

# The Role of Circumferential Cracking In Fracture Originating From Cylindrical Inclusions

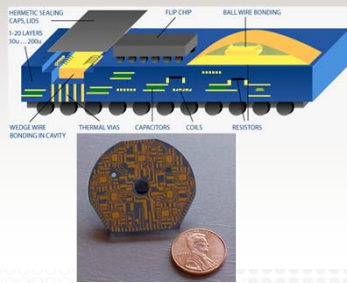
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**Materials Motivation:** Low temp. co-fired ceramic (LTCC) is versatile glass-filled ceramic composite packaging material. Cracking problems are sometimes observed during production and use.

**Mechanics Motivation:** Cylindrical inclusions, such as fibers, electrical vias, and metal connector pins are common features in many functional ceramic materials.

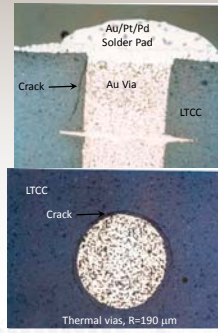
## Schematic of Technology



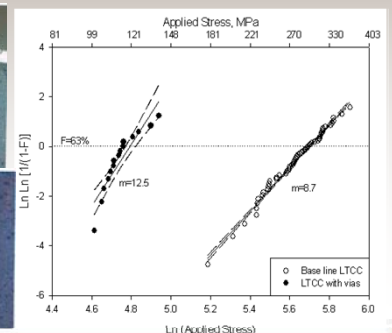
## LTCC Microstructure



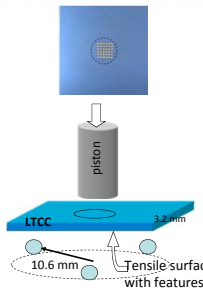
## Cracks at/near Vias



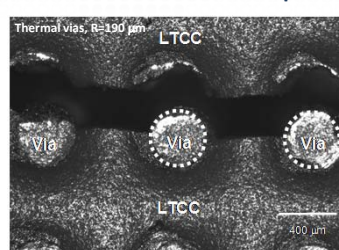
## ~60% Strength loss due to Vias



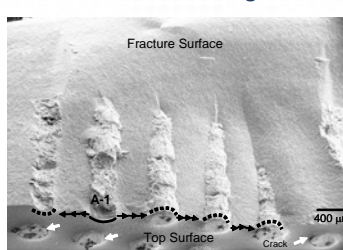
## Test Structure



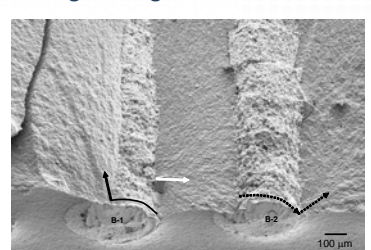
## Fractured Structure: Top



## Fracture Surface showing Crack Path

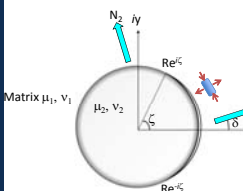


## Higher Mag. View of Fracture



Crack initiates at the via-LTCC boundary at several vias. These cracks grow as arc cracks along the interface as stress. At some load, one of the arc cracks kinks out into the matrix, links up with other cracks, and causes failure.

## Model: Interfacial Arc Crack



Driving Forces:  
-  $\sigma_{II}$ =tensile, as  $\alpha_2 > \alpha_1$   
- Biaxial loads,  $N_1$  and  $N_2$   
-  $\sigma_{\theta\theta}$ =compressive. In-plane stress for kinking analysis

## Stress Intensity Factors

**K due to thermal expansion mismatch**

$$K_{th} = F_{th} \sigma_{th} \sqrt{\pi R \sin \zeta} (1 - \alpha) (1 - 2i\gamma) e^{-\gamma \zeta} e^{i \left( \frac{\zeta}{2} - \gamma \ln(2R \sin \zeta) \right)}$$

$$F_{th} = \frac{f(\alpha, \beta, \gamma)}{(1 + \alpha)(1 - \beta) - (1 - \alpha)(1 + \beta) e^{-\gamma \zeta} (\cos \zeta + 2\gamma \sin \zeta)}$$

$$\sigma_{th} = -\frac{E_1}{1 + \nu_1} (A_1 - A_2) \Delta T \text{ where } A_1 = A_1 \text{ [for plane stress], or } = (1 + \nu_1) A_1 \text{ [for plane strain]}$$

W. H. Müller and S. Schmauder, Int. J. Sol. Struc., 29 (1992)

**K due to Biaxial Loading**

$$K_L = \left( \frac{b_1 - b_2 e^{i\zeta}}{2} \right) \sqrt{\pi R \sin \zeta} (1 - 2i\gamma) e^{-\gamma \zeta} e^{i \left( \frac{\zeta}{2} - \gamma \ln(2R \sin \zeta) \right)}$$

$$b_1 = \frac{(1 + \alpha)(N_1 + N_2) - (1 - \alpha)(N_1 - N_2)(0.25 + \gamma^2) \sin^2 \zeta \cos 2\zeta}{(1 + \alpha)(1 - \beta) - (1 - \alpha)(1 + \beta) e^{-\gamma \zeta} (\cos \zeta + 2\gamma \sin \zeta)}$$

$$b_2 = \frac{1(1 + \alpha)(N_1 - N_2)(0.25 + \gamma^2) \sin^2 \zeta \sin 2\zeta}{(1 - \beta)(1 + \beta) e^{-\gamma \zeta} (\cos \zeta + 2\gamma \sin \zeta)}$$

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## Interface K, G expressions

**Interface K, G expressions**

$$K_{Net} = K_I + iK_{II}$$

$$K_I = F_I \sqrt{\pi R \sin \zeta} (\cos \psi + 2\gamma \sin \psi) e^{-\gamma \zeta}$$

$$K_{II} = F_{II} \sqrt{\pi R \sin \zeta} (\sin \psi - 2\gamma \cos \psi) e^{-\gamma \zeta}$$

$$F_I = F_I(\sigma_{\theta\theta} + \sigma_{\alpha\alpha} + \alpha[\sigma_{\theta\theta} - \sigma_{\alpha\alpha}]); \psi = \frac{\zeta}{2} + \gamma \ln(2R \sin \zeta)$$

$$G = \frac{K_{Net} \overline{K_{Net}}}{4 \cosh^2 \frac{\pi \gamma}{2}} \left[ \frac{1}{\mu_1(1 + \nu_1)} + \frac{1}{\mu_2(1 + \nu_2)} \right]$$

R. Tandon, in preparation

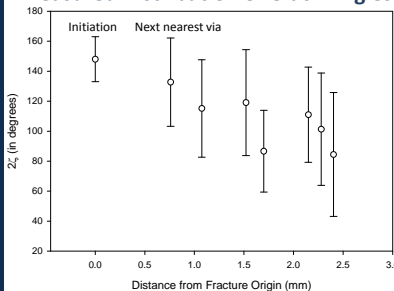
## Elastic Properties

Material, #	E (GPa)	$\nu$	$\mu$ (GPa)	$\alpha$ (ppm/°C)
LTCC, 1	116	0.24	47	7
Gold-Via fill, 2	78	0.44	27	15

Calculated Values (Plane-Stress)				
$\alpha$	$\beta$	$\gamma$	$\sigma_{\theta\theta}$ (MPa)	
-0.1958	-0.01464	0.0046	156	

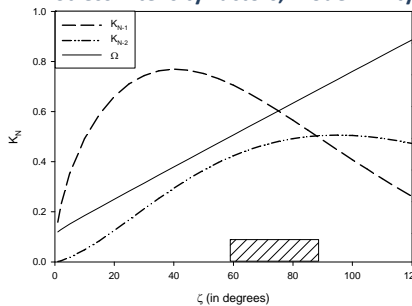
Note:  $\gamma$  is extremely small,  $\sim 0$   
 $K_I, K_{II} \sim K_I, K_{II}$

## Measured Distribution of Crack Angles



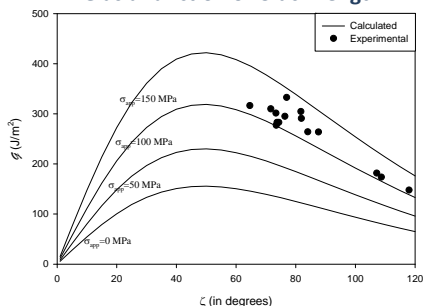
The via causing fracture was identified, and the crack angle at which kinking occurred was measured.  $2\zeta = 148 \pm 15^\circ$ .

## Stress Intensity Factors, Mode Mixity



Normalized K's:  $K_I$  rises and decays more rapidly than  $K_{II}$ . Mode mix increases with  $\zeta$ .

## G as a function of Crack Length



G has a maximum at  $\zeta \sim 50^\circ$ . Kinking occurs on the decreasing portion of G-curve.

## Crack kinking out of an interface

(He et al. J. Am. Ceram Soc. 74, (1991))

where  $G_{max,0}$  is the maximized energy release rate at kink angle  $\omega$ .

$$\frac{\Gamma_{interface}}{\Gamma_{substrate}} = \frac{G_{interface}}{G_{max,0}}$$

Non-dimensional length parameter which characterizes the stress parallel to the interface.

$$\eta = \frac{\sigma_0 \sqrt{a}}{\sqrt{(E, G)}}$$

$\eta$  plays an important role in determining kink condition

## For Arc Crack at LTCC-Au interface

- Crack kinking equality eq. satisfied due to decrease in  $\Gamma_{interface}$  with increasing  $\zeta$ .
- For interface arc crack, mode mixity  $\Omega$  at kinking  $\sim 40^\circ$ .
- Even though there is a compressive stress parallel to the interface, the contribution of the biaxial applied stress must exceed this value  $\Rightarrow$  stress parallel to the interface must be positive
- There does not appear to be any natural length parameter 'a' for calculation of  $\eta$
- Taking  $\eta=0$  will provide an upper bound for  $\Gamma_I$
- Therefore  $\Gamma_I/\Gamma_{II} \sim 0.6$  ( $\alpha=0$ ), or  $\sim 0.8$  ( $\alpha=0.5$ )
- $\Gamma_I \sim 9 \text{ J/m}^2$ . Max.  $\Gamma_I \sim 5.5-7.2 \text{ J/m}^2$  - very brittle

## Conclusions and Issue

Large amounts of stable crack growth followed by kinking lead to lower strength, and lower variability

As  $G_{interface}$  decreases with increasing  $\zeta$ , kinking is expected even if  $\Gamma_I$  is not a strong function of  $\Omega$

Increasing  $\Gamma_I$ , and stiffness of matrix will lead to higher strength

Is the interface fracture toughness a function of geometry as well?  $\Omega$  increases even though  $K_{II} \downarrow$