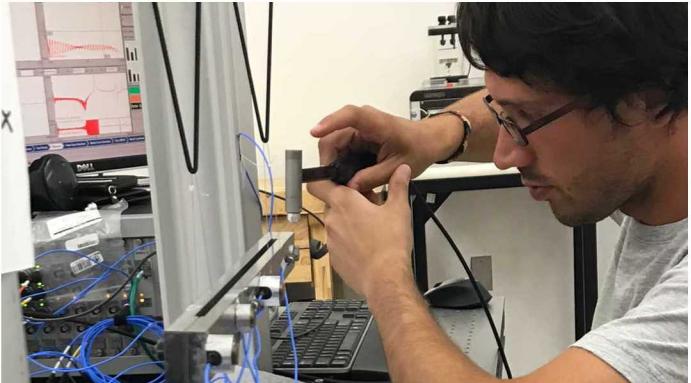
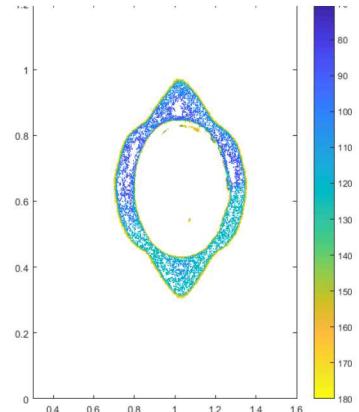
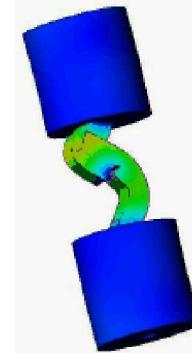


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A Priori Methods to Assess the Strength of Nonlinearities for Design Applications

Craig Broadman, She'ifa Punla-Green, Edward Rojas

Agenda

1. Introduction
2. Project Overview
3. Numerical Methodology
4. Experimental Methodology
5. Characterization of Strength of Nonlinearity
6. Conclusion



Research Team

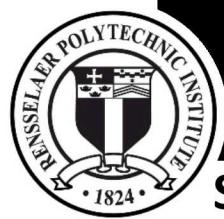
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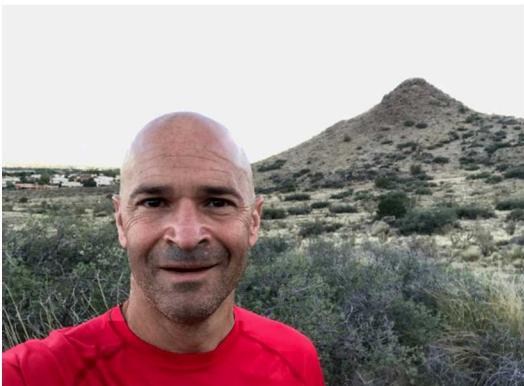
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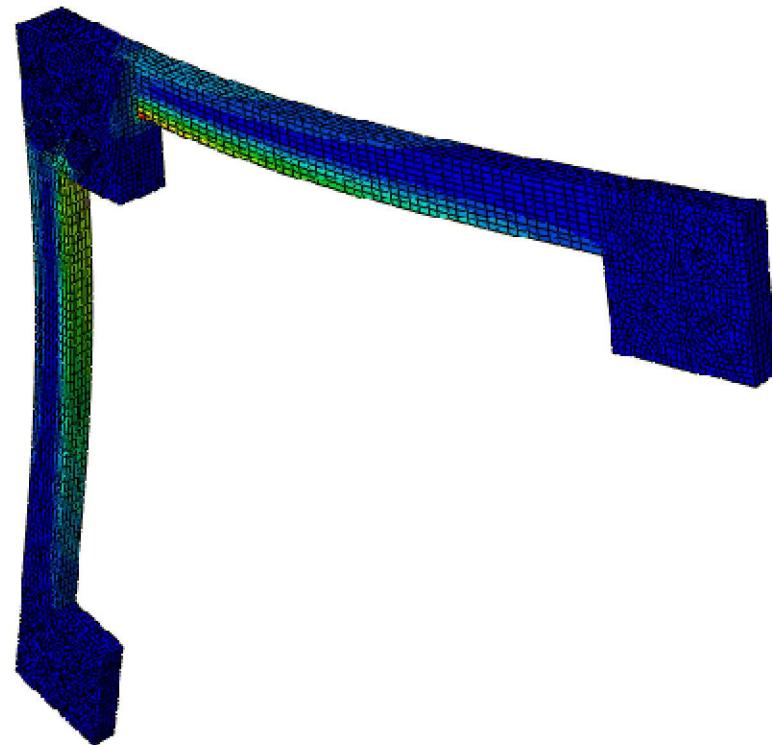


Project Overview

- It is hypothesized that the strength of a nonlinearity (SNL) in a jointed system can be predicted by quantifying the **magnitude and uniformity of contact pressure** within an interface and by assessing the **modal excitation of an interface**.

- Numerical Methods:

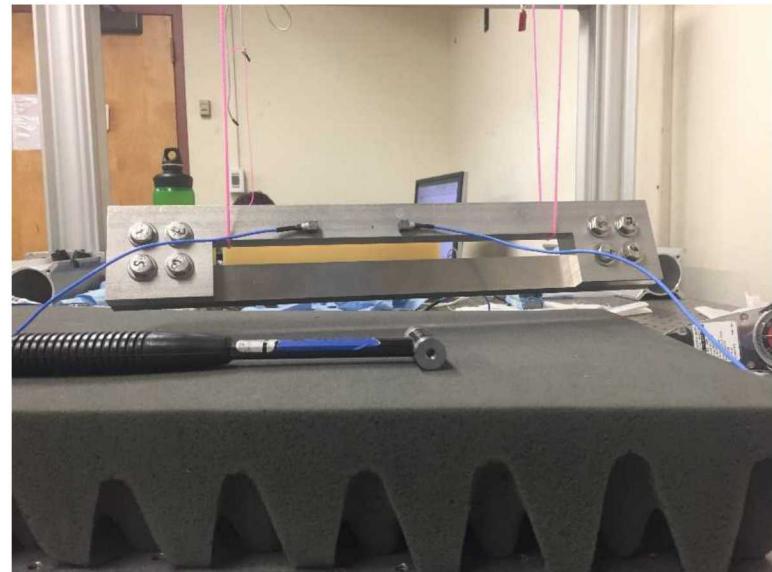
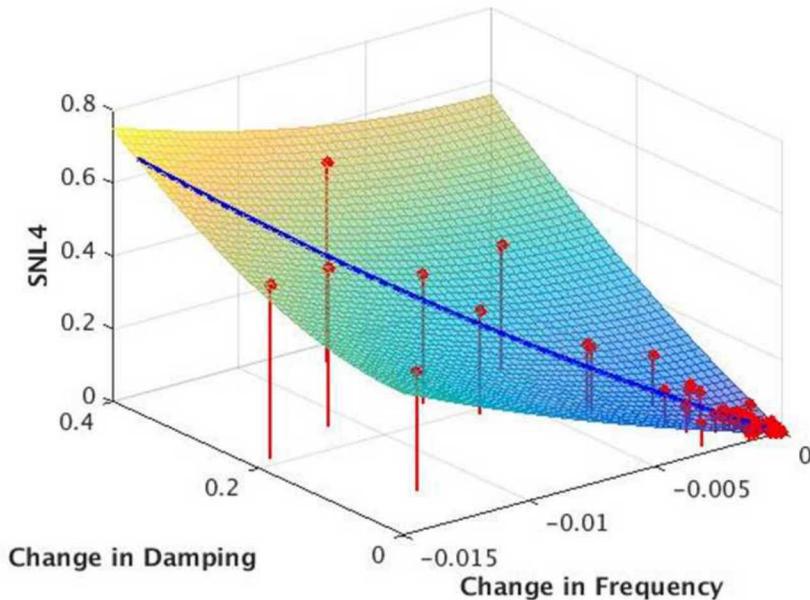
Using Abaqus, we calculated a variety of statistics regarding contact pressure and modal strain to utilize in developing a metric to predict strength of nonlinearity.



Project Overview

- Experimental Methods:

We obtained time response data for many beam configurations.

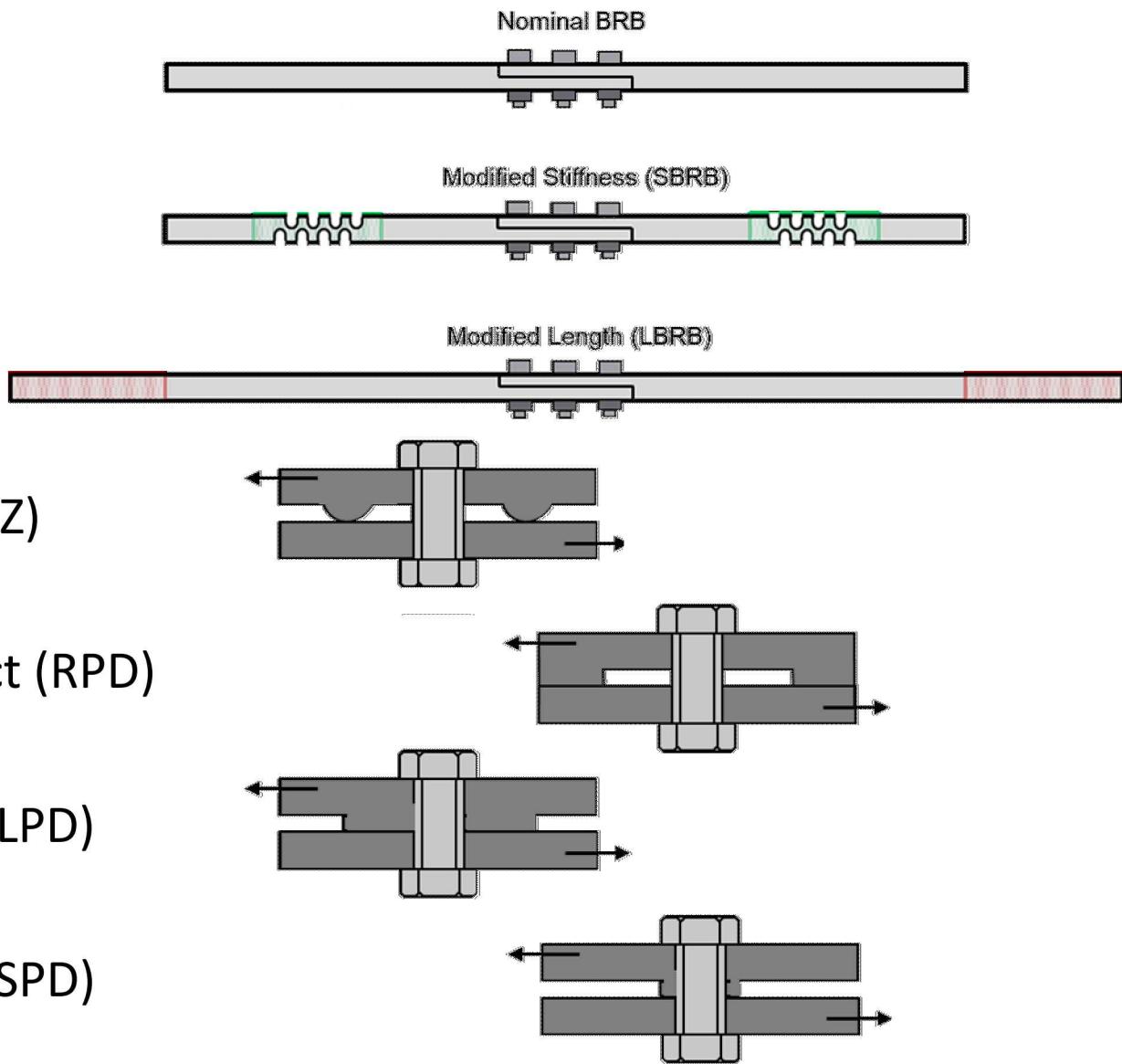


- Analysis:

We developed a definition for SNL based on change in damping ratio and change in frequency. Using machine learning, we assessed the importance of various statistics in predicting SNL and finalized a metric.

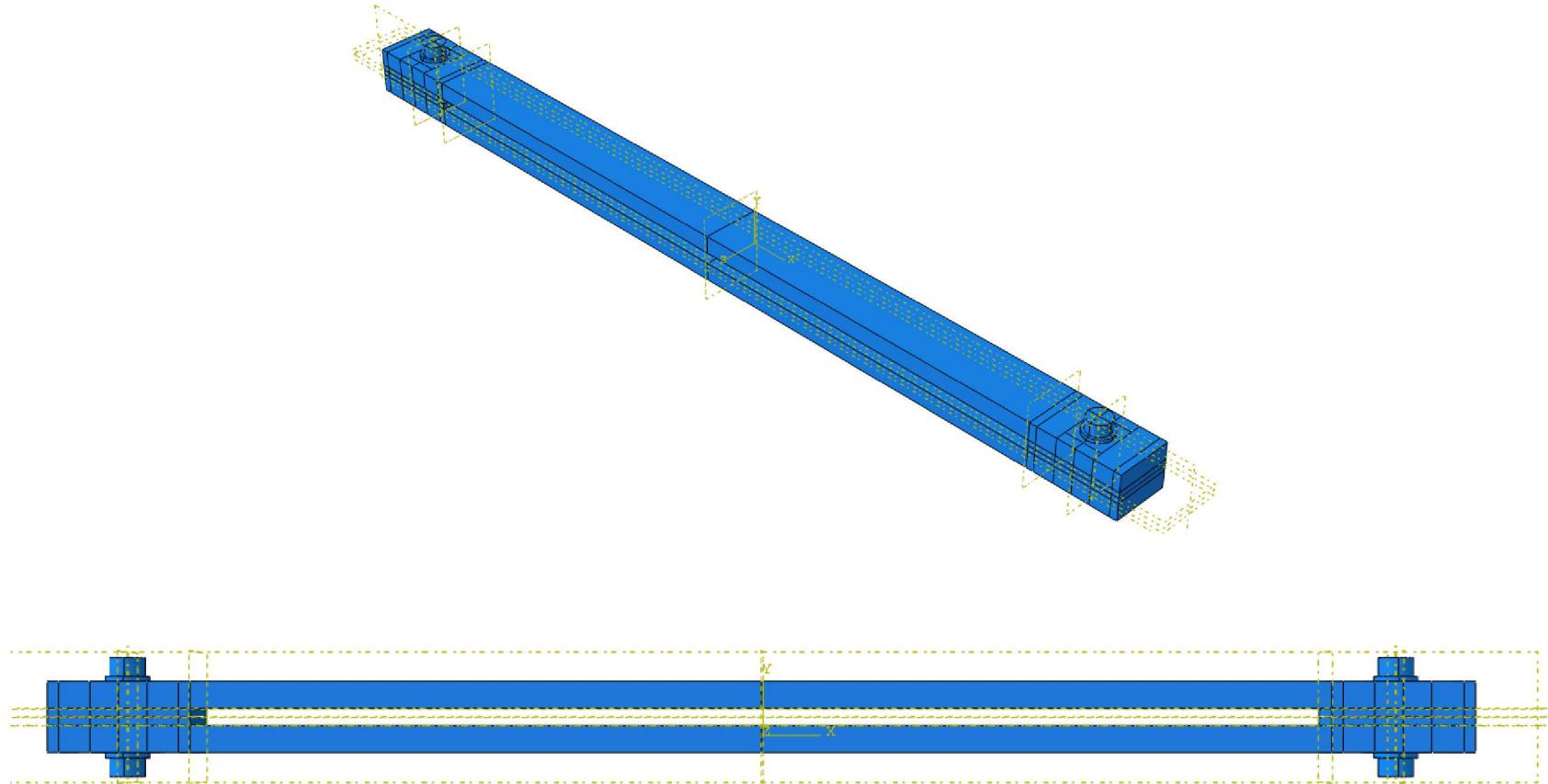
Configurations – Brake-Reuß Beam

- BRB
- Spring (SBRB)
- Long (LBRB)
- Hertzian Contact (HZ)
- Reverse Pad Contact (RPD)
- Large Pad Contact (LPD)
- Small Pad Contact (SPD)



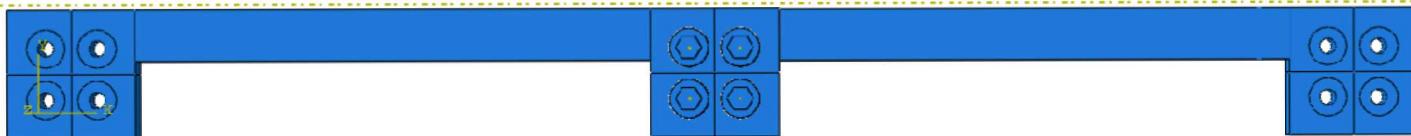
Configurations – C-Beam

- CBM (also known as the S4 or Sandwich beam)



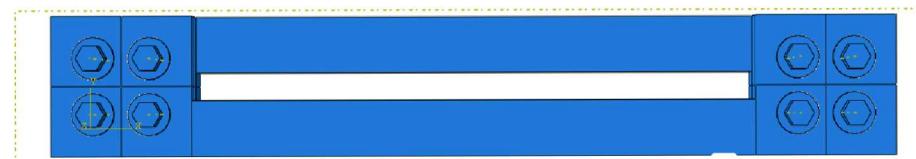
Configurations – 4-Bolt Beam

- 4LS



(4-bolt Long Same-side)

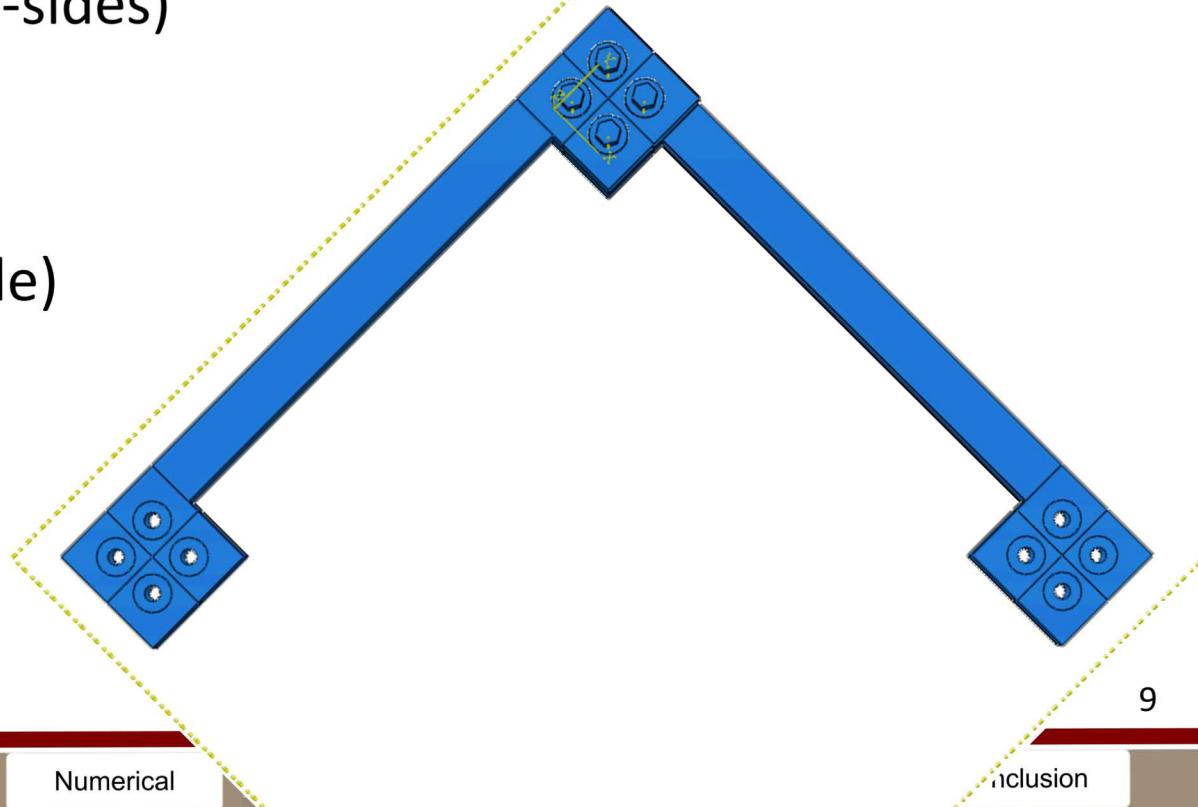
- 4SO



(4-bolt Short Opposite-sides)

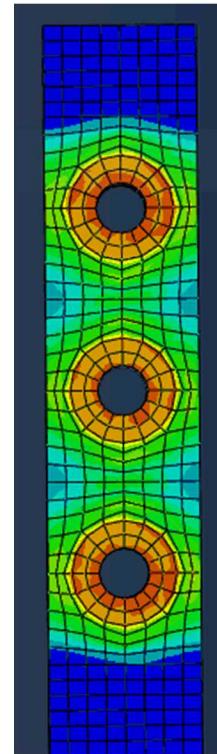
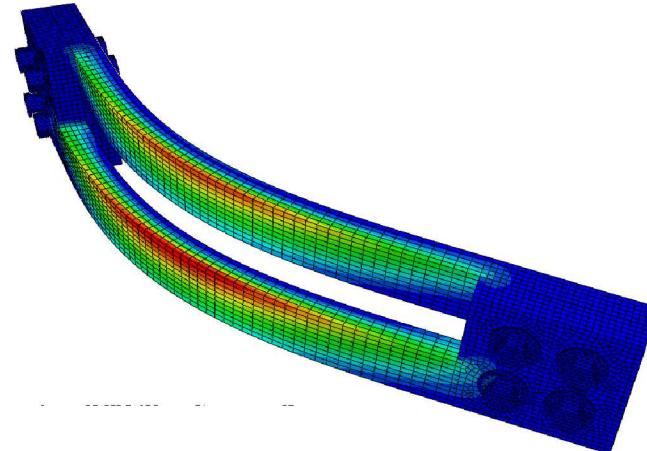
- 4VO

(4-bolt V-shape Outside)



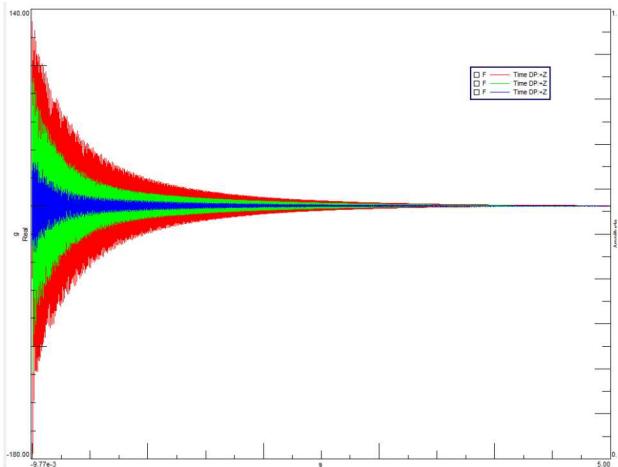
Numerical Methodology

- To develop the metric, we sought easy-to-access data from an FEA model: modal strain and contact pressure.
- A nonlinear frictionless interface implicit solver was used to determine contact pressure.
- A linearized eigen analysis was used to find mode shapes and strain.
- Various statistics were calculated based on the data:
 - Mean (Contact Pressure and Strain)
 - Max (Contact Pressure and Strain)
 - Standard Deviation (Contact Pressure and Strain)
 - Skew (Contact Pressure and Strain)
 - Kurtosis (Contact Pressure and Strain)
 - Contact Area

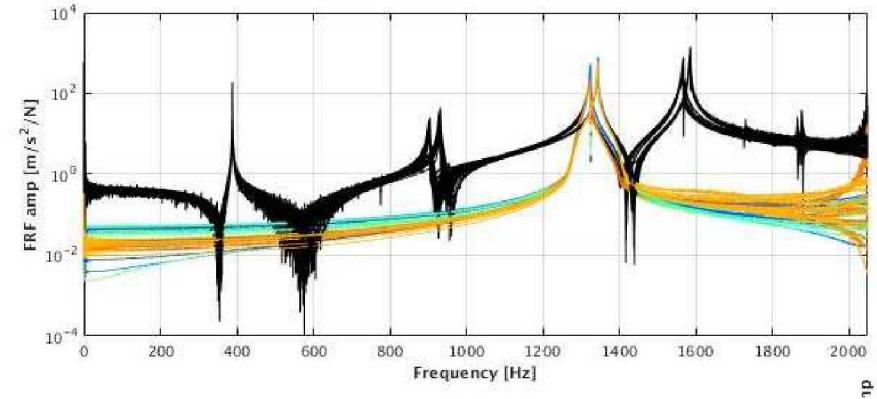


Experimental Methodology

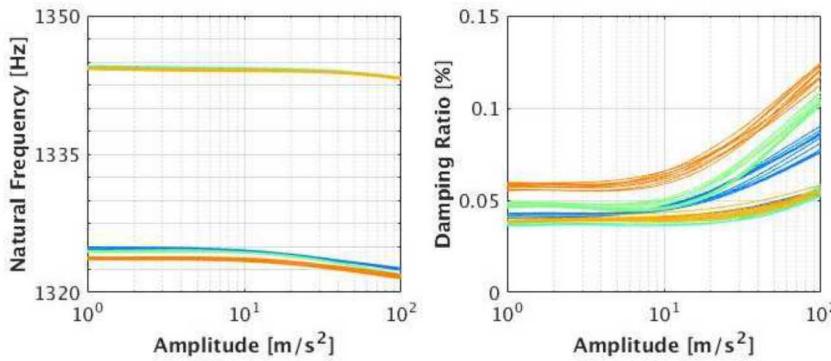
- Impact Testing



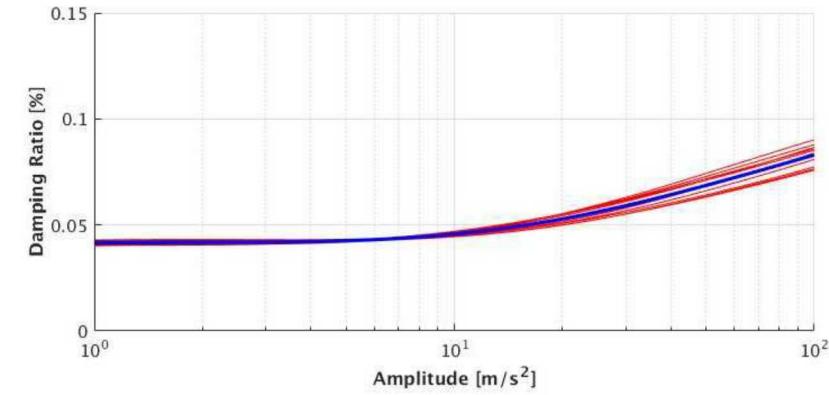
- Bandpass Filtering and Hilbert Transform



- Frequency and Damping vs Amplitude

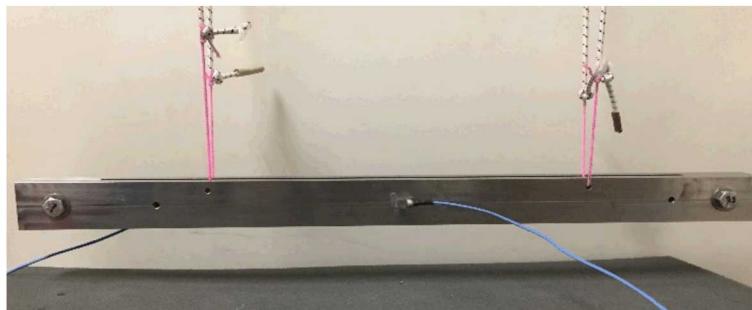


- Smooth and Average Curves

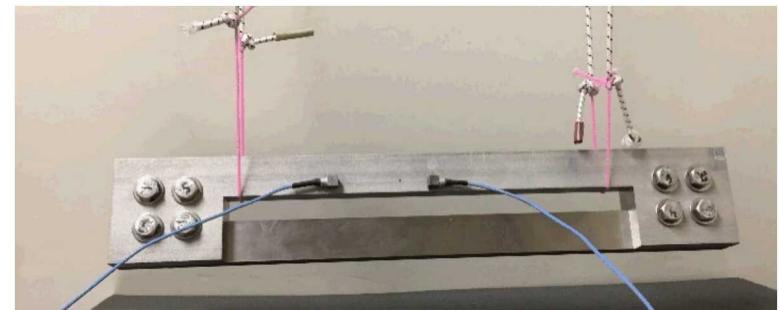


Experimental Methodology

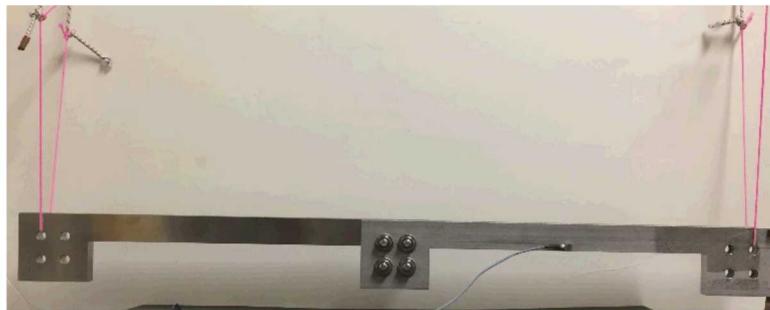
- CBM



- 4SO



- 4LS

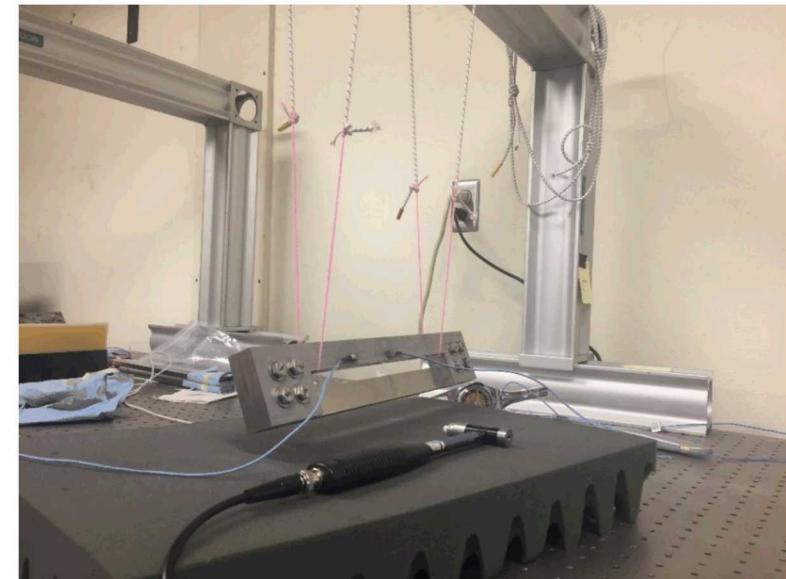
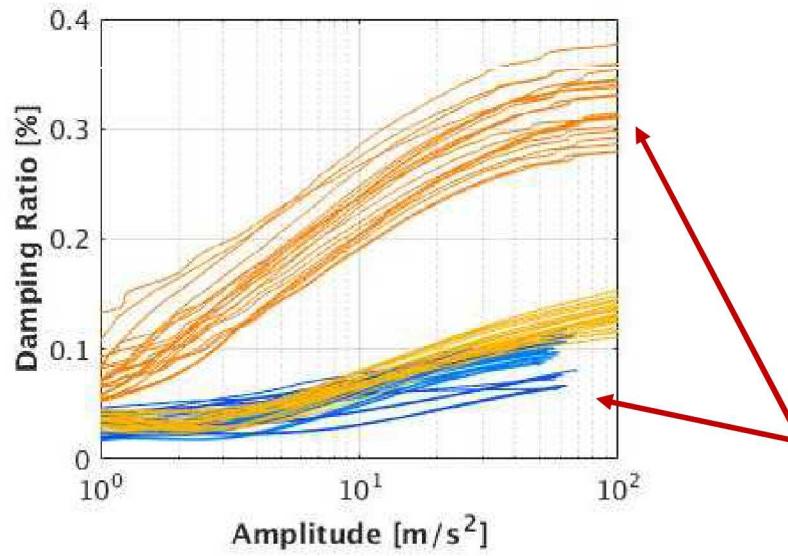


- 4VO



Experimental Methodology

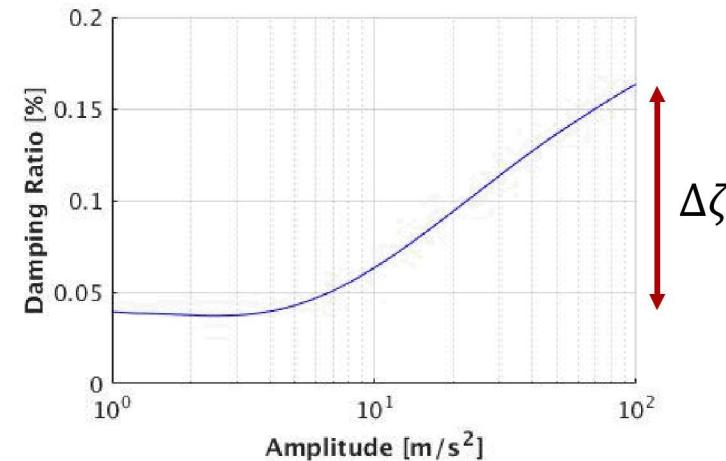
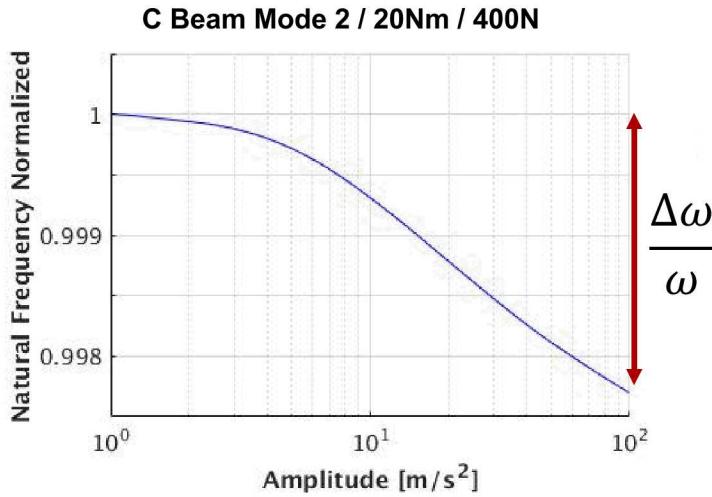
- Impact hammer testing using free-free boundary condition
- Bolt torques range from 5Nm → 20Nm
 - What is the effect of changing contact pressure within beam configurations?
- Impact Levels ranging from 60N to 900N
 - What is the effect of modal coupling?
 - How do we normalize force?
- Standardize by max mode shape



C Beam Mode 2: Extreme Modal Coupling

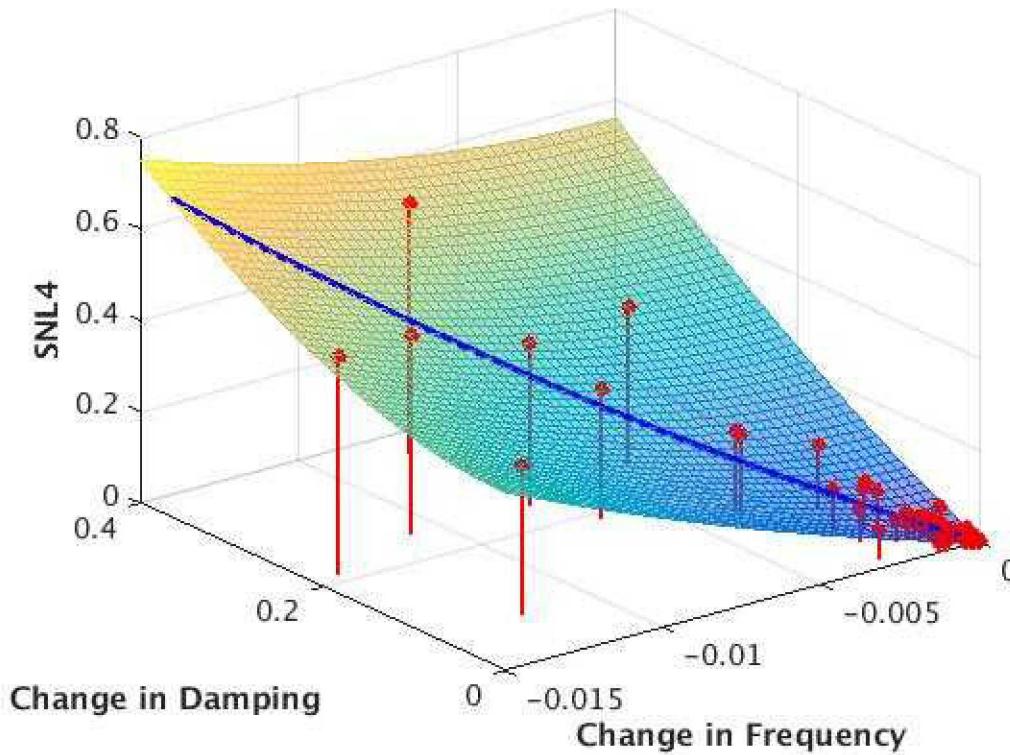
Defining Strength of Nonlinearity (SNL)

- Magnitude of shift in natural frequency and damping as the response amplitude of a structure is varied between two fixed bounds.
- $SNL = \alpha \frac{\Delta\omega}{\omega} + \beta \Delta\zeta$



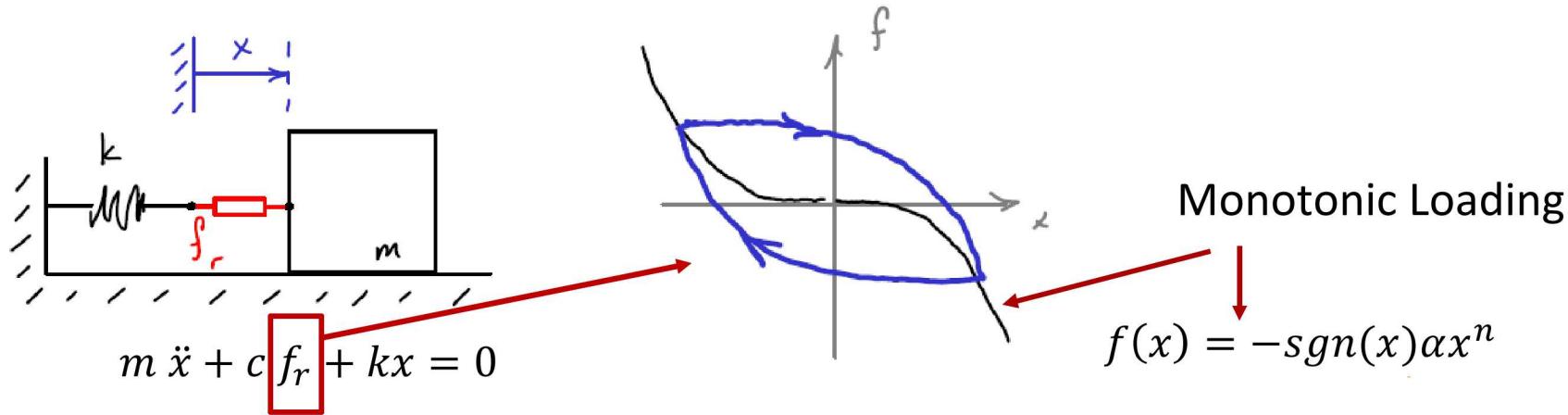
Defining SNL

$$SNL4 = \left(\left(20 * \frac{\Delta\omega}{\omega} \right)^2 + (\Delta\zeta)^2 \right)^{\frac{1}{2}} + \left(20 * \frac{\Delta\omega}{\omega} \right)^2 + (\Delta\zeta)^2$$



Defining SNL: Perturbations Approach

- Based off of a mass-spring-damper system



- General form of equivalent frequency and damping

$$\omega = 1 + C_1 A^{n-1}$$
$$\zeta = C_3 + C_2 A^{n-1}$$

Interface properties

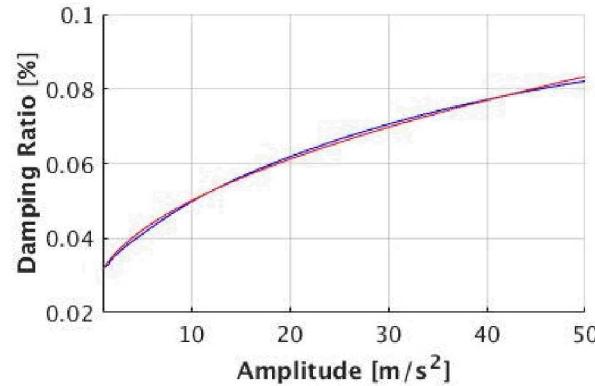
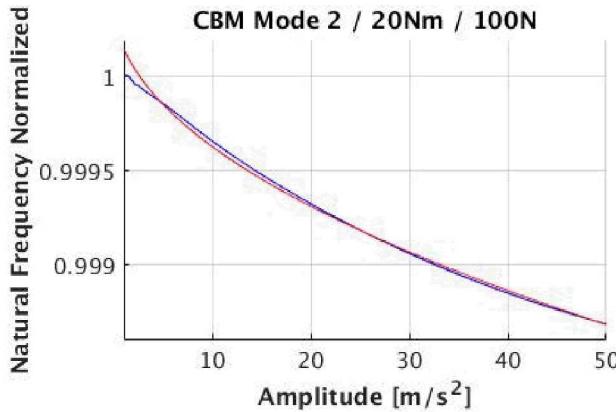
Linear Damping

Influence of joint on structure

Linear Frequency (Normalized)

Arrows from the text labels point to the terms $C_1 A^{n-1}$ and $C_2 A^{n-1}$ in the equations.

Defining SNL: Perturbations Approach



$$\zeta = -0.01 + \underline{0.0345} A^{0.25}$$
$$R^2 = 0.9931$$

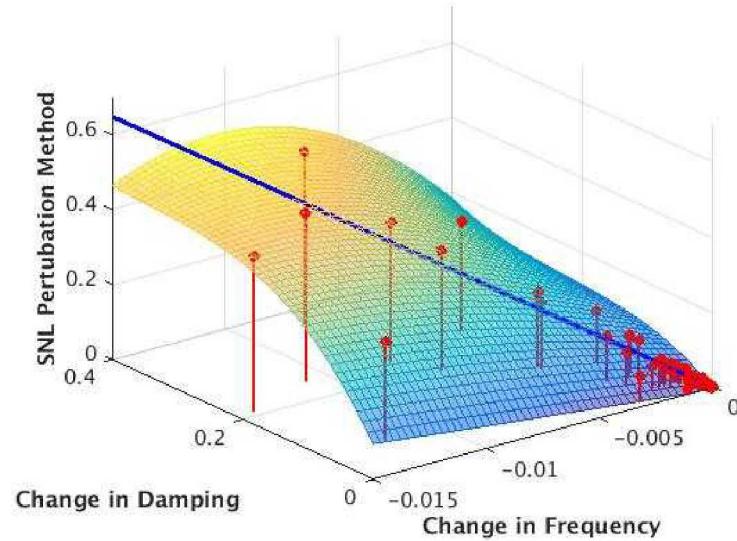
$$\omega = 1.003 - \underline{9.703 * 10^{-4}} A^{0.25}$$

$$R^2 = 0.9757$$

Overall:

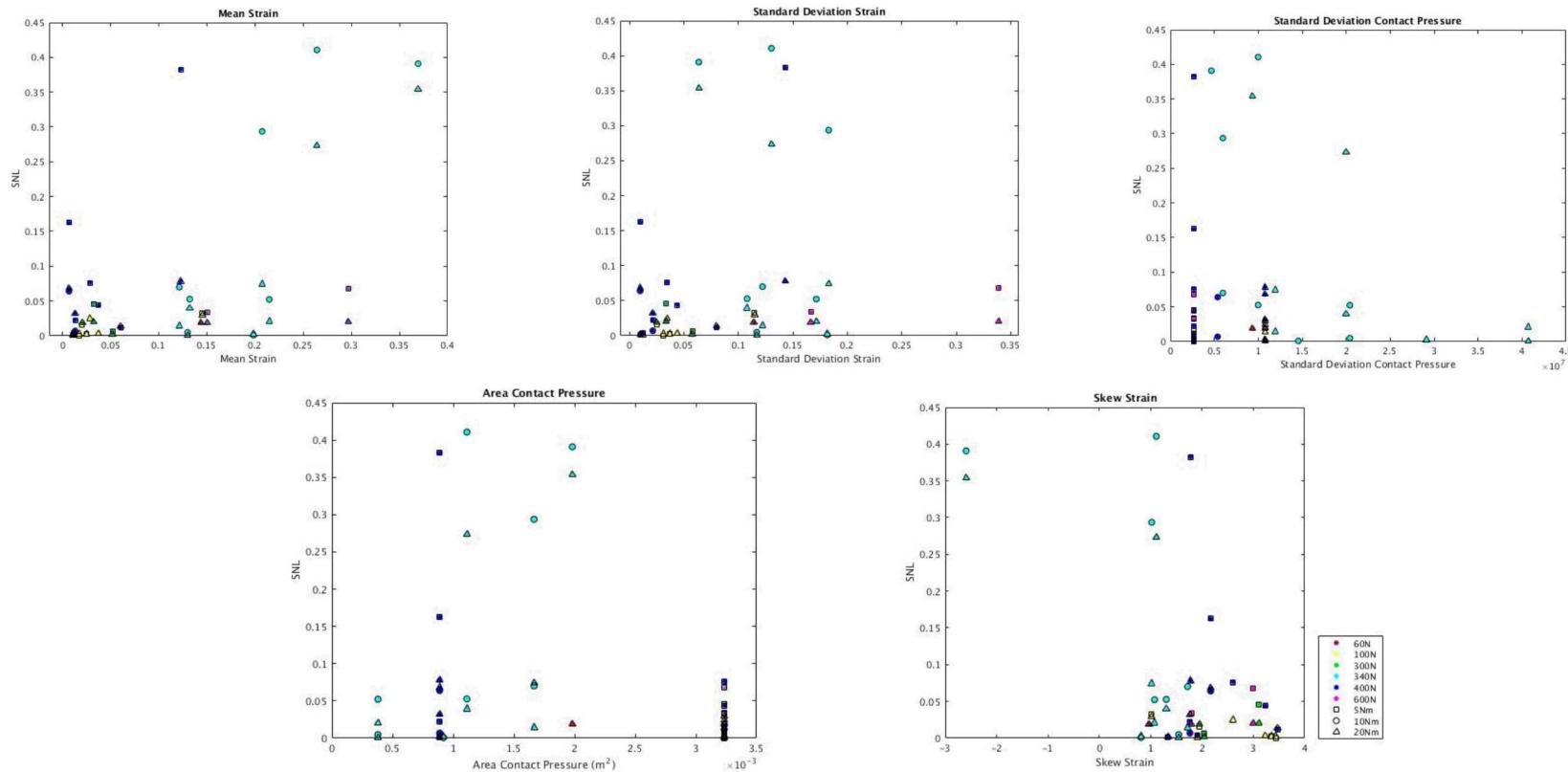
Frequency $R^2 = 0.8434$

Damping $R^2 = 0.7841$



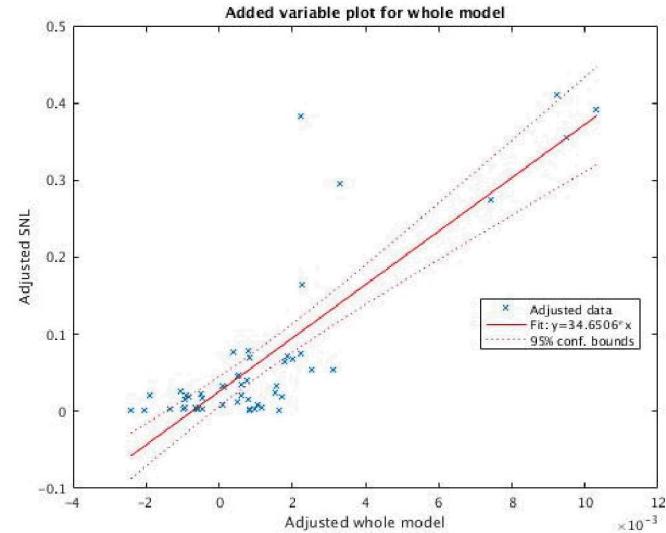
Machine Learning

- Correlation of parameters to SNL and frequency-only-based SNL by visual inspection and ANOVA
 - **70% of variance explained by 5 variables:** Mean Strain, Standard Deviation Strain, Standard Deviation Contact Pressure, Contact Area, Skew Strain



Machine Learning

- $$SNL = .065802 + 1.7405MeanE - 1.3022STDE - 5.7476 * 10^{-9}STDCP - 41.301AreaCP + .041298SkewE, p\text{-value}=1.17e-10, R^2 = .696$$
- MATLAB's built-in functions *fitlm*, *stepwiselm*, and *step* were utilized to create a linear regression model using the various statistics
 - stepwiselm* automatically tests the importance of each statistic to create the optimal metric
 - step* takes an existing model and checks whether additional terms should be added or existing terms should be removed
 - fitlm* fits a model using the parameters specified



Whole Model Parameters (R^2=0.696, p-value=1.17e-10)	p-value	Frequency Parameters (R^2=0.833, p-value=1.49e-15)	p-value
MeanE	6.5816e-08	MeanE	6.6878e-12
STDE	1.499e-05	STDE	2.8118e-10
STDCP	4.1232e-05	SkewE	2.6391e-06
AreaCP	0.00032329	STDCP	9.671e-06
SkewE	0.01428	KurtosisE	0.000461
(KurtosisE)	(0.09601)	AreaCP	.026855

Conclusions

- We were not able to produce a metric that could accurately predict the SNL metric using only contact pressure and modal strain. A future metric could possibly be determined if additional interface properties were also included.
- We were able to identify the key variables that explain variance in our SNL metric
 - Strain: Mean, Standard Deviation and Skew
 - Contact Pressure: Standard Deviation and Area
- We were able to identify areas of improvement for future research
 - Using modal acceleration instead of absolute acceleration
 - Better understand how force levels activate modal coupling
 - Implement genetic algorithms for model predictions
 - Use surface properties to explain damping variance

Acknowledgments

- This research was conducted at the 2018 Nonlinear Mechanics and Dynamics Research Institute hosted by Sandia National Laboratories and the University of New Mexico.
- This research was assisted by the research of previous NOMAD teams and research ongoing at University of Wisconsin-Madison and Rice University.
- Sandia National Laboratories is a multimission laboratory managed and operated by National Technology and Engineering Solutions of Sandia, LLC., a wholly owned subsidiary of Honeywell International, Inc., for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-NA-0003525.