

# **Thermal/Chemical Degradation of Ceramic Candle Filter Materials**

**Final Report  
September 1988 - October 1994**

January 1995

Work Performed Under Contract No.: DE-AC21-88MC25203

For  
U.S. Department of Energy  
Office of Fossil Energy  
Morgantown Energy Technology Center  
Morgantown, West Virginia

By  
Acurex Environmental Corporation  
Mountain View, California

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January 1995

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## SECTION 1

### EXECUTIVE SUMMARY

The development of advanced coal-fueled power generation systems, such as integrated gasification-combined cycles and pressurized fluidized-bed combustors (PFBC), can provide cost-effective alternatives for power generation, improve coal utilization, and decrease our dependence on oil and gas. The direct firing of coal produces particulate matter which must be removed for both environmental and process reasons. For example, in a PFBC, the gas must be cleaned of particulate matter before expanding through the turboexpander in order to protect the downstream turbine components. Under the auspices of the U.S. Department of Energy (DOE), ceramic candle filters are being developed for high-temperature applications to increase the efficiency of current and advanced fuel combustion processes. The filters protect downstream equipment from erosion and impingement of particulate matter in the hot exhaust gases. Ceramic candle filters are rigid, closed-ended porous cylinders with a flange on the open-ended side. The flange at the open end allows the candle to be suspended by a plate (tubesheet) in the filter vessel.

Ceramic candles have shown promise in hot gas filtration applications, but they have encountered durability problems during actual use in hostile, high-temperature environments. The objective of this project is to identify thermal and chemical degradation mechanisms in candle filters. Sub-objectives are to (a) develop test procedures, and (b) apply the test procedures to testing and predicting performance of representative candle materials.

At the start of this project, the mechanisms involved in the degradation of candle filters were not understood. The degradation mechanisms were broken into two main areas, thermal and chemical. The investigations in the chemical mechanisms were performed by Virginia Polytechnic



Institute (VPI) under the project under a major subcontract to Acurex Environmental. Acurex Environmental investigated the thermal and the combined thermal and chemical mechanisms. The primary tests completed were performed on candles composed of a support structure of silicon carbide grains held together by an alumino-silicate binder.

Most of the ceramic candle examinations conducted by Acurex Environmental were performed on Schumacher Diaschumalith (Schumacher) and Refractron candles. A report on the analysis of Schumacher candles exposed to the KRW gasifier is provided in Appendix F.

The Schumacher candle consists of two porous layers. The outer layer acts as the filtration surface and is made of a thin, relatively dense (wash coat) layer of alumino-silicate fibers in an alumino-silicate binder. The thick inner layer primarily supports the thin outer filtration layer of the candle and consists of a highly porous structure.

The two-layered Refractron 70/3, Refractron 50/5, and IFP LayCer candles are similar to the Schumacher candles in that they are made by bonding the silicon carbide (SiC) grains together with an alumino-silicate clay binder and they have two separate layers (one for filtration and one for structural support). Similar to the Schumacher, the 9 mm inner supporting layer consists of coarse SiC grains. The thin (1 mm) outer filtration layer is made of fine SiC grains.

The homogeneous single layer Didier SiC20 candle does not have two separate layers. The Didier candle uses the same sized SiC grains throughout the whole candle.

Acurex Environmental also investigated a candle that is not made of SiC grains. Industrial Filter and Pump (IFP) has developed a fibrous filter with a vacuum-formed body which is post-treated to strengthen the high stress areas and modify the surface pore size. Acurex Environmental examined the durability of the IFP Fibrosic candle and tubesheet in this project.

Under operating conditions, ceramic candle filter systems use pulse cleaning as the filter-cleaning mechanism. Pulse cleaning of candles is accomplished by introducing a sharp pulse of high-pressure gas into the individual candle via a manifold. The gas is usually delivered from the cleaning nozzle at near sonic velocity. Also, because of cost and material limitations, the pulse gas

is not heated and is therefore, at about room temperature. In addition, because of the expansion across the nozzle, the pulse gas is adiabatically cooled. The relatively cold gas is introduced into the candle, which is at operating temperature (typically, 370°C (700°F) to 1,000°C (1,830°F)).

The early method used for strength comparisons was four-point flexure tests, but C-ring testing was selected as the better method of testing the strength degradation of ceramic candles. Acurex Environmental developed and designed the testing system used for this project. C-rings are obtained by sectioning rings out of the candles and then making a longitudinal slit (perpendicular to the section direction) on one side of the ring (see Figure 1-1). C-ring testing is an excellent method because it leaves the candle surfaces (inner bore and the outer surface of the tube) intact, and thus leaves intact the surface flaw population.

Figure 1-1 also shows how the c-ring specimens are loaded (indicated by the arrows). Under tension, the inner surface of the candle at a point opposite the slit experiences the maximum

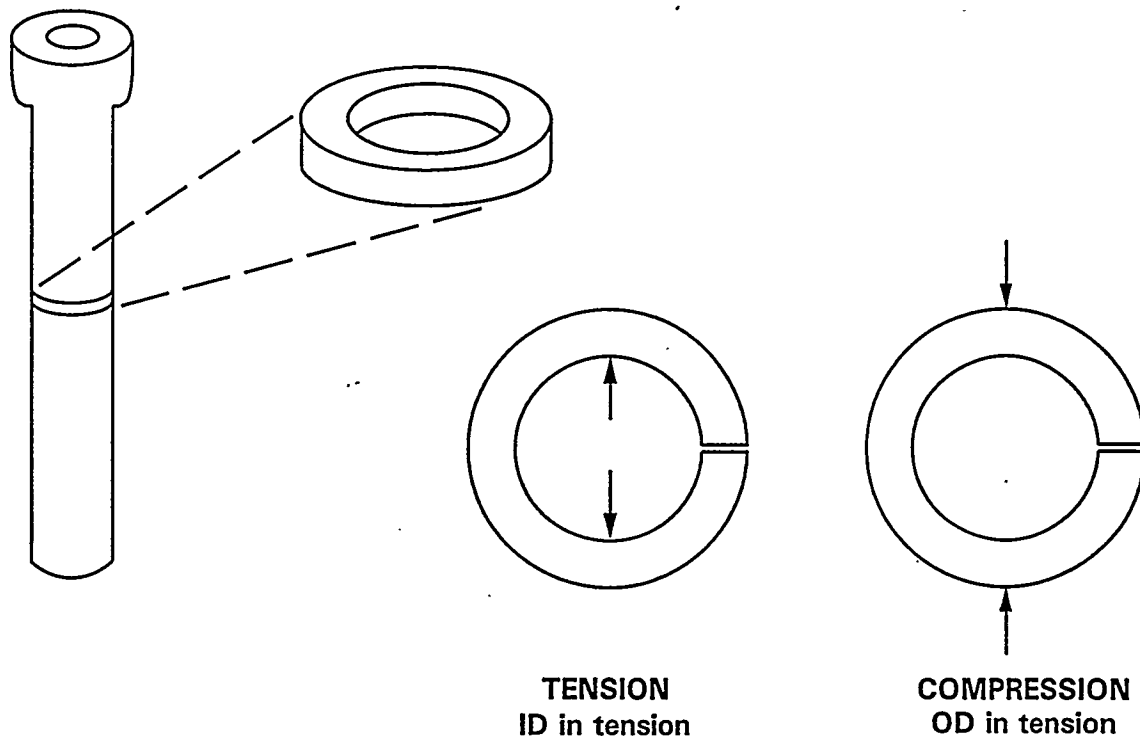


Figure 1-1. C-ring specimens from a ceramic candle

tensile stress, while the outer surface experiences the maximum compressive stress. Under compression, the outside surface (opposite the slit) bears the maximum tensile stress. Ceramics are much stronger under compressive than tensile stresses; therefore, failure/fracture of the c-ring will originate at the surface with the maximum tensile stress. This phenomenon allows focusing on a surface (inner or outer) to investigate its strength degradation under different environmental conditions. The areas of interest (opposite the slit in the c-ring) are the inner surface for c-ring tension testing ( $R=R_1$ ) and the outer surface for c-ring compression testing ( $R=R_2$ ).

The analytical expression for the maximum c-ring stress is based on the bending of curved beams, and is as follows:

$$\sigma = \frac{F T}{2 W R \left[ \frac{(R_1 + R_2) \ln(R_2/R_1)}{2} - T \right]},$$

where,

$\sigma$  = maximum stress

$F$  = maximum load

$R$  =  $R_1$  for tensile loading of c-ring (tension tests)

$R_2$  for compressive loading of c-ring (compression tests)

$R_1$  = Inside radius of c-ring

$R_2$  = Outside radius of c-ring

$W$  = width/thickness of c-ring

$T$  =  $R_2 - R_1$ .

C-ring testing was performed to justify the use of this approach for strength measurement of ceramic candle filters and to validate the use of the above equation for maximum stress calculations. The typical c-ring specimen width is 0.75 inches. Due to the design and size

limitations of the thermal exposure test facility, c-ring widths of 0.50 inches were used. To investigate the possibility of specimen size influencing the strength measurements, Acurex Environmental performed c-ring tension and compression tests on 0.50-, 0.75-, 1.00-, and 1.25-in. specimens on Schumacher and Refractron candles. Figures 1-2, 1-3, 1-4, and 1-5 summarizes these tests.

Figures 1-2 through 1-5 demonstrate that the maximum stress values experienced with 0.5-in c-ring specimens are representative of the strength of the candle. The optimum c-ring width would be 0.75-in because it would be an inefficient use of material to use c-ring widths greater than 0.75-in if 0.75-in specimens would result in the same maximum stress value as the larger specimens. 0.5-in specimens are acceptable, but they are less desirable because they are more fragile than the larger 0.75-in specimens. 0.5-in specimens are also harder to work with at high temperature.

To investigate why tension strength levels were different from compression strengths, Acurex Environmental tested an unused homogenous candle made by Didier. The homogenous candle does not have a filtration layer that is different from the supporting body such as the Schumacher and Refractron candles. The Didier candle should therefore have the same strength in c-ring tension as in compression because the material on the inner surface is the same as on the outer surface. Figure 1-6 shows graphically that the strength of the material at the inner bore of the candle (via the tension test) had the same material strength as the material at outer filtration surface (via the compression test). This verifies the use of the stress equation for determining the strength of a material irrespective of the surface being examined.

One of the possible causes of the durability problems encountered by candle filters is thermal shock/thermal fatigue. It has been observed that candles exposed to operating conditions resulted in loss of strength. Acurex Environmental investigated the effect of ambient air pulses on candle strength at elevated temperatures. Figures 1-7, 1-8, and 1-9 summarizes the tests conducted. The candles were pulsed 5,000 times with ambient air down the inner bore while the candle was at 870°C (1,600°F). No c-ring compression tests were conducted because any strength degradation

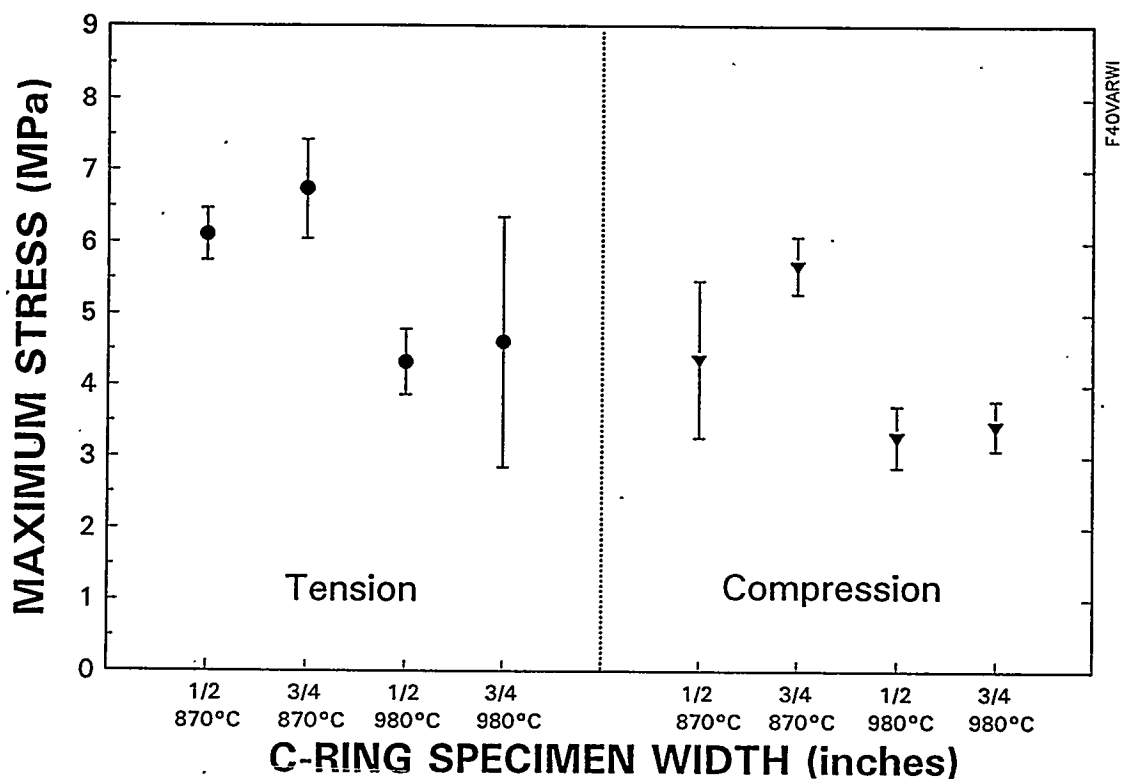


Figure 1-2. Maximum stress comparisons of 0.5-in and 0.75-in c-ring specimens: Schumacher Diaschumalith F40

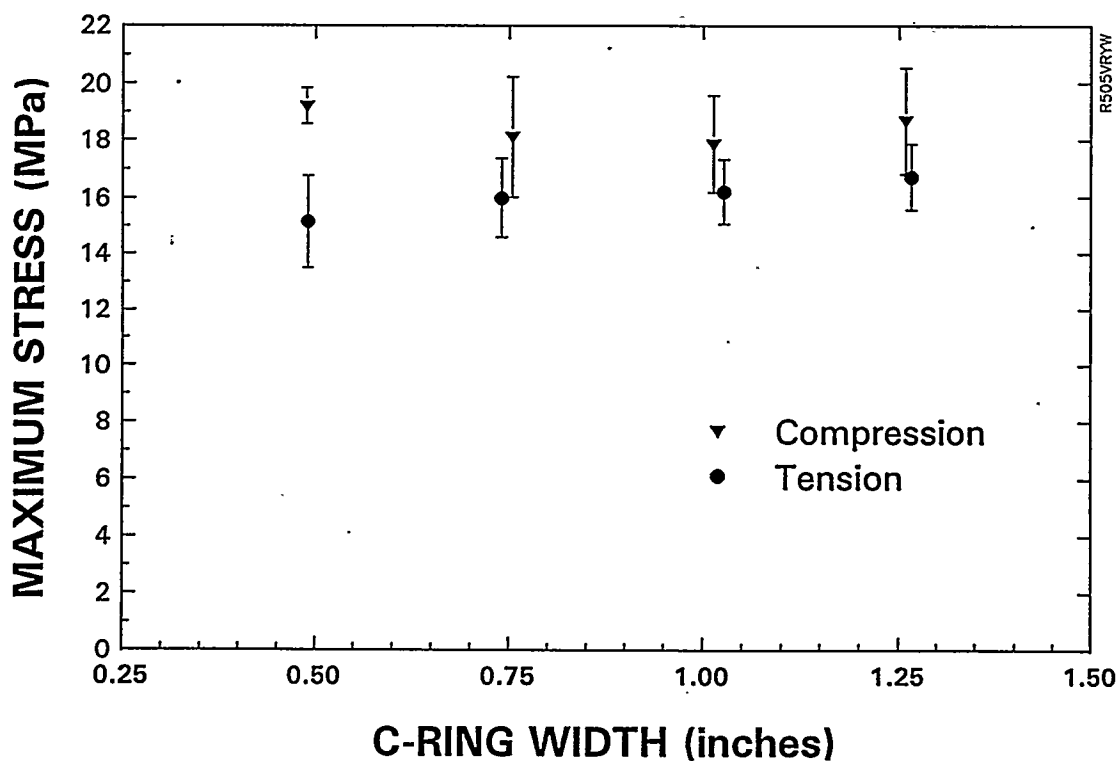


Figure 1-3. Maximum stress values for a Refractron 50/5 with 505 binder candle of various c-ring widths

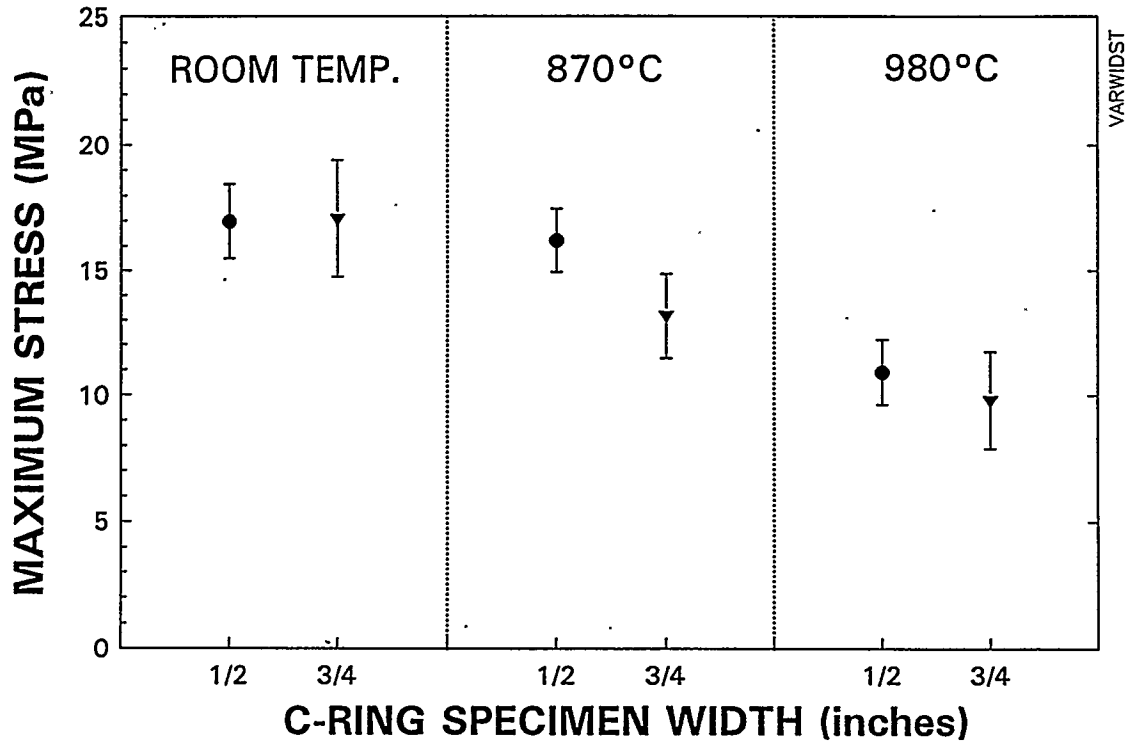


Figure 1-4. Maximum stress comparisons of 0.5-in and 0.75-in c-ring tension specimens: Refractron 70/3 with 442-T binder

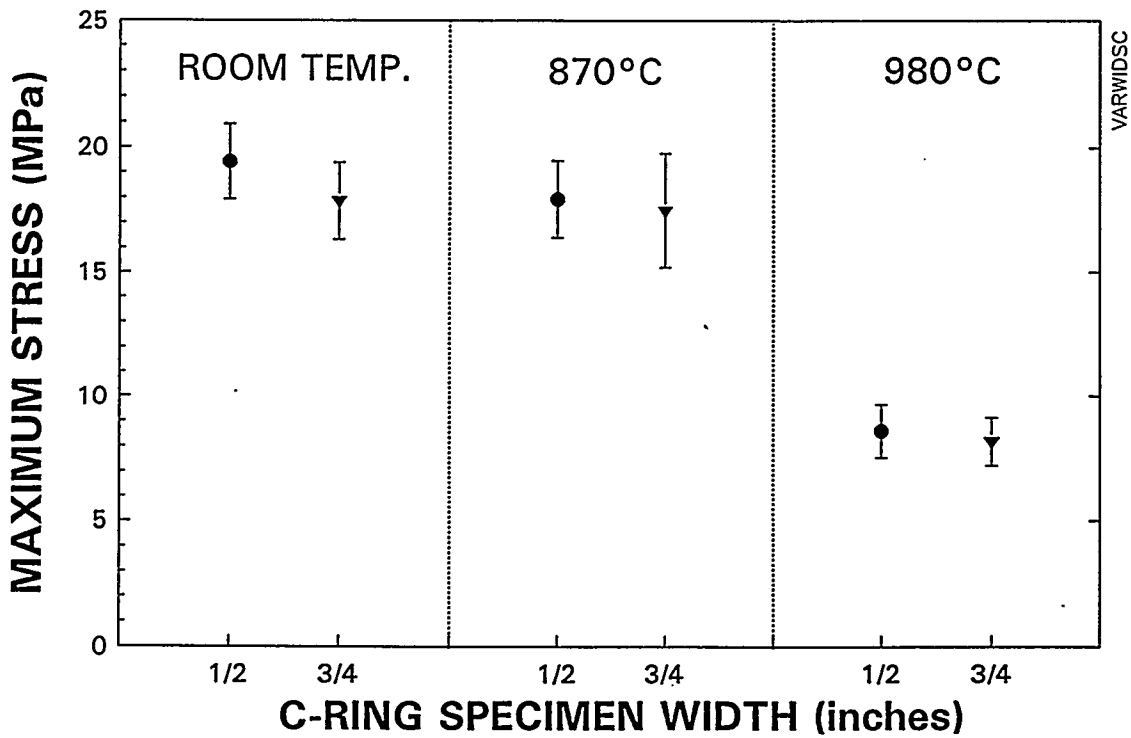


Figure 1-5. Maximum stress comparisons of 0.5-in and 0.75-in c-ring compression specimens: Refractron 70/3 with 442-T binder

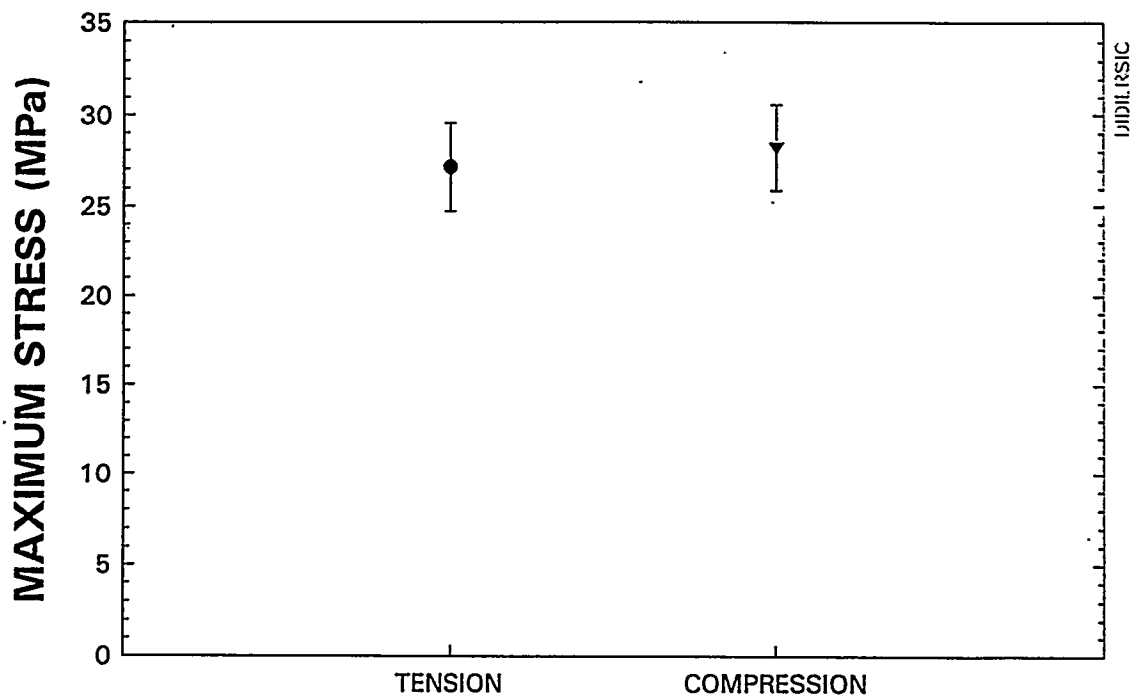
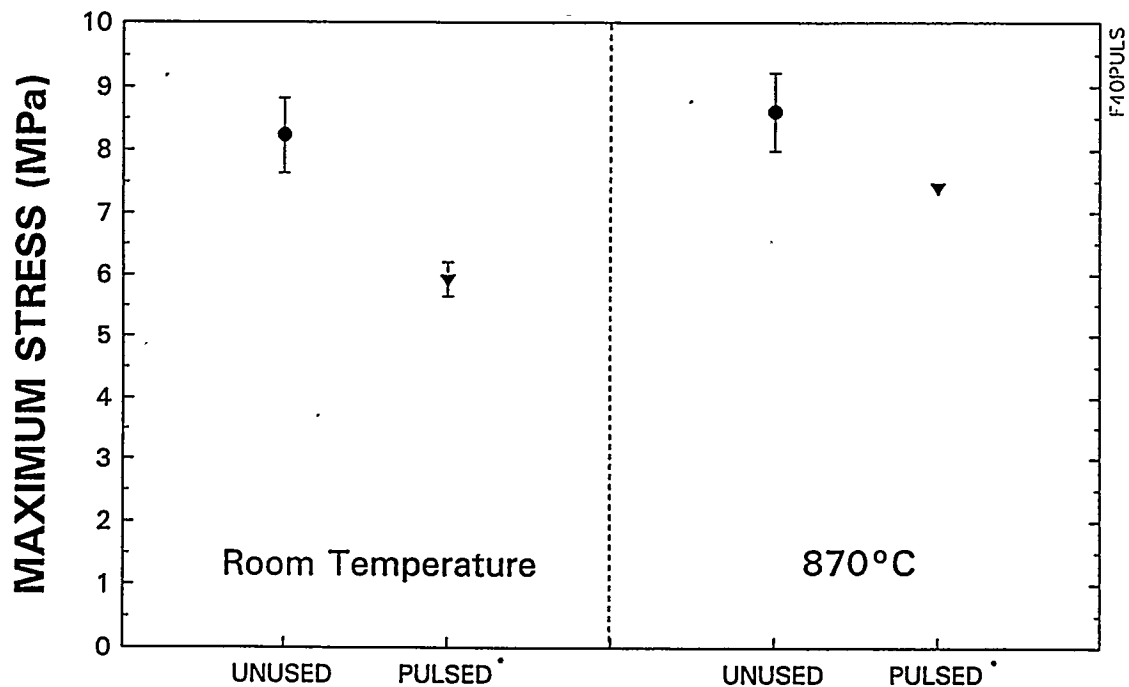
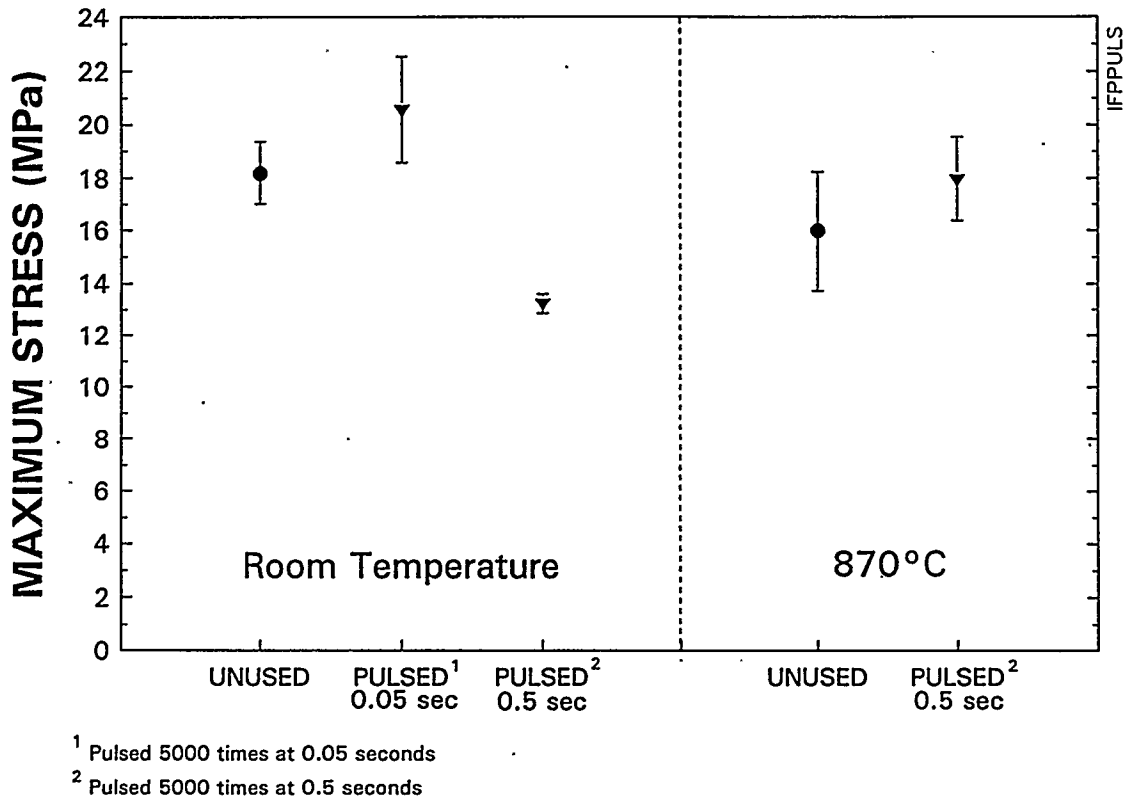


Figure 1-6. Maximum stress comparison of c-ring tension with c-ring compression testing: Didier SiC20

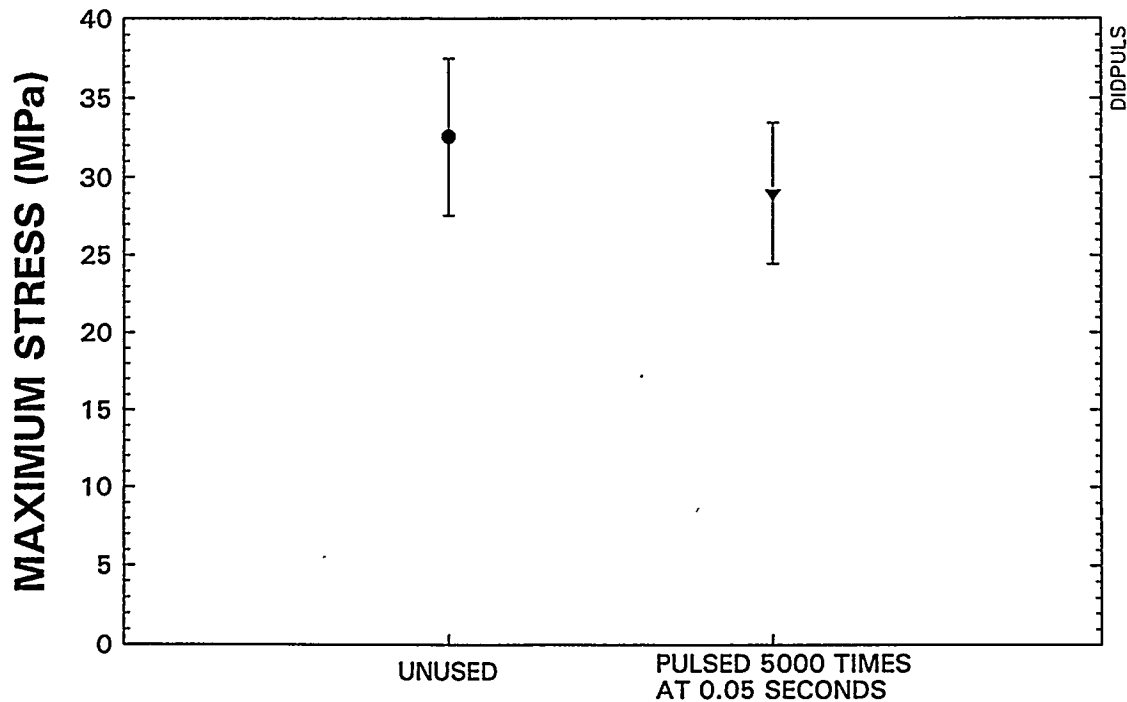


\* Pulsed 5000 times at 0.5 seconds

Figure 1-7. C-ring tension test results of a Schumacher Diaschumalith F40 candle pulsed 5,000 times with ambient air while at 870°C



**Figure 1-8. C-ring tension test results of an IFP Laycer candle pulsed 5,000 times with ambient air while at 870°C**



**Figure 1-9. Room temperature c-ring tension test results of a Didier SiC20 candle pulsed 5,000 times with ambient air while at 870°C**



due to thermal shock from the pulse air would be most apparent on the inner surface of the candle (c-ring tension).

The unused candles have no history of pulse cleaning nor of temperature cycling. It can be seen that the Schumacher and FP LayCer candles pulsed with time durations of 0.5 seconds exhibit room temperature strength loss. The strength loss due to these pulses can be attributed to thermal shock in which temperature gradients produce stresses under conditions that restrain the free expansion. The temperature distribution could be such that free expansion of each volume element would separate the individual volume elements in such a way that they would not fit together. If the maximum stress for a material is reached, failure (such as cracking and spalling) will result. The effect of thermal stresses depends not only on the absolute level of stresses, but also the stress distribution, stress duration, and material characteristics (such as ductility, homogeneity, porosity, and presence of pre-existing flaws).

As a control, the unused candles (in Figures 1-7 and 1-8) tested at high temperature resulted in stress values within the range of room temperature tests of the same unused candles. This suggests that the elevation of temperature will not affect the strength of c-ring specimens if there is no change in the surface characteristics of the candle. The control also acts as a reference to which strength values from exposed candles can be compared.

With stress duration also being a determining factor, it can be seen that the FP LayCer and Didier candles pulsed at a time duration of 0.05 seconds (refer to Figures 1-8 and 1-9) do not show any room temperature strength loss (the averages are within their standard deviations). This series of room temperature tension test comparisons does imply that ambient air pulses to candles at operating temperatures cause strength degradation (possibly via micro-cracking on the inner bore of the candle) when the pulse duration is at least 0.5 seconds.

The purpose of the long-term exposure testing at Acurex Environmental was to examine the synergistic effects of long term exposure to temperature, alkali vapor, steam, and pulse cleaning on candle filter materials.

Schumacher Diaschumalith F40 and Refractron 70/3 with 442-T binder candles were tested under the following conditions:

- 20/65/95 ppm NaCl (by volume)
- 15/20 % steam (by volume)
- Atmospheric pressure
- 870°C/980°C (1,600°F/1,800°F)
- With/without 0.5-second pulse cleaning
- 100/400/500/1,000 hours of exposure

The candle face velocities ranged between 8 to 17 ft/min.

The Schumacher and Refractron candles did not retain their integrity when tested at 980°C (1,800°F). Contamination from metals, such as Fe and Cr, from the corrosion byproducts of the vessel and pre-vessel piping may have contributed to the degradation of the exposure candle specimens.

Cristobalite formation was detected in granular aluminosilicate by VPI in their hydrothermal tests. Strength degradation of the Schumacher candle can be attributed to the formation of cristobalite that was detected in the exposure specimens.

The initial strength degradation and subsequent strength increase of the Refractron candle may be attributed to the formation of albite. More investigation is needed on the effects of albite on the characteristics of the candle filter.

The modeling of the thermal/chemical degradation mechanisms to predict filter lifetime was unsuccessful.

Industrial Filter and Pump (IFP) has developed a fibrous filter that is different from the rigid candles (typically SiC grains held together by an alumino-silicate binder) tested above. The filter is a vacuum-formed body which is post-treated to strengthen the high stress areas and modify the surface pore size. In late 1991 and early 1992, Acurex Environmental performed durability testing (funded by EPRI) of the Fibrosic candle under development by IFP. In operation, the

Fibrosic candles performed well (having high collection efficiency with low pressure drop), but the candle durability was suspect because the candles catastrophically failed during testing. These candles were assembled by gluing a flange onto the vacuum-formed body. The tests showed the glued joint to be the weak area. The burst tests of these filters showed the used candles to have a candle burst pressure of approximately half the unused candles. These measurements indicated there had been significant strength loss from use.

Since these original tests, IFP has developed the Fibrosics filter further and the preliminary indications are that the filter is much stronger than measured previously. Acurex Environmental assessed the durability of the filter by performing an accelerated pulse cleaning (15,000 pulses) test at 870°C (1,600°F) and a test with actual filtration operating conditions (4,000 ppm dust load at 870°C for 100 hours). Cylindrical burst pressure tests were performed to assess the effect of testing on the mechanical strength of the candle filters.

Both Fibrosic candles had collection efficiencies greater than 99.99% during and after the 15,000 pulse cleanings test and also the 100 hour operational filtration test. The unused Fibrosic candle (Fibro2800) tested in this project had a burst pressure of 20 psig. There was about a 25% drop in burst pressure for both of the Fibrosic candles tested as compared to the 50% burst pressure loss for the earlier version of the Fibrosic candle. The modifications by IFP have improved the durability of the Fibrosic candle.

IFP has also been developing a fibrous ceramic tubesheet and candle hold-down system. This approach has appeal by reducing the amount of metal subjected to high temperature gas. There was no long term testing data of the structural integrity of the Fibrosic tubesheet at high temperature prior to this project. Acurex Environmental tried to simulate the "worst" case structural loading conditions that the tubesheet would encounter during operation. The tubesheet was loaded with 113 pounds, similar to that expected from the filter, dust, and hold-down plate at 870°C for 500 hours. The rate at which the tubesheet deflected was highest during the first 24 hours. After 500 hours, the Fibrosic tubesheet has a deflection/sag of about 0.1 inch (measured

at the lowest point across the width of the tubesheet). Aside from a few minor cracks, the tubesheet retained its structural integrity.

To date, field demonstrations of ceramic candle filters have had limited success. Examination of SiC candle filters has shown that the filters experience significant strength degradation that limits the operational life of the candle filters. This project has provided a better understanding of the degradation mechanisms of candle filters which will allow for improved designs of candle filters and more effective adaptation to operational constraints. Before this project began, thermal degradation from pulse cleaning was thought to be insignificant. It obviously is a contributing factor in candle filter integrity. The information derived from this project has improved the level of knowledge concerning ceramic candle filters.

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## **SECTION 2**

### **INTRODUCTION**

Under the auspices of the U.S. Department of Energy (DOE), high-temperature ceramic candle filters are being developed for use in advanced power generation systems such as the Integrated Gasification Combined Cycle (IGCC), Pressurized Fluidized-Bed Combustor (PFBC), and Direct Coal-Fired Turbine (DCFT). The direct firing of coal produces particulate matter which must be removed to meet both environmental and process limitations. The ceramic candles increase the efficiency of the advanced power generation systems and protect downstream equipment from erosion and impingement of particulate matter in the hot exhaust gases. Ceramic candle filters are rigid, closed-ended (capped on one side) porous cylinders which generally have a flange on the open-ended side. The flange at the open end allows the candle to be suspended by a tubesheet in the filter vessel.

Candle filters have shown promise, but have also encountered durability problems during use in hostile, high-temperature environments. Limitations in the candle lifetime lower the economic advantages of using candle filters for this application. Candles typically fail by cracking at the flange or in the body of the candle.

#### **2.1 PROBLEM STATEMENT**

Ceramic candle filters have encountered durability problems during actual use in advanced coal utilization projects. Filters have cracked during use at Grimethorpe PFBC and at the KRW gasifier. It has been observed that candles exposed to operating conditions have experienced loss of material strength.<sup>1</sup>

To date, the mechanisms that contribute to ceramic candle degradation are not well understood, and data on the properties of the candle filters at elevated temperatures were limited. The major degradation mechanisms of the candles are thought to be thermal fatigue and corrosion.

## **2.2 PROJECT OBJECTIVE**

The objective of this project was to test and analyze ceramic candle filter materials and to evaluate the degradation mechanisms. The tests were conducted such that the effects of each degradation mechanism could be examined separately. The overall objective of the project was to (a) develop a better understanding of the thermal and chemical degradation mechanisms of ceramic candle filter materials in advanced coal utilization projects, (b) develop test procedures, and (c) recommend changes to increase filter lifetime.

## **2.3 TEST SCOPE**

In order to assess the effects of operating conditions on the candle filters, both unused and used candles were tested. This project primarily tested the rigid Schumacher Diaschumalith F40 and Refractron 70/3 with 442-T binder candles, and the fibrous Fibrosic (Fibro2800) candle. C-ring testing was developed to evaluate the mechanical strength of rigid candle filters after being exposed to various environmental/exposure conditions. Some candles were exposed to accelerated pulse cleaning conditions to examine the strength degradation due to thermal shock. Prior to the beginning of this project, strength degradation due to thermal shock was thought to be minimal or nonexistent. Thermal/chemical exposure tests were performed on the rigid candles. The Fibrosic material has the potential to be the next generation of hot gas filtration materials. The Fibrosic tubesheet was examined to see what the effects of operational load were on the tubesheet over time at operating temperature.

## SECTION 3

### TECHNICAL DISCUSSION

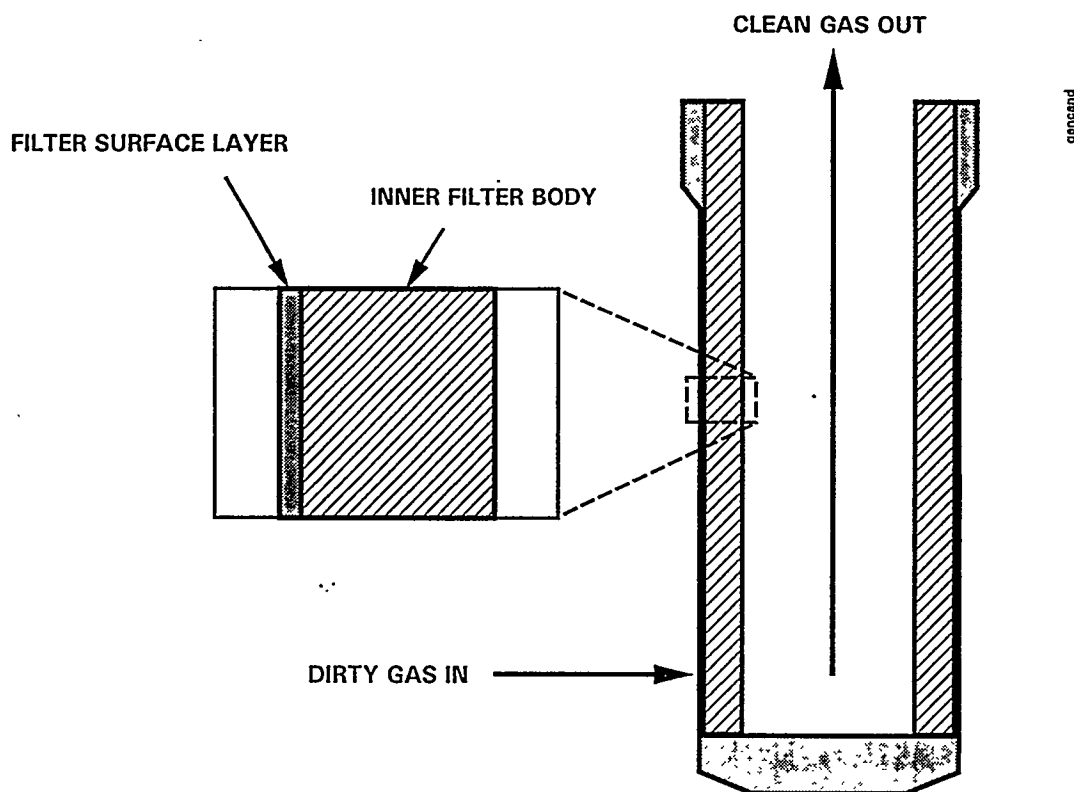
#### 3.1 BACKGROUND

There are two major types of ceramic filters currently under development for advanced coal utilization processes: ceramic candle filters and ceramic crossflow filters. The high-temperature filter systems in these processes typically operate at temperatures from 370°C (700°F) to 1,000°C (1,830°F) and at pressures from 10 to 30 atm. These processes may either be oxidizing or reducing; and the atmospheres contain alkali metals, carbon monoxide, sulfur, calcium, hydrogen, iron, and water vapor. This project focused on the investigation of ceramic candle filters only. Ceramic candle filters are rigid, tubular filters made by bonding ceramic fibers or grains, or a combination thereof. The most common forms of commercial candle filters for the applications of interest are composed of silicon carbide or alumina-silica grains held together with a clay binder.

A typical candle configuration consists of two porous layers. The outer filtration surface is a relatively dense thin layer with small pore openings. This surface layer is the primary filtration surface, and can be formed from small grains or fine fibers. This thin outer layer is supported by a thick, highly porous grain structure used mainly to impart overall strength (see Figure 3-1).

Pulse cleaning is used as the filter-cleaning mechanism for ceramic candle filter systems under operating conditions. Pulse cleaning is accomplished by a sharp pulse of high-pressure gas introduced into the individual candle via a manifold. The gas is usually delivered from the cleaning nozzle at near sonic velocity. Also, because of cost and material limitations, the pulse gas is not heated and is therefore, at about room temperature. In addition, because of the expansion across





**Figure 3-1. Ceramic Candle filter element**

the nozzle, the pulse gas is adiabatically cooled. The relatively cold gas is introduced into the candle, which is at operating temperature (typically, 370°C to 1,000°C).

Ceramic candle filters have been tested in a variety of PFBC and IGCC demonstration projects. In most of these demonstrations, the candles have failed during use. The failures have been caused primarily by thermal or mechanical shock. For example at the Grimethorpe PFBC<sup>2</sup>, ceramic filter candles failed twice. The first failure occurred when a pulse cleaning solenoid valve stuck open, dumping cold pulse air into a pulse cleaning manifold until the pulse reservoir pressure equaled the system pressure, thermally shocking the filter candles. In the second, separate failure, high face velocities created sufficient pressure drop to lift the candles from the tubesheet. The pressure released and the candles dropped back into the tubesheet, breaking some of the candles and damaging the flanges on many others. -

At KRW (refer to Appendix F), the same type of clay-bound silicon carbide (SiC) filter used at the Grimethorpe facility failed early in the testing because of thermal shock. A water quench

upstream of the filter malfunctioned and ceased operation. The temperature of gas through the filter vessel rose rapidly to about 925°C (1,700°F). When the water quench was reinitiated, the filters broke immediately. In a second episode at KRW, filters broke when a slug of solids moved through the system and impacted the filters.

As a part of this project, Acurex Environmental analyzed selected ceramic candle filters which have been exposed at the KRW gasifier. The purpose was to determine if exposure had been detrimental to the strength and filtration properties of the candle filters. A copy of this report, entitled "Analysis of Candle Filters from the KRW Gasifier," is presented in Appendix F. This project attempted to investigate thermal and chemical degradation mechanisms that affects the life expectancy of ceramic candle filters.

### 3.2 MATERIALS SELECTION

The initial candle examined by Acurex Environmental in this project was the Schumacher DiaSchumalith (Schumacher) F40. The Schumacher candle was being used at the American Electric Power Tidd PFBC Demonstration Project and were the most widely available candle filter at the time. The porous outer layer acts as the filtration surface and is made of a thin, relatively dense (wash coat) layer of alumino-silicate fibers in an alumino-silicate binder. The pore size of the filtration surface is approximately 15 to 30  $\mu\text{m}$ . The outer membrane retains particulates greater than 0.1  $\mu\text{m}$  in diameter. The thick inner layer primarily supports the thin outer filtration layer of the candle and consists of a highly porous structure (pore size of  $\approx 300$  to 500  $\mu\text{m}$ ).

Around mid-project, the Refractron 70/3 with 442-T binder candle was examined because it was a candidate for the next generation of candle filter design. The two-layered Refractron candles tested in this project had an outside diameter of 60 mm and a wall thickness of 10 mm. The candle filters are made by bonding the silicon carbide (SiC) grains together with an alumino-silicate clay binder. The 9 mm inner supporting layer consists of a coarse SiC grain (grade 70). The thin (1 mm) outer filtration layer has a fine SiC grain (grade 3). Hence, the designation "70/3" is derived from the grades of the inner and outer SiC grains.

The latter part of the project involved the preliminary assessment of Industrial Filter & Pump's (IFP) fibrous filter and tubesheet (Fibrosic). The fibrous filter is different from all the other rigid candles (typically SiC grains held together by an alumino-silicate binder) tested in this project. The filter is a vacuum-formed body which is post-treated to strengthen the high stress areas and modify the surface pore size. It is currently thought that fibrous candles have better thermal shock characteristics than rigid candles. Because of the unique way in which the fibrous candles are constructed, thermal stresses are not as great, probably due to low thermal expansion. The fibrous tubesheet was also examined for its structural integrity as a possible replacement for the standard metal tubesheet used in candle filter vessels.

The other candle materials examined in this project to a lesser extent than the above materials were the Refractron 50/5 with 505 binder, the IFP LayCer, and the Didier SiC20 candle. The Refractron 50/5 with 505 binder candle has the same dimensions as the previously mentioned Refractron 70/3 with 442-T binder. The difference is that the Refractron 50/5 with 505 binder candle has a 9-mm inner supporting layer which consists of grade 50 SiC grains and the thin 1-mm outer filtration layer of grade 5 SiC grains. The IFP LayCer candle has the same dimensions and grain sizes as the Refractron 50/5 with 505 binder candle because IFP was initially marketing this candle under the name of LayCer. The Didier SiC20 candle is a homogeneous single layer candle that does not have two separate layers (one for filtration and one for structural support). The Didier candle is made up of SiC grains of grade 20 throughout the whole candle.

### **3.3 TEST PROCEDURES**

#### **3.3.1 Long-Term Thermal Exposure Tests**

Porous ceramic candle filter material degradation may be caused by several factors working in combination. Pulse cleaning with room-temperature gas may cause temperature drops in the filter sufficient to cause alkali vapor to condense in the porous filter as well as thermally shocking/fatiguing the candle. The alkali deposition may provide areas for alkali/ceramic reactions to occur, leading to weakening of the filter material. Thermal shock may induce micro-cracking that

reduces the material strength. The purpose of the long-term exposure testing is to provide data on the synergistic or antagonistic effects on candle filter materials of long term exposure to temperature, alkali vapor, steam, and pulse cleaning.

Schumacher Diaschumalith and Refractron 70/3 with 442-T binder candles were tested under the following atmospheric conditions:

- 20/65/95 ppm NaCl (by volume)
- 15/20% steam (by volume)
- Atmospheric pressure
- 870°C/980°C (1,600°F/1,800°F)
- With/without pulse cleaning
- 100/400/500/1,000 hours of exposure

The pulse cleaned candles were exposed to pulse cleaning durations of 0.5 seconds. The pulse cleaning gas enters the candles through a quarter inch tube which is attached to a solenoid valve with a pressure reservoir (set at 80 psig) on the other side. The levels of alkali tested at atmospheric pressure simulate the operating alkali concentrations of 2 to 10 ppm at 10 atmospheres.

The candle face velocities ranged between 8 to 17 ft/min. Table 3-1 lists the different combinations tested and the letter designation used for identification in this report.

Tests A and B were conducted in the original exposure vessel made of RA 330. About midway through tests C and D, the RA 330 exposure vessel failed. The remainder of tests C and D were conducted in a stainless steel vessel. Tests E through M (all at 870°C) were also conducted in a stainless steel vessel. Tests N through Q were conducted in a Haynes 556 (a corrosion resistant superalloy) vessel. Figure 3-2 is an assembly drawing of the Haynes 556 vessel. Not shown in the figure is the alumina liner (tube) that was inserted inside the exposure vessel to minimize alkali corrosion of the superalloy. This also minimized vessel corrosion by-products from being deposited on the candle specimens. On Figure 3-2, the gas entered from the left side of the drawing and the gas exited out the right side. All of the design drawings for the Haynes 556 exposure vessel are

**Table 3-1. Thermal exposure test conditions**

Test	Pulse Cleaning	NaCl Concentration (ppm)	Steam (%)	Temperature (°C)	Exposure (hours)
<b>SCHUMACHER DIASCHUMALITH F40 CANDLE</b>					
A	No	95	20	870	100
B	Yes	65	20	980	100
C	Yes	65	20	980	475
D	Yes	65	20	980	575
E	Yes	20	15	870	100
F	Yes	20	15	870	400
G	Yes	20	15	870	500
H	Yes	20	15	870	1,000
<b>REFRACTRON 70/3 WITH 442-T BINDER CANDLE</b>					
I	No	20	15	870	100
J	Yes	20	15	870	100
K	Yes	20	15	870	400
L	Yes	20	15	870	500
M	Yes	20	15	870	1,000
N	Yes	20	15	980	100
O	Yes	20	15	980	400
P	Yes	20	15	980	500
Q	Yes	20	15	980	1,000

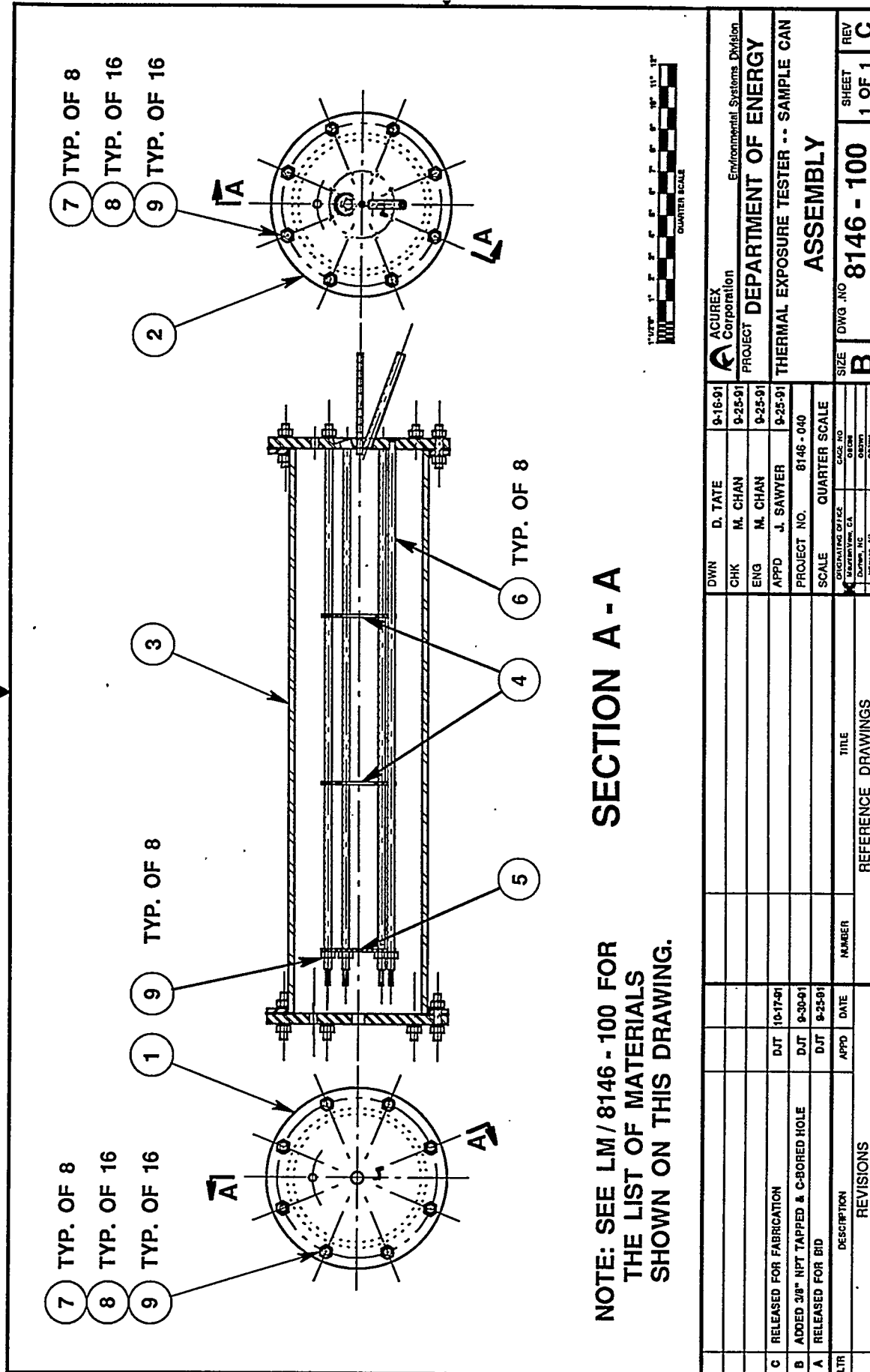


Figure 3-2. Thermal exposure sample vessel

included in appendix A. The RA 330 and stainless steel vessels were similar to the Haynes 556 exposure vessel except they were smaller, had no alumina liner, and only had flanges on the exit side of the vessel.

The Acurex Environmental thermal exposure system was designed to accommodate the running of two exposure vessels simultaneously. Figure 3-3 is the schematic of the dual high-temperature flow-through exposure test system. The design allowed the two exposure processes to operate with sufficient independence that if one of the processes had to be shut down, the other exposure process/test could continue running. Both systems shared the same air supply, air preheat ("A" system in Figure 3-3), and pulse cleaning system. This sharing lowered space and cost requirements.

The "B" system pertains to the stainless steel exposure vessel (which is inside the B-3 heater), and the "C" system denotes components of the Haynes exposure vessel (which is inside the C-3 heater). The B-x and C-x designations are clam shell heaters for the respective exposure processes/systems that heat up the gas lines and the exposure vessels. The only critical components that could shut down both processes at the same time are the air supply, the electrical power supply, the "A" system, the pulse vessel, or the timer for the pulse cleaning system.

In the C-system, the alkali was designed to be injected into C-2. This design was feasible due the size of the Haynes exposure vessel (it is twice the length of the B-system vessel) and the two heated sections (A-1 and C-1) that preheated the air which enabled the alkali solution to vaporize.

In the B-system, the alkali was injected into B-1 which is heated and then mixed with preheated air from A-1. The mixture was then heated some more by B-2 before entering the exposure vessel in B-3. The B-system design was deemed undesirable due to alkali deposition onto the inner walls of the lines leading into the exposure vessel, which sometimes resulted in tests shutting down due to blockages of the line. Alkali deposition was not constant nor predictable

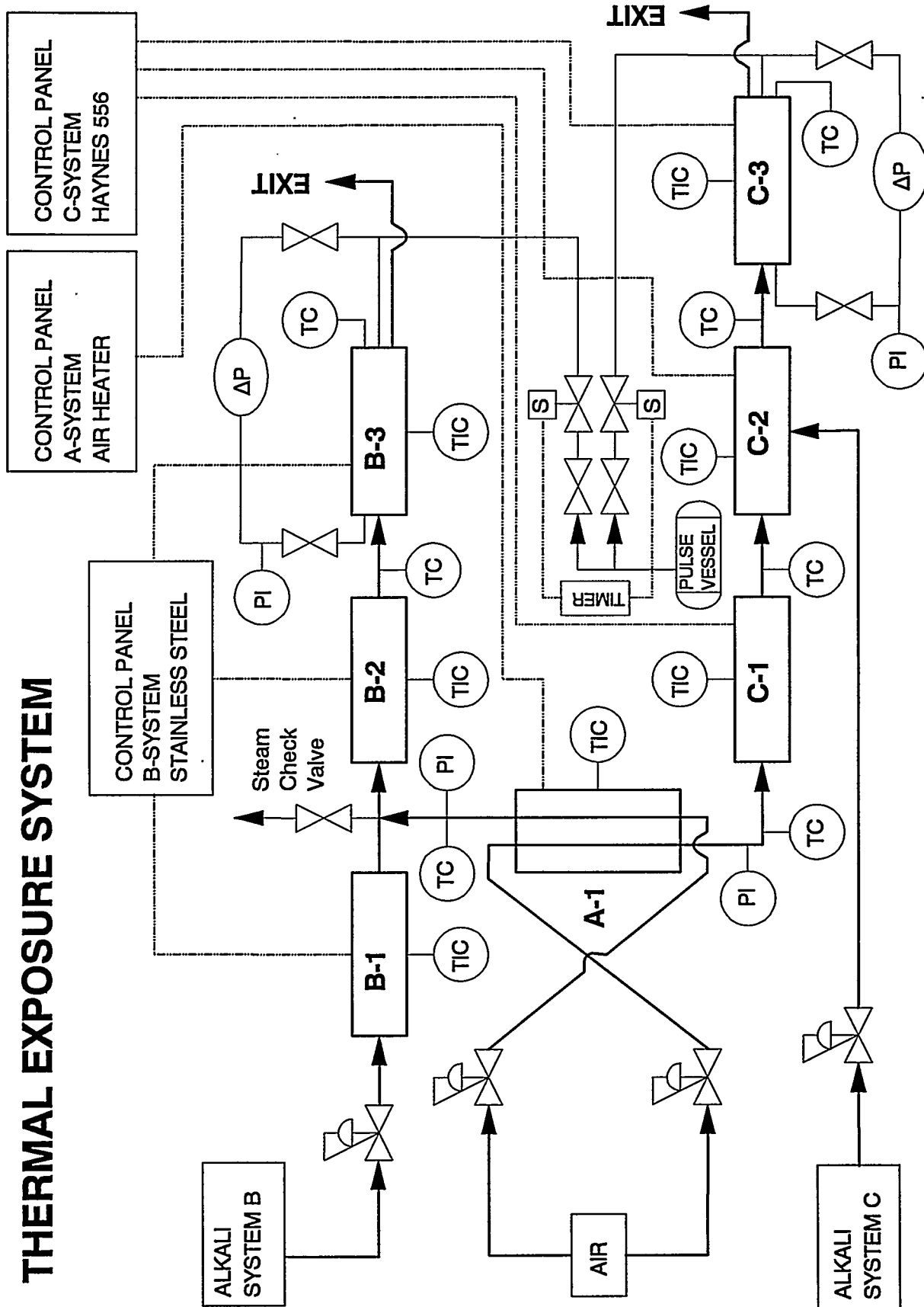


Figure 3-3. Schematic of dual flow-through thermal exposure system



because the process lines (and connectors) were stainless steel that exhibited scaling and becoming embrittled due to the alkali, steam and temperatures.

The C-system minimized alkali deposition into the process line by injecting the alkali solution into the pipe in a heater just prior to the exposure vessel, thereby reducing the length of the process line exposed to the alkali. The two heaters (A-1 and C-1) also helped by preheating the air close to the test temperature before even coming in contact with the alkali solution. There were no shut downs attributed to alkali blockage of the lines leading to the exposure vessel.

The methodology for exchanging samples consisted of putting the 1,000-hr specimen into the exposure vessel with the 100-hr specimen. After 100 hours of exposure, the 100-hr specimen was replaced with the 500-hr specimen (leaving the 1,000-hr specimen in the vessel). After 500 hours of exposure, the 500-hr specimen was replaced with a 400-hr specimen (the 1,000-hr specimen still remaining inside the vessel). After 400 more hours of exposure, the 1,000-hr specimen had accumulated about 1,000 hours of exposure. The 1,000-hr specimen was situated in the back of the vessel near the gas exit of the vessel, and the other specimens (100-hr, 500-hr, and 400-hr) were butted against the front of the 1,000-hr specimen which was nearer to the gas entrance into the vessel. Figure 3-4 shows the placement of the candle specimens inside the exposure vessel.

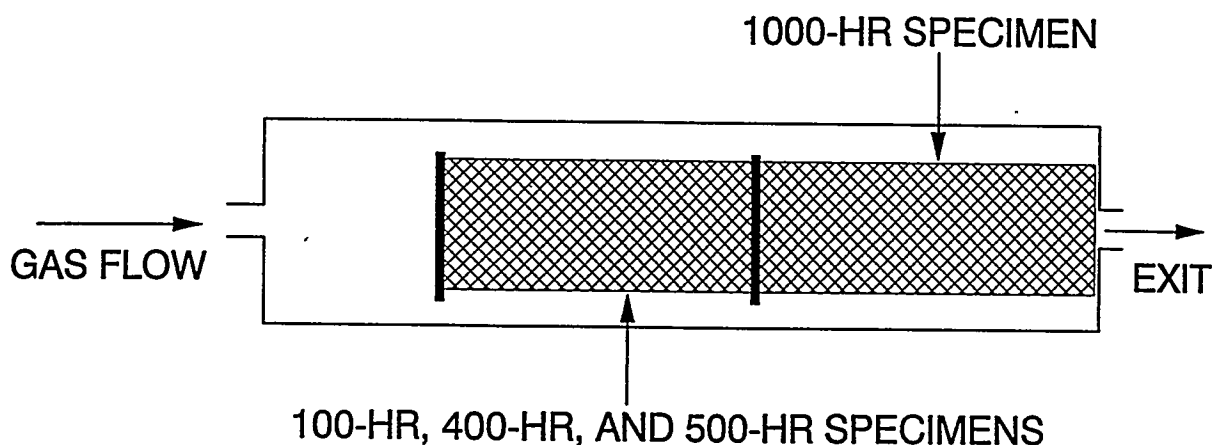


Figure 3-4. Candle specimen placement inside the exposure vessel

The pulse cleaning system pulsed ambient air into the bore of the candle specimens approximately every 30 minutes during testing. The duration of each pulse is 0.5 seconds. Table 3-2 summarizes the approximate total pulse cleaning history of each candle specimen.

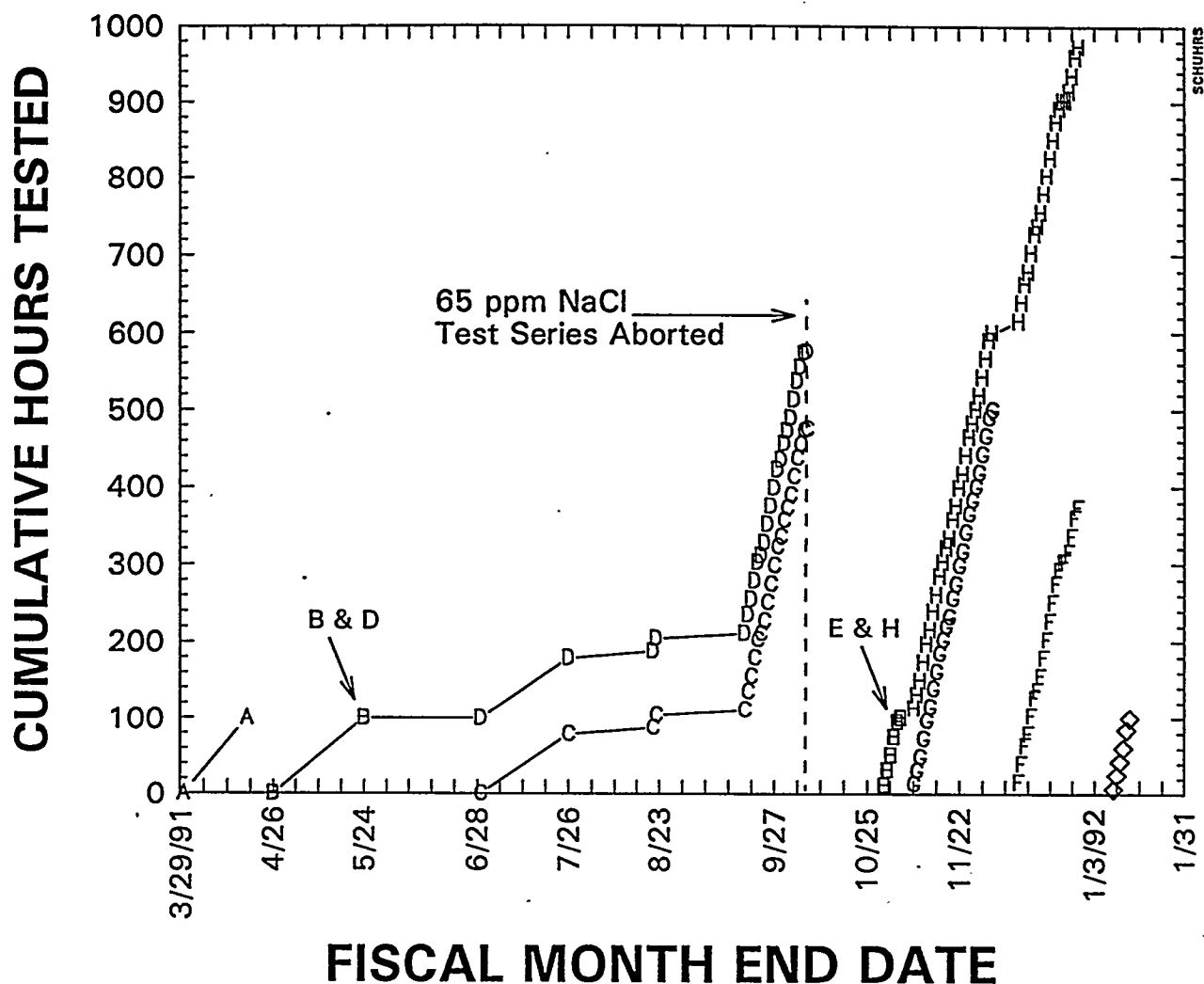
If the system ran perfectly, the number of pulses for each test would be double the number of hours tested because the system was pulsed every half hour. However, due to problems such as heater failures, blockage of alkali injection orifices, process line failures, and drops in alkali flow, the process down times were deducted from the cumulative exposure times. Therefore, the actual cumulative hours that the system was running are greater than the cumulative exposure times.

The alkali system for each process system (B or C in Figure 3-3) consisted of a dual pressurized reservoir for the NaCl/water mixture. The alkali solution was injected through a 0.005 inch orifice into the respective process system. The dual alkali reservoir system for each process allowed for continuous operation whereby the depressurization and filling of alkali solution into one reservoir could be accomplished while the other pressurized reservoir continued alkali injection.

Problems due to high temperature operation and obstruction of the alkali injection orifice caused delays in the beginning months of thermal exposure testing with the Schumacher Diaschumalith F40 candles. Figure 3-5 shows the cumulative hours of exposure at the stated conditions for the Schumacher candles and a Refractron 50/5 with 505 binder candle. The 65 ppm NaCl exposure testing condition was aborted because the level of NaCl was deemed to be excessive. The process line and exposure vessel failed prematurely. The testing of the Refractron 50/5 with 505 binder candle was aborted because the testing of the Refractron 70/3 with 442-T binder candle was considered more appropriate. At the time, the Refractron 50/5 candle was no longer going to be available commercially. Figure 3-6 shows the cumulative hours tested for the Refractron 70/3 with 442-T binder candles under the stated conditions. It can be seen in Figure 3-6 that the stainless steel system [operating at 870°C (1,600°F)] was running concurrently with the Haynes system [operating at 980°C (1,800°F)] and when one system was down, the other system was able to continue testing unaffected.

Table 3-2. Pulse cleaning history

Test	Pulse Cleaning	Test Description/Condition	Total Pulses
SCHUMACHER DIASCHUMALITH F40			
A	No	95 ppm NaCl/20%H <sub>2</sub> O/870°C/100 hrs	0
B	Yes	65 ppm NaCl/20%H <sub>2</sub> O/980°C/100 hrs	200
C	Yes	65 ppm NaCl/20%H <sub>2</sub> O/980°C/475 hrs	1,050
D	Yes	65 ppm NaCl/20%H <sub>2</sub> O/980°C/575 hrs	1,250
E	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	230
F	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/400 hrs	920
G	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/500 hrs	1,140
H	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/1,000 hrs	2,290
REFRACTRON 70/3 WITH 442-T BINDER			
I	No	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	0
J	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	240
K	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/400 hrs	980
L	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/500 hrs	1,170
M	Yes	20 ppm NaCl/15%H <sub>2</sub> O/870°C/1,000 hrs	2,390
N	Yes	20 ppm NaCl/15%H <sub>2</sub> O/980°C/100 hrs	230
O	Yes	20 ppm NaCl/15%H <sub>2</sub> O/980°C/400 hrs	840
P	Yes	20 ppm NaCl/15%H <sub>2</sub> O/980°C/500 hrs	1,060
Q	Yes	20 ppm NaCl/15%H <sub>2</sub> O/980°C/1,000 hrs	2,130



### Schumacher Dia-Schumalith

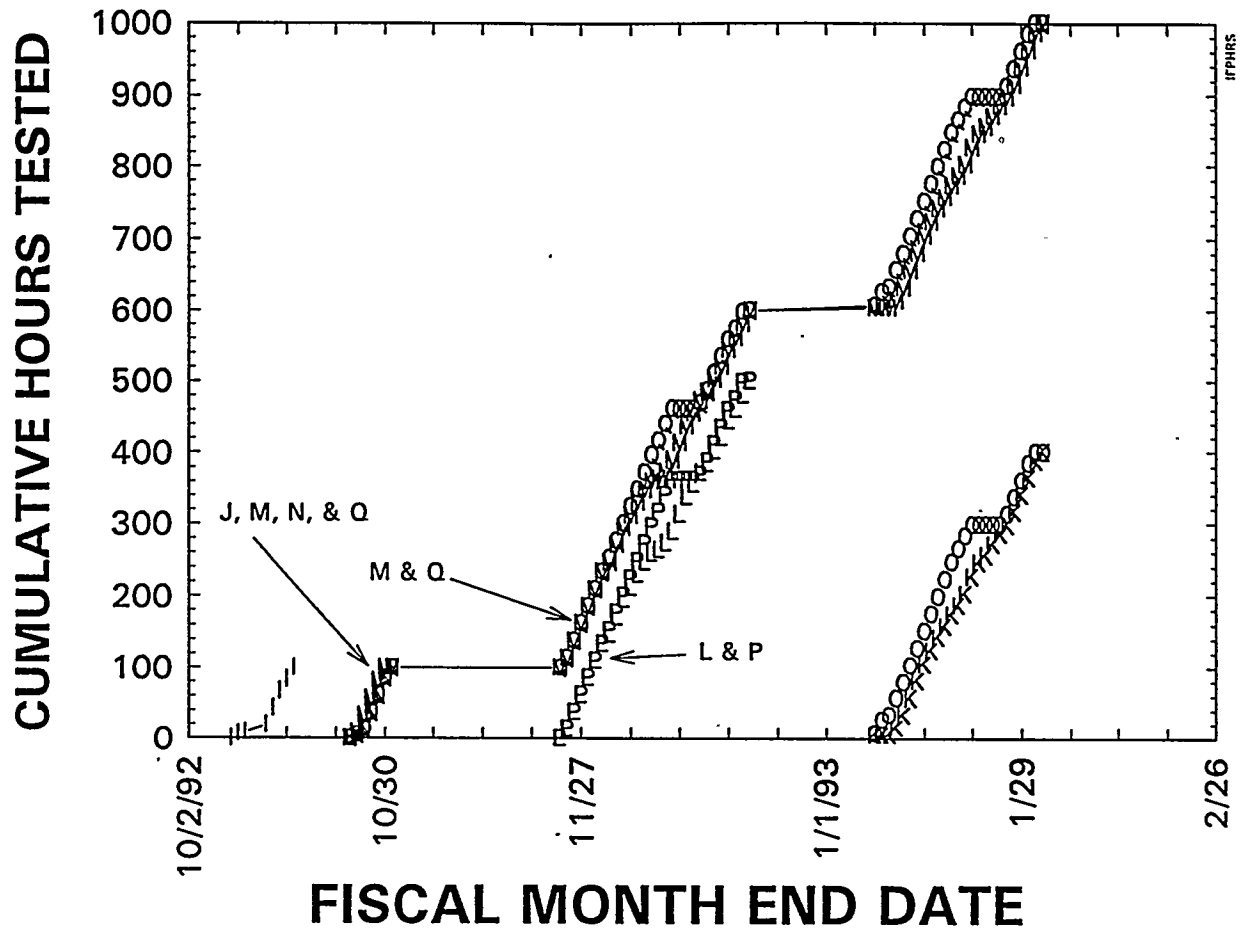
- A 100-hr Test at 870°C, 20% Steam, and 95 ppm NaCl \*
- B 100-hr Test at 980°C, 20% Steam, and 65 ppm NaCl
- C 500-hr Test at 980°C, 20% Steam, and 65 ppm NaCl (475-hr)
- D 1,000-hr Test at 980°C, 20% Steam, and 65 ppm NaCl (575-hr)
- E 100-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- F 400-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- G 500-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- H 1,000-hr Test at 870°C, 15% Steam, and 20 ppm NaCl

### Refractron 50/5 with 505 binder

- ◇ 100-hr Test at 870°C, 15% Steam, and 20 ppm NaCl \*

\* No Pulse Cleaning

**Figure 3-5. Graph of the cumulative thermal exposure system up-time for Schumacher candle specimens**



#### REFRACTRON 70/3 with 442-T binder

- I 100-hr Test at 870°C, 15% Steam, and 20 ppm NaCl\*
- J 100-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- K 400-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- L 500-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- M 1,000-hr Test at 870°C, 15% Steam, and 20 ppm NaCl
- N 100-hr Test at 980°C, 15% Steam, and 20 ppm NaCl
- O 400-hr Test at 980°C, 15% Steam, and 20 ppm NaCl
- P 500-hr Test at 980°C, 15% Steam, and 20 ppm NaCl
- Q 1,000-hr Test at 980°C, 15% Steam, and 20 ppm NaCl

\* No Pulse Cleaning

Figure 3-6. Graph of the cumulative thermal exposure system up-time for Refractron 70/3 with 442-T binder candle specimens

In order to track the operation of the exposure system overnight while still running, a data acquisition system (DAS) was implemented near the end of the exposure testing series. The DAS allowed for an estimate of the time when a problem occurred (such as when a heater failed) in order to adjust the cumulative time of exposure correctly. Figures 3-7 and 3-8 are examples of the temperature data obtained from the data acquisition system of the stainless steel system (B-system) and the Haynes system (C-system) respectively. The B-system is smaller and was therefore harder to keep at a steady temperature (as evident in Figure 3-7). The C-system was larger and therefore had a larger thermal mass to help keep the temperatures steady (see Figure 3-8). From examining the data in Figures 3-7 and 3-8, it appears the gas temperature in the rear of the vessel was lower than the gas temperature in the front of the vessel and inside the candle. The difference in temperature was about 100°C. This would imply that the outside surface of the 1,000-hr candle specimens (which is situated at the rear of both vessels) was not at the desired temperature, but the 100-hr, 400-hr, 500-hr candle specimens, and the inside surfaces of all the candles were at the desired temperatures. A possible cause for the differences in temperatures is that the large flanges at the ends of the vessels were at a much cooler temperature (although they are insulated by fibrous board and blanket insulation) than the body of the vessel because the flanges are not directly heated (only the cylindrical body of the vessel is directly heated by the heaters).

### 3.3.2 C-Ring Testing

The conventional methods of mechanical strength measurement/testing of candles are uniaxial tension or compression tests, burst tests, and three or four point bend tests. Burst tests, which entail the use of a bladder in the bore of the candle, are not easily performed at elevated temperature. Uniaxial tension or compression and flexural tests require the machining of the candles into specimen shapes such that the original exposed surfaces (outer filtration surface and the inner bore surface) will no longer be present. For example, flexural tests require a rectangular bar-shaped specimen that is cut longitudinally from the candle and then machined leaving parallel opposing faces. If one wishes to study the operational performance taking into account the different

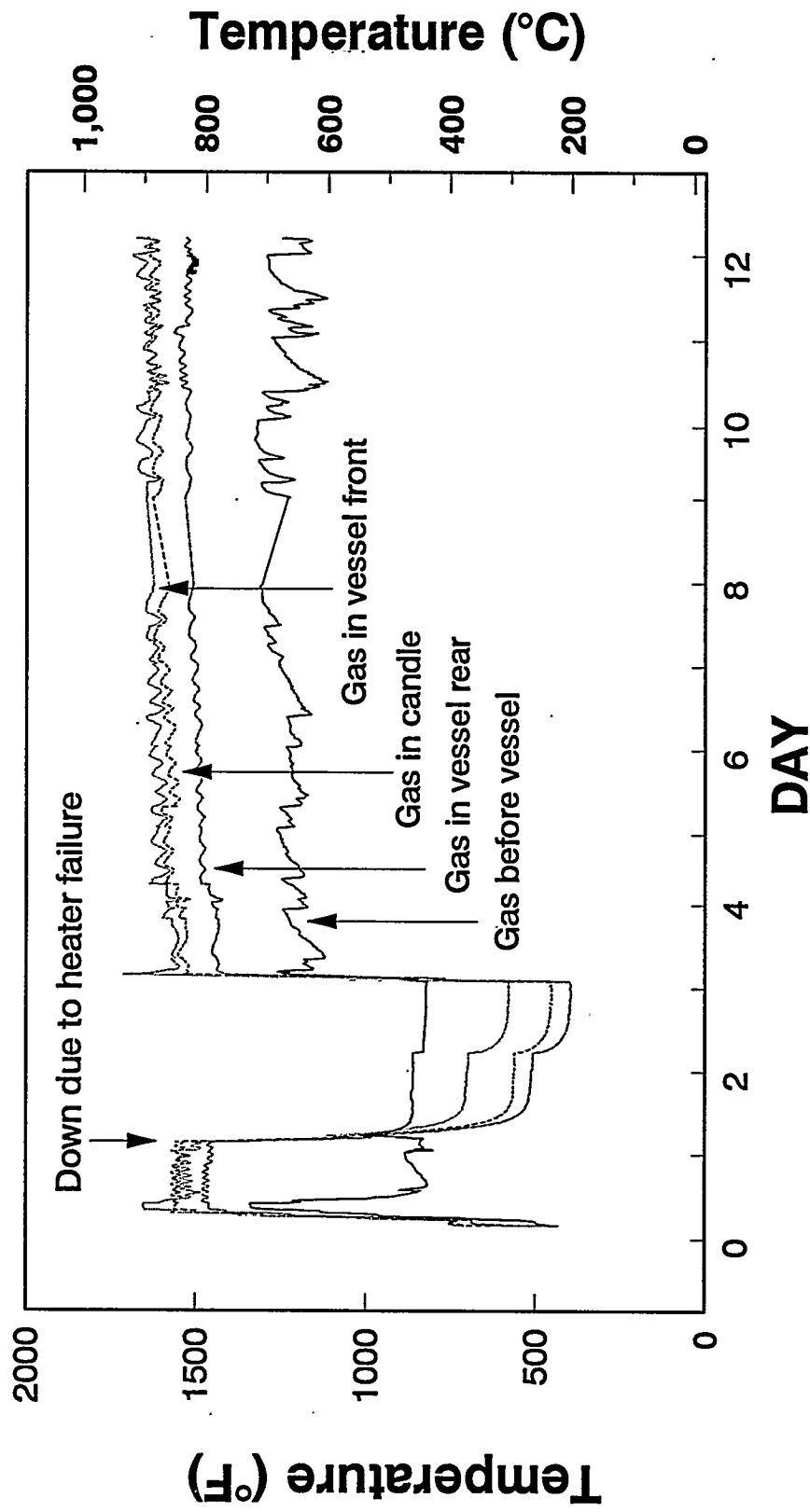


Figure 3-7. Temperature measurements from the data acquisition system of a 870°C exposure test in the stainless steel ("B") system

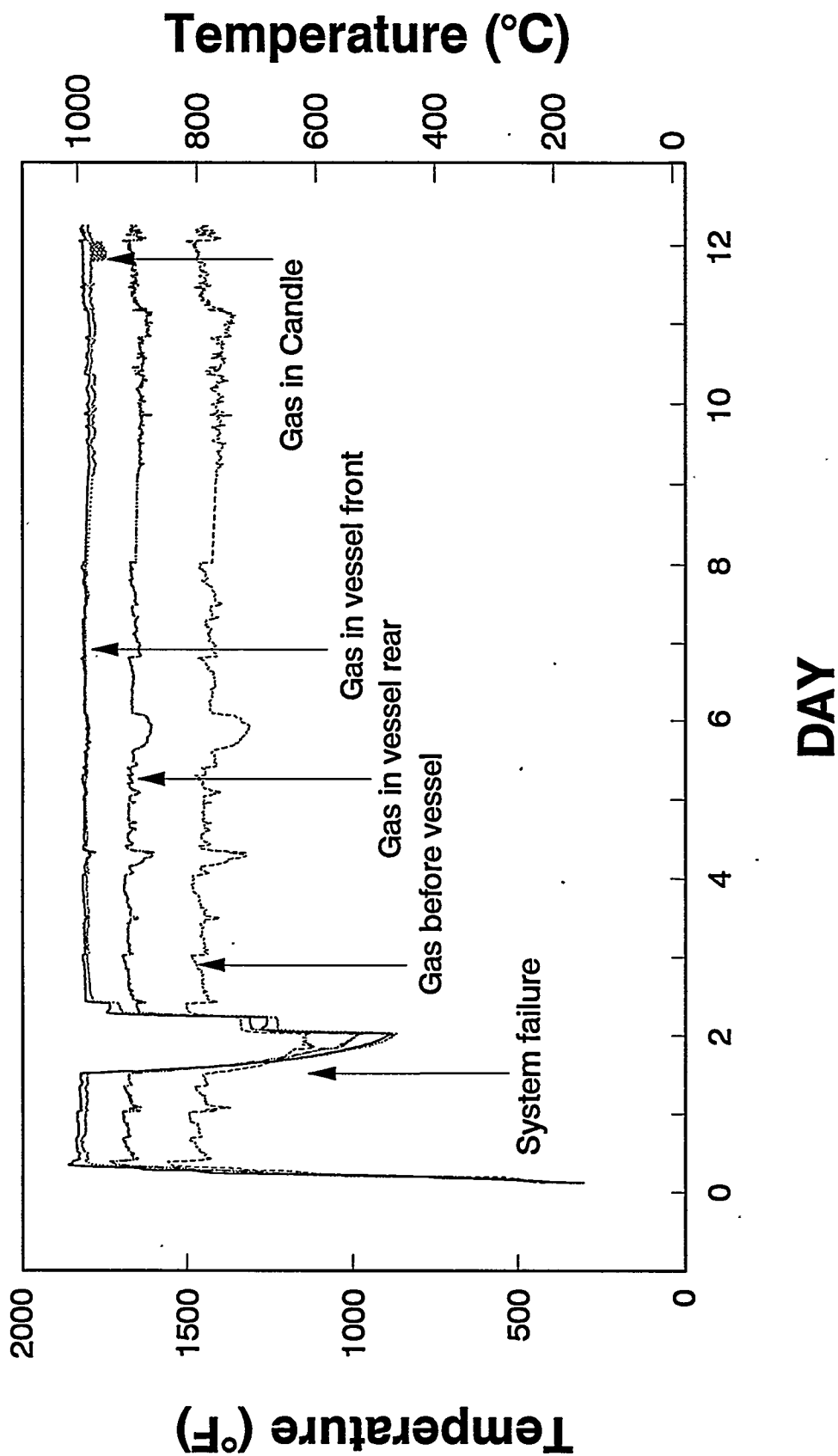


Figure 3-8. Temperature measurements from the data acquisition system of a 980°C exposure test in the Haynes ("C") system



surfaces of the candle that are exposed to environmental conditions, then conventional uniaxial and flexural tests cannot be used.

C-ring testing is the preferred testing method because the specimens retain the surface flaw population. The ability to test the outer surface (via compression testing) or inner surface (via tension testing) independently helps to eliminate variables that may influence the results. The equipment required to perform c-ring testing allows for easy insertion into a clam shell furnace which permits high temperature testing to more closely simulate operational temperatures.

C-ring testing was selected to be the major mechanical strength testing method to be used for this study. C-rings were obtained by sectioning rings out of the candles and then making a longitudinal slit (perpendicular to the sectioning direction) on one side of the ring (see Figure 3-9). C-ring testing is an excellent method because it leaves the surfaces of the candle (inner bore and the outer surface of the tube) intact and therefore the surface flaw population remains intact.

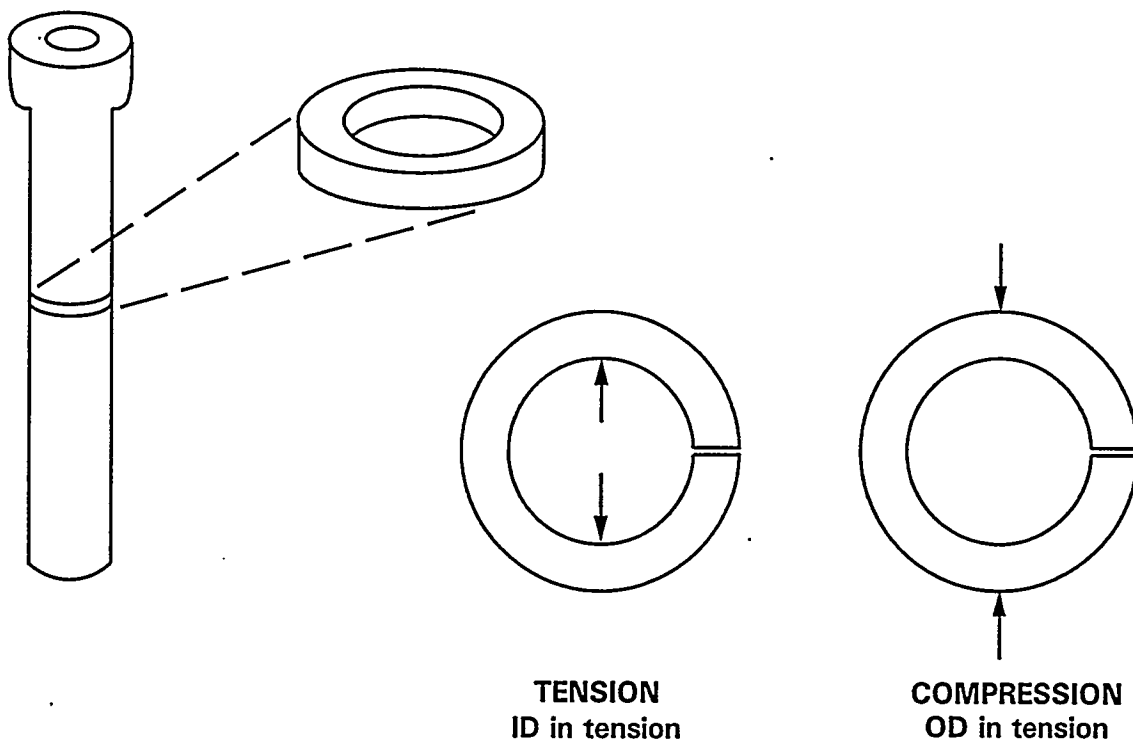


Figure 3-9. C-ring specimens from a ceramic candle

C-ring preparation requires minimal machining, thus reducing the sample preparation costs. The selection of multiple specimens from specific positions of the candles and for specific test conditions is easy to accomplish using the c-ring specimen design.

Figure 3-9 also shows how the c-ring specimens loaded (indicated by the arrows). In tension, the inner surface of the candle at a point opposite the slit experiencing the maximum tensile stress; the outer surface experiencing the maximum compressive stress. In compression, the outside surface (opposite the slit) is bearing the maximum tensile stress. When compared to tensile strength, ceramics are much stronger under compressive stresses, therefore, failure/fracture of the c-ring will originate at the surface with the maximum tensile stress. This phenomenon allows focussing on a surface (inner or outer) to investigate its strength degradation from different environmental conditions. The areas of interest (opposite the slit in the c-ring) are the inner surface for c-ring tension testing ( $R=R_1$ ) and the outer surface for c-ring compression testing ( $R=R_2$ ).

In order to validate the equation used for c-ring maximum stress calculations, Acurex Environmental examined the derivation of the equation (M. Ferber)<sup>3</sup>. Errors were discovered in a 1988 report by National Physical Laboratory (United Kingdom).<sup>4</sup> The equations from M. Ferber, NPL (corrected), and Mark's Standard Handbook for Mechanical Engineers<sup>5</sup> were found to be in agreement. The conclusion is that the correct expression for stress, after being simplified, is

$$\sigma = \frac{F T}{2 W R \left[ \frac{(R_1 + R_2) \ln(R_2/R_1)}{2} - T \right]},$$

Where,

$\sigma$  = maximum stress

F = maximum load

R =  $R_1$  for tensile loading of c-ring (tension tests)

R = R<sub>1</sub> for tensile loading of c-ring (tension tests)

R<sub>2</sub> for compressive loading of c-ring (compression tests)

R<sub>1</sub> = Inside radius of c-ring

R<sub>2</sub> = Outside radius of c-ring

W = width/thickness of c-ring

T = R<sub>2</sub> - R<sub>1</sub>

The above equation is to calculate the maximum tensile stresses endured by the inner or outer surface of the candle (depending upon c-ring tension or compression testing) opposite the slit (see Figure 3-9). The original equation included a "cos Θ" term (see Appendix C) that allowed for the calculation of stresses in other areas of the c-ring specimen between the two points in which the load is applied. The angle Θ is measured from the horizontal opposite the c-ring slit. The maximum stress (for both c-ring tension and compression tests) occurs at the angle Θ of 0°. Any angle other than 0° would result in smaller stress values and would not be at the site in which the c-ring would be experiencing the highest stress. Since the purpose of c-ring testing is to measure the maximum stress experienced by the specimen in order to ascertain the material strength, the angles other than 0° are of no interest.

The standard deviation equation used for each sample set is defined as follows:

$$s = \sqrt{\frac{\sum_{i=1}^N (X_i - \bar{X})^2}{N-1}},$$

Where,

s = standard deviation of a sample set

N = number of specimens in the sample set

X = a specimen value in the sample set

$\bar{X}$  = arithmetic mean of the sample set

The standard deviation gives a measure of how the specimen values in the sample set scatter or spread about the arithmetic mean (average). Note that the denominator has the (N-1) expression instead of the more commonly used (N) expression. The standard deviation equation that uses the (N) expression can be converted into the above equation by multiplying the equation or the resulting value by the square root of  $N/(N-1)$ . For example, the standard deviation for a sample set of 5 which uses the (N) expression would be multiplied by the square root of  $5/(5-1)$  to get the same standard deviation value as the equation given above. This new standard deviation is about 12 percent larger than the commonly used standard deviation. For small sample sets (such as the example given), the standard deviation that uses the (N-1) expression gives a better estimate on the dispersion of the data. The use of the above equation does not affect the arithmetic mean of the sample set. For large sample sets (greater than 30), there is essentially no difference between the two standard deviation definitions.<sup>6</sup>

Acurex Environmental utilizes a Tinius Olsen Four-Screw Electromatic (120,000 lb. max.) universal testing machine for c-ring testing. Compression testing is performed between the weighing table and the lower crosshead. Tension testing is achieved by movement between the upper crosshead and the lower crosshead. The upper crosshead is stationary, but can be adjusted to certain predetermined locations on the vertical support columns. Lower crosshead displacement is the means to apply load to the specimen. A LVDT was used to measure the displacement of the crosshead movement during testing. The load readings are from a Sensotec load cell (model #41/571-06). The Sensotec load cell has an accompanying digital meter that is accurate but has no capability for an output signal to a pen recorder. Some load versus displacement data were obtained from the output signal of the LVDT and the Tinius Olsen's analog dial (that reads load).

The crosshead speed for both tension and compression testing was 0.05 in./min. (1.3 mm/min.).

The universal test system had to be configured to test the ceramic c-ring specimens of this study. Haynes® Alloy 230 was chosen as the material for the c-ring fixtures. Haynes® Alloy 230

is a nickel-chromium-tungsten-molybdenum alloy that meets the required properties for 1,000°C testing. The design of the c-ring tension/compression fixtures for 3/4 inch max. and 1.25 inch max. c-ring widths are in Appendix B.

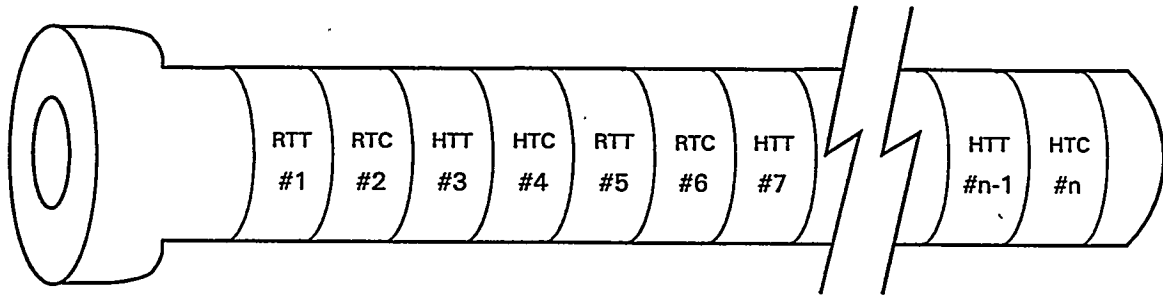
The pins (cylindrical rods) needed to assemble different parts of the c-ring fixture to allow articulation, and contact to the c-ring specimens from the inside surface for tension testing must exhibit limited flexure. Pure (99.8%)  $\text{Al}_2\text{O}_3$  was chosen for this purpose.

To test in the range 650°C to 1,000°C, a single zone vertical clam shell (split tube) furnace (with controller) was used. The ability to open up the furnace and install another specimen after each test was a requirement for expedient testing. A single zone furnace was selected based on the cost. The controller for this furnace allows varying power output that reduces the power output as the set-point temperature is approached. This prevents the overshoot and undershoot cycles characteristic of furnaces that only have on-off capability.

Since the fixtures inside the furnace will reach temperatures of 1,000°C, the ends of the fixtures that extend from the furnace were water cooled to protect the universal testing machine from damage.

The specimens were sectioned with a diamond blade saw and the slit in the c-ring was also made with the same diamond blade.

Specimens used for testing were obtained from each candle, in sequence, depending upon the amount of material available and the types of testing needed to be done. For example, the following sequencing scheme was adopted if there were enough material to test room temperature tension, high temperature tension, room temperature compression, and high temperature compression from one candle. Each candle was sequentially sectioned and each section labeled from 1 to 20, from top to bottom. Specimen #1 would be for room temperature tension, specimen #2 for room temperature compression, #3 for high temperature tension, and #4 for high temperature compression. This sequence would then be repeated starting with specimen #5 (see Figure 3-10). This sorting scheme resulted in a representative sampling of the strength of each



RTT = ROOM TEMPERATURE TENSION TESTING  
RTC = ROOM TEMPERATURE COMPRESSION TESTING  
HTT = HIGH TEMPERATURE TENSION TESTING  
HTC = HIGH TEMPERATURE COMPRESSION TESTING

**Figure 3-10. C-ring testing sequence from candle filter**

candle (even if there were strength variations along the length of the candle) when tested at the respective conditions.

For high temperature testing, a set of five specimens was put into the furnace, one positioned for testing and the other four preheating inside of the furnace. The thermal expansion of the rams during the heating cycle must be considered, requiring re-positioning of the moving crosshead. After the c-ring specimen is positioned between the tension fixtures, an alumina pin is inserted into the bottom fixture/ram (1) to keep the specimen from rolling off the fixture and (2) to allow the furnace to be closed.

For compression testing, the specimen was ready for testing as soon as the furnace stabilized at the target temperature for about 30 minutes. For tension testing, the furnace must be reopened to position the specimen and bottom ram (since they are on the moving crosshead) to allow the second alumina pin to be inserted into the top fixture.

Additional information concerning c-ring testing can be found in Appendix C.

### 3.3.3 XRD Analysis

The interaction between x-rays and the electrons of matter through which it passes will result in the diffraction of the x-rays. A given crystalline substance produces a characteristic diffraction pattern that can be used for identification in a specimen.

For the X-Ray Diffraction (XRD) analyses, a small cross-section of the candle is ground into a fine homogeneous powder. Each particle of the powder is a collection of crystals oriented at random with respect to the incident x-ray. These crystals produce the characteristic diffraction peaks and intensities which can be detected and compared to known standards to identify the crystal phases present. Although amorphous solids by definition have no crystallinity (lack of periodicity), they do have a tendency to "order" in the sense that the atoms are tightly packed together and show a statistical preference for a particular interatomic distance. Hence, the result is an x-ray scattering curve that shows a broad maxima that sometimes look like high background noise.<sup>7</sup>

XRD provides a macroscopic look at the candle to disclose the chemical combination of elements present, not merely the candle's constituent chemical elements as in other analyses. With sufficient presence of a phase or compound, differences in compounds and phase changes can be detected.

### 3.3.4 SEM/EDX Analysis

The scanning electron microscope (SEM) is used to visually examine surfaces at magnifications higher than that achievable by optical microscopes. In order to get an image, the specimen is scanned by an incident electron beam and the radiation that escapes (such as secondary electrons and characteristic x-rays) is collected and processed. Typically the secondary electron image is used for micrographs because of the considerable depth of focus that gives a "three-dimensional" appearance to rough specimens.

An energy dispersive x-ray (EDX) detector is used in conjunction with a SEM to provide quantitative analysis of specific areas being viewed by the SEM. This technique allows determination of the surface elemental composition. The EDX detector used will not determine

elements of atomic weights less than 23 (sodium). Elements such as carbon, oxygen, and nitrogen were therefore not determined.

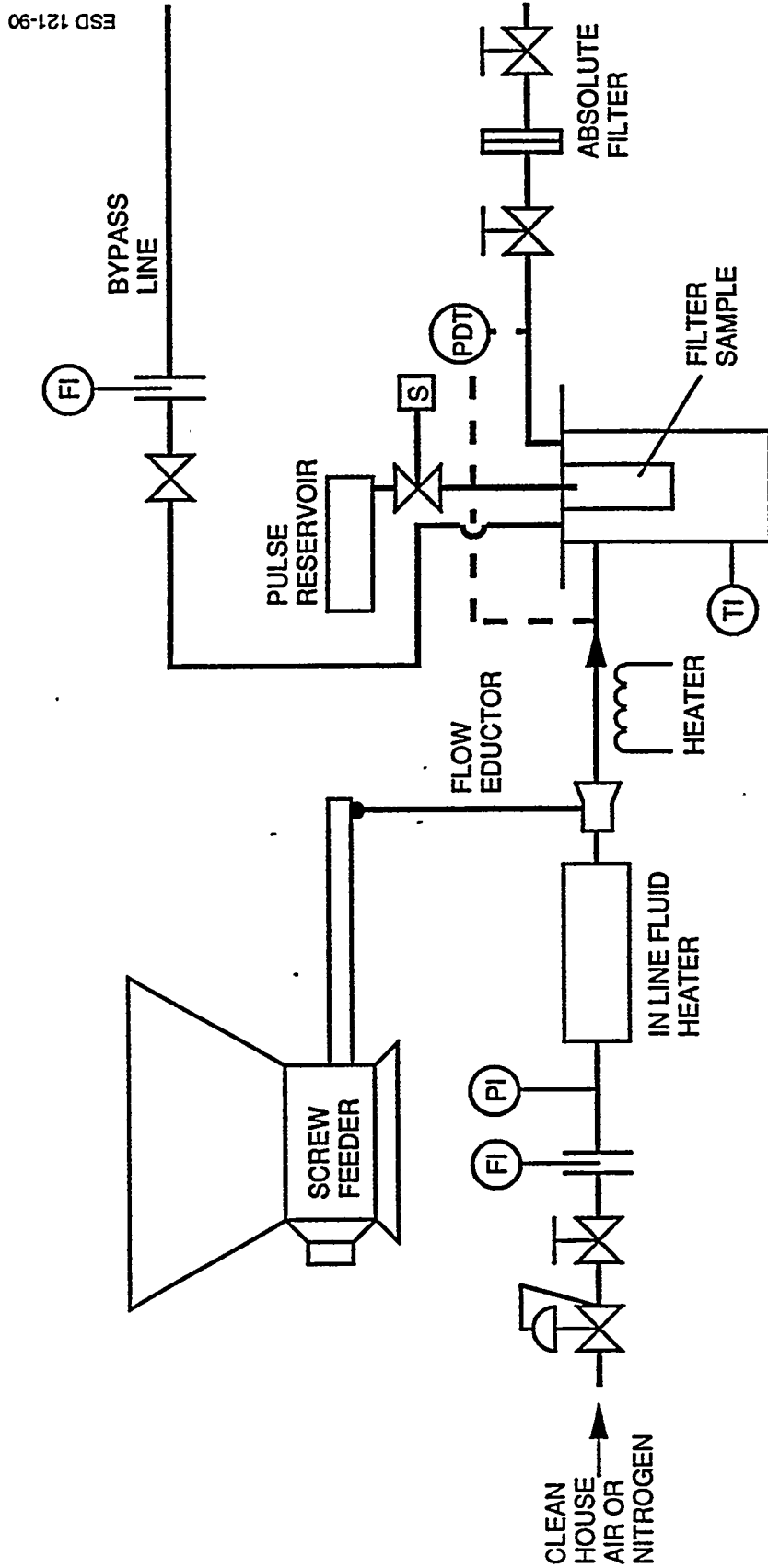
Visual SEM examination of the candle surface morphology provides insight into the binder and grain interaction that produces the filtration properties, as well as the physical changes due to operational conditions. EDX analysis can indicate the presence of elements that may have detrimental effects on the operational properties of the ceramic candles. Therefore, the SEM/EDX analyses performed in this project provided microscopic examination and documentation for visual, as well as elemental comparisons of used and unused candles.

### 3.3.5 Thermal Ratcheting

Currently, all ceramic candle filter systems use pulse cleaning as the filter-cleaning mechanism. Filters are usually pulse cleaned in either of two ways: pressure-triggered or time-triggered cleaning pulses. In either case, however, typical cleaning intervals (time between pulses) can range from 3 minutes to as long as 1 to 2 hours, depending on dust load and dust characteristics. The thermal ratcheting tests is to determine the extent to which pulse cleaning degrades the candle filter material.

Accelerated pulse cleaning (thermal ratcheting) and filter efficiency testing were performed on a bench-scale filter evaluation apparatus operating at 870°C and atmospheric pressure. Figure 3-11 is a schematic of the bench-scale filter test facility. Each of the specimens were exposed to a total of 5,000 pulses with a 30-second time interval between each pulse cleaning. Therefore the total time to complete each accelerated pulse cleaning test was about 42 hours. The pulse gas was at room temperature in the reservoir tank. The pulse gas reservoir was set at 80 psig and was released into the candle specimens (each specimen was 1-foot long) through a time-triggered solenoid valve. The materials exposed to the accelerated pulse cleaning conditions are the Schumacher Diaschumalith F40, IFP LayCer, Didier SiC20, and the IFP Fibrosics candles.





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Figure 3-11. Bench-scale filter test facility

### 3.3.6 Fibrosics Testing

#### 3.3.6.1 Fibrosic Candle

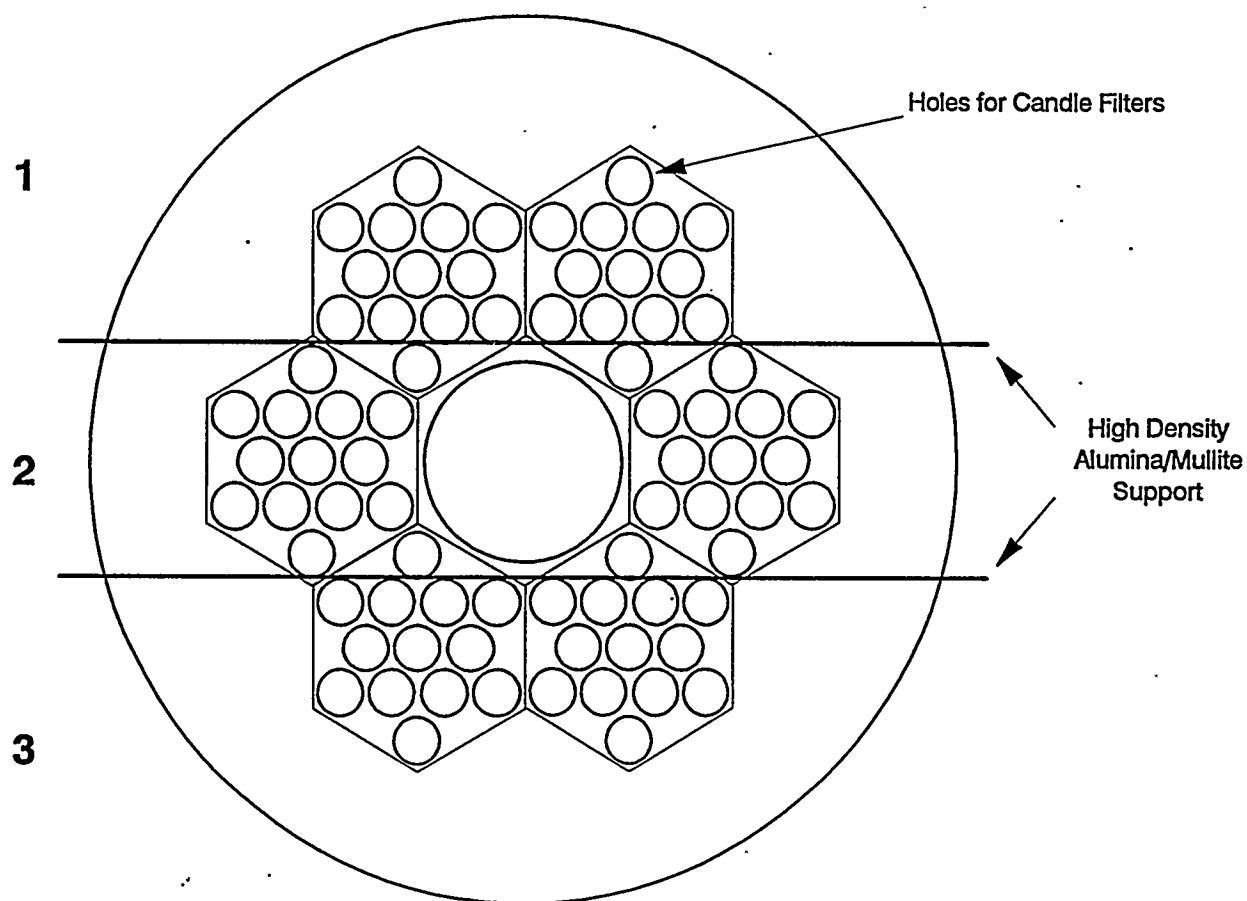
The IFP Fibrosics candle is a new type of candle for which there is little public information available. The bench-scale filter facility was used to perform collection efficiency and permeability testing. The collection efficiency tests were performed at 870°C with a dust loading of 4,000 ppm<sub>w</sub>. The dust used is Curtis Wright PFBC flyash ( $D_{50} = 4.7 \mu\text{m}$ ).

Since the IFP Fibrosic candle is relatively "soft" compared to the rigid candles, c-ring testing could not be performed to measure the strength of the candles. The strength measurements would be at the lower limits of the c-ring testing system. Hence, burst testing would be used to ascertain the strength of the candles. In burst testing, a rubber bladder is inserted into the bore of the candle. The candle and bladder (pre-filled with water) is then fitted into the burst tester before a hydraulic pump is activated to pressurize the bladder/candle assembly. The burst pressure of the candle is the specified pressure at which the candle fails.

#### 3.3.6.2 Fibrosic Tubesheet

IFP had also developed a fibrous ceramic tubesheet and candle hold-down system. This approach has appeal by reducing the amount of metal subjected to high temperature gas. The tubesheet/hold-down assembly had not been subjected to sufficient testing. There was no long term testing data of the structural integrity of the Fibrosic tubesheet at high temperature. This structural loading test entailed the use of a section of the Fibrosic tubesheet to simulate (when possible) the "worst" case structural conditions that the tubesheet can encounter.

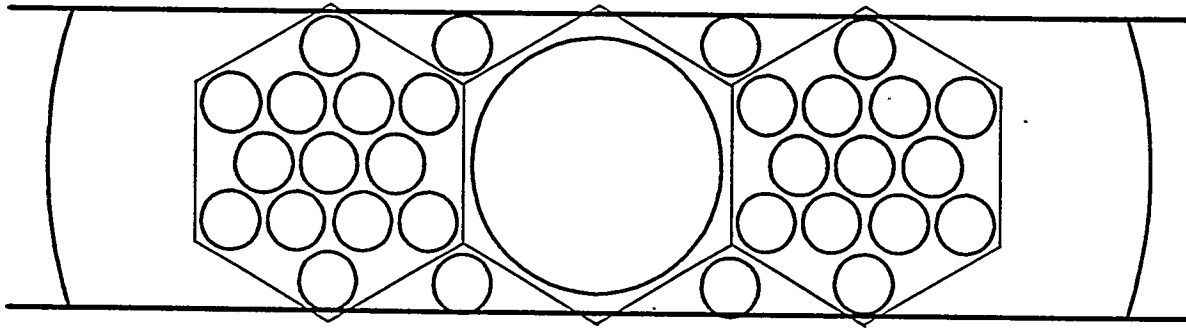
Figure 3-12 shows the Fibrosic tubesheet design that this experiment is based upon. The high density of candle filter holes would represent the least amount of supporting tubesheet material surrounding each candle filter. The two parallel alumina/mullite support design was chosen instead of the four parallel support design because the two support design maximizes the distance between the supports which would allow the tubesheet between the supports to experience the worst case scenario. Only a section of the tubesheet was tested. With the two support design, the tubesheet



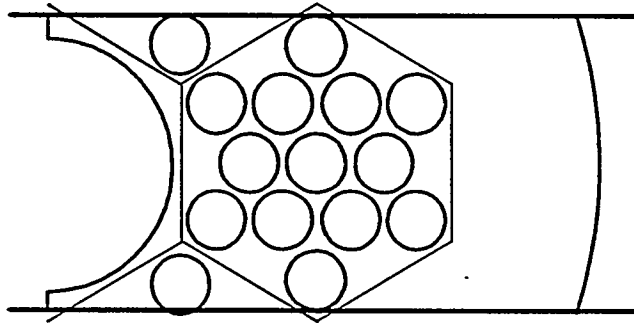
**Figure 3-12. Fibrosic tubesheet**

is divided into three sections denoted on to Figure 3-12. The middle section (labeled as number "2" in Figure 3-12) was selected because the center section has the most holes lying around the center line parallel to the supports. Having a high density of holes around the center line of the tubesheet would have the most effect on weakening the strength of the tubesheet.

Testing of the complete middle section of the tube sheet (see Figure 3-13) is not necessary because bending is expected about the lines in the axial direction parallel to the two supports. Bending of the supports is not expected due to the strength of the high density alumina/mullite. Therefore the only bending will be inwards and downwards between the two supports. Since the big hole in the center of the middle section essentially separates the section into two parts and there is symmetry between the two halves, then only half of the middle section is required to test the integrity of the tubesheet at high temperature. Figure 3-14 shows the tubesheet section procured



**Figure 3-13. Middle section of Fibrosic tubesheet**



**Figure 3-14. Section of Fibrosic tubesheet tested**

from IFP. In the oven, the tubesheet (supported by the two 1.5-inch wide alumina/mullite supports) was placed on four refractory bricks to elevate the tubesheet above the heaters. Each of the four bricks was placed under the one of the four ends of the two supporting members. Since the members were not fixed, the supports were allowed to rotate inwards and downwards between the supports (see Figure 3-15). This is a worst case situation because in reality, the tubesheet on the outside of the supports (sections "1" and "3" which are non existent in this test) would produce an outward and upward force on the middle section of the tubesheet that would reduce the inward and downward forces caused by the weight in the middle section of the tubesheet.

Figure 3-16 is a cross-sectional view of a candle filter hole in the Fibrosic tubesheet. To simulate the loading of the tubesheet encountered during operation, a ceramic disc with the same

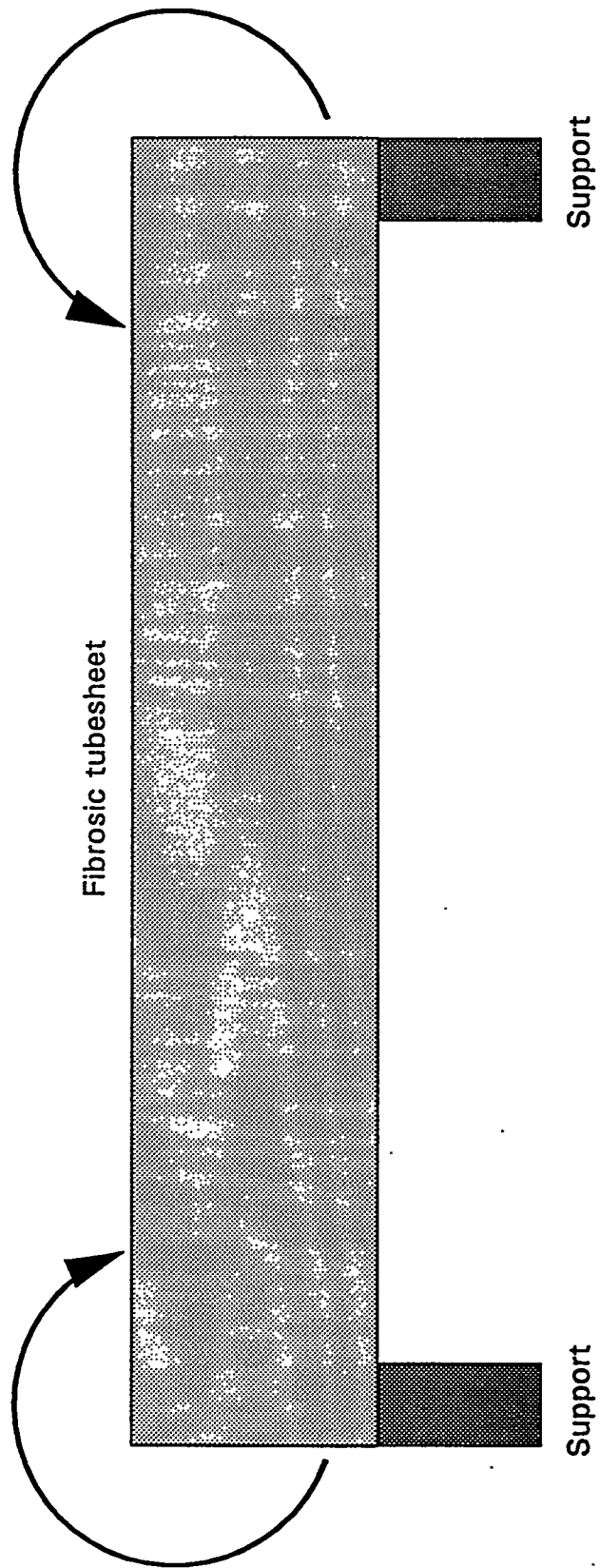


Figure 3-15. End-view of Fibrosic tubesheet tested

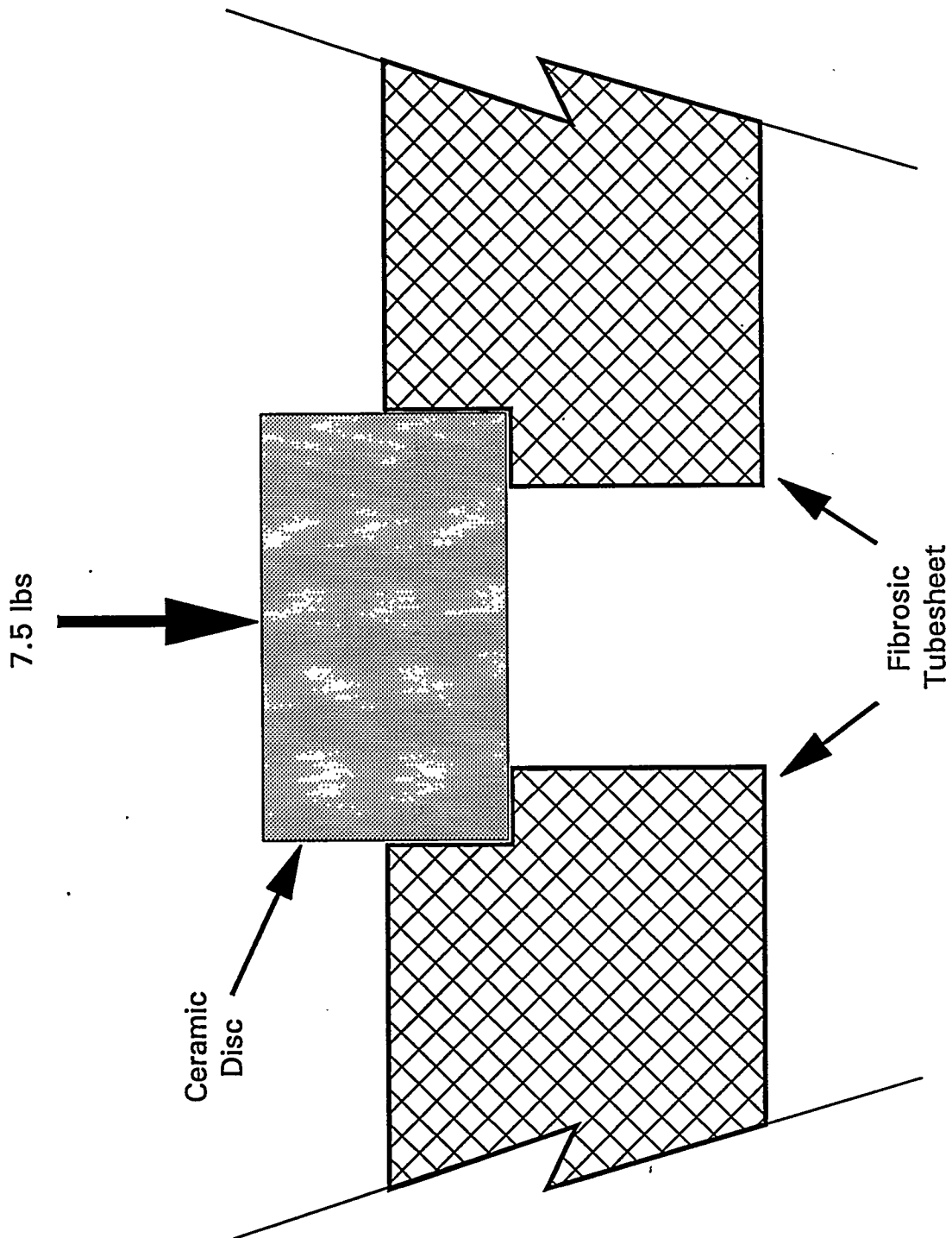


Figure 3-16. Cross-sectional view of candle hole in Fibrosic tubesheet

diameter as the outside diameter of a candle filter flange was inserted into the candle hole of the tubesheet (refer to Figure 3-16). The bottom of the disc was in contact with the same surfaces that the candle filter flange would be in contact with. The thickness of the disc is about double the thickness of the candle flange because the weights to be placed on the disc must not be in contact with the top of the tubesheet. In operation, the weight of the candle and any ash which is deposited on the candle is transmitted to the tubesheet primarily via the bottom surface of the candle flange. The weight of a 1.5 meter long candle filter is about 2.5 pounds. The weight of 7.5 pounds per candle hole is derived from an assumption of a dust dry bulk density of  $15 \text{ lb/ft}^3$  which is in a "bridged" (worst case) condition between the candles (as well as having significant ash penetration into the candle body). The total weight 113 lbs was used to load the tubesheet.<sup>8</sup>

In operation, the filter tubesheet is also supported in addition to the ceramic support beams by a peripheral "ledge" that extends several inches under the circumference of the tubesheet. To simulate the worst case scenario, no supporting "ledge" was used in the load test.

To measure deflection of the tubesheet at  $870^\circ\text{C}$  during the 500 hour test, thin alumina rods resting on the surface of the tubesheet were extended vertically through the top of the oven to allow for measurements of the vertical displacement of the alumina rods above the oven as a function of time.

## SECTION 4

### THERMAL/MECHANICAL TEST RESULTS AND ANALYSIS

#### 4.1 C-RING TESTING

The c-ring tension test results are presented in Tables 4-1, 4-2, and 4-3. The c-ring compression test results are presented in Tables 4-4, 4-5, and 4-6. Tables 4-1 and 4-4 are for the Schumacher Diaschumalith candles, Tables 4-2 and 4-5 show the Refractron candle results, and Tables 4-3 and 4-6 show the c-ring test results of the IFP LayCer and the Didier SiC20 candles. The row headings in the c-ring compression tables (Tables 4-4, 4-5, and 4-6) correspond to the same sequence of row headings in the c-ring tension tables. Hence, there are some rows in the compression tables that may not have any data because only c-ring tension tests were performed. These tables show the average maximum stress values for each individual candle tested (with its corresponding standard deviation). The raw data used for these results are detailed in Appendix D.

##### 4.1.1 C-Ring Testing: Preliminary Investigations

The typical c-ring specimen width is 0.75 inches. Due to the design and size limitations of the thermal exposure test facility, c-ring widths of 0.50 inches were used. To investigate the possibility of specimen size influencing the strength measurements, Acurex Environmental performed c-ring tension and compression tests on 0.50-, 0.75-, 1.00-, and 1.25-in. specimens on Schumacher and Refractron candles. The data obtained from these tests are summarized in Table 4-7. Figures 4-1, 4-2, 4-3, and 4-4 graphically displays the data presented in Table 4-7. The raw data can be found in Appendix D.

The above graphs demonstrate that the maximum stress values derived from 0.5-in c-ring specimens are representative of the strength of the candle. Although the average maximum stress



Table 4-1. C-ring tension testing: Schumacher Diaschumalith

Less than 5 specimens available for testing									
CANDLE DESCRIPTION	C-ring Width (inches)	C-RING TENSION TESTING							
		Room Temp		650°C		870°C		980°C	
		Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev
SCHUMACHER DIASCHUMALITH									
F30 CANDLE									
UNUSED									
For KRW test (7068)	3/4"	24.29	1.00	25.02	1.29				
USED/EXPOSED									
KRW (7052)	3/4"	18.21	1.38	19.60	0.99				
KRW (7067)	3/4"	18.10	0.83	21.27	0.49				
F40 CANDLE									
UNUSED									
SOLAR	3/4"	5.17	0.55					4.43	0.24
SOLAR	3/4"					6.74	0.69	4.60	1.75
SOLAR	1/2"					6.11	0.36	4.32	0.46
For 5000 pulse test	3/4"	8.23	0.59			8.60	0.62		
EXPOSURE TESTING									
Control: 500 hrs at 870°C only	1/2"					7.92	1.01		
95ppm NaCl/20%H <sub>2</sub> O/870°C									
100hr/No Pulse (A)	1/2"					4.85	0.33		
65ppm NaCl/20%H <sub>2</sub> O/980°C									
100hr/Pulse (B)	1/2"							1.79	0.73
475hr/Pulse (C)	1/2"							2.69	0.45
575hr/Pulse (D)	1/2"							2.51	0.25
20ppm NaCl/15%H <sub>2</sub> O/870°C									
100hr/Pulse (E)	1/2"					5.72	0.39		
400hr/Pulse (F)	1/2"					4.76	2.14		
500hr/Pulse (G)	1/2"					8.43	0.54		
1000hr/Pulse (H)	1/2"					4.64	0.01		
1/3									

Table 4-2. C-ring tension testing: Refractron

Less than 5 specimens available for testing									
CANDLE DESCRIPTION	C-ring Width (inches)	C-RING TENSION TESTING							
		Room Temp		650°C		870°C		980°C	
		Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev
REFRACTRON									
50/5 WITH 505 BINDER CANDLE									
UNUSED									
Varying width test 0.50": #8	1/2"	14.47	1.57						
SOLAR/Vary width 0.75": #8	3/4"	15.25	1.34					19.33	1.70
Varying width test 1.00": #8	1"	15.46	1.09						
Varying width test 1.25": #8	1-1/4"	15.97	1.10						
Heat to 980°C, Post-cut: #8	1/2"	12.91	0.51						
Precut, Refrac heat 1200°C: #8	1/2"	20.02	1.11						
SOLAR: #9	3/4"	16.18	0.41					15.64	1.98
SOLAR: #11	3/4"	15.92	0.68					17.46	1.08
Heat to 1200°C, Post-cut: #11	3/4"	12.52	1.56					18.73	0.80
Pre-cut, heat To 1200°C: #11	3/4"	11.68	1.60						
EXPOSURE TESTING									
20ppm NaCl/15%H2O/870°C									
100hr/Pulse (♦)	1/2"					13.72	1.43		
70/3 w/442-T BINDER CANDLE									
UNUSED									
Varying width test 0.50"	1/2"	16.97	1.47			16.21	1.27	10.91	1.28
Varying width test 0.75"	3/4"	17.08	2.32			13.15	1.70	9.82	1.91
EXPOSURE TESTING									
20ppm NaCl/15%H2O/870°C									
100hr/No Pulse/Front (I)	1/2"					15.96	0.15		
100hr/No Pulse/Rear (I)	1/2"					16.61	4.35		
100hr/Pulse (J)	1/2"					13.85	0.87		
400hr/Pulse (K)	1/2"					11.97	1.65		
500hr/Pulse (L)	1/2"					15.36	1.40		
1000hr/ Pulse (M)	1/2"					18.96	1.51		
20ppm NaCl/15%H2O/980°C									
100 hr/Pulse (N)	1/2"							8.52	0.64
400hr/Pulse (O)	1/2"							2.28	0.67
500hr/Pulse (P)	1/2"							1.68	0.17
1000hr/Pulse (Q)	1/2"							2.32	0.64
2/3									

**Table 4-3. C-ring tension testing: IFP LayCer and Didier SiC20**

Less than 5 specimens available for testing

[illegible]

Table 4-4. C-ring compression testing: Schumacher Diaschumalith

Less than 5 specimens available for testing

CANDLE DESCRIPTION	C-ring Width (inches)	C-RING COMPRESSION TESTING							
		Room Temp		650°C		870°C		980°C	
		Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev	Stress ± Std (MPa)	Dev
<b>SCHUMACHER DIASCHUMALITH</b>									
<b>F30 CANDLE</b>									
UNUSED									
For KRW test (7068)	3/4"	21.16	0.96	25.38	1.84				
USED/EXPOSED									
KRW (7052)	3/4"	17.58	2.18	18.89	1.48				
KRW (7067)	3/4"	20.81	1.08	21.36	0.79				
<b>F40 CANDLE</b>									
UNUSED									
SOLAR	3/4"	4.59	0.31					3.56	0.23
SOLAR	3/4"					5.68	0.40	3.44	0.35
SOLAR	1/2"					4.36	1.10	3.28	0.43
For 5000 pulse test	3/4"								
EXPOSURE TESTING									
Control: 500 hrs at 870°C only	1/2"					4.49	1.07		
95ppm NaCl/20%H <sub>2</sub> O/870°C									
100hr/No Pulse (A)	1/2"					3.93	0.59		
65ppm NaCl/20%H <sub>2</sub> O/980°C									
100hr/Pulse (B)	1/2"							1.29	0.21
475hr/Pulse (C)	1/2"							2.98	1.12
575hr/Pulse (D)	1/2"							1.46	0.30
20ppm NaCl/15%H <sub>2</sub> O/870°C									
100hr/Pulse (E)	1/2"					4.07	0.97		
400hr/Pulse (F)	1/2"					4.22	0.53		
500hr/Pulse (G)	1/2"					5.01	0.79		
1000hr/Pulse (H)	1/2"					3.37	0.26		
1/3									

Table 4-5. C-ring compression testing: Refractron

Less than 5 specimens available for testing									
CANDLE DESCRIPTION	C-ring Width (inches)	C-RING COMPRESSION TESTING							
		Room Temp		650°C		870°C		980°C	
		Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev	Stress (MPa)	± Std Dev
REFRACTRON									
50/5 WITH 505 BINDER CANDLE									
UNUSED									
Varying width test 0.50": #8	1/2"	19.20	0.64						
SOLAR/Vary width 0.75": #8	3/4"	18.12	2.10					16.82	1.36
Varying width test 1.00": #8	1"	17.87	1.69						
Varying width test 1.25": #8	1-1/4"	18.69	1.85						
Heat to 980°C, Post-cut: #8	1/2"	16.73	1.23						
Precut, Refrac heat 1200°C: #8	1/2"								
SOLAR: #9	3/4"	21.83	1.89					12.45	1.92
SOLAR: #11	3/4"	18.33	2.13					15.06	0.72
Heat to 1200°C, Post-cut: #11	3/4"	15.46	2.24						
Pre-cut, heat To 1200°C: #11	3/4"	13.20	0.81						
EXPOSURE TESTING									
20ppm NaCl/15%H <sub>2</sub> O/870°C									
100hr/Pulse (♦)	1/2"					13.85	0.41		
70/3 w/442-T BINDER CANDLE									
UNUSED									
Varying width test 0.50"	1/2"	19.42	1.50			17.94	0.90	8.61	1.05
Varying width test 0.75"	3/4"	17.86	1.52			17.49	2.27	8.19	0.95
EXPOSURE TESTING									
20ppm NaCl/15%H <sub>2</sub> O/870°C									
100hr/No Pulse/Front (I)	1/2"					16.64	1.77		
100hr/No Pulse/Rear (I)	1/2"					16.36	0.72		
100hr/Pulse (J)	1/2"					16.60	1.13		
400hr/Pulse (K)	1/2"					12.14	0.68		
500hr/Pulse (L)	1/2"					9.10	1.45		
1000hr/ Pulse (M)	1/2"					12.20	2.16		
20ppm NaCl/15%H <sub>2</sub> O/980°C									
100 hr/Pulse (N)	1/2"							5.76	0.67
400hr/Pulse (O)	1/2"							1.72	0.34
500hr/Pulse (P)	1/2"							1.17	0.20
1000hr/Pulse (Q)	1/2"							2.31	0.93
2/3									

**Table 4-6. C-ring compression testing: IFP LayCer/Didier SiC20**

**Less than 5 specimens available for testing**

[illegible]

**Table 4-7. Summary of c-ring tension/compression tests investigating the effect of sample width on c-ring strength measurement**

	C-Ring Width (inch)	Temperature (°C)	Tension		Compression	
			Stress (Mpa)	± Std Dev	Stress (Mpa)	± Std Dev
Schumacher Diaschumalith F40	0.50	870	6.11	0.36	4.36	1.10
	0.75	870	6.74	0.69	5.68	0.40
	0.50	980	4.32	0.46	3.28	0.43
	0.75	980	4.60	1.75	3.44	0.35
Refractron 50/5 with 505 Binder	0.50	Room	14.47	1.57	19.20	0.64
	0.75	Room	15.25	1.34	18.12	2.10
	1.00	Room	15.46	1.09	17.87	1.69
	1.25	Room	15.97	1.10	18.69	1.85
Refractron 70/3 with 442-T Binder	0.50	Room	16.97	1.47	19.42	1.50
	0.75	Room	17.08	2.32	17.86	1.52
	0.50	870	16.21	1.27	17.94	0.90
	0.75	870	13.15	1.70	17.49	2.27
	0.50	980	10.91	1.28	8.61	1.05
	0.75	980	9.82	1.91	8.19	0.95

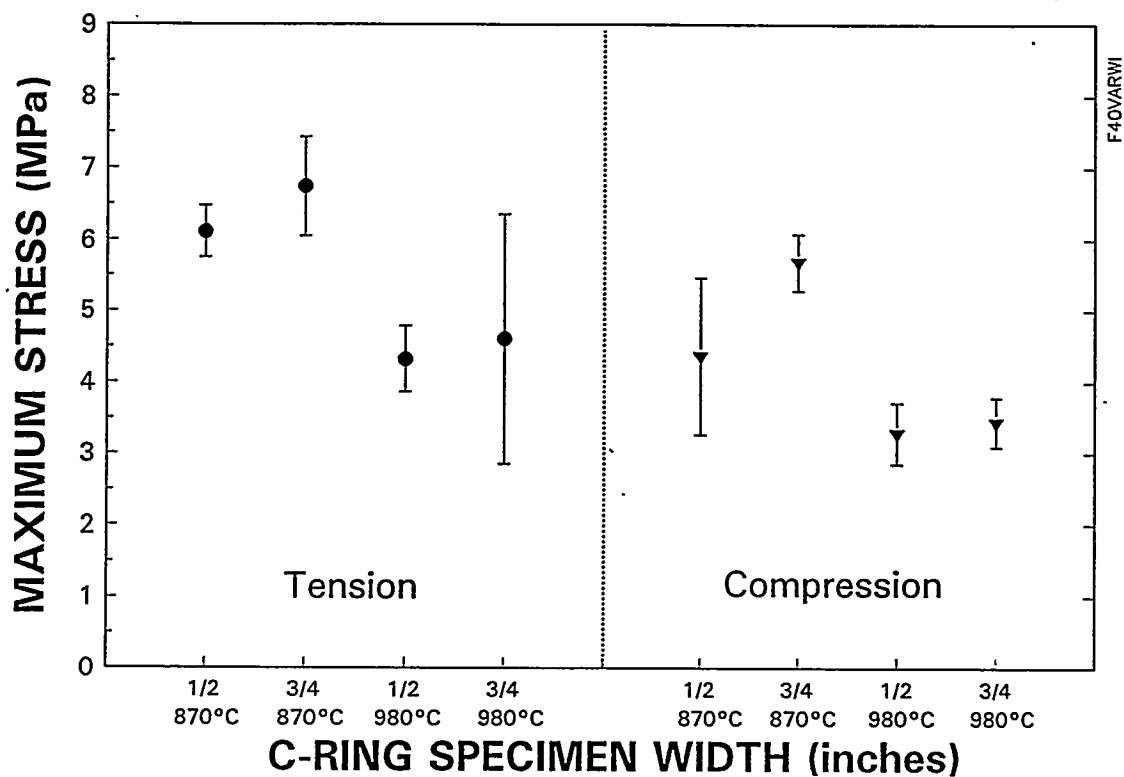


Figure 4-1. Maximum stress comparisons of 0.5-inch with 0.75-inch c-ring specimens for a Schumacher Diaschumalith F40 candle

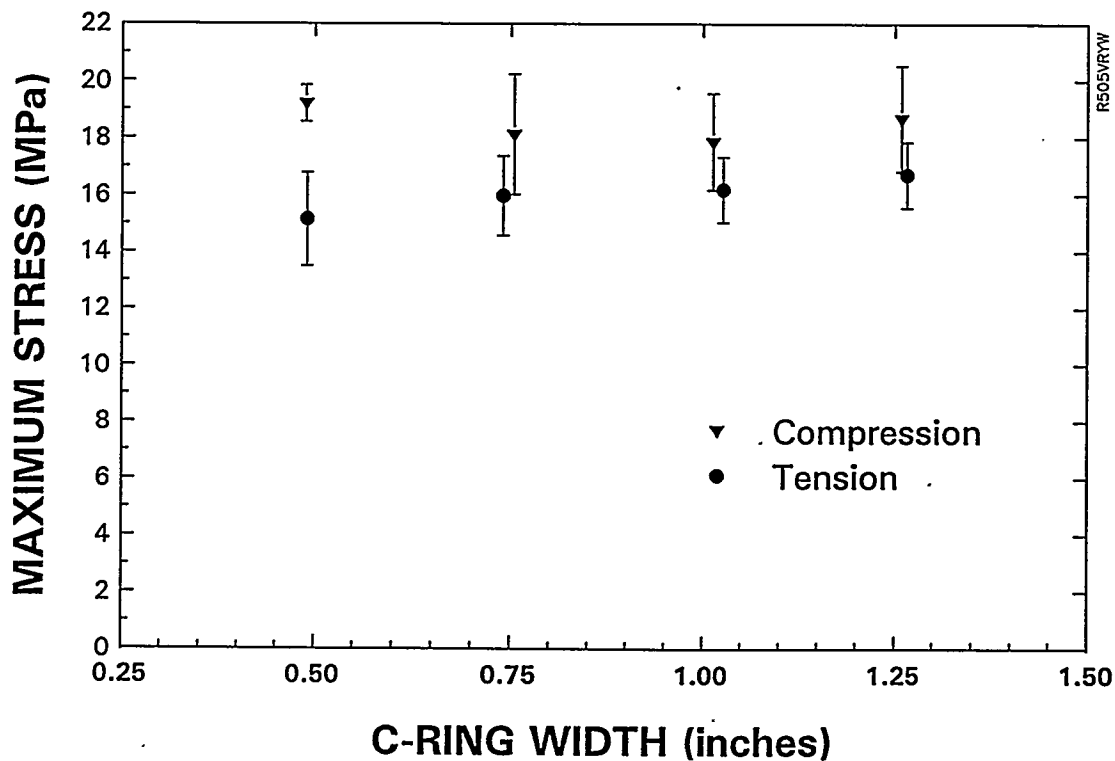


Figure 4-2. Maximum stress values for a Refractron 50/5 with 505 binder candle of various c-ring widths



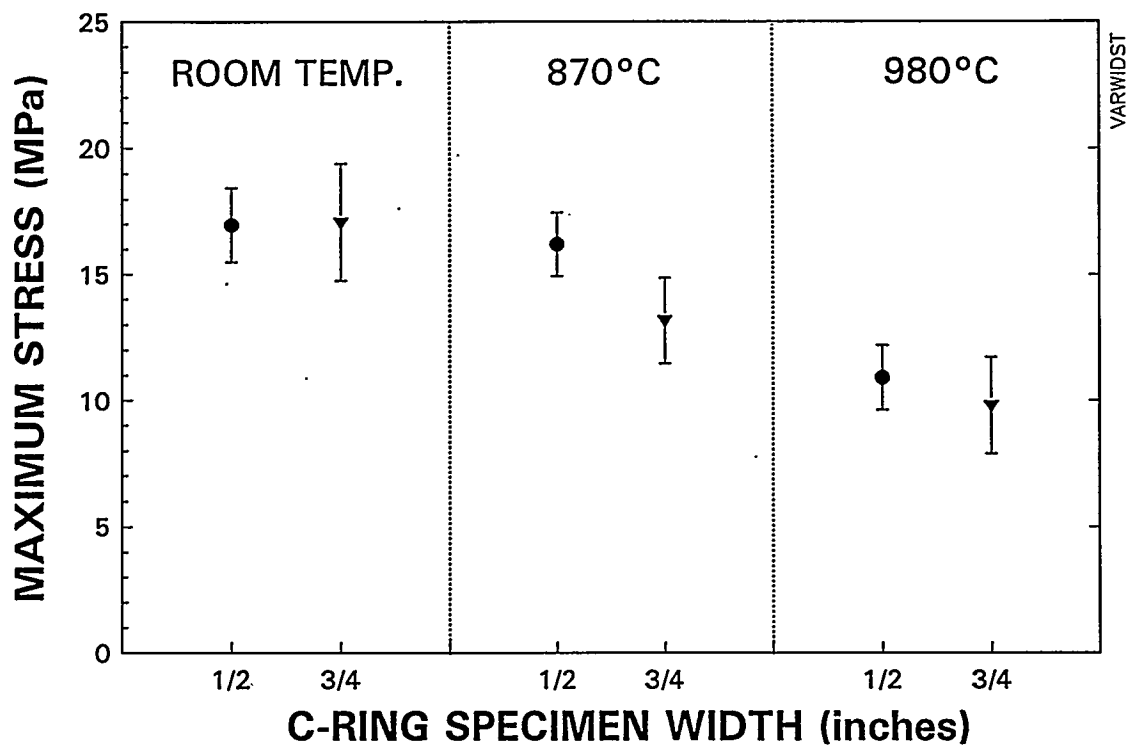


Figure 4-3. Maximum stress comparisons of 0.5-inch with 0.75-inch c-ring tension specimens for a Refractron 70/3 with 442-T binder candle

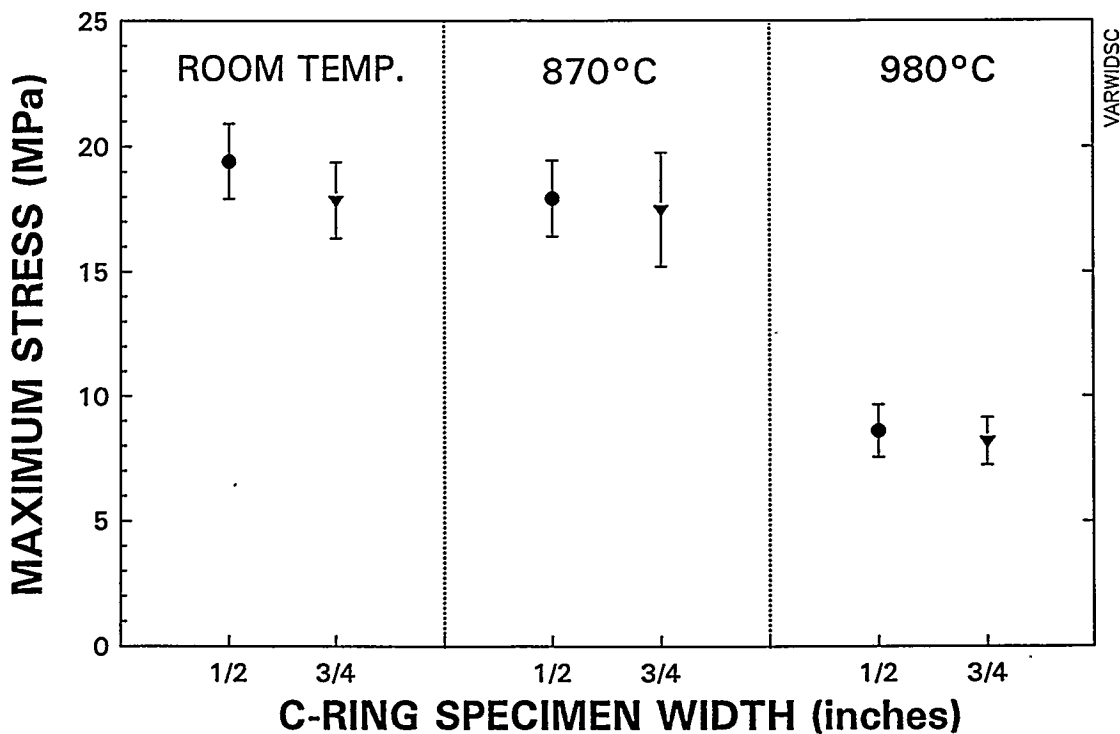


Figure 4-4. Maximum Stress comparisons of 0.5-inch with 0.75-inch c-ring compression specimens for a Refractron 70/3 with 442-T binder candle

values in Figure 4-2 fall within the standard deviations for each respective test (tension and compression), the average maximum stress values seem to be more stable with widths greater than or equal to 0.75 inch. The optimum c-ring width would be 0.75-in because it would be an inefficient use of material to use c-ring widths greater than 0.75-in if the 0.75-in specimens would result in the same maximum stress value. 0.5-in specimens are acceptable, but they are not desirable because they are more fragile than the larger 0.75-in specimens. 0.5-in specimens are also harder to work with at high temperature (during specimen set-up). The slight variability in the 0.5-in c-ring specimen widths (compared to the larger c-ring widths) may be attributed to the following:

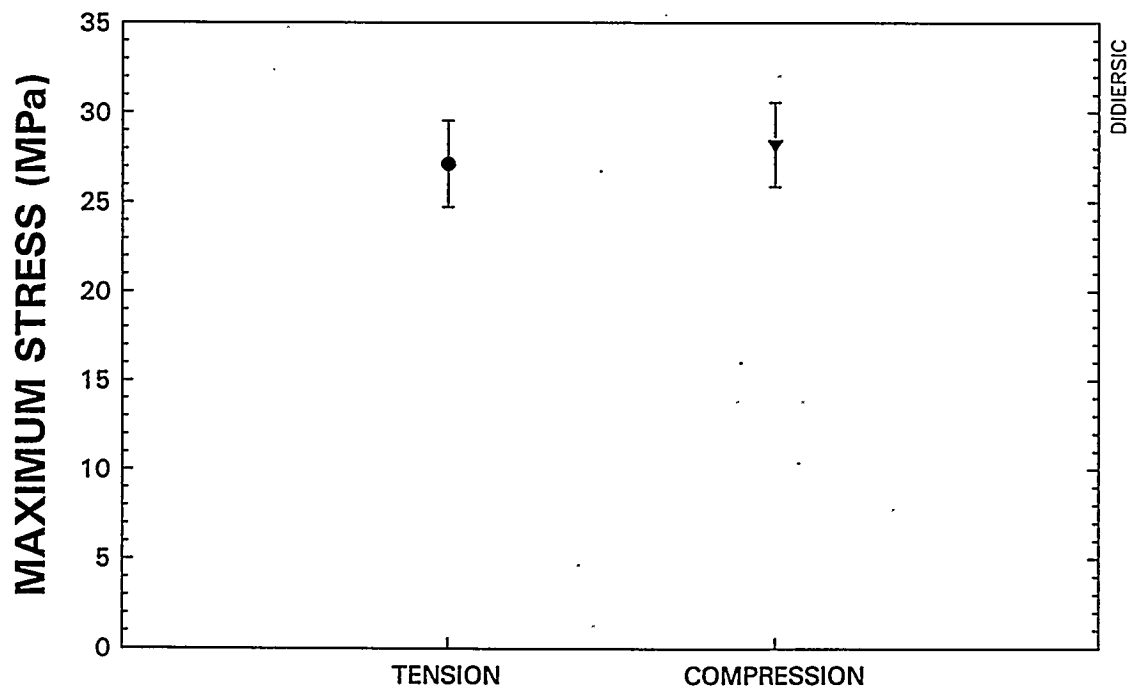
- Edge/surface defects, possibly caused during sample preparation, could have lowered the strength measurements because with a smaller sample volume, the edges would have a greater influence on the sample strength than on a bigger sample volume. This would explain the sometimes lower 0.5-in strength values.
- Smaller samples might have a lower failure probability and hence will result in higher maximum stress values. This is because smaller samples are less likely to have strength-lowering flaws contained in the smaller sample volume than larger samples. This would explain the sometimes higher 0.5-in strength values.

The above two strength affecting mechanisms are in conflict with each other and would result in the slight variability of 0.5-in specimens seen in Figures 4-1 through 4-4.

To investigate the cause of the discrepancy between tension strength and compression strength values due to candle geometry/constituent and not due to the stress equation being used, Acurex Environmental tested an unused homogenous candle made by Didier. The homogenous candle does not have a filtration layer that is different from the supporting body such as the Schumacher and Refractron candles. The Didier candle should have the same strength in c-ring tension as in c-ring compression because the material on the inner surface is the same as on the outer surface. Table 4-8 shows the results from c-ring tests at room temperature (raw data in Appendix D). Figure 4-5 shows graphically that the strength of the material at the inner bore of

**Table 4-8. Maximum stress values for a Didier SiC candle**

	C-Ring Width (inch)	Temperature (°C)	Tension		Compression	
			Stress (Mpa)	±Std Dev	Stress (Mpa)	±Std Dev
Didier SiC	0.75	room	27.13	2.41	28.22	2.35



**Figure 4-5. Maximum stress comparison of c-ring tension testing with c-ring compression testing for a Didier SiC candle**

the candle (via the tension test) has the same material strength as the material at outer filtration surface (via the compression test). This verifies the use of the stress equation for determining the strength of a material irrespective of the surface being examined.

The higher compression stresses compared to tension (at room temperature), for the Refractron candles are due to the candle design — an outer filtration layer that is denser than the inner supporting body of the candle filter. The denser material will result in a higher strength measurement at room temperature. The lower compression strength at elevated temperatures could be due to the denser material having a lower softening temperature than the less dense inner supporting candle material.

There had been some concern about the machining of the c-rings with a diamond blade saw. Machining processes may be a major contributor to the reliability of ceramic materials. Especially for very dense fine-grained ceramics where flaws created by machining are larger than pre-existing flaws in the material. However, candle filters by nature, are highly porous materials. Coarse-grained materials with uniformly distributed porosity are not affected to any great degree because the size of the flaws created by machining are comparable to or smaller than the existing flaws in the material.<sup>9</sup> Therefore, to insure that machining does not affect the strength measurements, c-rings were obtained by two different methods as follows:

- Postcut: A section of candle is heated to 1,200°C (2,200°F) for two hours and slowly cooled to room temperature before the candle is cut into 10 c-rings (5 for tension and 5 for compression testing)
- Precut: A section of candle (from the same candle as the postcut case) is cut into 10 c-rings and then heated to 1,200°C (the precut specimens were heat treated at the same time as the postcut specimens).

Chad Sheckler, Product Development Manager of Refractron Technologies Corporation, specified the temperature of 1,200°C as appropriate to anneal machined c-rings. If machining does produce strength affecting flaws, then the postcut c-rings (which would retain any built up stresses from

machining) should be noticeably weaker than the precut c-rings (which would have had some of the internal stresses relieved via the annealing process). Table 4-9 and Figure 4-6 displays the results of this test (raw data in Appendix D).

It can be seen that there is essentially no difference between the precut and postcut c-rings. Hence, machining during c-ring preparation does not significantly affect the strength measurement of the c-rings.

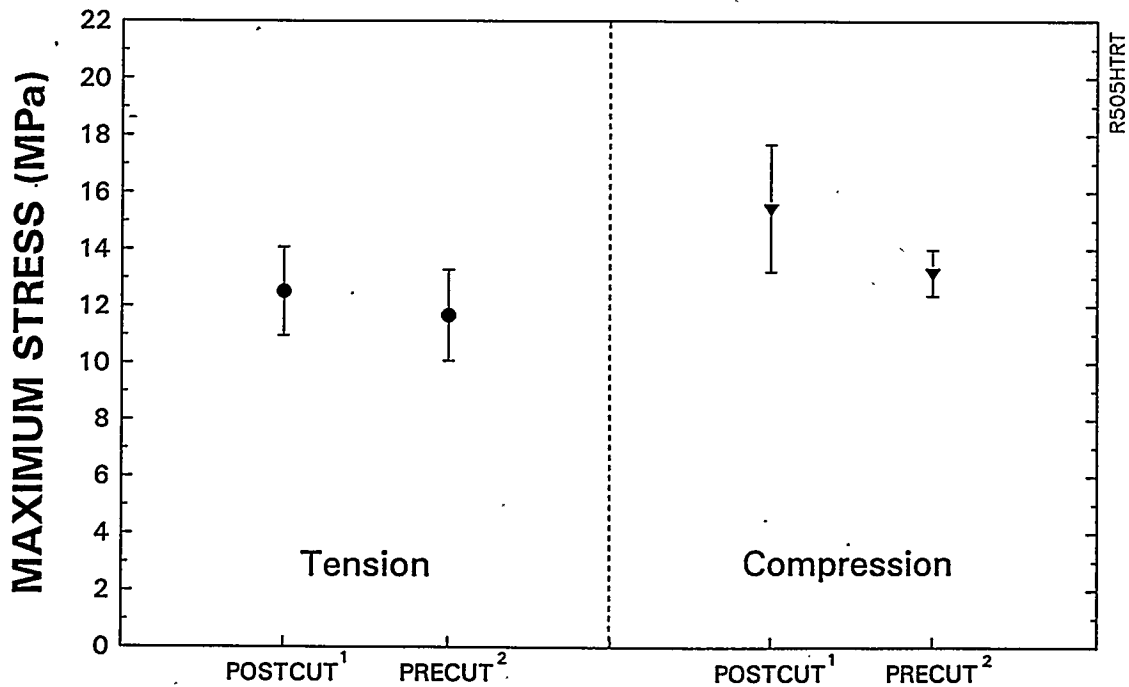
#### **4.1.2 C-Ring Testing: Thermal Ratcheting**

One of the possible causes of the durability problems encountered by candle filters is thermal shock/thermal fatigue. It has been observed that candles exposed to operating conditions resulted in loss of strength. Acurex Environmental investigated the effect on strength that ambient air pulses have on candles at an elevated temperature. Table 4-10 and Figures 4-7, 4-8, and 4-9 summarizes the tests conducted. The pulsed candles were pulsed 5,000 times with ambient air down the inner bore of the candle while the candle was at 870°C. Hence, no c-ring compression tests were conducted because if there were to be any strength degradation due to the thermal shock from the ambient air being blown down the inner bore of the candle, then it would be most apparent in tests examining the inner surface of the candle (c-ring tension).

The unused candles have no history of pulse cleaning nor of temperature cycling. It can be seen that the Schumacher and IFP LayCer candles pulsed with time durations of 0.5 seconds exhibit room temperature strength loss. The strength loss due to these pulses can be attributed to thermal shock in which temperature gradients produce stresses under conditions that restrain the free expansion of a body (candle). The temperature distribution could be such that free expansion of each volume element would separate the individual volume elements in such a way that they would not fit together. If the maximum stress for a material is reached, failure (such as cracking and spalling) will result. The effect of thermal stresses depends not only on the absolute level of stresses, but also the stress distribution, stress duration, and material characteristics (such as ductility, homogeneity, porosity, and presence of pre-existing flaws).<sup>10</sup>

**Table 4-9. Maximum stress values for an unused Refractron 50/5 with 505 binder that had c-rings cut before and after heat treatment at 1,200°C**

	C-Ring Width (inch)	Temperature (°C)	Tension		Compression	
			Stress (Mpa)	±Std Dev	Stress (Mpa)	±Std Dev
Postcut	0.75	room	12.52	1.56	15.46	2.24
Precut	0.75	room	11.68	1.60	13.20	0.81



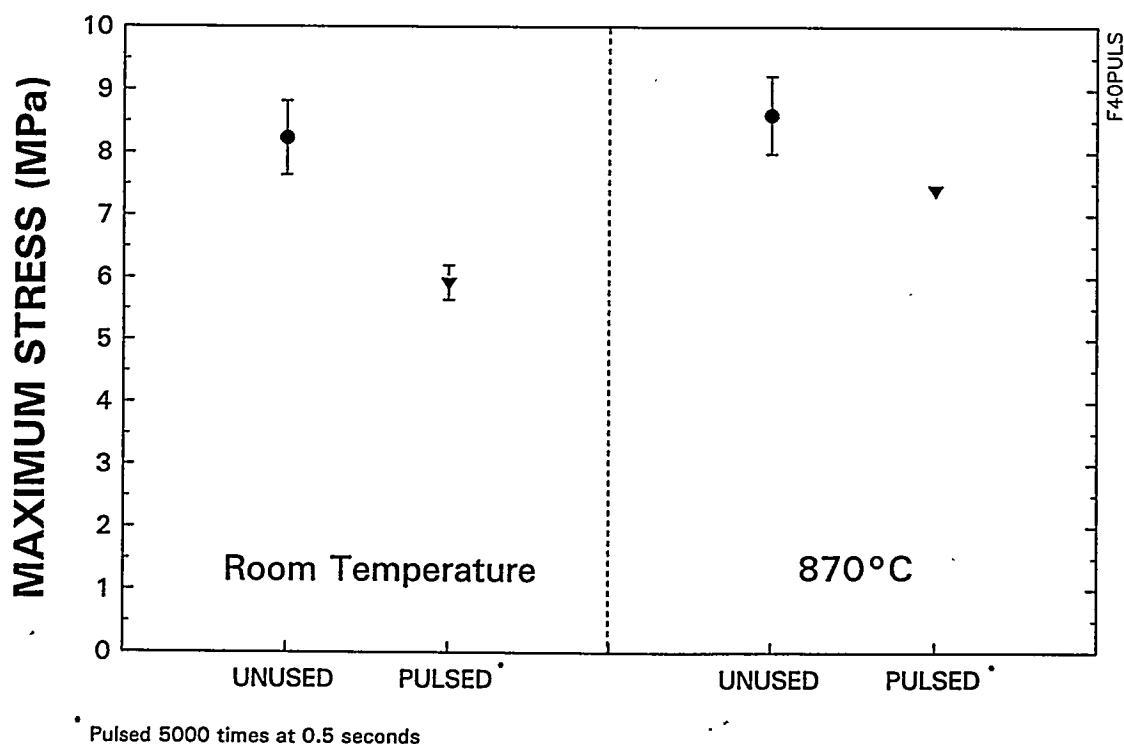
<sup>1</sup> Heated to 1200°C, then cut into C-rings

<sup>2</sup> Cut into C-rings, then heated to 1200°C

**Figure 4-6. Maximum stresses of a heat treated Refractron 50/5 with 505 binder candle c-rings to determine the effect of machining**

**Table 4-10. Maximum stress summary of c-ring tension tests investigating the effect of 5,000 ambient air pulses on candles at 870°C**

		Pulse Duration (sec)	C-Ring Test Temperature (°C)	Tension	
				Stress (Mpa)	±Std Dev
Schumacher Diaschumalith F40 Candle	Unused	n/a	Room	8.23	0.59
	Pulsed	0.5	Room	5.92	0.28
	Unused	n/a	870°C	8.60	0.62
	Pulsed	0.5	870°C	7.41	0.09
Industrial Filter & Pump (IFP) LayCer Candle	Unused	n/a	Room	18.18	1.17
	Pulsed	0.05	Room	20.56	1.97
	Pulsed	0.5	Room	13.22	0.38
	Unused	n/a	870°C	15.98	2.26
	Pulsed	0.5	870°C	17.96	1.59
Didier SiC20 Candle	Unused	n/a	Room	32.54	4.96
	Pulsed	0.05	Room	28.94	4.49



**Figure 4-7. C-ring tension test results of a Schumacher Diaschumalith F40 candle pulsed 5,000 times with ambient air while at 870°C**

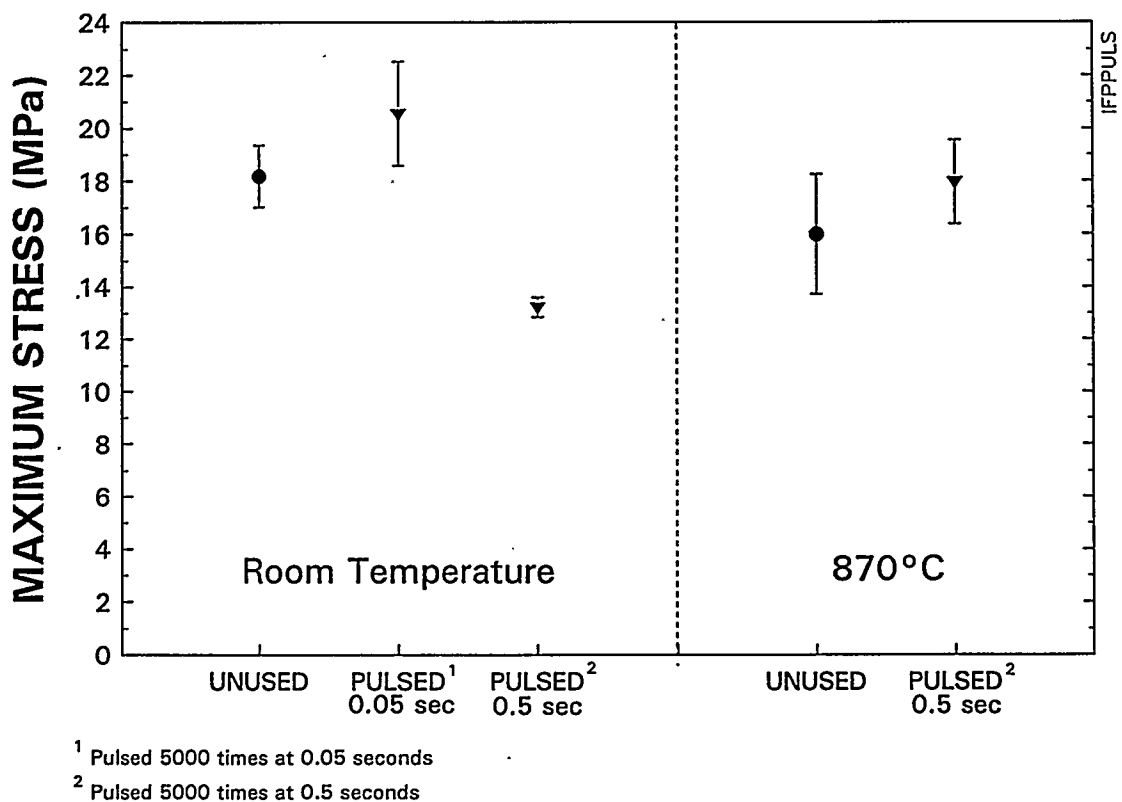


Figure 4-8. C-ring tension test results of an IFP LayCer candle pulsed 5,000 times with ambient air while at 870°C

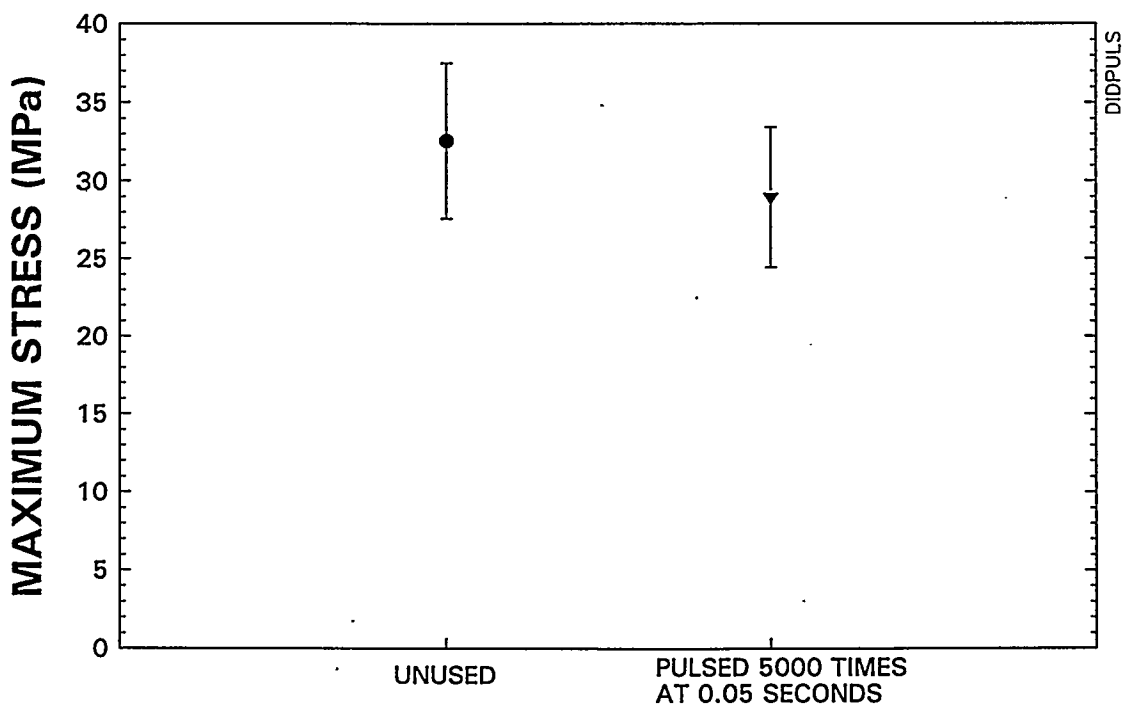


Figure 4-9. Room temperature c-ring tension test results of a Didier SiC20 candle pulsed 5,000 times with ambient air while at 870°C



As a control, the unused candles (in Figure 4-7 and 4-8) tested at high temperature resulted in stress values within the range of room temperature tests of the same unused candles. This suggests that the elevation of temperature will not affect the strength of c-ring specimens if there is no change in the surface characteristics of the candle. The control also acts as a reference to which strength values from exposed candles can be compared.

With stress duration also being a determining factor, it can be seen that the IFP LayCer and Didier candles pulsed at a time duration of 0.05 seconds (refer to Figures 4-8 and 4-9) do not show any room temperature strength loss (the averages are within the breadth of their standard deviations). This series of room temperature tension test comparisons do imply that ambient air pulses to candles at operating temperatures cause strength degradation (possibly via micro-cracking on the inner bore of the candle) when the pulse duration is at least 0.5 seconds.

An interesting aspect exhibited in Figures 4-7 and 4-8 is that at high temperature, the candles that showed strength loss (due to 0.5 second pulse durations) when tested at room temperature show improved strength (some to the point of being within the  $\pm 1$  standard deviation range for the unused candles). A possible explanation for this "healing" behavior is that the binder holding the SiC particles together is softening (increased ductility) and thereby recovering the base toughness of the candles at high temperature. A more ductile material has a larger plastic deformation energy value that dominates the high temperature fracture process and will result in increasing the critical strain-energy-release rate (also referred to as the critical crack driving force). This increases the strength because the stress at the crack tip is relieved by viscous deformation which reduces the stress at the crack tip. This also increases the critical stress value because a greater applied load is required to cause failure.<sup>11,12</sup> The raw data for this series of tests can be found in Appendix C.

#### 4.1.3 C-Ring Testing: Thermal Exposure Specimens

The summary of all c-ring tests on the Schumacher candle specimens and the Refractron candle specimens from the long term thermal exposure testing series are presented in Tables 4-11

**Table 4-11. C-ring test results summary of Schumacher candles used in the long term thermal exposure tests**

[illegible]

and 4-12, respectively. All of the raw c-ring data can be found in Appendix D. Figures 4-10 through 4-19 are the graphical representations of the c-ring data in Tables 4-11 and 4-12. Figures 4-10 through 4-14 are of the 0.50 inch Schumacher Diaschumalith F40 candle c-ring specimen strengths in Table 4-11. Figures 4-15 through 4-19 are of the 0.50 inch c-ring specimen data in Table 4-12 of the Refractron 70/3 with 442-T binder candle.

In Figure 4-10, there is no change in c-ring compression strength between 100 hours of exposure at 95 ppm NaCl, 20% steam, and 870°C (no pulse cleaning) with an unused Schumacher candle. This implies that there is no change in strength to the outer surface with exposure. There is about a 20% drop in strength from an unused candle when the candle is tested in tension. The drop in strength could not be attributed to pulse cleaning because this exposure test did not include any pulse cleaning. Because this was the first candle tested in the long term thermal exposure testing system, it is possible that thermal shock or other mechanical damage to the candle could have occurred because there were problems with the testing system at the beginning of testing which led to many shut down episodes.

Figure 4-11 shows a 40 to 60% drop in strength for the 100-, 475-, and 575-hr Schumacher specimens exposed to 65 ppm NaCl, 20% steam, and 980°C tested in c-ring tension. Figure 4-12 shows a 55 to 60% drop in strength for the 100- and 575-hr specimens tested in c-ring compression, but no significant drop in strength for the 475-hr specimen. The 575-hr candle specimen was situated near the rear of the vessel, and the 100- and 475-hr specimens were placed near the front of the vessel (not simultaneously). The tension test appears to confirm the strength degradation due to thermal shock from pulse cleaning theory, but the compression test results appears inconclusive.

There is no significant change in strength of all the Schumacher exposure candles tested at 20 ppm NaCl, 15% steam, and 870°C when tested in c-ring compression (refer to Figure 4-14). Figure 4-13 shows no significant difference in strength for the c-ring tension tests of the 100- and 400-hr exposure specimens. However, there is a 25% decrease in strength for the 1,000-hr specimen

Table 4-12. C-ring test results summary of Refractron candles used in the long term thermal exposure tests

CANDLE DESCRIPTION	C-ring Width (inches)	C-RING TENSION TESTING						C-RING COMPRESSION					
		Room Temp			870°C			Room Temp			870°C		
		Stress ± Std (MPa)	Dev	± Std	Stress ± Std (MPa)	Dev	± Std	Stress ± Std (MPa)	Dev	± Std	Stress ± Std (MPa)	Dev	± Std
REFRACTRON													
50/5 WITH 505 BINDER CANDLE													
EXPOSURE TESTING													
20ppm NaCl/15%H <sub>2</sub> O/870°C													
100hr/Pulse (♦)	1/2"				13.72	1.43					13.85	0.41	
70/3 w/442-T BINDER CANDLE													
UNUSED													
Varying width test 0.50" (2/93)	1/2"	16.97	1.47		16.21	1.27		19.42	1.50		17.94	0.90	8.61 1.05
Varying width test 0.75" (2/93)	3/4"	17.08	2.32		13.15	1.70		17.86	1.52		17.49	2.27	8.19 0.95
EXPOSURE TESTING													
20ppm NaCl/15%H <sub>2</sub> O/870°C													
100hr/No Pulse/Front (I)	1/2"				15.96	0.15					16.64	1.77	
100hr/No Pulse/Rear (I)	1/2"				16.61	4.35					16.36	0.72	
100hr/Pulse (J)	1/2"				13.85	0.87					16.60	1.13	
400hr/Pulse (K)	1/2"				11.97	1.65					12.14	0.68	
500hr/Pulse (L)	1/2"				15.36	1.40					9.10	1.45	
1000hr/Pulse (M)	1/2"				18.96	1.51					12.20	2.16	
20ppm NaCl/15%H <sub>2</sub> O/980°C													
100 hr/Pulse (N)	1/2"												5.76 0.67
400hr/Pulse (O)	1/2"												1.72 0.34
500hr/Pulse (P)	1/2"												1.17 0.20
1000hr/Pulse (Q)	1/2"												2.31 0.93

tetsumry

Less than 5 specimens available for testing

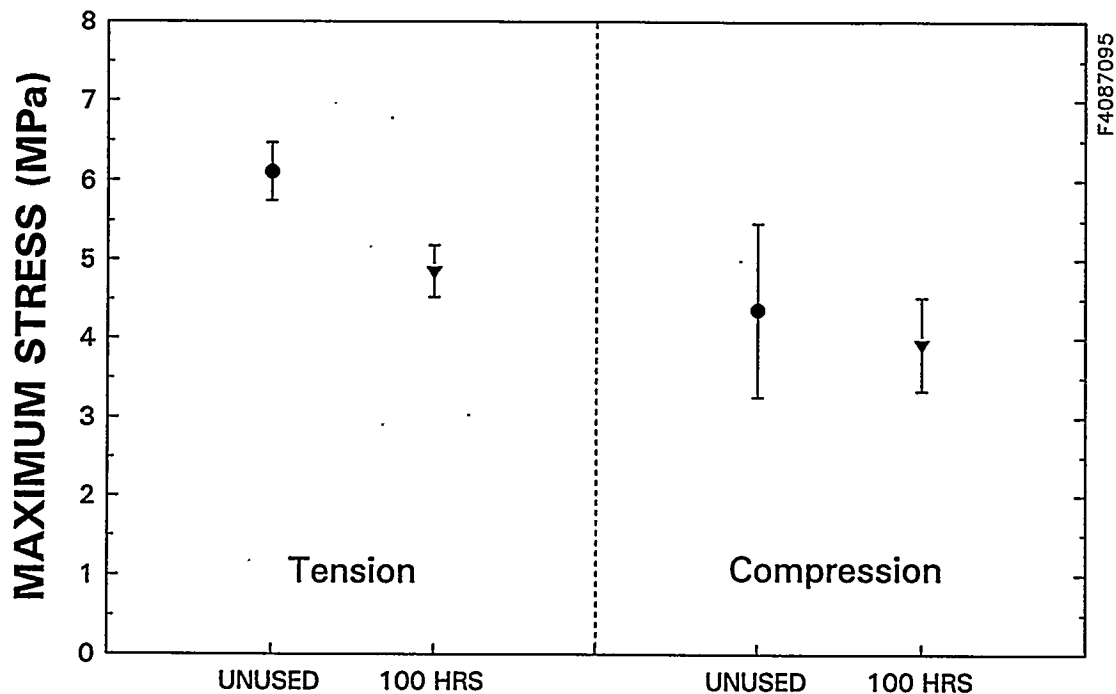


Figure 4-10. Schumacher Diaschumalith F40 candle exposed to 95 ppm NaCl and 20% steam at 870°C (no pulse cleaning): c-ring tested at 870°C

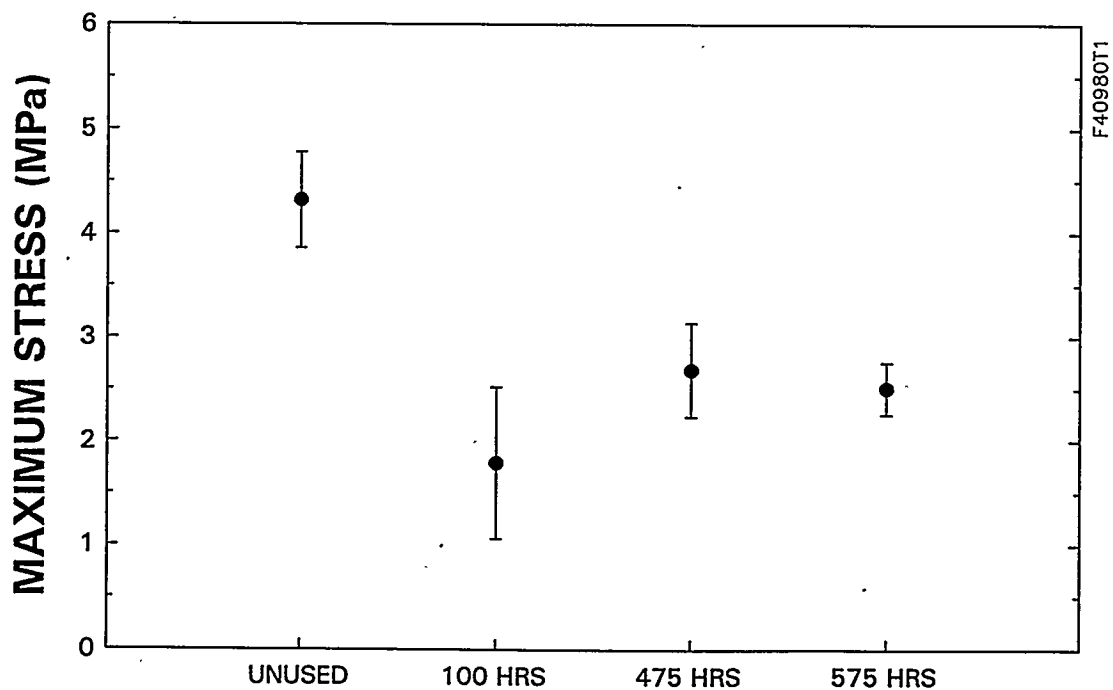


Figure 4-11. Schumacher Diaschumalith F40 candle exposed to 65 ppm NaCl and 20% steam at 980°C: c-ring tension test at 980°C

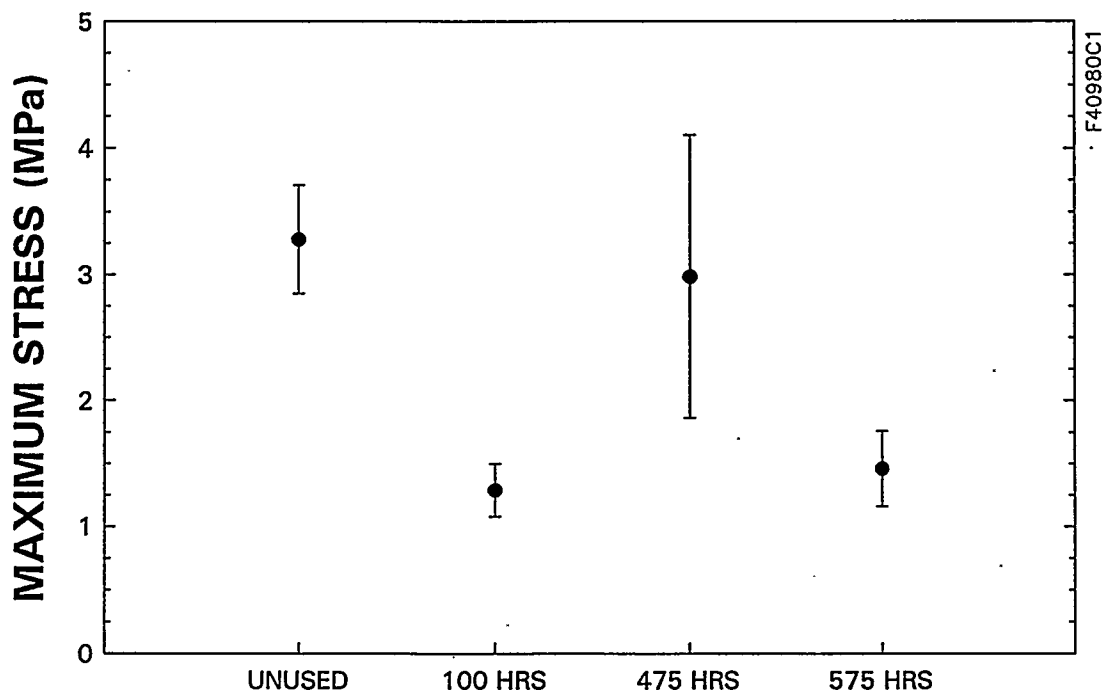


Figure 4-12. Schumacher Diaschumalith F40 candle exposed to 65 ppm NaCl and 20% steam at 980°C: c-ring compression test at 980°C

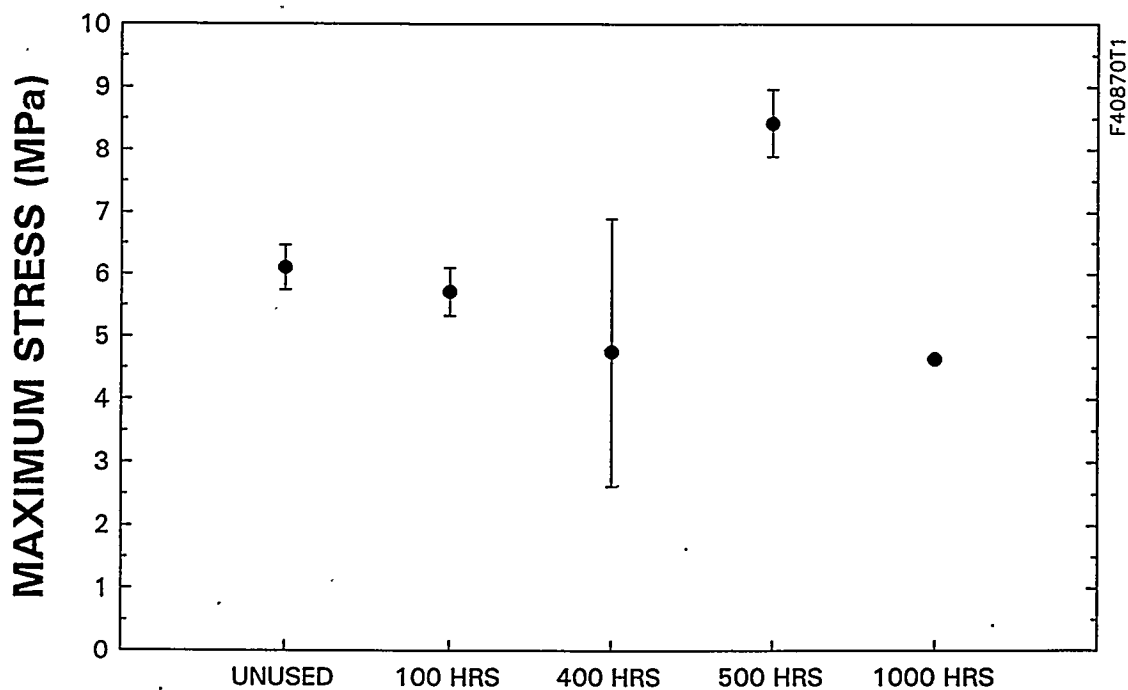


Figure 4-13. Schumacher Diaschumalith F40 candle exposed to 20 ppm NaCl and 15% steam at 870°C: c-ring tension test at 870°C

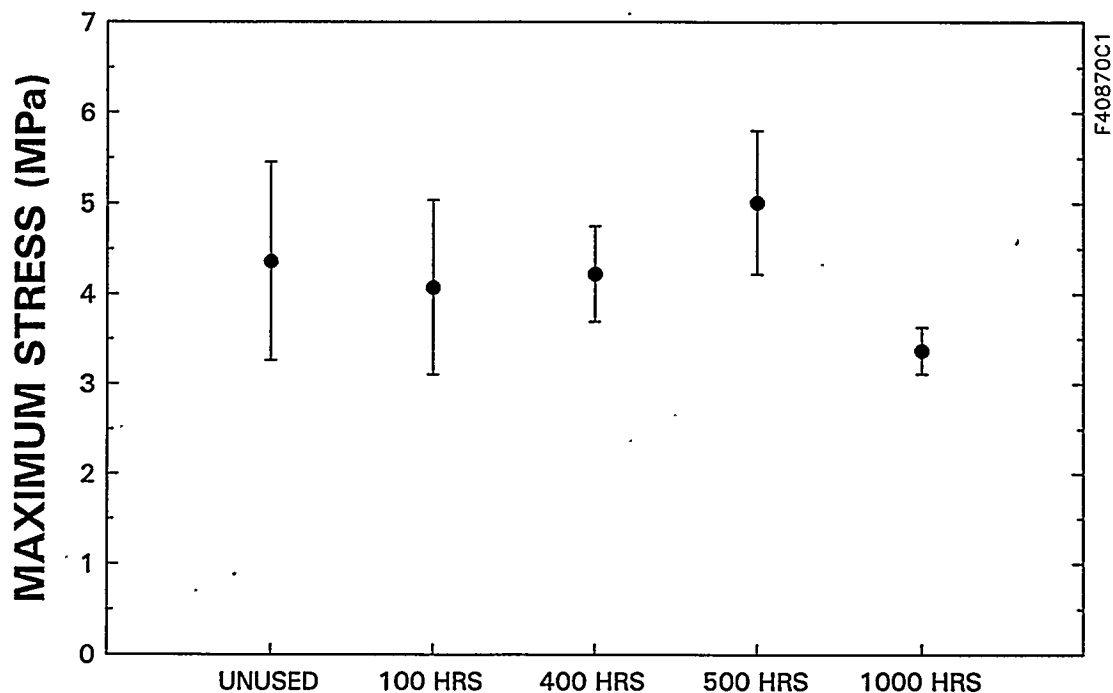


Figure 4-14. Schumacher Diäschumalith F40 candle exposed to 20 ppm NaCl and 15% steam at 870°C: c-ring compression test at 870°C

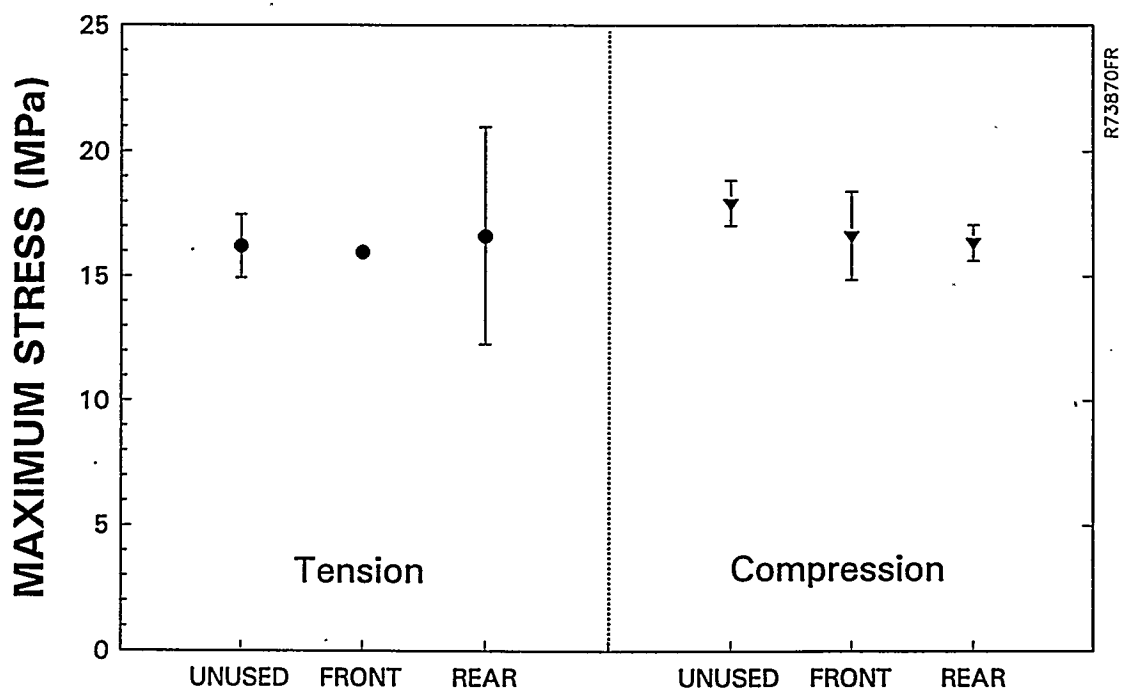


Figure 4-15. Refractron 70/3 with 442-T binder candle exposed to 20 ppm NaCl and 15% steam at 870°C for 100 hours (no pulse cleaning): c-ring tested at 870°C

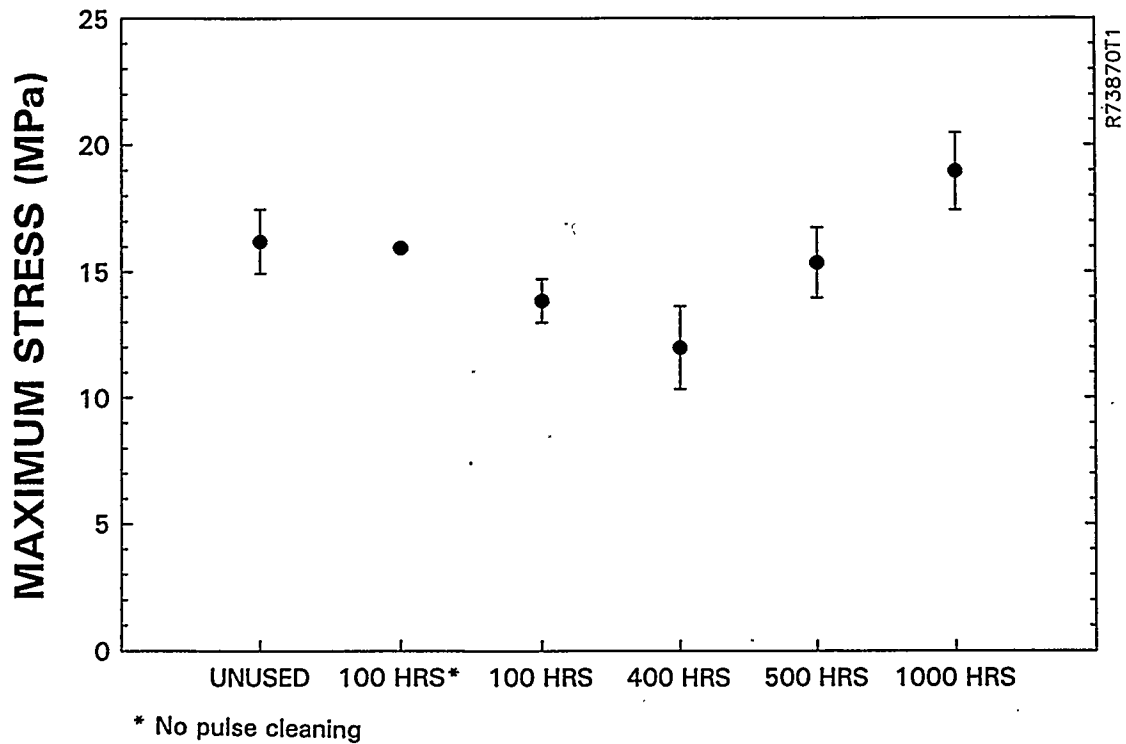


Figure 4-16. Refractron 70/3 with 442-T binder candle exposed to 20 ppm NaCl and 15% steam at 870°C: c-ring tension test at 870°C

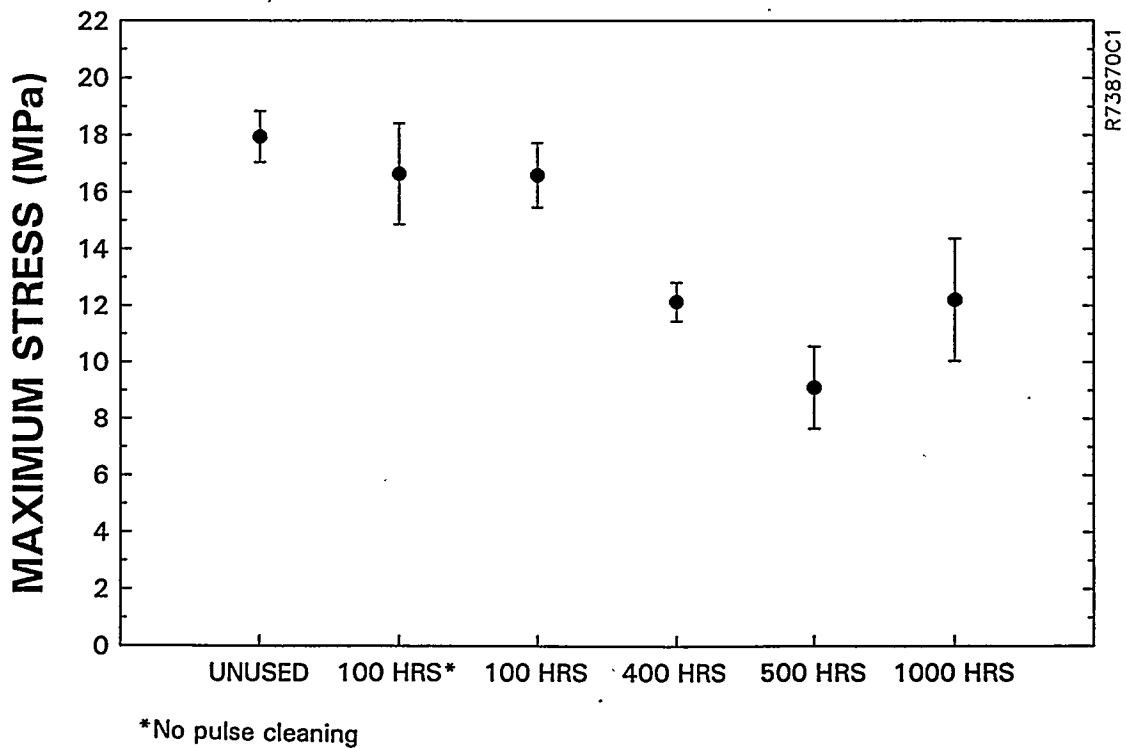


Figure 4-17. Refractron 70/3 with 442-T binder candle exposed to 20 ppm NaCl and 15% steam at 870°C: c-ring compression test at 870°C



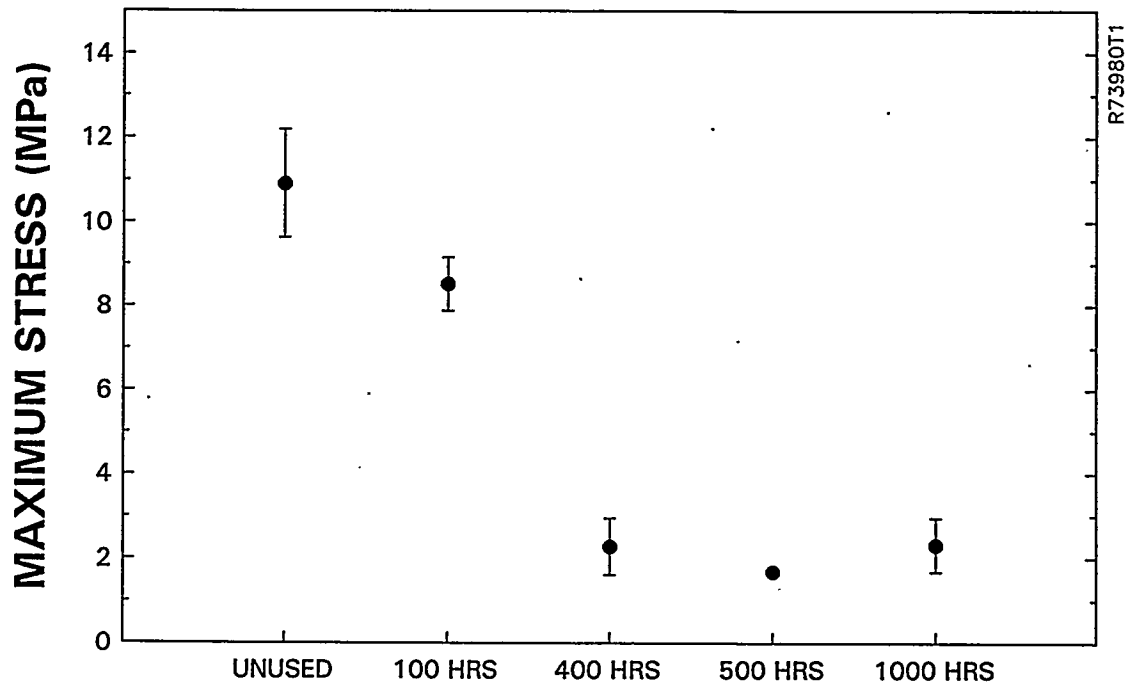


Figure 4-18. Refractron 70/3 with 442-T binder candle exposed to 20 ppm NaCl and 15% steam at 980°C: c-ring tension test at 980°C

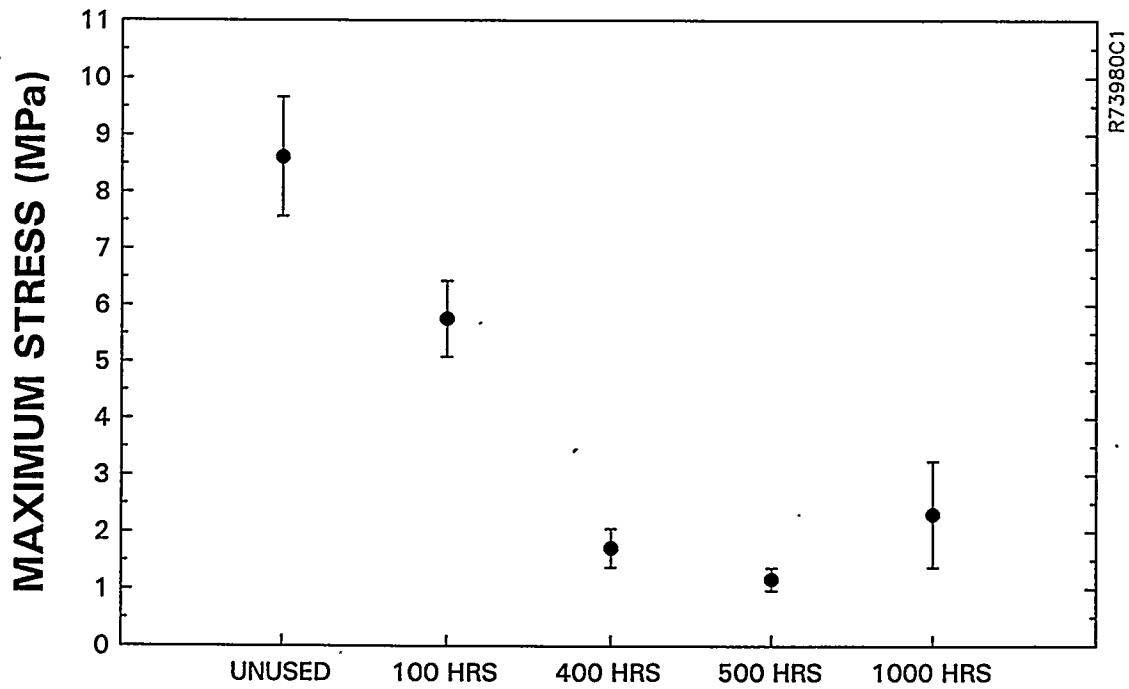


Figure 4-19. Refractron 70/3 with 442-T binder candle exposed to 20 ppm NaCl and 15% steam at 980°C: c-ring compression test at 980°C

and a 40% increase in strength for the 500-hr specimen. Other than noting that the 1,000-hr specimen is placed at the rear of the exposure vessel and the 500-hr specimen is placed at the front of the vessel, there is no apparent explanation of the difference in strengths of the 500-hr and the 1,000-hr candle specimens.

It can be seen in Figure 4-15 that there is no change in strengths for both surfaces (inner and outer) for the front and rear Refractron 70/3 with 442-T binder candles exposed for 100 hours (without any pulse cleaning) at 20 ppm NaCl, 15% steam, and 870°C. There does not appear to be any difference in the placement of the candle specimens in the exposure vessel. However, it is possible that at longer exposure times and with the inclusion of pulse cleaning, there may be a difference between the front and the rear candle.

Figures 4-16 and 4-17 show that there is essentially no difference in strength relative to unused for the Refractron 70/3 with 442-T binder 100-hr exposure specimens (both with and without pulse cleaning) at 20 ppm NaCl, 15% steam, and 870°C tested in c-ring compression. In c-ring tension, only the unpulsed 100-hr specimen had no change in strength (as mentioned above and in Figure 4-15). there is about a 15% drop in strength for the 100-hr pulsed specimen when tested in tension. The 400-hr specimen had a 25 to 30% decrease in strength for the inner and outer surfaces (tested in tension and compression). The 500- and 1,000-hr specimens showed no significant change in the tension test, but did show a 30 to 50% drop in strength in the compression test.

There is a 20 to 35% loss in strength on the inner and outer surfaces for the Refractron 70/3 with 442-T binder candle exposed for 100 hours at 20 ppm NaCl, 15% steam, and 980°C (refer to Figures 4-18 and 4-19). The inner and outer surfaces showed a tremendous 75 to 85% loss in strength for the 400-, 500-, and 1,000-hr exposure specimens. The rate of loss in strength for the outer surface is similar to the inner surface. It appears to be consistent in terms of major strength loss after 400 hours of exposure.

#### **4.1.4 C-Ring Testing: KRW Candle Strength**

In the report "Analysis of Candle Filters from the KRW Gasifier" (refer to Appendix F), 4-point flexure (bend) tests were used to ascertain the strength of the candle. At the time, Acurex Environmental had not yet finished development of the C-Ring Testing system. The report indicated that there had been no obvious strength degradation in material strength of the used candles from unused. After the c-ring testing system was developed, the KRW candles were tested to verify whether the 4-point flexure tests were appropriate for testing ceramic candle strength. Table 4-13 and Figures 4-20 and 4-21 displays the c-ring test results. Note that the high temperature (650°C) c-ring compression tests were based upon a c-ring sample population of less than 5 per candle due to the amount of material available.

It can be seen in Figure 4-20 (comparing only the unused candle with the average of the used candles) that there is about 15 to 25% drop in strength in the candle's inner surface that the 4-point flexure specimens (refer to Appendix F) could not detect. There is an obvious strength degradation that would be attributed to the pulse cleaning because the outer surface strength measurements does not show any strength loss (the high strength value for the unused candle at 650°C in Figure 4-21 is questionable due to the lack of enough material for 5 c-ring specimens per candle per test condition). The 4-point flexure tests confirm that the body of the candle between the inner and outer surfaces did not get weaker. Therefore the candles exposed to the operating conditions at KRW had the strength degradation confined to the inner surface where thermal shock from pulse cleaning would have the most effect.

#### **4.2 SEM/EDX ANALYSIS**

In the thermal exposure tests (see section 3.3.1), Tests A-D were aborted before the full 1,000 hours of testing because it was determined that the NaCl levels were higher than the 20 ppm NaCl target. The Schumacher Diaschumalith F40 candle was tested at 870°C for the full 1,000 hours with correct NaCl levels. The Refractron 70/3 with 442-T binder candle was tested at 870°C and 980°C for the full 1,000 hours also with correct NaCl levels. The candles tested at 980°C were

Table 4-13. C-ring tests on KRW candles

	C-Ring Width (inch)	Temperature (°C)	Tension		Compression	
			Stress (Mpa)	±Std Dev	Stress (Mpa)	±Std Dev
Unused (7068)	0.75	Room	24.29	1.00	21.16	0.96
Unused (7068)	0.75	650°C	25.02	1.29	25.38 <sup>a</sup>	1.84
Used (7052)	0.75	Room	18.21	1.38	17.58	2.18
Used (7052)	0.75	650°C	19.60	0.99	18.89 <sup>a</sup>	1.48
Used (7067)	0.75	Room	18.10	0.83	20.81	1.08
Used (7067)	0.75	650°C	21.27	0.49	21.36 <sup>a</sup>	0.79
Used (Average)	0.75	Room	18.16	1.11	19.20	1.63
Used (Average)	0.75	650°C	20.44	0.74	20.13 <sup>a</sup>	1.14

<sup>a</sup>The c-ring compression values at 650°C were based upon less than 5 specimens per candle

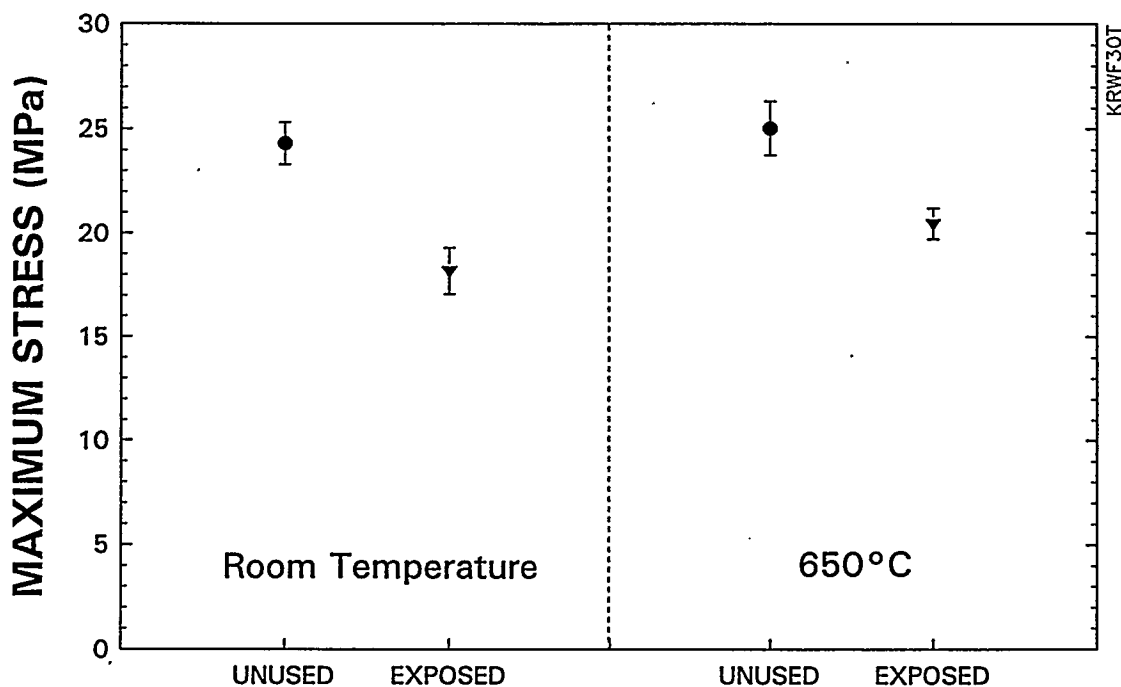
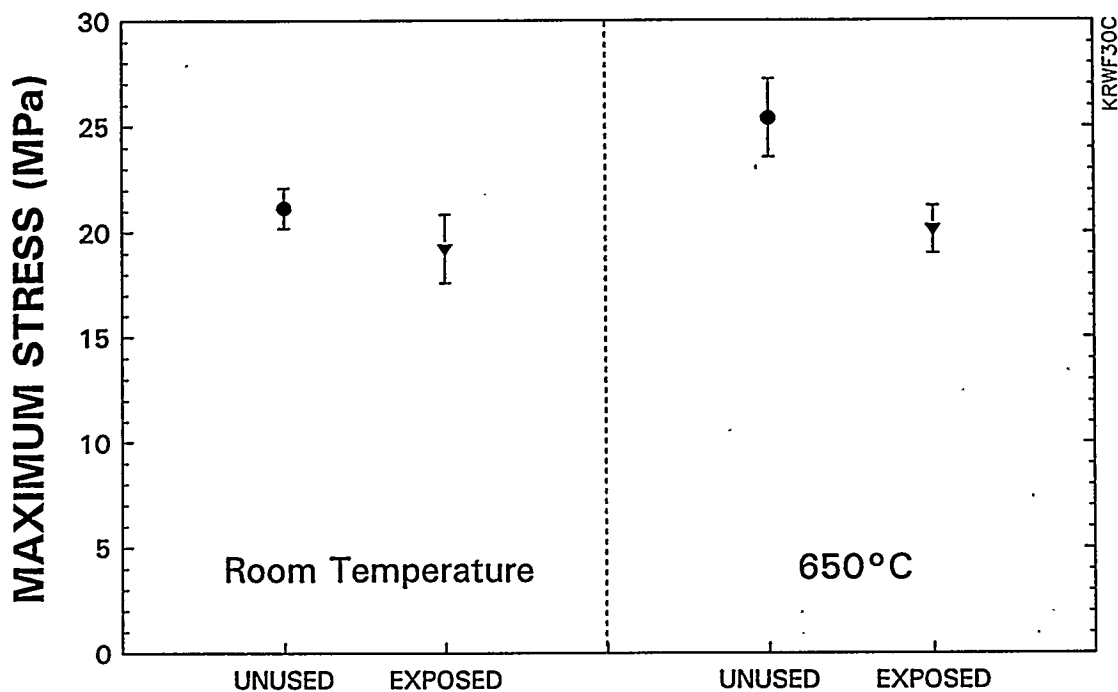


Figure 4-20. KRW candles: c-ring tension tests



**Figure 4-21. KRW candles: c-ring compression tests**

deformed and showed evidence of melting and flowing. The final condition of the candles was unacceptable for use in that environment (20 ppm NaCl/15% H<sub>2</sub>O/980°C).

The SEM/EDX analysis of an unused Schumacher candle showed a fibrous outer layer (see Figure 4-22) that consists of about 90% Si and 10% Al (the EDX detector used could not determine elements of atomic weights less than 23 (sodium)). Figure 4-23 shows the outer layer of the candle from Test H (20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours). The SEM micrograph shows an outer layer that had melted and flowed. The fibers that are easily recognized in Figure 4-22 are no longer evident in Figure 4-23. The Test H candle outer surface consisted of about 34% Si, 4% Al, 42% Fe, and 13% Cr (the balance are of other elements). The presence of Fe and Cr suggests that the high temperature air, steam, and alkali chemically attacked the pre-vessel piping. The corrosion byproducts were then deposited onto the candle. The inner candle surface revealed 46% Si, 6% Al, 24% Fe, 12% Cr, and 0.5% Na. This shows that the vessel corrosion byproducts reached all the

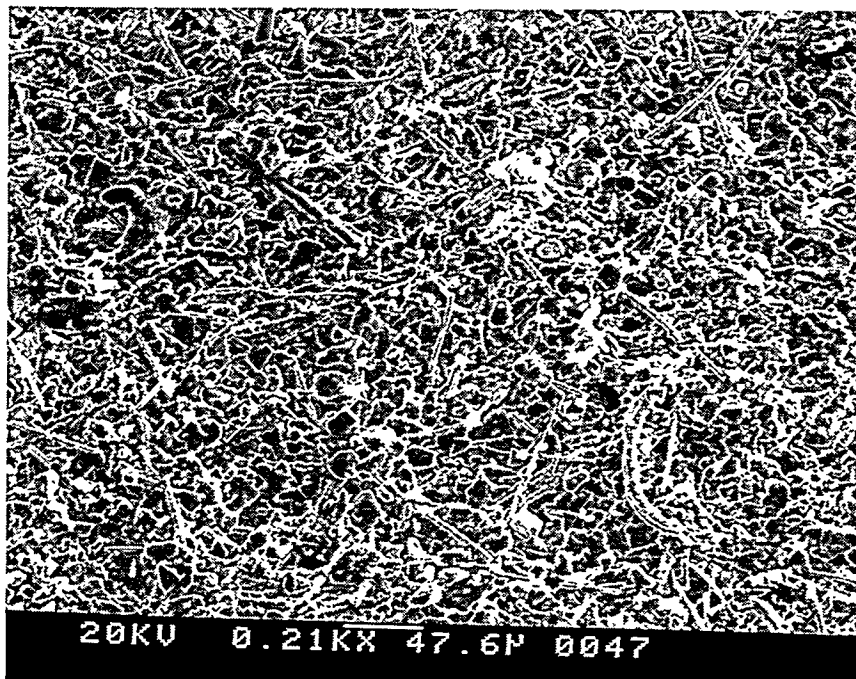


Figure 4-22. Micrograph of Schumacher F40/Unused: Outer surface (210X)

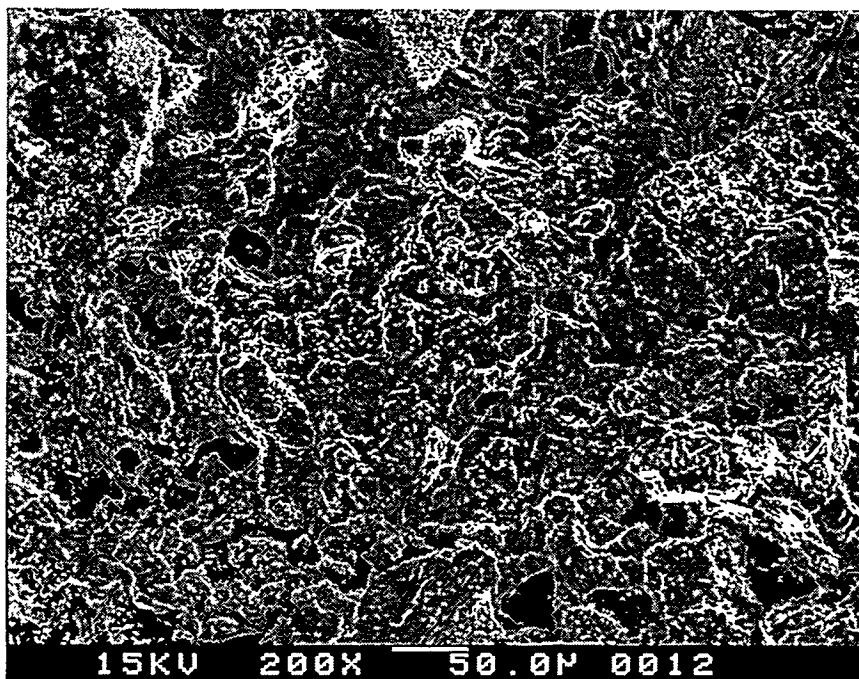


Figure 4-23. Micrograph of Schumacher F40/20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours: Outer surface (200X)

way to the interior of the candle. It also appears that the outer surface experienced blocking of pore openings by the melting and flowing of the surface. The shape of the candle had been retained and was not deformed as were the candles tested at 980°C.

Figure 4-24 shows the cross-sectional view of the unused Schumacher candle (outer surface on right side of micrograph). Figure 4-25 is the cross-sectional view (outer surface on right side of micrograph) of the Schumacher candle from Test H. In comparing the two cross-sectional views, it can be seen that the SiC grains that make up the body of the Schumacher candle is still present.

Using SEM/EDX analysis, the outer surface of an unused Refractron 70/3 with 442-T binder candle (see Figure 4-26) contains about 91% Si, 3% Al, and 3% Fe. The outer surface of the Refractron candle from Test M (20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours) is shown in Figure 4-27 and contains about 36% Si, 4% Al, 38% Fe, 9% Cr, and 0.5% Na. It can be seen in Figure 4-27 that there had been some melting and flow of the outer surface, but it appears that the pore structure has not been as blinded/plugged as in the Schumacher candle at the same operating conditions. The shape of the small grains on the outer surface can still be distinguished.

Figure 4-28 shows the inner surface of an unused Refractron candle from Test M which consists of about 70% Si, 13% Al, 3% Fe, and 0.8% Na. The inner surface of the Refractron candle in Test M (Figure 4-29) is comprised of about 77% Si, 6% Al, 6% Fe, 0.5% Cr, and 1.5% Na. It is evident that there is also binder melting and flowing on the interior of the candle like the exterior. At higher magnification of the inner surface of the Refractron candle (Figure 4-30), microcracks can be observed. This magnification of the inner surface is at the same magnification as the outer surface (200X). No microcracks can be seen on the outer surface (Figure 4-27) like in the inner surface (Figure 4-30).

From the EDX analysis, it appears that most of the corrosion byproducts from the metallic piping and vessel remained on the outer surface of the Refractron candles.

Examination of the exposed filters show that there had been some bond depletion and flowing of the candle material. The flowing of the binder (becoming viscous at high temperature)

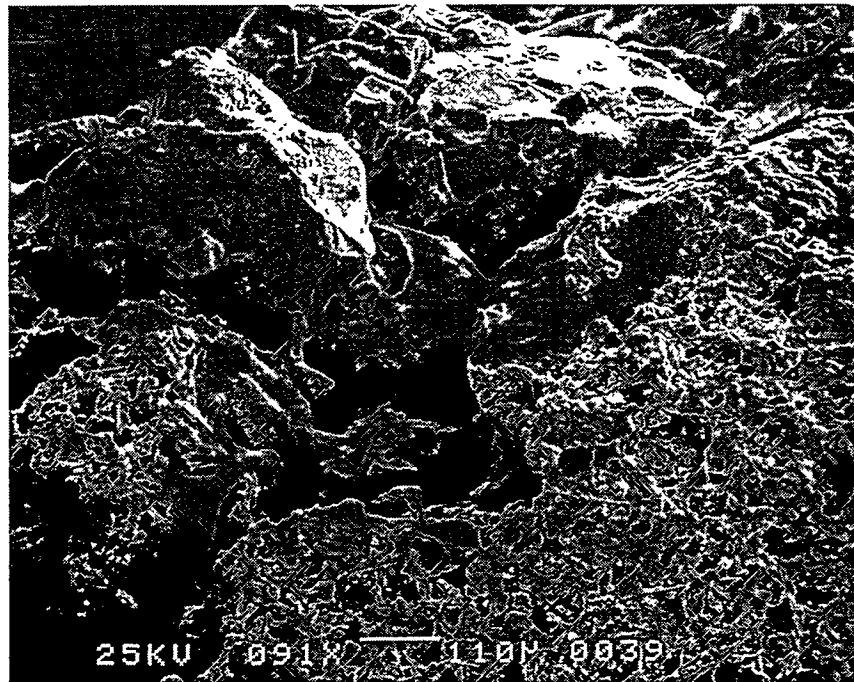


Figure 4-24. Micrograph of Schumacher F40/Unused: Cross-section (91X)

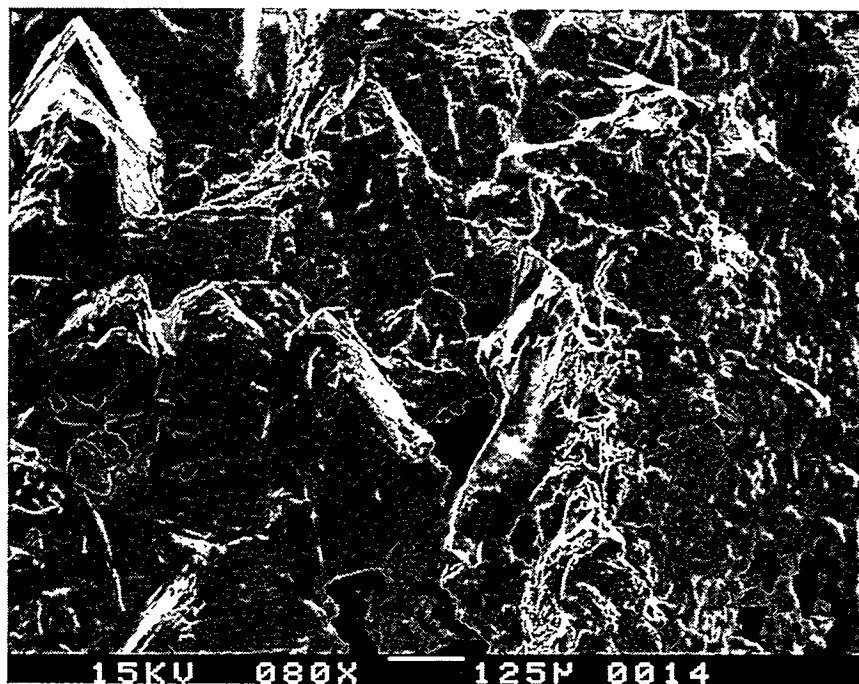


Figure 4-25. Micrograph of Schumacher F40/20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours: Cross-section (80X)





Figure 4-26. Micrograph of Refractron 70/3 with 442-T/Unused: Outer surface (200X)

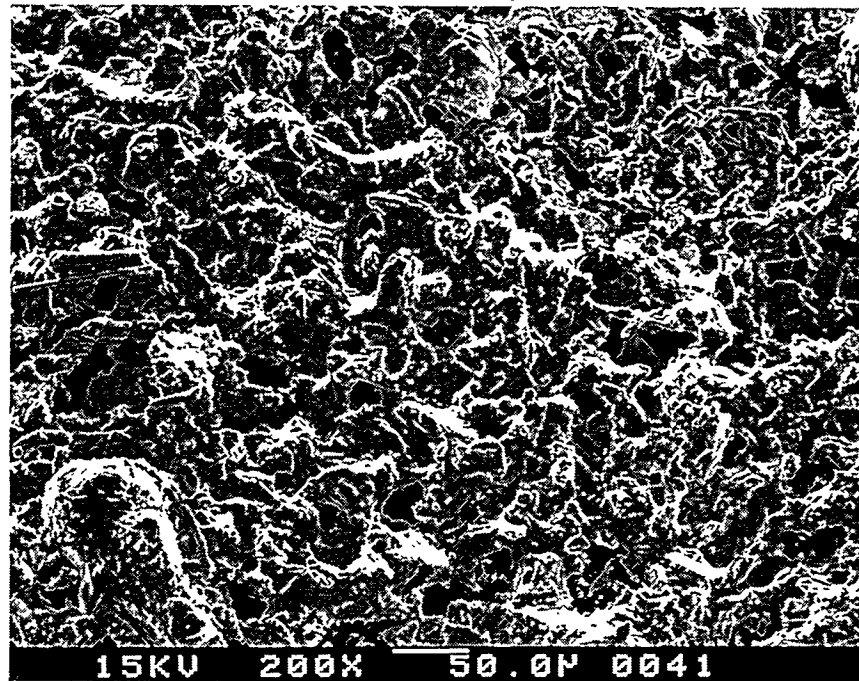


Figure 4-27. Micrograph of Refractron 70/3 with 442-T/20 ppm NaCl/15% H<sub>2</sub>O/870°C/1000 hours: Outer surface (200X)



Figure 4-28. Micrograph of Refractron 70/3 with 442-T/Unused: Inner surface (80X)

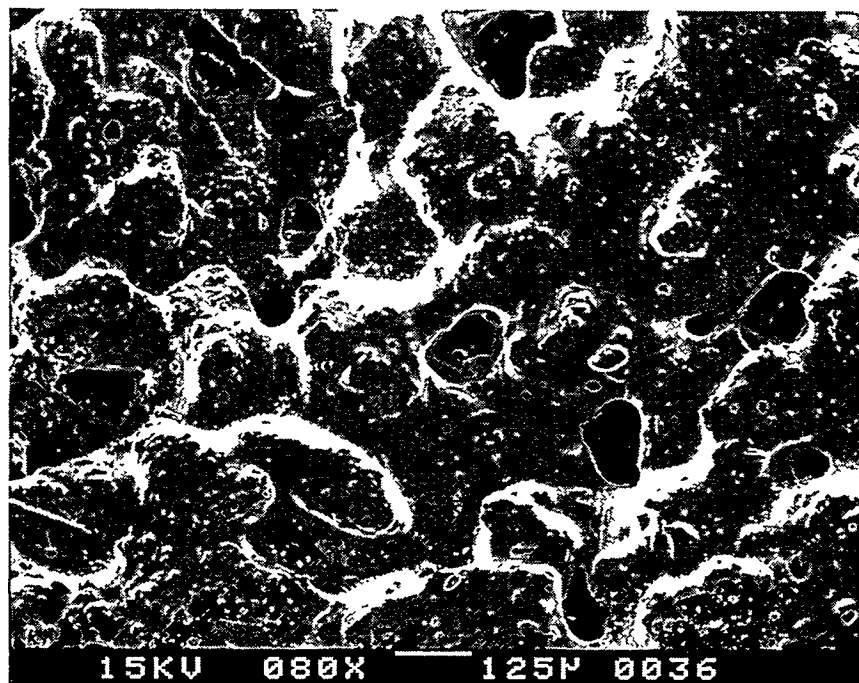


Figure 4-29. Micrograph of Refractron 70/3 with 442-T/20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours: Inner surface (80X)

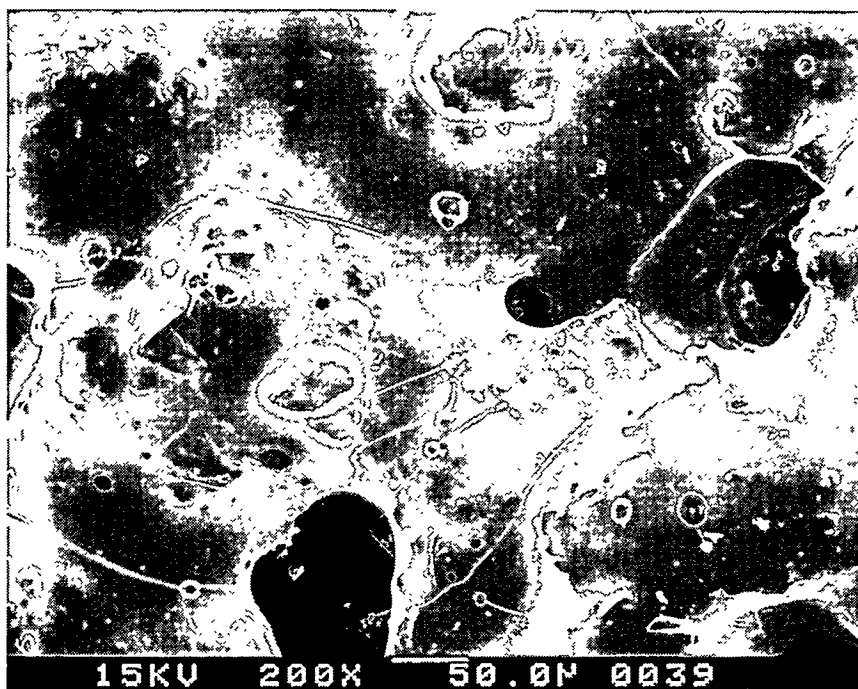


Figure 4-30. Micrograph of Refractron 70/3 with 442-T/20 ppm NaCl/15% H<sub>2</sub>O/870°C/1,000 hours: Inner surface (200X)

may have been due to gravity and aerodynamic drag of the hot gases flowing through the candle filters.

#### 4.3 XRD ANALYSIS

XRD of the Schumacher and Refractron candle filters revealed silicon carbide (SiC) as the major constituent. This is expected because SiC grains constitute more than 90% of the total candle material. The lack of enough binder material to analyze makes the detection of changes of phases in the binder material difficult. A summary of XRD analyses performed are presented in Table 4-14 (the raw XRD data is presented in Appendix E). Instead of showing all of the individual compounds, the types of phases are presented instead. The following is a list of the individual compound numbers (as they appear in the Powder Diffraction File of the Joint Committee on Powder Diffraction Standards) for each phase listed in Table 4-14 (each dark square in Table 4-14 represents one or more of the listed compounds).

Table 4-14. X-Ray diffraction summary

Test,	Test Description/Condition	Total Pulses	Silicon Carbide		Silicon Dioxide		NaAlSi <sub>3</sub> O <sub>8</sub> Albite	Al <sub>2</sub> O <sub>3</sub> Corundum
			SiC	Moissanite	SiO <sub>2</sub>	Cristobalite		
	Schumacher Diaschumalith F40							
	Unused Schumacher	0						
A	95 ppm NaCl/20%H <sub>2</sub> O/870°C/100 hrs	0						
B	65 ppm NaCl/20%H <sub>2</sub> O/980°C/100 hrs	200						
C	65 ppm NaCl/20%H <sub>2</sub> O/980°C/475 hrs	1,050						
D	65 ppm NaCl/20%H <sub>2</sub> O/980°C/575 hrs	1,250						
E	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	230	n/a	n/a	n/a	n/a	n/a	n/a
F	20 ppm NaCl/15%H <sub>2</sub> O/870°C/400 hrs	920	n/a	n/a	n/a	n/a	n/a	n/a
G	20 ppm NaCl/15%H <sub>2</sub> O/870°C/500 hrs	1,140						
H	20 ppm NaCl/15%H <sub>2</sub> O/870°C/1,000 hrs	2,290						
	Refractron 70/3 with 442-T Binder							
	Unused Refractron	0						
I	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	0	n/a	n/a	n/a	n/a	n/a	n/a
J	20 ppm NaCl/15%H <sub>2</sub> O/870°C/100 hrs	240	n/a	n/a	n/a	n/a	n/a	n/a
K	20 ppm NaCl/15%H <sub>2</sub> O/870°C/400 hrs	980	n/a	n/a	n/a	n/a	n/a	n/a
L	20 ppm NaCl/15%H <sub>2</sub> O/870°C/500 hrs	1,170						
M	20 ppm NaCl/15%H <sub>2</sub> O/870°C/1,000 hrs	2,390						
N	20 ppm NaCl/15%H <sub>2</sub> O/980°C/100 hrs	230	n/a	n/a	n/a	n/a	n/a	n/a
O	20 ppm NaCl/15%H <sub>2</sub> O/980°C/400 hrs	840	n/a	n/a	n/a	n/a	n/a	n/a
P	20 ppm NaCl/15%H <sub>2</sub> O/980°C/500 hrs	1,060	n/a	n/a	n/a	n/a	n/a	n/a
Q	20 ppm NaCl/15%H <sub>2</sub> O/980°C/1,000 hrs	2,130						

n/a = Data is not available because XRD analysis was not performed

Silicon Carbide (SiC)

SiC: 1-119, 4-0756, 22-1273, 22-1316, 29-1127, 29-1129

Moissanite: 19-1138, 29-1128, 29-1130, 29-1131

Silicon Oxide (SiO<sub>2</sub>)

SiO<sub>2</sub>: 18-1169, 27-0605, 34-1382

Cristobalite: 11-0695

Sodium Aluminum Silicate (NaAlSi<sub>3</sub>O<sub>8</sub>)

Albite: 10-0393, 20-0572

Aluminum Oxide (Al<sub>2</sub>O<sub>3</sub>)

Corundum: 10-0173

XRD analyses were not performed for the thermal exposure Tests E, F, I, J, K, N, O, and P.

Crystalline silica was detected in the used candles, but none was detected in the unused candles. This observation implies that the silicon was either tied up as another crystalline compound (such as SiC) or as a noncrystalline compound (such as amorphous silica) before exposure. After exposure, the amorphous silica may have devitrified or silicon compounds (such as SiC) may have reacted with oxygen to produce crystalline silica. Cristobalite was detected only in the used Schumacher candles exposed to 980°C (65 ppm NaCl) or 95 ppm NaCl (870°C). The used Schumacher candles exposed to 20 ppm NaCl (870°C) did not show the presence of cristobalite. The presence of cristobalite is undesirable because of the difference in volume expansion (thermal expansion) of cristobalite compared to the volume expansion of the common form of silica (quartz). Cristobalite, the high-temperature polymorphic form of silica, has a density of about 2.32 g/cm<sup>3</sup>, where as quartz has a density of 2.65 g/cm<sup>3</sup>.<sup>13</sup> This difference in density corresponds to differences in volume changes, which can lead to the cracking. The transformation temperature for cristobalite is 1,470°C. Since the thermal exposure tests never exceeded 1,000°C, then the presence of other elements (such as the Na modifier) has reduced the transition temperature to less than 1,000°C. This could mean that each time a "cold" air pulse cleaning

occurs, the drop in temperature could cause the formation of microcracks (due to the volume mismatch) which degrades the strength of the candles.

The cristobalite phase was only noticed in the Schumacher candles tested at 980°C and/or >15% steam and/or >20 ppm NaCl. No cristobalite was detected for the Schumacher candles tested at 870°C nor any of the Refractron candles tested (both at 870°C and 980°C).

Albite ( $\text{NaAlSi}_3\text{O}_8$ ) was detected only in the exposed Refractron candles, hence the formation of albite is as a result from the exposure testing at both 870°C and 980°C. The same type of argument for the formation of microcracking can be applied to albite as it did for cristobalite. However, the exact effect of albite on the candle properties are unknown. It is also a possibility that the formation of albite (which incorporates the Na modifier) would decrease the viscosity of the candle that would result in crack blunting.

Corundum ( $\text{Al}_2\text{O}_3$ ) was detected in all candles tested except for the unused Schumacher candle. It is assumed that the alumina in the unused Schumacher candle is in the form of amorphous alumina.

#### 4.4 FIBROSIC TEST RESULTS

Industrial Filter and Pump (IFP) has developed a fibrous filter that is different from the rigid candles (typically SiC grains held together by an alumino-silicate binder) tested above. The filter is a vacuum-formed body which is post-treated to strengthen the high stress areas and modify the surface pore size. In late 1991 and early 1992, Acurex Environmental performed durability testing (funded by the Electric Power Research Institute) of the Fibrosic candle under development by IFP. In operation, the Fibrosic candles performed well (having high collection efficiency with low pressure drop), but the candle durability was suspect because the candles catastrophically failed during testing. These candles were assembled by gluing a flange onto the vacuum-formed body. The tests showed the glued joint to be the weak area. The burst tests of these filters showed the used candles to have a candle burst pressure of approximately 8 psi whereas the unused candles

burst at approximately 16 psi. These measurements indicated there had been significant strength loss from use.

#### 4.4.1 Fibrosic Candle Test Results

Since the original tests, IFP has further developed the Fibrosics filter. A final part of the project assessed the durability of the new Fibrosic filter to accelerated pulse cleaning at high temperature and to actual filtration operating conditions. The Fibrosics candle test conditions were:

- Temperature: 1,600°F (870°C)
- Pressure: Atmospheric
- Dust: PFBC flyash (4.7  $\mu\text{m}$  D<sub>50</sub>)
- Dust load: 4,000 ppm<sub>w</sub>
- Pulse pressure: 60 psig
- Pulse duration: 0.5 seconds
- Face velocity: 8 ft/min

Table 4-15 summarizes the test results performed on the Fibrosic candles.

**Table 4-15. Fibrosics candle test summary**

	Unused	Accelerated Pulse Cleaning	Operational Filtration
Pulse Frequency (sec)		30-50	
$\Delta P$ Set Point (inches H <sub>2</sub> O)			15
Pulse Duration (sec)		0.5	0.5
Total Number of Pulses		15,000	20
Total Time (hours)		200	100
Collection Efficiency (%)		99.992% (0 pulses) 99.990% (5k pulses) 99.992% (10k pulses) 99.992% (15k pulses)	99.995 (10 hours) 99.995 (24 hours) 99.998 (53 hours) 99.998 (79 hours) 99.999 (100 hours)
Burst Pressure (psig)	20	16	15

Upon disassembly of the candle from the accelerated pulse cleaning test, a crack was discovered about 4 inches from the bottom of the candle flange. The crack was perpendicular to the axial length of the candle and extended about 3.5 inches along the candle surface. The crack depth (measured about mid-point along the crack) was estimated to be about 1/8 of an inch. The crack had not propagated to the interior wall. The collection efficiency results confirm this observation.

The cylindrical burst strength tests performed on the accelerated pulsed cleaned candle resulted in the failure of the candle at a location other than at the crack site. This is contrary to the expectation that the site of candle failure would be at the weakest part of the candle which is presumed to be the crack.

Figure 4-31 shows the Fibrosic candle permeability at 870°C during the accelerated pulse cleaning tests. The results show that initially, the permeability dropped (pressure drop across filter increased) at the 5,000 and 10,000 pulse mark, and then increased at the 15,000 (concluding) mark. The initial drop in permeability could be attributed to the fact that filter collection efficiencies were also being performed on the same candle. The flyash used during collection efficiency tests remained on the surface of the candle, affecting the permeability.

Figure 4-32 shows the Fibrosic candle permeability at 870°C during the 100 hour operational filtration testing. This candle test also shows the same initial drop in permeability followed by an increase after the 50 hour mark.

There was only one candle available for each test.

#### **4.4.2 Fibrosic Tubesheet Test Results**

IFP has also been developing a fibrous ceramic tubesheet and candle hold-down system. This approach has appeal by reducing the amount of metal subjected to high temperature gas. The tubesheet/hold-down assembly has not been subjected to sufficient testing. There is no long term testing data of the structural integrity of the Fibrosic tubesheet at high temperature. Simulation of the "worst" case structural conditions that the tubesheet can encounter was done (see Section 3



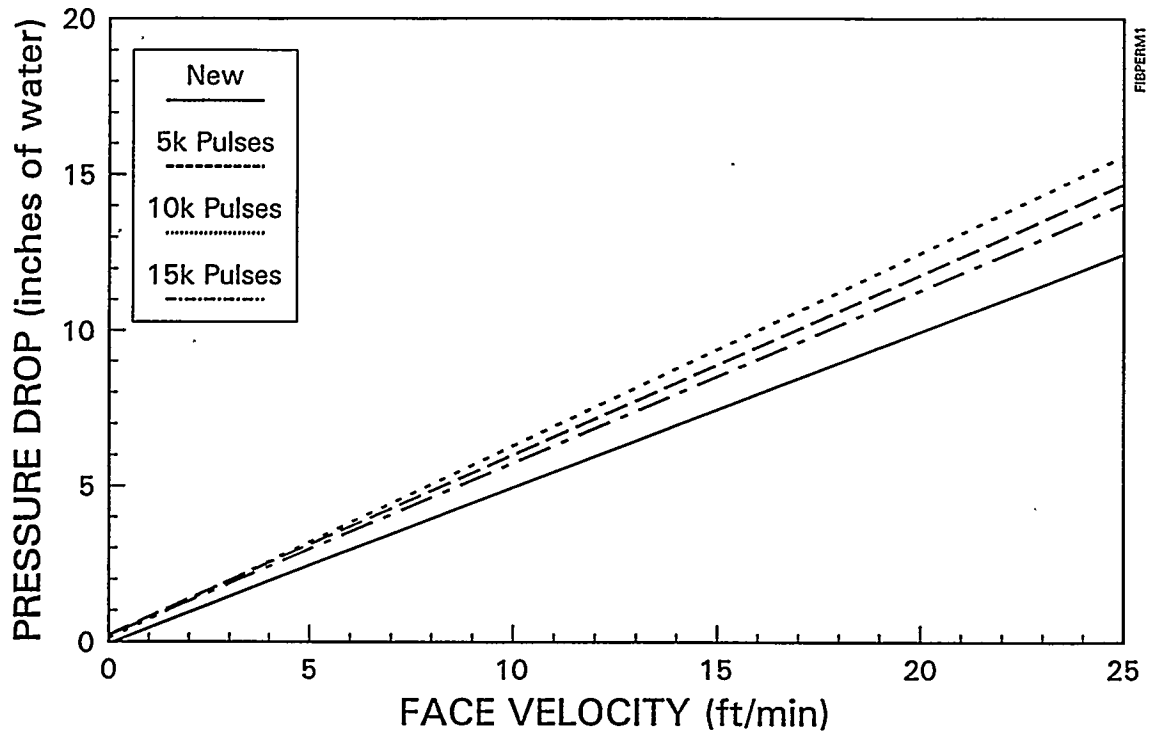


Figure 4-31. Fibrosic candle permeability: Accelerated pulse cleaning at 870°C

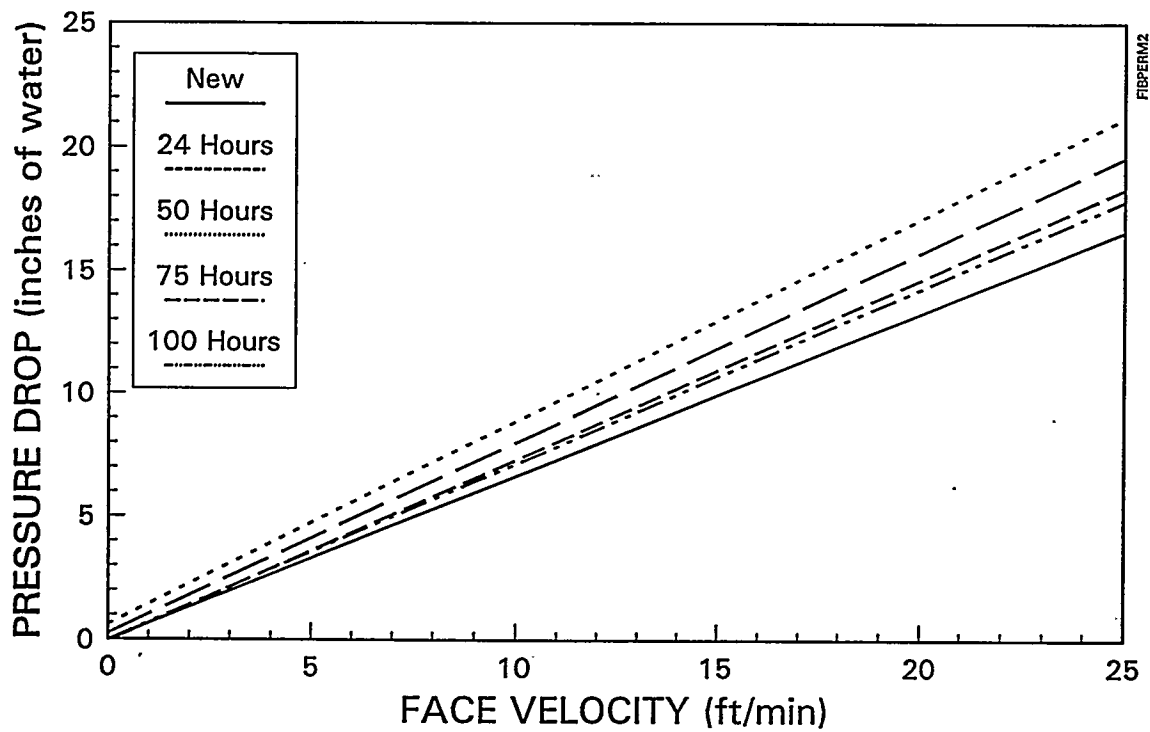


Figure 4-32. Fibrosic candle permeability: 100 hour operational filtration test at 870°C

above). The tubesheet subjected to a loading similar to that expected from the filter, dust, and hold-down plate at 870°C for 500 hours. During the test, measurements of deflection of the tubesheet (via alumina rods) from the as received tubesheet as a function of time were performed. The measurements were performed manually with a pair of calipers.

Figure 4-33 shows the absolute value of displacement over time of the alumina rod that was extending up from the surface of the Fibrosic tubesheet and through the top of the oven. Figure 4-34 displays the locations on the tubesheet in which the alumina rod was situated. It can be seen in Figure 4-33 that the major displacement in the Fibrosic tubesheet occurred within the first 24 hours at 870°C. Even after 500 hours, it appears there was still deflection (sagging) in the tubesheet. The raw data from the measurements of the alumina rods are in Appendix G.

Visual examination after removal of the tubesheet from the oven after load testing revealed some separation of the layers on the edge of the tubesheet. Since the tubesheet is in compression during operational use, however, this does not pose a problem. There were some small cracks observed on the bottom of the tubesheet as well as on the interior of some of the candle holes. Upon laying a straight edge across the width of the tubesheet section, a deflection of about 0.1 inches was observed.

Figure 4-35 shows the topographic map of the surface of the Fibrosic tubesheet. This topographic mapping is based on the edges of the Fibrosic tubesheet (across the width) defined as at the reference height of "0". Figure 4-36 shows an overlay of the topographic map on a drawing of the Fibrosic tubesheet to show which parts of the tubesheet are correlated by the topographic map. Figure 4-37 shows the three dimensional representation of the topographic map of the Fibrosic tubesheet surface (note the Z-axis is not in the same scale as the X- and Y- axes).

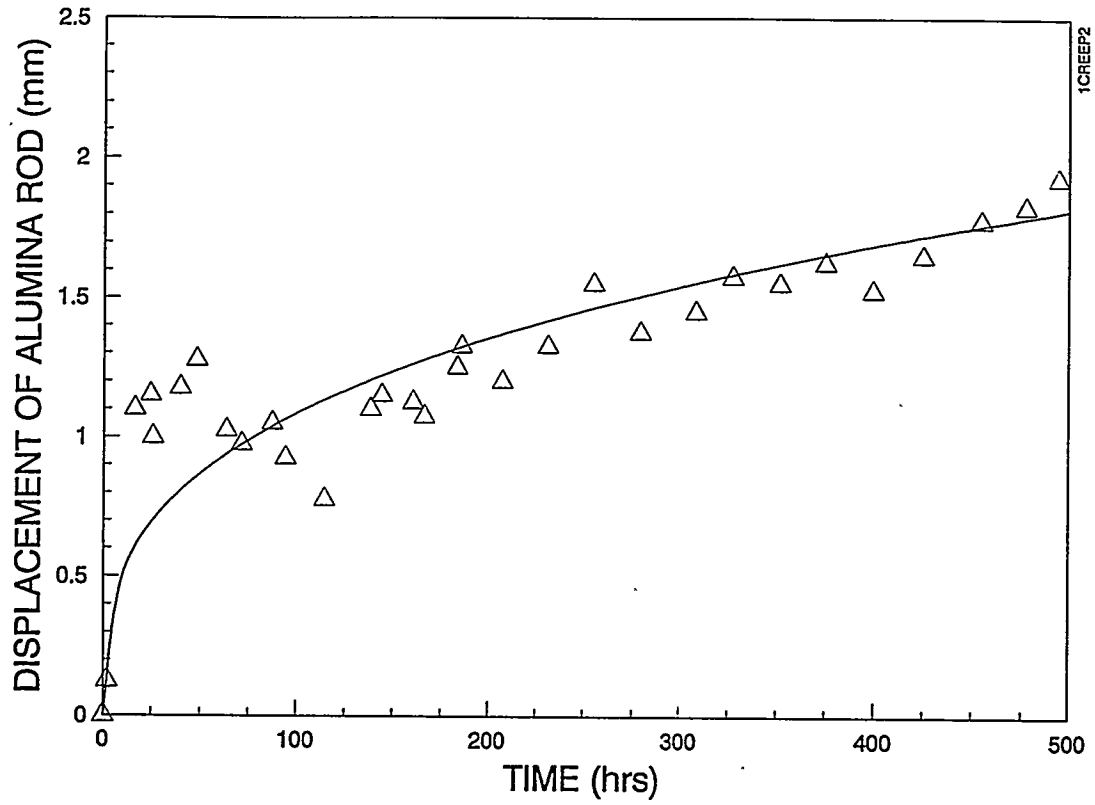


Figure 4-33. Displacement of an alumina rod on the surface of the Fibrosic tubesheet

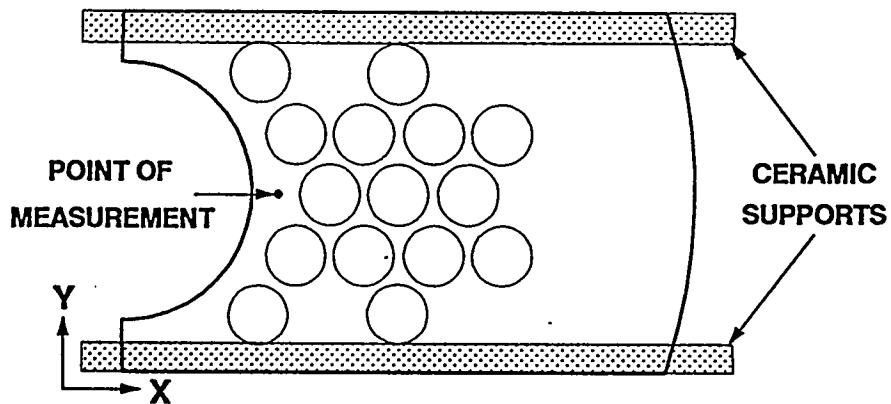


Figure 4-34. Location of alumina rod on Fibrosic tubesheet

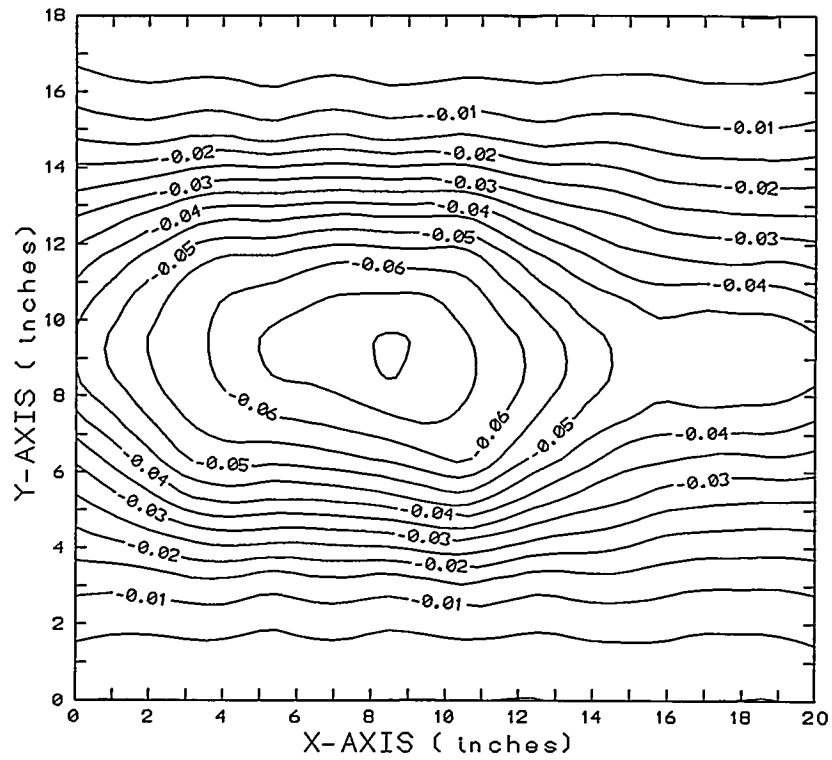


Figure 4-35. Topographic map of the Fibrosic tubesheet after load testing at 870°C

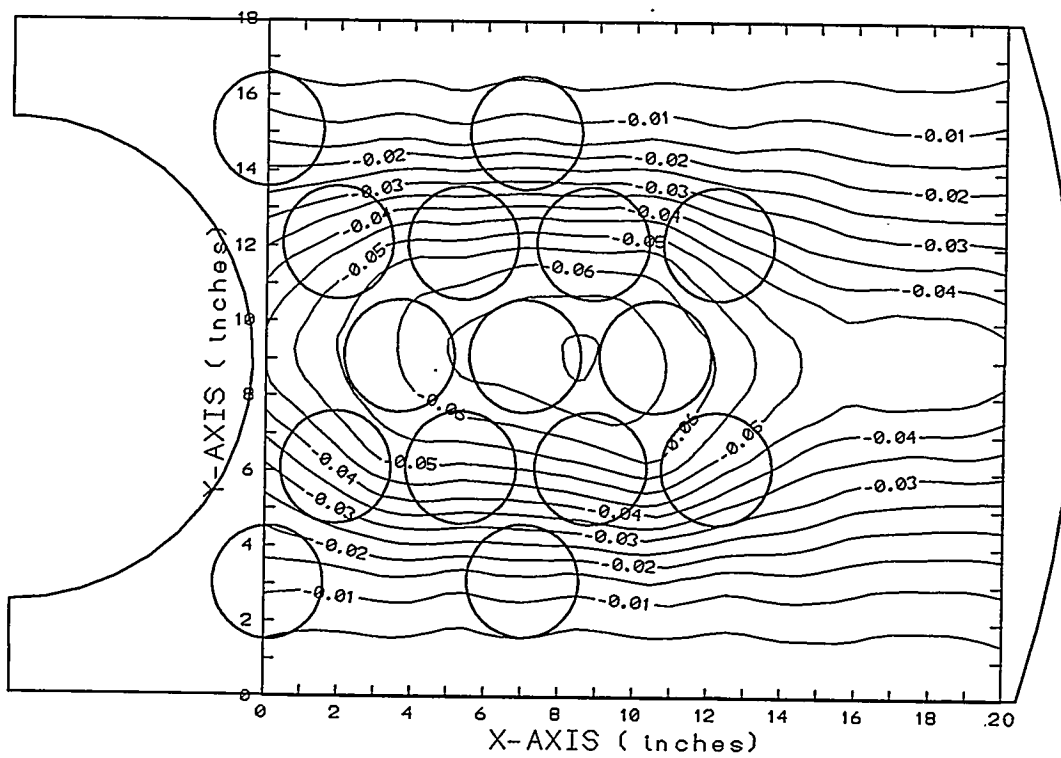


Figure 4-36. Overlay of topographic map with a drawing of the Fibrosic tubesheet

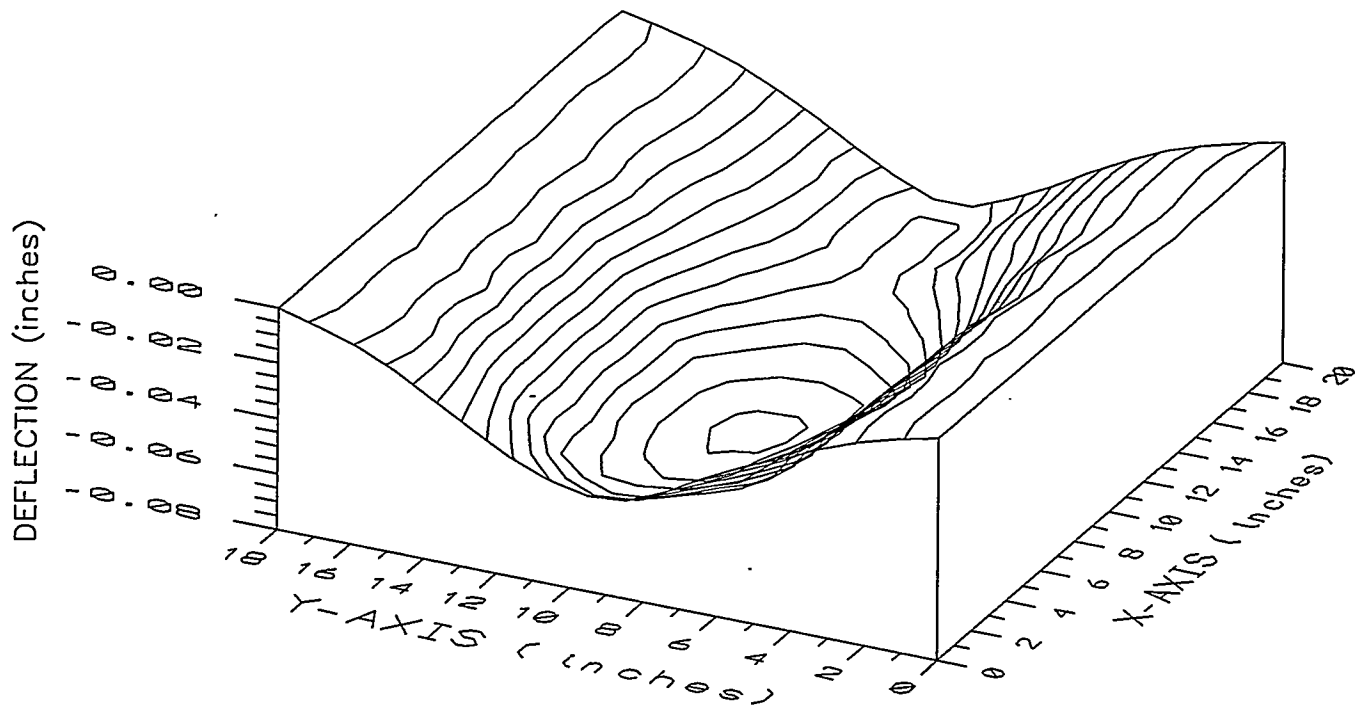


Figure 4-37. Three dimensional representation of the Fibrosic tubesheet surface

## SECTION 5

### CHEMICAL TEST RESULTS AND ANALYSIS

Experiments at Virginia Polytechnic Institute (VPI) were performed (VPI were subcontracted and provided support to this project) to determine the response of commercially available ceramic particulate filters to the high-temperature environments of advanced coal conversion systems. The final reports prepared by VPI are supplied in Appendix H. The following is a summary of some of the analyses performed by VPI. The SiC candles examined were the Schumacher Diaschumalith, Didier SiC, and the IFP LayCer candle filters.

The SiC candle filter specimens were exposed to alkali-containing atmospheres at temperature ranging from 450 to 1,225°C and pressures ranging from atmospheric to 1,000 psi (68 atm) for up to 24 hours. An alumina crucible containing either  $\text{Na}_2\text{CO}_3$ , or  $\text{K}_2\text{CO}_3$  was placed in a box furnace and surrounded by test specimens. Upon firing, the alkali vaporized within the test environment.

At the end of 6-, 12-, and 24-hour exposures, to the evaporation of  $\text{Na}_2\text{CO}_3$  and  $\text{K}_2\text{CO}_3$  at atmospheric pressure, and temperatures of 925 and 1,225°C, the SiC candle filter specimens displayed evidence of attack. The Didier specimens appeared slightly bloated and were coated with a glassy phase. The Diaschumalith specimens were also coated with a glassy phase and bloated, with the fibrous membrane areas of these specimens being more severely bloated. The membranes reacted more severely to the  $\text{K}_2\text{CO}_3$  environment than to the  $\text{Na}_2\text{CO}_3$  environment. Energy Dispersive X-ray analysis (EDX) identified the bloated membrane areas as 10 to 15% alkali, whereas little or no alkali was detected in interior portions of the samples.

In hydrothermal testing, a filter sample was sealed in a platinum tube containing the required amount of  $\text{H}_2\text{O}$  to generate the desired steam atmosphere. The sealed tube was placed in a pressure cell purged with argon gas, which served as a pressure medium and also prevented oxidation of the test cell. The pressure on the sample was applied through the steam generated inside the platinum tube. The total pressure at the testing temperature was calculated using the ideal gas law.

Following exposures to steam at 700 to 1,000°C and pressures of 200 psi, 500 psi, and 1,000 psi for up to 20 hours, no visual evidence of degradation of any SiC specimens was observed. However, SEM examination of the Didier specimens indicated that melting and flowing of the bond phase had occurred in the steam atmosphere at 700°C and 200 psi and became more prominent with increasing temperature at this pressure. SEM examination revealed that bond melting and depletion had occurred in the Diaschumalith, Didier, and LayCer specimens exposed to steam at 700°C, for the pressures of 500 and 1,000 psi.

Discoloration had occurred in the granular aluminosilicate following all steam exposure. However, XRD and SEM examination revealed no changes in specimens exposed at 700 to 1,000°C at 200 psi. Specimens exposed to 500 and 1,000 psi steam atmospheres at 1,000°C, however, were deteriorated and crumbled upon handling. Phase identification by XRD showed cristobalite formation.

The chemical degradation of clay-bonded SiC candle filters in alkali-containing atmospheres appears to result primarily from the alkali attack on the bonding phase of the filters. The filter specimens exposed to the alkali-containing atmospheres at temperatures from 700 to 1,000°C showed varying degrees of bond melting and flowing.

The kinetics of sodium diffusion in the bonding (binder) phase of the clay-bonded SiC filters were investigated to predict sodium penetration and accumulation in the binder phase when exposed to a sodium-containing atmosphere. The aluminosilicate binder phase was approximated by a 76  $\text{SiO}_2$  -16  $\text{Al}_2\text{O}_3$  -8  $\text{P}_2\text{O}_5$  (mol%) fireclay glass, and the specimens used were core-drilled glass rods

with dimensions of 6 mm in diameter and 5 mm long. Both ends of the glass pellets were polished flat to provide a defined boundary from which diffusion measurements were taken.

The alkali diffusion in the fireclay glass was evaluated at 800, 900, and 1,000°C. The thin film method was used for diffusion annealing at 800°C in which a drop of  $\text{Na}_2\text{CO}_3$  or  $\text{K}_2\text{CO}_3$  was evaporated on one end of the fireclay glass pellet, leaving a thin film containing Na or K sources. The coated pellets were heated isothermally at 800°C for 96 hours. The alkali diffusion at 900 and 1,000°C was carried out by exposing the fireclay glass pellets at a constant alkali partial pressure generated by either  $\text{Na}_2\text{CO}_3$  or  $\text{K}_2\text{CO}_3$ . The fireclay glass pellets in this case were wrapped in platinum foils so that diffusion occurred in only one direction. The alkali concentration of the exposed sample was measured as a function of diffusion distance using EDX. A series of spot analyses was performed along a line scan perpendicular to the exposed surface.

The resulting sodium diffusivities at 900 and 1,000°C are  $3.7 \times 10^{-8} \text{ cm}^2/\text{s}$  and  $7.1 \times 10^{-8} \text{ cm}^2/\text{s}$ , respectively. The potassium diffusivities at 900 and 1,000°C were determined to be  $7.5 \times 10^{-10} \text{ cm}^2/\text{s}$  and  $3.2 \times 10^{-9} \text{ cm}^2/\text{s}$ , respectively. The final reports supplied by VPI are in Appendix H.



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## SECTION 6

### MODELING

In attempts to provide modeling of the degradation mechanisms which predict candle degradation as well as candle lifetimes, attempts were made using theoretical framework and empirical observations of the data.

Solgasmix-PV is a chemical equilibrium program used for predicting the concentration of gas phase alkali species given the temperature, pressure, and the flue gas composition. The results of the program were to be used as input into the Virginia Polytechnic Institute (VPI) analyses to predict the lifetime of candle filters.

VPI developed a model based upon material loss due to reaction with alkalis. The reaction kinetics of mullite, cordierite, and aluminum titanate were examined in the presence of alkalis (in the form of  $\text{Na}_2\text{O}$ ) via solid state reactions. The full report by VPI is presented in Appendix H. The basic assumptions of VPI for this model were:

- (1) Alkali reactions originate from the surface
- (2) Alkali concentration in the atmosphere is equal to the alkali content in the solid state reactions
- (3) New surface is exposed due to the spalling of the reacted layer
- (4) No external forces applied
- (5) Material loss rate is equal to the alkali reaction rate

Summary of the VPI model are as follows:

- (1) Time for consumption of 50% of the available mullite at  $833^\circ\text{C}$ : 40 minutes
- (2) Time for consumption of 50% of the available mullite at  $1,065^\circ\text{C}$ : 2 minutes

- (3) Time for consumption of 50% of the available cordierite at 833°C: 20 minutes
- (4) Time for consumption of >60% of the available cordierite at >927°C: <2 minutes
- (5) Time for consumption of  $\approx$ 30% of the available cordierite at 833°C: 8 minutes
- (6) Time for consumption of 100% of the available cordierite at 927°C: 60 minutes

The VPI model predicted that major candle changes would occur within minutes of exposure to an alkali atmosphere at high temperature and implies that the lifetime for the candle filter may be very short. The VPI model showed a very rapid penetration and consumption of alkali into the candle material. Attempts to correlate the VPI model with results from the thermal exposure tests proved unsuccessful. The poor correlation may have been due in part to the assumption that alkali uptake was only diffusion limited. The VPI model predicted major candle degradation to occur within a time scale of minutes, but the thermal/chemical exposure tests showed the candle degradation to occur on a time scale of days. The application of the VPI model probably is appropriate for the initial degradation of the candle (which shows a rapid drop in strength), but it apparently does not describe the long term degradation mechanisms which occurs gradually over a period of days.

In addition to the analytical modeling, attempts were made to relate empirical observations from the c-ring tests to knowledge of fundamental material characteristics and degradation mechanisms. Examination of the data as well as the qualitative behavior of the candles during testing may provide insight into the characteristics of the candles during operational conditions which may in turn produce a useful model of the contribution of thermal/mechanical stresses compared to chemical degradation mechanisms.

In examining the strength of unused Schumacher and Refractron candles at various temperatures, it was determined that the candles did not retain structural integrity at 980°C. Figures 6-1 and 6-2 shows the strength (via c-ring tests) of unused Schumacher Diaschumalith F40 and Refractron 70/3 with 442-T binder candles at different temperatures. The decrease in strength could be attributed to the softening of the candle/binder material (refer to section 4.1.2 above).

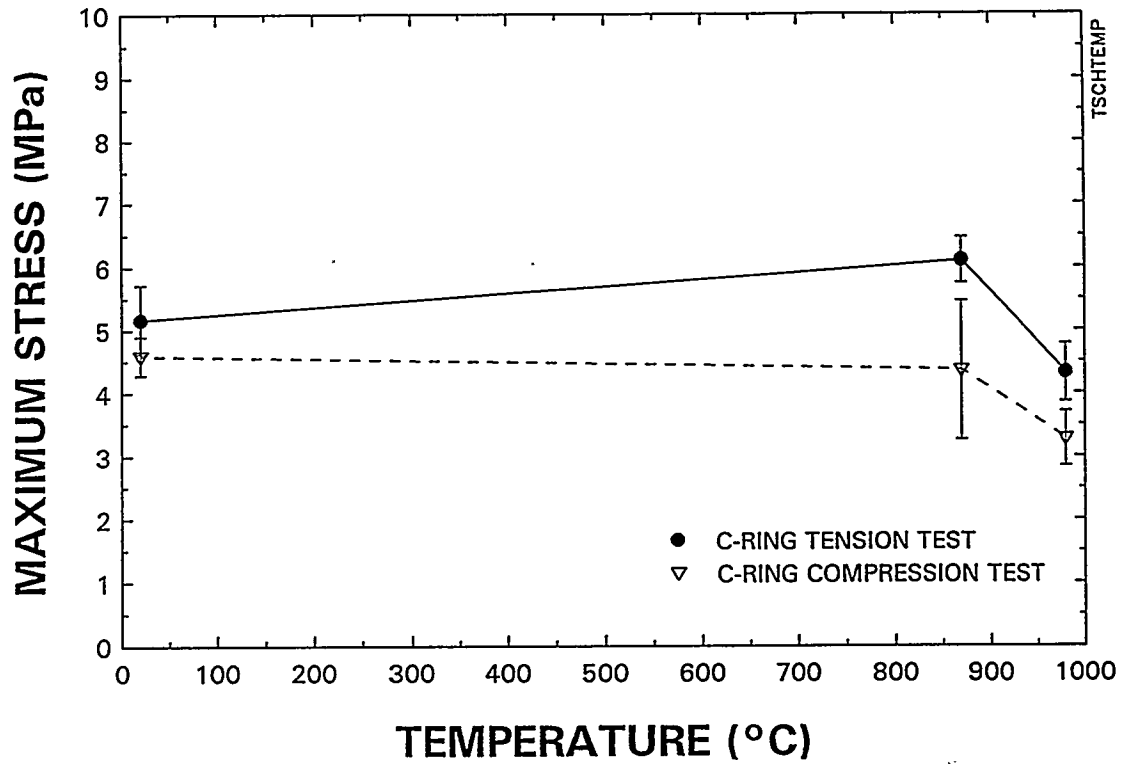


Figure 6-1. Unused Schumacher Diachumalith F40: c-ring tension and compression tests at various temperatures

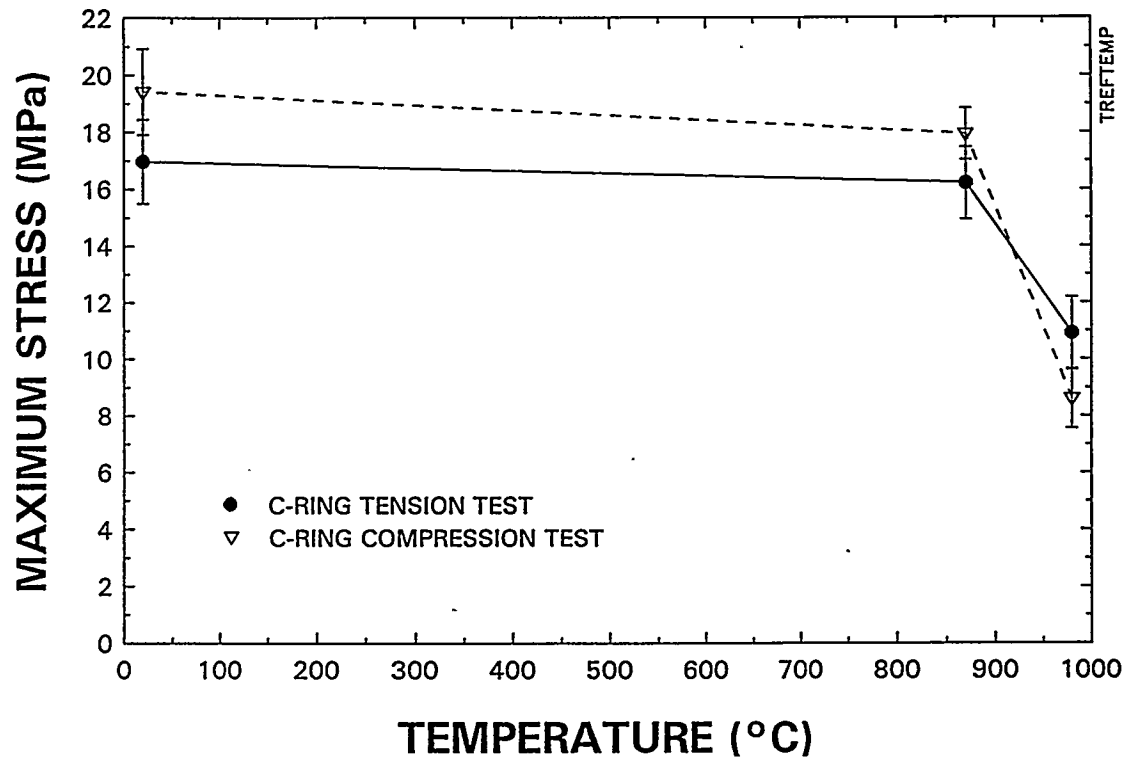
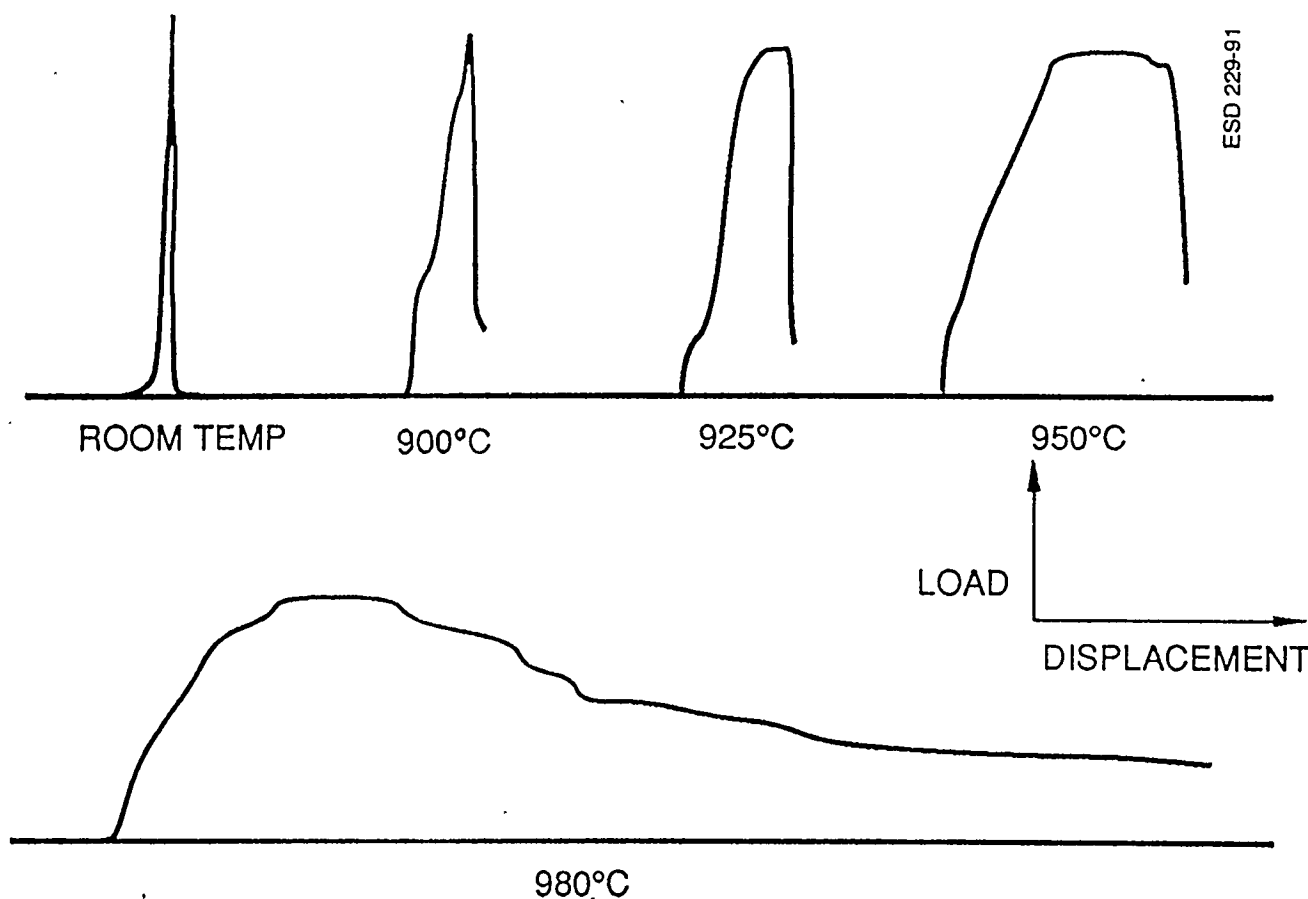


Figure 6-2. Unused Refraction 70/3 with 442-T binder: c-ring tension and compression at various temperatures

When the softening temperature of the candle material is exceeded, the microcracks may be blunted and increase the toughness, but the ultimate strength of the candle may also decrease as a result. This viscoplastic behavior of the candles was observed at 980°C in both the Schumacher and the Refractron candles. At 980°C, there was not the characteristic brittle failure (elastic up to the fracture stress) expected from a ceramic material. Brittle fractures of the c-rings were (generally) observed for specimens tested at 870°C and below. This would imply that the upper limit for the observed "healing" of candles (when tested at a crosshead rate of 0.05 in/min) may be limited to temperatures ranging from 870°C to 980°C. Further testing was performed to see if the transition temperature to viscoplastic behavior could be determined. A quick test in which one specimen was tested for each of the selected intermediate temperatures for the Schumacher candle were performed. Figure 6-3 shows the load versus displacement curves for this series of tests.

From Figure 6-3, it appears that  $\approx 920^{\circ}\text{C}$  is the transition temperature for viscoplastic flow at a rate of 0.05 in/min. A substance is viscoplastic if it exhibits both energy dissipation and energy storage in its mechanical properties (it would be viscoelastic if the deformation was recoverable after removal of all stress)<sup>14</sup>. For viscoplastic materials, the rate of testing would greatly influence the results.

Even at temperature as low as 500°C, stress relaxation in the c-rings was observed (this would make creep deformation during long-term use a concern). This observation is based on loading a c-ring to a stress level below the failure stress and stopping the crosshead motion. At room temperature the c-ring (an unused Schumacher candle) sustained the load; but above 500°C, the load dropped with time. When specimens display deformation (displacement of crosshead) without the expected linear increase in load characteristic of an elastic material, creep is said to have occurred. The flow of the material (probably due to binder softening) allowed stress release and thus reduced the structural integrity of the candle filters. Since this behavior is rate dependent, it is highly probable that changing the crosshead testing speed would result in a transition temperature different from the 920°C obtained from the crosshead speed of 0.05 in/min.



**Figure 6-3. Load versus displacement curves for the Schumacher F40 at various temperatures**

Figure 6-4 shows the percentage change of the strength of Schumacher candles from an unused Schumacher candle when tested at 870°C. The candle that is only pulse cleaned (not exposed to any chemical degradation) shows about a 15% decrease in strength after 5,000 "0.5 second" pulse cleanings. The candle that was thermal/chemically exposed to 20 ppm NaCl, 870°C, 15% steam, and pulse cleaned with 0.5 second pulses (every 30 minutes) shows a quick decrease in strength initially, but the degradation in strength appears to level off to approximately the same slope as the "pulsed only" candle. The thermal/chemical exposure candle specimens extends up to about 2,300 pulses (over 1,000 hours of exposure) and the remaining pulses on the graph were extrapolated out.

Figure 6-4 implies that the candle degradation is initially controlled by its chemical interaction with the candle environment, but as equilibrium is reached/approached, the thermal

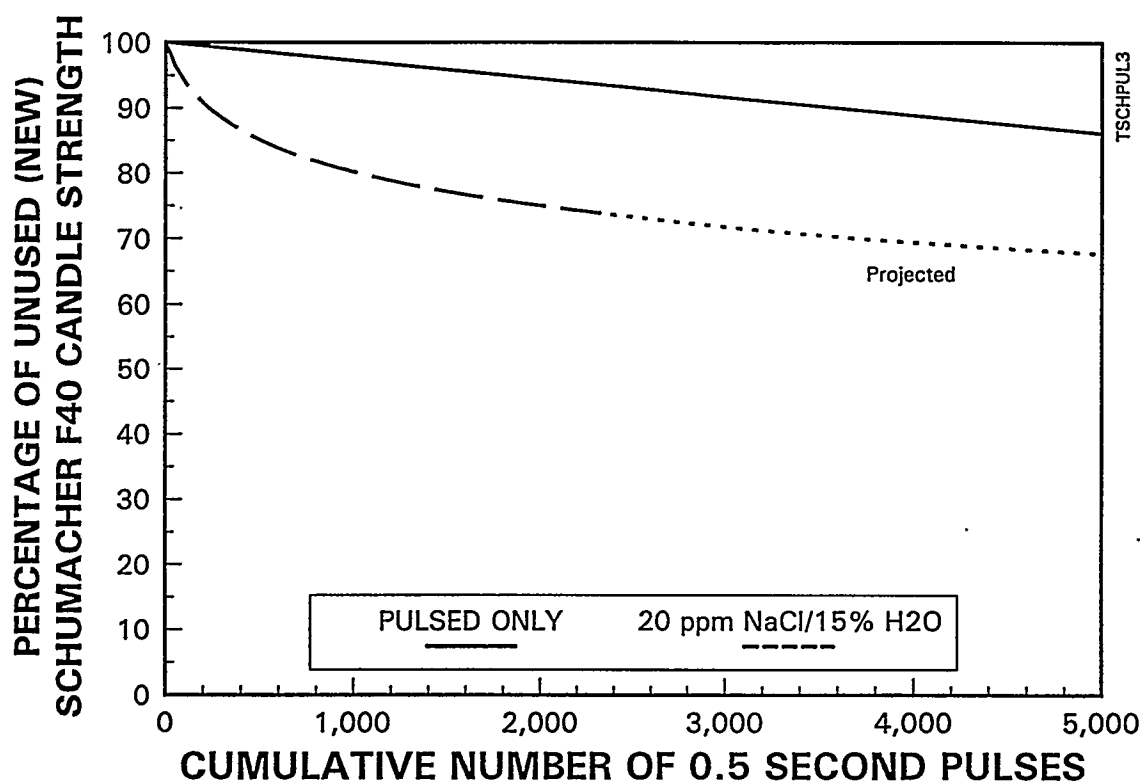
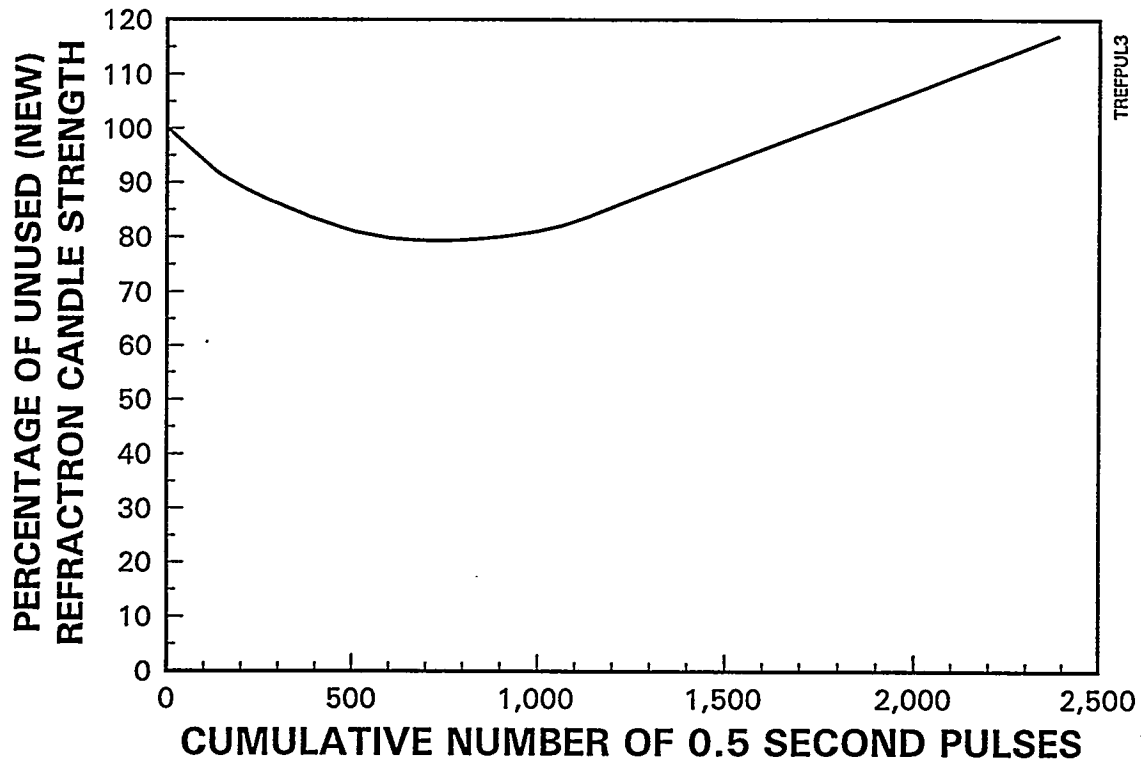


Figure 6-4. Schumacher F40/c-ring tension at 870°C: Strength loss due to thermal and thermal/chemical degradation

degradation contribution (from pulse cleaning) to the candle's overall degradation becomes more significant.

Modeling of the degradation mechanisms of the candle filters is complex. As it can be seen in Figure 6-5, the strength of the Refractron 70/3 with 442-T binder appears to decrease with exposure at the 20 ppm NaCl, 870°C, 15% steam, and pulse cleaned (2,400 times with 0.5 second pulses every 30 minutes) condition, but it increases up to a strength greater than an unused candle. A possible explanation of this phenomena is the formation of albite (see Section 4.3). Albite apparently affects the characteristics of the Refractron candles. It had not been detected in any of the previous studies with the Schumacher candles. Another possible explanation of the increase in strength of the Refractron candles with exposure is the possible increased softening of the candle to the point where the candle material flowed (although major deformation of the candle at 870°C was not observed). Material could have flowed to a point where the candle became denser thus



**Figure 6-5. Refractron 70/3 w/442-T/c-ring tension at 870°C: Strength change due to thermal and thermal/chemical degradation**

reducing porosity. This could cause the strength of the material to increase. Unfortunately, no tests were performed on the candle porosity and permeability. In another project performed by Acurex Environmental, Refractron 50/5 with 442-T binder candles exposed at the Solar Turbines Direct Coal-Fired Turbine Test Facility also showed the unexplained phenomena of increasing strength at high temperature.<sup>15</sup>



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## SECTION 7

### CONCLUSIONS AND RECOMMENDATIONS

#### 7.1 CONCLUSIONS

Ceramic materials used in hot gas cleaning applications will be exposed to conditions that may degrade strength and limit service life due to thermal and chemical degradation. Further, the mechanical limitations of ceramic materials pose significant problems when such materials are exposed to dynamic environments. Based upon the tests and analyses of this project, the following conclusions can be stated.

##### C-RING TESTS

- C-ring testing proved to be the desired mechanical testing method for rigid candle materials because of the c-ring specimens retain the surface flaw population. For example, the c-ring specimens showed strength degradation for the KRW exposed candles, whereas the 4-point flexure tests did not show any strength loss.
- C-ring testing allows for separate evaluation of inner and outer surface strengths to investigate strength degradation from different environmental conditions.
- C-ring widths from 0.5 to 1.25 inches are representative of the strength of the candle material. The average maximum stress values (strength values) are more stable with widths greater than or equal to 0.75 inch.
- The optimum c-ring width is 0.75 inch. The 0.5-inch specimen is more fragile than the 0.75-inch specimen. Using c-ring widths greater than 0.75 inch is an inefficient use of available material.

- C-ring testing of the homogenous Didier SiC20 candle validated the c-ring stress equation because the strength for the inner and outer surfaces is the same.
- Machining of c-ring specimens during c-ring preparation does not significantly affect the strength measurement of the c-rings.

#### **THERMAL RATCHETING**

- Strength degradation due to room temperature air pulse cleaning of candles at high temperatures (870°C) occurs with pulse durations greater than 0.05 second. Pulse durations of 0.05 second did not produce strength degradation.
- Strength degradation occurs with 0.5 second room temperature air pulse cleaning of candles at high temperatures (870°C).
- At high temperatures, the SiC based ceramic candles showed increased toughness due to binder softening which blunted microcracks (relieving stress at the crack tips) resulting in increased strength.

#### **THERMAL/CHEMICAL EXPOSURE**

- The flow-through alkali thermal/chemical exposure experiments showed contamination from metals, such as Fe and Cr, from the corrosion byproducts of the vessel and pre-vessel piping.
- There was evidence the binder became a viscous liquid at operating temperatures and flowed under gravity and aerodynamic drag to blind some of the pore structure of the exposed candles.
- The modeling of the thermal/chemical degradation mechanisms to predict filter lifetime was unsuccessful.
- Both the Schumacher Diaschumalith F40 and the Refractron 70/3 with 442-T binder candles did not retain their integrity when exposed at 980°C.

- Initially, the dominant degradation mechanism appears to be attributed to chemical degradation (alkali, metals, and/or steam), but the thermal/mechanical degradation (pulse cleaning) becomes the primary degradation mechanism.

#### Schumacher Diaschumalith F40

- Strength degradation was observed after 100 hours of exposure to 95 ppm NaCl, 20% steam, 870°C, and without pulse cleaning.
- Strength degradation was observed due to 65 ppm NaCl, 20% steam, 980°C, and with pulse cleaning.
- Strength degradation was observed only after 1,000 hours at 20 ppm NaCl, 15% steam, 870°C, and with pulse cleaning.
- Cristobalite was detected only in the exposed candles at 980°C. No cristobalite was detected in the unused Schumacher candle nor the candles exposed at 870°C.
- Crystalline silica was, however, detected in the exposed 870°C and 980°C candle specimens.

#### Refractron 70/3 with 442-T binder

- Strength degradation was observed after 100 hours at 20 ppm NaCl, 15% steam, 870°C, and with pulse cleaning. Increase in strength for the 500-hour and 1,000-hour specimens was seen at the inner surface only.
- Strength degradation was observed due to 20 ppm NaCl, 15% steam, 980°C, and with pulse cleaning.
- Albite formation was detected only in the exposed Refractron candles.

#### FIBROSIC CANDLE TESTS

- Both Fibrosic candles had collection efficiencies greater than 99.99% during and after the 15,000 pulse cleaning cycles and after 100-hour operational filtration testing.
- The unused Fibrosic candle (Fibro2800) tested in this project has a burst pressure of 20 psig as compared to the 16 psig tested in an earlier version of the Fibrosic candle.

- There was about a 25% drop in burst pressure for both of the Fibrosic candles tested as compared to the 50% burst pressure loss for the earlier version of the Fibrosic candle.
- Although a crack was discovered on the candle exposed to accelerated pulse cleaning, it did not affect the burst test strength of the candle (the failure occurred away from the crack).
- The permeability of the Fibrosic candle remained high throughout the tests.

#### **FIBROSIC TUBESHEET TEST**

- Minor cracks on the bottom side of the tubesheet were observed after loading with 113 pounds (using worst-case scenarios) and heating to 870°C for 500 hours.
- The major displacement of the tubesheet during load testing at 870°C occurred within the first 24 hours.
- Deflection in the tubesheet was still observed after 500 hours (but at a much slower rate than the first 24 hours).
- The Fibrosic tubesheet deflection/sag of about 0.1 inch (measured at the lowest point across the width of the tubesheet).

#### **7.2 RECOMMENDATIONS**

At the conclusion of this project, the following recommendations can be made:

- The results from the Fibrosics filtration testing look promising (high collection efficiencies, high permeability, and no catastrophic failure), but there was only one candle available for testing. Therefore the results are not statistically representative for describing the candle characteristics and properties under operating conditions. More candles should be tested to confirm the burst strength values and to determine if the crack discovered on the accelerated pulse cleaning candle specimen was an anomaly.
- Cylindrical burst strength tests need to be investigated to define the limitations, constraints, and applicability for strength measurements of candle filters. This is an

issue because the burst strength test performed on a cracked candle did not fail at the site of the crack.

- Collection efficiency tests should be performed separately from permeability measurements to examine if the flyash used during collection efficiency tests affects the permeability measurements.
- Future studies on the effect of thermal/chemical degradation on candle filters should also include the measurement of porosity and permeability (possibly including collection efficiency) to ascertain whether the candle material retains filtration properties.
- Chemical modeling studies need to consider more than the diffusion limited case in the uptake of alkali. The modeling studies should also include kinetic mechanisms such as retention of some corroded material which will affect the surface area of new material for the continuation of alkali corrosion.
- Contact of alkali with piping materials leading into the thermal exposure vessel should be minimized, thus minimizing the exposure to corrosion byproducts that may affect the candle filters being tested.
- Extended exposure times to flow through testing at temperature without pulse cleaning. This data can be compared to the thermal/chemical degradation testing of candle filters which received thermal pulse cleaning at high temperature.
- A minimum of 5 c-ring specimens should be tested at each test condition in order to have a statistically significant strength value (having more specimens is desirable).

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**APPENDIX A**  
**THERMAL EXPOSURE SAMPLE VESSEL DRAWINGS**

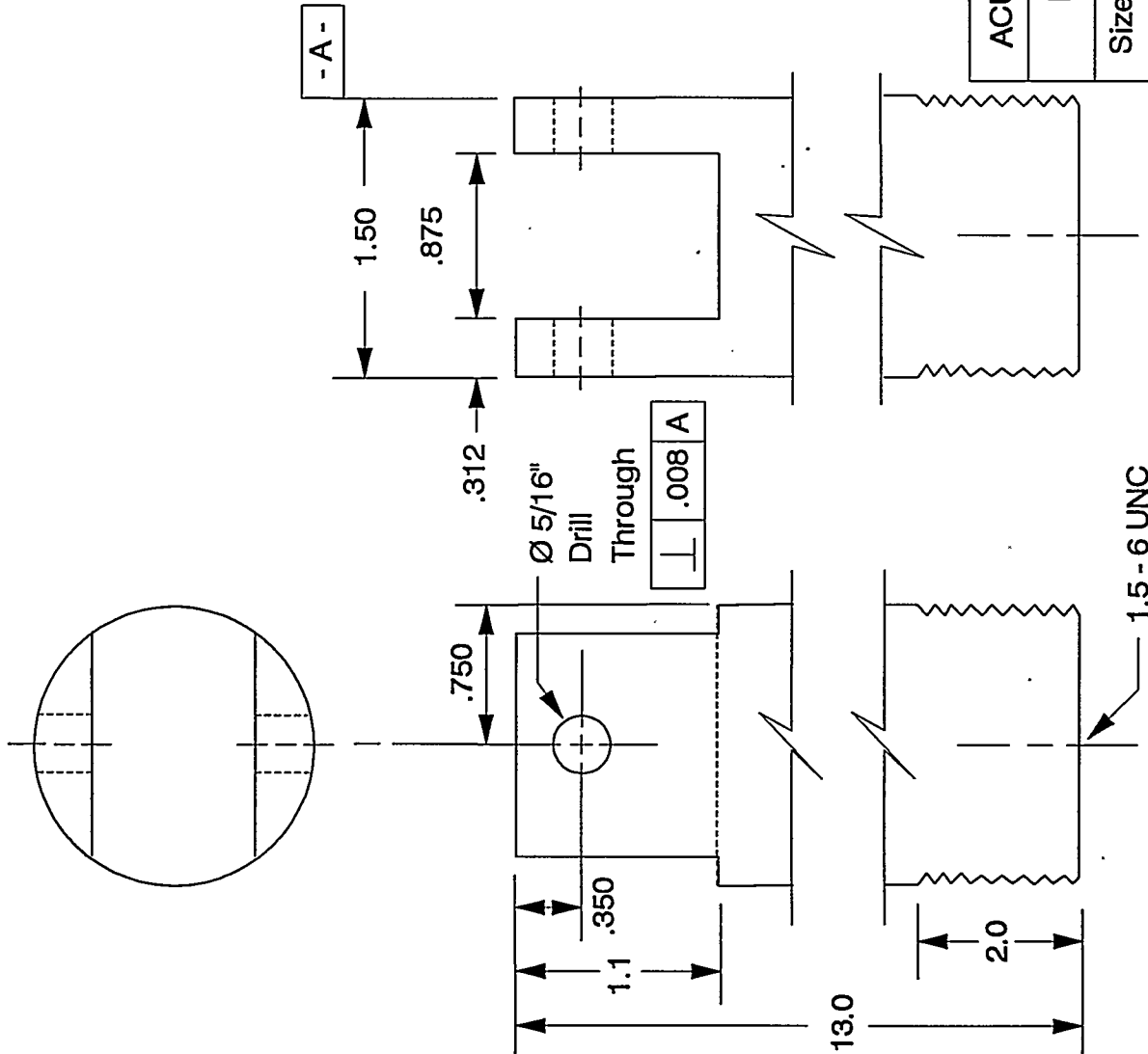
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**APPENDIX B**  
**DRAWINGS OF C-RING TESTING FIXTURES**

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otherwise specified:

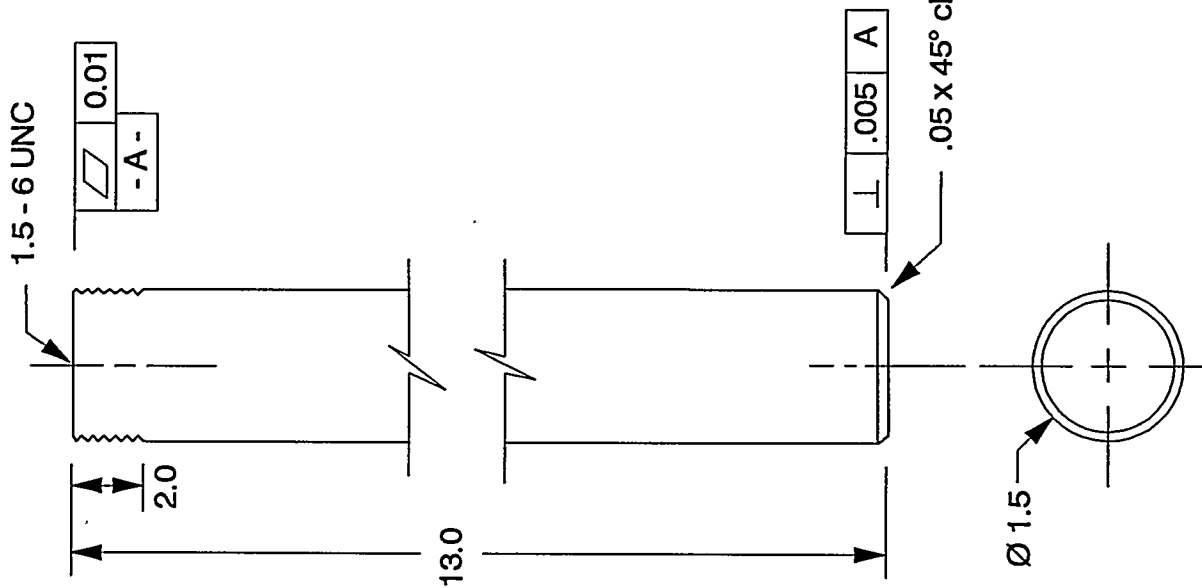
.XXX  $\pm$  .005  
.XX  $\pm$  .010  
.X  $\pm$  .10



Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
LOWER RAM	
Size A	Project Number: 8146
SHEET OF	DWN:



All tolerances unless  
otherwise specified:  
.XXX ± .005  
.XX ± .010  
.X ± .10

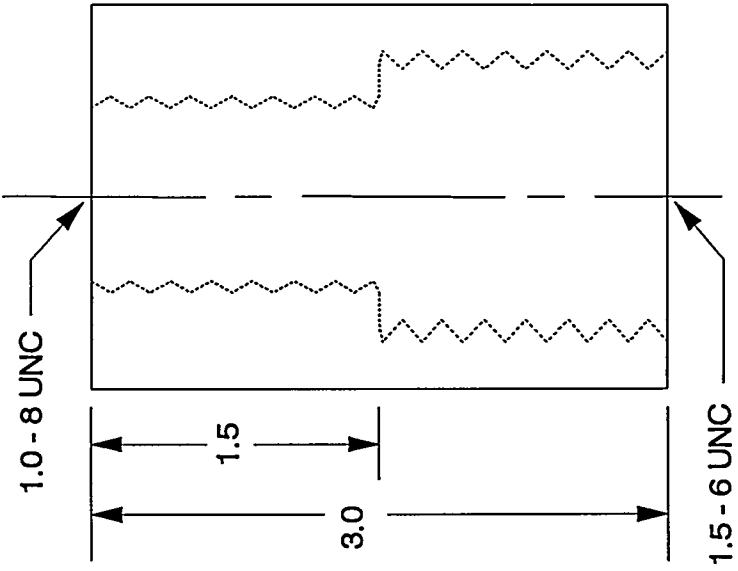
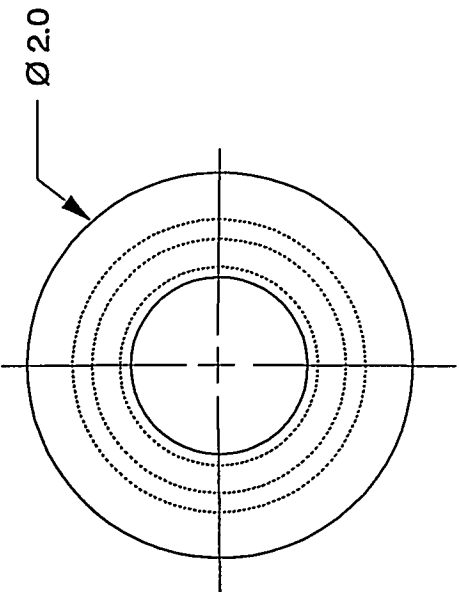
Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
UPPER RAM FOR COMPRESSION TEST	
Size A	Project Number: 8146
SHEET OF	DWN:

All tolerances unless  
otherwise specified:

- .XXX ± .005
- .XX ± .010
- X ± .10

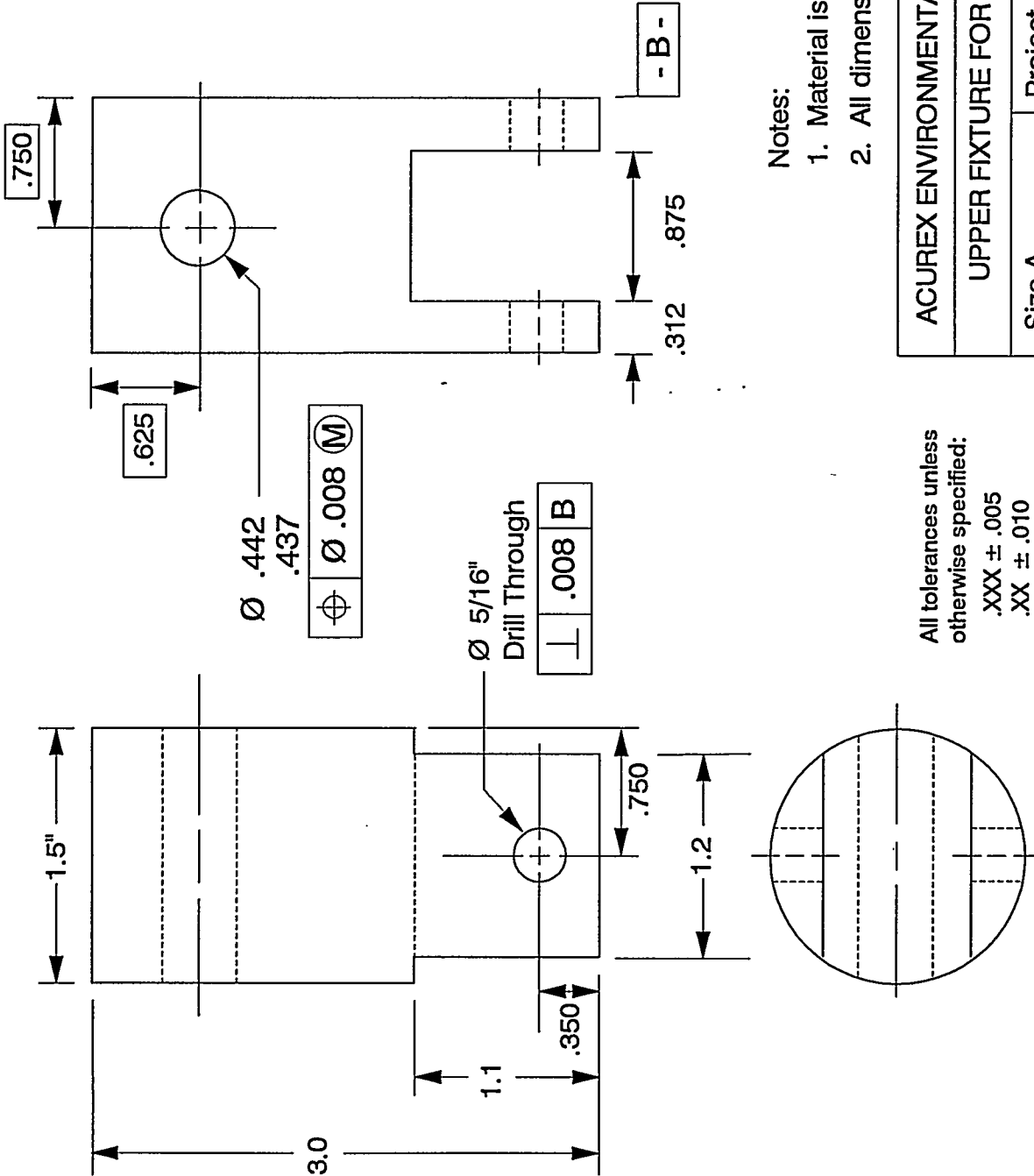


Notes:

1. Material is
2. All dimensions in inches
3. Quantity required is 2

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ADAPTER COUPLE FOR UPPER RAM (COMPR.)	
Size A	Project Number: 8146
SHEET OF	DWN:





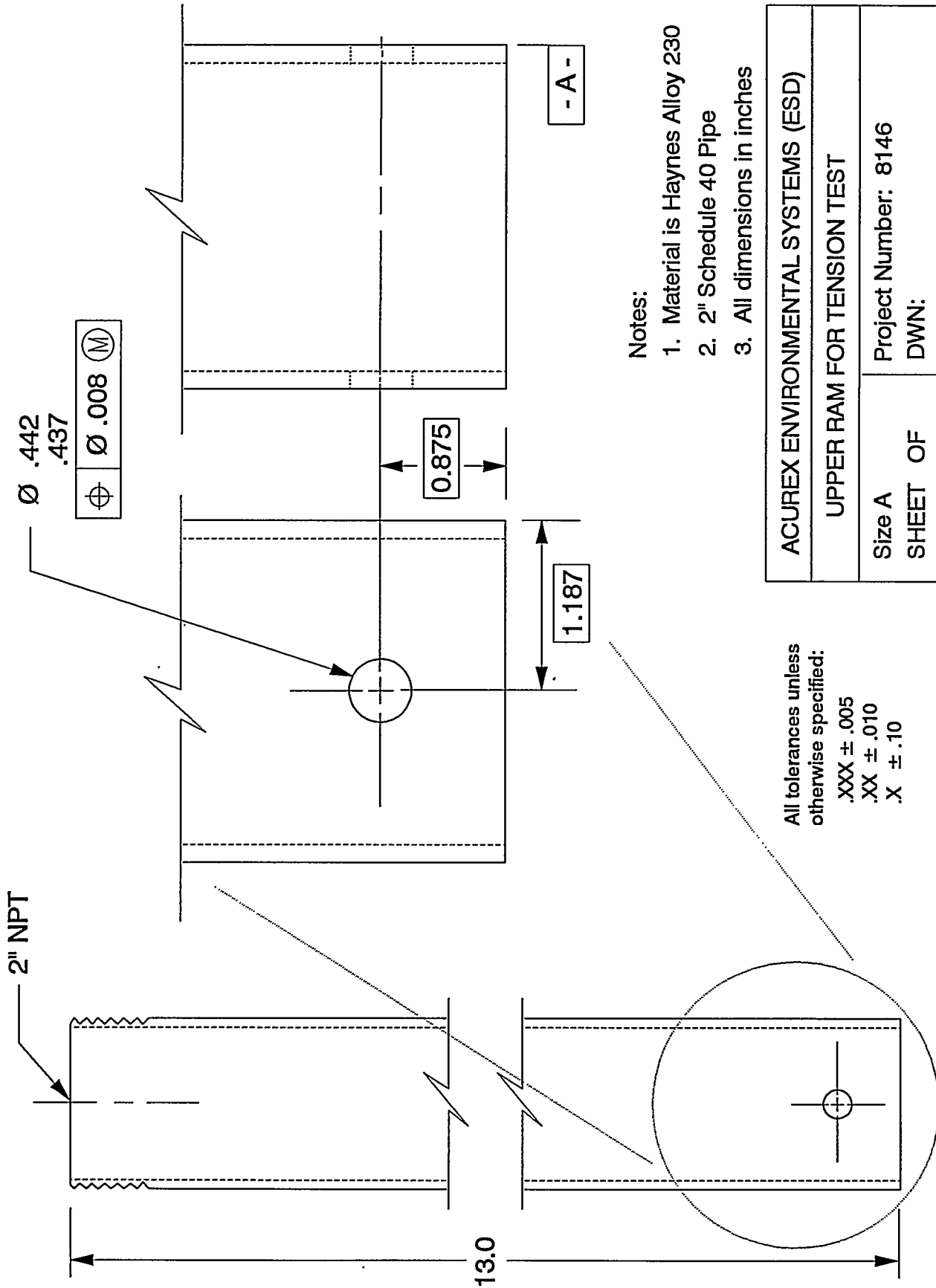
Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

All tolerances unless  
otherwise specified:

.XXX  $\pm .005$   
.XX  $\pm .010$   
.X  $\pm .10$

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
UPPER FIXTURE FOR TENSION TEST	
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SHEET OF	DWN:

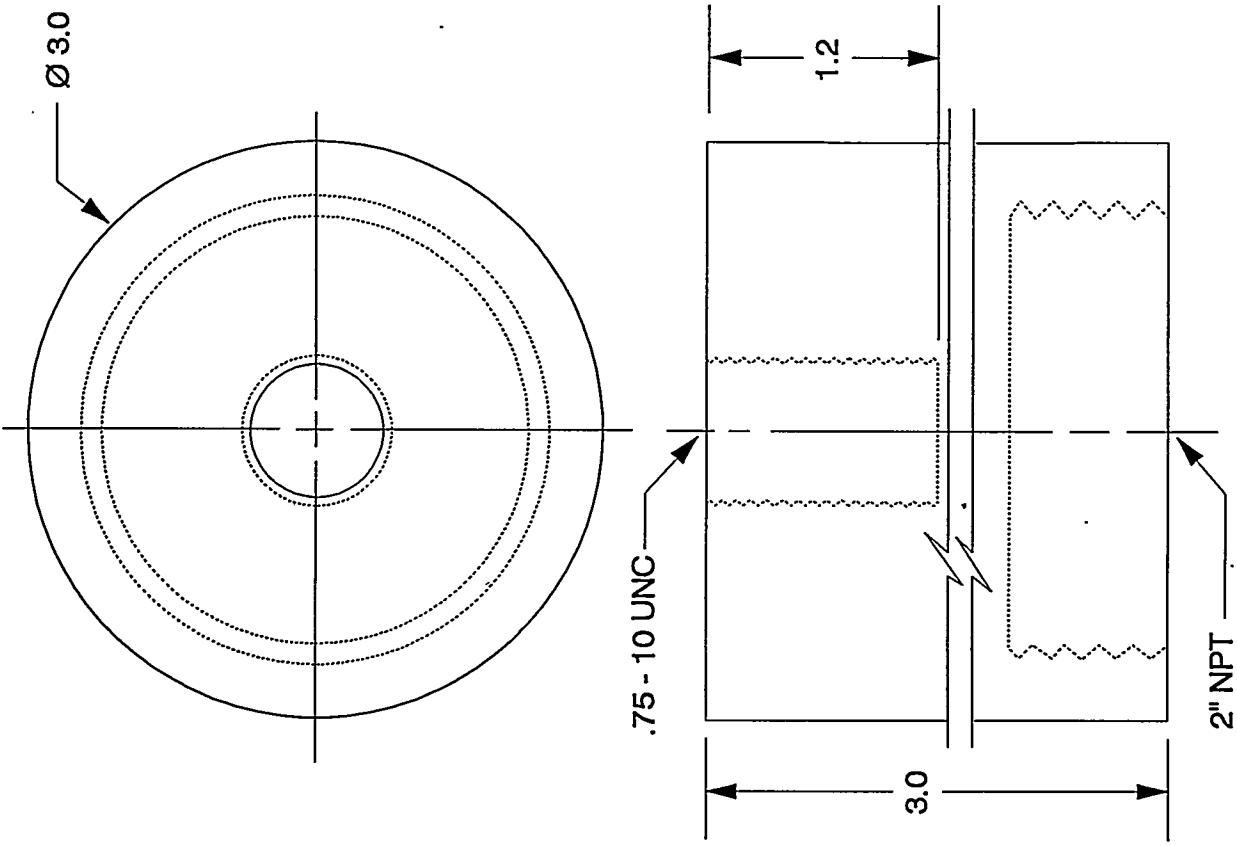


Notes:

1. Material is Haynes Alloy 230
2. 2" Schedule 40 Pipe
3. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)			
UPPER RAM FOR TENSION TEST			
Size A	Project Number: 8146		
SHEET OF	DWN:		

All tolerances unless  
otherwise specified:  
.XXX ± .005  
.XX ± .010  
.X ± .10



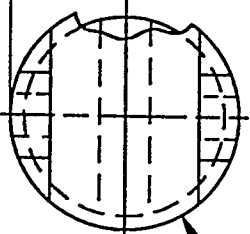
Notes:

- 1. Material is
- 2. Female 2" NPT for a 2" Schedule 40 Pipe
- 3. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
ADAPTER COUPLE FOR UPPER RAM (TENSION)	
Size A	Project Number: 8146
SHEET OF	DWN:

DECIMALS  
.XX ± .02 .XXX ± .005

ANGLES  
± 5 DEGREES



Ø 2.00 REF STOCK

1.00

.75

.38

.38

.35

-A-

Ø 2.00 STOCK

1.30

.62

4.00

Ø .38 THRU

Ø .44

Ø 1.75

1.88

2.13


1.25

30°

MATERIAL:  
HAYNES ALLOY 230

ACUREX Corporation		Environmental Systems Division	
PROJECT		C-RING TESTING	
DWN BOYD		10/3/91	
CHK		10/3/91	
ENG CHAN		10/3/91	
APPD		10/3/91	
PROJECT NO. 8146		FULL	
SCALE		FULL	
ORIGINATOR: 8146		DATE: 10/3/91	
DRAWN: BOYD		DATE: 10/3/91	
CHECKED: CHAN		DATE: 10/3/91	
APPROVED: CHAN		DATE: 10/3/91	
TITLE		REFERENCE DRAWINGS	
NUMBER		8146-250	
DATE		10/4/91	
APPROVED		ADB	
DESCRIPTION		A RELEASED FOR FABRICATION	
REVISIONS		8146-250/REV A	
SIZE		B	
TEST		8146-250	
SHEET		1 OF 1	
REV		A	

**MATERIAL:**  
**HAYNES ALLOY 230**

				DWN	BOYD	10/3/91	 ACUREX Environmental Systems Division
				CHK			
				ENG	CHAN	10/3/91	
				APPD			
				PROJECT NO. 8146			PROJECT C - RING TESTING
				SCALE	FULL		
A RELEASED FOR FABRICATION				ADB	10/4/91		
DESCRIPTION				APPD	DATE		
REVISIONS				REFERENCE DRAWINGS			
8146 - 8146-251				NUMBER	TITLE		
REV				DATE	DWG NO		REV
1 OF 1				SIZE	B		1 OF 1 A
8146-251							
A							
</							

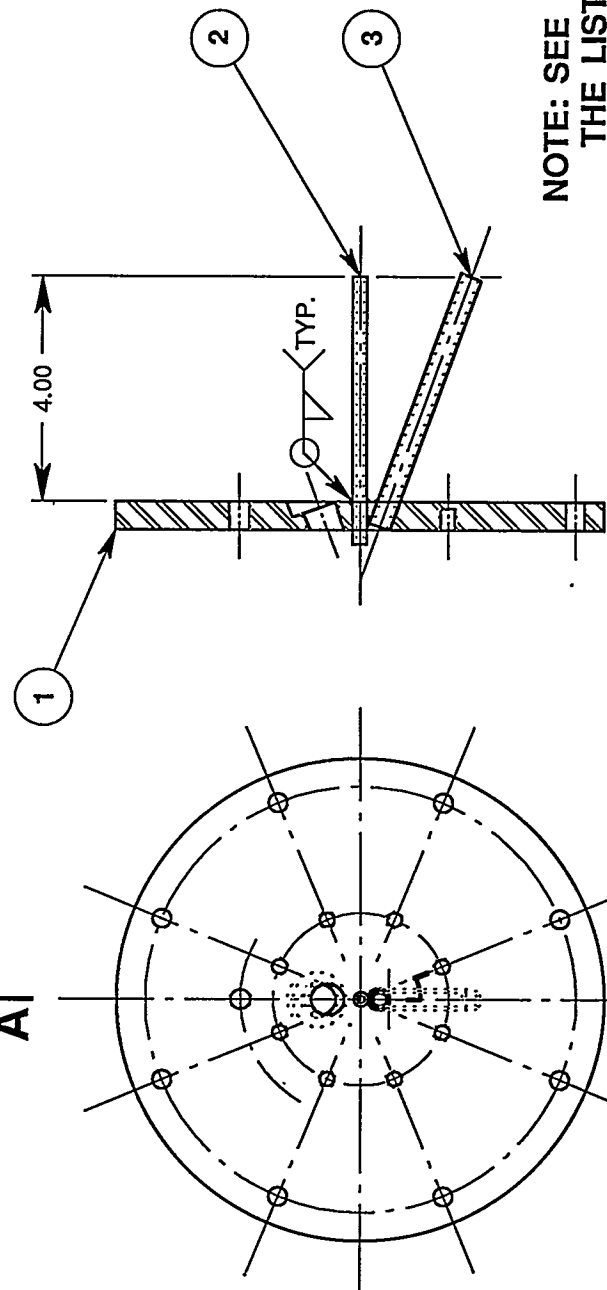


**NOTE: SEE LM/8146 - 100 FOR  
THE LIST OF MATERIALS  
SHOWN ON THIS DRAWING.**

# SECTION A - A

[illegible]





NOTE: SEE LM/8146-101 FOR THE LIST OF MATERIALS SHOWN ON THIS DRAWING.

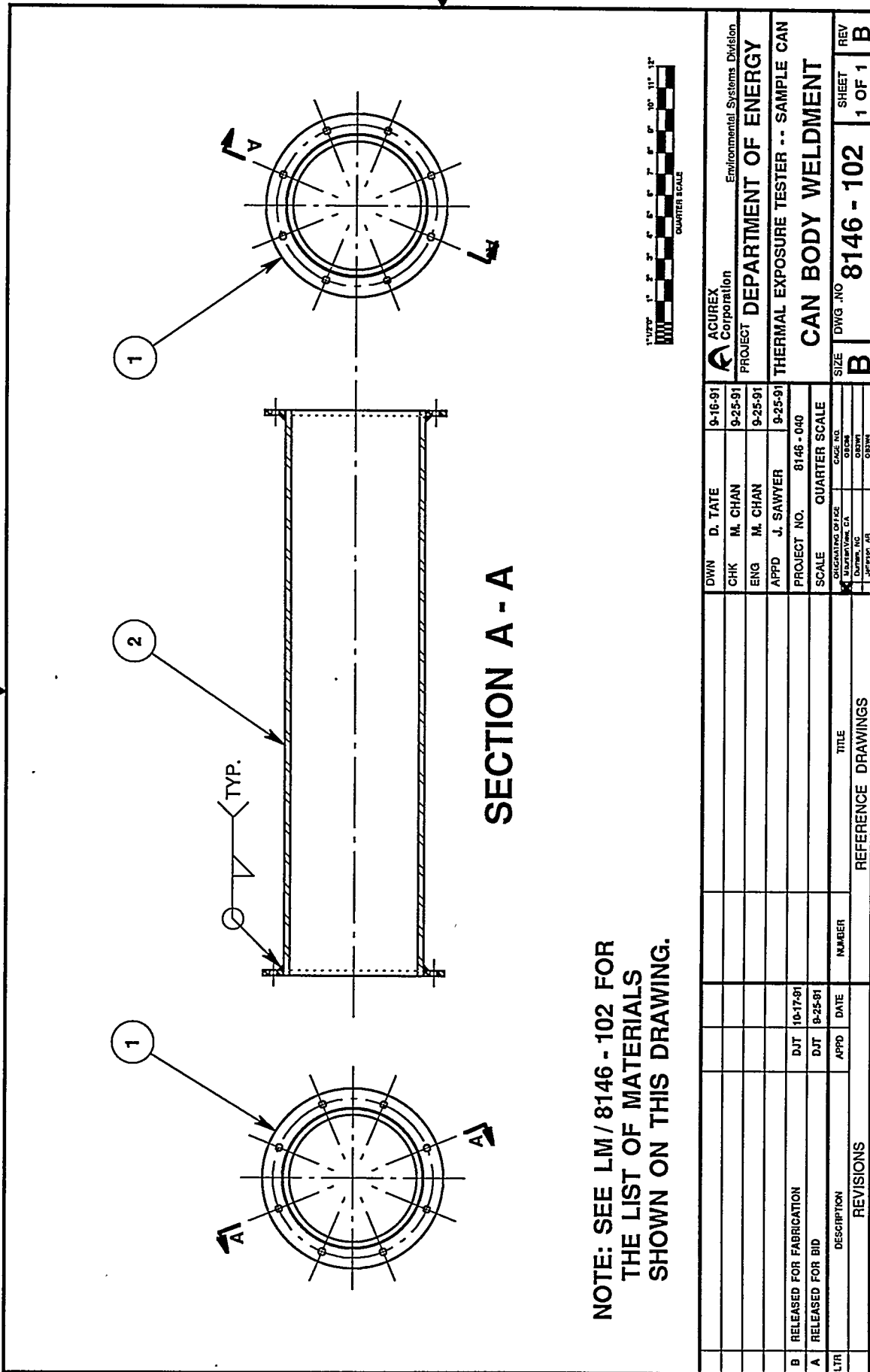
SECTION A - A




ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		OUTLET FLANGE WELDMENT	
SIZE		DWG. NO. 8146 - 101	
REV		SHEET 1 OF 1	
C		B	
DWN		D. TATE	
CHK		M. CHAN	
ENG		M. CHAN	
APPD		J. SAWYER	
PROJECT NO.		8146 - 040	
SCALE		HALF SIZE	
ORIGINATING OFFICE		CASE NO.	
Durham, NC		03304	
J. SAWYER, J.E.		03304	
TITLE		REFERENCE DRAWINGS	
NUMBER		DATE	
DESCRIPTION		APPROVED	
REVISIONS		REVISIONS	
LTH		C	
RELEASED FOR FABRICATION		DJT 10-17-91	
ADDED 3/8" NPT TAPPED & C-BORED HOLE		DJT 9-30-91	
RELEASED FOR BID		DJT 9-25-91	





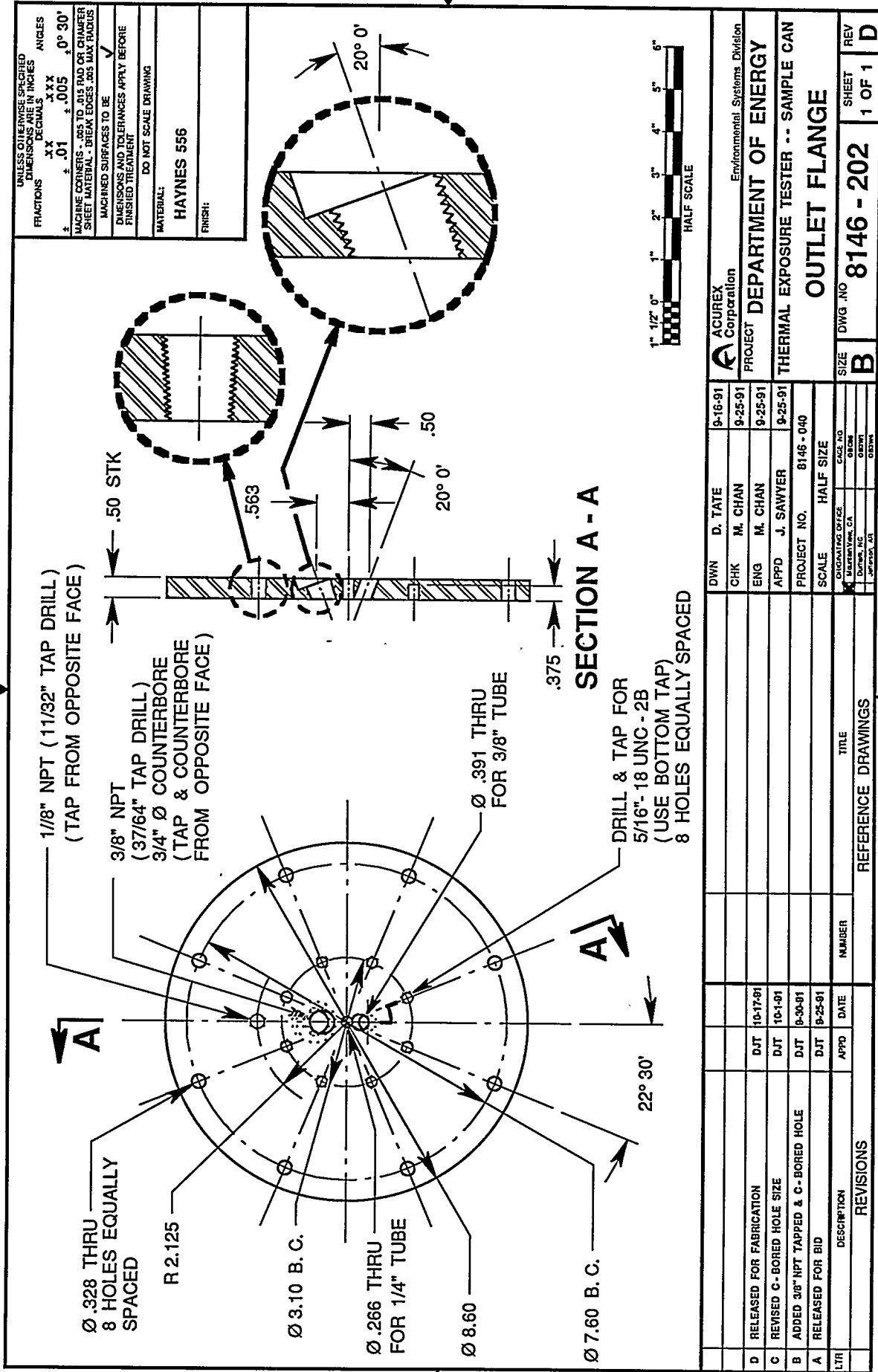


NOTE: SEE LM/8146-102 FOR  
THE LIST OF MATERIALS  
SHOWN ON THIS DRAWING.

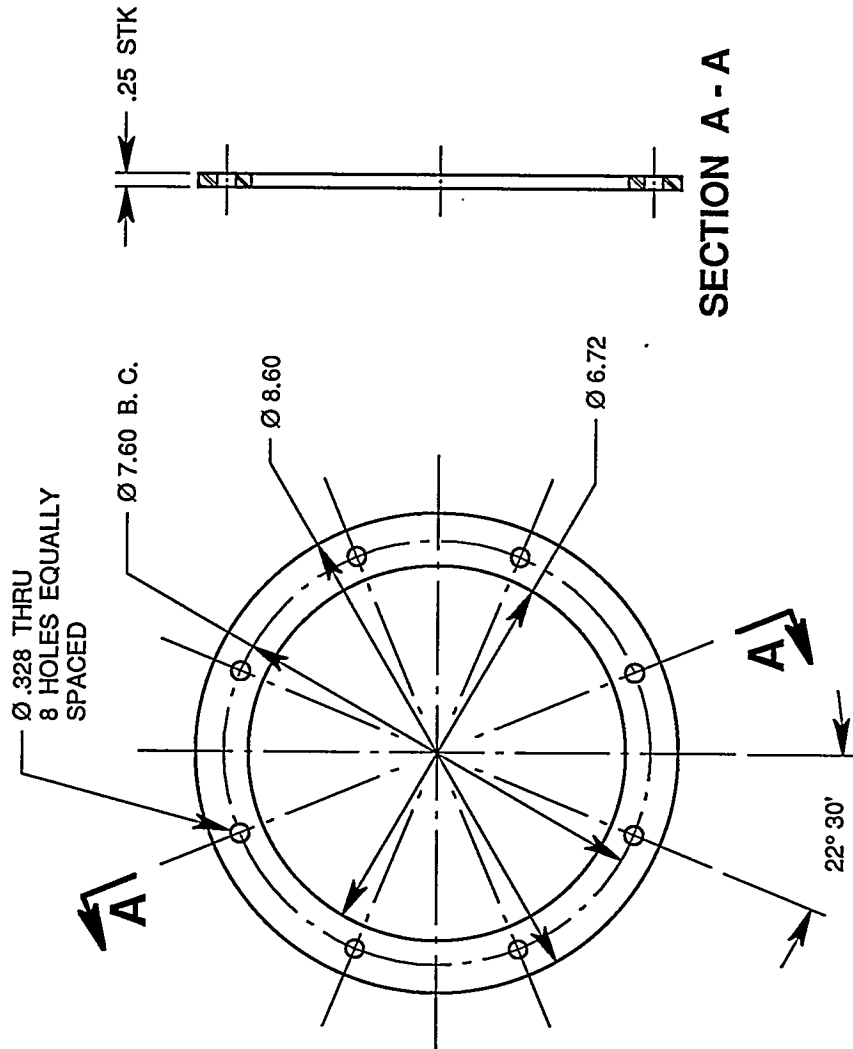
						DWN		D. TATE		9-16-91		 ACUREX Corporation		Environmental Systems Division	
						CHK		M. CHAN		9-25-91		PROJECT		DEPARTMENT OF ENERGY	
						ENG		M. CHAN		9-25-91		THERMAL EXPOSURE TESTER -- SAMPLE CAN			
						APPD		J. SAWYER		9-25-91		CAN BODY WELDMENT			
						PROJECT NO.		8146 - 040				SIZE		DWG. NO	
						SCALE		QUARTER SCALE				B		8146 - 102	
						OPERATING OFFICE						SHEET		REV	
						MOBILITY/CA		CA				1 OF 1		B	
						MOBILITY/CA		CA							
						MOBILITY/CA		CA							
						MOBILITY/CA		CA							
						MOBILITY/CA		CA							
						MOBILITY/CA		CA							
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						MOBILITY/CA		CA							
						MOBILITY/CA		CA							



UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES			
FRACTIONS	DECIMALS	ANGLES	
$\frac{1}{8}$	.125	30°	
$\frac{1}{16}$	.0625	15°	
$\frac{3}{16}$	.1875	7 1/2°	
$\frac{1}{4}$	.25	0°	
$\frac{5}{16}$	.3125		
$\frac{3}{8}$	.375		
$\frac{7}{16}$	.4375		
$\frac{1}{2}$	.5		
$\frac{9}{16}$	.5625		
$\frac{5}{8}$	.625		
$\frac{11}{16}$	.6875		
$\frac{3}{4}$	.75		
$\frac{13}{16}$	.8125		
$\frac{7}{8}$	.875		
$\frac{15}{16}$	.9375		
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$1 \frac{3}{8}$	1.375		
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$2 \frac{1}{8}$	2.125		
$2 \frac{1}{4}$	2.25		
$2 \frac{3}{8}$	2.375		
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$2 \frac{5}{8}$	2.625		
$2 \frac{3}{4}$	2.75		
$2 \frac{7}{8}$	2.875		
$3$	3		
$3 \frac{1}{8}$	3.125		
$3 \frac{1}{4}$	3.25		
$3 \frac{3}{8}$	3.375		
$3 \frac{1}{2}$	3.5		
$3 \frac{5}{8}$	3.625		
$3 \frac{3}{4}$	3.75		
$3 \frac{7}{8}$	3.875		
$4$	4		
$4 \frac{1}{8}$	4.125		
$4 \frac{1}{4}$	4.25		
$4 \frac{3}{8}$	4.375		
$4 \frac{1}{2}$	4.5		
$4 \frac{5}{8}$	4.625		
$4 \frac{3}{4}$	4.75		
$4 \frac{7}{8}$	4.875		
$5$	5		
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$5 \frac{1}{4}$	5.25		
$5 \frac{3}{8}$	5.375		
$5 \frac{1}{2}$	5.5		
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$6 \frac{1}{2}$	6.5		
$6 \frac{5}{8}$	6.625		
$6 \frac{3}{4}$	6.75		
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$7 \frac{1}{2}$	7.5		
$7 \frac{5}{8}$	7.625		
$7 \frac{3}{4}$	7.75		
$7 \frac{7}{8}$	7.875		
$8$	8		
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$8 \frac{5}{8}$	8.625		
$8 \frac{3}{4}$	8.75		
$8 \frac{7}{8}$	8.875		
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$9 \frac{1}{8}$	9.125		
$9 \frac{1}{4}$	9.25		
$9 \frac{3}{8}$	9.375		
$9 \frac{1}{2}$	9.5		
$9 \frac{5}{8}$	9.625		
$9 \frac{3}{4}$	9.75		
$9 \frac{7}{8}$	9.875		
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$19 \frac{7}{8}$	19.875		
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$23$	23		
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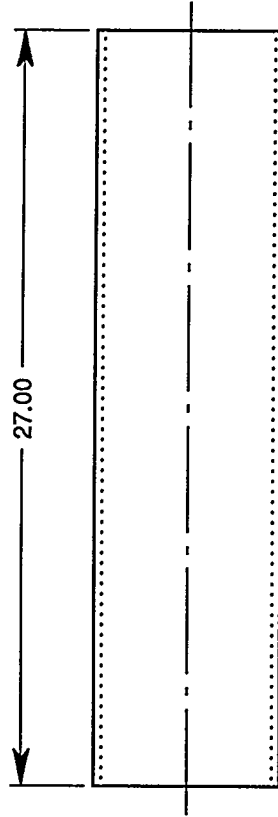


UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES		
FRACTIONS	XX	DECIMALS
ANGLES	XXX	0° 30'
±	.01	±.005
MACHINE CORNERS - .005 TO .015 RAD OR CHAMFER		
SHEET MATERIAL - BREAK EDGES .005 MAX RADIUS		
MACHINED SURFACES TO BE		
DIMENSIONS AND TOLERANCES APPLY BEFORE FINISHED TREATMENT		
DO NOT SCALE DRAWING		
MATERIAL:		
HAYNES 556		
FINISH:		



ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		SLIP - ON FLANGE	
PROJECT NO. 8146 - 040		SCALE HALF SIZE	
DWN D. TATE		9-16-91	
CHK M. CHAN		9-25-91	
ENG M. CHAN		9-25-91	
APPD J. SAWYER		9-25-91	
PROJECT NO. 8146 - 040		SCALE HALF SIZE	
ORIGINATING OFFICE		CASE NO.	
Durham, NC		ORIG	
Jefferson, NC		DEWT	
Jefferson, NC		ODWA	
REFERENCE DRAWINGS		TITLE	
NUMBER		DATE	
APPD		10-17-91	
DJT		9-25-91	
DJT		9-25-91	
REVISIONS		DESCRIPTION	
B		RELEASED FOR FABRICATION	
A		RELEASED FOR BID	
LTR		DATE	
DWG NO.		8146 - 203	
SIZE		B	
REV		1 OF 1	
SHEET		B	

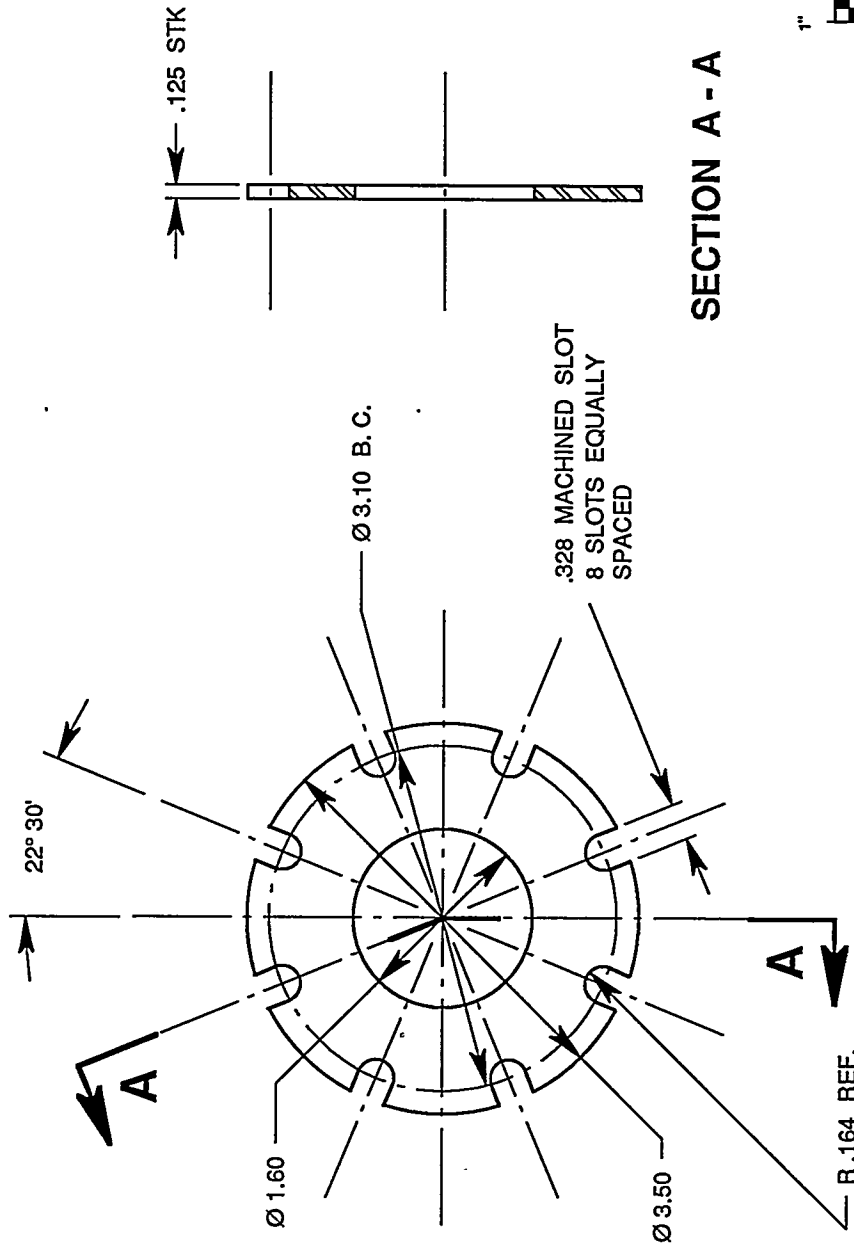
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES		
FRACTIONS	XX	ANGLES
± .01	± .005	± 0° 30'
MACHINE CRYSTAL GRADE TO 400 PSI OR CHAMFER SHEET MATERIAL - BREAK EDGES 200 PSI TYPICAL		
MACHINED SURFACES TO BE		
DIMENSIONS AND TOLERANCES APPLY BEFORE FINISHED TREATMENT		
DO NOT SCALE DRAWING		
MATERIAL:		
HAYNES 556		
FINISH:		



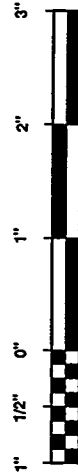
6" NOM. SCH. 40 PIPE



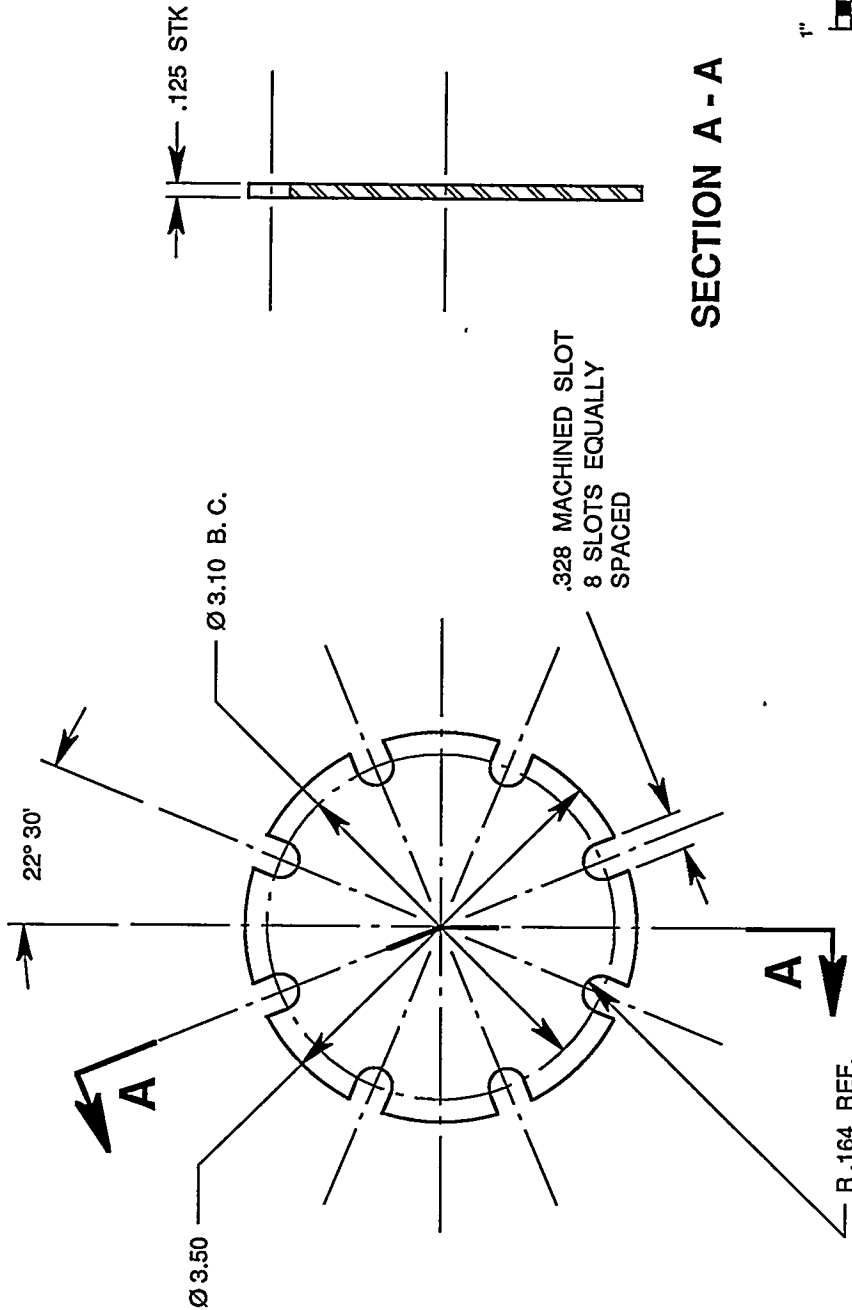
ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		CAN BODY	
SIZE	DWG. NO.	SHEET	REV.
B	8146 - 204	1 OF 1	B
DWN		9-16-91	9-16-91
CHK		M. CHAN	9-25-91
ENG		M. CHAN	9-25-91
APPD		J. SAWYER	9-25-91
PROJECT NO.		8146 - 040	
SCALE		QUARTER SCALE	
DRAWING OFFICE		CAGE NO.	
DRAWING DATE		DATE	
DRAWING BY		DATE	
DRAWING CHECKED BY		DATE	
DRAWING APPROVED BY		DATE	
DESCRIPTION		NUMBER	TITLE
B		RELEASED FOR FABRICATION	10-17-91
A		RELEASED FOR BID	9-25-91
LTR		DATE	
REVISIONS		REFERENCE DRAWINGS	



# SECTION A - A


[illegible]



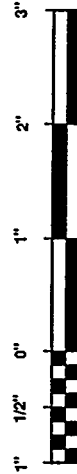
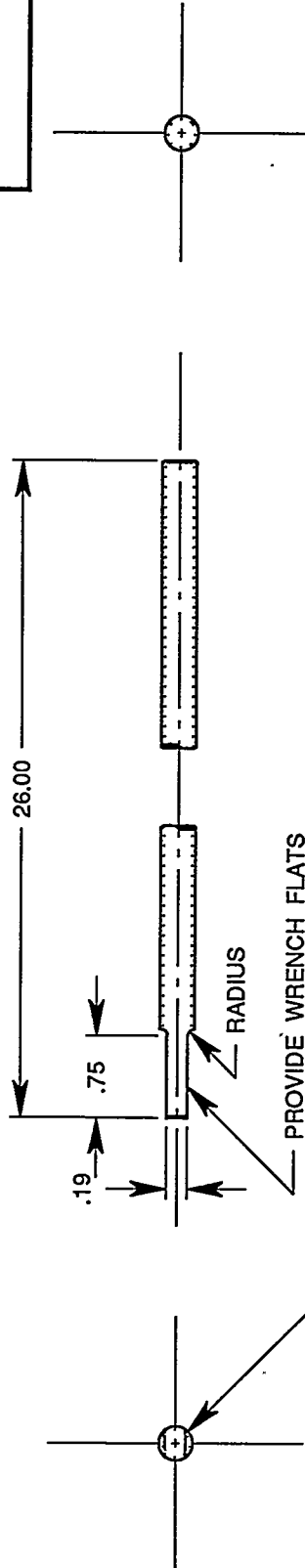


## SECTION A - A



		DWN	D. TATE	9-16-91	 <b>ACUREX Corporation</b> PROJECT <b>DEPARTMENT OF ENERGY</b> <b>THERMAL EXPOSURE TESTER -- SAMPLE CAN</b> <b>SAMPLE MOUNTING</b> <b>END FLANGE</b>
		CHK	M. CHAN	9-25-91	
		ENG	M. CHAN	9-25-91	
		APPD	J. SAWYER	9-25-91	
		PROJECT NO.		8146 - 040	SIZE <b>B</b> DWG. NO. <b>8146 - 206</b> SHEET <b>1</b> OF <b>1</b> REV <b>B</b>
		SCALE		FULL SCALE	
		ORIGINATING OFFICE		CAGE NO.	
		ALBUQUERQUE, NM BOSTON, MA BOSTON, MA JACKSON, MS JACKSON, MS		00000 00000 00000 00000 00000	
		TITLE		REFERENCE DRAWINGS	
		NUMBER			
		DATE			
		APPD			
B		RELEASED FOR FABRICATION		DJT	10-17-91
A		RELEASED FOR BID		DJT	9-25-91
LTR		DESCRIPTION			
		REVISIONS			

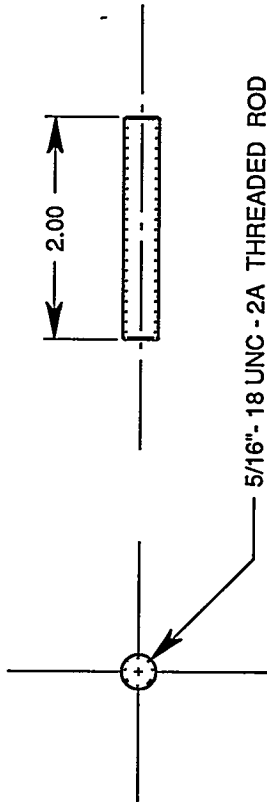
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES	
FRACTIONS	DECIMALS
XX	.XX
XXX	.XXX
± .01	± .005
± .005	± .001
± .001	± .0005
± .0005	± .0001
MACHINE CORNERS - .005 TO .015 RAD OR CHAMFER	
SHEET MATERIAL - BREAK EDGES .005 MAX RADIUS	
MACHINED SURFACES TO BE	
DIMENSIONS AND TOLERANCES APPLY BEFORE	
FINISH TREATMENT	
DO NOT SCALE DRAWING	
MATERIAL:	
R A - 330	
FINISH:	



FULL SCALE

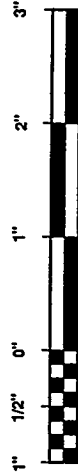
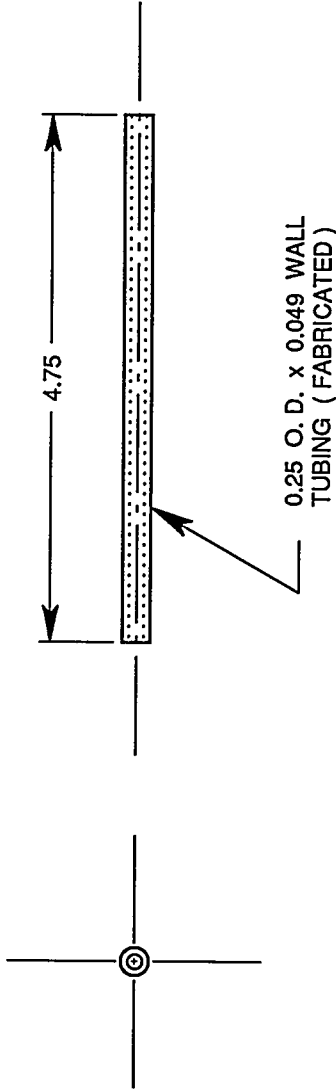
ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		SAMPLE MOUNTING ROD	
SIZE		DWG. NO. 8146 - 207	
REV		SHEET 1 OF 1	
B		B	
DWN		D. TATE	
CHK		M. CHAN	
ENG		M. CHAN	
APPD		J. SAWYER	
PROJECT NO.		8146 - 040	
SCALE		FULL SCALE	
ORIGINATING OFFICE		CASE NO.	
Durham, NC		032791	
APPROVAL		032794	
NUMBER		TITLE	
REVISIONS		REFERENCE DRAWINGS	
DESCRIPTION		DATE	
B		10-17-91	
A		9-25-91	
LTH		APPD	

UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES	
FRACTIONS	XX
DECIMALS	.XXX
ANGLES	± .01 ± .005 ± 0° 30'
MACHINE CORNERS - .005 TO .015 RAO OR CHAMFER	
SHEET MATERIAL - BREAK EDGES .005 MAX RADIUS	
MACHINED SURFACES TO BE	
DIMENSIONS AND TOLERANCES APPLY BEFORE FINISHED TREATMENT	
DO NOT SCALE DRAWING	
MATERIAL:	
R A - 330	
FINISH:	



ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		FLANGE STUD	
SIZE	DWG. NO.	SHEET	REV
B	8146 - 208	1 OF 1	B
DWN	D. TATE	9-16-91	
CHK	M. CHAN	9-25-91	
ENG	M. CHAN	9-25-91	
APPD	J. SAWYER	9-25-91	
PROJECT NO.	8146 - 040		
SCALE	FULL SCALE		
ORIGINATING OFFICE	DATE	TITLE	REFERENCE DRAWINGS
San Francisco, CA	10-17-91		
Durham, NC	9-25-91		
JPL/STP, ATL			
REVISIONS			
DESCRIPTION	DATE	NUMBER	
B RELEASED FOR FABRICATION	DJT		
A RELEASED FOR BID	DJT		
LTR	APPD		

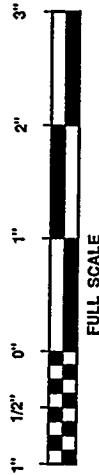
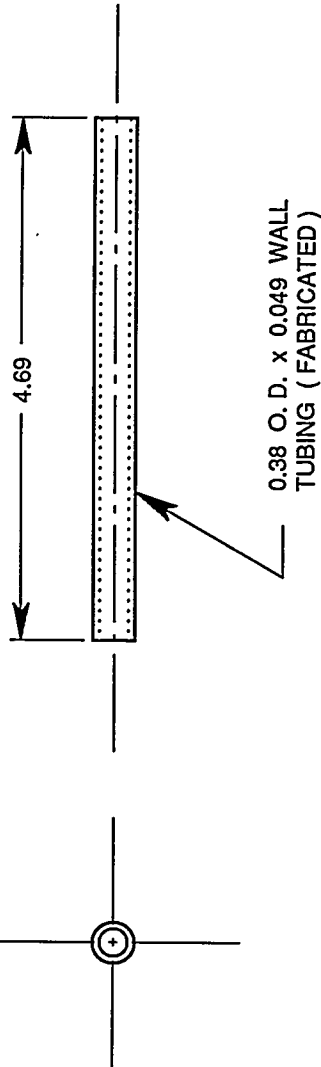
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES	
FRACTIONS	XX
DECIMALS	.XXX
ANGLES	± .01 ± .005 ± 0° 30'
MACHINE CORNERS - .005 TO .015 RAD OR CHAMFER	
SHEET MATERIAL - BREAK EDGES .005 MAX RADIUS	
MACHINED SURFACES TO BE	
DIMENSIONS AND TOLERANCES APPLY BEFORE	
FINISHED TREATMENT	
DO NOT SCALE DRAWING	
MATERIAL:	
HAYNES 556	
FINISH:	



FULL SCALE

ACUREX Corporation		Environmental Systems Division	
PROJECT		DEPARTMENT OF ENERGY	
THERMAL EXPOSURE TESTER -- SAMPLE CAN		1 / 4" EXTENSION TUBE	
SIZE	DWG. NO.	B	8146 - 209
REV	SHEET	1	1
C	1		
REVISIONS			
DESCRIPTION	DATE	APPROVED	DATE
A	10-1-81	DJT	10-1-81
B	10-1-81	DJT	10-1-81
C	10-17-91	DJT	10-17-91
RELEASED FOR FABRICATION			
RELEASED MATERIAL			
RELEASED FOR BID			
LTH			
NUMBER		TITLE	
REFERENCE DRAWINGS			
DWN		D. TATE	
CHK		M. CHAN	
ENG		M. CHAN	
APPD		J. SAWYER	
PROJECT NO.		8146 - 040	
SCALE		FULL SCALE	
DRAWING OFFICE		GAGE NO.	
Durham, NC		DURHAM, NC	
JULY 1981		JULY 1981	

UNLESS OTHERWISE SPECIFIED		
FRACTIONS	DECIMALS	ANGLES
$\frac{1}{2}$	.5	30°
$\frac{1}{4}$	.25	45°
$\frac{3}{4}$	.75	60°
$\frac{1}{8}$	.125	90°
$\frac{3}{16}$	.1875	120°
$\frac{1}{2}$	.5	150°
$\frac{3}{4}$	.75	180°
$\frac{1}{4}$	.25	210°
$\frac{3}{8}$	.375	240°
$\frac{1}{2}$	.5	270°
$\frac{3}{4}$	.75	300°
$\frac{1}{8}$	.125	330°
$\frac{3}{16}$	.1875	360°
MACHINE CORNERS - ROUNDED TO .015 RADIUS OR CHAMFER		
SHEET MATERIAL - BREAK EDGES AND TRIM PRODS		
MACHINED SURFACES TO BE		
FINISHED TREATMENT		
DO NOT SCALE DRAWING		
MATERIAL:		
HAYNES 556		
FINISH:		



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**APPENDIX C**

**REPORT: "THERMAL SHOCK-INDUCED STRENGTH  
DEGRADATION OF POROUS SiC CANDLE FILTERS"**

*Removed for separate cycling. al*

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## APPENDIX B: SKETCHES OF CERAMIC C-RING TESTING FIXTURES

The cost of the raw materials to make one set of fixtures out of Hexoloy SA SiC (from Carborundum<sup>a</sup>) was quoted to be about \$10,300. The cost of having the fixtures milled was not attempted because of the expensive cost of the raw materials. Ceramics are brittle materials. If a slight crack is inadvertently introduced into the fixture during manufacturing, it could cause the fixture to break. Since there would be no additional funds (nor time) to finish the Senior Project on schedule, this design route was aborted.

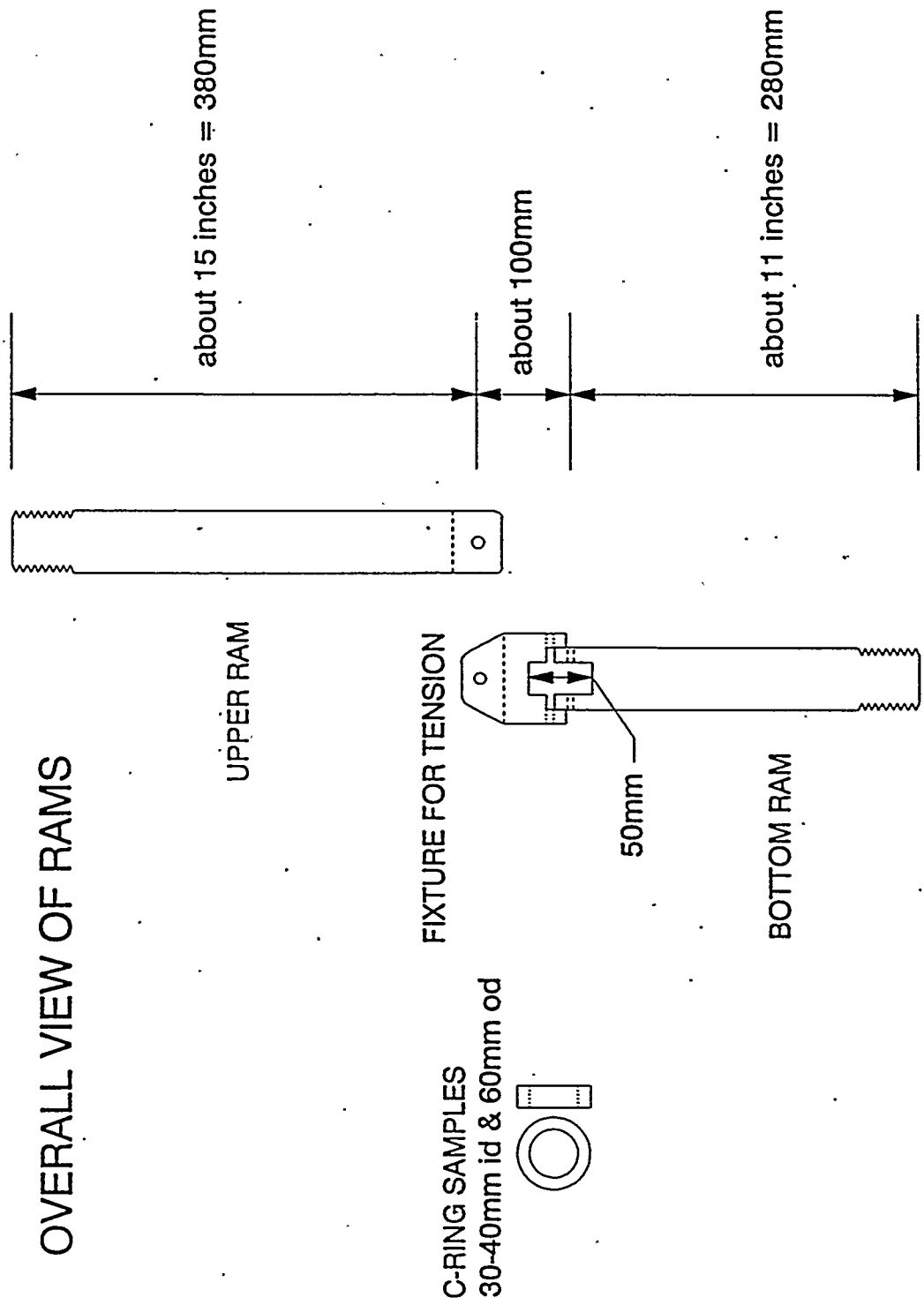
The following pages are preliminary sketches/designs for c-ring testing fixtures to be made out of a ceramic material for elevated temperature testing. They would exceed the testing capabilities of a super alloy ( $\approx >1200^{\circ}\text{C}$ , low thermal expansions, and high modulus).

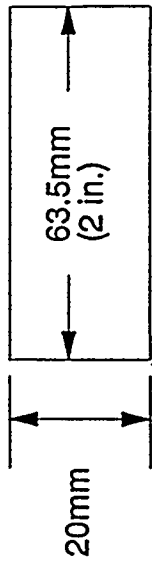
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<sup>a</sup> Carborundum Company of Niagara Falls, New York

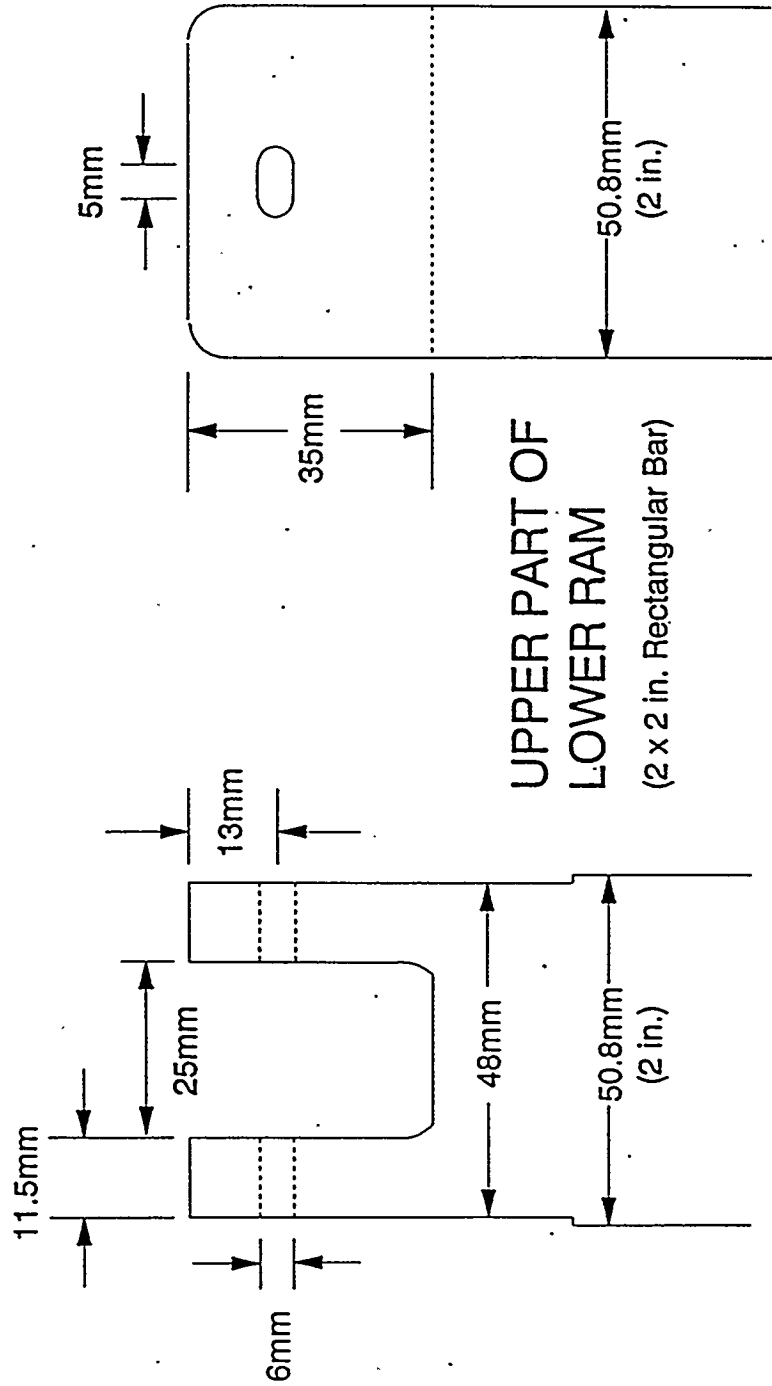
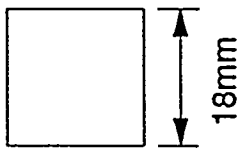


# OVERALL VIEW OF RAMS

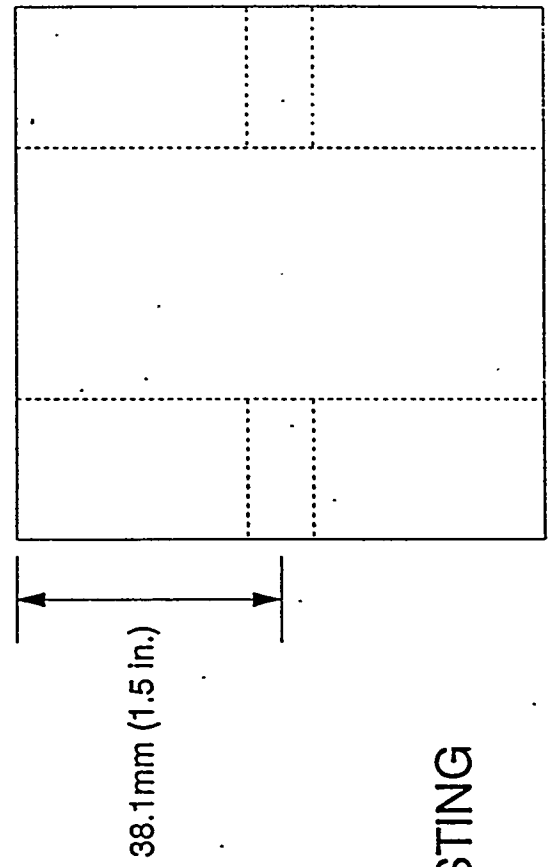
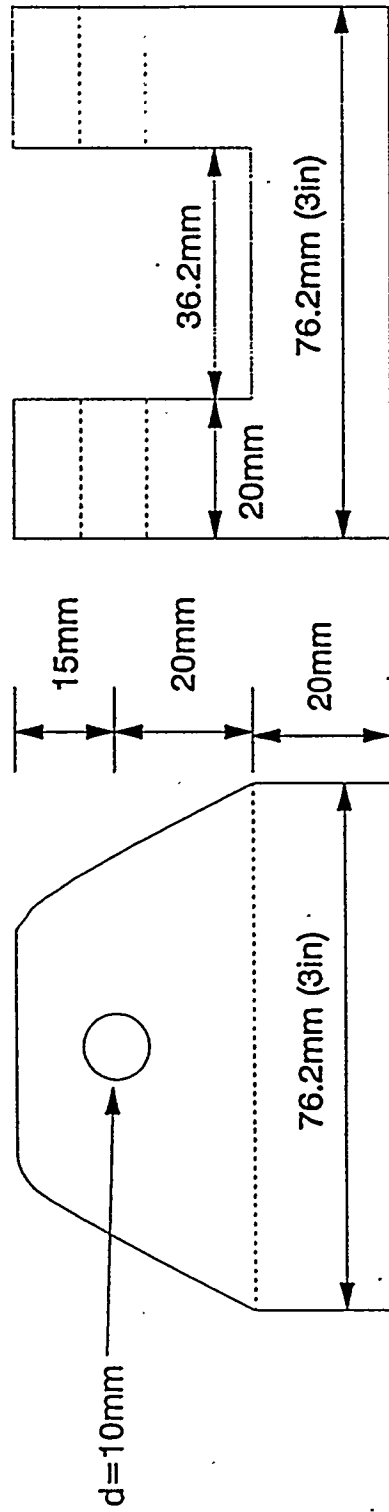




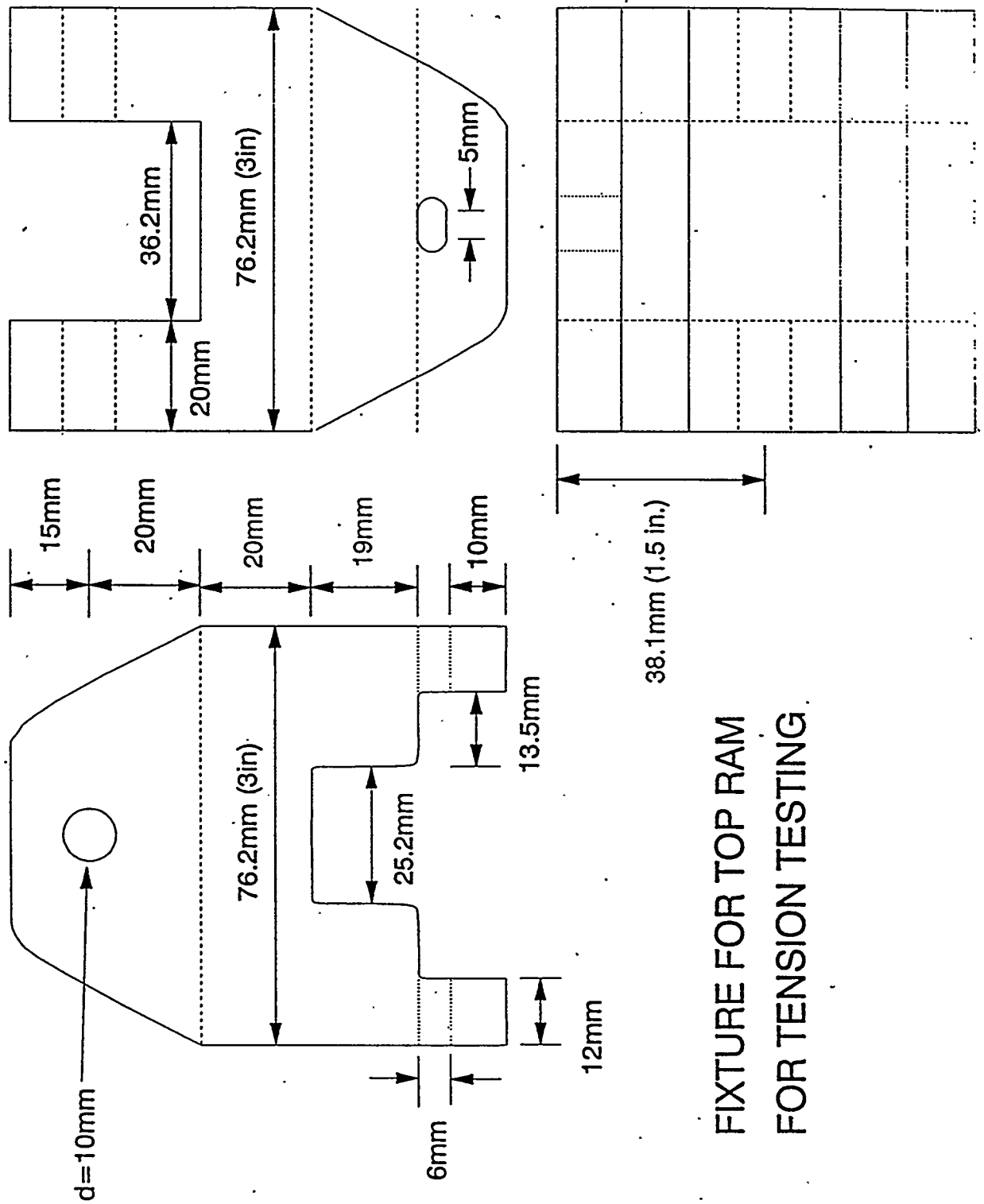
Bar used for  
Compression



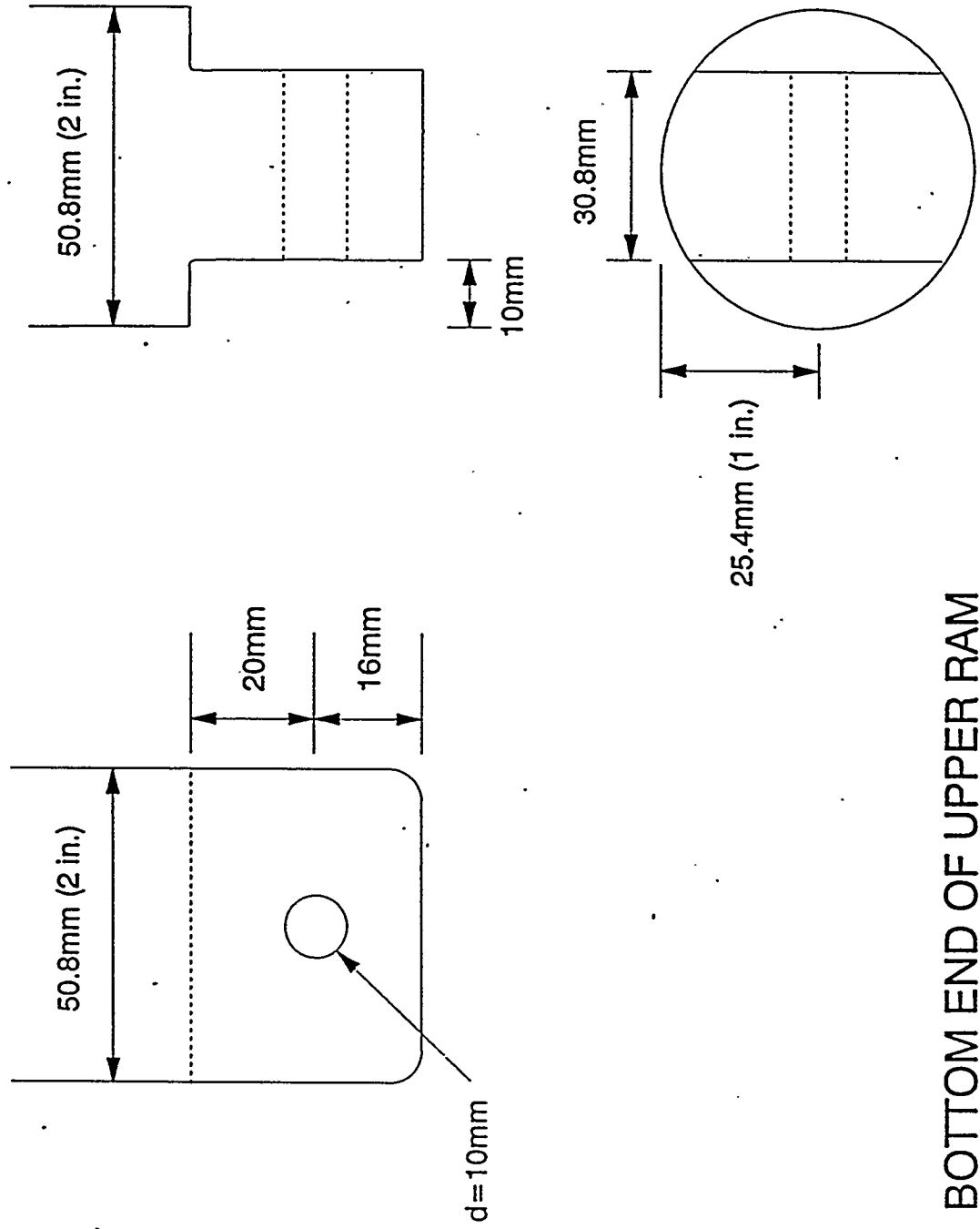
UPPER PART OF  
LOWER RAM  
(2 x 2 in. Rectangular Bar)



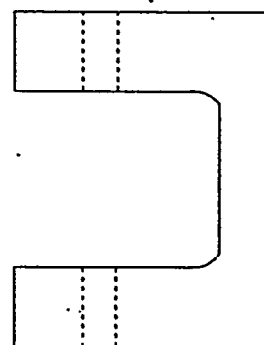
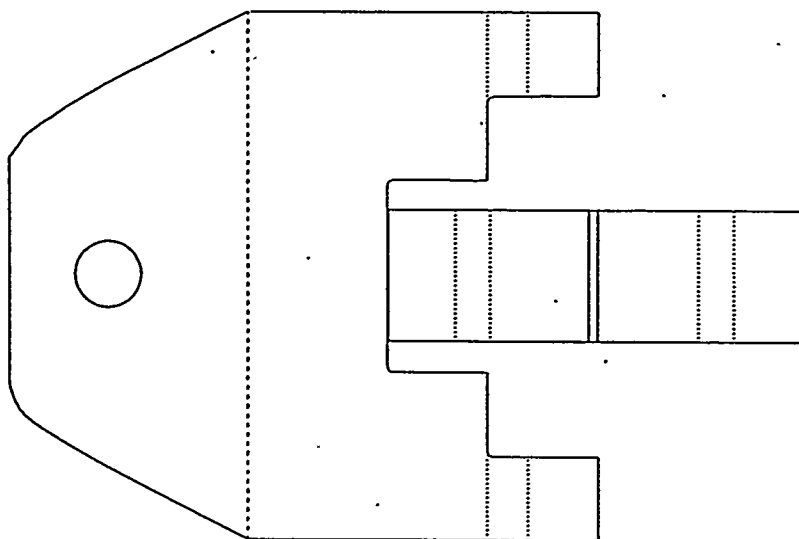
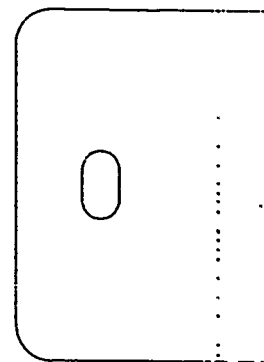
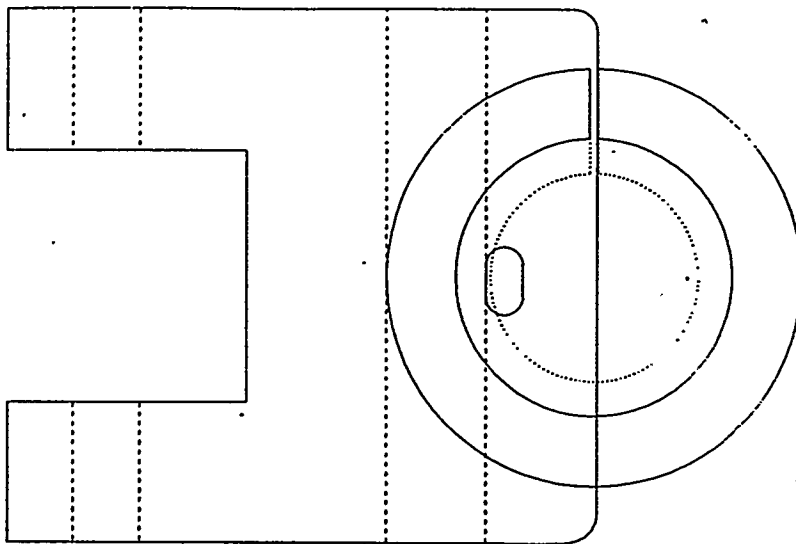
FIXTURE FOR TOP RAM  
FOR COMPRESSION TESTING

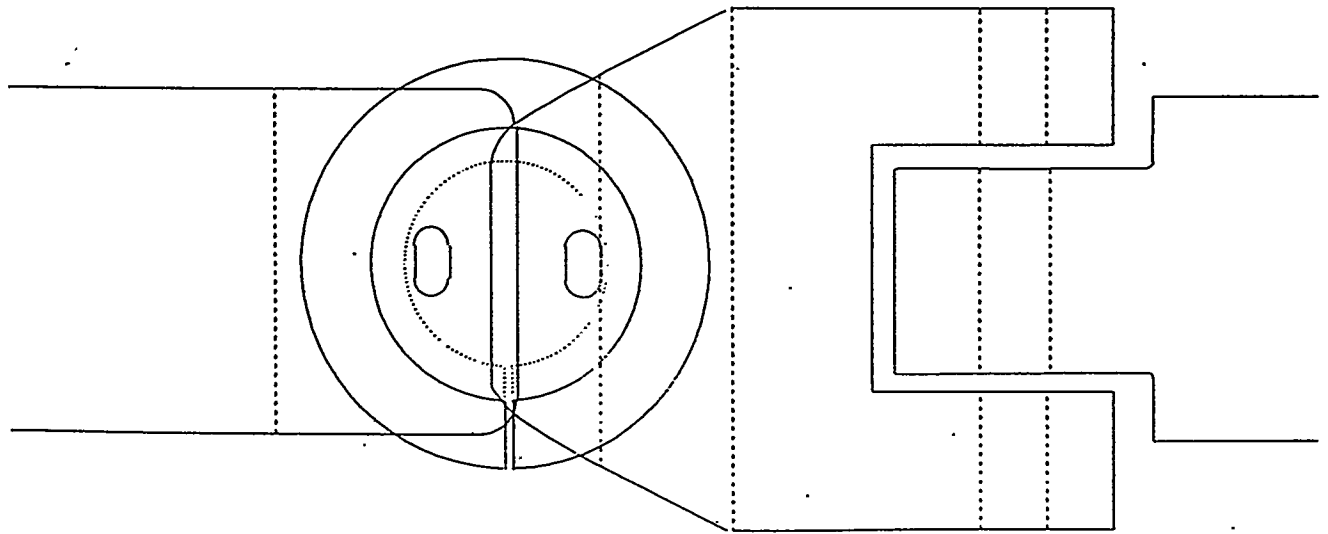
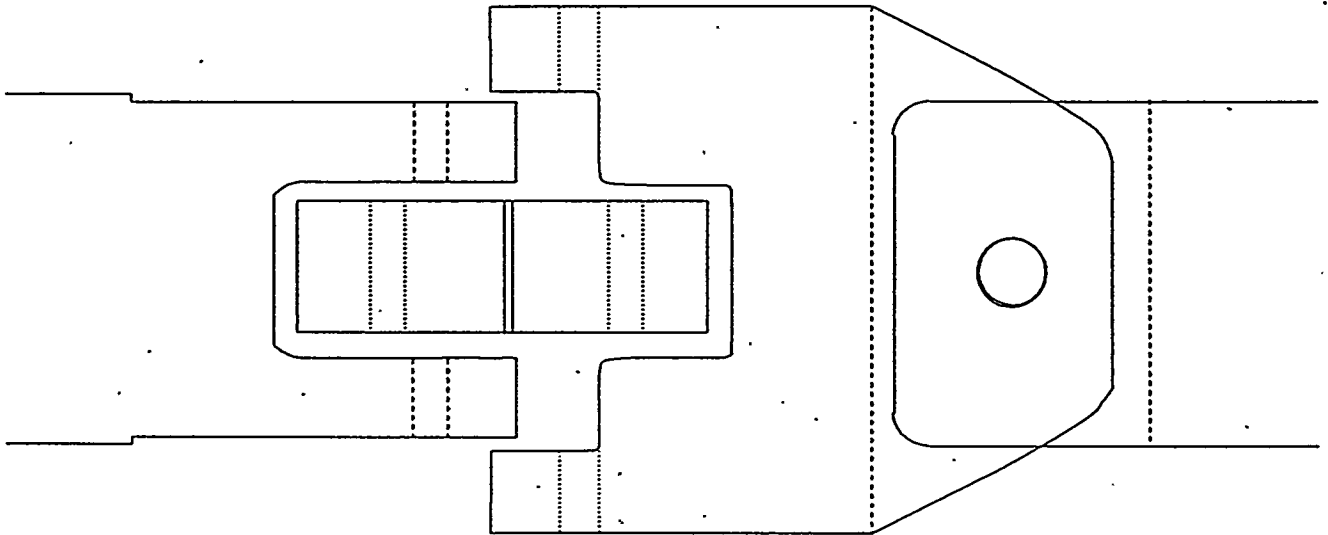


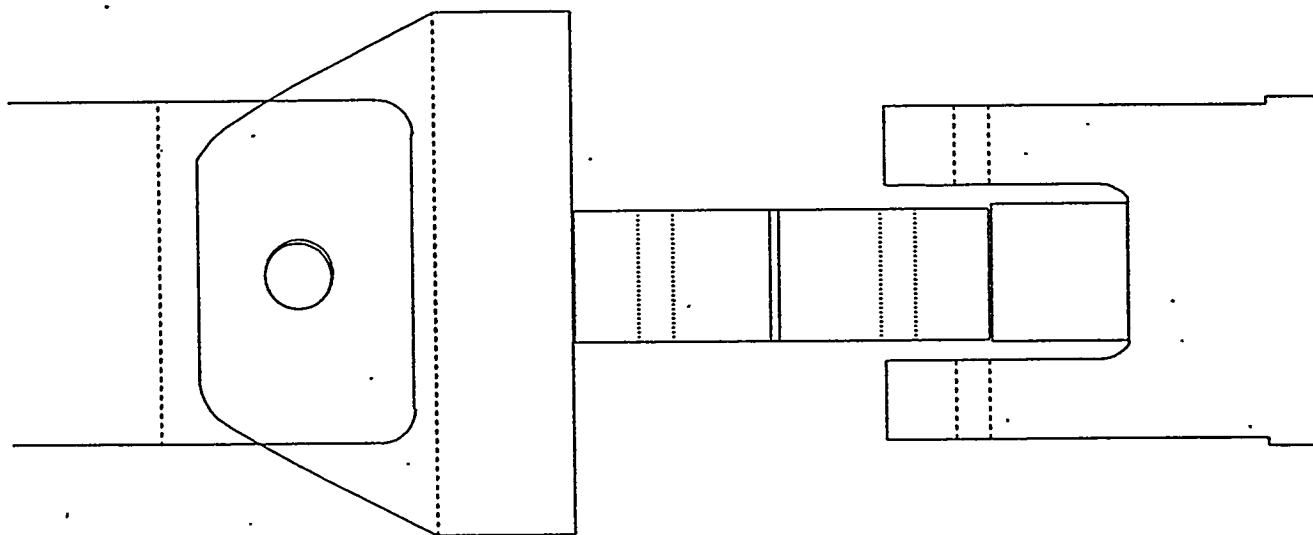
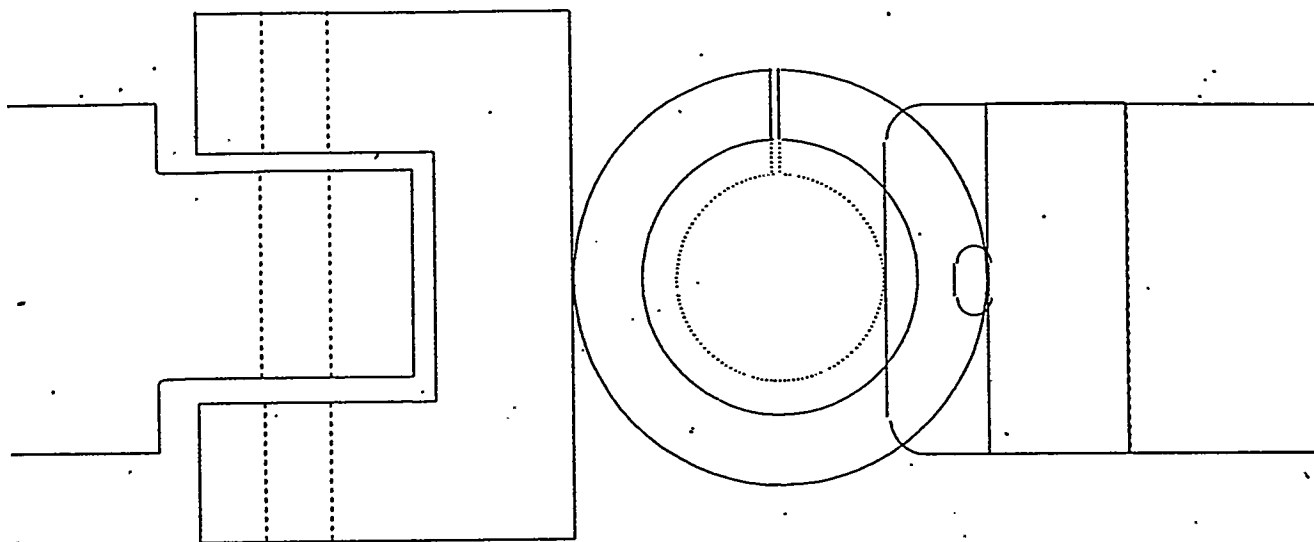
FIXTURE FOR TOP RAM  
FOR TENSION TESTING.



BOTTOM END OF UPPER RAM







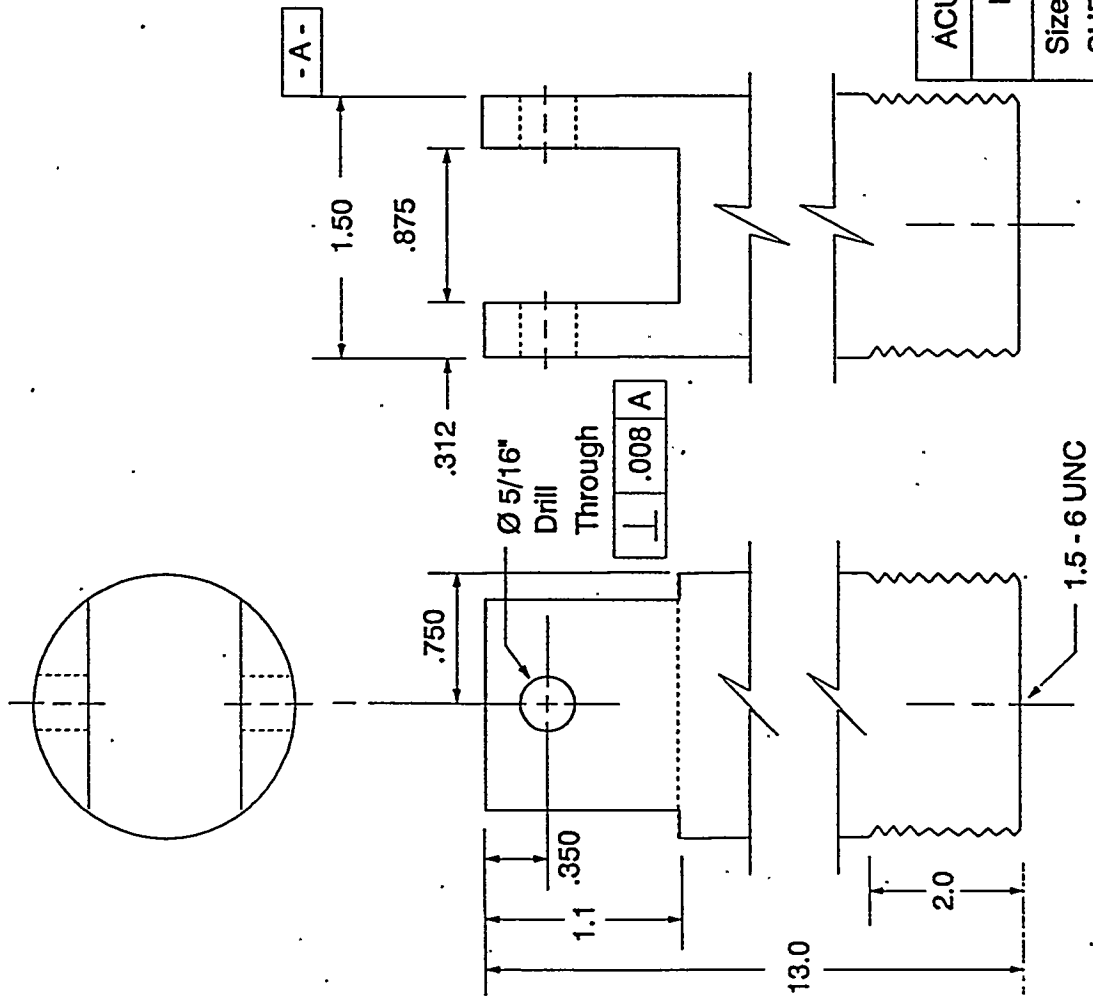


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APPENDIX C: DESIGNS OF THE SUPERALLOY C-RING TESTING FIXTURES

The following designs were drawn by the author and submitted to Modern Machine (in San Jose) to be fabricated from Haynes® Alloy 230 (a nickel-chromium-tungsten-molybdenum alloy). The adapter/cooling couples were made of standard stainless steel that Modern Machine supplied.

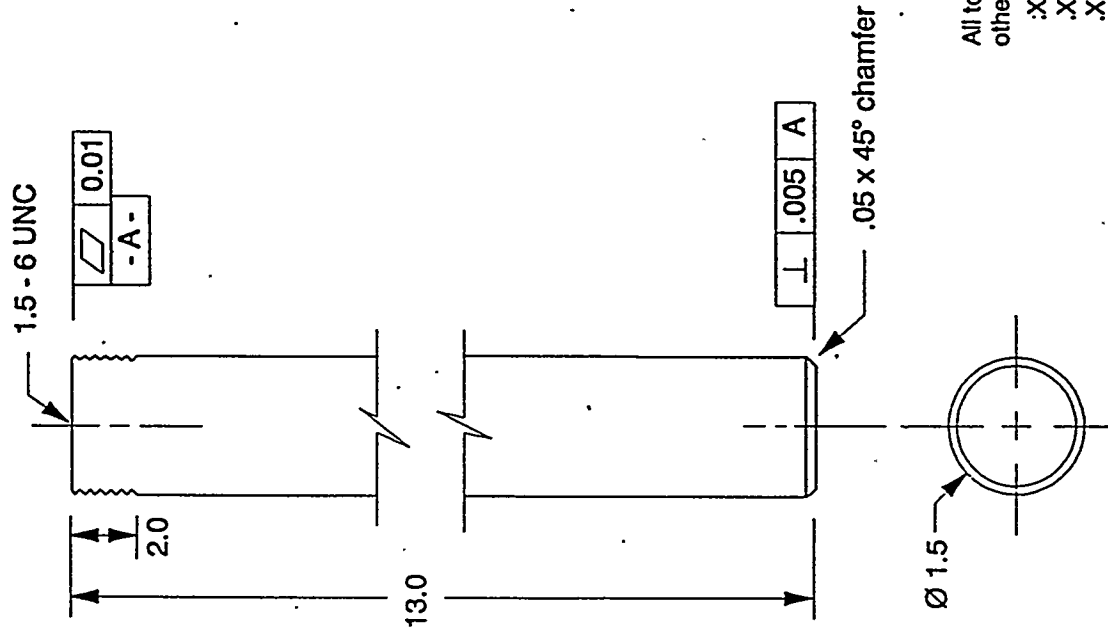
All tolerances unless  
otherwise specified:  
.XXX  $\pm$  .005  
.XX  $\pm$  .010  
.X  $\pm$  .10



Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
LOWER RAM	
Size A	Project Number: 8146
SHEET 1 OF 6	DWN: <i>[Signature]</i>



All tolerances unless  
otherwise specified:  
.XXX ± .005  
.XX ± .010  
.X ± .10

Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

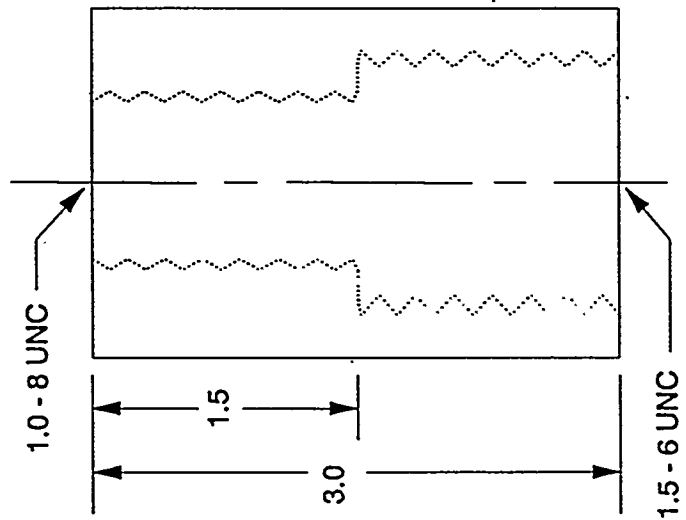
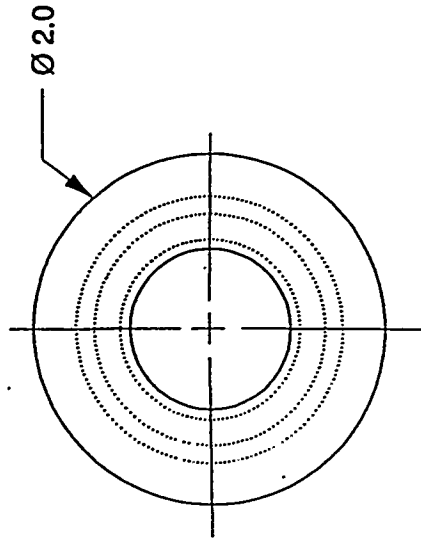
ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
UPPER RAM FOR COMPRESSION TEST	
Size A SHEET 2 OF 6	Project Number: 8146 DWN: <i>Michael Cho</i>

All tolerances unless  
otherwise specified:

.XXX  $\pm$  .005

.XX  $\pm$  .010

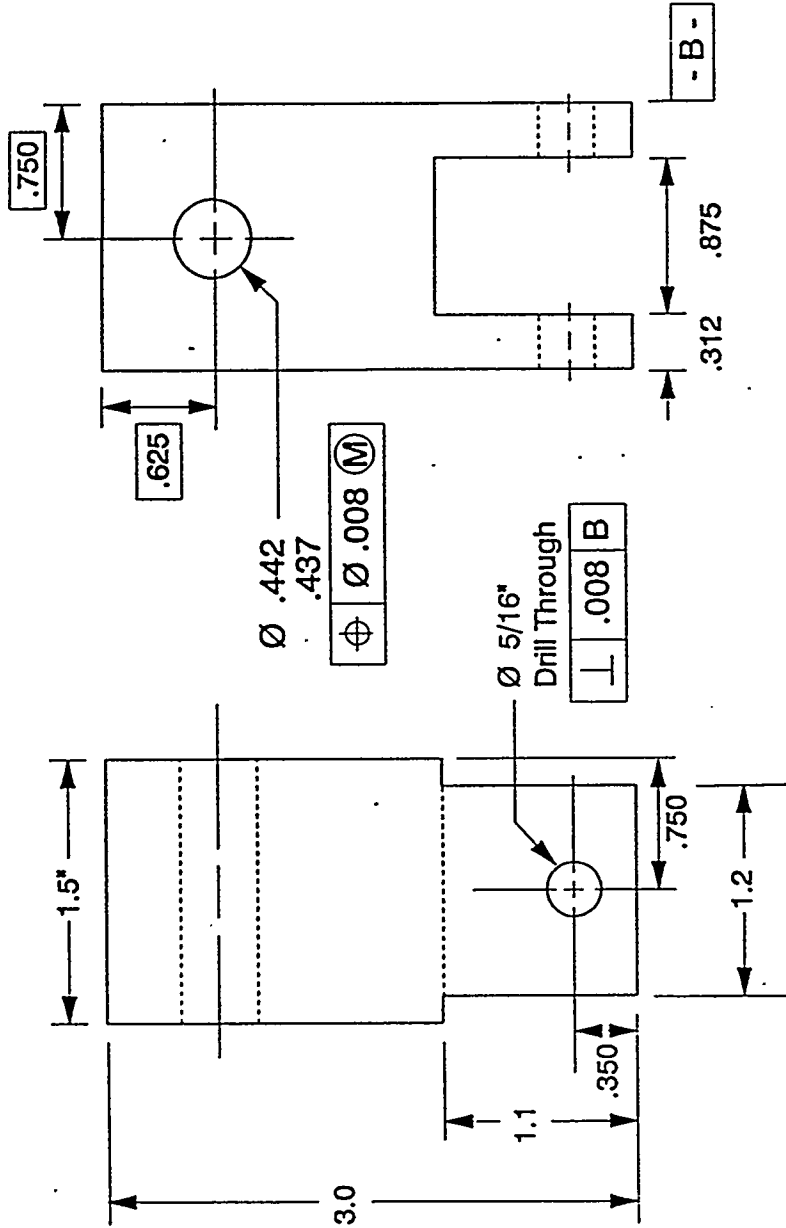
X  $\pm$  .10



Notes:

1. Material is
2. All dimensions in inches
3. Quantity required is 2

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
ADAPTER COUPLE FOR UPPER RAM (COMPR.)	
Size A	Project Number: 8146
SHEET 3 OF 6	DWN: <i>Michael Chen</i>



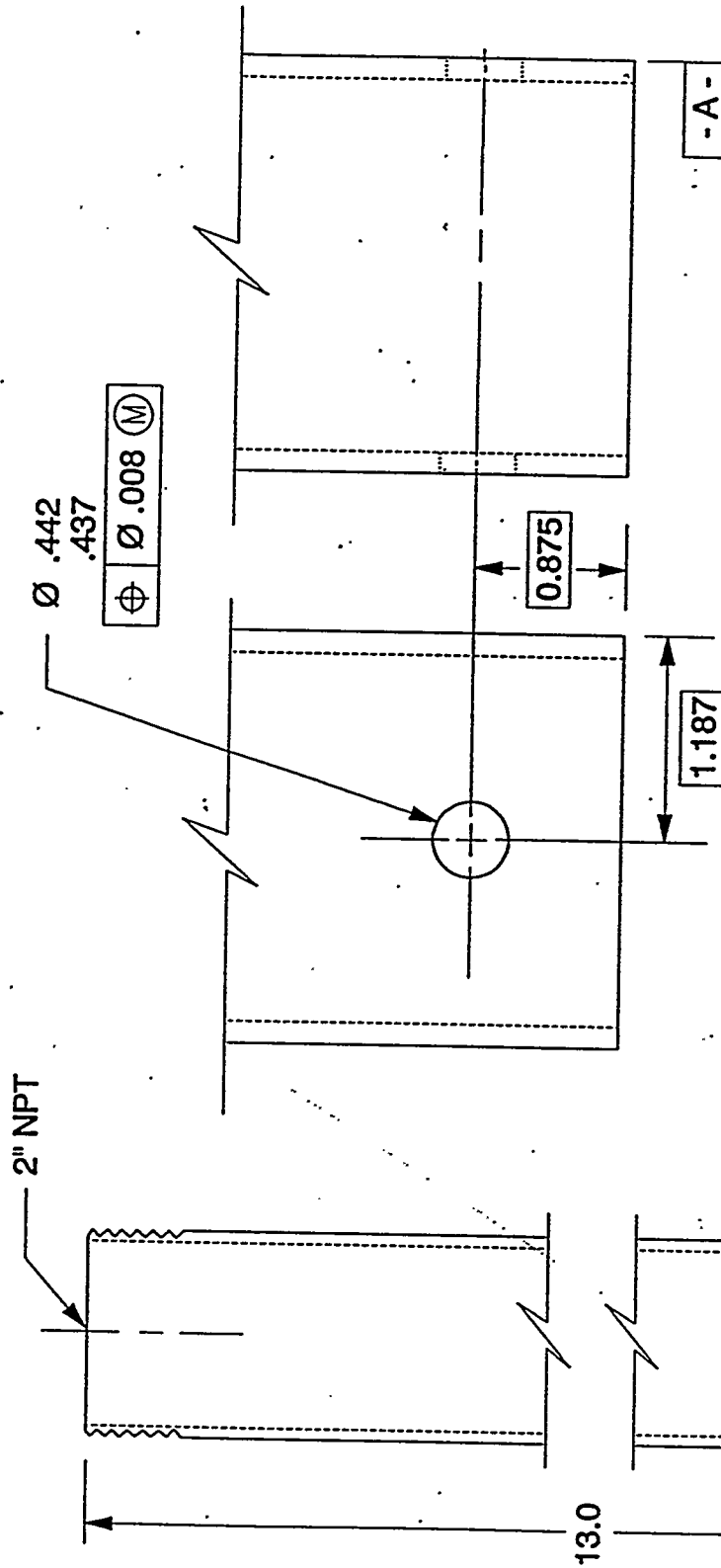
Notes:

1. Material is Haynes Alloy 230
2. All dimensions in inches

All tolerances unless  
otherwise specified:

.XXX  $\pm .005$   
.XX  $\pm .010$   
.X  $\pm .10$

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
UPPER FIXTURE FOR TENSION TEST	
Size A	Project Number: 8146
SHEET 4 OF 6	DWN: <i>Michael Chan</i>



Notes:

1. Material is Haynes Alloy 230
2. 2" Schedule 40 Pipe
3. All dimensions in inches

All tolerances unless  
otherwise specified:

XXX ± .005  
XX ± .010  
X ± .10

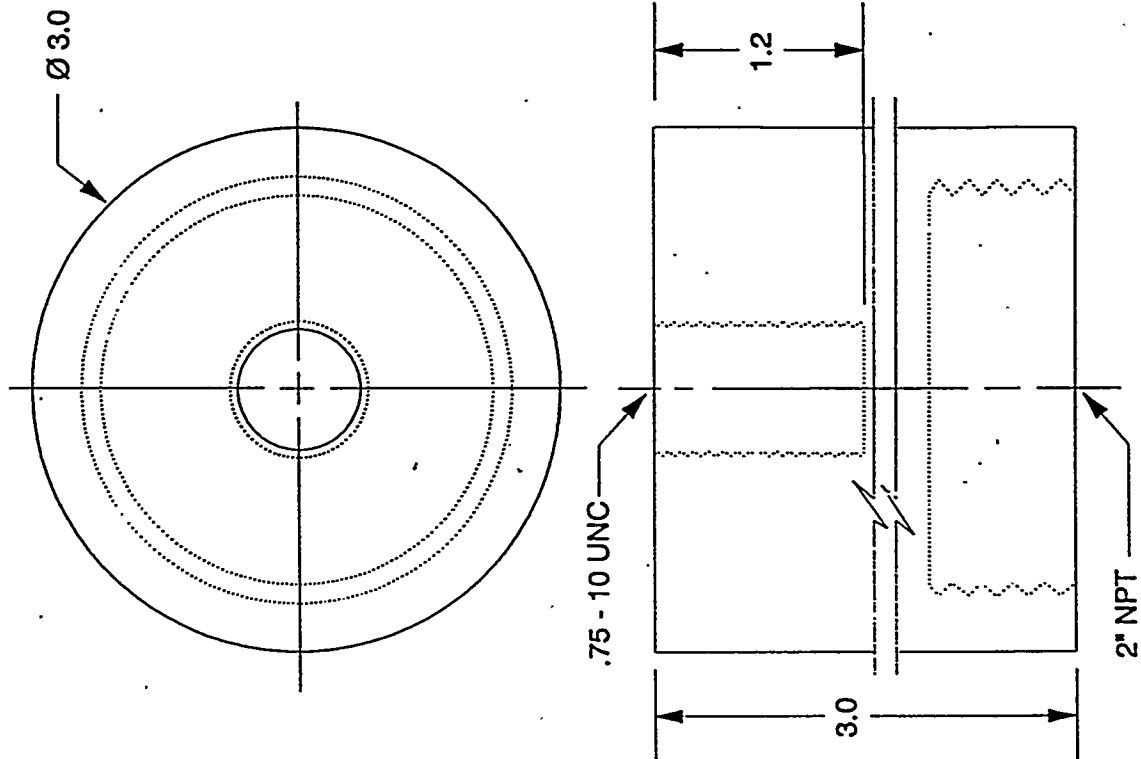
ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
UPPER RAM FOR TENSION TEST	
Size A:	Project Number: 8146
SHEET 5 OF 6	DWN: <i>Michael Chan</i>

All tolerances unless  
otherwise specified:

.XXX  $\pm$  .005

.XX  $\pm$  .010

X  $\pm$  .10



Notes:

1. Material is
2. Female 2" NPT for a 2" Schedule 40 Pipe
3. All dimensions in inches

ACUREX ENVIRONMENTAL SYSTEMS (ESD)	
ADAPTER COUPLE FOR UPPER RAM (TENSION)	
Size A SHEET 6 OF 6	Project Number: 8146 DWN: <i>Michael Chan</i>



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# APPENDIX D: RAW DATA AND MAXIMUM STRESS CALCULATIONS

The equation used for calculating the maximum stresses for c-ring compression and tension testing is:<sup>a</sup>

$$\sigma_{\theta} = \frac{P}{w(r_o - r_i)} \left[ \frac{\frac{r_o - r_i}{\ln\left(\frac{r_o}{r_i}\right)} \left( r - \frac{r_o + r_i}{2} \right)}{r \left[ \frac{r_o - r_i}{\ln\left(\frac{r_o}{r_i}\right)} - \frac{r_o + r_i}{2} \right]} \right] \cos\theta$$

Note:  $\cos \theta = 1$  for  $\theta = 0^\circ$

P = breaking load (negative for compressive loading)

w = c-ring thickness

r = radial distance of interest

$r_i$  = inner radius

$r_o$  = outer radius

The following pages are the raw data and stress calculation spreadsheets in Lotus 1-2-3. Two inner diameter (I.D.) (as well as two outer diameter) measurements were made and averaged. The averaged value was then corrected because the calipers the author used was not calibrated.

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<sup>a</sup> Ferber, M. K., V. J. Tennery, S. B. Waters, and J. Ogle, "Fracture Strength Characterization of Tubular Ceramic Materials Using a Simple C-ring Geometry", Journal of Materials Science, V. 21, 1986, pp. 2628-2632

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SPECIMEN	SPECIMEN NUMBER	CORRECTED			CORRECTED			WIDTH	TEST	LOAD (lbs)
		I.D. 1 (in.)	I.D. 2 (in.)	O.D. 1 (in.)	O.D. 2 (in.)	AVE. I.D. (in.)	I.D. (in.)	AVE. O.D. (in.)	O.D. (in.)	
SCHUMACHER	T2	1.220	1.231	2.390	2.380	1.226	1.246	2.385	2.425	0.775 RTT 22.75
SOLAR (UNUSED)	T13	1.190	1.202	2.386	2.376	1.196	1.216	2.381	2.421	0.770 RTT 28.25
	T17	1.185	1.190	2.378	2.364	1.188	1.208	2.371	2.411	0.755 RTT 26.25
	B3	1.155	1.142	2.361	2.345	1.149	1.168	2.353	2.393	0.740 RTT 31.75
	B7	1.160	1.148	2.356	2.338	1.154	1.174	2.347	2.387	0.745 RTT 27.00
SCHUMACHER	2	1.218	1.217	2.370	2.382	1.218	1.238	2.376	2.416	0.732 RTT 17.25
SOLAR (USED)	6	1.205	1.210	2.393	2.375	1.208	1.228	2.384	2.424	0.807 RTT 14.00
	10	1.183	1.187	2.389	2.372	1.185	1.205	2.381	2.421	0.796 RTT 25.50
	14	1.180	1.174	2.376	2.358	1.177	1.197	2.367	2.407	0.798 RTT 23.50
	18	1.179	1.170	2.386	2.395	1.175	1.194	2.391	2.431	0.798 RTT 25.25
SCHUMACHER (UNUSED)	1	1.210	1.200	2.392	2.388	1.205	1.225	2.390	2.431	0.741 RTT 40.50
	3	1.198	1.194	2.380	2.375	1.196	1.216	2.378	2.418	0.742 RTT 37.75
	5	1.199	1.203	2.386	2.386	1.201	1.221	2.386	2.426	0.766 RTT 47.25
	7	1.191	1.178	2.376	2.384	1.185	1.205	2.380	2.420	0.752 RTT 44.25
	9	1.158	1.170	2.340	2.345	1.164	1.184	2.343	2.382	0.730 RTT 43.25
SCHUMACHER	1	1.100	1.100	2.339	2.352	1.100	1.119	2.346	2.385	0.743 RTT 33.00
5000 PULSES	3	1.098	1.098	2.334	2.354	1.098	1.117	2.344	2.384	0.740 RTT 32.00
@ 0.5 sec	5	1.101	1.105	2.339	2.339	1.103	1.122	2.339	2.379	0.729 RTT 35.00
	9	1.091	1.105	2.368	2.362	1.098	1.117	2.365	2.405	0.746 RTT 35.75
	7	1.097	1.088	2.369	2.358	1.093	1.111	2.364	2.404	0.742 RTT 34.00
SCHUMACHER	1	1.155	1.167	2.349	2.346	1.161	1.181	2.348	2.387	0.709 RTT 119.00
KRW 7068	5	1.146	1.140	2.350	2.349	1.143	1.162	2.350	2.389	0.768 RTT 134.25
(UNUSED)	9	1.126	1.134	2.350	2.352	1.130	1.149	2.351	2.391	0.772 RTT 131.00
	12	1.128	1.109	2.360	2.358	1.119	1.137	2.359	2.399	0.751 RTT 140.75
	15	1.114	1.112	2.357	2.365	1.113	1.132	2.361	2.401	0.785 RTT 154.50
DIDIER	3	1.598	1.590	2.361	2.360	1.594	1.621	2.361	2.401	0.754 RTT 88.00
(UNUSED)	5	1.511	1.492	2.361	2.355	1.502	1.527	2.358	2.398	0.752 RTT 101.75
	6	1.500	1.495	2.365	2.347	1.498	1.523	2.356	2.396	0.748 RTT 85.25
	8	1.489	1.493	2.350	2.358	1.491	1.516	2.354	2.394	0.775 RTT 77.25
	9	1.481	1.478	2.348	2.352	1.480	1.505	2.350	2.390	0.786 RTT 95.75
DIDIER	2	1.490	1.477	2.331	2.336	1.484	1.509	2.334	2.373	0.764 RTT 103.25
5000 PULSES	3	1.510	1.519	2.358	2.368	1.515	1.540	2.363	2.403	0.742 RTT 65.75
@ 0.05 sec	4	1.510	1.500	2.362	2.351	1.505	1.531	2.357	2.396	0.770 RTT 83.00
	5	1.508	1.504	2.355	2.352	1.506	1.532	2.354	2.393	0.773 RTT 78.75
	6	1.500	1.494	2.350	2.345	1.497	1.522	2.348	2.387	0.773 RTT 78.50
IFP LAYCER	2	1.575	1.574	2.321	2.321	1.575	1.601	2.321	2.360	0.761 RTT 36.75
(UNUSED)	4	1.572	1.565	2.320	2.318	1.569	1.595	2.319	2.358	0.756 RTT 41.50
	6	1.578	1.573	2.328	2.328	1.576	1.602	2.328	2.367	0.751 RTT 42.00
	8	1.561	1.554	2.320	2.312	1.558	1.584	2.316	2.355	0.744 RTT 43.25
	10	1.542	1.542	2.310	2.313	1.542	1.568	2.312	2.351	0.752 RTT 40.25
IFP LAYCER	2	1.520	1.512	2.282	2.264	1.516	1.542	2.273	2.312	0.731 RTT 45.25
5000 PULSES	3	1.532	1.515	2.286	2.266	1.524	1.549	2.276	2.315	0.778 RTT 50.25
@ 0.05 sec	4	1.537	1.517	2.295	2.278	1.527	1.553	2.287	2.325	0.767 RTT 50.00
	6	1.521	1.532	2.312	2.322	1.527	1.552	2.317	2.356	0.741 RTT 41.75
	5	1.512	1.530	2.293	2.320	1.521	1.547	2.307	2.346	0.750 RTT 54.25
IFP LAYCER	3	1.556	1.562	2.309	2.322	1.559	1.585	2.316	2.355	0.756 RTT 29.50
5000 PULSES	5	1.555	1.560	2.324	2.328	1.558	1.584	2.326	2.365	0.759 RTT 31.00
@ 0.5 sec	7	1.548	1.554	2.312	2.324	1.551	1.577	2.318	2.357	0.745 RTT 29.00
	9	1.560	1.553	2.331	2.316	1.557	1.583	2.324	2.363	0.775 RTT 31.75

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SPECIMEN	SPECIMEN TEST NUMBER		Ri (in.)	Ro (in.)	w (in)	R=Ri(T) R=Ro(C)	STRESS (psi)	STRESS (MPa)		STRESS (psi)	STRESS (MPa)
SCHUMACHER	T2	RTT	0.62	1.21	0.78	0.62	642	4.43	MAX=	864	5.96
SOLAR (UNUSED)	T13	RTT	0.61	1.21	0.77	0.61	770	5.31	MIN=	642	4.43
	T17	RTT	0.60	1.21	0.76	0.60	728	5.02	AVE=	749	5.17
	B3	RTT	0.58	1.20	0.74	0.58	864	5.96	STD=	72	0.49
	B7	RTT	0.59	1.19	0.75	0.59	741	5.11			
SCHUMACHER	2	RTT	0.62	1.21	0.73	0.62	515	3.55	MAX=	661	4.56
SOLAR (USED)	6	RTT	0.61	1.21	0.81	0.61	369	2.55	MIN=	369	2.55
	10	RTT	0.60	1.21	0.80	0.60	661	4.56	AVE=	558	3.85
	14	RTT	0.60	1.20	0.80	0.60	610	4.20	STD=	106	0.73
	18	RTT	0.60	1.22	0.80	0.60	635	4.38			
SCHUMACHER	1	RTT	0.61	1.22	0.74	0.61	1150	7.93	MAX=	1296	8.94
(UNUSED)	3	RTT	0.61	1.21	0.74	0.61	1072	7.39	MIN=	1072	7.39
	5	RTT	0.61	1.21	0.77	0.61	1296	8.94	AVE=	1194	8.23
	7	RTT	0.60	1.21	0.75	0.60	1214	8.37	STD=	77	0.53
	9	RTT	0.59	1.19	0.73	0.59	1238	8.54			
SCHUMACHER	1	RTT	0.56	1.19	0.74	0.56	839	5.78	MAX=	918	6.33
5000 PULSES	3	RTT	0.56	1.19	0.74	0.56	816	5.62	MIN=	816	5.62
@ 0.5 sec	5	RTT	0.56	1.19	0.73	0.56	918	6.33	AVE=	859	5.92
	9	RTT	0.56	1.20	0.75	0.56	883	6.09	STD=	37	0.25
	7	RTT	0.56	1.20	0.74	0.56	839	5.79			
SCHUMACHER	1	RTT	0.59	1.19	0.71	0.59	3470	23.92	MAX=	3724	25.68
KRW 7068	5	RTT	0.58	1.19	0.77	0.58	3506	24.17	MIN=	3331	22.97
(UNUSED)	9	RTT	0.57	1.20	0.77	0.57	3331	22.97	AVE=	3523	24.29
	12	RTT	0.57	1.20	0.75	0.57	3583	24.71	STD=	130	0.89
	15	RTT	0.57	1.20	0.79	0.57	3724	25.68			
DIDIER	3	RTT	0.81	1.20	0.75	0.81	5619	38.74	MAX=	5619	38.74
(UNUSED)	5	RTT	0.76	1.20	0.75	0.76	5237	36.11	MIN=	3796	26.17
	6	RTT	0.76	1.20	0.75	0.76	4388	30.25	AVE=	4719	32.54
	8	RTT	0.76	1.20	0.78	0.76	3796	26.17	STD=	643	4.43
	9	RTT	0.75	1.19	0.79	0.75	4554	31.40			
DIDIER	2	RTT	0.75	1.19	0.76	0.75	5257	36.25	MAX=	5257	36.25
5000 PULSES	3	RTT	0.77	1.20	0.74	0.77	3500	24.13	MIN=	3500	24.13
@ 0.05 sec	4	RTT	0.77	1.20	0.77	0.77	4218	29.08	AVE=	4192	28.91
	5	RTT	0.77	1.20	0.77	0.77	4018	27.70	STD=	582	4.01
	6	RTT	0.76	1.19	0.77	0.76	3968	27.36			
IFP LAYCER	2	RTT	0.80	1.18	0.76	0.80	2409	16.61	MAX=	2805	19.34
(UNUSED)	4	RTT	0.80	1.18	0.76	0.80	2708	18.67	MIN=	2409	16.61
	6	RTT	0.80	1.18	0.75	0.80	2755	18.99	AVE=	2637	18.18
	8	RTT	0.79	1.18	0.74	0.79	2805	19.34	STD=	152	1.05
	10	RTT	0.78	1.18	0.75	0.78	2506	17.28			
IFP LAYCER	2	RTT	0.77	1.16	0.73	0.77	2945	20.31	MAX=	3246	22.38
5000 PULSES	3	RTT	0.77	1.16	0.78	0.77	3113	21.46	MIN=	2508	17.29
@ 0.05 sec	4	RTT	0.78	1.16	0.77	0.78	3099	21.37	AVE=	2982	20.56
	6	RTT	0.78	1.18	0.74	0.78	2508	17.29	STD=	256	1.76
	5	RTT	0.77	1.17	0.75	0.77	3246	22.38			
IFP LAYCER	3	RTT	0.79	1.18	0.76	0.79	1892	13.05	MAX=	1988	13.70
5000 PULSES	5	RTT	0.79	1.18	0.76	0.79	1929	13.30	MIN=	1839	12.68
@ 0.5 sec	7	RTT	0.79	1.18	0.75	0.79	1839	12.68	AVE=	1918	13.22
	9	RTT	0.79	1.18	0.78	0.79	1940	13.38	STD=	50	0.34

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SPECIMEN	SPECIMEN NUMBER					CORRECTED		CORRECTED		WIDTH (in.)	TEST	LOAD (lbs)
		I.D. 1 (in.)	I.D. 2 (in.)	O.D. 1 (in.)	O.D. 2 (in.)	AVE. I.D. (in.)	I.D. (in.)	AVE. O.D. (in.)	O.D. (in.)			
IFP LAYCER 1000 PULSES @ 0.05 sec	11	1.553	1.548	2.325	2.330	1.551	1.577	2.328	2.367	0.762	RTT	32.75
	2	1.515	1.531	2.330	2.335	1.523	1.549	2.333	2.372	0.755	RTT	49.50
	3	1.545	1.526	2.350	2.330	1.536	1.562	2.340	2.380	0.772	RTT	43.75
	4	1.532	1.519	2.345	2.359	1.526	1.551	2.352	2.392	0.758	RTT	46.75
	5	1.532	1.547	2.345	2.354	1.540	1.566	2.350	2.389	0.760	RTT	49.25
	6	1.535	1.528	2.356	2.341	1.532	1.557	2.349	2.388	0.758	RTT	52.50
IFP LAYCER-DOE (USED)	3	1.526	1.512	2.304	2.301	1.519	1.545	2.303	2.342	0.778	RTT	32.25
	7	1.521	1.520	2.310	2.300	1.521	1.546	2.305	2.344	0.751	RTT	26.00
	11	1.500	1.506	2.298	2.304	1.503	1.528	2.301	2.340	0.758	RTT	22.25
	15	1.494	1.498	2.301	2.309	1.496	1.521	2.305	2.344	0.776	RTT	26.50
	21	1.436	1.442	2.295	2.299	1.439	1.463	2.297	2.336	0.745	RTT	49.25
SCHUMACHER KRW 7067 (USED)	1	1.163	1.160	2.347	2.330	1.162	1.181	2.339	2.378	0.761	RTT	96.75
	5	1.164	1.163	2.333	2.349	1.164	1.183	2.341	2.381	0.764	RTT	94.25
	9	1.132	1.131	2.343	2.339	1.132	1.151	2.341	2.381	0.760	RTT	97.25
	13	1.128	1.129	2.348	2.341	1.129	1.148	2.345	2.384	0.752	RTT	107.50
	17	1.110	1.110	2.353	2.349	1.110	1.129	2.351	2.391	0.750	RTT	99.25
SCHUMACHER KRW 7052 (USED)	1	1.165	1.163	2.347	2.342	1.164	1.184	2.345	2.384	0.753	RTT	94.25
	5	1.159	1.162	2.342	2.341	1.161	1.180	2.342	2.381	0.759	RTT	96.75
	9	1.139	1.144	2.346	2.349	1.142	1.161	2.348	2.387	0.756	RTT	110.75
	13	1.122	1.124	2.349	2.352	1.123	1.142	2.351	2.390	0.752	RTT	101.75
	17	1.112	1.110	2.348	2.348	1.111	1.130	2.348	2.388	0.759	RTT	94.25
SCHUMACHER SOLAR (UNUSED) 980 C	T3	1.167	1.177	2.325	2.317	1.172	1.192	2.321	2.360	0.744	HTT	20.50
	T14	1.161	1.172	2.352	2.354	1.167	1.186	2.353	2.393	0.750	HTT	24.00
	T18	1.139	1.142	2.330	2.315	1.141	1.160	2.323	2.362	0.771	HTT	24.75
	B4	1.129	1.116	2.341	2.315	1.123	1.142	2.328	2.367	0.752	HTT	25.50
	B8	1.127	1.122	2.330	2.312	1.125	1.144	2.321	2.360	0.743	HTT	22.50
SCHUMACHER SOLAR (USED) 980 C	3	1.188	1.193	2.368	2.355	1.191	1.211	2.362	2.402	0.796	HTT	20.75
	7	1.181	1.180	2.362	2.345	1.181	1.201	2.354	2.393	0.786	HTT	15.50
	11	1.157	1.159	2.360	2.349	1.158	1.178	2.355	2.394	0.795	HTT	25.00
	15	1.155	1.154	2.361	2.345	1.155	1.174	2.353	2.393	0.798	HTT	22.00
	19	1.111	1.111	2.344	2.369	1.111	1.130	2.357	2.396	0.797	HTT	14.75
SCHUMACHER (UNUSED) 870C	2	1.179	1.185	2.365	2.369	1.182	1.202	2.367	2.407	0.748	HTT	45.50
	4	1.184	1.179	2.362	2.364	1.182	1.202	2.363	2.403	0.761	HTT	42.25
	6	1.164	1.177	2.360	2.355	1.171	1.190	2.358	2.397	0.763	HTT	45.25
	8	1.160	1.155	2.362	2.359	1.158	1.177	2.361	2.401	0.760	HTT	44.50
	10	1.142	1.145	2.342	2.342	1.144	1.163	2.342	2.382	0.720	HTT	49.50
SCHUMACHER 5000 PULSES @ 0.5 sec 870 C	2	1.100	1.100	2.330	2.356	1.100	1.119	2.343	2.383	0.755	HTT	43.25
	4	1.097	1.080	2.332	2.349	1.089	1.107	2.341	2.380	0.723	HTT	41.25
IFP LAYCER-DOE (USED) 870 C	4	1.530	1.522	2.302	2.293	1.526	1.552	2.298	2.336	0.750	HTT	26.75
	8	1.520	1.511	2.310	2.303	1.516	1.541	2.307	2.346	0.760	HTT	21.50
	12	1.499	1.507	2.296	2.304	1.503	1.528	2.300	2.339	0.760	HTT	24.50
	16	1.488	1.500	2.300	2.314	1.494	1.519	2.307	2.346	0.772	HTT	31.50
	22	1.435	1.440	2.291	2.298	1.438	1.462	2.295	2.333	0.752	HTT	45.25
IFP LAYCER (UNUSED) 870 C	3	1.566	1.569	2.321	2.321	1.568	1.594	2.321	2.360	0.755	HTT	30.00
	5	1.543	1.563	2.316	2.314	1.553	1.579	2.315	2.354	0.745	HTT	31.00
	7	1.563	1.559	2.320	2.316	1.561	1.587	2.318	2.357	0.759	HTT	38.25

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SPECIMEN	SPECIMEN TEST NUMBER		Ri (in.)	Ro (in.)	w (in)	R=Ri(T) R=Ro(C)	STRESS (psi)	STRESS (MPa)		STRESS (psi)	STRESS (MPa)
	11	RTT	0.79	1.18	0.76	0.79	1988	13.70			
IFP LAYCER	2	RTT	0.77	1.19	0.76	0.77	2804	19.33	MAX=	2928	20.19
1000 PULSES	3	RTT	0.78	1.19	0.77	0.78	2461	16.97	MIN=	2461	16.97
@0.05 sec	4	RTT	0.78	1.20	0.76	0.78	2553	17.60	AVE=	2707	18.66
	5	RTT	0.78	1.19	0.76	0.78	2788	19.22	STD=	173	1.19
	6	RTT	0.78	1.19	0.76	0.78	2928	20.19			
IFP LAYCER-DOE (USED)	3	RTT	0.77	1.17	0.78	0.77	1867	12.87	MAX=	2488	17.15
	7	RTT	0.77	1.17	0.75	0.77	1557	10.73	MIN=	1274	8.79
	11	RTT	0.76	1.17	0.76	0.76	1274	8.79	AVE=	1726	11.90
	15	RTT	0.76	1.17	0.78	0.76	1446	9.97	STD=	427	2.94
	21	RTT	0.73	1.17	0.75	0.73	2488	17.15			
	1	RTT	0.59	1.19	0.76	0.59	2659	18.33	MAX=	2821	19.45
SCHUMACHER	5	RTT	0.59	1.19	0.76	0.59	2580	17.79	MIN=	2521	17.38
KRW 7067	9	RTT	0.58	1.19	0.76	0.58	2547	17.56	AVE=	2626	18.10
(USED)	13	RTT	0.57	1.19	0.75	0.57	2821	19.45	STD=	108	0.74
	17	RTT	0.56	1.20	0.75	0.56	2521	17.38			
SCHUMACHER	1	RTT	0.59	1.19	0.75	0.59	2609	17.99	MAX=	2938	20.26
KRW 7052	5	RTT	0.59	1.19	0.76	0.59	2652	18.29	MIN=	2377	16.39
(USED)	9	RTT	0.58	1.19	0.76	0.58	2938	20.26	AVE=	2641	18.21
	13	RTT	0.57	1.20	0.75	0.57	2630	18.13	STD=	178	1.23
	17	RTT	0.56	1.19	0.76	0.56	2377	16.39			
SCHUMACHER	T3	HTT	0.60	1.18	0.74	0.60	599	4.13	MAX=	676	4.66
SOLAR (UNUSED)	T14	HTT	0.59	1.20	0.75	0.59	663	4.57	MIN=	599	4.13
980 C	T18	HTT	0.58	1.18	0.77	0.58	662	4.57	AVE=	642	4.43
	B4	HTT	0.57	1.18	0.75	0.57	676	4.66	STD=	31	0.21
	B8	HTT	0.57	1.18	0.74	0.57	610	4.21			
SCHUMACHER	3	HTT	0.61	1.20	0.80	0.61	555	3.83	MAX=	642	4.42
SOLAR (USED)	7	HTT	0.60	1.20	0.79	0.60	417	2.88	MIN=	351	2.42
980 C	11	HTT	0.59	1.20	0.80	0.59	642	4.42	AVE=	505	3.48
	15	HTT	0.59	1.20	0.80	0.59	560	3.86	STD=	106	0.73
	19	HTT	0.56	1.20	0.80	0.56	351	2.42			
SCHUMACHER	2	HTT	0.60	1.20	0.75	0.60	1269	8.75	MAX=	1392	9.60
(UNUSED)	4	HTT	0.60	1.20	0.76	0.60	1163	8.02	MIN=	1163	8.02
870C	6	HTT	0.60	1.20	0.76	0.60	1229	8.48	AVE=	1248	8.60
	8	HTT	0.59	1.20	0.76	0.59	1185	8.17	STD=	81	0.56
	10	HTT	0.58	1.19	0.72	0.58	1392	9.60			
SCHUMACHER	2	HTT	0.56	1.19	0.76	0.56	1085	7.48	AVE=	1075	7.41
5000 PULSES	4	HTT	0.55	1.19	0.72	0.55	1066	7.35	STD=	10	0.07
@ 0.5 sec											
870 C											
IFP LAYCER-DOE (USED)	4	HTT	0.78	1.17	0.75	0.78	1652	11.39	MAX=	2267	15.63
870 C	8	HTT	0.77	1.17	0.76	0.77	1252	8.64	MIN=	1252	8.64
	12	HTT	0.76	1.17	0.76	0.76	1402	9.67	AVE=	1657	11.43
	16	HTT	0.76	1.17	0.77	0.76	1712	11.81	STD=	348	2.40
	22	HTT	0.73	1.17	0.75	0.73	2267	15.63			
IFP LAYCER (UNUSED)	3	HTT	0.80	1.18	0.76	0.80	1946	13.42	MAX=	2631	18.14
870 C	5	HTT	0.79	1.18	0.75	0.79	1989	13.71	MIN=	1946	13.42
	7	HTT	0.79	1.18	0.76	0.79	2443	16.85	AVE=	2318	15.98

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SPECIMEN	SPECIMEN NUMBER	CORRECTED		CORRECTED		AVE. I.D. (in.)	I.D. (in.)	AVE. O.D. (in.)	O.D. (in.)	WIDTH (in.)	TEST	LOAD (lbs)
		I.D. 1 (in.)	I.D. 2 (in.)	O.D. 1 (in.)	O.D. 2 (in.)							
IFP LAYCER 5000 PULSES @0.5 sec 870 C	9	1.558	1.550	2.313	2.307	1.554	1.580	2.310	2.349	0.751	HTT	40.00
	11	1.545	1.544	2.304	2.300	1.545	1.571	2.302	2.341	0.754	HTT	41.25
	2	1.554	1.556	2.306	2.318	1.555	1.581	2.312	2.351	0.750	HTT	38.00
	4	1.542	1.561	2.309	2.317	1.552	1.578	2.313	2.352	0.739	HTT	37.50
	6	1.562	1.547	2.325	2.327	1.555	1.581	2.326	2.365	0.765	HTT	47.25
	8	1.553	1.560	2.322	2.330	1.557	1.583	2.326	2.365	0.759	HTT	39.25
	10	1.550	1.544	2.326	2.315	1.547	1.573	2.321	2.360	0.743	HTT	45.00
	2	1.165	1.158	2.350	2.350	1.162	1.181	2.350	2.390	0.773	HTT	131.75
SCHUMACHER KRW 7068	6	1.144	1.148	2.346	2.348	1.146	1.165	2.347	2.387	0.755	HTT	132.00
(UNUSED) 650 C	10	1.122	1.132	2.358	2.357	1.127	1.146	2.358	2.397	0.773	HTT	141.50
	13	1.120	1.112	2.361	2.359	1.116	1.135	2.360	2.400	0.768	HTT	144.75
	16	1.111	1.112	2.356	2.363	1.112	1.130	2.360	2.399	0.770	HTT	161.25
SCHUMACHER	2	1.163	1.155	2.345	2.341	1.159	1.179	2.343	2.383	0.760	HTT	102.00
KRW 7052	6	1.150	1.149	2.347	2.340	1.150	1.169	2.344	2.383	0.758	HTT	113.75
(USED) 650 C	10	1.128	1.137	2.348	2.344	1.133	1.152	2.346	2.386	0.758	HTT	105.25
	14	1.118	1.114	2.354	2.348	1.116	1.135	2.351	2.391	0.763	HTT	115.50
	18	1.110	1.113	2.349	2.348	1.112	1.130	2.349	2.388	0.768	HTT	108.75
SCHUMACHER	2	1.158	1.167	2.340	2.326	1.163	1.182	2.333	2.373	0.753	HTT	110.50
KRW 7067	6	1.148	1.145	2.341	2.330	1.147	1.166	2.336	2.375	0.760	HTT	112.25
(USED) 650 C	10	1.128	1.128	2.347	2.334	1.128	1.147	2.341	2.380	0.758	HTT	119.50
	14	1.108	1.116	2.349	2.341	1.112	1.131	2.345	2.385	0.764	HTT	119.25
	18	1.111	1.113	2.348	2.352	1.112	1.131	2.350	2.390	0.757	HTT	125.75
SCHUMACHER	T4	1.190	1.203	2.365	2.349	1.197	1.217	2.357	2.397	0.753	RTC	-48.00
SOLAR (UNUSED)	T15	1.165	1.173	2.353	2.343	1.169	1.189	2.348	2.388	0.780	RTC	-48.25
	T19	1.155	1.163	2.358	2.345	1.159	1.179	2.352	2.391	0.764	RTC	-53.50
	B5	1.143	1.132	2.354	2.337	1.138	1.157	2.346	2.385	0.754	RTC	-52.75
	B9	1.125	1.141	2.313	2.339	1.133	1.152	2.326	2.365	0.742	RTC	-45.00
SCHUMACHER	4	1.187	1.191	2.366	2.352	1.189	1.209	2.359	2.399	0.804	RTC	-22.75
SOLAR (USED)	8	1.170	1.176	2.366	2.352	1.173	1.193	2.359	2.399	0.785	RTC	-15.00
	12	1.150	1.159	2.366	2.355	1.155	1.174	2.361	2.401	0.795	RTC	-49.00
	16	1.150	1.155	2.364	2.354	1.153	1.172	2.359	2.399	0.793	RTC	-48.50
	20	1.110	1.105	2.345	2.332	1.108	1.126	2.339	2.378	0.790	RTC	-45.50
IFP LAYCER-DOE	9	1.516	1.512	2.305	2.306	1.514	1.540	2.306	2.345	0.759	RTC	-80.75
(USED)	13	1.500	1.503	2.302	2.308	1.502	1.527	2.305	2.344	0.752	RTC	-89.25
	17	1.493	1.493	2.303	2.303	1.493	1.518	2.303	2.342	0.782	RTC	-79.00
	23	1.434	1.441	2.292	2.299	1.438	1.462	2.296	2.334	0.760	RTC	-108.25
	27	1.422	1.420	2.292	2.291	1.421	1.445	2.292	2.330	0.732	RTC	-99.00
SCHUMACHER	3	1.158	1.163	2.342	2.347	1.161	1.180	2.345	2.384	0.761	RTC	-227.50
KRW 7068	7	1.138	1.146	2.347	2.350	1.142	1.161	2.349	2.388	0.780	RTC	-236.75
(UNUSED)	11	1.113	1.120	2.349	2.351	1.117	1.135	2.350	2.390	0.773	RTC	-252.50
	14	1.116	1.111	2.363	2.362	1.114	1.132	2.363	2.403	0.772	RTC	-262.00
	17	1.110	1.111	2.357	2.361	1.111	1.129	2.359	2.399	0.760	RTC	-282.50
SCHUMACHER	3	1.162	1.163	2.347	2.342	1.163	1.182	2.345	2.384	0.762	RTC	-169.00
KRW 7052	7	1.139	1.148	2.346	2.342	1.144	1.163	2.344	2.384	0.757	RTC	-194.25
(USED)	11	1.113	1.130	2.350	2.347	1.122	1.141	2.349	2.388	0.761	RTC	-191.75
	15	1.107	1.107	2.348	2.348	1.107	1.126	2.348	2.388	0.766	RTC	-214.25
	19	1.113	1.111	2.354	2.354	1.112	1.131	2.354	2.394	0.760	RTC	-260.50
SCHUMACHER	3	1.161	1.160	2.337	2.325	1.161	1.180	2.331	2.371	0.754	RTC	-225.25
KRW 7067	7	1.150	1.155	2.343	2.336	1.153	1.172	2.340	2.379	0.759	RTC	-225.75

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SPECIMEN	SPECIMEN TEST NUMBER		Ri (in.)	Ro (in.)	w (in)	R=Ri(T) R=Ro(C)	STRESS (psi)	STRESS (MPa)	STRESS (psi)	STRESS (MPa)
	9	HTT	0.79	1.17	0.75	0.79	-2580	17.79	STD=	293 2.02
	11	HTT	0.79	1.17	0.75	0.79	2631	18.14		
IFP LAYCER	2	HTT	0.79	1.18	0.75	0.79	2450	16.89	MAX=	2895 19.96
5000 PULSES	4	HTT	0.79	1.18	0.74	0.79	2427	16.73	MIN=	2427 16.73
@0.5 sec	6	HTT	0.79	1.18	0.77	0.79	2895	19.96	AVE=	2605 17.96
870 C	8	HTT	0.79	1.18	0.76	0.79	2436	16.80	STD=	207 1.42
	10	HTT	0.79	1.18	0.74	0.79	2818	19.43		
SCHUMACHER	2	HTT	0.59	1.19	0.77	0.59	3516	24.24	MAX=	3961 27.31
KRW 7068	6	HTT	0.58	1.19	0.76	0.58	3533	24.36	MIN=	3516 24.24
(UNUSED) 650 C	10	HTT	0.57	1.20	0.77	0.57	3550	24.48	AVE=	3629 25.02
	13	HTT	0.57	1.20	0.77	0.57	3586	24.73	STD=	167 1.15
	16	HTT	0.57	1.20	0.77	0.57	3961	27.31		
SCHUMACHER	2	HTT	0.59	1.19	0.76	0.59	2781	19.17	MAX=	3062 21.11
KRW 7052	6	HTT	0.58	1.19	0.76	0.58	3062	21.11	MIN=	2711 18.69
(USED) 650 C	10	HTT	0.58	1.19	0.76	0.58	2752	18.97	AVE=	2843 19.60
	14	HTT	0.57	1.20	0.76	0.57	2910	20.06	STD=	128 0.88
	18	HTT	0.57	1.19	0.77	0.57	2711	18.69		
SCHUMACHER	2	HTT	0.59	1.19	0.75	0.59	3095	21.34	MAX=	3178 21.91
KRW 7067	6	HTT	0.58	1.19	0.76	0.58	3028	20.88	MIN=	3003 20.70
(USED) 650 C	10	HTT	0.57	1.19	0.76	0.57	3123	21.53	AVE=	3085 21.27
	14	HTT	0.57	1.19	0.76	0.57	3003	20.70	STD=	63 0.44
	18	HTT	0.57	1.19	0.76	0.57	3178	21.91		
SCHUMACHER	T4	RTC	0.61	1.20	0.75	1.20	700	4.82	MAX=	708 4.88
SOLAR (UNUSED)	T15	RTC	0.59	1.19	0.78	1.19	644	4.44	MIN=	600 4.14
	T19	RTC	0.59	1.20	0.76	1.20	708	4.88	AVE=	666 4.59
	B5	RTC	0.58	1.19	0.75	1.19	678	4.67	STD=	40 0.27
	B9	RTC	0.58	1.18	0.74	1.18	600	4.14		
SCHUMACHER	4	RTC	0.60	1.20	0.80	1.20	304	2.10	MAX=	607 4.19
SOLAR (USED)	8	RTC	0.60	1.20	0.79	1.20	197	1.36	MIN=	197 1.36
	12	RTC	0.59	1.20	0.80	1.20	607	4.19	AVE=	447 3.08
	16	RTC	0.59	1.20	0.79	1.20	601	4.15	STD=	166 1.15
	20	RTC	0.56	1.19	0.79	1.19	525	3.62		
IFP LAYCER-DOE	9	RTC	0.77	1.17	0.76	1.17	3088	21.29	MAX=	3354 23.13
(USED)	13	RTC	0.76	1.17	0.75	1.17	3317	22.87	MIN=	2764 19.06
	17	RTC	0.76	1.17	0.78	1.17	2764	19.06	AVE=	3117 21.49
	23	RTC	0.73	1.17	0.76	1.17	3354	23.13	STD=	212 1.46
	27	RTC	0.72	1.17	0.73	1.17	3062	21.11		
SCHUMACHER	3	RTC	0.59	1.19	0.76	1.19	3068	21.15	MAX=	3306 22.80
KRW 7068	7	RTC	0.58	1.19	0.78	1.19	2959	20.40	MIN=	2959 20.40
(UNUSED)	11	RTC	0.57	1.19	0.77	1.19	2989	20.61	AVE=	3069 21.16
	14	RTC	0.57	1.20	0.77	1.20	3024	20.85	STD=	124 0.86
	17	RTC	0.56	1.20	0.76	1.20	3306	22.80		
SCHUMACHER	3	RTC	0.59	1.19	0.76	1.19	2287	15.77	MAX=	3083 21.26
KRW 7052	7	RTC	0.58	1.19	0.76	1.19	2529	17.44	MIN=	2287 15.77
(USED)	11	RTC	0.57	1.19	0.76	1.19	2339	16.12	AVE=	2549 17.58
	15	RTC	0.56	1.19	0.77	1.19	2510	17.30	STD=	283 1.95
	19	RTC	0.57	1.20	0.76	1.20	3083	21.26		
SCHUMACHER	3	RTC	0.59	1.19	0.75	1.19	3134	21.61	MAX=	3206 22.11
KRW 7067	7	RTC	0.59	1.19	0.76	1.19	3018	20.81	MIN=	2830 19.51



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SPECIMEN	SPECIMEN NUMBER	I.D. 1 (in.)	I.D. 2 (in.)	O.D. 1 (in.)	O.D. 2 (in.)	CORRECTED		CORRECTED		WIDTH (in.)	TEST	LOAD (lbs)
						AVE. I.D. (in.)	I.D. (in.)	AVE. O.D. (in.)	O.D. (in.)			
(USED)	11	1.125	1.125	2.346	2.340	1.125	1.144	2.343	2.383	0.767	RTC	-260.50
	15	1.108	1.104	2.345	2.341	1.106	1.125	2.343	2.383	0.761	RTC	-238.75
	19	1.102	1.110	2.362	2.355	1.106	1.125	2.359	2.398	0.760	RTC	-250.25
SCHUMACHER	T5	1.191	1.197	2.369	2.355	1.194	1.214	2.362	2.402	0.748	HTC	-39.00
SOLAR (UNUSED)	T16	1.162	1.167	2.354	2.344	1.165	1.184	2.349	2.389	0.748	HTC	-36.50
980 C	T20	1.155	1.156	2.342	2.348	1.156	1.175	2.345	2.385	0.754	HTC	-36.50
	B10	1.141	1.160	2.333	2.355	1.151	1.170	2.344	2.384	0.752	HTC	-40.25
	B2	1.140	1.135	2.352	2.323	1.138	1.157	2.338	2.377	0.743	HTC	-37.00
SCHUMACHER	5	1.173	1.190	2.373	2.362	1.182	1.202	2.368	2.408	0.808	HTC	-30.75
SOLAR (USED)	9	1.168	1.150	2.332	2.327	1.159	1.179	2.330	2.369	0.803	HTC	-26.75
980 C	13	1.138	1.132	2.347	2.332	1.135	1.154	2.340	2.379	0.803	HTC	-31.75
	17	1.146	1.156	2.341	2.354	1.151	1.171	2.348	2.387	0.775	HTC	-24.75
	21	1.108	1.109	2.342	2.360	1.109	1.127	2.351	2.391	0.790	HTC	-34.50
IFP LAYCER-DOE	6	1.525	1.513	2.300	2.289	1.519	1.545	2.295	2.333	0.745	HTC	-69.50
(USED)	10	1.501	1.511	2.298	2.303	1.506	1.532	2.301	2.339	0.762	HTC	-73.25
870 C	14	1.492	1.500	2.301	2.305	1.496	1.521	2.303	2.342	0.763	HTC	-59.00
	20	1.434	1.444	2.293	2.298	1.439	1.463	2.296	2.334	0.759	HTC	-83.00
	24	1.429	1.443	2.308	2.311	1.436	1.460	2.310	2.349	0.773	HTC	-89.75
SCHUMACHER	4	1.147	1.143	2.342	2.346	1.145	1.164	2.344	2.384	0.759	HTC	-268.00
KRW 7068	8	1.136	1.134	2.355	2.352	1.135	1.154	2.354	2.393	0.747	HTC	-304.00
(UNUSED) 650 C												
SCHUMACHER	4	1.158	1.158	2.343	2.342	1.158	1.178	2.343	2.382	0.751	HTC	-181.50
KRW 7052	8	1.140	1.147	2.355	2.342	1.144	1.163	2.349	2.388	0.760	HTC	-206.75
(USED) 650 C	12	1.130	1.133	2.349	2.346	1.132	1.151	2.348	2.387	0.759	HTC	-238.00
	16	1.113	1.111	2.352	2.348	1.112	1.131	2.350	2.390	0.768	HTC	-241.00
SCHUMACHER	4	1.168	1.169	2.331	2.338	1.169	1.188	2.335	2.374	0.759	HTC	-223.25
KRW 7067	8	1.143	1.142	2.343	2.338	1.143	1.162	2.341	2.380	0.763	HTC	-247.50
(USED) 650 C	12	1.129	1.131	2.335	2.337	1.130	1.149	2.336	2.376	0.773	HTC	-250.25
	16	1.108	1.111	2.346	2.341	1.110	1.128	2.344	2.383	0.764	HTC	-246.25
EXTRA TESTING =====												
SCHUMACHER	T9	1.158	1.162	2.314	2.324	1.160	1.180	2.319	2.358	0.736	RTT(E)	27.00
SOLAR (UNUSED)												
IFP LAYCER	12	1.565	1.582	2.328	2.335	1.574	1.600	2.332	2.371	0.786	RTT(E)	45.00
(UNUSED)												
SCHUMACHER	B13 870	1.146	1.128	2.348	2.319	1.137	1.156	2.334	2.373	0.748	HTT(E)	34.25
SOLAR (UNUSED)	B14 870	1.146	1.125	2.353	2.320	1.136	1.155	2.337	2.376	0.726	HTT(E)	30.75
	B15 870	1.128	1.144	2.344	2.316	1.136	1.155	2.330	2.369	0.744	HTT(E)	28.50

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SPECIMEN	SPECIMEN TEST NUMBER		Ri (in.)	Ro (in.)	w (in)	R=Ri(T) R=Ro(C)	STRESS (psi)	STRESS (MPa)		STRESS (psi)	STRESS (MPa)
(USED)	11	RTC	0.57	1.19	0.77	1.19	3206	22.11	AVE=	3018	20.81
	15	RTC	0.56	1.19	0.76	1.19	2830	19.51	STD=	140	0.97
	19	RTC	0.56	1.20	0.76	1.20	2900	20.00			
SCHUMACHER	T5	HTC	0.61	1.20	0.75	1.20	564	3.89	MAX=	564	3.89
SOLAR (UNUSED)	T16	HTC	0.59	1.19	0.75	1.19	502	3.46	MIN=	489	3.37
980 C	T20	HTC	0.59	1.19	0.75	1.19	490	3.38	AVE=	516	3.56
	B10	HTC	0.58	1.19	0.75	1.19	536	3.70	STD=	29	0.20
	B2	HTC	0.58	1.19	0.74	1.19	489	3.37			
SCHUMACHER	5	HTC	0.60	1.20	0.81	1.20	396	2.73	MAX=	396	2.73
SOLAR (USED)	9	HTC	0.59	1.18	0.80	1.18	349	2.41	MIN=	319	2.20
980 C	13	HTC	0.58	1.19	0.80	1.19	385	2.65	AVE=	368	2.54
	17	HTC	0.59	1.19	0.78	1.19	319	2.20	STD=	30	0.20
	21	HTC	0.56	1.20	0.79	1.20	391	2.70			
IFP LAYCER-DOE	6	HTC	0.77	1.17	0.75	1.17	2828	19.50	MAX=	2828	19.50
(USED)	10	HTC	0.77	1.17	0.76	1.17	2755	19.00	MIN=	2135	14.72
870 C	14	HTC	0.76	1.17	0.76	1.17	2135	14.72	AVE=	2589	17.85
	20	HTC	0.73	1.17	0.76	1.17	2587	17.84	STD=	242	1.67
	24	HTC	0.73	1.17	0.77	1.17	2637	18.18			
SCHUMACHER	4	HTC	0.58	1.19	0.76	1.19	3492	24.08	AVE=	3681	25.38
KRW 7068	8	HTC	0.58	1.20	0.75	1.20	3870	26.69	STD=	189	1
(UNUSED) 650 C											
SCHUMACHER	4	HTC	0.59	1.19	0.75	1.19	2473	17.05	MAX=	2986	20.58
KRW 7052	8	HTC	0.58	1.19	0.76	1.19	2662	18.35	MIN=	2473	17.05
(USED) 650 C	12	HTC	0.58	1.19	0.76	1.19	2986	20.58	AVE=	2740	18.89
	16	HTC	0.57	1.19	0.77	1.19	2840	19.58	STD=	192	1.33
SCHUMACHER	4	HTC	0.59	1.19	0.76	1.19	3129	21.57	MAX=	3207	22.11
KRW 7067	8	HTC	0.58	1.19	0.76	1.19	3207	22.11	MIN=	2930	20.20
(USED) 650 C	12	HTC	0.57	1.19	0.77	1.19	3128	21.57	AVE=	3098	21.36
	16	HTC	0.56	1.19	0.76	1.19	2930	20.20	STD=	103	0.71

## EXTRA TESTING =====

SCHUMACHER	T9	RTT(E	0.59	1.18	0.74	0.59	784	5.40			
SOLAR (UNUSED)											
IFP LAYCER	12	RTT(E	0.80	1.19	0.79	0.80	2784	19.20			
(UNUSED)											
SCHUMACHER	B13	870HTT(E	0.58	1.19	0.75	0.58	927	6.39	AVG=	853	5.88
SOLAR (UNUSED)	B14	870HTT(E	0.58	1.19	0.73	0.58	853	5.88	STD=	61	0.42
	B15	870HTT(E	0.58	1.18	0.74	0.58	778	5.36			

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SPECIMEN	SPECIMEN NUMBER	CORRECTED			CORRECTED			WIDTH	TEST	LOAD
		I.D. 1 (in.)	I.D. 2 (in.)	O.D. 1 (in.)	O.D. 2 (in.)	AVE. I.D. (in.)	I.D. (in.)	AVE. O.D. (in.)	O.D. (in.)	
	B22 925	1.135	1.157	2.312	2.341	1.146	1.165	2.327	2.366	0.736 HTT(E 35.00
	B23 900	1.165	1.167	2.349	2.368	1.166	1.186	2.359	2.398	0.785 HTT(E 35.75
	B11 910	1.126	1.139	2.313	2.335	1.133	1.152	2.324	2.363	0.751 HTT(E 36.75
	B12 920	1.147	1.155	2.339	2.358	1.151	1.171	2.349	2.388	0.743 HTT(E 35.75
	T21 925	1.151	1.161	2.358	2.342	1.156	1.176	2.350	2.390	0.750 HTT(E 38.50
	T23 950	1.131	1.133	2.338	2.317	1.132	1.151	2.328	2.367	0.725 HTT(E 28.25
	T22 980	1.147	1.150	2.358	2.345	1.149	1.168	2.352	2.391	0.756 HTT(E 21.75
DIDIER UNUSED	1 980	1.511	1.505	2.354	2.346	1.508	1.534	2.350	2.390	0.776 HTT(E 14.00
IFP LAY UNUSED	1 980	1.578	1.576	2.311	2.308	1.577	1.604	2.310	2.349	0.761 HTT(E 14.75

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SPECIMEN	SPECIMEN TEST NUMBER	Ri (in.)	Ro (in.)	w (in)	R=Ri(T) R=Ro(C)	STRESS (psi)	STRESS (MPa)	STRESS (psi)	STRESS (MPa)
	B22 925HTT(E	0.58	1.18	0.74	0.58	985	6.79		
	B23 900HTT(E	0.59	1.20	0.79	0.59	936	6.46		
	B11 910HTT(E	0.58	1.18	0.75	0.58	995	6.86		
	B12 920HTT(E	0.59	1.19	0.74	0.59	978	6.74		
	T21 925HTT(E	0.59	1.19	0.75	0.59	1050	7.24		
	T23 950HTT(E	0.58	1.18	0.73	0.58	789	5.44		
	T22 980HTT(E	0.58	1.20	0.76	0.58	581	4.00		
DIDIER UNUSED	1 980HTT(E	0.77	1.19	0.78	0.77	720	4.96		
IFP LAY UNUSED	1 980HTT(E	0.80	1.17	0.76	0.80	999	6.89		

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**APPENDIX D**  
**C-RING TESTING DATA**

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05/01/93  
schvrywd

# VARYING WIDTH TESTS OF SCHUMACHER

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample Standard Deviation
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.75" Compression @980°C	5	1.137	1.166	2.340	2.376	0.744	HTC	30.5	-30.5	410	2.82	Ave MPa= 3.44	
	13	1.136	1.150	2.371	2.334	0.753	HTC	41	-41	538	3.71	STD = 0.31	0.35
	21	1.110	1.105	2.378	2.333	0.792	HTC	45	-45	513	3.54		
	29	1.096	1.092	2.385	2.365	0.771	HTC	47	-47	518	3.57	Ave Width 0.770	
	35	1.095	1.091	2.364	2.386	0.791	HTC	48	-48	514	3.55	STD = 0.019	0.022
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.50" Compression @980°C	6	1.150	1.132	2.346	2.353	0.480	HTC	21	-21	432	2.98	Ave MPa= 3.28	
	14	1.148	1.164	2.370	2.335	0.503	HTC	22	-22	446	3.07	STD = 0.39	0.43
	22	1.124	1.112	2.376	2.330	0.500	HTC	28.5	-28.5	530	3.65		
	30	1.101	1.106	2.386	2.340	0.530	HTC	25	-25	417	2.88	Ave Width 0.505	
	36	1.102	1.105	2.390	2.363	0.514	HTC	33	-33	556	3.84	STD = 0.016	0.018
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.75" Compression @870°C	7	1.166	1.147	2.377	2.348	0.733	HTC	59.5	-59.5	815	5.62	Ave MPa= 5.68	
	15	1.142	1.336	2.343	2.382	0.744	HTC	54.5	-54.5	899	6.20	STD = 0.34	0.40
	23	1.134	1.130	2.384	2.345	0.743	HTC	64.5	-64.5	819	5.65		
	31	1.119	1.076	2.388	2.341	0.733	HTC	64	-64	760	5.24	Ave Width 0.738	
							HTC					STD = 0.005	0.006
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.50" Compression @870°C	8	1.142	1.108	2.368	2.329	0.495	HTC	21	-21	404	2.78	Ave MPa= 4.36	
	16	1.138	1.135	2.349	2.369	0.509	HTC	35	-35	662	4.56	STD = 0.95	1.10
	24	1.113	1.109	2.330	2.354	0.519	HTC	38.5	-38.5	690	4.76		
	32	1.102	1.106	2.341	2.379	0.518	HTC	45	-45	773	5.33	Ave Width 0.510	
							HTC					STD = 0.010	0.011
							RTT		0	ERR	ERR	Ave MPa= ERR	
							RTT		0	ERR	ERR	STD = ERR	ERR
							RTT		0	ERR	ERR		
							RTT		0	ERR	ERR	Ave Width ERR	
							RTT		0	ERR	ERR	STD = ERR	ERR
							RTT		0	ERR	ERR	Ave MPa= ERR	
							RTT		0	ERR	ERR	STD = ERR	ERR
							RTT		0	ERR	ERR		
							RTT		0	ERR	ERR	Ave Width ERR	
							RTT		0	ERR	ERR	STD = ERR	ERR



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# VARYING WIDTH TESTS OF SCHUMACHER

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample Standard Deviation
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.75" Tension @980°C	1	1.145	1.151	2.351	2.331	0.788	HTT	20	20	527	3.63	Ave MPa= 4.60	
	9	1.120	1.131	2.354	2.332	0.768	HTT	22	22	573	3.95	STD = 1.52	1.75
	17	1.470	1.183	2.369	2.341	0.791	HTT	30	30	1047	7.22		
	25	1.200	1.111	2.335	2.383	0.789	HTT	20	20	521	3.59	Ave Width 0.778	
	33	1.090	1.103	2.368	2.346	0.752	HTT		BROKE	DURING	INST.	STD = 0.015	0.018
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.50" Tension @980°C	2	1.164	1.134	2.390	2.348	0.509	HTT	14.5	14.5	573	3.95	Ave MPa= 4.32	
	10	1.160	1.121	2.388	2.353	0.511	HTT	17.5	17.5	679	4.68	STD = 0.40	0.46
	18	1.128	1.122	2.367	2.329	0.492	HTT	17	17	687	4.74		
	26	1.214	1.091	2.361	2.343	0.512	HTT	14	14	564	3.89	Ave Width 0.511	
	34	1.102	1.095	2.385	2.236	0.530	HTT		BROKE	DURING	INST.	STD = 0.012	0.014
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.75" Tension @870°C	3	1.165	1.143	2.362	2.337	0.743	HTT	34	34	949	6.55	Ave MPa= 6.74	
	11	1.162	1.142	2.331	2.372	0.758	HTT	37	37	1007	6.94	STD = 0.60	0.69
	19	1.122	1.119	2.378	2.356	0.755	HTT	33.5	33.5	858	5.91		
	27	1.223	1.191	2.387	2.342	0.757	HTT	37.5	37.5	1097	7.57	Ave Width 0.753	
							HTT					STD = 0.006	0.007
SCHUMACHER F40 (FROM SOLAR) VARY WIDTHS UNUSED 0.50" Tension @870°C	4	1.135	1.143	2.335	2.341	0.495	HTT	20.5	20.5	851	5.87	Ave MPa= 6.11	
	12	1.136	1.142	2.345	2.330	0.518	HTT	21	21	833	5.75	STD = 0.31	0.36
	20	1.128	1.137	2.386	2.372	0.505	HTT	24	24	922	6.36		
	28	1.127	1.102	2.397	2.341	0.520	HTT	25.5	25.5	937	6.46	Ave Width 0.510	
							HTT					STD = 0.010	0.012
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	ERR
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	ERR
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	ERR
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	ERR

02/25/93

505vrywd

## VARYING WIDTH TESTS FROM 0.50" TO 1.25"

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample Standard Deviation
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 0.50" UNUSED #8	2	1.593	1.581	2.347	2.337	0.508	RTT	20.2	20.2	1990	13.72	Ave MPa= 14.47	
	13	1.563	1.559	2.337	2.343	0.482	RTT	21.7	21.7	2118	14.60	STD = 1.40	1.57
	23	1.575	1.579	2.368	2.377	0.475	RTT	18.4	18.4	1772	12.22		
	33	1.568	1.573	2.364	2.370	0.510	RTT	26.3	26.3	2348	16.19	Ave Width 0.490	
	43	1.534	1.532	2.339	2.342	0.475	RTT	24.5	24.5	2262	15.59	STD = 0.016	0.018
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 0.75" UNUSED #8	3	1.583	1.591	2.343	2.352	0.740	RTT	30.4	30.4	2032	14.01	Ave MPa= 15.25	
	14	1.570	1.562	2.350	2.346	0.742	RTT	32.7	32.7	2064	14.23	STD = 1.19	1.34
	24	1.541	1.545	2.334	2.334	0.743	RTT	35.5	35.5	2176	15.00		
	34	1.543	1.533	2.339	2.332	0.739	RTT	41.4	41.4	2512	17.32	Ave Width 0.741	
	44	1.550	1.544	2.349	2.349	0.740	RTT	37.8	37.8	2278	15.71	STD = 0.001	0.002
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 1.00" UNUSED #8	5	1.558	1.560	2.320	2.321	1.121	RTT	49.5	49.5	2154	14.85	Ave MPa= 15.46	
	15	1.559	1.559	2.342	2.342	1.001	RTT	47.0	47	2188	15.09	STD = 0.98	1.09
	25	1.545	1.545	2.335	2.335	0.990	RTT	50.4	50.4	2325	16.03		
	35	1.546	1.533	2.337	2.330	1.019	RTT	46.7	46.7	2071	14.28	Ave Width 1.026	
	45	1.529	1.533	2.330	2.334	1.001	RTT	55.8	55.8	2475	17.06	STD = 0.048	0.054
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 1.25" UNUSED #8	6	1.557	1.564	2.321	2.332	1.277	RTT	62.7	62.7	2374	16.37	Ave MPa= 15.97	
	16	1.570	1.577	2.358	2.364	1.254	RTT	56.9	56.9	2107	14.53	STD = 0.98	1.10
	26	1.571	1.560	2.367	2.356	1.288	RTT	62.0	62	2190	15.10		
	36	1.559	1.553	2.356	2.351	1.259	RTT	69.0	69	2476	17.07	Ave Width 1.267	
	46	1.530	1.539	2.334	2.344	1.256	RTT	69.2	69.2	2432	16.77	STD = 0.013	0.015
REFRACTRON 50/5 WITH 505 BINDER HEAT TREATED TO 980°C (0.50") UNUSED #8	1	1.571	1.571	2.322	2.323	0.490	RTT	18.4	18.4	1882	12.97	Ave MPa= 12.91	
	12	1.562	1.570	2.340	2.347	0.513	RTT	19.3	19.3	1779	12.26	STD = 0.46	0.51
	22	1.567	1.575	2.359	2.368	0.495	RTT	19.6	19.6	1818	12.54		
	32	1.534	1.539	2.332	2.341	0.493	RTT	21.3	21.3	1926	13.28	Ave Width 0.491	
	42	1.533	1.532	2.337	2.337	0.464	RTT	20.6	20.6	1958	13.50	STD = 0.016	0.018
							RTT		0	0	0.00	Ave MPa= 0.00	
							RTT		0	0	0.00	STD = 0.00	0.00
							RTT		0	0	0.00		
							RTT		0	0	0.00	Ave Width 0.478	
							RTT		0	0	0.00	STD = 0.005	0.006

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505vrywd

# VARYING WIDTH TESTS FROM 0.50" TO 1.25"

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample Standard Deviation
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 0.50" UNUSED #8	8	1.558	1.553	2.323	2.319	0.481	RTC	40.7	-40.7	2739	18.88	Ave MPa= 19.20	
	18	1.580	1.577	2.369	2.368	0.499	RTC	44.3	-44.3	2739	18.89	STD = 0.57	0.64
	28	1.566	1.570	2.359	2.365	0.492	RTC	45.6	-45.6	2813	19.40		
	38	1.549	1.545	2.345	2.349	0.486	RTC	48.3	-48.3	2933	20.22	Ave Width 0.489	
	48	1.549	1.533	2.351	2.343	0.485	RTC	45.2	-45.2	2700	18.62	STD = 0.006	0.007
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 0.75" UNUSED #8	9	1.567	1.554	2.328	2.324	0.758	RTC	54.2	-54.2	2322	16.01	Ave MPa= 18.12	
	19	1.558	1.572	2.346	2.356	0.766	RTC	68.9	-68.9	2780	19.17	STD = 1.88	2.10
	29	1.564	1.553	2.355	2.347	0.769	RTC	71.1	-71.1	2800	19.31		
	39	1.550	1.552	2.351	2.354	0.736	RTC	57.0	-57	2283	15.74	Ave Width 0.754	
	49	1.518	1.517	2.325	2.323	0.742	RTC	76.9	-76.9	2955	20.37	STD = 0.013	0.015
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 1.00" UNUSED #8	10	1.599	1.592	2.371	2.366	1.022	RTC	85.9	-85.9	2735	18.86	Ave MPa= 17.87	
	20	1.553	1.548	2.340	2.337	1.021	RTC	75.6	-75.6	2256	15.56	STD = 1.51	1.69
	30	1.542	1.536	2.338	2.329	1.013	RTC	81.8	-81.8	2404	16.57		
	40	1.551	1.548	2.348	2.344	0.997	RTC	93.9	-93.9	2808	19.36	Ave Width 1.013	
	50	1.561	1.556	2.369	2.361	1.014	RTC	95.4	-95.4	2753	18.98	STD = 0.009	0.010
REFRACTRON 50/5 WITH 505 BINDER VARY WIDTHS 1.25" UNUSED #8	11	1.597	1.590	2.376	2.368	1.263	RTC	107.0	-107	2716	18.72	Ave MPa= 18.69	
	21	1.582	1.579	2.375	2.369	1.264	RTC	118.3	-118	2880	19.86	STD = 1.65	1.85
	31	1.565	1.554	2.362	2.350	1.265	RTC	98.9	-98.9	2346	16.17		
	41	1.551	1.550	2.351	2.351	1.246	RTC	108.6	-109	2575	17.75	Ave Width 1.259	
	51	1.510	1.519	2.311	2.327	1.258	RTC	133.6	-134	3037	20.94	STD = 0.007	0.008
REFRACTRON 50/5 WITH 505 BINDER HEAT TREATED TO 980°C (0.50") UNUSED #8	7	1.555	1.568	2.322	2.330	0.529	RTC	36.8	-36.8	2266	15.62	Ave MPa= 16.73	
	17	1.556	1.557	2.340	2.342	0.494	RTC	41.0	-41	2561	17.66	STD = 1.10	1.23
	27	1.573	1.568	2.370	2.363	0.501	RTC	38.0	-38	2294	15.82		
	37	1.539	1.549	2.341	2.346	0.474	RTC	37.6	-37.6	2340	16.13	Ave Width 0.495	
	47	1.524	1.519	2.327	2.323	0.479	RTC	44.4	-44.4	2669	18.40	STD = 0.019	0.022
							RTC		0	0	0.00	Ave MPa= 0.00	
							RTC		0	0	0.00	STD = 0.00	0.00
							RTC		0	0	0.00		
							RTC		0	0	0.00	Ave Width 0.478	
							RTC		0	0	0.00	STD = 0.005	0.006

## VARYING WIDTH TESTS

[illegible]

## VARYING WIDTH TESTS

[illegible]

## VARYING WIDTH TESTS

[illegible]

05/01/93  
703vrywd

# UNUSED REFRACTRON 70/3 w/442-T BINDER

## VARYING WIDTH TESTS

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION-----		Small Sample STD
										STRESS psi	STRESS MPa	
REFRACTRON 70/3 WITH 442-T BINDER 980°C (1800°F) 0.50 inch Compression	8	1.548	1.568	2.319	2.337	0.485	HTC	16.5	-16.5	1090	7.52	Ave MPa= 8.61
	20	1.523	1.553	2.328	2.340	0.488	HTC	20.7	-20.7	1257	8.67	STD = 0.94
	32	1.552	1.534	2.358	2.332	0.490	HTC	21.0	-21.0	1256	8.66	
	44	1.529	1.535	2.358	2.339	0.484	HTC	19.8	-19.8	1149	7.92	Ave Width 0.486
	56	1.520	1.514	2.357	2.332	0.484	HTC	26.6	-26.6	1489	10.27	STD = 0.002
REFRACTRON 70/3 WITH 442-T BINDER 870°C (1600°F) 0.50 inch Compression	10	1.578	1.541	2.352	2.325	0.478	HTC	36.7	-36.7	2407	16.60	Ave MPa= 17.94
	22	1.563	1.525	2.333	2.350	0.477	HTC	44.3	-44.3	2753	18.98	STD = 0.80
	34	1.546	1.545	2.334	2.365	0.494	HTC	43.2	-43.2	2553	17.60	
	46	1.524	1.551	2.340	2.365	0.484	HTC	45.5	-45.5	2659	18.33	Ave Width 0.482
	58	1.537	1.521	2.334	2.346	0.477	HTC	44.3	-44.3	2638	18.19	STD = 0.007
REFRACTRON 70/3 WITH 442-T BINDER ROOM TEMP 0.50 inch Compression	12	1.546	1.577	2.319	2.326	0.464	RTC	44.0	-44.0	3117	21.49	Ave MPa= 19.42
	24	1.553	1.539	2.357	2.317	0.485	RTC	41.6	-41.6	2587	17.84	STD = 1.35
	36	1.541	1.518	2.348	2.330	0.483	RTC	48.7	-48.7	2875	19.83	
	48	1.509	1.515	2.328	2.362	0.497	RTC	48.8	-48.8	2617	18.05	Ave Width 0.481
	60	1.533	1.498	2.339	2.352	0.475	RTC	50.9	-50.9	2883	19.88	STD = 0.011
							HTC		0.0	ERR	ERR	Ave MPa= ERR
							HTC		0.0	ERR	ERR	STD = ERR
							HTC		0.0	ERR	ERR	
							HTC		0.0	ERR	ERR	Ave Width ERR
							HTC		0.0	ERR	ERR	STD = ERR
							RTT		0.0	ERR	ERR	Ave MPa= ERR
							RTT		0	ERR	ERR	STD = ERR
							RTT		0	ERR	ERR	
							RTT		0	ERR	ERR	Ave Width ERR
							RTT		0	ERR	ERR	STD = ERR

05/01/93  
505htttt

# REFRACTRON 50/5 w/505: HEAT TREATED UNUSED #11

>1 HOUR AT 1200°C

PRECUT A (1-5) POSTCUT A (6-12) PRECUT B (1-5) POSTCUT B (6-12)	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi   STRESS MPa			Small Sample STD
REFRACTRON 50/5 WITH 505 BINDER UNUSED (#11) PRECUT C-rings HEAT TO 1200°C	A1	1.536	1.534	2.362	2.344	0.729	RTT	30.9	30.9	1822	12.56	Ave MPa= 11.68	
	A3	1.507	1.523	2.346	2.348	0.741	RTT	31.2	31.2	1746	12.04	STD = 1.43	1.60
	A5	1.545	1.515	2.346	2.360	0.746	RTT	31.0	31.0	1765	12.17		
	B2	1.444	1.489	2.341	2.364	0.749	RTT	26.2	26.2	1286	8.87	Ave Width 0.744	
	B4	1.543	1.515	2.344	2.351	0.754	RTT	32.6	32.6	1852	12.77	STD = 0.009	0.010
REFRACTRON 50/5 WITH 505 BINDER UNUSED (#11) HEAT TO 1200°C POSTCUT C-ring	A6	1.538	1.542	2.361	2.359	0.755	RTT	33.4	33.4	1897	13.08	Ave MPa= 12.52	
	A8	1.530	1.537	2.360	2.367	0.740	RTT	34.2	34.2	1939	13.37	STD = 1.39	1.56
	A10	1.535	1.528	2.362	2.352	0.756	RTT	34.6	34.6	1935	13.34		
	B7	1.517	1.516	2.359	2.367	0.751	RTT	26.3	26.3	1413	9.74	Ave Width 0.751	
	B9	1.515	1.507	2.357	2.361	0.753	RTT	35.5	35.5	1893	13.05	STD = 0.006	0.006
REFRACTRON 50/5 WITH 505 BINDER UNUSED (#11) PRECUT C-rings HEAT TO 1200°C	A2	1.509	1.529	2.343	2.358	0.728	RTC	49.3	-49.3	1820	12.55	Ave MPa= 13.20	
	A4	1.537	1.545	2.354	2.346	0.743	RTC	54.3	-54.3	2102	14.49	STD = 0.73	0.81
	B1	1.460	1.540	2.368	2.335	0.747	RTC	57.0	-57.0	1933	13.33		
	B3	1.491	1.503	2.369	2.347	0.757	RTC	58.4	-58.4	1908	13.16	Ave Width 0.740	
	B5	1.540	1.475	2.376	2.336	0.725	RTC	51.2	-51.2	1810	12.48	STD = 0.012	0.013
REFRACTRON 50/5 WITH 505 BINDER UNUSED (#11) HEAT TO 1200°C POSTCUT C-ring	A7	1.518	1.528	2.353	2.357	0.741	RTC	52.1	-52.1	1892	13.04	Ave MPa= 15.46	
	A9	1.515	1.537	2.373	2.358	0.753	RTC	56.7	-56.7	1994	13.75	STD = 2.00	2.24
	B6	1.533	1.544	2.348	2.364	0.759	RTC	59.7	-59.7	2213	15.26		
	B8	1.528	1.539	2.353	2.362	0.739	RTC	64.6	-64.6	2413	16.64	Ave Width 0.748	
	B10	1.514	1.524	2.365	2.347	0.748	RTC	76.1	-76.1	2698	18.61	STD = 0.007	0.008
REFRACTRON 50/5 w/505 (Unused #11) HEAT TO 1200°C POSTCUT C-ring HTT at 980°C	A11	1.513	1.526	2.357	2.356	0.700	HTT	45.6	45.6	2680	18.48	Ave MPa= 18.73	
	A12	1.533	1.545	2.364	2.381	0.758	HTT	52.4	52.4	2886	19.90	STD = 0.69	0.80
	B11	1.493	1.506	2.356	2.347	0.751	HTT	49.7	49.7	2625	18.10		
	B12	1.530	1.523	2.341	2.364	0.810	HTT	51.4	51.4	2675	18.44	Ave Width 0.755	
							HTT					STD = 0.039	0.045
REFRACTRON 50/5 w/505 (Unused #11) Refractron heated cut c-rngs 1200°C POSTCUT C-ring	1	1.499	1.565	2.314	2.373	0.477	RTT	Load cell off				Ave MPa= 20.02	
	2	1.493	1.518	2.329	2.353	0.484	RTT	35.2	35.2	2984	20.58	STD = 0.96	1.11
	3	1.509	1.559	2.332	2.376	0.486	RTT	32.8	32.8	2888	19.91		
	4	1.513	1.518	2.322	2.370	0.479	RTT	35.2	35.2	3057	21.08	Ave Width 0.480	
	5	1.496	1.543	2.321	2.374	0.474	RTT	30.4	30.4	2685	18.52	STD = 0.004	0.005



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05/01/93 EXPOSURE TESTS: SCHUMACHER F40 DIA-SCHUMALITH

tetsch

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION-----		Small Sample STD
										STRESS psi	STRESS MPa	
SCHUMACHER TET CONTROL SAMPLE Heated to 870°C for 500 hours TENSION @ 870°C	2	1.175	1.187	2.381	2.376	0.508	HTT	25.4	25.4	1045	7.20	Ave MPa= 7.92
	4	1.176	1.172	2.381	2.385	0.489	HTT	BROKE				STD = 0.71
	6	1.173	1.184	2.373	2.386	0.480	HTT	28.9	28.9	1252	8.63	
							HTT					Ave Width 0.492
							HTT					STD = 0.012
SCHUMACHER TET CONTROL SAMPLE Heated to 870°C for 500 hours COMPR. @ 870°C	1	1.137	1.153	2.388	2.381	0.504	HTC	25.2	-25.2	472	3.25	Ave MPa= 4.49
	3	1.200	1.173	2.394	2.368	0.490	HTC	39.7	-39.7	848	5.85	STD = 0.93
	5	1.177	1.181	2.380	2.378	0.473	HTC	30.7	-30.7	670	4.62	
	7	1.190	1.147	2.395	2.373	0.478	HTC	29.4	-29.4	614	4.23	Ave Width 0.486
							HTC					STD = 0.012
SCHUMACHER TET 100hr/870°C/20%H2O 95ppm NaCl/No Pulse TENSION @ 870°C	1 A	1.183	1.189	2.353	2.376	0.504	HTT	17.0	17.0	723	4.98	Ave MPa= 4.85
	3 A	1.194	1.171	2.383	2.356	0.522	HTT	16.0	16.0	649	4.48	STD = 0.27
	5 A	1.186	1.169	2.372	2.332	0.494	HTT	17.0	17.0	738	5.09	
	A						HTT					Ave Width 0.507
	A						HTT					STD = 0.012
SCHUMACHER TET 100hr/870°C/20%H2O 95ppm NaCl/No Pulse COMPR @ 870°C	2 A	1.177	1.191	2.382	2.358	0.494	HTC	28.5	-28.5	611	4.22	Ave MPa= 3.93
	4 A	1.183	1.195	2.361	2.384	0.476	HTC	28.0	-28.0	628	4.33	STD = 0.48
	6 A	1.154	1.175	2.335	2.376	0.505	HTC	23.0	-23.0	471	3.25	
	A						HTC					Ave Width 0.492
	A						HTC					STD = 0.012
							RTT		0.0	ERR	ERR	Ave MPa= ERR
							RTT		0	ERR	ERR	STD = ERR
							RTT		0	ERR	ERR	
							RTT		0	ERR	ERR	Ave Width ERR
							RTT		0	ERR	ERR	STD = ERR

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05/01/93  
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# EXPOSURE TESTS: SCHUMACHER F40 DIA-SCHUMALITH

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample STD
SCHUMACHER TET	1 D	1.206	1.195	2.388	2.403	0.431	HTT	7.8	7.8	382	2.63	Ave MPa= 2.51	
575hr/980°C/20%H <sub>2</sub> O	3 D	1.207	1.206	2.386	2.418	0.486	HTT	8.9	8.9	387	2.67	STD = 0.20	0.25
65ppm NaCl(*1000hr*)	5 D	1.208	1.180	2.373	2.353	0.559	HTT	8.3	8.3	323	2.23		
	D						HTT					Ave Width 0.492	
TENSION @ 980°C	D						HTT					STD = 0.052	0.064
SCHUMACHER TET	2 D	1.208	1.185	2.386	2.413	0.488	HTC	11.4	-11.4	243	1.68	Ave MPa= 1.46	
575hr/980°C/20%H <sub>2</sub> O	4 D	1.178	1.200	2.383	2.401	0.517	HTC	11.4	-11.4	228	1.57	STD = 0.24	0.30
65ppm NaCl(*1000hr*)	6 D	1.151	1.146	2.393	2.377	0.520	HTC	8.9	-8.9	163	1.12		
	D						HTC					Ave Width 0.508	
COMPR @ 980°C	D						HTC					STD = 0.014	0.018
							HTT		0.0	ERR	ERR	Ave MPa= ERR	
							HTT		0.0	ERR	ERR	STD = ERR	ERR
							HTT		0.0	ERR	ERR		
							HTT		0.0	ERR	ERR	Ave Width ERR	
							HTT		0.0	ERR	ERR	STD = ERR	ERR
							HTC		0.0	ERR	ERR	Ave MPa= ERR	
							HTC		0.0	ERR	ERR	STD = ERR	ERR
							HTC		0.0	ERR	ERR		
							HTC		0.0	ERR	ERR	Ave Width ERR	
							HTC		0.0	ERR	ERR	STD = ERR	ERR
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	

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	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS	LOAD lb	LOAD lb	-----FERBER EQUATION-----		Small Sample STD
								LÓAD lb			STRESS psi	STRESS MPa	
SCHUMACHER TET 100hr/870°C/15%H2O 20ppm NaCl TENSION @ 870°C	2 E	1.178	1.159	2.389	2.344	0.484	HTT	17.8	17.8	765	5.27	Ave MPa= 5.72	
	4 E	1.168	1.171	2.384	2.381	0.469	HTT	19.8	19.8	863	5.95	STD = 0.32	0.39
	6 E	1.161	1.185	2.371	2.344	0.482	HTT	19.6	19.6	861	5.93		
	E						HTT					Ave Width 0.478	
	E						HTT					STD = 0.007	0.008
SCHUMACHER TET 100hr/870°C/15%H2O 20ppm NaCl COMPR @ 870°C	1 E	1.173	1.179	2.381	2.364	0.481	HTC	18.1	-18.1	390	2.69	Ave MPa= 4.07	
	3 E	1.182	1.160	2.376	2.365	0.460	HTC	27.8	-27.8	620	4.28	STD = 0.84	0.97
	5 E	1.155	1.171	2.388	2.380	0.459	HTC	29.5	-29.5	633	4.37		
	7 E	1.158	1.175	2.399	2.368	0.529	HTC	38.2	-38.2	718	4.95	Ave Width 0.482	
	E						HTC					STD = 0.028	0.033
SCHUMACHER TET 400hr/870°C/15%H2O 20ppm NaCl TENSION @ 870°C	2 F	1.156	1.165	2.351	2.358	0.453	HTT	7.4	7.4	340	2.35	Ave MPa= 4.76	
	4 F	1.170	1.185	2.359	2.385	0.463	HTT	20.6	20.6	932	6.43	STD = 1.75	2.14
	6 F	1.183	1.164	2.391	2.383	0.462	HTT	18.0	18.0	797	5.49		
	F						HTT					Ave Width 0.459	
	F						HTT					STD = 0.004	0.006
SCHUMACHER TET 400hr/870°C/15%H2O 20ppm NaCl COMPR @ 870°C	1 F	1.165	1.168	2.341	2.368	0.509	HTC	32.0	-32.0	655	4.52	Ave MPa= 4.22	
	3 F	1.169	1.179	2.352	2.390	0.533	HTC	34.0	-34.0	659	4.54	STD = 0.46	0.53
	5 F	1.163	1.157	2.389	2.369	0.503	HTC	32.4	-32.4	635	4.38		
	7 F	1.157	1.162	2.373	2.388	0.458	HTC	23.2	-23.2	498	3.43	Ave Width 0.501	
	F						HTC					STD = 0.027	0.031
							RTT		0.0	ERR	ERR	Ave MPa= ERR	
							RTT		0	ERR	ERR	STD = ERR	
							RTT		0	ERR	ERR		
							RTT		0	ERR	ERR	Ave Width ERR	
							RTT		0	ERR	ERR	STD = ERR	

05/01/93 EXPOSURE TESTS: SCHUMACHER F40 DIA-SCHUMALITH  
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	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample STD
SCHUMACHER TET 500hr/870°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 870°C	2 G	1.206	1.202	2.395	2.395	0.475	HTT	28.9	28.9	1292	8.91	Ave MPa= 8.43	
	4 G	1.205	1.198	2.393	2.383	0.486	HTT	25.9	25.9	1137	7.84	STD = 0.44	0.54
	6 G	1.186	1.181	2.379	2.394	0.451	HTT	26.9	26.9	1240	8.55		
	G						HTT					Ave Width 0.471	
	G						HTT					STD = 0.015	0.018
SCHUMACHER TET 500hr/870°C/15%H <sub>2</sub> O 20ppm NaCl COMPR @ 870°C	1 G	1.198	1.210	2.378	2.399	0.462	HTC	26.0	-26.0	607	4.19	Ave MPa= 5.01	
	3 G	1.197	1.190	2.395	2.380	0.464	HTC	30.0	-30.0	681	4.70	STD = 0.69	0.79
	5 G	1.195	1.192	2.398	2.391	0.494	HTC	41.7	-41.7	880	6.06		
	7 G	1.191	1.197	2.394	2.383	0.506	HTC	35.4	-35.4	737	5.08	Ave Width 0.482	
	G						HTC					STD = 0.019	0.022
SCHUMACHER TET 1000h/870°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 870°C	2 H	1.118	1.146	2.419	2.358	0.528	HTT	18.5	18.5	672	4.64	Ave MPa= 4.64	
	4 H	1.158	1.188	2.398	2.388	0.488	HTT	16.2	16.2	674	4.65	STD = 0.00	0.01
	6 H	1.172	1.179	2.354	2.406	0.475	HTT	BROKE					
	H						HTT					Ave Width 0.497	
	H						HTT					STD = 0.023	0.028
SCHUMACHER TET 1000h/870°C/15%H <sub>2</sub> O 20ppm NaCl COMPR @ 870°C	1 H	1.185	1.166	2.402	2.377	0.487	HTC	21.0	-21.0	434	2.99	Ave MPa= 3.37	
	3 H	1.181	1.161	2.396	2.351	0.482	HTC	24.2	-24.2	513	3.54	STD = 0.23	0.26
	5 H	1.147	1.170	2.367	2.362	0.488	HTC	23.8	-23.8	490	3.38		
	7 H	1.175	1.184	2.395	2.380	0.528	HTC	26.8	-26.8	517	3.57	Ave Width 0.496	
	H						HTC					STD = 0.018	0.021
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	

[illegible]

05/01/93 tetref EXPOSURE TESTS: REFRACTRON 70/3 w/442-T BINDER

	SPECIMEN		ID 1	ID 2	OD 1	OD 2	WIDTH	TEST	ABS	LOAD	-----FERBER EQUATION-----		Small
	NUMBER		in.	in.	in.	in.	in.		lb	lb	STRESS	STRESS	Sample
											psi	MPa	STD
Refract. 70/3 w/442-T	2	I	1.544	1.548	2.345	2.354	0.484	HTT	25.3	25.3	2323	16.02	Ave MPa= 15.96
100hr/870°C/15%H2O	4	I	1.547	1.545	2.348	2.352	0.473	HTT	24.4	24.4	2290	15.79	STD = 0.12
20ppm NaCl/No Pulse	6	I	1.539	1.553	2.355	2.362	0.517	HTT	27.6	27.6	2330	16.06	0.15
FRONT CANDLE		I						HTT					Ave Width 0.491
TENSION @ 870°C		I						HTT					STD = 0.019
													0.023
Refract. 70/3 w/442-T	1	I	1.558	1.544	2.358	2.370	0.504	HTC	37.3	-37.3	2121	14.63	Ave MPa= 16.64
100hr/870°C/15%H2O	3	I	1.543	1.540	2.345	2.357	0.491	HTC	46.0	-46.0	2692	18.56	STD = 1.53
20ppm NaCl/No Pulse	5	I	1.557	1.544	2.358	2.348	0.482	HTC	41.8	-41.8	2549	17.58	1.77
FRONT CANDLE	7	I	1.536	1.538	2.357	2.359	0.499	HTC	41.0	-41.0	2290	15.79	Ave Width 0.494
COMPR. @ 870°C		I						HTC					STD = 0.008
													0.010
Refract. 70/3 w/442-T	2	I	1.556	1.567	2.358	2.365	0.486	HTT	33.6	33.6	3114	21.47	Ave MPa= 16.61
100hr/870°C/15%H2O	4	I	1.549	1.555	2.356	2.370	0.486	HTT	24.5	24.5	2212	15.25	STD = 3.55
20ppm NaCl/No Pulse	6	I	1.552	1.555	2.362	2.351	0.552	HTT	23.5	23.5	1900	13.10	4.35
REAR CANDLE		I						HTT					Ave Width 0.508
TENSION @ 870°C		I						HTT					STD = 0.031
													0.038
Refract. 70/3 w/442-T	1	I	1.554	1.570	2.367	2.371	0.494	HTC	40.1	-40.1	2377	16.39	Ave MPa= 16.36
100hr/870°C/15%H2O	3	I	1.550	1.511	2.347	2.364	0.490	HTC	44.9	-44.9	2519	17.37	STD = 0.62
20ppm NaCl/No Pulse	5	I	1.551	1.558	2.356	2.385	0.485	HTC	39.1	-39.1	2299	15.85	0.72
REAR CANDLE	7	I	1.548	1.543	2.370	2.351	0.489	HTC	39.5	-39.5	2296	15.83	Ave Width 0.490
COMPR. @ 870°C		I						HTC					STD = 0.003
													0.004
								RTC		0	ERR	ERR	Ave MPa= ERR
								RTC		0	ERR	ERR	STD = ERR
								RTC		0	ERR	ERR	
								RTC		0	ERR	ERR	Ave Width ERR
								RTC		0	ERR	ERR	STD = ERR



05/01/93 tetref EXPOSURE TESTS: REFRACTRON 70/3 w/442-T BINDER

	SPECIMEN NUMBER	ID 1 In.	ID 2 In.	OD 1 In.	OD 2 In.	WIDTH In.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION-----		Small Sample STD
										STRESS psi	STRESS MPa	
Refract. 70/3 w/442-T 100hr/870°C/15%H2O 20ppm NaCl  TENSION @ 870°C	2 J	1.544	1.508	2.335	2.345	0.466	HTT	22.9	22.9	2121	14.62	Ave MPa= 13.85
	4 J	1.541	1.524	2.347	2.339	0.484	HTT	20.8	20.8	1873	12.91	STD = 0.71
	6 J	1.524	1.551	2.339	2.332	0.500	HTT	22.7	22.7	2033	14.02	
	J						HTT					Ave Width 0.483
	J						HTT					STD = 0.014
Refract. 70/3 w/442-T 100hr/870°C/15%H2O 20ppm NaCl  COMPR. @ 870°C	1 J	1.521	1.540	2.349	2.324	0.499	HTC	45.8	-45.8	2642	18.21	Ave MPa= 16.60
	3 J	1.535	1.526	2.325	2.345	0.491	HTC	40.3	-40.3	2371	16.35	STD = 0.98
	5 J	1.535	1.520	2.343	2.321	0.501	HTC	41.0	-41.0	2359	16.27	
	7 J	1.528	1.535	2.333	2.329	0.570	HTC	44.0	-44.0	2259	15.57	Ave Width 0.515
	J						HTC					STD = 0.032
Refract. 70/3 w/442-T 400hr/870°C/15%H2O 20ppm NaCl  TENSION @ 870°C	2 K	1.518	1.522	2.365	2.362	0.467	HTT	21.8	21.8	1897	13.08	Ave MPa= 11.97
	4 K	1.515	1.515	2.351	2.386	0.480	HTT	22.3	22.3	1849	12.75	STD = 1.35
	6 K	1.523	1.515	2.363	2.365	0.488	HTT	17.6	17.6	1461	10.07	
	K						HTT					Ave Width 0.478
	K						HTT					STD = 0.009
Refract. 70/3 w/442-T 400hr/870°C/15%H2O 20ppm NaCl  COMPR. @ 870°C	1 K	1.514	1.501	2.362	2.355	0.490	HTC	31.1	-31.1	1617	11.15	Ave MPa= 12.14
	3 K	1.514	1.526	2.356	2.386	0.488	HTC	33.8	-33.8	1779	12.27	STD = 0.59
	5 K	1.527	1.518	2.375	2.374	0.516	HTC	36.9	-36.9	1836	12.66	
	7 K	1.501	1.520	2.358	2.373	0.488	HTC	34.9	-34.9	1809	12.47	Ave Width 0.496
	K						HTC					STD = 0.012
							RTC		0	ERR	ERR	Ave MPa= ERR
							RTC		0	ERR	ERR	STD = ERR
							RTC		0	ERR	ERR	
							RTC		0	ERR	ERR	Ave Width ERR
							RTC		0	ERR	ERR	STD = ERR

05/01/93 EXPOSURE TESTS: REFRACTRON 70/3 w/442-T BINDER

tetref

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample STD
Refract. 70/3 w/442-T 500hr/870°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 870°C	2 L	1.533	1.513	2.357	2.345	0.502	HTT	28.3	28.3	2364	16.30	Ave MPa= 15.36	
	4 L	1.533	1.519	2.367	2.370	0.514	HTT	29.3	29.3	2327	16.04	STD = 1.15	1.40
	6 L	1.535	1.518	2.356	2.388	0.515	HTT	25.3	25.3	1994	13.75		
	L						HTT					Ave Width 0.510	
	L						HTT					STD = 0.006	0.007
Refract. 70/3 w/442-T 500hr/870°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 870°C	1 L	1.520	1.469	2.352	2.366	0.524	HTC	21.5	-21.5	1005	6.93	Ave MPa= 9.10	
	3 L	1.530	1.531	2.358	2.366	0.502	HTC	26.1	-26.1	1407	9.70	STD = 1.25	1.45
	5 L	1.530	1.520	2.384	2.392	0.501	HTC	28.6	-28.6	1431	9.87		
	7 L	1.517	1.531	2.387	2.383	0.479	HTC	27.3	-27.3	1434	9.89	Ave Width 0.502	
	L						HTC					STD = 0.016	0.018
Refract. 70/3 w/442-T 1000h/870°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 870°C	2 M	1.516	1.526	2.359	2.339	0.496	HTT	34.6	34.6	2923	20.15	Ave MPa= 18.96	
	4 M	1.546	1.508	2.353	2.356	0.542	HTT	36.4	36.4	2823	19.47	STD = 1.24	1.51
	6 M	1.525	1.519	2.328	2.333	0.510	HTT	29.3	29.3	2503	17.26		
	M						HTT					Ave Width 0.516	
	M						HTT					STD = 0.019	0.024
Refract. 70/3 w/442-T 1000h/870°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 870°C	1 M	1.502	1.539	2.341	2.355	0.493	HTC	30.8	-30.8	1696	11.70	Ave MPa= 12.20	
	3 M	1.538	1.541	2.360	2.346	0.484	HTC	38.0	-38.0	2231	15.39	STD = 1.87	2.16
	5 M	1.543	1.526	2.346	2.332	0.493	HTC	27.0	-27.0	1586	10.94		
	7 M	1.519	1.509	2.322	2.354	0.484	HTC	27.8	-27.8	1566	10.80	Ave Width 0.489	
	M						HTC					STD = 0.004	0.005
							RTT		0.0	ERR	ERR	Ave MPa= ERR	
							RTT		0	ERR	ERR	STD = ERR	
							RTT		0	ERR	ERR		
							RTT		0	ERR	ERR	Ave Width ERR	
							RTT		0	ERR	ERR	STD = ERR	

05/01/93 EXPOSURE TESTS: REFRACTRON 70/3 w/442-T BINDER

tetref

	SPECIMEN NUMBER	ID 1 in.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample STD
Refract. 70/3 w/442-T 100hr/980°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 980°C	2 N	1.515	1.510	2.360	2.355	0.504	HTT	15.3	15.3	1227	8.46	Ave MPa= 8.52	
	4 N	1.512	1.529	2.356	2.370	0.502	HTT	14.0	14.0	1136	7.83	STD = 0.55	0.64
	6 N	1.502	1.527	2.364	2.373	0.491	HTT	15.1	15.1	1223	8.43		
	8 N	1.515	1.506	2.352	2.377	0.508	HTT	17.4	17.4	1360	9.37	Ave Width 0.501	
	N						HTT					STD = 0.006	0.007
Refract. 70/3 w/442-T 100hr/980°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 980°C	1 N	1.525	1.514	2.358	2.344	0.511	HTC	14.0	-14.0	736	5.08	Ave MPa= 5.76	
	3 N	1.512	1.516	2.356	2.367	0.452	HTC	14.7	-14.7	839	5.79	STD = 0.55	0.67
	5 N	1.523	1.503	2.371	2.354	0.493	HTC	17.9	-17.9	932	6.42		
	7 N	1.511	1.525	2.364	2.355	0.509	HTC	BROKE				Ave Width 0.491	
	N						HTC					STD = 0.024	0.027
Refract. 70/3 w/442-T 400hr/980°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 980°C	2 O	1.484	1.488	2.411	2.405	0.493	HTT	5.1	5.1	360	2.48	Ave MPa= 2.28	
	4 O	1.514	1.519	2.392	2.410	0.504	HTT	5.6	5.6	418	2.88	STD = 0.58	0.67
	6 O	1.515	1.490	2.387	2.418	0.491	HTT	4.8	4.8	356	2.45		
	8 O	1.548	1.526	2.415	2.387	0.516	HTT	2.5	2.5	191	1.32	Ave Width 0.501	
	O						HTT					STD = 0.010	0.012
Refract. 70/3 w/442-T 400hr/980°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 980°C	1 O	1.510	1.508	2.416	2.434	0.494	HTC	4.9	-4.9	219	1.51	Ave MPa= 1.72	
	3 O	1.506	1.508	2.365	2.405	0.426	HTC	4.9	-4.9	276	1.90	STD = 0.31	0.34
	5 O	1.509	1.513	2.409	2.399	0.436	HTC	5.7	-5.7	304	2.09		
	7 O	1.514	1.519	2.414	2.416	0.515	HTC	4.0	-4.0	179	1.23	Ave Width 0.469	
	9 O	1.501	1.480	2.397	2.411	0.473	HTC	5.8	-5.8	269	1.85	STD = 0.034	0.038
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	

05/01/93 EXPOSURE TESTS: REFRACTRON 70/3 w/442-T BINDER

tetref

	SPECIMEN NUMBER	ID 1 In.	ID 2 in.	OD 1 in.	OD 2 in.	WIDTH in.	TEST	ABS LOAD lb	LOAD lb	-----FERBER EQUATION----- STRESS psi	STRESS MPa		Small Sample STD
Refract. 70/3 w/442-T 500hr/980°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 980°C	2 P	1.517	1.481	2.368	2.389	0.477	HTT	3.3	3.3	261	1.80	Ave MPa= 1.68	
	4 P	1.504	1.498	2.386	2.372	0.482	HTT	2.7	2.7	212	1.46	STD = 0.15	0.17
	6 P	1.506	1.506	2.381	2.371	0.491	HTT	3.4	3.4	266	1.84		
	8 P	1.494	1.422	2.375	2.387	0.492	HTT	3.4	3.4	237	1.64	Ave Width 0.486	
	P						HTT					STD = 0.006	0.007
Refract. 70/3 w/442-T 500hr/980°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 980°C	1 P	1.497	1.507	2.365	2.378	0.482	HTC	3.7	-3.7	187	1.29	Ave MPa= 1.17	
	3 P	1.501	1.495	2.372	2.387	0.482	HTC	3.0	-3.0	147	1.01	STD = 0.18	0.20
	5 P	1.505	1.487	2.371	2.389	0.489	HTC	3.8	-3.8	182	1.26		
	7 P	1.506	1.512	2.366	2.386	0.456	HTC	3.7	-3.7	200	1.38	Ave Width 0.506	
	9 P	1.441	1.429	2.347	2.379	0.621	HTC	4.0	-4.0	132	0.91	STD = 0.059	0.066
Refract. 70/3 w/442-T 1000h/980°C/15%H <sub>2</sub> O 20ppm NaCl TENSION @ 980°C	2 Q	1.523	1.476	2.541	2.477	0.459	HTT	6.5	6.5	430	2.96	Ave MPa= 2.32	
	4 Q	1.540	1.513	2.516	2.469	0.506	HTT	6.2	6.2	402	2.77	STD = 0.55	0.64
	6 Q	1.510	1.543	2.461	2.463	0.496	HTT	3.6	3.6	251	1.73		
	8 Q	1.556	1.503	2.480	2.441	0.513	HTT	3.9	3.9	265	1.83	Ave Width 0.494	
	Q						HTT					STD = 0.021	0.024
Refract. 70/3 w/442-T 1000h/980°C/15%H <sub>2</sub> O 20ppm NaCl COMPR. @ 980°C	1 Q	1.515	1.471	2.426	2.437	0.517	HTC	6.4	-6.4	258	1.78	Ave MPa= 2.31	
	3 Q	1.540	1.549	2.514	2.426	0.520	HTC	7.0	-7.0	298	2.05	STD = 0.84	0.93
	5 Q	1.516	1.531	2.470	2.480	0.554	HTC	6.4	-6.4	239	1.65		
	7 Q	1.495	1.517	2.434	2.459	0.531	HTC	7.8	-7.8	307	2.12	Ave Width 0.526	
	9 Q	1.512	1.548	2.405	2.418	0.506	HTC	12.0	-12.0	572	3.94	STD = 0.016	0.018
							RTC		0	ERR	ERR	Ave MPa= ERR	
							RTC		0	ERR	ERR	STD = ERR	
							RTC		0	ERR	ERR		
							RTC		0	ERR	ERR	Ave Width ERR	
							RTC		0	ERR	ERR	STD = ERR	

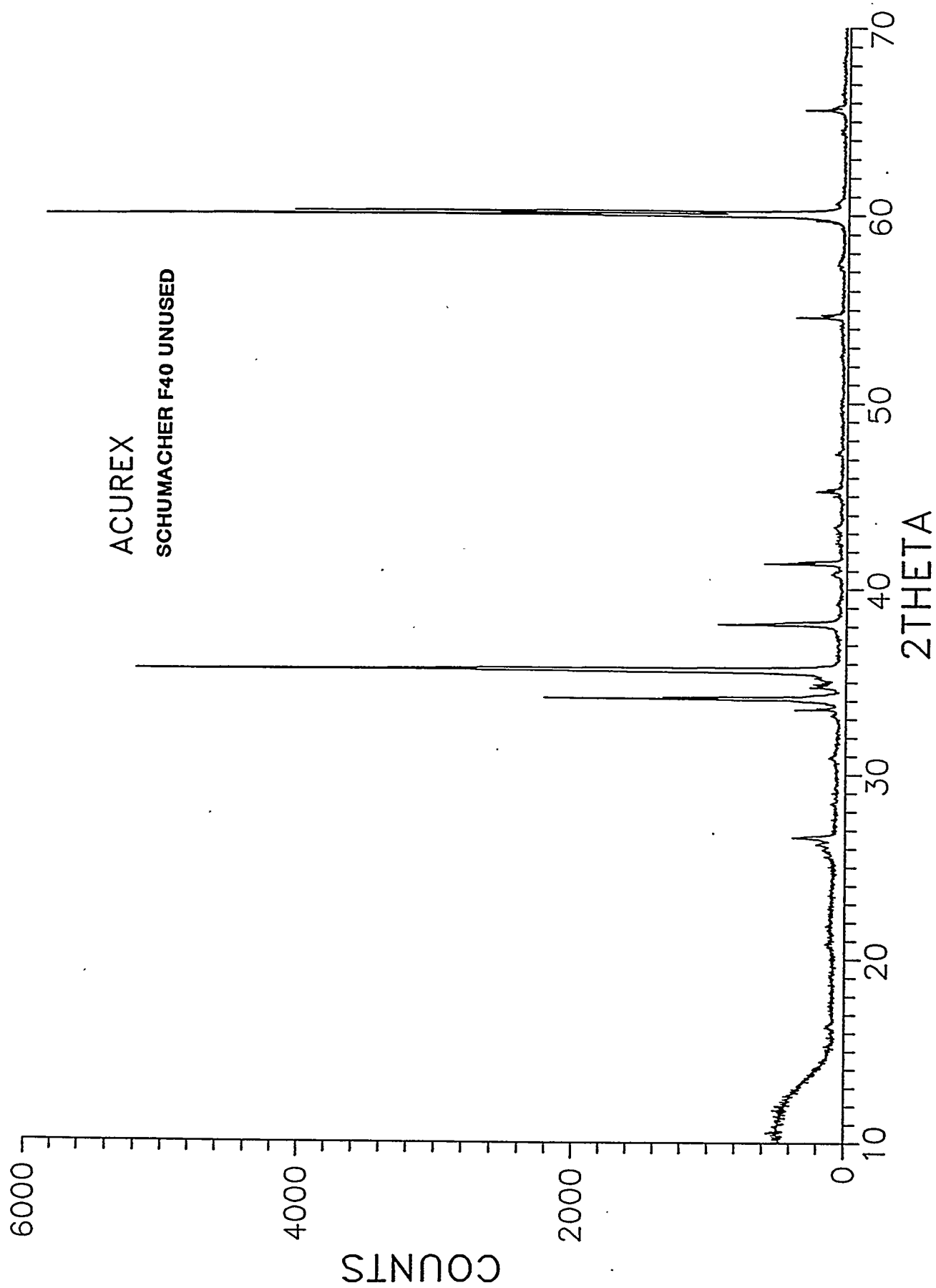
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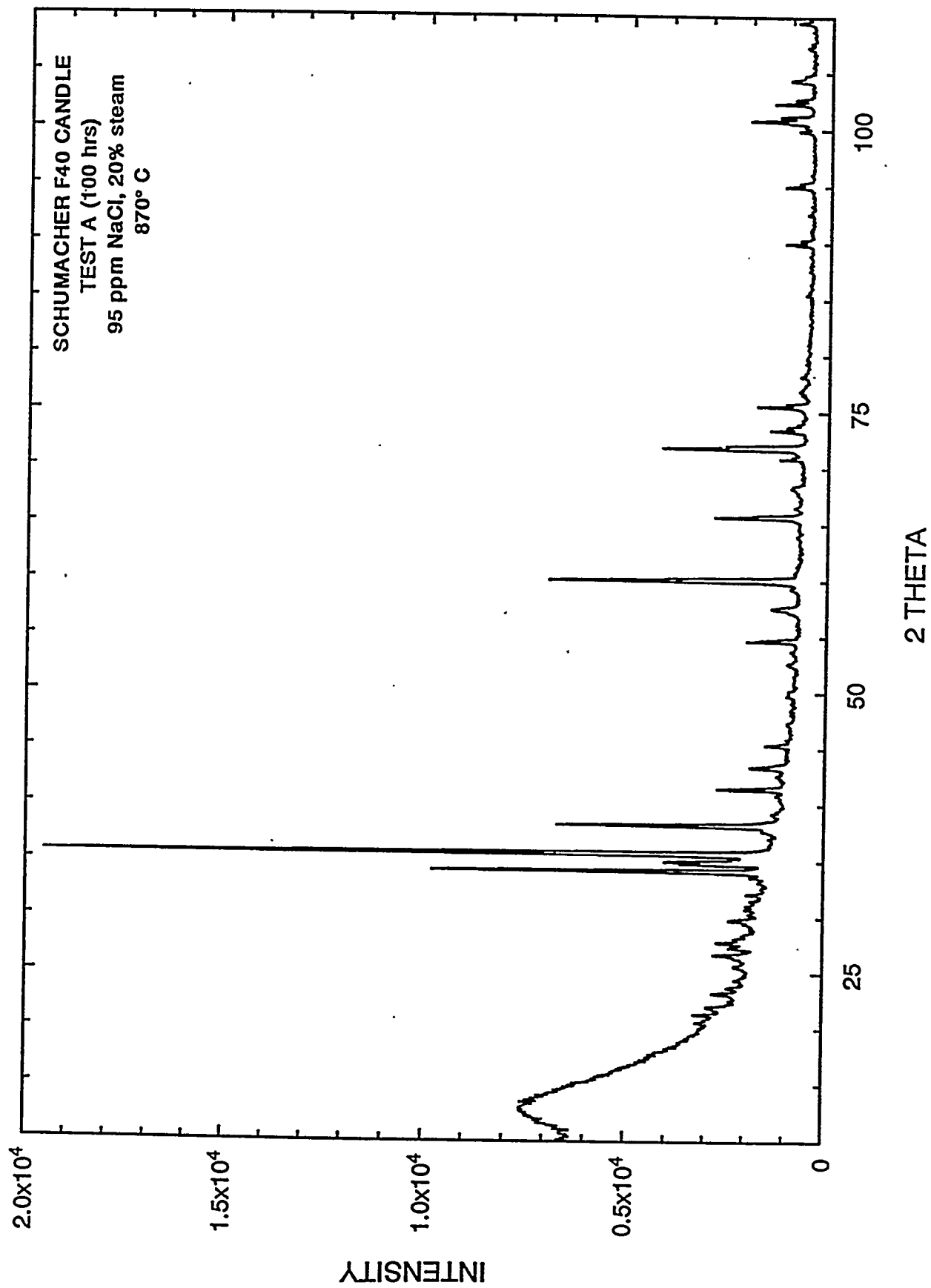
**APPENDIX E**  
**X-RAY DIFFRACTION DATA**

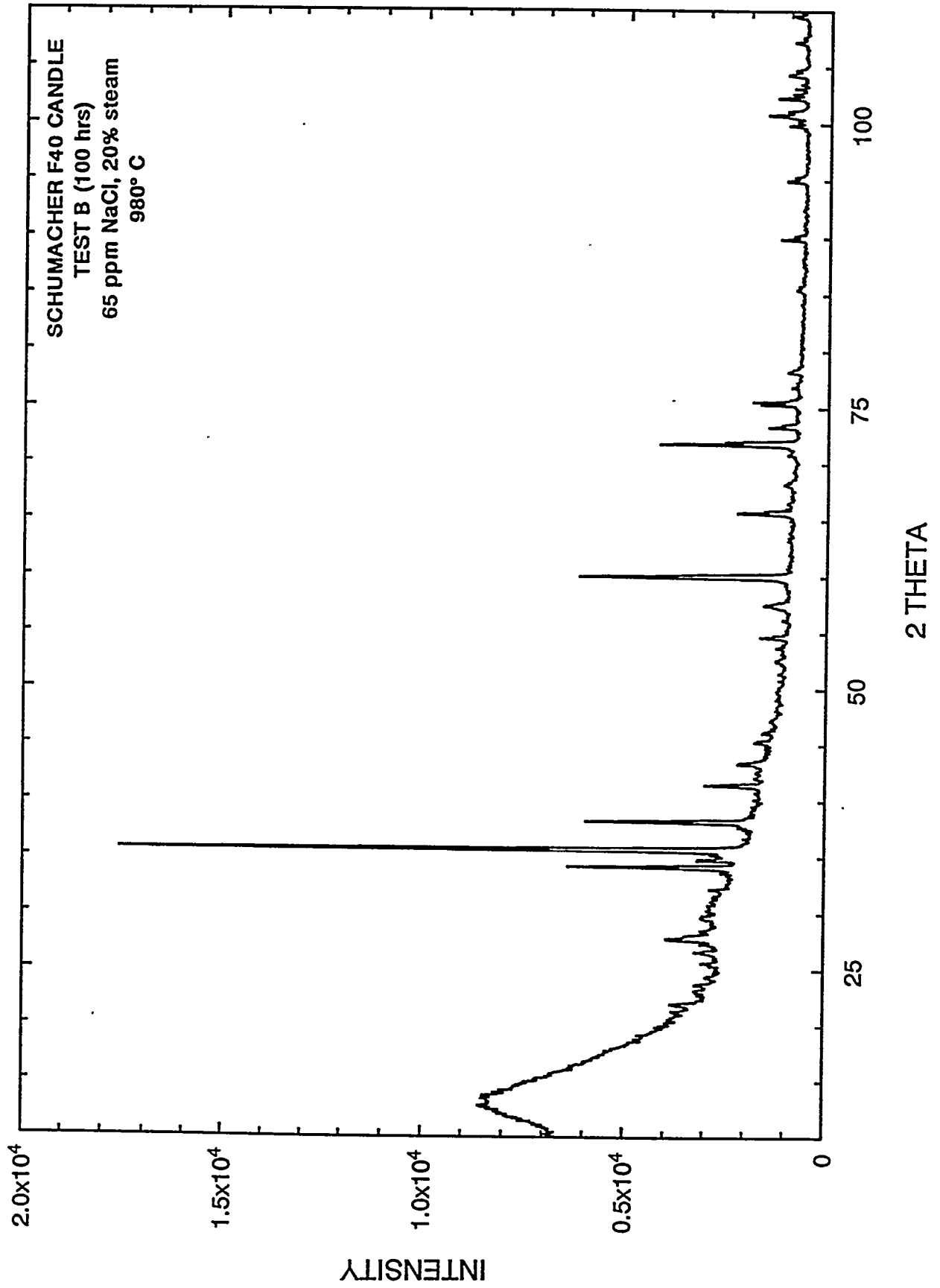
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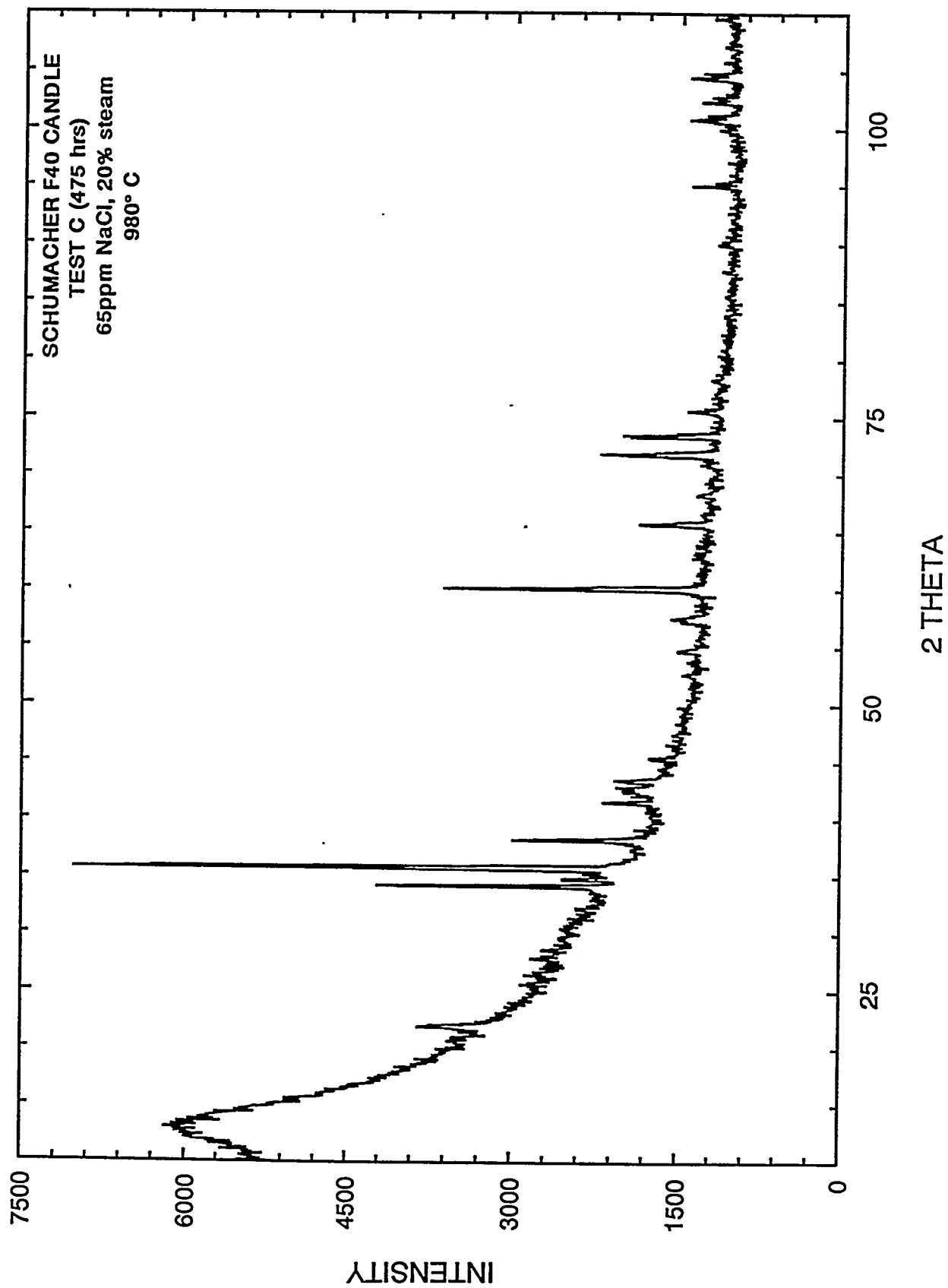
TEST IDENTIFICATION		SILICON CARBIDE: SiC												SILICA: SiO2				NaAlSi3O8		Al2O3
		1-1119 SILICON CARBIDE	4-0756 SILICON CARBIDE	22-1273 SILICON CARBIDE	22-1316 SILICON CARBIDE	29-1127 SILICON CARBIDE	29-1129 SILICON CARBIDE	19-1138 MOISSANITE	29-1128 MOISSANITE	29-1130 MOISSANITE	29-1131 MOISSANITE	18-1169 SILICON DIOXIDE	27-0605 SILICON DIOXIDE	34-1382 SILICON DIOXIDE	11-0665 CRISTOBALITE	10-0393 ALBITE	20-0572 ALBITE	10-0173 CORUNDUM		
SCHUMACHER UNUSED																				
F40 CANDLE																				
TEST A (100 hrs) 95 ppm NaCl 20% steam 870° C		1-1119													11-0665 11-0665 11-0665			10-0173 10-0173 10-0173		
TEST B (100 hrs) 65 ppm NaCl 20% steam 980° C															11-0665 11-0665 11-0665			10-0173 10-0173 10-0173		
TEST C (475 hrs) 65ppm NaCl 20% steam 980° C			4-0756 4-0756 4-0756										27-0605 27-0605 27-0605		11-0665 11-0665 11-0665			10-0173 10-0173 10-0173		
TEST D (575hrs) 65ppm NaCl 20% steam 980° C															11-0665 11-0665 11-0665			10-0173 10-0173 10-0173		
TEST G 20ppm 15% steam 870°C 500 hrs						29-1127 29-1127 29-1127									11-0665			10-0173 10-0173 10-0173		
TEST H 20ppm 15% steam 870°C 1000 hrs																		10-0173 10-0173 10-0173		
REFRACTRON UNUSED																		10-0173 10-0173 10-0173		
70/3 WITH 442-T																		10-0173 10-0173 10-0173		
TEST L 20ppm 15% steam 870°C 500 hrs				22-1273 22-1273 22-1273		29-1127 29-1127 29-1127										10-0393 10-0393 10-0393	20-0572 20-0572 20-0572	10-0173 10-0173 10-0173		
TEST M 20ppm 15% steam 870°C 1000 hrs																10-0393 10-0393 10-0393	20-0572 20-0572 20-0572	10-0173 10-0173 10-0173		
TEST Q 20ppm 15% steam 980°C 1000 HRS						29-1127 29-1127 29-1127										10-0393 10-0393 10-0393	20-0572 20-0572 20-0572	10-0173 10-0173 10-0173		

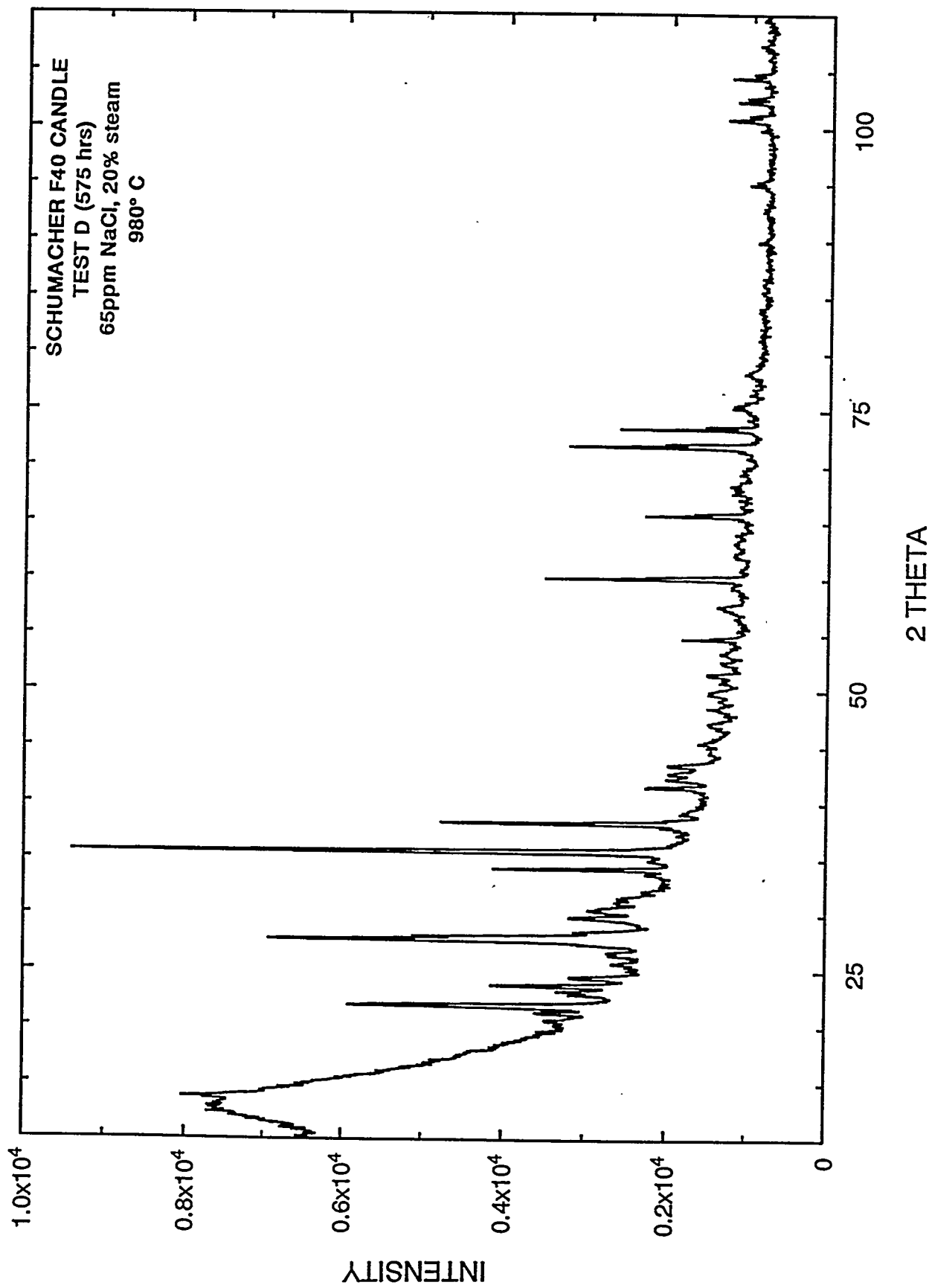


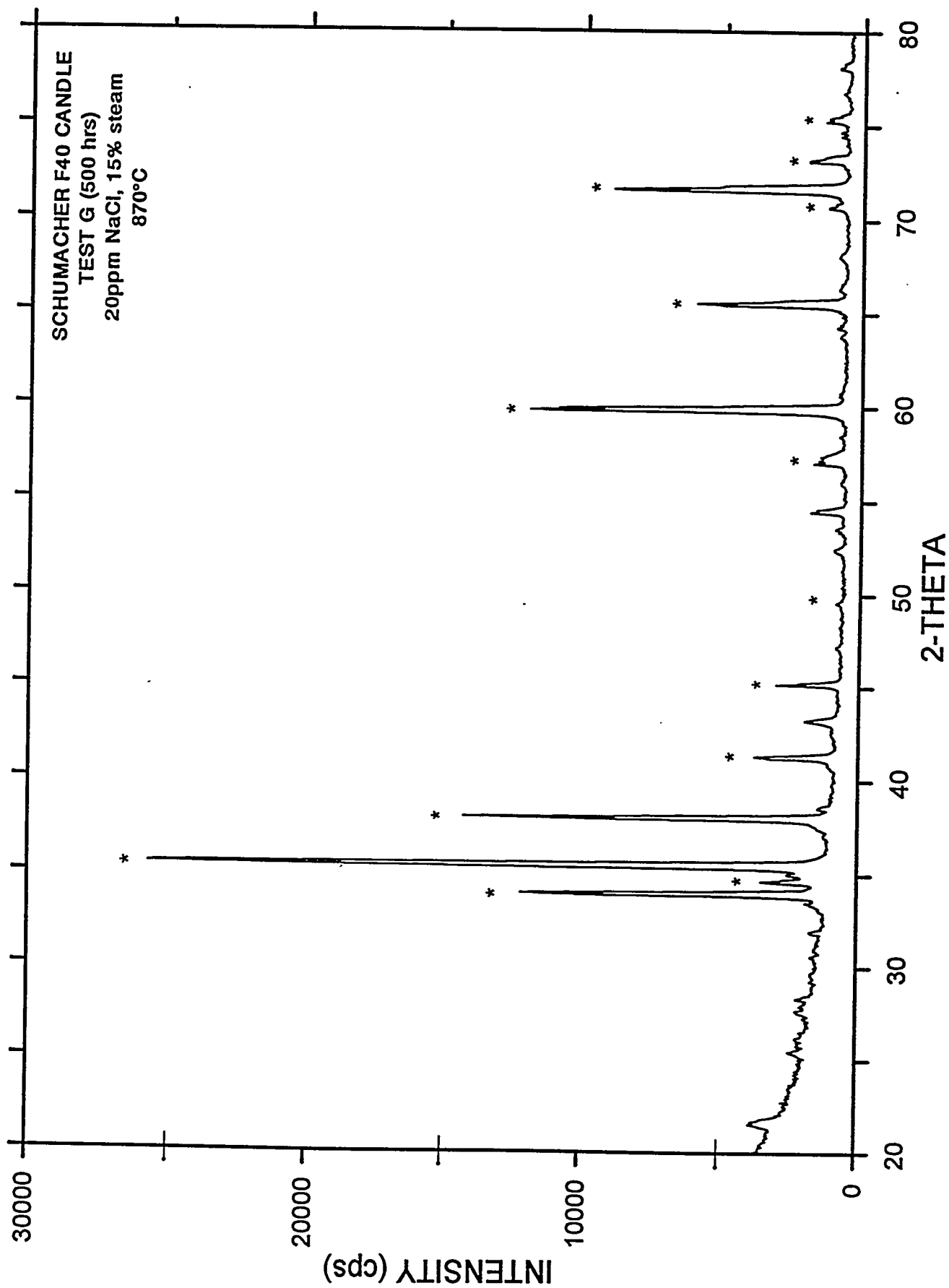


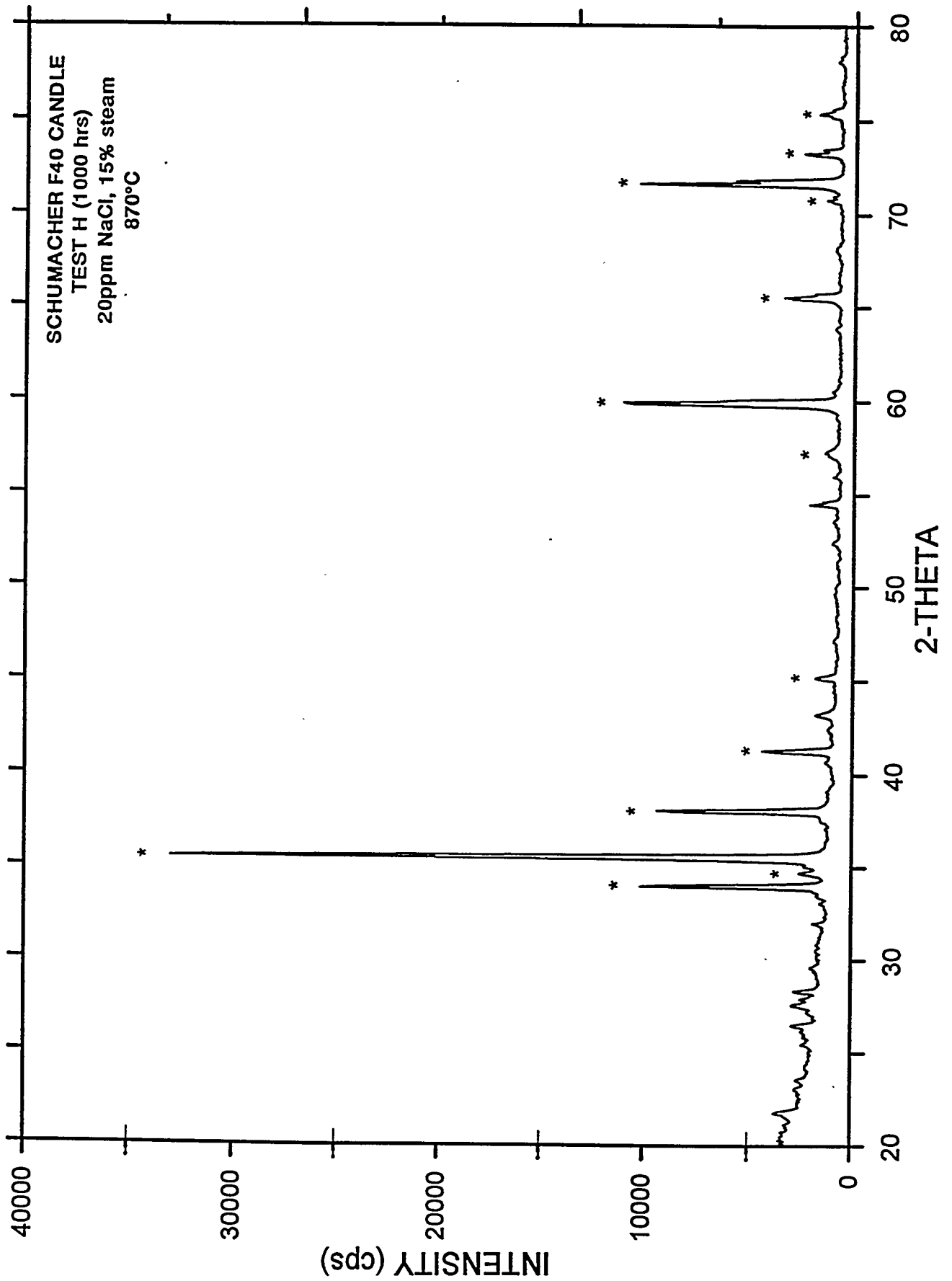


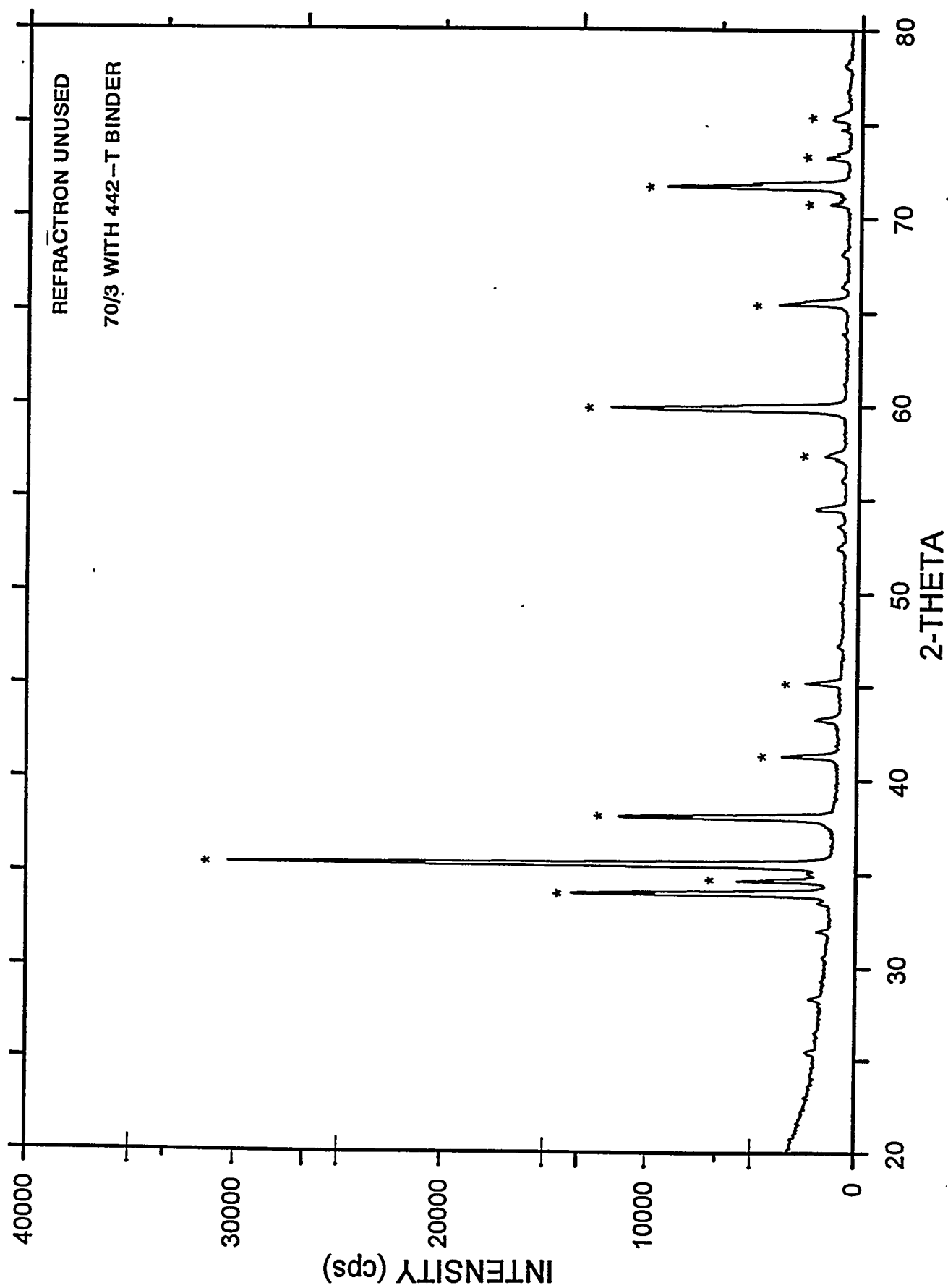




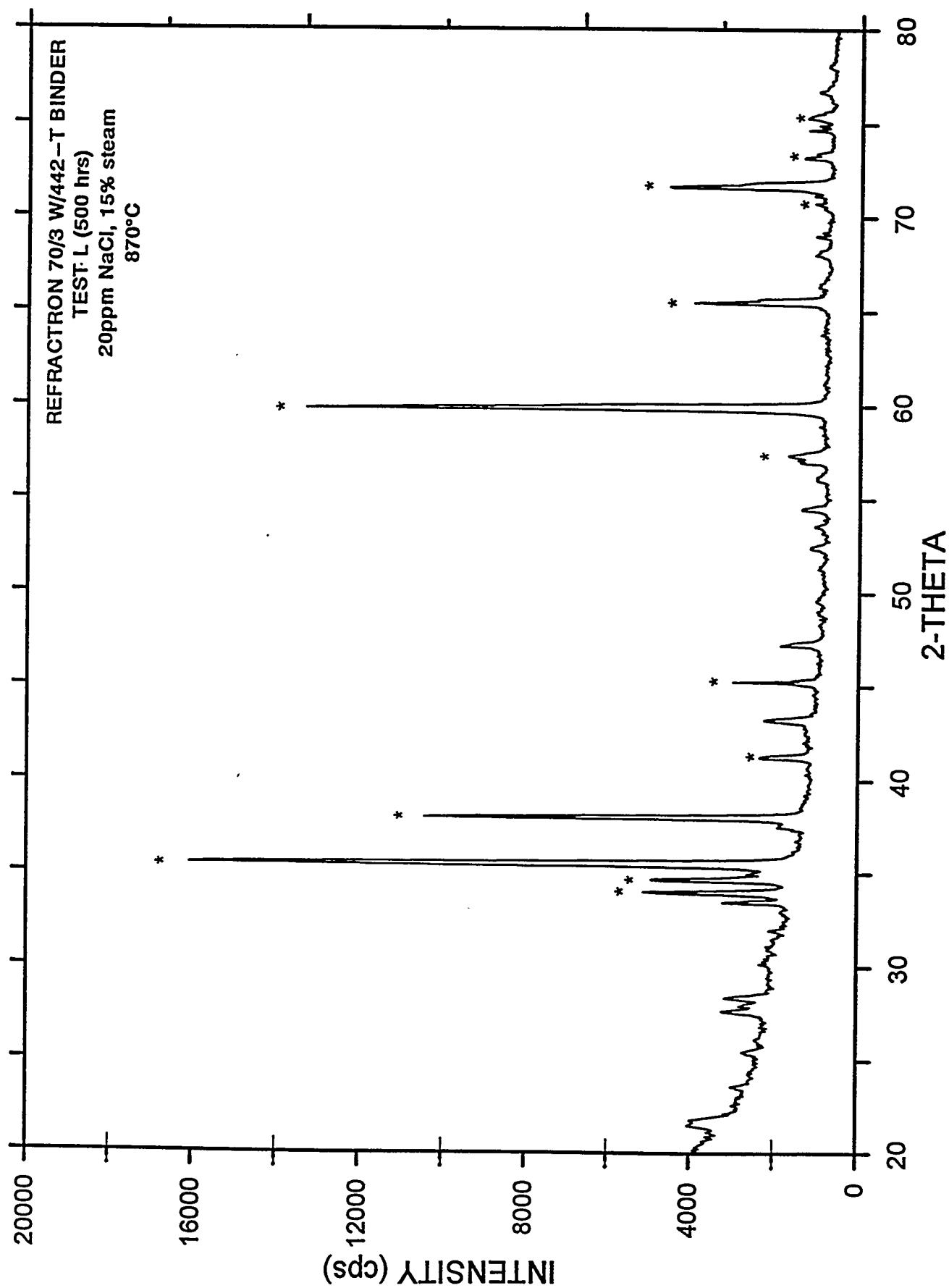


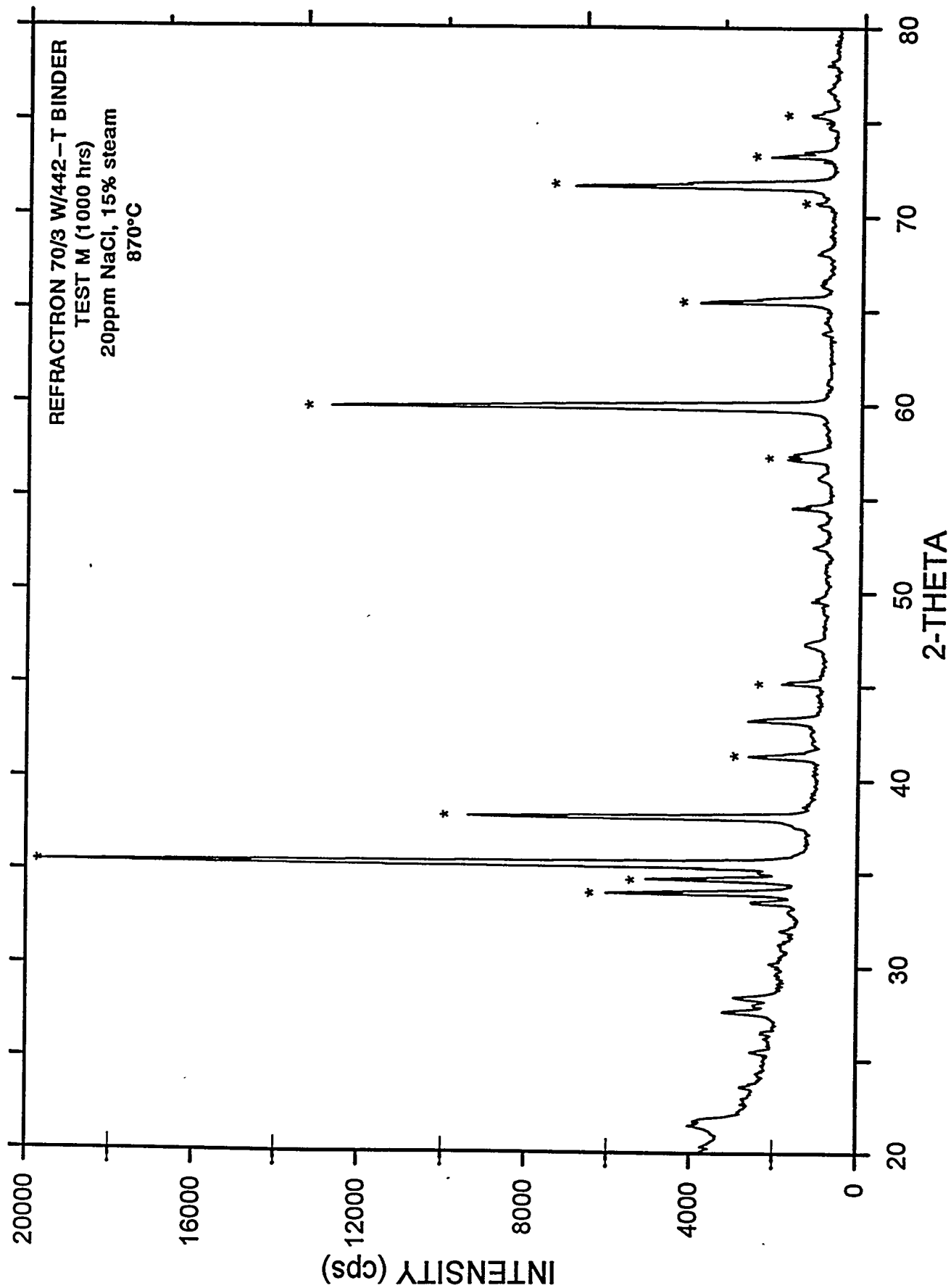


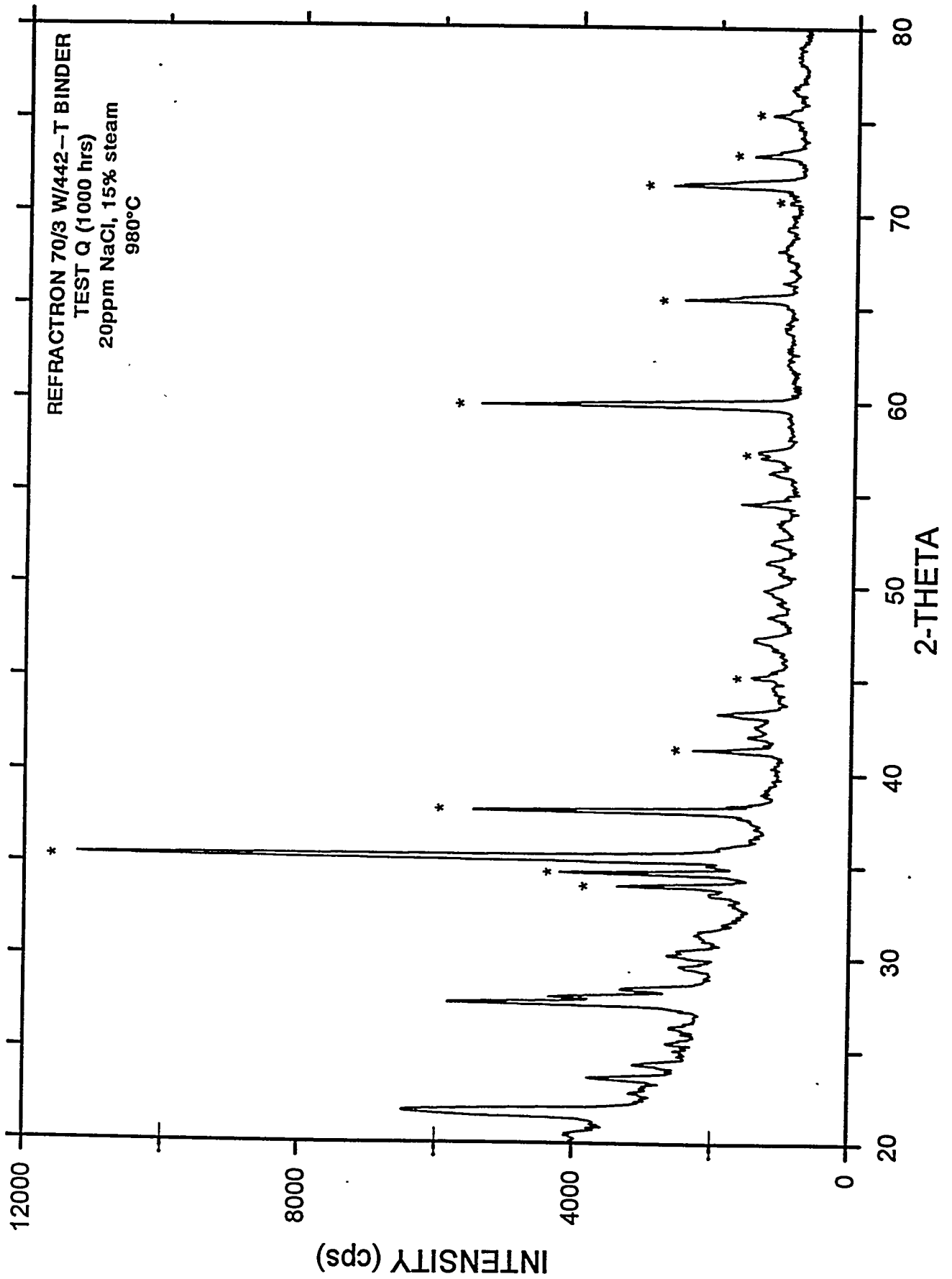












**APPENDIX F**

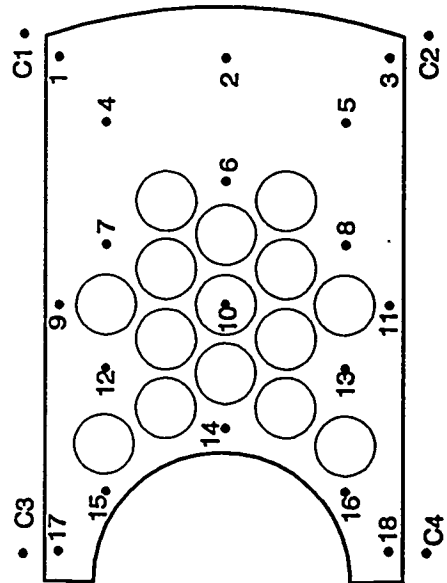
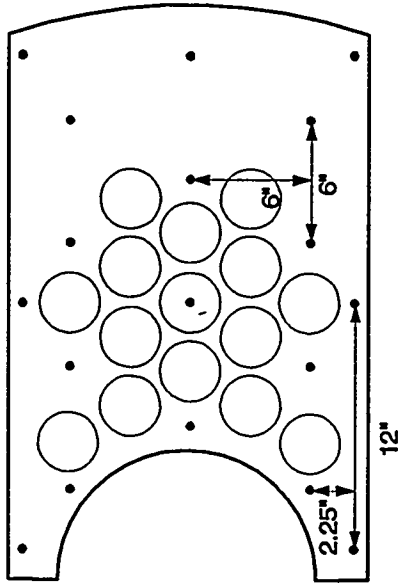
**REPORT: "ANALYSIS OF CANDLE FILTERS FROM THE KRW GASIFIER"**

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**APPENDIX G**  
**RAW DATA FROM ALUMINA ROD MEASUREMENTS**  
**FOR TUBESHEET DEFLECTION DETECTION**

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# ALUMINA ROD POSITIONS



Raw Alumina Rod Height Measurements (mm)

temp (F) or  
elapsed  
hours

Alumina rod positions

	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	C1	C2	C3	C4
room temp 04/08/94 16:09	44.1	44.1	45.2	43.2	44.4	43.3	42.7	44.1	41.9	43.5	44.1	42.6	44.3	42.8	42.7	43.6	42.4	43.3	29.1	31.1	26.5	28.8
room temp 04/09/94 08:23	44.3	44.6	45.6	43.4	44.8	43.7	42.8	44.6	42.3	43.5	44.4	42.6	44.3	43.1	42.8	43.9	41.9	43.4	29.1	31	26.4	28.9
room temp 04/10/94 12:09	44.3	44.5	45.6	43.4	44.6	43.6	42.5	44.4	42.2	43.8	44.4	42.5	44.3	43	42.8	43.9	42.2	43.7	29.2	30.9	26.4	28.7
room 04/11/94 08:35	44.3	44.5	45.6	43.6	44.6	43.2	42.8	44.3	42.1	43.2	44.3	42.5	43.9	43	42.5	44.5	42.9	44.3	29.1	30.9	26.3	28.5
1400 04/11/94 14:20	44.6	44.7	46.5	44.5	45.9	44.6	43	45	42.7	43.2	44.3	42.5	44.1	43.6	42.9	44.5	42.9	44.3	29.1	31.7	26.3	28.5
1500 04/11/94 15:50	44.1	45.8	45.7	43.8	45.2	43.5	42.3	44.2	41.6	41	43.9	41.6	43.5	42.6	42.2	43.5	42.5	43.5	28.8	31.1	25.8	28.2
1570 04/11/94 17:00	43.9	44	45.5	42.9	44.4	42.5	41.4	43.4	41.4	39.3	43.2	41.1	42.9	42.1	41.9	43.1	42	42.8	28.5	30.9	25.6	28.1
1619 04/12/94 09:12	42.8	42.9	44.3	41.7	43.2	41.2	40.1	41.9	40	38.5	43.2	40.9	42.6	41.8	41.7	42.8	42	42.9	28.6	30.6	25.3	27.9
24.42 04/12/94 17:26	42.8	42.5	44.1	41.7	43.2	41.1	40.2	41.7	39.5	33.7	41.5	39.6	41.5	40.2	40.2	41.4	40.5	41.4	27.8	30	24.1	27
25.66 04/12/94 18:40	42.9	42.5	44.3	42	43.5	41.1	40.2	41.7	40.2	33.7	41.5	39.6	41.5	40.2	40.2	41.4	40.9	40.2	28	29.9	24.5	27.1
39.55 04/13/94 08:33	42.3	42.6	44.1	42.2	43.3	41.2	40	41.6	40	33.3	41.3	39.6	41.3	39.8	40	41.3	40.8	41.8	27.5	29.8	24.4	26.9
48.22 04/13/94 17:14	42.1	41.9	43.8	41.9	42.5	40.9	40	41.2	39.9	33.2	41.1	39.6	41.1	39.6	40	41.1	40.5	41.5	27.5	29.8	24.2	26.9
63.48 04/14/94 08:29	42.3	42.2	43.7	42.1	43.2	40.5	39.6	41.6	40.1	33.4	41.5	39.6	41.2	40	40	41.1	40.7	41.5	27.6	29.9	24.4	26.9
71.32 04/14/94 16:20	42.2	42.2	43.6	41.9	42.9	40.3	39.5	41.3	39.8	33	41.2	39.5	41.1	40	39.9	41	40.5	41.5	27.6	29.9	24.2	26.9
87.16 04/15/94 08:10	42.2	42.3	43.7	41.9	43.2	40.6	39.5	41.4	39.7	32.9	41	39.3	41.2	40	39.9	41.2	40.4	41.4	27.7	29.9	24.3	26.9
94.29 04/15/94 15:18	42.2	42.2	43.8	42.1	43.1	40.7	39.7	41.4	39.8	32.9	41.3	39.2	41.3	40.1	39.9	41.2	40.4	41.4	27.7	29.9	24.3	26.7
114.45 04/16/94 11:27	42	42.2	43.6	41.7	43.2	40.7	39.5	41.4	39.9	32.9	41.3	39.2	41.3	40.1	39.9	41.2	40.4	41.6	27.6	29.9	24.1	26.6
138.35 04/17/94 11:22	42	42	43.6	42	43.1	40.5	39.7	41.3	39.3	32.6	41.3	39.6	41.1	39.8	40	41	40	41.6	27.7	29.9	24.4	26.7
144.28 04/17/94 17:17	41.9	41.9	43.6	42	43.1	40.7	39.7	41.4	39.8	32.6	41.3	39.6	41.1	39.8	40	41	40	41.6	27.7	29.9	24.3	26.7
160.30 04/18/94 09:19	41.9	41.9	43.5	41.2	42.8	40.4	39.6	41.3	39.3	32.6	41.3	39.6	41.1	39.8	39.9	41	40.8	41.4	27.5	29.9	24.3	26.7
166.37 04/18/94 15:23	41.9	42	43.4	42	42.9	40.5	39.5	41.1	39.6	32.7	40.9	39.3	41.1	39.8	39.8	41	40.8	41.4	27.5	29.9	24.3	26.7
183.65 04/19/94 08:39	41.9	41.8	43.3	41.8	42.8	40.6	39.5	41.4	39.9	32.6	41.7	39.1	41.4	39.7	40.3	41.2	40.5	41.5	27.5	29.8	24.3	26.9
186.08 04/19/94 11:05	41.9	41.8	43.3	41.6	42.8	40.6	39.6	41.4	39.9	32.6	41.7	39.1	41.4	39.7	40.3	41.2	40.5	41.5	27.5	29.8	24.3	26.9
207.59 04/20/94 08:36	42.1	41.9	43.6	41.5	43	40.7	39.8	41.4	40	32.6	42	39.3	41.3	39.7	40.4	41.3	40.4	41.6	27.7	29.6	24.1	26.9
231.25 04/21/94 08:16	42.2	41.8	43.7	41.5	43.1	40.8	39.9	41.6	40.3	32.4	41.9	39.3	41.2	39.6	40.3	41.2	40.7	41.8	27.7	29.6	24.2	26.9
255.39 04/22/94 08:24	42.3	41.9	43.7	41.4	43	40.7	40	41.4	40.2	32.2	41.9	39.4	41.2	39.4	40.3	41.1	40.4	41.7	27.8	29.8	24.2	27.1
279.18 04/23/94 08:12	42.3	41.9	43.7	41.4	43	40.6	39.7	41.5	40.3	31.9	42	39.4	41.2	39.6	40.2	40.9	40.4	41.7	27.9	29.6	24.2	27
308.45 04/24/94 13:28	42.3	41.6	43.9	41.4	42.9	40.7	39.8	41.5	40.2	32.3	42.1	39.4	41.3	39.5	40.2	41.1	40.5	41.7	27.7	29.7	24.2	27
327.45 04/25/94 08:28	42.4	41.7	43.8	41.4	43	40.7	39.6	41.4	40.3	32.3	41.9	39.3	41.2	39.4	40.1	40.9	40.5	41.8	27.8	29.7	24.3	27.1
351.55 04/26/94 08:34	42.4	41.7	43.8	41.4	43	40.7	39.6	41.4	40.3	32.3	41.9	39.3	41.2	39.4	40.2	41.1	40.5	41.8	27.8	29.7	24.3	27
375.26 04/27/94 08:16	42.3	41.7	43.8	41.5	42.9	40.5	39.7	41.5	39.6	32.2	41.9	39.4	41.2	39.4	40.2	41.1	40.5	41.6	27.8	29.8	24.2	27
399.46 04/28/94 08:28	42.3	41.6	43.6	41.4	42.9	40.5	39.5	41.3	39.6	32.3	41.9	39.3	41.2	39.5	40.1	41.1	40.8	41.7	27.8	29.7	24.3	27
425.06 04/29/94 10:04	42.3	41.5	43.5	41.4	42.9	40.5	39.4	41.4	39.6	32.1	41.9	39.1	41.2	39.4	40.1	40.9	40.4	41.6	27.8	29.7	24.3	27
454.54 04/30/94 15:33	42.2	41.6	43.5	41.3	42.8	40.5	39.3	41.3	39.4	32.1	41.9	39.3	41.1	39.3	40.1	41	40.5	41.6	27.9	29.9	24.2	27
477.08 05/01/94 14:05	42.2	41.6	43.6	41.4	43	40.3	39.4	41.3	39.6	32	41.9	39.1	41	39.2	40	40.8	40.3	41.6	27.9	29.8	24.3	27
494.61 05/02/94 07:37	42.2	41.7	43.7	41.4	42.9	40.4	39.4	41.2	39.4	31.9	41.8	39.1	41	39.1	40.1	40.9	40.4	41.6	27.8	29.8	24.2	27
501.33 05/02/94 14:20	42.2	41.6	43.7	41.3	42.9	40.4	39.4	41	39.6	31.9	41.9	39.1	40.9	39.1	39.9	40.8	40.4	41.6	27.8	29.9	24.1	27
503.99 05/02/94 17:01	41.9	41.5	42.8	41	42.9	39.5	38.9	40.5	39.2	31.6	41.3	39.1	40.4	38.6	39.5	39.7	39.6	41.6	27.3	29.1	23.4	26.4
504.98 05/02/94 17:59	40.9	40.6	42.8	40.6	41.8	39.2	38.6	40.5	39.3	30.8	40.9	39.1	40.5	38.2	39.2	39.1	39.1	40.2	27.2	28.8	23.2	25.9
519.61 05/03/94 08:37	40.1	38.8	41.3	38.5	41	38	37.4	39.2	37.1	30.1	40.8	37.6	39.5	36.8	37.4	38.4	37.8	38.4	26.4	28.1	21.9	25
522.89 05/03/94 11:54	40	38.4	41.3	38.6	40.5	37.5	36.8	38.8	36.9	28.4	39.8	37.3	38.7	36.5	36.9	37.5	36.6	39.3	26.1	27.5	21.4	24.9
528.18 05/03/94 17:12	39.9	38.2	41.2	38.1	40.5	37.3	36.4	37.2	36.4	28.3	39.5	37.3	38.4	35.7	36.7	37.6	36.3	38.3	25.9	27.3	21.4	24.6
544.23 05/04/94 09:14	40	38.2	41.2	39.2	40.5	37.2	36.6	38.5	36.9	29.3	39.7	37.2	39.1	36.3	36.7	38.1	37.1	39.2	26.1	27.4	21.5	24.7
568.61 05/05/94 09:37	40.2	37.9	41.2	39.2	40.5	36.8	36.7	37.6	36.5	28.4	39.8	37.2	38.3	36.3	36.7	38.3	37.3	39	25.9	27.3	21.5	24.5



Corrected alumina rod displacement measurements (mm)

measured at 870 C

Alumina Rod Positions

elapsed hours	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	C1	C2	C3	C4
0 04/11/94 17:00	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
1.93 04/11/94 18:56	0.225	0	0.125	0	0	0	0.025	0.025	0.025	0.625	0	0.025	0.125	0.125	0.025	0	0	0	-0.1	0.3	0.3	0
16.19 04/12/94 09:12	0	0	0.1	0.1	0.1	0.2	0.2	0.4	0.3	4.1	0.7	0.1	0.5	1.1	0.9	0.8	0.4	1.1	0.7	0.9	1.5	1.3
24.42 04/12/94 17:26	0.05	0.25	0.35	0.15	0.15	0.35	0.35	0.65	0.85	4.55	1.05	0.85	0.65	1.15	0.95	0.95	0.75	1.65	0.7	0.9	1.5	1.1
25.66 04/12/94 18:40	0.1	0.6	0.3	0	0	0.5	0.3	0.8	0.3	4.7	0.8	0.6	0.5	1	0.8	0.8	0.2	0.7	0.5	1	1.1	1.1
39.55 04/13/94 08:33	0.475	0.275	0.275	0	0	0.175	0.275	0.675	0.275	4.875	0.775	0.375	0.475	1.175	0.775	0.675	0.075	0.675	1	1.1	1.2	1.2
48.22 04/13/94 17:14	0.575	0.875	0.475	0	0.675	0.375	0.175	0.975	0.275	4.875	0.875	0.775	0.575	1.275	0.675	0.775	0.275	0.875	1	1.1	1.4	1.4
63.48 04/14/94 08:29	0.525	0.725	0.725	0	0.125	0.925	0.725	0.725	0.225	4.825	0.625	0.425	0.625	1.025	0.825	0.925	0.225	1.025	0.9	1	1.2	1.2
71.32 04/14/94 16:20	0.575	0.675	0.775	0	0.375	1.075	0.775	0.975	0.475	5.175	0.875	0.475	0.675	0.975	0.875	0.975	0.375	0.975	0.9	1	1.4	1.2
87.16 04/15/94 08:10	0.65	0.65	0.75	0	0.15	0.85	0.85	0.95	0.65	5.35	1.15	0.75	0.65	1.05	0.95	0.85	0.55	1.15	0.7	1	1.3	1.2
94.29 04/15/94 15:18	0.525	0.625	0.525	0	0.125	0.625	0.525	0.825	0.425	5.225	0.825	0.525	0.625	0.925	0.825	0.825	0.425	1.025	0.8	1.2	1.3	1.4
114.45 04/16/94 11:27	0.675	0.575	0.675	0	0	0.575	0.675	0.775	0.275	5.175	0.675	0.675	0.375	0.775	0.775	0.675	0.375	0.775	0.9	1	1.5	1.5
138.35 04/17/94 11:22	0.8	0.9	0.8	0	0.2	0.7	0.6	0.9	0.5	5.5	0.8	0.4	0.6	1.1	0.5	0.9	0	0.6	0.8	1	1.2	1.4
144.28 04/17/94 17:17	0.85	0.95	0.75	0	0.15	0.85	0.55	0.95	0.95	5.55	0.75	0.35	0.65	1.15	0.75	0.95	0.85	0.85	0.8	1.2	1.2	1.4
160.30 04/18/94 09:19	0.825	0.925	0.825	0.575	0.425	0.925	0.625	0.925	0.925	5.525	0.925	0.325	0.625	1.125	0.825	0.925	0.825	0.825	0.8	1.2	1.3	1.4
166.37 04/18/94 15:23	0.775	0.775	0.875	0	0.275	0.775	0.675	1.075	0.575	5.375	1.075	0.575	0.575	1.075	0.875	0.875	0	0.975	1	1.2	1.3	1.4
183.65 04/19/94 08:39	0.85	1.05	1.05	0	0.45	0.75	0.75	1.05	0.35	5.55	0.35	0.85	0.35	1.25	0.45	0.75	0.35	0.95	1	1.1	1.3	1.2
186.08 04/19/94 11:05	0.925	1.125	1.125	0.225	0.525	0.825	0.725	0.925	0.325	5.625	0.125	0.525	0.525	1.325	0.725	0.725	0.425	1.225	0.7	1.1	1.2	1.3
207.59 04/20/94 08:36	0.6	0.9	0.7	0.2	0.2	0.6	0.4	0.8	0.2	5.3	0	0.6	0.4	1.2	0.3	0.6	0.4	0.8	0.8	1.3	1.5	1.2
231.25 04/21/94 08:16	0.525	1.025	0.625	0.225	0.125	0.525	0.325	0.625	0	5.725	0.125	0.625	0.525	1.325	0.425	0.725	0.125	0.625	0.8	1.3	1.4	1.2
255.39 04/22/94 08:24	0.45	1.05	0.55	0.35	0.25	0.65	0.25	0.85	0.05	5.95	0.15	0.55	0.55	1.55	0.45	0.85	0.45	0.75	0.7	1.1	1.6	1.2
279.18 04/23/94 08:12	0.475	0.975	0.675	0.375	0.275	0.775	0.575	0.775	0	6.275	0.075	0.575	0.575	1.375	0.575	1.075	0.475	0.775	0.6	1.3	1.4	1.2
308.45 04/24/94 13:28	0.45	1.25	0.45	0.35	0.35	0.65	0.45	0.75	0.05	5.85	0	0.55	0.45	1.45	0.55	0.85	0.35	0.75	0.8	1.2	1.4	1.2
327.45 04/25/94 08:28	0.375	1.175	0.875	0.475	0.475	0.875	0.775	0.875	0.075	6.175	0.175	0.675	0.575	1.575	0.675	1.075	0.375	0.775	0.7	1.3	1.4	1.1
351.55 04/26/94 08:34	0.45	1.25	0.65	0.45	0.35	0.75	0.75	0.95	0.05	5.95	0.25	0.75	0.65	1.55	0.75	1.05	0.45	0.75	0.7	1.2	1.3	1.1
375.26 04/27/94 08:16	0.525	1.225	0.625	0.325	0.425	0.925	0.625	0.825	0.725	6.025	0.225	0.625	0.625	1.625	0.625	0.925	0.425	0.925	0.7	1.1	1.4	1.1
399.46 04/28/94 08:28	0.565	1.325	0.825	0.425	0.425	0.925	0.825	1.025	0.725	5.925	0.225	0.725	0.625	1.525	0.725	1.125	0.525	0.825	0.7	1.2	1.3	1.1
425.06 04/29/94 10:04	0.55	1.45	0.95	0.45	0.45	0.95	0.95	0.95	0.75	6.15	0.25	0.95	0.65	1.65	0.75	1.25	0.45	0.95	0.7	1.1	1.3	1.1
454.54 04/30/94 15:33	0.675	1.375	0.975	0.575	0.575	0.975	1.075	1.075	0.975	6.175	0.275	0.775	0.775	1.775	0.775	1.075	0.475	0.975	0.6	1	1.4	1.1
477.08 05/01/94 14:05	0.625	1.325	0.825	0.425	0.325	1.125	0.925	1.025	0.725	6.225	0.225	0.925	0.825	1.825	0.825	1.225	0.625	0.925	0.8	1.1	1.3	1.1
494.61 05/02/94 07:37	0.625	1.225	0.725	0.425	0.425	1.025	0.925	1.125	0.925	6.325	0.325	0.925	0.825	1.925	0.725	1.125	0.525	0.925	0.7	1.1	1.4	1.1
501.33 05/02/94 14:20	0.625	1.325	0.725	0.525	0.425	1.025	0.925	1.325	0.725	6.325	0.225	0.925	0.925	1.925	0.925	1.225	0.525	0.925	0.7	1	1.5	1.1