

FINITE ELEMENT STRESS ANALYSIS OF ITER MODULE 13

**Elastic Finite Element Stress Analysis of Modules 13 Shield Block and
First Walls Under Operational Pressure and Temperature Distribution**

June 20, 2007

Joe Garde, Dennis Youchison, Greg Natoni, Jim Bullock, T.J. Tanaka
Sandia National Laboratories

Manmeet Narula, Alice Ying
University of California, Los Angeles

M. Sawan, P. Wilson,
Fusion Technology Institute
University of Wisconsin-Madison

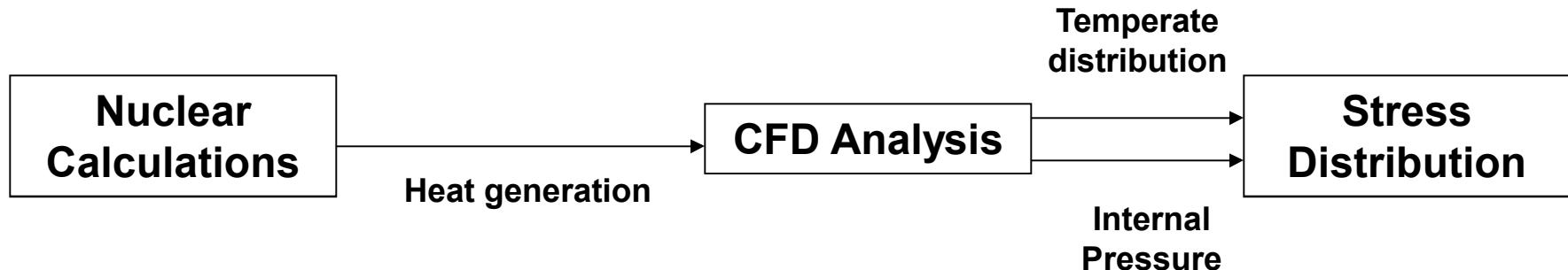


Sandia is a multiprogram laboratory operated by Sandia Corporation, a Lockheed Martin Company, for the United States Department of Energy's National Nuclear Security Administration under contract DE-AC04-94AL85000.



Purpose Of Analysis

- Demonstrate FEA stress results based on nuclear heating and CFD analysis

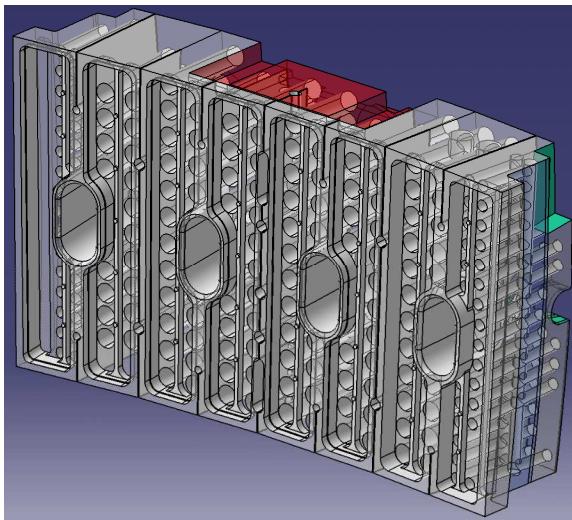


- Compare initial stress results to ITER criteria for stress limits during operating conditions

- **Acceptable stress intensity is dictated by S_m for elastic analysis.**
 - S_m is defined for each material based on a percentage of minimum, temperature dependant, ultimate and yield strengths.
- **S_m is the stress limit for primary stresses.**
 - Stresses caused by internal coolant pressure are classified as primary stresses
- **$3S_m$ is the stress limit for the combination of primary and secondary stresses.**
 - Stresses caused by temperature distribution classify as secondary stresses.

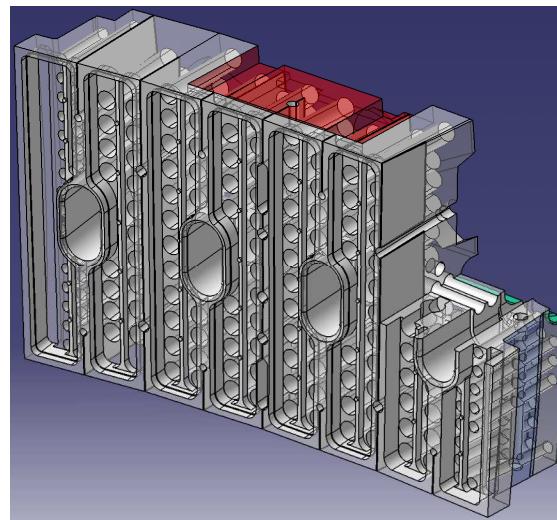
FEA Geometry For Shield Modules

- Linear Tetrahedral meshes created in CFDesign
- Temperature distribution taken from CFDesign results
- Thermal stresses based on initial temperature of 100°C
- Pressure stresses based on 3MPa coolant pressure
- ABAQUS used for elastic static FEA

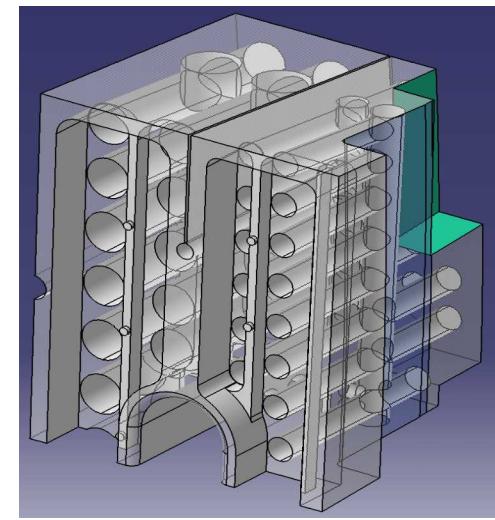


Full Geometry of
Shield Block

4

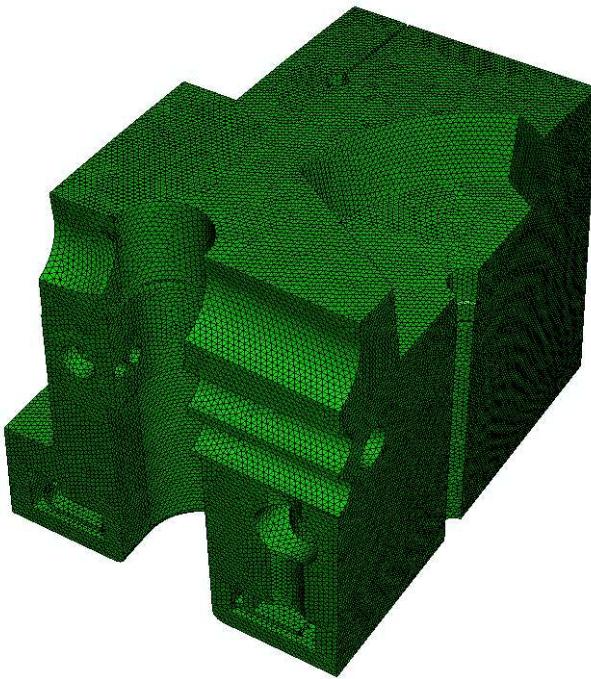


Analyzed Section
Missing

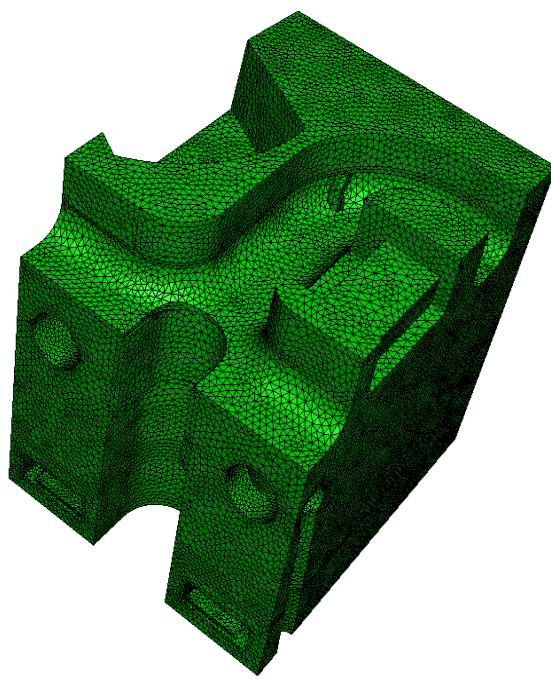


Analyzed Section

FEA Meshes For Shield Modules



Module 13 Corner

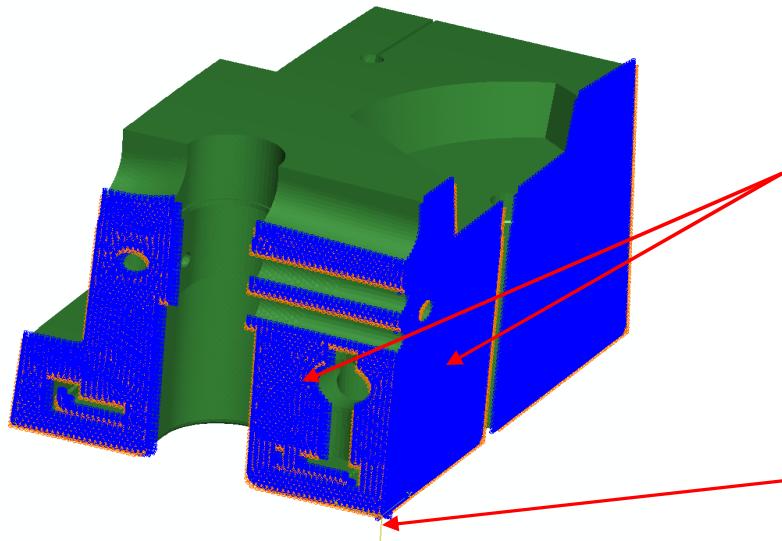


Module 13 Mid 1/8th



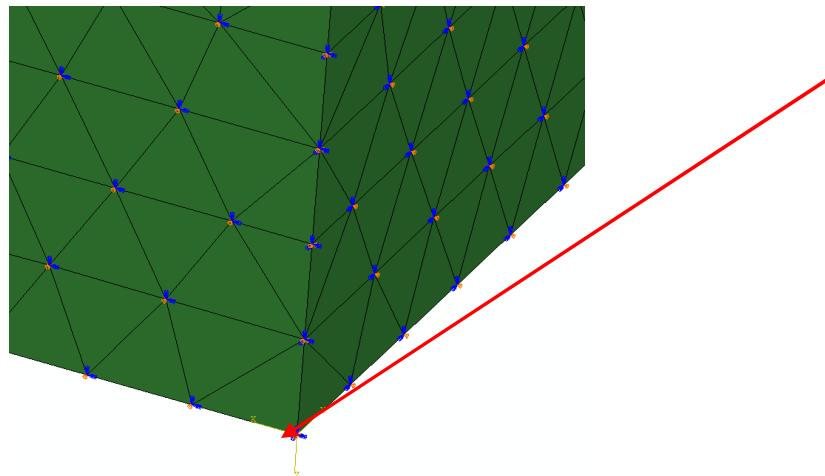
Module 13 First Wall

Boundary Conditions



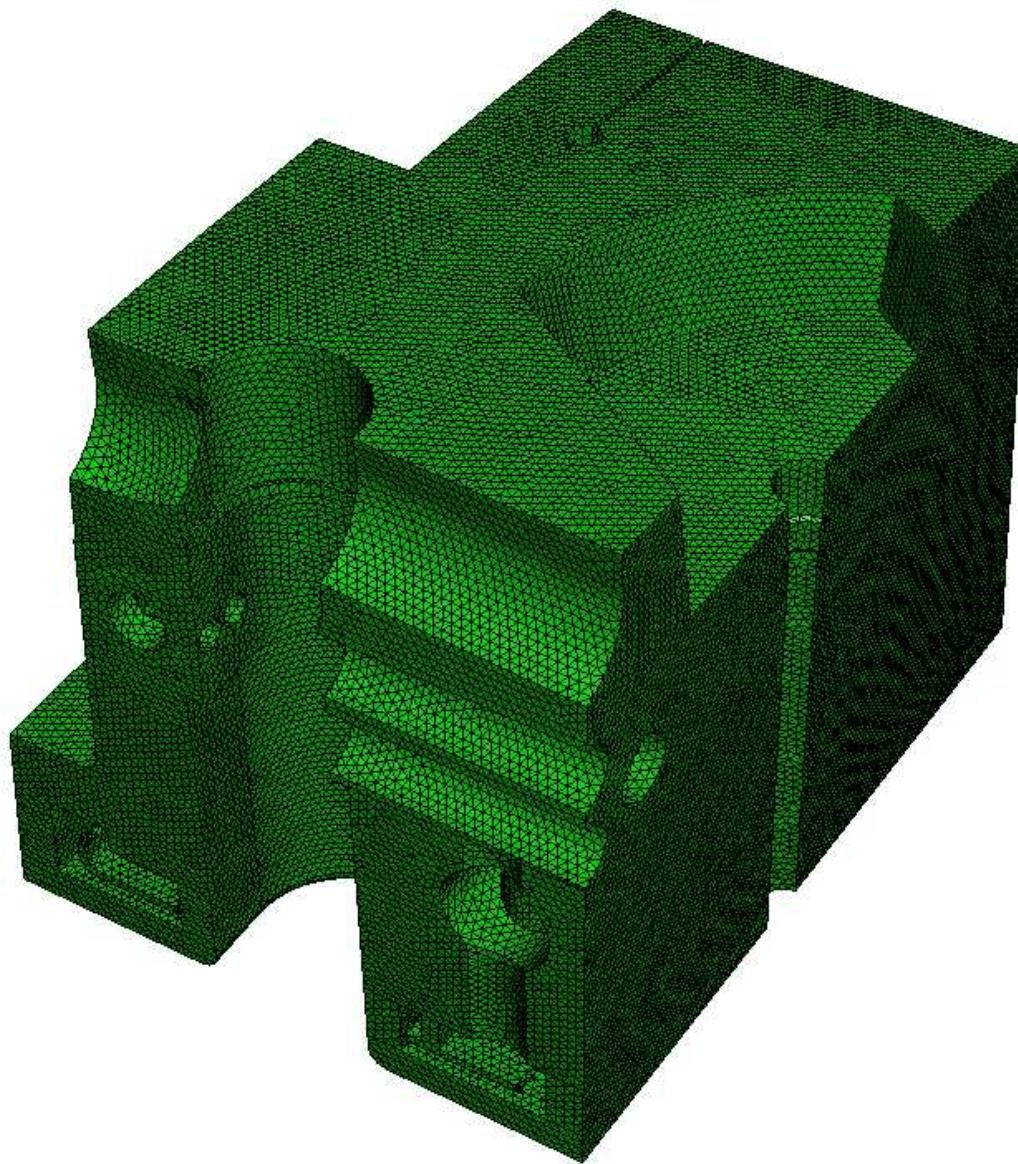
Cut faces (shown in blue) were restrained in 1 DOF relative to local coordinate system

Pinned Corner



Local coordinate system located at pinned node

FEA Mesh of Shield Module 13 Corner Section

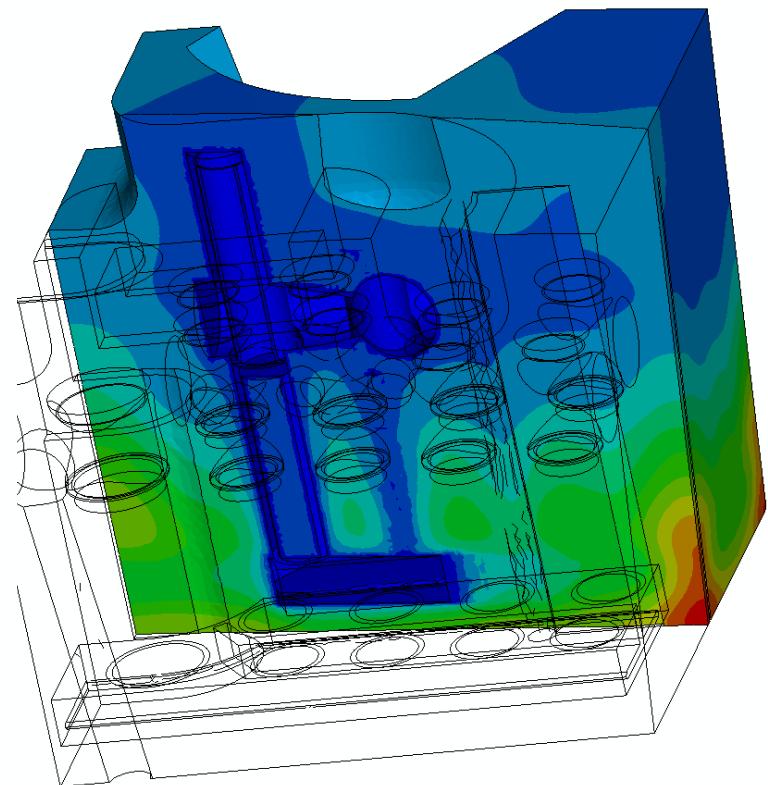
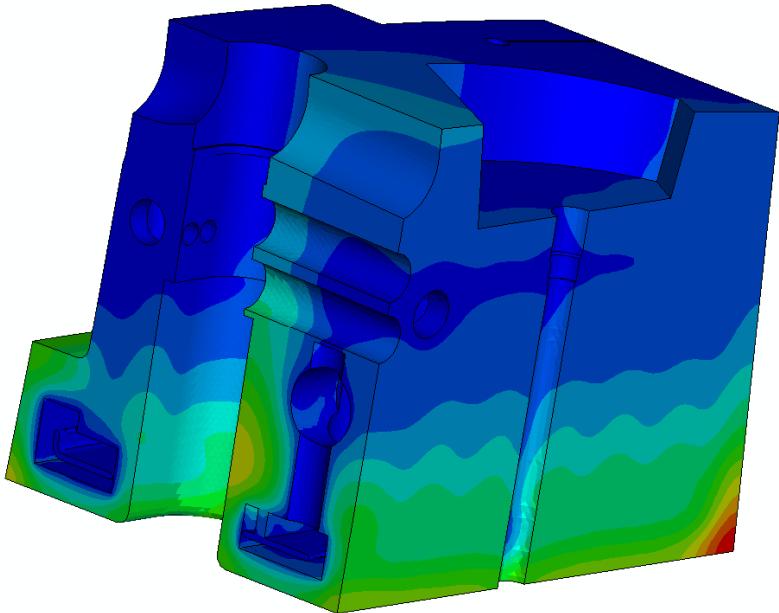


Temperature Distribution for Shield Module

13 Corner section °C

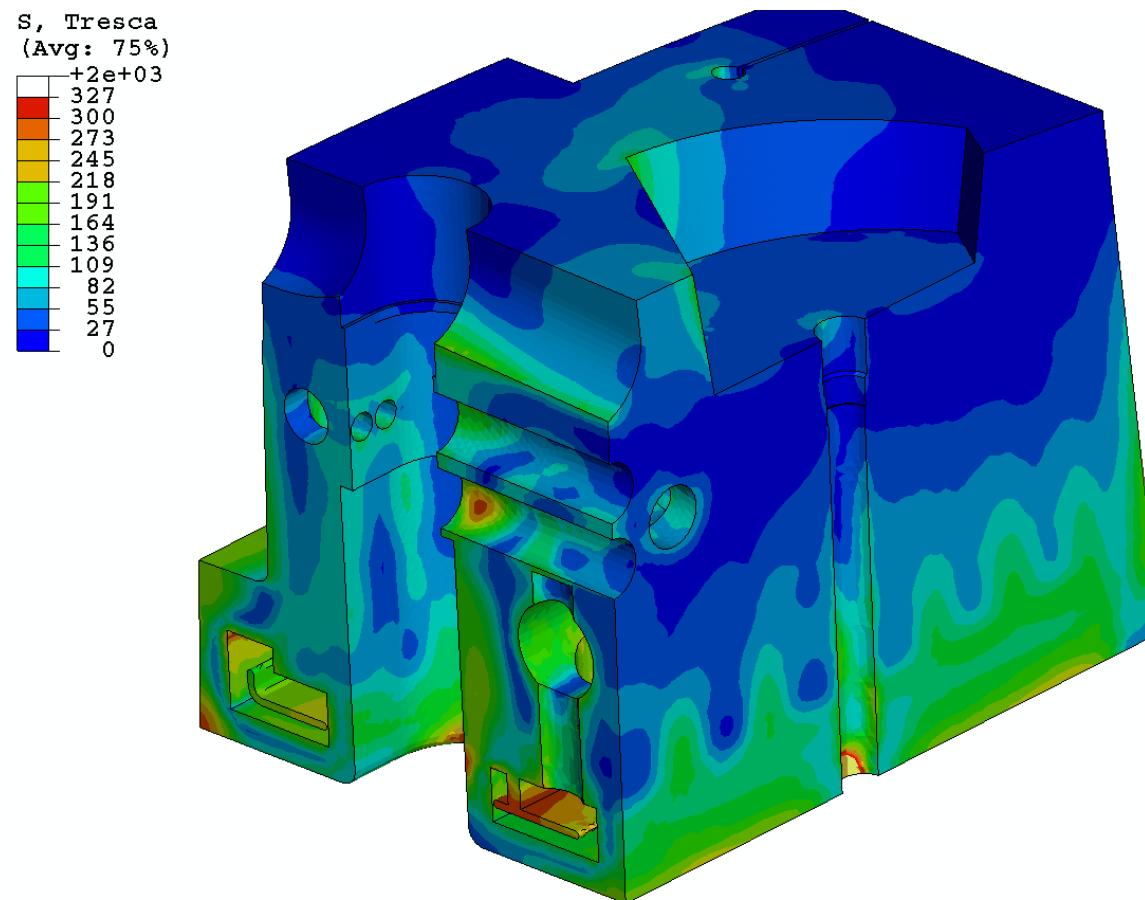
NT11

341
322
304
285
267
249
230
212
194
175
157
138
120



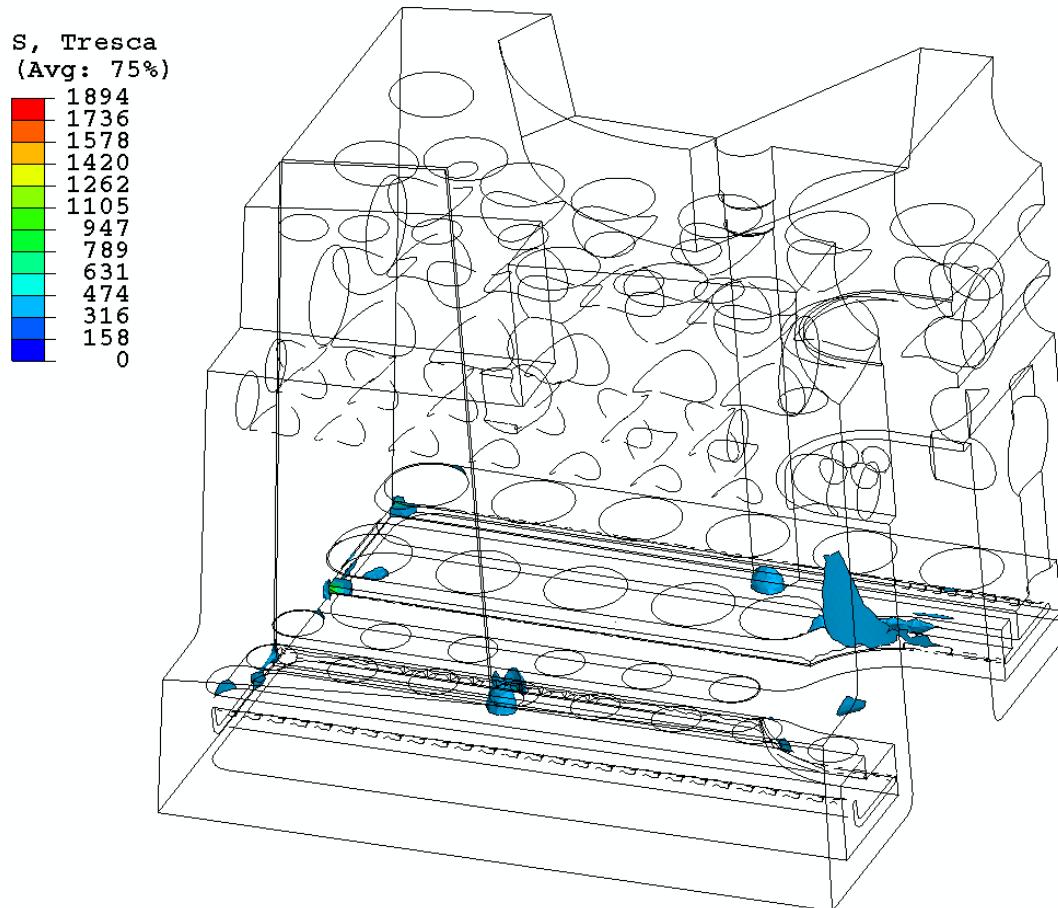
FEA Thermal Stress Results for Shield Module 13 Corner section

Tresca stress distribution (MPa) with upper
contour limit set to 3Sm

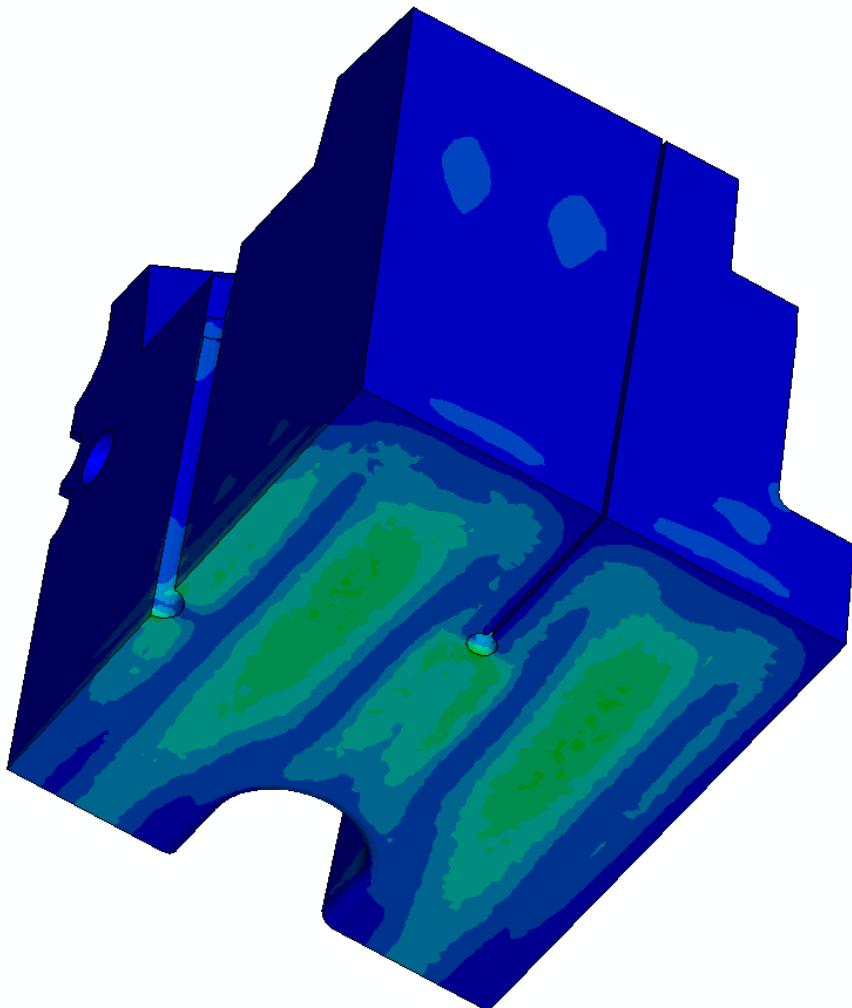


FEA Thermal Stress Results for Shield Module 13 Corner section

Tresca stress results (MPa) showing
regions over 3Sm



FEA Pressure Stress Results for Shield Module 13 Corner section

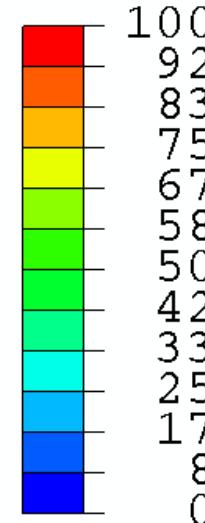


Max stress due to 3MPa coolant
pressure=100MPa, Tresca

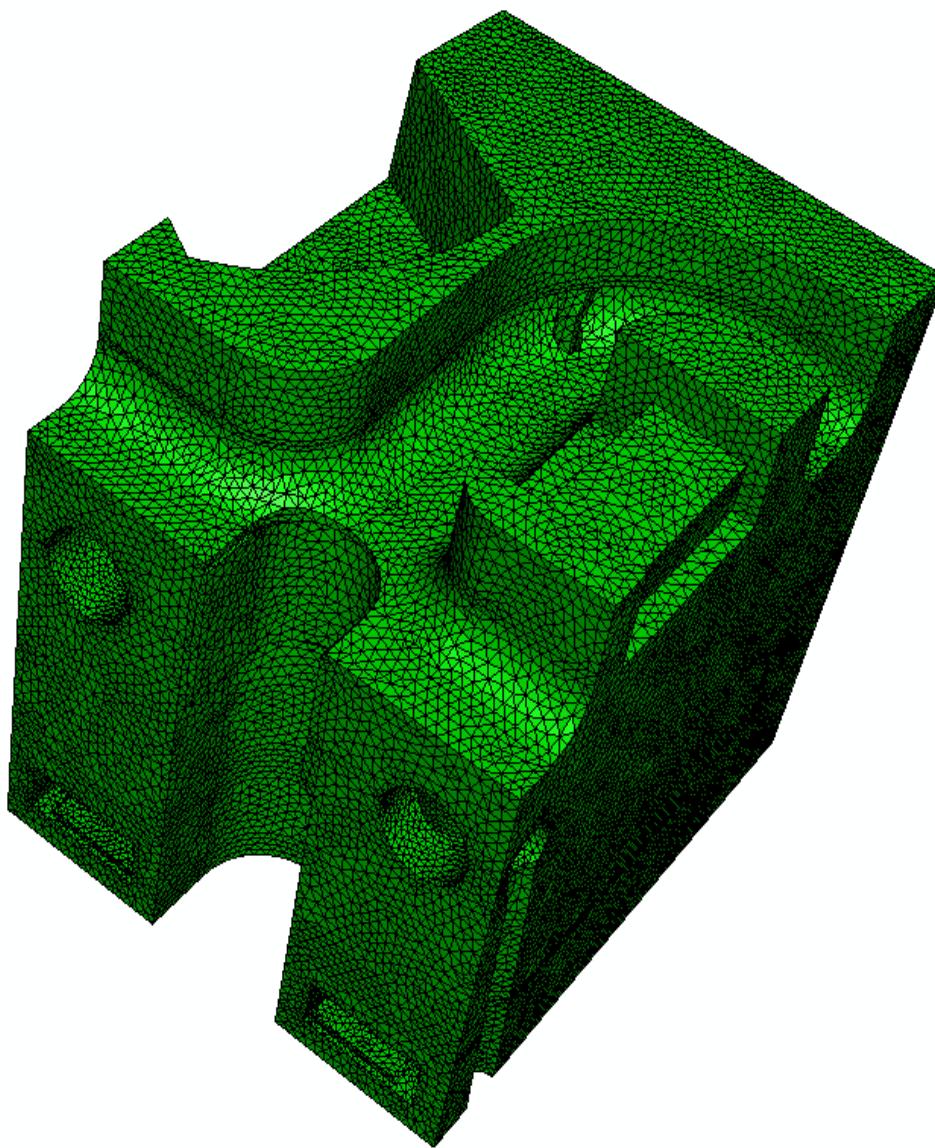
(less than 50MPa away from
stress concentrations and
discontinuities)

$S_m=109\text{ MPa}$

S , Tresca
(Avg: 75%)

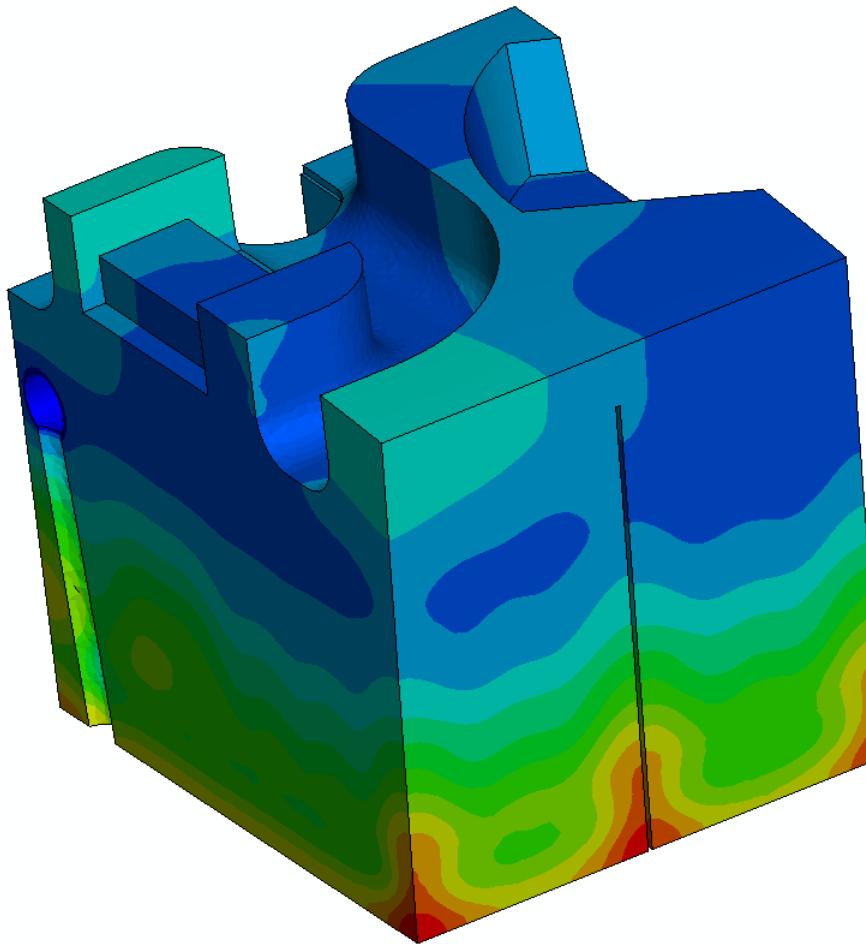
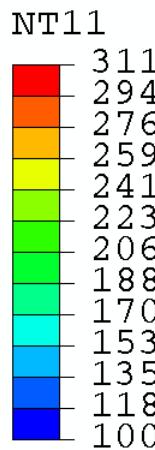


FEA Mesh of Shield Module 13 Mid Section



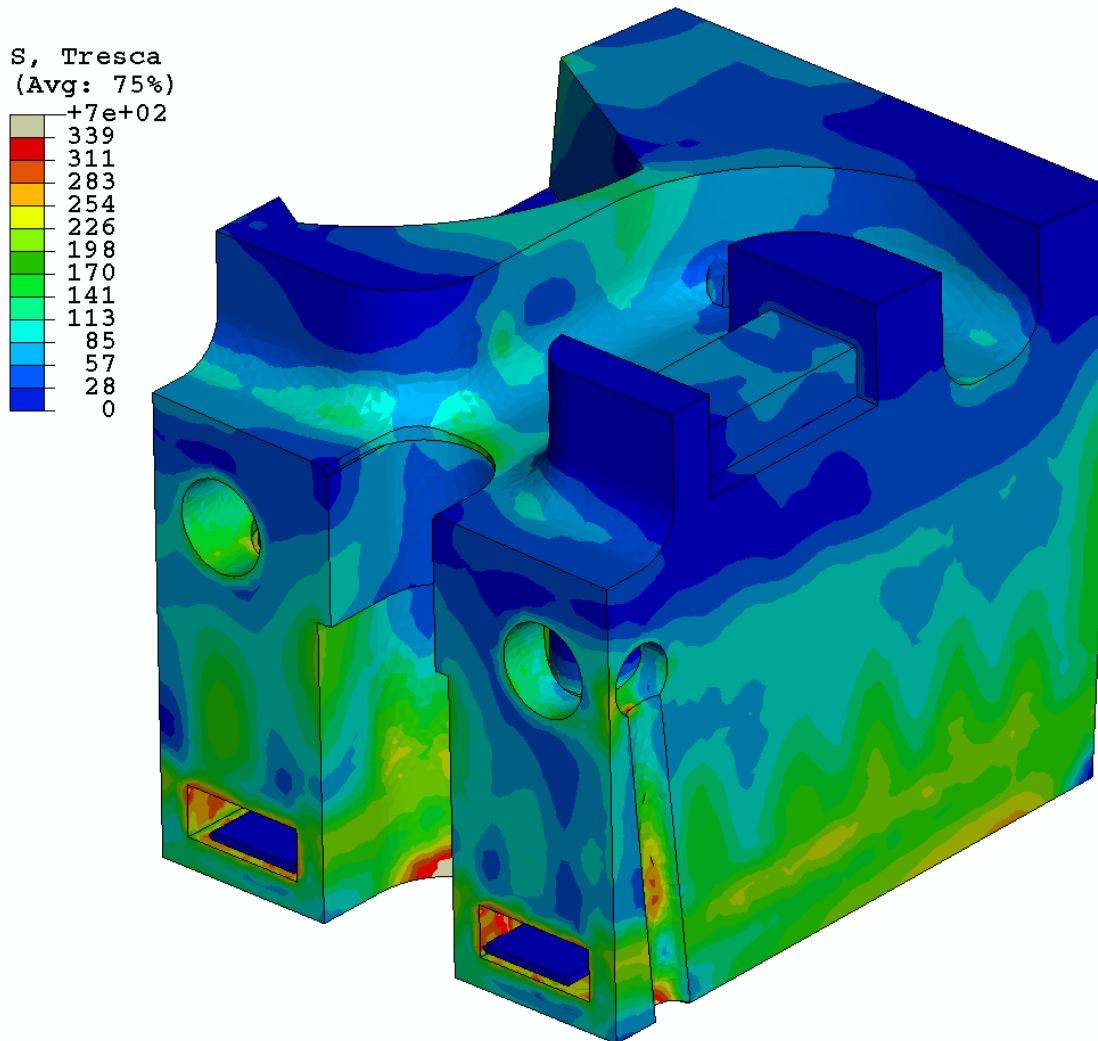
Temperature Distribution for Shield Module 13

Mid Section °C



FEA Thermal Stress Results for Shield Module 13 Mid Section

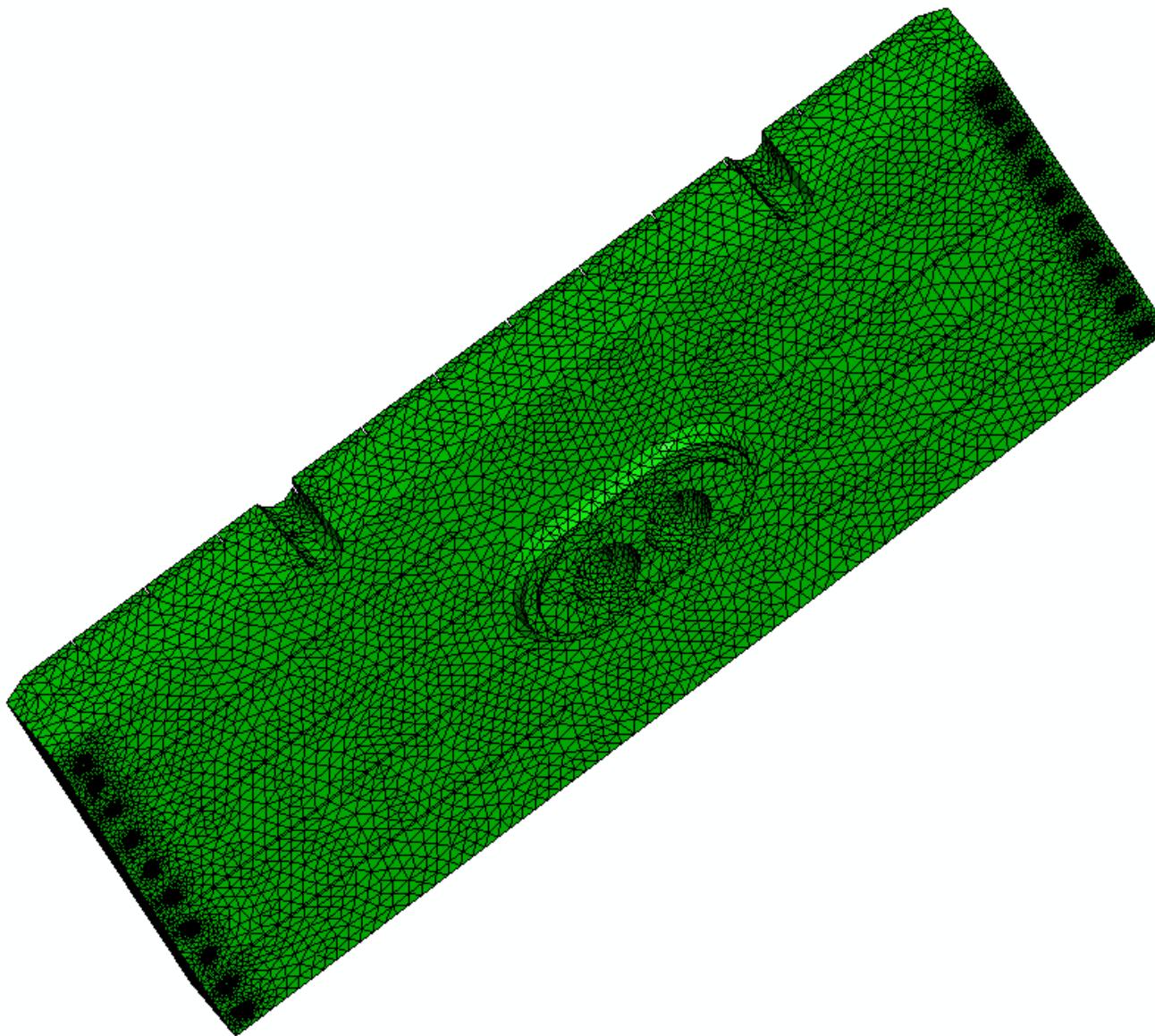
Tresca stress distribution (MPa) with upper
contour limit set to 3Sm



Thermal Stress FEA Of First Walls

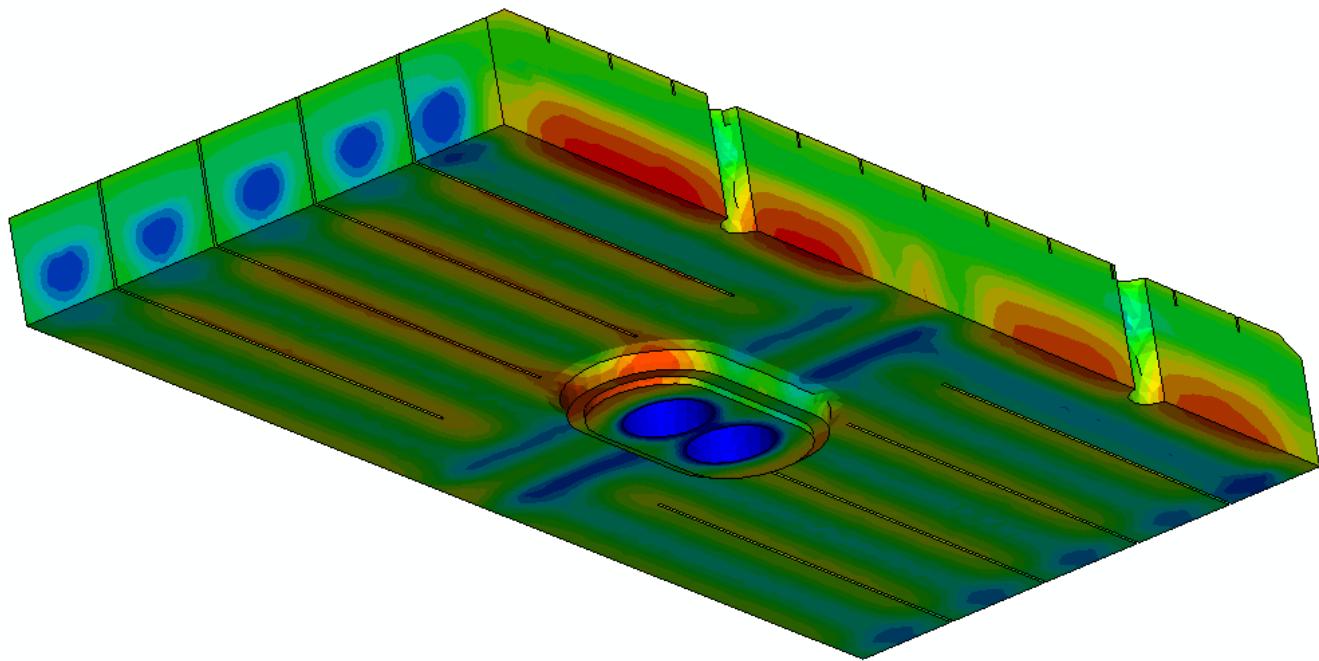
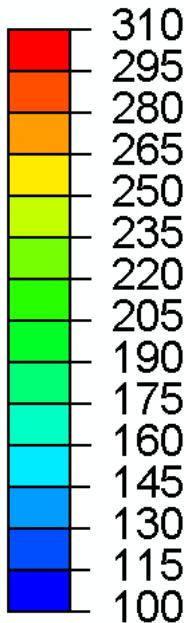
- Tetrahedral meshes created in CFDesign
- Temperature distribution taken from CFDesign results
- Thermal stresses based on initial temperature of 100°C
- Used “Stabilize”- no other boundary conditions
- ABAQUS used for elastic static FEA

FEA Mesh of Module 13 First Wall

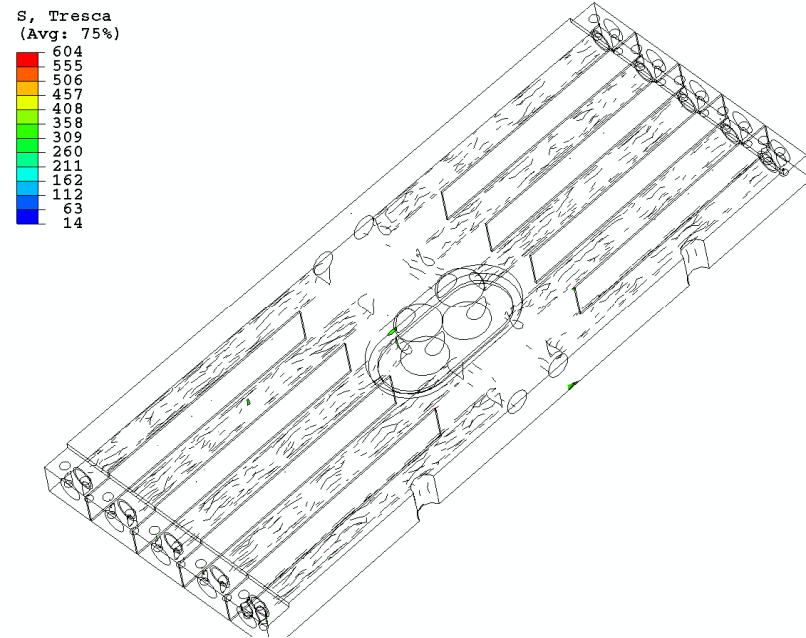
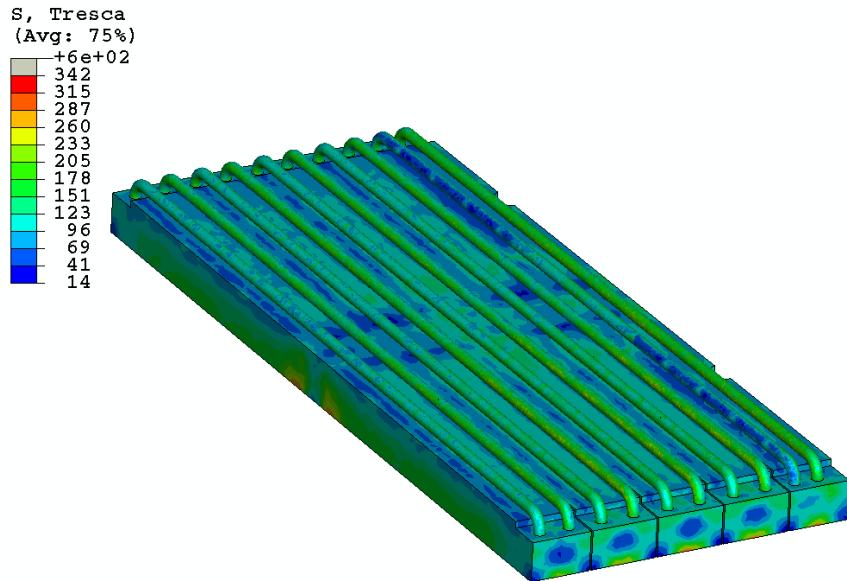


Temperature Distribution In Module 13 First Wall

NT11



Module 13 First Wall Thermal Stress Analysis Results in SS316L(N)-IG

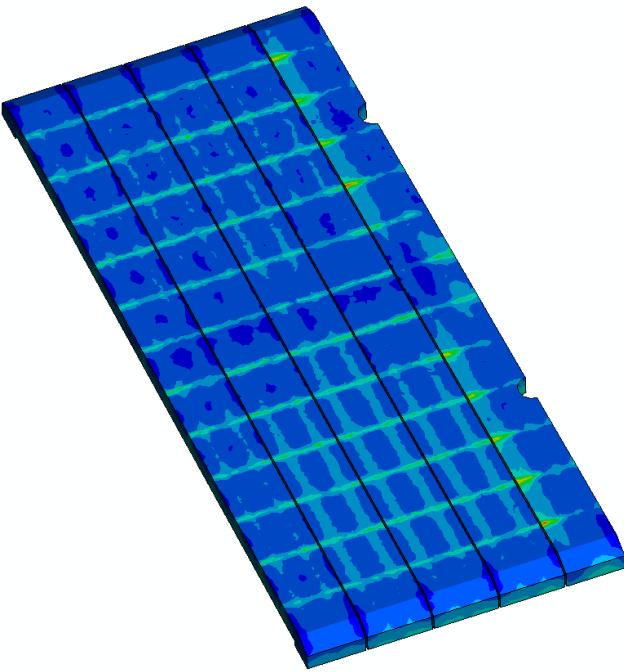


Tresca stress distribution (MPa) with upper contour limit set to 3Sm

Tresca stress results (MPa) showing regions over 3Sm

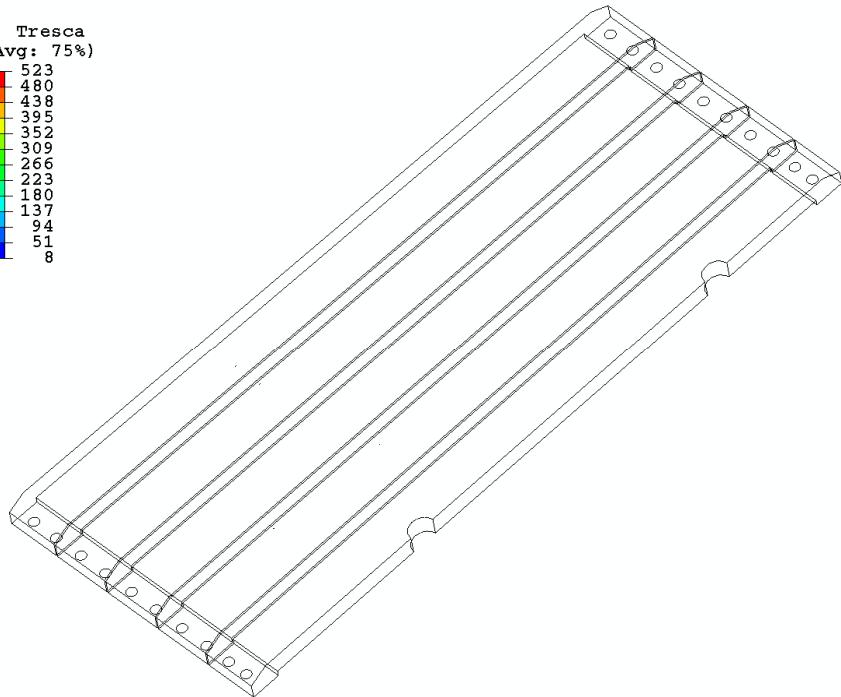
Module 13 First Wall Thermal Stress Analysis Results in CuCrZr

S , Tresca
 (Avg: 75%)
 $+5e+02$
 312
 287
 261
 236
 211
 186
 160
 135
 110
 84
 59
 34
 8



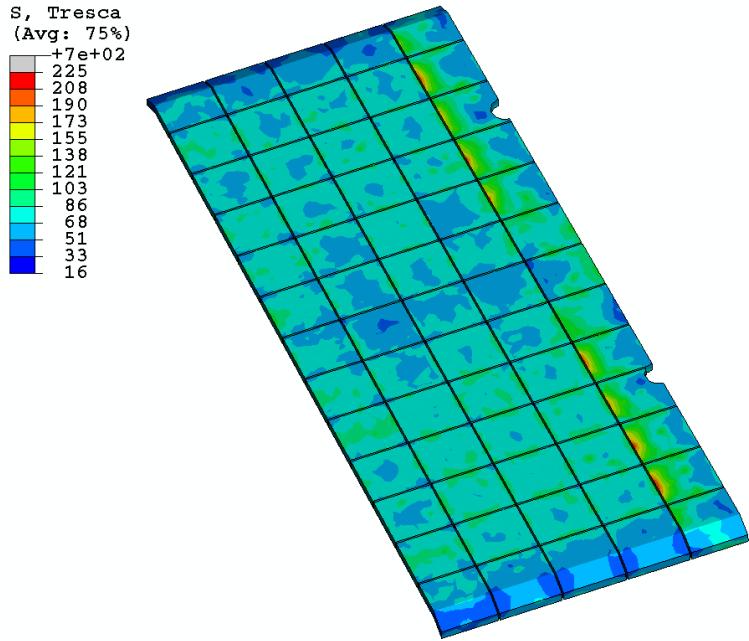
Tresca stress distribution (MPa)
 with upper contour limit set to
 $3S_m$

S , Tresca
 (Avg: 75%)
 523
 480
 438
 395
 352
 309
 266
 223
 180
 137
 94
 51
 8

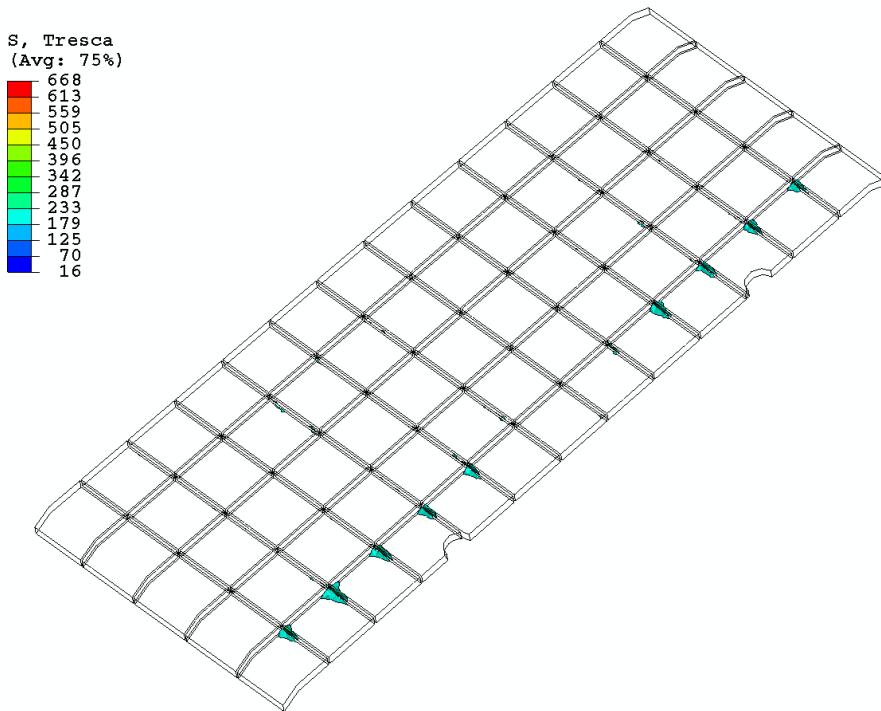


Tresca stress results (MPa)
 showing regions over $3S_m$

Module 13 First Wall Thermal Stress Analysis Results in Be(S65C)



**Tresca stress distribution (MPa)
with upper contour limit set to
3Sm**



**Tresca stress results (MPa)
showing regions over 3Sm**

Conclusion

- FEA Stress results were successfully produced based on nuclear heating calculations and the corresponding CFD thermal and pressure results.
- Stress levels in module 13 were shown to be generally under the limits set by ITER IO for elastic analysis of components during operating conditions.

Next Steps

- **Analyze larger sections of modules**
 - Better fit for symmetry boundary conditions
- **Optimize meshes for mechanical stress analyses**
 - Mesh refinement near smaller geometric features
- **Include electromagnetic forces**