

# 1 Tailoring Ductile-Phase Toughened Tungsten Hierarchical Microstructures 2 for Plasma-Facing Materials

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5

6 **Abstract:** Biological materials, such as bones, nacre, etc., have been found to exhibit attractive  
7 and unique combinations of stiffness, strength and fracture toughness. Research has been  
8 conducted in various engineering areas to mimic the naturally hierarchical microstructures or  
9 nanostructures of these materials to produce materials with optimum stiffness, high strength and  
10 fracture toughness for their intended structural applications. In the same objective, this work  
11 applies a multiscale microstructural approach recently developed (*J. Nucl. Mater.*, (2018) 508:  
12 371-384) to investigate the deformation and fracture behavior of ductile phase toughened tungsten  
13 (W) materials such as tungsten-nickel/iron (W-Ni/Fe) composites that possess lamellar-like and  
14 hierarchical “brick-and-mortar” (BAM) microstructures. First, the approach is used to simulate  
15 tensile loading of W-Ni/Fe specimens cut out from hot-rolled W-Ni/Fe plates which possess  
16 lamellar-like microstructures. The finite element model of the gage section of the specimen  
17 consists of a W-Ni/Fe dual-phase microstructural domain in which the constitutive behaviors of  
18 W and Ni/Fe phases are described by an elastic-plastic damage model. The predicted material  
19 stress-strain response and crack pattern development as a function of loading are then compared  
20 to the corresponding experimental results to determine the constitutive model parameters for W  
21 and Ni/Fe. Subsequently, the model parameters are used to analyze W-Ni/Fe BAM microstructures  
22 that are artificially created to investigate the effects of microstructural features and morphologies  
23 on the composite stress-strain response, damage, and fracture patterns. The modeling shows that  
24 regular BAM microstructures that experience bridging mechanism combined with crack  
25 penetration across W-phase regions exhibit significantly higher strengths than the rather random  
26 lamellar-like microstructures. More significantly, however adjusting the brick’s length-to-height  
27 ratio, the BAM microstructure can be designed to allow a more distributed damage zone which  
28 leads to increased strength, ductility and fracture energy.

29 *Keywords:* Plasma-facing materials, hierarchical microstructure, ductile-phase toughening,  
30 damage modeling, finite element.

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31 **1. Introduction**

32 Tungsten (W) alloys and W composites are currently the prime candidate materials for plasma-  
33 facing components (PFCs) of future fusion reactors such as the International Thermonuclear  
34 Experimental Reactor (ITER) and Demonstration Power Plant (DEMO). These W-based materials  
35 possess a high melting point, good strength at high temperatures, high thermal conductivity, low  
36 coefficient of thermal expansion, and low sputtering yields [1-3] that appear necessary for fusion  
37 energy applications. However, pure W and most W alloys can exhibit rather brittle behavior and  
38 suffers from high ductile-to-brittle transition temperatures. One way to mitigate the ductility issue  
39 is employing the concept of ductile-phase toughening (DPT), in which W brittle phase is embedded  
40 in a second ductile phase. Our previous results and report from other groups both confirmed  
41 significant improvement in DPT W composite materials [4-5].

42 In this paper we consider a variant of conventional DPT W composites, a hierarchical brick-  
43 and-mortar (BAM) composite, inspired by nature, where , biological materials such as nacre, teeth,  
44 shells, etc. have been found to exhibit attractive and unique combinations of stiffness, strength,  
45 and fracture toughness. Research has already been conducted in various engineering areas to  
46 mimic the kind of naturally occurring hierarchical microstructures or nanostructures of biological  
47 materials to produce materials with improved mechanical properties for their intended structural  
48 applications [6-8]. Gao [6] applied fracture mechanics concepts to hierarchical biomechanics of  
49 bone and bone-like materials to understand the nanoscale effect and mechanisms responsible for  
50 the optimized mechanical properties of these materials. This study indicates that understanding the  
51 mechanics or engineering principles in these naturally hierarchical materials may provide guidance  
52 to develop biologically inspired novel materials with unique properties. By exploring the features  
53 of the BAM structures found in enamel or nacre, Brandt et al. [7] proposed a multi-step processing  
54 procedure for hierarchical polymer composites by combining different coating techniques on  
55 different length scales. Henri and Pimenta [8] studied the damage tolerance that could be realized  
56 in a carbon-fiber polymer composite conceived as a hierarchical material by means of modeling  
57 and experiments. These authors found that the presence of discontinuities in the form of  
58 discontinuous fibers and hierarchical features promoted stable energy dissipation before failure.  
59 Combined, these microstructural features enabled disperse damage throughout the specimen  
60 leading to delays in damage localization that could otherwise occur in typically brittle composites.

61 Similar to these previous studies on other BAM materials, our work investigates the  
62 mechanical properties of ductile phase toughened (DPT) tungsten (W) composites that possess  
63 hierarchical BAM microstructures to elucidate the toughening mechanisms that are responsible for  
64 their mechanical properties in the hopes of further tailoring and improving them. To accomplish  
65 this, we examine the role of microstructural features and parameters that can contribute to the  
66 occurrence of a given or several toughening mechanisms and seek to understand the effect of the  
67 combined mechanisms on the composite stress-strain response. At the time of this paper, BAM  
68 materials have not yet been developed for high temperature applications such as PFCs for fusion  
69 energy applications. Particularly, no models exist that can guide the development of BAM  
70 materials for fusion energy science applications, although there have been general  
71 micromechanics-based models developed for these natural or synthetic materials that predict their  
72 mechanical properties including elastic stiffness, stress-strain response, fracture toughness etc.  
73 These models help understand the microstructural features, deformation and damage mechanisms  
74 that could lead to optimum combinations of mechanical properties [6-13]. Gorbatikh et al. [11]  
75 investigated in detail the failure mechanisms in hierarchical materials, including naturally  
76 occurring composites, and found that the structure of the reinforcement phase could change the  
77 failure initiation mechanism in a material with two levels of hierarchy by controlling the structure  
78 of the reinforcement phase on the lower level. More recently, Radi et al. [12-13] modeled BAM  
79 materials using direct element simulations and reported microcracking and crack deflection  
80 mechanisms that controlled strength, fracture toughness, and crack-growth toughness of these  
81 materials. In this work, we further elucidate these important mechanisms in our investigation of  
82 W-hierarchical composites relevant to fusion energy material application. Specifically, we apply  
83 a multiscale microstructural approach recently developed in [4] based on a W-copper (Cu)  
84 composite to investigate the tensile deformation and fracture behavior of DPT-W materials such  
85 as W-Ni/Fe composites that possess hierarchical lamellar-like and BAM microstructures.  
86 Previously, this approach was also successfully applied to simulate microcracking patterns and  
87 crack propagation in  $Ti_3SiC_2/SiC$  joints subjected to thermomechanical loading in addition to  
88 irradiation-induced swelling [14-15].

89 Our study does not focus on particular selection and qualification of W-Ni/Fe for PFMs.  
90 Research for testing W-Ni/Fe in fusion relevant conditions can be found in Refs [16-18]. The W-  
91 Ni/Fe composites have been adopted in this work primarily for the purposes of model

92 development, validation and illustration of the predictive capability needed to support the material  
93 design for PFCs. Predictive models that capture the toughening and damage mechanisms and their  
94 interactions are crucial to tailoring the mechanical properties to meet structural requirements for  
95 these applications. The same methodology for developing a predictive model of BAM W-NiFe  
96 composite as described in this work can then be applied to design other heavy W alloys of interest.

97 The use of Ni in fusion may raise a radiological concern. The plasma facing materials in  
98 divertor are expected to withstand the high heat flux and neutron irradiation doses in service for  
99 several years. The addition of Ni could lead to significant transmutation of Ni and production of  
100 helium, since Ni has a higher neutron interaction cross section than W and Fe in the relevant  
101 neutron spectrum [19]. Previous study to assess the activity of 97W-2.1Ni-0.9Fe composite  
102 (weight composition) under neutron irradiation condition at first wall of DEMO shows that the  
103 activity of the 97W with 5 years of service is very similar to that of pure W after 1 year cooldown,  
104 and it is less than two times higher than pure W after 10,000 years [16]. Another study advises that  
105 the limitation of Ni is 10 at.% to qualify as a class C waste [20-21]. All these studies suggest that  
106 W-Ni/Fe composites containing a few percentages of Ni are acceptable for fusion application.  
107 Nevertheless, further neutronics studies are clearly needed to assess the overall volume limit of Ni  
108 based on the designs of the fusion devices. Such studies are beyond the scope of this work.

109 The structure of this paper is as follows. Section 1 provides an introduction. Section 2 describes  
110 the microstructural approach [4] involving finite element (FE) discretization of the microstructure  
111 and constitutive modeling. This section then describes the application of the modeling approach  
112 to simulate tensile loading of W-Ni/Fe specimens cut out from hot-rolled W-Ni/Fe plaques. The  
113 predicted material stress-strain response and crack pattern development as a function of loading  
114 are then compared to the corresponding experimental results to determine the constitutive model  
115 parameters for W and Ni/Fe. Section 3 describes the design of BAM microstructure to achieve  
116 improved mechanical properties using the constitutive model parameters obtained in Section 2.  
117 Then, Section 3 discusses the effects of microstructural parameters on the damage and crack  
118 development that affect mechanical properties of BAM composites. Section 4 draws conclusions  
119 from this work.

## 120 **2. Multiscale microstructural modeling**

### 121 *2.1 Finite element discretization of the microstructure*

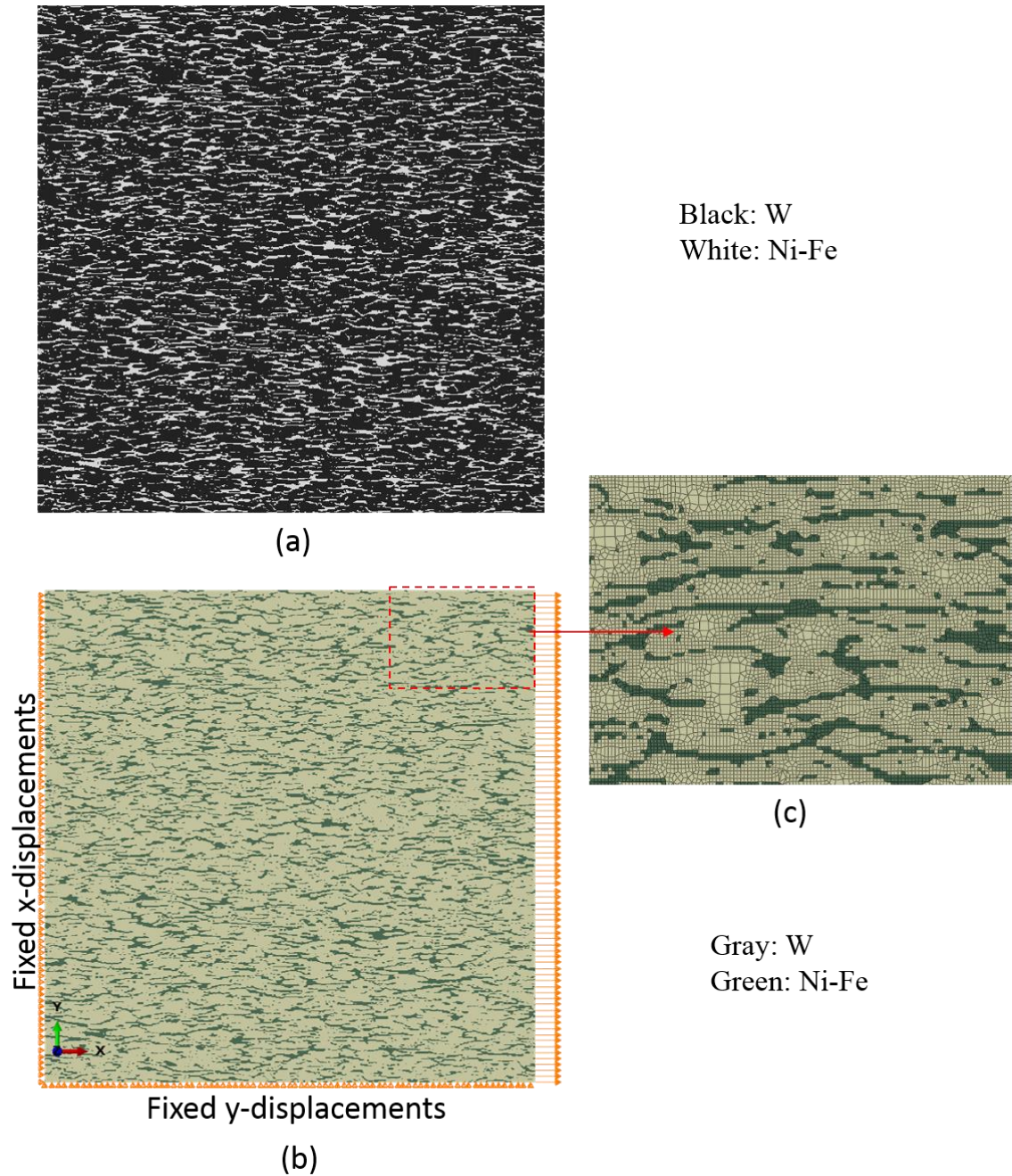
122 We investigate the deformation behavior of ductile phase toughened W materials such as W-Ni/Fe  
123 composites using the Nguyen et al.'s multiscale microstructural approach [4] that involves a dual-  
124 phase model where the two constituent phases (i.e., W and Ni/Fe) are finely discretized in finite  
125 elements (FEs) and described by a continuum damage mechanics (CDM) model. This approach is  
126 suitable for modeling deformation, cracking, and crack bridging for W-Ni/Fe, and other ductile  
127 phase toughened W-composites, or more generally, any multi-phase composite structure where  
128 two or more phases undergo cooperative deformation in a composite system.

129 The hot-rolled W-Ni/Fe material (weight composition of 90W-7Ni-3Fe) studied and defined  
130 as the baseline material in this work has a lamellar-like microstructure as presented in Fig. 1a. The  
131 image shown in Fig. 1a corresponds to a central part of a W-Ni/Fe tensile specimen located on the  
132 specimen's symmetry axes away from the gripped ends. This part is 0.584-mm long, 0.584-mm  
133 wide and 0.25-mm thick. It belongs to the specimen portion of interest that does not include the  
134 entailed sections and is 5-mm long, and 1.168-mm wide. The nominal W volume fraction in the  
135 composite is about 0.8. To construct a 3D model from a 2D image, the 2D image is simply  
136 extended in the thickness direction (out-of-plane) to form a uniform microstructure along this  
137 direction. Since the specimen was subjected to a uniform tensile loading, only a representative  
138 domain located on the symmetry axes which covered half of the specimen width was modeled.

139 The discretization of the 2D finite elements was achieved based on a digital image using the  
140 OOF2<sup>1</sup> software. The 2D FE model with boundary conditions illustrated in Fig. 1b contains the  
141 dual-phase W-Ni/Fe microstructural domain created using the Nguyen et al. method [4] that  
142 captures the constitutive behaviors of W and of the Ni/Fe alloy described by an elastic-plastic  
143 damage model. Although OOF2 can capture the actual microstructure obtained with SEM at some  
144 level of details, information such as chemical, compositional variations, impurities, and  
145 dislocation, etc. are not included. The model assumes composition and other factors are uniform  
146 within each phase boundary. Since the modeling domain is located on the specimen's symmetry  
147 axes, the corresponding displacements along the  $x$ - and  $y$ -directions on the domain left and bottom  
148 boundaries are fixed, respectively, while a uniform displacement parallel to the  $x$ -direction is  
149 applied on the domain right boundary (Fig. 1b). A magnified view of a local region inside the  
150 domain is given in Fig. 1c showing the detailed mesh.

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<sup>1</sup>Software developed at the National Institute of Standards and Technology.



151

152 **Figure 1.** (a) The digital image of the as-formed dual-phase microstructure from a tensile  
 153 specimen, (b) The associated FE model with boundary conditions for analysis, and (c) a magnified  
 154 view showing the mesh details.

155 *2.2 Constitutive relations*

156 The constitutive responses of W and Ni/Fe in the dual-phase region are described by a CDM  
 157 formulation previously used [4]. The model uses the  $J_2$  deformation theory of plasticity and the

158 principle of strain equivalence (Lemaitre [22]) which states that the elastic or plastic constitutive  
 159 equations of a damaged material derived from the same formalism as the virgin material by  
 160 expressing these equations in terms of the effective stress tensor as:

$$\tilde{\sigma}_{ij} = \frac{\sigma_{ij}}{1 - D} \quad (1)$$

161 where  $\sigma_{ij}$  is the stress tensor and  $D$  is the isotropic damage variable.

162 The modified Ramberg-Osgood relation (Nguyen and Kunc [23]) that describes the material  
 163 nonlinear stress-strain response affected by damage is expressed in terms of the equivalent  
 164 effective stress as:

$$\bar{\varepsilon} = \frac{\bar{\sigma}}{E(1-D)} + \frac{\sigma_0}{E} \left( \frac{\bar{\sigma}}{\sigma_0(1-D)} \right)^n = \frac{\bar{\tilde{\sigma}}}{E} + h \bar{\tilde{\sigma}}^n \quad (2)$$

165 where  $\bar{\varepsilon}$ ,  $\bar{\sigma}$ , and  $\sigma_0$  are the total equivalent strain, equivalent stress and reference stress,  
 166 respectively. The first and second terms of the right-hand side of Eq. (2) represent the equivalent  
 167 elastic and plastic strains, respectively. The coefficient  $h$  is given by  $h = \sigma_0^{1-n}/E$  where  $E$  and  $n$  are  
 168 the elastic modulus of the undamaged material and power-law exponent. The “tilde” symbol  
 169 denotes an effective quantity based on the principle of strain equivalence as in Eq. (1). For an  
 170 elastic-plastic material without consideration of damage, the modified Ramberg-Osgood relation  
 171 (Eq. 2) reduces to the well-known Ramberg-Osgood relation (Ramberg and Osgood, 1943) [24].  
 172 In Eq. (2), the equivalent stress is expressed as a function of the equivalent plastic strain,  $\bar{\varepsilon}^p$ :

$$\bar{\varepsilon}^p = h \left( \frac{\bar{\sigma}}{1 - D} \right)^n \quad (3)$$

$$\bar{\sigma} = (1 - D)k(\bar{\varepsilon}^p)^{1/n} \quad (4)$$

173 where  $k = (1/h)^{1/n}$  is the hardening coefficient. Next, in the framework of the deformation theory  
 174 of plasticity and considering proportional loading, the Lemaitre-Chaboche damage evolution law  
 175 for isotropic hardening materials (Lemaitre and Chaboche, [25]) is used to express the damage  
 176 variable increment in terms of the equivalent plastic strain increment:

$$dD = \frac{D_c}{\bar{\varepsilon}_p^R - \bar{\varepsilon}_p^D} \left( \left[ \frac{2}{3}(1+\nu) + 3(1-2\nu) \left( \frac{\sigma^h}{\bar{\sigma}} \right)^2 \right] d\bar{\varepsilon}^p \right) \quad (5)$$

177 where  $\bar{\varepsilon}_p^D$  and  $\bar{\varepsilon}_p^R$  are the equivalent plastic strains at damage initiation and at rupture, respectively,  
 178 and  $D_c$  is the value of  $D$  at rupture,  $\nu$  is Poisson's ratio, and  $\sigma^h$  the hydrostatic stress. The  
 179 effective stress increment is related to the total strain increment by the constitutive equation:

$$d\tilde{\sigma}_{ij} = C_{ijkl}(D)d\varepsilon_{kl} \quad (6)$$

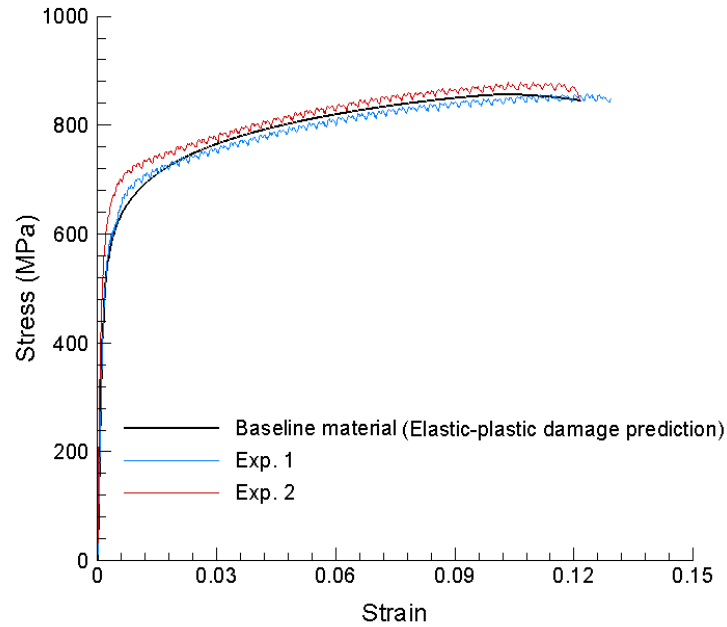
180 where the material tangent stiffness tensor  $C_{ijkl}(D)$  reflects both plasticity and damage.

181 Damage evolves with deformation according to the damage evolution law Eq. (5) until a  
 182 critical (saturation) state occurs for which  $D = D_c$  ( $0 < D_c \leq 1$ ) and failure results. In the current  
 183 model, damage is not activated under a compressive state of stress, rather the material is modeled  
 184 as elastic-plastic and obeys a classical Ramberg-Osgood relation [24] under compression. The  
 185 occurrence of failure implies that the failed material at a given location can no longer carry loads.  
 186 In this work, failure at damage saturation leading to crack initiation and propagation is modeled  
 187 by a vanishing finite element method [26, 4]. This damage model has been implemented in the  
 188 Abaqus FE package. The detailed of the computational procedure is given in Nguyen et al. [4].

### 189 190 2.3 Analysis results for the baseline W-Ni/Fe microstructure

191 The 2D plane-stress model (Fig. 1b) of the W-Ni/Fe specimen subjected to tensile loading was  
 192 analyzed using Abaqus and the elastic-plastic damage model described in the previous section to  
 193 depict the constitutive behaviors of W and Ni/Fe phases in the microstructural domain. A series of  
 194 FE analyses was first conducted to determine the constitutive parameters for the damage model.  
 195 Typical constitutive parameters of W and of Ni alloys were used as the initial guesses. However,  
 196 since the actual properties of the two phases are not known, the constitutive parameters are adjusted  
 197 through FE analyses until the predicted stress-strain response and loading-dependent damage  
 198 pattern development reasonably agree with the corresponding experimental results. Fig. 2a shows  
 199 the predicted engineering stress-strain response that agrees well with the corresponding  
 200 experimental data (also plotted on this figure). The damage model parameters for W and Ni/Fe  
 201 determined through these analyses are given in Table 1 and it should be noted that the fitted

202 parameters are reasonable for both W and Ni/Fe phases. In the next section, these parameters are  
 203 used as constitutive material inputs for W and Ni/Fe in BAM microstructural models for studying  
 204 the hierarchical microstructure effect on the stress-strain response of the as-designed BAM  
 205 composite.



206

207 **Figure 2.** Predicted true tensile stress-strain response for the baseline W-Ni/Fe composite  
 208 compared to the corresponding experimental data obtained using digital image correlation (DIC)  
 209 data from the tensile gage section of miniature tensile specimens described in Section 2.1.

210

211 **Table 1.** Parameters of the elastic-plastic damage model for W and Ni/Fe in the dual-phase  
 212 microstructural domain.

Material	$E$ (MPa)	$\nu$	$\sigma_0$ (MPa)	$n$	$\bar{\epsilon}_p^D$	$\bar{\epsilon}_p^R$	$D_c$
Ni/Fe	203570	0.304	250	7	0.2	0.65	0.98
W	383000	0.28	760	12.5	0.25	0.5	0.98

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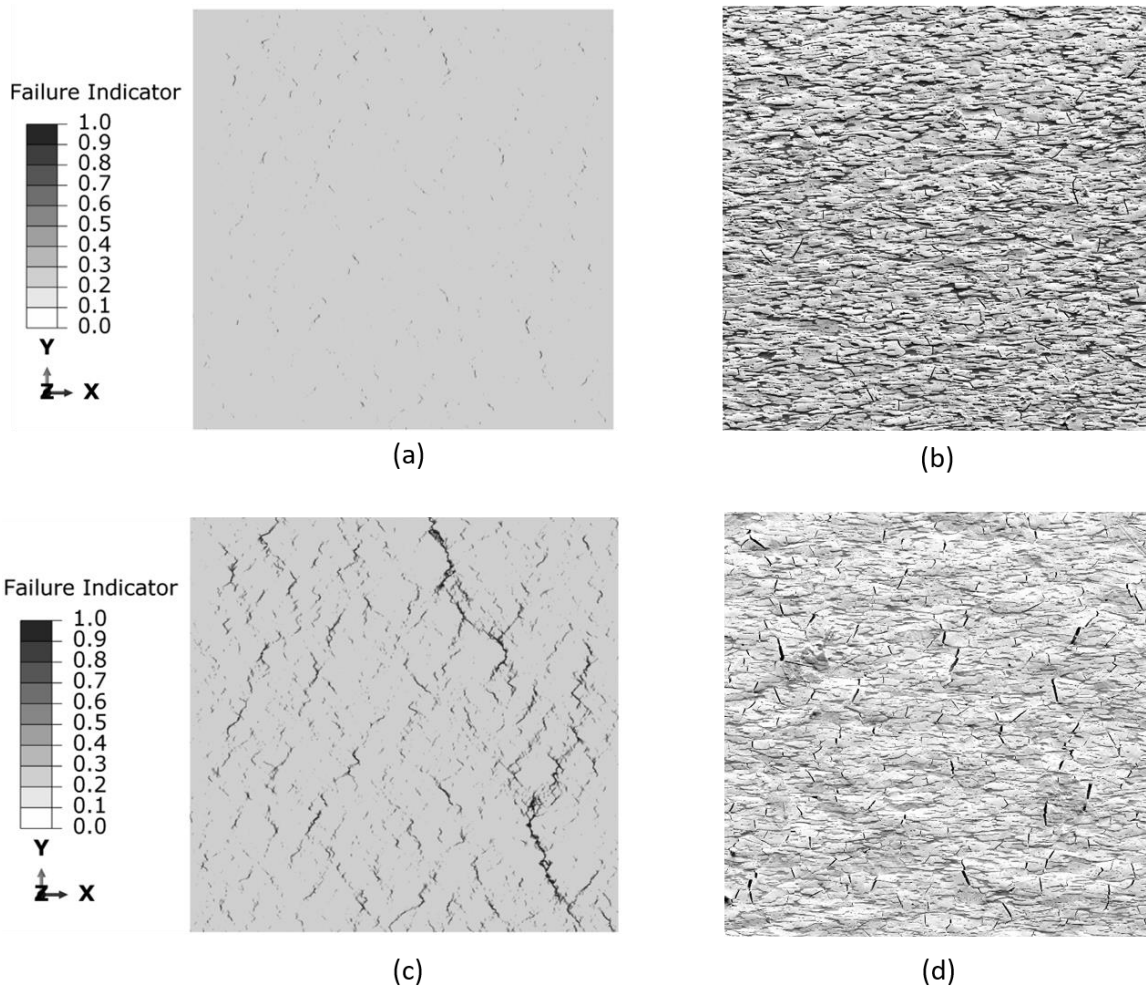
214 Figs 3a and 3c illustrate the predicted damage distributions and crack patterns at 0.04 and 0.12  
215 overall strains, respectively. At 0.04 strain, significant damage has been predicted to accumulate  
216 in the composite. At this loading level, damage has preferentially developed inside the Ni/Fe  
217 regions. However, at 0.12 strain that corresponds to about the elongation limit, many macrocracks  
218 were formed by the linking-up of failed Ni/Fe and W elements. The predicted damage and crack  
219 patterns reasonably agree with the experimental observations illustrated in Figs. 3b and 3d that  
220 show the SEM images of a tested sample region at 0.04 and 0.125 overall strains, respectively.  
221 Figs 4(a-d) illustrate in detail the occurrence of damage mechanisms as a function of loading. Fig.  
222 4a shows a magnified view of a sub-domain inside the modeling domain where the areas colored  
223 in gray and green depict the W and Ni-Fe phases, respectively. Fig. 4a is used to identify the  
224 regions for damage and fracture occurrences. At 0.04 applied strain (Fig. 4b), damage has already  
225 initiated in some Ni/Fe regions since a number of Ni/Fe elements have experienced nonzero to  
226 relatively high values of the failure indicator in these regions. At 0.085 strain these Ni/Fe elements  
227 start failing (failure indicator = 1) and microcracks are formed by the linking-up of failed Ni/Fe  
228 elements as clearly shown in Fig. 4c. Figs 4b and 4c illustrate the DPT mechanism caused by  
229 damage and failure of the ductile Ni/Fe elements between W elements. When the applied strain is  
230 increased to 0.12, the microcracks are also able to penetrate the W regions causing the W elements  
231 to fail. The linking-up of microcracks through both W and Ni/Fe regions forms a visible  
232 macrocrack shown in Fig. 4d at this loading level that triggers a combination of two failure  
233 mechanisms: the DPT and the fracture of W mechanisms. Fig. 4e shows an SEM of a local crack  
234 that propagated through W regions adjacent to Ni/Fe regions and appears similar to the predicted  
235 crack paths in Fig. 4d.

236 The agreement between predicted and experimental crack development results can only be  
237 qualitative due to the following reasons. First, due to the statistical nature of the microstructure,  
238 the microstructure used to develop the FE model of the dual-phase domain for the tensile specimen  
239 was not identical to the one from the tested specimen. We can see clearly on Fig. 4a compared to  
240 Fig. 4e that the FE model microstructure contains a weak path (involving many Ni/Fe elements)  
241 that favored crack propagation along this path while the tested specimen microstructure does not  
242 have such a path or the same path. Second, the experimental crack patterns at two different  
243 deformation levels were captured at a specimen's surface. Three-dimensional crack patterns might  
244 reveal more complex microcrack distributions and higher crack density closer to the predicted

245 crack patterns; and third, the 2D FE model could overpredict damage and crack development.  
246 Third, the FE model does not contain the W-W grain boundaries that are present in the  
247 microstructure. Finally, the current model does not consider effects such as compositional  
248 variations and impurities that influence on the mechanical properties. The influence of composition  
249 variation and impurity can only be incorporated in the FE model through the composition and  
250 impurity dependence of the constitutive behaviors. However, such input information is currently  
251 unavailable and more detailed and accurate experimental measurements are needed in the future  
252 to improve the model.

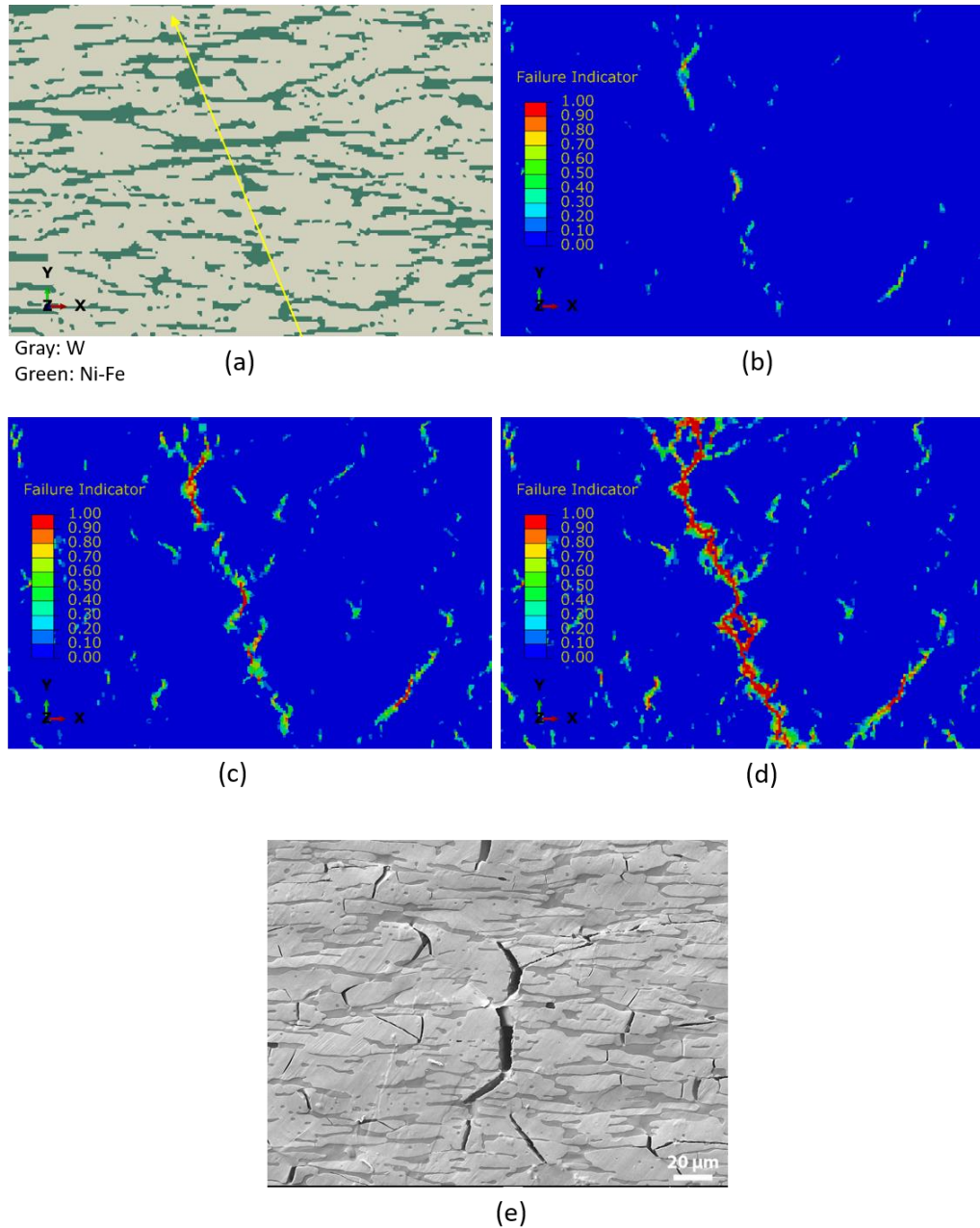
253 However, in addition to the DPT mechanism, we have identified two other important  
254 toughening mechanisms, namely distributed damage zone over the dual-phase domain and crack  
255 penetration across the W phases at high loading levels. The multiplication and distribution of  
256 microcracks that develop over a large area of the dual-phase domain (as shown in Fig. 3)  
257 correspond to increased strength and ductility when the applied deformation is increased from 0.04  
258 to  $\sim 0.12$  (as shown Fig. 2).

259



260

261 **Figure 3.** Predicted damage distribution and fracture patterns at (a) 0.04 and (b) 0.12 overall  
 262 strains. Optical images of surface cracks on experimental tensile specimens strained to (c) 0.04  
 263 and (d) 0.125 overall strains.



264

265 **Figure 4.** (a) A magnified view of a local area inside the microstructural domain (yellow arrow  
 266 indicating a preexisting weak path favoring crack propagation), (b) Predicted local damage  
 267 distribution at 0.04 strain, (c) Predicted local damage distribution and crack propagation at 0.085  
 268 strain, (c) Predicted cracks formed by the linking-up of failed Ni-Fe and W elements at 0.12 strain,  
 269 and (e) SEM image of cracks near the fracture path in the tensile sample strained to failure.

270

271

### 272 3. Tailoring W-Ni/Fe microstructure for improved mechanical properties

#### 273 3.1 “Brick-and-mortar” microstructures

274 Hierarchical “brick-and-mortar” (BAM) microstructures have been shown to be able to  
275 combine high stiffness, strength and fracture toughness [6-13]. In this section, a series of regular  
276 W-Ni/Fe BAM microstructures are generated to evaluate the potential of such microstructures to  
277 achieve improve strength, ductility and fracture energy in comparison to the baseline lamellar-like  
278 microstructure analyzed in the previous section. Various BAM microstructures can be generated  
279 by varying key microstructural features such as brick’s volume faction, length, height, width,  
280 length-to-height aspect ratio  $L/H$ , brick’s spacing and relative positions, etc. In this work, for the  
281 sake of the modeling approach illustration and meaningful comparisons with the baseline  
282 microstructure in terms of W content, microcrack distributions and their effects on the BAM  
283 composite’s mechanical properties, all the generated BAM microstructural domains contain the  
284 same volume fraction of W (0.8) and have the same dimensions (0.584 x 0.584 x 0.025 mm) as  
285 the baseline microstructure. In addition, they have the same topology in terms of bricks’ relative  
286 positions from one to another. These considerations are the constraints of our problem that aims at  
287 tailoring the BAM microstructure by varying the aspect ratio  $L/H$  in order to maximize the fracture  
288 energy.

289 Fig. 5 shows an example of a BAM microstructure having two hierarchical levels: Level 0  
290 corresponds to a single W “brick” with length  $L$  and height  $H$ ; Level 1 includes the representative  
291 W-brick-and-Ni/Fe “mortar” unit cell that is repeated in both horizontal ( $x$ ) and vertical ( $y$ )  
292 direction to build the microstructural domain. The thickness  $t$  of the Ni/Fe mortar is adjusted such  
293 that the W volume fraction is kept at 0.8 for the modeling domain. Figs 6(b-f) show the as-  
294 generated BAM microstructures with  $L/H = 5, 10, 20, 30$  and  $40$ , and Fig. 6a presents the baseline  
295 lamellar-like microstructure for a comparison with the BAM ones.

#### 296 3.2 Elastic modulus computation

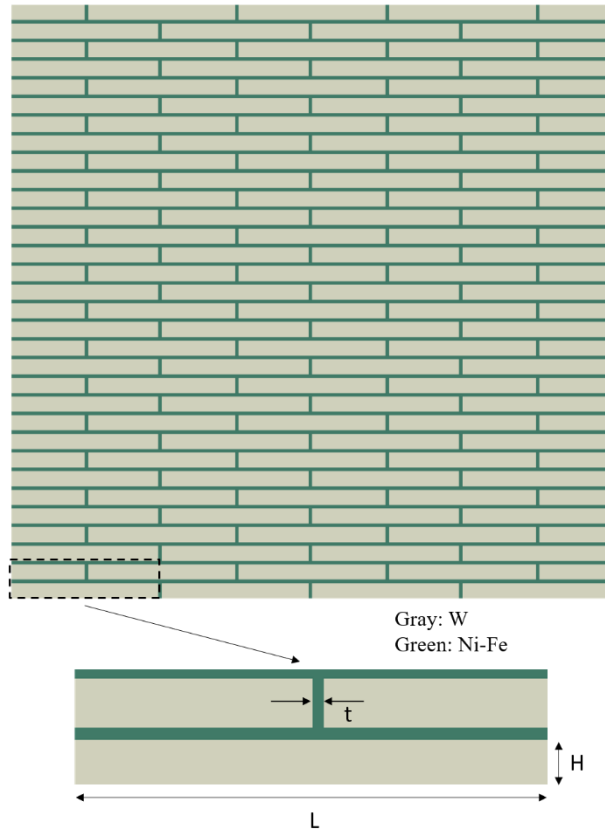
297 First, we studied the evolution of the elastic modulus of the BAM microstructures in the  
298 reinforcement direction of interest which is the  $x$ -direction along the brick length. To this end, the  
299 BAM microstructures were discretized in 2D finite elements (about 250000 elements) working  
300 under plane-stress or plane-strain assumptions subject to similar boundary conditions to those  
301 illustrated in Fig. 1b for the baseline microstructure. Figs 7a and 7b respectively report the plane-

302 strain and plane-stress FE solutions for the normalized composite elastic modulus,  $E_{xx}^c/E_b$  ( $E_b$   
303 being the W brick elastic modulus) as a function of the brick aspect ratio  $L/H$  for the prescribed  
304 values of the constituent modulus ratio,  $E_m/f_m E_b$ . Since the brick modulus (i.e.,  $E_b= 383$  GPa)  
305 and the brick volume fraction ( $f_b= 0.8$ ) are given, the mortar elastic modulus,  $E_m$  is determined  
306 from the modulus ratio and  $f_m = 1 - f_b$ . For a W-Ni/Fe composite studied in this paper or a W-  
307 Cu composite with the same W volume fraction and known constituent elastic modulus, the  
308 modulus ratios are 2.66 and 1.44, respectively. Figs 7a and 7b show that if the ratio,  $E_m/f_m E_b$  is  
309 small ( $\ll 1$ ), the composite modulus increases substantially with the increasing brick aspect ratio  
310  $L/H$  and tends to saturate for large aspect ratios ( $> 90$ ). In fact, the limit or saturation of the  
311 composite elastic modulus for the inclusion aspect ratio tending toward infinite is a known problem  
312 in composite materials. For larger and larger values of  $E_m/f_m E_b$ , the effect of  $L/H$  is smaller and  
313 smaller and becomes negligible for the W-Ni/Fe composite.

314 The plane-strain FE solutions agree very well with the numerical predictions by an Eshelby-  
315 Mori-Tanaka approach (EMTA [27-29]) only for small modulus ratios ( $< 0.1$ ), while the plane-  
316 stress FE results agree globally well with the EMTA predictions for all constituent modulus ratios  
317 considered including the values for the W-Ni/Fe and W-Cu composites. In the EMTA  
318 computation, ellipsoidal brick inclusions were assumed. To further investigate the elastic modulus  
319 computations, our FE plane-strain results and EMTA solutions presented in Fig. 7a are compared  
320 to the Begley et al.'s plane-strain micromechanical solutions [7] for the same modulus ratios (0.02,  
321 0.05 and 0.1). It is seen in Fig. 8 that Begley et al.'s results agree fairly well with the FE solutions  
322 for small to moderate brick aspect ratios ( $L/H < 20$ ), but the results diverge for larger brick aspect  
323 ratios. The same findings are found when comparing EMTA's solutions to Begley et al.'s results.  
324 In fact, the modulus saturation for large brick aspect ratios is not realized in Begley et al.'s model,  
325 and that explains the significant divergence of our solutions compared to Begley et al.'s for large  
326 brick aspect ratios.

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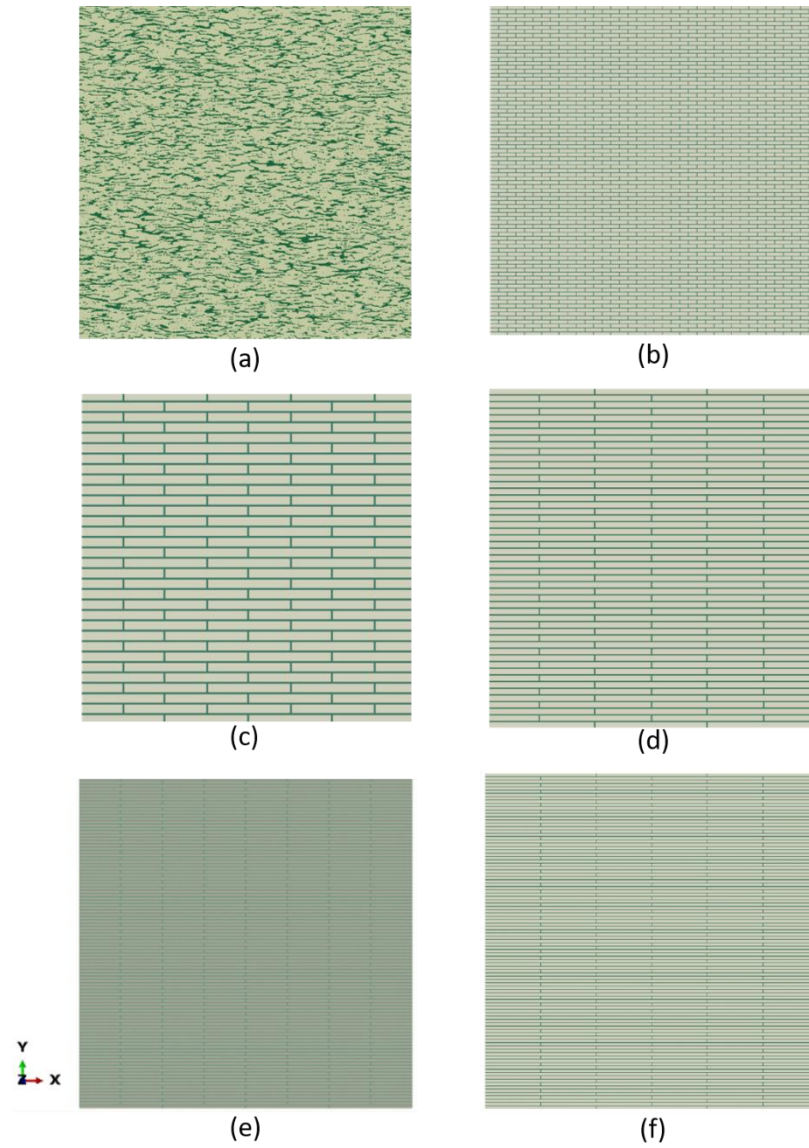
330 **Figure 5.** An example of BAM microstructure for W/Ni-Fe showing two hierarchical levels:

331 Level 0 corresponds to a single W “brick” with length  $L$  and height  $H$ ; Level 1 includes the

332 representative W-brick-and-Ni/Fe “mortar” unit cell that is repeated in both horizontal (x) and

333 vertical (y) direction to build the microstructural domain.

334



335

336 **Figure 6.** (a) Baseline lamellar-like microstructure - Generated BAM W/Ni-Fe microstructures

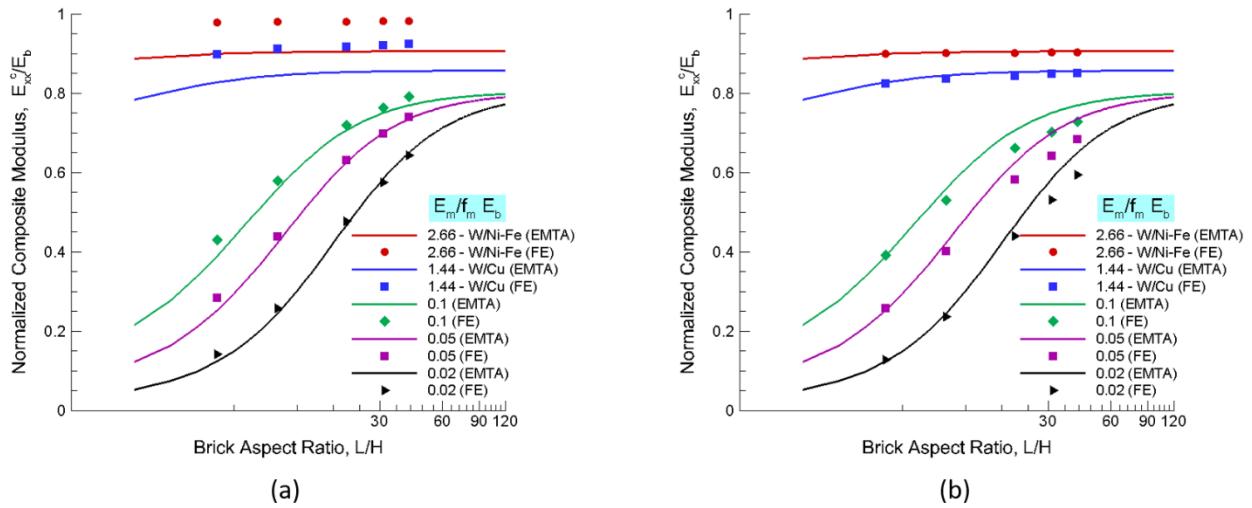
337 (gray: W, green: Ni/Fe): (b)  $L/H=5$ , (c)  $L/H=10$ , (d)  $L/H=20$ , (e)  $L/H=30$ , and (f)  $L/H=40$ .

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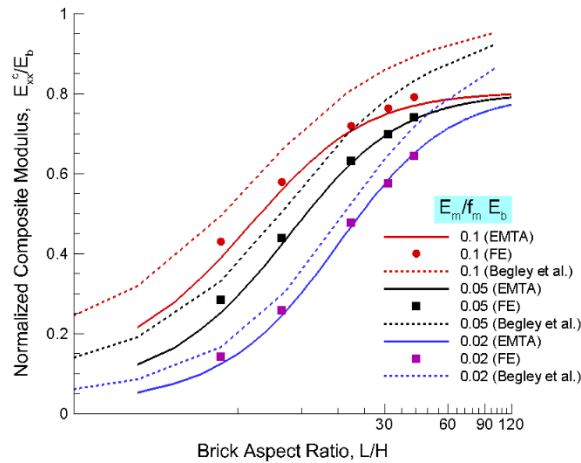
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342

343 **Figure 7.** Normalized composite modulus as a function of the brick aspect ratio: (a) Plane-strain  
 344 FE and (b) Plane-stress FE results compared to EMTA solutions.



345

346 **Figure 8.** Normalized composite modulus as a function of the brick aspect ratio: plane-strain FE  
 347 results and EMTA predictions compared to Begley et al.'s results [7].

348

349 **3.2 Stress-strain response and damage predictions**

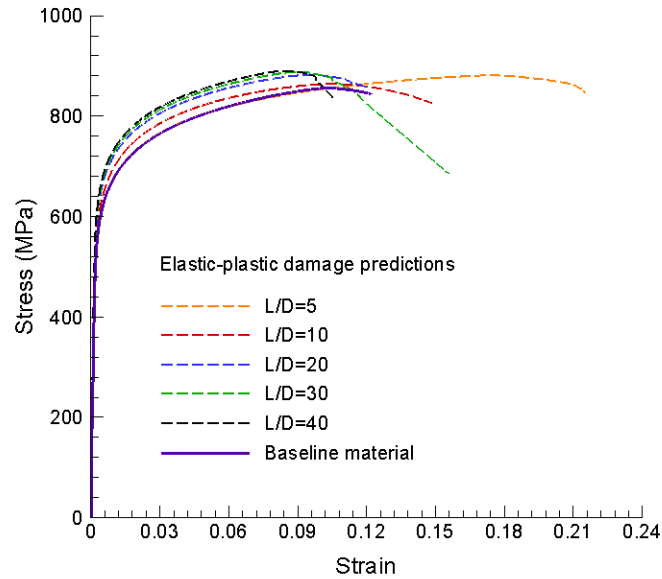
350 A series of FE analyses of the BAM models illustrated in Fig. 6 was conducted to compare the  
 351 mechanical performance of the BAM microstructures in terms of the overall stress-strain response  
 352 to that of the actual hot-rolled microstructure (Fig. 6a). The model parameters given in Table 1

353 were used for W and Ni/Fe in all the FE analyses of the BAM microstructural models to investigate  
 354 the W brick aspect ratio and microstructural effects on the stress-strain behavior, damage and crack  
 355 development in these composites. Fig. 9 reports the stress-strain curves up to failure of these  
 356 composites compared to the predicted response of the baseline composite shown in Fig. 2. The  
 357 results on this figure show that increasing the brick aspect ratio  $L/H$  from 10 to 40 results in a  
 358 significant increase in strength but a reduction in fracture strain. In addition, strength saturation  
 359 with increasing  $L/H$  from  $L/H = 30$  has been found. For small brick aspect ratios such as the case  
 360 for  $L/H=5$  illustrated in Fig. 6b, the model predicts a significant increase in strength combined  
 361 with an important increase in fracture strain. Table 2 gives the predicted strengths and fracture  
 362 strains for all the cases. The fracture strains correspond to the strains at the maximum stresses  
 363 (strength) on the stress-strain responses presented in Fig. 9. It is seen in this table that *under the*  
 364 *prescribed constraints and selected design parameters for our microstructural tailoring problem,*  
 365 the BAM with  $L/H=40$  possesses the highest strength (889 MPa) but the lowest fracture strain  
 366 (0.0848) while the BAM which  $L/H=5$  appears to have an optimum solution with a high strength  
 367 (880.4 MPa) and the highest fracture strain (0.1737).

368 **Table 2.** Predicted strengths and fracture strains for the BAM microstructures compared to  
 369 the predicted and experimental results for the baseline hot-rolled microstructure.

Microstructure	Strength (MPa)	Fracture Strain
Hot-rolled (predicted)	856.6	0.1064
Hot-rolled (experimental*)	867.8	0.112
BAM ( $L/H = 5$ )	880.4	0.1737
BAM ( $L/H = 10$ )	863.	0.1054
BAM ( $L/H = 20$ )	881.2	0.0954
BAM ( $L/H = 30$ )	887.3	0.0904
BAM ( $L/H = 40$ )	889.	0.0848

370 (\*) Average values from two experiments

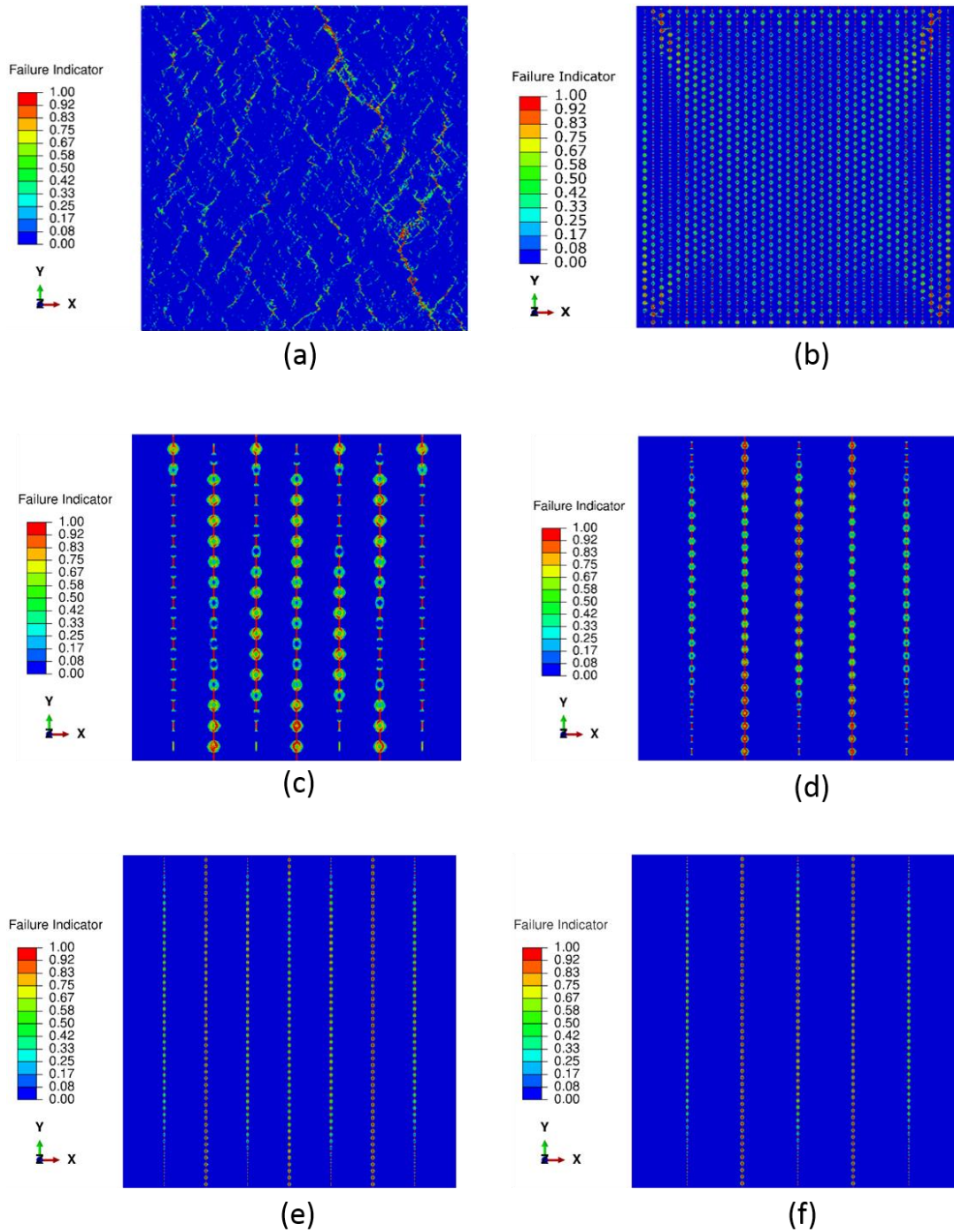


371

372 **Figure 9.** Predicted tensile stress-strain responses for the BAM composites shown in Figs 6(b-f)  
 373 compared to the predicted response for the baseline material (Fig. 6a).

374

375 The damage and fracture patterns for all the cases presented in Figs 10(a-f) elucidate the stress-  
 376 strain responses in Fig. 9. For comparison, Fig. 10a shows the results for the baseline hot-rolled  
 377 material already discussed in Section 2. Figs 10(b-f) illustrate the damage and fracture patterns in  
 378 the BAM microstructures with aspect ratios 5, 10, 20, 30 and 40, respectively. As previously  
 379 mentioned, in addition to the *DPT mechanism* caused by failed Ni/Fe elements, there are two other  
 380 mechanisms found: *crack penetration across tungsten phase* and *multiplication of cracks* over the  
 381 domain. More importantly these mechanisms combined occur during loading, more fracture energy  
 382 is dissipated leading to higher strength and fracture strain. This is realized with the  $L/H = 5$  BAM  
 383 microstructure. The fracture patterns in the other BAM microstructures (with  $L/H = 10, 20, 30, 40$ )  
 384 are quite similar: increasing  $L/H$  has led to increased strength. More damage distributed in the  
 385  $L/H=10$  BAM has yielded higher fracture strain for this case. All the BAM materials exhibit higher  
 386 strength than the baseline composite's predicted strength.

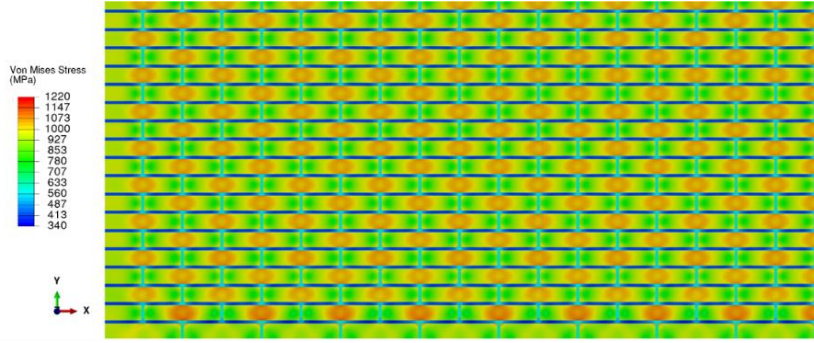


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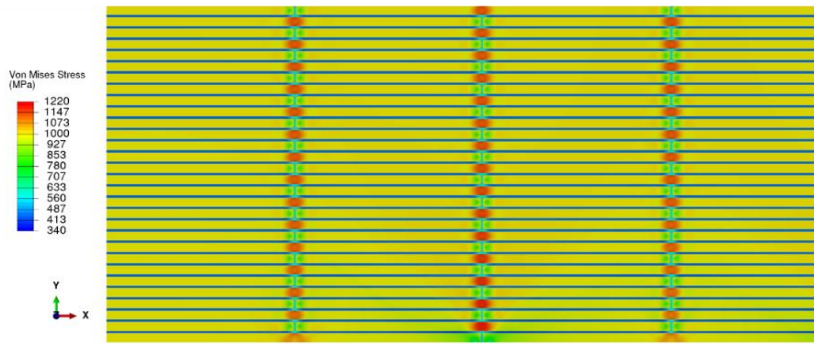
388 **Figure 10.** Predicted damage distributions and fracture patterns at final failure for: (a) the baseline  
 389 lamellar-like microstructure, (b)  $L/H=5$  (c)  $L/H=10$ , (d)  $L/H=20$ , (e)  $L/H=30$ , and (f)  $L/H=40$  BAM  
 390 microstructures.

391 The equivalent Von Mises stress distributions in the BAMs with  $L/H = 5$  and 40 presented in  
 392 Figs 11a and 11b, respectively, for a magnified zone at a stage of loading prior to the W onset of

393 cracking, further elucidate the origin of damage and failure mechanisms. The applied strain at this  
394 stage is 0.08. The size of this zone from the respective BAMs is the same. The stress concentrations  
395 build up in all the Ni/Fe ligaments and expand from the junction locations to the W bricks. Under  
396 increasing loading, the Ni/Fe ligaments fail first and the stress concentrations continue to build up  
397 in the W bricks until the bricks break and the cracks propagate through them. Similar to the  
398 mechanisms observed for the baseline material discussed previously (Fig. 4), the DPT mechanism  
399 due to failure of Ni/Fe elements combined with the crack-penetration-in-W-phase mechanism has  
400 led to an increased strength for the BAM composite. Comparing Fig. 11a to Fig. 11b reveals that  
401 there are substantially more stress concentration sites in the  $L/H=5$  BAM than in the  $L/H=40$  BAM.  
402 These stress concentration sites in the  $L/H=5$  BAM produces many microcracks that are distributed  
403 over the domain while more limited cracks and crack distribution occur in the  $L/H=40$  domain.  
404 Compared to the baseline microstructure, the  $L/H=5$  BAM also has many distributed microcracks  
405 but the regular lattice microstructure does not contain a weak path like that occurring in the  
406 baseline one that favors crack propagation along such a path. More cracks distributed over the  
407 domain have triggered crack interactions at advanced stages of loading, and the combined crack  
408 distribution/interaction effect has produced increased fracture energy leading to substantially  
409 increased fracture strain for  $L/H=5$  case. The strength obtained in the  $L/H=5$  case is also  
410 significantly increased compared to the  $L/H=10$  case (Table 2).



(a)



(b)

411  
 412 **Figure 11.** Magnified views of Von Mises stress distribution inside (a)  $L/H=5$  BAM and (b)  
 413  $L/H=40$  BAM composites subject to tensile loading at 0.08 strain.

414 **4. Conclusions**

415 We have applied a multiscale microstructural approach to simulate tensile loading of W-Ni/Fe  
 416 specimens cut out from a hot-rolled W-Ni/Fe composite with 0.8 W volume fraction. The predicted  
 417 material stress-strain response and crack patterns were then compared to the corresponding  
 418 experimental results to determine the elastic-plastic damage model parameters for W and Ni/Fe  
 419 phases in this domain. Subsequently, the model parameters were used to analyze W-Ni/Fe BAM  
 420 microstructures (arbitrarily designed) to investigate the effects of microstructural features on the  
 421 composite stress-strain response, strength, damage, and fracture patterns. The cracking and  
 422 fracture of the rolled W-Ni/Fe alloy (defined as the baseline material) appeared to be related to a  
 423 BAM microstructure consisting of weak but stiff bricks, or nearly brittle bricks, embedded in a  
 424 tough but less stiff mortar. We imagined building such a hierarchical microstructure using a simple  
 425 half-stagger BAM model. We were able to assign the W-bricks and Ni-Fe mortar in the BAM  
 426 microstructure the same constitutive properties as our baseline material using our developed FE

427 CDM model and the resulting stress-strain curves suggested that a finely divided but regular BAM  
428 microstructure was significantly better than our rolled W-Ni/Fe microstructure, but overall very  
429 similar. This type of correspondence needs to be further explored in the future but our model  
430 appears promising as a method to explore and optimize BAM architectures even though the FE  
431 discretization has some shortcomings in capturing some of the details of our rolled microstructure,  
432 such as unable to account for the weak W-W grain boundaries present in the rolled material.

433 The focus of this study is on tailoring the DPT W hierarchical microstructure to increase the  
434 resulting composite strength, ductility and fracture energy. Thus, the use of an elastic-plastic  
435 damage model with the rate-independent plasticity is justified for this purpose. However, other  
436 relevant types of constitutive behavior such as i.e., high-temperature and time-dependent behavior,  
437 irradiation hardening behavior, etc. can be modeled at the constituent phase level and incorporated  
438 in the developed multiscale modeling approach. For instance, the high-temperature response of the  
439 composite can be predicted by the same approach using the thermomechanical properties and  
440 stress-strain data at the corresponding high temperatures for the model parameter determination.  
441 In addition, thermal stresses due to the mismatch of coefficients of the thermal expansion between  
442 phases must be included in the constitutive equations. Regarding the application of this multiscale  
443 microstructural approach to high-temperature behaviors, the reader is referred to our previous work  
444 [14-15] in which it was applied to simulate microcracking of  $Ti_3SiC_2/SiC$  joints subjected to  
445 thermomechanical loading in addition to irradiation-induced swelling. To account for the loading  
446 rate effect, a rate-dependent plasticity constitutive model for the constituent phases must be used,  
447 and the model parameters should be determined for a given loading rate and then used to validate  
448 the model for other loading rates.

449 In this work, the model parameters have been determined from a hot-rolled W-Ni/Fe  
450 composite. Applying the developed multiscale approach to an unrolled W-Ni/Fe composite or a  
451 similar composite obtained under other rolling conditions (i.e., different rolling reduction ratios)  
452 requires the careful meshing of the as-formed microstructure (rolled or unrolled) by the same  
453 method described in Section 2.1 (see also Ref. [4]). Tensile stress-strain data obtained from  
454 unrolled or rolled materials from other rolling conditions should then be used to determine the  
455 model parameters since rolling conditions do have an important effect on the constituent phase  
456 material plasticity.

457 The following insights are drawn from this study:

- 458 - Different modeling scales from the micron scale of the constituent phases to a macroscale  
459 domain of a specimen are effectively bridged through the FE analysis that accounts for the  
460 constituent constitutive behavior.
- 461 - The microstructure study reveals the important effects of toughening mechanisms  
462 controlling the composite material mechanical properties that have been investigated in  
463 this work in terms of strength and failure strain. In addition to the DPT mechanism, crack  
464 penetrating into W phases combined with crack distribution and interaction over the  
465 domain can significantly contribute to the increased fracture energy, thus both increased  
466 strength and fracture strain for the composite.
- 467 - The predictive modeling tool has been applied to tailor the mechanical properties of W-  
468 BAM composites with the brick aspect ratio  $L/H$  as the selected design parameter under  
469 specific constraints on W volume fraction (0.8), the same dimension of the modeling  
470 domain, and regular lattice topology. Among the explored aspect ratios, BAM composite  
471 with  $L/H = 5$  exhibits the most preferred combination of strength and fracture strain.
- 472 - The predictive modeling capability can be applied to explore additional material  
473 combinations, other microstructural topologies and parameters to determine optimal  
474 combinations of stiffness, strength, ductility and toughness for BAM composites.
- 475 - This capability can provide guidance to assist in processing BAM composites via a process  
476 such as e.g., 3D printing.

477 Our future work will include the application and extension of the developed approach to  
478 investigate W-composite high-temperature mechanical behaviors and irradiation effects.

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