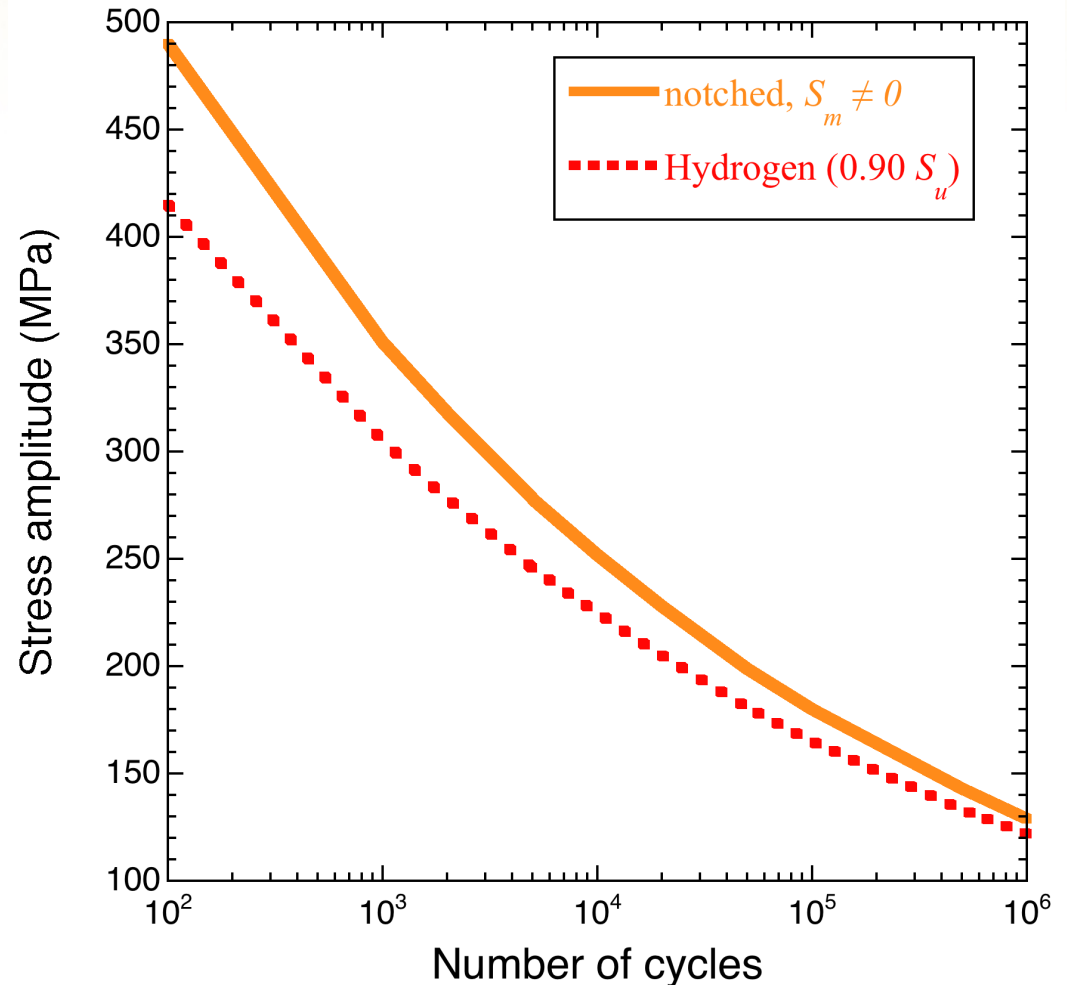
Data for Cr-Mo steels in tension-compression fatigue suggests

$$S_f(\text{H}_2) \approx S_f(\text{air})^\dagger$$

Implication: at low stress hydrogen does not affect fatigue crack initiation

Conservative assumption based on notched tension‡ in gaseous hydrogen:

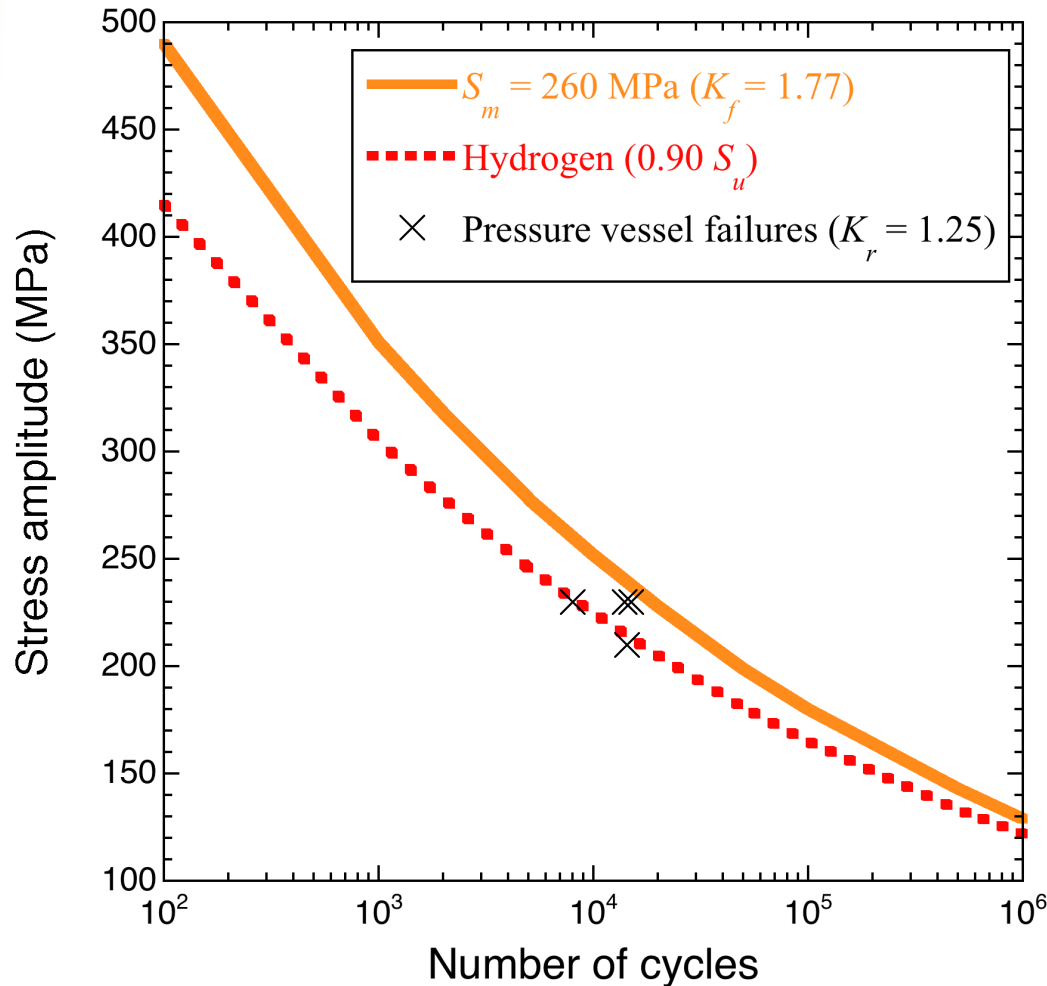
$$S_u(\text{H}_2) \sim 0.9 S_u$$



† Ref. Wada et al. ICES 2005

‡ Ref. Steinman et al. Welding J Res Supp 44 (1965)

Fatigue of pressure vessels with engineered defects compare favorably with predictions



Materials properties:

$$S_u = 750 \text{ MPa}$$

$$S_f = 350 \text{ MPa (est.)}$$

Geometry and loading:

$$K_f = 1.77 \text{ (Neuber est.)}$$

$$S_m = 260 \text{ MPa } (K_r = 1.25)$$

Fatigue notch sensitivity (K_f)

$$K_f = f(K_t, \rho, S_u)$$

ρ = notch root radius

K_t estimated from FEA

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Draft CSA Standard for Compressed Hydrogen Powered Industrial Truck On-Board Fuel Storage and Handling Components (HPIT1)

Performance requirements

- Leak-before-break requirements
 - type 1, 2 and 3: ASME VIII.3 KD-141 using K_{IC}
 - type 4: ISO 15869 Annex B.8
- Two performance options:
 - Fatigue life verification by *testing*
 - OR
 - Fatigue life qualification by *analysis*

Draft CSA Standard for Compressed Hydrogen Powered Industrial Truck On-Board Fuel Storage and Handling Components (HPIT1)

- Fatigue life verification by *testing*

Requirement: 3X maximum fill cycles specified by manufacturer

- Pressure cycling with gaseous hydrogen
- Artificial defect: depth \geq NDE; aspect ratio $>3:1$ (length:depth)
- 10 to 125% service pressure

- Fatigue life qualification by *analysis*

Requirement: maximum fill cycles determined from ASME VIII.3 KD-3

- Design pressure = 125% service pressure (25 or 35 MPa service)
- DOT 3AA 4130X or ASME SA-372 (Cr-Mo) steels
- $S_u < 890$ MPa
- Wall stress (hoop stress) $< 0.4 S_u$
- Surface roughness: $R_a < 6.4$ μm and $R_{max} < 20$ μm

Fatigue life qualification by analysis

Proposed requirements for type 1 steel pressure vessels in gaseous hydrogen service

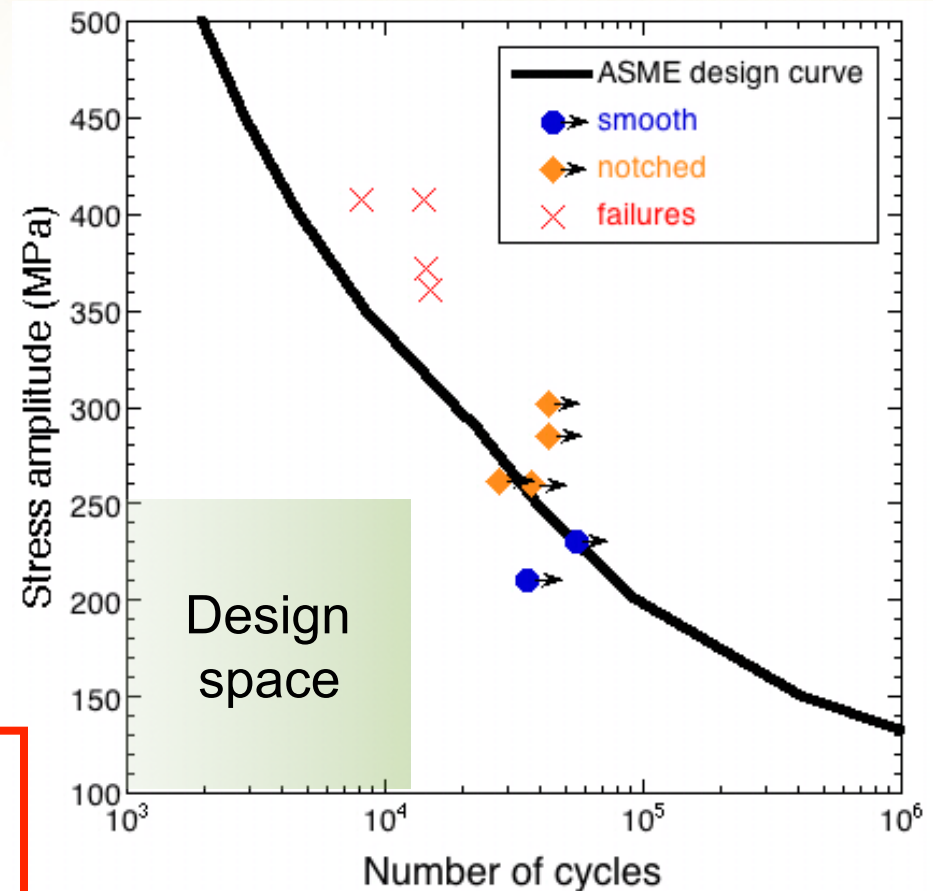
- $S_u < 890$ MPa
- hoop stress $\leq 0.4 S_u = 356$ MPa

From ASME VIII.3 KD-3

- assume: $K_r = 1.25$
- T1 design
- $S_m \approx 260$ MPa & $S_a \approx 230$ MPa

Engineering Significance of these requirements

- Stress intensity $< \sim 400$ MPa
- Stress amplitude $< \sim 250$ MPa
- Design life $> 40,000$ cycles



ASME design curve: carbon and low alloy steels with UTS = 620 MPa

Summary

- Vessels being used for hydrogen storage were subjected to more than 55,000 pressure cycles with gaseous hydrogen at a peak pressure of 43.5 MPa
- Engineered defects with depth $>6\%$ of the wall thickness failed between 8,000 and 15,000 cycles
- Leak-before-burst was observed for all failures
- Fatigue crack growth assessment is very conservative for idealized defects
 - Cycles to failure due to engineered defects is >4 times design calculation using ASME VIII.3
 - Crack initiation dominates the cycle life even with internal defects
- Fatigue life curves based on testing in air are being considered for design of hydrogen pressure vessels (CSA HPIT1)