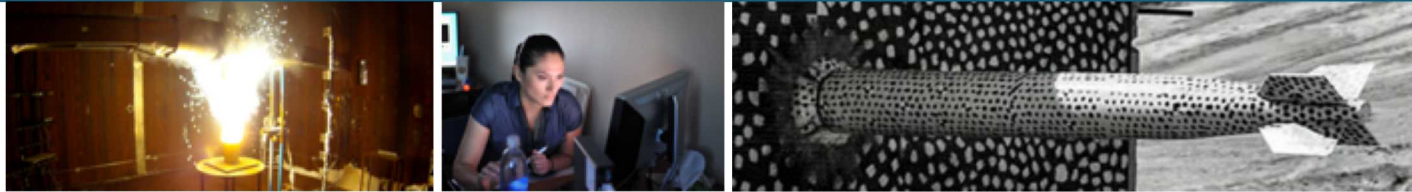


Machine-learned reduced-order modeling



PRESENTED BY

Patrick Blonigan

Collaborators: Kevin Carlberg, Brian Freno, Francesco Rizzi, Chi Hoang, Kookjin Lee, Eric Parish, Jaideep Ray, Yukiko Shimizu, John Tencer, Irina Tezaur

CIS External Review, August 26-29, 2019

SAND 2019XXX-XX



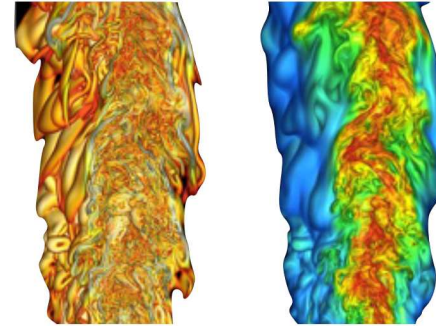
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High-fidelity simulations are crucial, but often too costly for rigorous use within Sandia's mission

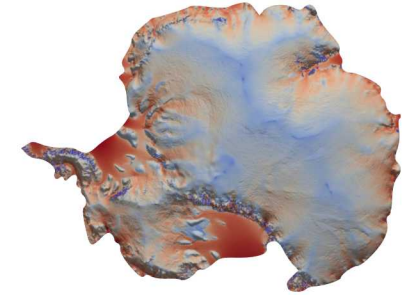


•High-fidelity simulation:

- Extreme-scale nonlinear computational models,
- Indispensable for a wide range of engineering and scientific applications.
- Example: captive carry aerodynamics simulation
 - Extreme-scale: 100 million cells, 200,000 time steps.
 - High cost: 6 weeks on 5000 cores.



*Turbulent reacting flows
courtesy J. Chen*



*Antarctic ice sheet modeling
courtesy R. Tuminaro*

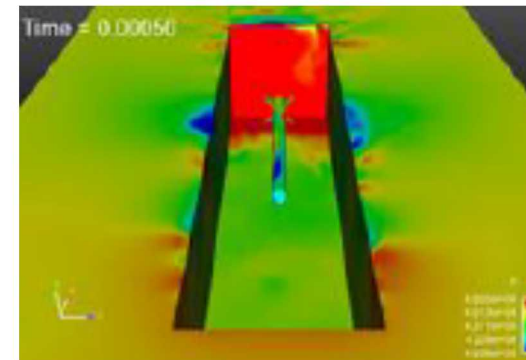
•Mission problems:

• Time-critical:

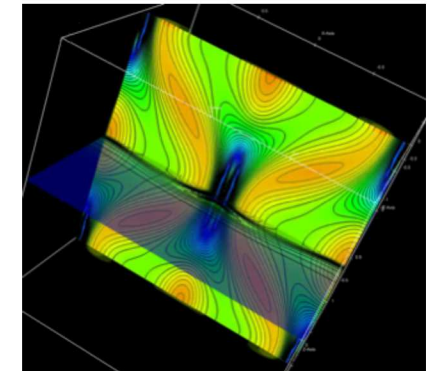
- Model predictive control
- Health monitoring

• Many-query:

- Uncertainty quantification
- Design optimization



*Captive carry aerodynamics
courtesy M. Barone*



*Magnetohydrodynamics
courtesy J. Shadid*

We use model reduction to exploit high-fidelity simulation data for use within many-query and time-critical mission applications

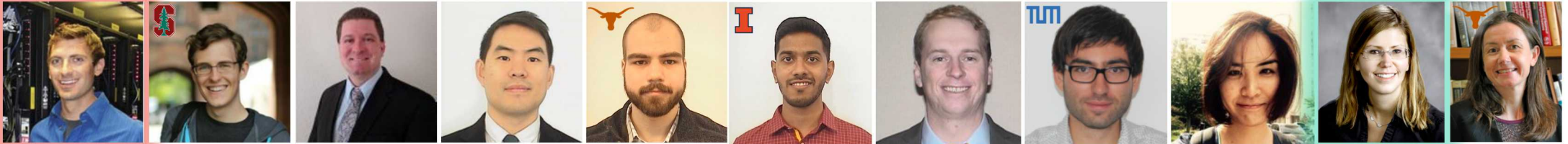


Model Reduction Criteria

1. **Accuracy:** achieves less than 1% error
2. **Low cost:** achieves at least 100x computational savings
3. **Property preservation:** preserves important physical properties
4. **Generalization:** should work even in difficult cases
5. **Certification:** accurately quantify the reduced order model (ROM) error
6. **Extensibility:** should work for many application codes

Model reduction at Sandia is a large multidisciplinary effort supported by researchers spanning centers and institutions

- Started by **Kevin Carlberg** with 3 people in FY12, focused on applied math research
- Grown to 21 in FY19, with a **leadership team** spanning the Computing and Information Sciences (CIS) and Engineering Science (ES) research foundations and institutions
- **Applied Math (CIS+ES)**: method development and analysis



- **Computer Science (CIS)**: generalized, minimally intrusive model reduction implementation



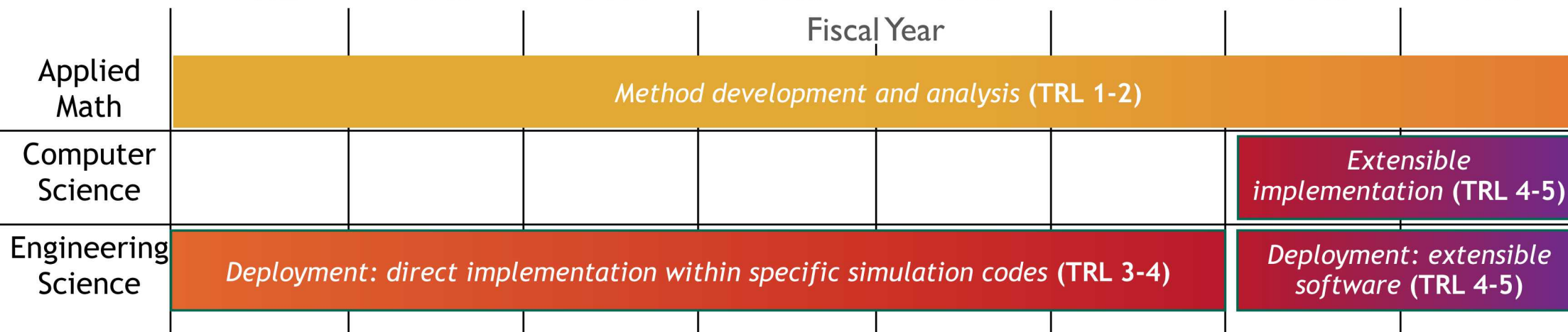
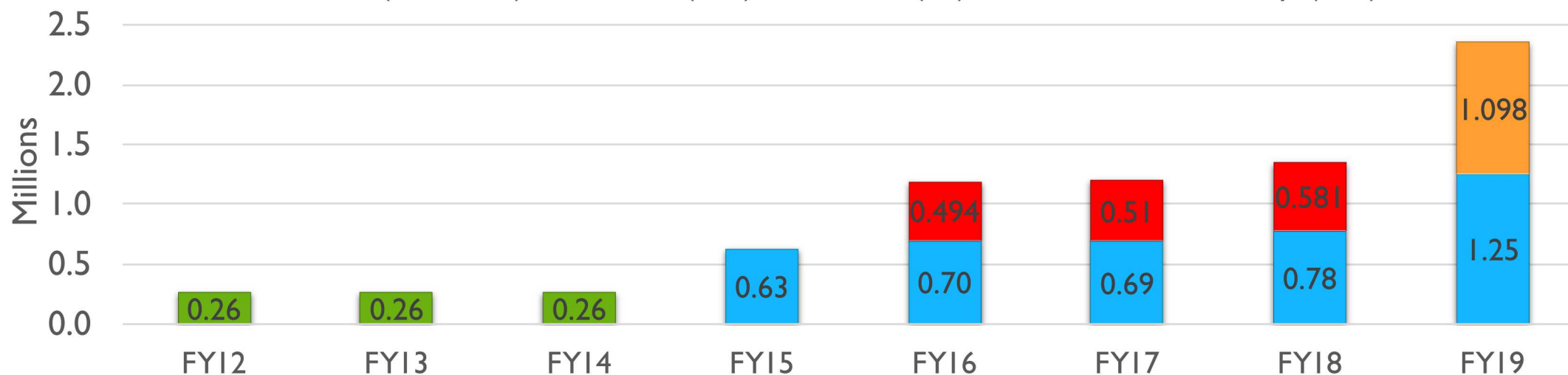
- **Engineering science (ES)**: deployment of model reduction in engineering applications and analysis



The funding scope and technology readiness level of our work are growing



■ ASC (CIS + ES) ■ LDRD (CIS) ■ LDRD (ES) ■ Truman Fellowship (CIS)



Historical model reduction approaches are ineffective for nonlinear dynamical systems, which arise often in Sandia's mission applications



Historical model reduction work external to Sandia

- **Linear time-invariant systems: mature** [Antoulas, 2005]
 - Balanced truncation [Moore, 1981; Willcox and Peraire, 2002; Rowley, 2005]
 - Transfer-function interpolation [Bai, 2002; Freund, 2003; Gallivan et al, 2004; Baur et al., 2011]
 - ✓ **Accurate, generalizes, certified:** sharp *a priori* error bounds
 - ✓ **Inexpensive:** pre-assemble operators
 - ✓ **Property preservation:** guaranteed stability
- **Elliptic/parabolic PDEs: mature** [Prud'Homme et al., 2002; Barrault et al., 2004; Rozza et al., 2007]
 - Reduced-basis method
 - ✓ **Accurate, generalizes, certified:** sharp *a priori* error bounds
 - ✓ **Inexpensive:** pre-assemble operators
 - ✓ **Property preservation:** preserve operator properties
- **Nonlinear dynamical systems: ineffective**
 - Proper orthogonal decomposition (POD)–Galerkin [Sirovich, 1987; Colonius, 2004]
 - ✗ **Inaccurate, doesn't generalize:** often unstable
 - ✗ **Not certified:** error bounds grow exponentially in time
 - ✗ **Expensive:** projection insufficient for speedup
 - ✗ **Structure not preserved:** physical properties ignored
 - ✗ **Not extensible:** highly intrusive implementation required

Model Reduction Criteria

1. **Accuracy:** achieves less than 1% error
2. **Low cost:** achieves at least 100x computational savings
3. **Property preservation:** preserves important physical properties
4. **Generalization:** should work even in difficult cases and for many application codes
5. **Certification:** accurately quantify the ROM error
6. **Extensibility:** should work for many application codes

Our research is focused on satisfying model reduction criteria for nonlinear dynamical systems



Our model reduction research at Sandia

- **Accuracy**
 - Least-Squares Petrov—Galerkin (LSPG) projection: *our baseline approach* [Carlberg, Bou-Mosleh, Farhat, 2011; Carlberg, Barone, Antil, 2017]
- **Low cost**
 - Sample mesh: *use a fraction of the data for evaluating nonlinear functions* [Carlberg, Farhat, Cortial, Amsallem, 2013]
 - Space-time LSPG projection: *learn and exploit structure in spatial and temporal data* [Carlberg, Ray, van Bloemen Waanders, 2015; Carlberg, Brencher, Haasdonk, Barth, 2017; Choi and Carlberg, 2019]
- **Property preservation**
 - Impose additional physical constraints (e.g. conservation) [Carlberg, Tuminaro, Boggs, 2015; Peng and Carlberg, 2017; Carlberg, Choi, Sargsyan, 2018]
- **Generalization**
 - Projection onto nonlinear manifolds: *high capacity nonlinear approximation* [Lee, Carlberg, 2018]
 - *b*-adaptivity: *trade cost for accuracy* [Carlberg, 2015; Etter and Carlberg, 2019]
- **Certification**
 - Machine learning error model: *quantify reduced model uncertainties* [Drohmann and Carlberg, 2015; Trehan, Carlberg, Durlofsky, 2017; Freno and Carlberg, 2019; Pagani, Manzoni, Carlberg, 2019; Parish and Carlberg, 2019]
- **Extensibility**
 - Pressio software: *deploy methods for many application codes*

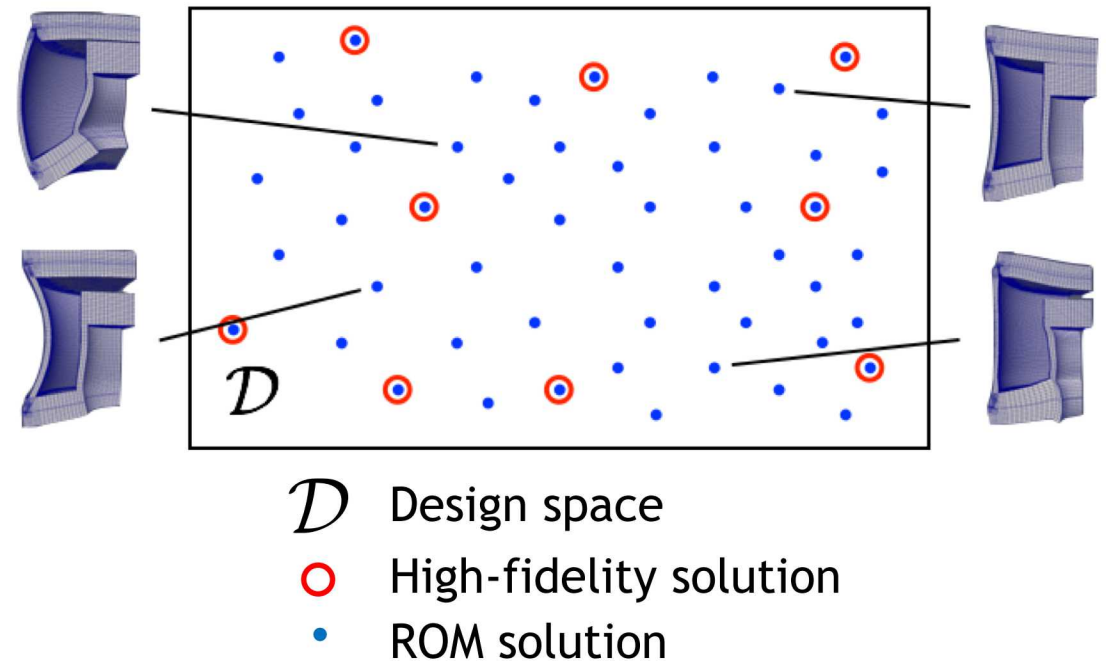
Model Reduction Criteria

1. **Accuracy**: achieves less than 1% error
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We employ a machine-learned model reduction approach that has four stages



1. **Acquisition:** Run high-fidelity simulation at a few design points, save simulation data
2. **Learning:** Use machine learning techniques to identify low-dimensional structure in the high-fidelity simulation data
3. **Reduction:** Build a reduced-order model (ROM) with extracted data structures, high-fidelity governing equations
4. **Deployment:** Use ROM at remaining design points

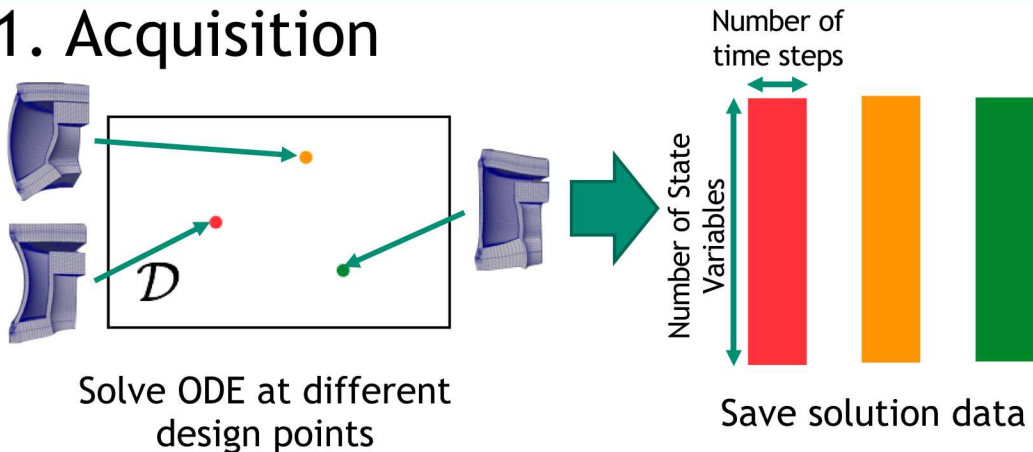


Our baseline approach* leverages a linear basis computed with unsupervised learning



- High-fidelity simulation = Ordinary Differential Equation (ODE): $\frac{d\mathbf{x}}{dt} = \mathbf{f}(\mathbf{x}; t, \mu)$

1. Acquisition



2. Learning

Unsupervised Learning with Principal Component Analysis (PCA):

$$\mathbf{X} = \begin{bmatrix} \text{red bar} & \text{orange bar} & \text{green bar} \end{bmatrix} = \Phi \mathbf{U} \quad \Sigma \quad \mathbf{V}^T$$

3. Reduction

Choose ODE
Temporal
Discretization

$$\frac{d\mathbf{x}}{dt} = \mathbf{f}(\mathbf{x}; t, \mu)$$



$$\mathbf{r}^n(\mathbf{x}^n; \mu) = \mathbf{0}, \quad n = 1, \dots, T$$

Reduce the
number of
unknowns

$$\mathbf{x}(t) \approx \tilde{\mathbf{x}}(t) = \Phi \hat{\mathbf{x}}(t)$$

Minimize the
Residual

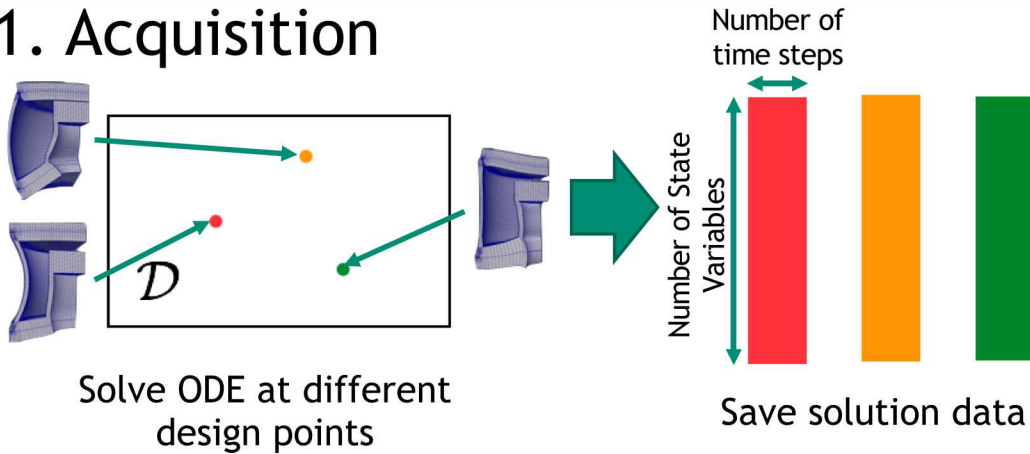
$$\underset{\hat{\mathbf{v}}}{\text{minimize}} \left\| \begin{bmatrix} \mathbf{A} \\ \mathbf{r}^n(\Phi \hat{\mathbf{v}}; \mu) \end{bmatrix} \right\|_2$$

Property preservation is enforced with additional constraints



- High-fidelity simulation = Ordinary Differential Equation (ODE): $\frac{dx}{dt} = f(x; t, \mu)$

1. Acquisition



2. Learning

Unsupervised Learning with Principal Component Analysis (PCA):

$$\mathbf{X} = \begin{bmatrix} \text{red bar} & \text{orange bar} & \text{green bar} \end{bmatrix} = \Phi \mathbf{U} \quad \Sigma \quad \mathbf{V}^T$$

3. Reduction

Choose ODE
Temporal
Discretization

$$\frac{dx}{dt} = f(x; t, \mu)$$



$$\mathbf{r}^n(\mathbf{x}^n; \mu) = \mathbf{0}, \quad n = 1, \dots, T$$

Reduce the
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unknowns

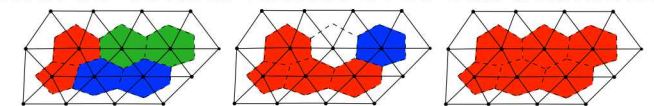
$$\mathbf{x}(t) \approx \tilde{\mathbf{x}}(t) = \Phi \hat{\mathbf{x}}(t)$$

Minimize the
Residual

$$\underset{\hat{\mathbf{v}}}{\text{minimize}} \quad \|\mathbf{A} \mathbf{r}^n(\Phi \hat{\mathbf{v}}; \mu)\|_2^2$$

$$\text{s.t. } \mathbf{C} \mathbf{r}^n(\Phi \hat{\mathbf{v}}; \mu) = \mathbf{0}$$

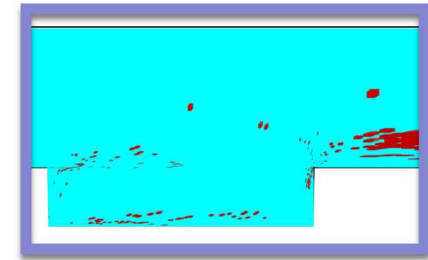
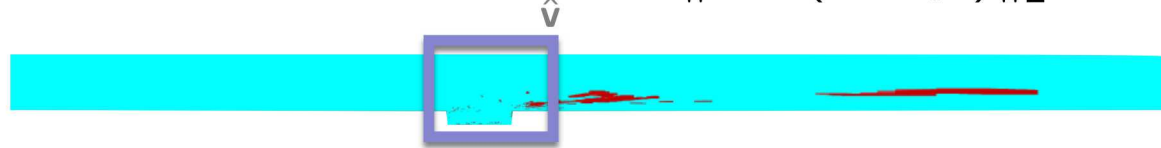
Enforce conservation over subdomains:



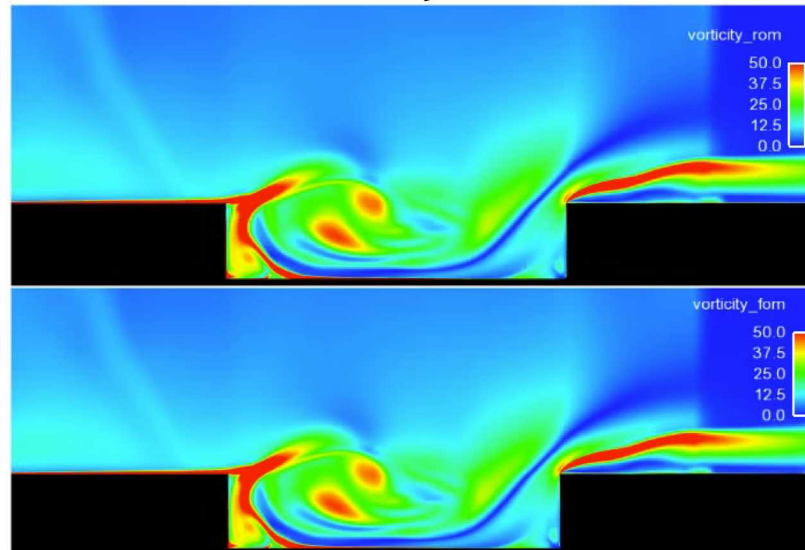
Our baseline approach achieves high accuracy at a low cost for captive carry application

$$\text{LSPG: minimize } \|\mathbf{A} \mathbf{r}^n(\Phi \hat{\mathbf{v}}; \mu)\|_2^2$$

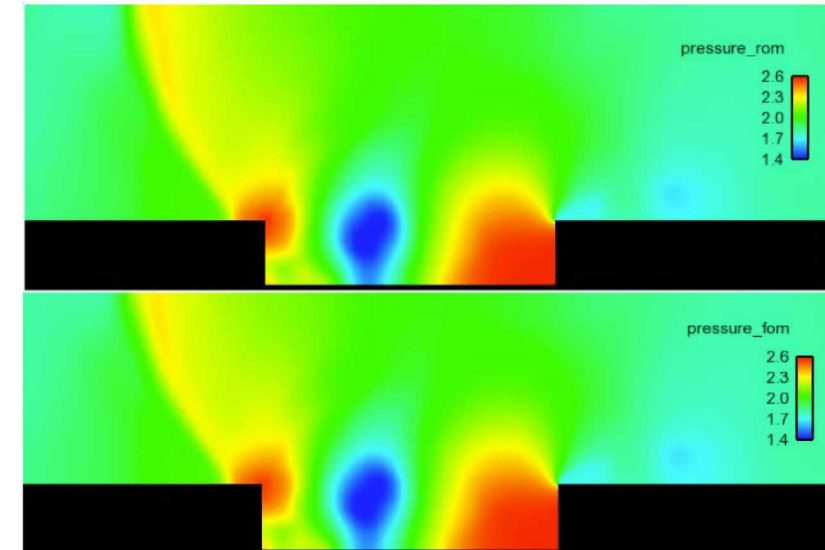
Sample mesh



Vorticity Field



Pressure Field



LSPG ROM

- 32 min, 2 cores

High-fidelity

- 5 hours, 48 cores

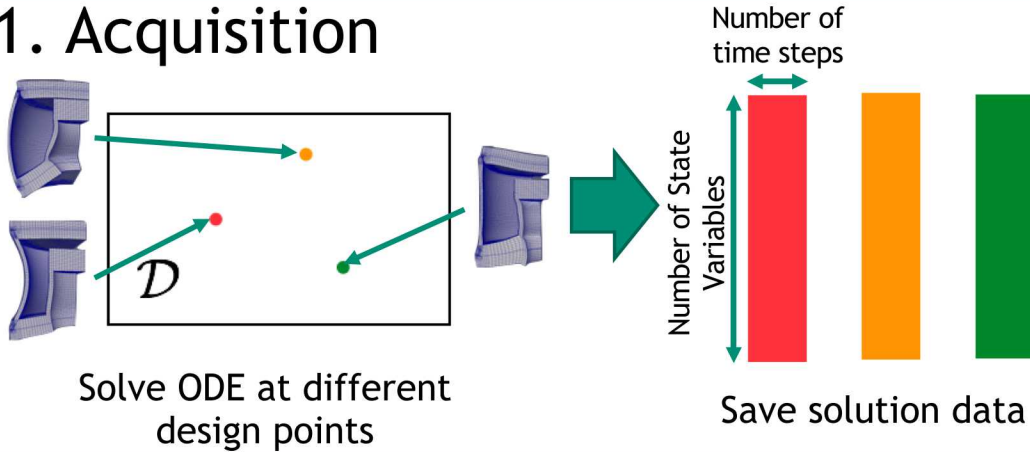
229x savings in core-hours
< 1% error in time-averaged drag

[Carlberg, Barone, Antil, 2017]

Manifold model reduction uses a nonlinear function instead of a linear basis

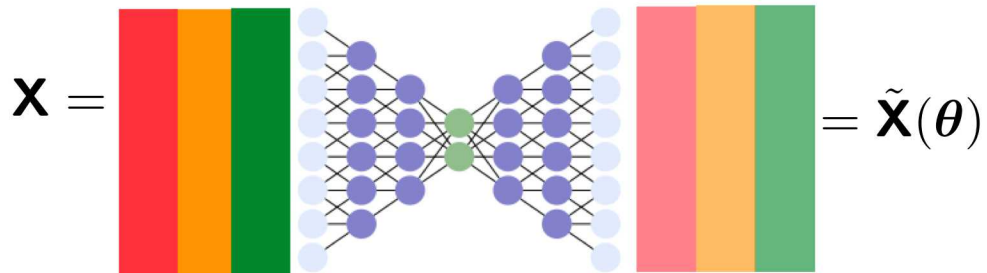
- High-fidelity simulation = Ordinary Differential Equation (ODE): $\frac{d\mathbf{x}}{dt} = \mathbf{f}(\mathbf{x}; t, \boldsymbol{\mu})$

1. Acquisition



2. Learning

Unsupervised Learning with non-linear manifold approach (e.g. deep autoencoder):



3. Reduction

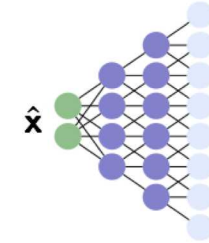
Choose ODE
Temporal
Discretization

$$\frac{d\mathbf{x}}{dt} = \mathbf{f}(\mathbf{x}; t, \boldsymbol{\mu})$$



$$\mathbf{r}^n(\mathbf{x}^n; \boldsymbol{\mu}) = \mathbf{0}, \quad n = 1, \dots, T$$

Reduce the
number of
unknowns



$$\mathbf{x}(t) \approx \tilde{\mathbf{x}}(t) = \mathbf{g}(\hat{\mathbf{x}}(t)) \in \mathcal{S}$$

Minimize the
Residual

$$\underset{\hat{\mathbf{v}}}{\text{minimize}} \left\| \begin{bmatrix} \mathbf{A} \end{bmatrix} \mathbf{r}^n(\mathbf{g}(\hat{\mathbf{v}}); \boldsymbol{\mu}) \right\|_2$$

We achieve large improvements in the generalization criteria with manifold model reduction



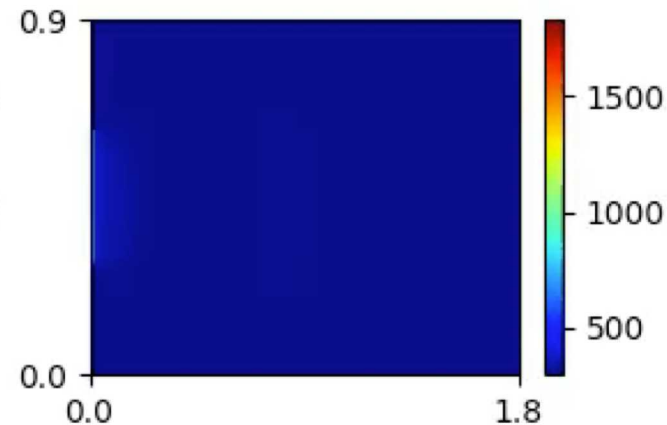
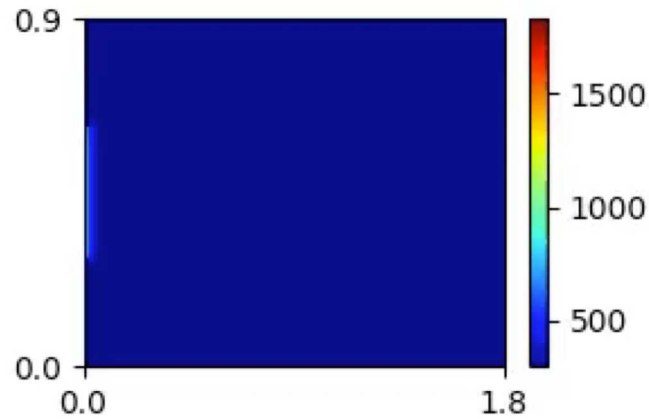
2D Chemically reacting
flow

$$\frac{\partial \mathbf{w}(\vec{x}, t; \mu)}{\partial t} = \nabla \cdot (\kappa \nabla \mathbf{w}(\vec{x}, t; \mu)) - \mathbf{v} \cdot \nabla \mathbf{w}(\vec{x}, t; \mu) + \mathbf{q}(\mathbf{w}(\vec{x}, t; \mu); \mu)$$

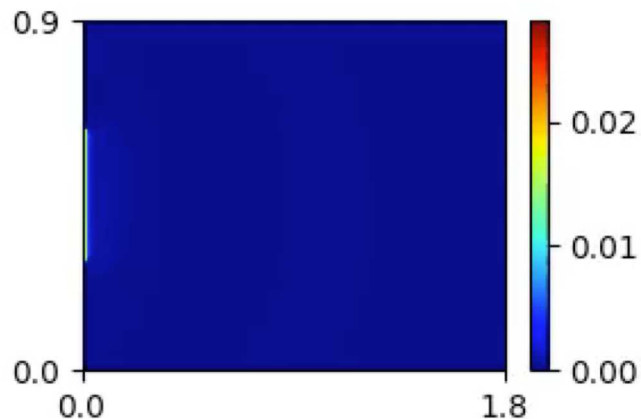
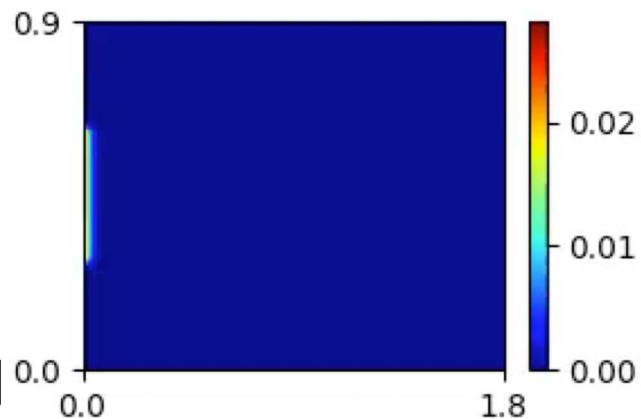
high-fidelity
model

LSPG w/ PCA
ROM dimension=5

temperature



H2 fraction

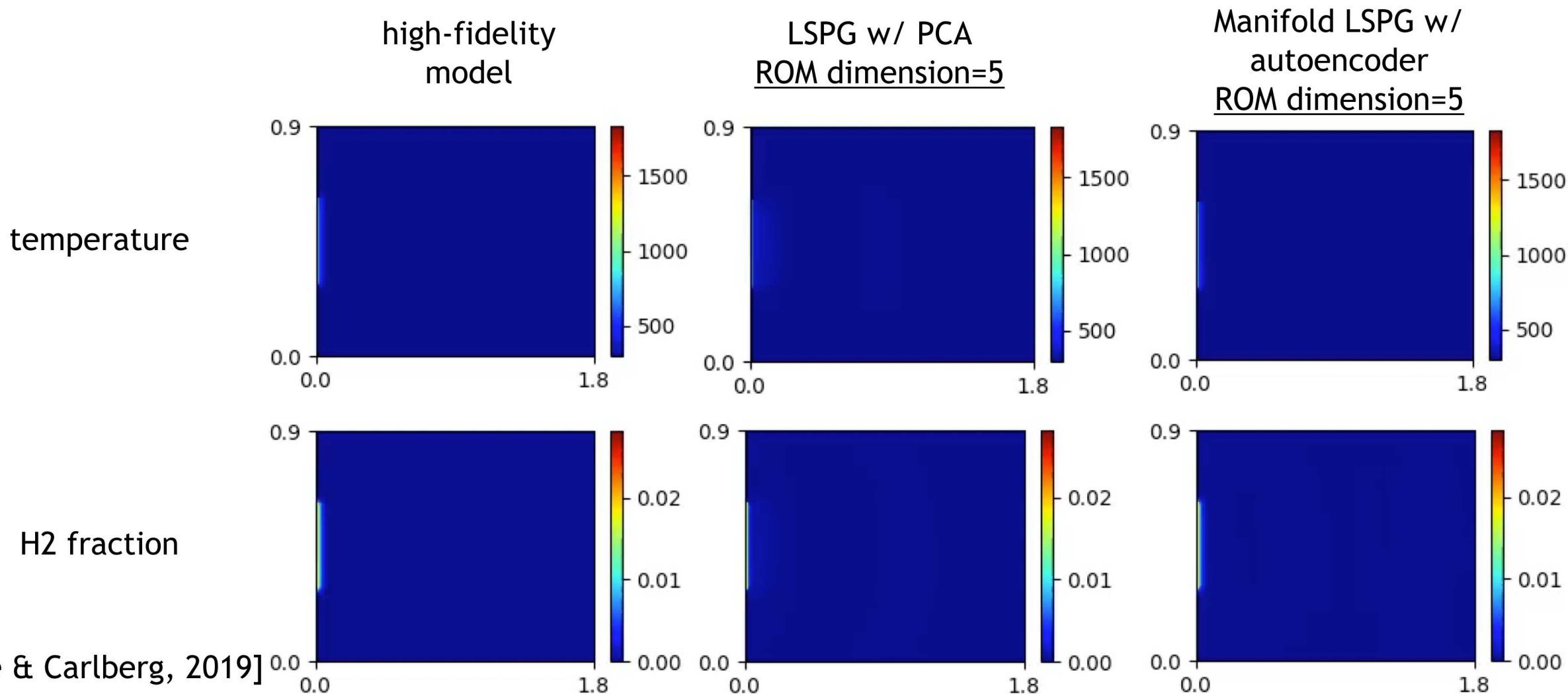


We achieve large improvements in the generalization criteria with manifold model reduction



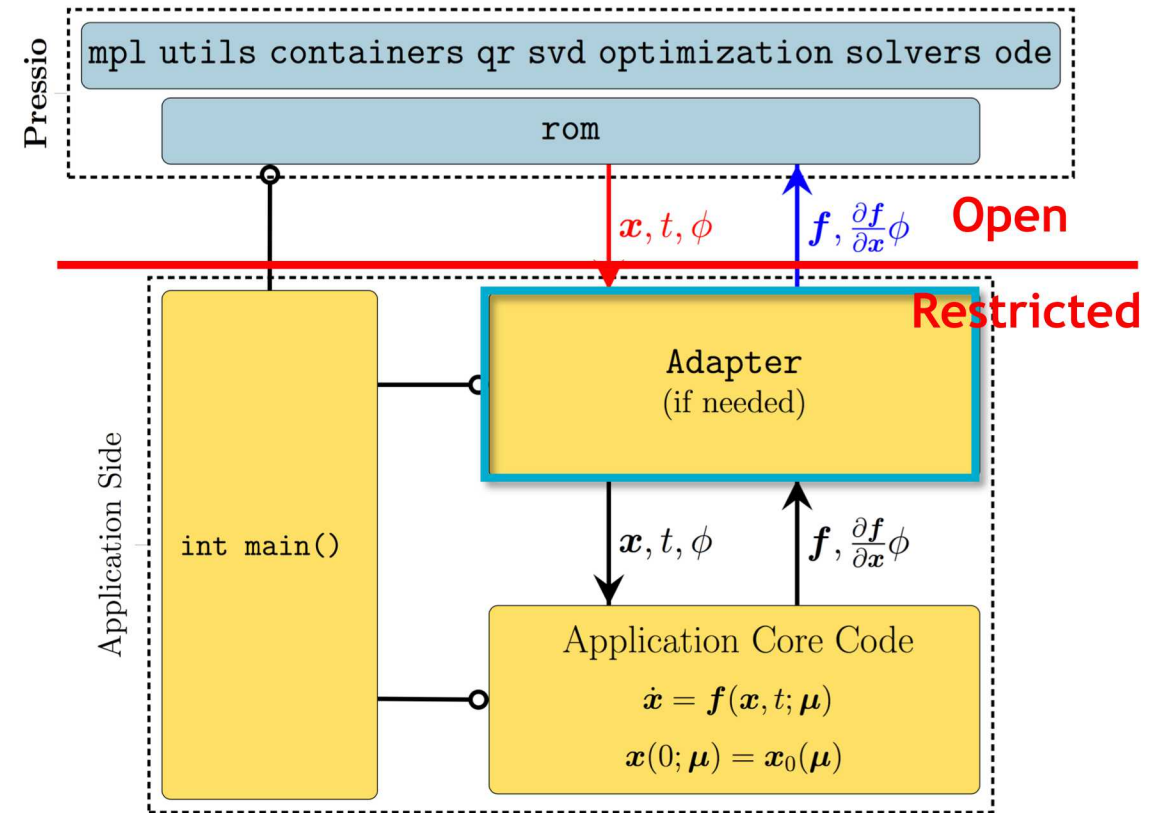
2D Chemically reacting
flow

$$\frac{\partial \mathbf{w}(\vec{x}, t; \mu)}{\partial t} = \nabla \cdot (\kappa \nabla \mathbf{w}(\vec{x}, t; \mu)) - \mathbf{v} \cdot \nabla \mathbf{w}(\vec{x}, t; \mu) + \mathbf{q}(\mathbf{w}(\vec{x}, t; \mu); \mu)$$



Pressio enables deployment of model reduction methods to a range of simulation codes

- Previous ROM methods were implemented directly in multiple application codes
 - ✗ **Highly intrusive**: major changes to application code
 - ✗ **Not extensible**: individual ROM implementation for each application
 - ✗ **Access requirements**: developers need direct access to application
- Pressio, a software package that addresses all three of these issues:
 - ✓ Minimally intrusive method implementation.
 - ✓ Leverages modern software engineering practices (e.g. C++ template-metaprogramming)
 - Portable implementation that works on different architectures, including GPUs
 - Restricted to practices used by mission application partners
 - ✓ Facilitates contributions from external partners
 - Undergoing open source copyright assertion
 - ✓ Clear separation between methods and application
 - Enables methods work without access to restricted applications (ITAR, Classified, etc.)



Schematic of Pressio software workflow (F. Rizzi)

We are building Pressio adapters for three simulation codes



SPARC

Hypersonic Aerothermodynamics

- Key Personnel: P. Blonigan, M. Howard, J. Fike, F. Rizzi
- Progress: creating and running ROMs for aerodynamics



ARIA

Heat Transfer

- Key Personnel: J. Tencer, F. Pierce, F. Rizzi
- Progress: interface complete, setting up basis computation



Sierra Aero

Compressible Aerodynamics

- Key Personnel: J. Tencer, C. Proctor, M. McWherter-Payne, P. Blonigan
- Progress: creating high fidelity models



We are integrating our model reduction tools with many-query and time-critical applications



❖ Multi-fidelity Uncertainty Quantification

- P. Blonigan, G. Geraci, and M. Eldred



❖ Network Uncertainty Quantification with ROMs for system-component design:

- J. Tencer, K. Carlberg, C. Proctor, M. McWherter-Payne, and P. Blonigan



❖ Optimization under uncertainty

- External collaboration, key personnel: M. Zahr, K. Carlberg, and D. Kouri
- [Zahr, Carlberg and Kouri, 2019].



UNIVERSITY OF
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⌚ Autonomy for hypersonics: path planning and adaptive control

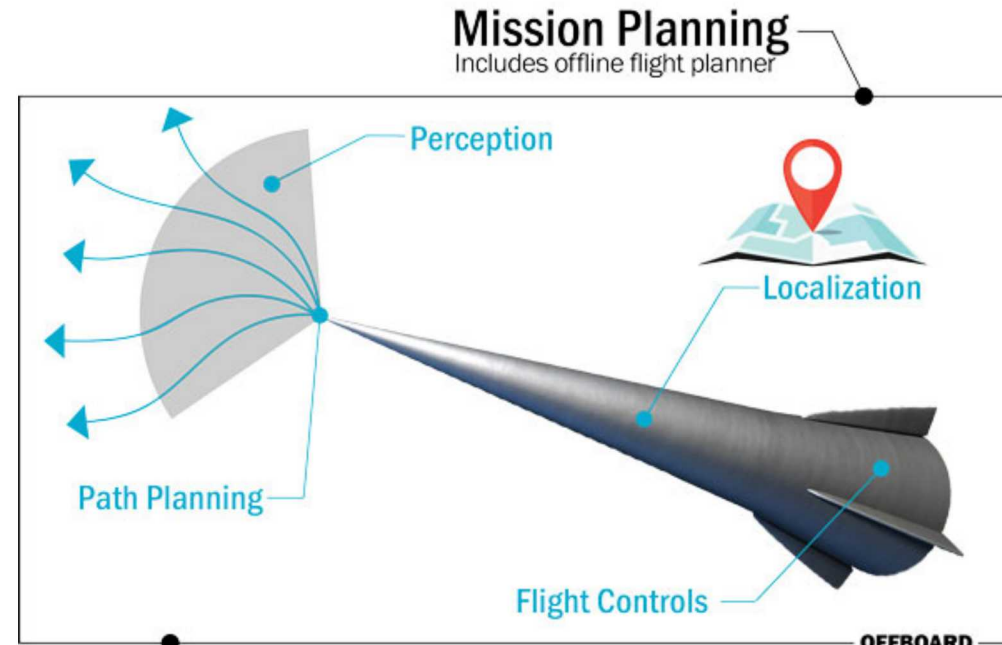
- P. Blonigan, K. Carlberg, M. Howard, J. Fike, and F. Rizzi



The Autonomy For Hypersonics (A4H) mission campaign focuses on time-critical problems for hypersonic vehicles



Autonomy for Hypersonics



Mission Analytics

Inform tactics & engagement strategies

(Image by Hannah Stangebye)

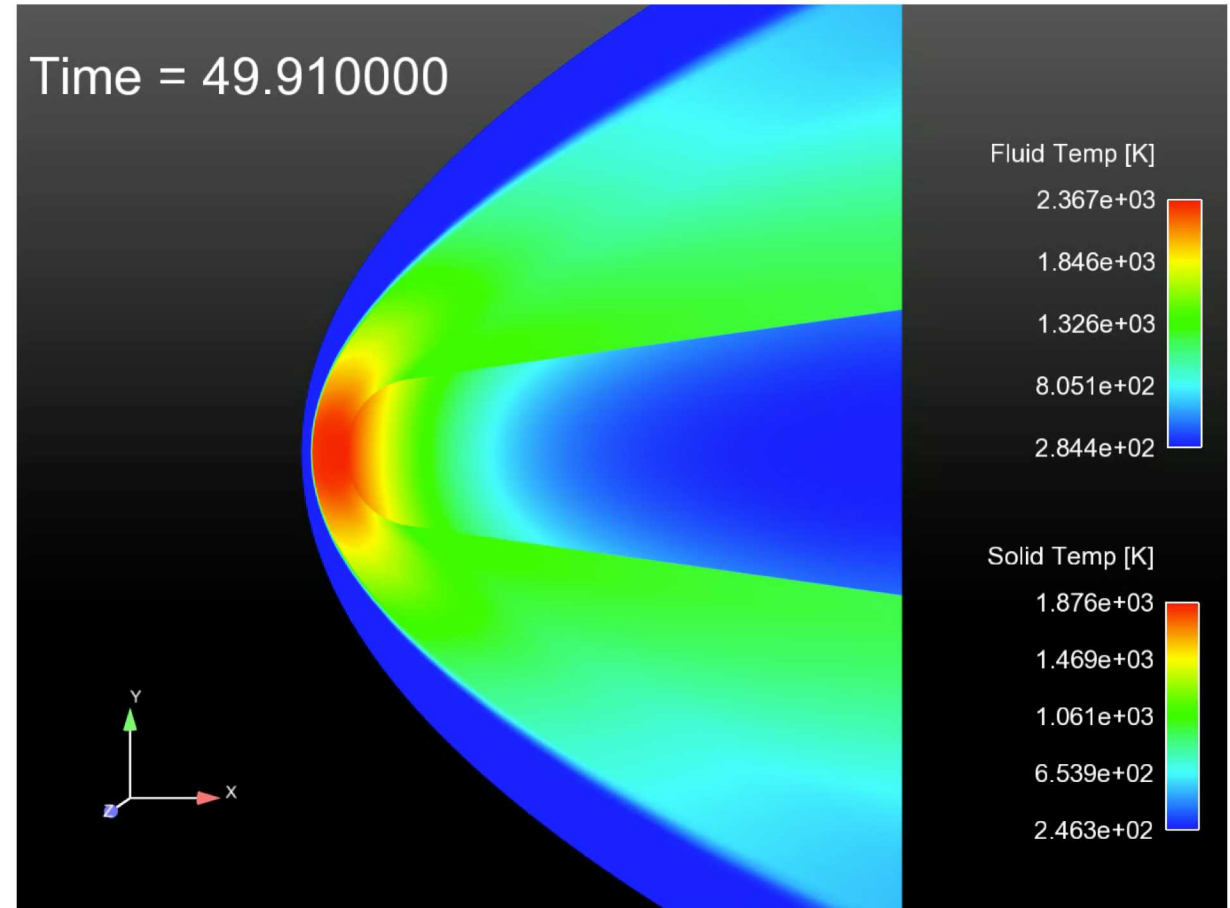
Sandia's A4H Mission Campaign seeks to:

- Significantly decrease the time required for hypersonic missile flight planning using artificial intelligence.
- Enable semi-autonomous hypersonic missiles to self-correct in flight to compensate for unexpected flight conditions or a change in the target's location.

➤ <http://www.sandia.gov/news/publications/labnews/articles/2019/04-26/hypersonics.html>

Our A4H project* will use our model reduction tool chain to accelerate path planning, design, and control of hypersonic vehicles

- Our project uses model reduction to:
 1. Generate large databases with quantified uncertainties for path planning.
 2. Enable rapid interactive simulation for vehicle design and control.
- We use Pressio (F. Rizzi) and SPARC (M. Howard) to create ROMs for hypersonic aerodynamics
- Joint work with aerosciences team (M. Howard, J. Fike) and UT Austin (K. Willcox, S. Majors)

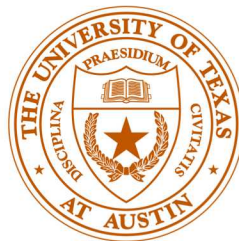


A slender body in hypersonic flow simulated with SPARC (courtesy M. Howard)



LDRD

Laboratory Directed Research and Development

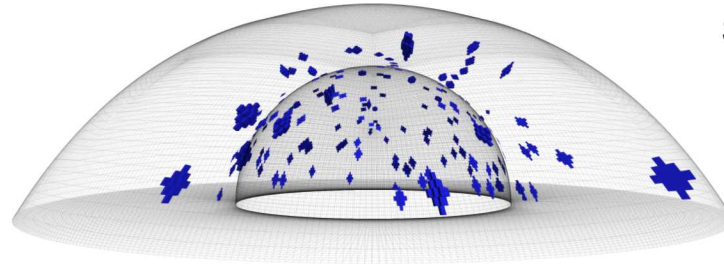


*Rapid high-fidelity aerothermal responses with quantified uncertainties via reduced-order modeling

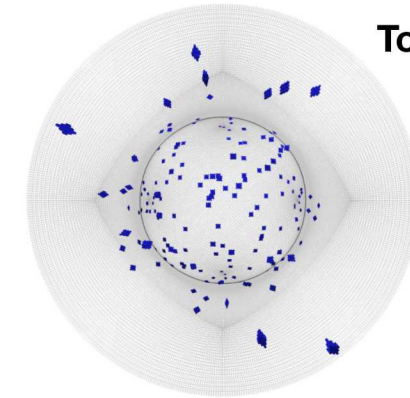
Preliminary results show that our model reduction tool chain will be effective for this application space

Blottner Sphere: ▸ Unsteady* Navier–Stokes ▸ $Re = 1.89 \times 10^6$ ▸ $M_\infty = 5.0$

Sample mesh



Side view



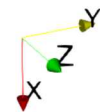
Top view

LSPG ROM:

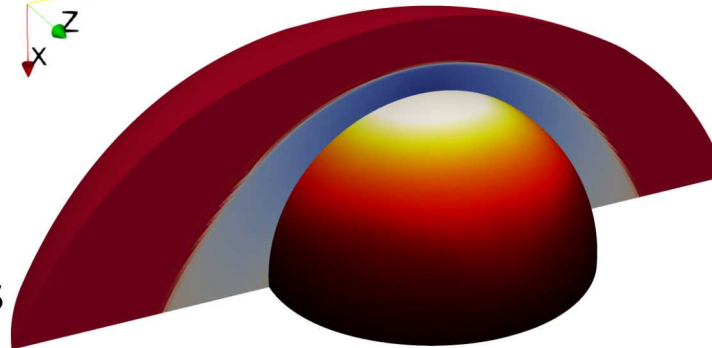
- Sample mesh: 4,150 cells= 20,750 DoFs
- 1 MPI rank, ~18 seconds

High-fidelity:

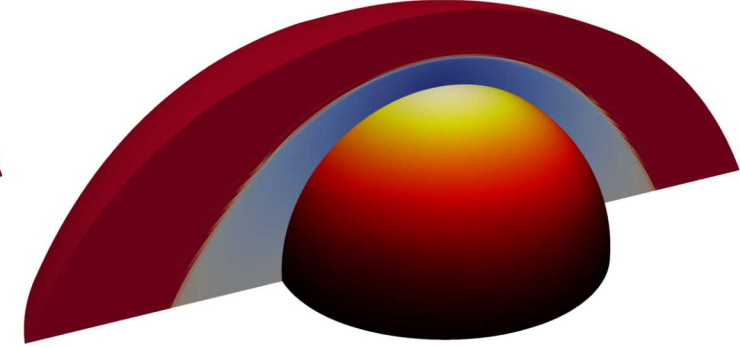
- 4,194,304 cells=20,971,520 DoFs
- 128 MPI ranks, ~147 seconds



LSPG ROM



High-fidelity



Ma

wall-heat-flux

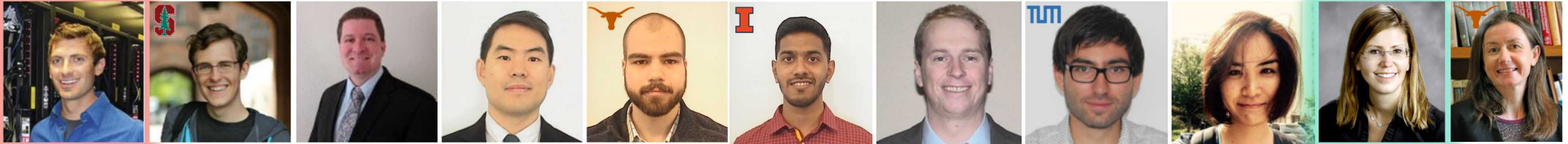
1060x savings in core-hours

< 1% error in density, Mach number, and temperature fields

< 1% error in axial force, heat flux

Model reduction at Sandia is a large multidisciplinary effort supported by researchers spanning centers and institutions

- **Applied Math:** method development and analysis



- **Computer Science:** generalized, minimally intrusive model reduction implementation



- **Engineering science:** deployment of model reduction in engineering applications and analysis



Future work will continue to span computer science, engineering science, and applied math research and development



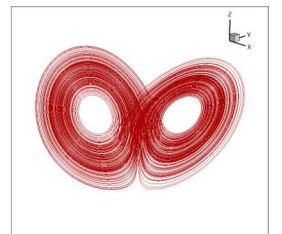
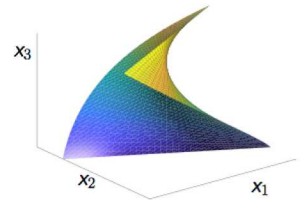
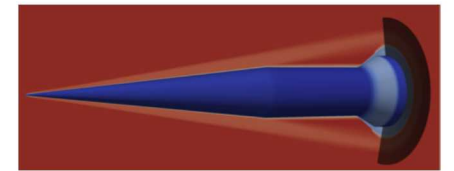
•Computer Science R&D:

- Adding existing methods to Pressio
- Pathway to production for Pressio



•Engineering Science R&D:

- Apply Pressio to increasingly complex physical systems
- New Pressio adaptors for additional simulation capabilities
- Integration of model reduction techniques with time-critical/many-query methods/frameworks:
 - Network UQ
 - Multi-fidelity UQ
 - Verification and Validation, Other UQ/Optimization approaches



•Applied Math R&D:

- Nonlinear manifold model reduction methods
- Develop methods for chaotic dynamical systems

•Maintain current projects and create new projects with internal and external collaborators.

Summary of accomplishments and key references



Team grew from 3 in FY12 to 21 in FY19



2 keynote presentations
34 conference presentations
32 invited talks



23 Journal publications (7 in review)
6 Conference publications



#1 most-cited paper, 2011, IJNME
#1 most-cited paper, 2013, JCP
Featured article, 2015, SIAM JSC
Top 5 most-cited paper, 2017, JCP

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Contact information:

- Patrick Blonigan
- pblonig@sandia.gov / 925-294-6707
- <https://www.sandia.gov/~pblonig>
- ROM group email: wg-rom-group@sandia.gov

Backup slides



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Manifold LSPG projection uses a nonlinear function instead of a linear basis, resulting in more capacity [Lee & Carlberg, 2019]

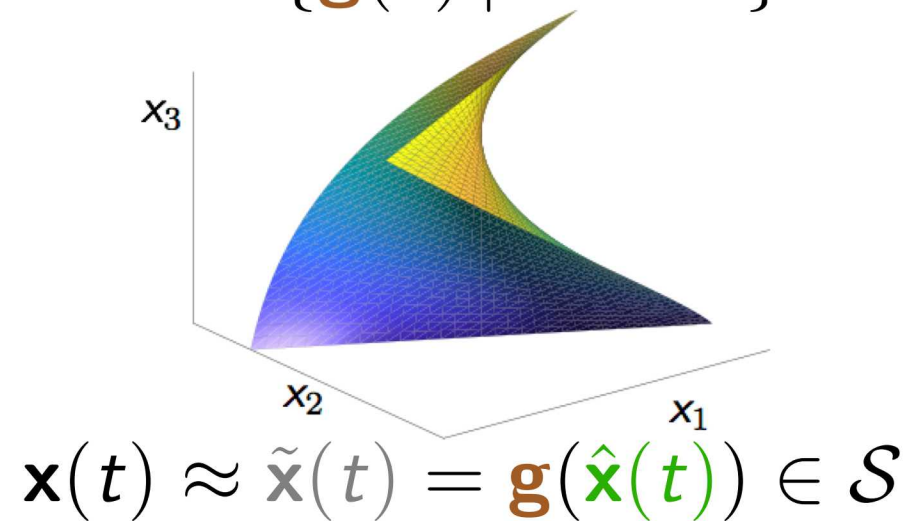
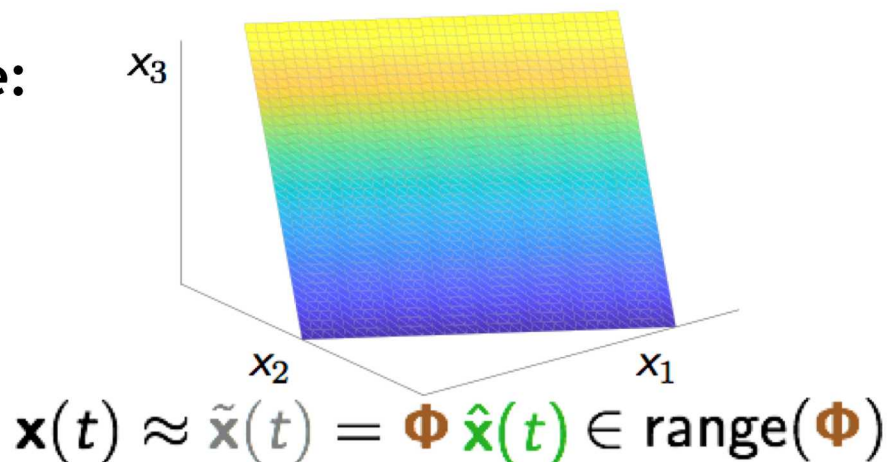
$$\text{range}(\Phi) := \{\Phi \hat{\mathbf{x}} \mid \hat{\mathbf{x}} \in \mathbb{R}^p\}$$

$$\mathcal{S} := \{\mathbf{g}(\hat{\mathbf{x}}) \mid \hat{\mathbf{x}} \in \mathbb{R}^p\}$$

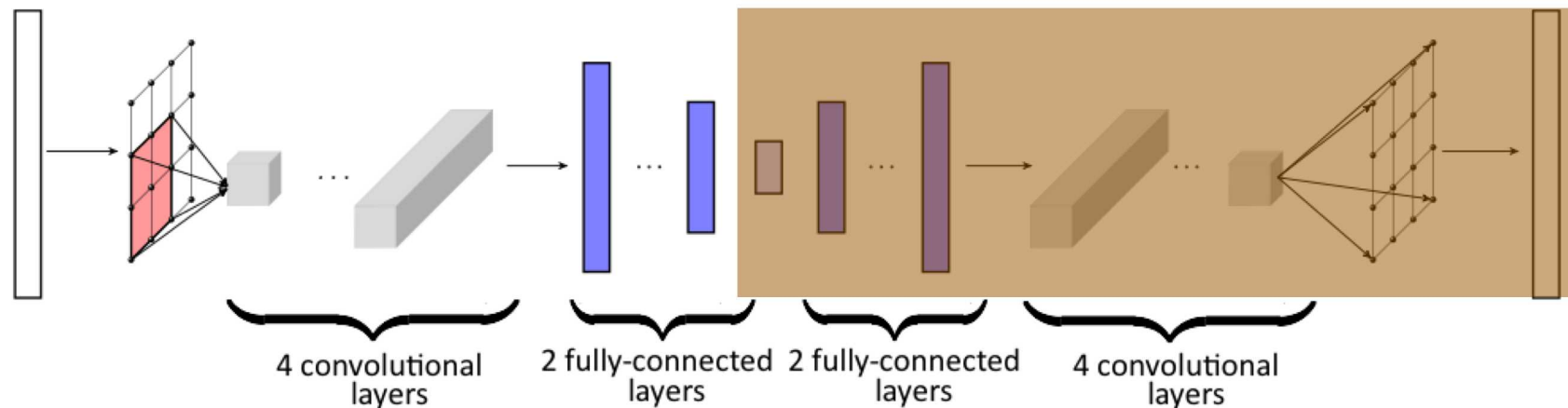
Example:

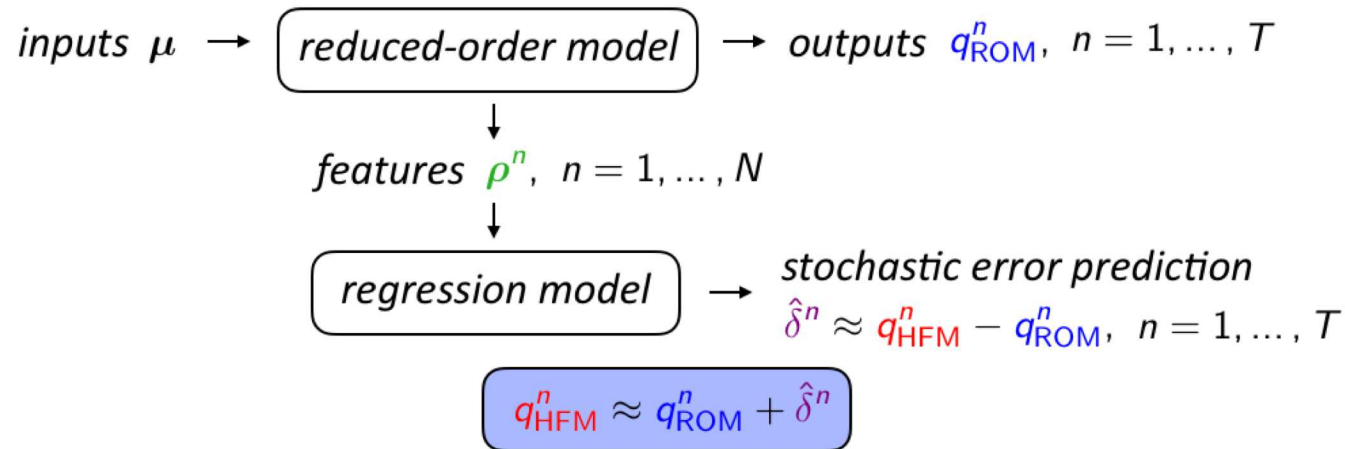
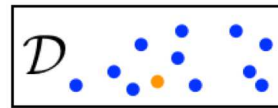
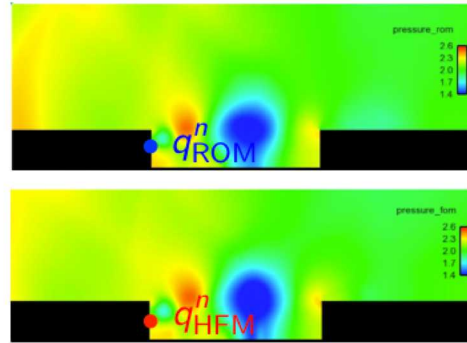
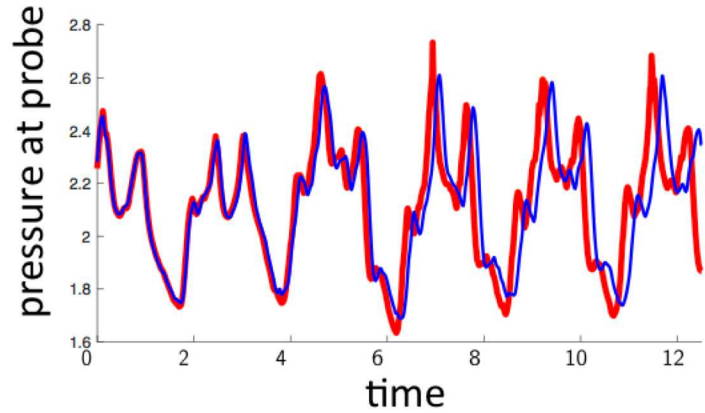
N=3

P=2



Decoder:
One choice of
nonlinear
function

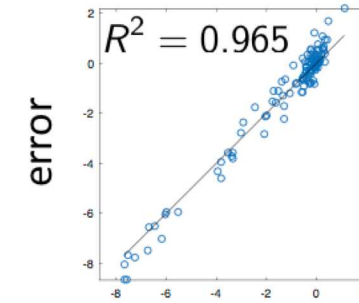




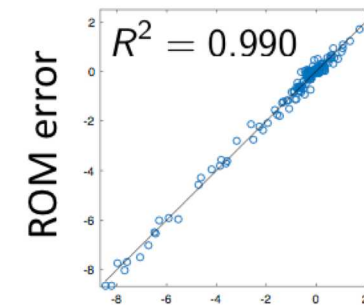
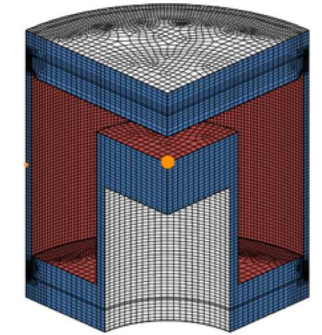
+ Statistical model of high-fidelity-model output

Physics-based feature engineering to determine ρ^n

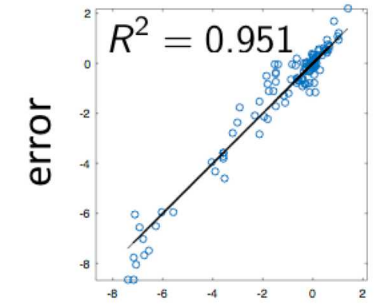
Example: Component Structural Model



random forest
error prediction



support vector machine
error prediction



k-NN
error prediction

ML methods yield low-variance error predictions

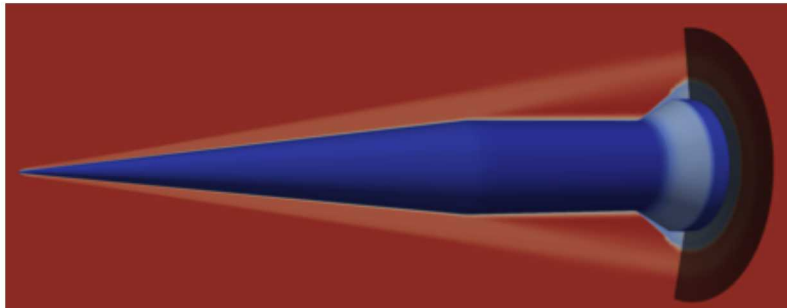
Our vision statement informs our current and future research

Establish a research-to-production capability, based on projection-based reduced-order models (ROMs), that enables deployment of high-fidelity physics & engineering simulations in time-critical (e.g., control, rapid analysis) and many-query applications (e.g., uncertainty quantification, design optimization, parameter-space exploration), in support of the Department of Energy mission.



Software-specific Technology Readiness Level (TRL) <https://trl.sandia.gov>

$$\mathbf{x} = \begin{bmatrix} \text{red} \\ \text{orange} \\ \text{green} \end{bmatrix} \rightarrow \text{Neural Network} \rightarrow \begin{bmatrix} \text{pink} \\ \text{orange} \\ \text{green} \end{bmatrix} = \tilde{\mathbf{x}}(\theta)$$

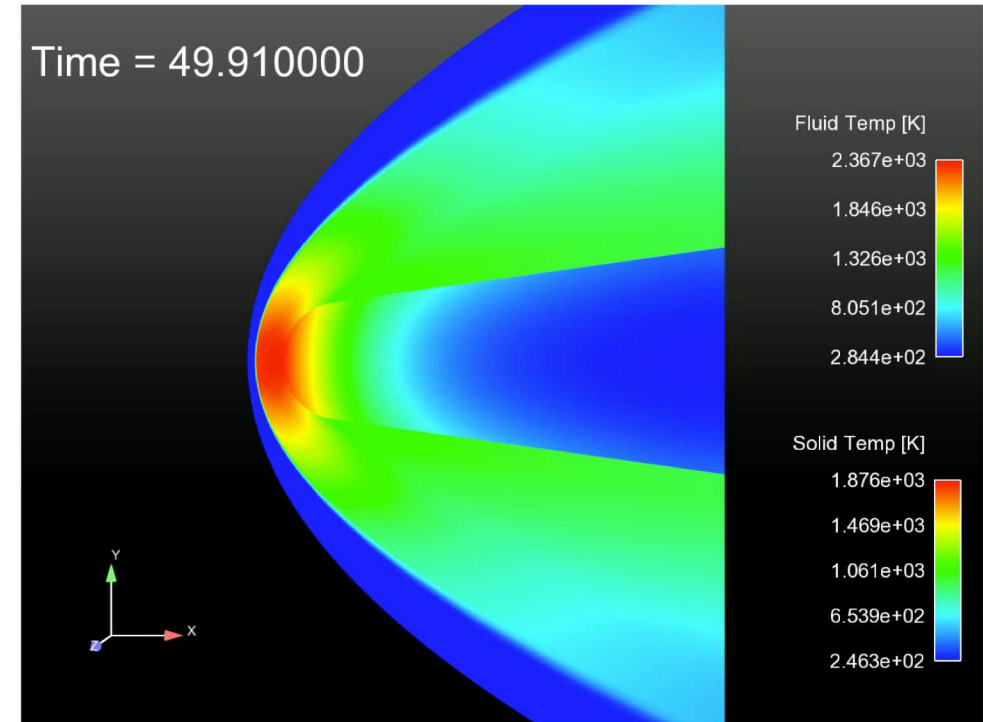


U.S. DEPARTMENT OF
ENERGY

Sandia Parallel Aerodynamics and Reentry Code (SPARC)



- Compressible CFD code focused on aerodynamics and aerothermodynamics in the Transonic and Hypersonic regimes
 - Being developed to run on today's leadership-class supercomputers and exascale machines.
 - Performance portability: SPARC leverages Kokkos to run on multiple machines with different architectures (e.g. CPU vs. CPU/GPU)
- Physics Capabilities include:
 - **Navier—Stokes, cell-centered finite volume method**
 - **Reynolds-Averaged Navier—Stokes (RANS) , cell-centered finite volume method**
 - Transient Heat Equation, Galerkin finite element method.
 - Decomposing and non-decomposing ablation equations, Galerkin finite element method.
 - One and two-way coupling between ablation, heat equation, RANS.



A slender body in hypersonic flow simulated with SPARC