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Title: The Bipolar Behavior in Richtmyer-Meshkov Instability (U)

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THE BIPOLAR BEHAVIOR IN RICHTMYER-MESHKOV INSTABILITY (U)

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Abstract

A numerical study of the evolution of the multimode planar Richtmyer-Meshkov instability (RMI) in a light-heavy (Air-SF₆, Atwood number $A=0.67$) configuration involving a Mach number $Ma=1.5$ shock is carried out. Our results demonstrate that the initial material interface morphology controls the evolution characteristics of RMI, and provide a significant basis to develop metrics for transition to turbulence. Depending on the initial *rms* slope of the interface, RMI evolves into linear or nonlinear (mode-coupled) regimes, with distinctly different flow features and growth rates, turbulence statistics and material mixing rates. *Some of our findings are not consistent with heuristic notions of mixing in equilibrium turbulence.* The more turbulent the flow – as measured by spectral bandwidth – the higher the material mixing and, paradoxically, the smaller the turbulent kinetic energy and the slower the growth of the mixing layer. On the other hand, the least turbulent flow has more turbulent kinetic energy and higher mixing layer growth rate, with less material mixing.

The bipolar behavior in Richtmyer Meshkov Instability (RMI)



Akshay A. Gowardhan

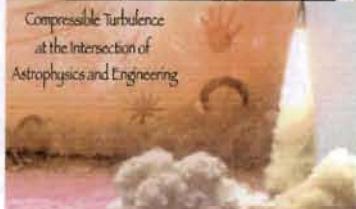
J. Ray Ristorcelli

Fernando F. Grinstein

Los Alamos National Laboratory

Compressible Turbulence at the Intersection of
Astrophysics and Engineering

April 25, 2011 - April 28, 2011, Santa Fe, NM



Outline

- **Motivation**
- **Setup & Simulation**
- **Results**
 - mixing width
 - turbulence metrics: k , w^2 , Re_t
- **Conclusion**
- **Future work**

Motivation



LDRD-DR "*Turbulence by Design*"

Effect of initial condition on late time turbulence

PI: Malcolm Andrews

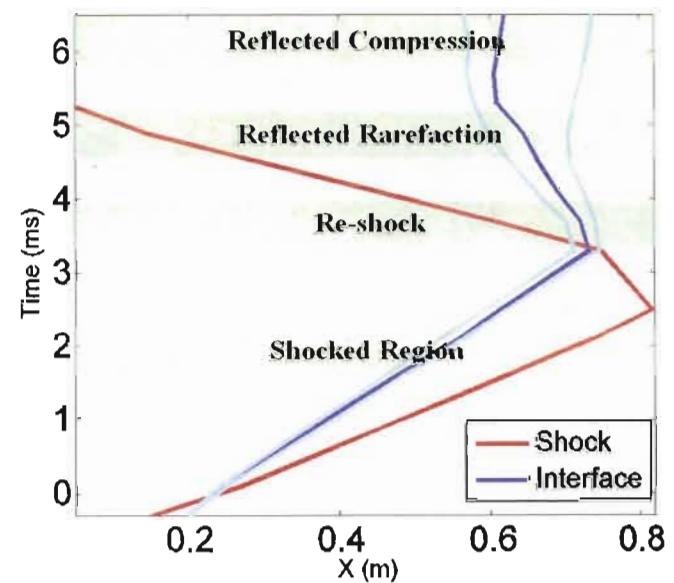
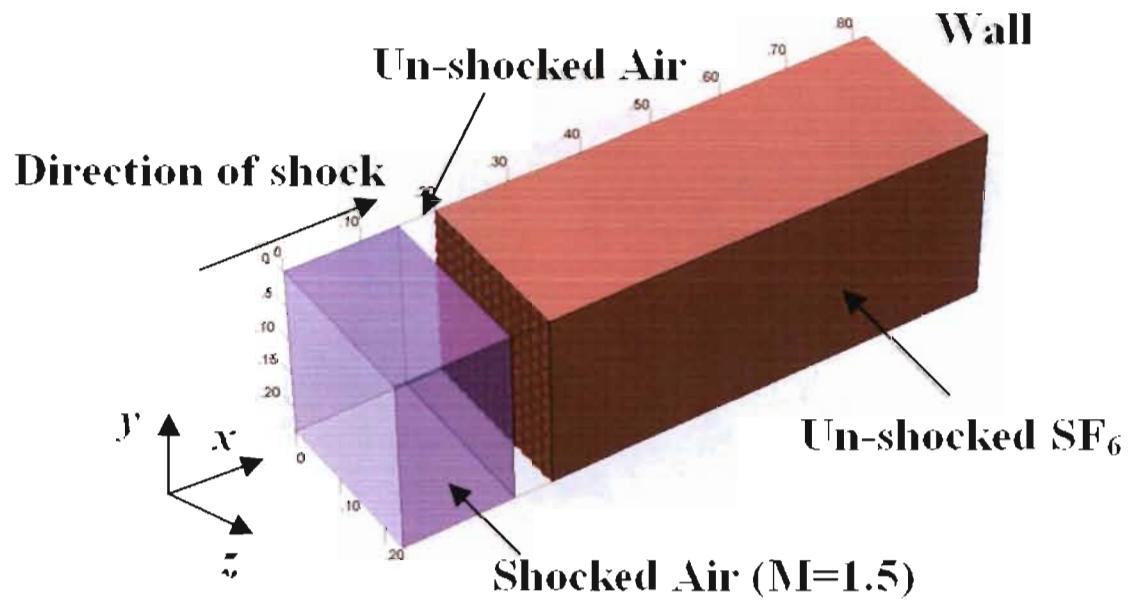
- **How does interfacial morphology control the evolution of RMI?**
- **Can turbulence transition be achieved with a single shock?**
- **Is there a controlling parameter in RMI?**

Point in scientific process

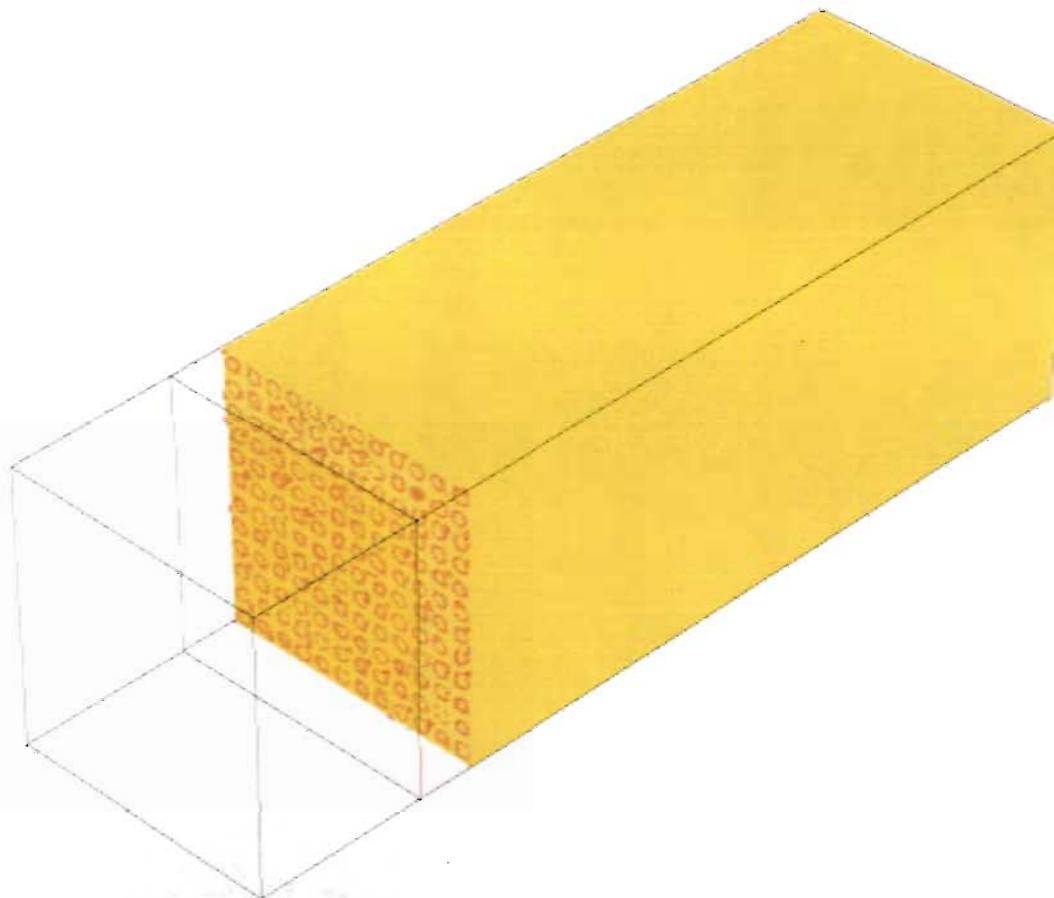


- 1. Data (numerical) collection**
- 2. Observation**
- 3. Hypothesis formulation**
- 4. Hypothesis verification**

Planar Richtmyer Meshkov Instability



Planar Richtmyer Meshkov Instability



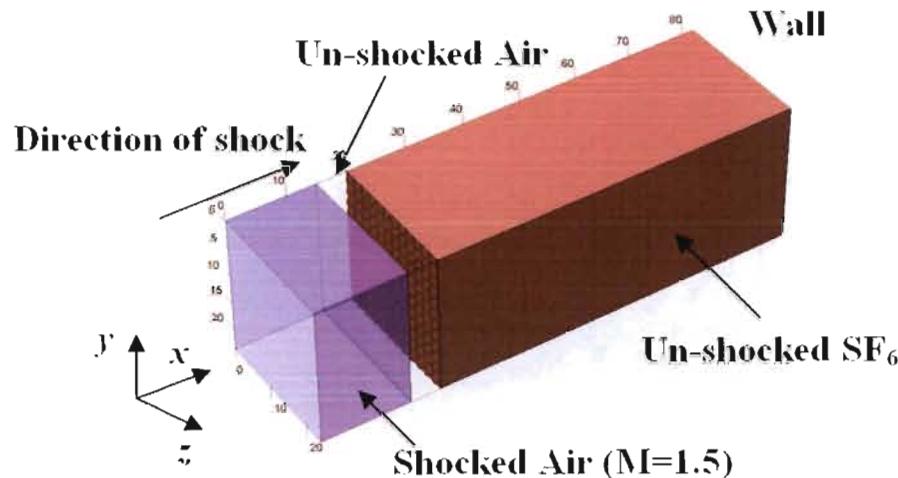
LANL's RAGE

Gittings et al. LA-UR Report 2006



- **solves multimaterial compressible-Euler equations**
- **high resolution (2nd order) Godunov scheme**
- **ideal gas equation of state**
- **adaptive mesh refinement capability**
- **gradient terms (limiters) and interface treatment.**
 - **for the gradient:** min-mod (MM) ; **Van Leer (VL)**
 - **for the interface treatment:**
 - **(no IP) – monotonized centered VL limiter.**
 - **(IP) interface preserver, width ~3 cells.**

Problem Setup & Validation



3D RAGE simulation details

Domain Dimension: 82cm x 24cm x 24cm

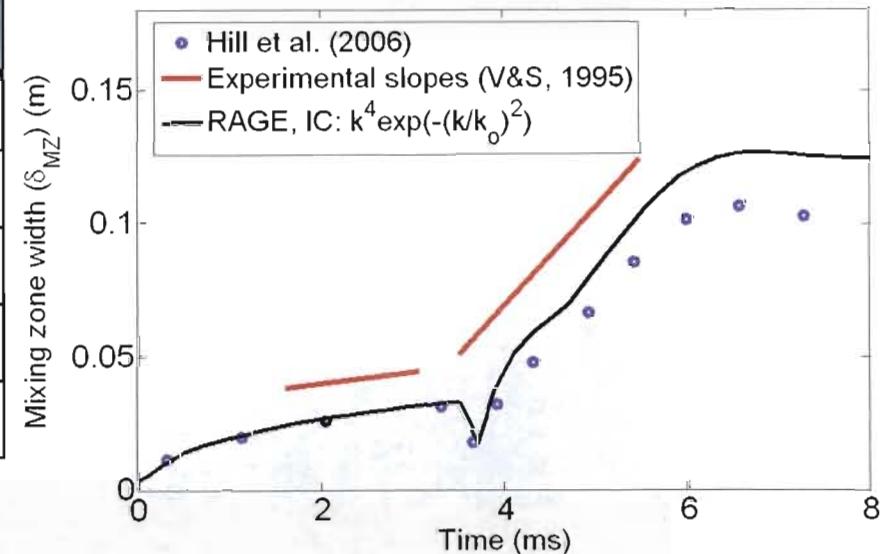
Mach number: 1.5

Simulations with and without reshock performed

Grid Resolution: $dx=dy=dz = 0.1\text{cm}$ (1 level AMR)

Periodic B.C. in y & z dir., inflow-wall in x dir.

Validation



Does interfacial morphology control the evolution of RMI?

RMS Slope of the interface $\kappa_o \delta_o$ or η_o

if $X_s(y,z)$ is the interface, then

$$\eta_o = \kappa_o \delta_o = \langle \nabla X_s \nabla X_s \rangle^{1/2}$$

$$\delta$$

$\lambda_o = 2\pi/\kappa_o$

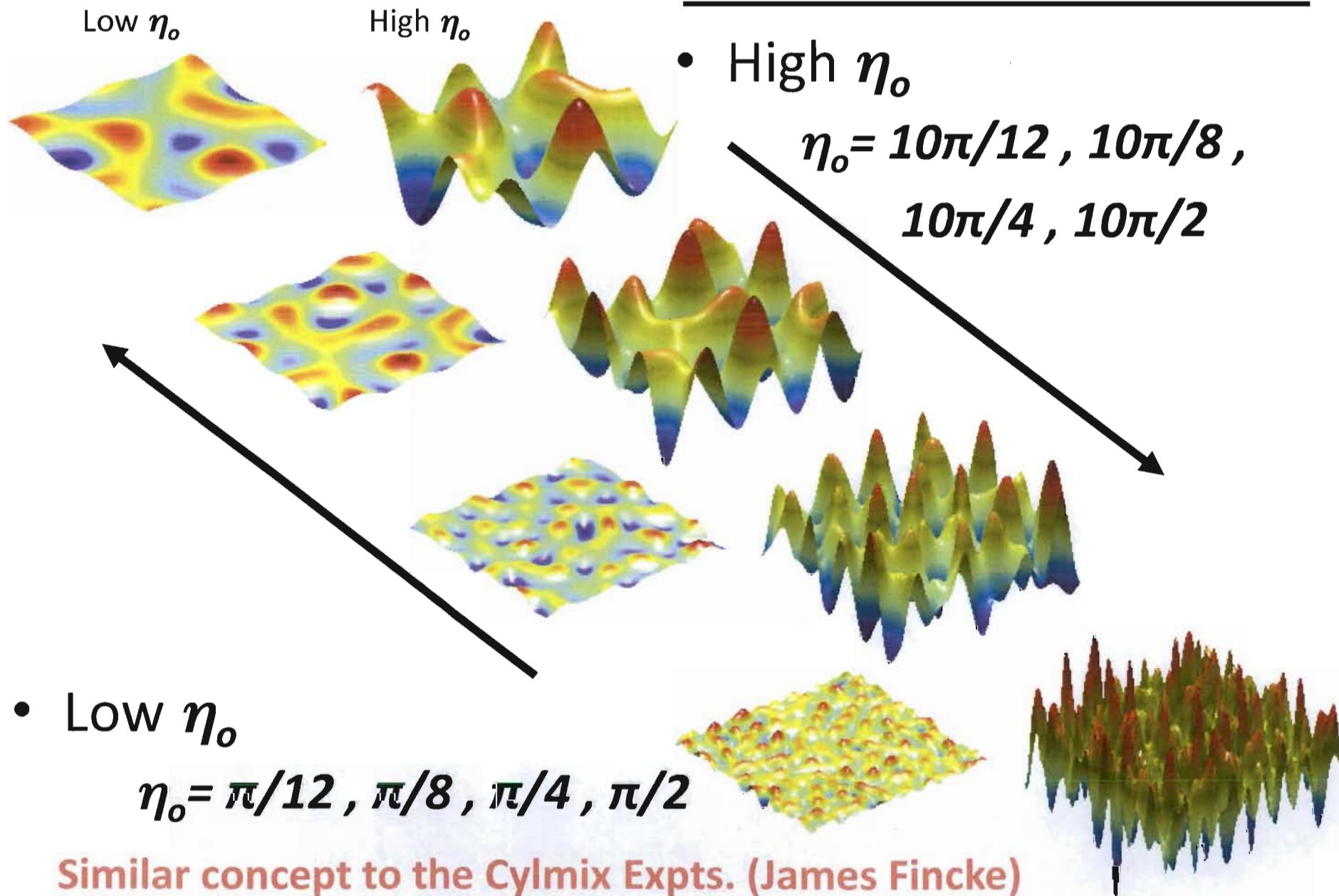


Low η_o



High η_o

Initial interface



Planar RMI: Low η_o

$$\langle f \rangle(x) = \frac{1}{A} \int f(x, y, z) dy dz, \quad A = \int dy dz,$$

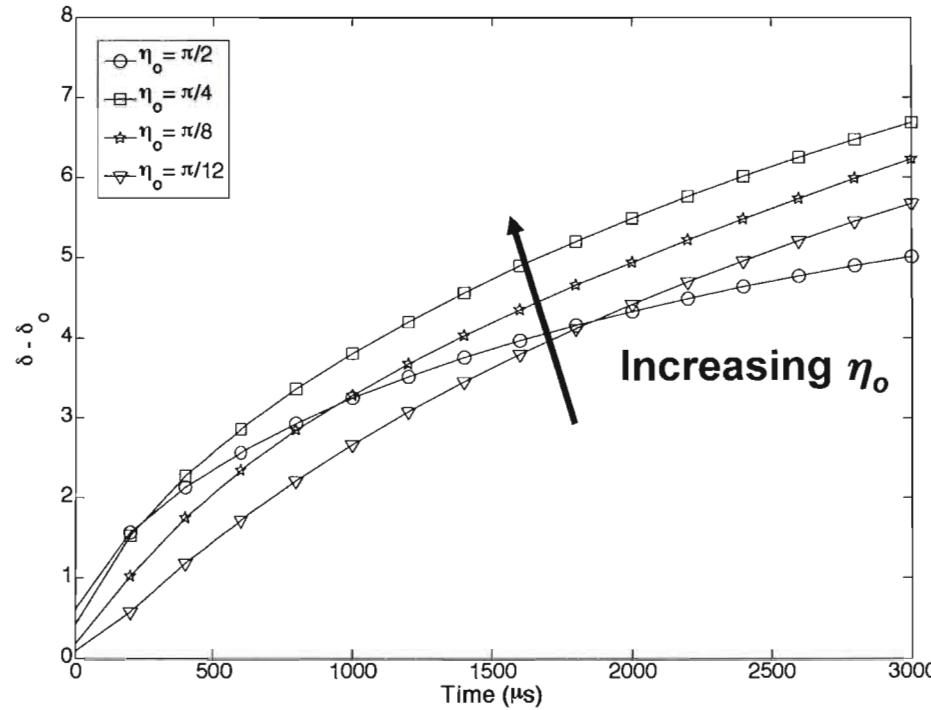
$$Y_{SF_6} = \rho_{SF_6} / \rho,$$

$$\theta = 4Y_{SF_6} [1 - Y_{SF_6}],$$

$$\psi(x) = \langle Y_{SF_6} \rangle,$$

$$M(x) = 4\psi(x)[1 - \psi(x)],$$

$$\delta_{MZ} = \int M(x) dx,$$



Richtmyer 1960
theory for η_o

$$V = A \Delta U \eta_o$$

$$\delta = A \Delta U \eta_o t + \delta_o$$

$$\delta - \delta_o = A \Delta U \eta_o t$$

For relatively low value of η_o , the Richtmyer's theory (growth is proportional to κ_o) is valid, however for the case $\eta_o = \pi/2$, the Richtmyer's theory is valid for a very short time after the first shock and soon the growth rate drops.

Consistent with Classical RMI theory!

Planar RMI: High η_o

$$\langle f \rangle(x) = \frac{1}{A} \int f(x, y, z) dy dz, \quad A = \int dy dz,$$

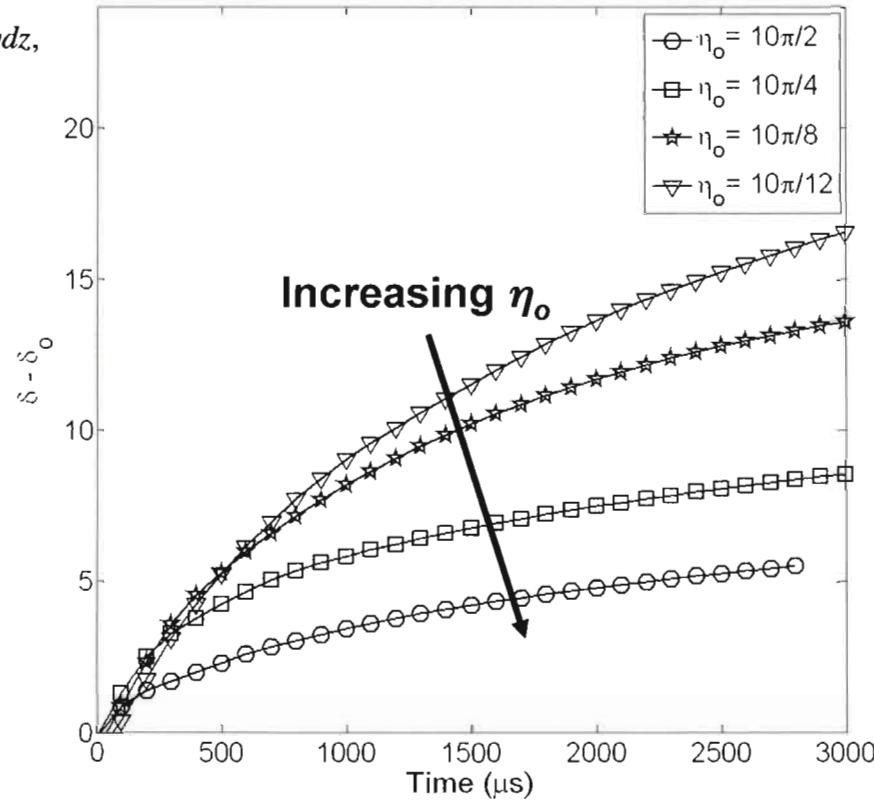
$$Y_{SF_6} = \rho_{SF_6} / \rho,$$

$$\theta = 4Y_{SF_6} [1 - Y_{SF_6}],$$

$$\psi(x) = \langle Y_{SF_6} \rangle,$$

$$M(x) = 4\psi(x)[1 - \psi(x)],$$

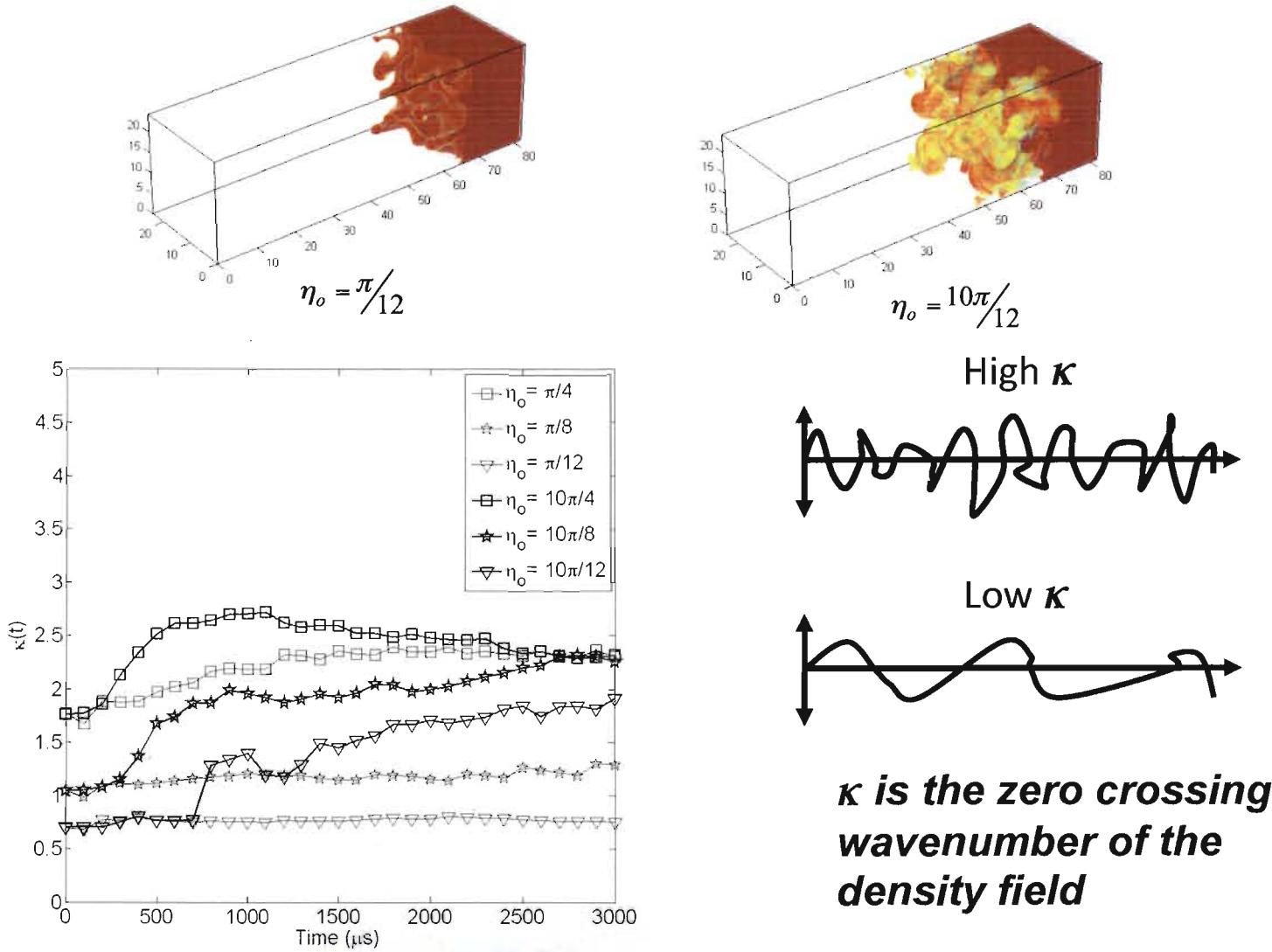
$$\delta_{MZ} = \int M(x) dx,$$



At very short time, for all the cases, the Richtmyer's theory (growth is proportional to κ_o) is valid. Soon after, the growth is actually **INVERSELY** proportional to κ_o .

NOT consistent with Classical RMI theory!

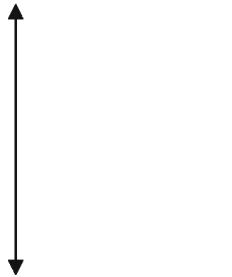
Planar RMI: Effect of η_o



- Higher κ – smaller the small scales, flow is more “turbulent”
- Higher η_o produces more small scales

κ is the zero crossing wavenumber of the density field

Planar RMI: Theory

$$\chi = \frac{\delta}{2} \sin(\kappa_o y)$$


Richtmyer 1960
theory for *low* $\kappa_o \delta$

$$V = A \Delta U \eta_o$$
$$\delta = A \Delta U \eta_o t + \delta_o$$
$$\delta - \delta_o = A \Delta U \eta_o t$$

Case I: Low η_o ($\eta_o \ll \pi$) - Jekyll

- increasing η_o \rightarrow more baroclinic vorticity is deposited \rightarrow higher growth.
- growth is mostly *ballistic* (no mode coupling)
- **Higher the η_o higher the growth**

Case II: High η_o ($\eta_o > \pi$) - Hyde

- increasing η_o , growth saturates due to secondary instabilities ...
- flow becomes more turbulent (non linear, chaotic, mode coupling)
- **Higher the η_o lower the growth**

Planar RMI evolves into different class of flows

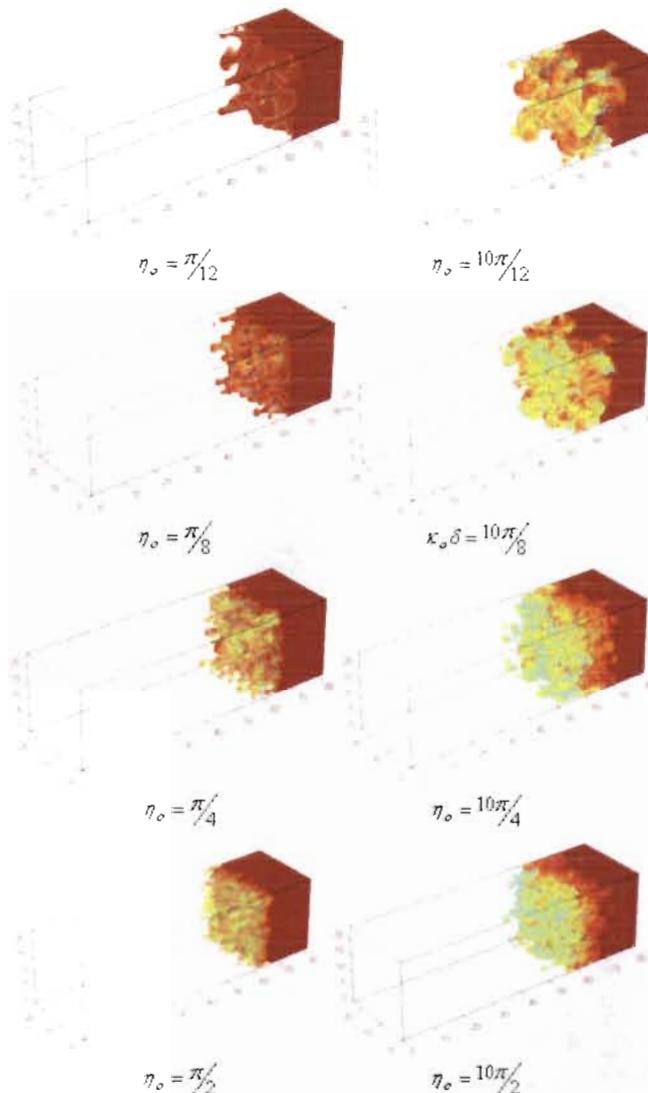
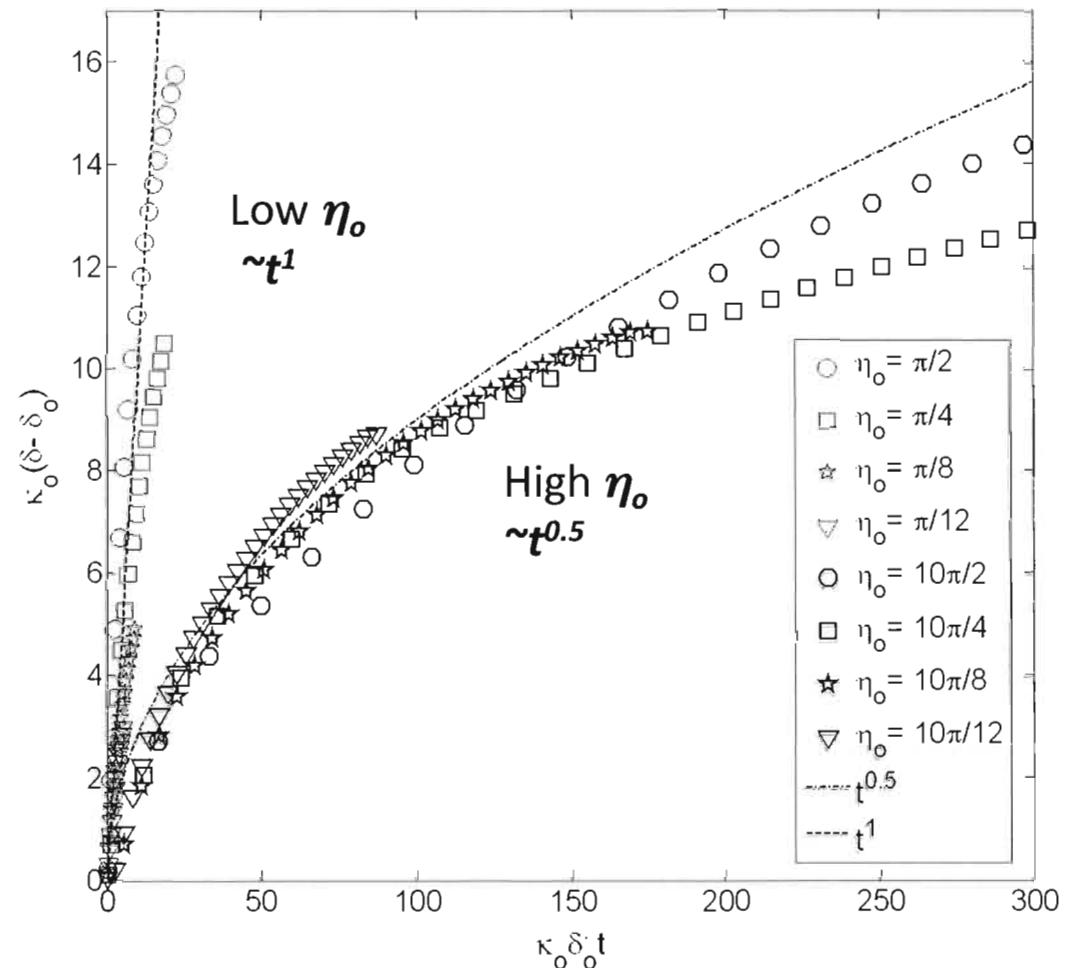


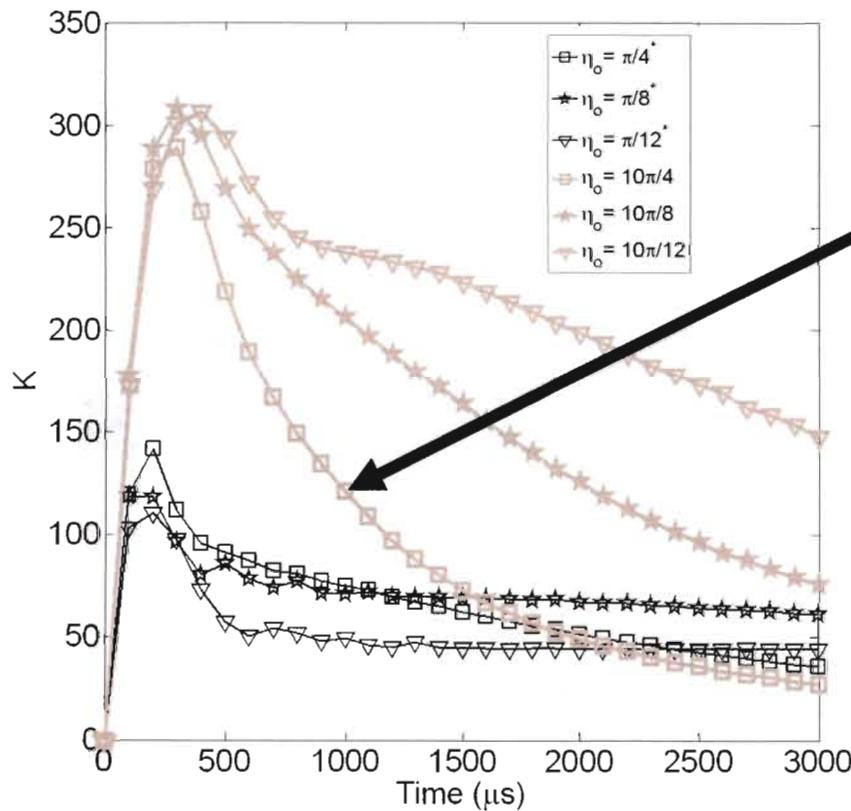
Figure 4. Volume visualization of Y_{SP6} at time $t = 3000 \mu s$ after the first shock.



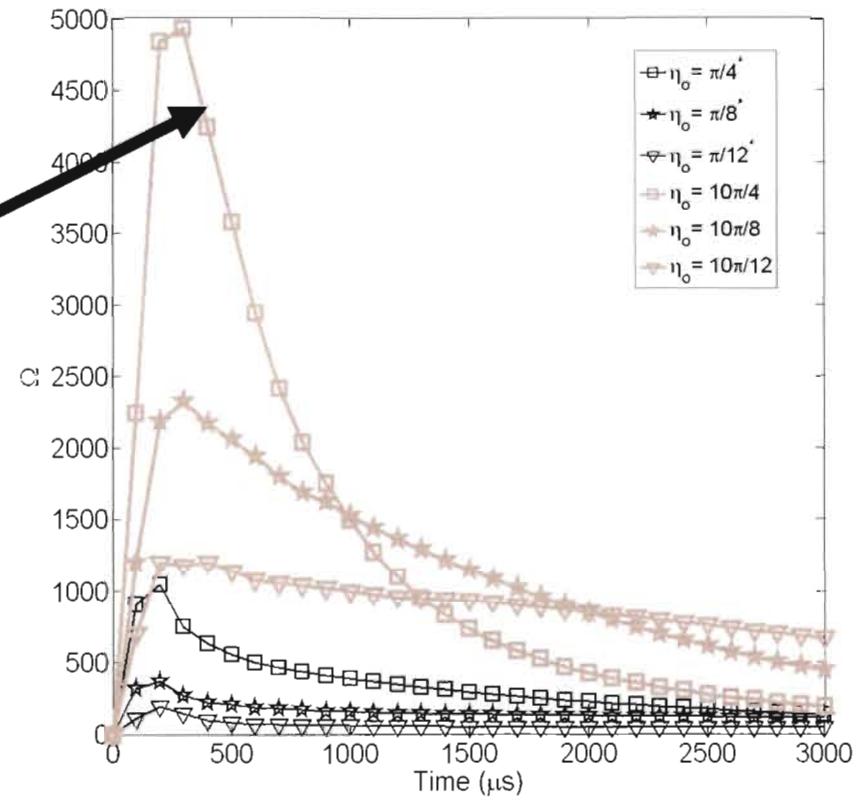
Low η_o : *Linear*
High η_o : *Non-linear*

Turbulence metrics....

Turbulent kinetic energy



Enstrophy

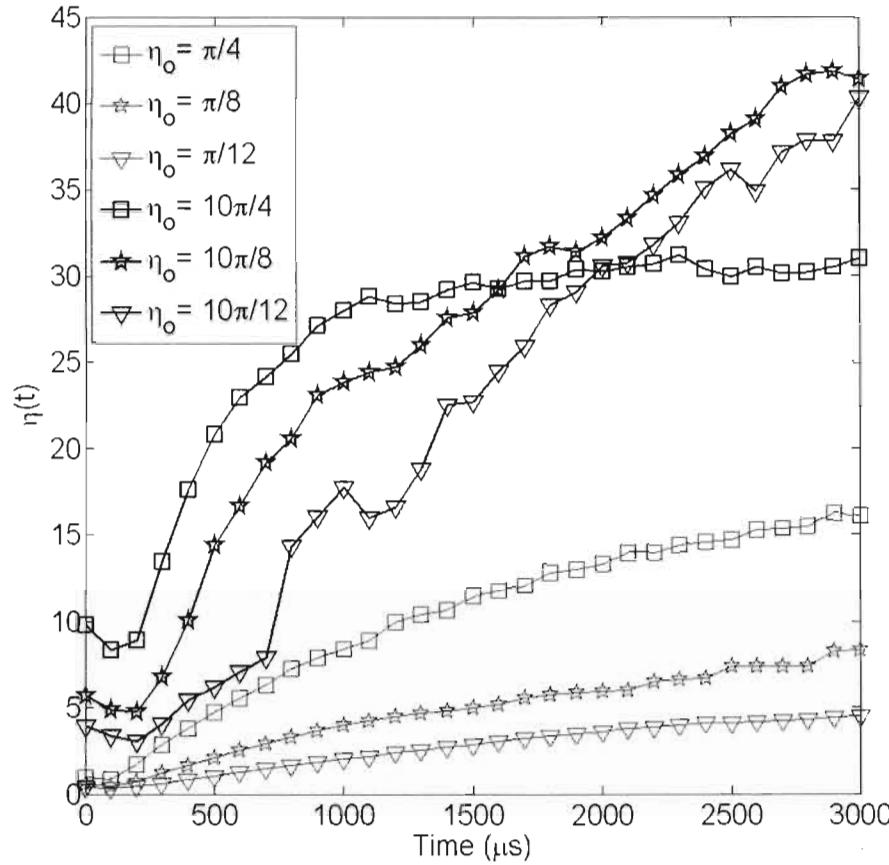


more enstrophy - less energy

“Spectral Bandwidth”

$$\eta_o = \frac{\delta}{\lambda} = \frac{\text{Integral lengthscale}}{\text{Taylor lengthscale}}$$

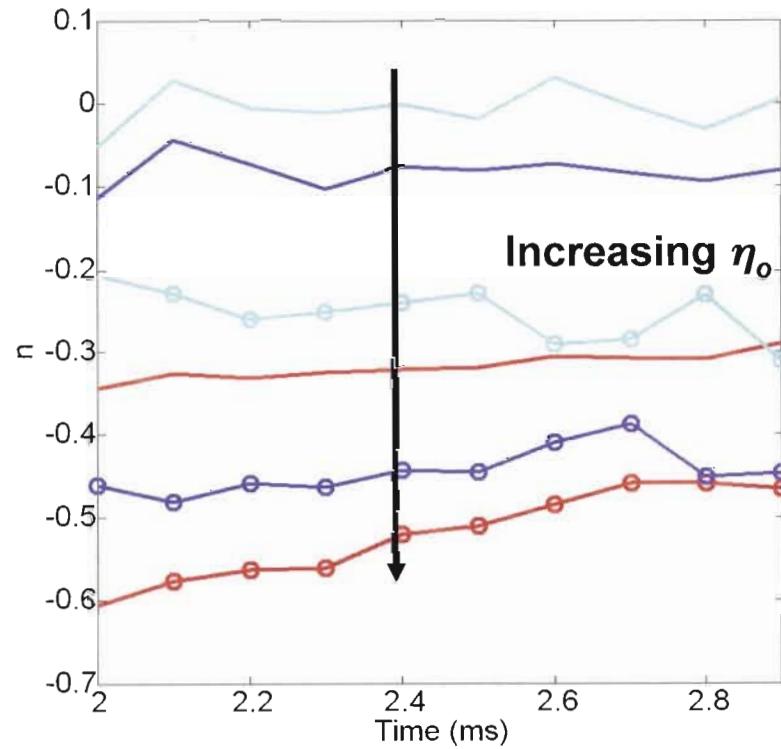
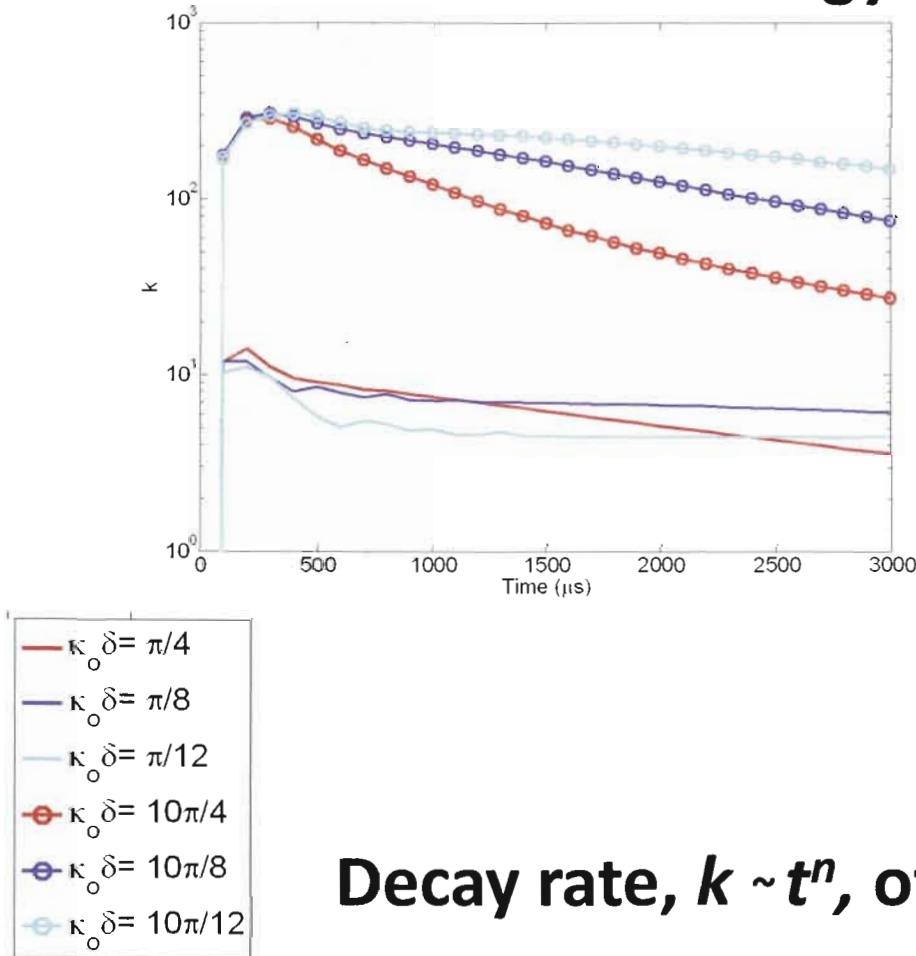
Spectra bandwidth can be thought of as a measure of how turbulent the turbulence is, loosely!



High initial η_o leads to high value at late times, until the decay is far enough along...

Rate Decay of TKE

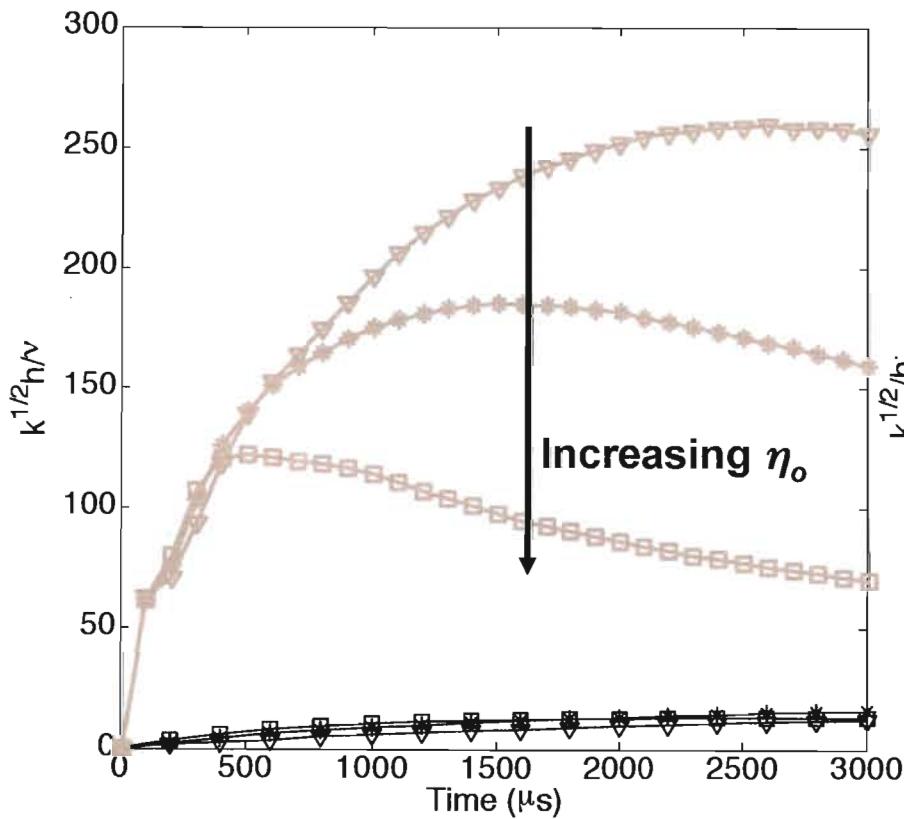
Turbulent kinetic energy



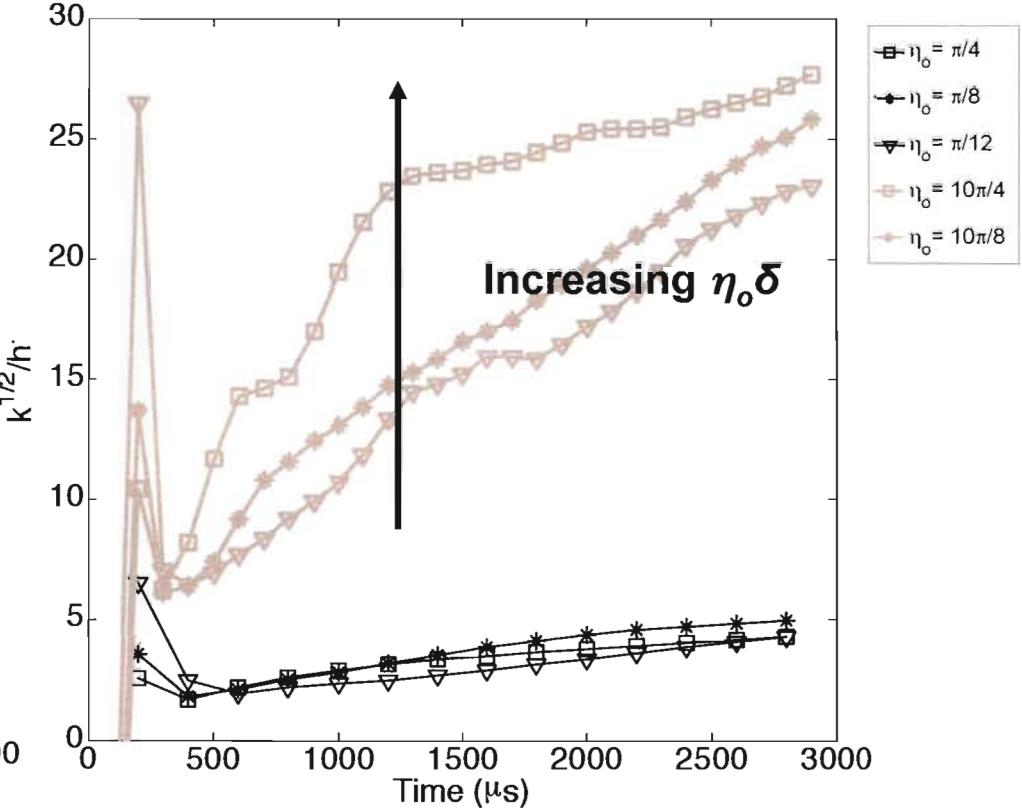
Decay rate, $\kappa \sim t^n$, of TKE depends on $\kappa_o \delta$

Turbulence metrics....

Turbulent Re



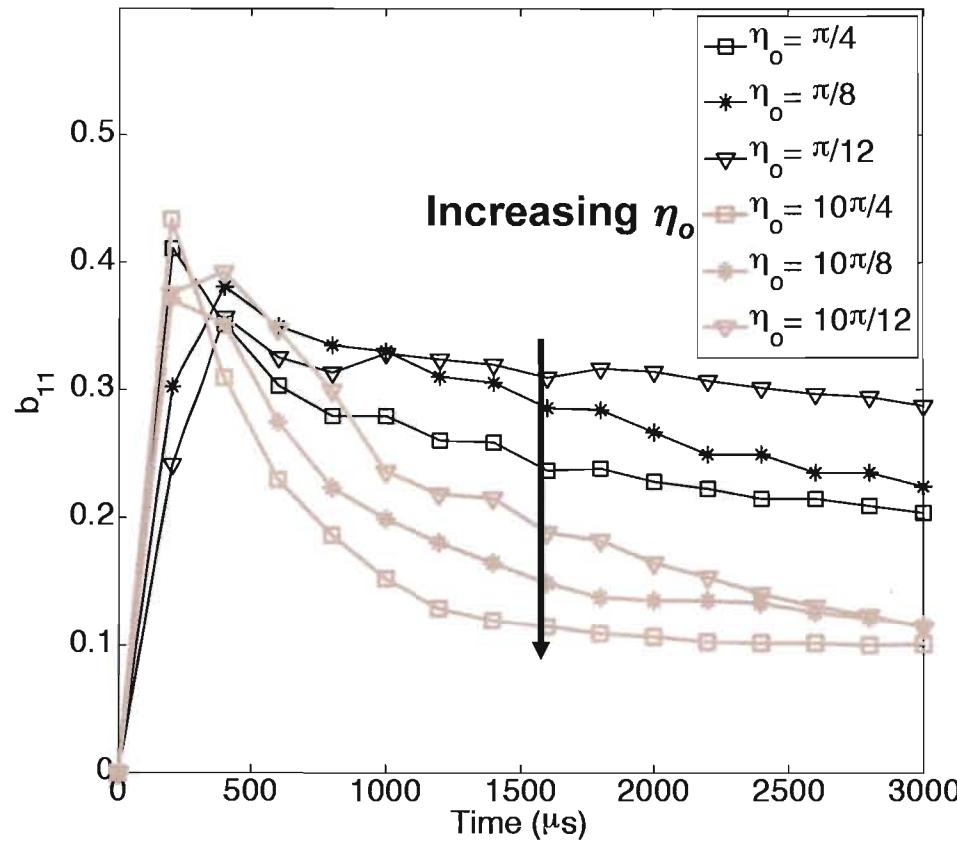
Turbulence intensity



Higher η_0 – smaller Turbulent Re number, BUT higher turbulent intensity

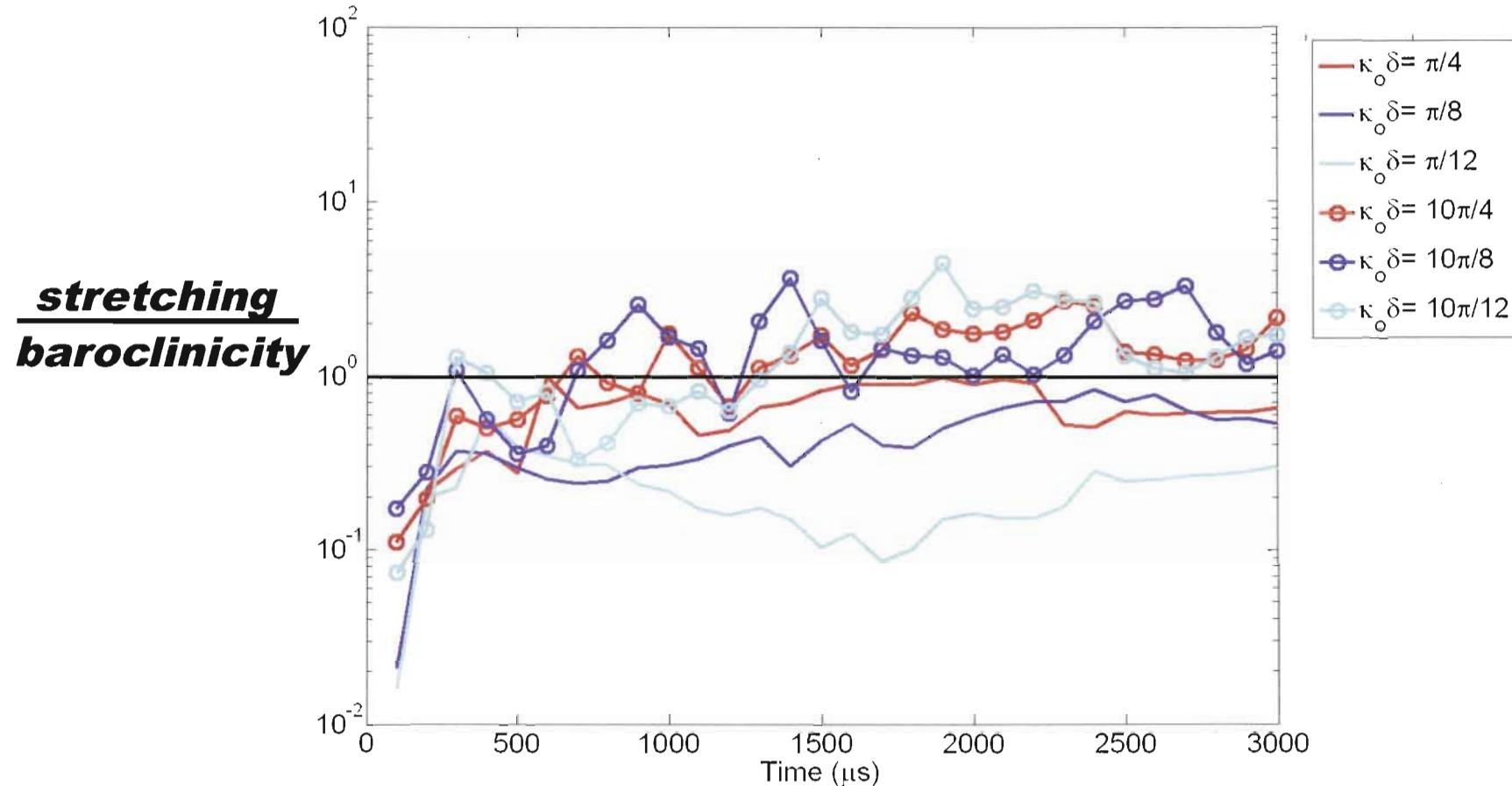
Turbulence metrics....

Anisotropy $b_{11} = \frac{\langle u_1 u_1 \rangle}{2k} - \frac{1}{3}$



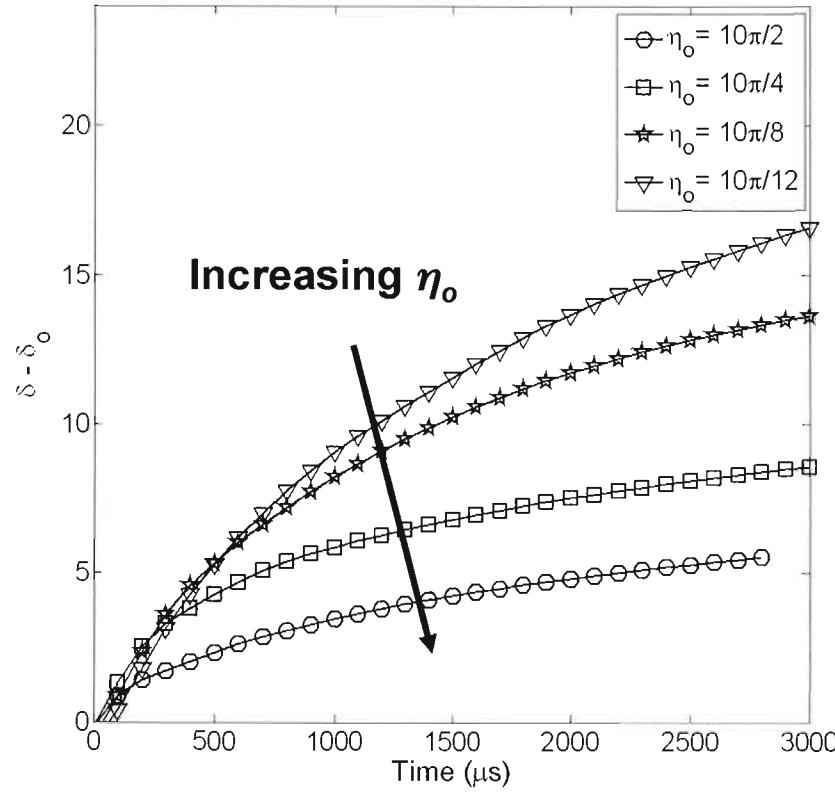
Higher η_o – more mixing, more isotropy

Ratio of enstrophy generation mechanisms

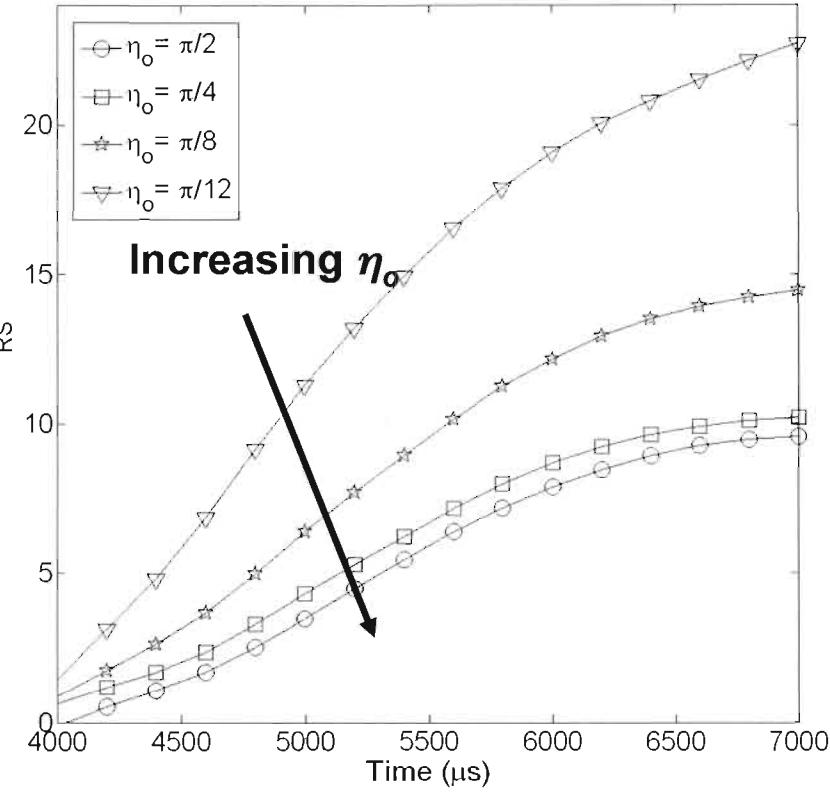


- Baroclinicity is important at all times for enstrophy production
- Higher η_o produces more enstrophy by “stretching”
- Baroclinicity is much higher initially

Can turbulence transition be achieved with a single shock?



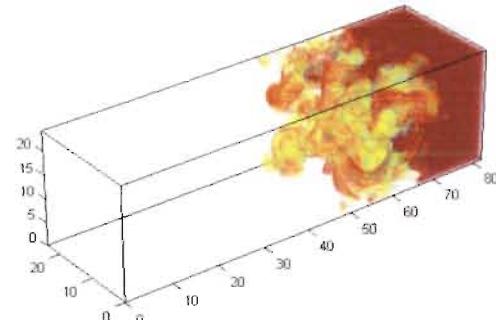
High η_0 before reshock



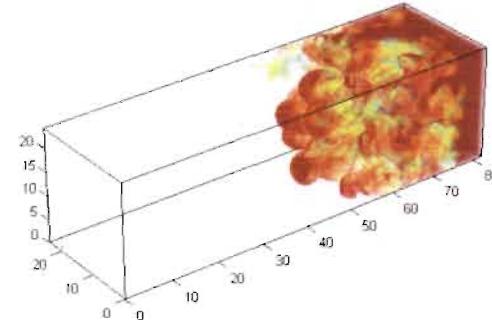
Low η_0 after reshock

Shocked high η_0 case behaves like reshocked low η_0 case

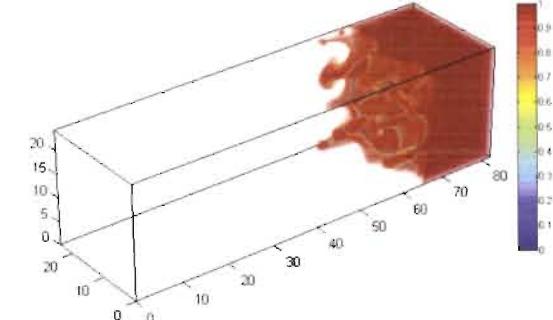
Can turbulence transition be achieved with a single shock?



High η_0 shocked

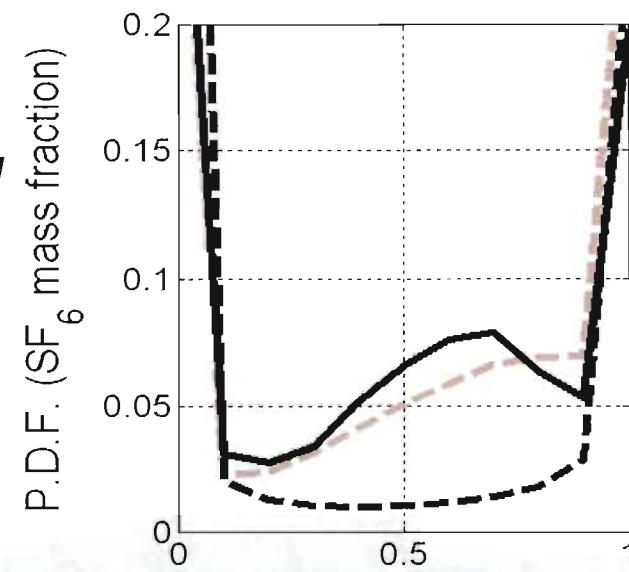


Low η_0 reshocked



Low η_0 shocked

- Low η_0 shocked
- - - Low η_0 reshocked
- High η_0 shocked



Shocked high η_0 case behaves like reshocked low η_0 case

Questions answered



- Does interfacial morphology control the evolution of RMI?
 - Yes
- Can turbulence transition be achieved with a single shock?
 - Yes
- Is there a controlling parameter in RMI?
 - $\kappa_o \delta$

Conclusion



- RMS slope η_o of the interface affects the evolution of RMI
- Reshock effect can be achieved with single shock, if $\eta_o > 1$.
- Higher η_o behaves like the reshock problem
- Higher η_o leads to faster dissipation of TKE because of presences on more small scale structure

Counter-intuitive results



- In non-linear regime, higher η_o leads to more material mixing, has smaller scales and is more isotropic

BUT

1. Lower TKE
2. Lower Re_T
3. Smaller mix width

Future work



- Study the effect of Mach number and Atwood number
- Identify scaling parameters ($\kappa_o \delta_o$ or $\kappa_o^\alpha \delta_o^\alpha$?)
- Enstrophy and TKE budget
- Identify parameters which can predict the transition to turbulence