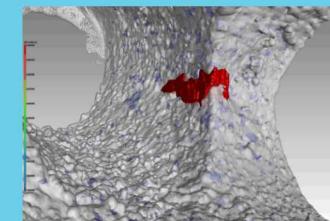
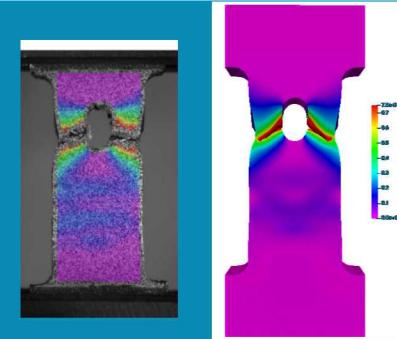
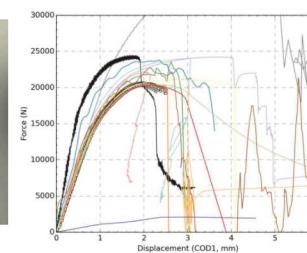
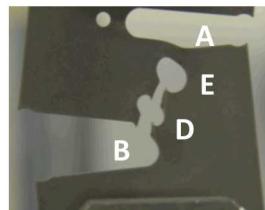
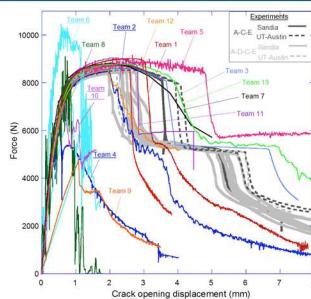
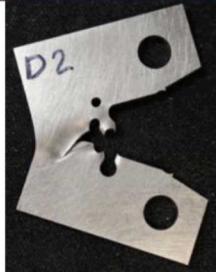


# The Third Sandia Fracture Challenge: A Deep Dive into the Experimental Results



Charlotte Kramer, Brad Boyce, **Amanda Jones**,  
Jhana Gearhart, Brad Salzbrenner, Thomas Ivanoff,  
Jonathan Madison, and Andrew Lentfer

12 June 2019

CFRAC 2019- Braunschweig, Germany

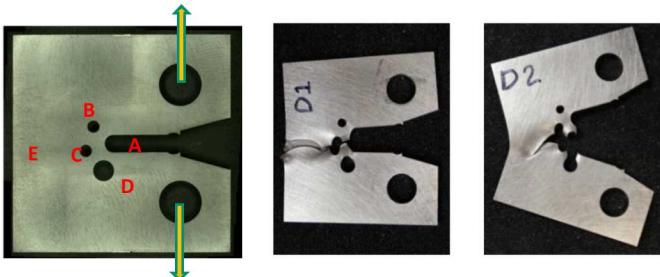


Sandia National Laboratories is a multimission laboratory managed and operated by National Technology & Engineering Solutions of Sandia, LLC, a wholly owned subsidiary of Honeywell International Inc., for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-NA0003525.

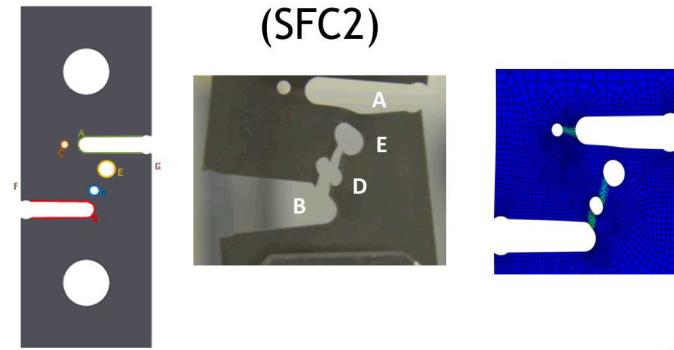
# Presentation Organization

## What is the Sandia Fracture Challenge?

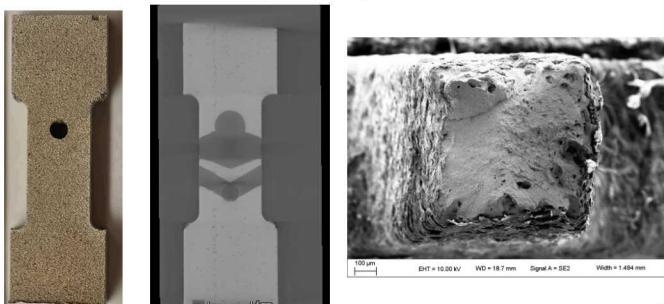
### First Sandia Fracture Challenge (SFC1)



### Second Sandia Fracture Challenge (SFC2)



### Third Sandia Fracture Challenge (SFC3)



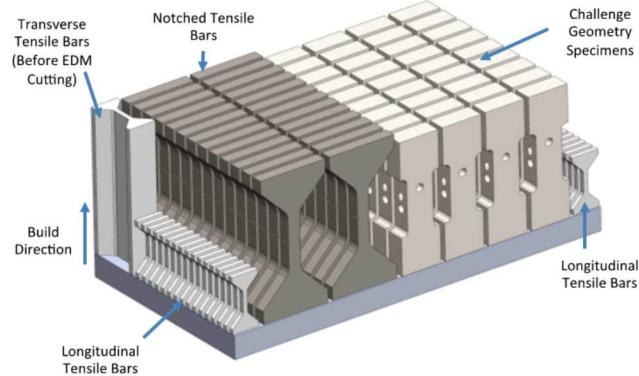
### Structural Reliability Partnership (SRP)



# SFC3: Failure in an AM Metal

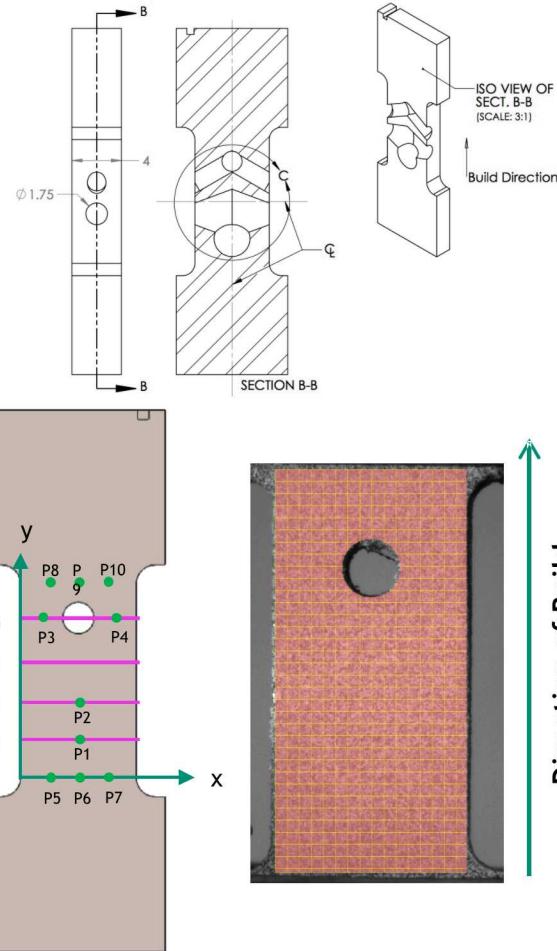
The Third Sandia Fracture Challenge explores the experiments and model methods required to predict ductile failure in AM metal parts.

SFC3 AM Build Plate



## Description of Challenge

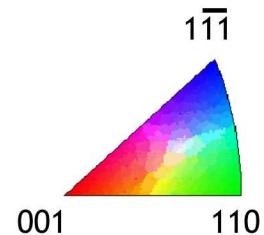
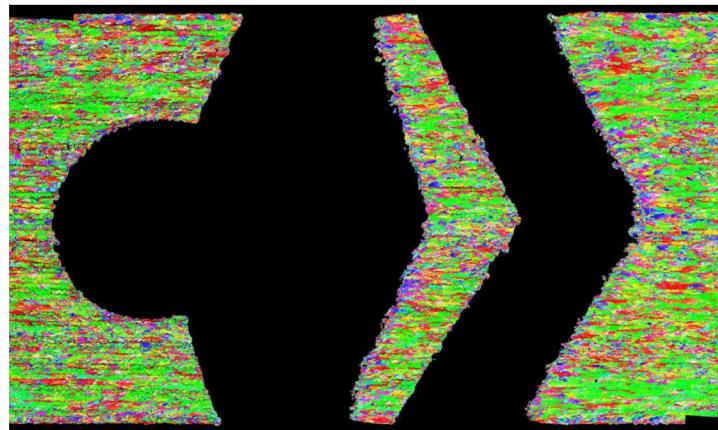
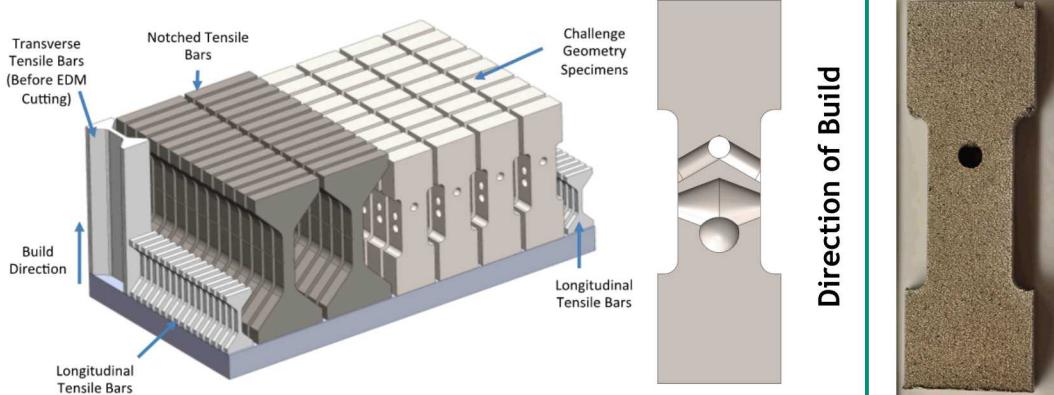
- Material:** Additively manufactured 316L stainless steel
- Extensive material property information**
- Six Questions of global and local measures** based on load and DIC measurements
  - Required nominal response and requested 20<sup>th</sup> and 80<sup>th</sup> percentile bounds as optional
- Challenge Issuance:** December 15<sup>th</sup>, 2016
- Prediction Deadline:** July 15<sup>th</sup>, 2017



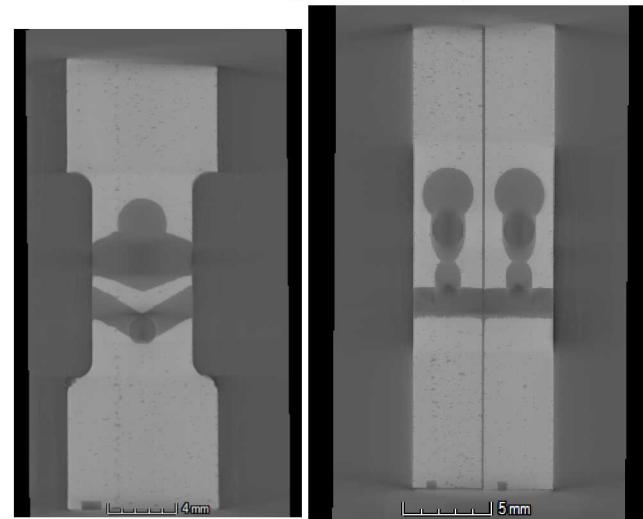
Challenge Geometry: Drawing (Top Image), Front View Schematic of Points and Lines of Interest (Bottom Left Image), and Representative DIC Area of Interest 41-pix Subsets (Bottom Right Image)

# Examples of Provided Data

AM Build of All Test Specimens (Left); Central Cross-Section Schematic (Center) and Front Views of the Challenge Geometry (Right)

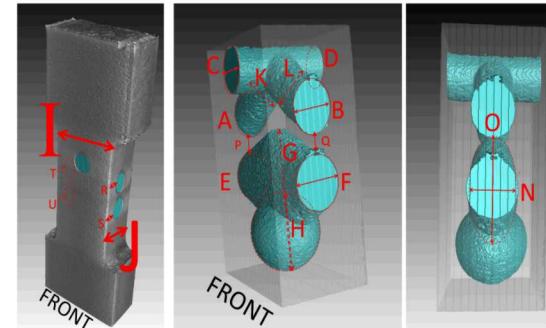


Example of micro-CT Scans of Challenge Geometry



A32 Front View  
"Thick Slab" Image

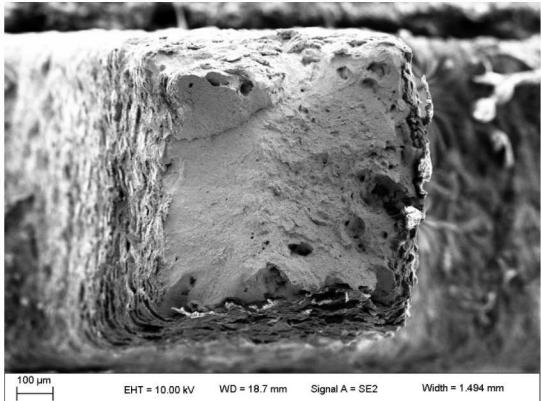
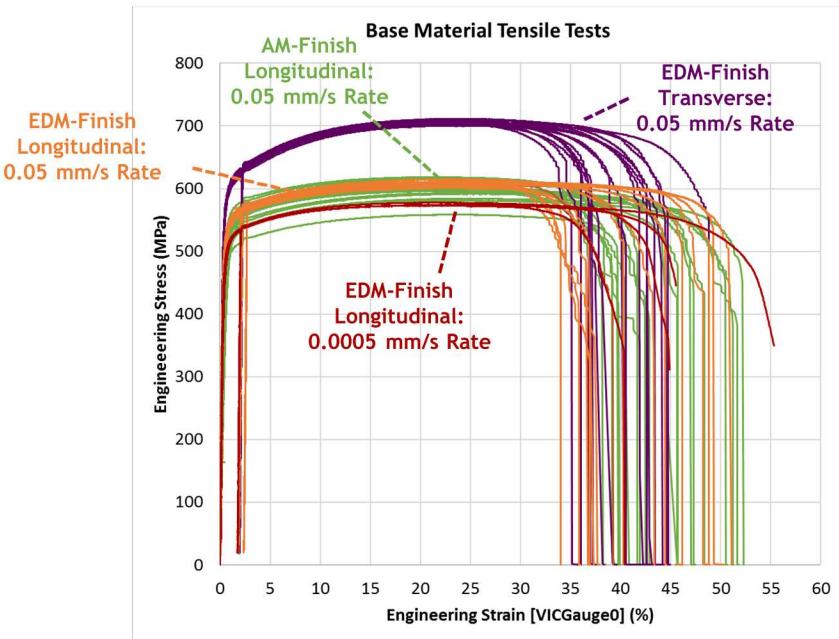
Side View Image of  
A32 (left) and A15 (Right)



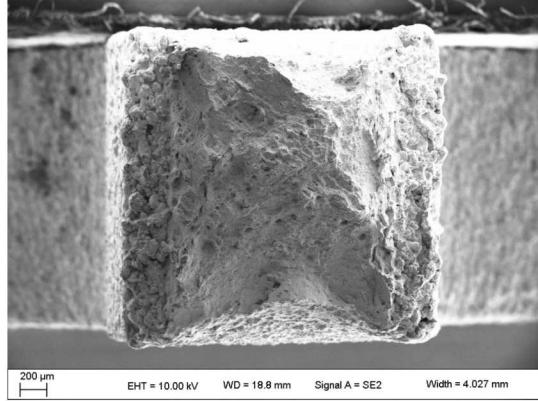
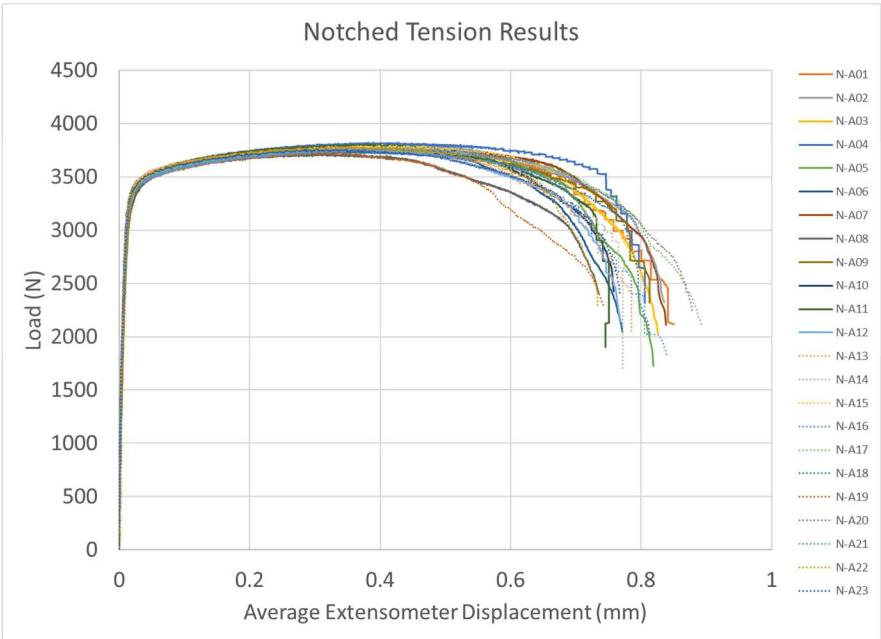
Reconstructed micro-CT Scans with  
Provided Feature Measurements

# Examples of Provided Data

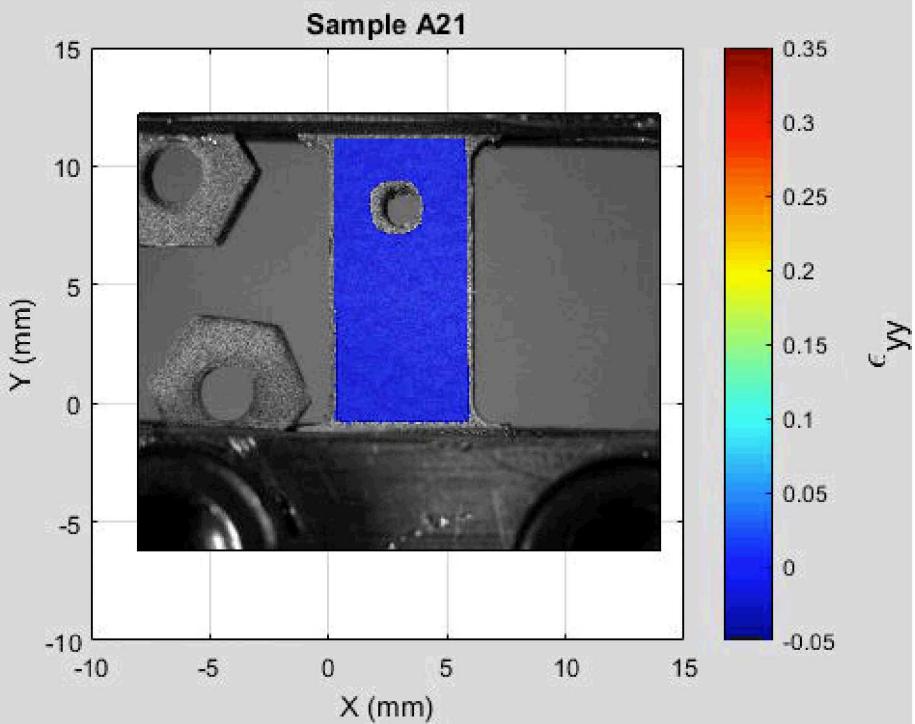
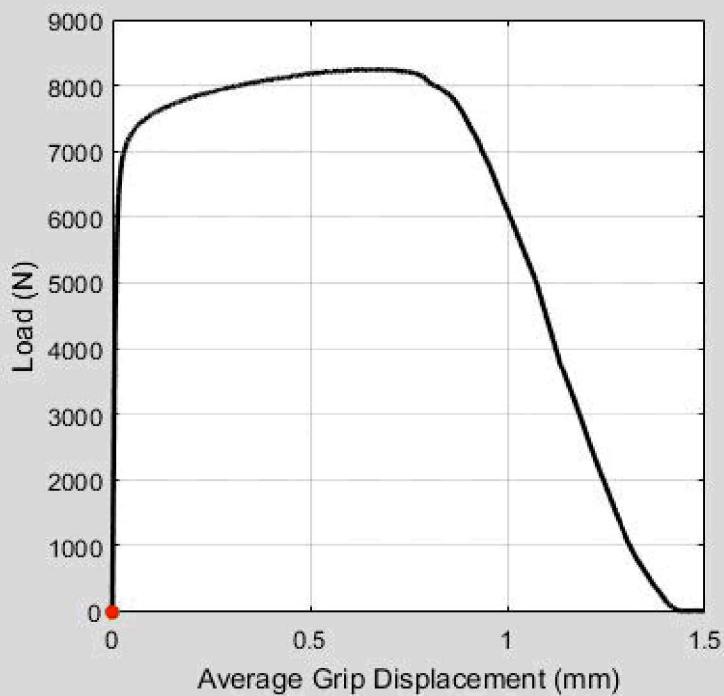
Base Material Test Data (Top); SEM Image of Fracture Surface of Tensile Specimen LTA04 (Bottom)



Notched Tensile Test Data (Top); SEM Image of Fracture Surface of Notched Tensile Specimen NA05 (Bottom)

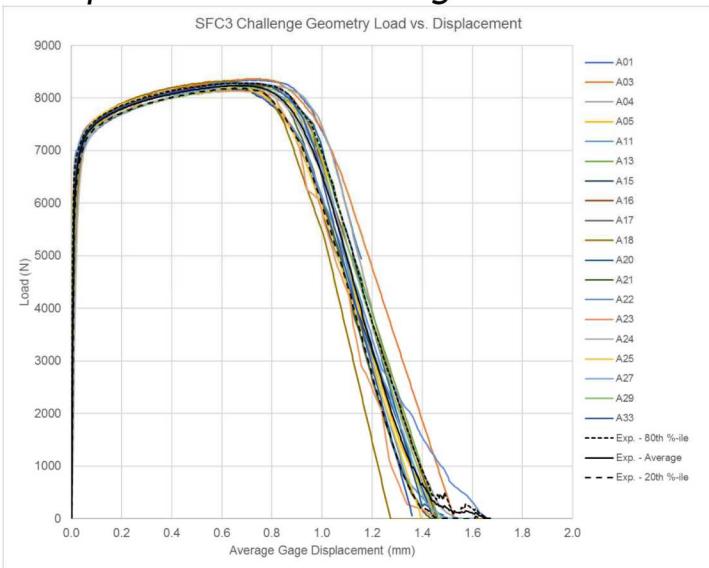


# Challenge Geometry Experimental Result



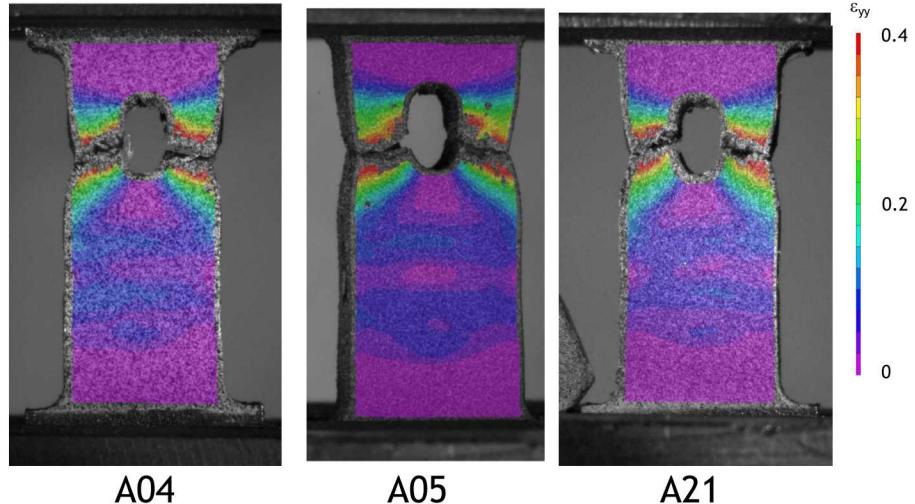
# Experiments: Question 3 & 6

## Force vs. Gage Displacement D 19 Specimens with Average and Bounds



- Relatively repeatable experimental data set with all specimen failing in nominally the same location
- Experimental data from two testing laboratories overlap
- Crack path is similar for each specimen, but not necessarily following the angled channels

## Hencky Tensile Strain Fields at Failure

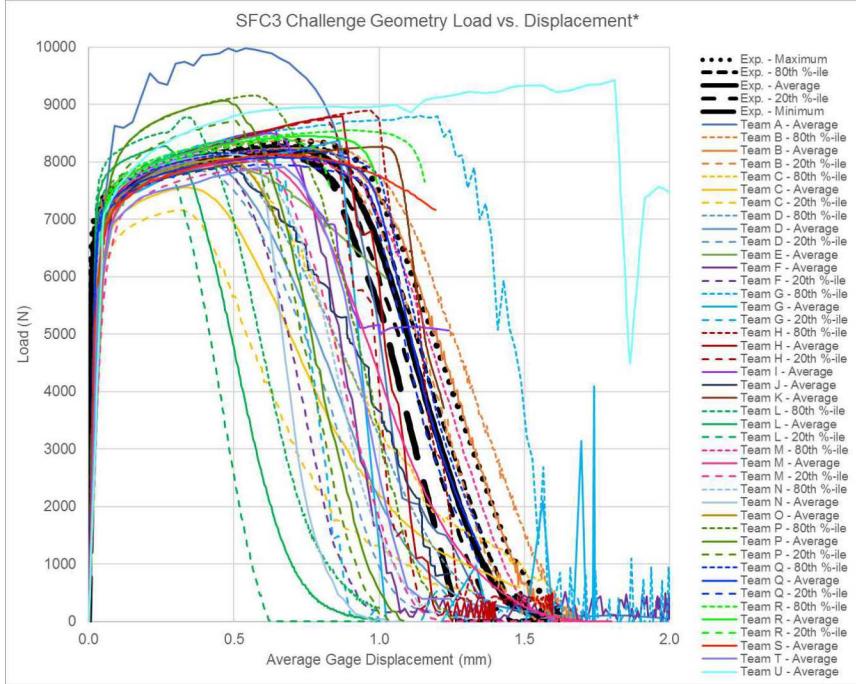


## Digital Image Correlation Parameters

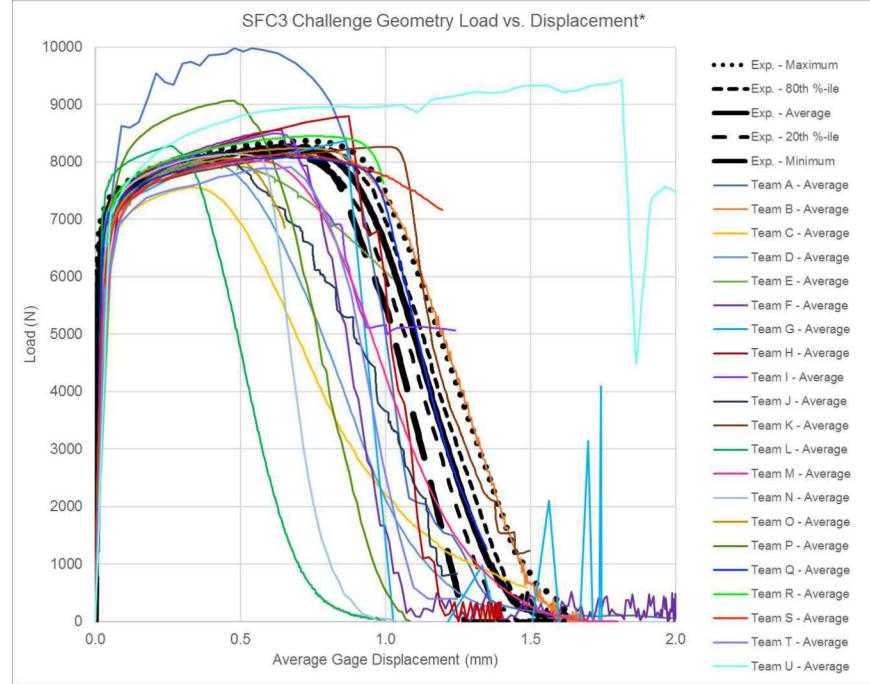
Subset	41 pixel x 41 pixel	0.368 mm x 0.368 mm
Step	7 pixels	0.063 mm
Strain Window	9 step	--
Virtual Strain Gage	57 pixel	0.512 mm

# Predictions: Global Measures

## 21 Predictions and Bounds with Exp. Average and Bounds

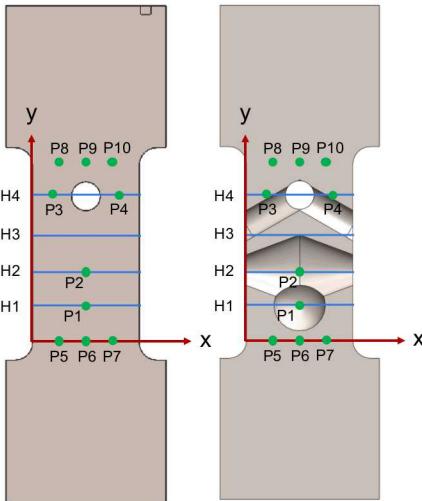


## 21 Nominal Predictions with Exp. Average and Bounds



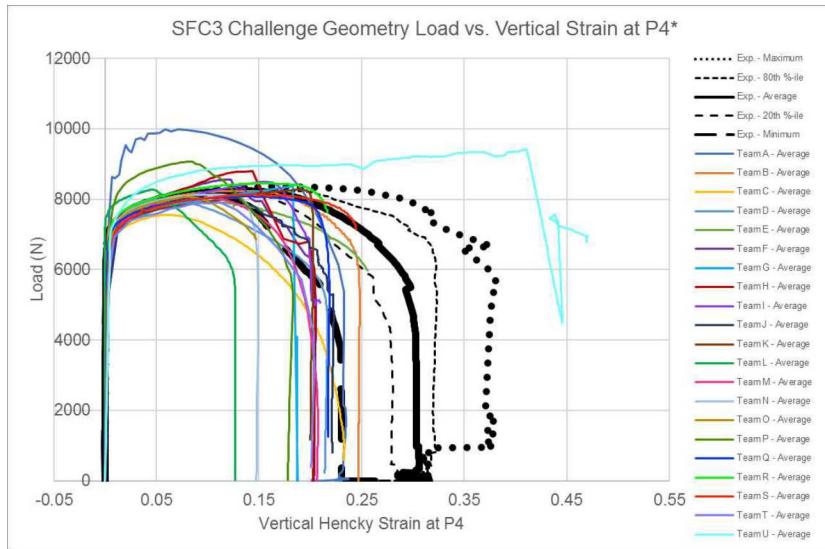
- All 21 predictions correctly identified the nominal crack path with initiation at the through-thickness hole
- More teams under-predicted the failure displacement than over-predicted.
- There were only two teams whose nominal prediction fell within the bounds of the experimental data (Teams B and Q)
- The uncertainty bounds on predictions ranged from too small to too large

# Predictions: Local Measures



Challenge Geometry  
(Surface - Left,  
Cutaway - Right)

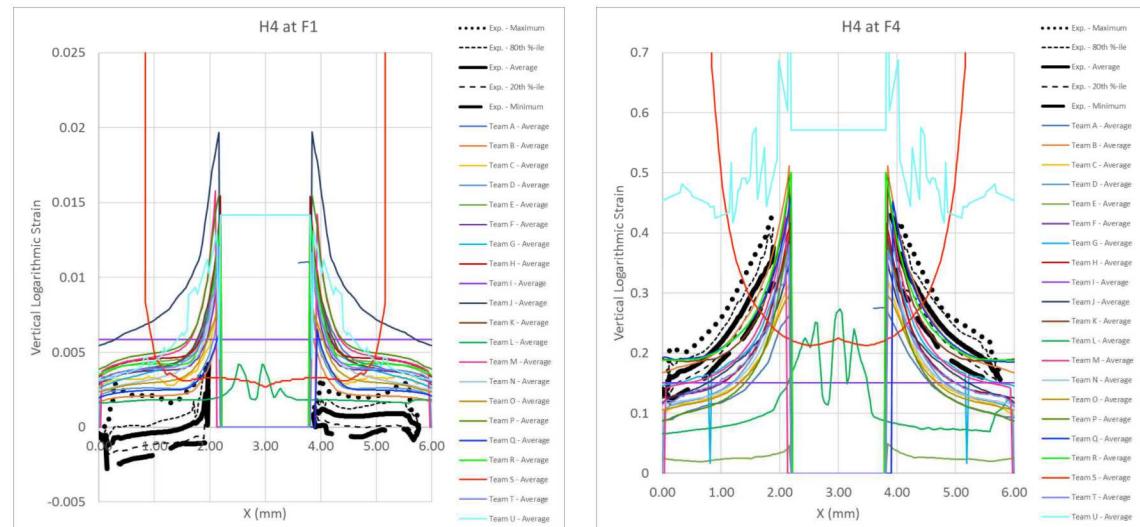
Question 4: Report force vs. Hencky strain in the vertical direction ( $\epsilon_{yy}$ ) at P4.



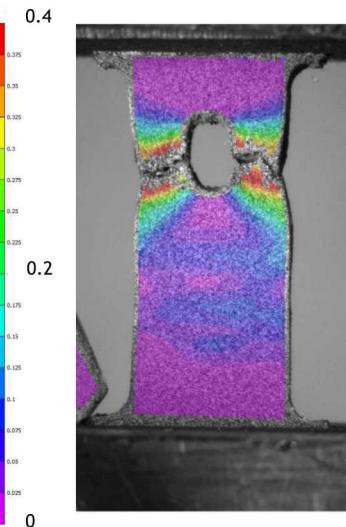
Strain predictions close to failure were generally too low.

Question 5: Report force and Hencky strain in the vertical direction ( $\epsilon_{yy}$ ) along horizontal line H4 on the surface at forces F1, F2, F3, and F4.

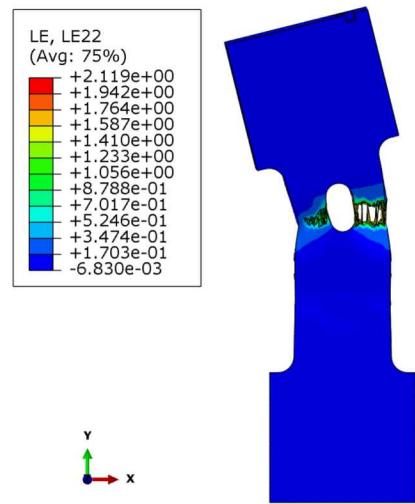
Most teams over-predicted the nominal strain at F1, but many teams' predictions improved for higher forces.



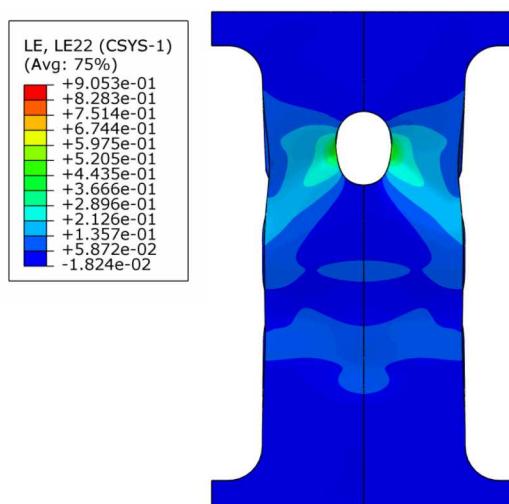
# Example Predictions: Question 6 Failure - Strain



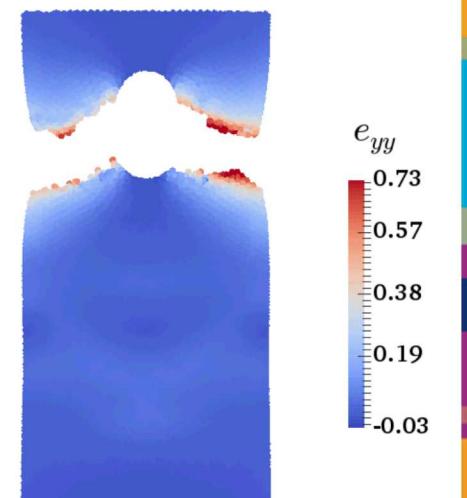
A13



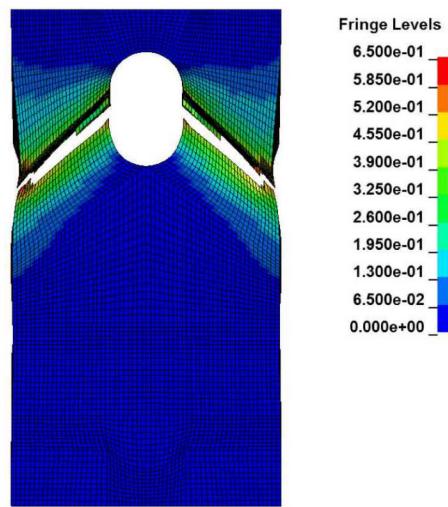
Team A



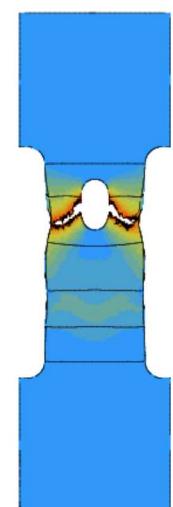
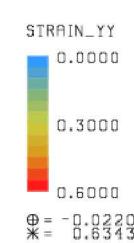
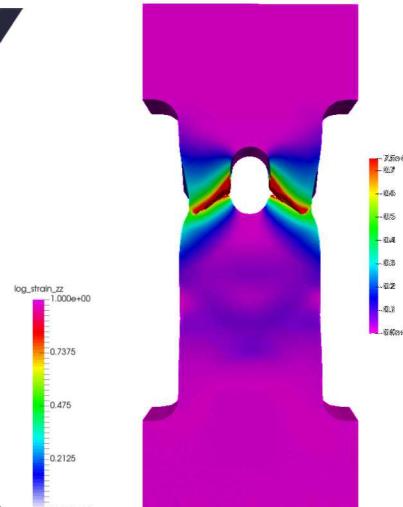
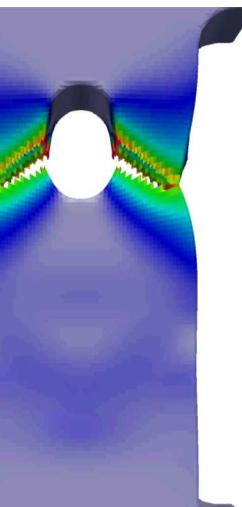
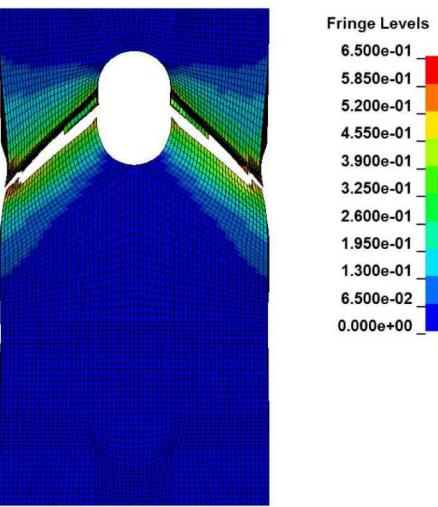
Team B



Team C



Time: 111.7 Seconds



Team L

Team M

Team Q

Team T

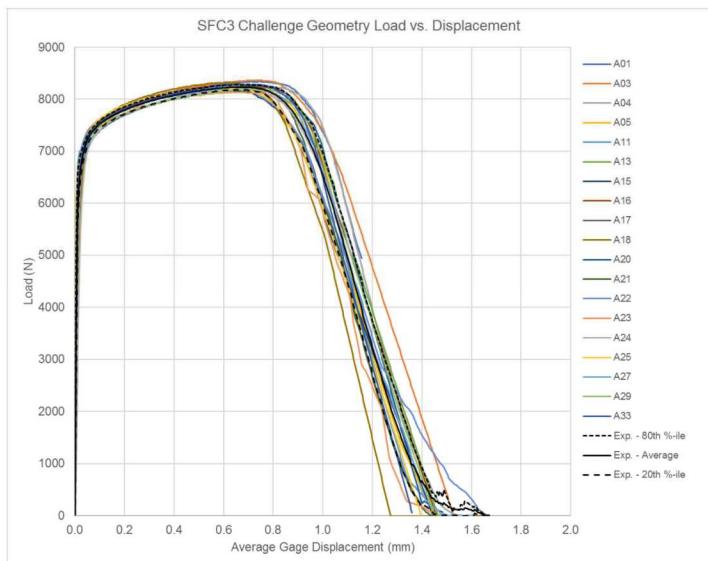
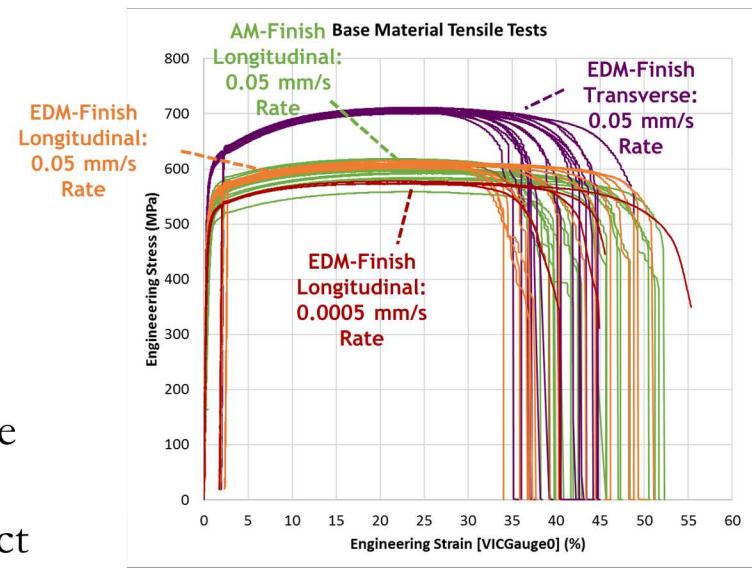
# SFC3 Observations

## Predictions

- All teams predicted the correct crack path
- Two teams predicted a nominal load-displacement behavior within experimental bounds
- All teams used provided **tensile data of AM-manufactured specimens** with similar void structure as the Challenge geometry
- Predictions of **surface strains** (local measures) were generally different than in experiments
- The teams took vastly different approaches to predict **uncertainty bounds** in their models
- Few teams considered the geometric variation and pore structures characteristic of AM metals, despite considerable data provided to aid that effort
- The post-blind assessment revealed several **clerical errors** in the predictions

## Overall

- **Large features in the AM part overwhelmed the overall response** of the structure compared to the unintentional voids, a hope to use of AM parts
- **We need further research to understand the role of the voids of the local fracture processes**



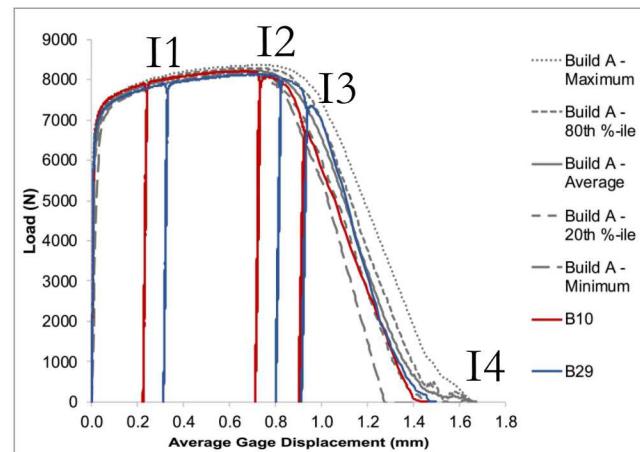
## SFC3 Experimental Reinvestigation

**Question: What is the effect of pre-existing voids on deformation, damage, and failure in AM metallic structures like the SFC3-geometry specimens?**

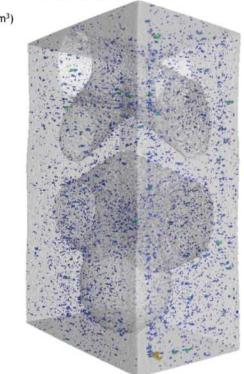
We will look at the global behavior and local crack initiation and growth relative to pre-test void population and the evolution of void growth and crack evolution via interrupted testing with ex situ micro-CT imaging.

- Tested 6 “Build B” specimens to failure to see overlap with original “Build A” specimens
- Interrupted test intervals for six specimens:
  - I1 – To middle of hardening
  - I2 – Peak load
  - I3 – Visible crack
  - I4 - Failure
- All interrupted test specimens had the same failure mode as tests to complete failure

Example of  
Interrupted Response



Pre-test  $\mu$ -CT  
Data Showing  
Voids



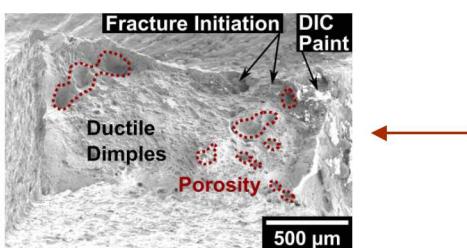
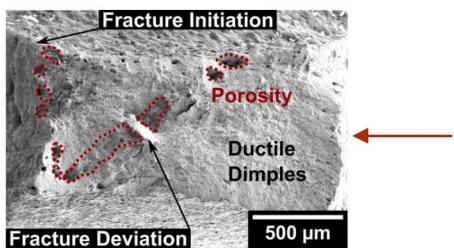
### Geometry Dominates Global Behavior in SFC3 Specimens:

Metrics of the pre-existing void population do not correlate with the global mechanical behavior of the SFC3 specimens, but rather the large stress concentrations from the geometry overwhelmingly dominate the global behavior.

### Voids Influence Local Crack Initiation and Growth:

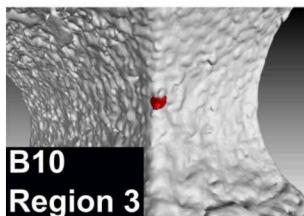
Voids and surface defects influence local crack initiation and growth by introducing variation in crack initiation site in some cases and deviation from initial crack path to intersect voids.

#### Fracture Surfaces



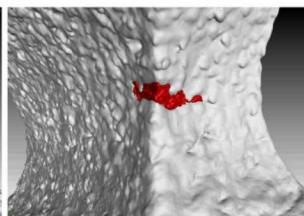
#### 3D Reconstructions Highlighting Crack Volume

Strain Interval 1



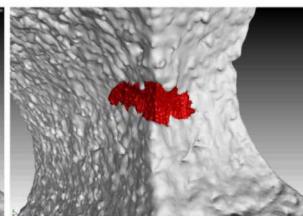
(a1)

Strain Interval 2



(a2)

Strain Interval 3

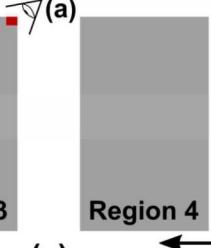


(a3)

Viewing Orientations



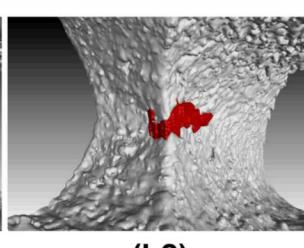
(b)



(a)

B33  
Region 1

(b1)



(b2)



(b3)

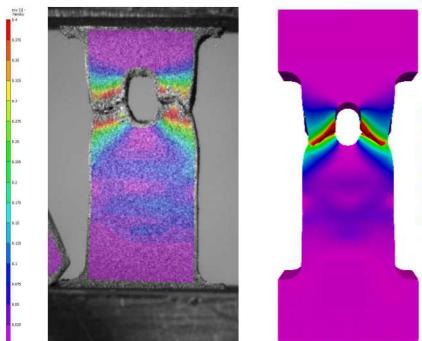
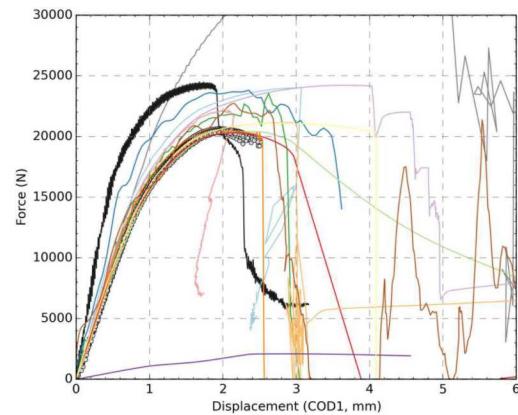
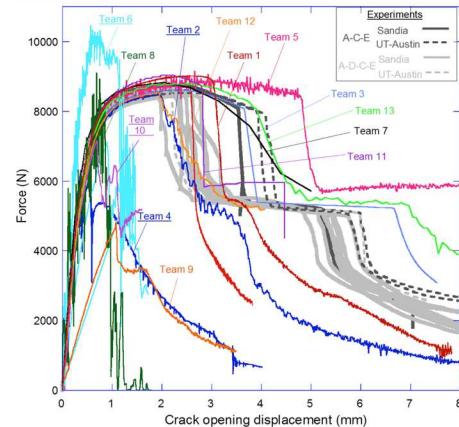
(c)

$x$   
 $-z$

- Fracture at high-stress intersection point or at large surface defect
- Crack initiation at a surface defect occurs in I1, while otherwise can happen at I1-I3
- Cracks deviate to intersect voids
- Cracks initiating at surface defects grow faster, but total specimen crack volume growth does not vary much

# SFC: What We Have Learned So Far

- A **good calibration of plasticity** is a prerequisite for predicting ductile failure
- **No single numerical method, constitutive model, or failure criterion** dominates amongst the more successful predictions
- **Experience and peer review** matter
- We need considerable **research on ductile failure** including:
  - **Stochastic and local nature of ductile failure**
  - Accepted **shear-dominated experiments** for model calibration
  - Theoretical and experimental work on **thermomechanical coupling**
  - Effect of material **anisotropy** on failure
  - **Efficient constitutive and failure models** for large structures
  - **Quantitative comparison** of full-field experimental and computational data
  - Methods for **uncertainty quantification** on engineering time-scales



# STRUCTURAL RELIABILITY PARTNERSHIP

The purpose of the Structural Reliability Partnership is to coordinate research, share best practices, and leverage investments from multiple institutions on areas of mutual interest in the domain of structural reliability.

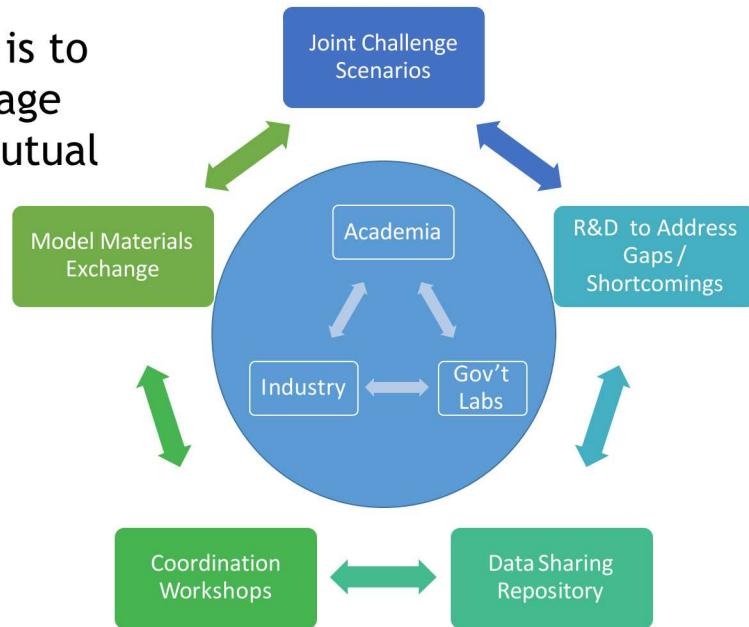
## PILOT PARTNERS:



## Initial Participant Institutions:

Lawrence Livermore National Lab  
 University of Utah  
 Purdue University  
 National Science Foundation  
 Pratt & Whitney  
 Pacific Northwest National Lab  
 NIST  
 University of Illinois

Georgia Tech University  
 Boeing  
 Mississippi State University  
 Oak Ridge National Lab  
 Naval Research Labs  
 Johns Hopkins University  
 University of California at Davis  
 Michigan State University



As a result of the August 2017 Workshop, the Partnership launched two initial focus areas:  
 (1) Reliability in additive manufacturing, and  
 (2) Hydrogen-assisted fracture in pipeline welds

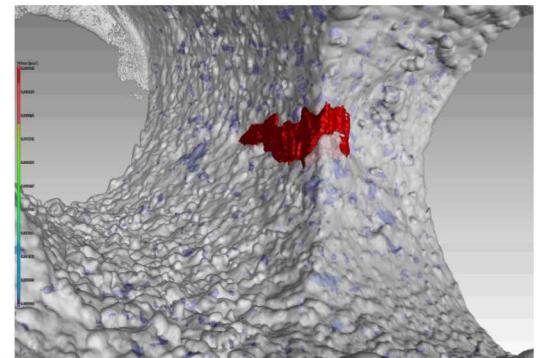
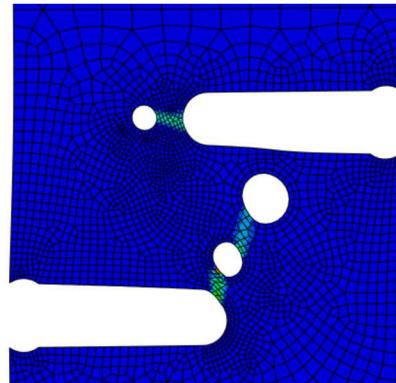
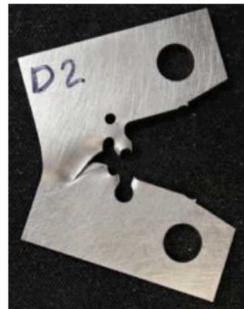
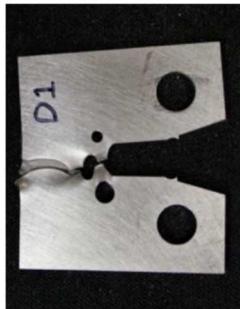
# Summary



## Sandia Fracture Challenge:

- Provides a **collaborative environment** for capability assessments
- Provides documentation of '**state-of-the-art**' in ductile failure predictions
- Illustrates **key deficiencies** in ductile failure predictions
- Raises international **awareness on the need for improved simulation capabilities**

**Structural Reliability Partnership** is a new paradigm of **cooperative research** using the “Challenge scenario” as the medium to research capabilities in experiments and computations **in structural mechanics**.



# Questions

Boyce, B.L., Kramer, S.L.B., *et al.*, "The Sandia Fracture Challenge: blind round robin predictions of ductile tearing," *International Journal of Fracture*, vol. 186, pp. 5-68, 2014.

Boyce, B.L., Kramer, S.L.B., *et al.*, "The second Sandia Fracture Challenge: predictions of ductile failure under quasi-static and moderate-rate dynamic loading," *International Journal of Fracture*, vol. 198, pp. 5-100, 2016.

**More details on SFC3 will be available in the summer in a special volume of the *International Journal of Fracture*.**

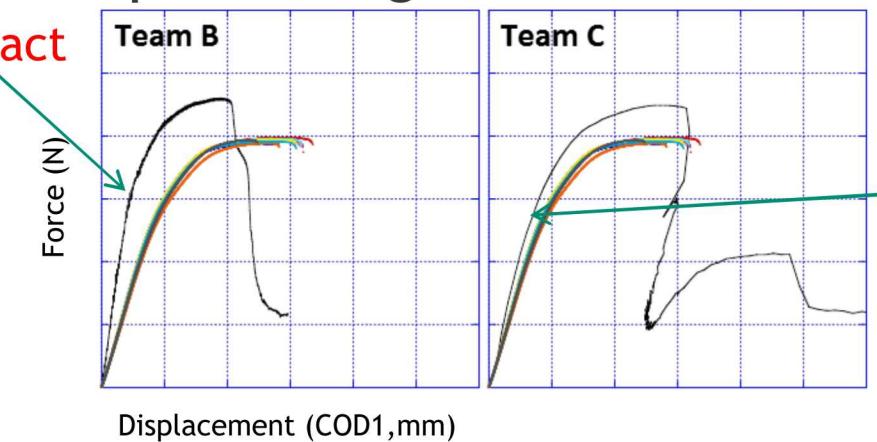
Kramer, S.L.B. *et al.*, "The third Sandia Fracture Challenge: predictions of ductile fracture in additively manufactured metal" (DOI: 10.1007/s10704-019-00361-1), (in press).

Kramer, S.L.B. *et al.*, "Evolution of Damage and Failure in an Additively Manufactured 316L SS Structure: Experimental Reinvestigation of the Third Sandia Fracture Challenge" (DOI: 10.1007/s10704-019-00357-x), Published Online March 2019.

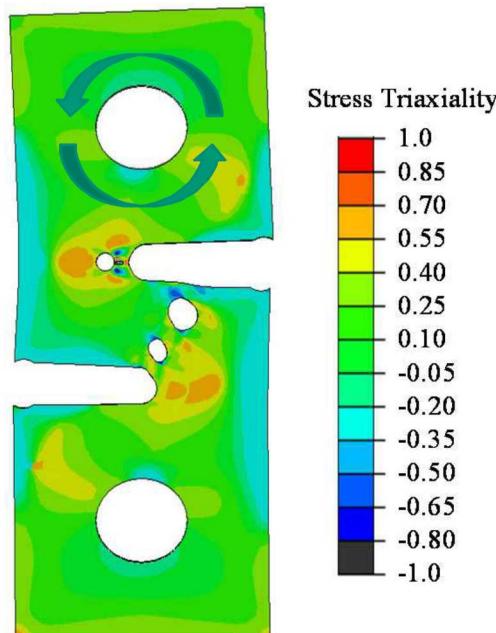
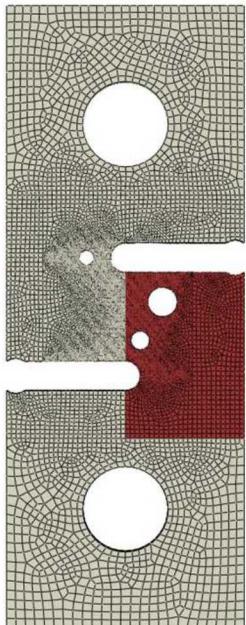


# SFC2 Gap 1: Representing surface contact and friction

Rigid, fixed pin contact



Rigid pin, friction coefficient

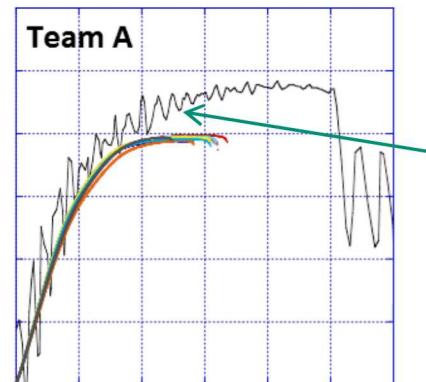
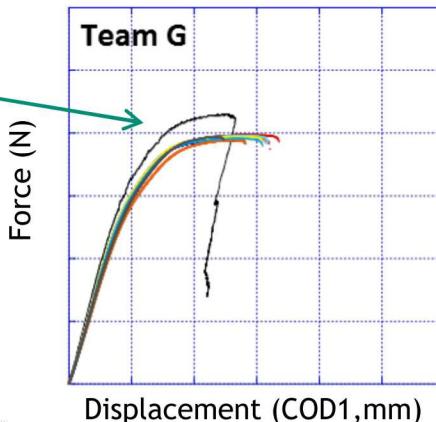


Model From Team C

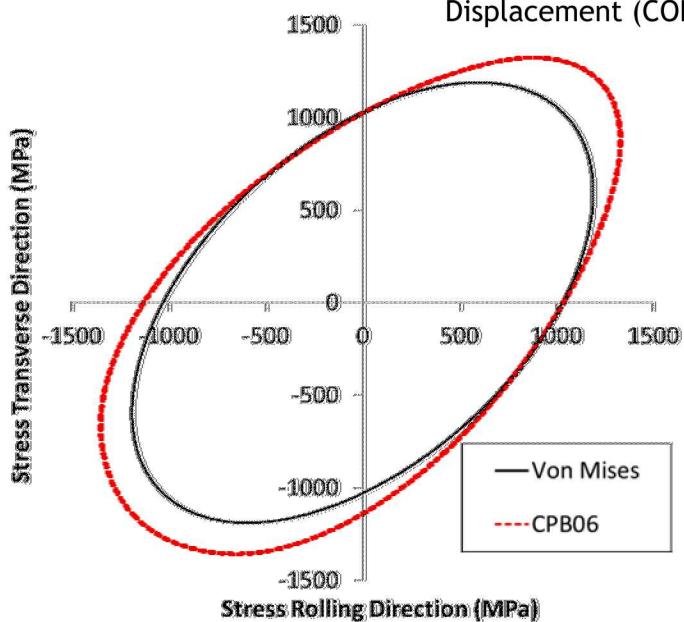
- Apparent stiffness was overpredicted by ~1/2 the teams. In some cases the predicted stiffness was 2X the experimental result!
- Teams that chose fully constrained non-sliding pin contact tended to overpredict stiffness and peak forces.
- Frictionless or free-rotating pin contact appeared to mimic experiments most closely.

# SFC2 Gap 2: Accounting for sheet anisotropy and Lode Angle dependence

No anisotropy or lode angle dependence



$J_2$  plasticity law,  
No calibration to shear data



Team B: Comparison of yield surfaces at zero plastic strain

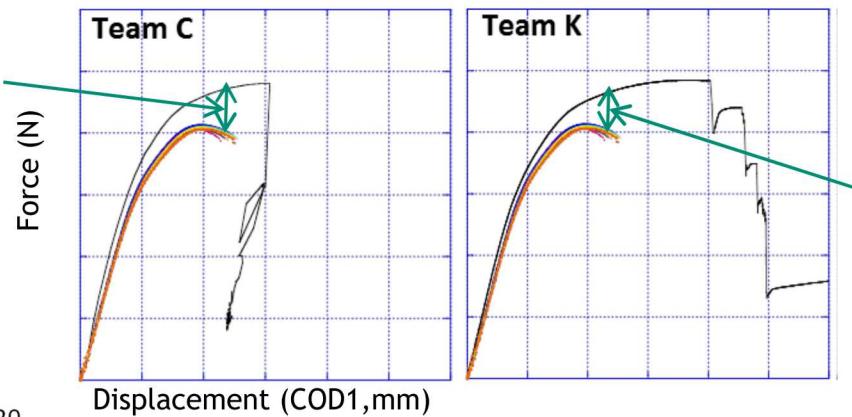
Id in shear was 0.88X the value of predicted by on Mises yield surface.

s effect was only observable by comparing the sile yield points to the yield point of the non-ndard shear test.

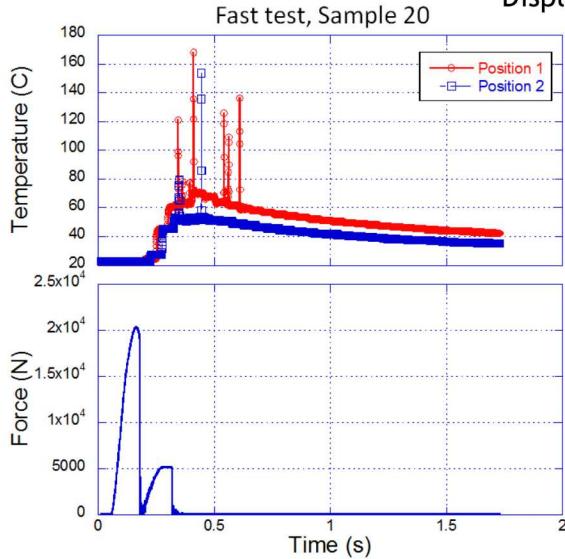
ms that used a simple  $J_2$  (von Mises) plasticity del tended to overpredict yield/hardening ravior since the yield in shear was softer than in tension.

# SFC2 Gap 3: Estimating thermal work coupling factor

No thermal coupling



No thermal coupling



- There was a  $>60^{\circ}\text{C}$  temperature rise in the necking ligament under the faster loading condition.
- Many teams ignored the plastic-work induced thermal softening that occurs under modest dynamic loading.
- Teams that chose either an adiabatic condition or some coupling parameter tended to capture some degree of extended necking behavior.

Volumetric heating rate

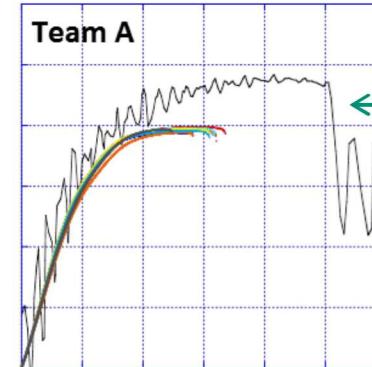
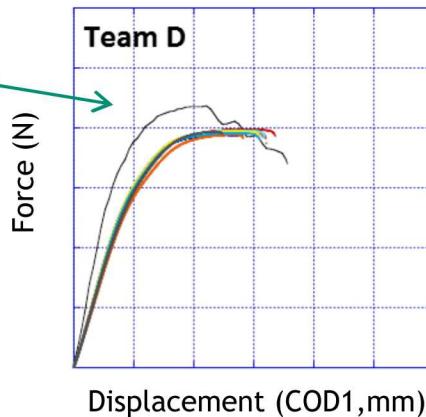
$$\dot{Q} = \eta \dot{W}^p = \eta \sigma : \dot{\varepsilon}^{in}$$

Taylor-Quinney coefficient

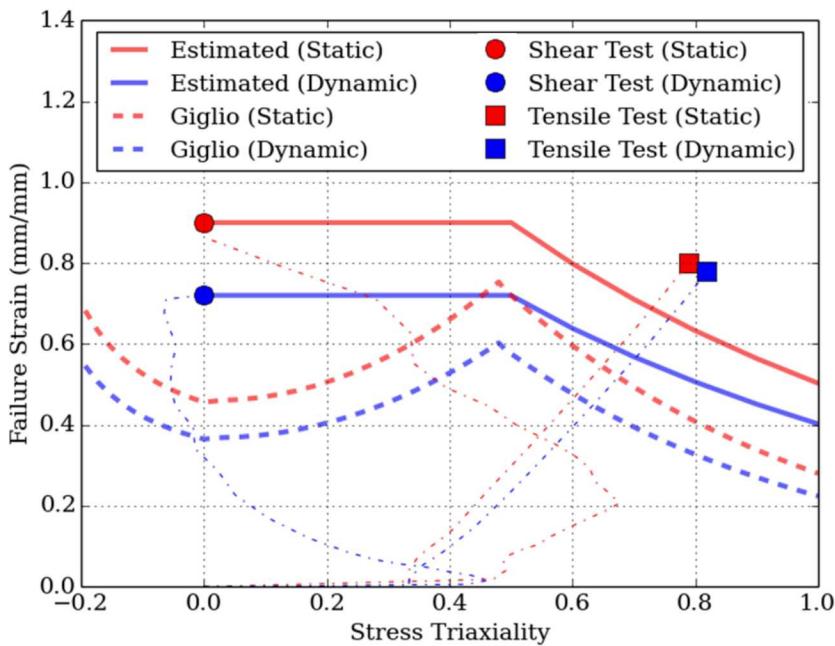
- There is little data (and even contradictory data from the same group) on the plastic work thermal coupling parameter (Taylor-Quinney coefficient).

# SFC2 Gap 4: Choosing a realistic failure parameter

Plastic strain criterion does not account for triaxiality / shear



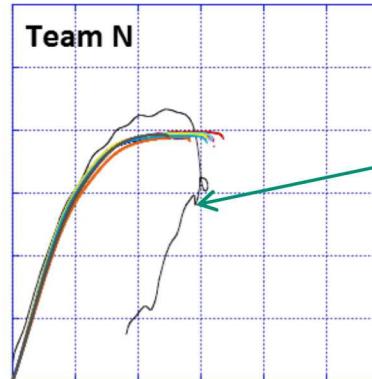
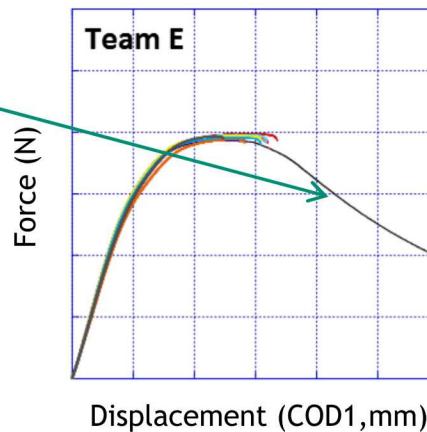
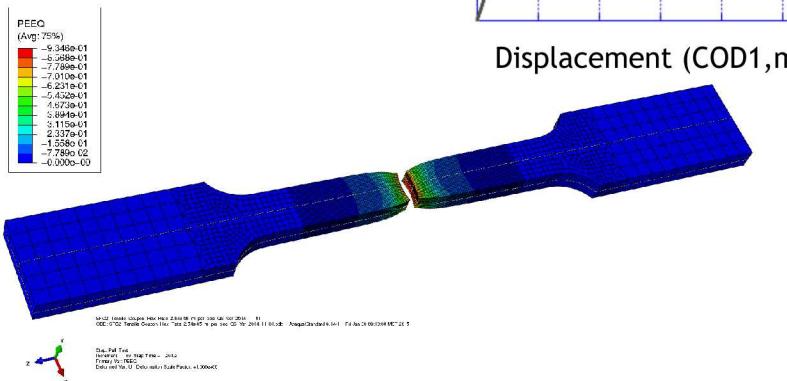
Failure parameter calibration only used tension data, not shear data.



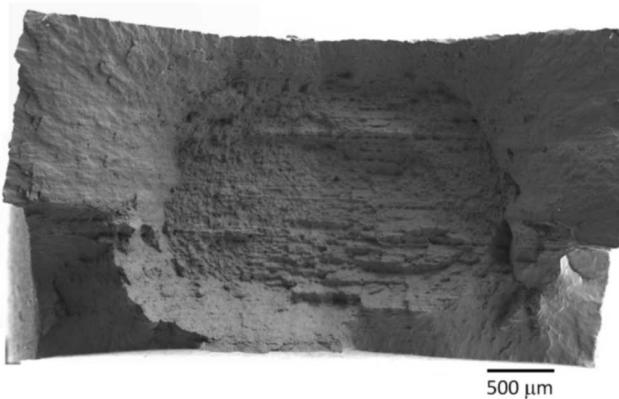
- There is no consensus on a realistic model for crack initiation. (**Gurson is not sufficient**)
- Predictions tended to be more accurate if they used shear data and calibrated a triaxiality-dependent failure model.
- While a suite of various loading paths and triaxiality conditions is needed, there are no standards for such material testing. (**A tension test is not sufficient**)

# SFC2 Gap 5: Damage progression / fracture morphology

Quasi-static cohesive zone law did not predict unstable fracture



Incorrect crack path



- Some approaches have difficulty capturing the unstable nature of crack propagation.
- Some approaches do not properly characterize the resistance of the material to crack growth (and hence crack path).
- None of the approaches accurately capture the macroscopic 3D profile of a real fracture surface, such as shear lips vs flat fracture seen in the calibration fracture surfaces.

# Challenge Questions

Questions regarding point measures and for the entire test, utilizing load and DIC measurements with uncertainty bounds.

**Question 1:** Report the force at the following displacements **D**: 0.25 mm, 0.5 mm, 0.75 mm, and 1.0 mm.

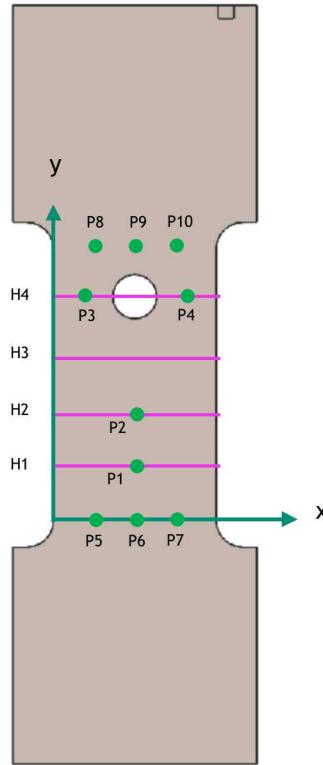
**Question 2:** Report force and Hencky (logarithmic) strain in the vertical direction ( $\epsilon_{yy}$ ) at four points, P1, P2, P3, and P4, on the surface at 75% of peak load before peak load (F1), 90% of peak load before peak load (F2), at peak load (F3), and at 90% of peak load after peak load (F4).

**Question 3:** Report the force vs. gage displacement **D** for the test.

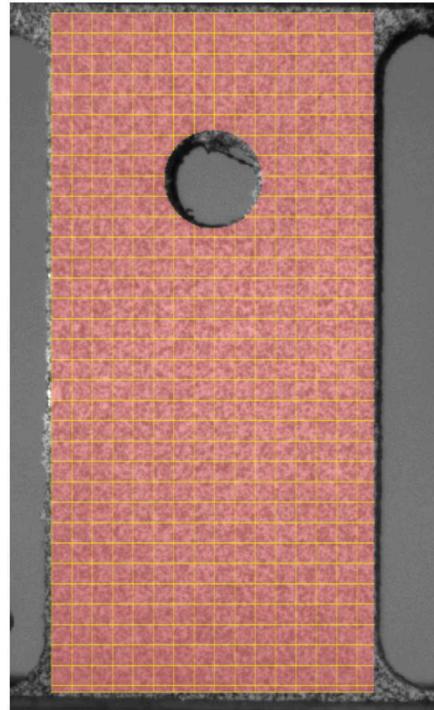
**Question 4:** Report force vs. Hencky (logarithmic) strain in the vertical direction ( $\epsilon_{yy}$ ) at four points, P1, P2, P3, and P4, on the surface for the test.

**Question 5:** Report force and Hencky (logarithmic) strain in the vertical direction ( $\epsilon_{yy}$ ) along four horizontal lines, H1, H2, H3, and H4 on the surface at forces F1, F2, F3, and F4. Line scan data should be provided with a data spacing of  $\Delta x=0.030$  mm.

**Question 6:** Provide images of the model directly viewing the front surface (same as the side for DIC) at crack initiation and at complete failure, showing contours of Hencky (logarithmic) strain.



Front View of Challenge Geometry

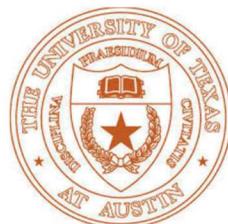
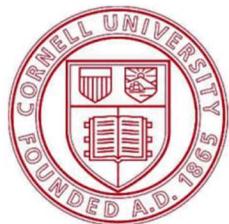


Representative DIC Area of Interest Showing a Grid of 41 pixel x 41 pixel Subsets

\*For Questions 1-4, please report nominal (average) value and optionally report the 80-percentile upper bound and 20-percentile lower bound values to compare to a population of 19 experimental observations.

# 21 Teams of Challenge Participants

## Universities



OSTBAYERISCHE  
TECHNISCHE HOCHSCHULE  
REGENSBURG



## Industry



SOUTHWEST RESEARCH INSTITUTE



## Government Laboratories



Sandia  
National  
Laboratories

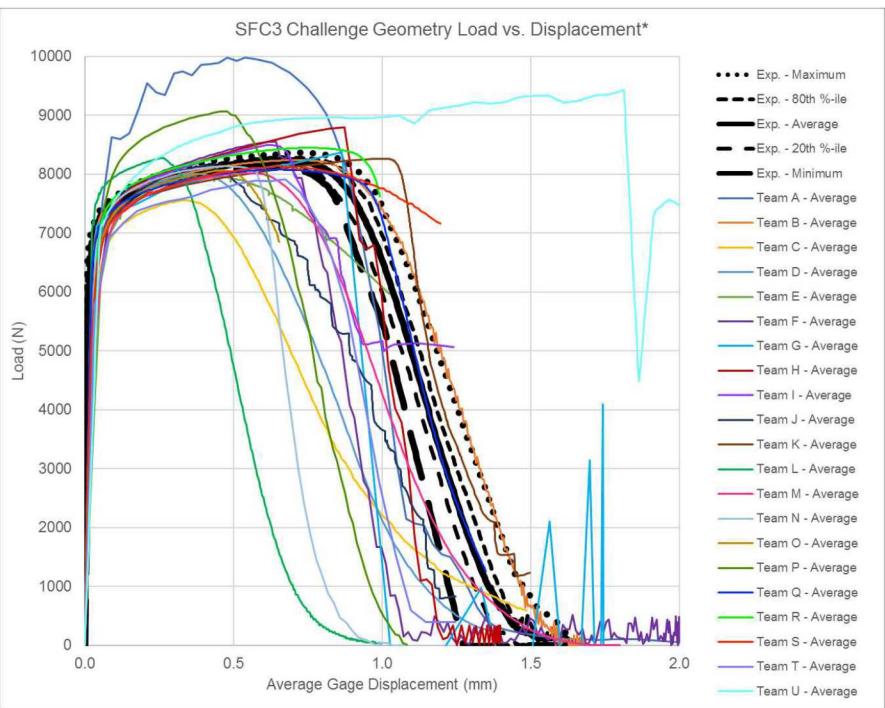


US Army Corps of Engineers

Natural Resources Canada   Ressources naturelles Canada

**CanmetÉNERGIE**  
Leadership en écoInnovation

# Prediction Methods



The 21 predictions were obtained from a variety of methods, for example:

- **Solvers:** Explicit vs. Implicit; Quasi-statics vs. Dynamics
- **Fracture Method:** Element deletion, Peridynamics with bond damage, XFEM, Damage (stiffness degradation), and Adaptive remeshing
- **Uncertainty:** Material and geometric
- **Plasticity:** J2 plasticity or Hill yield with Isotropic hardening, mixed Swift-Voce hardening, kinematic hardening, or custom hardening curves
- **Fracture Criteria:** GTN model, Hosford-Coulomb, triaxiality-dependent strain, critical fracture energy, damage-based model, critical void volume fraction, and Johnson-Cook model
- **Damage Evolution:** Damage accumulation / evolution, crack band model, fracture energy, displacement value threshold, incremental stress triaxiality, Cocks-Ashby void growth, and void nucleation / growth / coalescence
- **Calibration Data:** Various combinations of the tensile specimens, the notched tensile specimens, and literature data

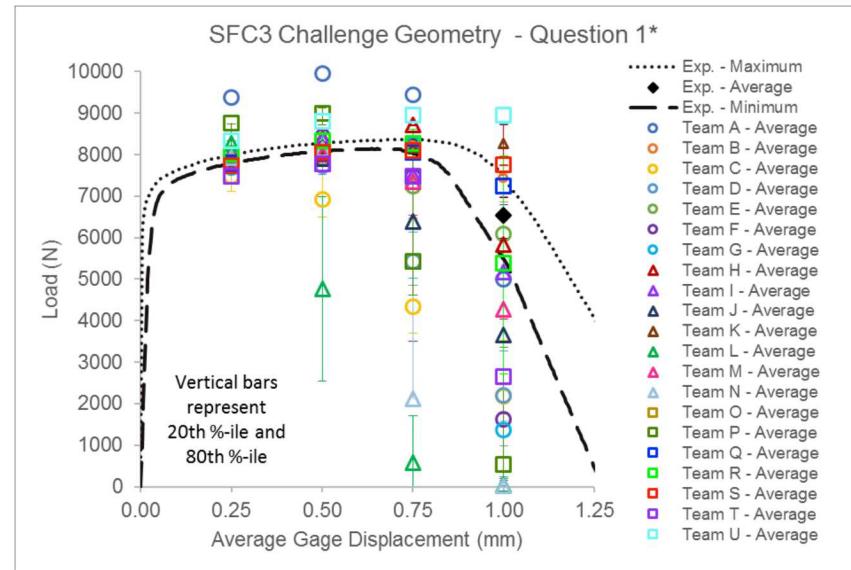
All 21 predictions correctly identified the nominal crack path with initiation at the through-thickness hole

13 teams offered uncertainty bounds on their predictions

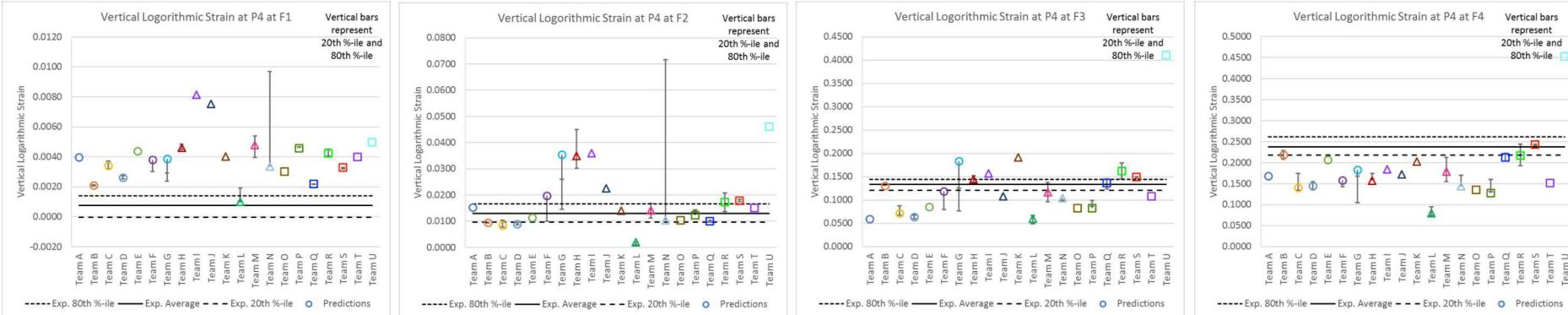
# SFC3 Predictions: Question 1 & 2

**Question 1:** Report the force at the following displacements D: 0.25 mm, 0.5 mm, 0.75 mm, and 1.0 mm.

Several teams predicted the initial structural yield, but the variation broadened with increasing displacement.



**Question 2:** Report force and Hencky (logarithmic) strain in the vertical direction ( $\epsilon_{yy}$ ) at four points, P1, P2, P3, and P4, on the surface at 75% of peak load before peak load (F1), 90% of peak load before peak load (F2), at peak load (F3), and at 90% of peak load after peak load (F4).

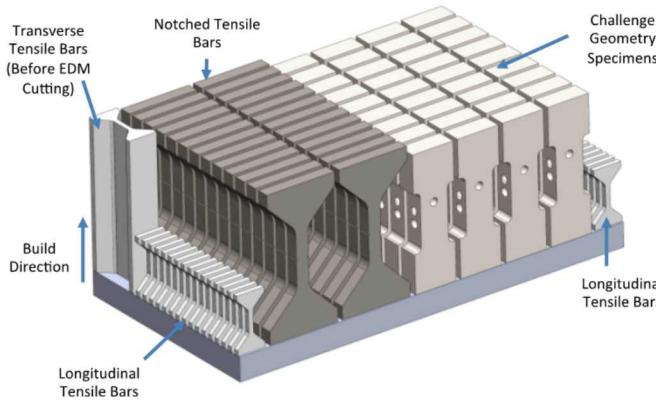


Generally, the predictions were initially too high, particularly for F1, and then were under-predicting by F4.

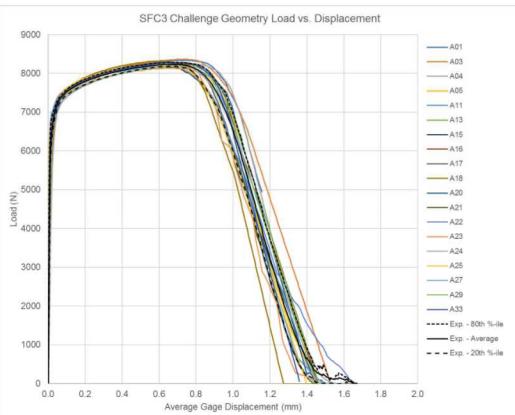
# Motivation: Understanding Failure in AM Metal Structures and The Third Sandia Fracture Challenge

The Third Sandia Fracture Challenge explores the experiments and model methods required to predict ductile failure in AM metal parts.

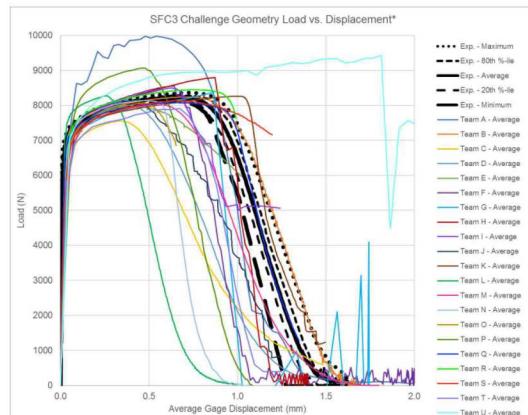
AM 316L SS Build Plate:  
Laser Powder Bed Fusion



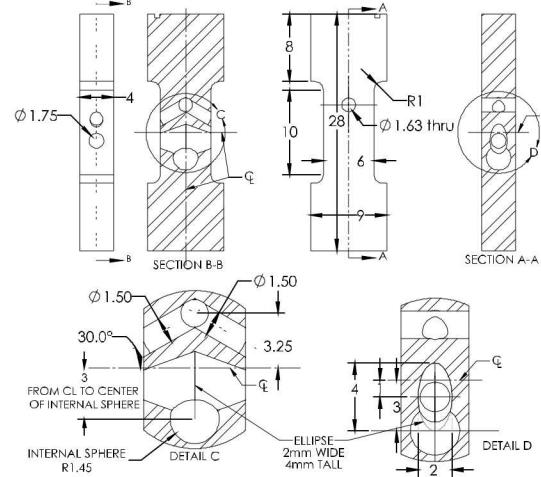
Load-Displacement Response of SFC3-Geometry Specimens



21 Nominal Predictions with  
Exp. Average and Bounds



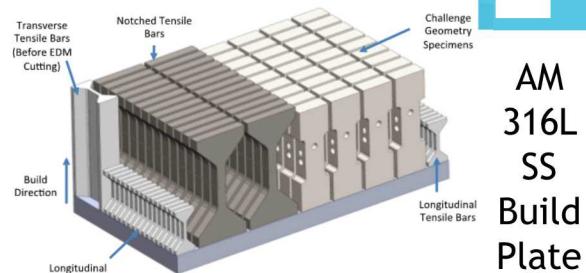
SFC3-Geometry Specimens



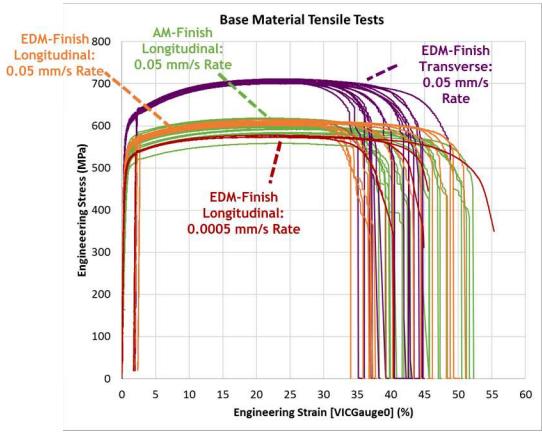
Most predictions did not consider the AM porosity, but yet did reasonably well in predicting the global response. This implies that geometry, not porosity, dominates global behavior.

# Motivation: Understanding Failure in AM Metal Structures and The Third Sandia Fracture Challenge

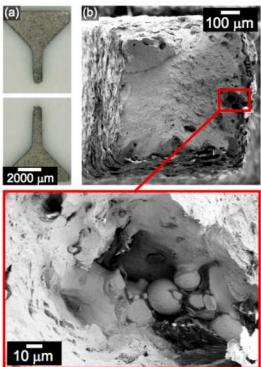
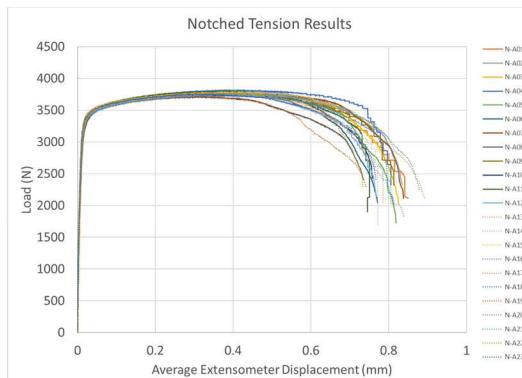
**Question: What is the effect of pre-existing voids on deformation, damage, and failure in AM metallic structures like the SFC3-geometry specimens?**



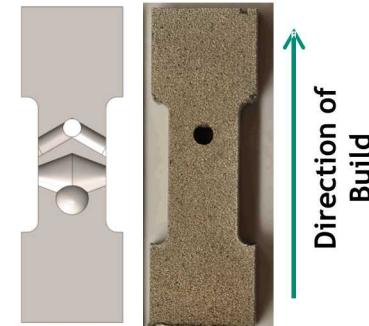
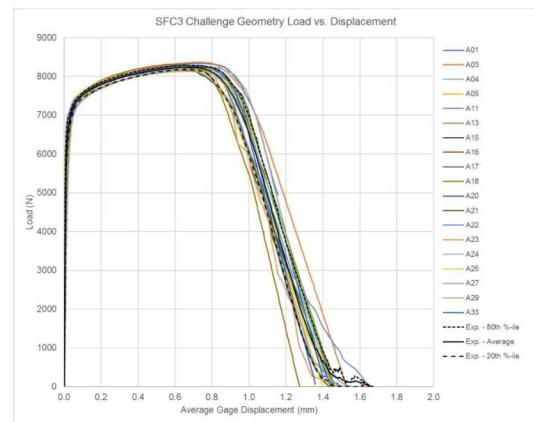
## Tensile Specimens



## Notched Tensile Specimens



## SFC3-Geometry Specimens

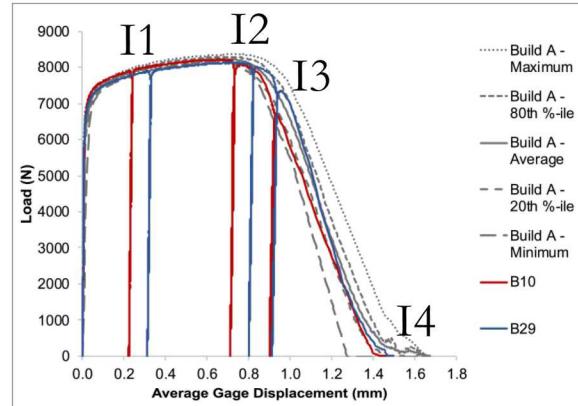


We will look at the global behavior and local crack initiation and growth relative to pre-test void population and the evolution of void growth and crack evolution.

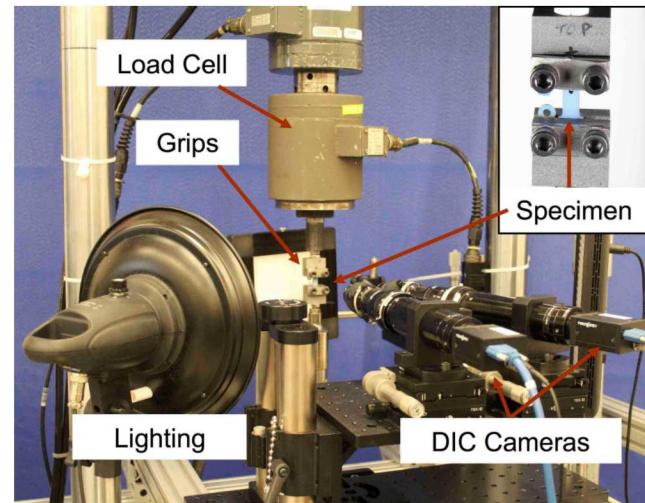
# Experimental Approach: Interrupted Testing with Micro-CT

- Tested 6 “Build B” specimens to failure to see overlap with original “Build A” specimens
- Interrupted test intervals for six specimens:
  - I1 – To middle of hardening
  - I2 – Peak load
  - I3 – Visible crack
  - I4 - Failure
- All interrupted test specimens had the same failure mode as tests to complete failure

## Interrupted Response

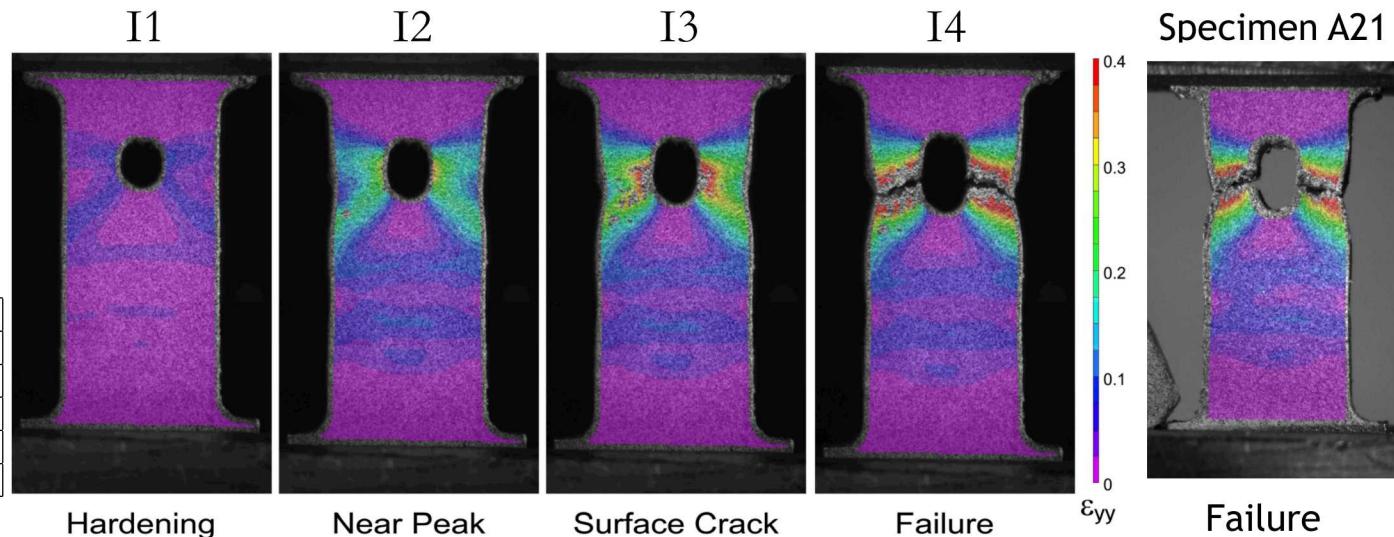


MTS 100-kN load frame with custom AM grips and Correlated Solutions Stereo DIC system (VIC3D)



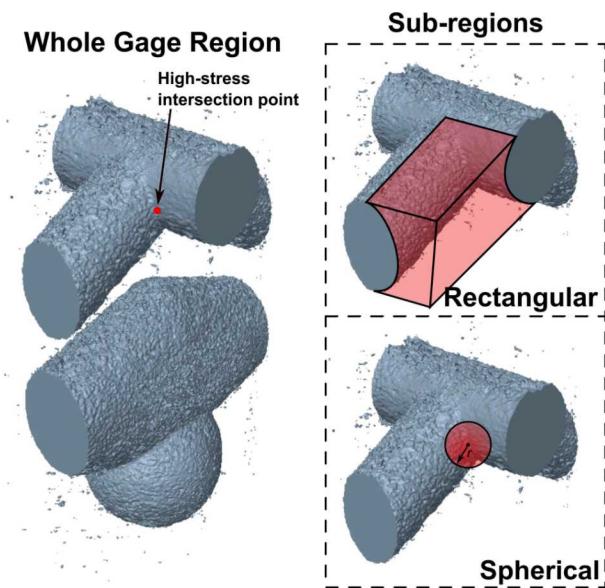
## Hencky Tensile Strain Fields for B28

Subset Size (pix)	41
Step Size (pix)	8
Strain Window (pix)	9
Virtual Strain Gage (pix)	65
Virtual Strain Gage (mm)	0.517
Pixel to Length Ratio (pix/mm)	125.8



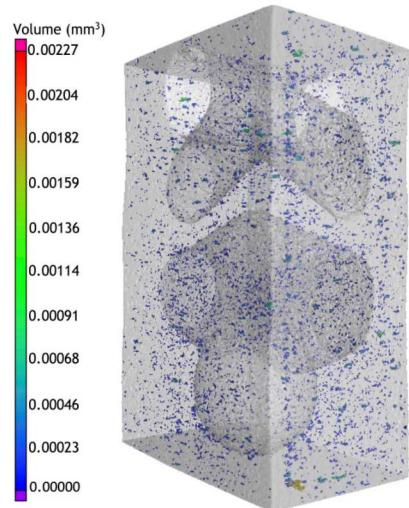
# Micro-CT and Void Analysis Methods

- X-Ray Worx 225kV tubehead with a Varian cesium iodide 2520DX detector using North Star Imaging software
- **Voxel resolution of  $6.2+/-0.6 \mu\text{m}$**
- 16-bit tiff images reconstructed with Volume Graphics 3.2 Max software
- Image processing in FIJI and MATLAB
- Void analysis performed using IDL software with a requirement of **at least 8-connected voxels to count as a void with a minimum Equivalent Spherical Diameter (ESD) of 13.9-16.9  $\mu\text{m}$**

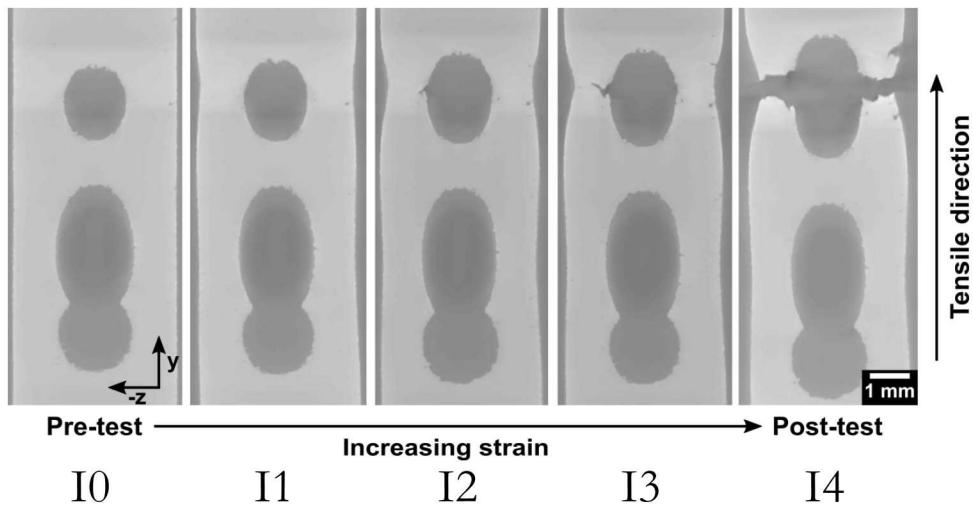


Isolation of voids with high-stress intersection point identified and different volumes for pre-test void analysis

Pre-test data showing voids



Ex situ micro-CT internal slices for Specimen B33



# Mechanical Response

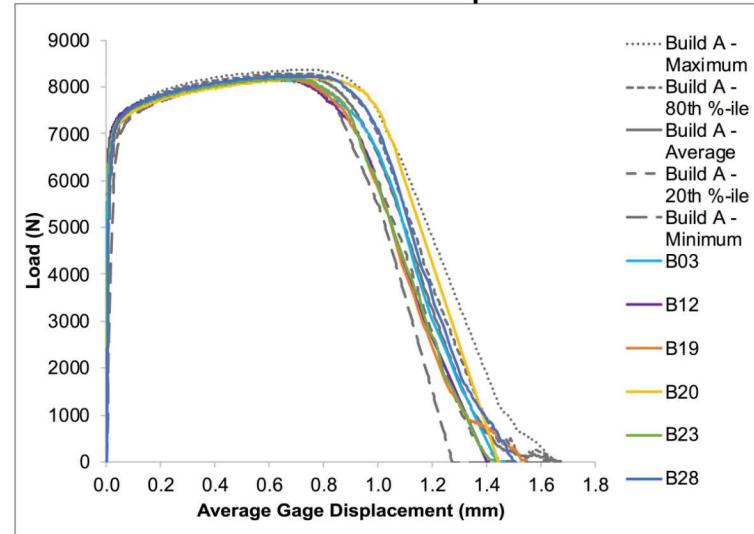
**SFC3-Geometry specimens from “Build B” behaved similarly to those from “Build A”, so analysis of “Build B” specimens is assumed represent that of all SFC3 specimens.**

Global measures considered:

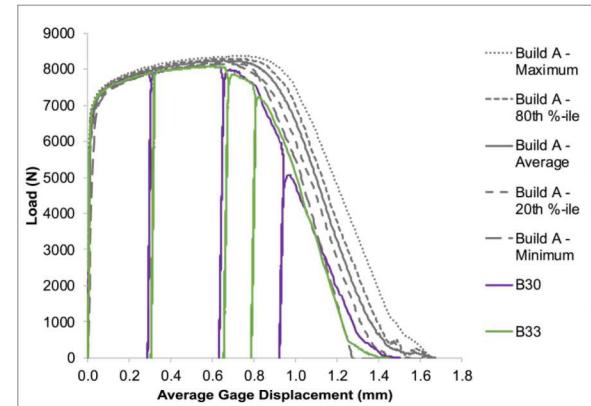
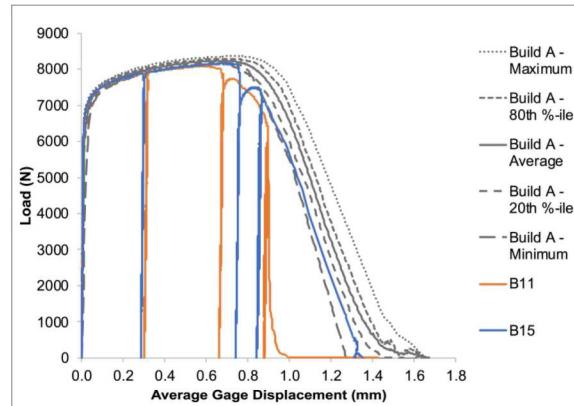
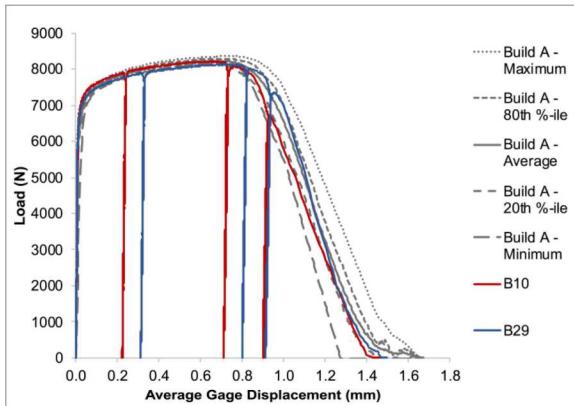
- Peak load
- Displacement at peak load
- Displacement at failure
- Maximum unloading rate

All measures for monotonic and interrupted “Build B” specimens were similar to that of the monotonic “Build A” specimens.

Monotonic Response



Interrupted Response



# Influence of Aggregate Void Metrics on Mechanical Response

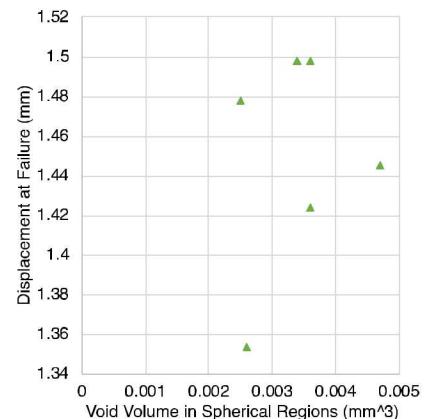
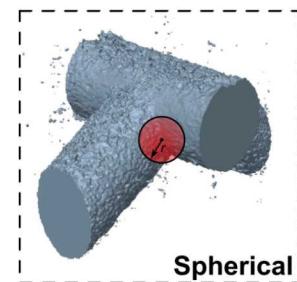
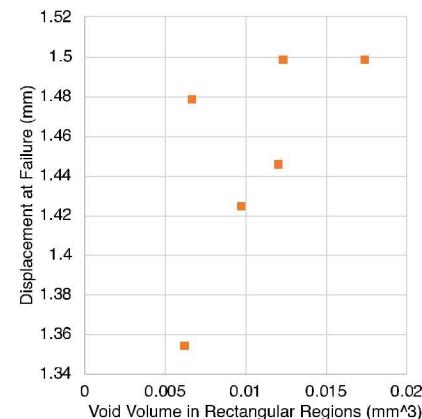
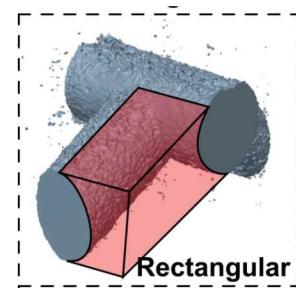
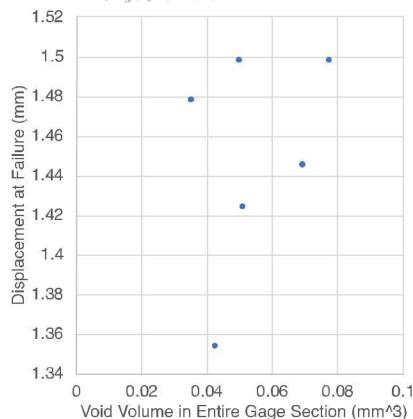
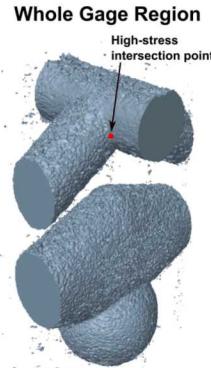
## Original Hypothesis:

Metrics of aggregate pre-test void population will correspond to mechanical behavior

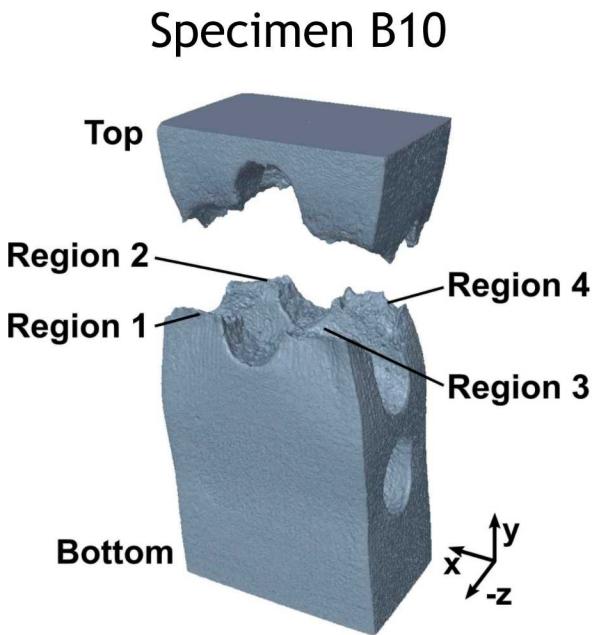
## Finding:

Metrics of pre-test void population do not strongly correspond to variations seen in mechanical performance

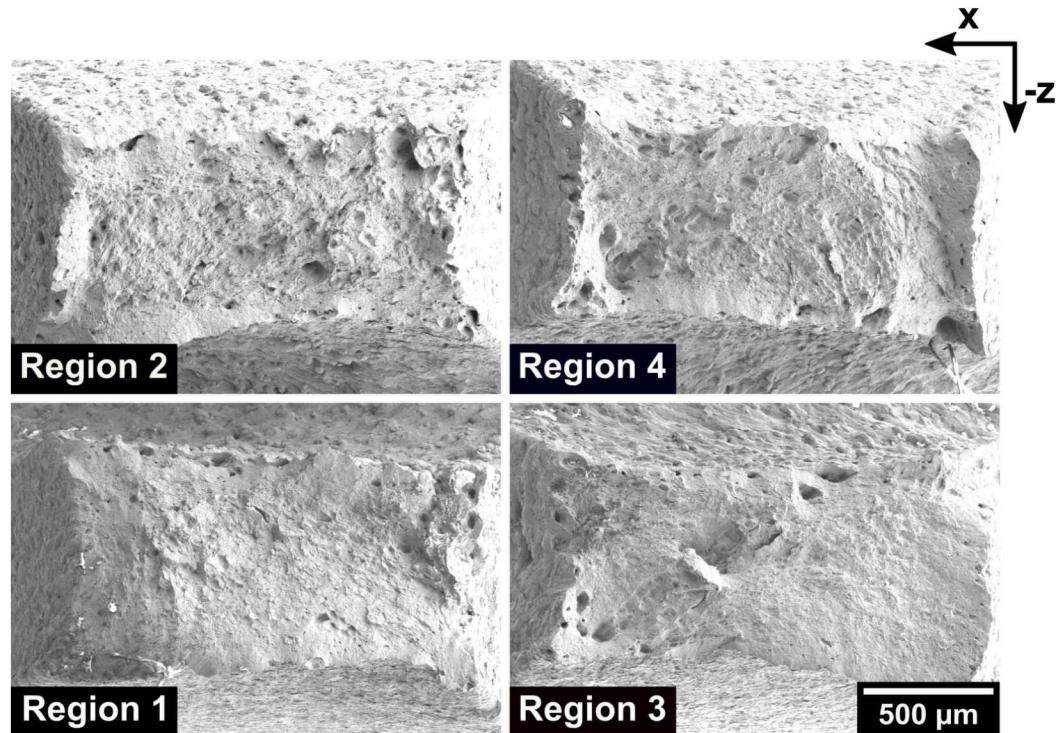
### Example: Displacement to Failure Versus Void Volume Over Different Regions



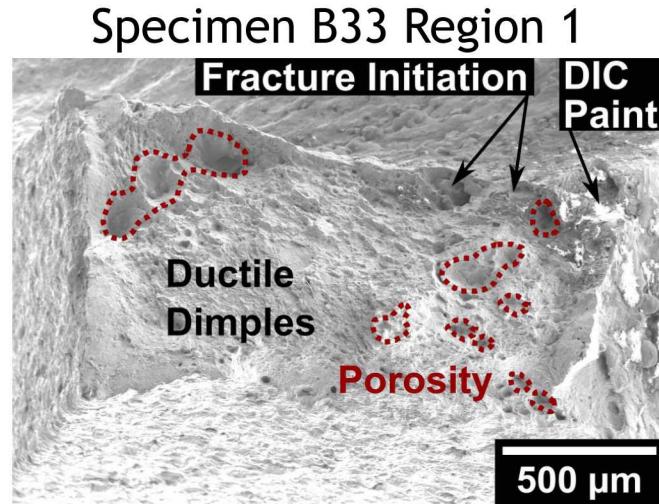
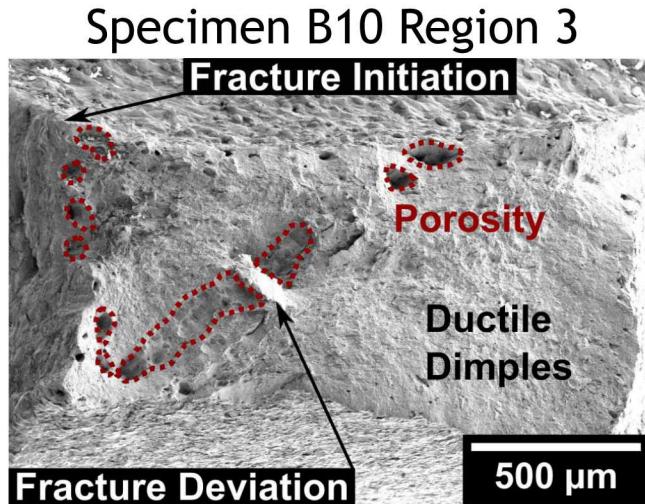
# Influence of Void Presence on Fracture



- Ductile dimples
- Intersected voids

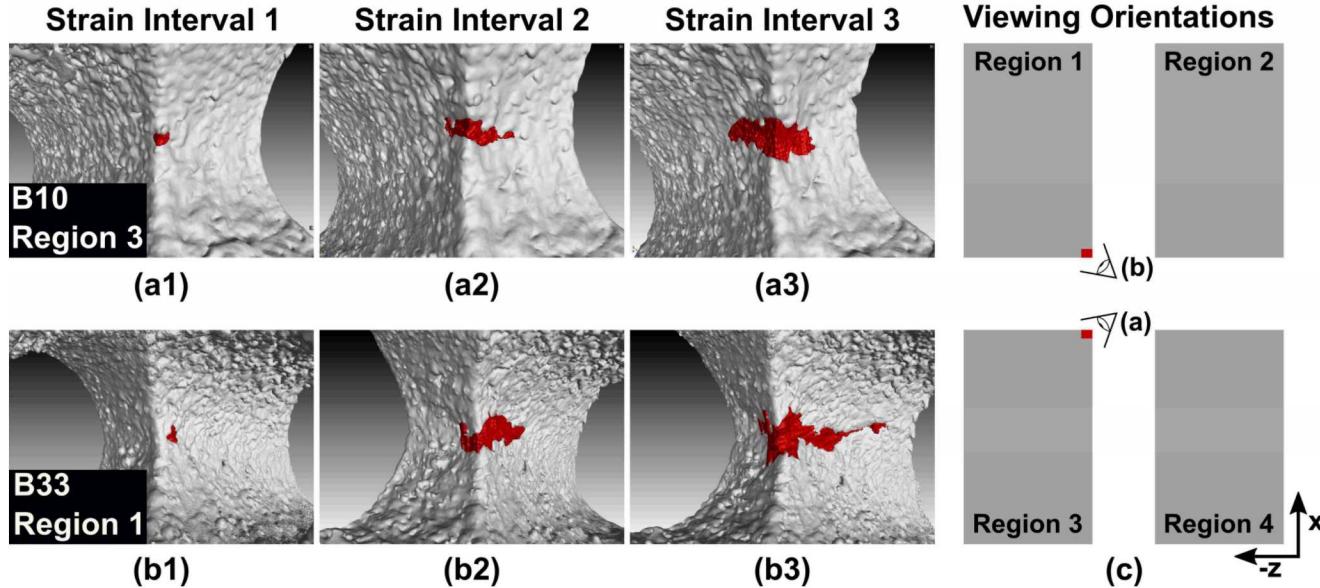


# Influence of Void Presence on Fracture



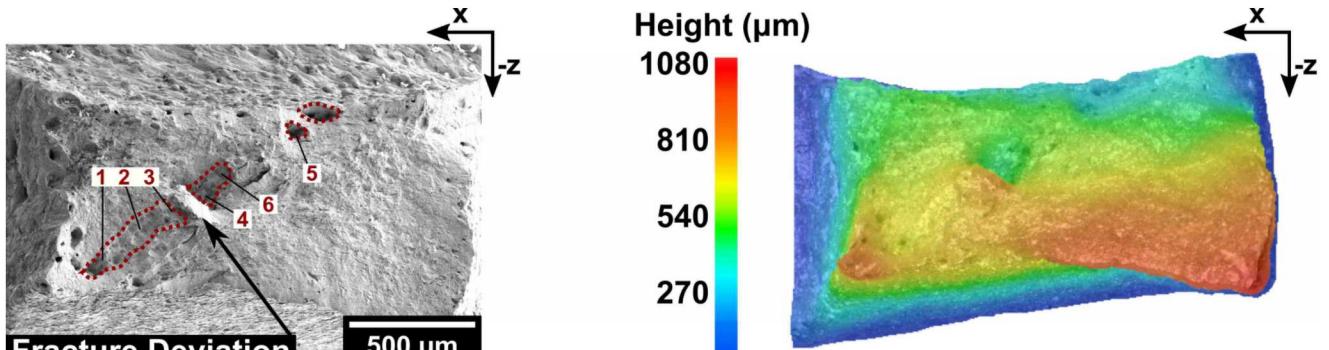
- Ductile dimples
- Intersected voids
- Fracture deviation
- Different crack initiation locations (Surface defect or geometric intersection point)

## 3D Reconstructions Highlighting Crack Volume



# Void Evolution Under Increasing Plastic Strain

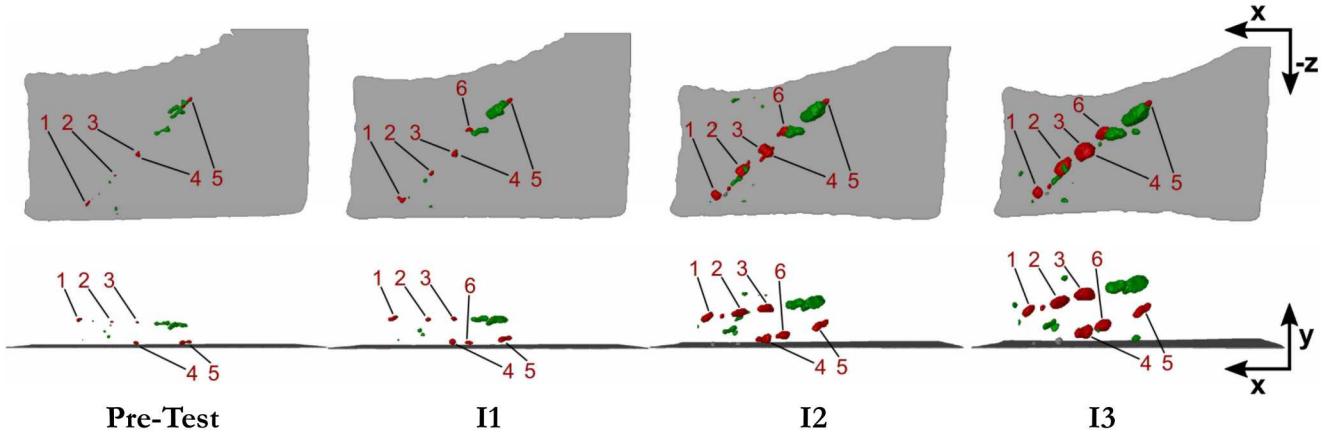
Specimen B10  
Region 3



Fracture Surface with Voids Highlighted

Fracture Surface Height Map

Void Evolution:  
3D Reconstruction  
of Voids Near the  
Fracture Surface (Red  
Voids Intersected  
Fracture Surface)



- All voids grew regardless of their involvement with the fracture surface.
- New voids (or voids too small to be resolved in pre-test scan) such as void 6 **appeared due to deformation** and grew larger than many voids observed in pre-test.
- The fracture surface did not intersect **some of the largest pre-test voids** in this region (see green).
- During I4, the crack deviated from the plane of voids 1-3 down to voids 4-6 (or from void 4-6 up to voids 1-3), avoiding the large green voids nearby.

# Influence of Local Porosity on Fracture Initiation

Local porosity can change the fracture initiation location and timing.

- **Surface Defect:** Depressions with depths >50  $\mu\text{m}$
- **Surface Roughness:** Smooth depressions with depths between 11-33  $\mu\text{m}$

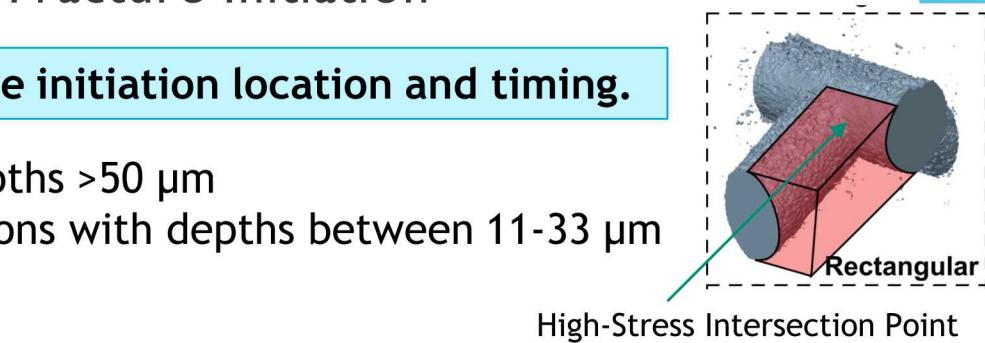
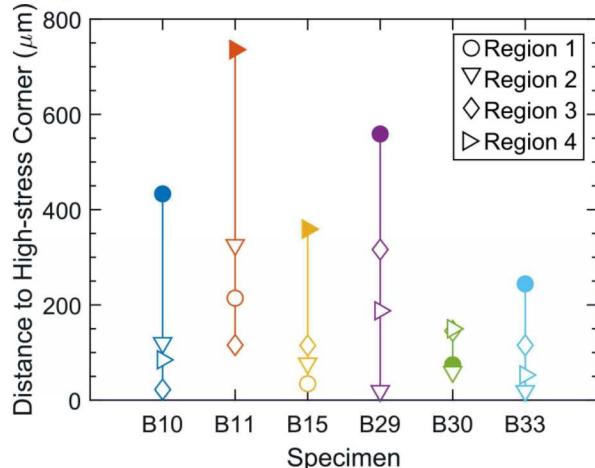
## Loading Interval of Fracture Initiation

Specimen	Region 1	Region 2	Region 3	Region 4
B10	1*	2	1	2
B11	2	2	1	1*
B15	2	3	2	1*
B29	1*	1	3	2
B30	1*	2	3	3
B33	1*	1	2	2

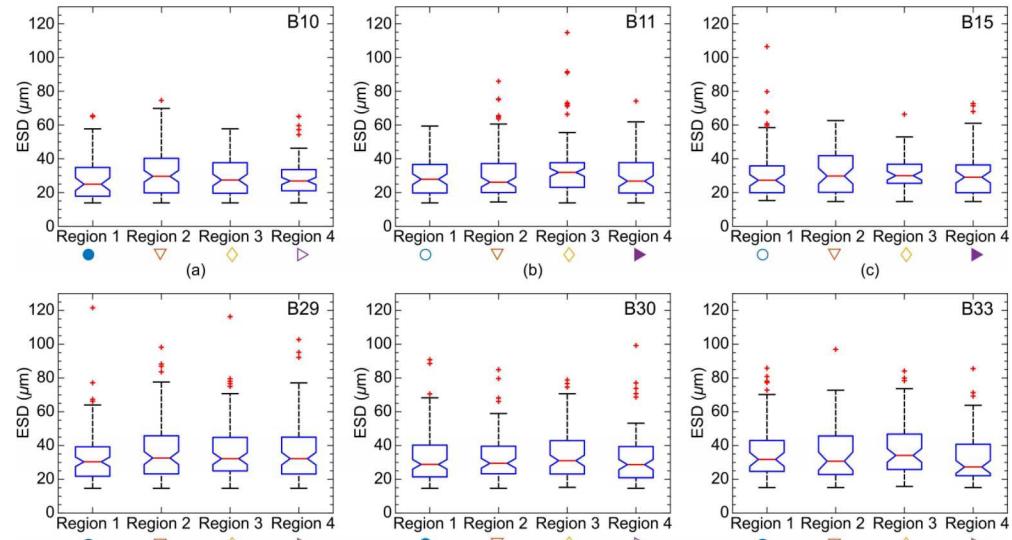
\* denotes fracture initiation at a surface defect

## Distance of Fracture Initiation Site to High-Stress Intersection Point

(Filled symbol denotes initiation at a surface defect)



Box and Whisker Plots of Equivalent Spherical Diameter (ESD) of Pre-Test Void Population in Rectangular Sub Regions and Table of Pre-test Size of Surface Defects Initiating Fracture

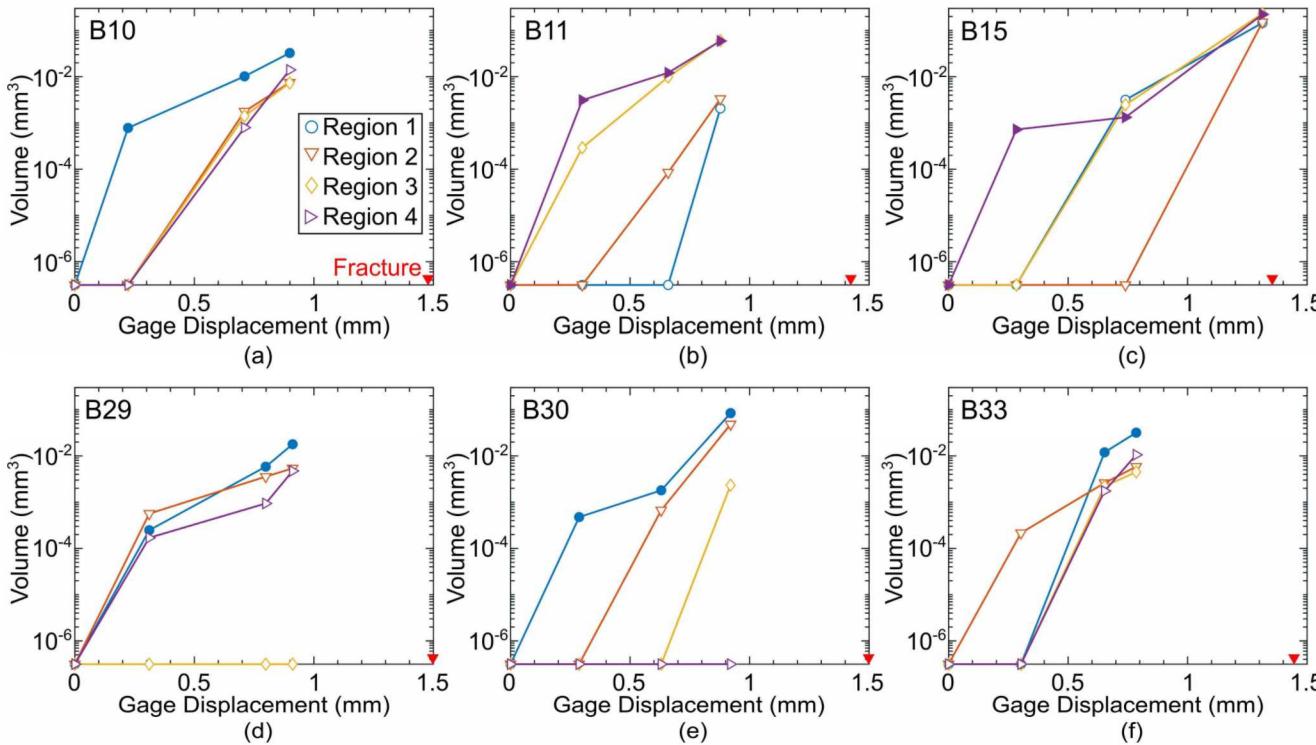


Specimen	B10	B11	B15	B29	B30	B33
Surface Defect ESD ( $\mu\text{m}$ )	102	169	80	57	73	113
Region	1	4	4	1	1	1

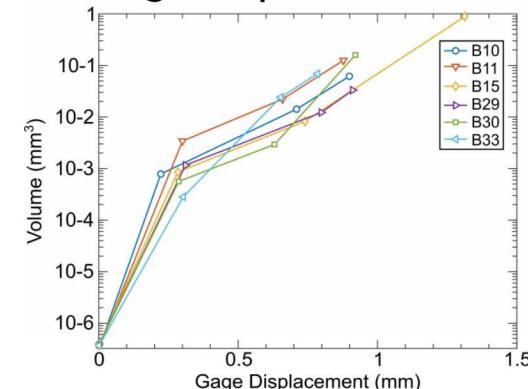
# Influence of Local Porosity on Fracture Growth

Cracks that initiated at a surface defect tend to grow faster than those that initiated at the high-stress intersection point.

Unloaded Crack Volume After Each Interval for Each Region  
Versus Unloaded Gage Displacement  
(Filled symbol denotes initiation at a surface defect)



Total Unloaded Crack Volume Versus Unloaded Gage Displacement



Despite variation in crack volume evolution between regions, agglomerate volume evolution does not greatly vary between specimens, much like mechanical response.

# Future Work

## Goal:

Deconvolve influence of several variables including void size, void location, void population, surface roughness, and geometric features on overall part performance.

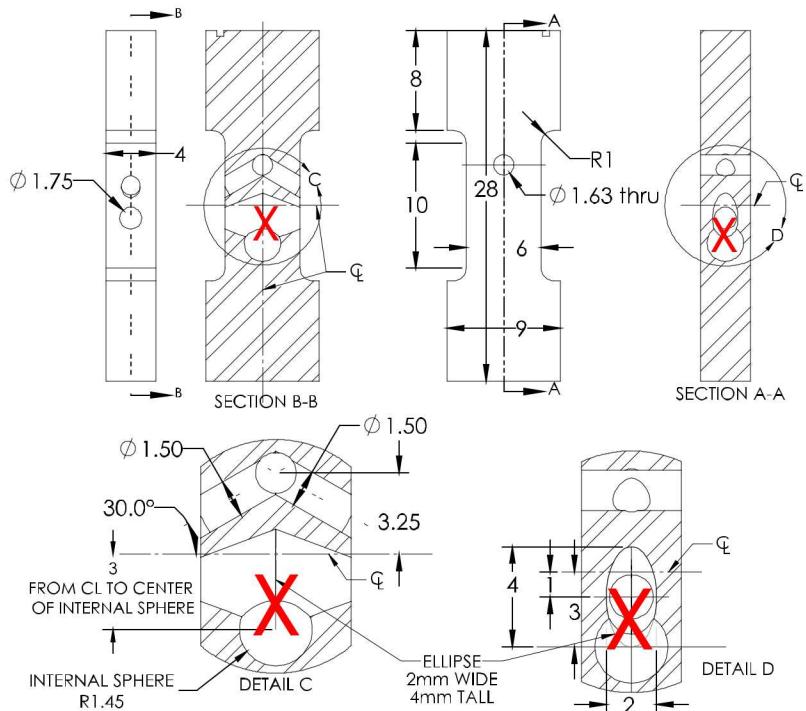
### Various SFC3 Cases to Experimentally Study:

- Case 1: AM-built structure with only the through-hole and angled channel features;
- Case 2: AM-built tensile bar with surface roughness removed and the through-hole and angled channel features machined into the part;
- Case 3: a wrought-metal tensile bar with the through-hole and angled channel features machined into the part;
- Case 4: Case 1 that has undergone Hot Isostatic Pressing (HIP); and
- Case 5: Case 2 that has undergone HIP.

### Additional Cases:

- Different geometric feature sizes relative to void sizes;
- Geometries with only one or two feature; and
- Many more!

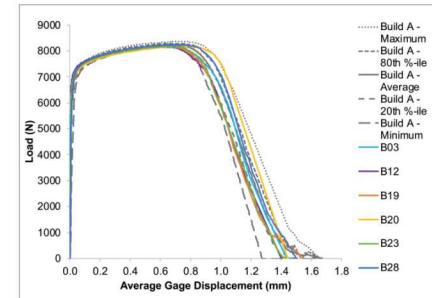
### SFC3 Geometry Denoted Features To Be Removed



## Conclusions

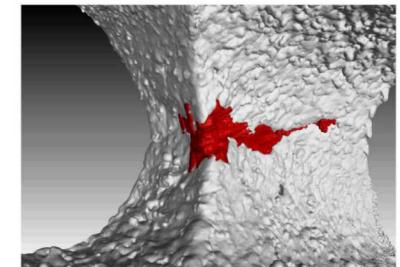
### Geometry Dominates Global Behavior in SFC3 Specimens:

Metrics of the pre-existing void population do not correlate with the global mechanical behavior of the SFC3 specimens, but rather the large stress concentrations from the geometry overwhelmingly dominate the global behavior.



### Voids Influence Local Crack Initiation and Growth:

Voids and surface defects influence local crack initiation and growth by introducing variation in crack initiation site in some cases and deviation from initial crack path to intersect voids.



### Open Question: When Do Voids or Geometry Dominate?

Future work is required to deconvolve influence of several variables including void size, void location, void population, surface roughness, and geometric features on overall part performance.

