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CNLM-2283

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Date

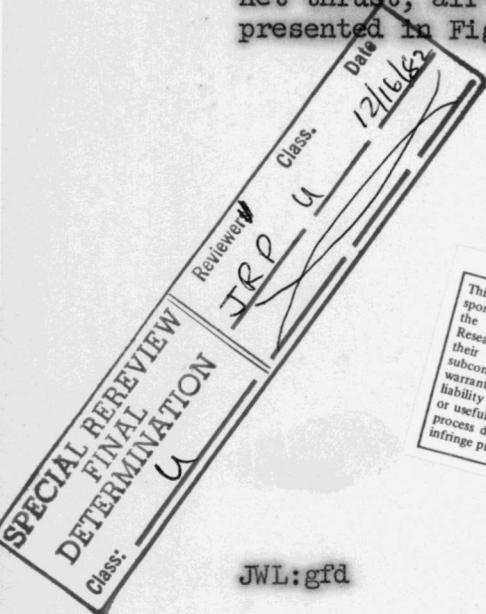
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TI84 027579

PRATT & WHITNEY AIRCRAFT NUCLEAR J-58 TURBOJET ENGINE

PERFORMANCE VARIATION WITH RADIATOR DIAMETER

The variation of engine performance with liquid metal radiator diameter and flight altitude has been estimated for both the 1600F NaK and 1800F NaK radiators at Mach 0.6 and hot day atmospheric conditions. The net thrust, air flow and reactor power is presented in Figs 1 to 3 for the Pratt & Whitney Aircraft J-58 engine with the 1600F NaK radiator. The net thrust, air flow and reactor power for the 1800F NaK radiator is presented in Figs 4 to 6.



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John W. Larson
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CLASSIFICATION CANCELLED

BY AUTHORITY OF DIVISION OF CLASSIFICATION
BY: **TED REDMON** DATE: **12-8-75**

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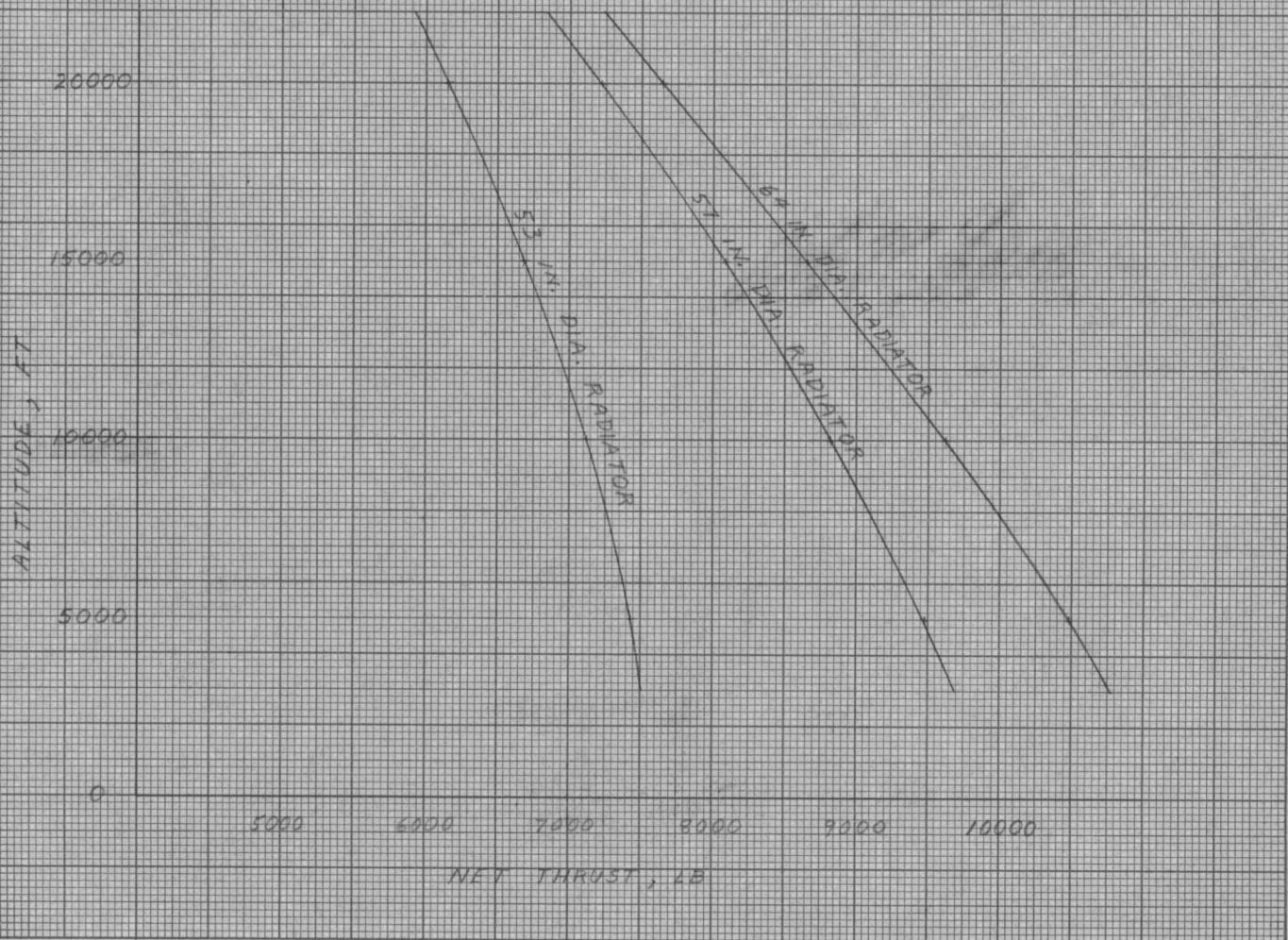
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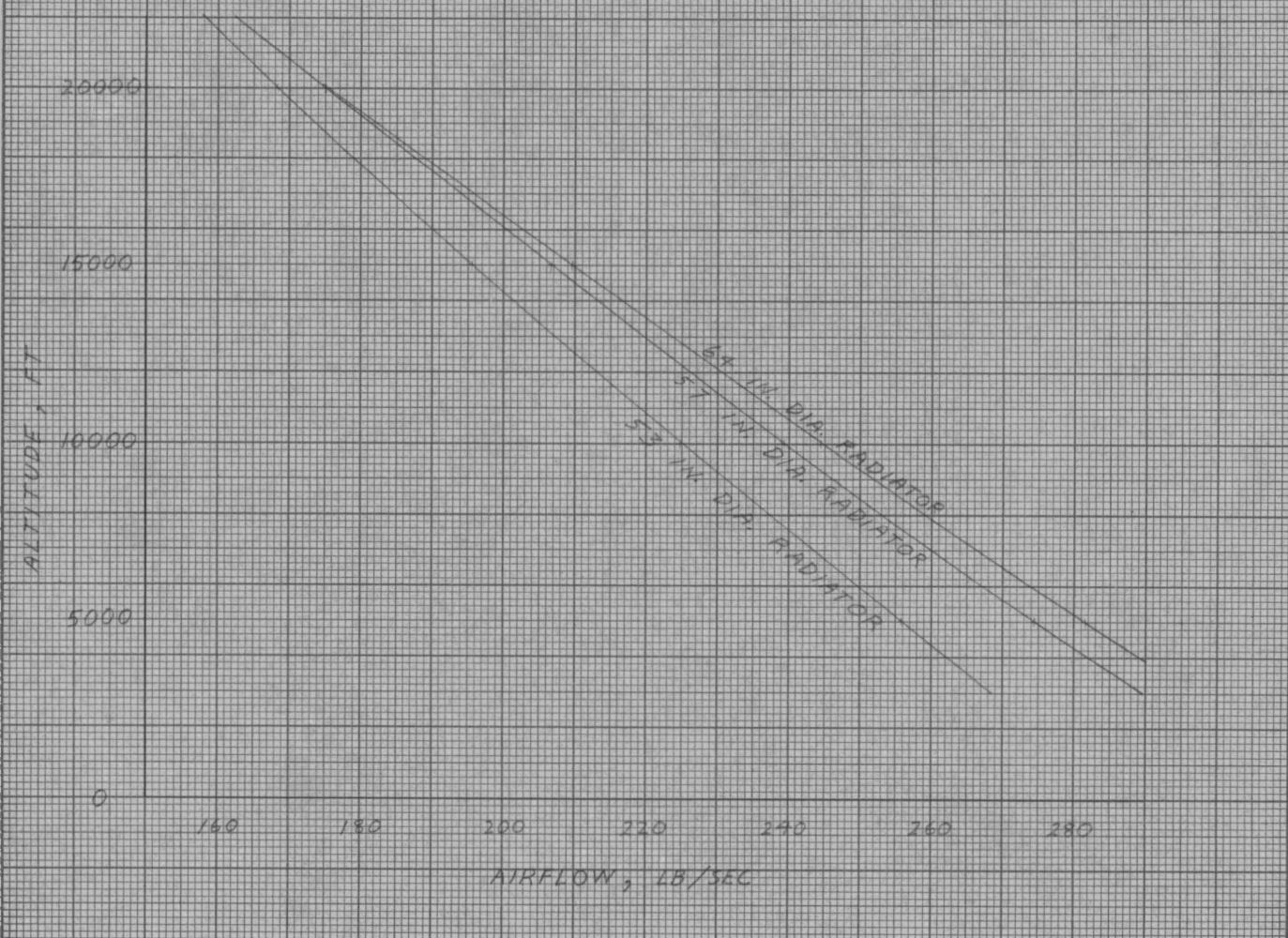
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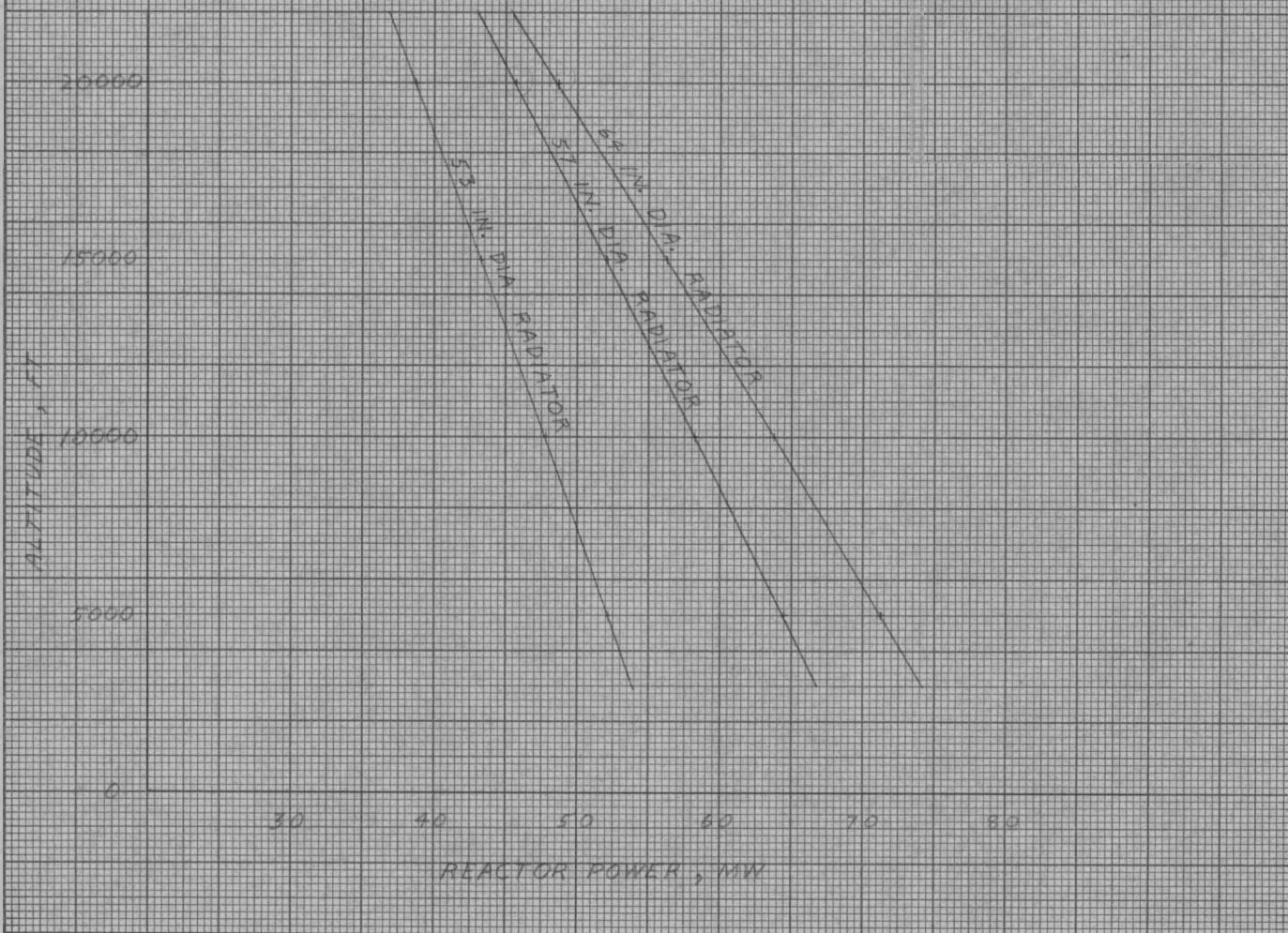
OFFICIAL USE ONLYPRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat OperationESTIMATED NET THRUST
MACH 0.6, HOT DAY ATMOSPHERE
1600F NaK RADIATOR

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PRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat Operation

ESTIMATED AIRFLOW
MACH 0.6, HOT DAY ATMOSPHERE
1600F N₂K RADIATOR

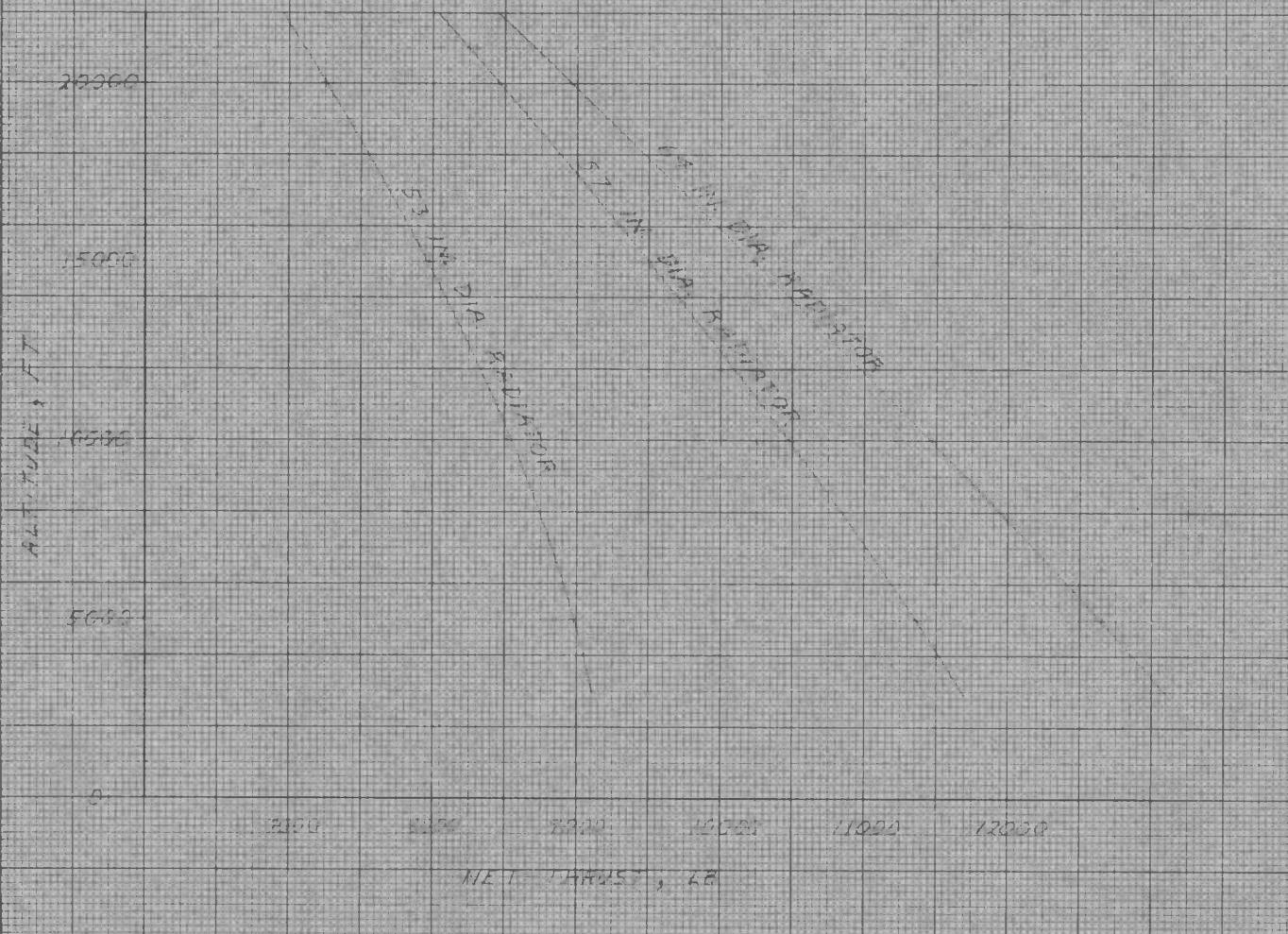


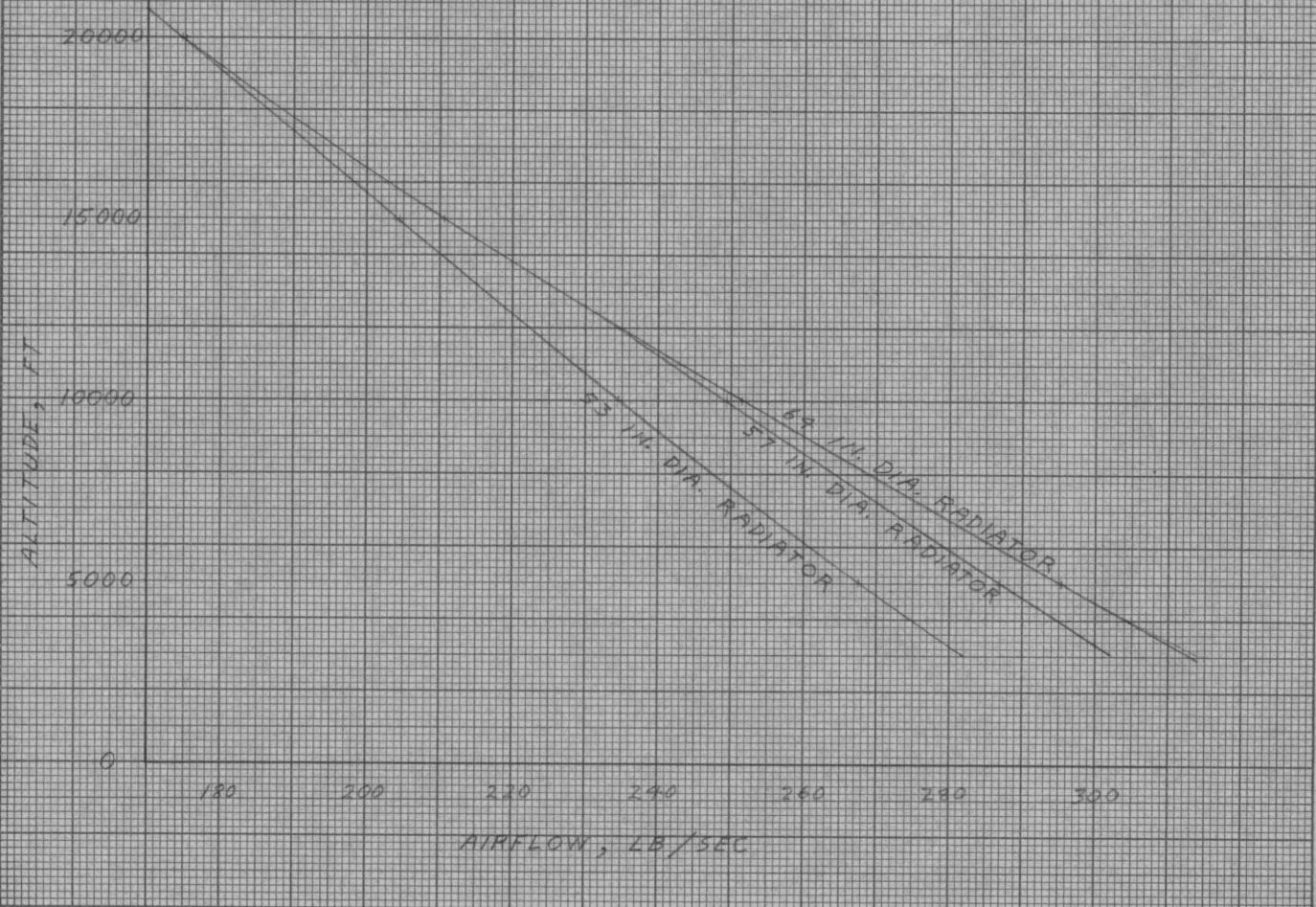
OFFICIAL USE ONLY**PRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat Operation****ESTIMATED REACTOR POWER
MACH 0.6, HOT DAY ATMOSPHERE
1600F NaK RADIATOR**

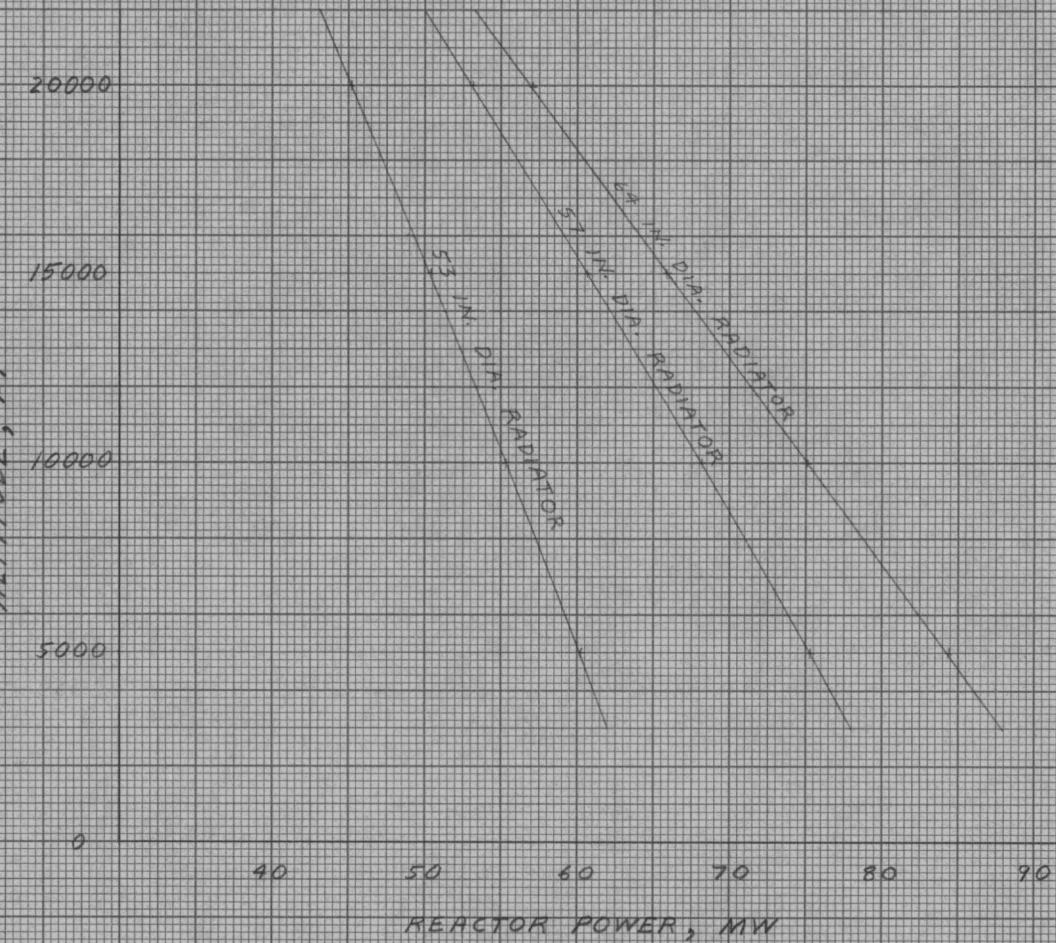
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PRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat Operation

ESTIMATED NET THRUST
MACH 0.6, HOT DAY ATMOSPHERE
1800F Nuk RADIATOR



PRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat OperationESTIMATED AIRFLOW
MACH 0.6, HOT DAY ATMOSPHERE
1800F NaK RADIATOR

OFFICIAL USE ONLYPRATT & WHITNEY AIRCRAFT NUCLEAR J58 TURBOJET ENGINE
Nuclear Heat OperationESTIMATED REACTOR POWER
MACH 0.6, HOT DAY ATMOSPHERE
1800F NaK RADIATOR**OFFICIAL USE ONLY**