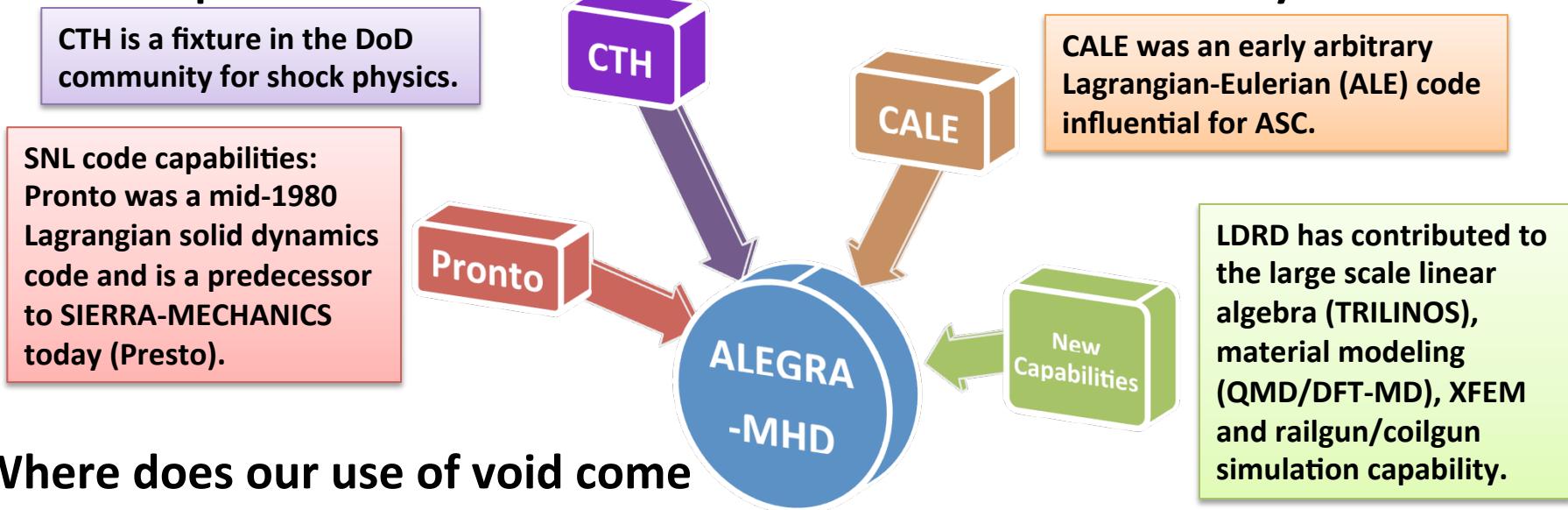


Void's Unphysical Response In Hydrocodes

Bill Rider
Sandia National Laboratories,
Albuquerque

“A vacuum is a hell of a lot better than some of the stuff that nature replaces it with.”—Tennessee Williams

ALEGRA is a hydrocode with specific multi-physics capabilities develop over the last 20+ years.



Where does our use of void come from? CTH

- **Void is used in CTH in several capacities.**
- **It is introduced to make fracture and spall volume filling.**
- **If material is discarded, it is replaced with void (cell doctor)**

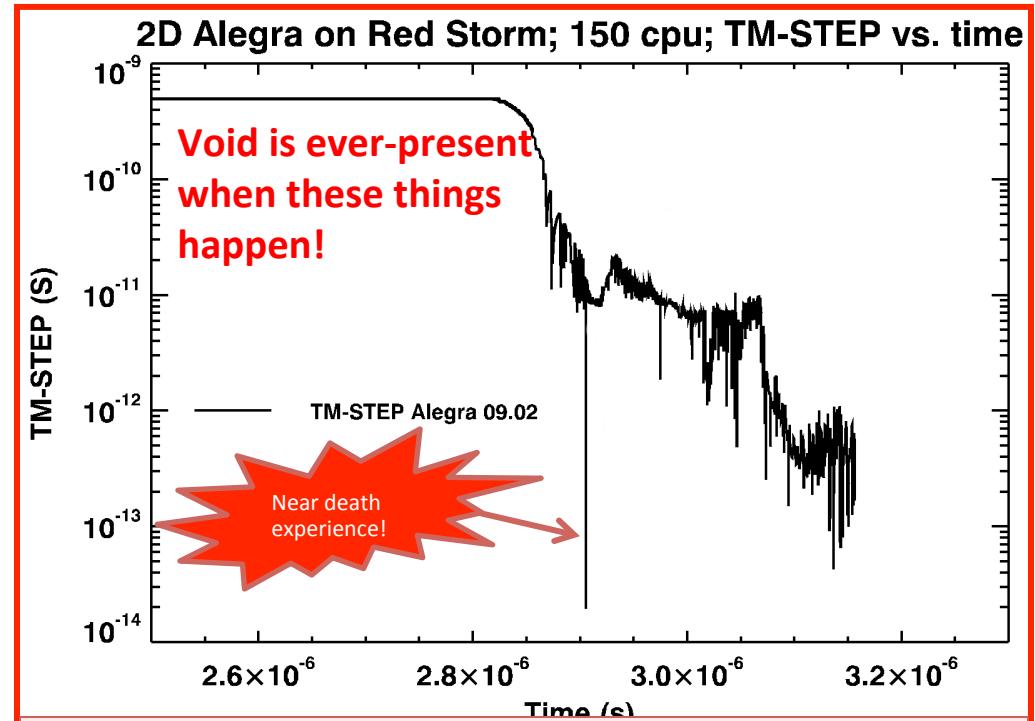
What key defining capabilities does ALEGRA possess?

- Mimetic magneto-hydrodynamics (MHD), circuit modeling
- Extended finite element (XFEM) under development
- **High fidelity material modeling** (especially for MHD)
- Ceramic material models
- Optimization and UQ linkage to DAKOTA
- **Robust modeling of high-strain rate deformation.**

Six years ago: With growing use on more complex problems, significant issues arose due to code robustness.

- Users wanted calculations to be reliable.
- Our continued support depended upon improving reliability and resilience.
- Many important calculations did not complete due to a variety of issues.

- Users had grown to expect the code to not reliably work for very challenging problems.
 - They expected the code to fail!
 - Some users wouldn't even report problems because it was the norm.



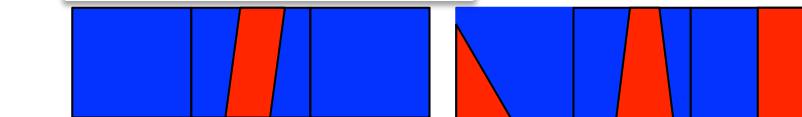
This plot shows a typical time step trace for ALEGRA in this time period. The time step “dropouts” were common as was the general decay in the magnitude of the time step. Calculations either failed or became untenable due to small time step size.

We made improvements in the remap and multimaterial methods plus the stability criteria.

Summary of remap changes: Detect the local multimaterial flow topology



Problem configurations



Third-order remap based on three element parabolic conservative interpolation.

- For robustness, the edge values are third-order, but bounded by neighbors,

$$\phi_{j+1/2} = \frac{1}{6}(2\phi_{j+1} + 5\phi_j - \phi_{j-1}) \rightarrow \frac{1}{2}(\phi_{j+1} + \phi_j) - \frac{1}{6}(\Delta_{j+1/2} \phi - \Delta_{j-1/2} \phi)$$

$$\Delta_{j-1/2} \phi = \min \text{mod}[\phi_j - \phi_{j-1}, 4(\phi_{j+1} - \phi_j)]$$

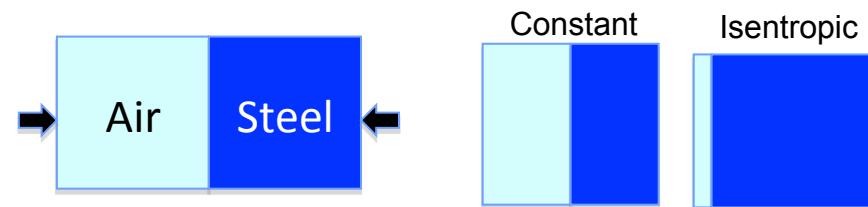
Mixed cell remap is now lower order

- ALEGRA uses the minmod scheme (the most dissipative second order “TVD” method)

$$\phi_{j+1/2} = \phi_j + \frac{1}{2} \min \text{mod}[\phi_{j+1} - \phi_j, \phi_j - \phi_{j-1}]$$

- Effectively uses one-sided differencing in mixed cells, only differencing into the pure material region (closer values).

Summary of multimaterial Lagrangian closure algorithm changes provide a physically based stable model (void is a “thorn in the side” of this algorithm)



$$\text{constant volume} \quad \frac{df_k}{dt} = 0$$

$$\frac{df_k}{dt} = f_k \left(\frac{\bar{B} - B_k}{B_k} \right) \nabla \cdot u - \frac{f_k}{\bar{p}} \frac{dp_k}{dt}$$

Summary of time step size calculations: based upon the Fourier analysis of the Lagrangian step with dissipation.

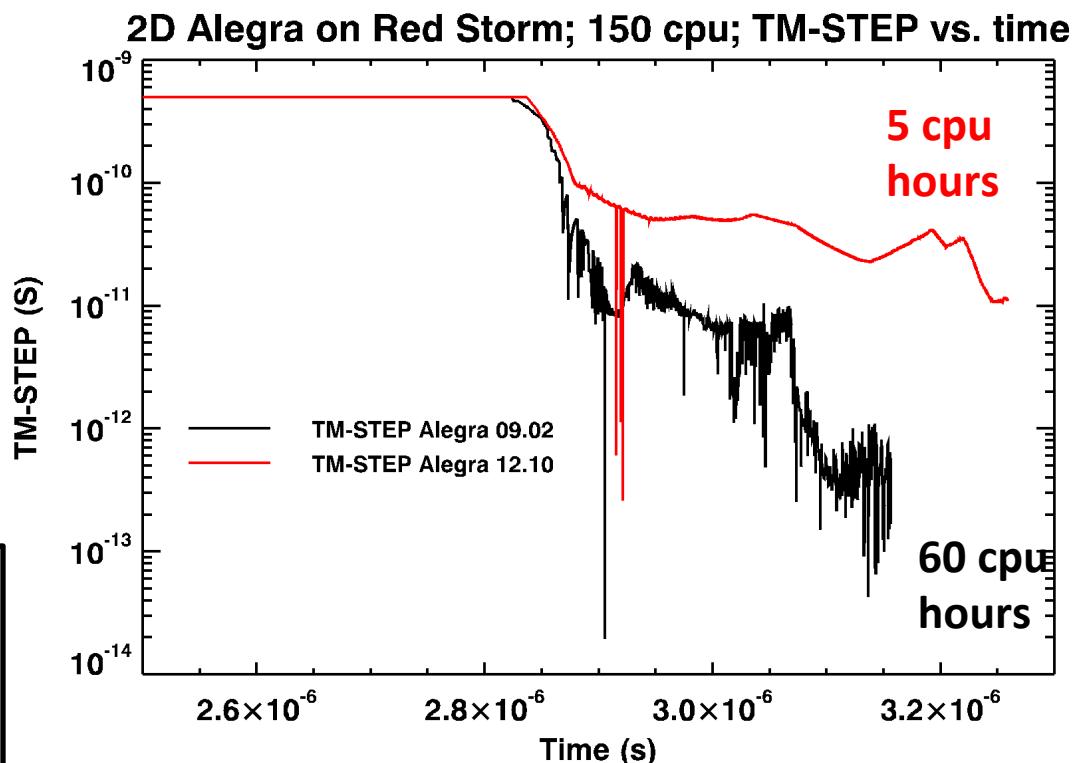
$$\Delta t_2 = \frac{h}{c \left(\sqrt{1 + \eta_{\max}^2} + \eta_{\max} \right)} \quad \eta_{\max} = c_1 + c^{-1} c_2 h |\nabla \cdot v|$$

Our existing user base stood up and took notice of the changes we made.

- The HEDP continued to rely upon ALEGRA for experimental design despite lack of direct support.
- In December 2008, I received the following e-mail from the lead designer (Ray Lemke, 1641) for EM flyer experiments:

Excerpted From Ray Lemke, Dec 11, 2008 e-mail:

...I thought you would be interested in this timing result. A large 2D Alegra MHD/thermal-conduction simulation (655,000 elements) I've been running on 150 nodes of Red Storm completes **more than 10 time faster** ... (~5 hrs completion time vs. ~60 hrs, respectively)...



ALEGRA was 12 times faster than before!

There are big
problems
with void



Lunch with Misha

- Allen and I had lunch with Misha at Multimat last year
- He is interested in exploring how everyone does void (and everyone is doing it in an ad hoc manner)
- Misha outlined several test problems we should run:
 - Void closure, 1 and 2-D
 - Free expansions, and shock through material and into void (JOWOG, NECDC possibilities)

Misha Shashkov proposed a set of problems to us at Multimat 2013 in San Francisco

- He set four problems:
 - Free expansion
 - Blast wave propagating into void
 - Void closure in 1-D
 - Void closure in 2-D
- We have started to set up and solve these with ALEGRA.
- We know the free expansion problem is an issue... it does not converge to the right answer
- We strongly suspect other codes have similar issues and we can interact on this topic.



Misha Shashkov
LANL Fellow

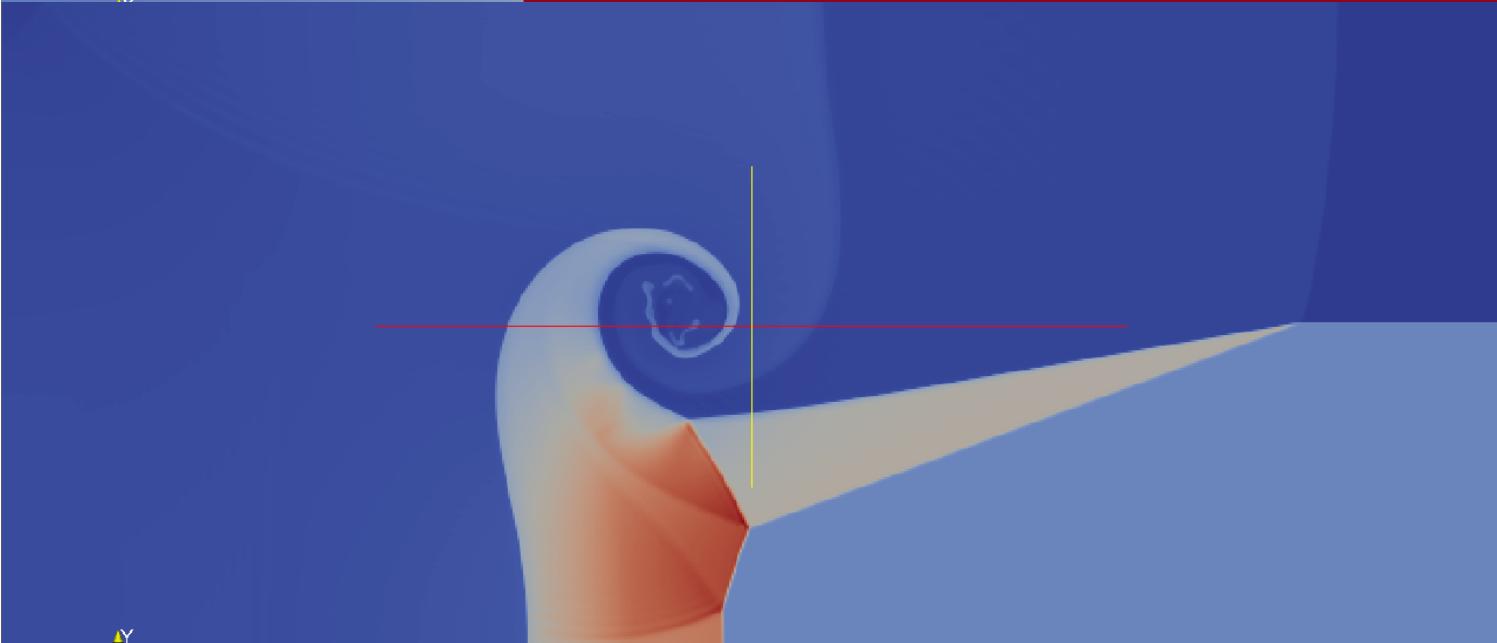
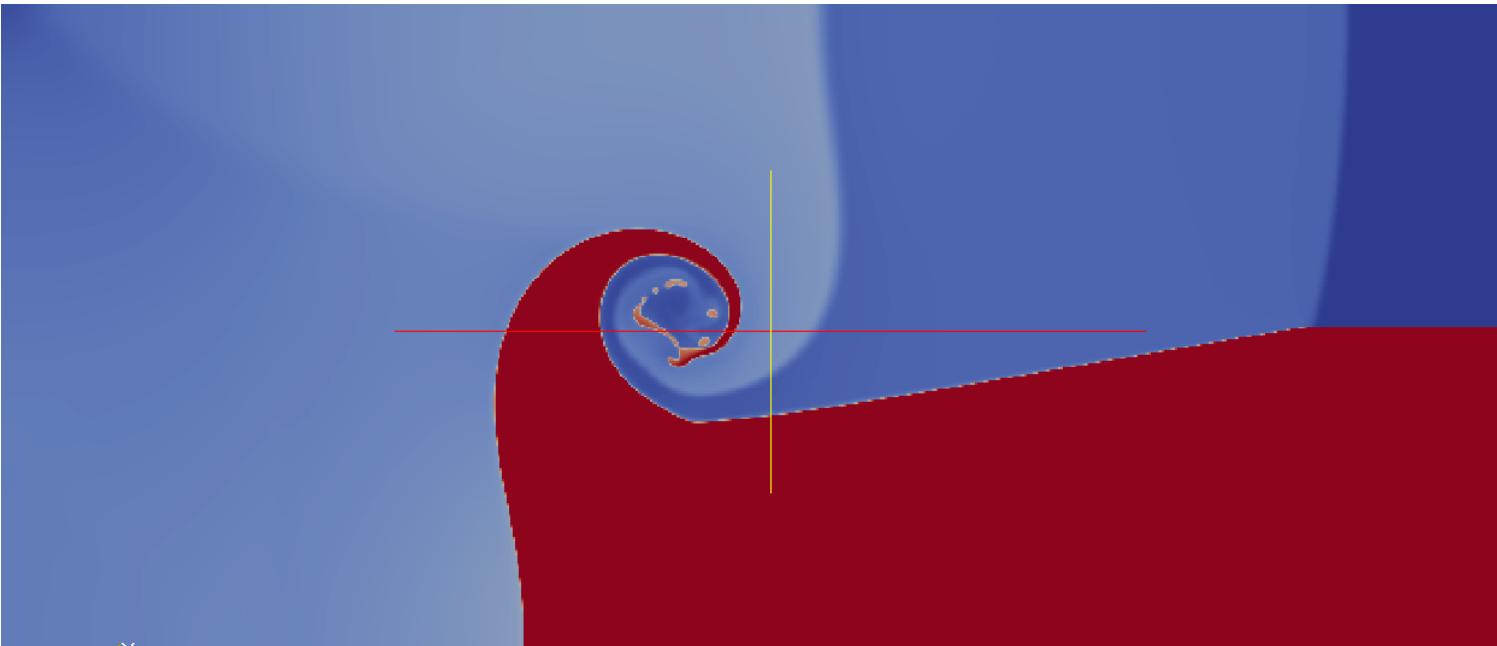
“I’m gonna make him an
offer he can’t refuse”

– The Godfather

New Test Problems

- Triple Point Series
 - Single material
 - Multimaterial
 - Asymptotically strong
- Void problems
 - Void Closure (material closes out void, results are good)
 - Void Expansion (no version of ALEGRA solves correctly, other codes share this problem)
 - Blast to void

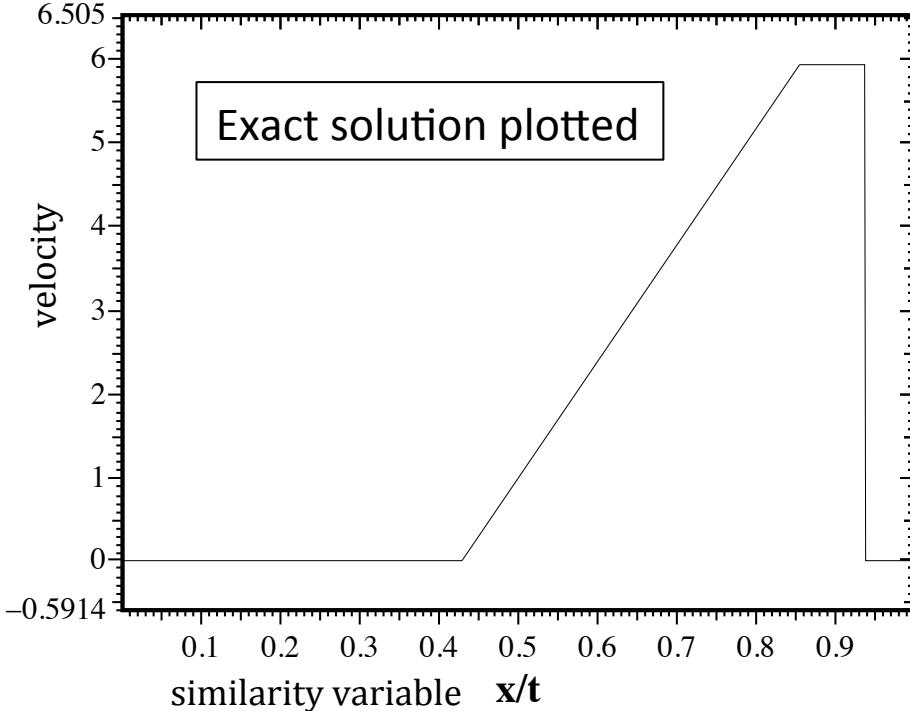
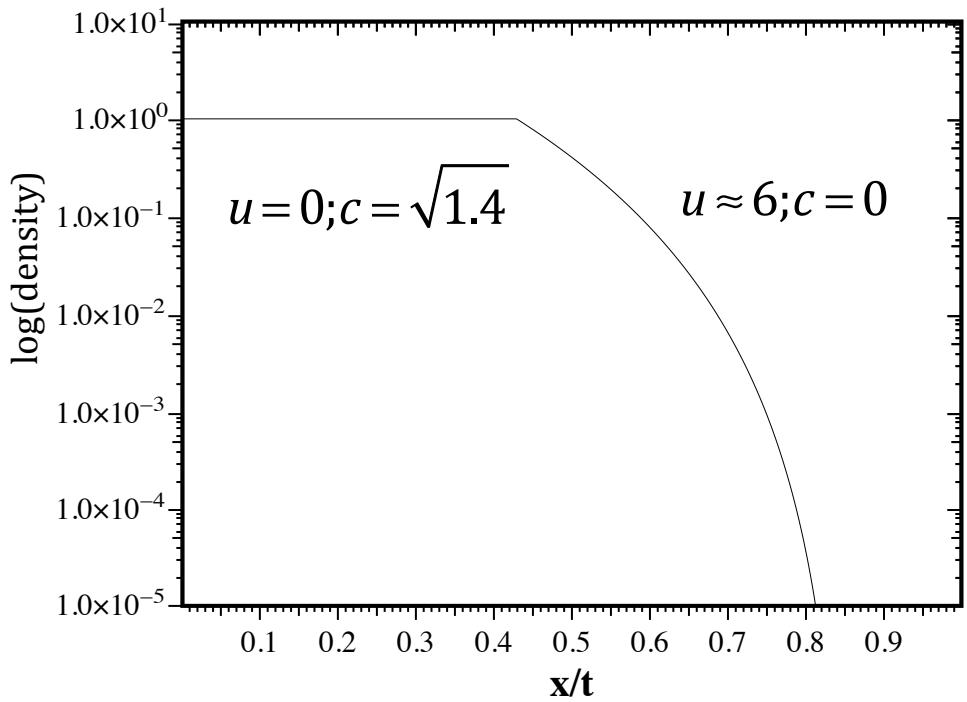
Triple Point Results are reasonable



"Just because there's an exact formula
doesn't mean it's necessarily a good
idea to use it." - Nick Trefethen

Void expansion results have asked more fundamental questions of the code

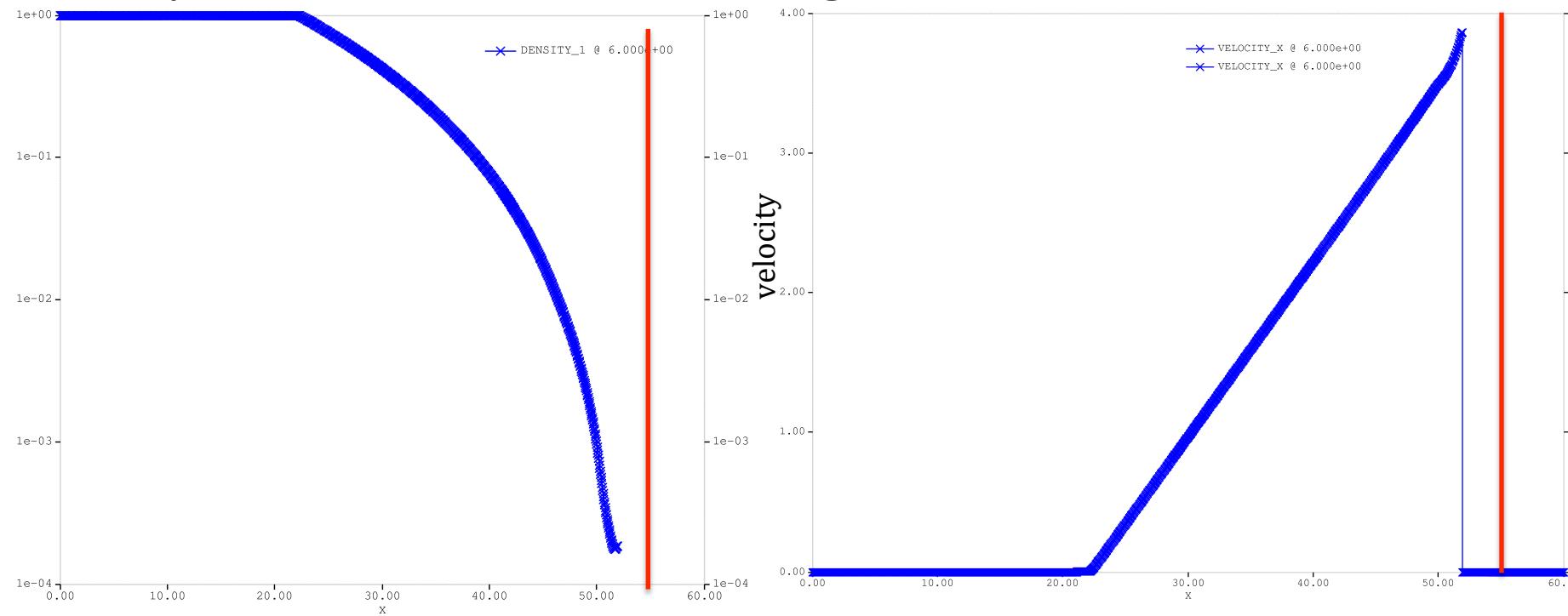
- Stability issues associated with the expansion problem highlight a fundamental oversight



The characteristic speed at the free surface is 5 times larger than the initial conditions arising from a change in pressure (Δp)

ALEGRA solutions do not converge to the analytical solution

- Nothing else does either as far as we know...
- The solution is fundamentally flawed, no set of options helps. The solution will never get to the exact one.

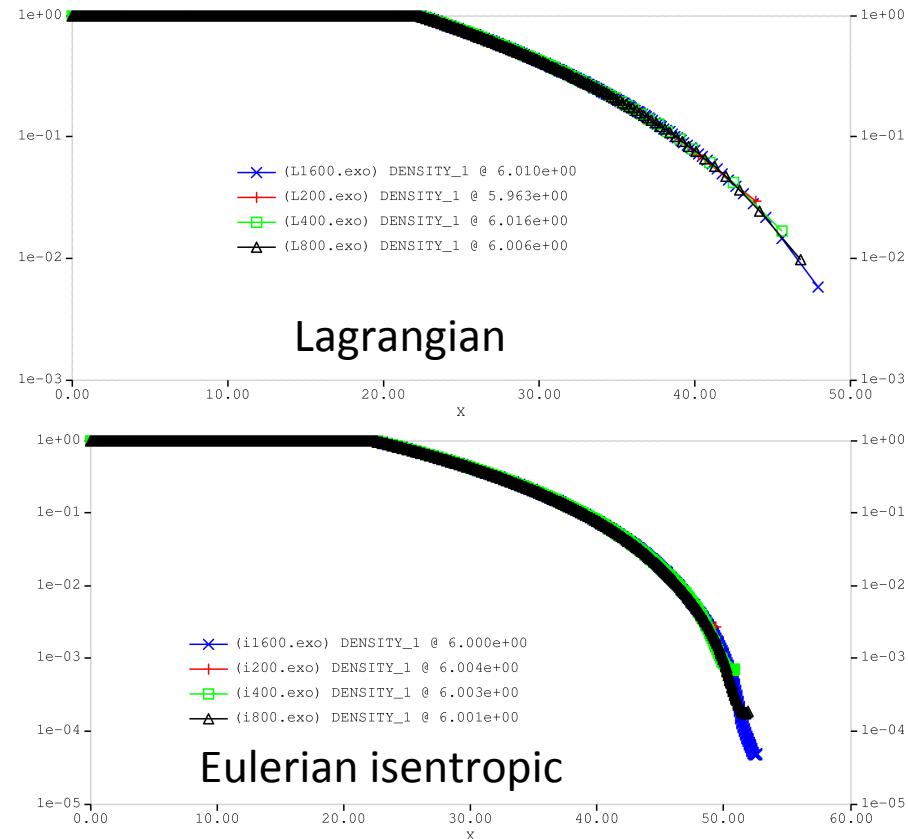
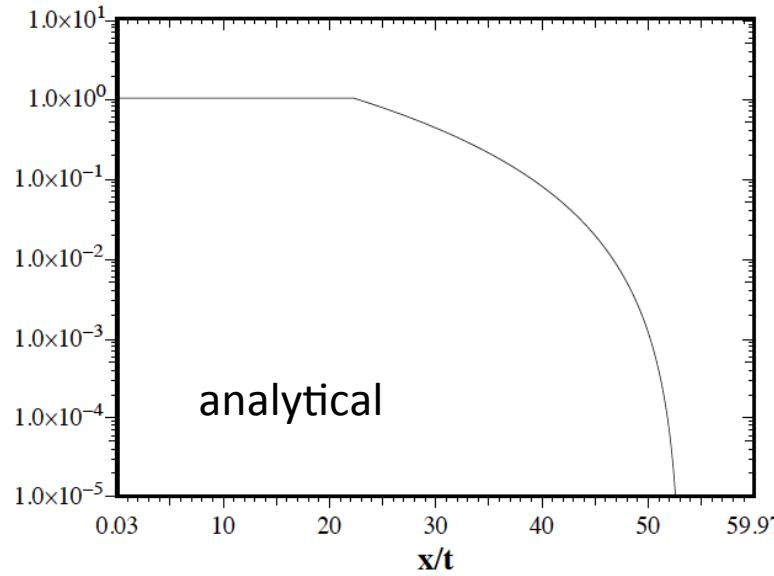


- XFEM is actually worse. Something really bad is happening, and we don't know what it is.

The free expansion problem

- We have an exact solution via the asymptotic limit of a Riemann solution
- The velocity of the free surface is analytical,

$$V_{F.S.} = V^0 + \frac{2a^0}{\gamma - 1}$$



Convergence of the free expansion

- We look at the problem on several meshes at $t=6.0$, $V_{F.S.}=3.87$. $X_{F.S.}=53.24$
- Default (isentropic MM)

$$54.24 - 8.78 \Delta x^{0.51}$$

- Lagrangian

$$51.75 - 8.69 \Delta x^{0.42}$$

- Default (CV MM)

$$54.88 - 8.71 \Delta x^{0.45}$$

- XFEM 1st order remap
 $54.65 - 12.53 \Delta x^{0.37}$

- 2nd order remap
 $54.59 - 11.13 \Delta x^{0.48}$

Almost identical issues show up with a cell-centered Eulerian calculation

- When the expansion into vacuum problem is run with an Eulerian cell-centered code using Riemann solvers, and Piecewise linear or parabolic reconstruction, the problems persist.
- No convergence
- Stability problems
- This isn't just a Lagrangian or ALE or staggered mesh issue.

The Expansion into void may induce numerical instabilities

- The standard time step estimate may be inadequate because the expansion into void induces dynamics that are faster than the stability estimate.
- Analysis is ongoing to produce a practical solution to this problem.
- The basic idea is to use existing velocity and pressure differences to estimate changes in the characteristic speeds.
- These estimates can then be used to produce a stable time step size estimate.
- This is straightforward for ideal gases, but harder for real materials. We don't want to get an overly conservative time step estimate either.

“Von Neumann told Shannon to call his measure entropy, since ‘no one knows what entropy is, so in a debate you will always have the advantage.’ ”

— Jeremy Campbell

Stability and dissipation are intimately related to each other

- Theory of hyperbolic conservation laws
$$\frac{\partial U}{\partial t} + \frac{\partial F(U)}{\partial x} = 0 \rightarrow \frac{\partial U}{\partial t} + \frac{\partial F(U)}{\partial U} \frac{\partial U}{\partial x} = 0 \rightarrow \frac{\partial U}{\partial t} + A \frac{\partial U}{\partial x} = 0$$
- This linearization can be analyzed to give the eigenvalues. These eigenvalues are used to define the stability limit.
$$A = R\lambda L; \lambda = (-c, 0, c)$$
- We could get the full stability limit through solving the Riemann problem exactly
 - what was used to create the plots on the previous slide, this takes about 16 iterations of Newton's method in that case! Always converging from below

A second order expansion of the analysis offers an answer to the issue

- Take the nonlinear flux and expand it,

$$F(U) = F(U_0) + \frac{\partial F(U_0)}{\partial U} \delta + \frac{1}{2} \frac{\partial^2 F(U_0)}{\partial U^2} \delta^2 + O(\delta^3)$$

- Plug this back into the PDE we started with

$$\frac{\partial U}{\partial t} + \frac{\partial F(U)}{\partial x} = \frac{\partial(U_0 + \delta)}{\partial t} + \frac{\partial F(U_0)}{\partial x} + \frac{\partial F(U_0)}{\partial U} \frac{\partial \delta}{\partial x} + \frac{1}{2} \frac{\partial^2 F(U_0)}{\partial U^2} \frac{\partial \delta^2}{\partial x} + O(\delta^3)$$

- Analyze – usually on the first term is carried,

$$\frac{\partial F(U_0)}{\partial U} = R_k \lambda_k L_k; \frac{\partial^2 F(U_0)}{\partial U^2} = \frac{1}{2} (\lambda_j - \lambda_k) L_i (R_k \cdot \nabla_U R_j - R_j \cdot \nabla_U R_k) \xrightarrow{\approx 0} \delta_{ik} (R_j \cdot \nabla_U \lambda_i)$$

- We expand the second order term (see e.g., Lax, Dafermos, and Roe & Balsara)

$$\lambda_k(U) = \lambda_k(U_0) + (R_k(U_0) \cdot \nabla_U \lambda_k(U_0)) L_k(U_0) \partial_x U$$

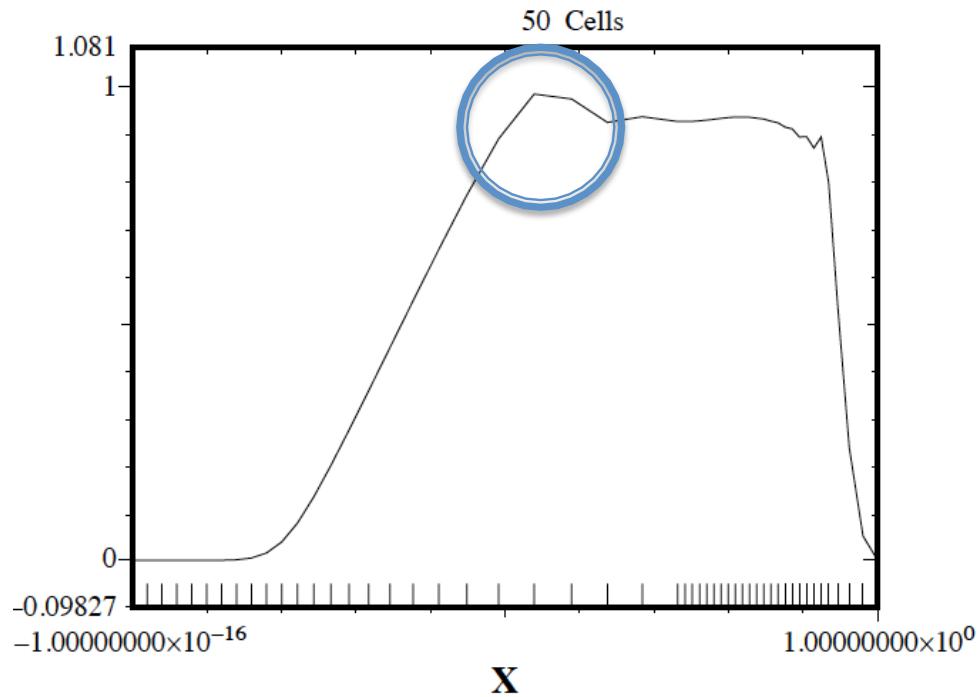
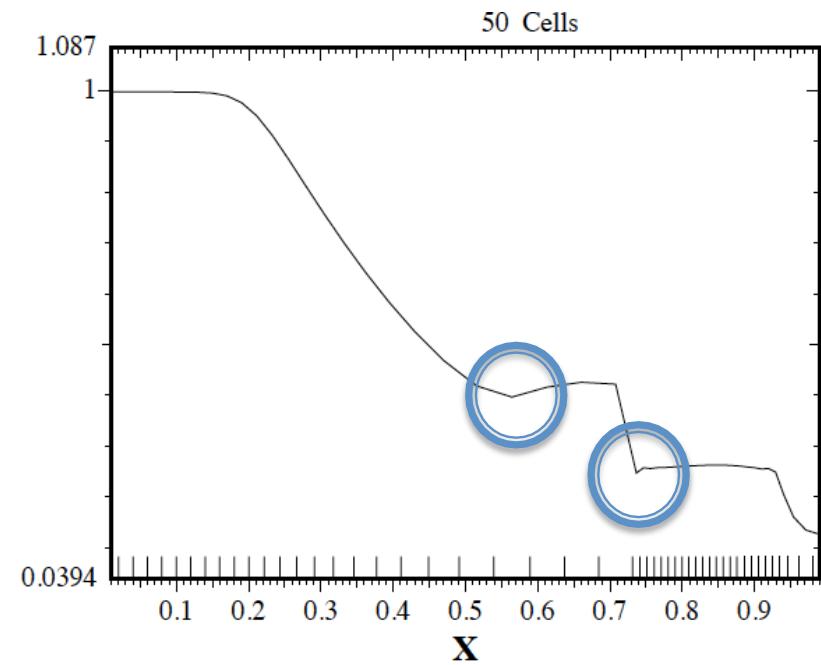
Diagnosis: Is this an entropy violating solution?

Lax Entropy Condition – Valid Shock

$$\lambda_L > \lambda > \lambda_R$$

Lax Entropy Condition – Valid Expansion

$$\lambda_L < \lambda < \lambda_R$$



Connection to artificial viscosity

- Usually the first order term defines the stability of the integration, i.e.,

$$\Delta t \leq \frac{\Delta x}{|u| + c} \text{ or } \Delta t \leq \min \left(\frac{\Delta x}{|u|}, \frac{\Delta x}{c} \right)$$

- We are proposing including the second order term, (for u & p , Δ 's in space)

$$\Delta t \leq \frac{\Delta x}{|u| + c + \frac{\gamma+1}{4} \left(|\Delta u| + \frac{|\Delta p|}{\rho c} \right)} \text{ or } \Delta t \leq \min \left(\frac{\Delta x}{|u|}, \frac{\Delta x}{c + \frac{\gamma+1}{4} \left(|\Delta u| + \frac{|\Delta p|}{\rho c} \right)} \right)$$

- These same ideas are used to define the level of numerical dissipation, usually the first term is used, but the second term also defines the quadratic part of the Q .

$$Q = c \Delta u + \frac{\gamma+1}{4} \left(|\Delta u| + \frac{|\Delta p|}{\rho c} \right) \Delta u$$

For dissipation via artificial viscosity, Riemann solvers and time step estimates

- We might do well to work with estimates of wave speeds that are bounding the “truth” from above, not below
- Riemann solver iterations might start with estimates that are larger than the solution
- The same for time step sizes.

“Nature abhors a vacuum.”
–Francois Rabelais

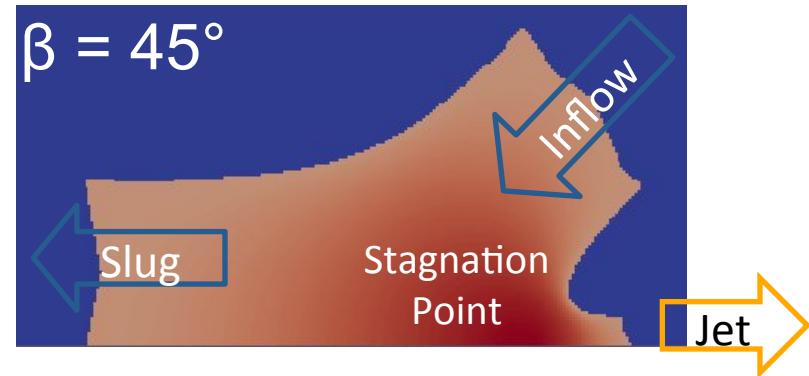
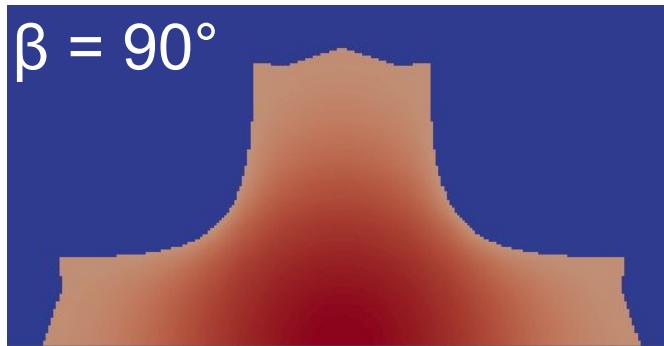
Recent progress has focused on the nature of the velocity at nodes

- The nodal “velocity” is really a conserved quantity, the momentum of the node
- We can recover the velocity at the nodal point and use this to move the nodes
$$u = \langle u \rangle - \frac{h^2}{24} \Delta^2 u$$
- This improves the solution notably and lessens the entropy violations.
- It does not fix things, it makes them better.
- Should this velocity be used for the energy?

“We become aware of the void as we fill it.” — Antonio Porchia

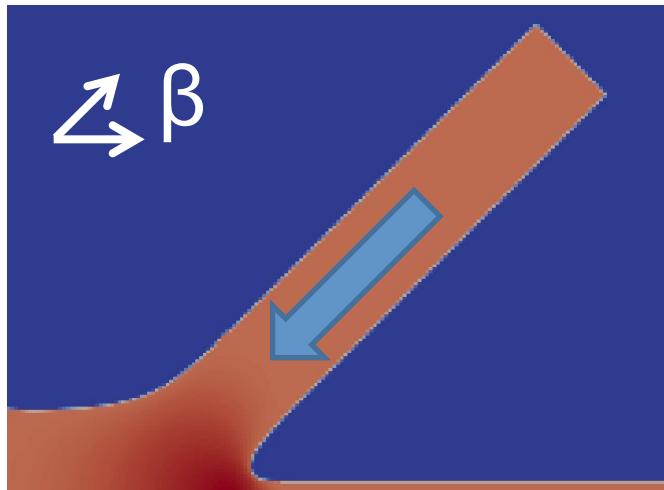
We are using a complex test problem with an exact solution to run our code through the “ringer”

CJETB code exodus solutions

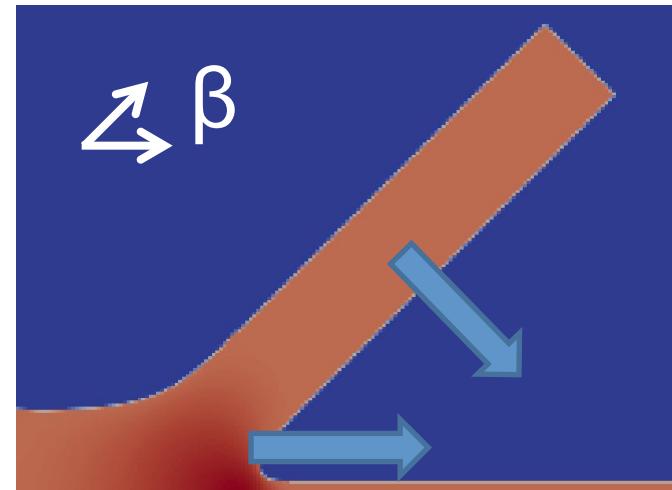


Simulations – 2 incident angles, 2 frames of reference

Stagnation Point
Frame of Reference

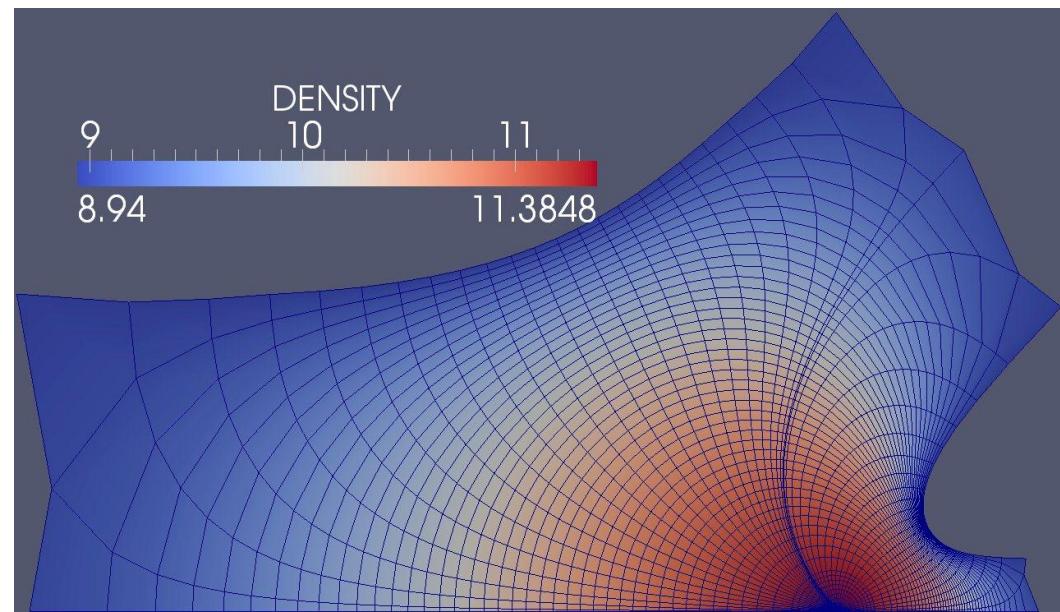


Laboratory Frame of Reference



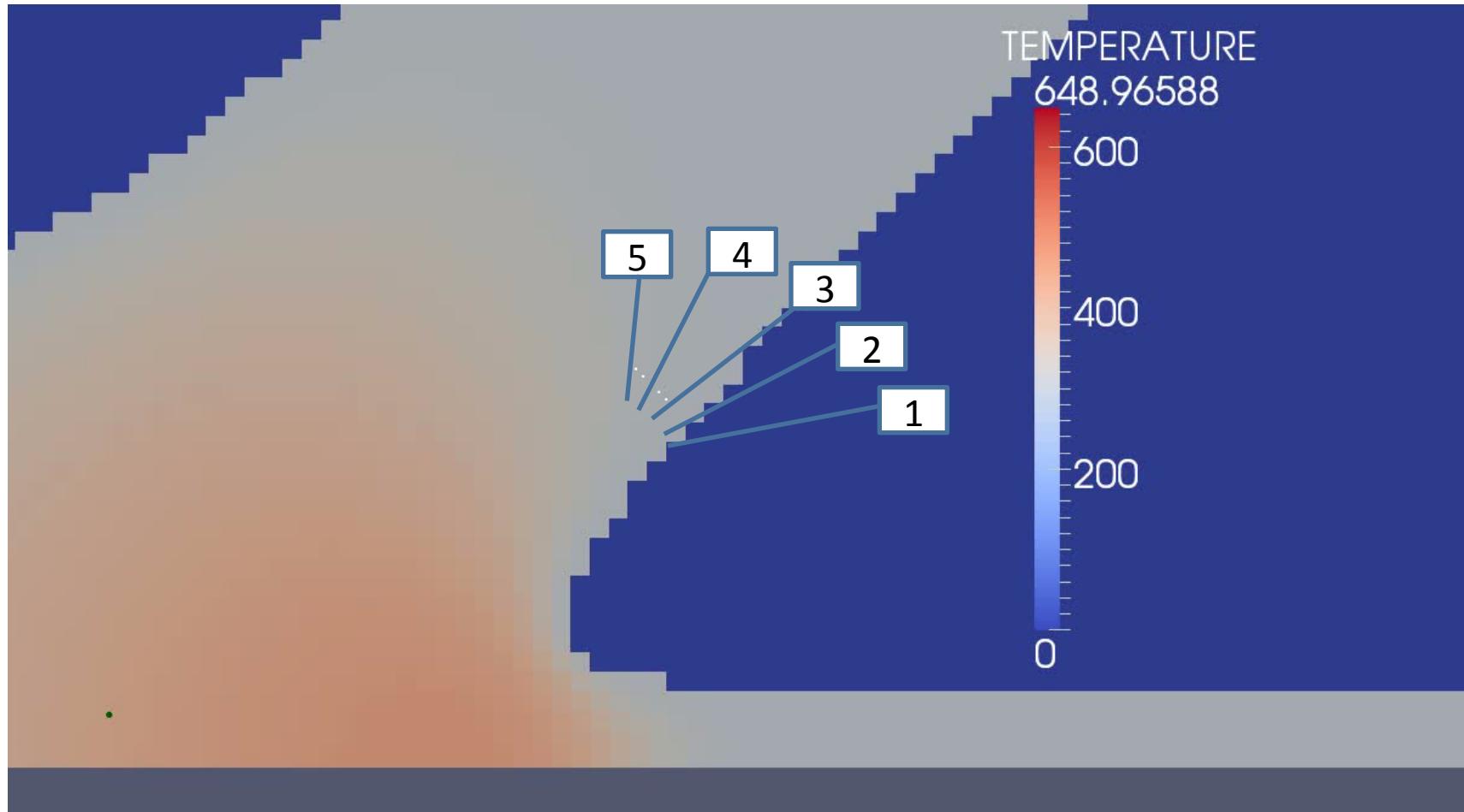
Exact Solution

- Complex difficult problem with simple characteristics
 - Steady Plane Subsonic Isentropic Fluid Flow (no strength)
 - Complex analytical representation (cjebt.f code) provides a solution on an exodus mesh. Robinson SAND2002-1015.
 - Solution imported to ALEGRA using diatoms exodus solution import
 - MG Murnaghan EOS Model
 - Two free reference curve parameters, valid for low compression
 - Simplified version of MG US UP useful for V&V



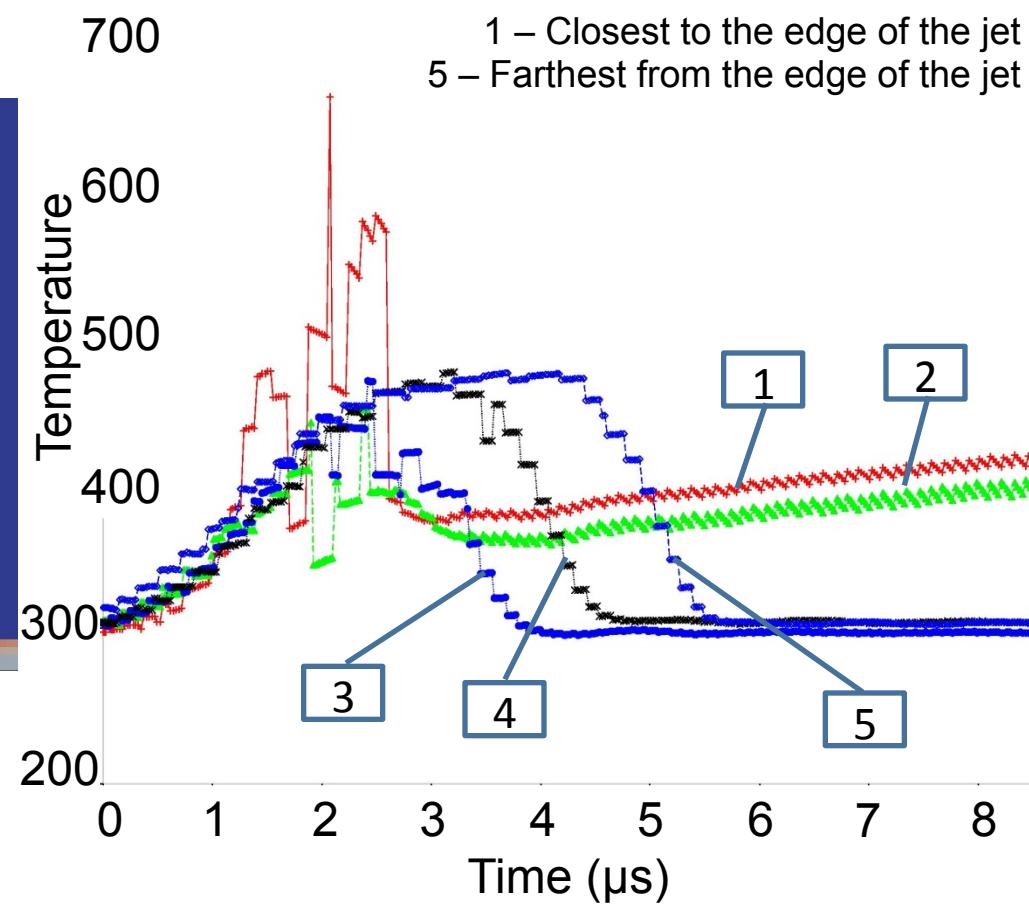
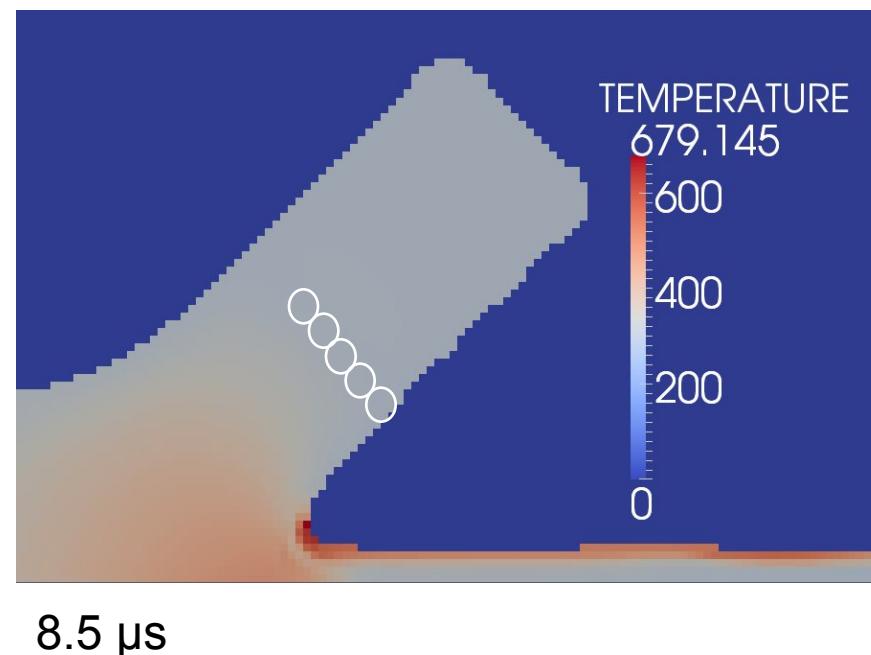
Lagrangian Tracers

In future slides, numbered tracers refer to the distance from the edge of the jet



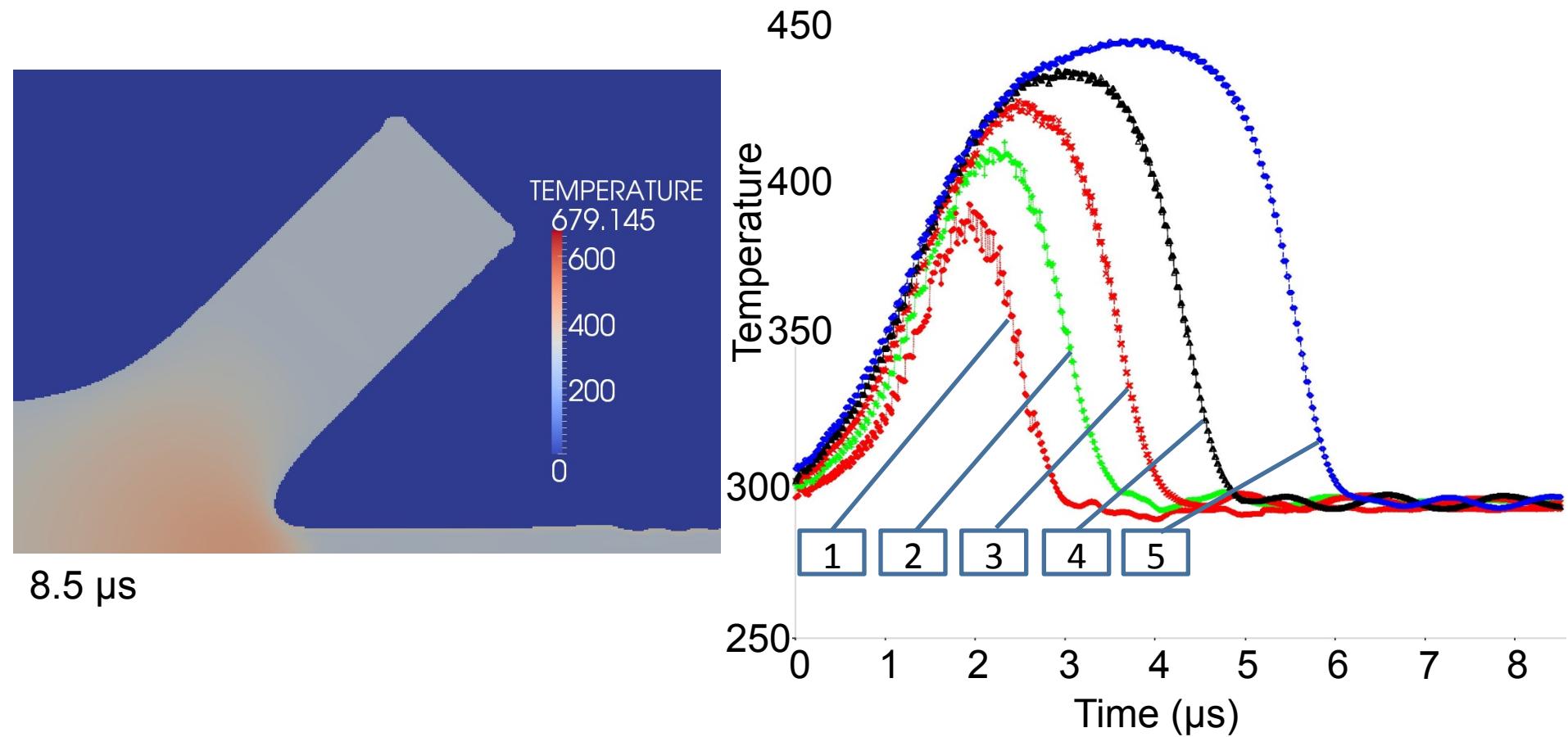
Heating Along the Jet – Under Resolved

- Default settings of artificial viscosity cause heating along the edge of the jet in under resolved cases



Resolution Study

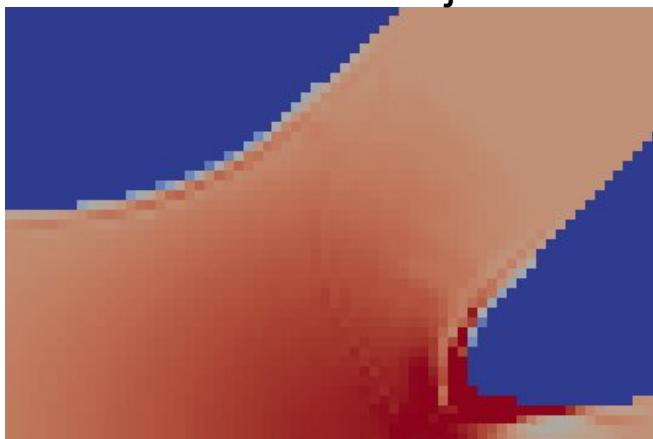
A resolved mesh completely reduces the heating along the edge of the jet, results consistent with about a first-order convergence.



Debar Energy Advection – Resolved

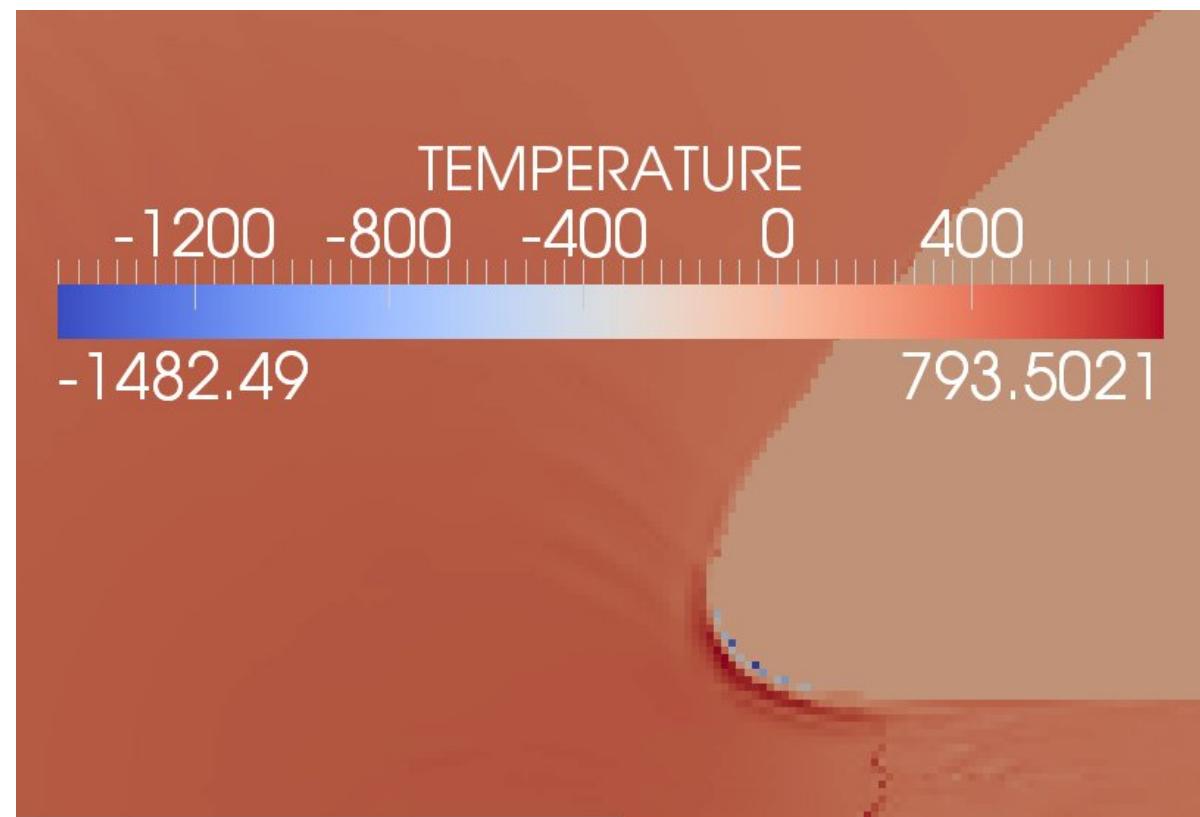
Temperature irregularities are still seen with a resolved mesh

4 elements across jet



Stagnation Point Frame

16 elements across jet



Laboratory Frame

Standard Artificial Viscosity Limiter and Hyperviscosity ON/OFF

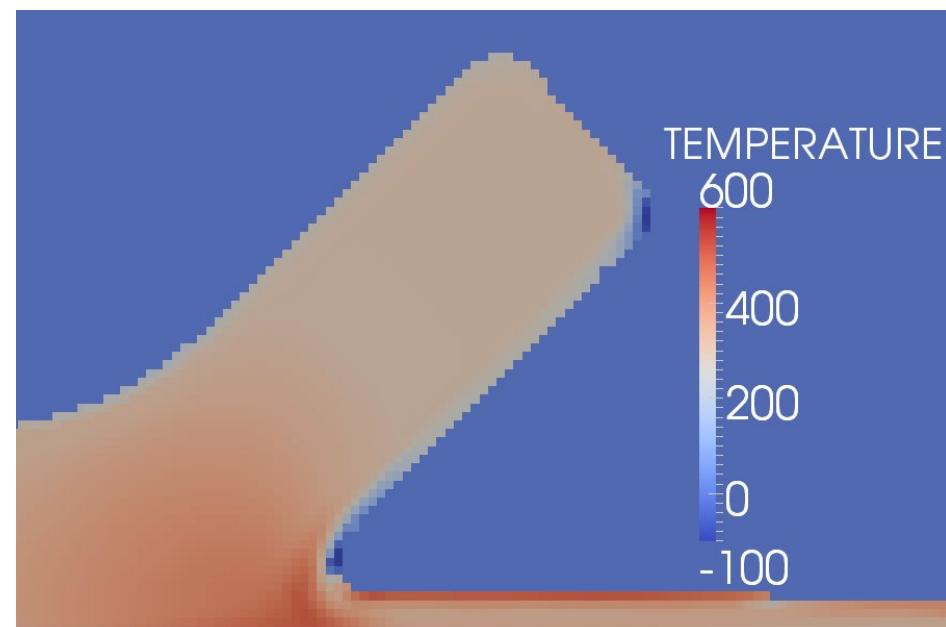
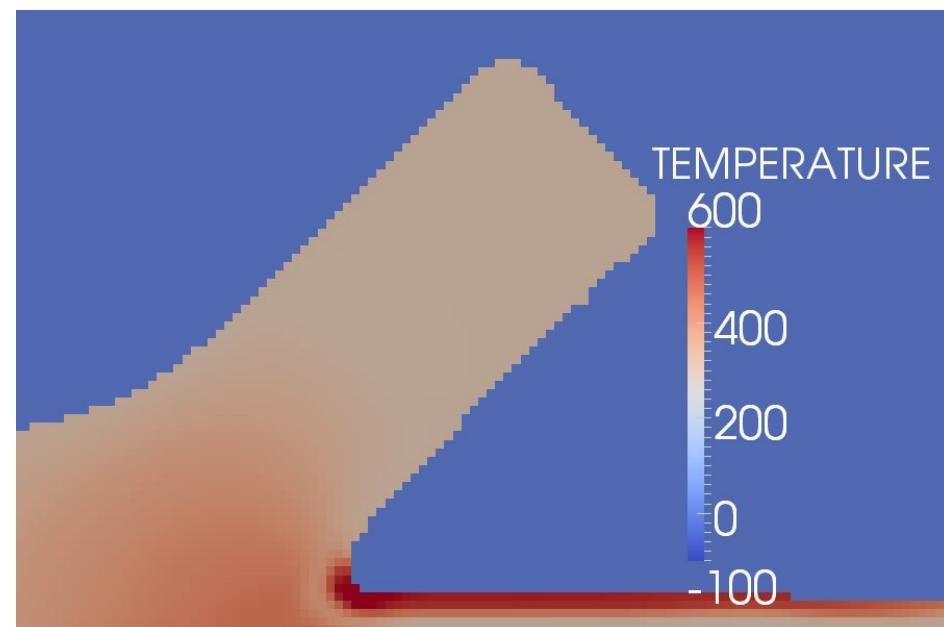
45° Lab Frame case shown below at 8.5 μ s

Standard Artificial Viscosity – Default Settings

Linear 0.15
Quadratic 2.0
Expansion Linear= OFF
Expansion Quadratic= OFF
Limiter = OFF
Hyperviscosity = 0.0

Standard Artificial Viscosity – New Settings

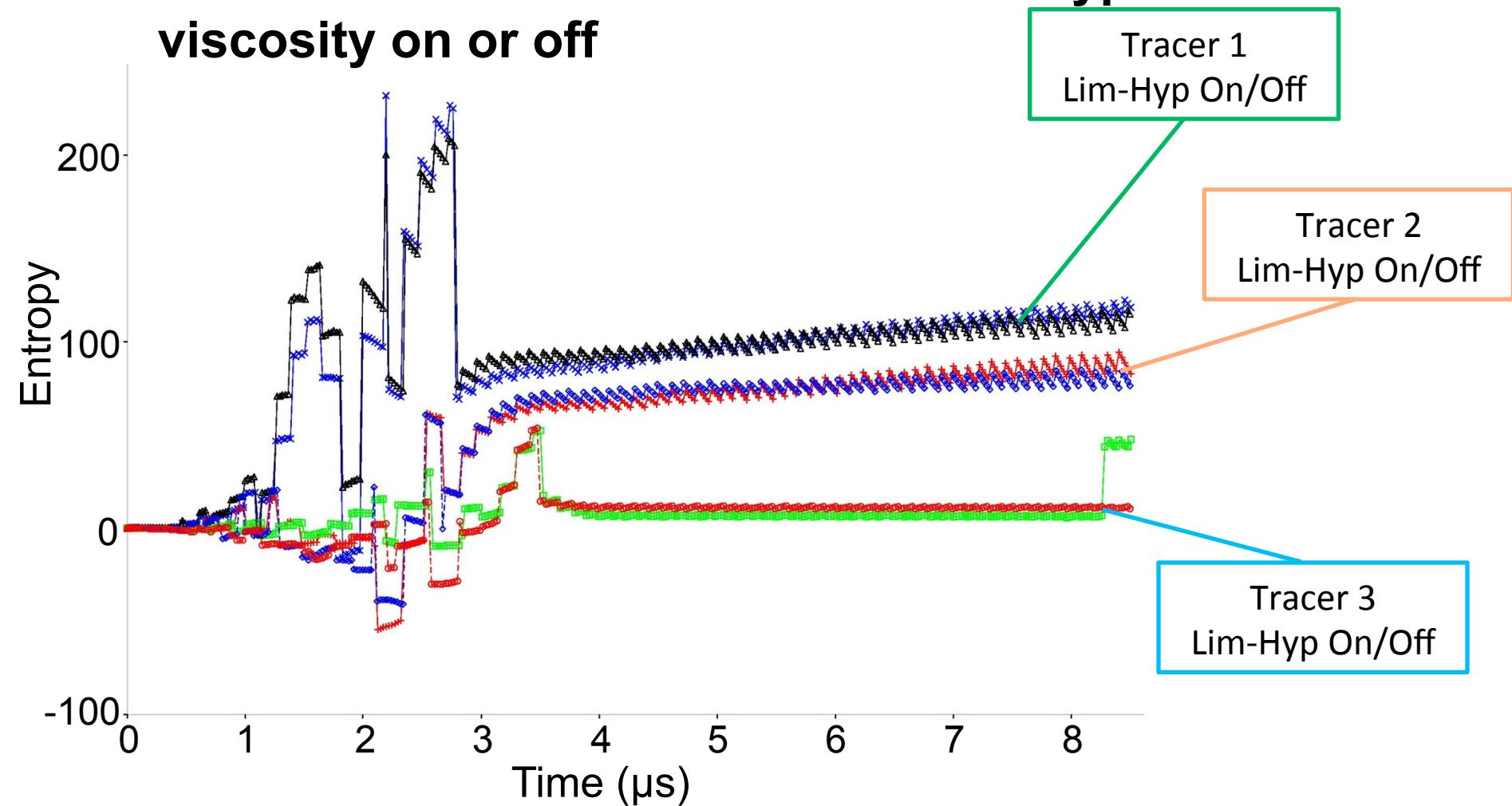
Linear 1.00
Quadratic 2.5
Expansion Linear= ON
Expansion Quadratic= OFF
Limiter = ON
Hyperviscosity = 1.0



Entropy

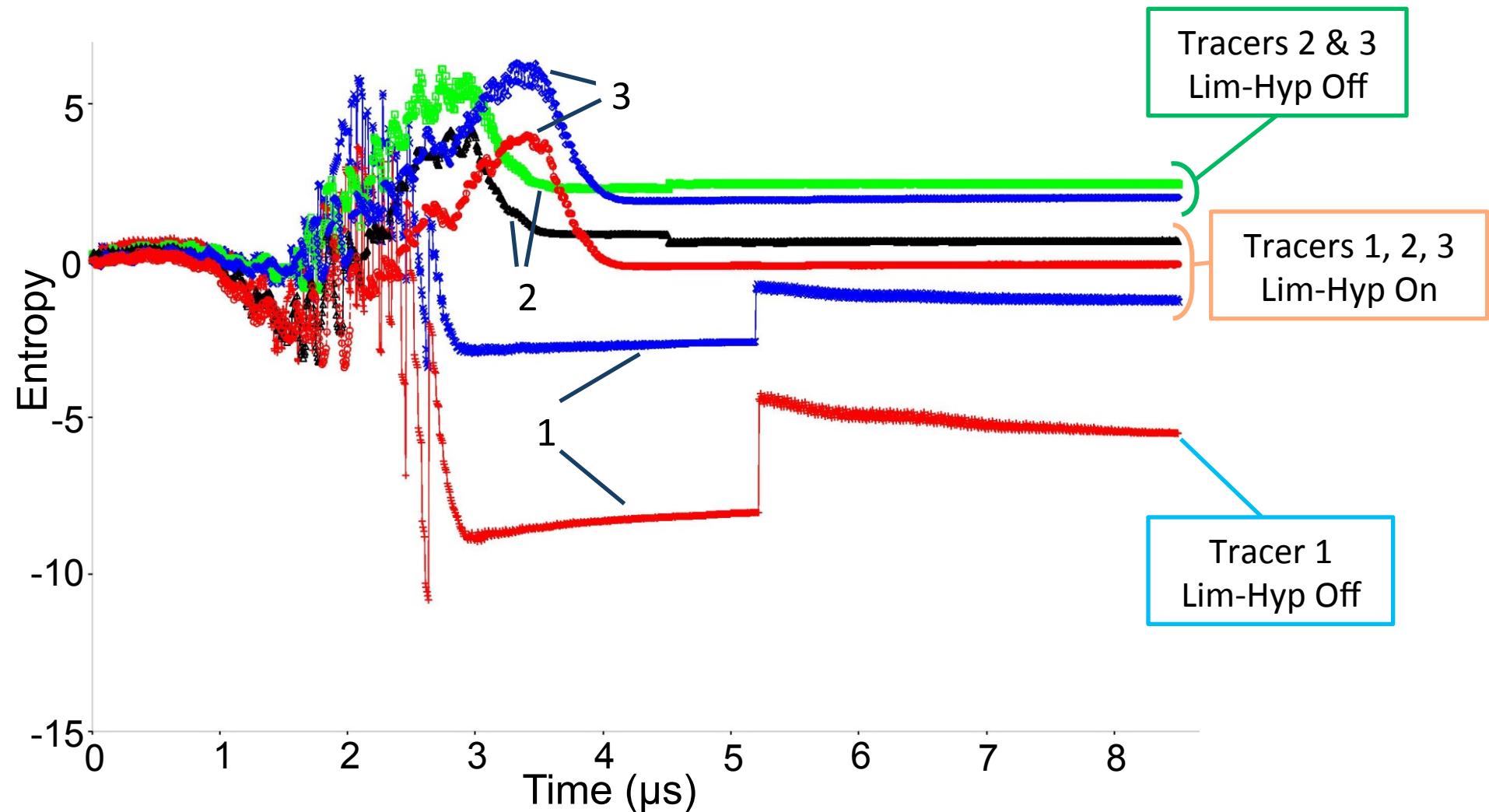
Limiter and Hyper Viscosity ON/OFF

With low resolution, little difference is noticeable between tracers with the limiter and hyper viscosity on or off



Entropy – High Resolution Limiter and Hyper Viscosity ON/OFF

With a resolved mesh, the limiter and hyper viscosity improve the results of the simulation



“I love to talk about nothing. It's the
only thing I know anything about.”

— Oscar Wilde

Conclusions

- Efforts to improve the modernity and resilience of the code is in mid-stream.
- The complexity of the effort has made it difficult
- The standard that we apply to code results makes changing the defaults difficult but drives a detailed look at old and new algorithms
- We have to balance a number of requirements:
 - Regression Testing Suite
 - Quality Embedded Verification Expectations
 - Prototype Problems Requirements
 - Customer Needs and Expectations
 - Performance Characteristics

References

- Thompson S. L., "Improvements in the CHARTD Radiation- Hydrodynamics Code II: A Revised Program", SC-RR-710713, Sandia National Laboratories (1972).
- Thompson S. L., "CSQ – A Two-Dimensional Hydrodynamic Program with Energy Flow and Material Strength", SAND74-0122, Sandia National Laboratories (1975).
- Crawford, D. A., et al., "Adaptive Mesh Refinement in the CTH Shock Physics Hydrocode" in New Models and Hydrocodes for Shock Wave Processes in Cond. Matter, Edinburgh, U.K., May 19-24, (2002).
- McGlaun, J. M., Thompson, S. L. and Elrick, M. G. 1990. "CTH: A three dimensional shock wave physics code", Int. J. Impact Engng., Vol. 10, 351 – 360.
- A. Robinson, et al., ALEGRA: An Arbitrary Lagrangian-Eulerian Multimaterial, Multiphysics Code, 46th AIAA Aerospace Sciences Meeting and Exhibit 7 - 10 January 2008, Reno, Nevada, AIAA 2008-1235
- [Gottlieb, J. J.; Groth, C. P. T.](#), Assessment of Riemann solvers for unsteady one-dimensional inviscid flows of perfect gases, Journal of Computational Physics (ISSN 0021-9991), vol. 78, Oct. 1988, p. 437-458.
- E. M. Landau and L. D. Lifshitz, Fluid Mechanics, Second Edition: Volume 6 (Course of Theoretical Physics), Elsevier, 1980.
- M. Shashkov, Personal Communication, San Francisco, September 2013.
- J. Greenough, W. Rider, A quantitative comparison of numerical methods for the compressible Euler equations: fifth-order WENO and piecewise-linear Godunov, J. Comp. Phys., 196(1), p. 259-281, 2004.
- W. J. Rider, Ed Love, Michael K Wong, O. Erik Strack, Sharon V. Petney, Duane A Labreche, "Adaptive methods for multi-material ALE hydrodynamics," Int. J. Num. Meth. Fluids, Vol. 65, (11-12), pp. 1325–1337, April 2011.