



Supporting Efforts in Deploying Large Eddy Simulation for Production Abnormal/Thermal Environment Analysis

*Exceptional
service
in the
national
interest*

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1541, Computational Thermal and Fluids Mechanics

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Outline

Description of the Abnormal/Thermal Environment

Verification & Validation and Physics & Engineering
Modeling FY16 Project Problems of interest

MMS Description and Order-of-Accuracy

Model Form Error Study

Incorporation of Buoyant Effects in LES Subgrid Models

Unstructured higher order

Path Forward/Conclusions

Abnormal/Thermal Environment

SNL TTC

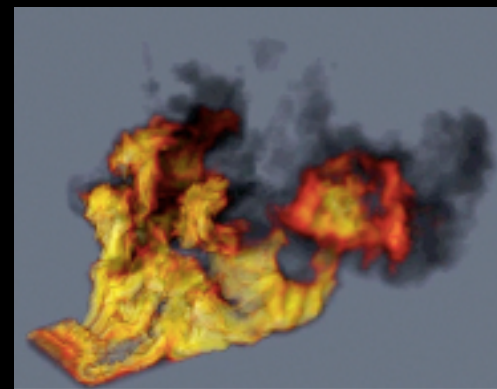


10 meter outdoor JP-8
(Nakos, lead experimentalist)



XTF for Cross
Wind Fire

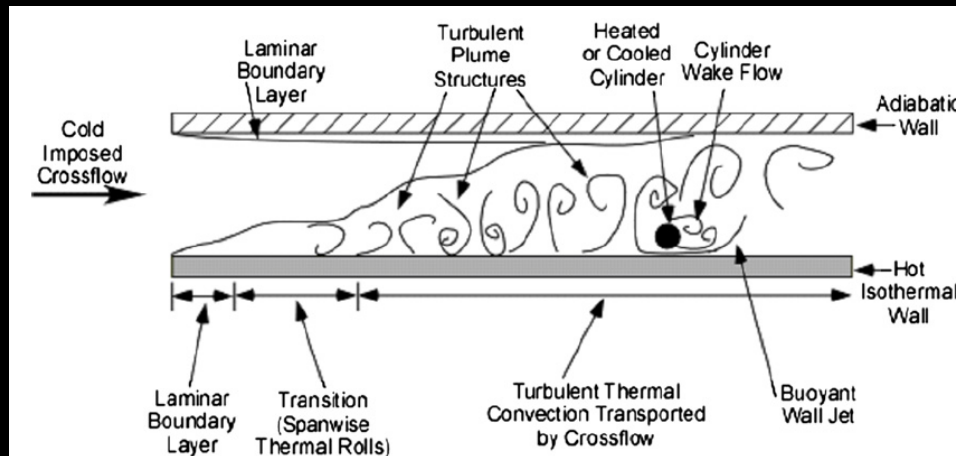
FLAME for
Quiescent Fire



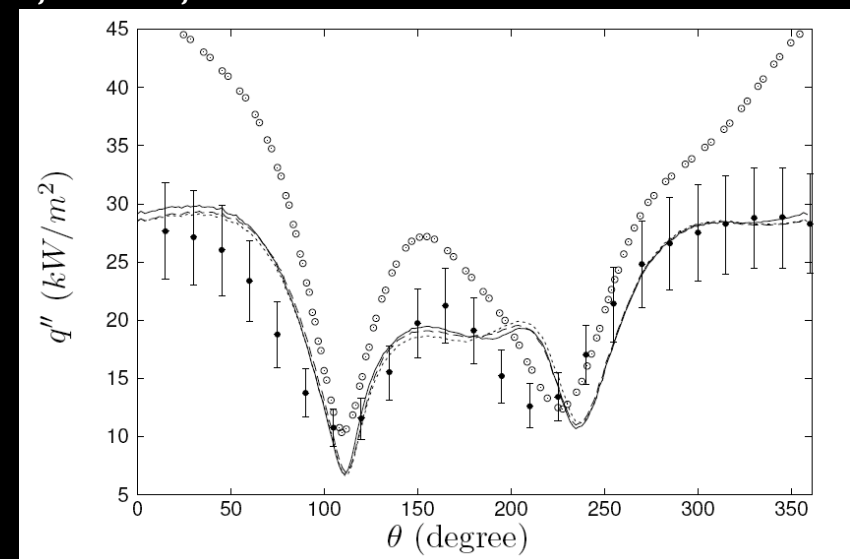
76-1 Qualification study

Towards Model Form Error (V&V¹)

Model Configuration: SNL-based Sean Kearney
Experiment, “Experimental investigation of a cylinder in turbulent convection with an imposed shear flow”, AIAA, 2005



Kearney experimental configuration



Laskowski et al. RANS / Kang et al. DNS study
(taken with permission from G. Iaccarino)

RANS-based simulation (v2-f, k-e) study conducted by Laskowski et al., AIAA 2007

DNS-based simulation study conducted by Kang et al., JCP, 2009

1. V&V Approaches to support the Abnormal/Thermal Environment; PI, Domino (V&V)

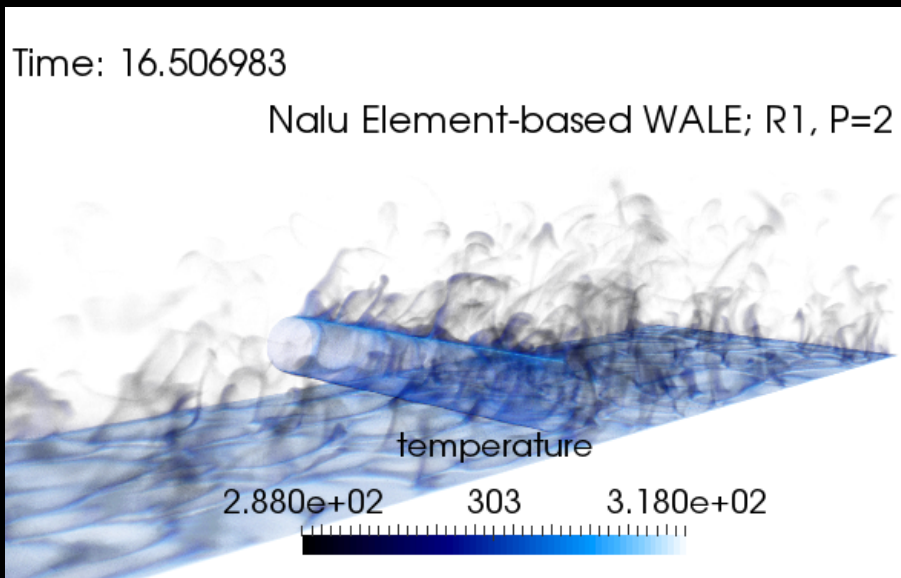
Towards Model Form Error (1/2)

Goal 1: Deploy standard V&V study (h- and P-refinement) with current set of LES models

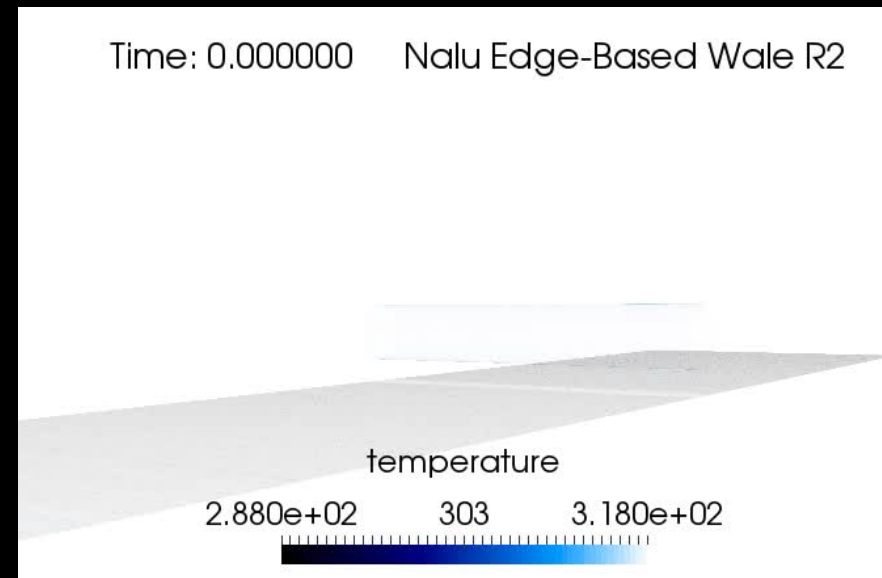
WALE, Ksgs, and Smagorinsky model

P=1 R0, R1 and R2; R2 $\sim O(100)$ million nodes

P=2 R1



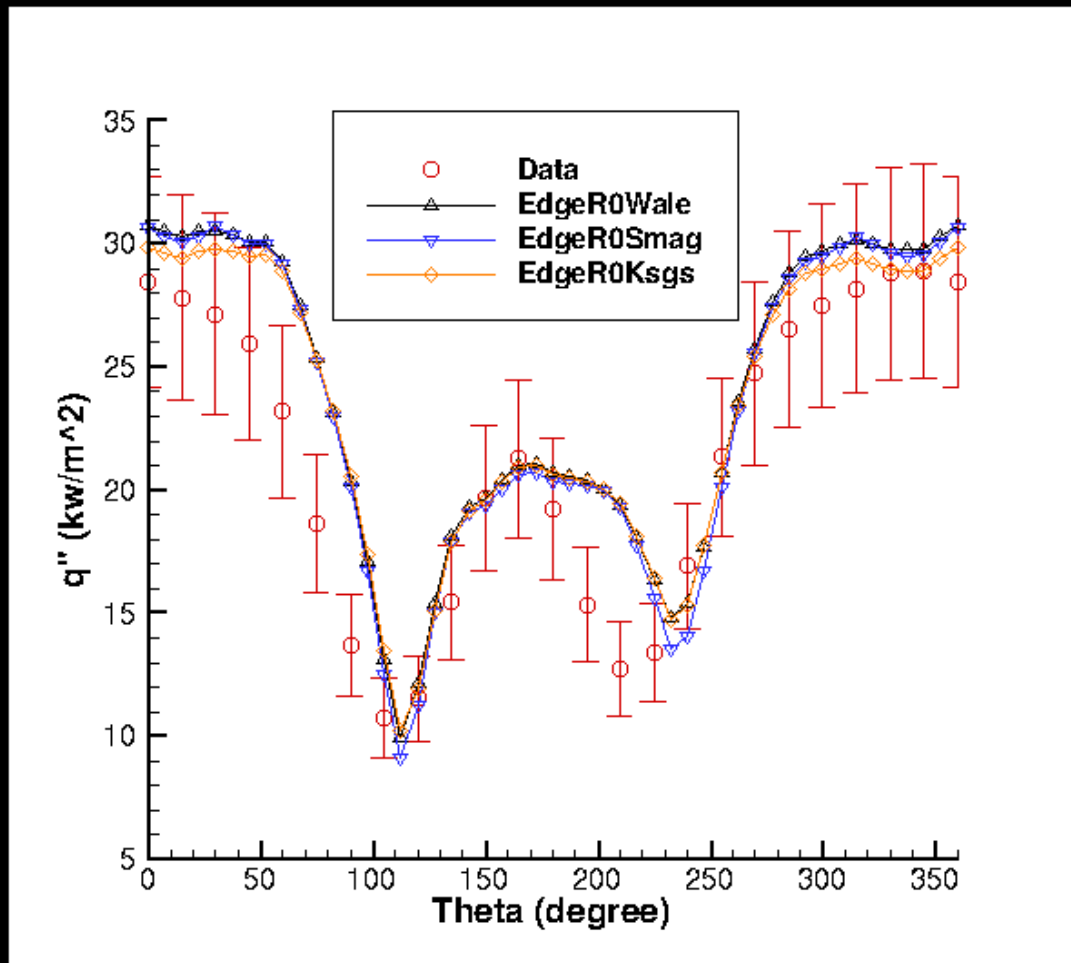
Typical Volume-rendered Temperature Field



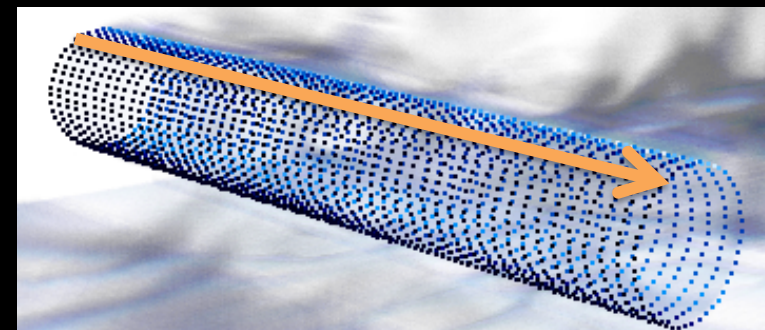
..... and P=1, EBVC movie

Towards Model Form Error (2/2)

Model form comparison using common LES sgs closure



Line-of-side post-processing



- Unstructured formulation Requires a line-of-site post processing technique
- 1% standard deviation in q''

LES using P=1 edge-based scheme with low-dissipation operators, BDF2

Goal 2: Advanced Model-Form Uncertainty Techniques

“A framework for epistemic UQ for turbulent scalar flux models for RANS”, C. Gorle, 2013; Q3 activity

$$\rho \langle u'_i h \rangle = - \frac{\mu_t}{\rho \kappa} \frac{\partial h}{\partial x_j}$$

Gradient diffusion hypothesis
(fails for a simple shear flow)

$$\rho \langle u'_i h \rangle = -C \langle u'_i u'_j \rangle \frac{\partial h}{\partial x_j}$$

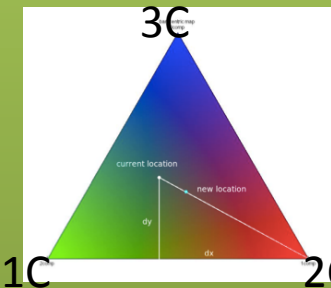
Generalized
gradient diffusion
hypothesis;
Daly and Harlow,
1970

$$\langle u'_i u'_j \rangle = 2k \left(\frac{1}{3} \delta_{ij} + v_{in} \Gamma_{nl} v_{jl} \right)$$

* = perturb

QL-decomposition
and GGDH
implemented and
about to be tested

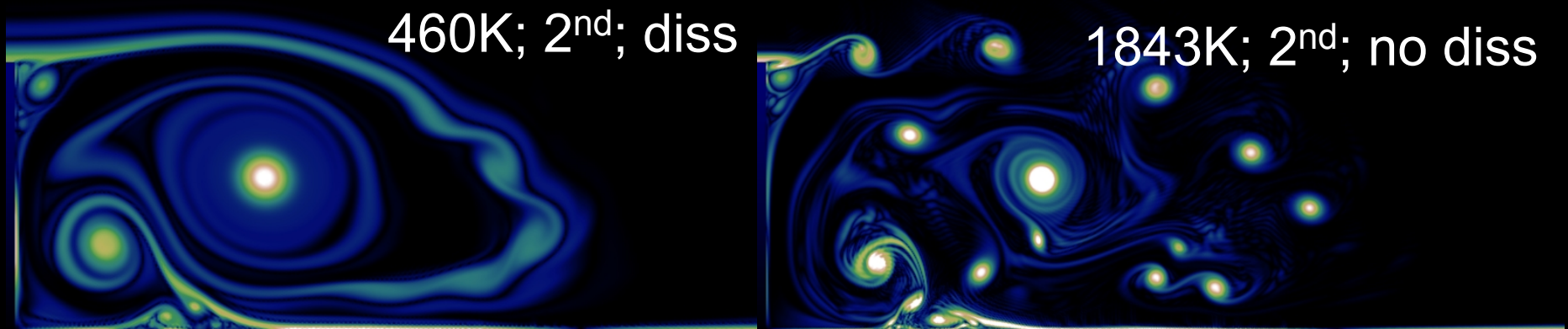
$$= 2k^* \left(\frac{1}{3} \delta_{ij} + v_{in}^* \Gamma_{nl}^* v_{jl}^* \right)$$



Barycentric map (Emory)

The low-Mach application space has extensive experience with low-order, low-dissipation schemes

Production low Mach algorithm for Abnormal/Thermal LES-based simulations is characterized as a low order ($P=1$), low dissipation (k.e. conserving) generalized unstructured CVFEM formulation



Re \sim 45K turbulent back step; vorticity magnitude

Question: What is the Viability of high-order schemes in low-Mach?

ASC Research project on continuous higher order methods with efforts on linear solver strategies and low Mach discretization (J. Hu, PI); Advantageous for NGP?

Goal 1: Implement and verify low Mach higher order method building from base CVFEM (P-based promotion; Domino, 2014) [Sweet spot of $P=3$? PC: M. Ihme; K. Jansen]

Goal 2: Are common pressure stabilization approaches used in low Mach equal-order interpolation design order?

Goal 3: Efficient linear solver interface/performance

Goal 4: Develop advanced stabilization approaches

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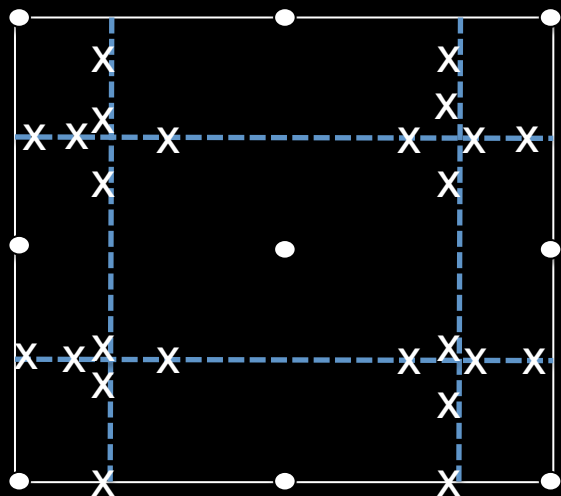
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Location, location, location!

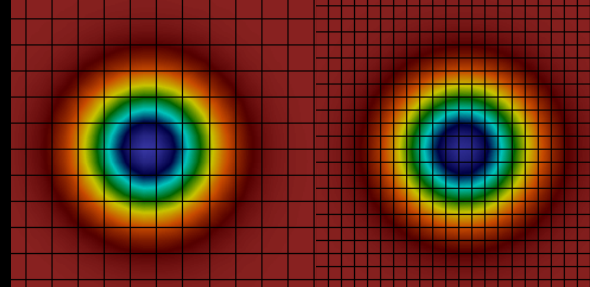
Quadrature locations are of the utmost importance!



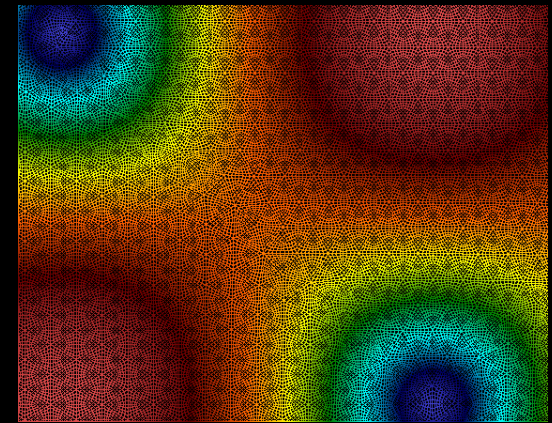
Dual volume definition; $P=2$
 2×2 integration at scs/scv

HOCVFEM, $P=2$

$P=1$



Inviscid Taylor Vortex



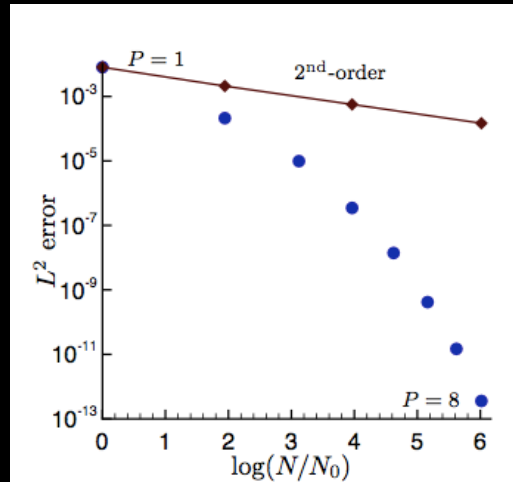
Steady Taylor Vortex

Challenge: $pStab \sim \tau(G_j \phi - d\phi/dx_j)n_j dS$

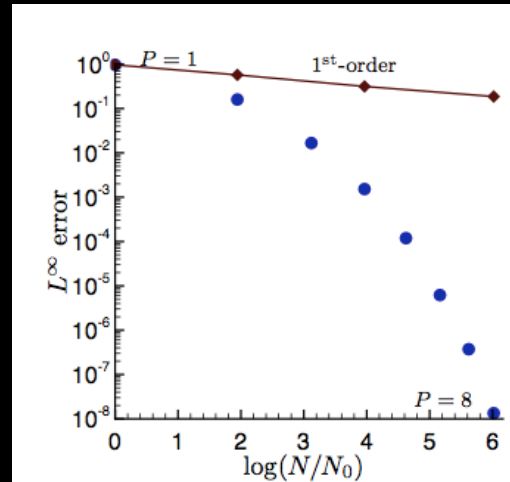
Finding: Current dual volume definition requires consistent integration for time, source, and $L2 G_j \phi$

Most unstructured CV-based methods are difficult to extend to H.O.

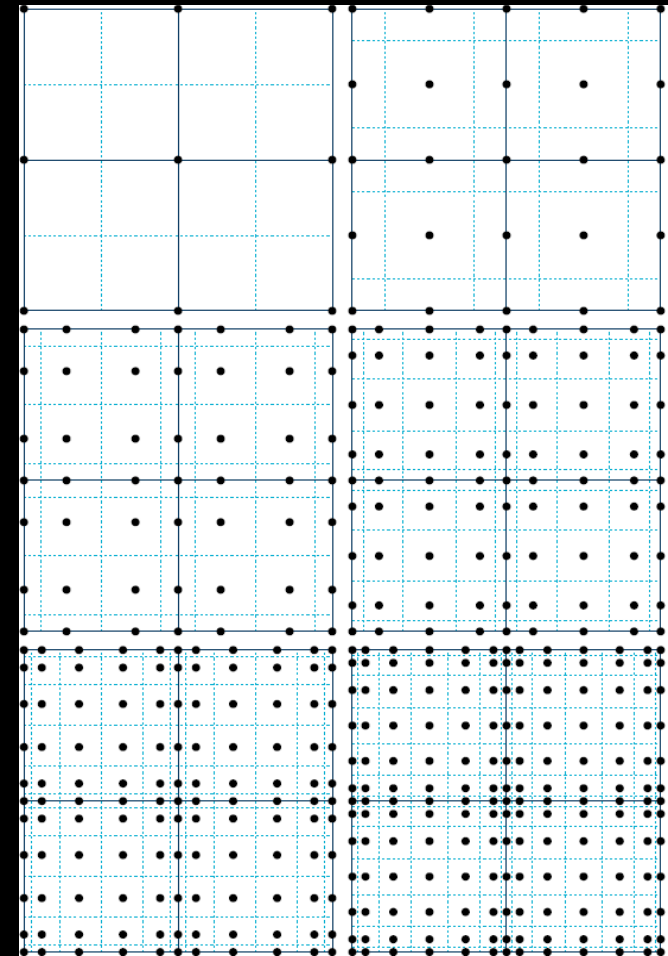
Spectral CVFEM demonstrated



$d-o-f; O(P+1)$

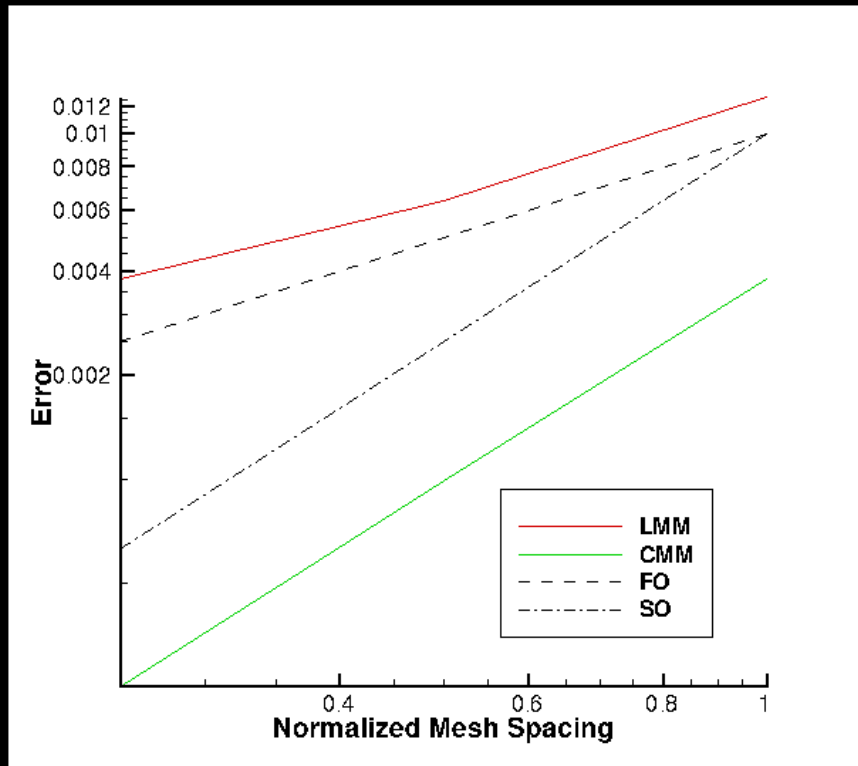


$G_j \phi; O(P)$

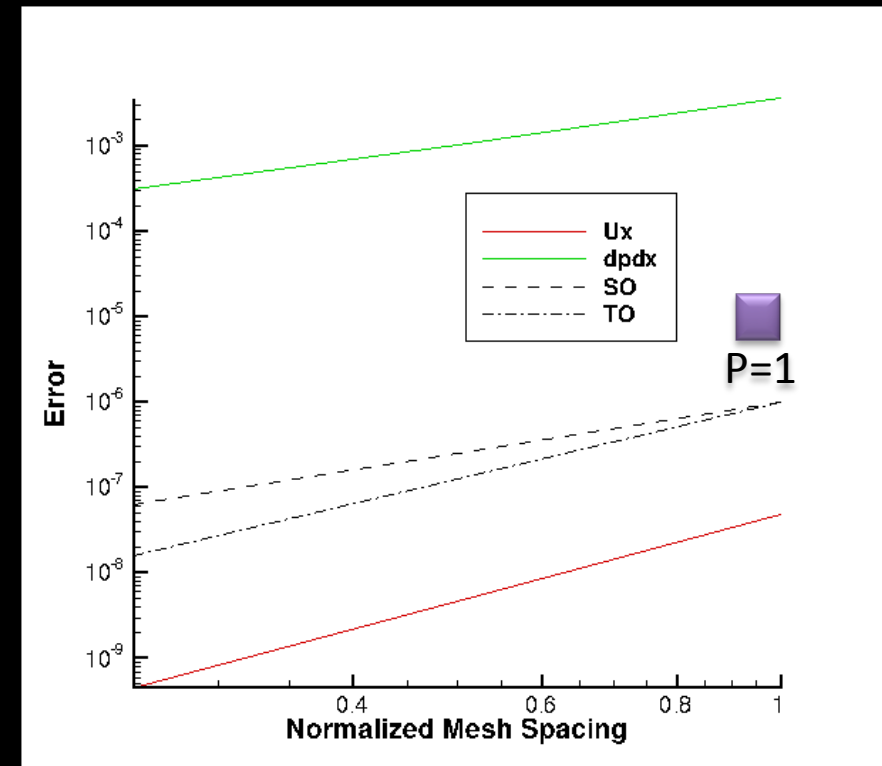


$P=1:6$ element integration rules (GLL/GL)

Code Verification (P1 vs P2)



L2 projected nodal gradients must be $O(P)$
 CMM L2 Projection Equation must be solved!



Velocity verified to be $O(3)$
 Pressure gradient $O(2)$



Lesson learned: If one is verifying an unstructured scheme
 then use unstructured meshes !

Advanced Stabilization Approaches

Goal: Advanced the state-of-the-art nonlinear stabilization operators (NSO) (ASC Algorithms)

$$\sum_e \int_{\Omega} \nu(\mathbf{R}) \frac{\partial w}{\partial x_i} g^{ij} \frac{\partial \phi}{\partial x_j} d\Omega, \quad g^{ij} = \frac{\partial x_i}{\partial \xi_k} \frac{\partial x_j}{\partial \xi_k}, \quad g_{ij} = \frac{\partial \xi_k}{\partial x_i} \frac{\partial \xi_k}{\partial x_j}$$

FEM-based DCO (Shakib, 91)

Co-variant and contra-variant metric tensors

$$\nu(\mathbf{R}) = \sqrt{\frac{\mathbf{R}_k \mathbf{R}_k}{\frac{\partial \phi}{\partial x_i} g^{ij} \frac{\partial \phi}{\partial x_j}}} \quad \text{or min.....} \nu^{upw} = C_{upw} (\rho^2 u_i g_{ij} u_j)^{\frac{1}{2}}$$

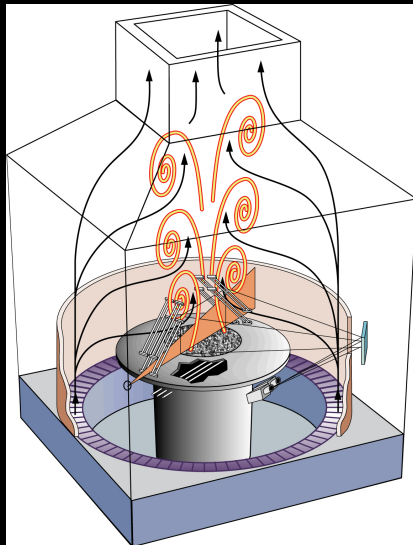
Coefficient based on local residual

Limit the coefficient based on an upwind analogy

$$- \sum_e \int_{\Gamma} \nu(\mathbf{R}) g^{ij} \left(\frac{\partial \phi}{\partial x_j} - G_j \phi \right) n_i dS$$

Piecewise-constant test function converts to the CVFEM form of interest

Helium Plume DNS



Reduced Geometry from
FLAME (above)
to simple cylinder



P=1 (left); P=4 (right); same number of nodes

Conclusions

A variety of innovations are required to deploy LES to production ATE simulation studies

Discretization, solver, V&V and PEM activities are in progress to advance the state of the art

MMS First (not shown), model validation second

LES non-isothermal flow V&V study in transit; Model form uncertainties will be explored via λ decomp

Higher order methods look promising; compare to P=1

Models to incorporate buoyancy in LES simulations will be explored; will drive novel DNS study for a 1-m plume