



Sandia
National
Laboratories

SAND2019-7991PE

Selective Lattice Destruction

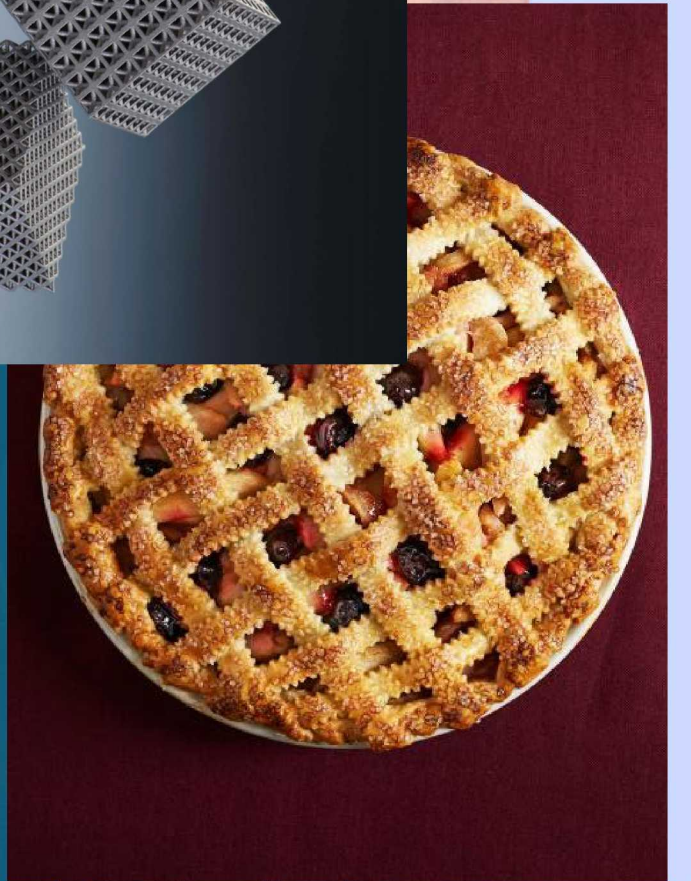
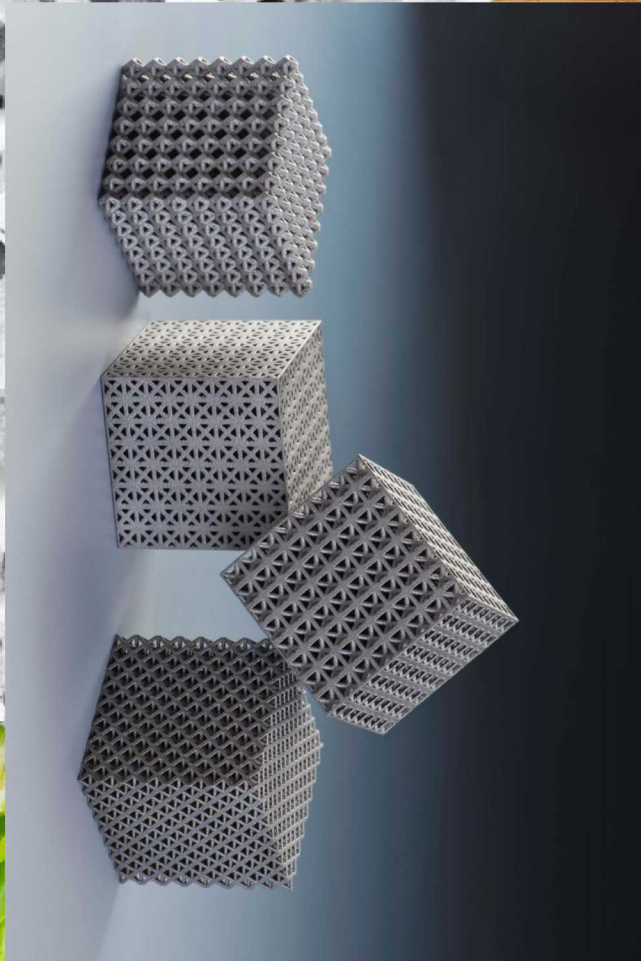
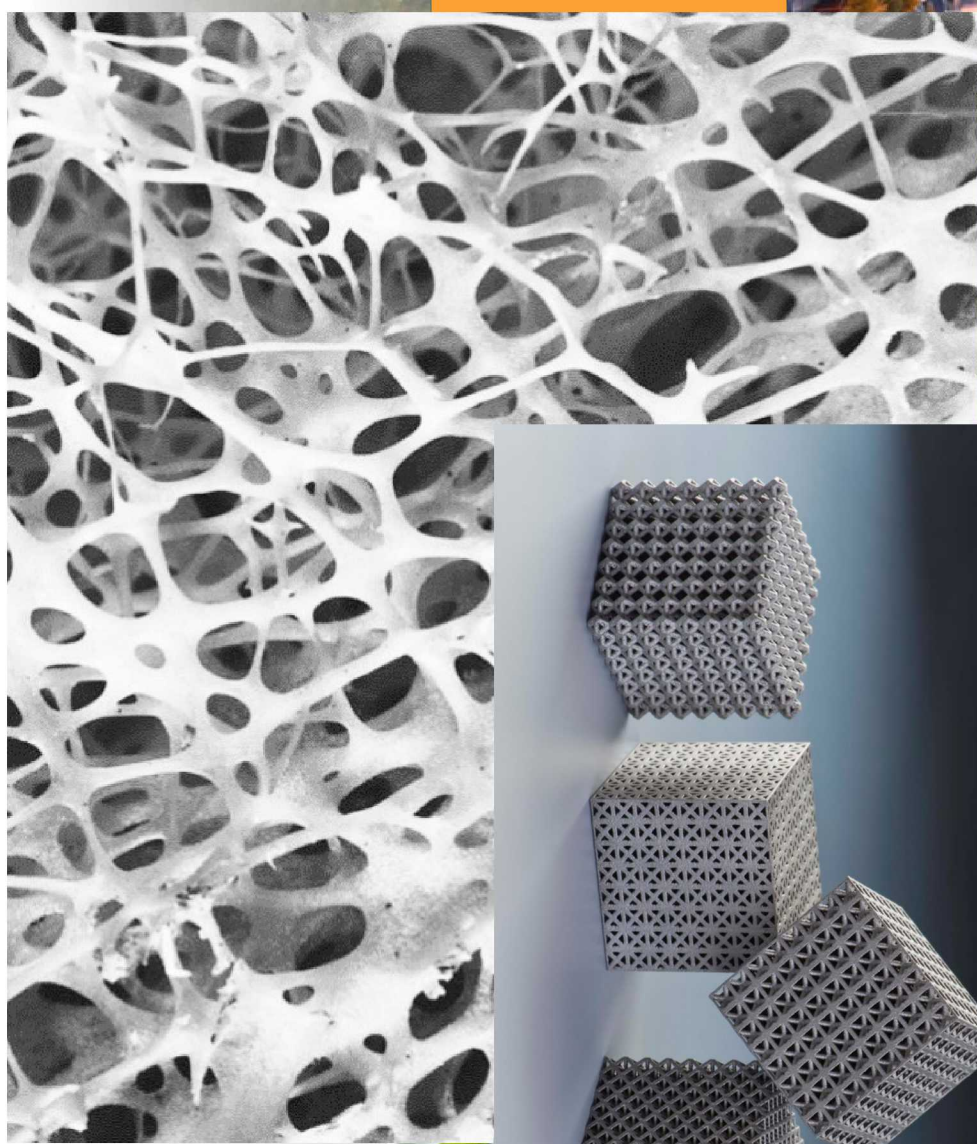


PRESENTED BY

1/c Patrick Irwin, USCGA



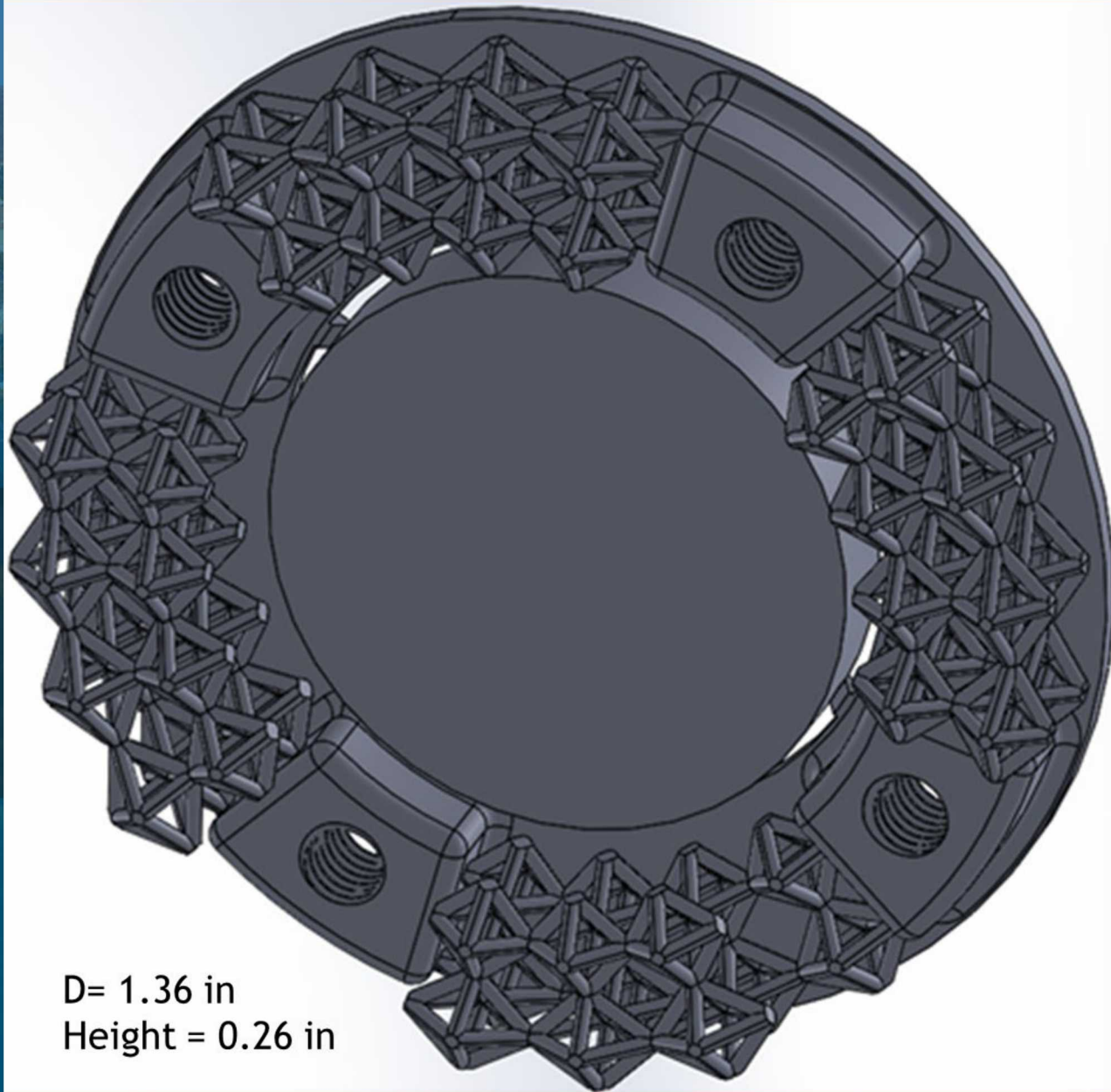
Sandia National Laboratories is a multimission laboratory managed and operated by National Technology & Engineering Solutions of Sandia, LLC, a wholly owned subsidiary of Honeywell International Inc., for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-NA0003525.



Manufacturing Lattices







D= 1.36 in
Height = 0.26 in

Motivation & Agenda



Stiffness Equation:



$$(1) \mathbf{k} = \frac{\mathbf{F}}{\delta}$$

Where
 k is stiffness (lb/in)
 F is force applied (lb)
 δ is displacement (in)

$$(2) \mathbf{\sigma} = \frac{\mathbf{F}}{\mathbf{A}}$$

Where
 σ is stress (psi)
 F is force applied (lb)
 A is cross sectional area (in²)

$$(3) \frac{\mathbf{\sigma}}{\mathbf{\epsilon}} = \mathbf{E}$$

Where
 σ is stress (psi)
 ϵ is strain
 E is Young's Modulus (psi)

$$(4) \mathbf{\epsilon} = \frac{\Delta \mathbf{L}}{\mathbf{L}}$$

Where
 ϵ is strain
 ΔL is change in length
 L is the length

$$\mathbf{E} = \frac{\mathbf{F/A}}{\mathbf{\epsilon}}$$

$$\frac{\mathbf{F/A}}{\mathbf{E}} = \frac{\Delta \mathbf{L}}{\mathbf{L}}$$

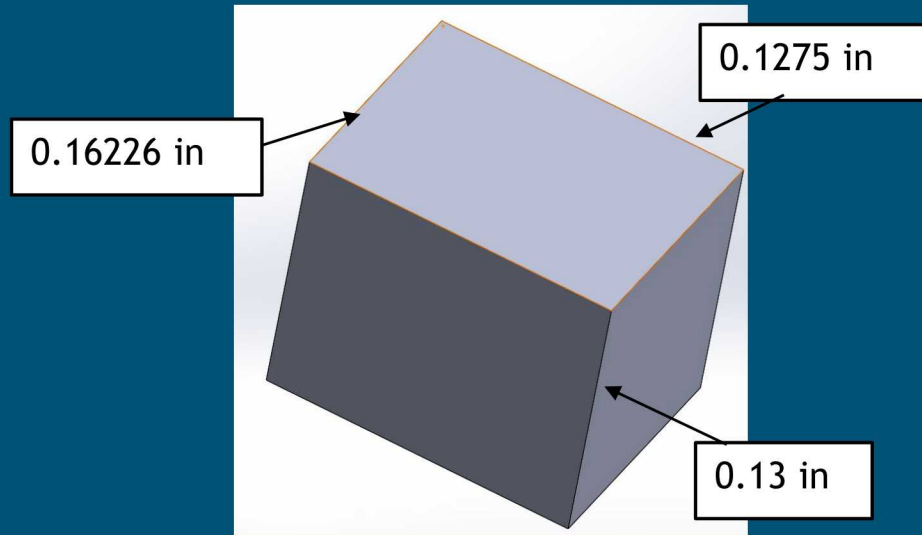
$$\Delta \mathbf{L} = \frac{\mathbf{LF}}{\mathbf{EA}}$$

$$\mathbf{k} = \frac{\mathbf{F}}{\frac{\mathbf{LF}}{\mathbf{EA}}}$$

$$\mathbf{k} = \frac{\mathbf{EA}}{\mathbf{L}}$$

Stiffness is found
 by dividing force
 and displacement

Model Validation:



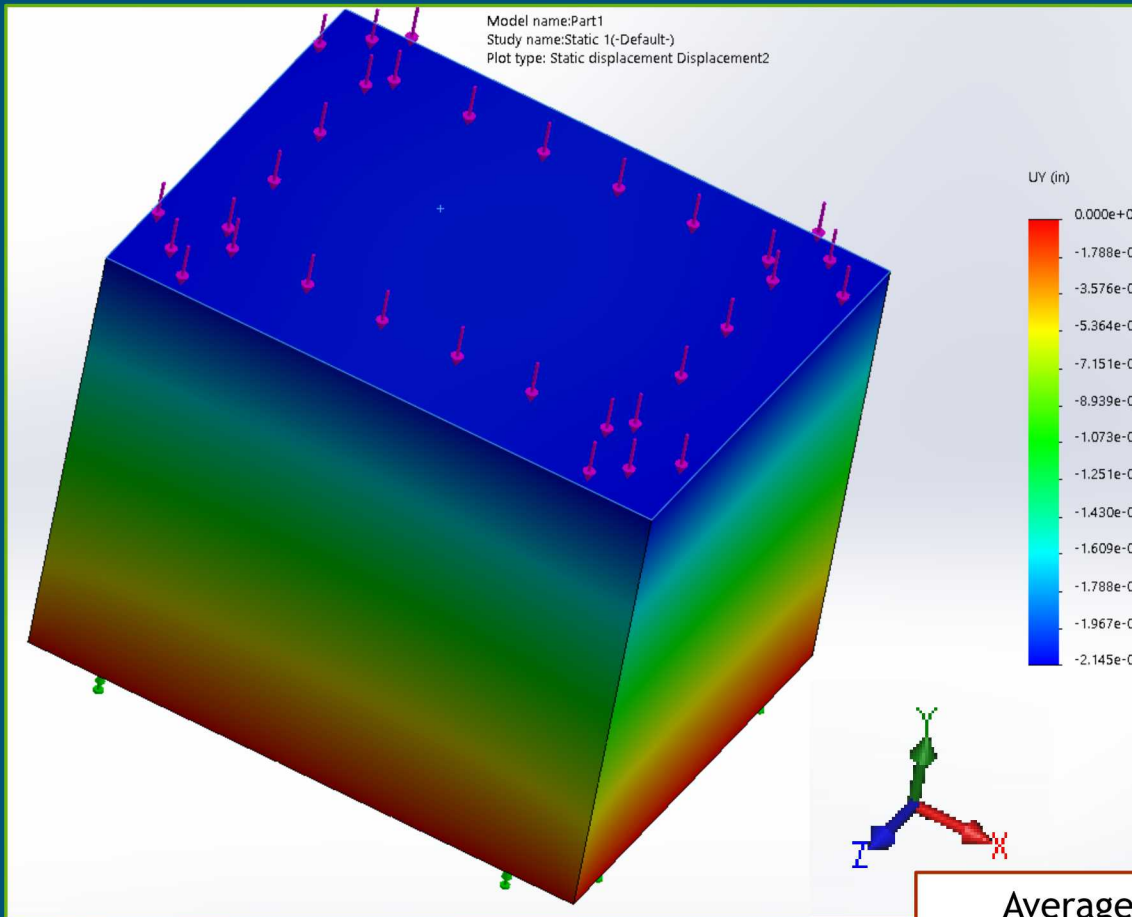
$$\begin{aligned} A &= (0.1275)(0.16226) \\ &= 0.02069 \text{ square inches} \end{aligned}$$

$$E = 29,000,000 \text{ psi}$$

$$k = \frac{(29 \times 10^6 \text{ psi})(0.02069 \text{ in}^2)}{0.13 \text{ in}}$$

$$\underline{k = 4.456 \times 10^6 \text{ lb/in}}$$

Model Validation:



Summary

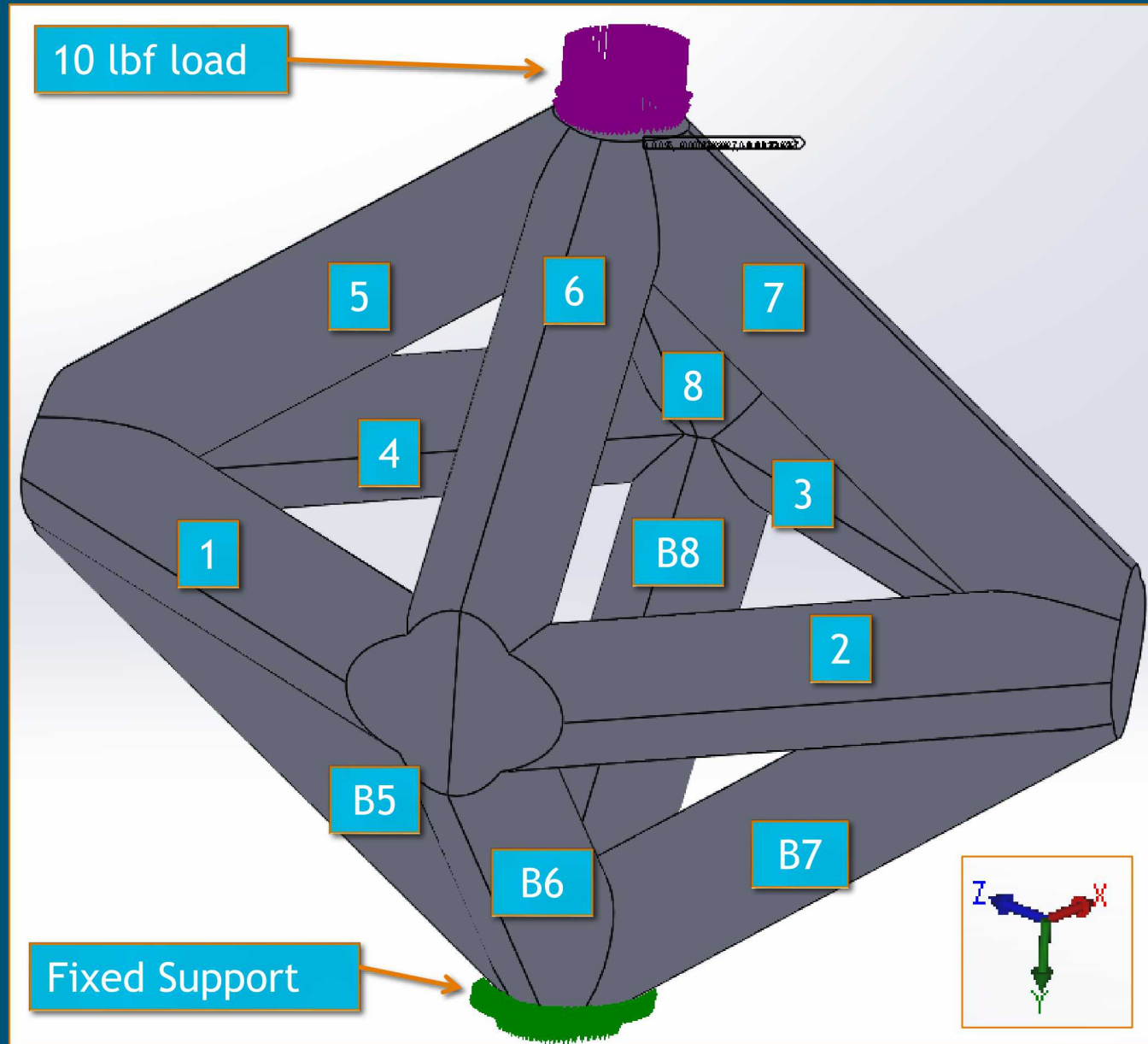
	Value	
Sum	-3.842e-03	in
Avg	-2.112e-06	in
Max	-2.089e-06	in
Min	-2.145e-06	in
RMS	2.112e-06	in

Average displacement in y direction is -
 2.112×10^{-6} , thus the stiffness is:

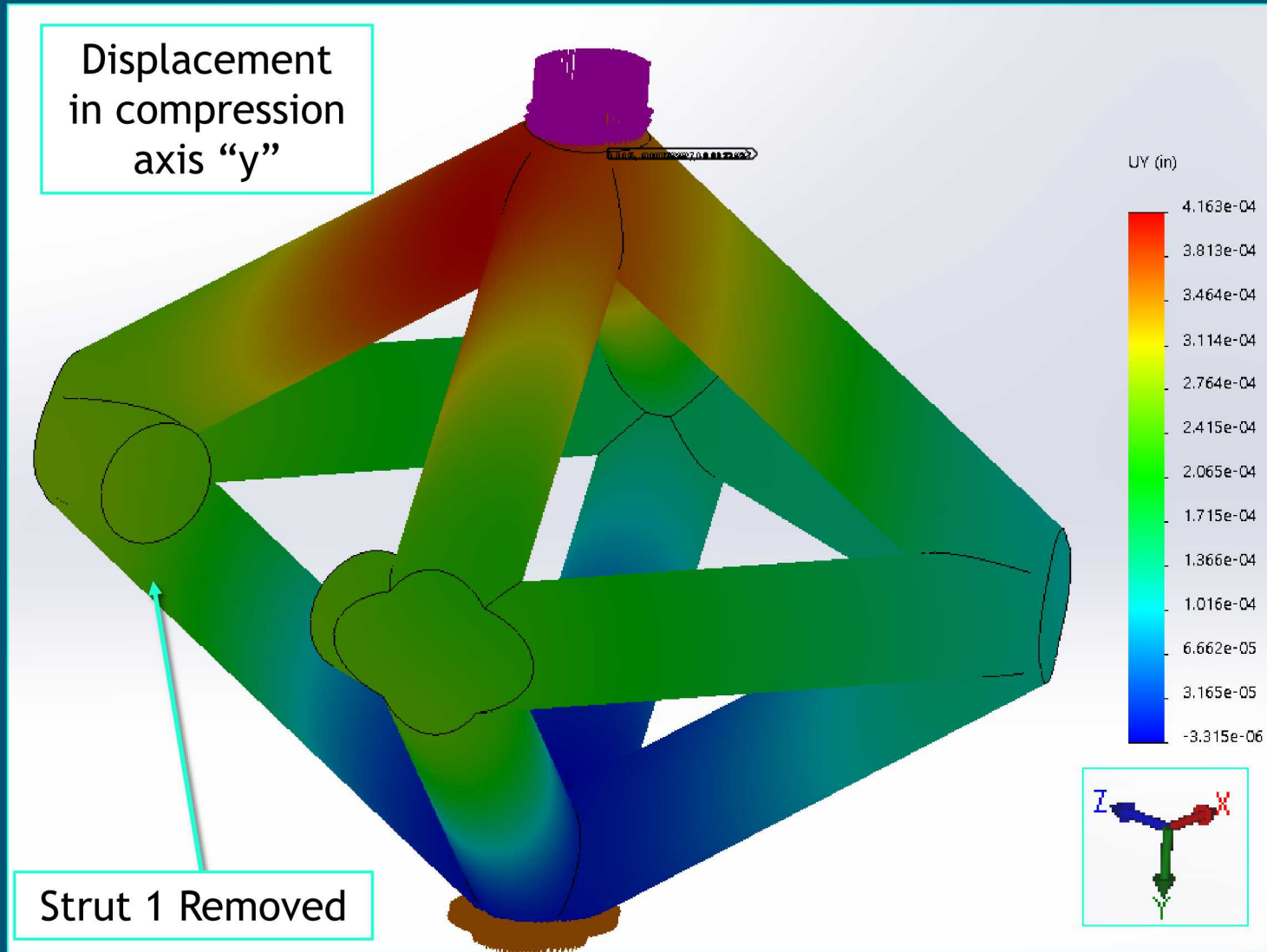
$$k = \frac{10 \text{ pounds}}{2.112 \times 10^{-6} \text{ inches}}$$

$$k = 4.734 \times 10^6 \text{ lb/in}$$

Single Unit Cell Support/Load Scheme



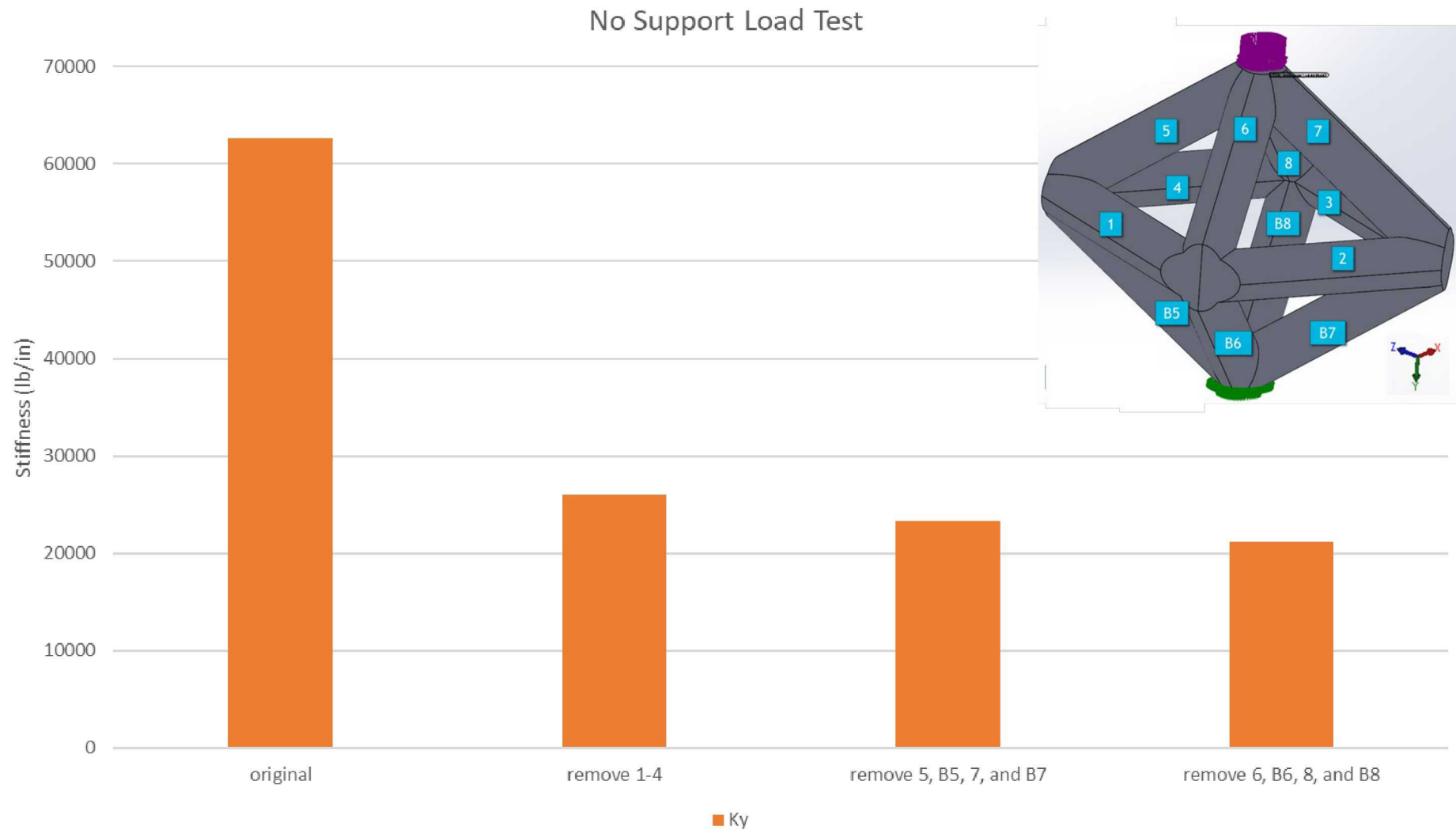
Preliminary Results: Single Unit Cell FEA



Stiffness was calculated by dividing the applied force by the displacement at the loaded face: $k = F/\delta$

Preliminary Results: Single Unit Cell, One Strut Removed

12

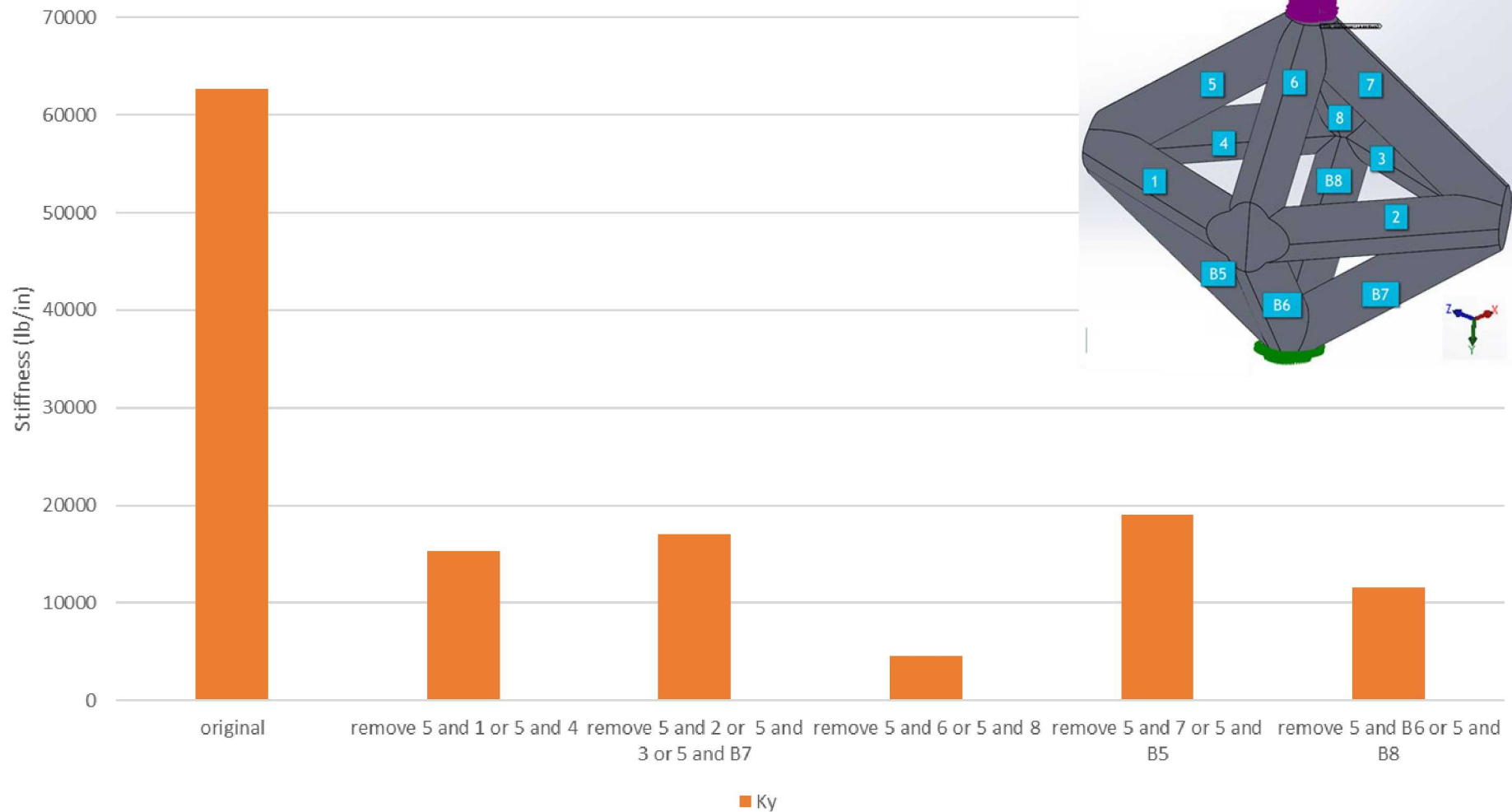


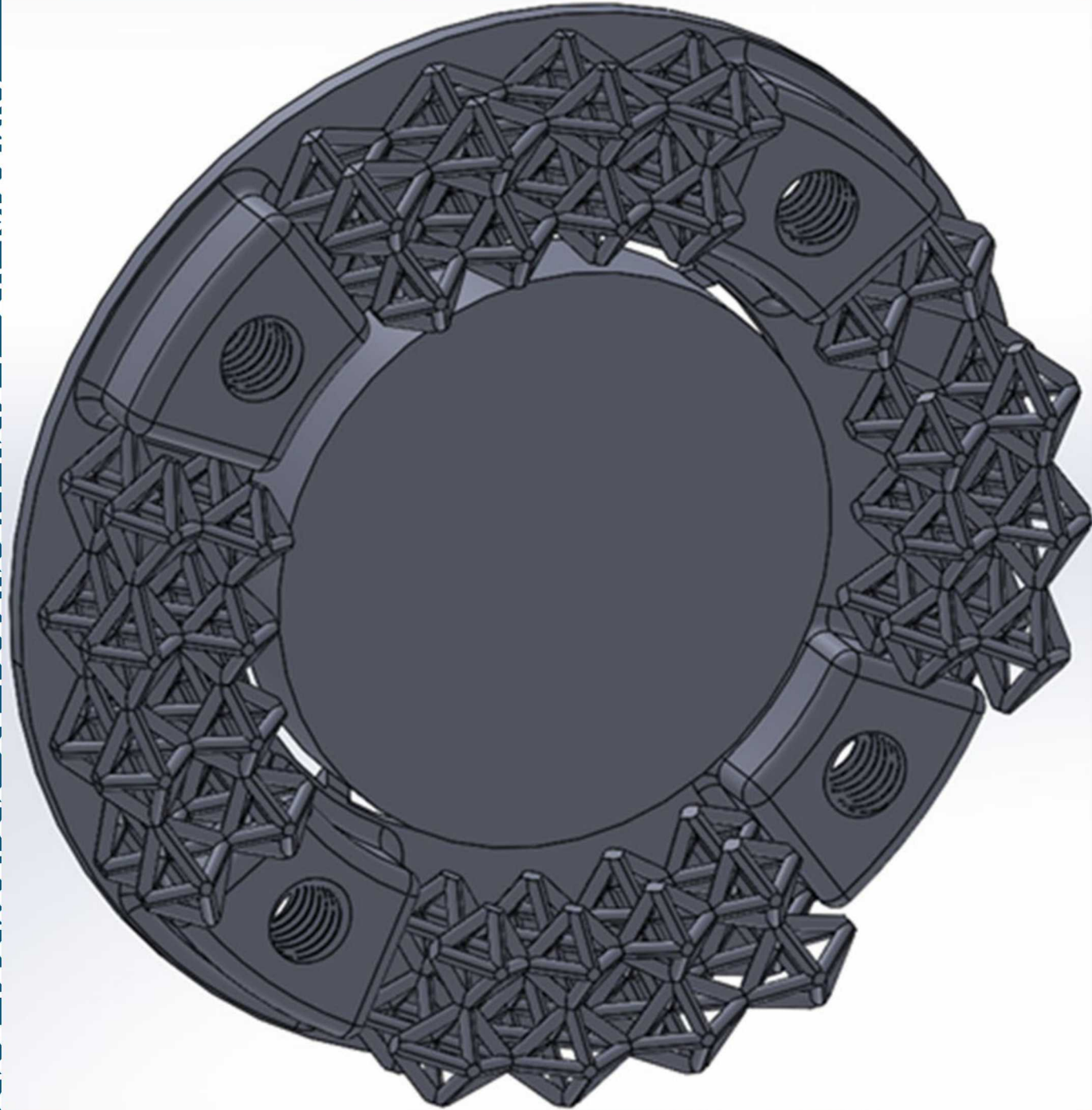
Preliminary Results: Single Unit Cell, Multiple Struts Removed

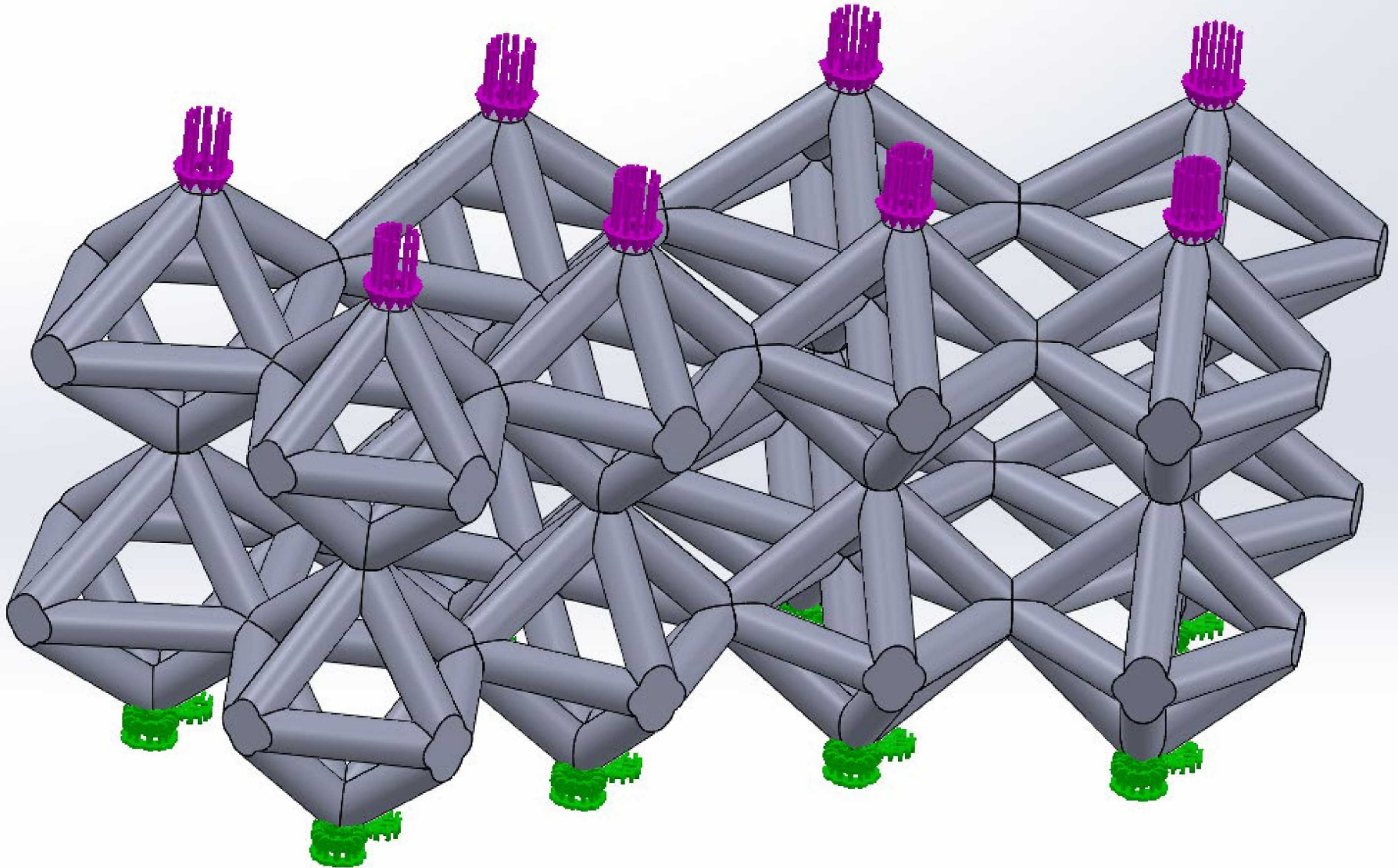
13



No Support Load Test, Multiple Strut Removal





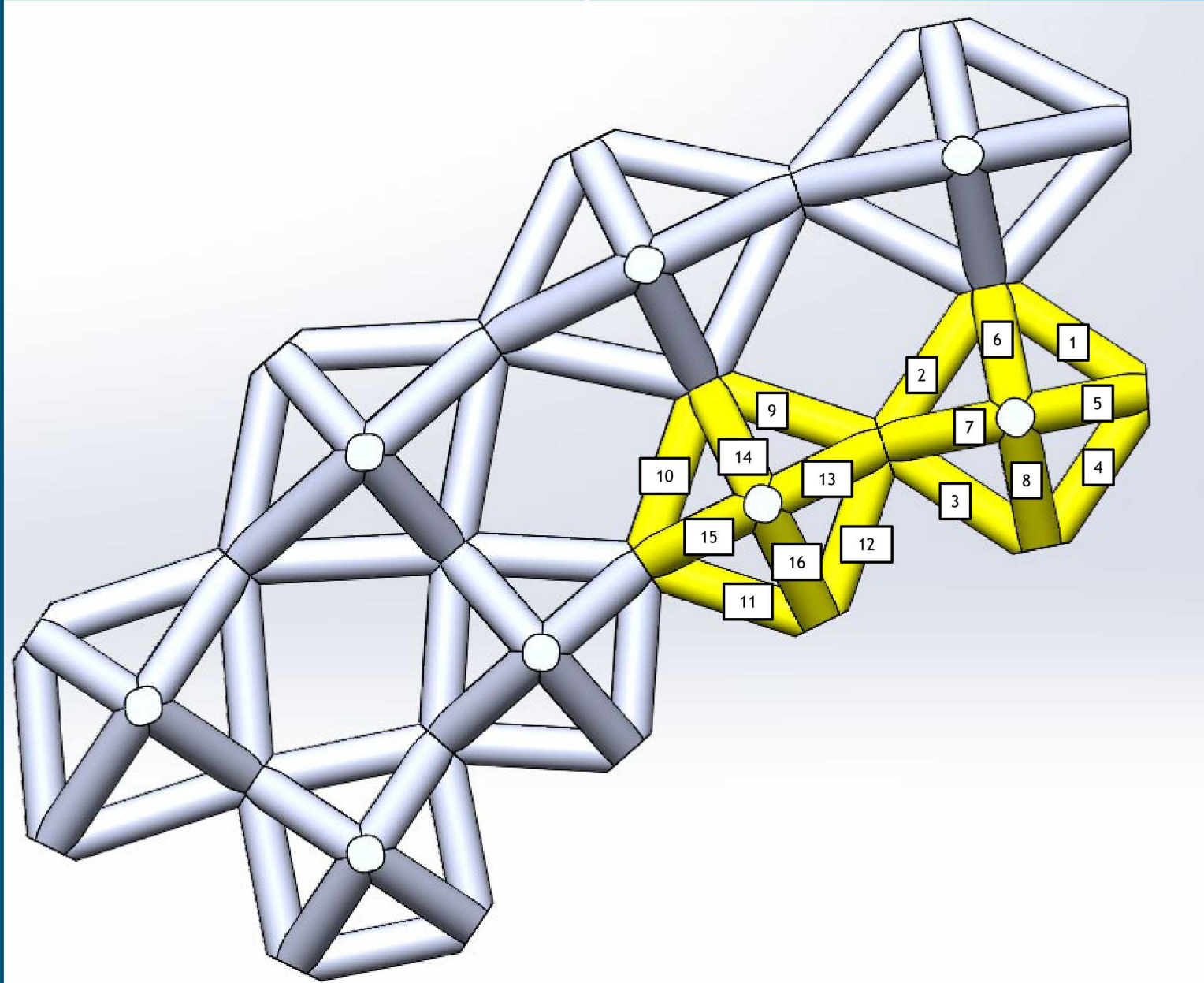


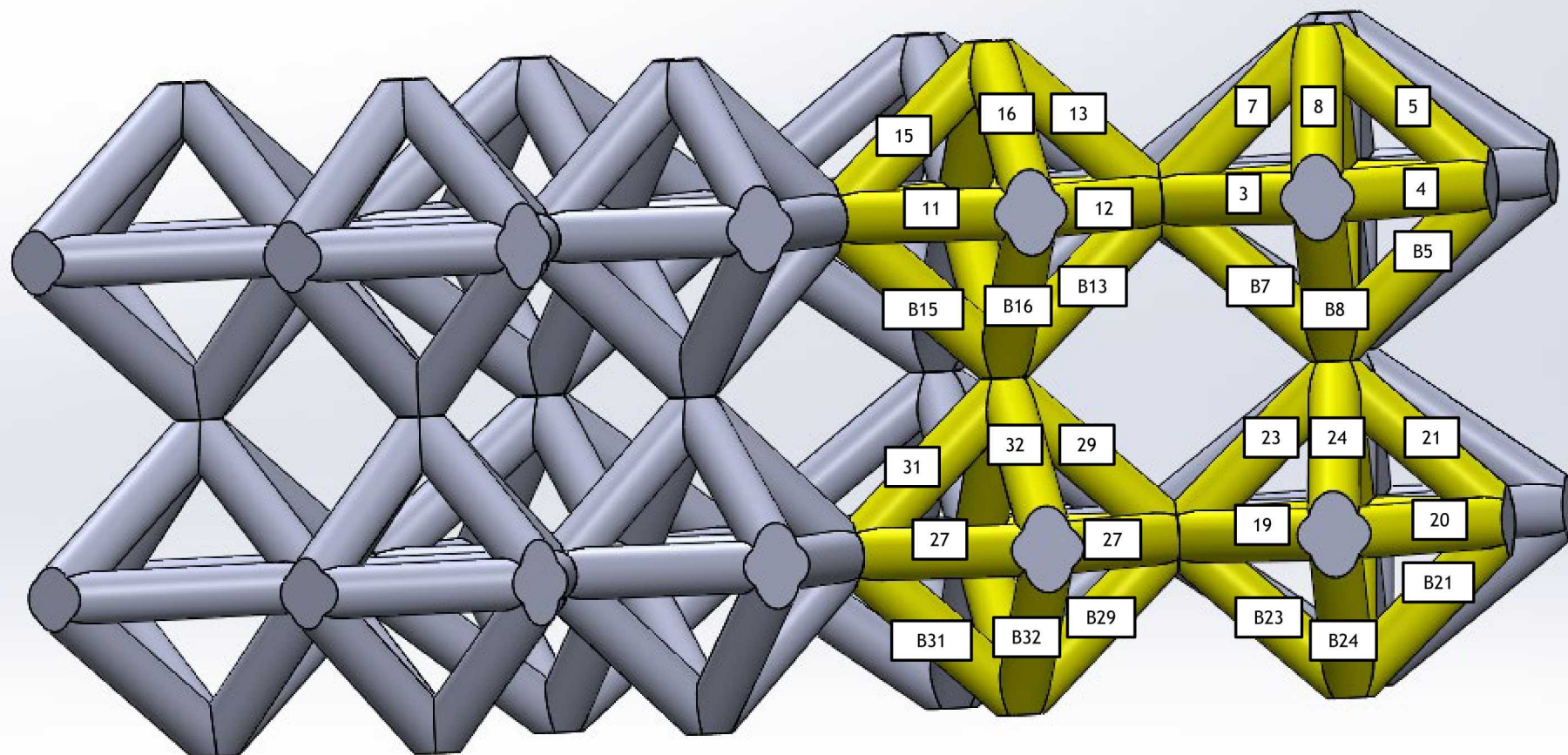
No Struts Removed



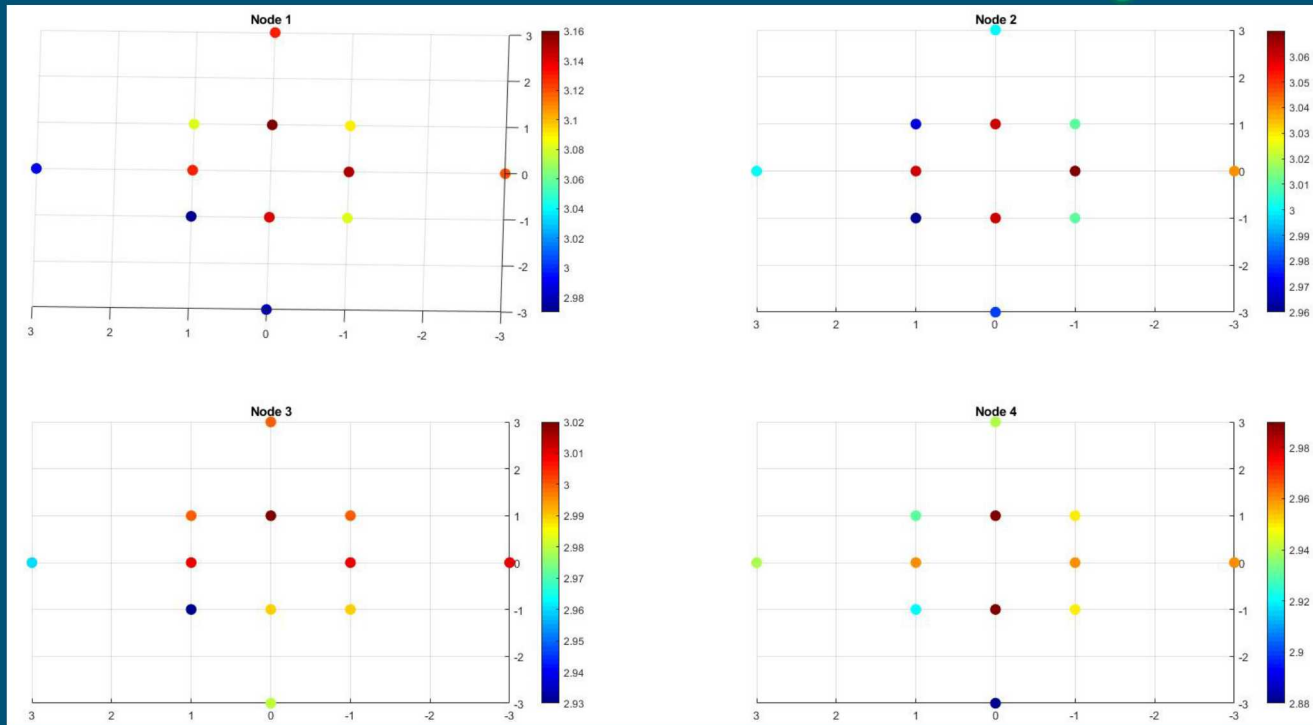
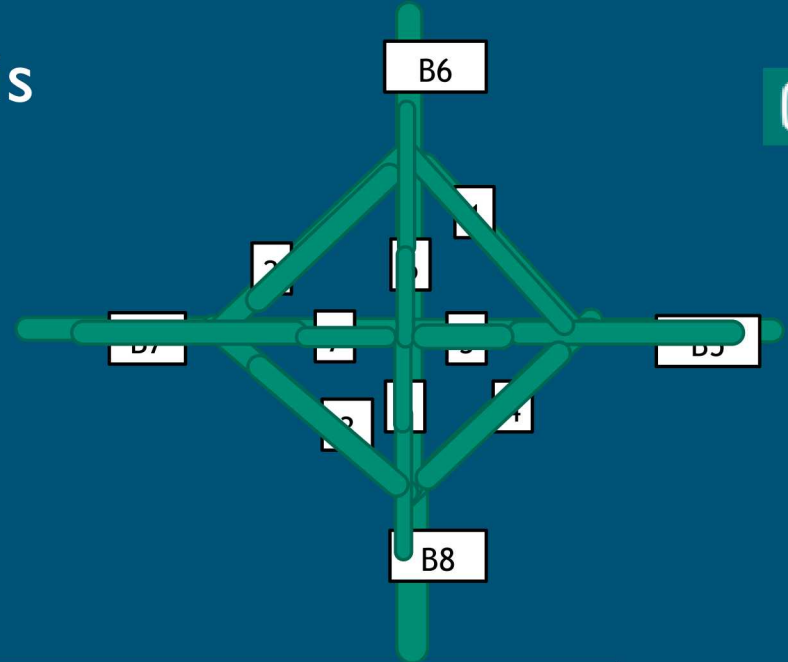
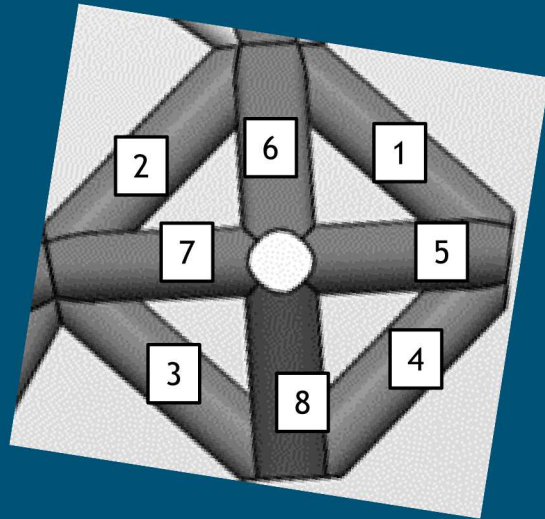


- Selectively remove struts from different locations
- Measure Displacement from the 8 top faces
- Compare for different removal locations





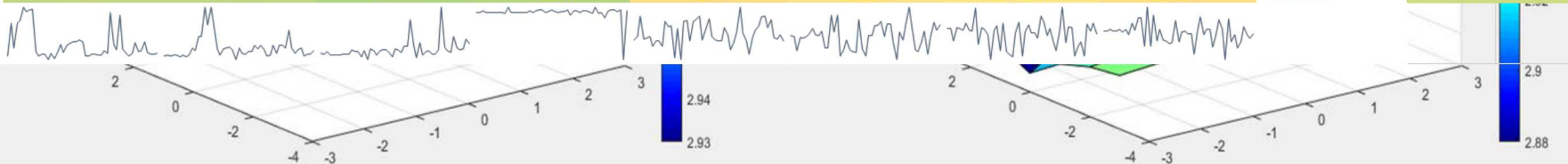
Data Collection and Analysis



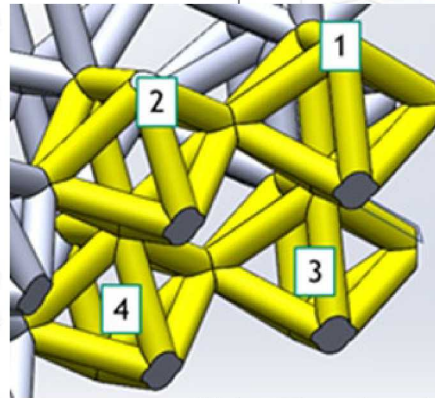
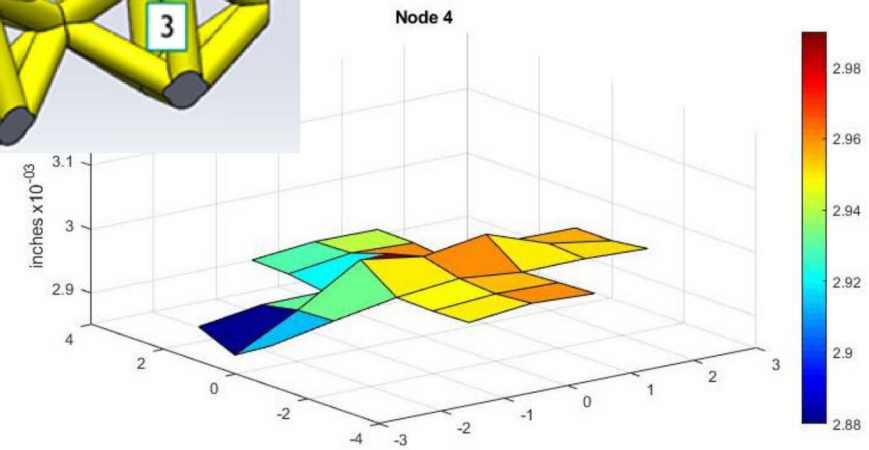
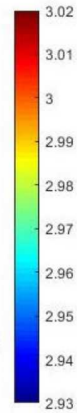
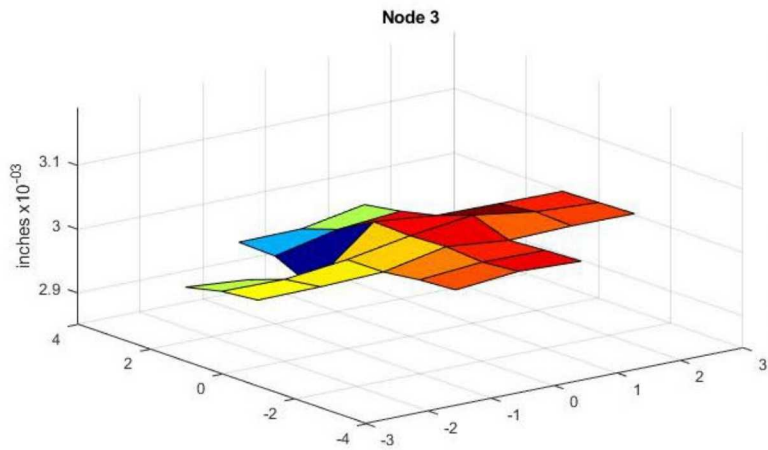
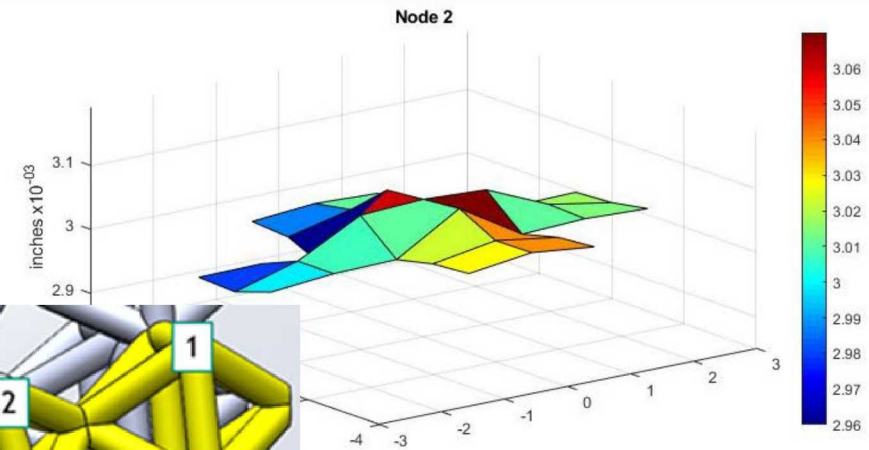
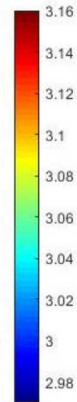
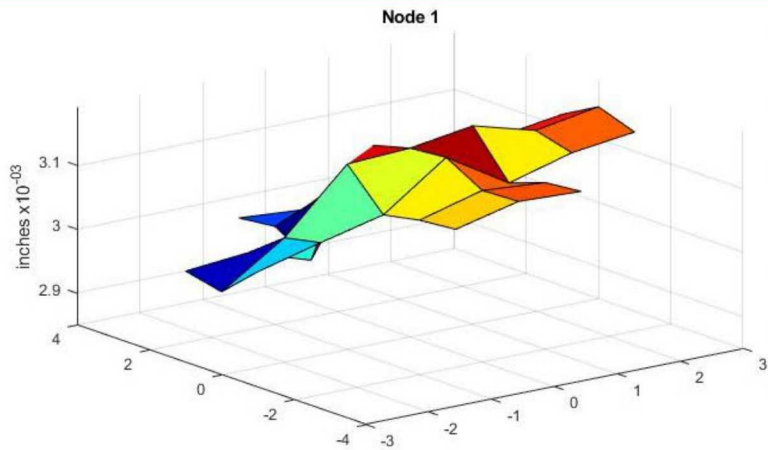
Initial Results



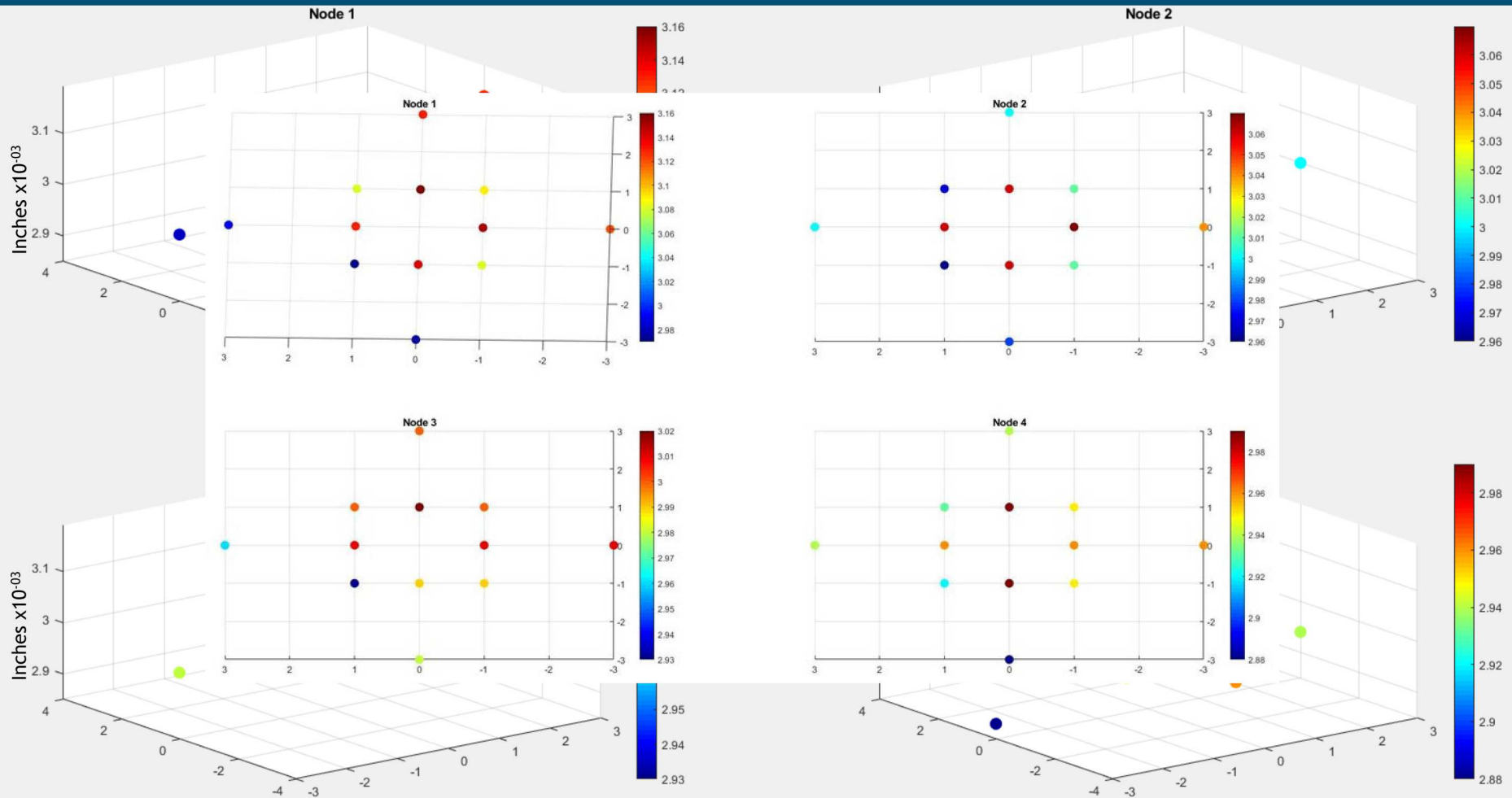
	1	2	3	4	5	6	7	8		SUM	AVERAGE
Original	2.63E-04	2.53E-04	2.53E-04	2.63E-04	3.31E-04	3.18E-04	3.18E-04	3.31E-04		2.33E-03	2.91E-04
Remove 1	3.96E-04	2.55E-04	2.53E-04	2.63E-04	3.33E-04	3.18E-04	3.18E-04	3.31E-04		2.47E-03	3.08E-04
Remove 2	2.94E-04	2.55E-04	2.55E-04	2.64E-04	3.47E-04	3.12E-04	3.14E-04	3.32E-04		2.37E-03	2.97E-04
Remove 3	3.95E-04	2.57E-04	2.52E-04	2.62E-04	3.32E-04	3.18E-04	3.19E-04	3.31E-04		2.47E-03	3.08E-04
Remove 4	4.06E-04	2.56E-04	2.52E-04	2.63E-04	3.32E-04	3.17E-04	3.18E-04	3.31E-04		2.48E-03	3.09E-04
Remove 5	4.60E-04	2.52E-04	2.52E-04	2.63E-04	3.33E-04	3.18E-04	3.18E-04	3.32E-04		2.53E-03	3.16E-04
Remove 6	4.40E-04	2.56E-04	2.52E-04	2.62E-04	3.27E-04	3.19E-04	3.20E-04	3.30E-04		2.51E-03	3.13E-04
Remove 7	4.46E-04	2.50E-04	2.53E-04	2.62E-04	3.30E-04	3.21E-04	3.18E-04	3.32E-04		2.51E-03	3.14E-04
Remove 8	4.53E-04	2.58E-04	2.53E-04	2.63E-04	3.30E-04	3.19E-04	3.17E-04	3.29E-04		2.52E-03	3.15E-04
Remove 9	2.81E-04	2.78E-04	2.53E-04	2.61E-04	3.27E-04	3.19E-04	3.25E-04	3.35E-04		2.38E-03	2.97E-04
Remove 10	2.62E-04	2.71E-04	2.57E-04	2.69E-04	3.43E-04	3.20E-04	3.13E-04	3.31E-04		2.37E-03	2.96E-04
Remove 11	2.71E-04	3.12E-04	2.55E-04	2.62E-04	3.40E-04	3.13E-04	3.19E-04	3.38E-04		2.41E-03	3.01E-04
Remove 12	2.71E-04	3.12E-04	2.55E-04	2.63E-04	3.41E-04	3.13E-04	3.18E-04	3.38E-04		2.41E-03	3.01E-04
Remove 13	2.49E-04	4.03E-04	2.50E-04	2.62E-04	3.19E-04	3.27E-04	3.16E-04	3.23E-04		2.45E-03	3.06E-04
Remove 14	2.72E-04	3.48E-04	2.56E-04	2.62E-04	3.39E-04	3.20E-04	3.08E-04	3.44E-04		2.45E-03	3.06E-04
Remove 15	2.51E-04	4.03E-04	2.49E-04	2.62E-04	3.18E-04	3.27E-04	3.18E-04	3.23E-04		2.45E-03	3.06E-04
Remove 16	2.74E-04	3.49E-04	2.57E-04	2.62E-04	3.44E-04	3.10E-04	3.19E-04	3.39E-04		2.45E-03	3.07E-04
Remove 17	3.08E-04	2.75E-04	2.56E-04	2.65E-04	3.45E-04	3.12E-04	3.11E-04	3.26E-04		2.40E-03	3.00E-04
Remove 18	2.77E-04	2.57E-04	2.52E-04	2.63E-04	3.30E-04	3.16E-04	3.20E-04	3.32E-04		2.35E-03	2.93E-04
Remove 19	3.07E-04	2.75E-04	2.56E-04	2.65E-04	3.44E-04	3.12E-04	3.11E-04	3.27E-04		2.40E-03	2.99E-04
Remove 20	3.13E-04	2.77E-04	2.56E-04	2.65E-04	3.45E-04	3.11E-04	3.10E-04	3.26E-04		2.40E-03	3.00E-04
Remove 21	3.19E-04	2.81E-04	2.56E-04	2.65E-04	3.48E-04	3.11E-04	3.09E-04	3.25E-04		2.41E-03	3.02E-04
Remove 22	3.19E-04	2.72E-04	2.54E-04	2.62E-04	3.31E-04	3.25E-04	3.17E-04	3.25E-04		2.40E-03	3.01E-04
Remove 23	3.27E-04	2.43E-04	2.51E-04	2.63E-04	3.38E-04	3.13E-04	3.27E-04	3.37E-04		2.39E-03	2.99E-04
Remove 24	3.20E-04	2.79E-04	2.57E-04	2.66E-04	3.45E-04	3.10E-04	3.09E-04	3.26E-04		2.41E-03	3.01E-04
Remove 25	2.63E-04	2.62E-04	2.56E-04	2.63E-04	3.33E-04	3.18E-04	3.17E-04	3.30E-04		2.34E-03	2.93E-04
Remove 26	2.65E-04	2.59E-04	2.54E-04	2.62E-04	3.31E-04	3.18E-04	3.19E-04	3.31E-04		2.34E-03	2.92E-04
Remove 27	2.70E-04	2.66E-04	2.59E-04	2.64E-04	3.30E-04	3.23E-04	3.17E-04	3.30E-04		2.36E-03	2.95E-04
Remove 28	2.70E-04	2.66E-04	2.59E-04	2.64E-04	3.30E-04	3.23E-04	3.17E-04	3.30E-04		2.36E-03	2.95E-04
Remove 29	2.66E-04	2.85E-04	2.80E-04	2.66E-04	3.39E-04	3.24E-04	3.15E-04	3.19E-04		2.39E-03	2.99E-04
Remove 30	2.66E-04	2.75E-04	2.60E-04	2.59E-04	3.35E-04	3.18E-04	3.22E-04	3.32E-04		2.37E-03	2.96E-04
Remove 31	2.85E-04	2.85E-04	2.50E-04	2.64E-04	3.24E-04	3.23E-04	3.21E-04	3.38E-04		2.39E-03	2.99E-04
Remove 32	2.73E-04	2.72E-04	2.62E-04	2.65E-04	3.30E-04	3.23E-04	3.16E-04	3.29E-04		2.37E-03	2.96E-04
Remove B5	4.36E-04	2.53E-04	2.52E-04	2.63E-04	3.36E-04	3.16E-04	3.18E-04	3.32E-04		2.51E-03	3.13E-04
Remove B6	3.09E-04	2.57E-04	2.55E-04	2.65E-04	3.54E-04	3.08E-04	3.14E-04	3.32E-04		2.39E-03	2.99E-04
Remove B7	2.88E-04	2.87E-04	2.56E-04	2.64E-04	3.36E-04	3.20E-04	3.11E-04	3.23E-04		2.38E-03	2.98E-04
Remove B8	4.26E-04	2.62E-04	2.53E-04	2.63E-04	3.31E-04	3.16E-04	3.18E-04	3.30E-04		2.50E-03	3.12E-04
Remove B13	2.93E-04	2.85E-04	2.53E-04	2.61E-04	3.21E-04	3.23E-04	3.25E-04	3.37E-04		2.40E-03	3.00E-04
Remove B14	2.75E-04	2.87E-04	2.56E-04	2.69E-04	3.33E-04	3.31E-04	3.14E-04	3.32E-04		2.40E-03	3.00E-04
Remove B15	2.63E-04	2.77E-04	2.90E-04	2.61E-04	3.38E-04	3.22E-04	3.19E-04	3.16E-04		2.39E-03	2.98E-04
Remove B16	2.74E-04	3.31E-04	2.58E-04	2.62E-04	3.41E-04	3.12E-04	3.19E-04	3.38E-04		2.43E-03	3.04E-04
Remove B21	3.16E-04	2.78E-04	2.56E-04	2.56E-04	3.47E-04	3.12E-04	3.09E-04	3.25E-04		2.40E-03	3.00E-04
Remove B22	2.82E-04	2.62E-04	2.54E-04	2.64E-04	3.47E-04	3.16E-04	3.13E-04	3.29E-04		2.37E-03	2.96E-04
Remove B23	2.85E-04	2.71E-04	2.67E-04	2.65E-04	3.39E-04	3.20E-04	3.14E-04	3.22E-04		2.38E-03	2.98E-04
Remove B24	3.18E-04	2.79E-04	2.58E-04	2.66E-04	3.45E-04	3.11E-04	3.09E-04	3.25E-04		2.41E-03	3.01E-04
Remove B29	2.76E-04	2.54E-04	2.56E-04	2.63E-04	3.32E-04	3.18E-04	3.20E-04	3.33E-04		2.35E-03	2.94E-04
Remove B30	2.69E-04	2.60E-04	2.57E-04	2.65E-04	3.32E-04	3.24E-04	3.17E-04	3.29E-04		2.35E-03	2.94E-04
Remove B31	2.69E-04	2.60E-04	2.65E-04	2.08E-04	3.35E-04	3.21E-04	3.20E-04	3.27E-04		2.30E-03	2.88E-04
Remove B32	2.73E-04	2.70E-04	2.61E-04	2.65E-04	3.30E-04	3.23E-04	3.17E-04	3.29E-04		2.37E-03	2.96E-04



MatLab surf function

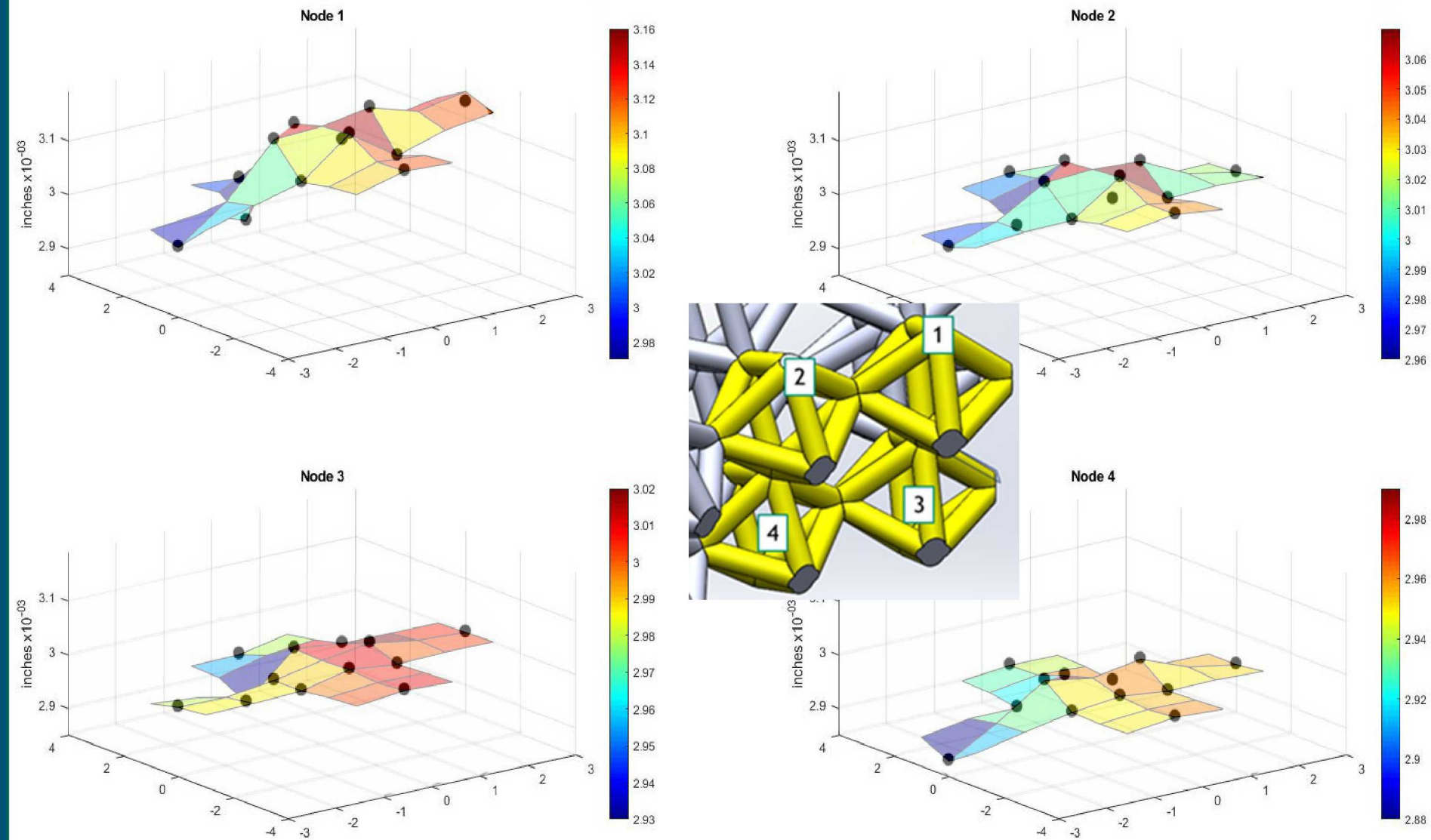


MatLab scatter3 function

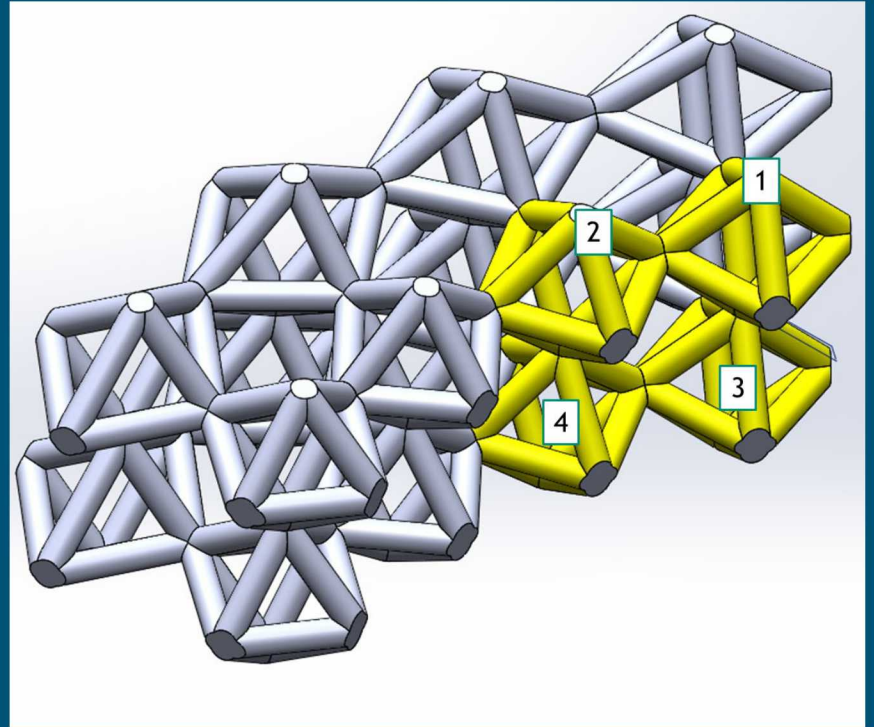
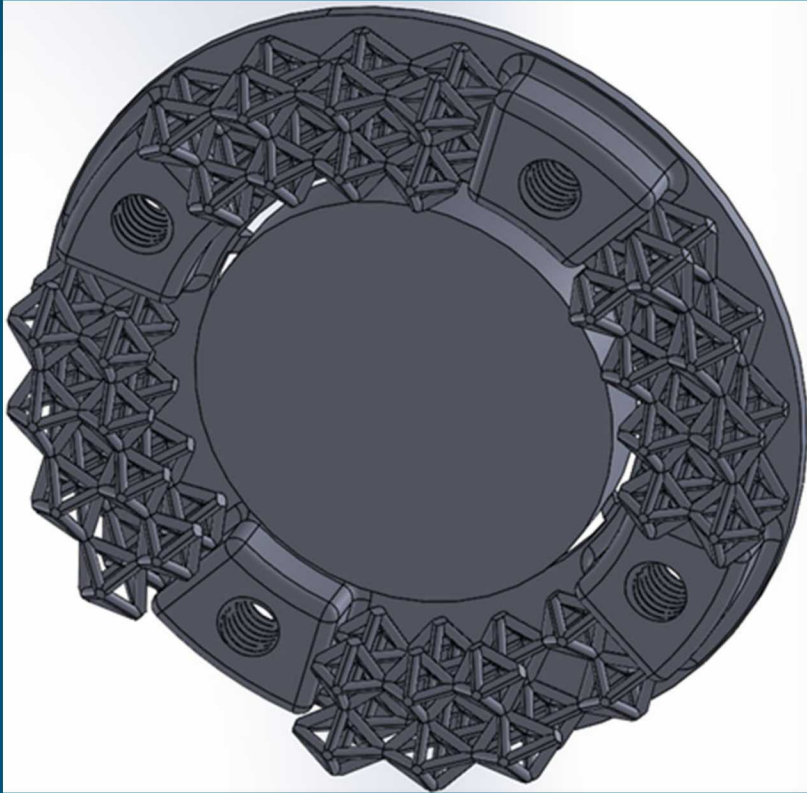


Surf and Scatter3 Transposed

24



Conclusions





Verify the computer simulation

- Destructive model testing
- Rerun with mirror unit cells
- Hand Calculations

Changing the strut size

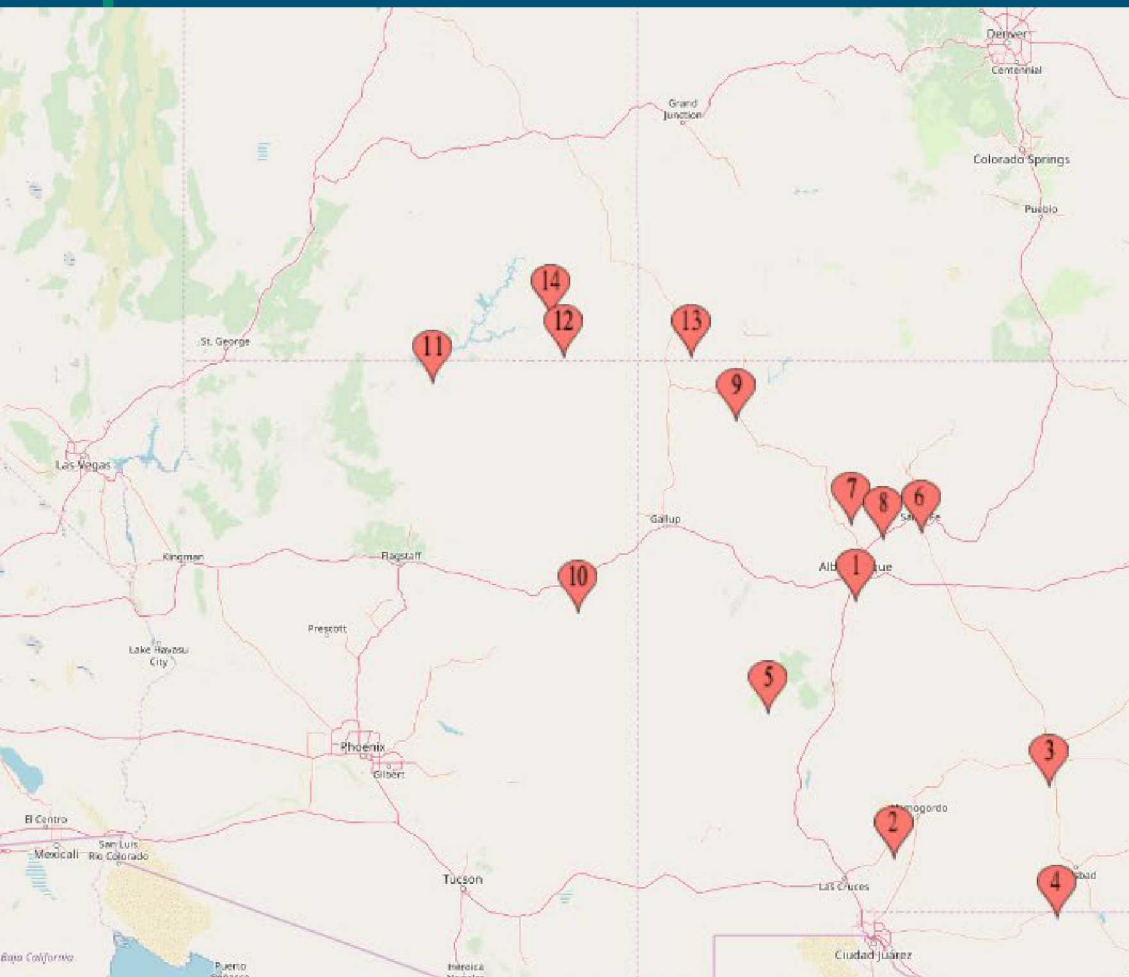
768 total struts

- How does one affect other sections

Is this the best lattice arrangement for the loading parameters

Engineering Does not Happen in a Vacuum

Technology $\neq$ People



Acknowledgements



- Deidre, Bradley, Shaun, Andrew, Michael, David, Ray, Matthew, Will
- MAC Coordinators - Staci and Jaime
- USCGA - LCDR Daeffler
- Val
- Dylan



**Sandia
National
Laboratories**



**UNITED STATES
COAST GUARD
ACADEMY**