

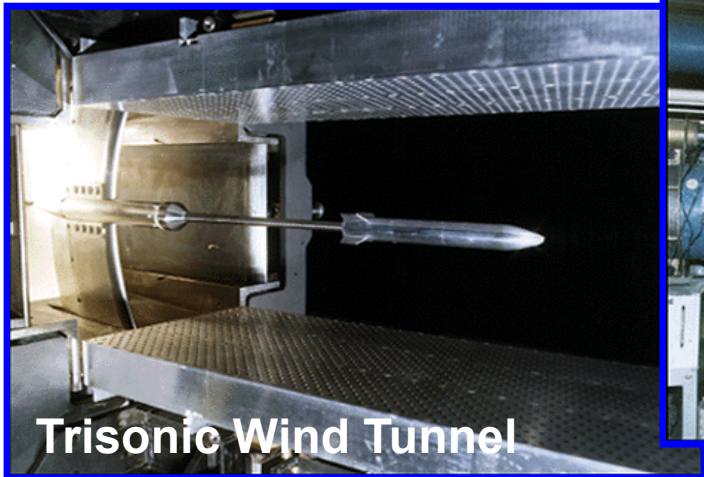
# Pulse-Burst PIV in High-Speed Flows

**Steven Beresh, Justin Wagner, Ed DeMauro,  
John Henfling, Rusty Spillers, and Paul Farias**

**Sandia National Laboratories  
Albuquerque, NM**

**Michigan State University Seminar  
February 25, 2016**

# Experimental Aerosciences Facility



**Trisonic Wind Tunnel**

## **Trisonic Wind Tunnel (TWT)**

- Mach 0.5 – 3
- Gravity bombs, missiles

## **Hypersonic Wind Tunnel (HWT)**

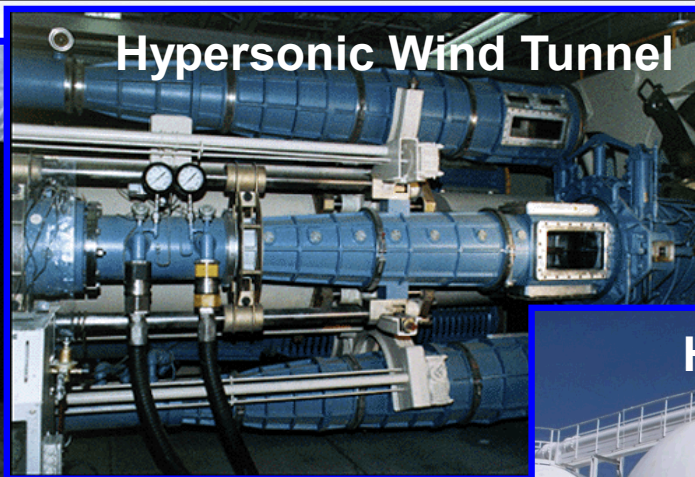
- Mach 5, 8, 14
- Re-entry vehicles, rockets

## **High-Altitude Chamber (HAC)**

- Satellite components

## **Multi-Phase Shock Tube (MST)**

- Explosives research



**Hypersonic Wind Tunnel**



**High-Altitude Chamber**

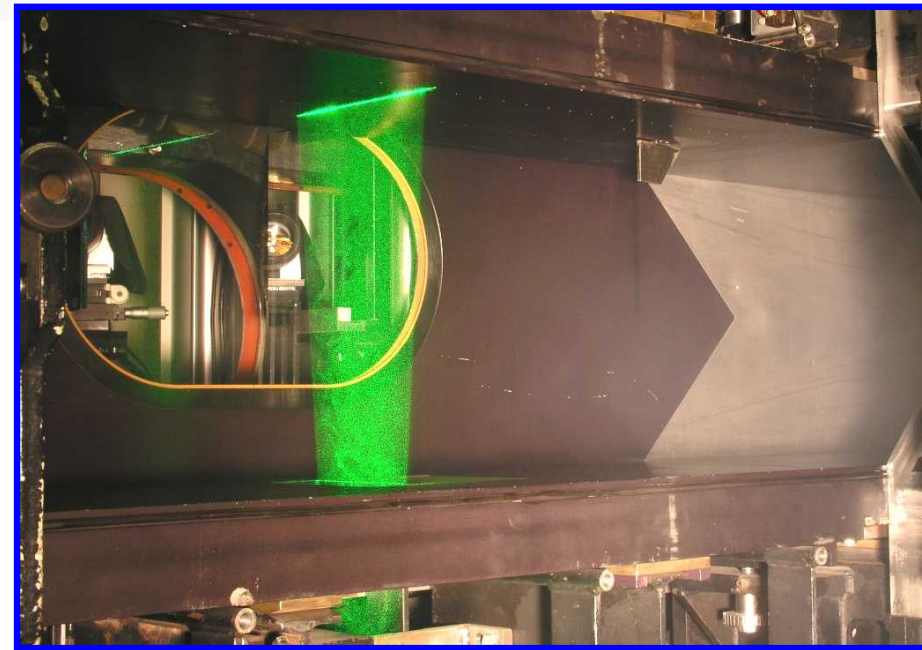


**Multi-Phase Shock Tube**

# Much of the focus in our laboratory is on advanced diagnostics.



Data acquisition ~1950



Data acquisition ~21<sup>st</sup> century

**Experimental data are necessary to develop and validate Sandia's modeling and simulation capability.**

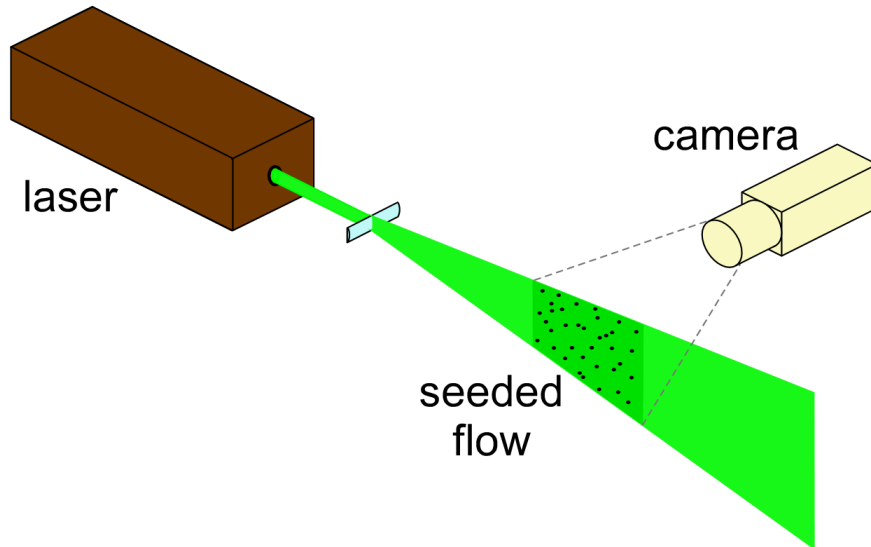
- Provide scientific discovery as well as validation data.

**High-fidelity flowfield data are needed, not just aerodynamic coefficients and surface measurements.**

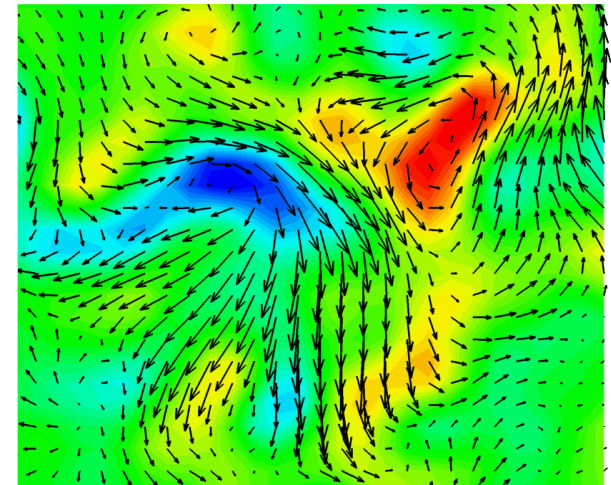
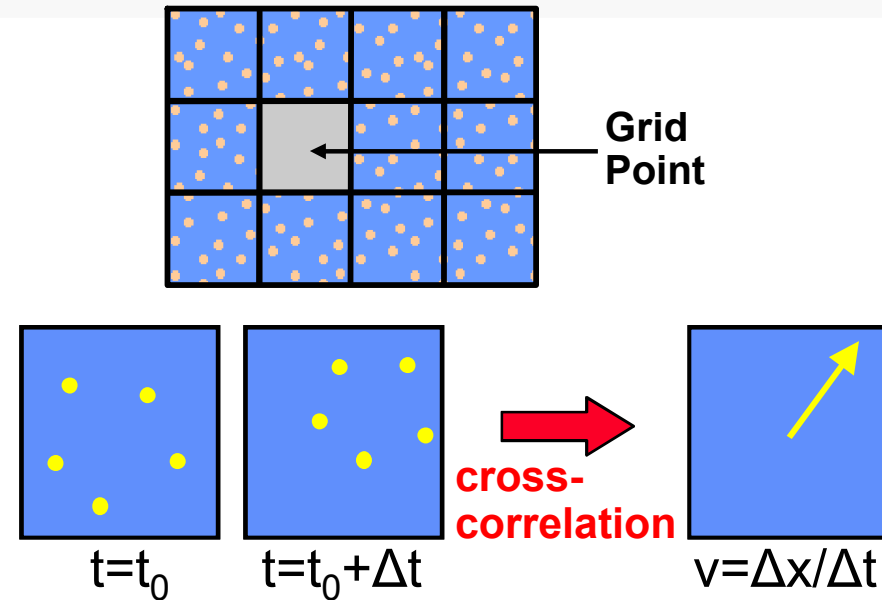
- Adapt new diagnostic technologies to aero applications.



# What is Particle Image Velocimetry (PIV)?



- Seed a large quantity of small **particles** into the wind tunnel
- Illuminate with a double-pulsed laser sheet and **image** with a specialized digital camera
- Grid the images into smaller windows
- In each grid window, track a pattern of particles as they move from the first exposure to the second
- Compute a field of **velocity** vectors



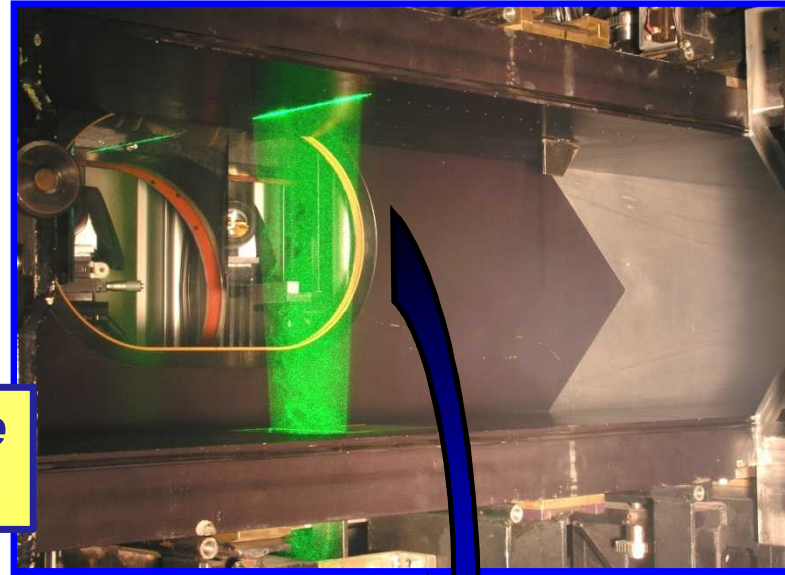


# Particle Image Velocimetry (PIV) has emerged as our most widely used measurement.

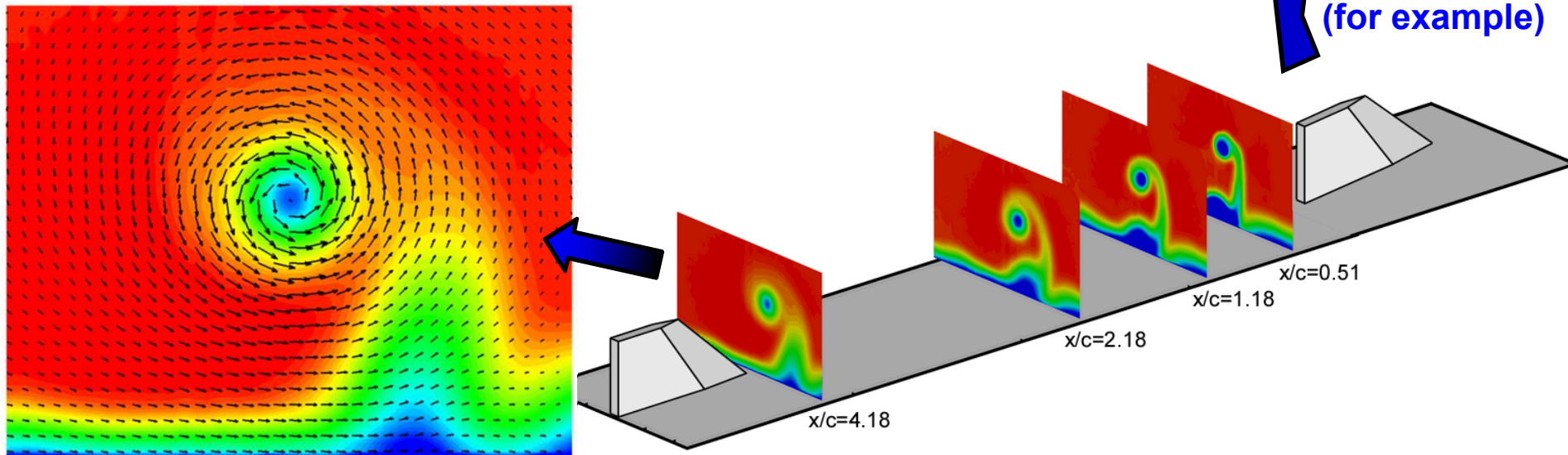
PIV data have been key to improving our understanding of the physics our codes must simulate.

- Can be reduced to mean velocity fields, turbulence quantities, flow structures, etc.
- Combined with a force balance or pressure sensors, it can explain aerodynamic forces.

*But conventional PIV does not provide the time component to flow velocimetry.*



(for example)



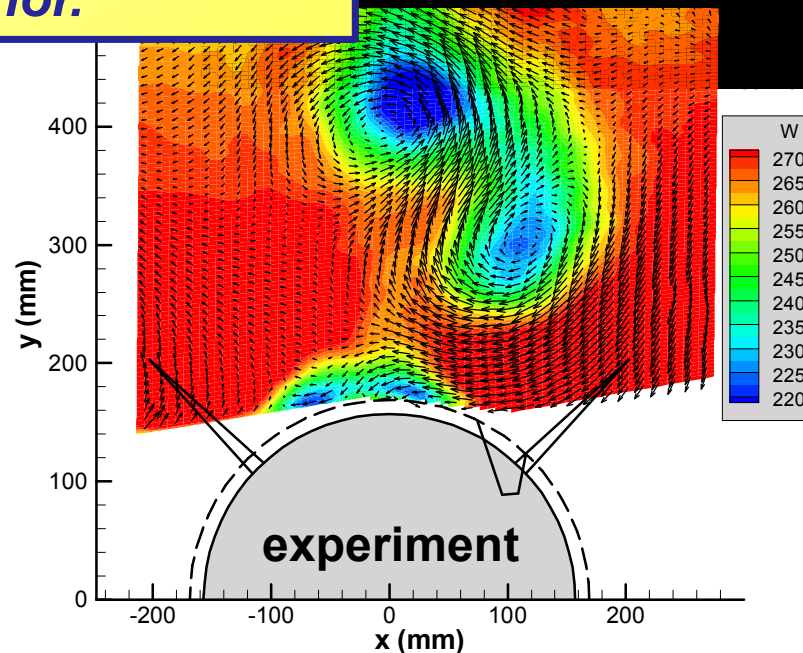
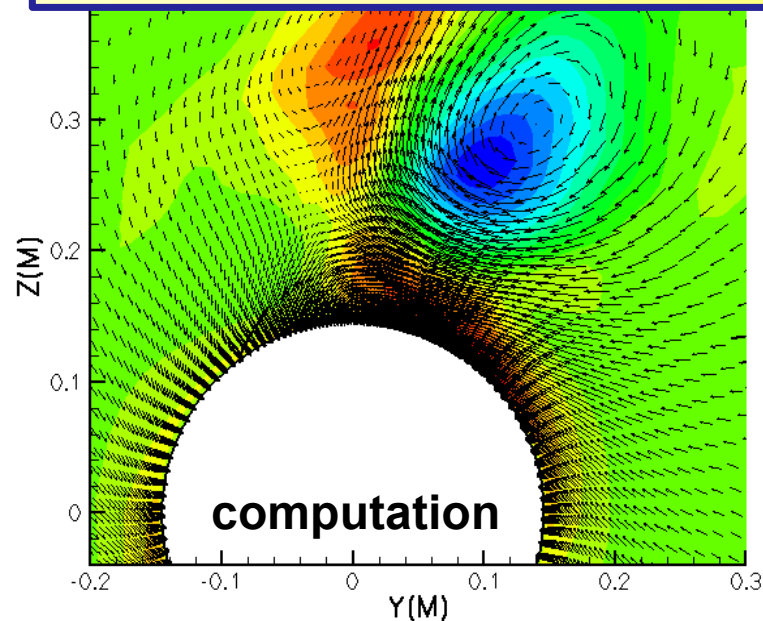
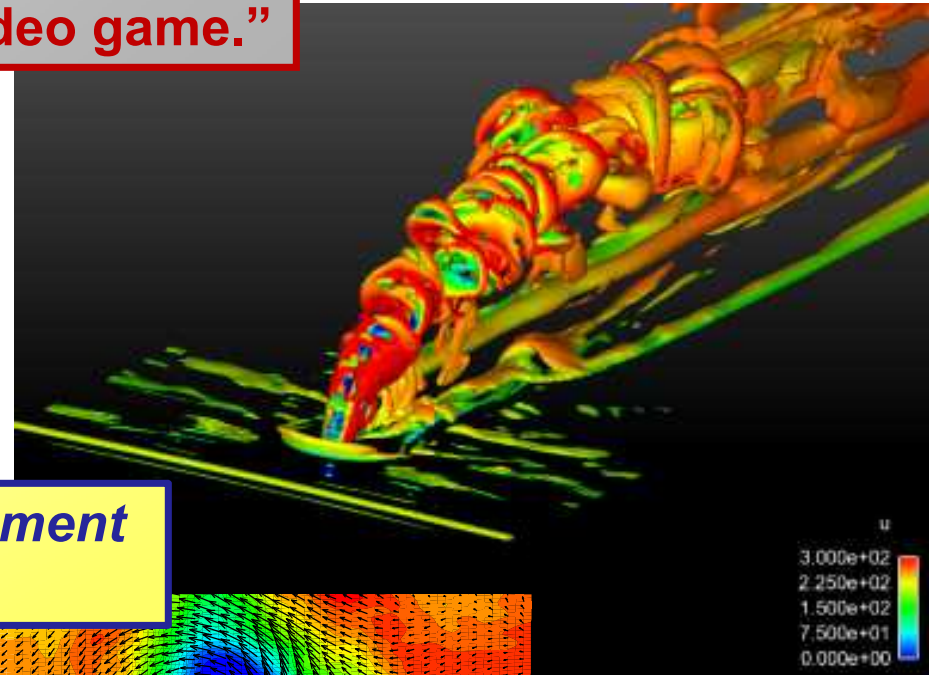
# PIV Data for CFD Validation

**“Without validation, CFD is just a video game.”**

The physical models in our CFD codes must be shown to produce accurate results before we may use them to design flight vehicles.

Our PIV experiments provide key validation data.

*Recent applications require development of models of unsteady behavior.*





# Time-Resolved PIV (TR-PIV)

*Provide temporally correlated velocity fields – that is, PIV movies.*

## The current state-of-the-art in TR-PIV:

- Diode-pumped solid-state (DPSS) lasers
  - Typically 1-10 kHz (16 kHz max)
  - Only a few mJ at high kHz
- Fast CMOS cameras to 20 kHz at 1 Megapixel
- Works fine for low-speed flows and small field of view.

## This isn't good enough for a high-speed wind tunnel:

- Faster repetition rates for briefer time scales.
- Higher energy required.
  - Scatter light off smaller particles
  - Expand laser sheet for larger field of view

**A pulse-burst laser is necessary.**



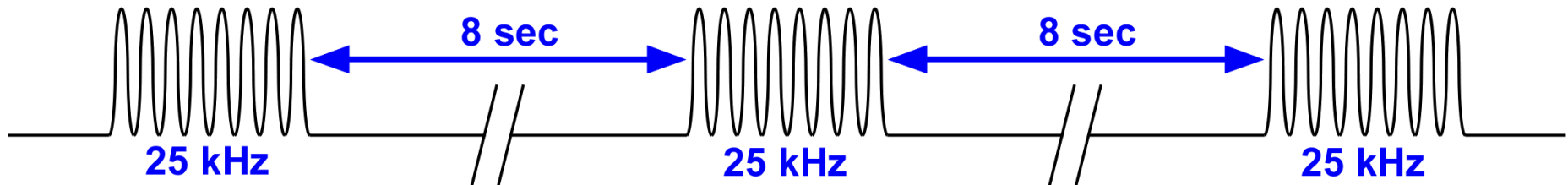
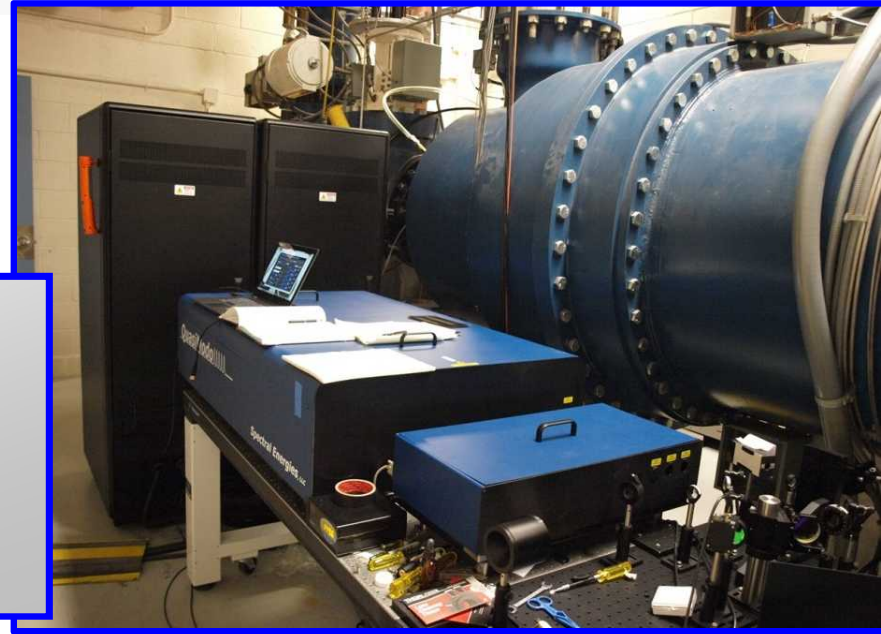
# Pulse-Burst Laser

*A pulse-burst laser allows high energy and high repetition rates.*

*But a very low duty cycle.*

## Pulse-Burst Laser:

- Manufactured by Spectral Energies, LLC
- Bursts of pulses for 10.2 ms
- Up to 500 kHz of pulse pairs, 20-500 mJ
- But only one burst every 8 sec



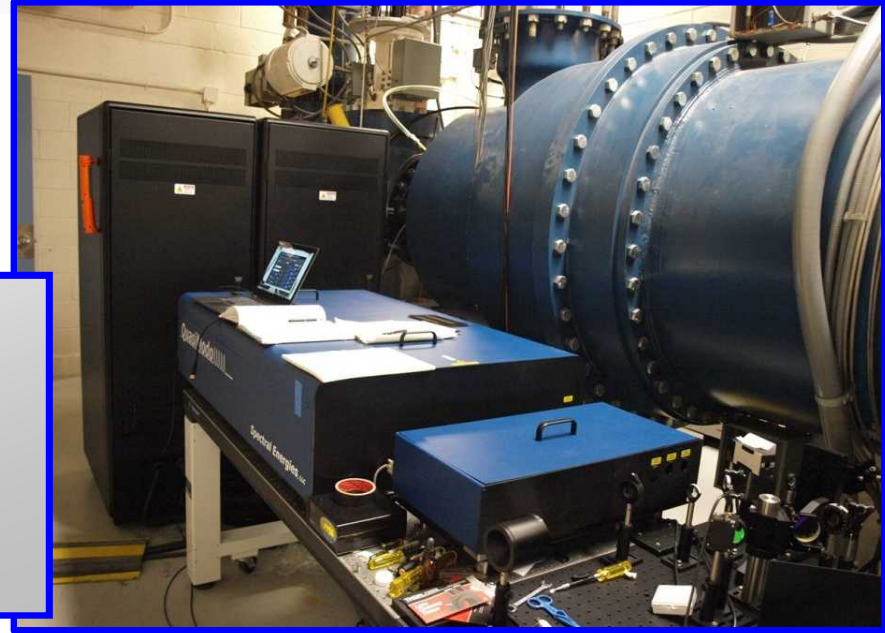
# Pulse-Burst Laser

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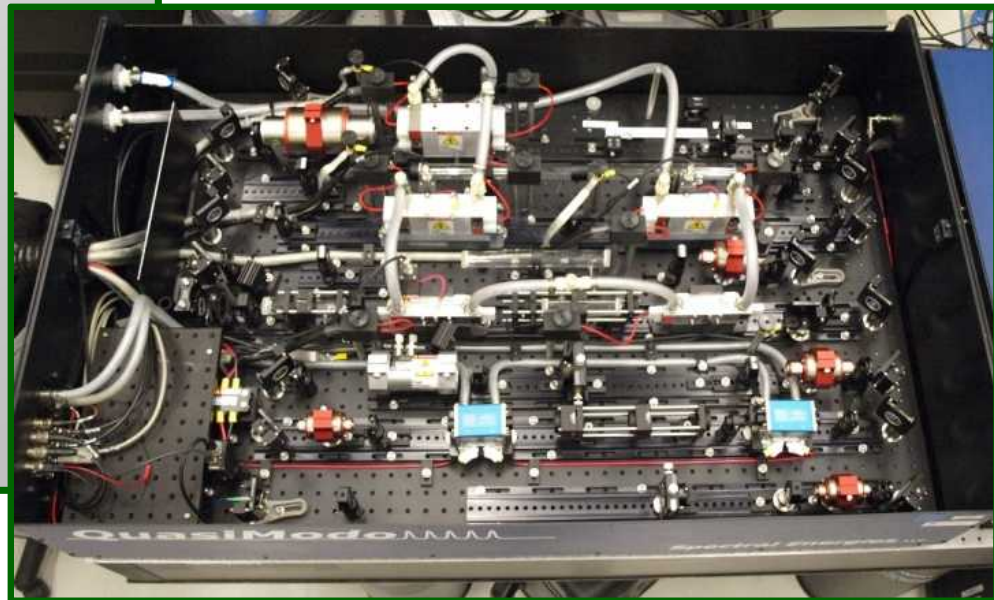
## Pulse-Burst Laser:

- Manufactured by Spectral Energies, LLC
- Bursts of pulses for 10.2 ms
- Up to 500 kHz of pulse pairs, 20-500 mJ
- But only one burst every 8 sec



## Laser Design:

- CW diode laser at 1064 nm
- Sliced by combined acousto-optic and electro-optic modulators
- Four diode-pumped amplification stages
- Four flashlamp-pumped amplification stages

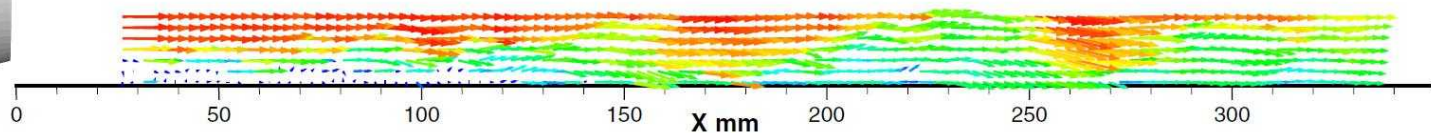


# A Brief History of Pulse-Burst PIV

**Wernet (2007)**

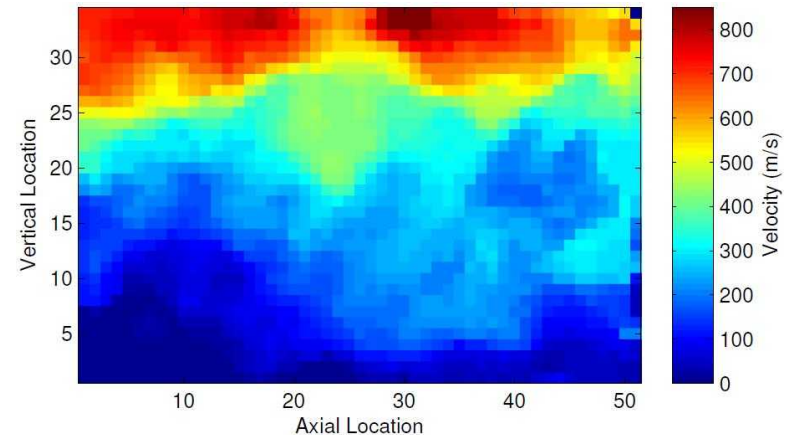
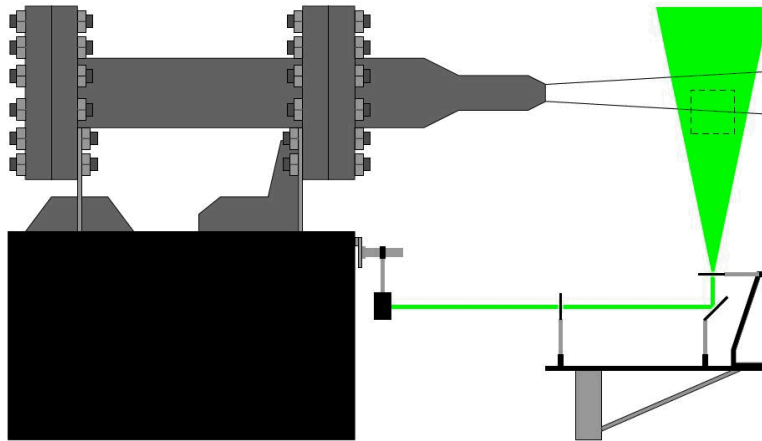
2-inch high-speed jet  
25 kHz data

Velocity [m/s]: 10 30 50 70 90 110 130 150 170



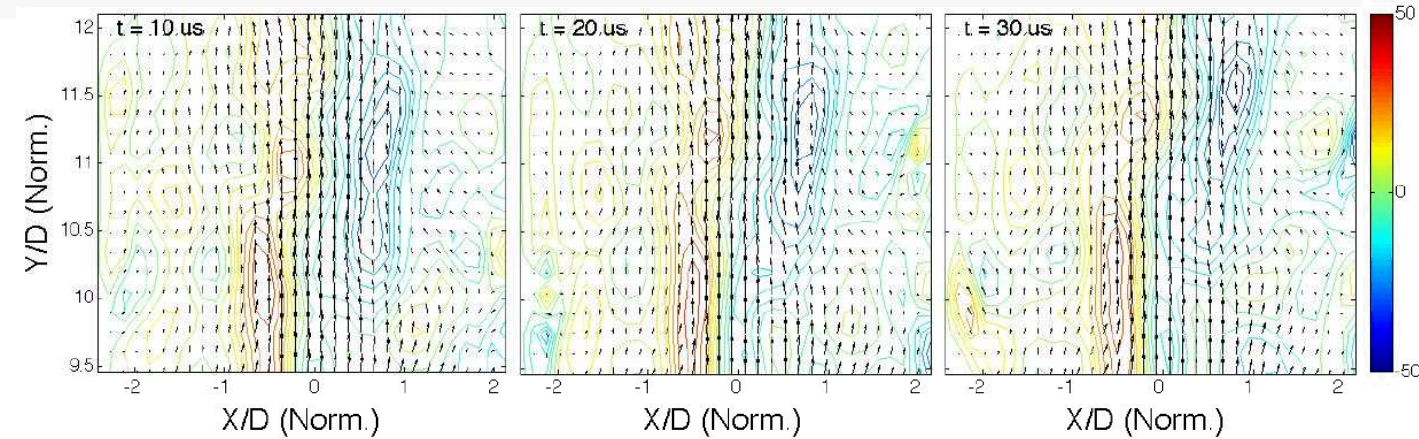
**Brock et al (2014)**

1 MHz data in a supersonic jet  
But only 13 images of poor quality





# A Brief History of Pulse-Burst PIV



**Miller et al (2013, 14)**

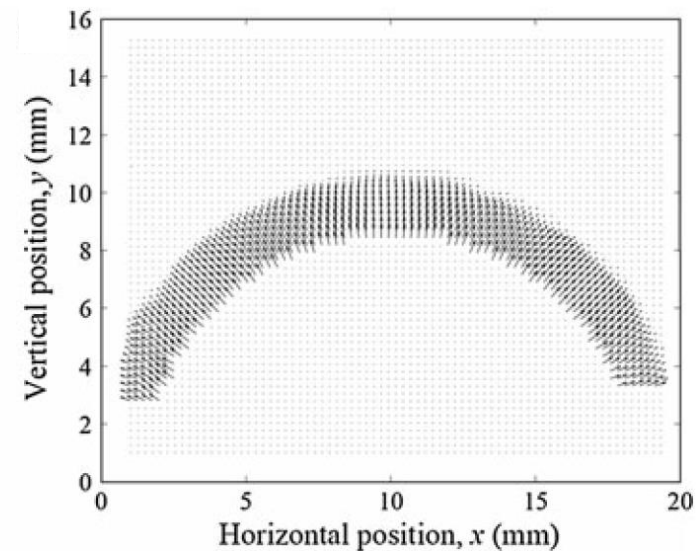
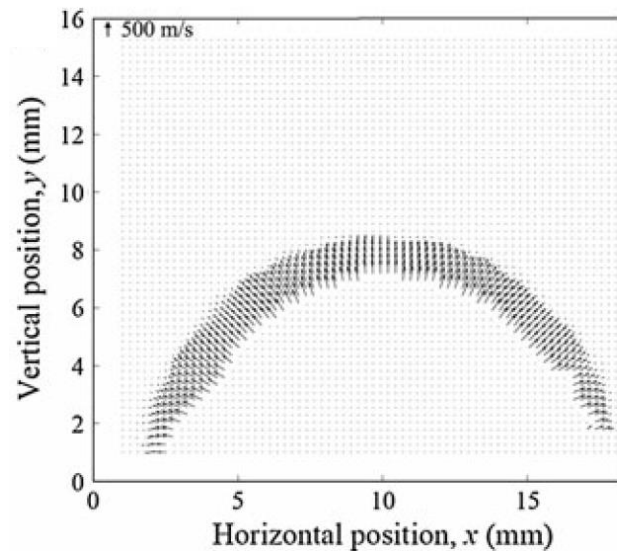
First at 10 kHz,  
then 100 kHz

4.6 mm jet at  
Mach 0.3

**Murphy and Adrian  
(2011)**

Chain together  
8 Nd:YAG's

300 kHz of a  
blast wave



***Our work is the first application of pulse-burst PIV in a ground-test facility.***

# Trisonic Wind Tunnel (TWT)

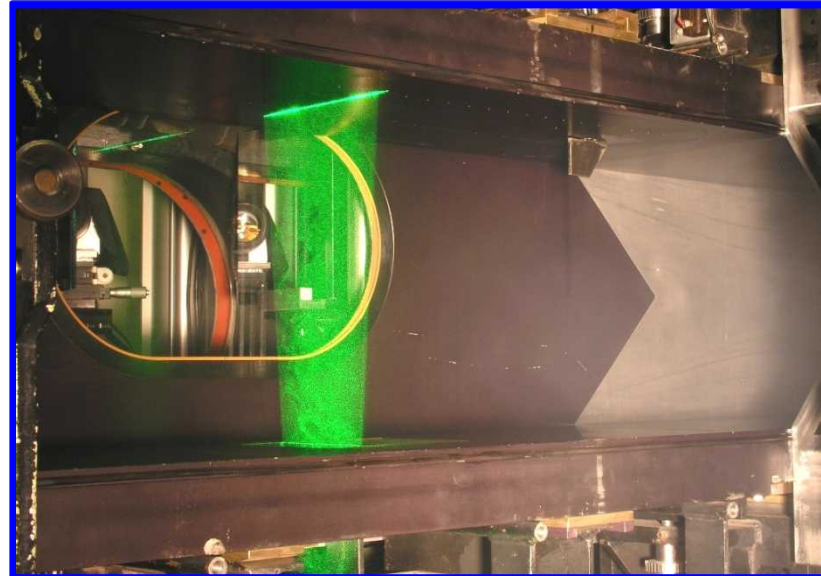
## Technical Characteristics

- Blowdown to atmosphere
- $M_\infty = 0.5 - 1.3, 1.5, 2.0, 2.5, 3.0$
- $Re = 3 - 20 \times 10^6 / ft$
- Run times: 20 - 120 seconds at 20 - 30 minute intervals
- 12 × 12 inch test section



## Transonic Test Section

- Multiple configurations
  - 4 porous walls
  - 3 porous & 1 solid wall (half-body models)
  - 2 porous walls, 2 solid walls (imaging)
  - 4 solid walls **Typical PIV Configuration**
- Test section enclosed in pressurized plenum





# The Particle Seeder

Particles generated from mineral oil and delivered into the TWT stagnation chamber.

In situ measurements of the particle size show  $d_p = 0.7 - 0.8 \mu\text{m}$ .

Is the particle response time good enough?

Stokes Number = 0.04

- $St < 1$  is acceptable
- $St < 0.1$  is very good

particle response  
is excellent





# The Particle Seeder

## Corona ViCount smoke generator:

- Particle diameter 0.2-0.3  $\mu\text{m}$ 
  - Generated from mineral oil
- Placed in pressure vessel to drive particles into stagnation chamber.

## Particle Injection

- Deliver particles upstream of the TWT flow conditioning.
- Insert tubes slotted with holes to deliver particles into TWT.



# The Particle Seeder

## Corona ViCount smoke generator:

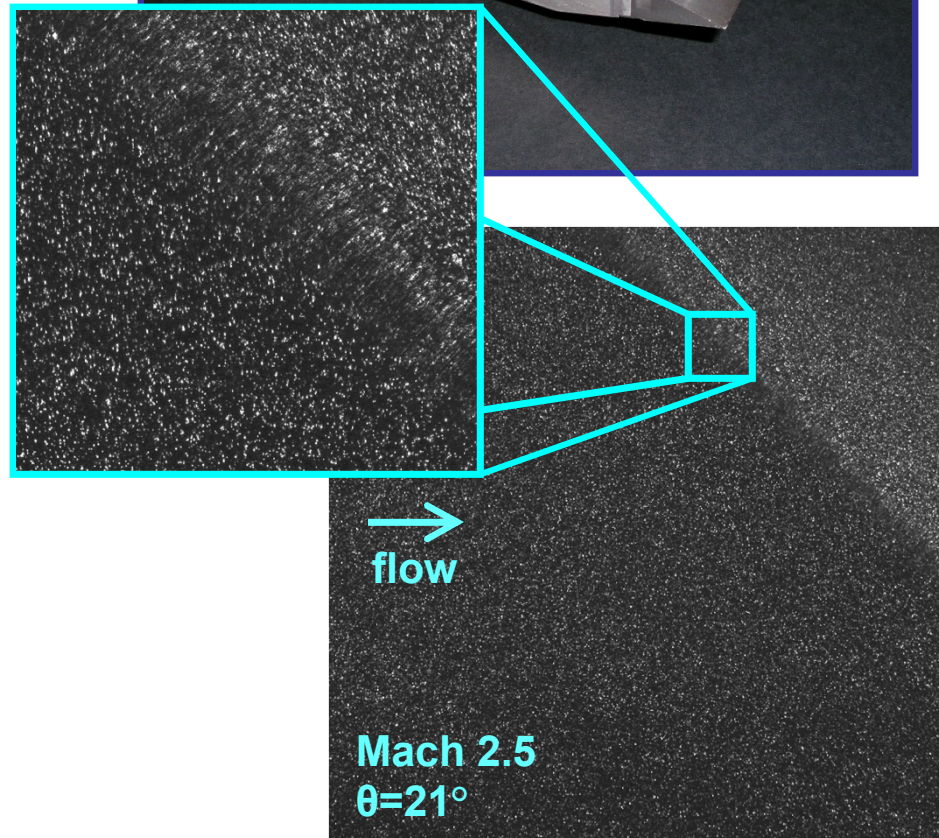
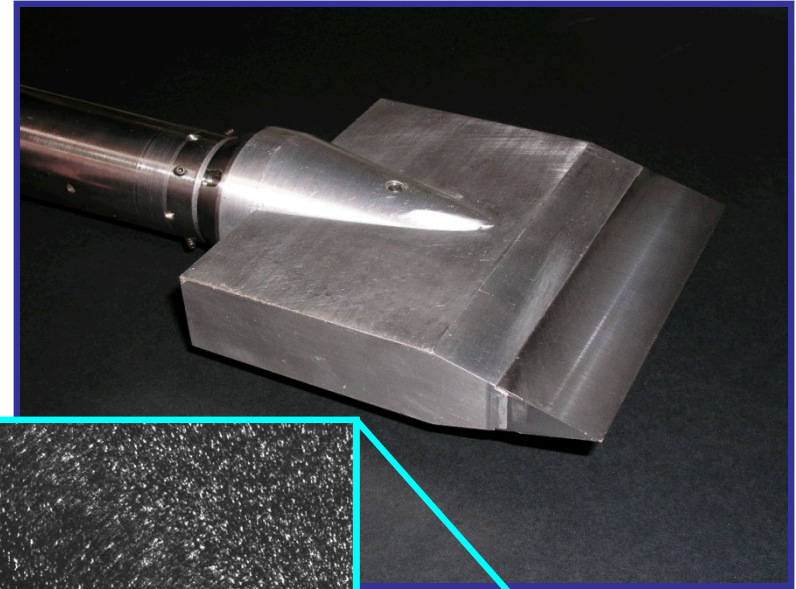
- Particle diameter 0.2-0.3  $\mu\text{m}$ 
  - Generated from mineral oil
- Placed in pressure vessel to drive particles into stagnation chamber.

## Particle Injection

- Deliver particles upstream of the TWT flow conditioning.
- Insert tubes slotted with holes to deliver particles into TWT.

## Test the particle response across a shock generated by a 15° wedge.

- Machs 1.5, 2, and 2.5
- Pitch wedge to get different shock angle  $\theta$





# Particle Response

Over a range of Machs and shock angles:

$$\tau_p = 1 - 2 \mu s$$

$$d_p = 0.7 - 0.8 \mu m$$

Particle diameter is larger than the manufacturer specification.

- Probably due to agglomeration when the smoke is ducted to the stagnation chamber.

Is this particle size and response time good enough?

What is a typical turbulent velocity gradient?

- $(du/dx)_{max} \approx 3\%$  of the interrogation window
- At Mach 2.5, this yields  $\tau_f = 50 \mu s$

Stokes Number =  $\tau_p / \tau_f = 0.04$

- $\tau_p / \tau_f < 1$  is acceptable
- $\tau_p / \tau_f < 0.1$  is very good

particle response  
is excellent





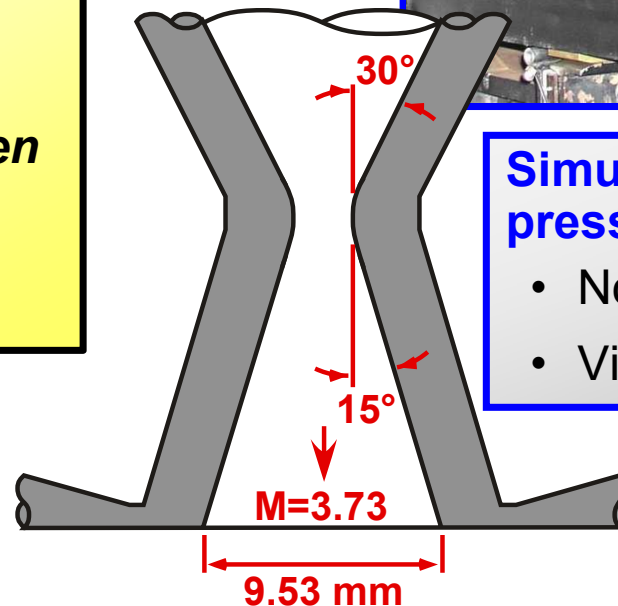
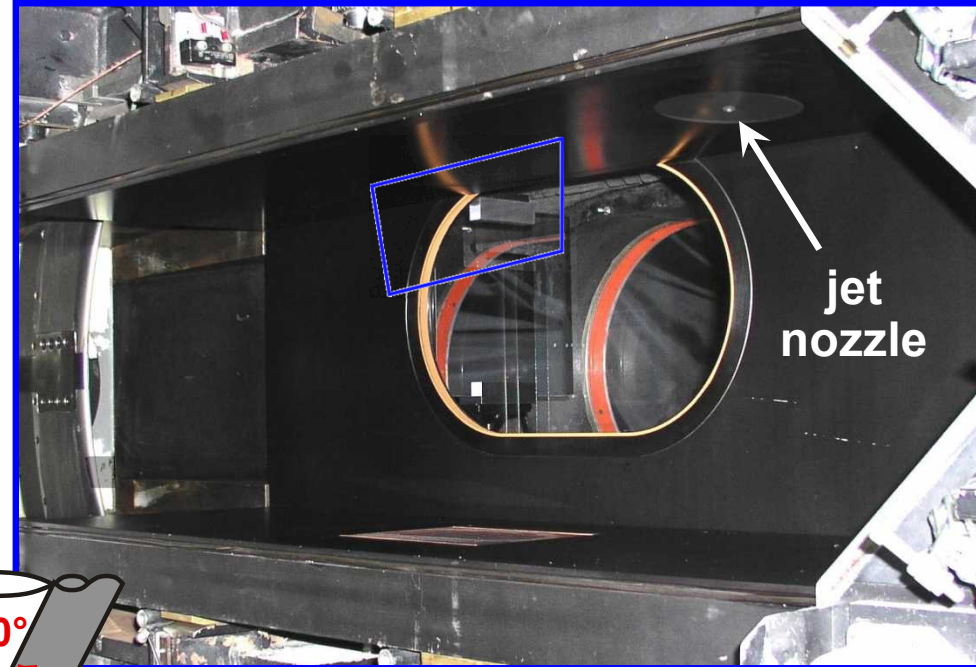
# Our First Test Application

## Supersonic Jet in Transonic Crossflow



**Spin rockets control vehicle rotation.**

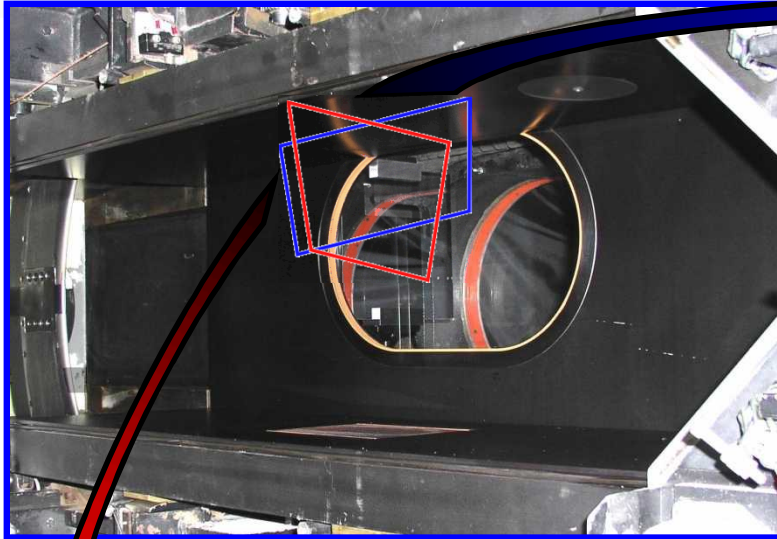
*An interaction between the jet plume and the fins alters the aerodynamics.*



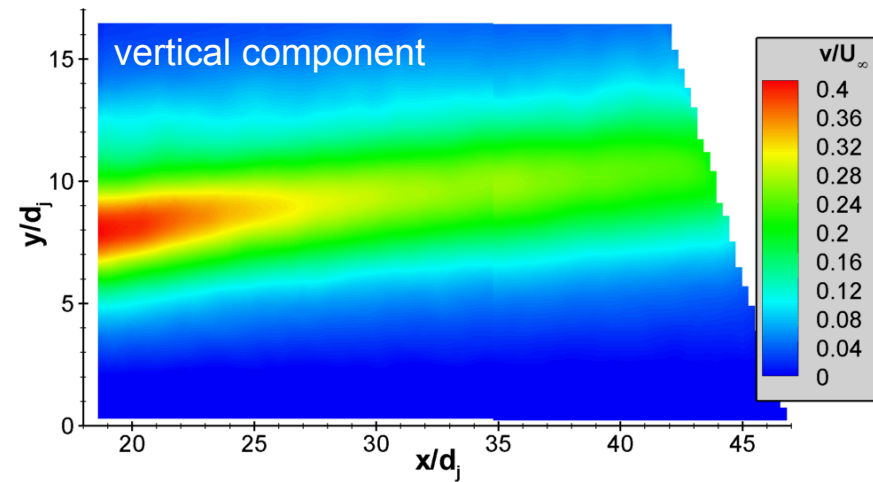
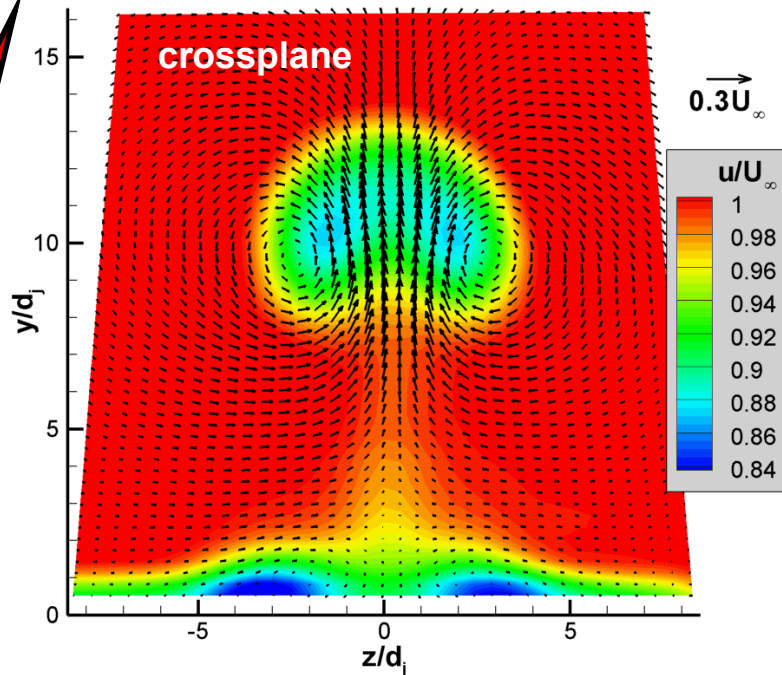
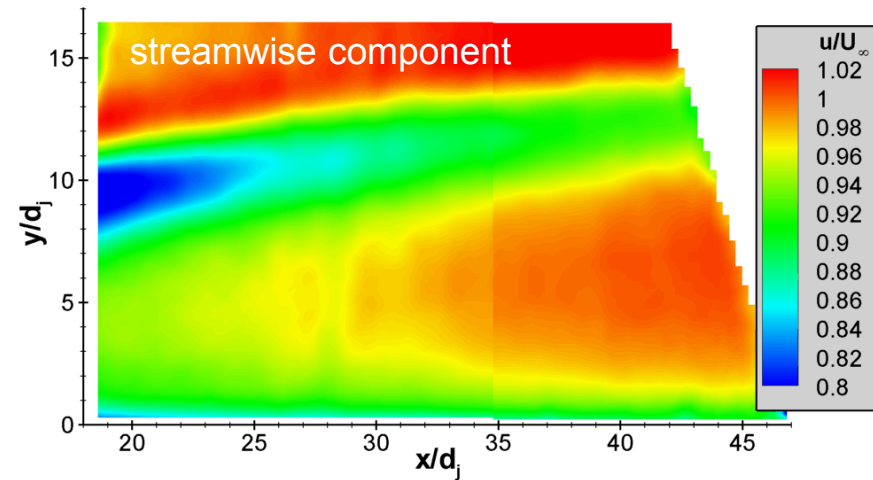
**Simulate the spin rockets with high pressure air through a nozzle.**

- Nozzle mounts on top wall of TWT.
- View the far-field of the interaction.

We have a lot of previous data on this configuration.



streamwise plane



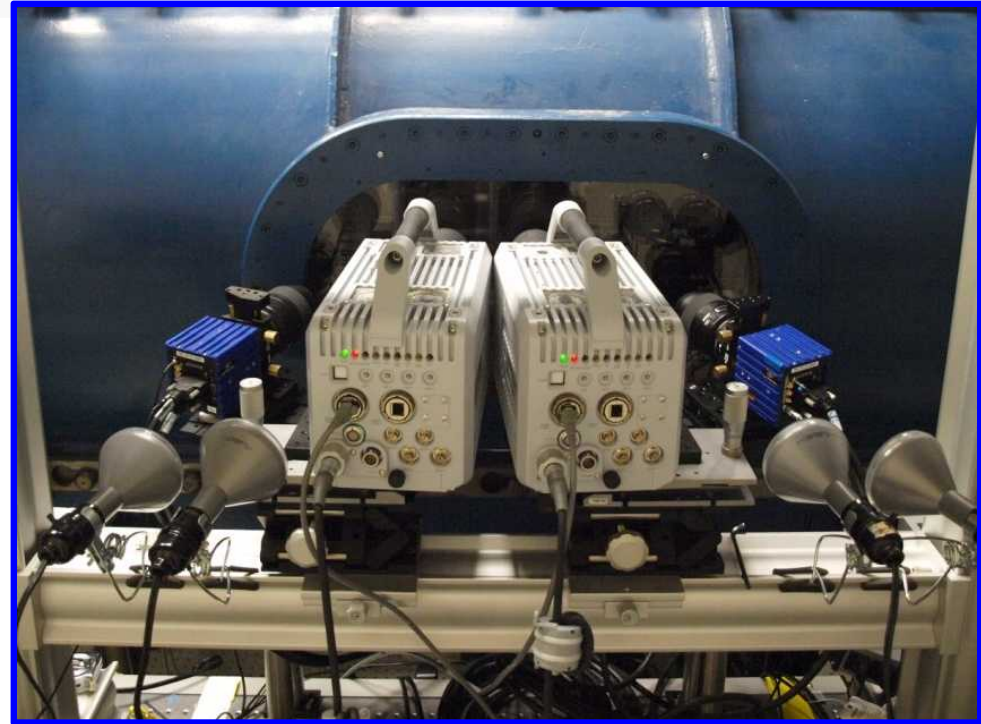
# High-Speed Cameras

## High-Speed Cameras

- Photron SA-X2
- Two side-by-side for wider field of view
- Two-component PIV

## Camera Orientation

- Cameras canted at  $5^\circ$  due to large size of camera body.
- Max error in streamwise component is  $< 2\%$ .



## Present experiments:

- 50 kHz framing rate
- $640 \times 384$  pixels
- Frame straddle pulse pairs

## Present laser settings:

- 25 kHz of pulse pairs
- $\Delta t = 2.00 \mu\text{s}$
- 2.5 ms burst, 175 mJ/pulse



# Field of View

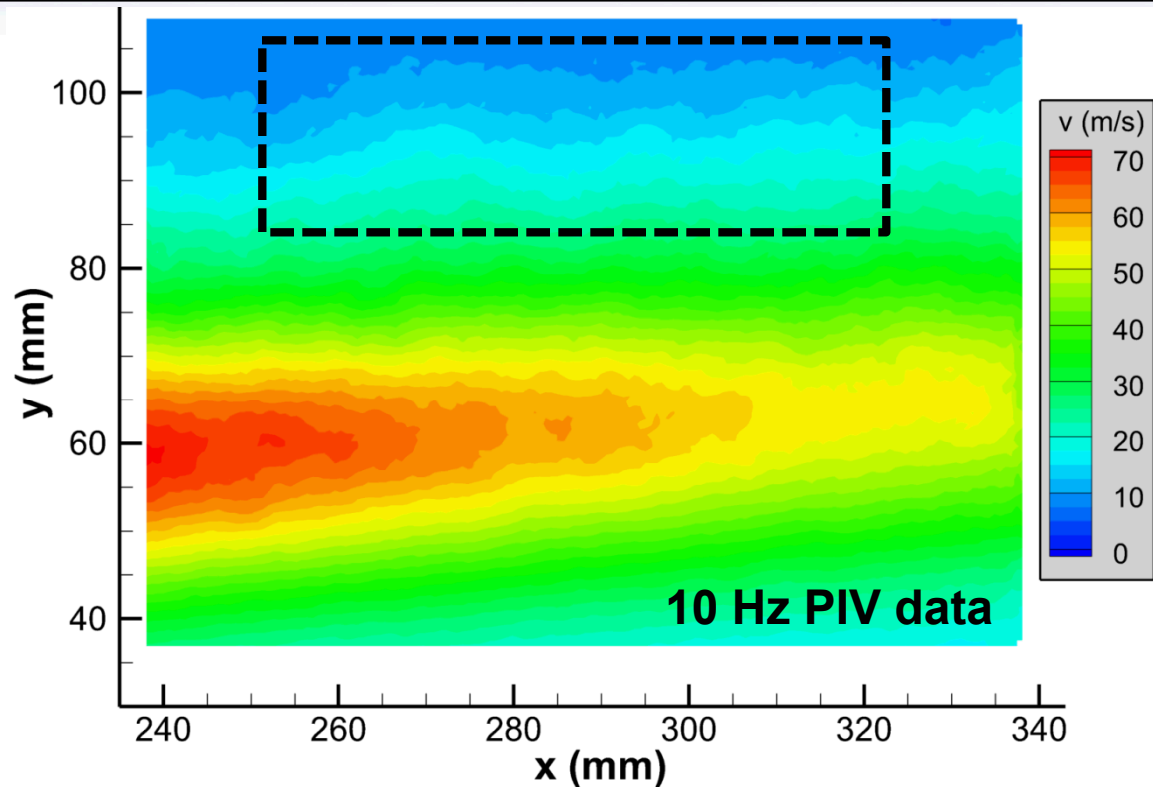
## Combined field of view:

Image turbulent eddies at the outward mixing layer.

## Today's data at J=8.1

Far from jet core and sparser turbulent eddies.

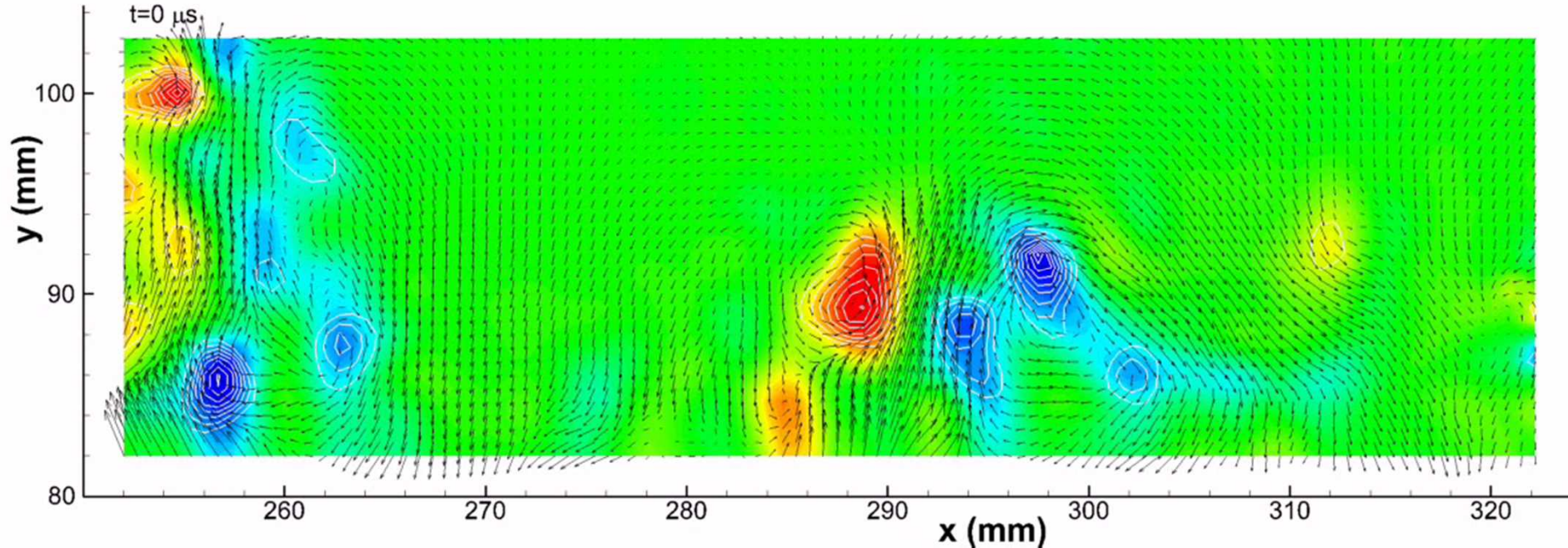
Makes data more visually interpretable.



← jet exit

# A Sample Pulse-Burst PIV Movie

*This is a 2.5 ms movie with 63 vector fields acquired at 25 kHz.*



(920)

Velocity fluctuations are shown.

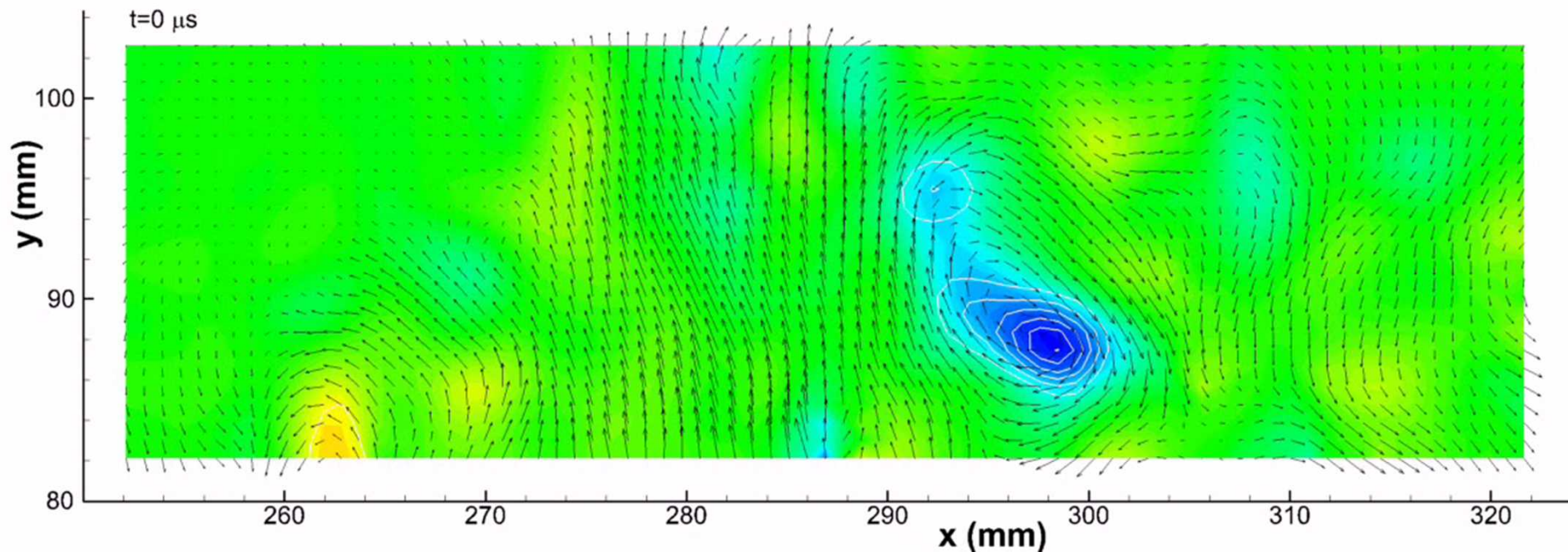
Final pass uses  $24 \times 24$  pixel interrogation windows.

Counter-rotating eddies convect past, typically in pairs.

- About 8-10 mm separating eddies in a pair
- About 20-30 mm separating pairs

# Increase the Frequency with Double Exposures

*Run the laser at 50 kHz, double-exposing pulse pairs on single images.*



(1700)

**Process using autocorrelations.**

This works because there's no directional ambiguity.

**Final pass is  $32 \times 32$  pixel due to increased noise.**

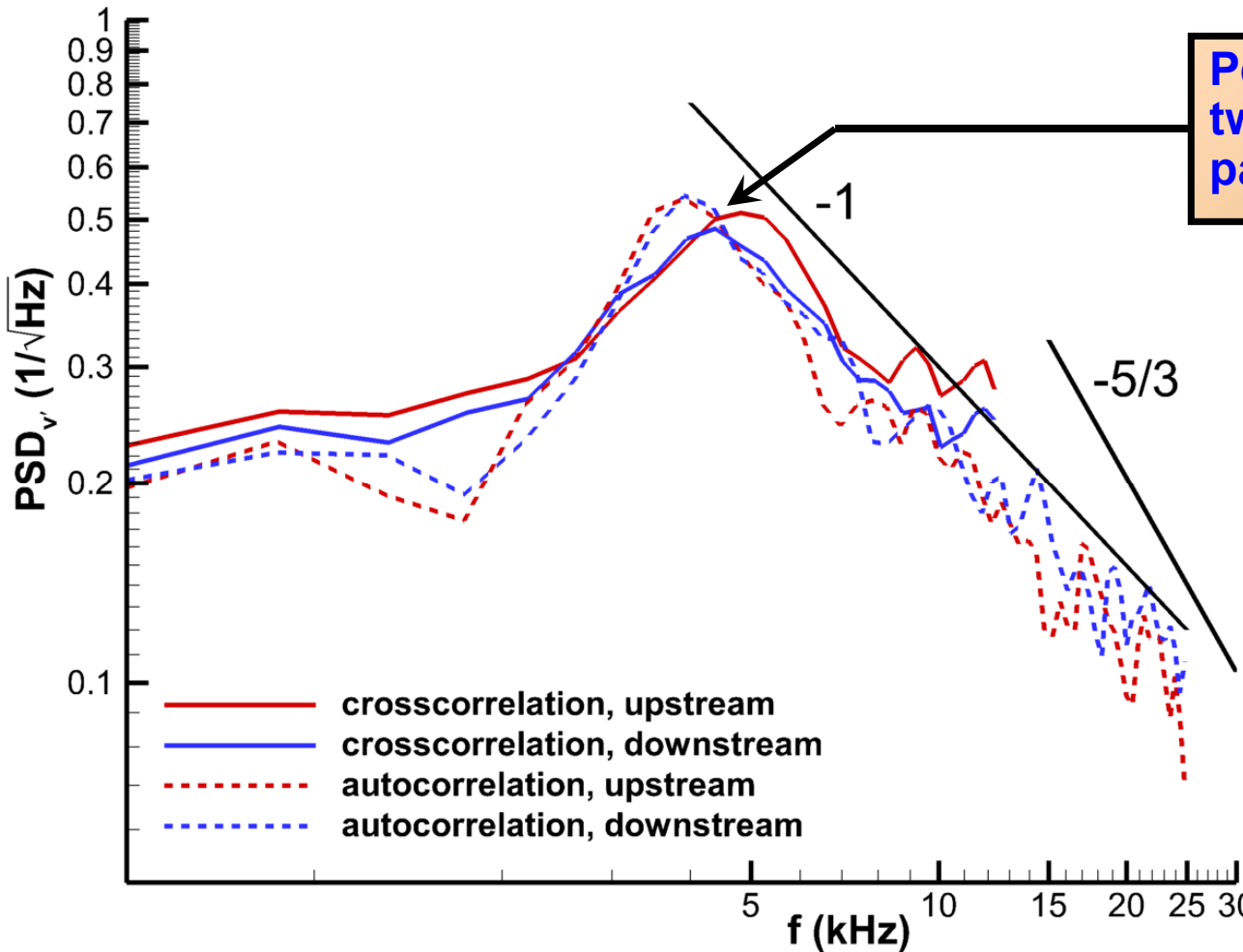
**Increased framing rate helps visualize vortex coalescence and decay.**

**Single eddies can become stable and long-lived.**



# What else can we do with Pulse-Burst PIV?

*Compute power spectra from the time signal of each vector.*



Peak corresponds to about twice the spacing of eddy pairs.

Inertial subrange should show -5/3 slope.

Does not begin until about 20-30 kHz.

But we do see an apparent “-1” power law.

Historically elusive and controversial for velocity fields.

Or is it a measurement artifact?

Assembled from 53 bursts of 25 kHz data,  
25 bursts of 50 kHz data.

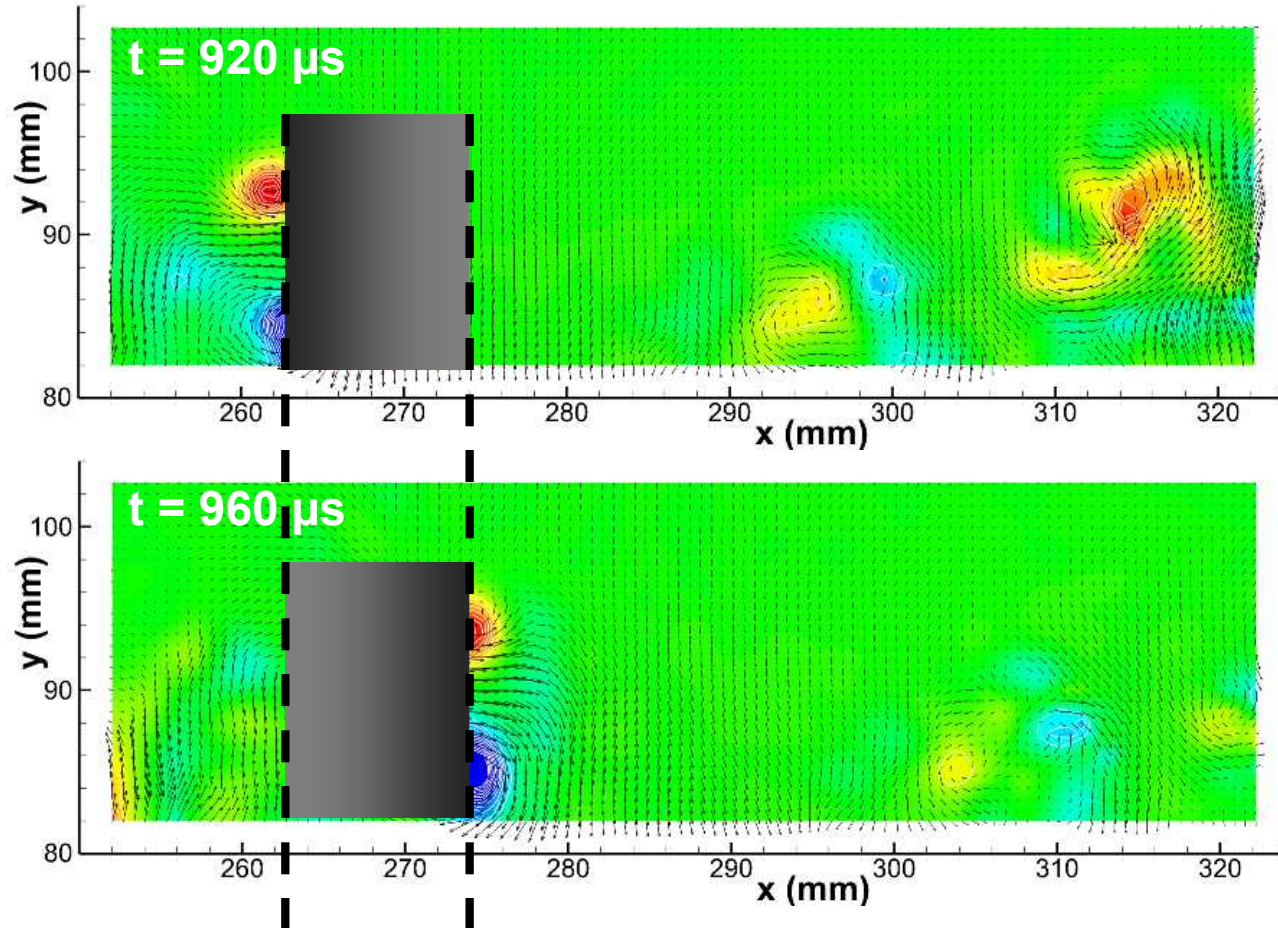
# Velocity Supersampling

*Use the supersampling algorithm of Scarano and Moore (2012).*

## **“Pour space into time”**

Between successive velocity fields, the flow convects by 16 vector spacings.

Use local convection velocity and Taylor's hypothesis to convert the intervening 15 vectors from space to time.

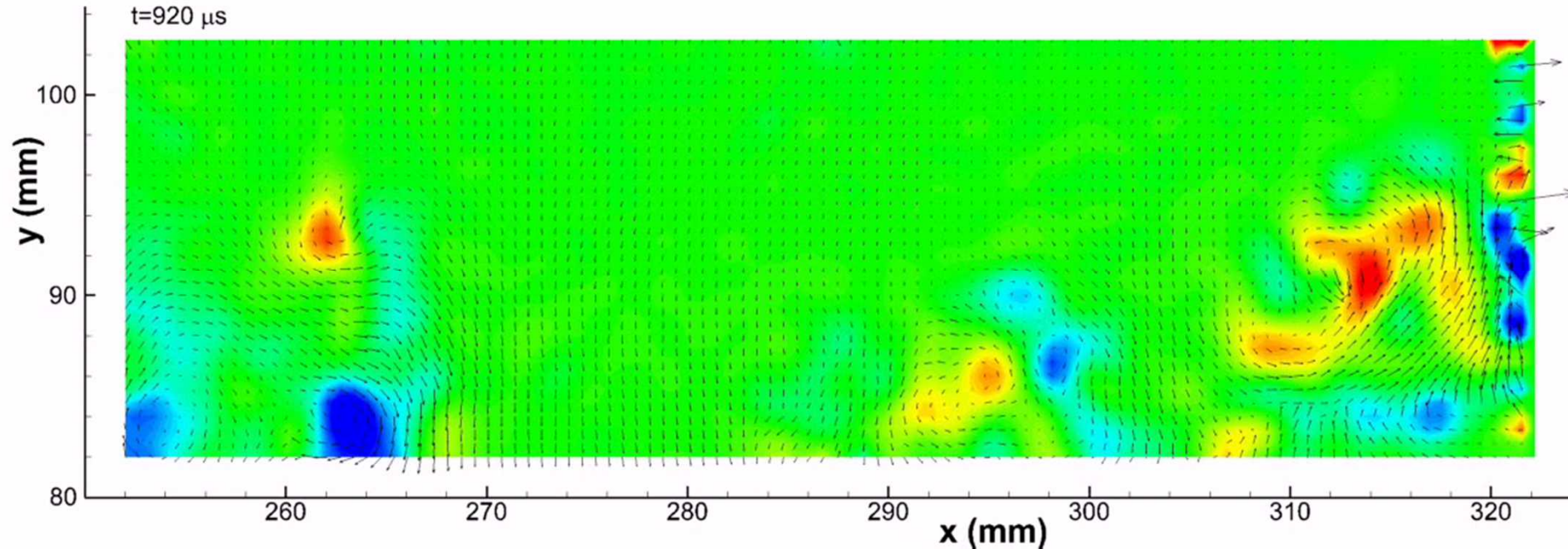


**Follow the local streamlines**

**Interpolate into new  
intervening vector fields**

# Velocity Supersampling

*Use the supersampling algorithm of Scarano and Moore (2012).*



We see a much smoother movie with more detail showing vortex rotation and deformation.

End effects are an artifact of extrapolating beyond the field of view.

What does supersampling reveal in frequency space?



# Velocity Supersampling Power Spectrum

*This will extend the power spectrum to much higher frequencies.*

**The -1 region is substantiated.**

Lasts for one full decade.

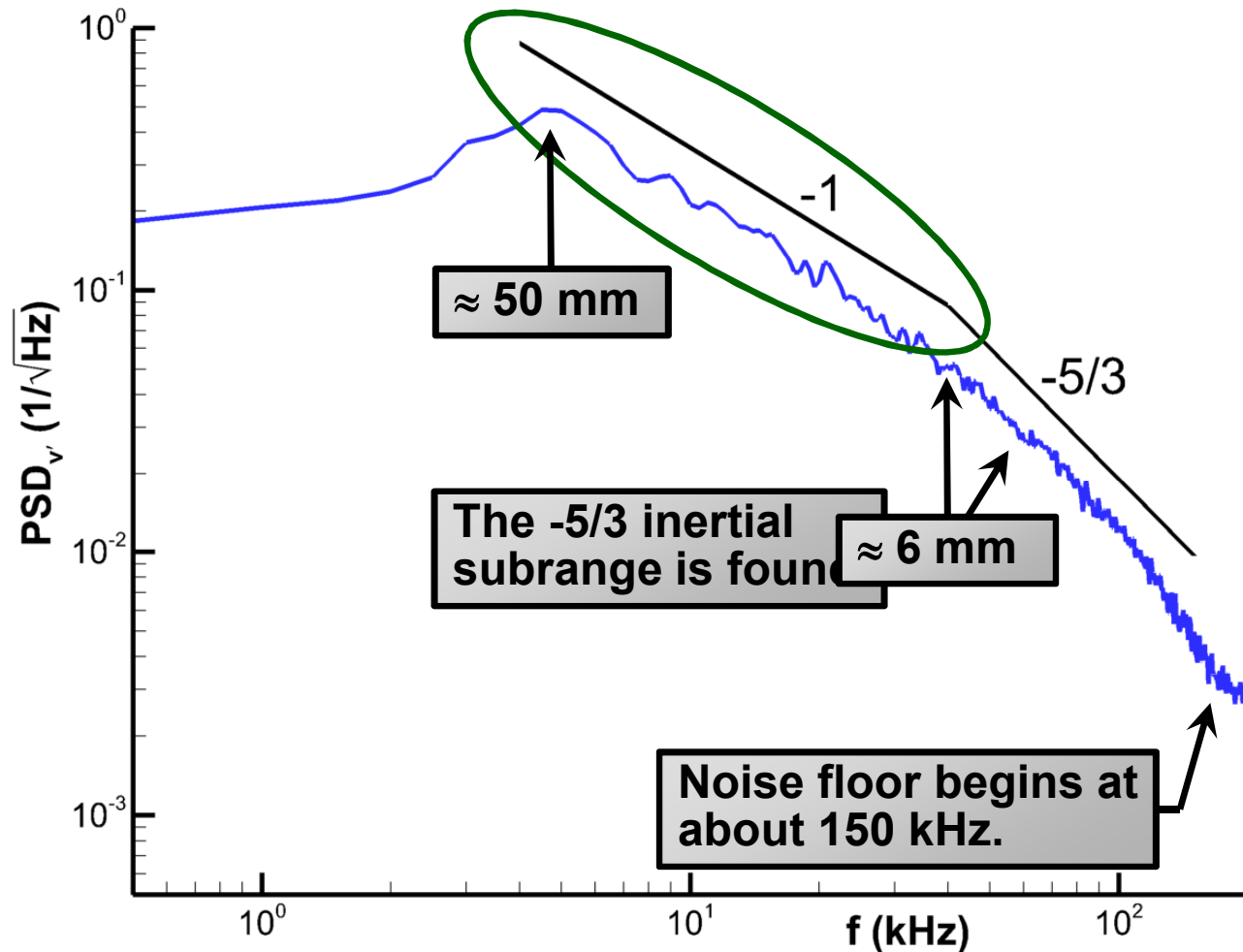
Any remaining aliasing or denoising effects  $\geq 100$  kHz.

**Scales of the -1 regime:**

Pope predicts inertial subrange starts at  $\Lambda/6 = 40 \text{ kHz} \approx 6 \text{ mm}$ .

PIV spatial resolution is about 1 mm.

**Corresponds to the dominant turbulent eddies measured by PIV.**

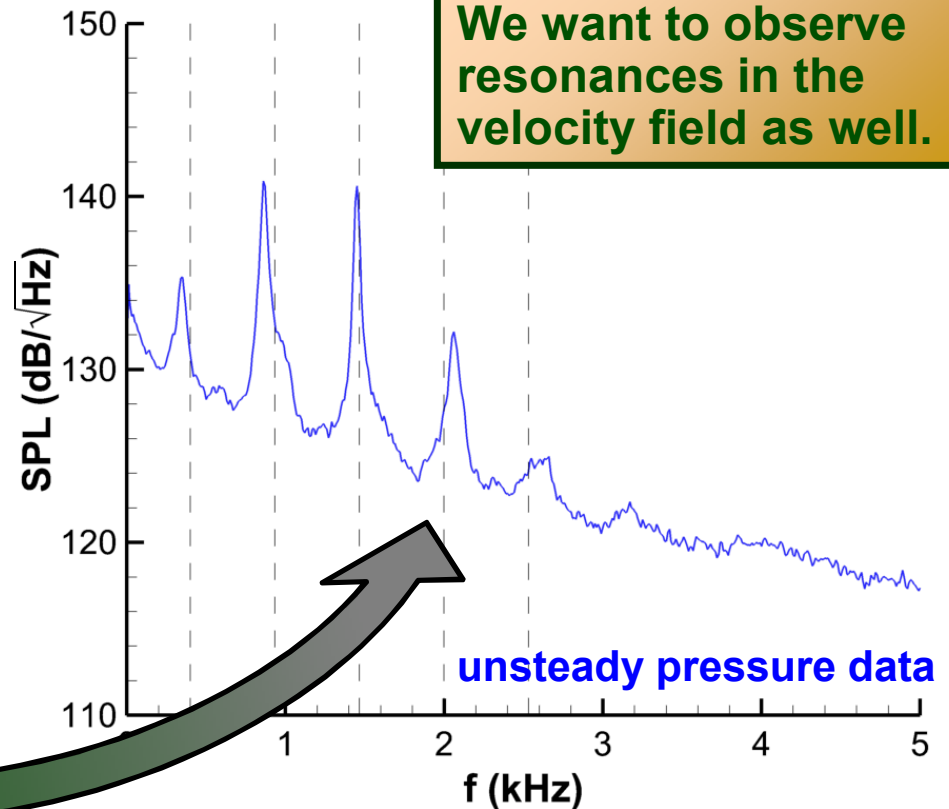


# Next Application: The Acoustics of an Aircraft Bay

Flow over an aircraft bay creates a harsh aeroacoustic environment.

Multiple strong resonance tones.

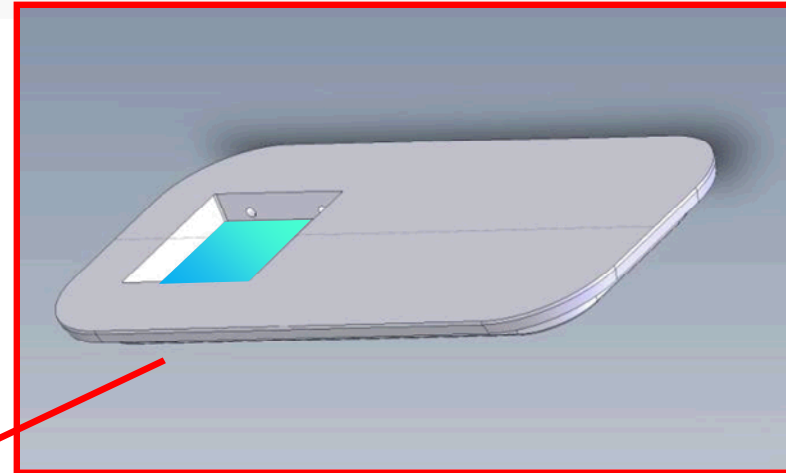
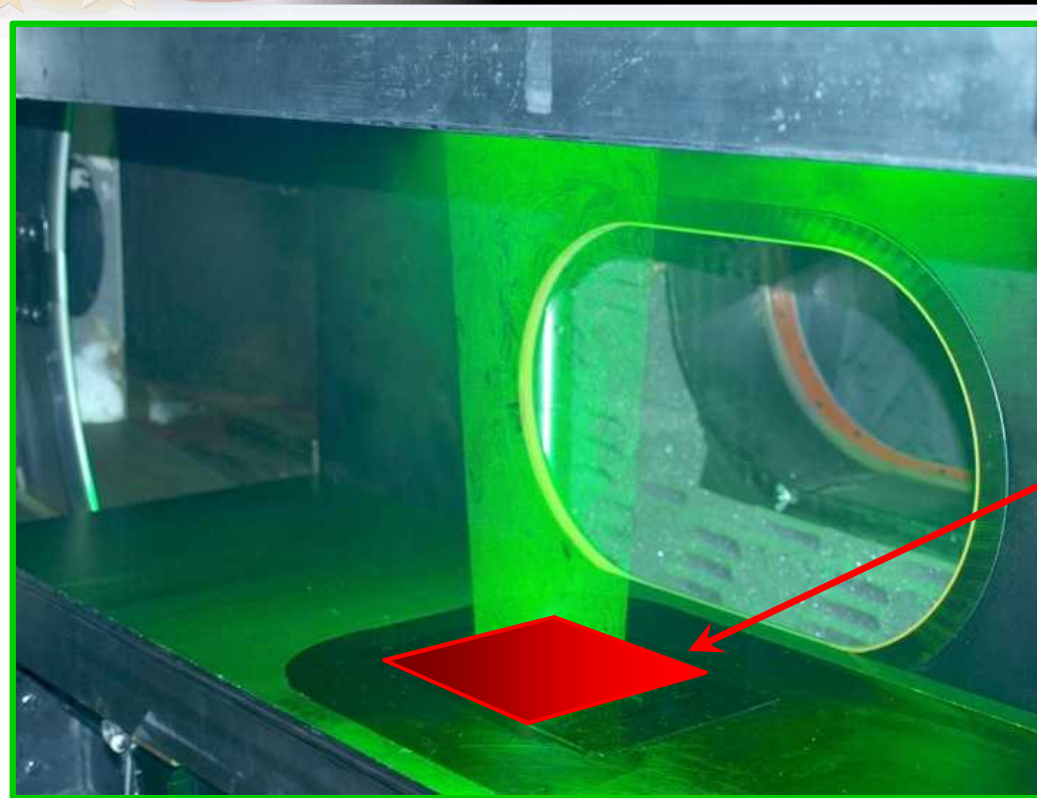
We want to observe resonances in the velocity field as well.



F-35 Weapons Bay  
(photo: Lockheed Martin)

Aircraft bays can be well represented by a simple rectangular cavity.

# Cavity Flow



**Build a cavity into the test section wall.**

**Floor is glass for laser access.**

**Our cavity is a rectangular cutout:**

- 5" long × 5" wide × 1" deep

**Tested at Mach 0.6, 0.8, and 0.94.**

***We have much acoustic and PIV data on this flow field.***



# New High-Speed Cameras

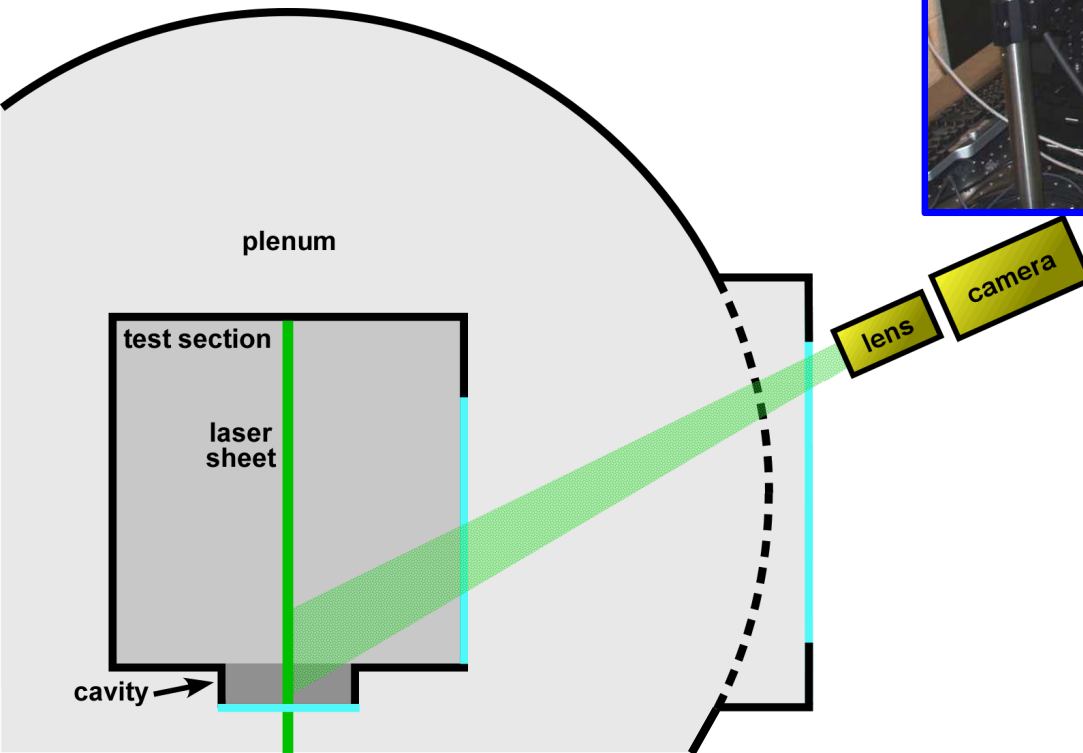
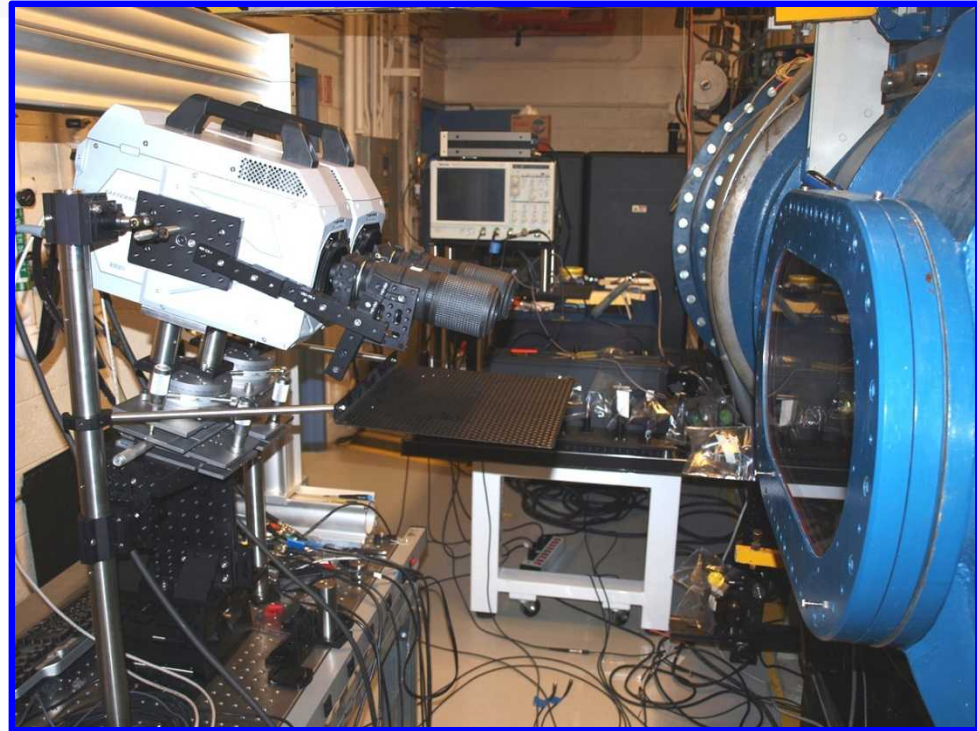
Two *Photron SA-Z's* placed adjacent for dual two-component PIV.

PIV framing rate now 37.5 kHz.

Tip cameras down by  $12^\circ$  to peer into the cavity.

Can reach about 55% depth.

Creates a bias error in vertical component.

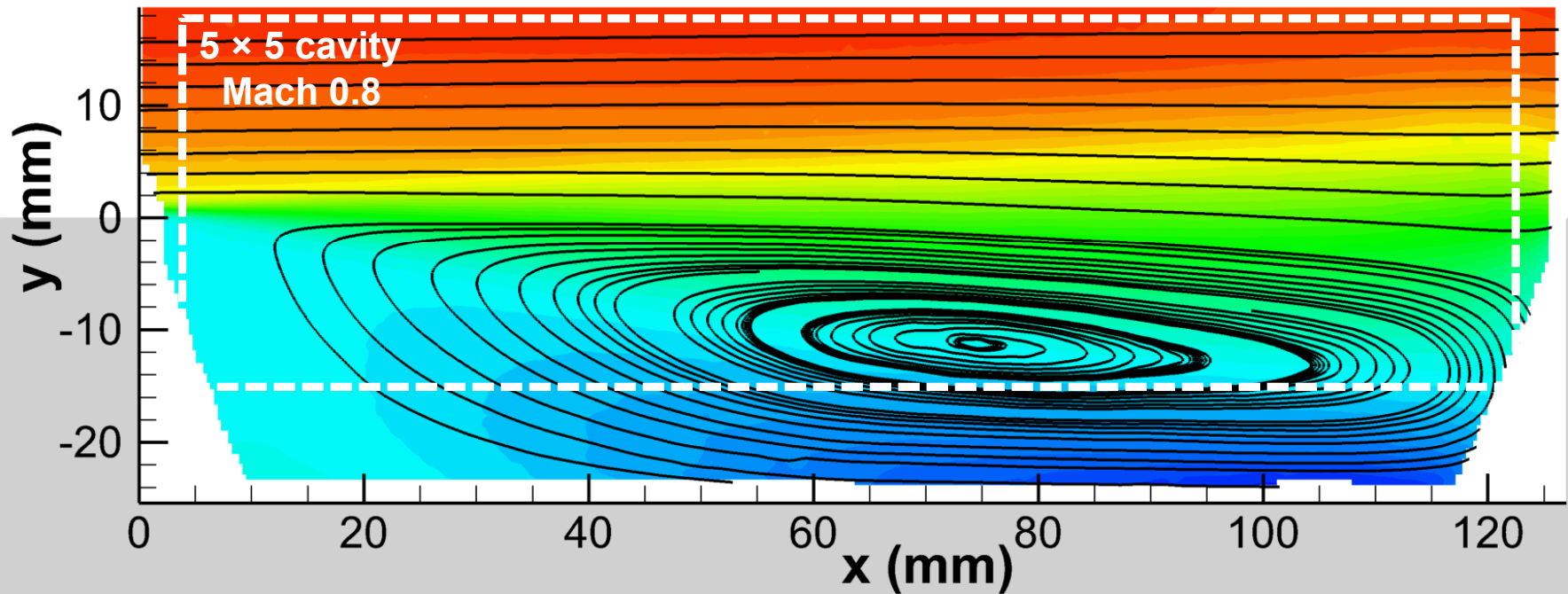
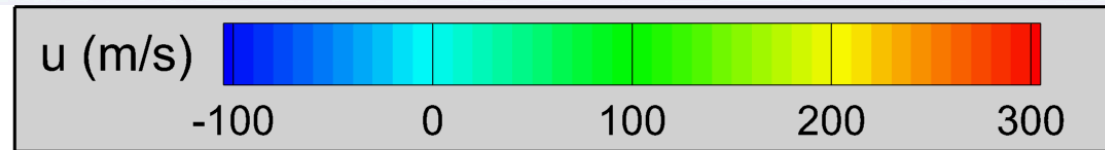


Previous 10-Hz PIV data were acquired similarly.

Bias error does not hinder visualization of the cavity flow or vortex detection.

**We have a lot of 10-Hz PIV data on this flow.**

**mean streamwise  
velocity field**



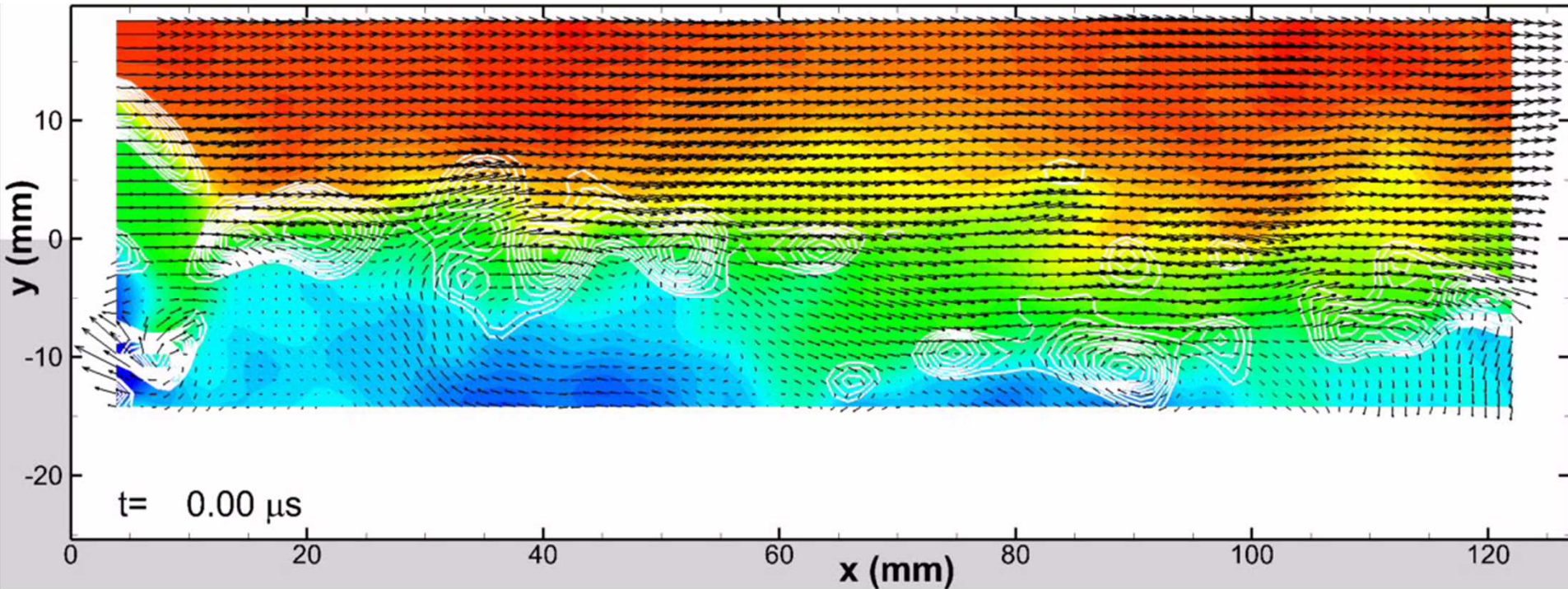
**Streamlines clearly visualize the recirculation region and strong reverse velocities are evident.**

**The pulse-burst PIV field of view visualizes most of the recirculation region and will capture reverse velocities.**

**The behavior of large-scale structures is key to the acoustic tones produced by the cavity resonance**

# A Sample Pulse-Burst PIV Movie

*This is a 10.2 ms movie with 386 vector fields acquired at 37.5 kHz.*



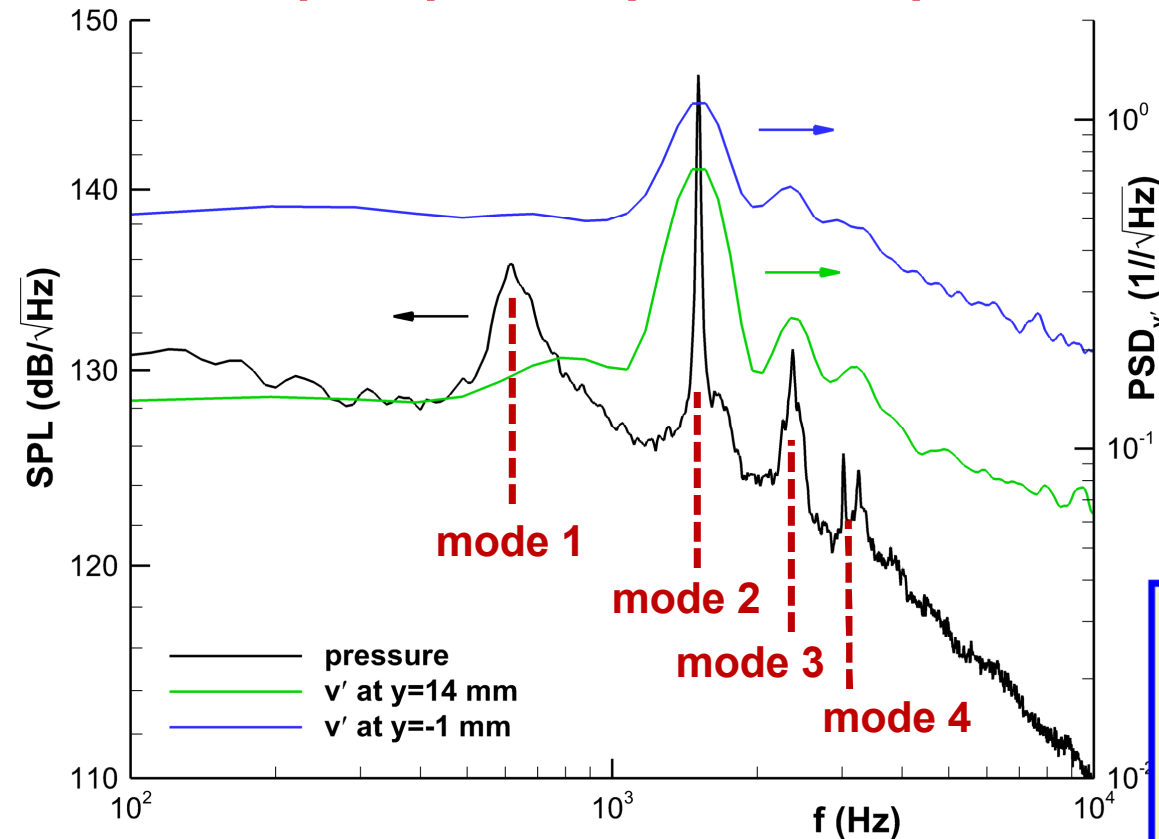
## We can visualize:

- Recirculation region shifting position.
- Ejection and impingement events at aft end of cavity.
- Recirculation events enhancing shear layer flapping.
- Growth of shear layer structures and their recirculation.



# Can we identify the cavity resonances using Pulse-Burst PIV?

**Compare power spectra to a pressure sensor in the aft wall.**



Extract two velocity signals:

- One above the shear layer
- One within the shear layer

**Velocity peaks broadened due to 100 Hz frequency resolution.**

Pressure frequency resolution is 10 Hz.

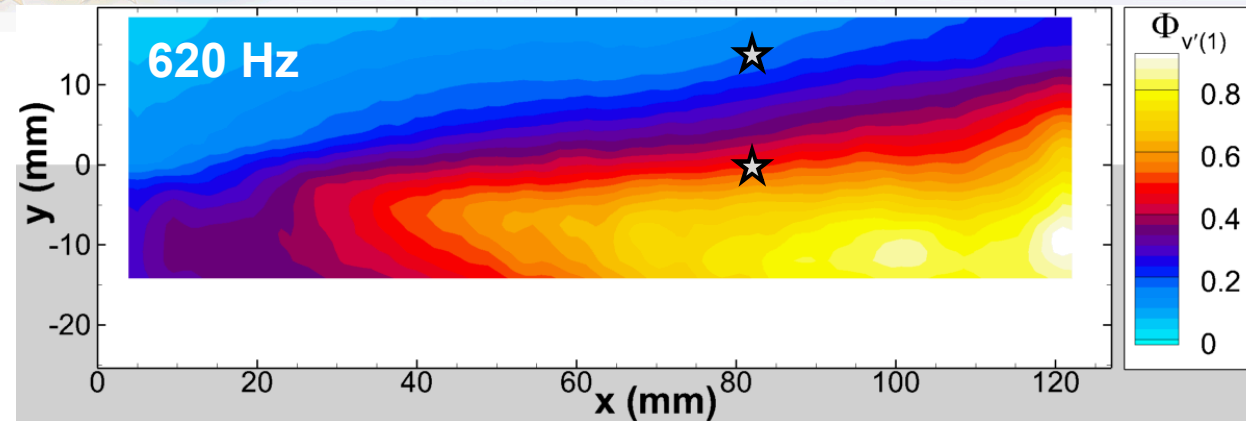
**Modes 2 – 4 match very well between pressure and velocity.**

We can even see the bifurcated mode 4 peak in the shear layer velocity data.

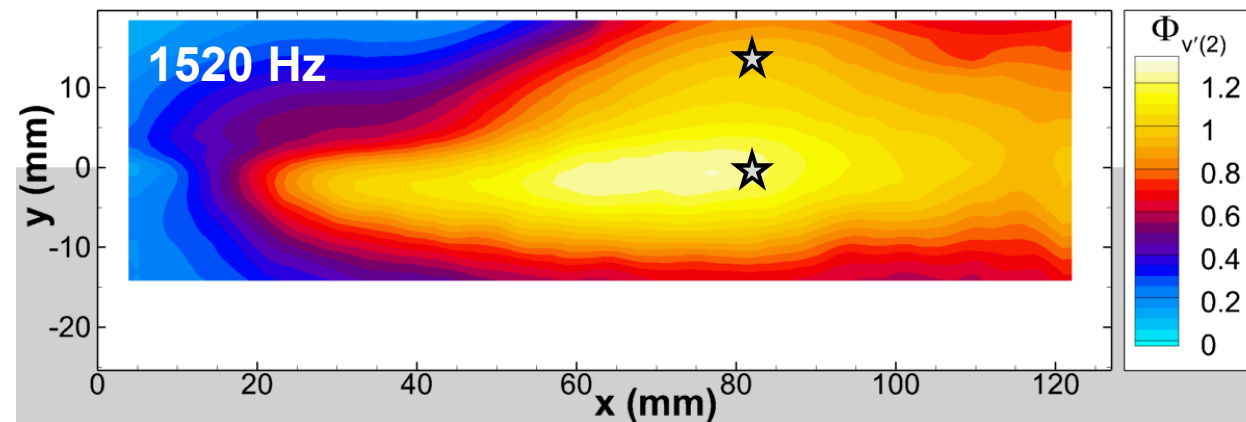
**Mode 1 is largely absent in the velocity data.**

Pulse-burst PIV allows us to look at the *spatial distribution* of the resonance modes.

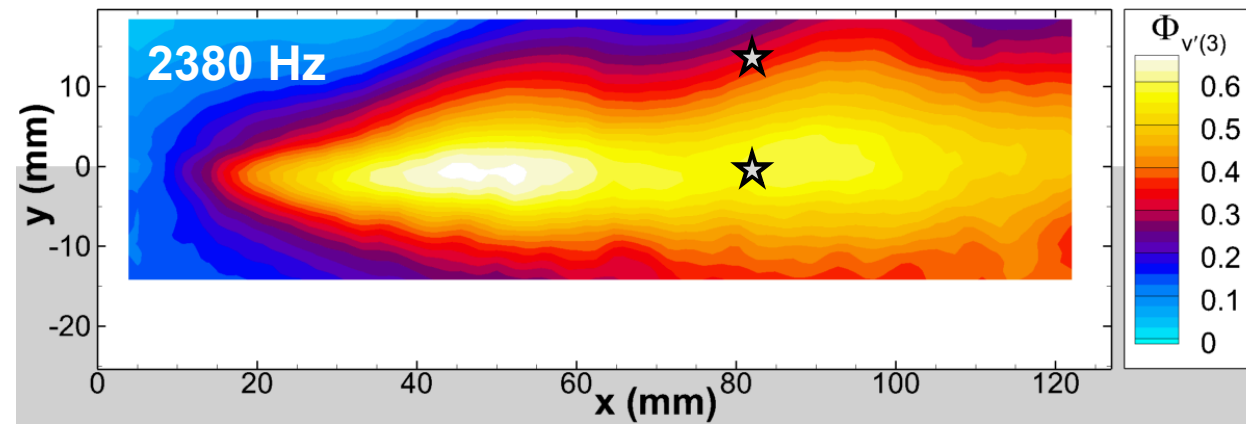
# Spatial Distribution of Resonance Modes



**Mode 1 concentrated in the recirculation region.**



**Mode 2 concentrated in the streamwise center of shear layer.**



**Mode 3 concentrated in the upstream shear layer.**

# Can we identify the turbulent structures responsible for resonance?

## Spatio-temporal correlations

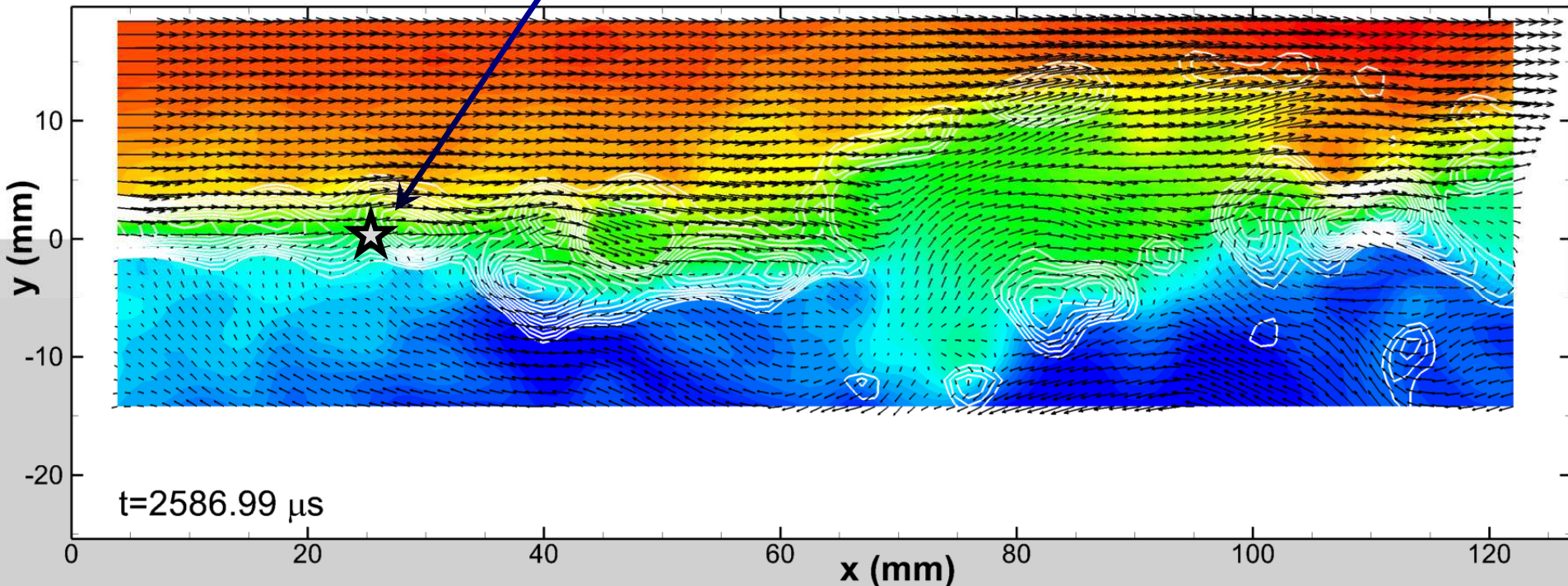
Correlate a reference vector with every vector in the velocity field.

Track correlation fields through space and time.

Choose a reference vector in the upstream shear layer.

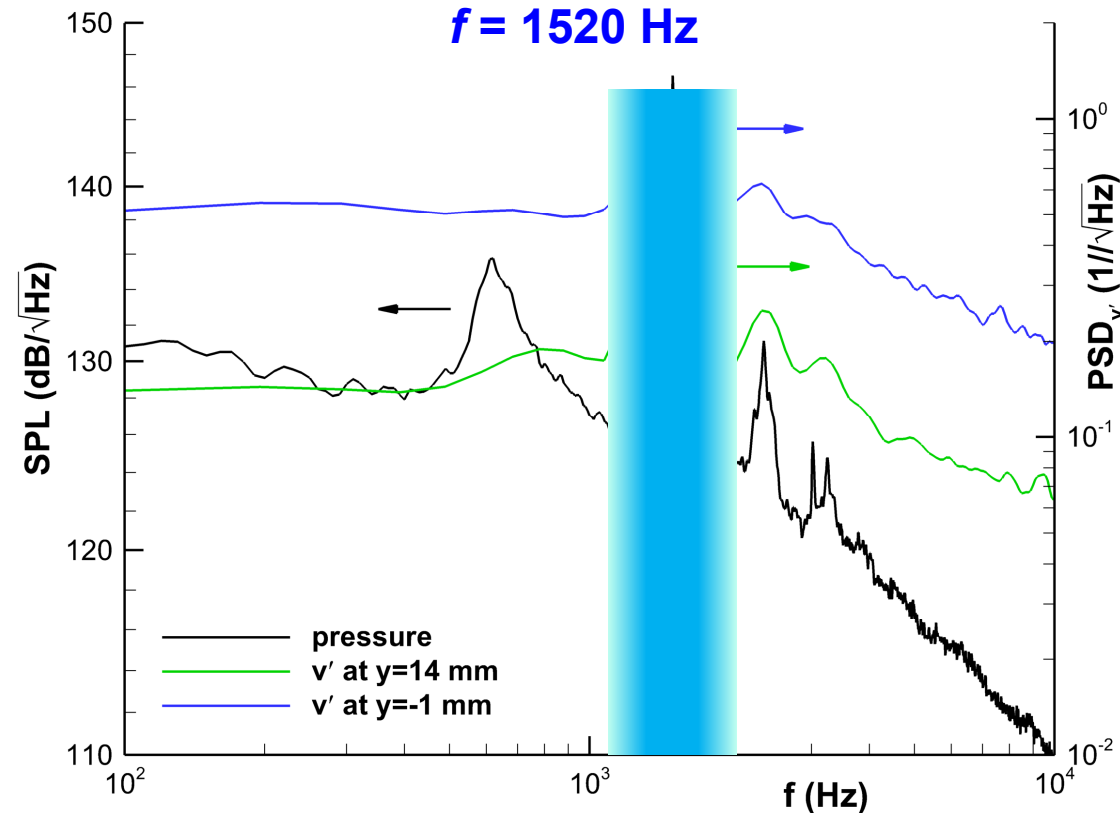
But it turns out our conclusions are not dependent upon the reference.

Correlations of the vertical component turn out to be most useful.





# Bandpass filter the velocity data.



**Mach 0.8 is dominated by the second mode.**

**But all modes may be simultaneously active, plus turbulent fluctuations.**

**Clouds detection of flowfield associated with each mode.**

**Filter each velocity signal for a specific resonance mode.**

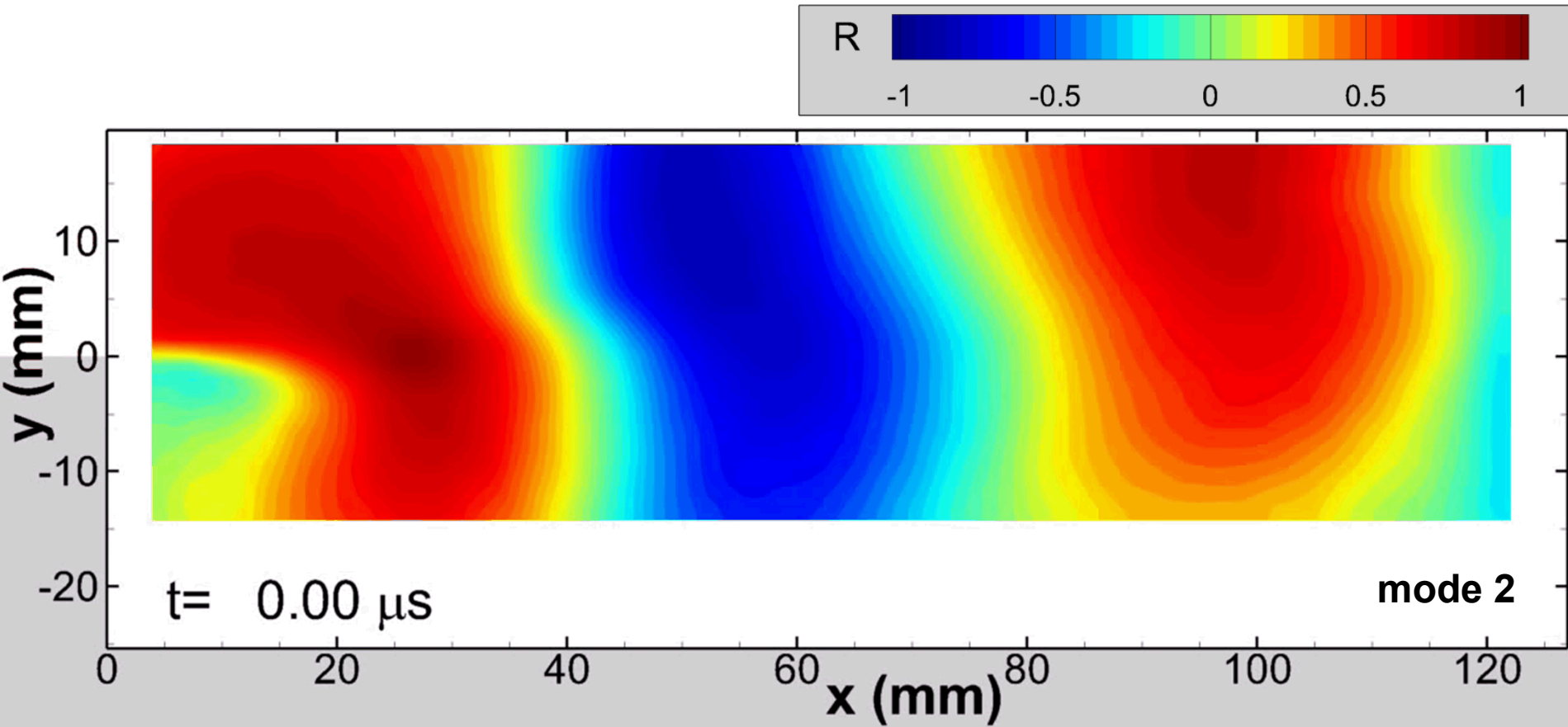
**Then, perform the correlations.**

**It turns out *mode 1* behaves differently than the rest, so let's start by discussing *mode 2*.**

# Mode 2 Correlation Field

In a narrow frequency band, we can see the characteristic structures in the flow responsible for resonance.

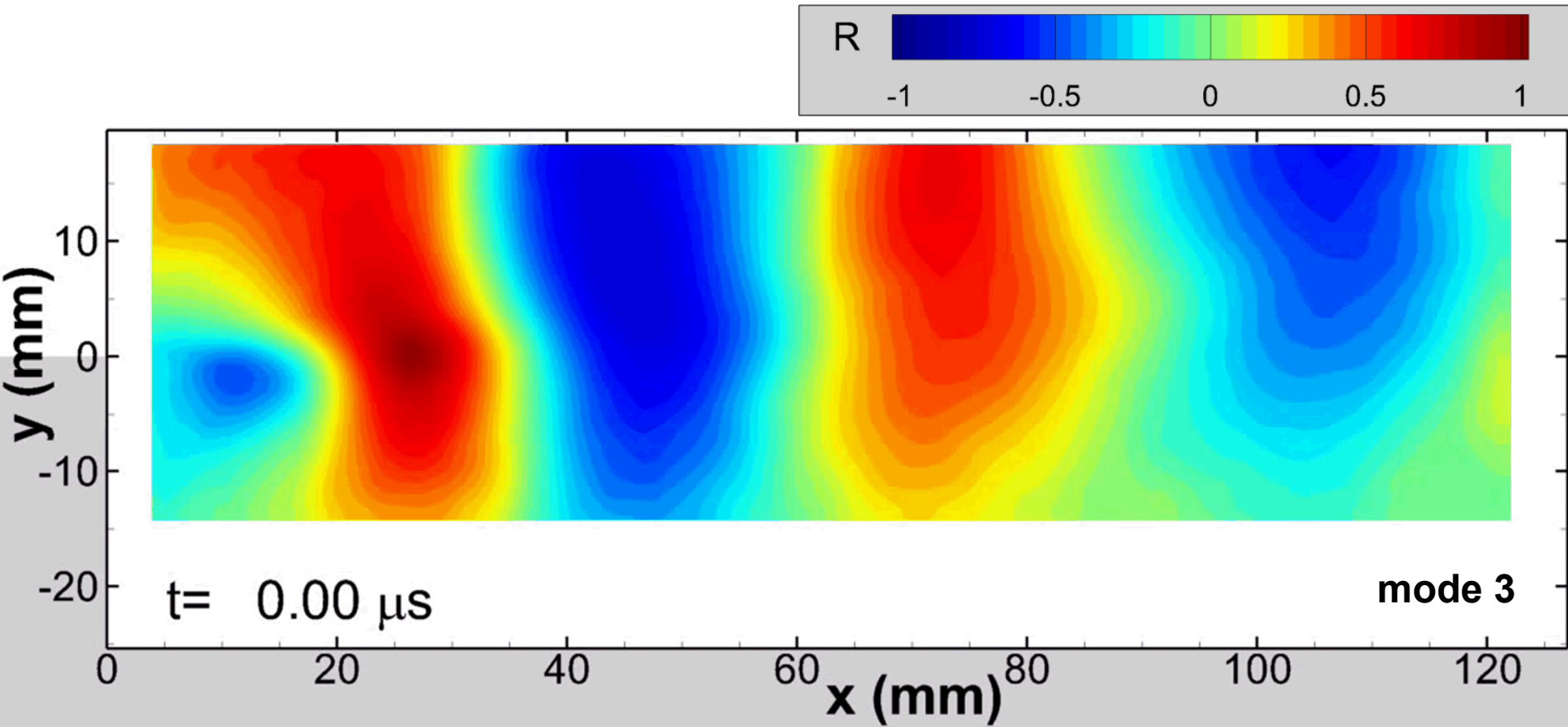
- The spacing between the structures increases as they convect downstream.
- The structures tend to drift outward.



# Mode 2 Correlation Field

In a narrow frequency band, we can see the characteristic structures in the flow responsible for resonance.

- The spacing between the structures increases as they convect downstream.
- The structures tend to drift outward.
- The spatial scale of the coherent structures shortens as the mode increases.

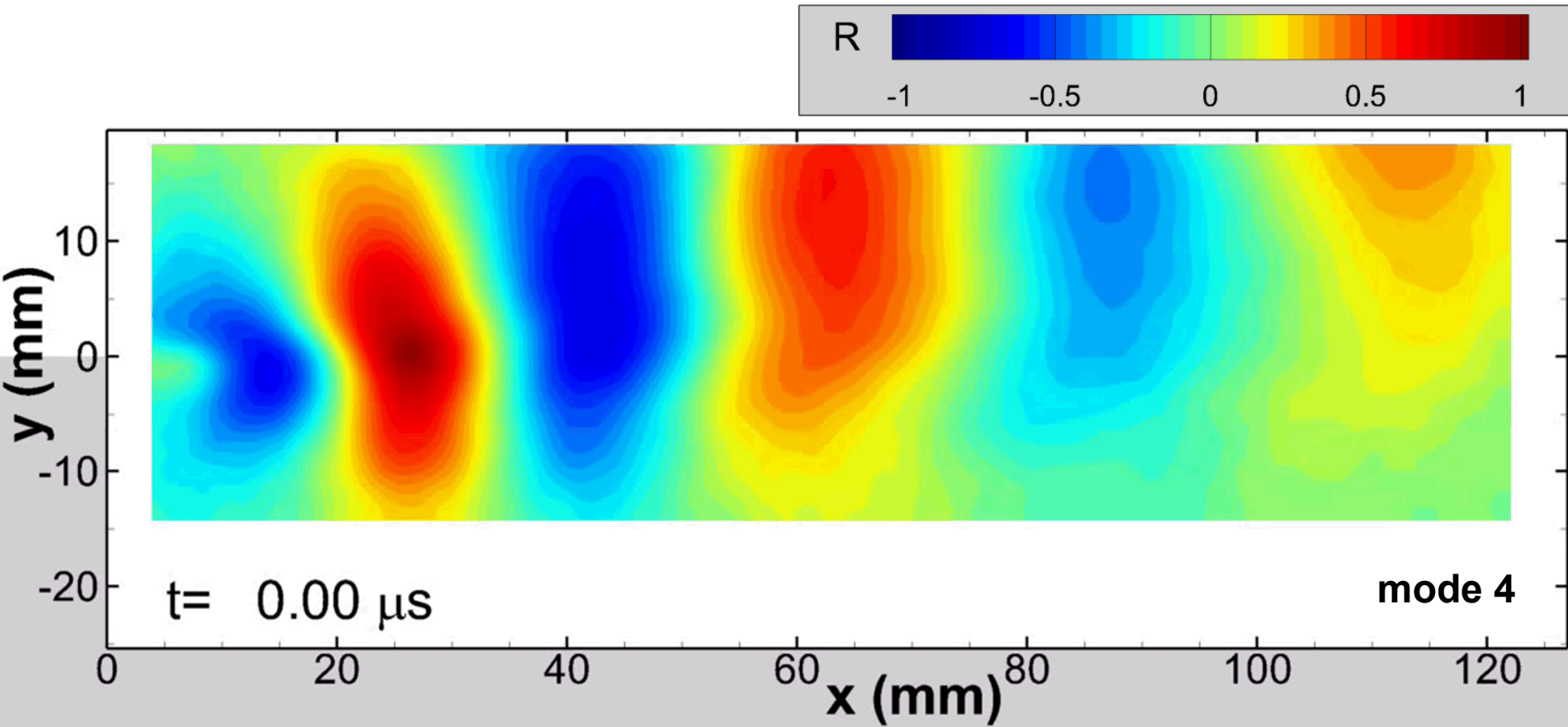




# Mode 3 Correlation Field

In a narrow frequency band, we can see the characteristic structures in the flow responsible for resonance.

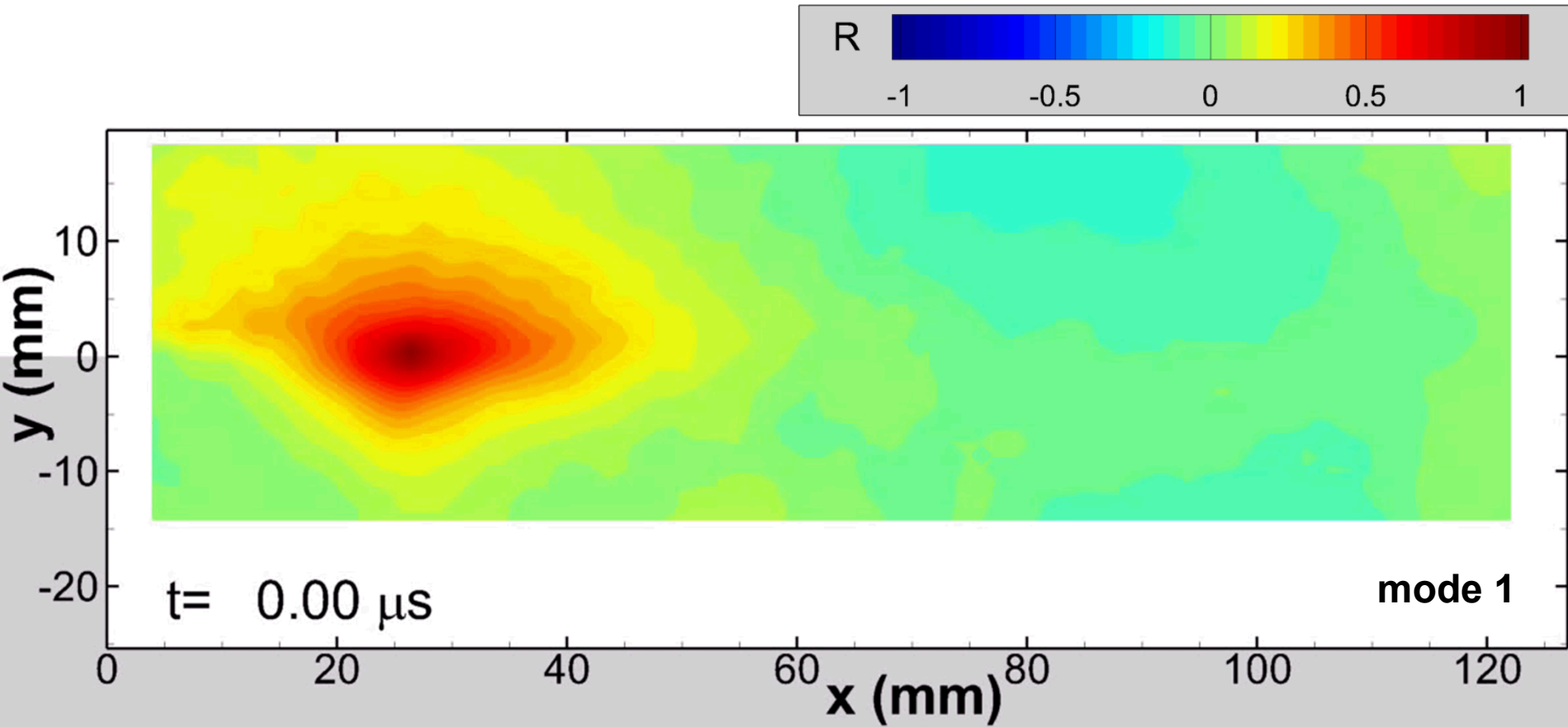
- The spacing between the structures increases as they convect downstream.
- The structures tend to drift outward.
- The spatial scale of the coherent structures shortens as the mode increases.



# Mode 4 Correlation Field

Mode 1 shows properties of a localized standing wave.

No apparent spatial resonance or convection.



# Mode 1 Correlation Field

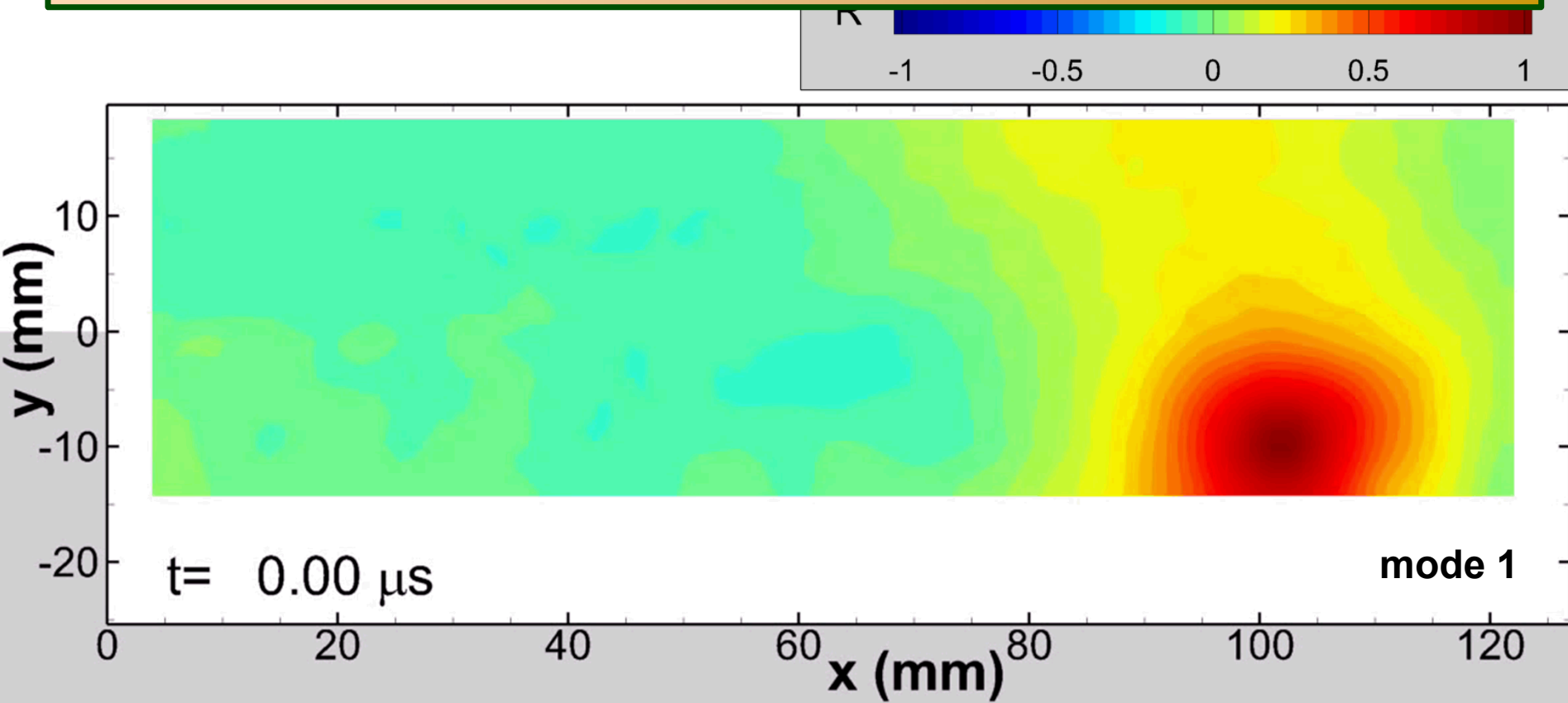
Mode 1 shows properties of a localized standing wave.

No apparent spatial resonance or convection.

Bandpass-filtered correlations do not detect mode 1 turbulent structure motion.

Because the mode 1 wavelength is longer than the cavity length.

*Pulse-burst PIV provides unprecedented data for studying resonance in compressible cavity flows, and we're full of new ideas for it.*







# What's next in the wind tunnel?

**Data up to Mach 2.5.**

**We're working on it right now!**

**Add more cameras!**

**Boost the framing rate without sacrificing spatial resolution.**

**Data analysis possibilities seem nearly endless:**

**Conditional analysis and time/space correlation**

**Bandpass-filtered movies for specific modes**

**Joint Time Frequency Analysis**

**Dynamic Mode Decomposition**

**We also are examining uncertainty issues.**

**Peak locking is a notorious difficulty, but not the only one.**

**Pulse-burst lasers make TR-PIV feasible for high-speed flows.**

***This is the first application of Pulse-Burst PIV to a ground test facility.***

# Multiphase Shock Tube (MST)

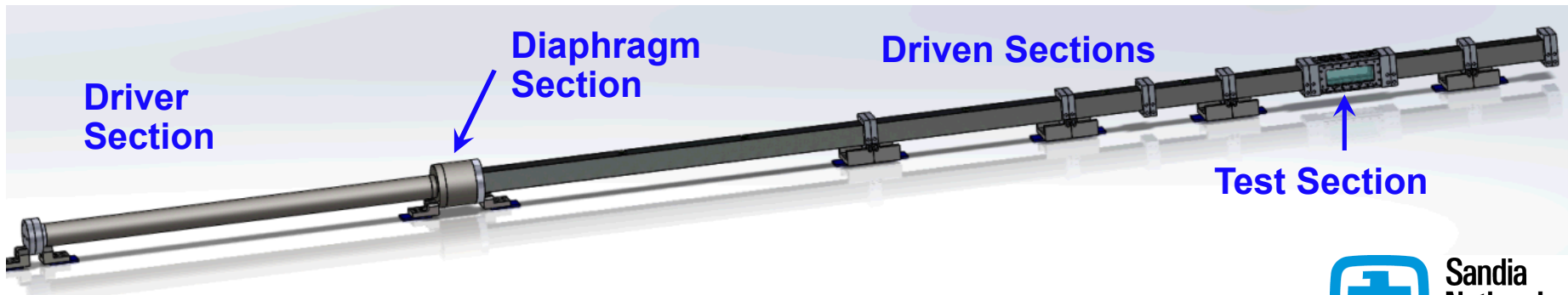
*Can we implement pulse-burst PIV in our shock tube?*

Shock Mach numbers  $M_s$  up to about 2.

Driven section is air initially at atmosphere.

Test section width  $D = 76$  mm.

*MST typically used for shock-particle interaction and liquid breakup experiments.*

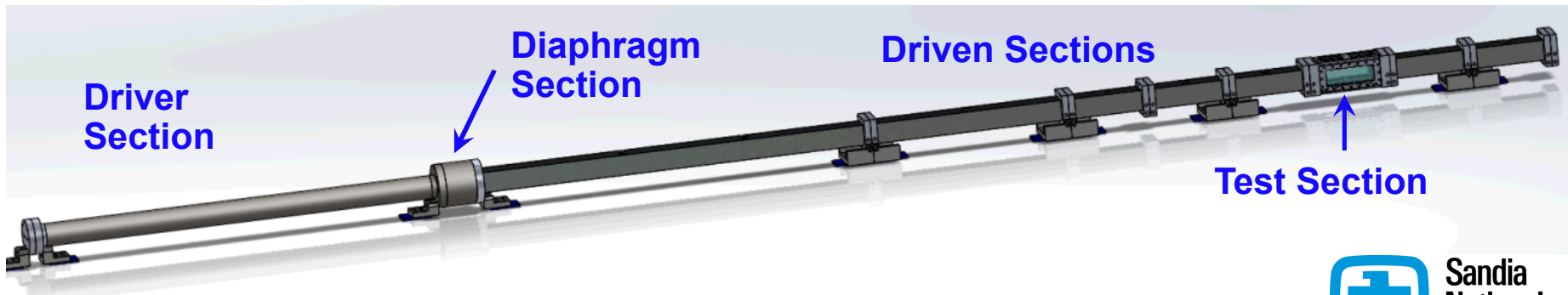


# Multiphase Shock Tube (MST)

*Can we implement pulse-burst PIV in our shock tube?*

Conventional PIV gives only one realization in the  $\sim 1$  ms test time of a shock tube.

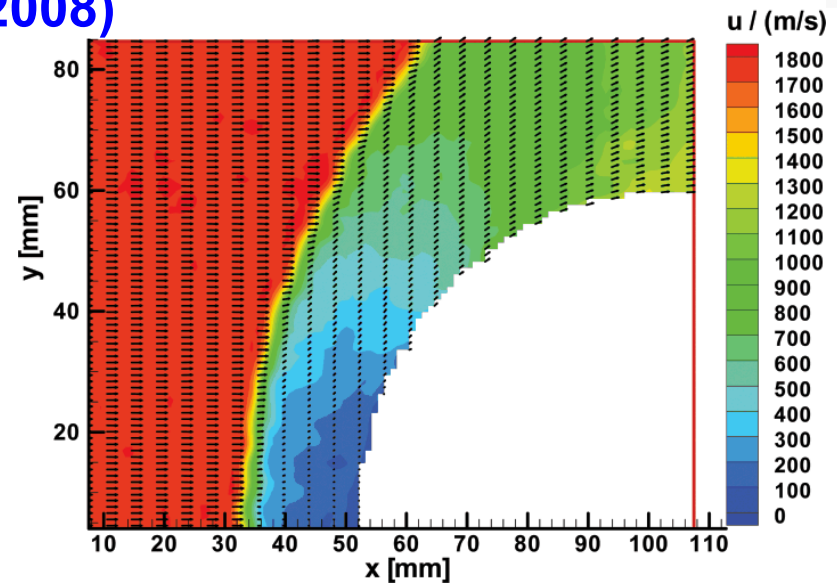
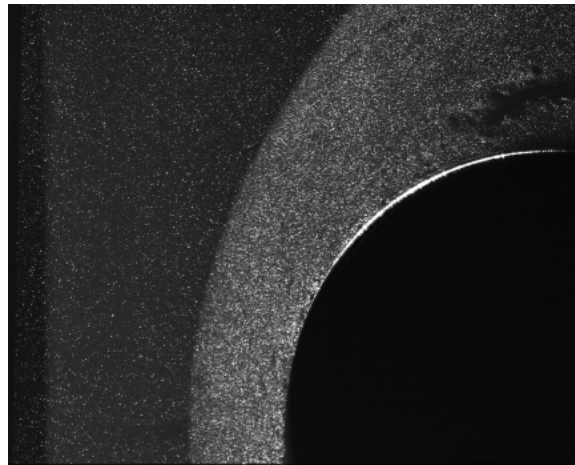
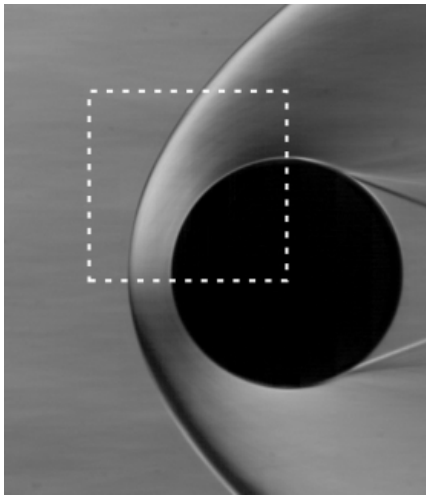
PIV pulse bursts last up to 10.2 ms, plenty long in a shock tube.



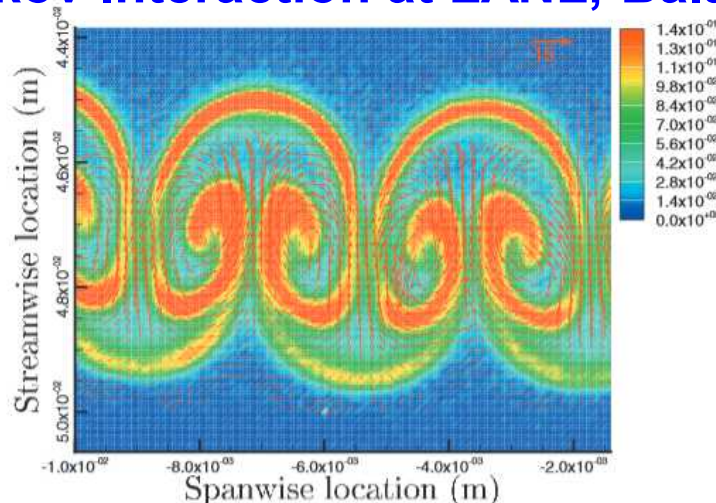


# Previous PIV in Shock Tubes and Shock Tunnels

## Cylinder Flow at ISL, Havermann et al. (2008)



## Richtmeyer-Meshkov Interaction at LANL, Balakumar et al. (2008)



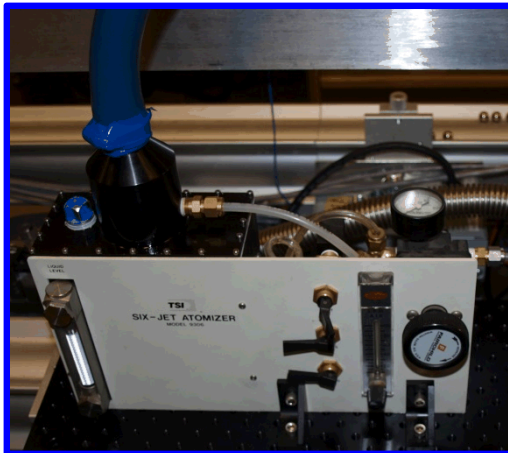
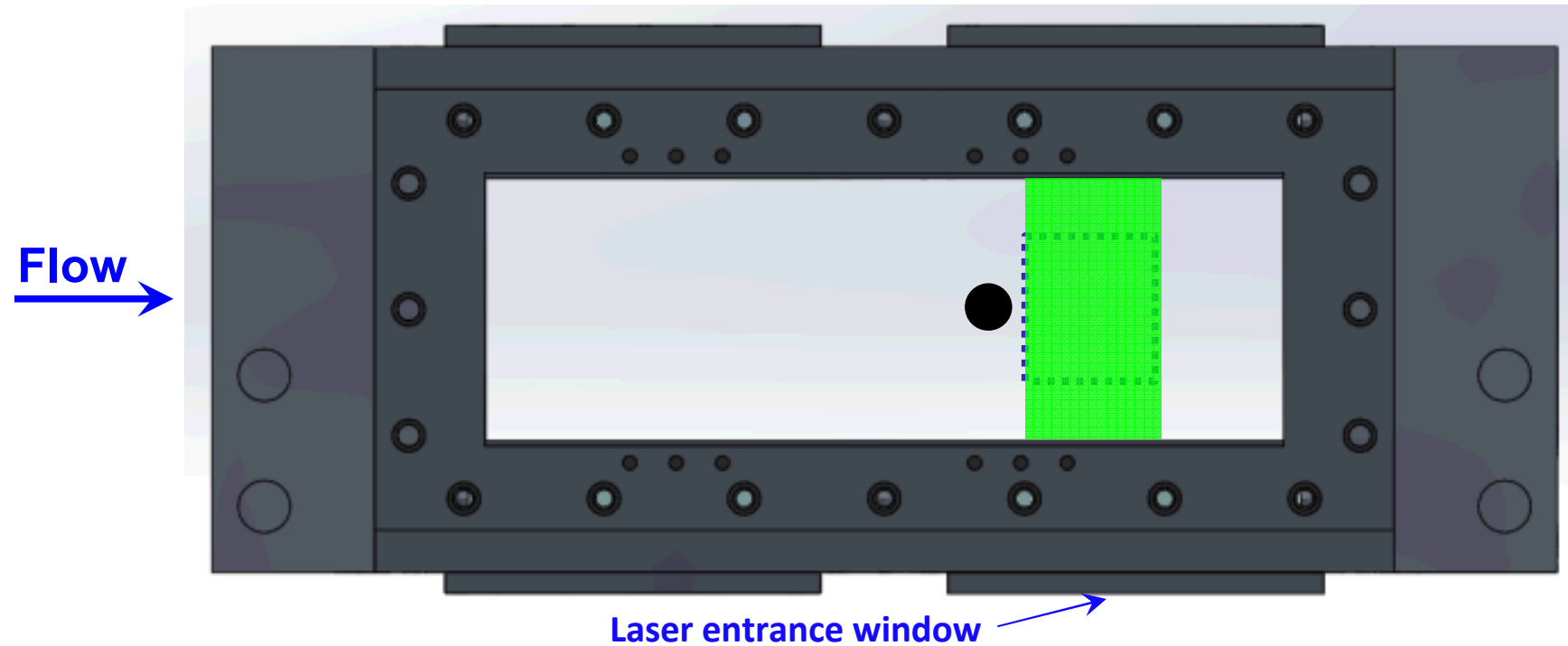
Insightful results, but conventional PIV gives only one realization in the  $\sim 1$  ms test time of a shock tube.

PIV pulse bursts last up to 10.2 ms, plenty long in a shock tube.

Also: Bonazzo et al at Wisconsin, Jacobs et al at Arizona, Vorobieff et al at New Mexico

# Cylinder Wake Flow

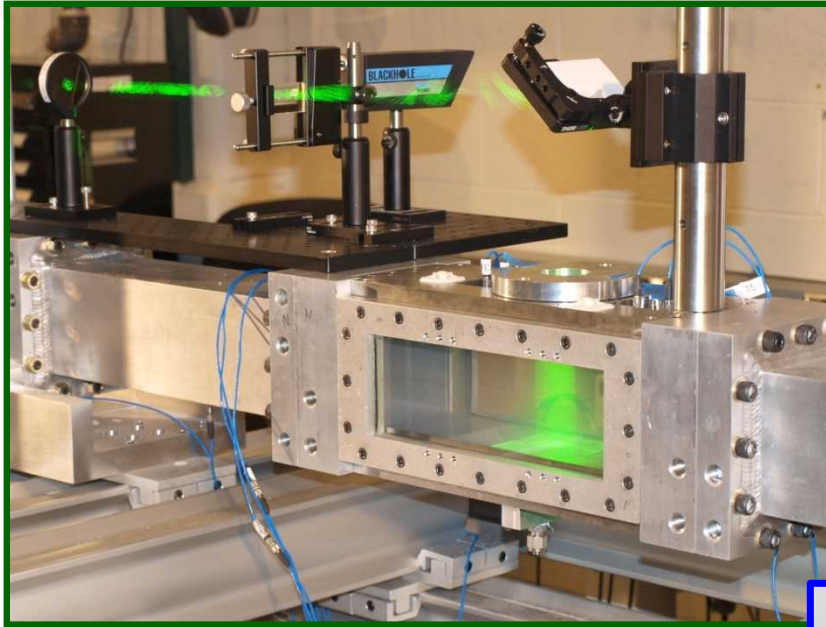
*Transient wake growth of a cylinder after an impulsively started flow.*



## Particle seeding:

- TSI Six-Jet Atomizer
- Particles mixed into driver section prior to run.
- Particle size:  $d_p = 1.6 \mu\text{m}$
- Stokes Number: 0.05 – 0.50

# Imaging Details



## Pulse-Burst Laser Settings:

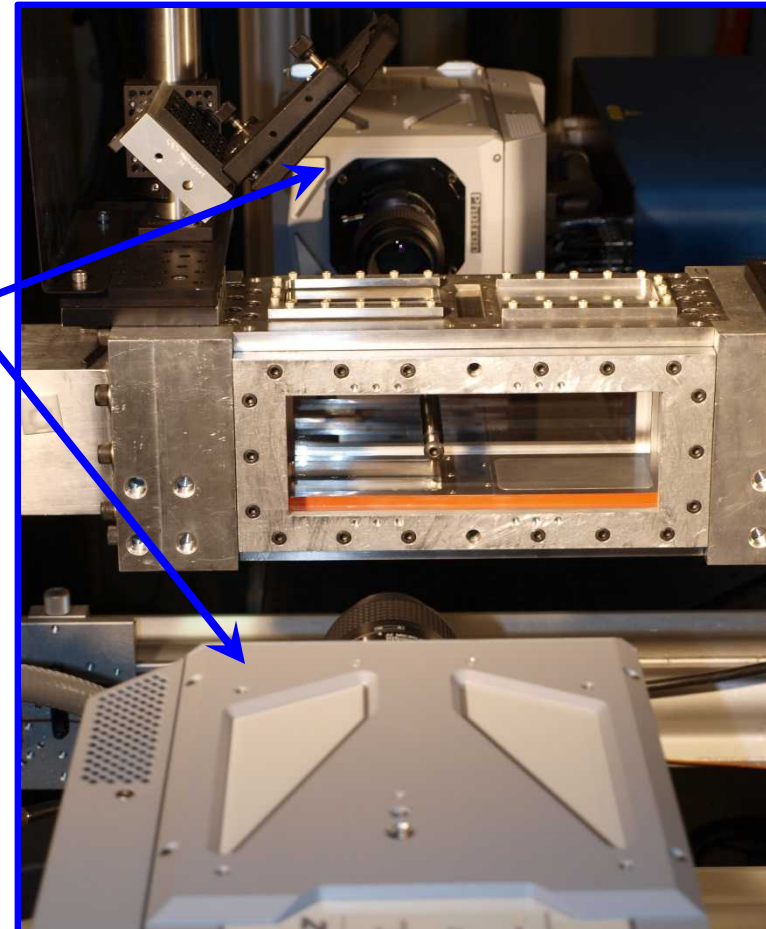
- Pulse pairs at 50 kHz
- 20 mJ per pulse
- Burst duration = 10.2 ms

cameras

Two *Photron SA-Z's* placed adjacent to extend field of view

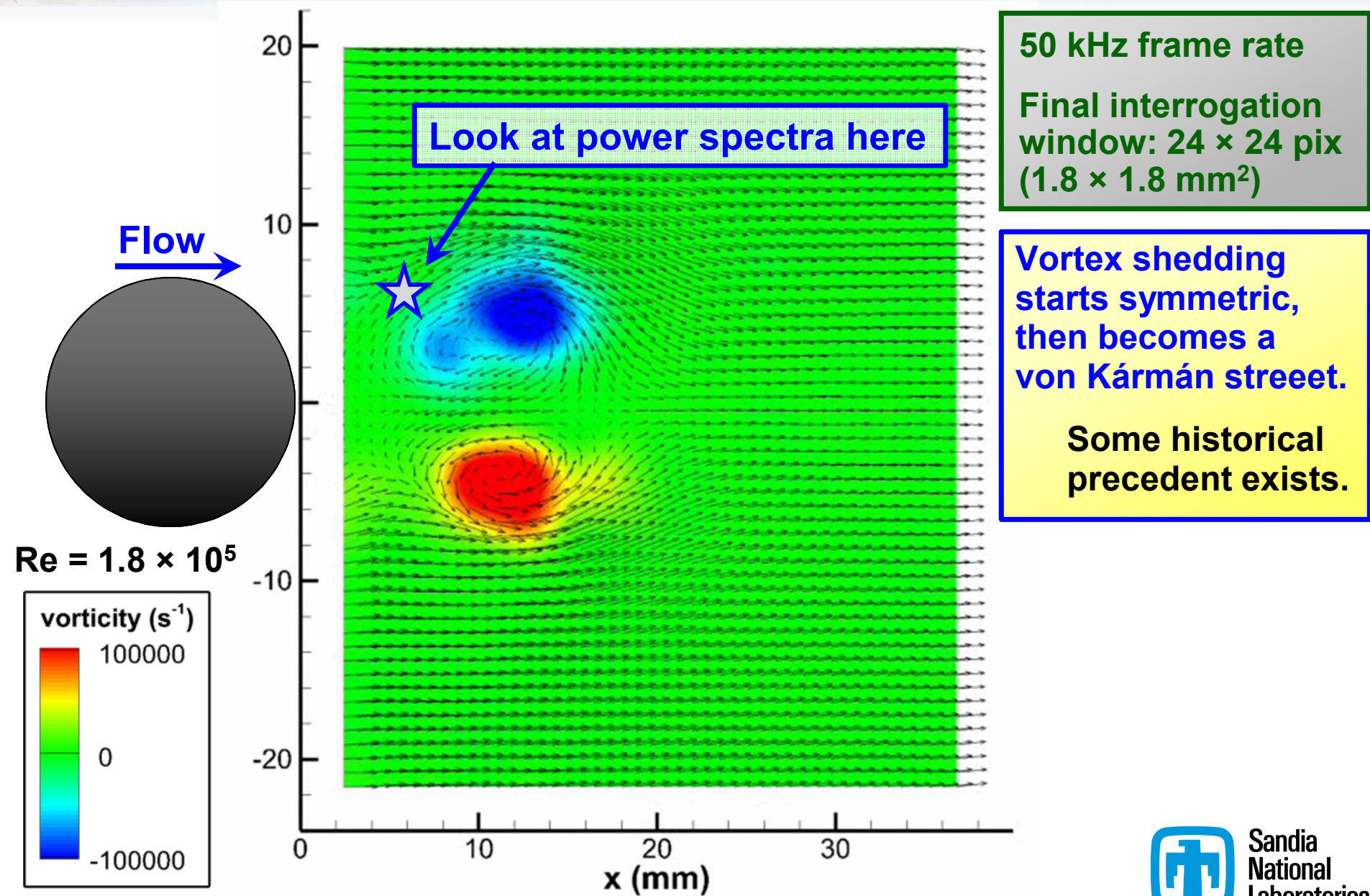
- Each 680 × 340 pixels
- Two-component vectors

Laser and cameras triggered off shock passage at upstream location.



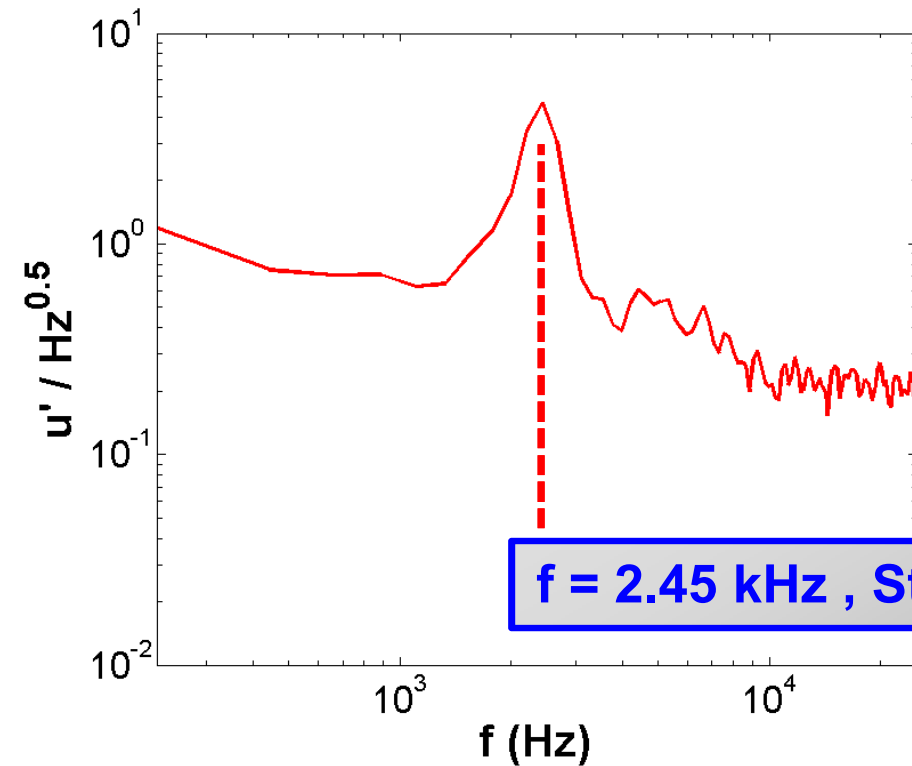


# Transient Wake Vorticity ( $M_s = 1.32$ , $M_2 = 0.43$ )

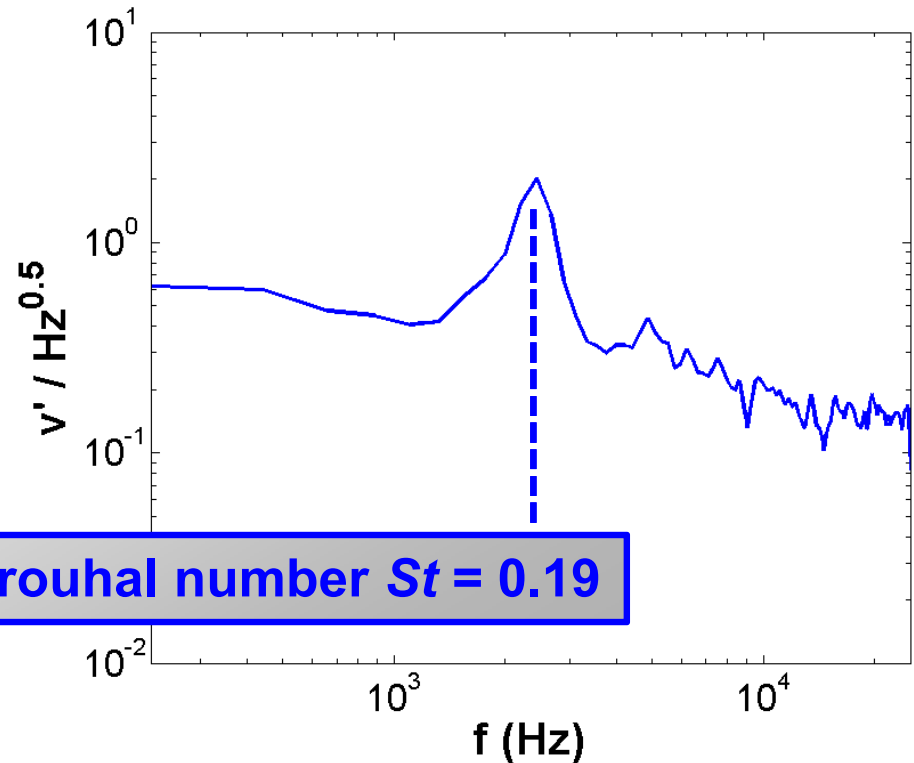


# Power Spectra of Vortex Shedding

## Streamwise Velocity



## Wall-Normal Velocity

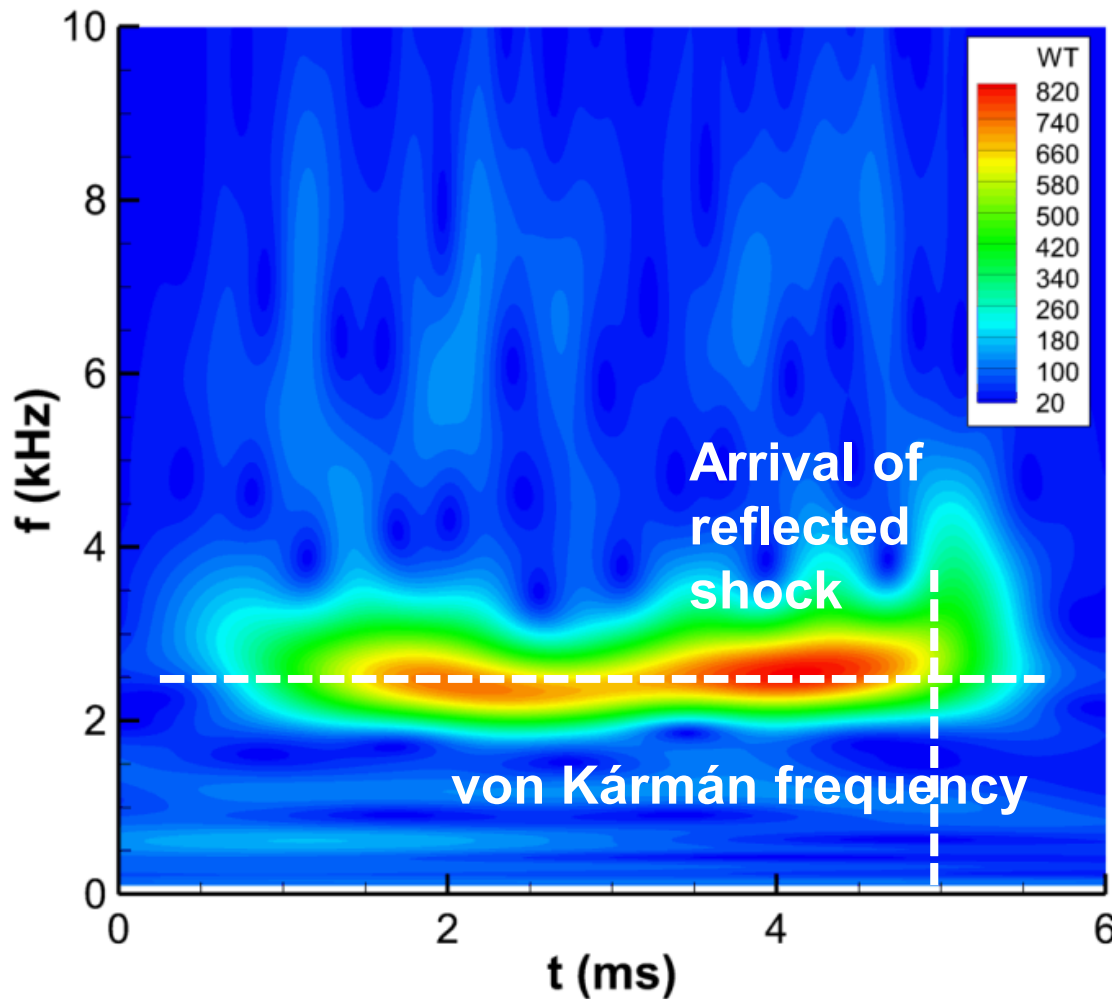


Similar  $St$  to previous studies at this  $Re$  (Roshko, 1961).

This is a time average. How does the frequency of vortex shedding change in time?

# Joint Time Frequency Analysis (JTFA)

## Wavelet Transform of Wall-Normal Velocity



- It takes  $\approx 0.5$  ms for vortex street to become active.
- Street reaches local max at  $\approx 2$  ms, remains near maximum until  $\approx 5$  ms
- After the reflected shock at 5 ms, it takes  $\approx 0.5$  ms for shedding to dampen.

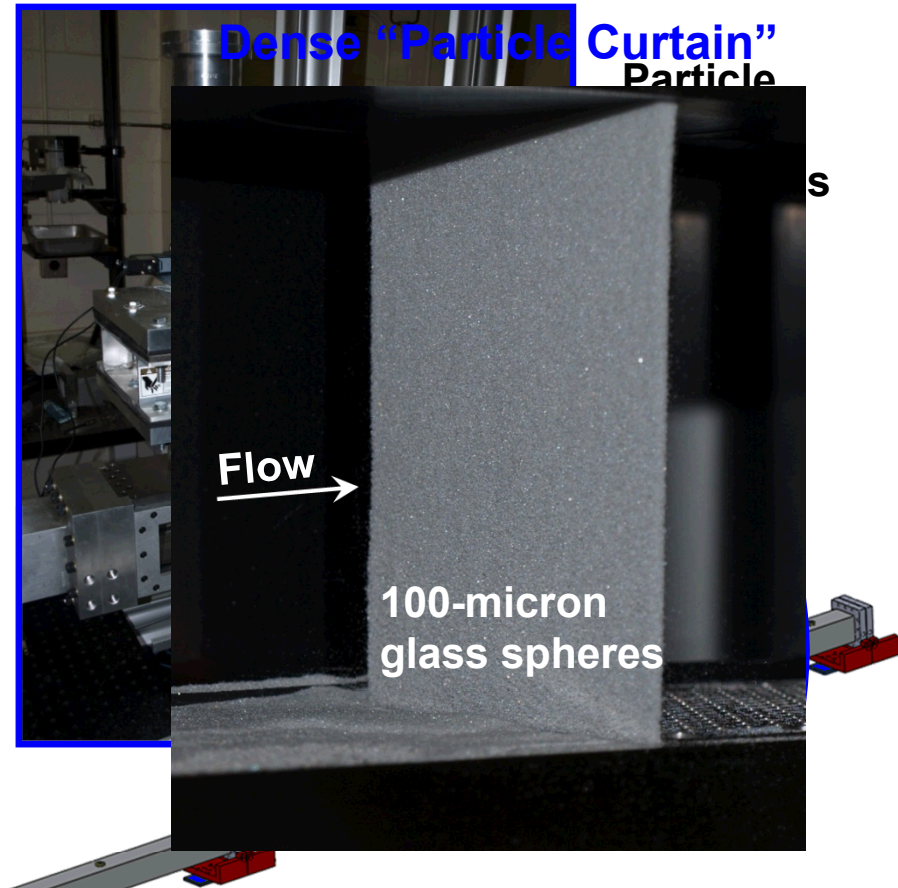
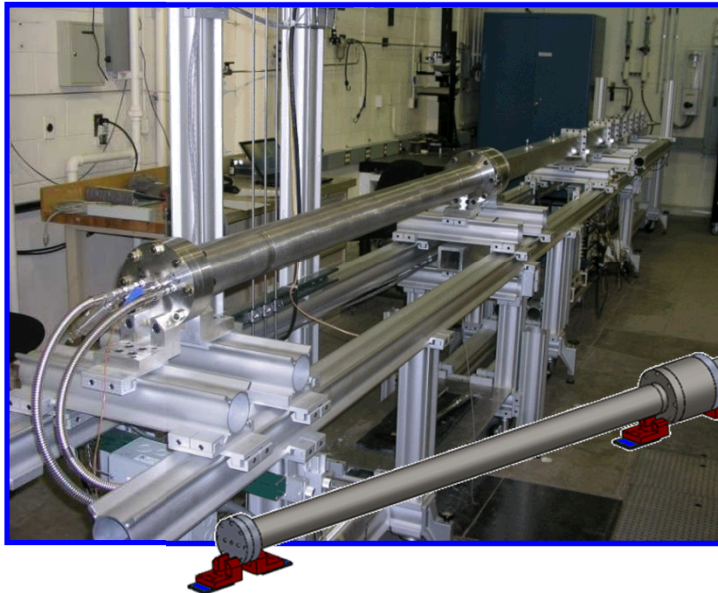
*Pulse-burst PIV quantifies the transient nature of vortex shedding in a shock tube.*



# Particle Curtain Experiments in MST

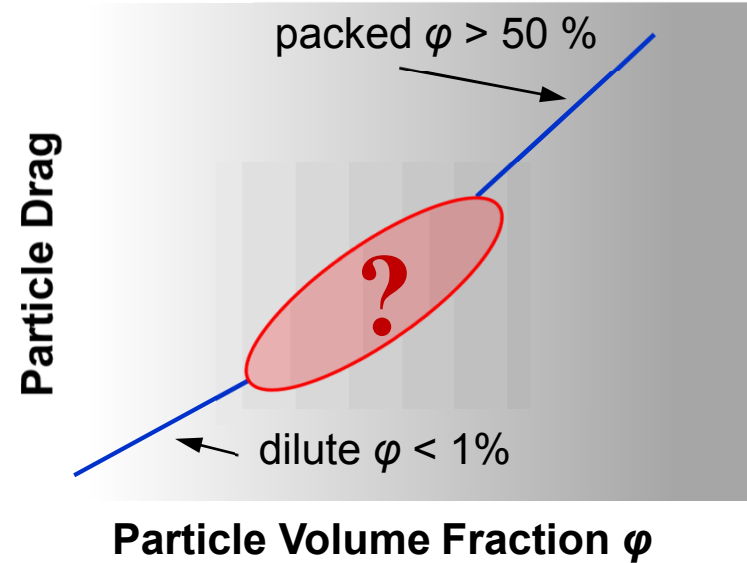
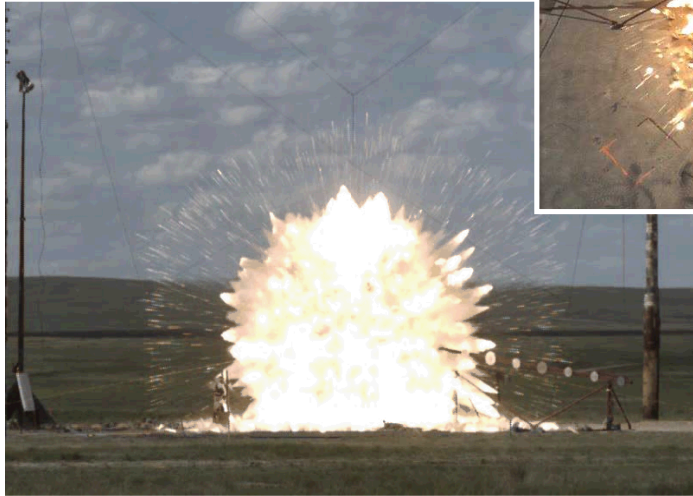
## Multiphase shock tube facility

MST built to study shock / particle interactions in dense gas-solid flows.



Particle volume fraction  $\approx 20\%$

# Why do we care about a particle curtain?



## Explosive Particle Dispersal

Dynamics of densely packed particles influence explosive processes.

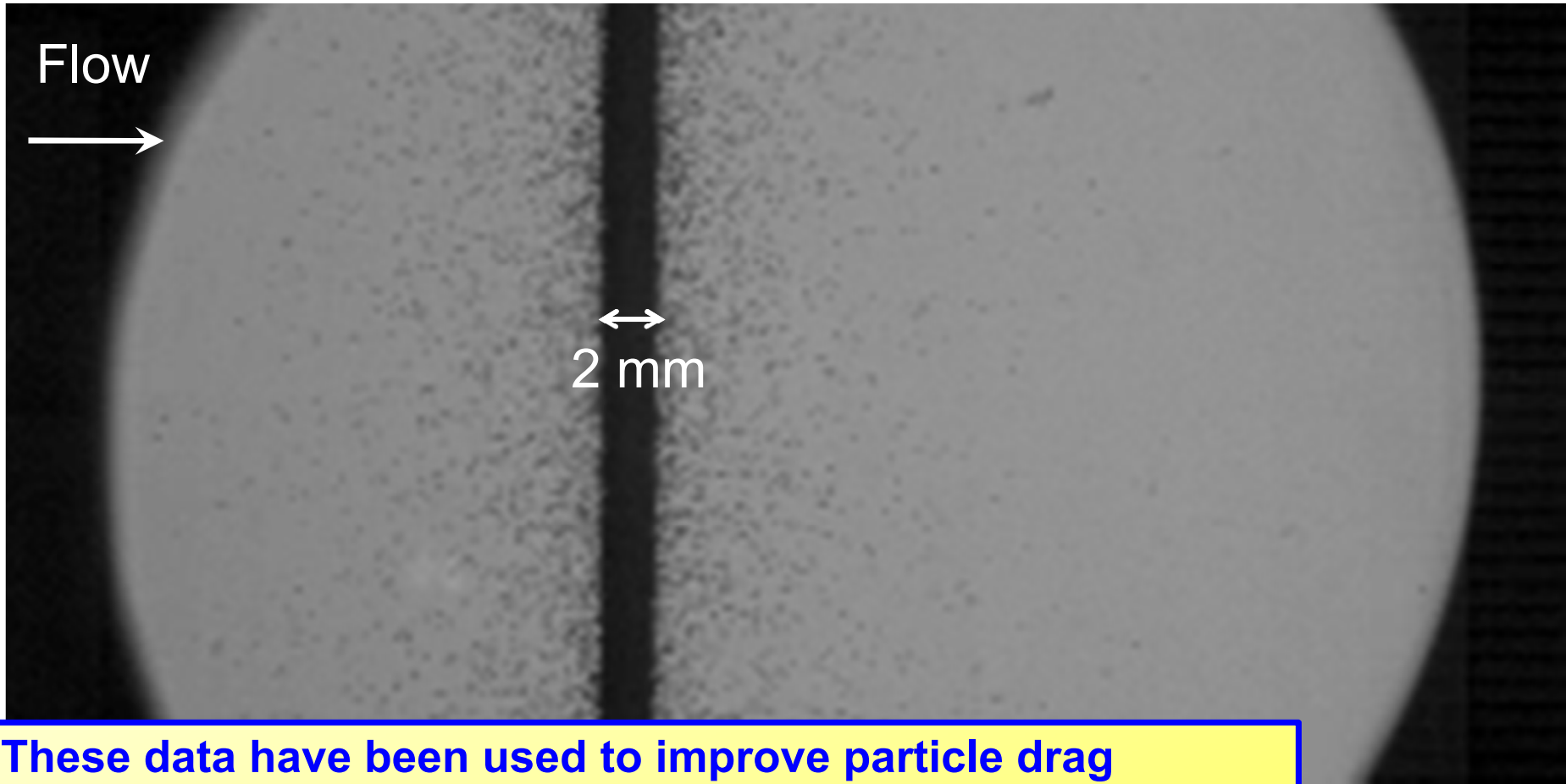
## Particle Dynamics

- Dynamics governed by volume fraction  $\phi$
- Very little data in “dense” regime ( $1\% < \phi < 50\%$ )



# High-Speed Schlieren (130 kHz)

## Interaction at shock Mach number = 1.67



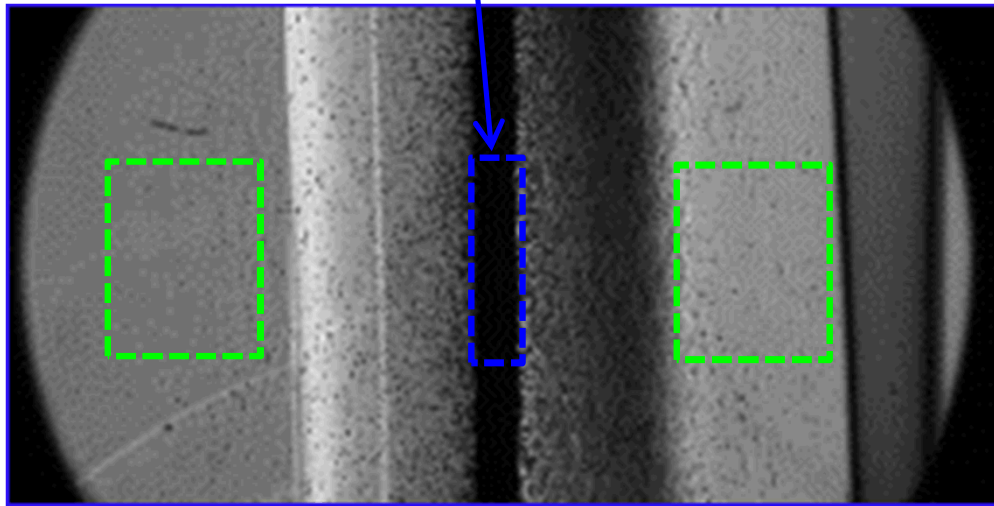
These data have been used to improve particle drag models for prediction of explosive processes.

But we need *gas-phase velocities* to accurately provide drag coefficients.



# Pulse-Burst PIV can probe much deeper physics.

We've previously focused on the solid particles.

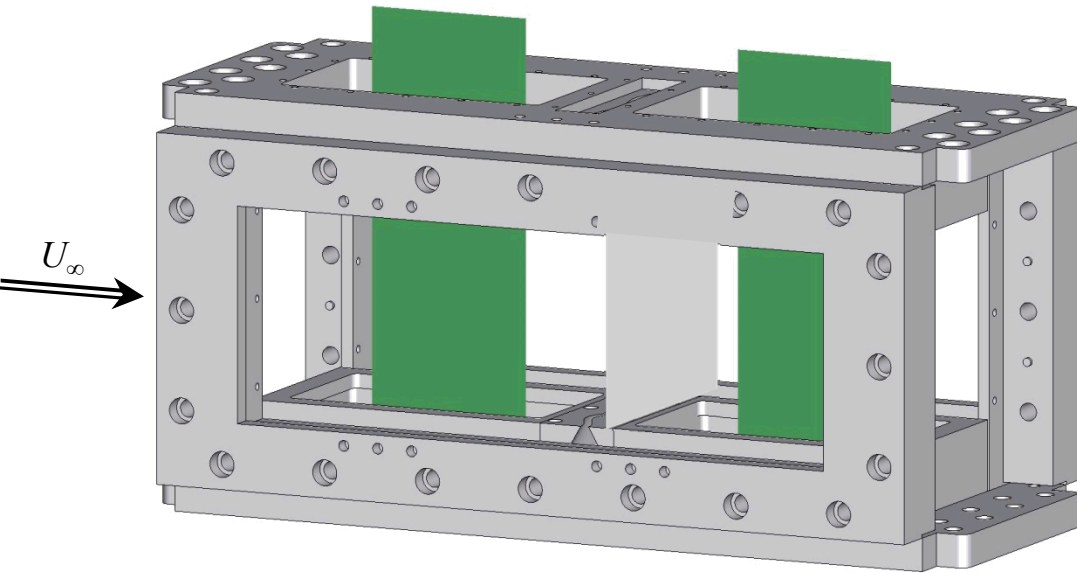


A conventional PIV system allows only one realization in the millisecond test times of a shock tube.

**Time-resolved gas phase data can measure:**

- Interaction Unsteadiness
- Interphase Momentum Transfer
- Particle-Induced Turbulence

# Particle Curtain Pulse-Burst PIV



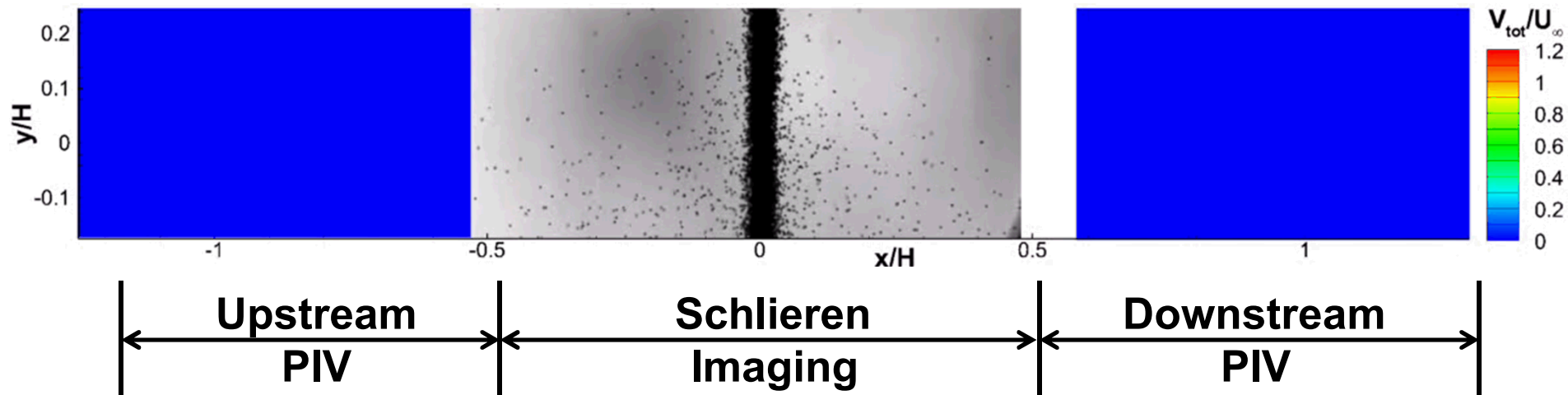
**Split the laser into upstream and downstream sheets.**

**Image each using synchronized cameras.**

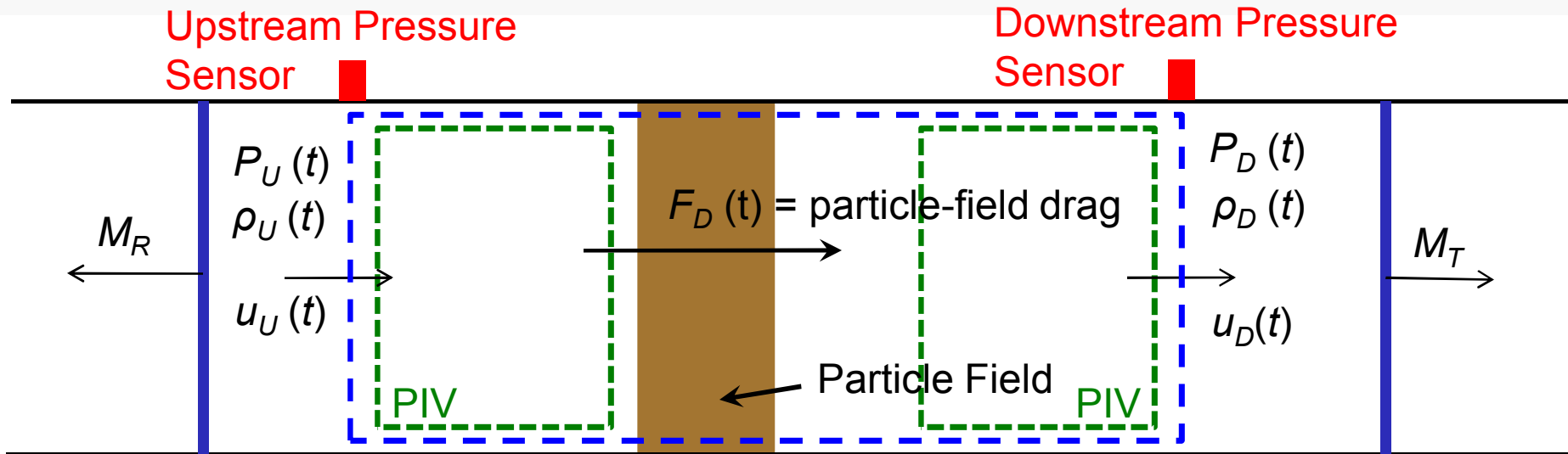
**Data capture gas jetting through the curtain and angled shock waves.**

**Control volume analysis to determine the particle drag.**

**Mach 1.4,  $U=200$  m/s**



# Calculating the Particle Drag

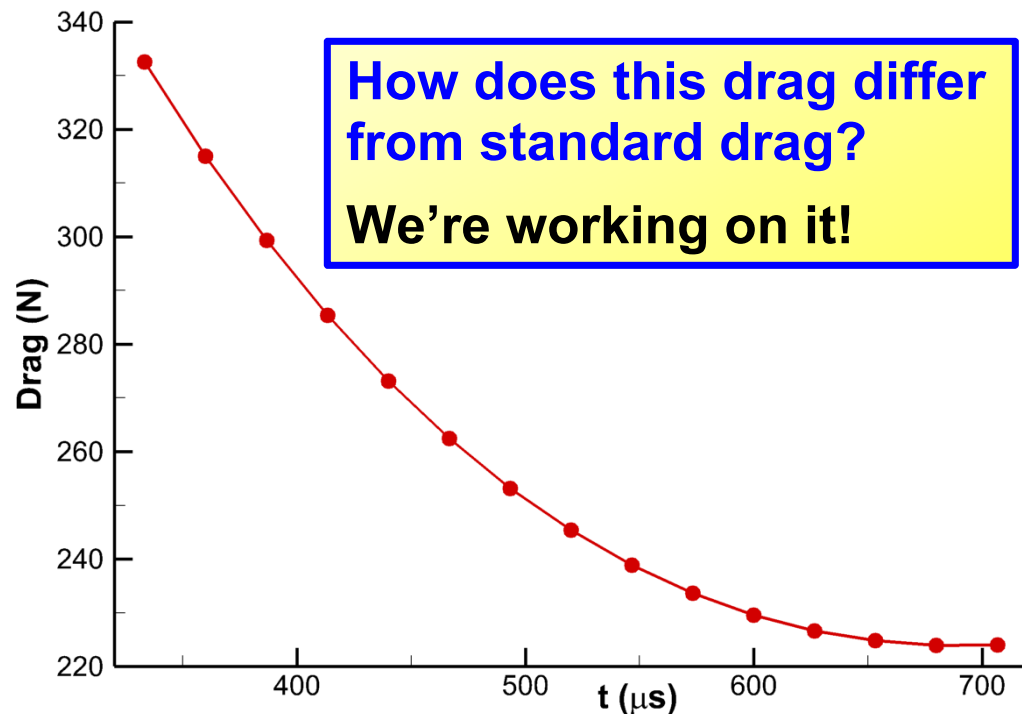


## Control volume balance of momentum conservation

Get pressure  $P_U(t)$  and  $P_D(t)$  from pressure sensors.

Get density  $\rho(x,y)$  from shock equations.

Get velocity  $u(x,y,t)$  from pulse-burst PIV.

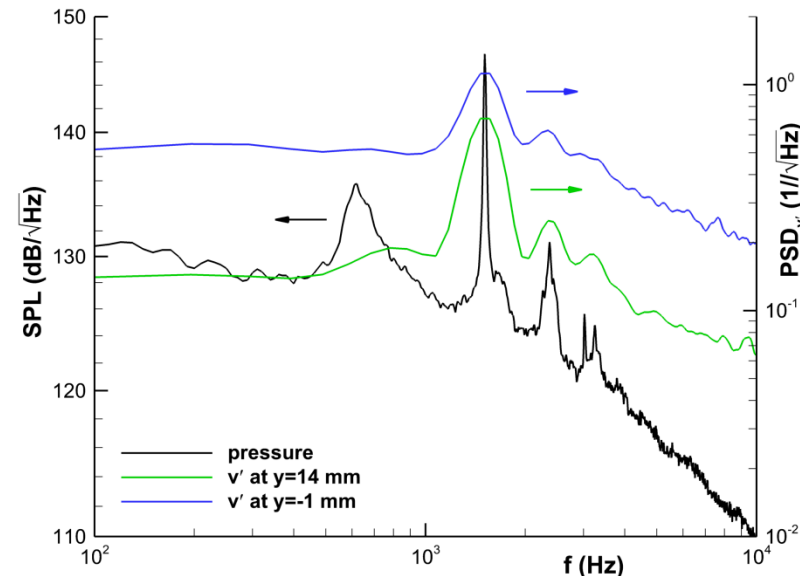
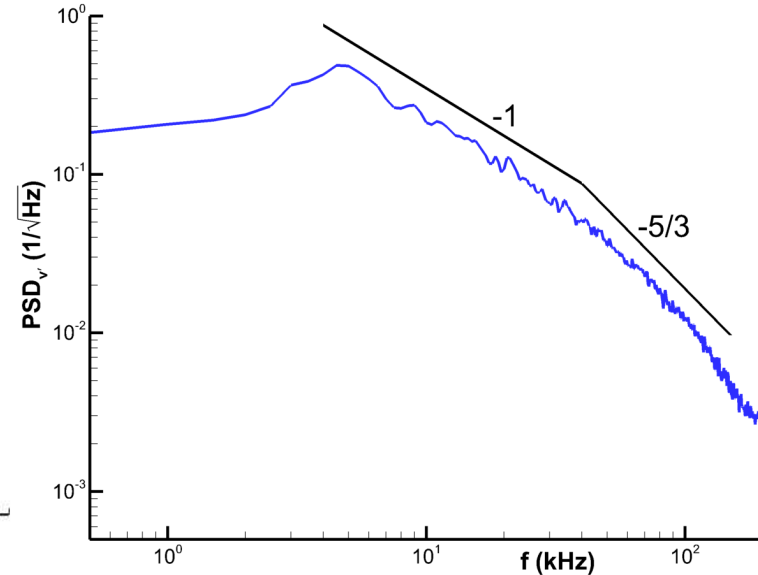
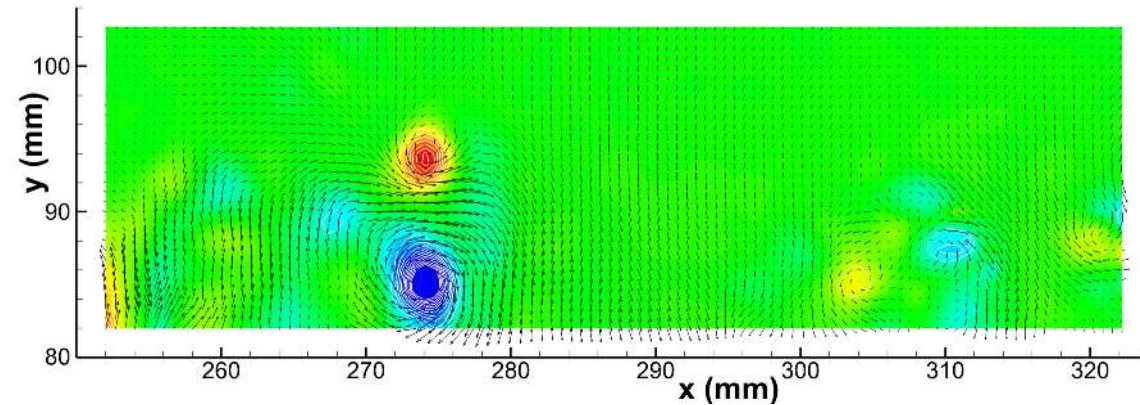




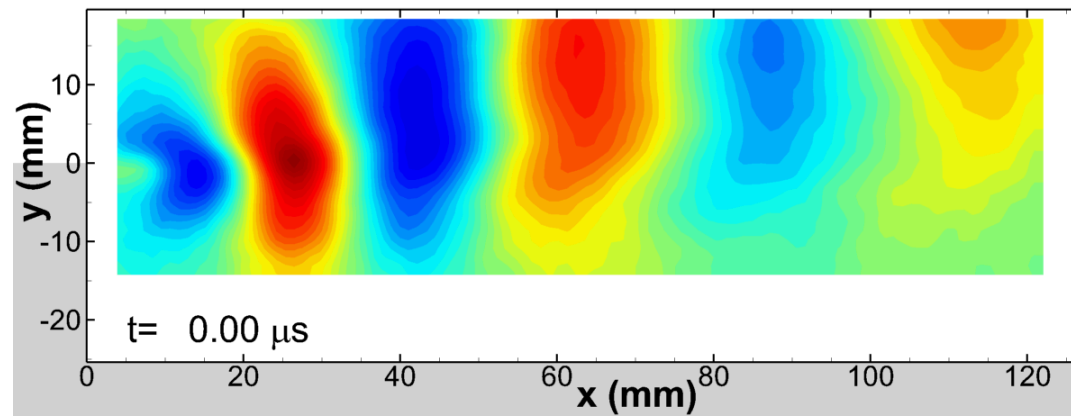
# Pulse-burst lasers make TR-PIV feasible for high-speed flows.

*This is the first application of Pulse-Burst PIV to a ground test facility.*

Supersampling of jet-in-crossflow data has revealed turbulent scaling laws.



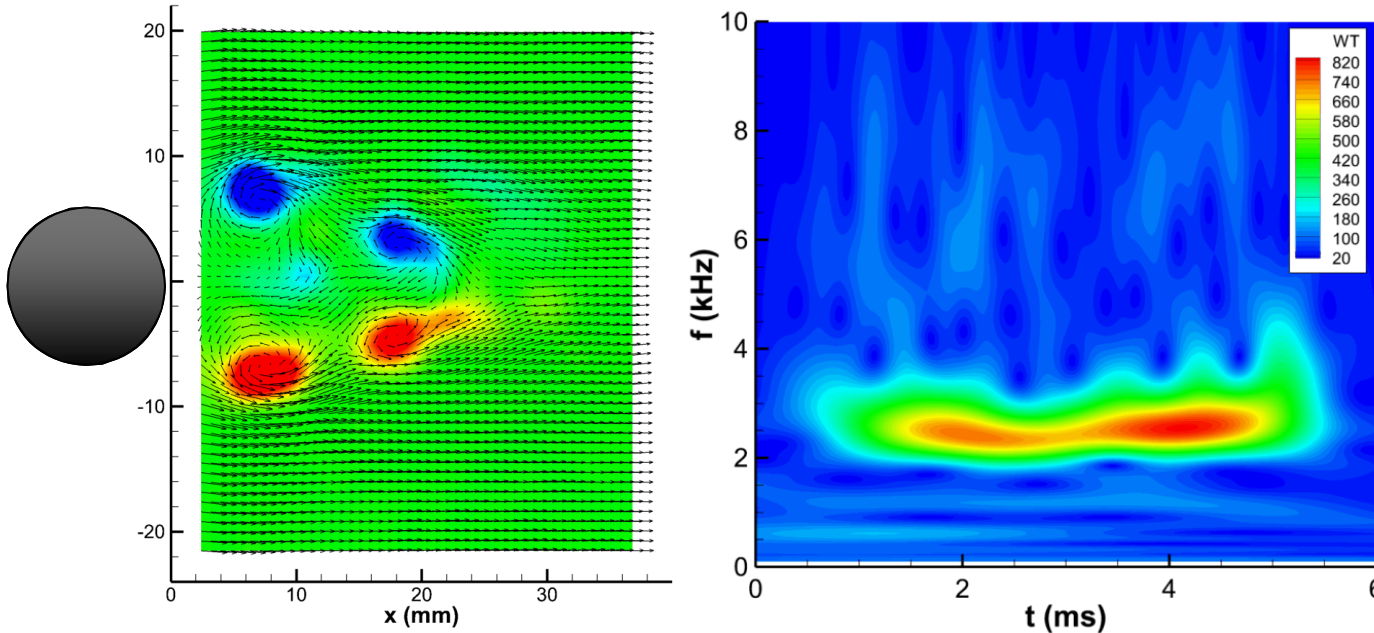
Cavity data show flow structures that create aeroacoustic resonances.



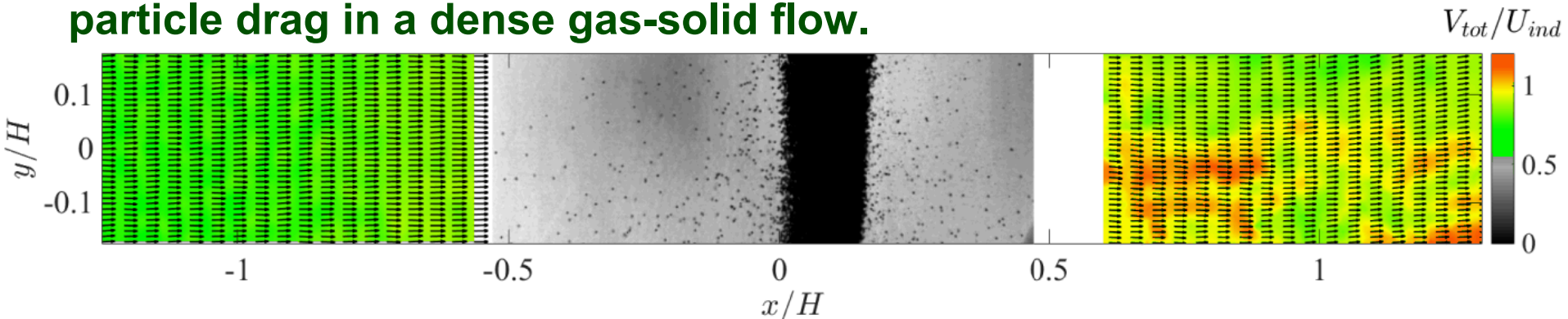
# Pulse-burst lasers make TR-PIV feasible for high-speed flows.

*This is the first application of Pulse-Burst PIV to a ground test facility.*

Shock tube data reveal the transient start of cylinder vortex shedding.



Shock-particle interaction data can be used to determine particle drag in a dense gas-solid flow.



*More physics to be revealed as we continue to analyze these data sets!*