

Dynamic Characterization of IM7 Composites in Support of M&S Efforts

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**IM Hazards M&S PA Meeting
December 3-4, 2014
Los Alamos, NM**



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Dynamic Behavior of Composite Materials

Goals:

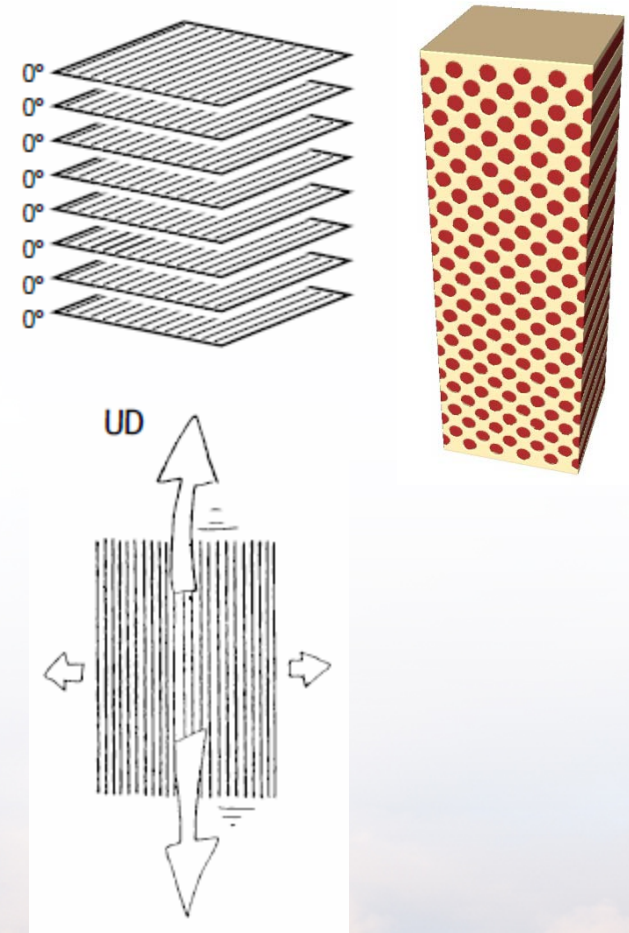
Obtain shock data to characterize the anisotropic response of fiber composite materials for development of advanced EOS and constitutive models

Utilize simulation results to better understand observed response

Test Material:

Hexcel IM7/8552
unidirectional / laminate
vary volume fraction (62, 65, 68%)

Extremely low porosity material
(ultrasound / x-ray confirmed)

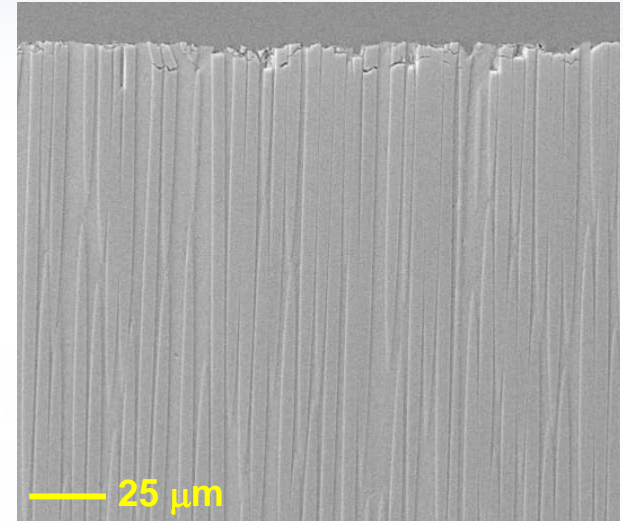
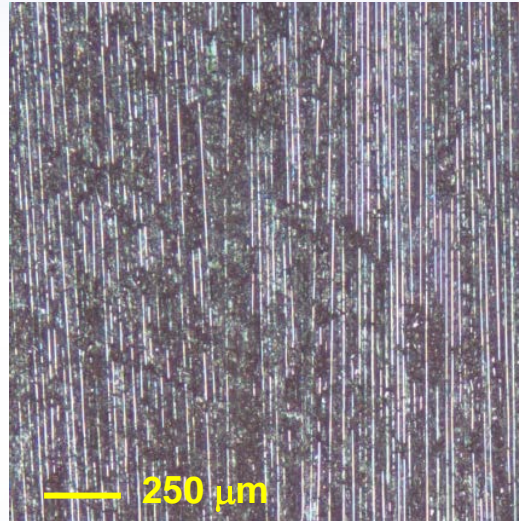
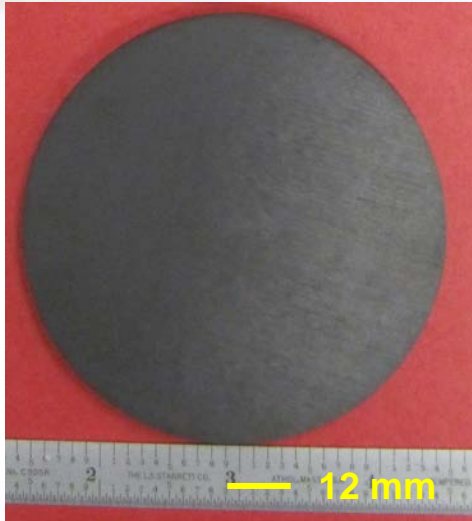


Hexcel Prepreg Technology Guide
Publication No. FGU 017b
March 2005



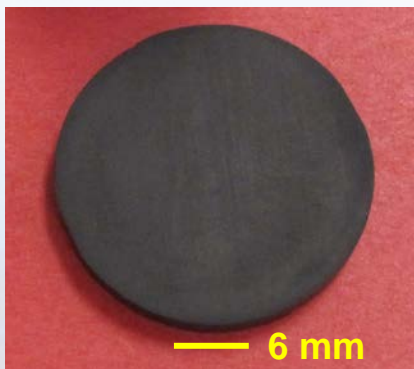
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Material is engineered to exhibit anisotropic response



T = 10.00 kV WD = 12.9 mm Signal A = SE2 File Name = CFE-68-0-21_longitudinal

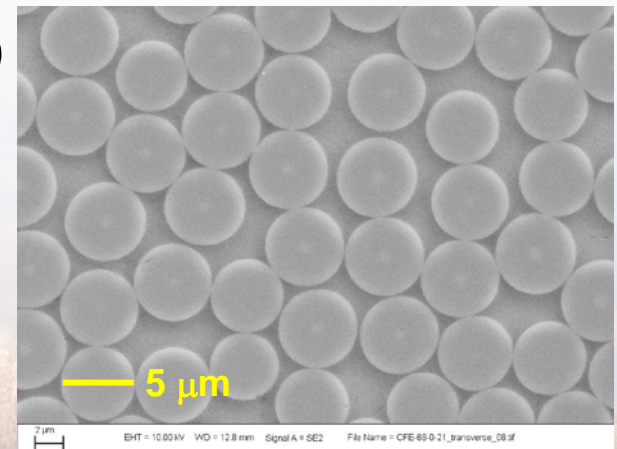
Cross-fiber orientation (90°)



On-fiber orientation (0°)

Ultrasonic Wave Speeds

0°	10.763 km/s
90°	3.042 km/s

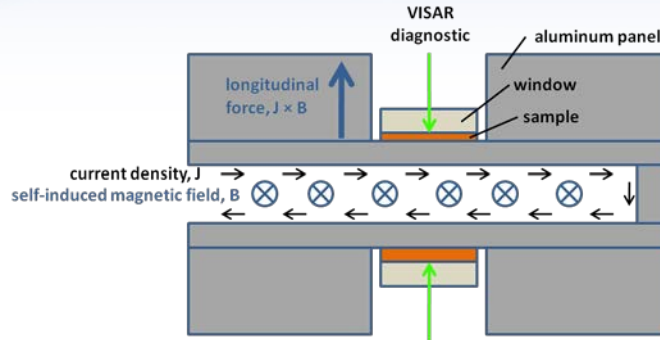


2 μm EHT = 10.00 kV WD = 12.9 mm Signal A = SE2 File Name = CFE-68-0-21_transverse_08.tif

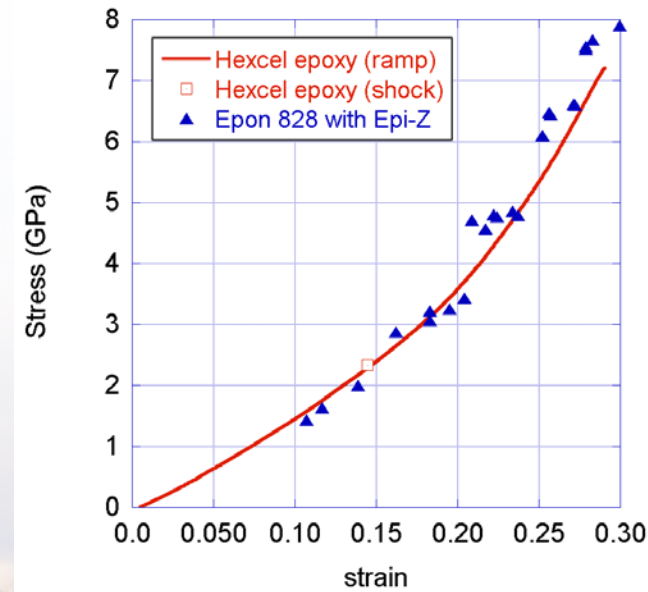
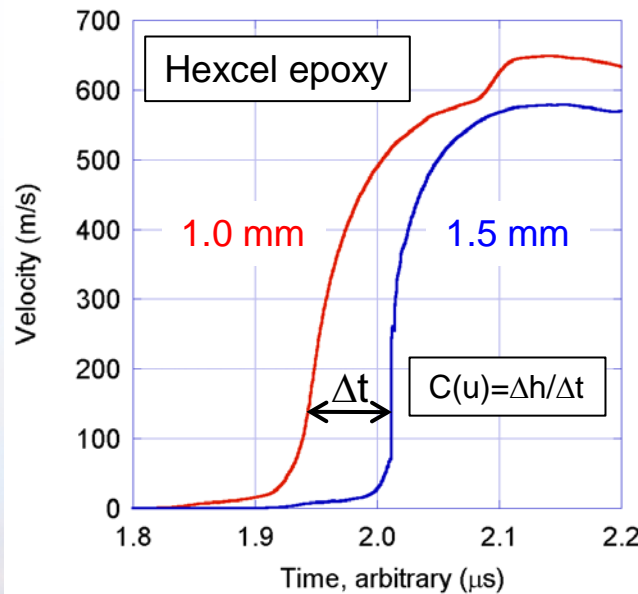
Unreinforced epoxy was characterized using both shock and ramp loading



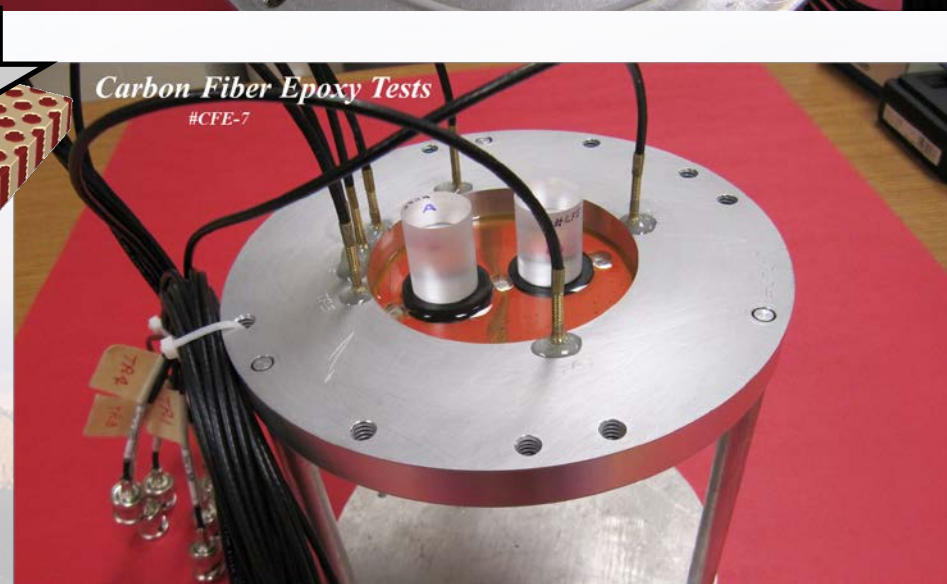
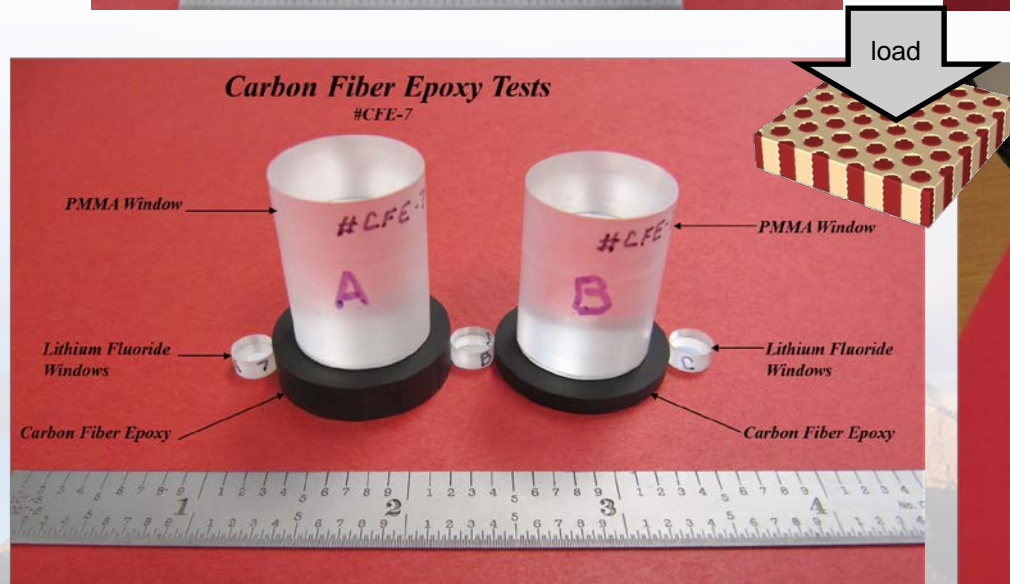
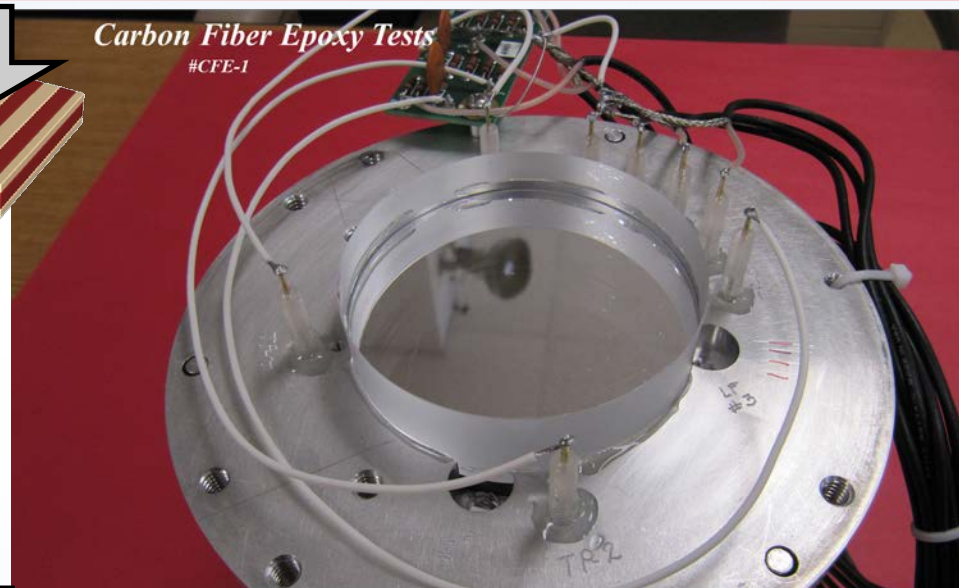
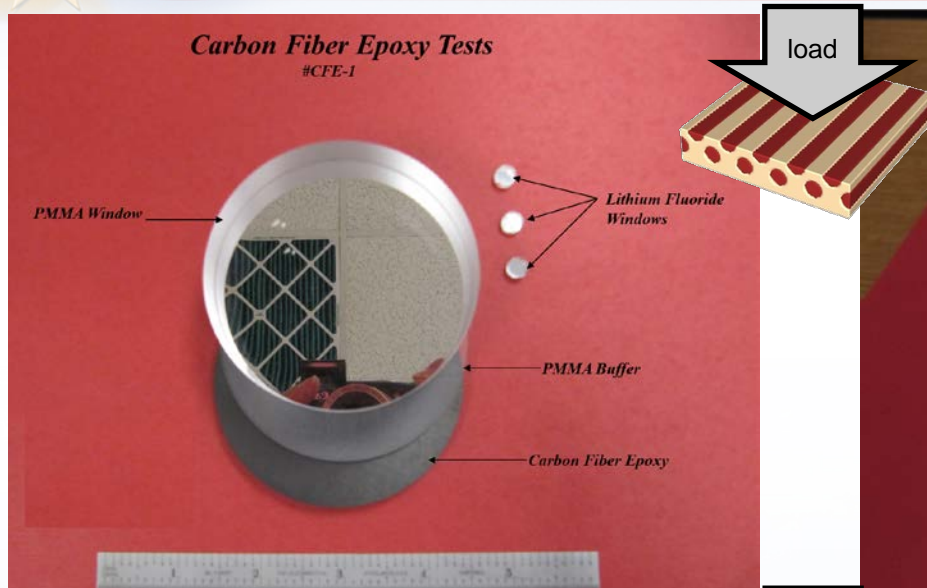
Hexcel 8552 epoxy test sample



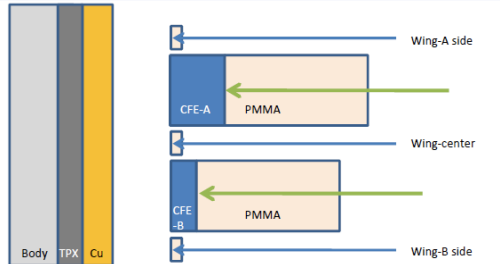
- Epoxy response is needed by model development effort
- Response measured via shock and ramp loading techniques
- Data similar to another common (well characterized) epoxy



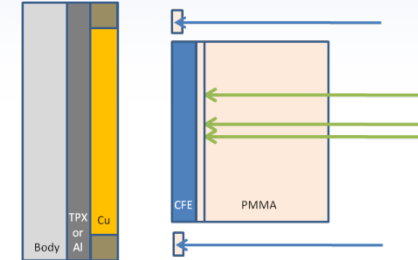
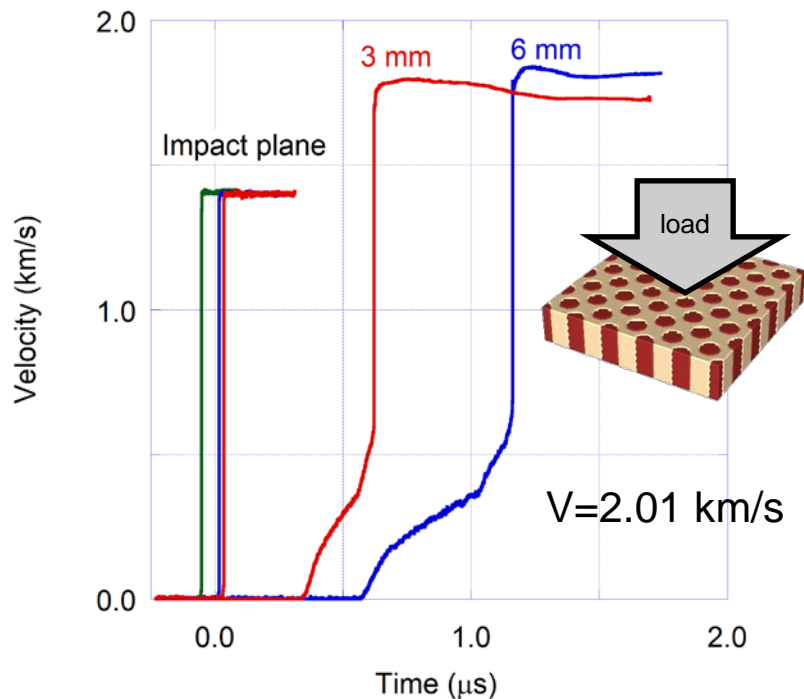
Experimental designs vary slightly depending on fiber orientation



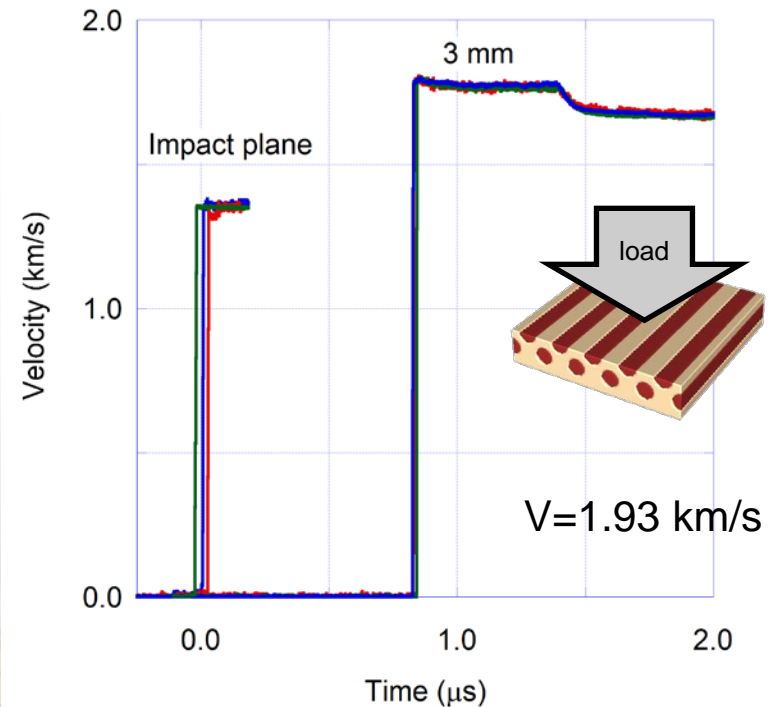
Response depends on fiber orientation



0° (shock along fiber direction)



90° (shock normal to fiber direction)



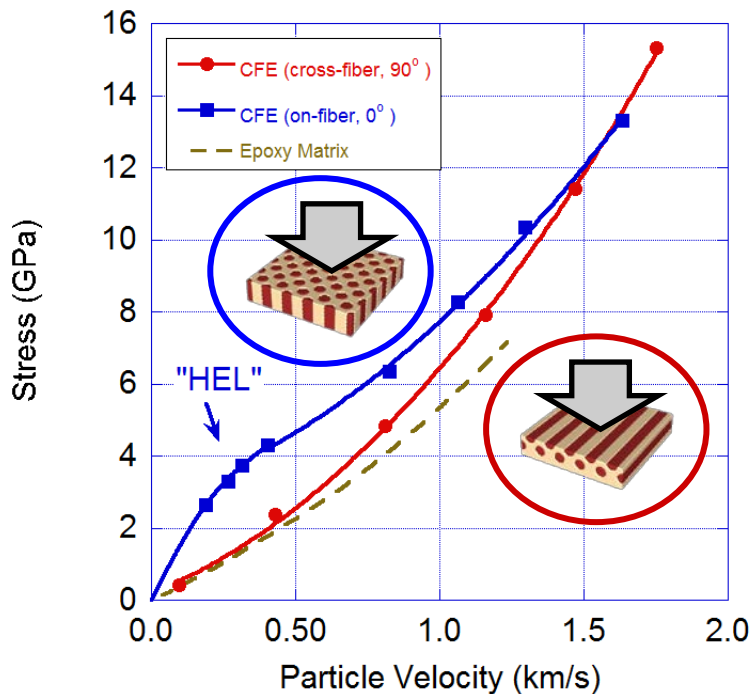
Data for 68% fiber FV



Shock response of CFE is anisotropic up to ~10-12 GPa

The vast majority of models in the dynamic regime assume **isotropic behavior**, especially for shock response

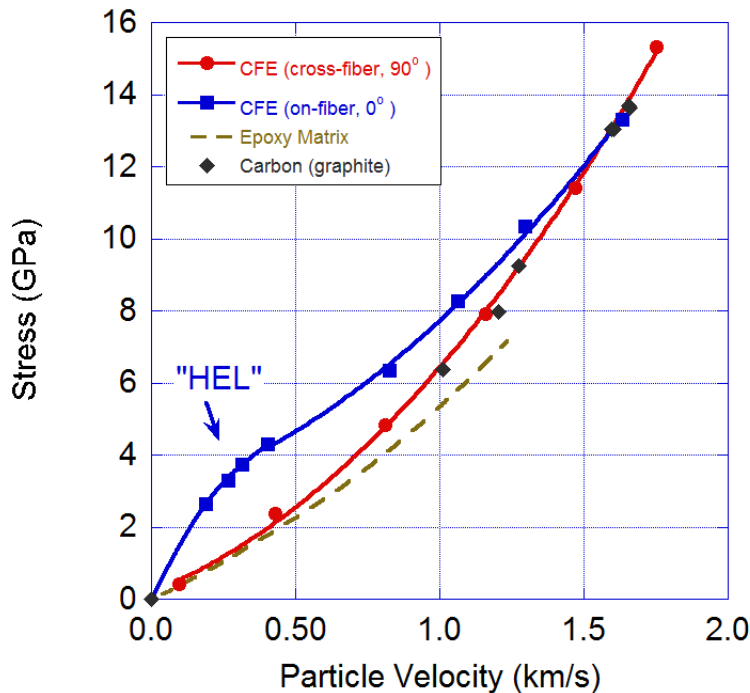
- Fiber stiffens matrix at low-intermediate pressures (up to ~10-12 GPa)
- Response appears to become isotropic at higher pressures
 - Epoxy disassociates at 10-20 GPa



Cross-fiber material response is similar to pressed graphite

Cross fiber response is similar to carbon, stiffer than epoxy

- Fiber stiffens matrix at low-intermediate pressures (up to ~10-12 GPa)
- Response appears to become isotropic at higher pressures
 - Epoxy disassociates at 10-20 GPa
- Comparison with pressed carbon of similar density shows like response in cross-fiber orientation



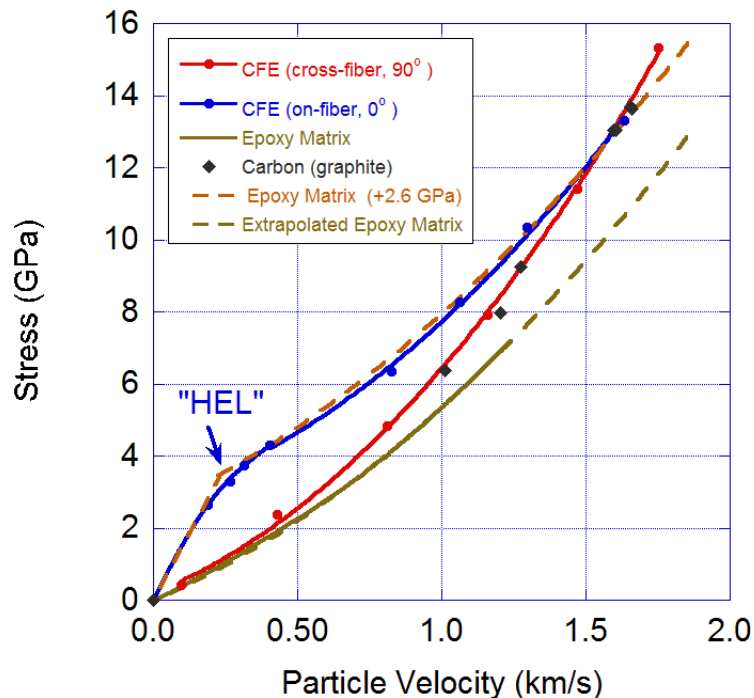
Pressed graphite $\rho_0 = 1.77 \text{ g/cm}^3$

CFE $\rho_0 = 1.58 \text{ g/cm}^3$



Fibers provide an elastic component to the epoxy response (HEL = 3.5 GPa) on-fiber

On-fiber bulk response is similar to epoxy but with an HEL of 3.5 GPa

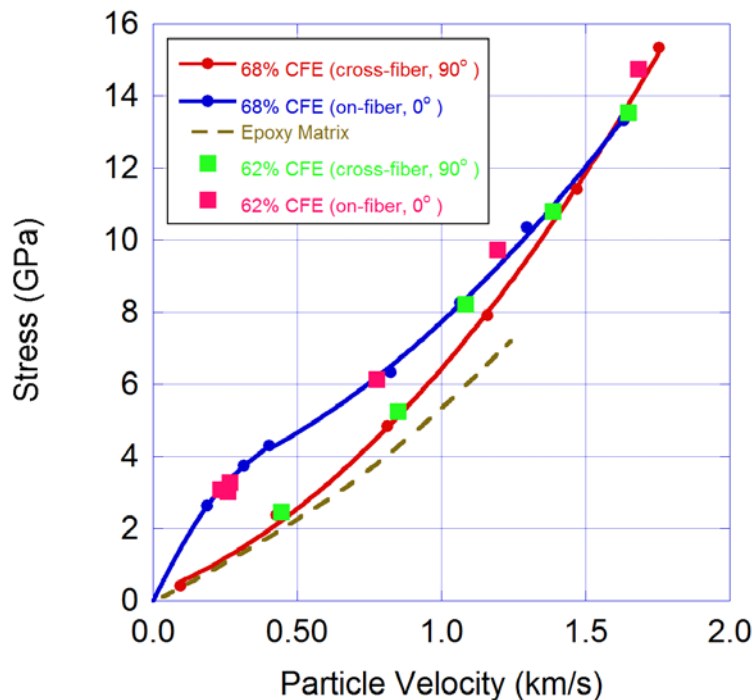


- Fiber stiffens matrix at low-intermediate pressures (up to ~10-12 GPa)
- Response appears to become isotropic at higher pressures
 - Epoxy disassociates at 10-20 GPa
- Comparison with pressed carbon of similar density shows like response in cross-fiber orientation
- On-fiber response shows a stiffening of the epoxy matrix with an HEL of 3.5 GPa
- Different fundamental mechanisms in play for different orientations

Dynamic response is not sensitive to exact fiber fill volume

No observed difference in response of 62 – 68 % fiber fill volumes

- 62% fill material (large squares) shows similar dynamic response to 68% fill
- Samples tested are maximum and minimum possible fill volumes for this composite



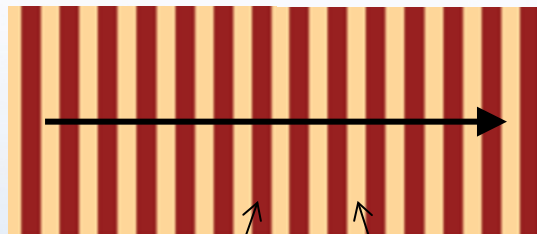
Understanding complex material response is improved through numeric simulation



Longitudinal Shock



Transverse Shock



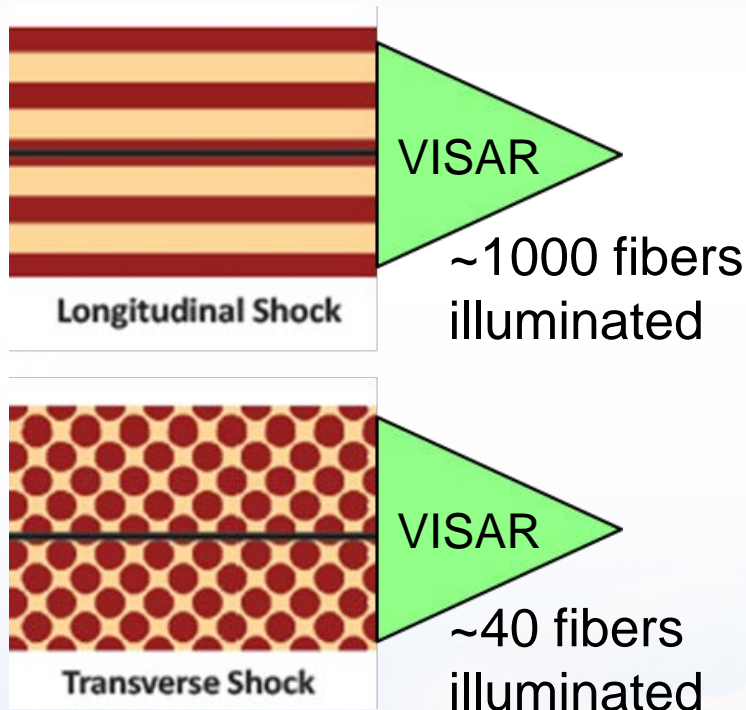
carbon fiber

epoxy

- Hydrocode simulations of multiple shock-fiber orientations
- Explicit modeling of all constituents
 - Small mesh to resolve fibers
 - ≥ 10 cells/fiber ($\Delta x \leq 500$ nm)
 - Limited domain sufficient to achieve stable waves

	Fiber Longitudinal	Fiber Transverse	Matrix
Mie-Gruneisen EOS	$\rho_o=1.694$ g/cc $C_o=4.50$ km/s $S=1.478$	$\rho_o=1.694$ g/cc $C_o=2.35$ km/s $S=1.478$	$\rho_o=1.305$ g/cc $C_o=2.35$ km/s $S=1.604$
Elastic-Perfectly Plastic Strength	$\nu=0.050$ $Y_o=3.5$ GPa	$\nu=0.352$ $Y_o=3.5$ GPa	$\nu=0.393$ $Y_o=0.1$ GPa

Simulation data is interpreted consistent with VISAR diagnostic averaging



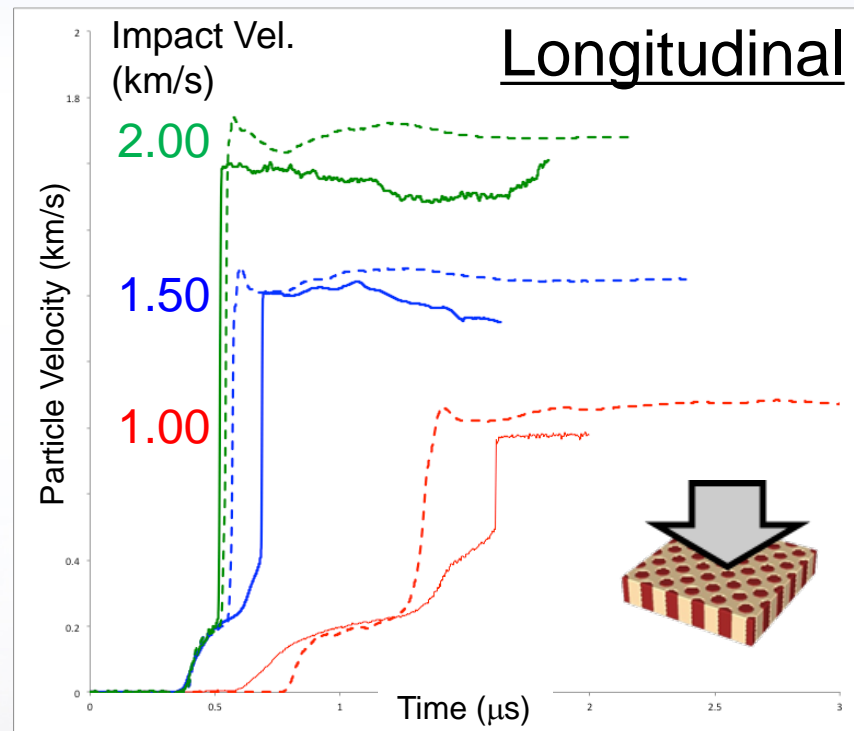
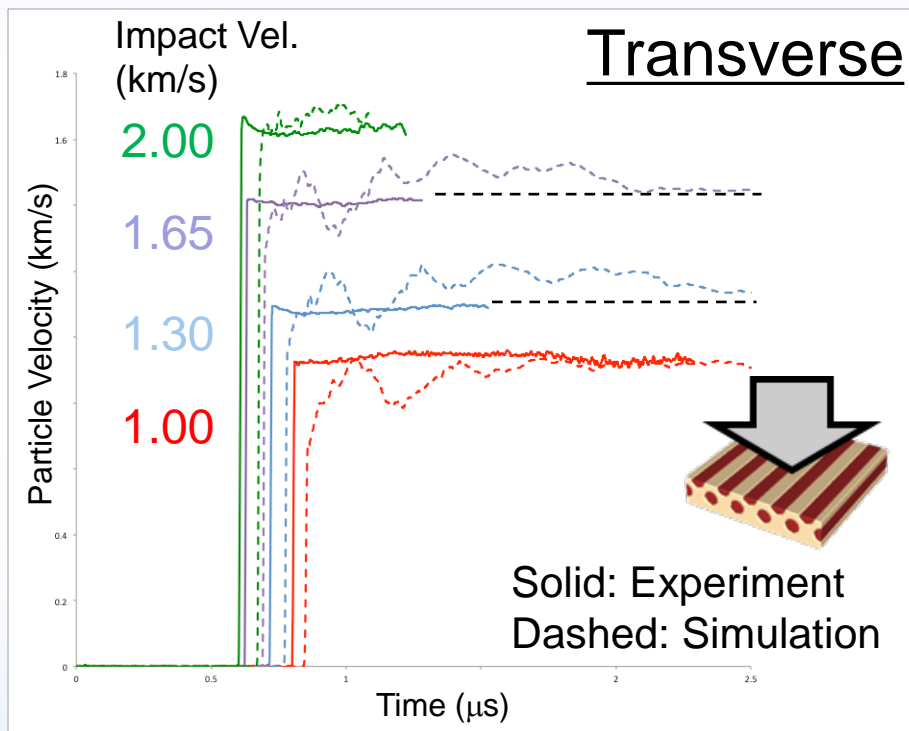
- VISAR diagnostics use $\sim 200\mu\text{m}$ spot size
- Multiple fibers and surrounding matrix are illuminated
- Volume fraction weighting used to mimic VISAR detection

$$\begin{aligned} Vel_{composite} &= \phi_{fiber} Vel_{fiber} \\ &+ \phi_{matrix} Vel_{matrix} \end{aligned}$$

- Periodicity of fibers in simulation domain is known to result in oscillatory response*



Simulation results are in good agreement with experiments

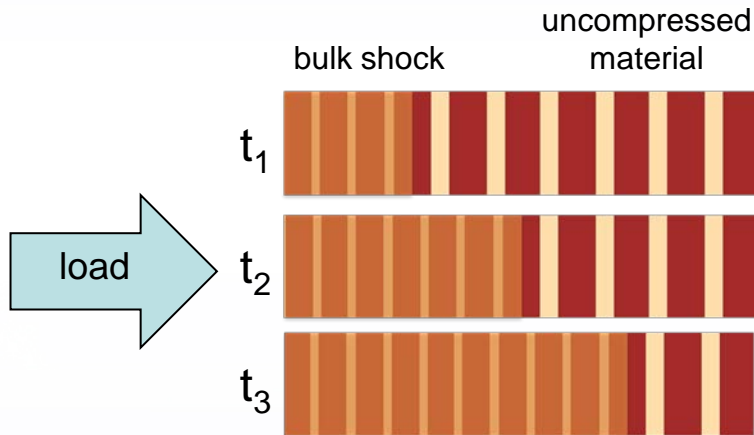


- Shock velocities agree within 10%
- Particle velocities agree within $\leq 5\%$ (generally better)

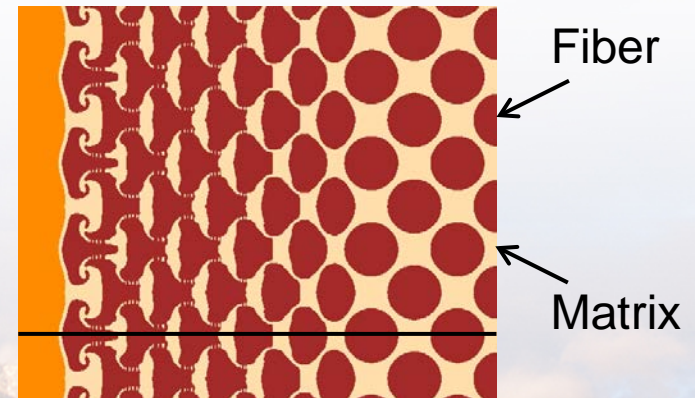
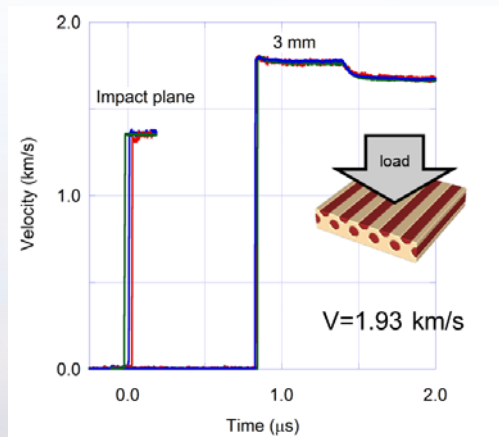
- Shock velocities agree within 10% of linear ramp arrival
- Particle velocities agree within $\leq 10\%$ (generally better)

Simulation results help to visualize and understand observed shock response

Cross-fiber orientation (90°)

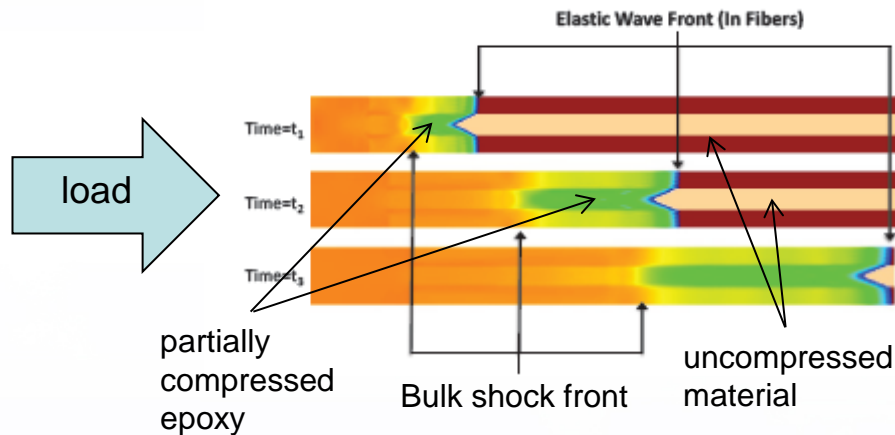


- Epoxy is compressed
 - Jet formation due to wave speed differences
- Fiber-on-fiber contact may occur
- Shock wave moves comparable to bulk carbon response at a similar density

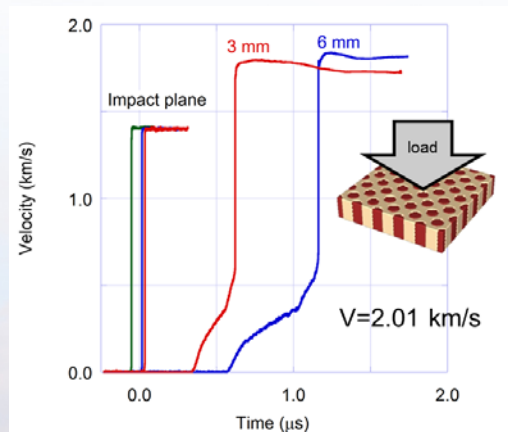


Simulation results help to visualize and understand observed shock response

On-fiber orientation (0°)



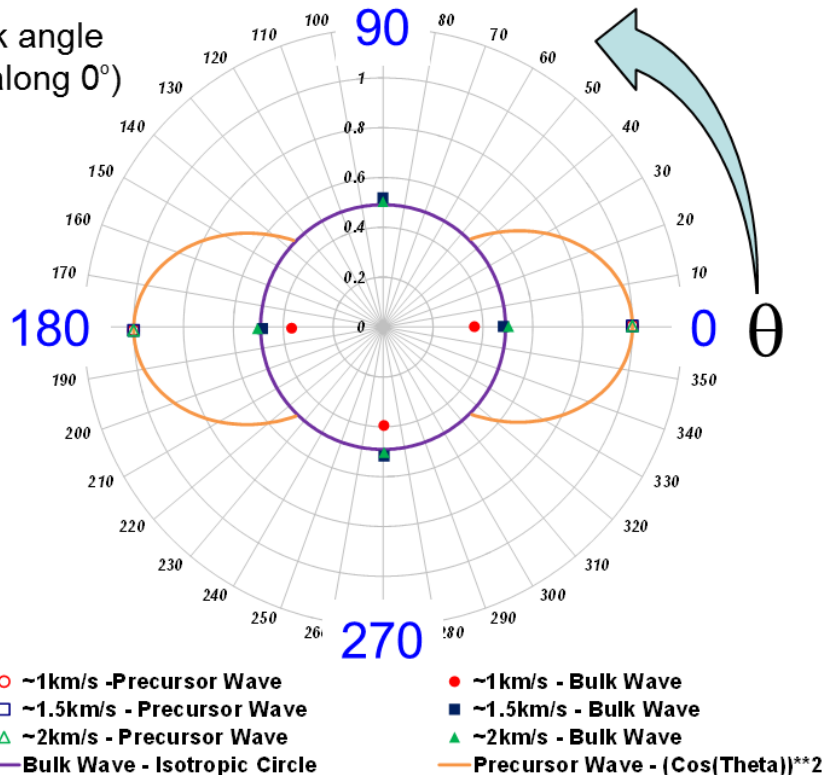
- Elastic waves travel along fibers
 - Observed elastic response entirely in fibers
- Mach wave features observed in epoxy behind elastic waves
- Bulk shock travels at slower epoxy velocity
- Results in observed complex loading structure



Shocks in real world applications rarely travel only along principal axes

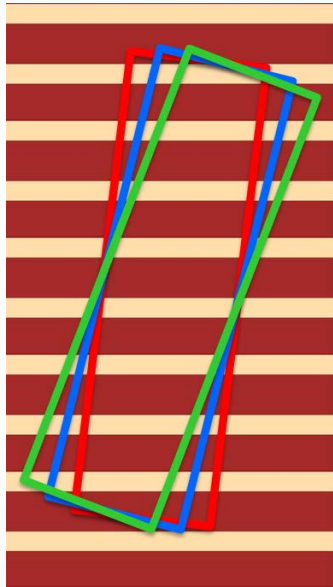
Polar Plot of Normalized Shock Velocity for IM7/8552 Unidirectional Composite

θ =shock angle
(fibers along 0°)

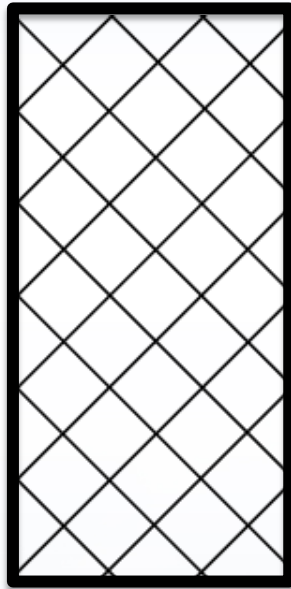


- Wave speeds best observed in polar space
 - Measurements made in $0-90^\circ$
 - Symmetry used to complete plot
- Velocities are normalized based on 0° precursor velocity
- Theory (solid curves) predicts
 - Elastic waves speeds will follow a $\cos^2(\theta)$ relation
 - Bulk shock will be isotropic
 - Elastic wave overdriven at 45°
- Experiments are required to verify these predictions

Off-principle-axis testing



7, 14, 21°
samples

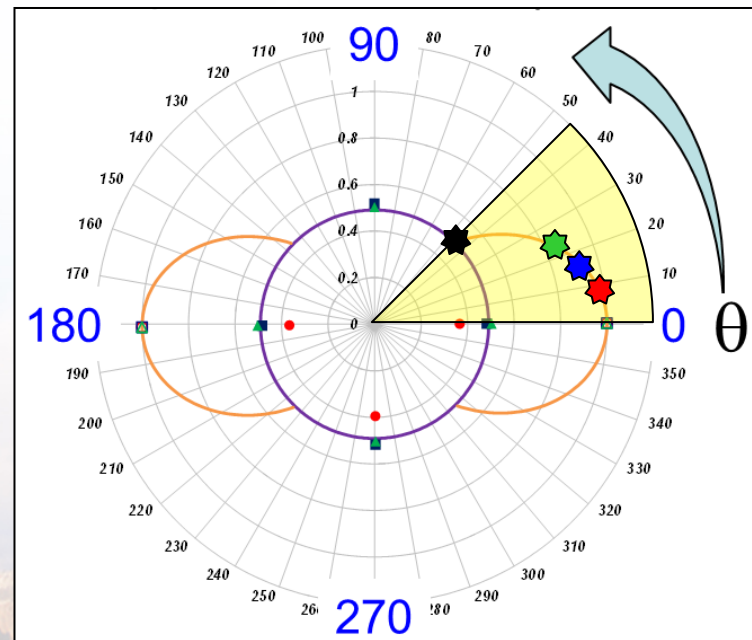


± 45°
samples

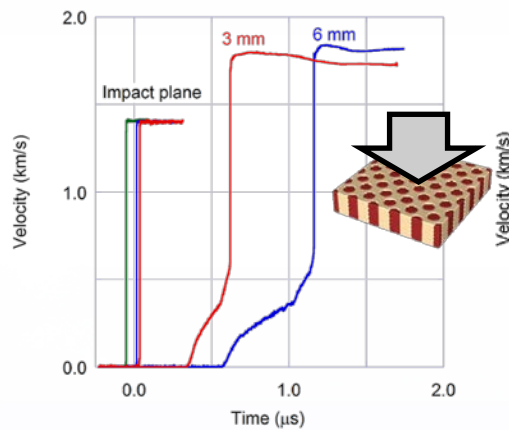


Example 21° sample

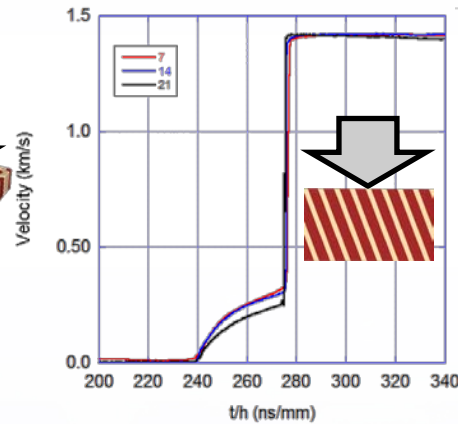
- Material manufactured as previously
- Samples cut from oversized parts at fixed angles (7, 14, 21 degrees)
- ±45 degree samples cut from 0-90 cross-ply material



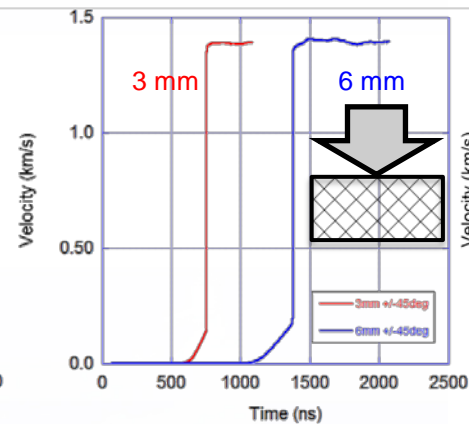
Off-principle-axis testing reveals trends



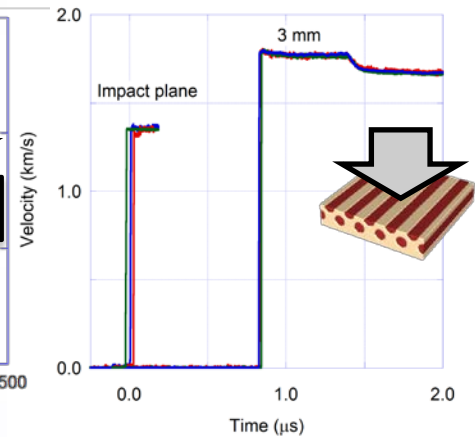
0°



7, 14, 21°



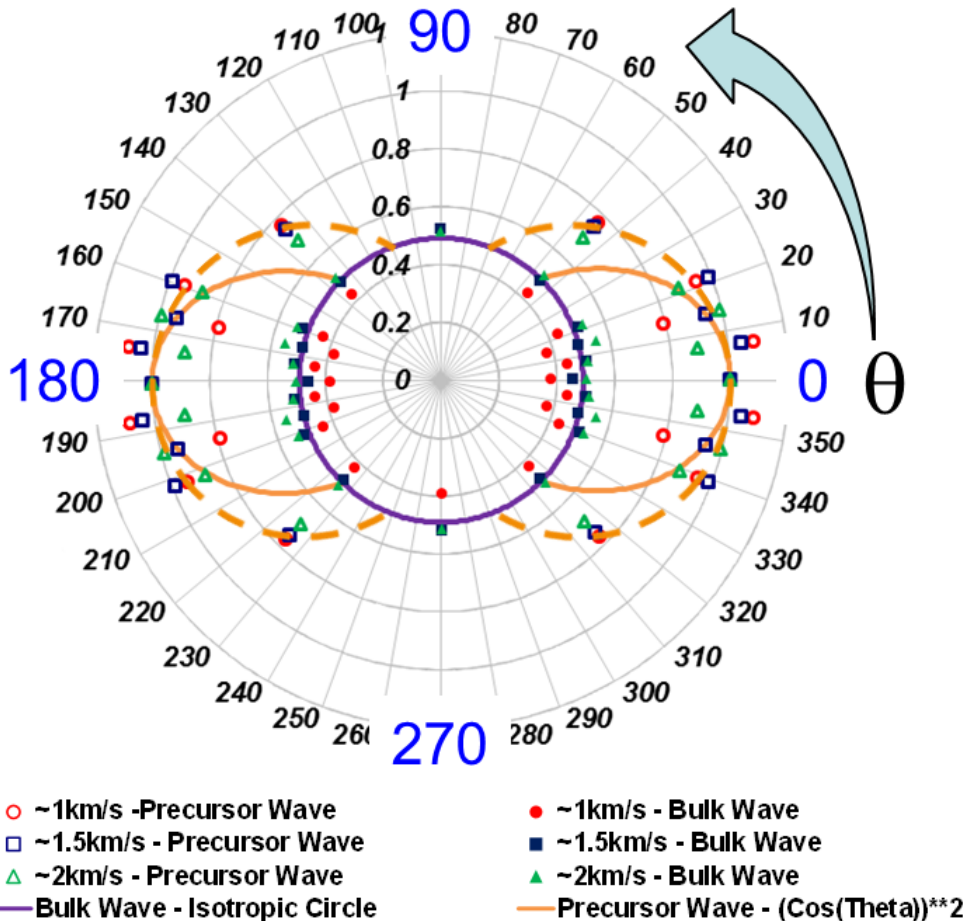
$\pm 45^\circ$



90°

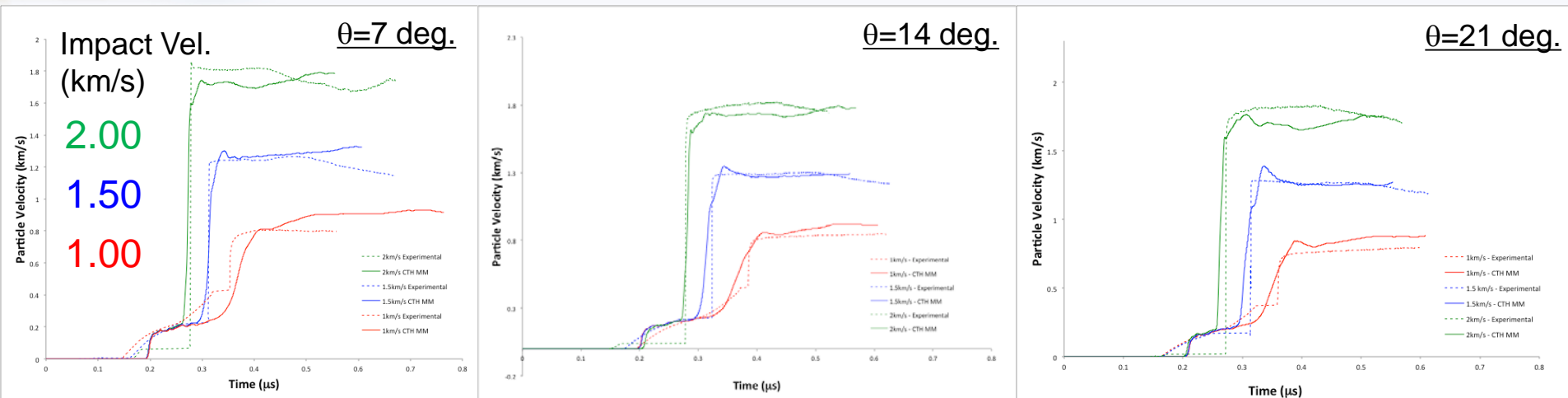
- Wave profiles show systematic trends
- Less pronounced precursor with increasing angle

Off-principle-axis testing verifies premise of predictions, suggests revision



- Solid curves represent isotropic (circle) response and predicted $\cos^2(\theta)$ response for precursor
- Isotropic bulk wave was observed
- Results suggest revised (dashed) precursor velocity curve matching bulk velocity at around 70°

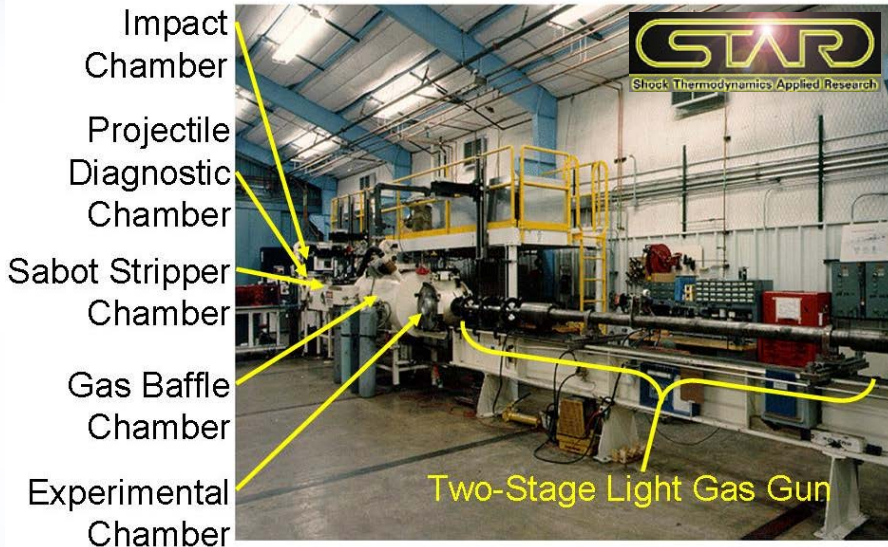
Model predictions (explicit) agree well with off-axis test results



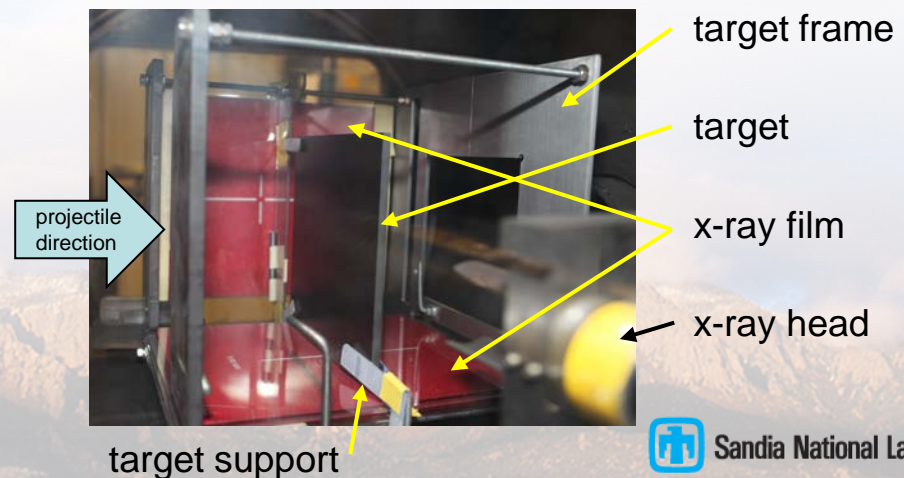
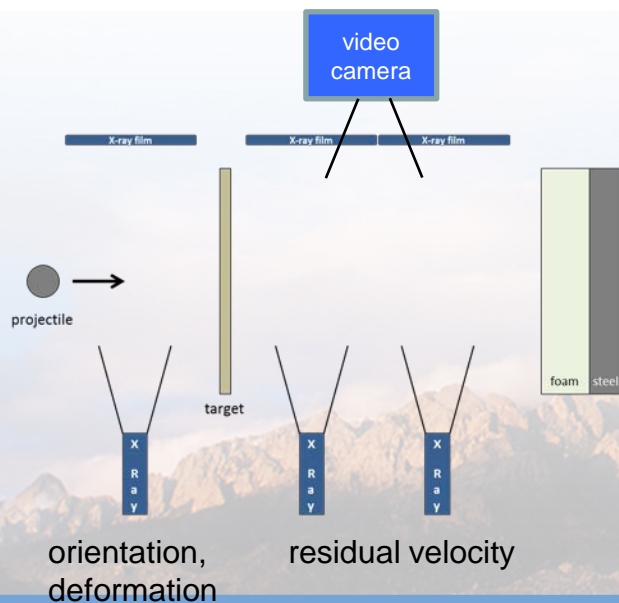
- Shock velocities agree within 10% (generally better)
- Particle velocities agree within $\leq 10\%$ (generally better)
- Consistent with on-axis results, discrepancies exist in detailed wave structure of precursor

Solid: Simulation
Dashed: Experiment

Testing was performed to support validation of advanced composite models



- Impact 0, 90, +45, -45 layup composite with standard projectiles
 - Hard (440C) steel 1/4" sphere
 - Soft (1018) steel 1/4" sphere
 - Sourced from Bal-tec
 - Previously characterized by Tom Mason
- Impact velocities similar to typical threats
 - 250, 500, 1000 m/s
- Vary composite thickness
 - 0.25, 0.50, 1.00 inch
- Measure impact and residual velocities
- Post shot damage characterization via ultrasound and/or x-ray CT



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X-ray images provide residual velocity and projectile shape information

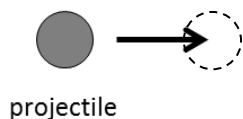
X-ray film

X-ray film

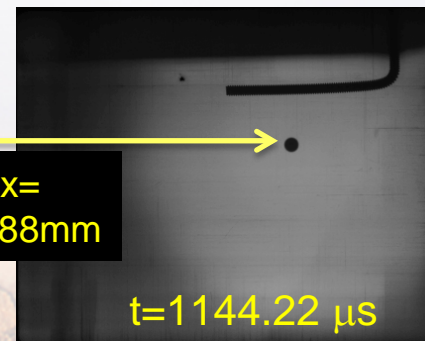
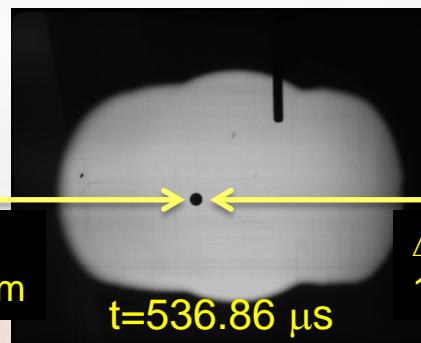
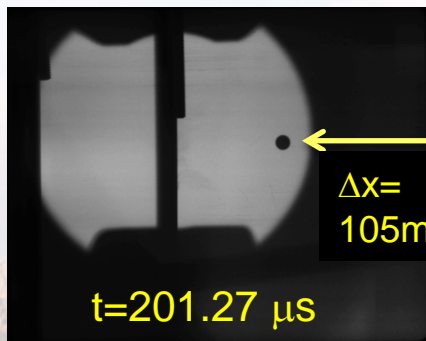
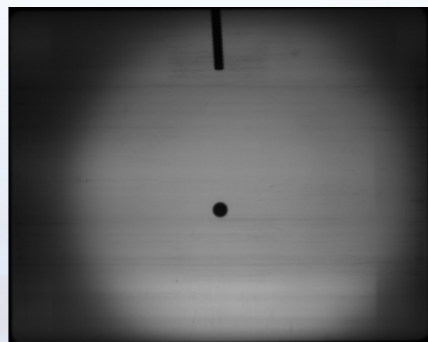
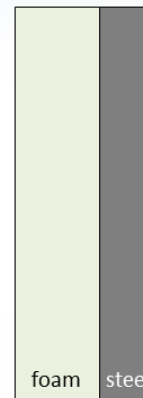
X-ray film

X-ray film

Impact velocity
= 501 m/s



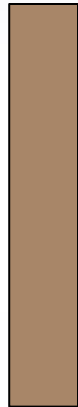
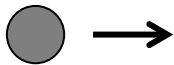
Residual velocity
= 310 m/s



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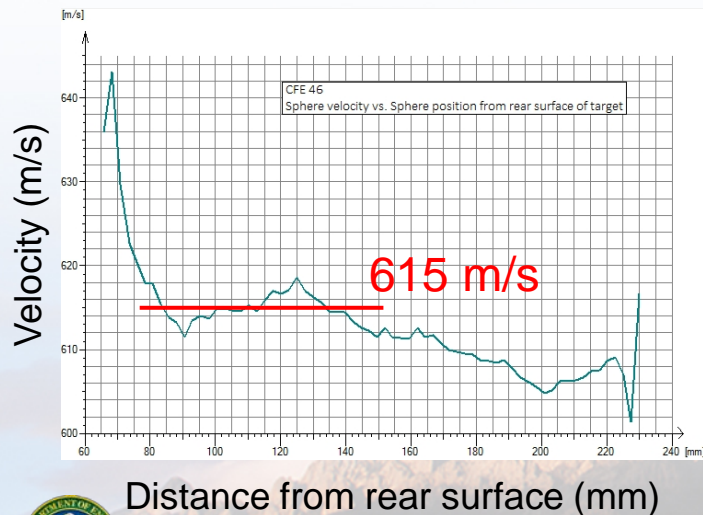
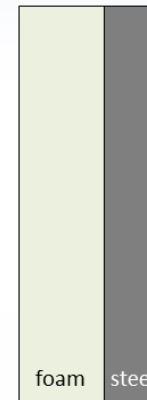
High speed video used in place of x-ray diagnostics on some shots

Impact velocity
= 1091m/s

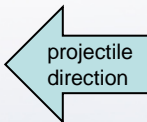


target

Residual velocity
= 615 m/s



CFE Test Series
May 5 - 7, 2014
STAR Facility/TBF Gun
CFE-46
Average velocity 610 m/s



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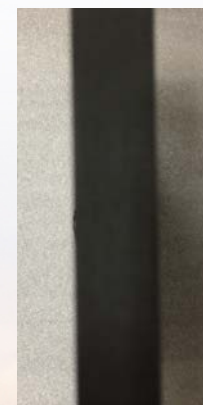
Test matrix includes range of impact conditions resulting in varying damage

Target Thickness (in)	Impactor	Impact Velocity (m/s)	Residual Velocity (m/s)	Diagnostics
1/4	soft	282	0	x-ray
1/4	soft	320	0	x-ray
1/4	soft	501	310	x-ray
1/4	hard	263	0	x-ray
1/4	hard	309	0	x-ray
1/4	hard	503	unknown*	x-ray
1/2	soft	507	0	x-ray
1/2	soft	517	0	x-ray
1/2	soft	1073	562	video
1/2	hard	502	0	x-ray
1/2	hard	520	0	x-ray
1/2	hard	1091	615	video
1	soft	1073	0	video
1	hard	1072	0	video

* Trigger failure on x-ray diagnostic system

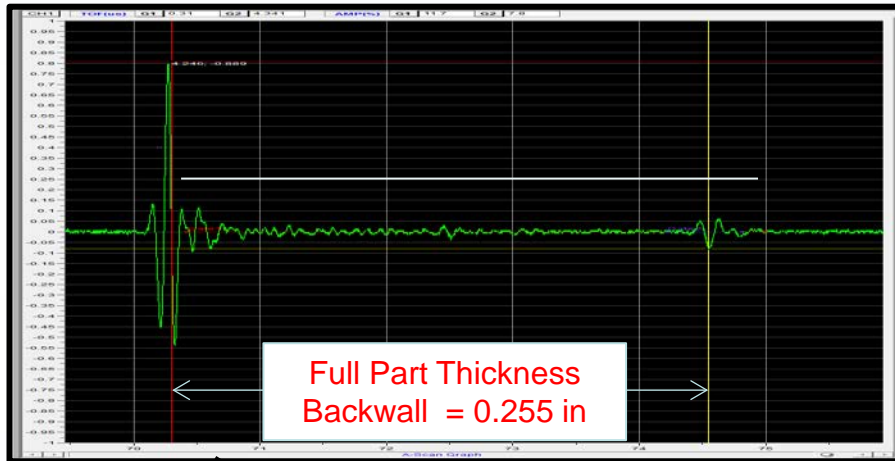
Penetration?

velocity thick	300 m/s	500 m/s	1000 m/s
1/4"	no	yes	
1/2"		no	yes
1"			no

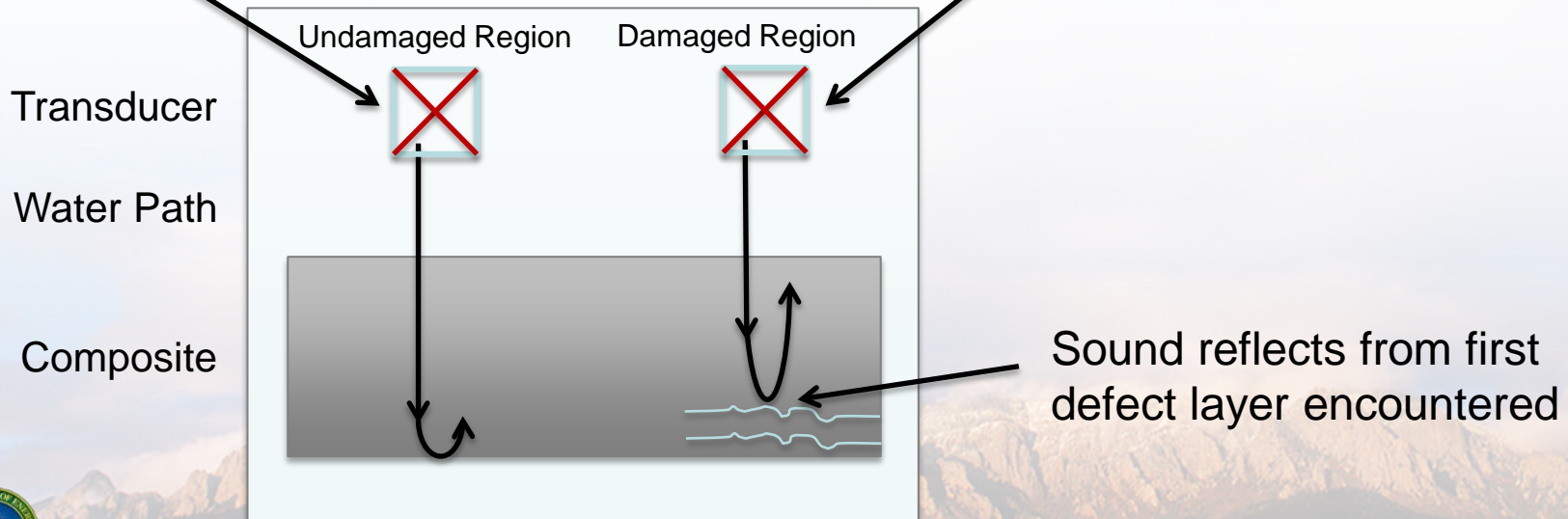
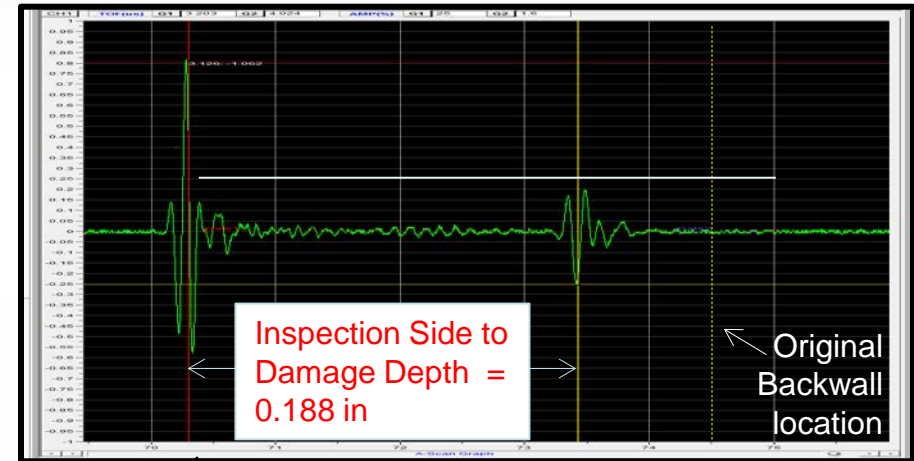


Ultrasound utilizes sound waves to characterize internal damage

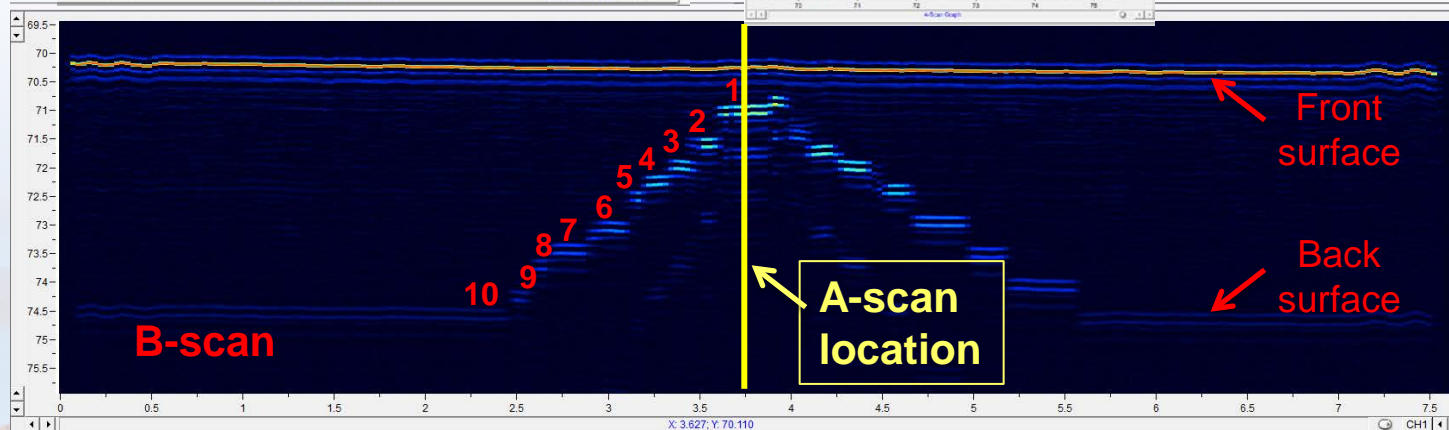
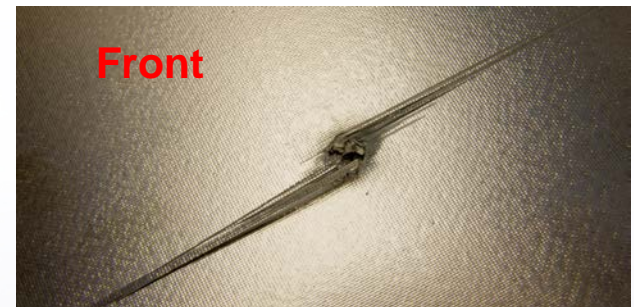
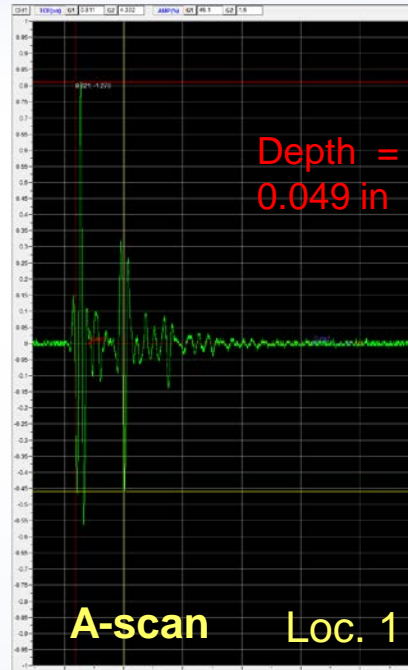
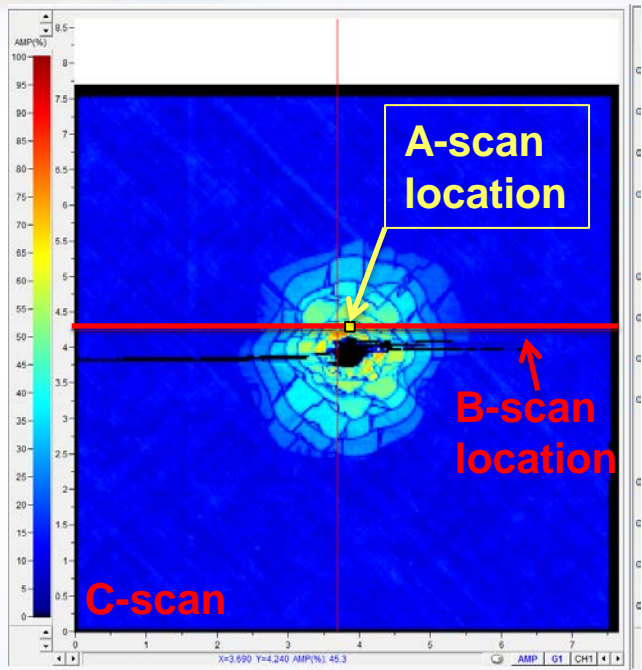
Undamaged ¼" Panel A-Scan Signal



Damaged ¼" Panel A-Scan Signal



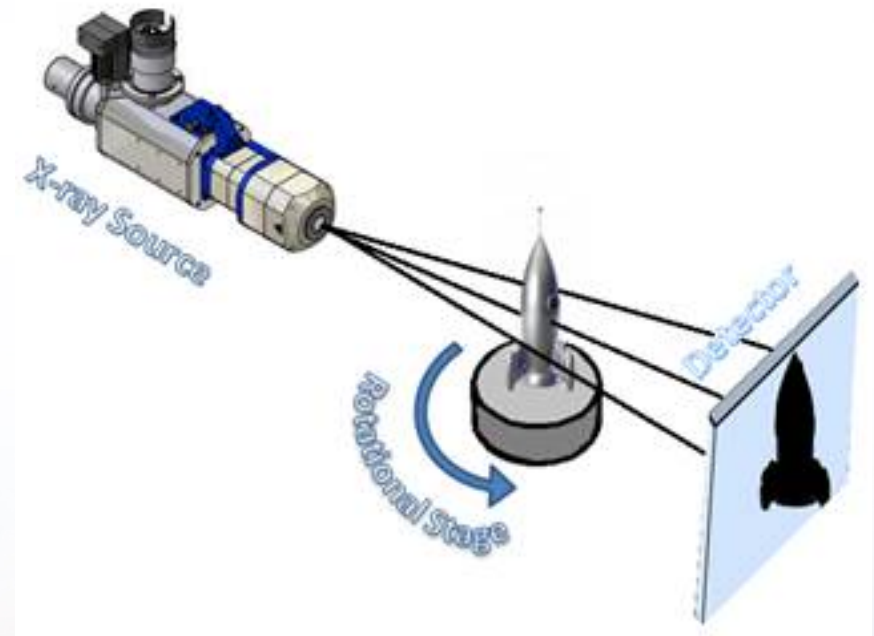
Ultrasound data used to determine depth of leading edge damage in target



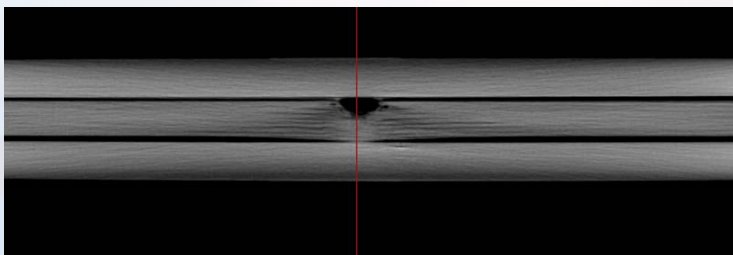
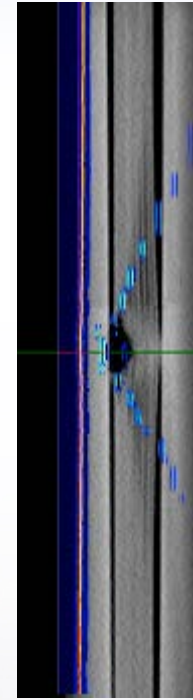
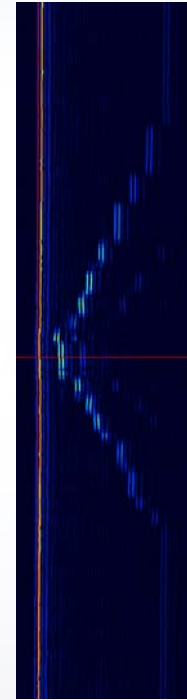
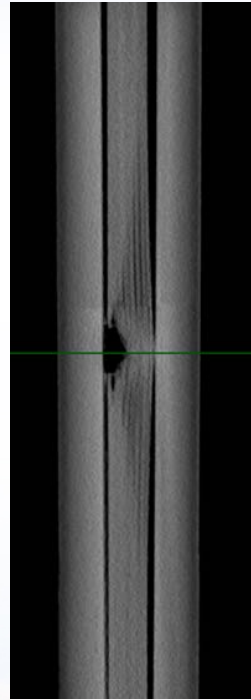
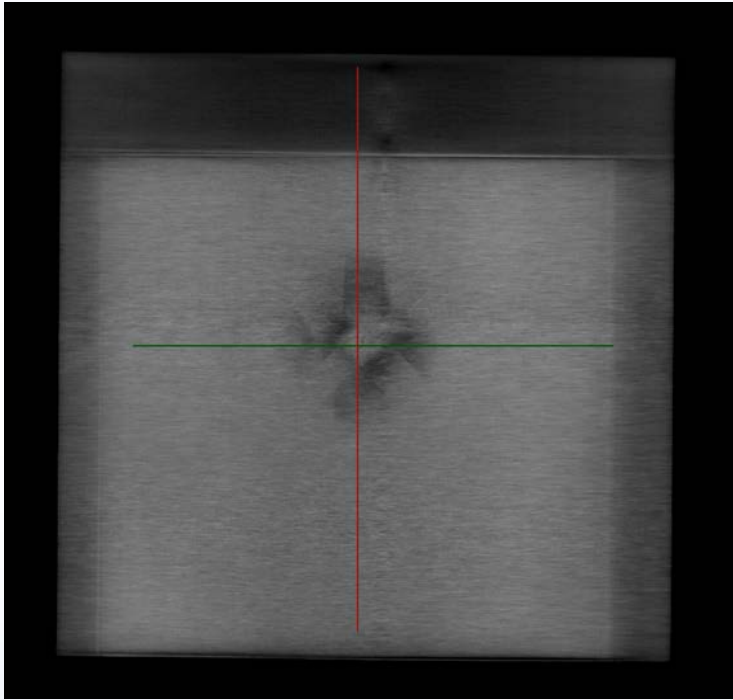
Point	Depth into Part (in)
1	0.049
2	0.079
3	0.101
4	0.118
5	0.132
6	0.163
7	0.188
8	0.204
9	0.236
10	0.255

X-ray computed tomography (CT) provides 3D view of damage patterns

- Penetrating radiation (X-Ray) attenuated by composite material
- Digital sampling of the radiation
- Multiple images taken at different angles while spinning 360 degrees.
- Images mapped into a three dimensional data set
- Data shown as single “slice” or rendered



Combination of x-ray CT and ultrasound data will be compared to simulated damage



- Non-penetrating shot showing characteristic damage pattern
- 1/4" panel, 309 m/s impact, hard sphere



Summary of results

- **Dynamic response of composite is anisotropic**
 - highlights need for advanced models
- **Compression across fibers shows:**
 - no elastic precursor
 - bulk response similar to pressed carbon
- **Compression along fibers shows:**
 - an elastic precursor traveling along fibers
 - a bulk response similar to the epoxy binder
- **Explicit numeric simulations confirm experimental observations and offer new insight**
 - transverse simulations in excellent agreement with data
 - longitudinal simulations match all but fine detail in precursor wave
- **Off-axis response consistent with predictions**
 - require minor alterations to form of $f(\theta)$
 - simulations in good agreement with data
- **Testing has been performed to provide a basis for advanced model validation**
 - Impact and residual velocities known
 - Damage characterized with x-ray CT and ultrasound

