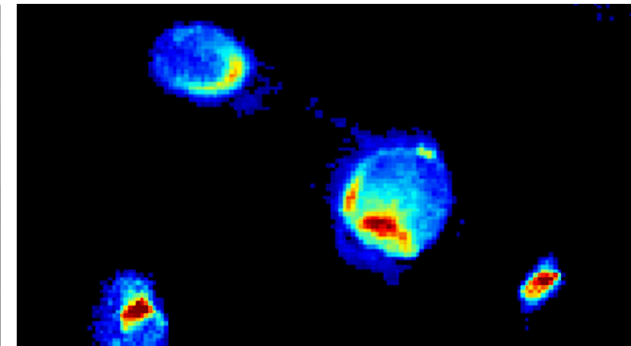
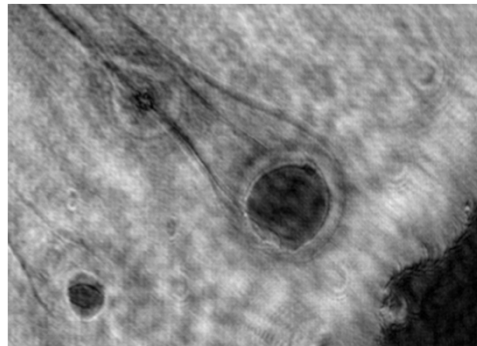
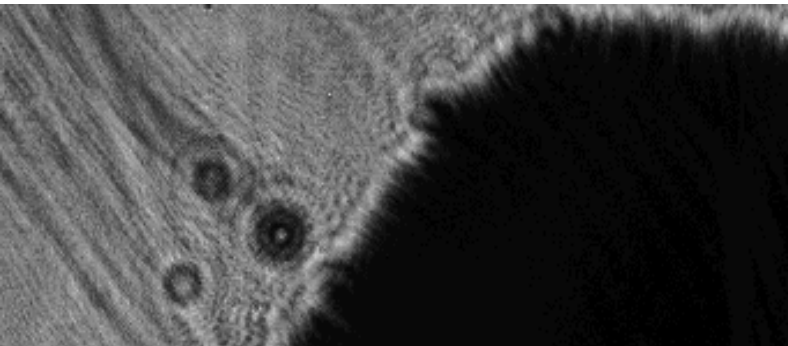


Exceptional service in the national interest



Digital Imaging Holography and Pyrometry of Aluminum Drop Combustion in Solid Propellant Plumes

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Presentation LT4F.2



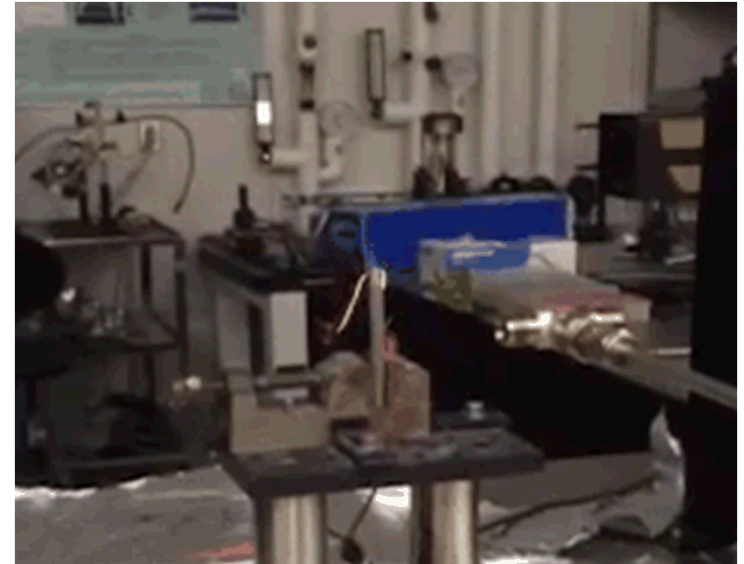
Sandia National Laboratories is a multi-program laboratory managed and operated by Sandia Corporation, a wholly owned subsidiary of Lockheed Martin Corporation, for the U.S. Department of Energy's National Nuclear Security Administration under contract DE-AC04-94AL85000. SAND NO. 2011-XXXXP

Motivation

<http://www.cbsnews.com/news/rocket-crash-no-immediate-threat-to-station-but-cause-is-unknown/>



Color video of burning propellant



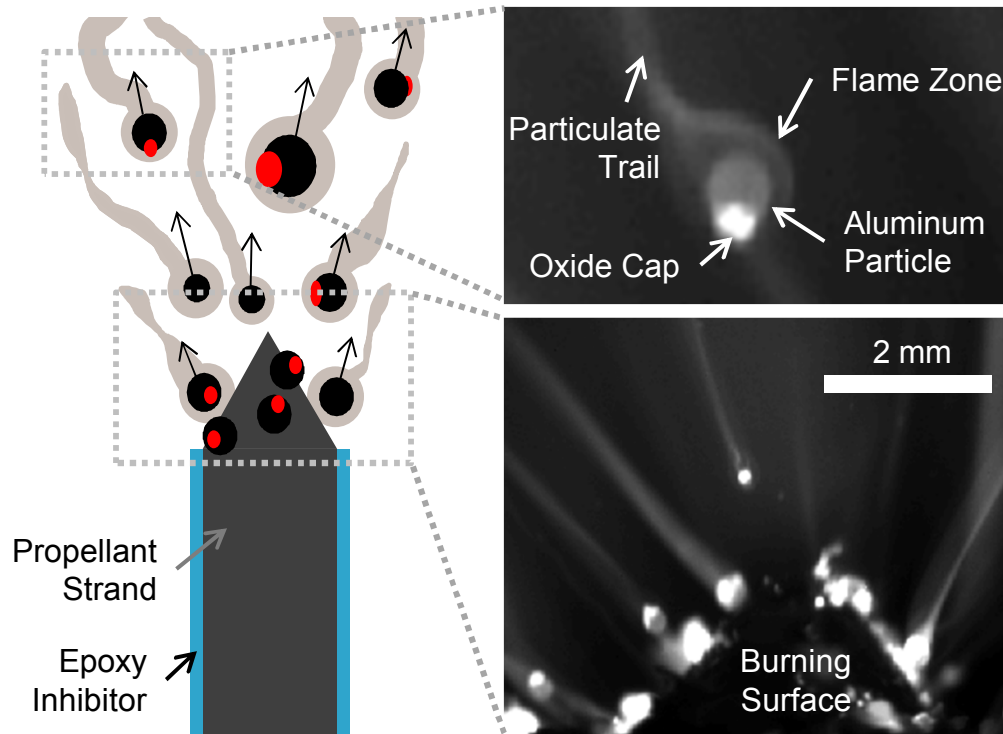
Problem:

- Rocket failures can lead to propellant fires
- Aluminum agglomeration at the surface yields large reacting drops with high damage potential
- Prediction requires knowledge of particle ***size, position, and temperature***

Goals:

- Making measurements in real propellant sticks
- Measuring the size of small particles that are out-of-focus

Approach



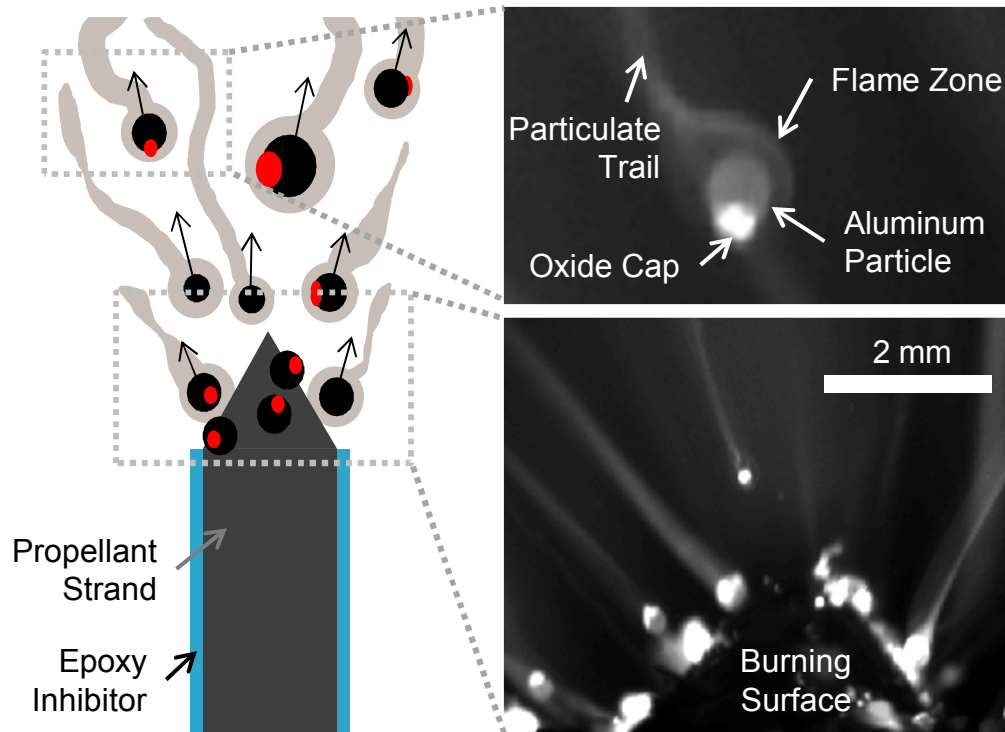
Propellant

- 6 mm diameter, 50 mm long
- 70 wt.% ammonium perchlorate oxidizer (AP)
- 20 wt.% aluminum particulate
- 10 wt.% inert hydroxyl terminated polybutadiene binder (HTPB)
- Epoxy inhibitor outer coating
- Burn rate ~ 1.4 to 1.6 mm/s

Combustion Process

- Propellant ignited with hot nichrome wire
- AP melts at 830 K, adiabatic flame temp 1205 K
- Al_2O_3 creates a flame holding effect, raising the flame temperature by ~ 1000 K
- Some Al particles (~ 30 μm) and Al_2O_3 shells melt at 2345 K
- These agglomerate into balls (~ 100 to 500 μm) and oxidizes to Al_2O_3 (caps and smoke) producing a flame zone (Al_2O_3 vaporizes at 3240 K)

Approach



High speed video of burning propellant

- Particle Size and Position?
- Particle Velocity?
- Particle Temperature?
- Gas Temperature and Composition?

Digital In-Line Holography (DIH)

Double Frame DIH

Two-Color Imaging Pyrometer

Coherent Anti-Stokes Raman
Scattering (CARS)

DTh2E.3 (Thurs @ 12:15 PM)

D. Guildenbecher and P. Sojka

Digital in-line holography (DIH) to quantify the impact
of a viscous drop on a thin film

LW5G.3 (Wed @ 5:45 PM)

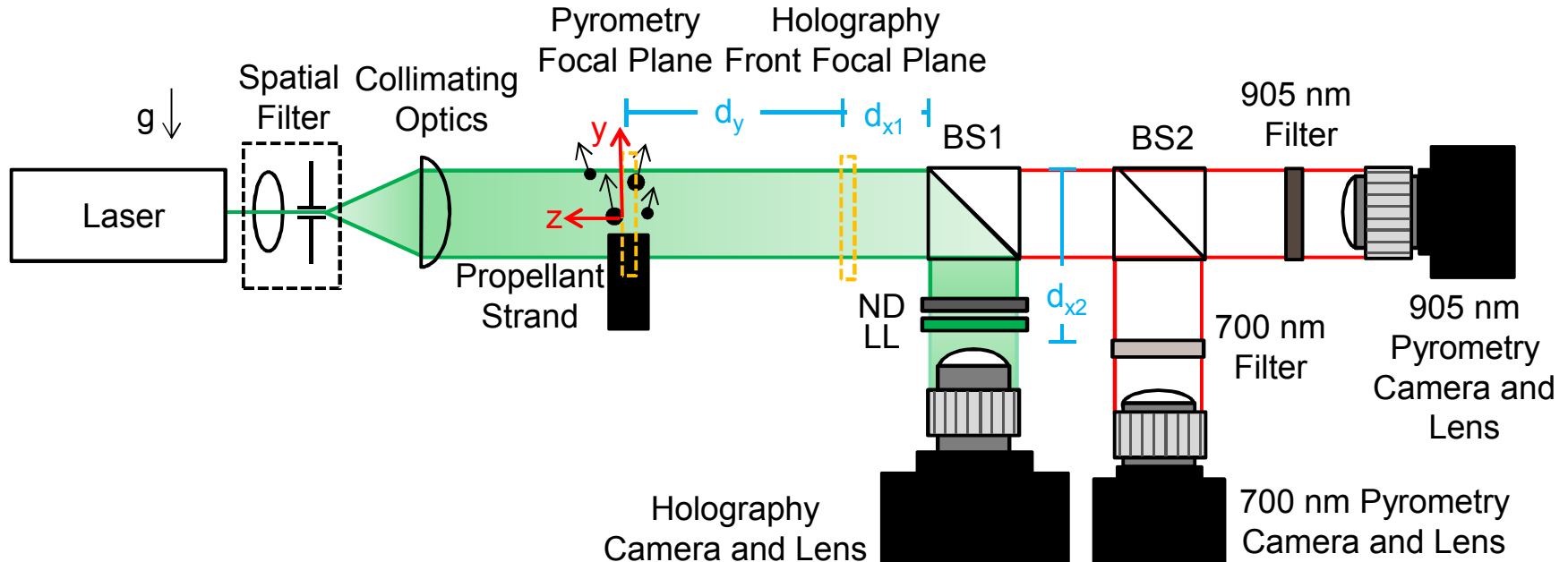
S. Kearney and D. Guildenbecher

Temperature and oxygen measurements in a
metallized propellant flame by hybrid fs/ps rotational
coherent anti-Stokes Raman scattering

Experimental Setup

Digital In-Line Holography

Two Color Pyrometer



- $d_x = d_{x1} + d_{x2} = \sim 170$ mm, $d_y = 50$ mm
- Double-pulsed ND:YAG laser (Continuum MiniLite PIV, 532 nm, 5 ns pulse duration, 40 μ s pulse separation)
- Holography camera LaVision sCMOS (2560x2160 pixels, 6.5 μ m pixel pitch, 16 bit depth, global shutter mode) and lens (Infinity K2 DistaMax with CF2 objective)
- Pyrometer cameras LaVision ProX 4M (2048x2048 pixels, 7.4 μ m pixel pitch, 14 bit depth, global shutter mode) and lens (Navitar Zoom 7000 lenses, aperture full open)
- ND – Neutral Density Filter (OD2)
- LL – Laserline filter at 532 nm with bandwidth of 1 nm (Andover 532FS02-50)
- BS1 – Beam splitter, reflection @ 532 nm and transmission @ 650 to 900 nm
- BS2 – Beam splitter, reflection @ 650 to 750 nm and transmission @ 850 to 950 nm
- 700 and 905 nm filters with bandwidth of 10 nm

Digital In-Line Holography

Light propagation is described by the diffraction integral equation:

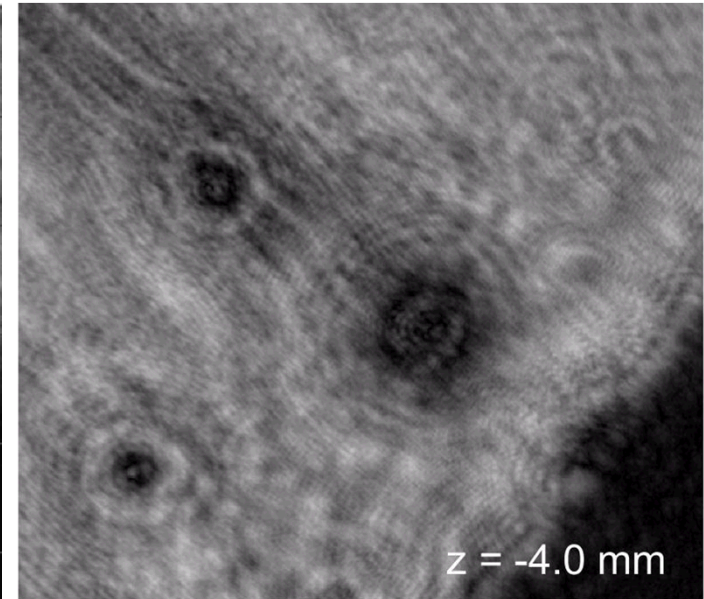
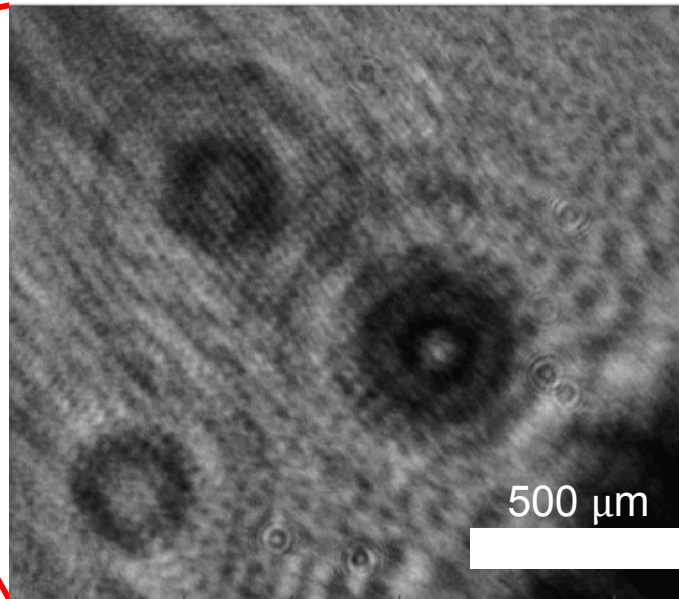
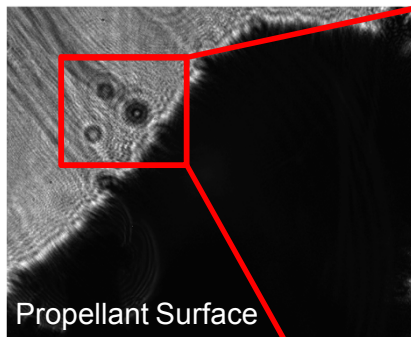
$$E(x, y, z) = \frac{1}{\lambda} \iint E(\xi, \eta, z=0) \frac{e^{-jkr}}{r} d\xi d\eta \quad \text{where: } r = \sqrt{(\xi - x)^2 + (\eta - y)^2 + z^2}$$

$E(x, h, 0) \equiv$ complex amplitude at hologram plane = $h(x, h) \cdot E_r^*$

$E(x, y, z) \equiv$ refocused complex amplitude at optical depth z

Raw Hologram

Refocused Hologram



Digital In-Line Holography

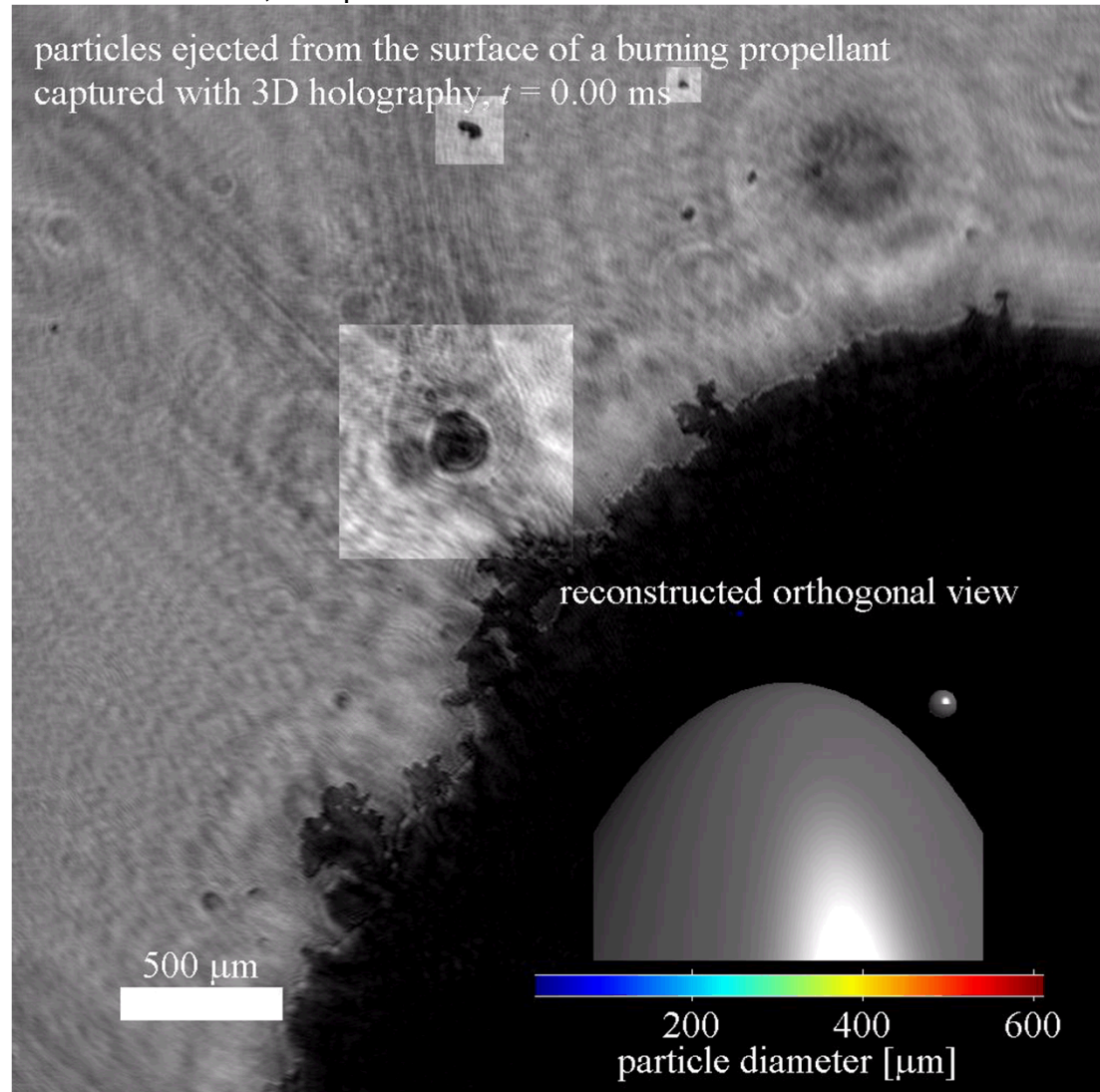
Recorded at 20,000 fps on Photron SA-Z with Coherent Verdi V6 Laser

Calibration

- Use a 1mm spacing dot grid
- Dewarp the images on 3 cameras to the same field of view

Detection Methods

- Find particle z-position by minimizing intensity and maximizing edge sharpness across a series of z-slices
- Find particle diameter by counting the number of dark pixels (minimum $d = 11 \mu\text{m}$)



D. Guildenbecher et. al., Applied Optics Vol. 55 (11), 2016.

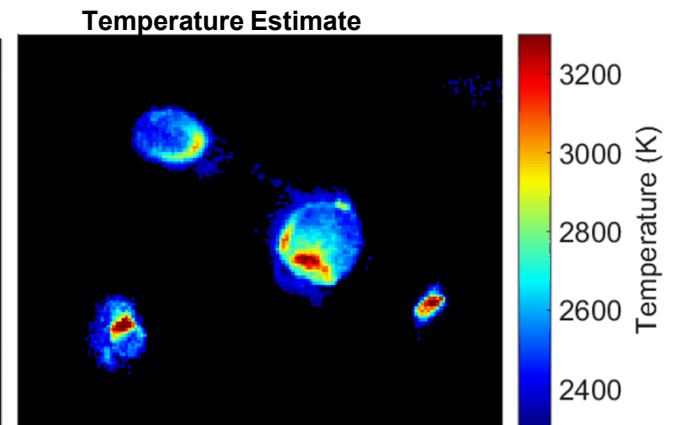
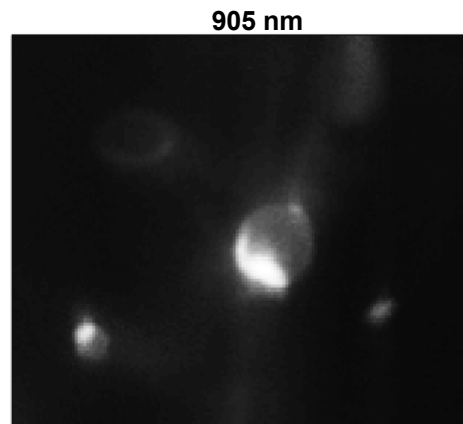
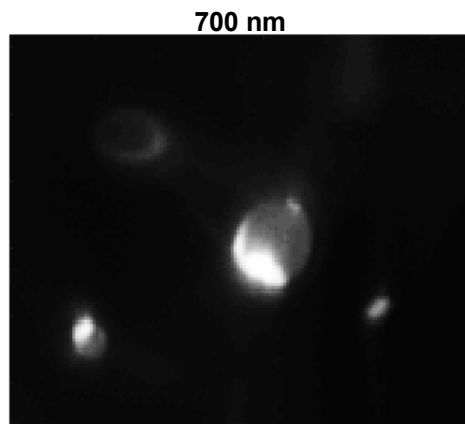
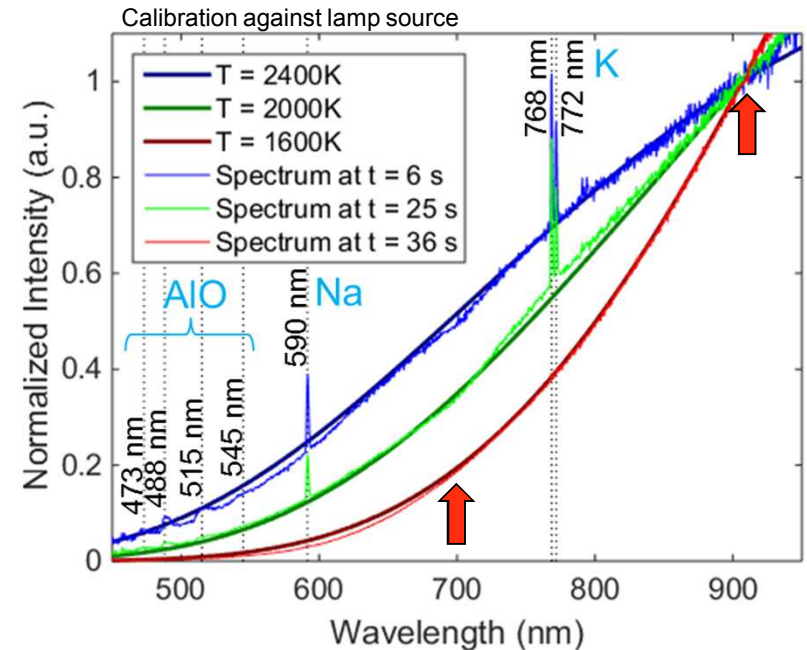
Two-Color Pyrometer

- Can use Plank's Law, assuming grey body emission
- Choose wavelengths to avoid combustion peaks
- If $hc/\lambda \gg kT$, use Wein's approximation for each pixel:

$$T = \left[\frac{k}{hc} \frac{\lambda_1 \lambda_2}{\lambda_2 - \lambda_1} \left(\ln(R) - 5 \ln \left(\frac{\lambda_1}{\lambda_2} \right) \right) \right]^{-1}$$

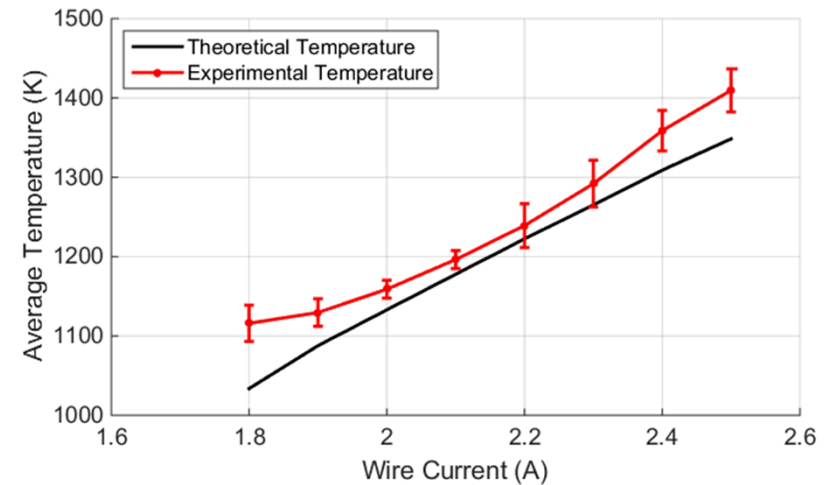
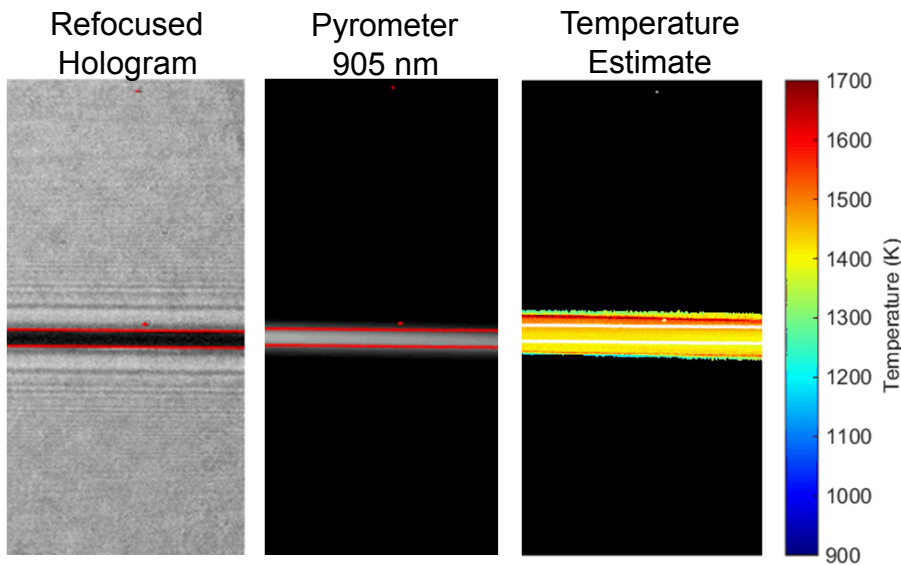
$$\text{where } R = \frac{I_2}{I_1} \frac{E_1 \eta_1}{E_2 \eta_2}$$

- Determine the matching pixels from DIH camera to pyrometer cameras (low intensity pixels removed)

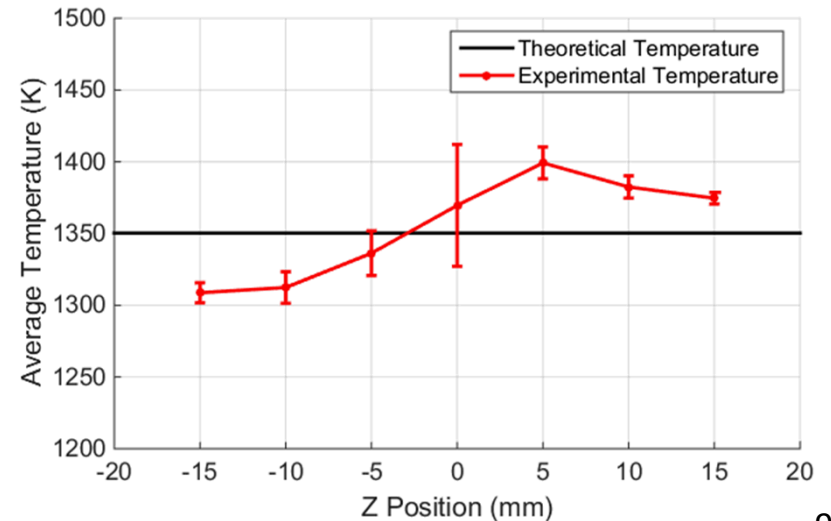


Calibration against Princeton Instruments LED source

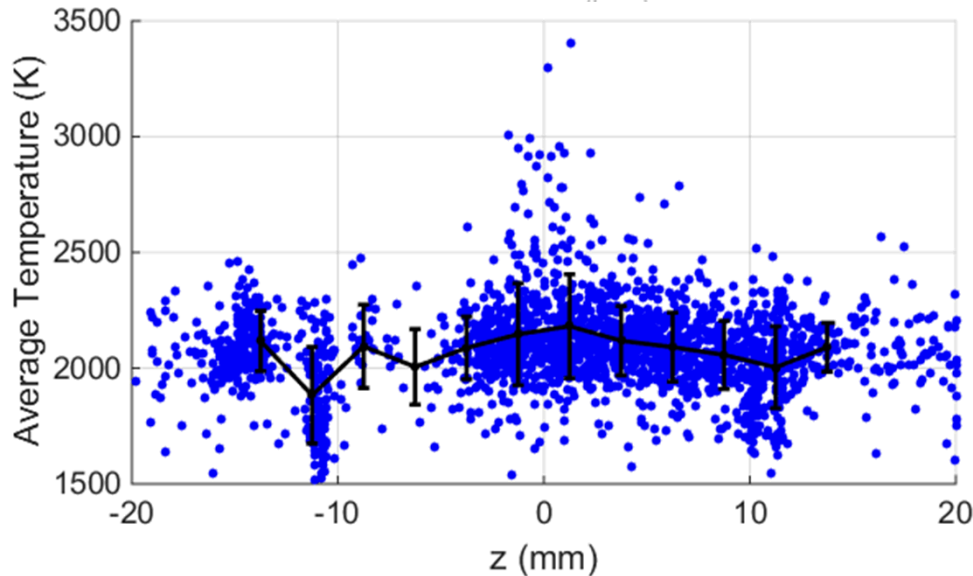
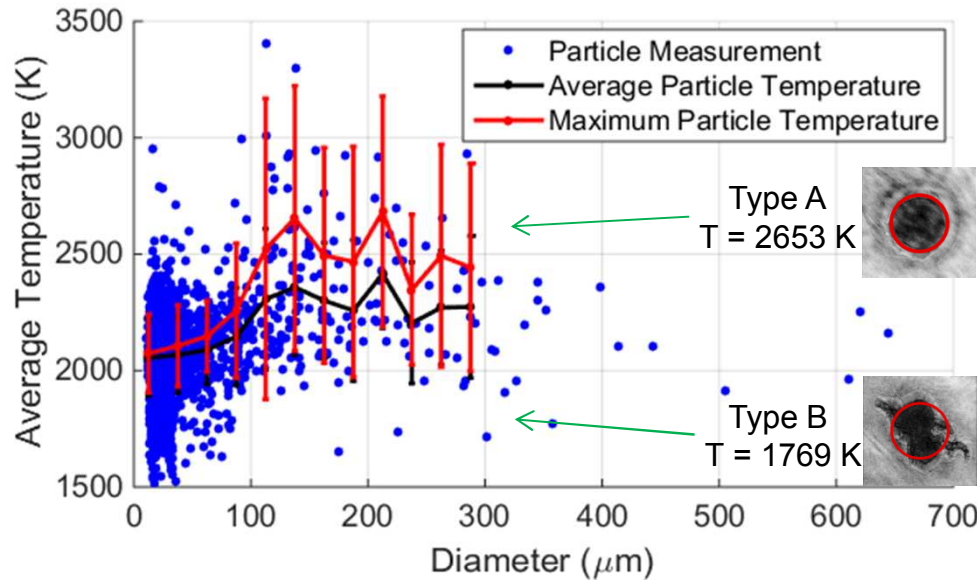
Pyrometer Calibration



- Checked temperature calibration against a 240 μm hot nichrome wire
- Determined the effect of focus on temperature estimate
 - In-focus features have more temperature variation
 - Out-of-focus features have less variation
 - Overall temperature variation ± 50 K



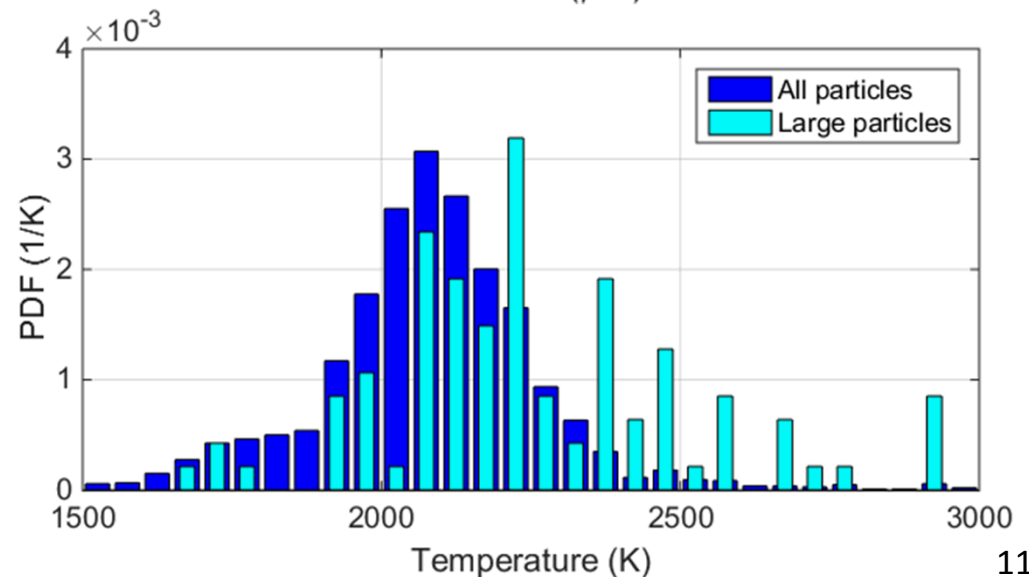
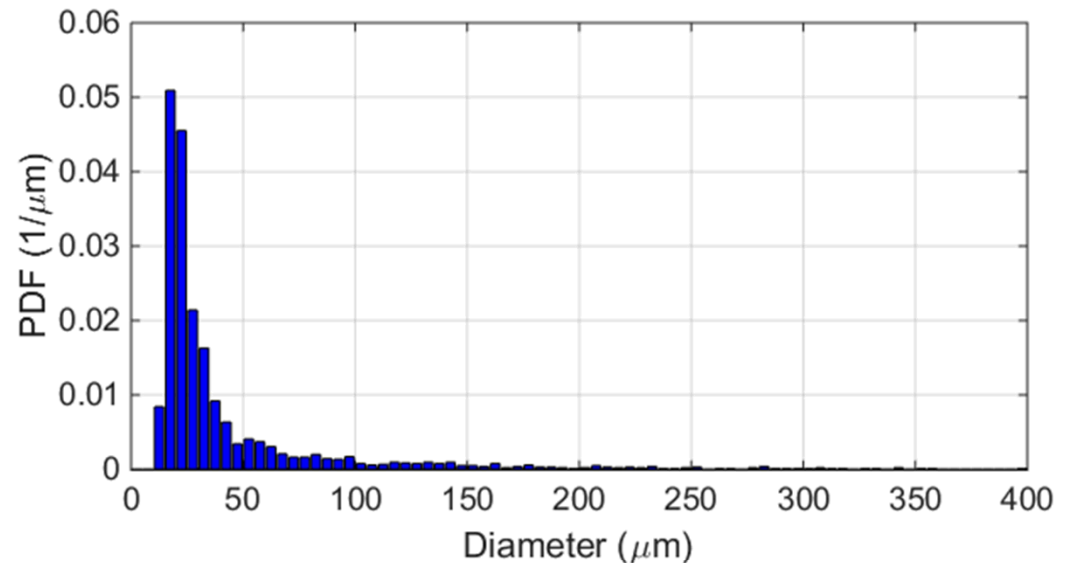
Measured Statistics



- One propellant stick burn with 2120 particles across 143 frames
- Sampled at 14 fps, each frame is uncorrelated
- Temperature of small particles approaches the temperature of the gas quickly
- Average particle temperatures are less than boiling point for Al (2792 K)
- Maximum particle temperatures less than boiling point for Al_2O_3 caps (3240 K)
- Temperature variation near the focal point is the greatest,
 - Similar to the calibration
 - May be evidence of hotter particles in the center of the flame

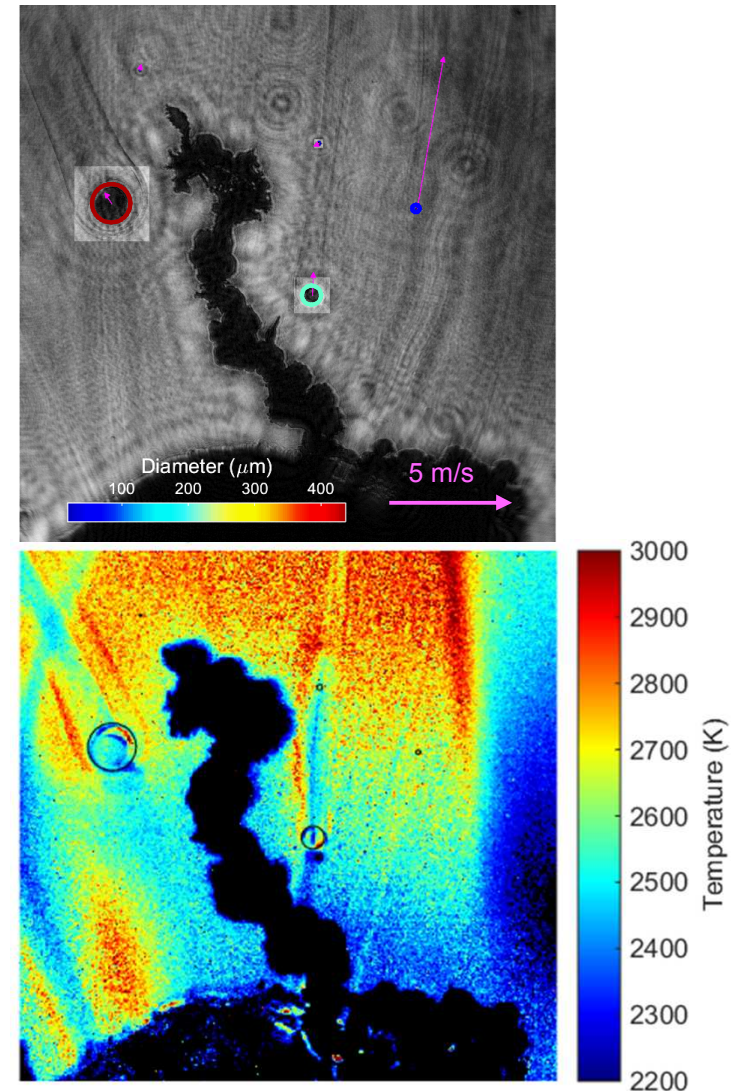
Measured Statistics

- The majority of the particles are small
- $D_{10} = 43.0 \mu\text{m}$
(mean diameter)
- $D_{32} = 249 \mu\text{m}$
(Sauter or surface area weighted)
- $D_{43} = 377 \mu\text{m}$
(volume weighted)
- Particle temperatures are a complex mixture of particles in different stages of combustion
- Since most of the particles measured are small, we expect average temperatures to be similar to gas temperatures
- For all particles
 $T_{\text{ave}} = 2084 \pm 189 \text{ K}$
- For large particles ($>150 \mu\text{m}$)
 $T_{\text{ave}} = 2131 \pm 256 \text{ K}$
- Similar to average gas temperatures measured with CARS



Conclusions and Future Work

- Demonstrated simultaneous DIH and Two-Color Pyrometer measurements of aluminized ammonium perchlorate propellant sticks
 - Position
 - Size
 - Temperature
- Sources of error and bias
 - Temperature calibration accuracy
 - Camera readout smear for short exposures
 - Gray body assumption
 - Overlap of particles and soot at different temperatures
- Future work
 - Additional propellant stick burns for better statistics
 - Velocity statistics to correct probability distributions

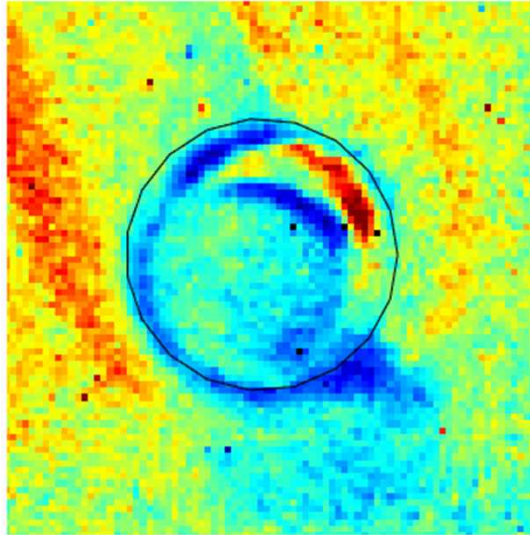


Acknowledgements

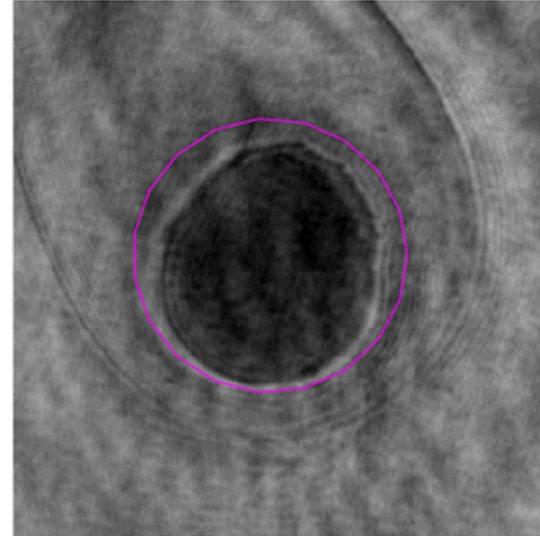
- The Weapons Systems Engineering Assessment Technology program
- The Laboratory Directed Research and Development program
- Howard Lee Stauffacher for his work with the propellant igniter system
- Thomas W. Grasser for his work with the optical system construction



Temperature Estimate



Refocused Hologram



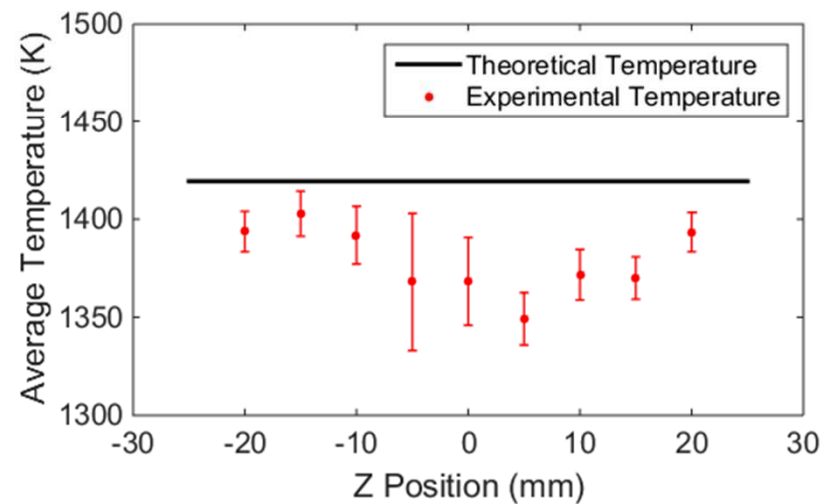
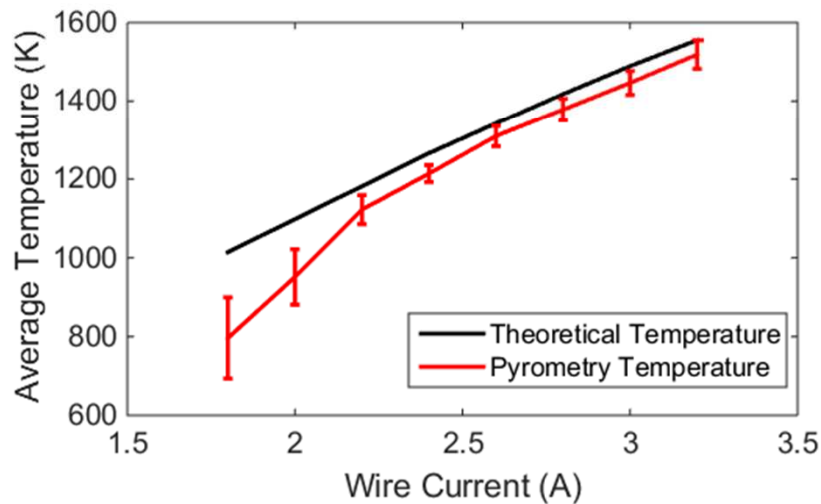
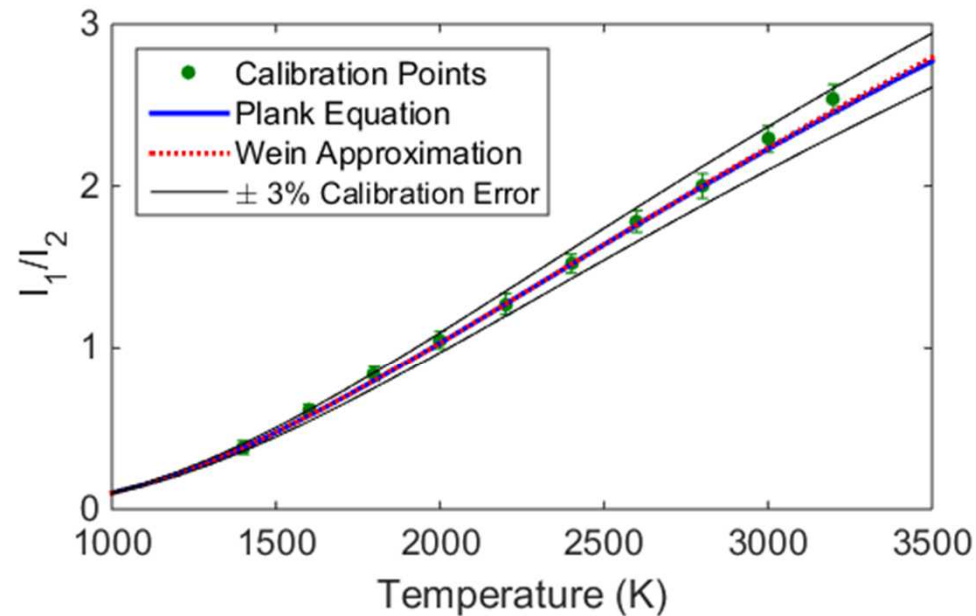
QUESTIONS?

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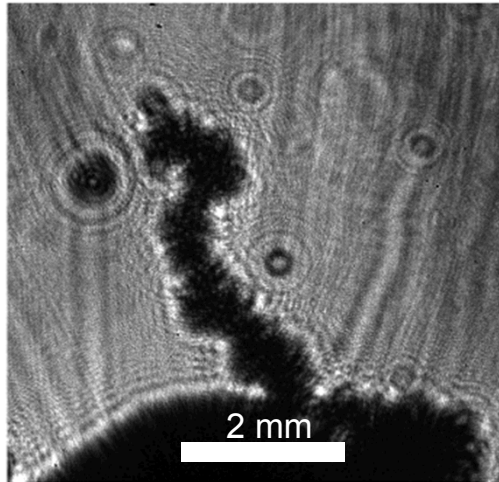
BACKUP SLIDES

Black Body Calibration

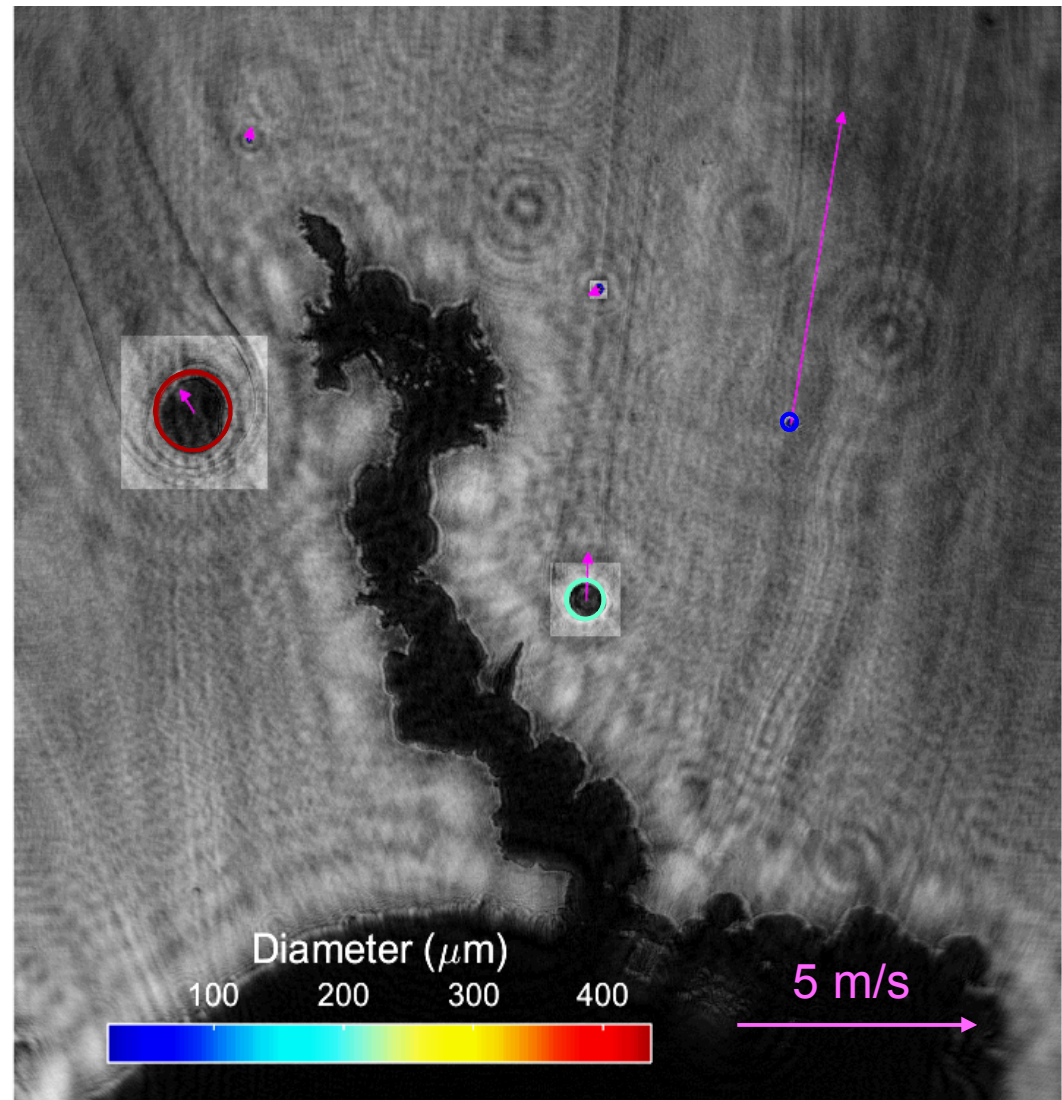
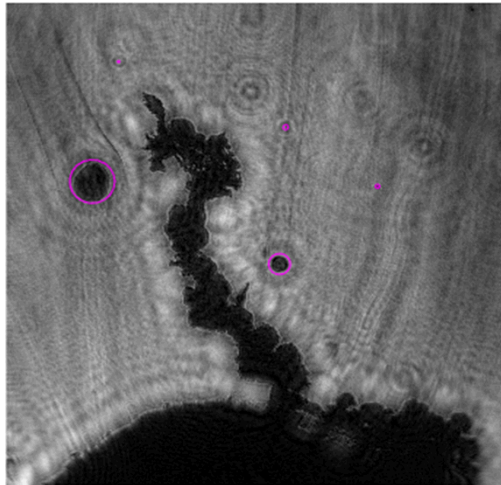


Velocity Measurement

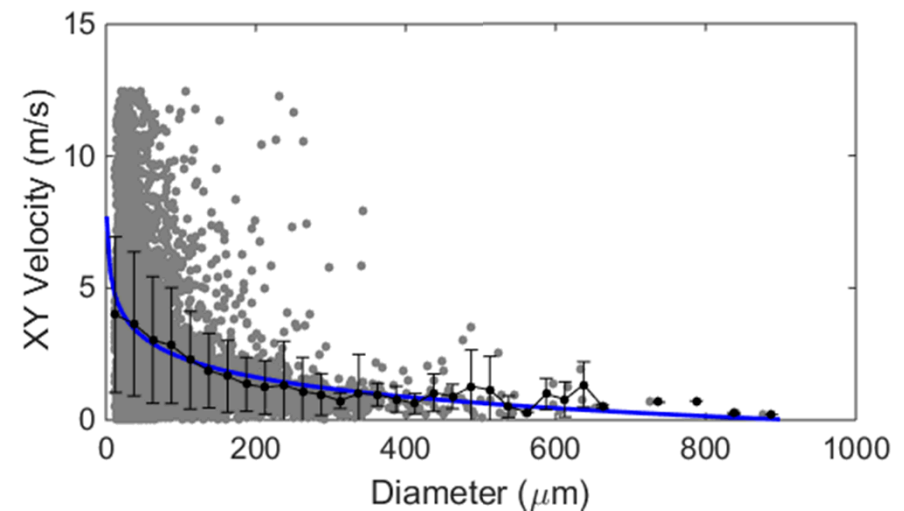
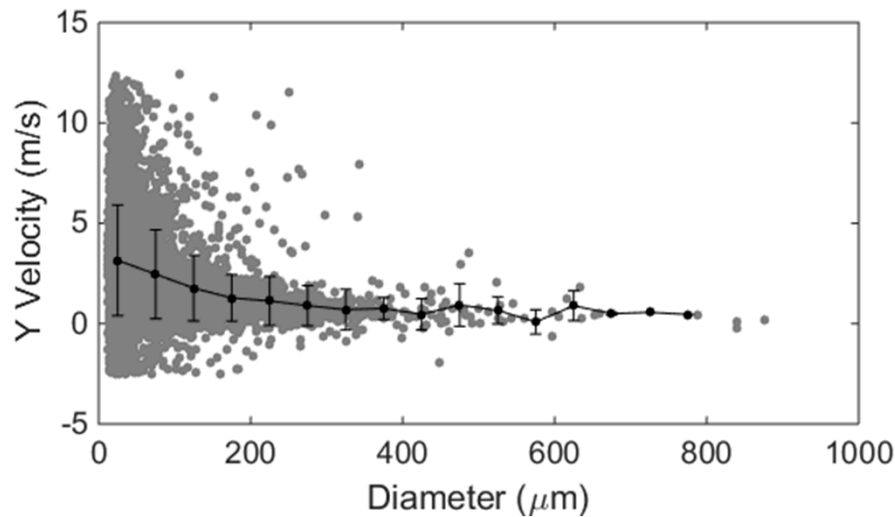
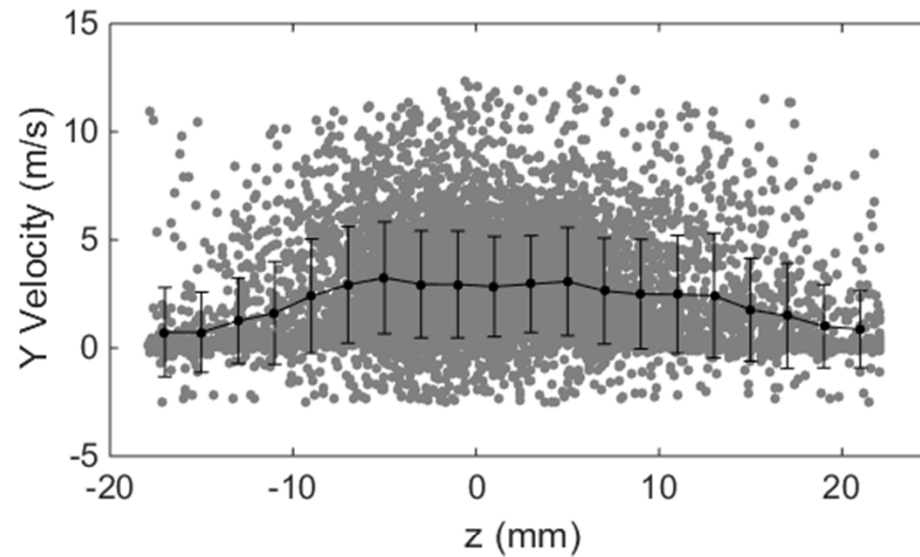
Raw Hologram



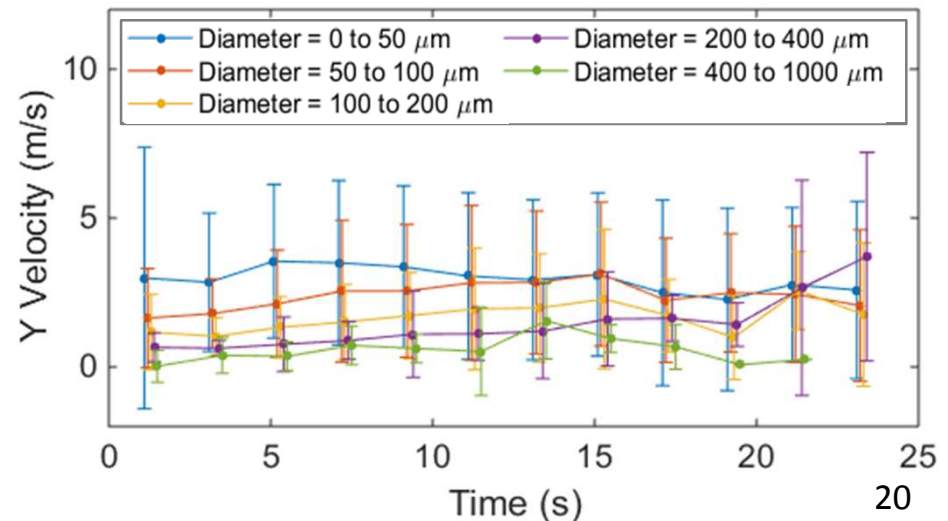
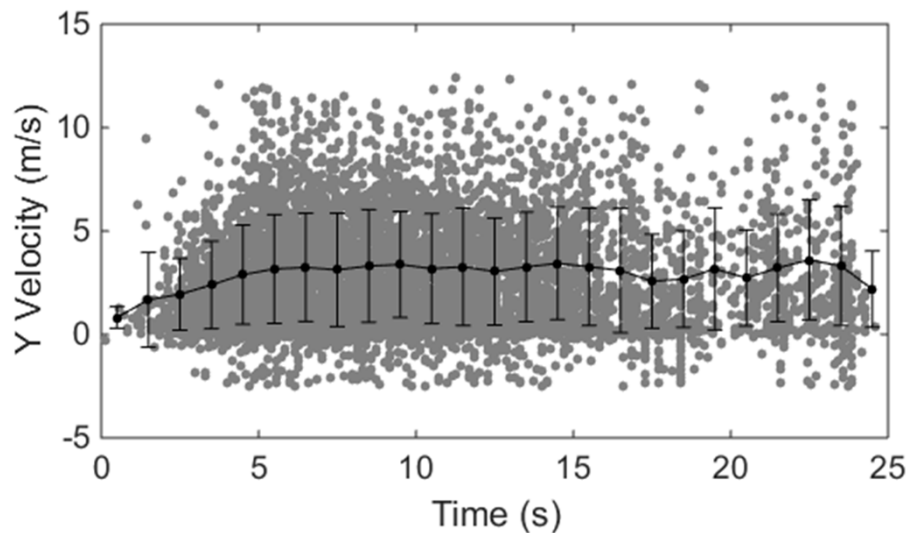
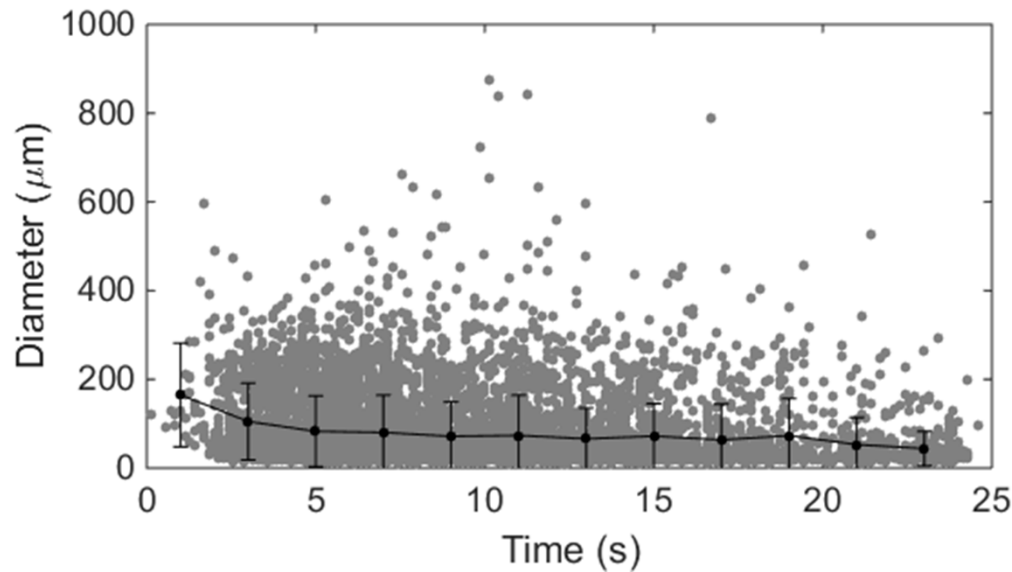
Refocused Hologram



Velocity Measurement

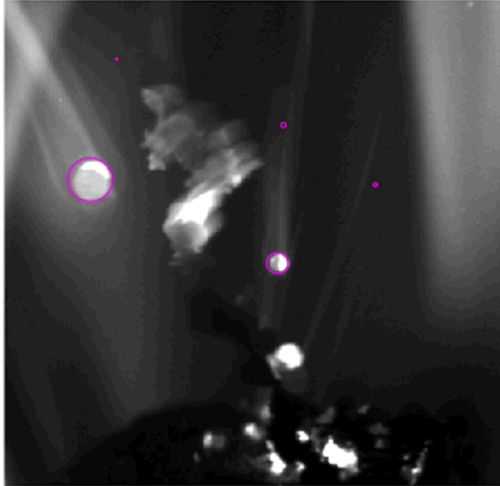


Time Evolution

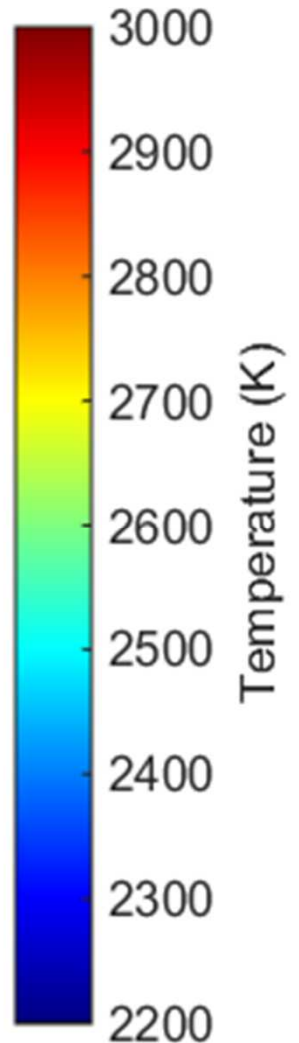
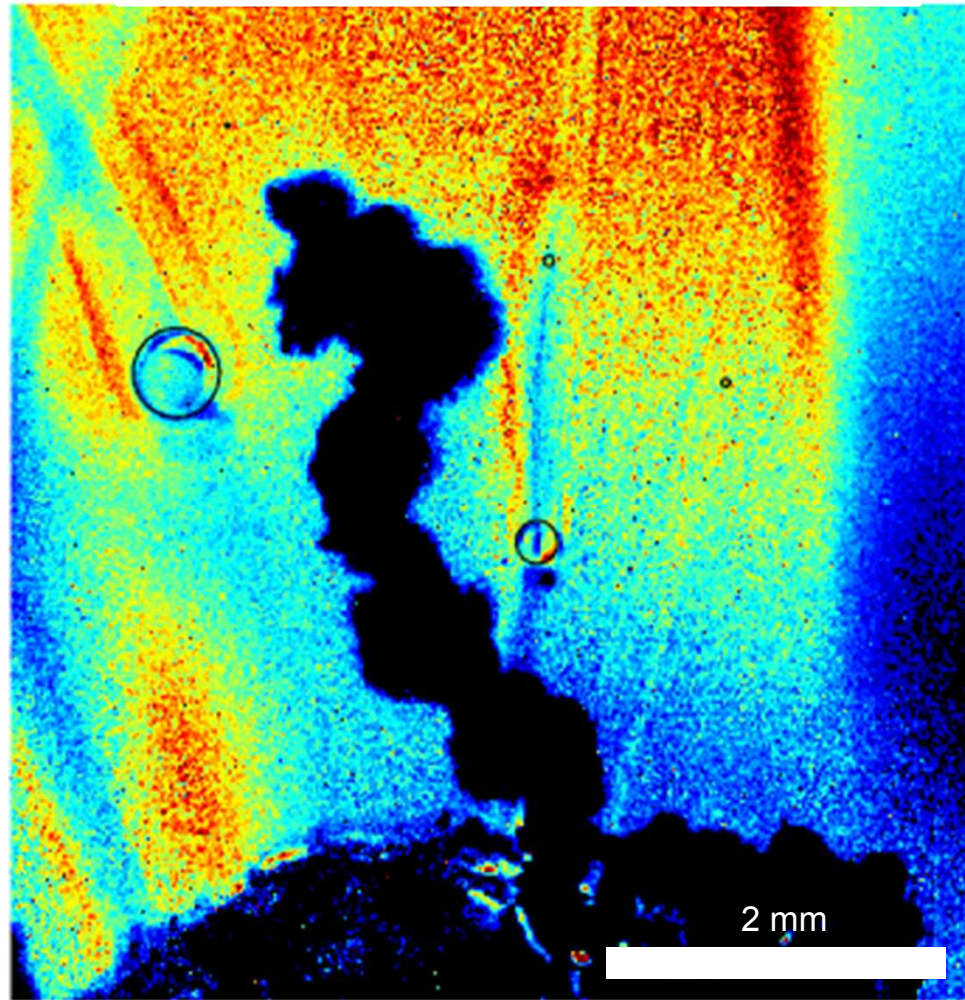
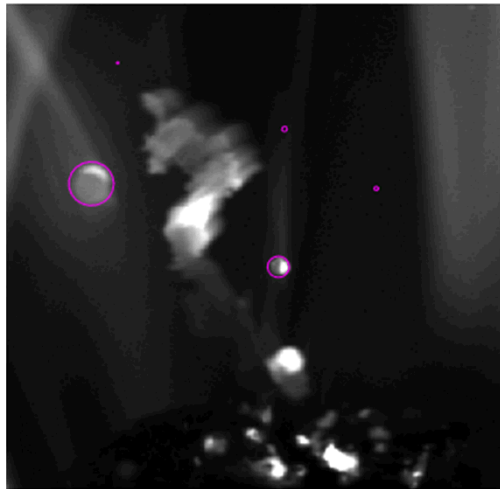


Temperature Measurement

700 nm

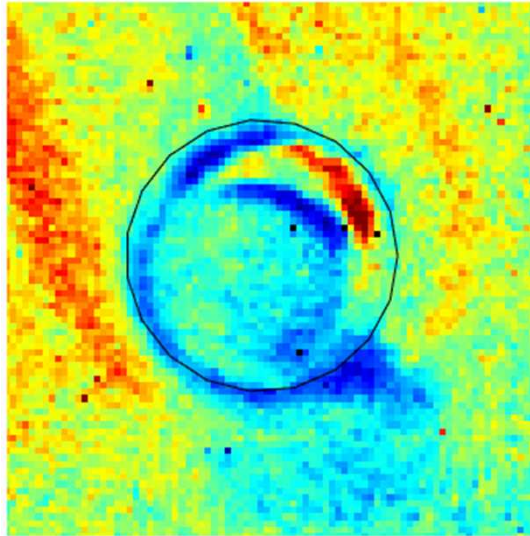


905 nm

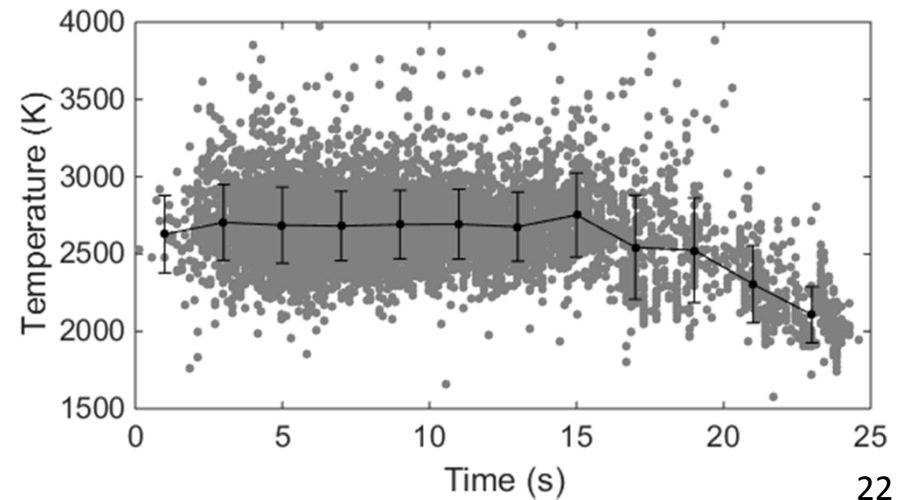
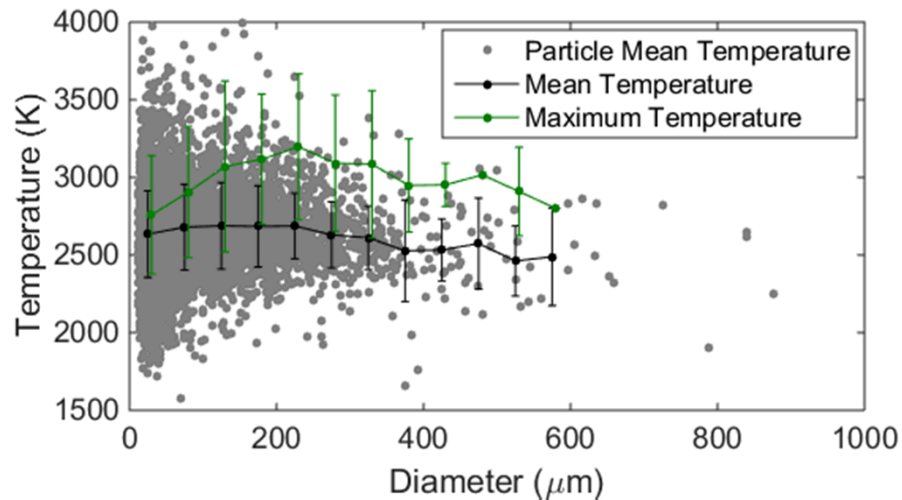
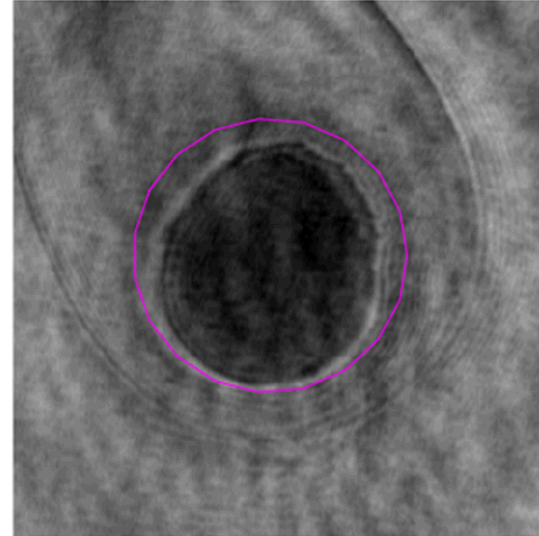


Temperature Measurement

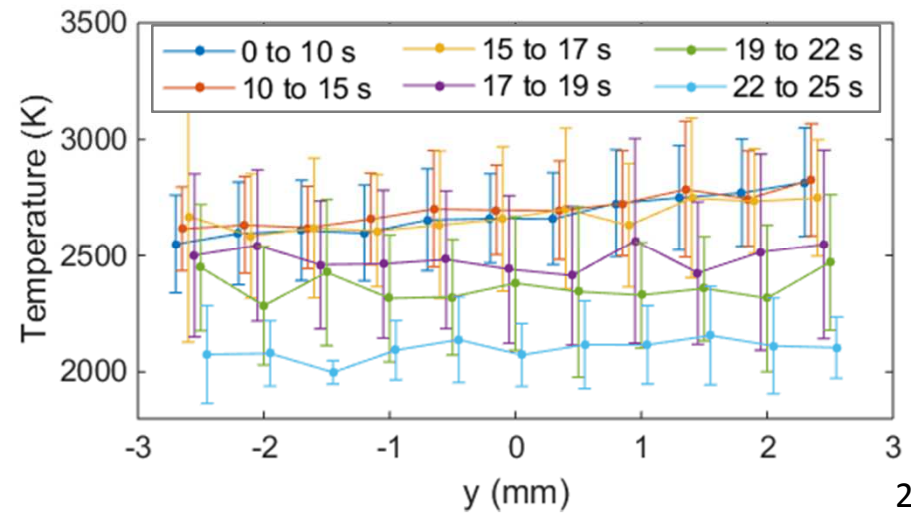
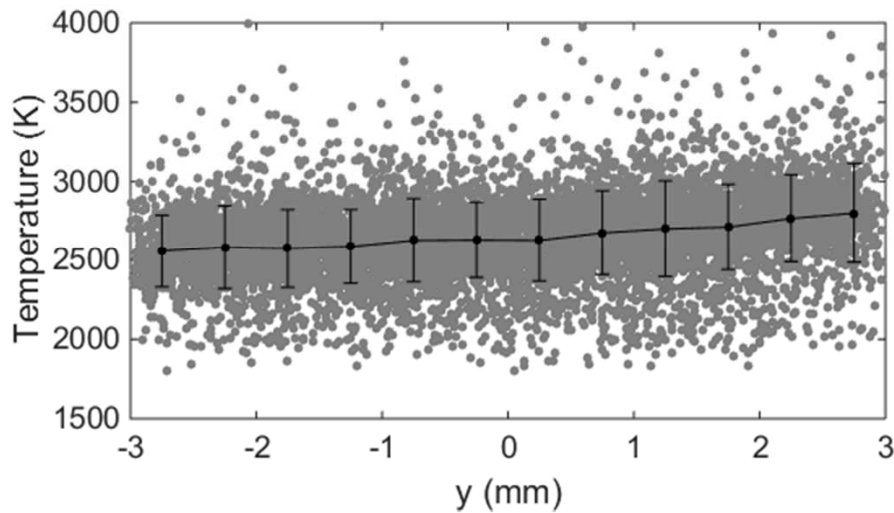
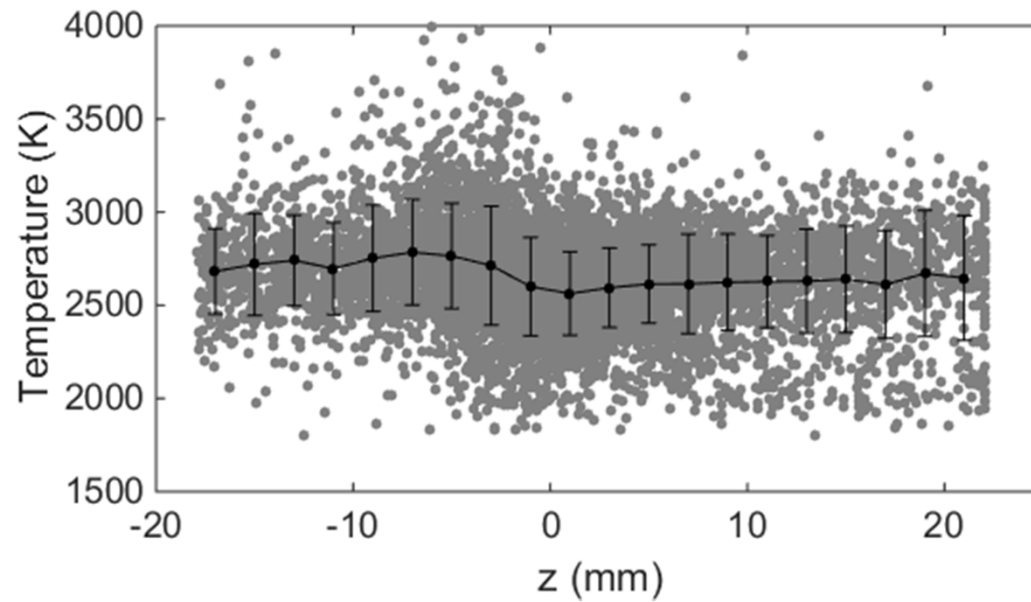
Temperature Estimate



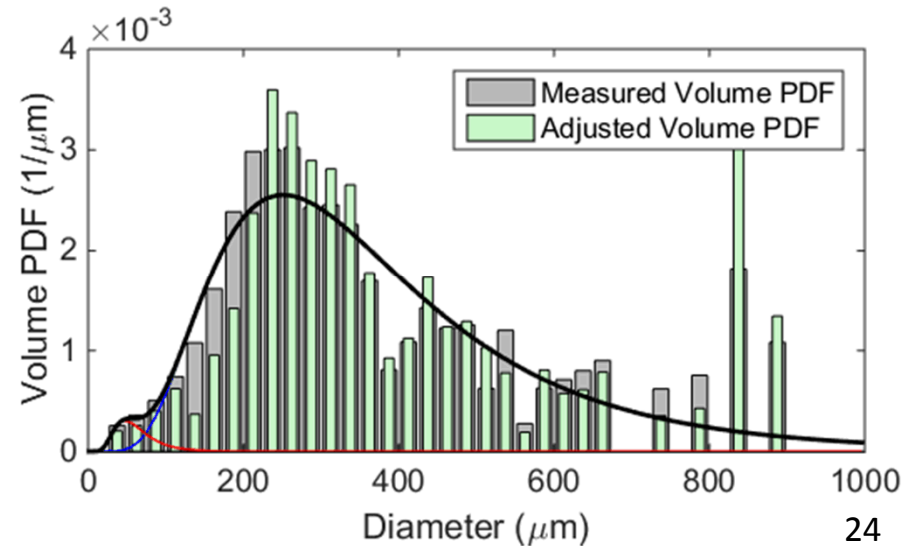
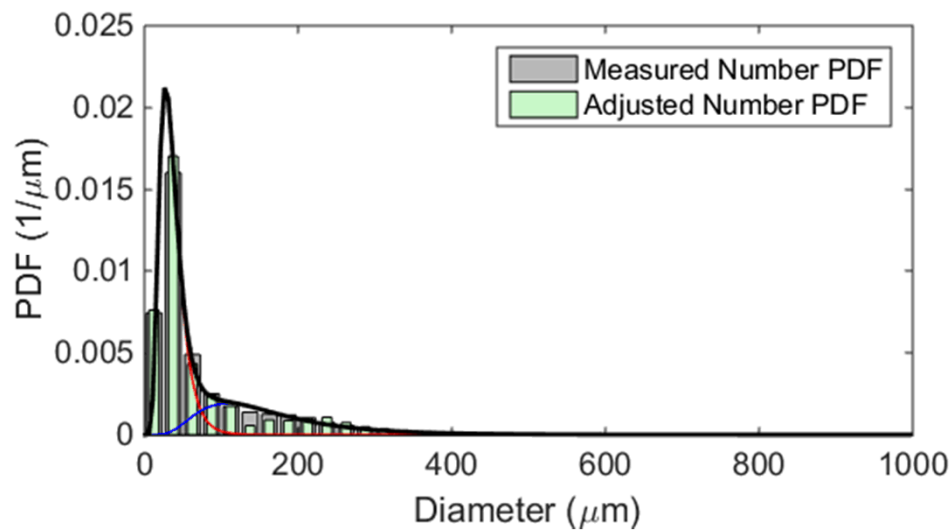
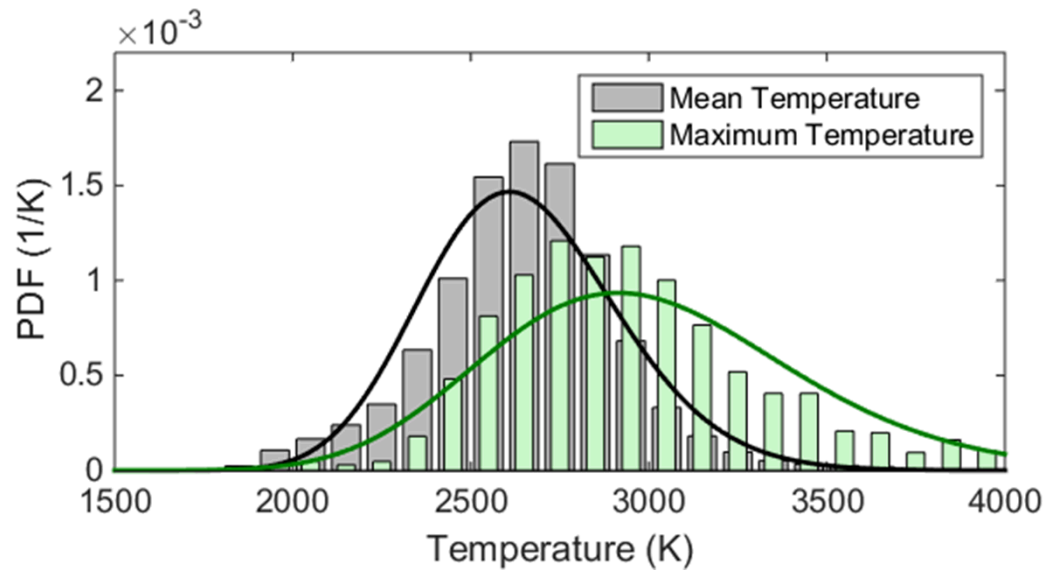
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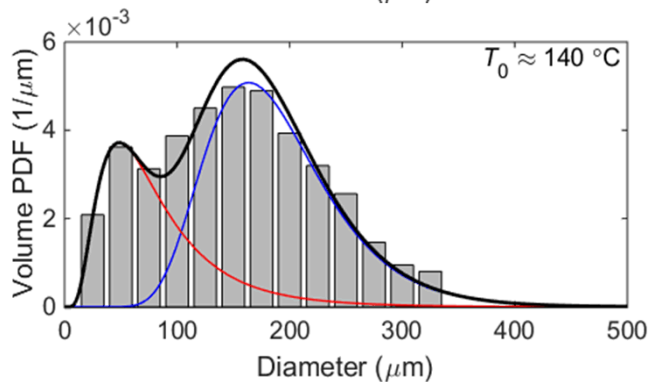
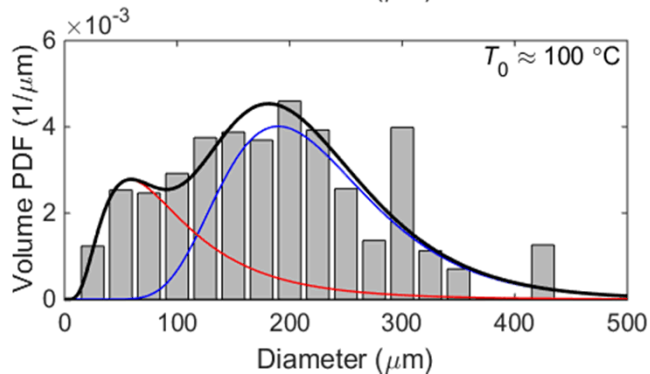
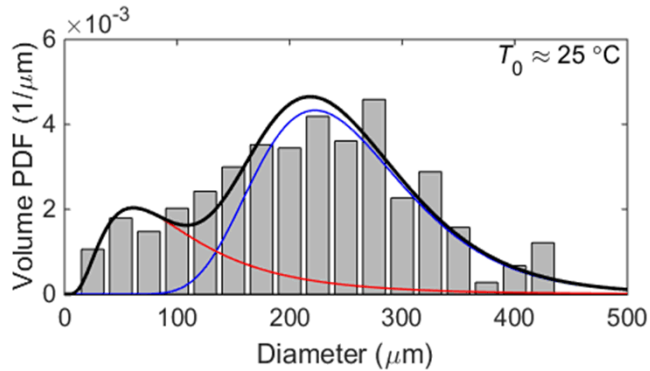
Temperature Measurement



Particle Statistics



Initial Propellant Temperature



T_0 ($^{\circ}\text{C}$)	N	D_{10} (μm)	D_{32} (μm)	D_{43} (μm)	MMD (μm)
25	17496	37.9	150.7	217.6	220.5
100	6890	39.0	127.4	187.6	185.7
140	15346	35.2	104.5	153.8	151.6

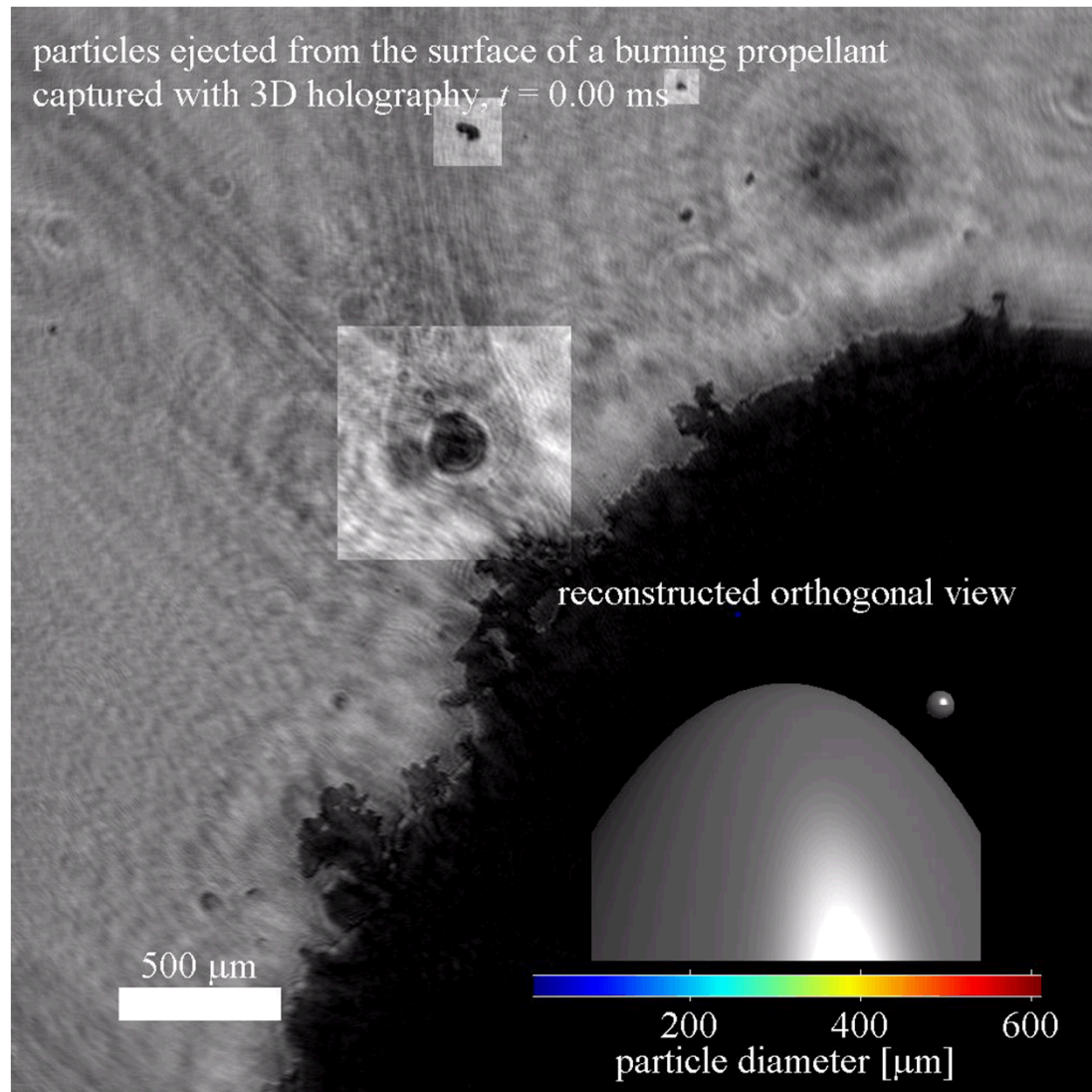
Bi-modal log-normal fits

$$f(d) = \frac{1-w}{d\sigma_a\sqrt{2\pi}} e^{-\frac{(\ln d - \mu_a)^2}{2\sigma_a^2}} + \frac{w}{d\sigma_b\sqrt{2\pi}} e^{-\frac{(\ln d - \mu_b)^2}{2\sigma_b^2}}$$

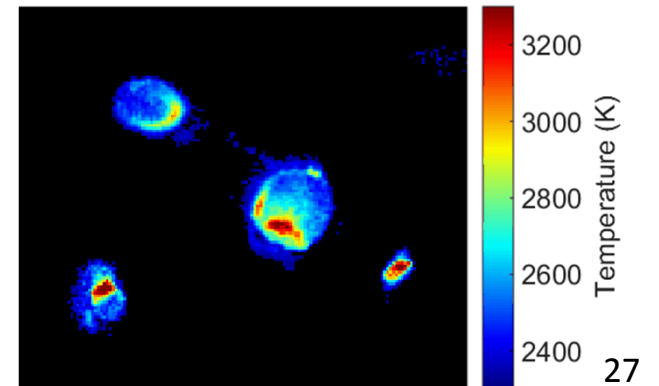
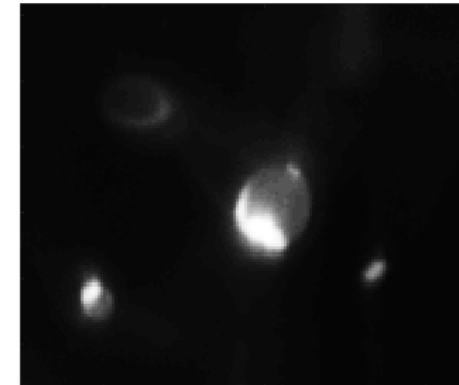
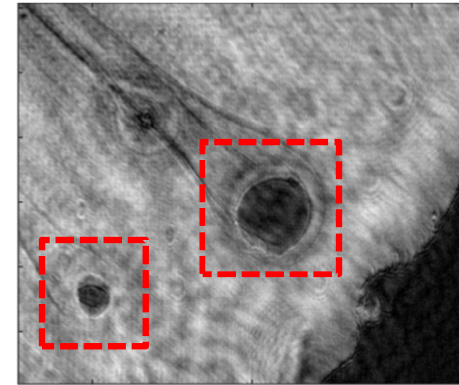
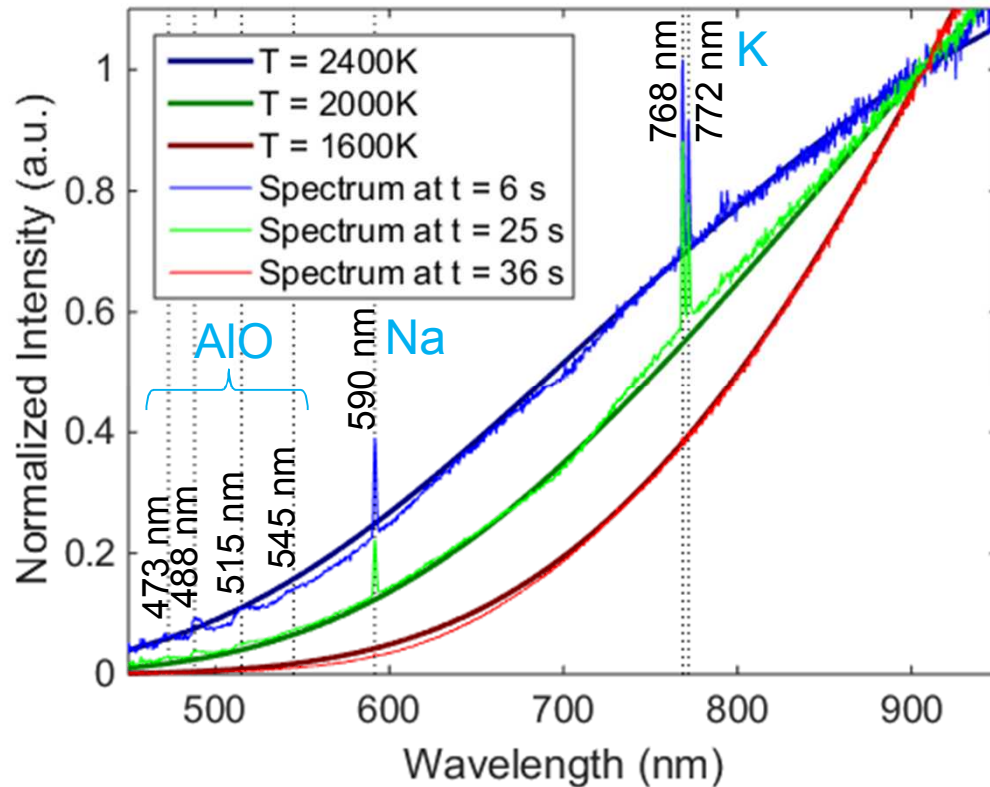
T_0 ($^{\circ}\text{C}$)	w	μ_a ($\ln[\mu\text{m}]$)	σ_a ($\ln[\mu\text{m}]$)	μ_b ($\ln[\mu\text{m}]$)	σ_b ($\ln[\mu\text{m}]$)
25	0.74	4.56	0.67	5.49	0.29
100	0.69	4.47	0.64	5.36	0.34
140	0.67	4.25	0.61	5.19	0.31

As the initial propellant stick temperature increases, the minimum size of agglomerated particle decreases and the weight fraction of larger particles decreases

High Speed Video



Two-Color Pyrometry



Equations

Diffraction Integral Equation

$$E(x, y, z) = \frac{1}{\lambda} \iint E(\xi, \eta, z=0) \frac{e^{-jkr}}{r} d\xi d\eta \quad \text{where: } r = \sqrt{(\xi - x)^2 + (\eta - y)^2 + z^2}$$

Two-Color Pyrometer Equation using Wein's Approximation

$$T = \left[\frac{k}{hc} \frac{\lambda_1 \lambda_2}{\lambda_2 - \lambda_1} \left(\ln(R) - 5 \ln \left(\frac{\lambda_1}{\lambda_2} \right) \right) \right]^{-1}$$

$$\text{where } R = \frac{I_2}{I_1} \frac{E_1 \eta_1}{E_2 \eta_2}$$