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Fatigue Life of Austenitic Stainless Steel for Hydrogen Service

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Relevance

Challenge:

- Balance of plant (BOP) onboard vehicles accounts for:
 - *30-57% of total system cost*
 - *15-20% of total system mass*
- Structural metals for hydrogen storage typically include
 - *Annealed type 316L austenitic stainless steel (Ni content >12 wt%)*
 - *A286 precipitation-strengthened austenitic stainless steel (Ni ~30 wt%)*
- Relatively little data in literature to identify component relevant metrics for performance

Project Outline

Objectives:

- *Identify alternatives to high-cost metals for high-pressure BOP components*
- *Use CHMC 1-2014 methodology to efficiently establish relevant selection metrics to compare materials*
- *Identify alternatives to high-cost alloys for high-pressure BOP components*

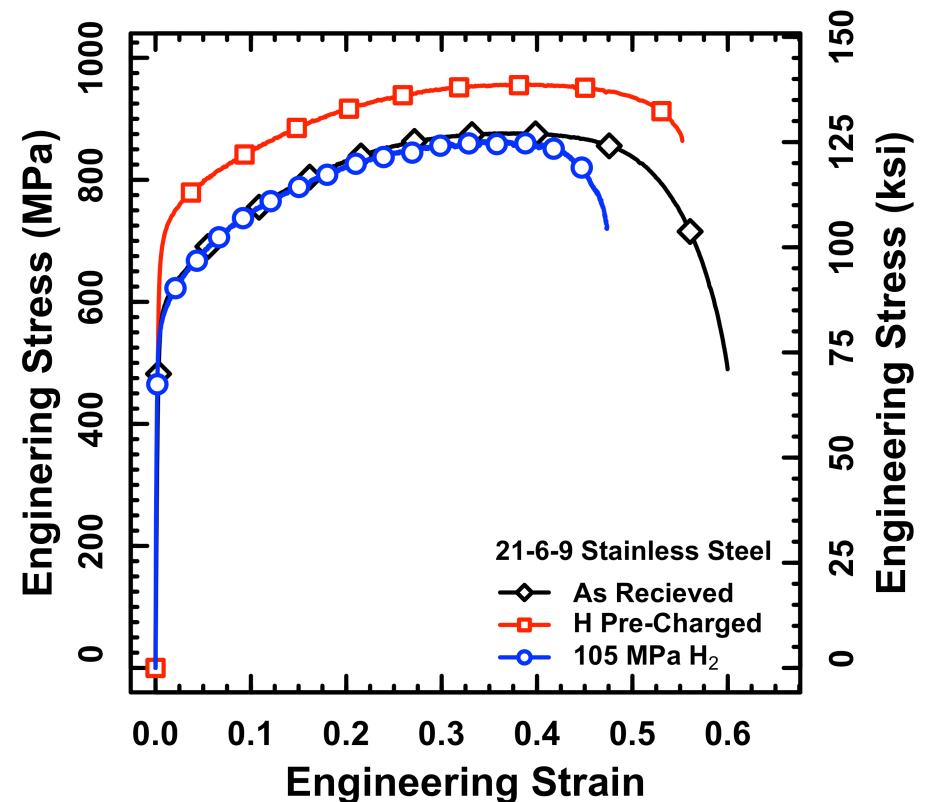
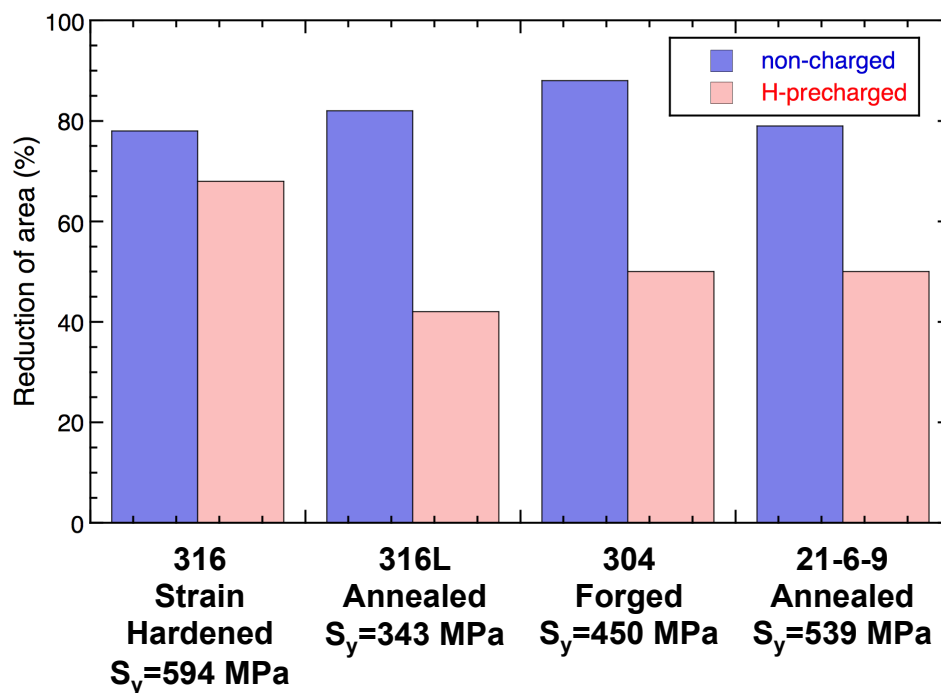
Outstanding questions:

- How effectively can stress-based fatigue testing be used to compare material performance in presence of hydrogen?
- Can thermally pre-charged samples be used in place of testing in external hydrogen?

Most hydrogen compatibility decisions based on tensile data

- Strength is unchanged by hydrogen
- Ductility is decreased by hydrogen

How well do metrics match component service?



C. San Marchi, et al., Int. Hydrogen Conf. 2012, Jackson, WY

Experimental Methods: Stainless Steels of Interest

Composition

Alloy ID	Cr	Ni	Mn	Mo	C	N	Si	<i>H₂ Charged</i>
316L	17.54	12.04	1.15	2.05	0.020	0.04	0.51	150 wppm
21Cr-6Ni-9Mn	20.45	6.15	9.55	NR	0.033	0.265	0.52	~210 wppm

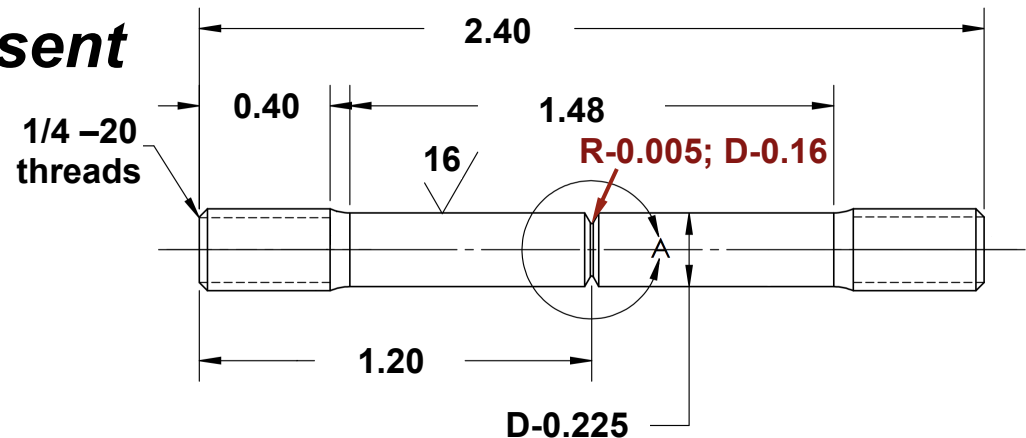
Tensile properties

Alloy ID	S _Y (MPa)	S _U (MPa)	S _Y (ksi)	S _U (ksi)	RA (pct.)	El _T (pct.)
316L - Cold Worked	589	967	85.5	140.5	74	44
21Cr-6Ni-9Mn - Annealed	539	881	78.2	127.8	79	61
21Cr-6Ni-9Mn - H Charged	669	977	104	142	-	55

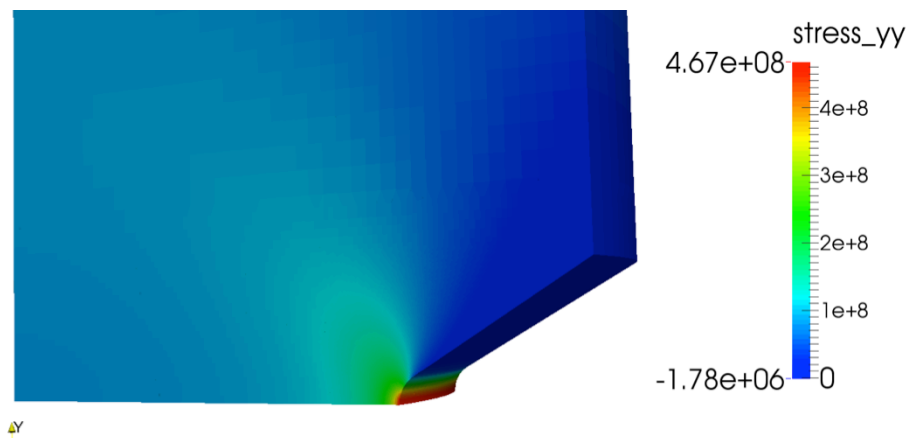
Experimental Methods: Fatigue Conditions

- ***Testing conditions represent pressure loading:***

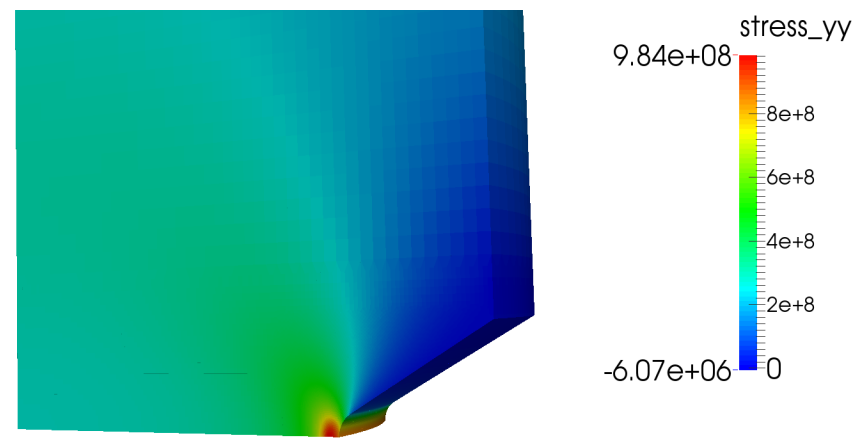
- ***$R = 0.1$ (tension-tension)***
- ***Notched specimens***
- ***Net-section $S_{max} < S_Y$***



$$K_{t,el} = \frac{\sigma_{max}}{\sigma_{min}} = \frac{467}{119.7} = 3.90$$

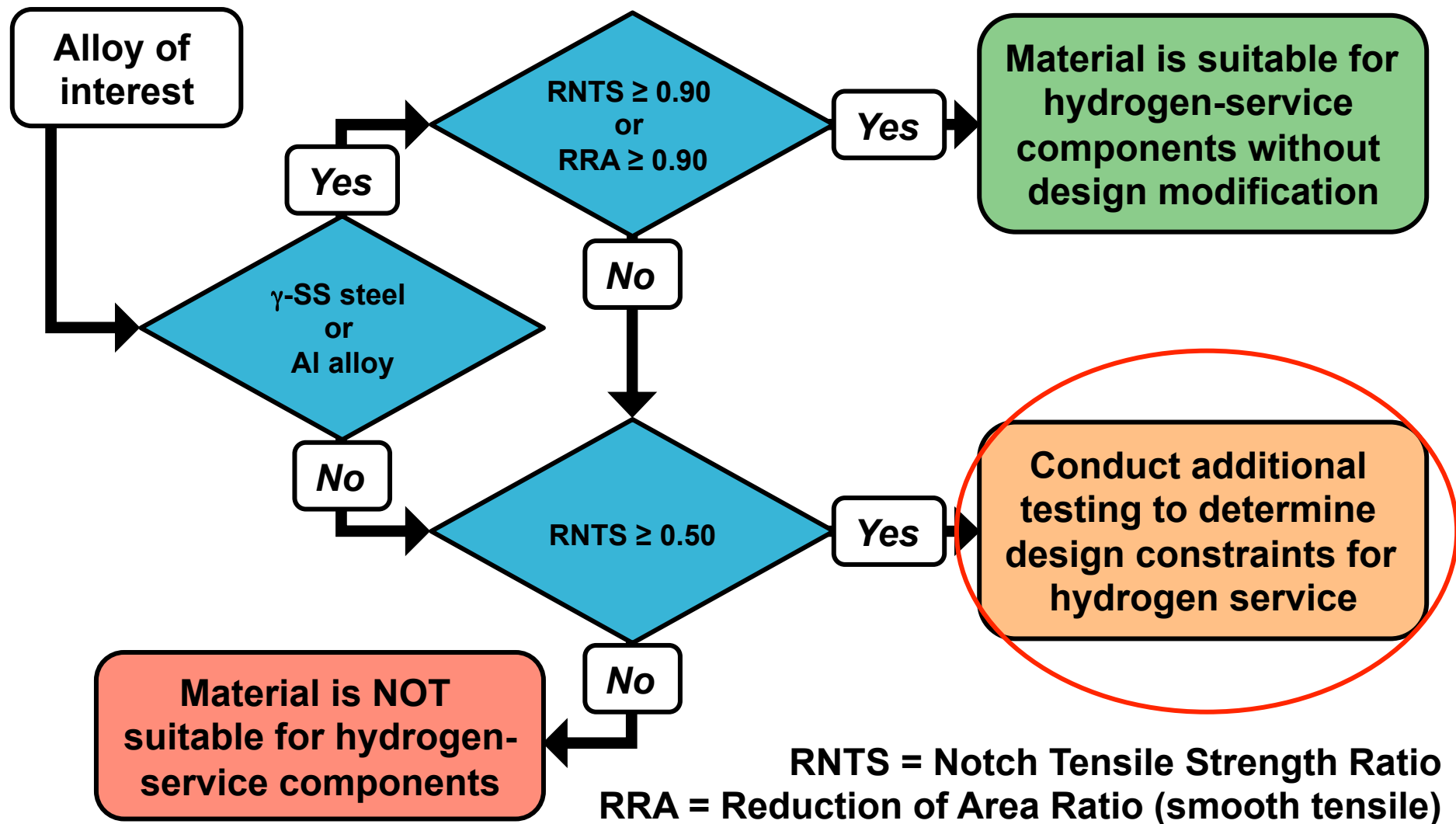


$$K_{t,el-pl} = \frac{\sigma_{max}}{\sigma_{min}} = \frac{984}{406.4} = 2.42$$



CSA CHMC1 – 2014:

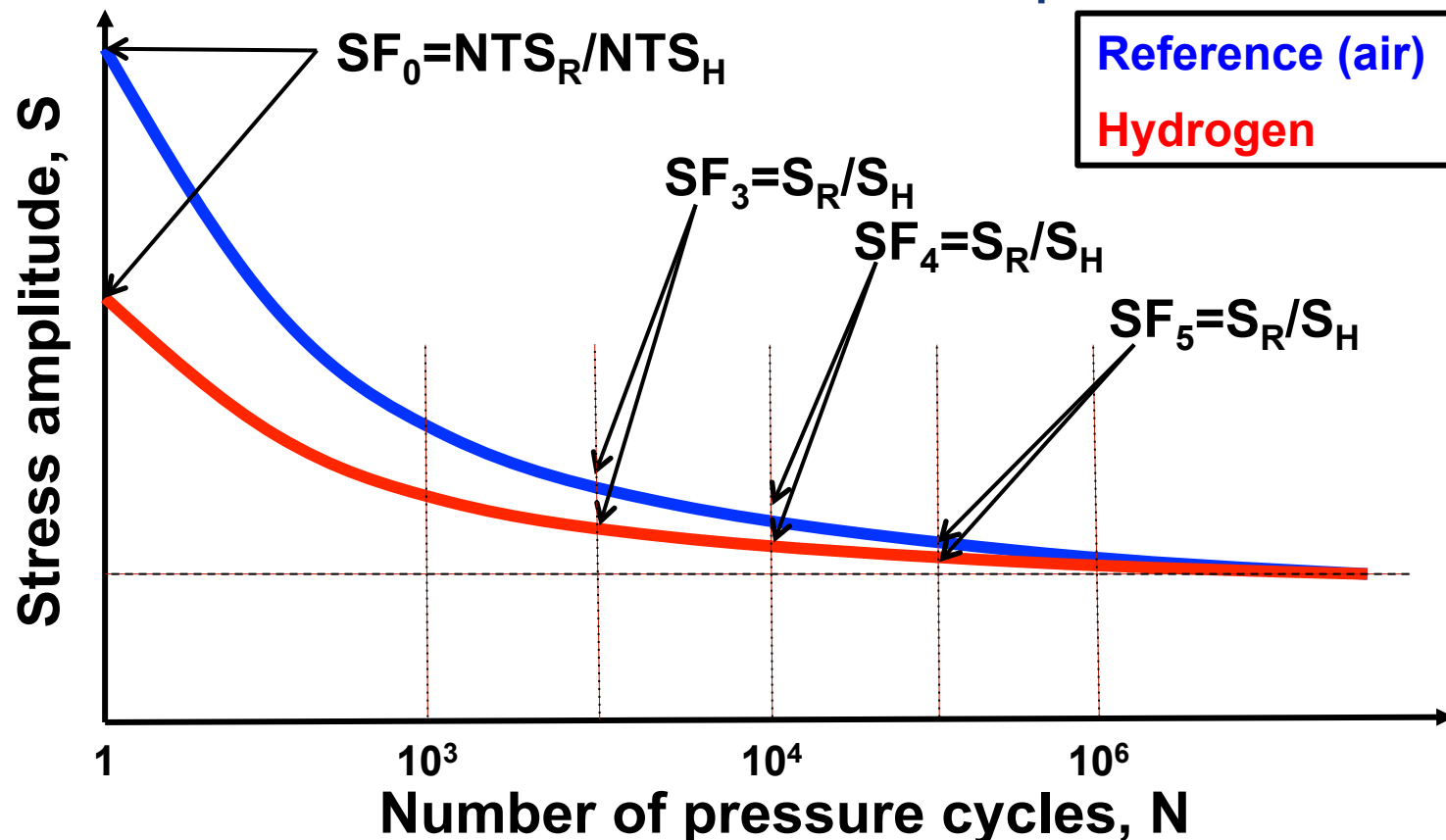
Part 3: Material qualification



Safety Factor Multiplier Method

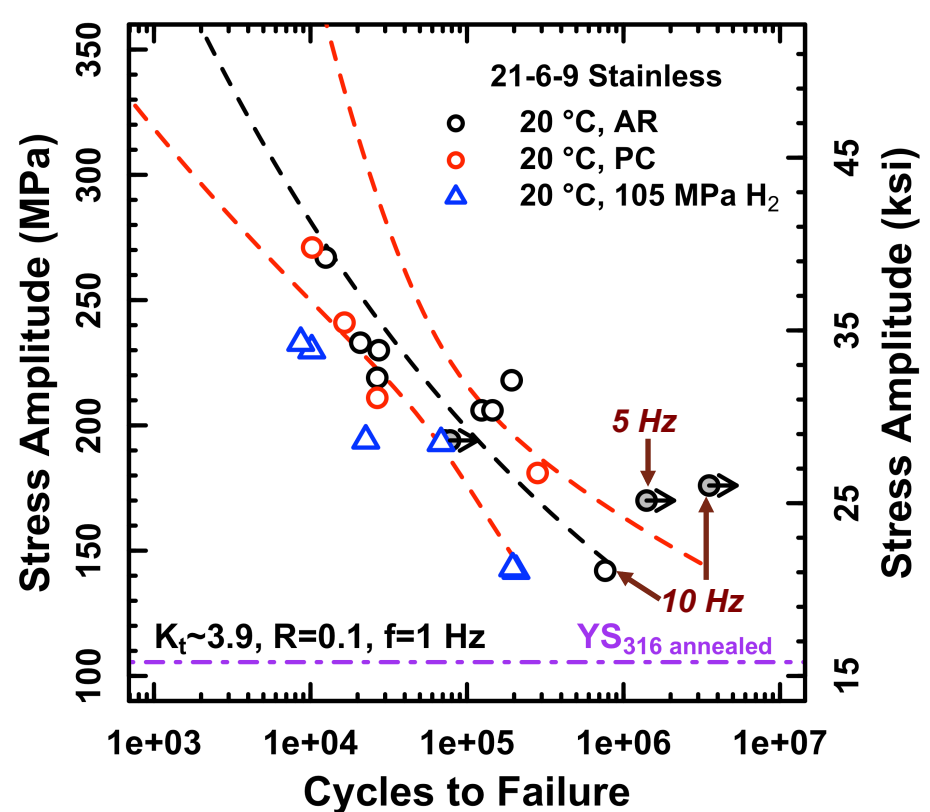
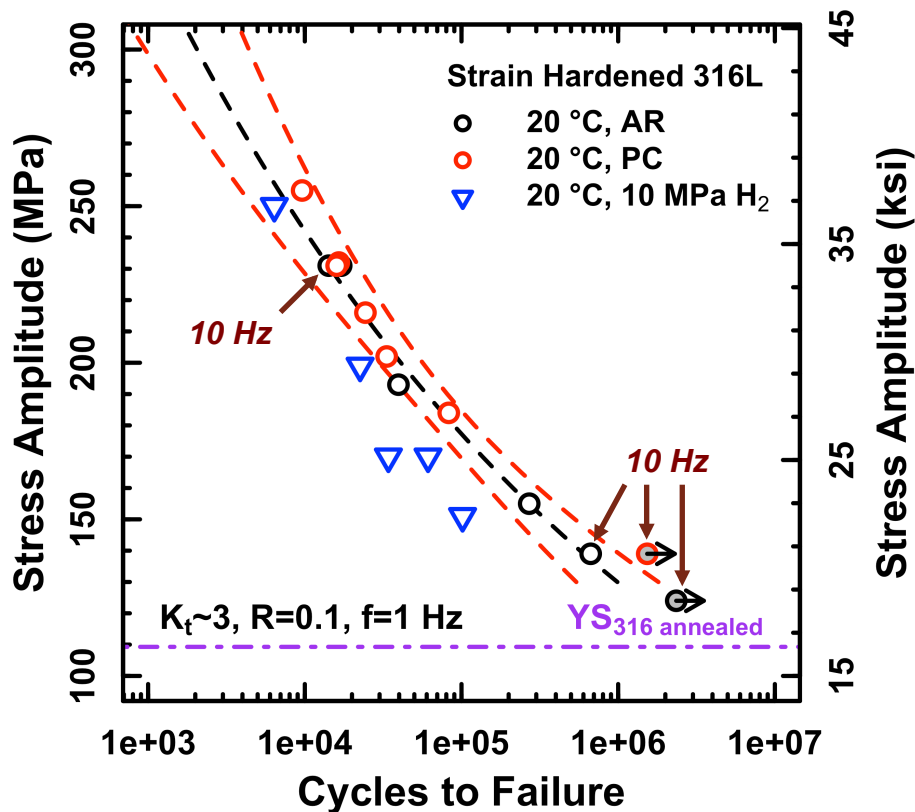
Hydrogen safety factor: $SF_H = \max(SF_0, SF_3, SF_4, SF_5)$

Safety factor for design $\rightarrow SF_{\text{component}} \times SF_H$



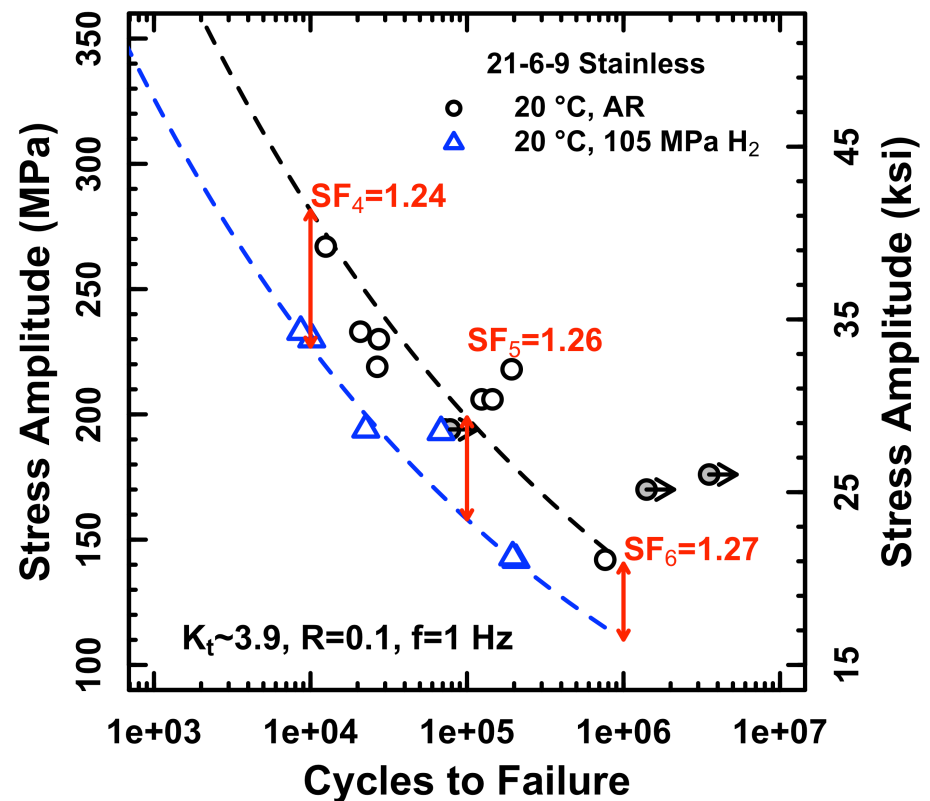
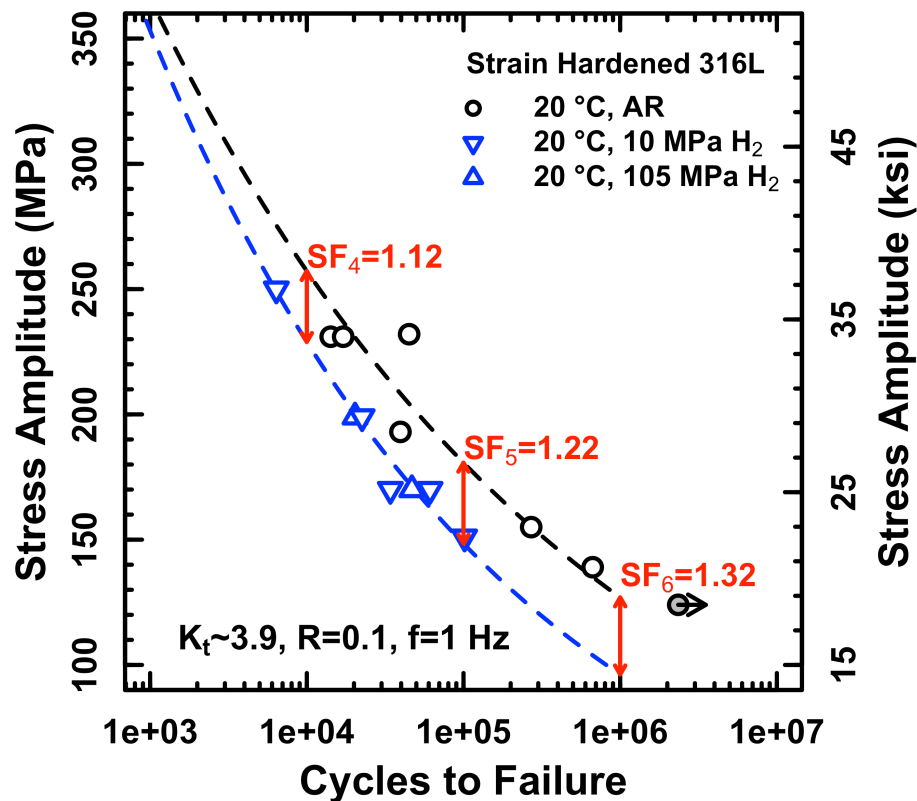
Fatigue Life

- No *apparent* change in performance with internal hydrogen
- Decrease in fatigue life in External hydrogen



Safety Factor Multiplier: External Hydrogen

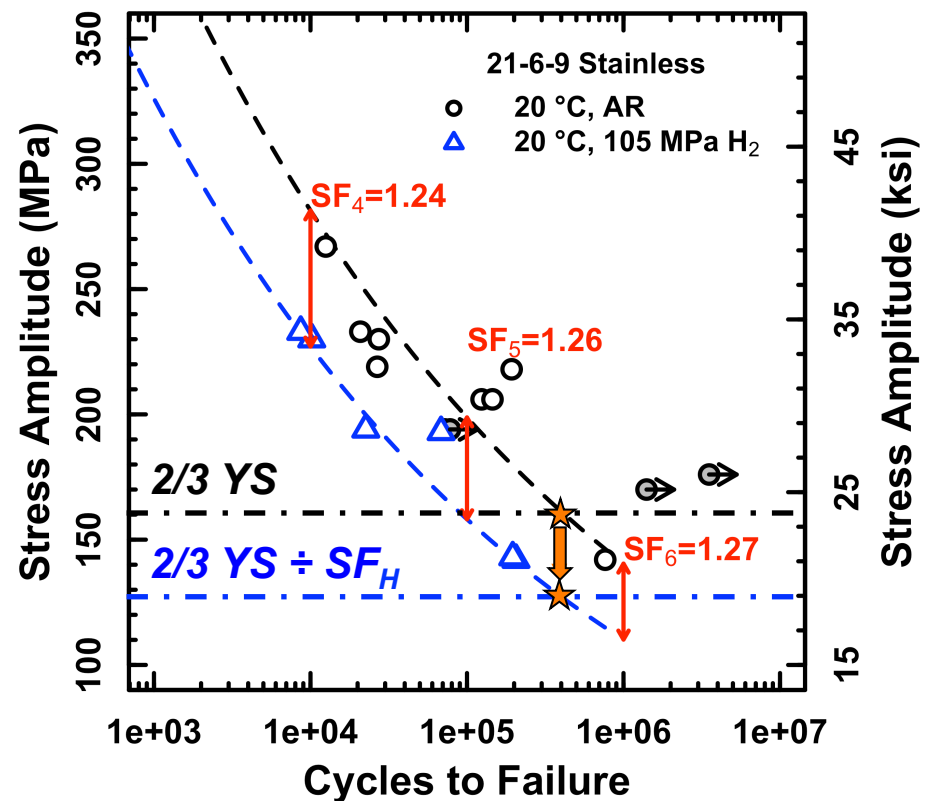
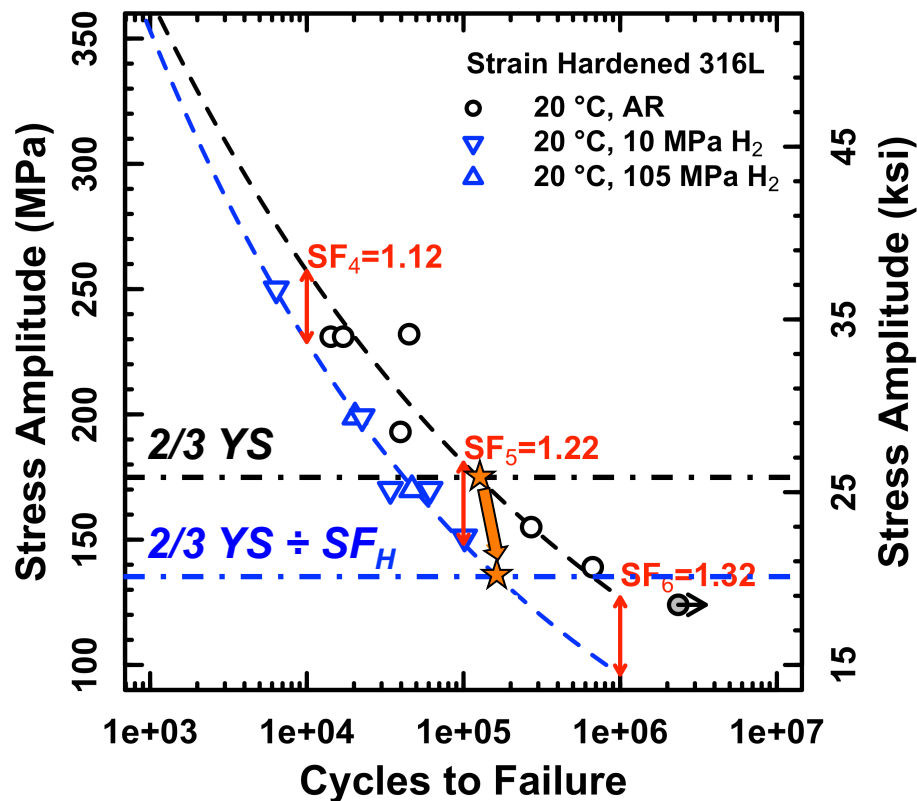
- 316L converges at low cycle regime
- 21-6-9 essentially constant SFM



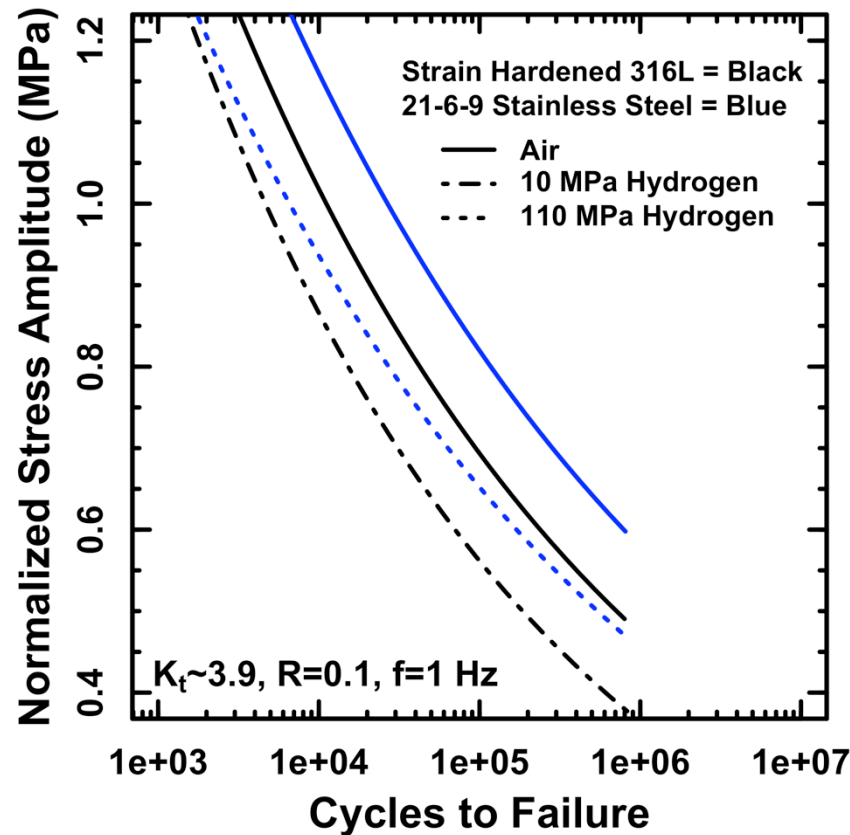
Safety Factor Multiplier: External Hydrogen

- Multiplier on *existing safety factor* results in consistent fatigue life in hydrogen

Anticipated component life >30k cycles



Comparison between alloys



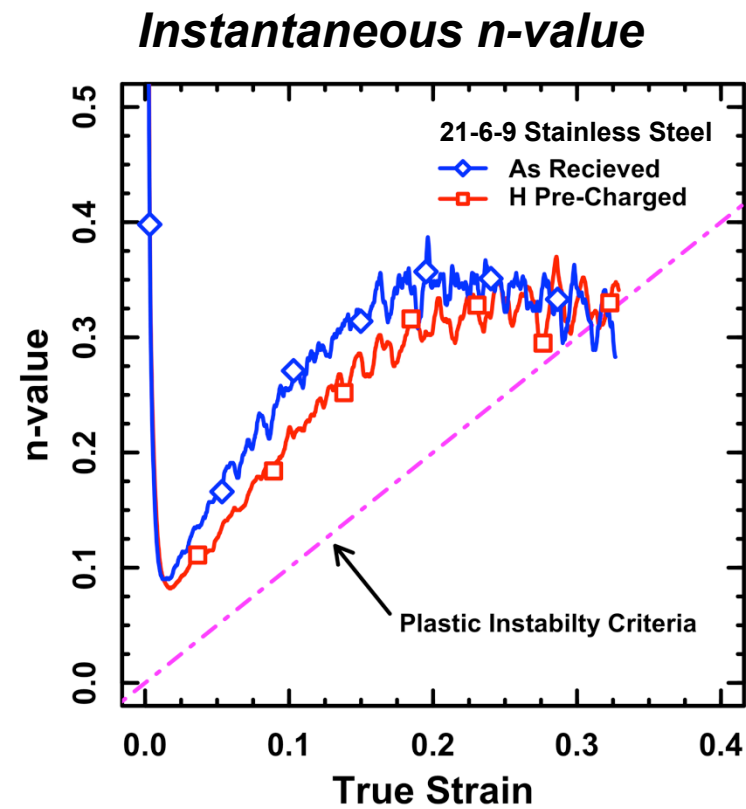
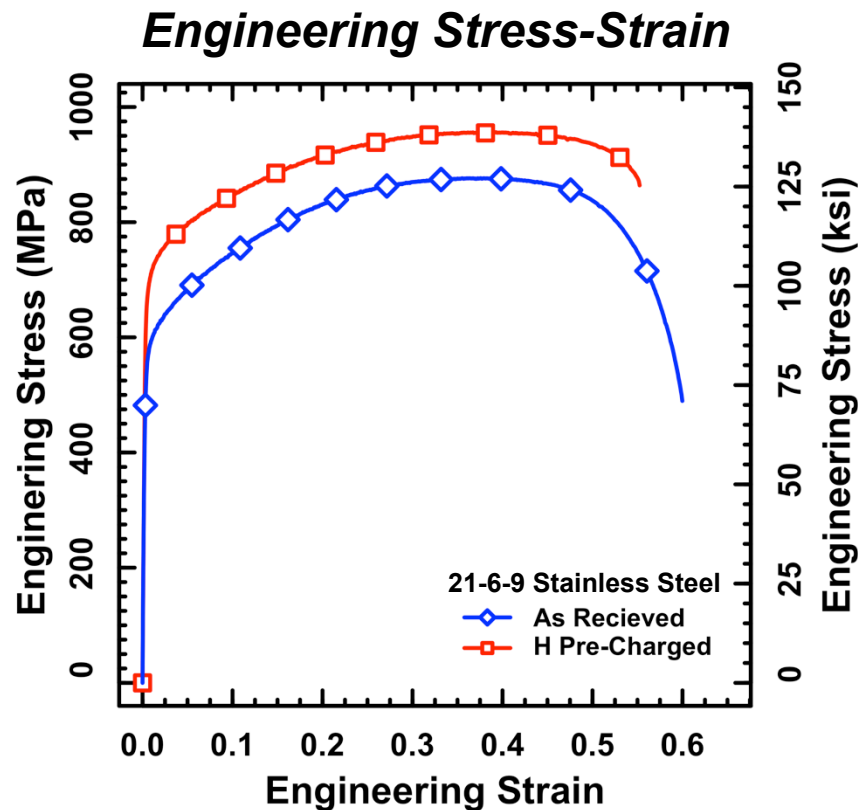
- 21-6-9 demonstrates improved fatigue life compared to hardened 316L
- YS of *annealed 21-6-9 less than YS of strain-hardened 316L*
- *Cost of 21-6-9 bar material is ~80% of type 316L bar*

Stress Amplitude Normalization

$$SA_N = \frac{1}{2} \frac{(S_{Max} - S_{Min})}{S_{Yield}} (1 - R)$$

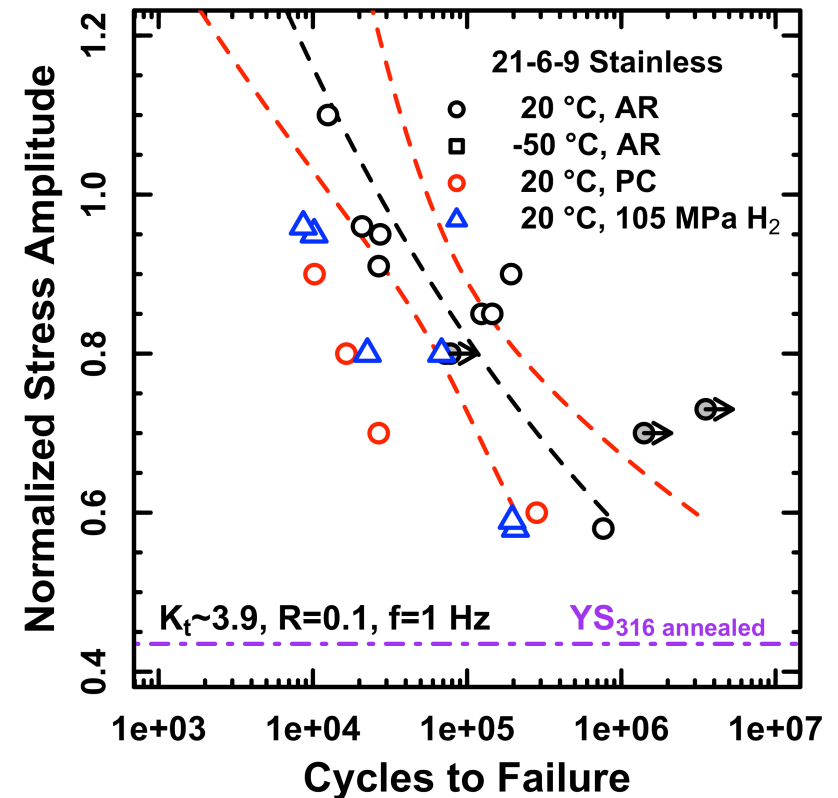
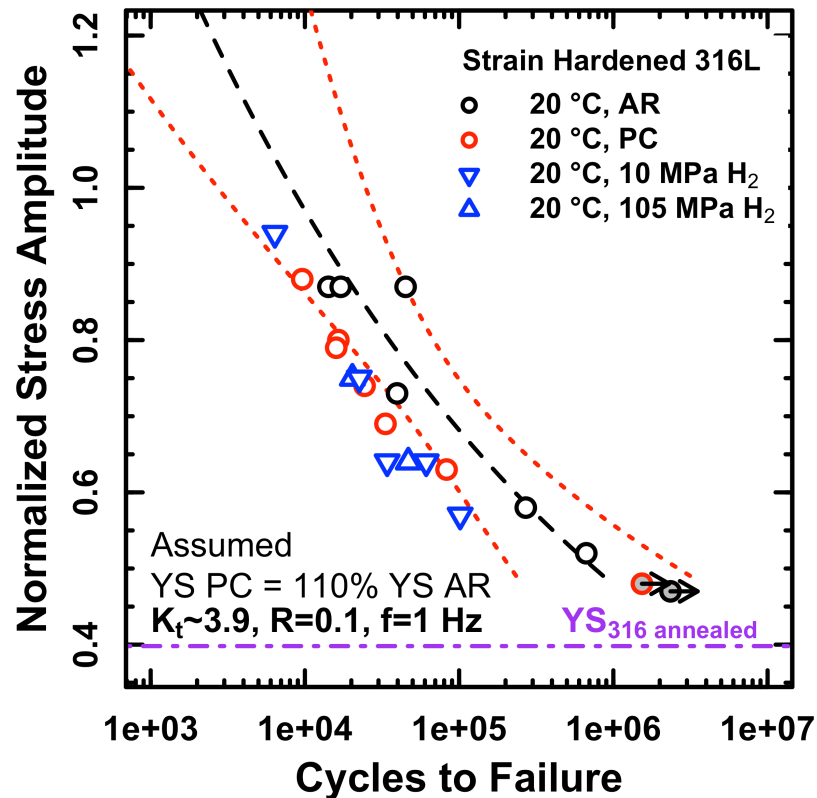
Hydrogen pre-charging to high H-levels changes the quasi-static tensile behavior

- ~5-10% increase in *Yield Strength*, minor change in *Hardening Rate*
- *May not be representative of in-service properties*



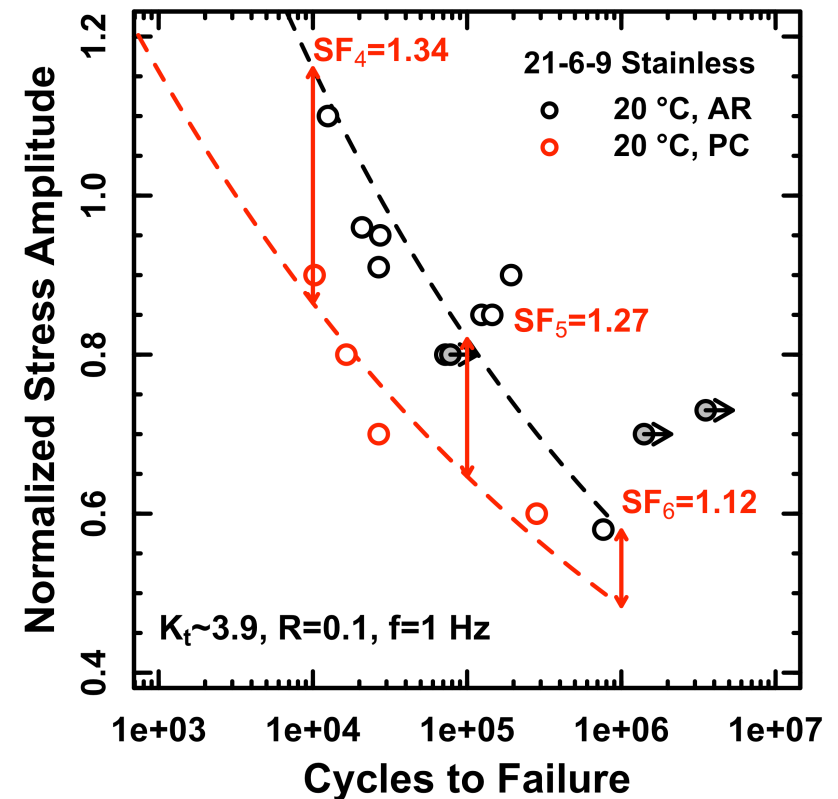
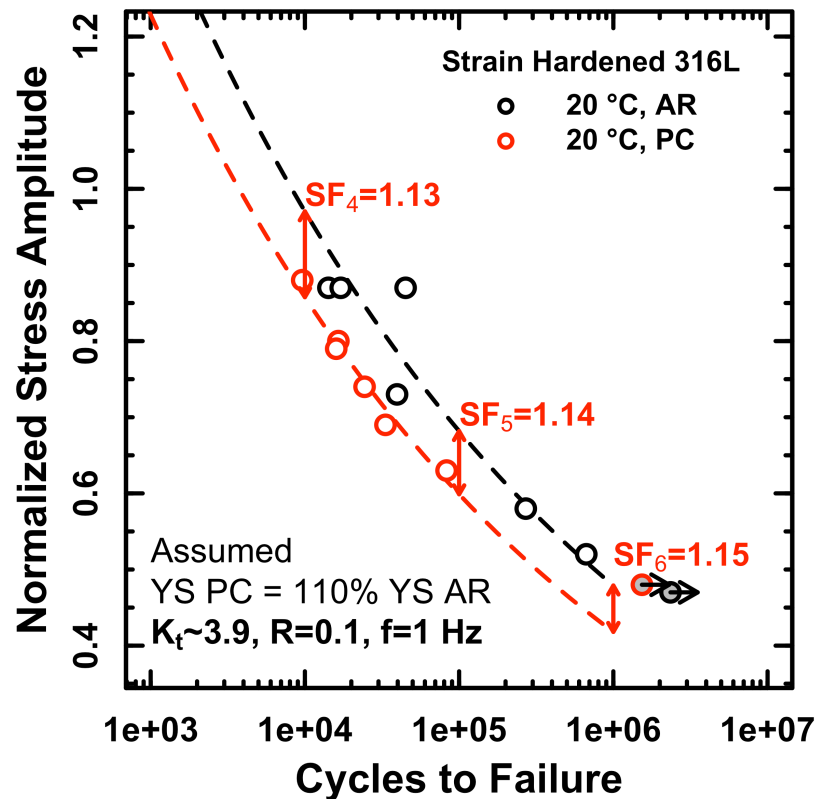
Safety Factor Multiplier: Internal Hydrogen

- Accounting for yield strength increase during thermal pre-charging results in all hydrogen data agreeing



Safety Factor Multiplier: Internal Hydrogen

- Safety factor determined using pre-charged samples similar to testing in external hydrogen



Summary

- Measurement of hydrogen effects on fatigue life enable materials selection based on relevant performance metric
 - **Fatigue highlights change in stress-performance not necessarily captured in quasi-static stress-strain data**
- High-strength austenitic stainless steels offer potential cost *and* weight savings
 - *Both 316L and 21-6-9 show measurable fatigue life change in presence of hydrogen (external and internal)*
- Safety factor multiplier results in equivalent fatigue life for a given criteria in air and hydrogen
- Thermally pre-charged samples could be an alternative to testing in external hydrogen if yield strength change accounted for

Acknowledgements

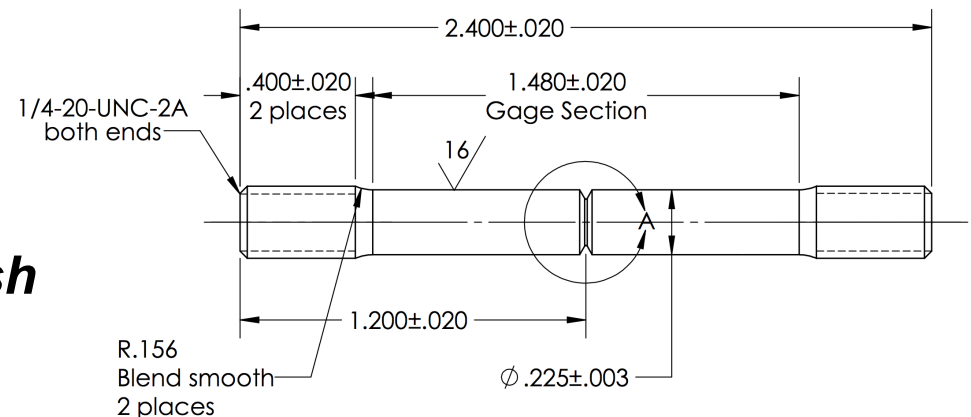
- **Jeff Campbell, Ken Lee, Brenden Davis, and Mark Zimmerman for experimental support**

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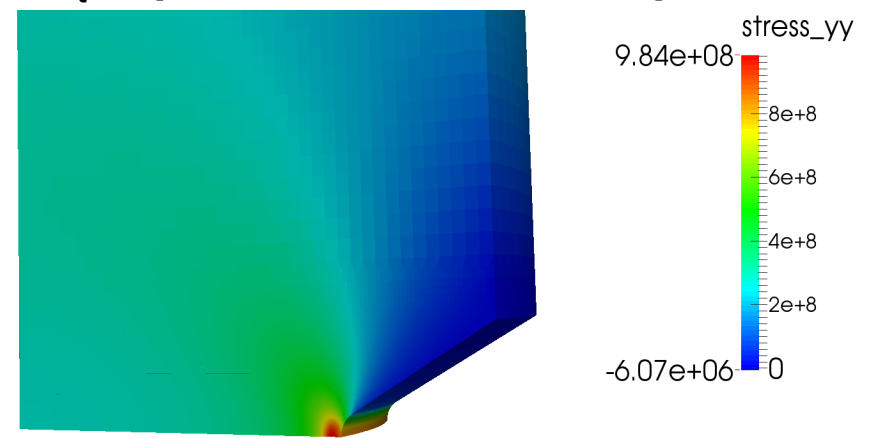
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Notched Sample Discussion

- Relatively straightforward testing
- **However:**
 - *Difficult to control surface finish of notch (scatter in data, like-like-testing)*
 - *No analytic solution for K_t (FEM)*
 - *Notch sensitivity may not be constant (material qualification)*
 - *Never know how much plasticity is present (artificially 'elastic' stresses) → Notch changes?*
- *Could smooth samples be tested to avoid these?*



$K_{t, \text{elastic}} \sim 3.9$ $K_{t, \text{elastic-plastic}} \sim 2.4$
 K_t dependent on stress-amplitude



Estimation of cost and weight reduction

- Relative component cost is estimated from the relative weight of material and material cost:

- *Weight is determined from required component thickness*
- *Cost estimated from price of bar stock*

ASME design equation

$$t = \frac{PD}{2(SE + PY)}$$

- **H₂ performance is not limiting use *Yield Strength* for design**

Material	Relative Material Cost	Yield Strength (MPa)	Relative Weight	Relative Cost Savings
316L	1.0	140	1.0	1.0
CW316L	2	589	0.24	0.52
21-6-9	0.79	539	0.26	0.79
304L	0.84	140	1.0	0.16
CW 21-6-9	1.6	620	0.17	0.73