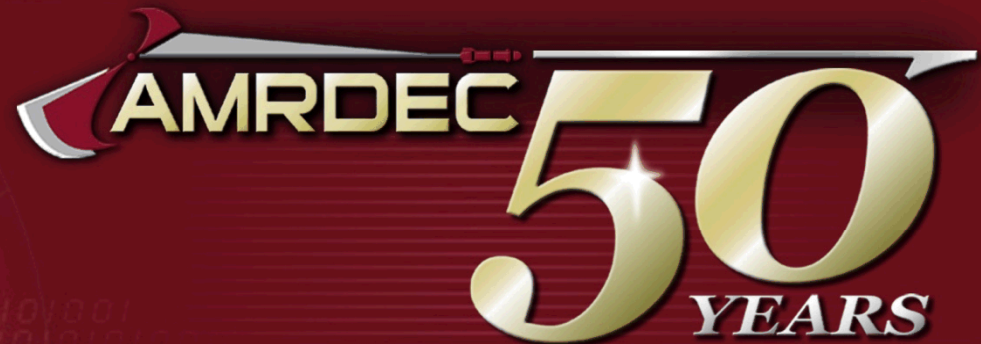




*High Performance Propellant  
Fragment Impact Testing: Small-scale  
and Full-scale*



*Staggering Accomplishments...*

*Limitless Possibilities*

**TECHNOLOGY DRIVEN. WARFIGHTER FOCUSED.**

**SEPT 2016**

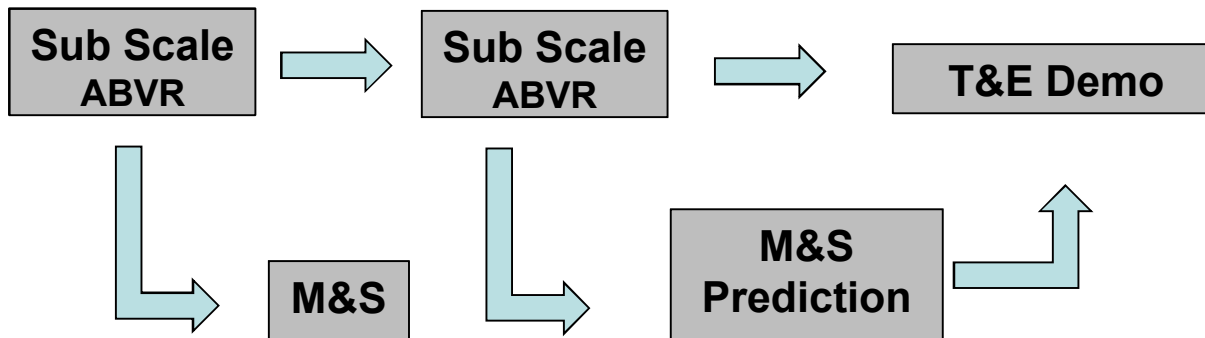
Jessica A. Stanfield, Jamie B. Neidert,  
Eric N. Harstad, Bradley W. White, and  
H. Keo Springer

- Full-scale rocket motor assets are expensive to test. The development of predictive tools to help predict/understand the response of propellants (non-ideal explosives) would lower overall cost and provide useful IM tools.
- Goal: Predictive capabilities for IM threats on energetics in representative systems

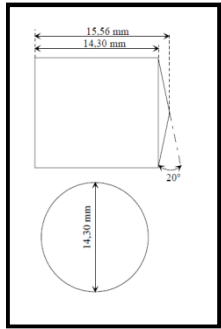
**ABVR (Army Burn to Violent Reaction)**- Sub scale fragment impact tests representing full scale; data provided for M&S; component tests performed for material characterization and model calibration

**M&S**- Modeling and simulation iterations to design a full scale fragment impact prediction tool; Integrated analog T&E Demo pre-test predictions

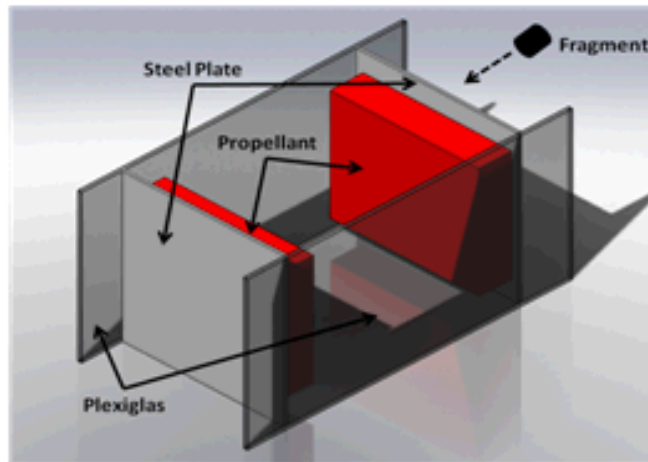
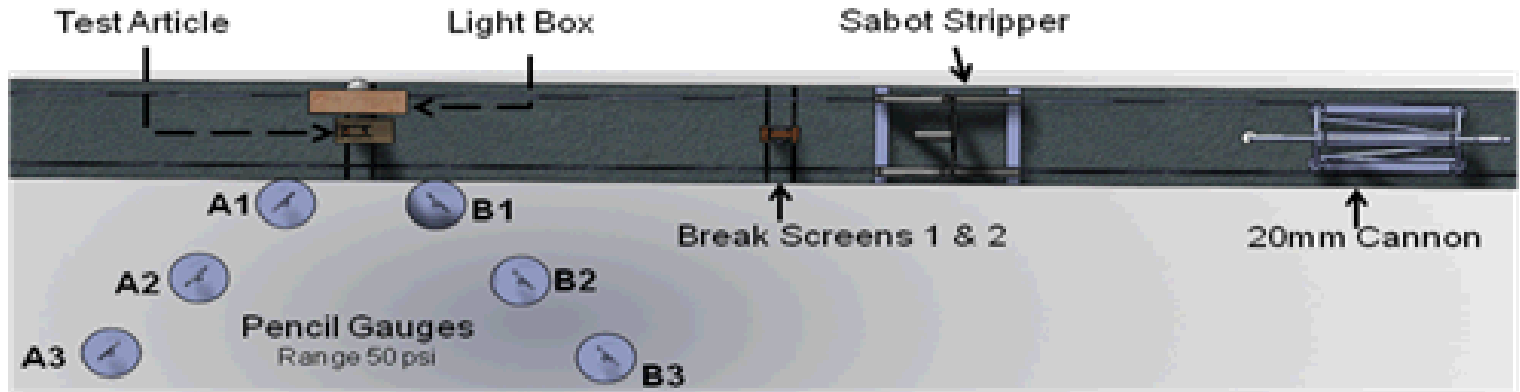
**Test and Evaluation (T&E) Analog Demo**– Full scale fragment impact test with analog rocket motor; Integrated analog T&E Demo test materials & test article fabricated, test range configured and test executed



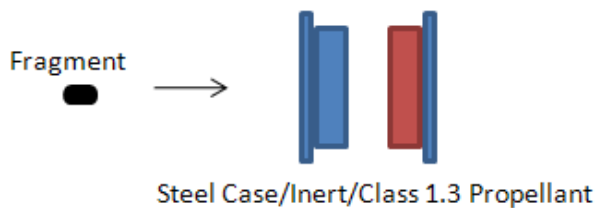
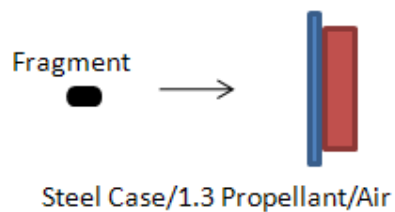
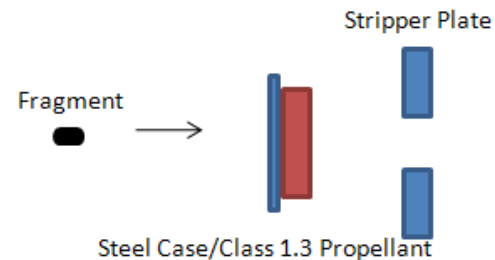
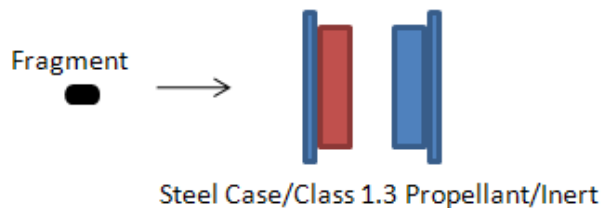
- High Performance Propellant (HPP)-Ammonium Perchlorate (AP) and aluminum powder bonded by hydroxyl-terminated butadiene

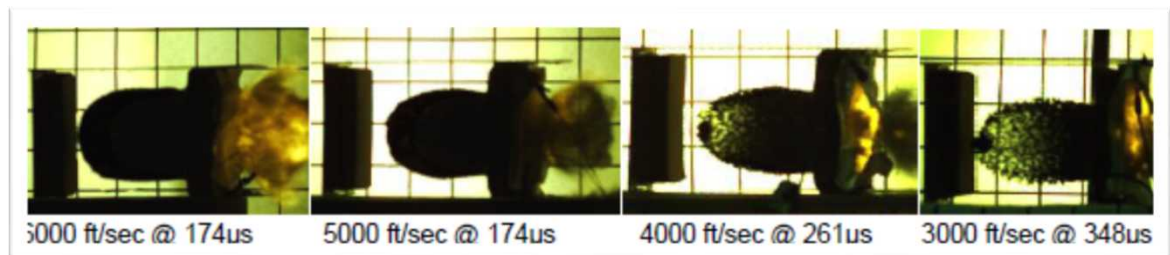
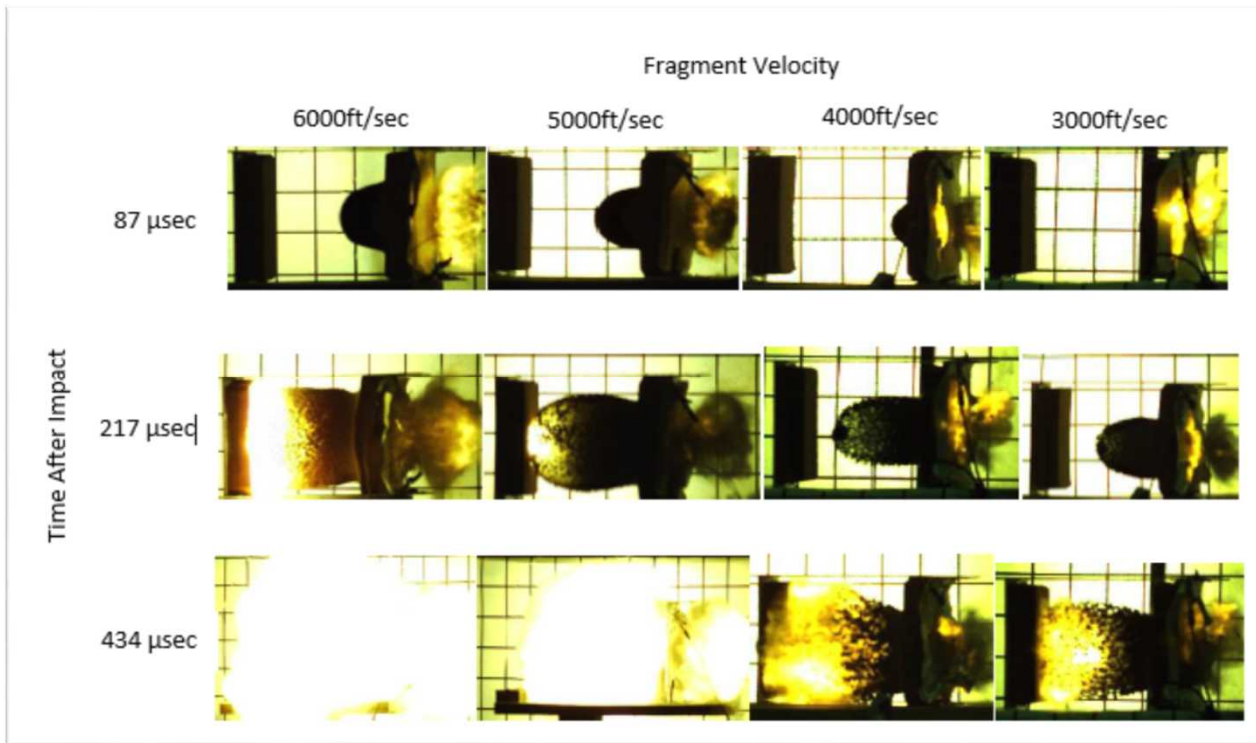


Fragment

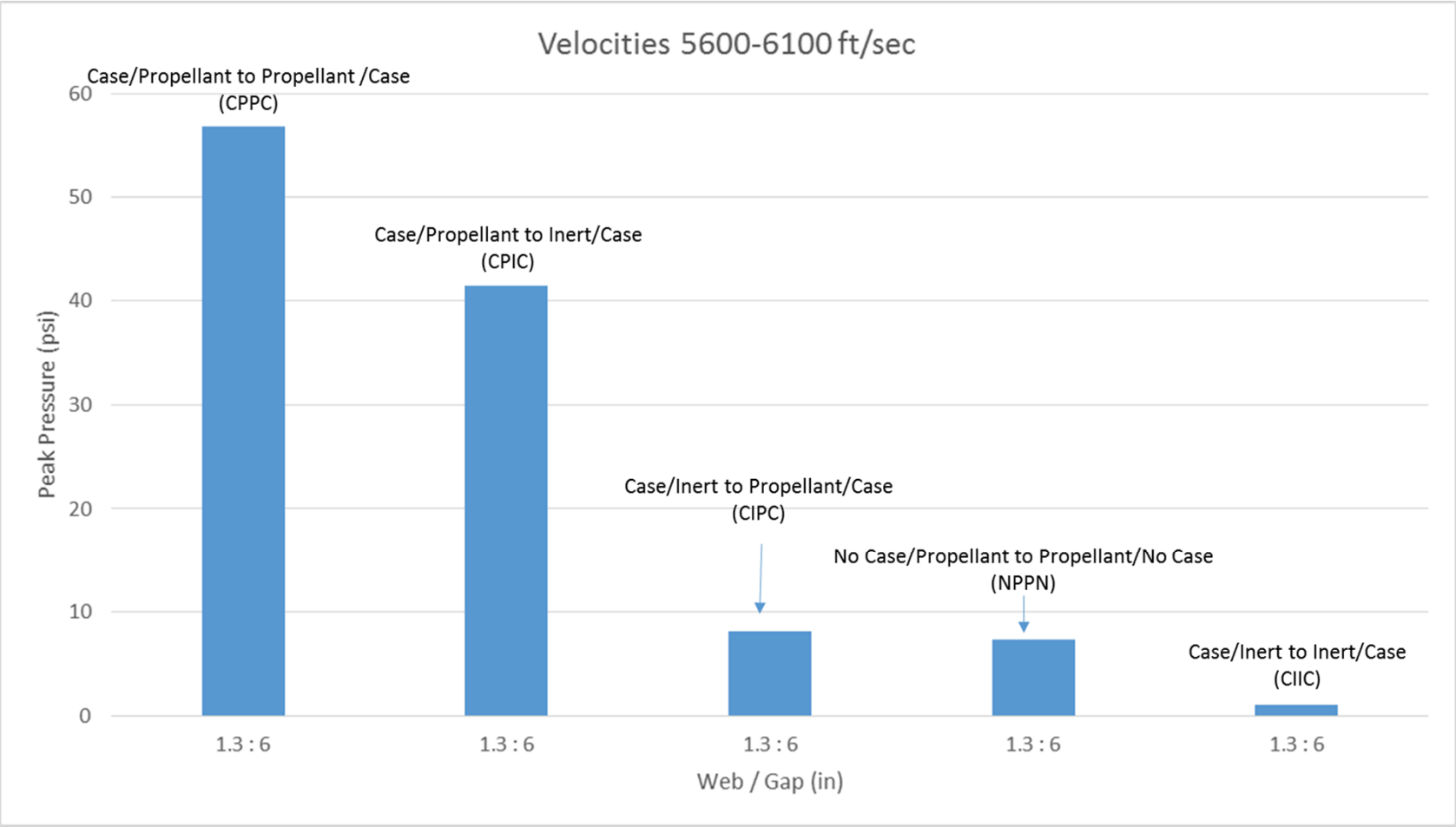


Standard Configuration









- Test articles represent the dimensions and materials used for T&E Demonstration.

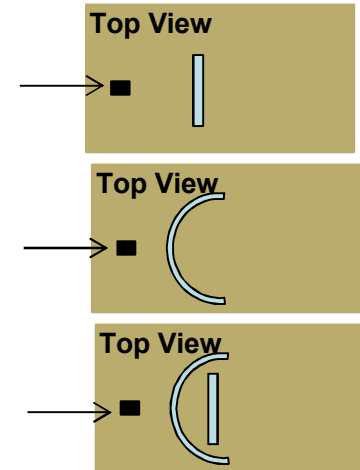
### -4 ABVR Tests

- Composite panels
- 3.65" thick HPP propellant slabs
- 1.5" Air Gap
- Fragment Impact Velocity (4000 to 6000ft/sec)

### -4 Inert Impact Tests

- Composite Panel (Bare)
- Composite Panel with Insulation
- Canister and Composite Panel (Bare) Canister
- Fragment Impact Velocity 6000ft/sec

- Supply data to modelers for T&E demonstration predictions



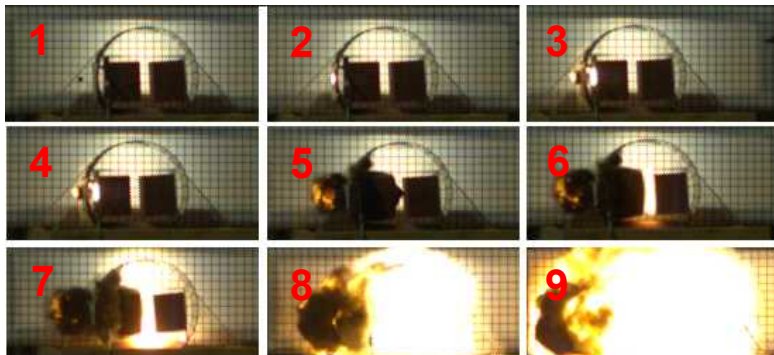


# ABVR Tests

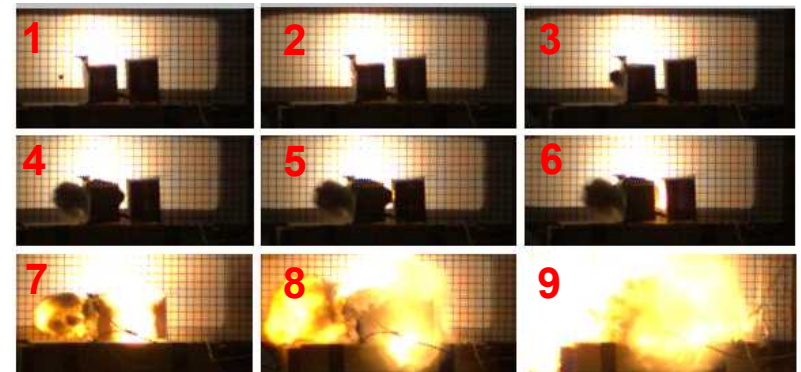
## Preparations for Analog T&E

- Reduction in velocity due to canister, case, and insulation material was significantly more than anticipated (see chart, 5-15% reduction).
- Increasing impact velocity increased pressure reading; with the exception of the added canister

Test Number	Canister	Composite Panel	Insulation	Test Article	Initial Impact Velocity, ft/sec	Velocity Reduction, %	Reaction Type	Peak Pressure, psi
1	X				6211	7	None	N/A
2		X			6374	5	None	N/A
3		X	X		6250	8	None	N/A
4	X	X			6179	15	None	N/A
5	X			X	6237	N/A	Burn	11
6				X	6217	N/A	Burn	26
7				X	5177	N/A	Burn	15
8				X	3993	N/A	Burn	4.5



Test 5

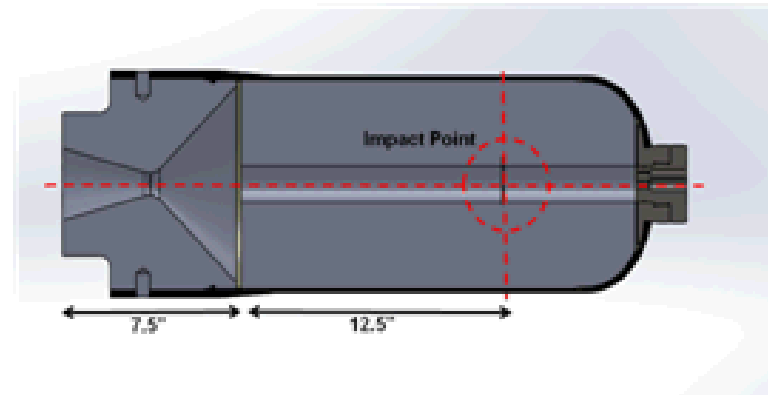
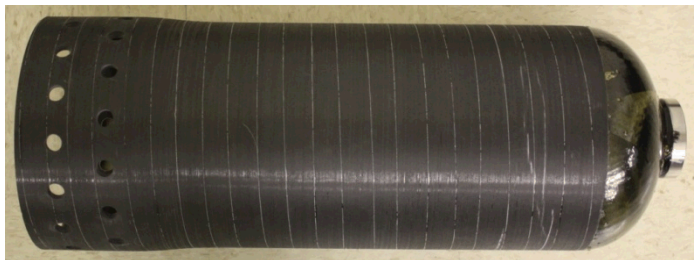


Test 6

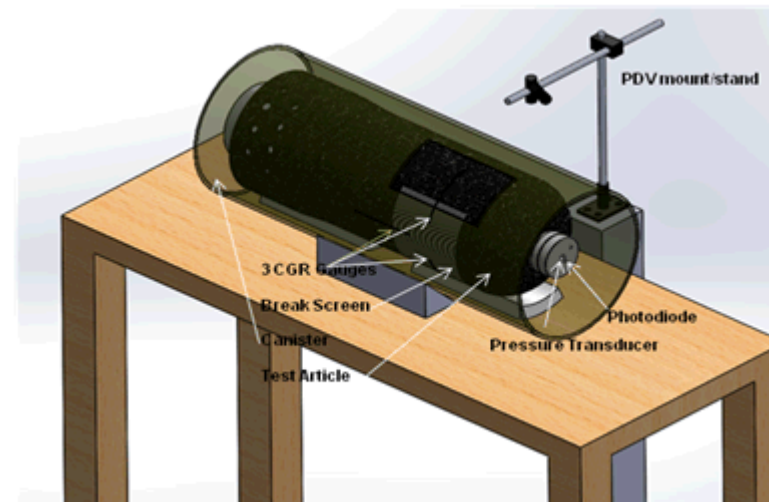
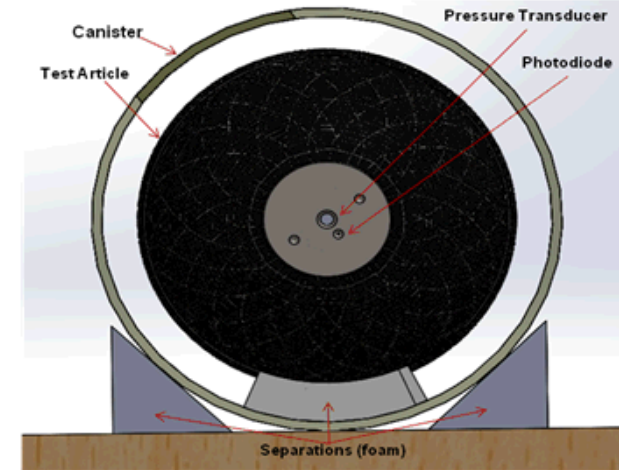


- **ALE3D multi-physics code with the PERMS reaction/burn model**
  - Arbitrary Lagrangian Eulerian Three Dimensional (ALE3D) code
  - Propellant Energetic Response to Mechanical Stimuli (PERMS) material model with Equivalent Plastic Strain (EPS)-enhanced burning parameters to explore reactivity
  - Used ABVR test results to calibrate the models for Demo tests
  - Performed sensitivity studies on model parameters due to uncertainties in the HPP fragmentation response and its central role in capturing reaction violence
- **CTH shock hydro-code with two propellant models**
  - Initial model was Coupled Damage and Reaction with Kinetics (CDAR-K) but was not well suited to HPP material
  - Propellant Model (PMOD) was used effectively starting in 2012
  - PMOD parameters calibrated from ABVR results for Demo tests
- **Material models for reactive & inert constituents were used extensively in both codes**
  - ABVR-related experiments helped team to better understand physics

Test Number	Planned Impact Velocity, ft/sec	Bore Dimension, in	Configuration	Test Description
1	8300	2	No Canister	Baseline
2	8300	4	No Canister	Bore Variation
3	8300	2	Canister	Canister Influence

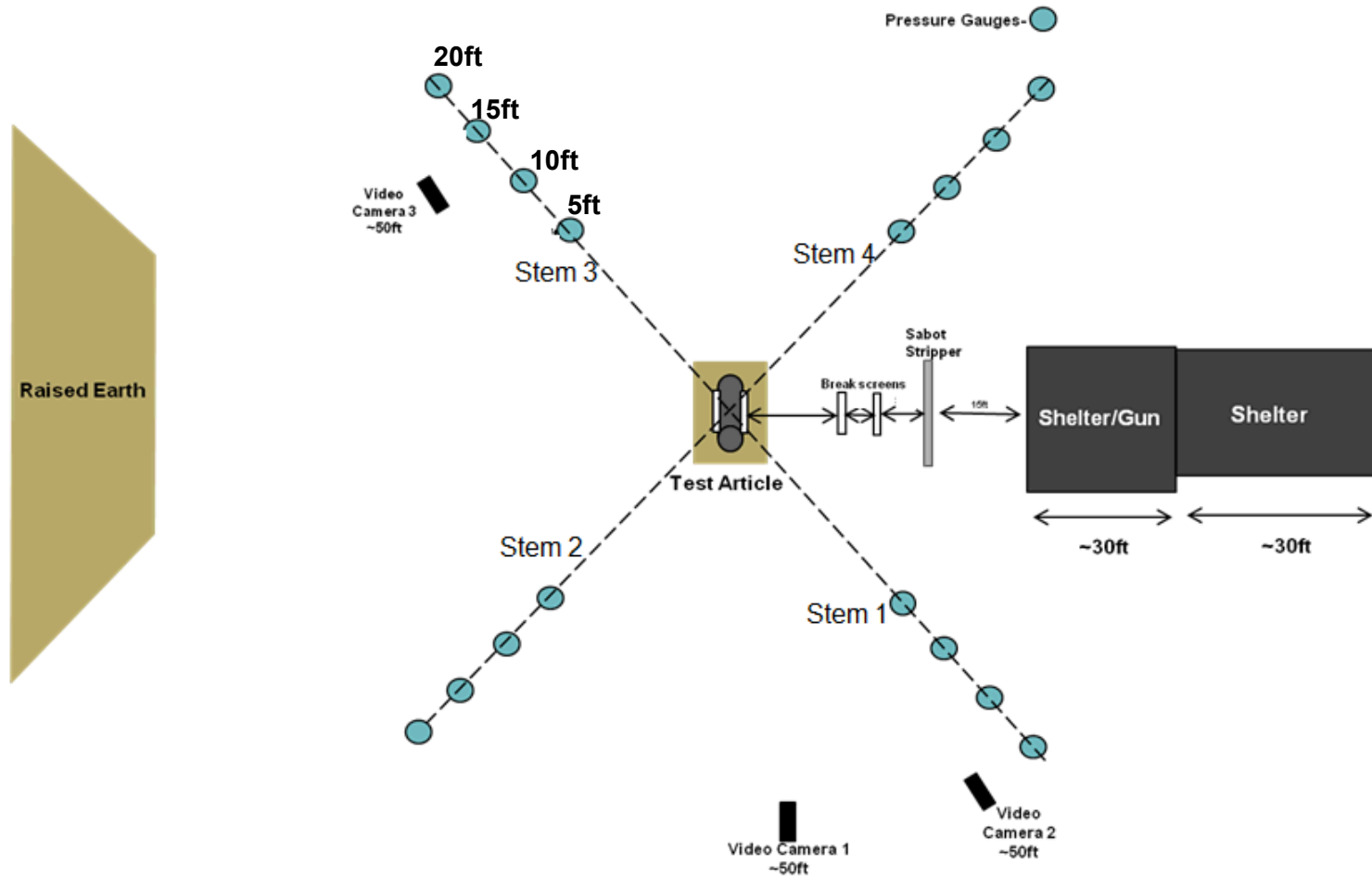


- In-Bore Pressure Transducer
- Open Air Over-Pressure (OP) Gauges
- Breakscreens (6) near and on Test Article
  - Measure Fragment Velocity,  $V_o$
  - Time,  $T_o$ , for Fragment Impact on Test Article
  - Time,  $T_f$ , for (potential) Fragment Exit
- Standard Video (3 views)
- High Speed Video (3 views)
- Still Photography
- Photodiode
- Photonic Doppler Velocimetry (PDV)





# HPP IM T&E Demonstration Test Setup







# IM Analog Demo Rocket Motor Test Data

Test	Velocity (ft/s)	Max.@5ft OP, (psi)	In-Bore Pressure, psi	Reaction Type
1	7989	12	>10K	IV
2	8399	20	>10K	IV
3	8279	11	8400	IV

Test Number	Description	Distance, ft	Location, degree
1	Aft End of Motor	22	220
1	Dome and Case Material	35	60
1	Propellant and Case Material	95	50
1	Firebrand	249	225
1	Forward Closure	300	20
2	Aft End of Motor	33	170
2	Firebrand	230	215
2	Case Material	2	225
3	Motor and Canister (minus Forward End Cap)	N/A	N/A

# Test Data to Post-test M & S Results Comparison

	Case Velocity (PDV 90° probe) (ft/s)	Photodiode (time to 1 <sup>st</sup> light) (µsec)	Max. OP, Stem 4 (psi)	In-bore pressure (psi )	Penetration through Test Article
<b>Test 1</b>					
Test data	43	No Data	12 at 5ft 5 at 10ft	>10K	No
CTH	140	N/A	16 at 5ft	40K	No
ALE3D	590	110	N/A	25K	No
<b>Test 2</b>					
Test data	No Data	No Data	20 at 5ft 9 at 10ft	>10K	Unknown
CTH	100	N/A	32 at 5ft	13.5K	Yes
ALE3D	295	102	N/A	13.6K	Yes
<b>Test 3</b>					
Test data	7.5	213	11 at 5ft 5 at 10ft	8400	Unknown
CTH	75	260	N/A	6700	No
ALE3D	280	165	N/A	6000	No

- **ABVR tests provided useful velocity, pressure, and visual data to make pre-test prediction simulations for the analog demonstration rocket motor IM tests**
- **ABVR tests provided velocity reduction information**
  - **Canister reduced fragment speed by approx. 7% (from ABVR)**
  - **Composite with insulation reduced fragment speed by approx. 8% (from ABVR)**
- **Pre-test predictive simulations of the analog demonstration rocket motor tests suggested bore size would influence the violence of the reaction**
  - **Bore size did influence violence of the reaction**
  - **As anticipated, data confirmed a more violent reaction for the larger bore diameter**
- **Pre-test prediction modeling was important to the analog demo RM design and the test matrix**

- Canister appeared to mitigate the reaction of the motor to fragment impact
- Placement of over pressure gauges closer to target was important to provide meaningful data as suggested by simulations
- 10K in-bore pressure gauge was not rated high enough for actual pressures
- Placement and type of break screens is critical to accurate time and velocity measurements
- Refined post-test ALE3D and CTH model simulations provided values that were improvements compared to the original predictions
  - Gaps in the test data and needed improvements in the M&S technology
  - Further experimental work and modeling enhancements are needed to continue to evolve predictive capabilities

- **Lawrence Livermore –H. Keo Springer, Lara Leininger, and Tony Whitworth**
- **Los Alamos -Thomas Mason and Paul Butler**
- **Sandia - Eric Harstad, Ken Chavez, and Michael Kaneshige**
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- **AMRDEC Aerospace Materials function-Cheryl Steele and Robert Esslinger**
- **AMRDEC Missile Sustainment- Justin Grissim**
- **AMRDEC Propulsion Technology- Joey Reed, Bill Delaney and Brian Curtis**
- **RTC- Jerry Webb and Justin Merritt**
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