

Nozzle Tip Wetting in Gasoline Direct Injection Injector and Its Link with Nozzle Internal Flow

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Abstract

Fuel film in the GDI Injector tip, or so-called nozzle tip wetting, has been found to be an important contributor of particle emissions. Attempts have been made to reduce the nozzle tip wetting by optimizing nozzle geometry designs. However, the inherent mechanism of the nozzle tip wetting formation and its link with nozzle internal flow is still unclear yet due to the lack of direct observations. To overcome this insufficiency, the nozzle internal flow and the formation process of the nozzle tip wetting were visualized in the real-scale aluminum nozzles using the X-ray phase-contrast technique. Results showed that the needle bouncing, injection pressure, and hole configuration affect the formation of the nozzle tip wetting, while the influence of needle bouncing is the most critical. A further study was conducted to examine the effect of nozzle counterbore diameter on the nozzle tip wetting. It was found that with an increase in counterbore diameter, the nozzle tip wetting slightly increased first and then decreased sharply after the counterbore diameter exceeded 0.40 mm. The mechanisms of the aforementioned phenomena were discussed in detail, which can contribute to the better understandings and control strategies of nozzle tip wetting.

Keyword: GDI engines, Particle emission, nozzle tip wetting, nozzle internal flow, needle motion

1. Introduction

Particle emissions from GDI engines have become a subject of concern. In the past emission regulations, particle emissions were considered as a matter of diesel engines only. In recent years, experimental evidence has shown that GDI engines may release more particle numbers (PN) than those of diesel engines in testing cycles and real driving conditions [1-3]. Consequently, the European emission regulations have started to limit

33 PN also for gasoline vehicles since 2014 [4]. In 2016, China has also announced its new emission standards with
34 a requirement of limiting the PN emission applying to light-duty vehicles and taking effect beginning on July
35 1, 2020 [5].

36 To comply with increasingly stringent emission regulations, the controlling particle emissions of GDI
37 engines has become a challenging issue [6-9]. The gasoline particulate filters (GPFs) has been considered
38 seriously as one of the essentials in GDI engines to meet the emission standards. However, investigations have
39 presumed that particle emissions might be controllable by the improved designs of the injector nozzle, and fuel
40 film in the GDI injector tip otherwise known as the nozzle tip wetting has been found to be an important
41 contributor of particle emissions [10-11]. There are two pathways that the nozzle tip wetting has been considered
42 to affect particle emissions. First, experiments have suggested that when the flame front of the main combustion
43 reaches the injector, the wetting fuel on the injector nozzle tip and in the counterbore could result in diffusion
44 flame due to the local rich air/fuel mixture [12]. This phenomenon, regarded as the tip sooting, could contribute
45 to the increased particle emissions. Second, injector deposit precursors are easily produced because of the
46 uncompleted combustion of the tip-wetting fuel [13-15]. The presence of injector deposits is able to absorb and
47 store more fuel during and after injection, leading to the even higher particle emissions [15]. Moreover, injector
48 deposits could cause the injector fouling eventually, which changes the supplying fuel flow rate and spray
49 pattern significantly in some cases, resulting in the further increased particle emissions. As a result, reducing
50 the nozzle tip wetting is one of the keys to controlling the particle emissions in GDI engines.

51 Studies have shown that the nozzle tip wetting mainly depends on the flow conditions inside the nozzle hole
52 and counterbore, which further relates to fuel properties, nozzle geometries, injection pressures, and nozzle
53 temperature conditions [16-20]. It was reported that increasing injection pressures could improve PN emissions
54 through reduced tip wetting [18]. *Oh et al.* reported that a decreased counterbore surface area could reduce the
55 nozzle tip wetting effectively [19]. *Fischer et al.* mentioned that rising fuel temperatures can lead to a reduction
56 in the nozzle tip wetting [20]. Although many parameters have been examined, the formation mechanism of tip
57 wetting and the influencing mechanism due to the parameter changes have been rarely discussed yet. This
58 insufficiency is mainly because of the lack of an experimental investigation to visualize the internal flow and
59 nozzle tip wetting behavior simultaneously.

60 The flow characteristics inside engine nozzles are inaccessible for the measurements using the conventional
61 light sources. Many attempts to visualize the nozzle internal flow have been conducted in the modeling nozzles,
62 which either have a different nozzle dimension or are studied in the very different boundary conditions from

63 engine nozzles. It has made difficult to derive the clear understandings of phenomena in engine nozzles through
64 the results of modeling nozzles. To overcome the above uncertainty, the X-ray phase-contrast imaging technique
65 was applied in this study to ensure direct observations of the flow characteristics both inside and outside the
66 nozzle. The nozzle tip-wetting phenomenon was examined in the real-size aluminum nozzles under the practical
67 injection conditions of GDI engines. The effects of counterbore diameter on the nozzle tip wetting were further
68 investigated. It was expected that the results obtained in this study can provide comprehensive understandings
69 of the formation mechanism of the nozzle tip wetting and its control strategies.

70 **2. Description of Experiments**

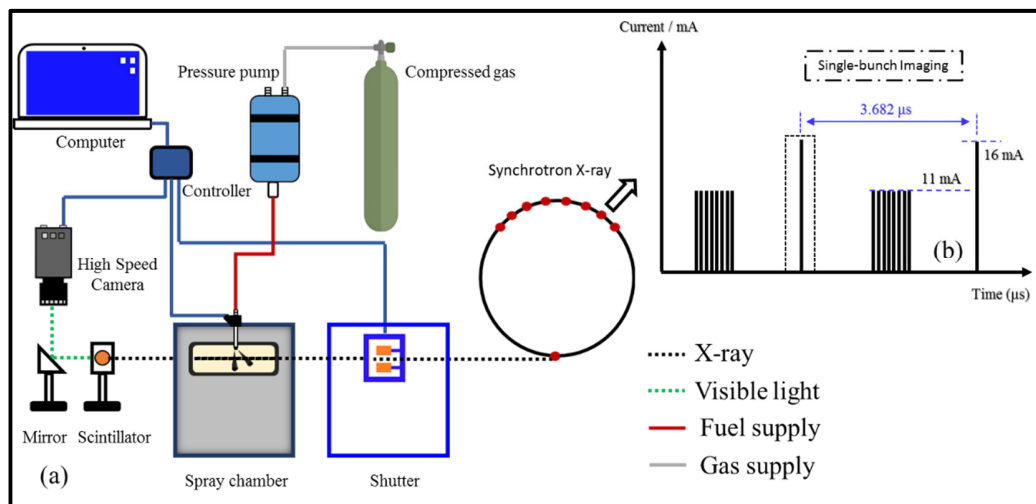
71 **2.1 X-ray Phase-Contrast Imaging**

72 When X-rays pass through an object, not only X-ray amplitude but X-ray phase is altered. According to an ideal
73 simulation, the phase-shift of an X-ray beam propagating through an object is much larger than the loss in
74 intensity thus making the phase-contrast imaging more sensitive to the density variations in the object than
75 absorption imaging ^[21]. This advantage could well contribute to imaging of the fuel injection process in the
76 near-nozzle field and to the revealing of the perturbations in fuel structures during breakups. The propagation
77 based phase-enhanced x-ray imaging was used in this study, which setup is very simple and basically the same
78 as conventional radiography. The only difference is that the detector is not placed immediately behind the
79 sample, but in some distance, so that the interference patterns between diffracted and undiffracted beam can be
80 recorded ^[22]. Our experiments were performed at the 7ID-B station at Advanced Photon Source (APS), where
81 provides a synchrotron X-ray source having the features of high brilliance, high collimation, and multiple pulse-
82 selections. Owing to the high brilliance of the synchrotron X-ray beam, the imaging of fuel injection in a
83 hundred-kilohertz rate with up to one-micrometer spatial resolution is achieved.

84 **2.2 Experimental setup**

85 Figure 1(a) shows the experimental setup in current study that resembles the one described in many of our
86 previous investigations ^[23-26]. In the hybrid-fill beam mode of APS, a single bunch containing 16 mA isolated
87 from the remaining bunches by symmetrical 1.594 microseconds gaps. The remaining current is distributed in
88 8 group of 7 consecutive bunches with a maximum of 11 mA per group, a periodicity of 68 ns, and a gap of 51
89 ns between groups, as presented in Figure 1(b). To protect the imaging system from the heat-load of X-ray
90 beams, a mechanical shutter is used to ensure that X-ray beams pass through the shutter only at the imaging
91 instant (8 ms opening duration). X-rays after penetrating through the sprays are converted to visible light using

92 a scintillator crystal (LuAg:Ce), which is then reflected by a 45° mirror and captured by a high-speed camera
 93 (Model SA-Z, Photron Inc.). In this study, the single bunch of the hybrid-fill beam mode was adopted for
 94 imaging because it provides a sufficiently high time resolution (150 ps). The camera frame rate was fixed to
 95 67889 Hz (one-fourth of the revolution frequency of the electron bunches in the storage ring) as regards a
 96 compromise between a wide view field and a fast frame rate.



97
 98 **Figure 1.** (a) Experimental setup for the X-ray phase-contrast imaging and (b) the X-ray electron bunches
 99 pattern of the hybrid mode in APS.
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101 2.3 Experimental nozzles

102 In this investigation, the GDI nozzles made by the aluminum were used. The nozzle holes and counterbore
 103 were made by drilling and without abrasive flow manufacturing (AFM). Using the aluminum nozzles ensures
 104 the sufficient image brightness to visualize the flow inside the nozzle due to its higher X-ray transmittance
 105 compared to the stainless nozzles while keeping the similar performance in terms of the metal surface as well
 106 as sufficient strength to endure engine-like pressure. These nozzles have only two holes, and the holes have a
 107 different configuration, as seen in Figure 2. The counterbore diameter effect was discussed in this study using
 108 seven nozzles in total. The counterbore diameter of these nozzles is varied with a step of 0.05mm from 0.25
 109 mm to 0.55 mm. Detailed descriptions of the nozzle geometries can be found in Table 1.

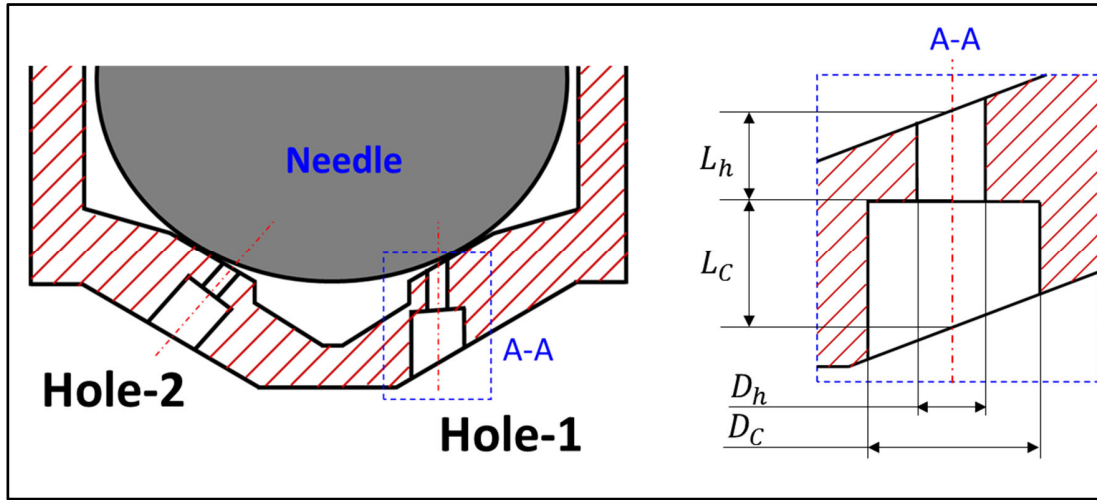


Figure 2. The nozzle internal structure and the definitions of the geometric parameters.

Table 1. Geometric parameters of test nozzles

Nozzle hole	Hole Diameter (D_h / mm)	Counterbore Diameter (D_c / mm)	Hole length (L_h / mm)	Counterbore length (L_c / mm)
Hole 1	0.16	0.25 ~ 0.55, 0.05/step	0.26	0.35
Hole 2	0.16		0.18	0.42

2.4 Analysis method

The quantitative analysis was conducted as regards the needle motions, the flow separations in nozzle hole and the projected area of the nozzle tip wetting. With these results, the tip wetting formation and its influencing parameters are discussed. The definitions and calculations of these parameters are described below.

Needle Motions

The needle lift can be obtained using the cross-correlation analysis. The calculation process is described below. First, a raw needle image before injection was selected as the base image. A region of interest covering the needle tip (base window) was set in the base image, standing for the initial needle location. Second, a searching window having the same size as the base window was set in the needle image to be calculated (at a different timing, regarded as the calculation image). The similarity factor (r) between the images cropped from these two windows could be obtained using the two-dimensional cross-correlation calculation, as equation (1).

$$r = \frac{\sum_m \sum_n (A_{mn} - \bar{A})(B_{mn} - \bar{B})}{\sqrt{(\sum_m \sum_n (A_{mn} - \bar{A})^2)(\sum_m \sum_n (B_{mn} - \bar{B})^2)}} \quad (1)$$

127 where A and B are the cropped images of the based window and the searching window, respectively, which have
128 a image size of $m \times n$. \bar{A} and \bar{B} stand for the averaged grey value of A and B . This similarity factor was
129 relevant to the location of the searching window, and its maximum would appear at where the cropped image
130 from the searching window perfectly matched with that cropped from the base window. Then, the needle lift
131 and wobbling could be derived based on the displacement of the search window relative to the base window.
132 The details on needle motion can also refer to our previous studies [27, 28].

133 Flow separations

134 In this study, the flow separations inside the nozzle hole have been observed. To quantify the characteristics,
135 an imaging processing method and calculation were used. Firstly, a background normalization together with the
136 contrast adjustment was applied to the raw images of the nozzle internal flow, as shown in Figure 3(a). Then,
137 the images at the steady state of injection and between shots were averaged as equation (2).

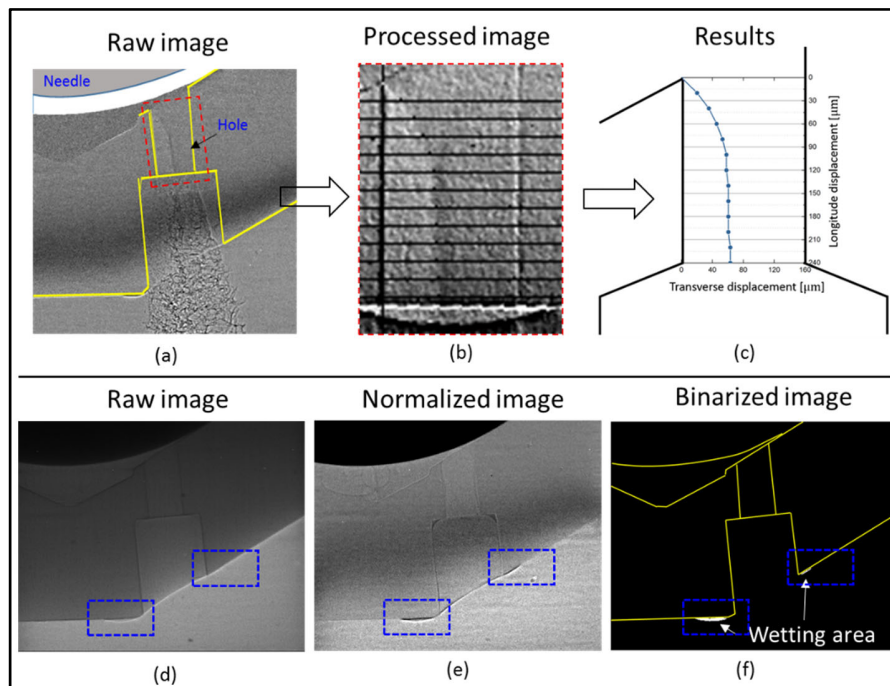
$$138 \quad I = \sum_{shots} \sum_{t_1}^{t_2} I(t) \times \frac{1}{shots} \times \frac{1}{n}, \quad [t_1 = 1.0 \text{ ms } t_2 = 2.0 \text{ ms}] \text{ and } shots \text{ equal to } 5. \quad (2)$$

139 where n stands for the number of images within the steady state (1.0 ms – 2.0ms). Finally, in the processed
140 image I , the profiles of flow separation were measured manually at different locations along the hole axis, as
141 shown in Figure 3(b) and (c). It should be explained that the flow separation fluctuates slightly during the
142 injection. Hence, the current method of measurement was used to guarantee the reliability of the results.

143 Projected area of the nozzle tip wetting

144 The characteristics of the nozzle tip wetting were quantified by examining its projected areas in the imaged
145 direction. It should be explained that the holes of GDI nozzles normally are concentrated to the center of nozzle
146 sac, and it can be observed that tip wetting occurs mostly at the nozzle exit. As a result, the projected area of tip
147 wetting can be regarded proportional to the practical tip wetting amount, and thus using the projected area might
148 as well ensure a valid examination of the tip wetting characteristics in this study. The calculation of the projected
149 area of tip wetting is described below. Firstly, similar to the flow-separation detection, the raw images were
150 processed by a background normalization together with the contrast adjustment, as seen in Figure 3(d) and (e).
151 Then, the normalized image was further converted to a binarized image with an identical threshold applied for
152 all the nozzles, as shown in Figure 3(f). Tip wetting was highlighted as the white region at the hole exit, and a
153 consistent trend can be confirmed by comparing the images of Figure 3(e) and (f). At last, tip wetting area can
154 be calculated by multiplying the total pixel numbers of the white region in Figure 3(e) with the pixel resolution.
155 In this study, the imaging setup with an image resolution of 2.53 μm per pixel was achieved.

156 In addition, it might be of concern that a white fringe appeared at the liquid surface due to the phase contrast
 157 mechanism, which can affect the boundary definition of the tip wetting fuel. It is confirmed that the white fringe
 158 thickness is almost the same regardless of the tip wetting amount, This might be because a short sample-to-
 159 lense distance (30cm) is used in this study, where the phase contrast is relatively weak. Besides, the same
 160 threshold have been applied to define the fuel boundary of tip wetting during image processings, and multiple-
 161 shots averaged results have been used for discussions.



162 **Figure 3.** parameter definitions and calculation methods. (a-c) stand for a calculation of flow separation,
 163 while (d-f) show the calculation of the nozzle tip wetting area.
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166 2.5 Experimental conditions

167 The experimental conditions were summarized in Table 2. Three different injection pressures were used to
 168 examine the injection pressure effect on tip wetting, including the injection pressures of 40 bar, 100 bar, and
 169 200 bar. The spray chamber was filled with Nitrogen. The ambient gas pressure and temperature in the spray
 170 chamber were fixed to the atmospheric and room condition, respectively. The injection-pulses duration was 2
 171 ms and N-heptane was used instead of gasoline in consideration of the experimental safety. N-heptane has a
 172 similar surface tension value with gasoline that is very important regarding the tip wetting investigation.
 173 Another reason for utilizing n-heptane is because n-heptane has been historically used as a representative of the

174 n-alkane group due to its octane number and its concentration in gasoline fuel gasoline, which can mimic the
175 combustion characteristics of gasoline fuel, especially in low-temperature combustion regime [29]. This feature
176 can also benefit to a link with the low-temperature combustion study in engines. Measurements were repeated
177 five times at each testing condition and the shot-averaged results were obtained for the analysis below.

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179 **Table 2.** Experimental conditions.

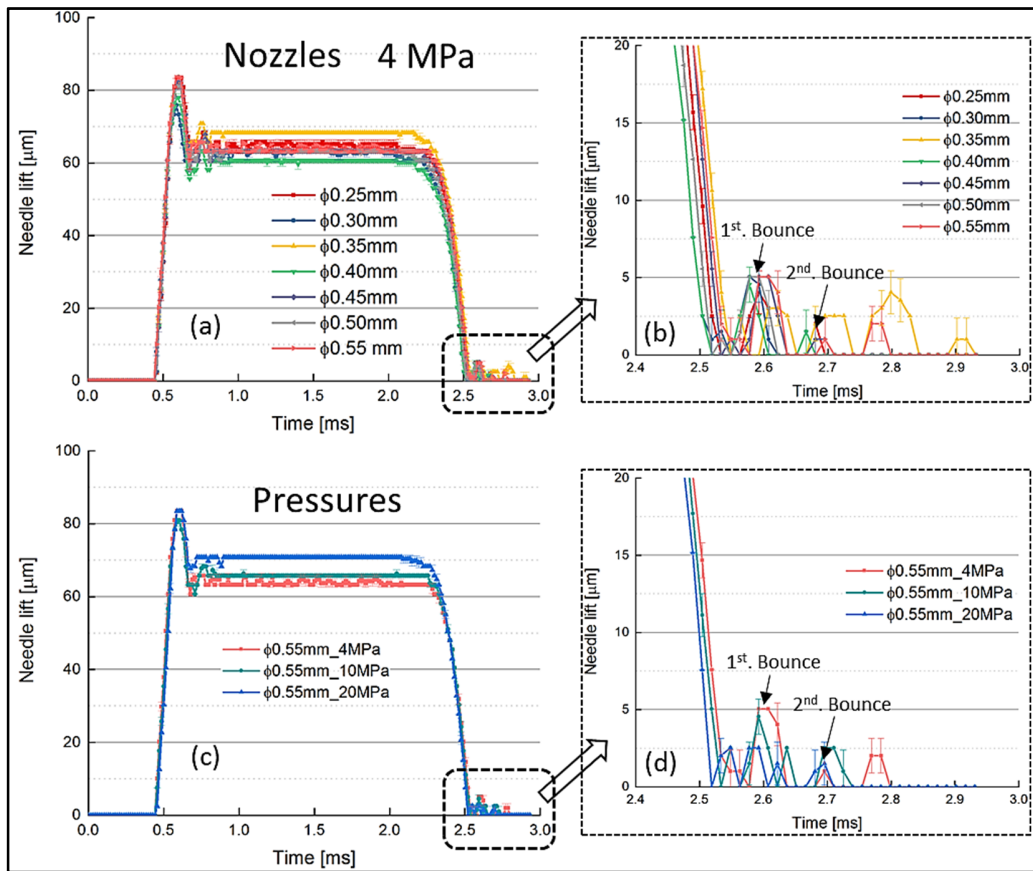
Injection (bar)	40, 100, 200
Ambient gas pressure (bar)	1.0
Ambient gas temperature (K)	Room condition (298)
Ambient gas density (kg/m ³)	1.25 kg/m ³ (N ₂)
Fuel	n-heptane
Injection pulse (ms)	2.0 ms

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181 **3. Results and Discussions**

182 **3.1 Needle motion of test nozzles**

183 Figure 4 shows the needle-motion results. These results are averaged between shots and the error bars are
184 presented as well. Several features are evident in Figure 4. First, it is clear that the error bars are short, indicating
185 good repeatability of the nozzle needle motion between shots. Second, Figure 4(a) and (b) show that the needle
186 motions are slightly different among the nozzles as regards the height of the initial lifting peak, at the steady
187 state, and at the end of injection. There is not a clear pattern linking the needle-motion differences to counterbore
188 diameters so that the difference is considered to result from manufacturing tolerance. More discussions can be
189 found in section 3.2.2. Third, the needle bouncing at the end of injection is observed in the nozzles with a similar
190 amplitude. Most of the nozzles bounce twice during needle closing. Fourth, Figure. 4(c) and (d) reveal that
191 increasing injection pressures result in a higher needle-lift height at the steady state. In addition, the needle lift
192 declines slightly faster at higher injection pressures, however, the needle-bouncing amplitude becomes smaller.
193 This phenomenon may result from that the higher injection pressure could press the upper part of the needle tip
194 with a stronger force so that the needle closes faster and the afterward needle bouncing attenuates. The injection-
195 pressure effect is consistent in the nozzles.



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Figure 4. The results of needle motion. The counterbore diameter effect was presented in a full injection event as (a) and focused at the end of injection as (b), while the injection pressure effect was presented in a full injection event as (c) and focused at the end of injection as (d).

201 3.2 Nozzle tip wetting characteristics

202 3.2.1 Effects of needle motion, injection pressures, and hole configuration

203 Nozzle tip wetting results are presented in this section. Figure 5 shows the formation process of tip wetting with
 204 the images shown at different needle lifts. The 'first close' in Figure 5 means the time when the needle-lift
 205 height declines to zero before the 1st needle bouncing. The results shown below come from the nozzle with a
 206 counterbore diameter of 0.3 mm. The identical result trend can be found in the other nozzles as well. First, it is
 207 found that the nozzle tip wetting in Hole-1 appears soon after the beginning of injection, while interestingly in
 208 Hole-2 the nozzle tip wetting does not appear until the needle-bouncing period. A large number of fuel drippings
 209 come out during the needle-bouncing period in both nozzle holes. Hole-1 has a much severer tip wetting
 210 compared to Hole-2.

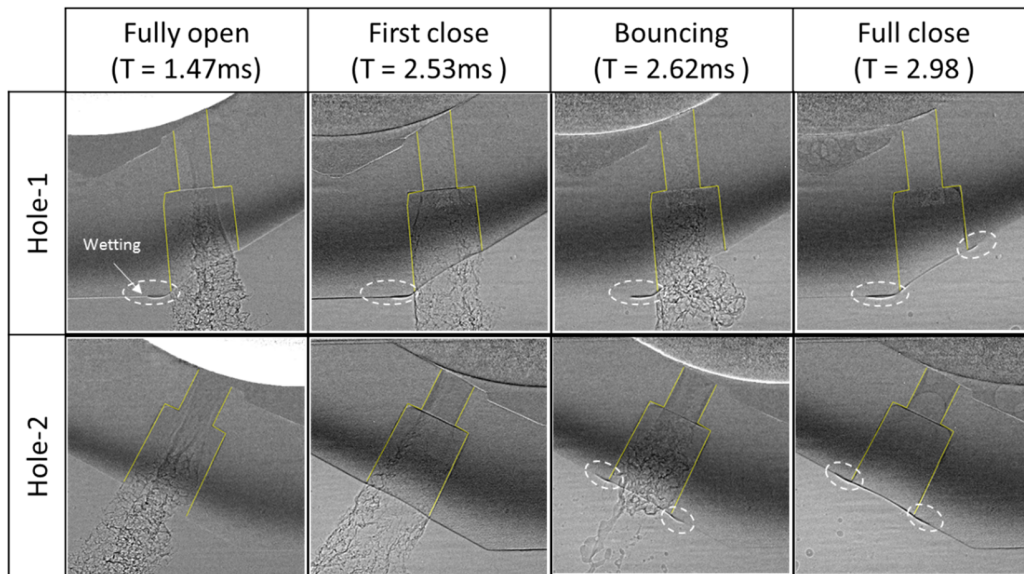
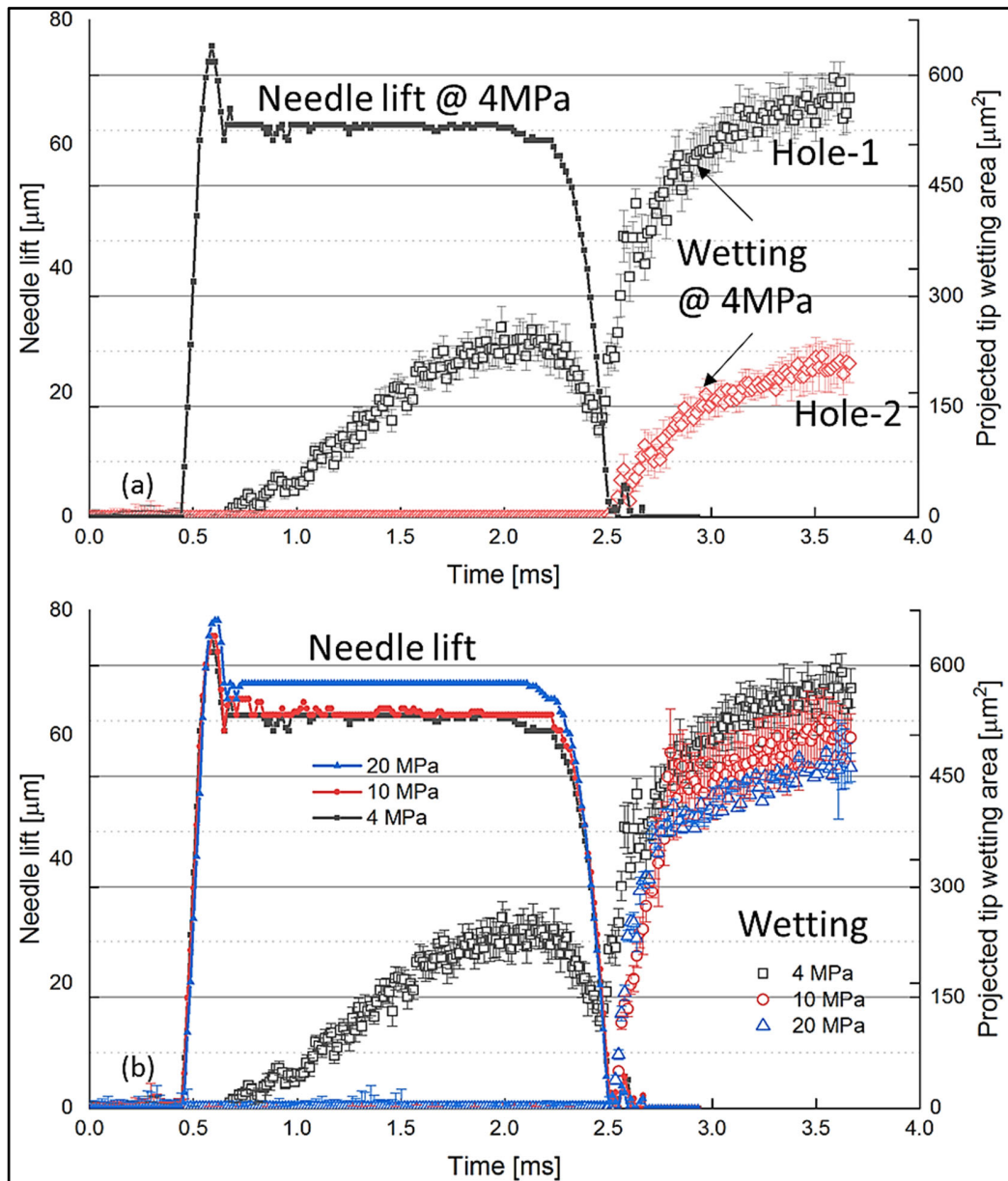


Figure 5. The nozzle tip wetting at the different needle lifts. Results from the nozzle with a counter diameter of $\varnothing 0.30$ mm and under an injection pressure of 4 MPa are presented as a sample.

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215 A quantitative analysis of the aforementioned phenomenon was conducted as regards the projected tip-
216 wetting area at the nozzle exit. Figure 6(a) shows that the tip-wetting amount of Hole-1 increases gradually
217 during injection, then decreases as the needle is declining, and sharply increases again during needle bouncing.
218 It might be of particular interest that the nozzle tip wetting decreases when the needle is declining, and this
219 phenomenon can be observed in all the nozzles. A discussion on this will be presented later. After the needle is
220 fully closed (the needle bouncing is terminated, when is approximately 2.7 ms in Figure 6), the nozzle tip
221 wetting still keeps increasing but much slower compared to that during needle bouncing. The higher injection
222 pressures result in the less tip wetting, as shown in Figure 6(b). The nozzle tip wetting during the injection
223 happens in Hole-1 and under the injection pressure of 4 MPa only. The tip-wetting amount of Hole-1 is
224 approximately three times larger than that of Hole-2. The results above indicate that the needle bouncing,
225 injection pressure, and hole configuration affect the formation of tip wetting, while the influence of needle
226 bouncing is the most critical.



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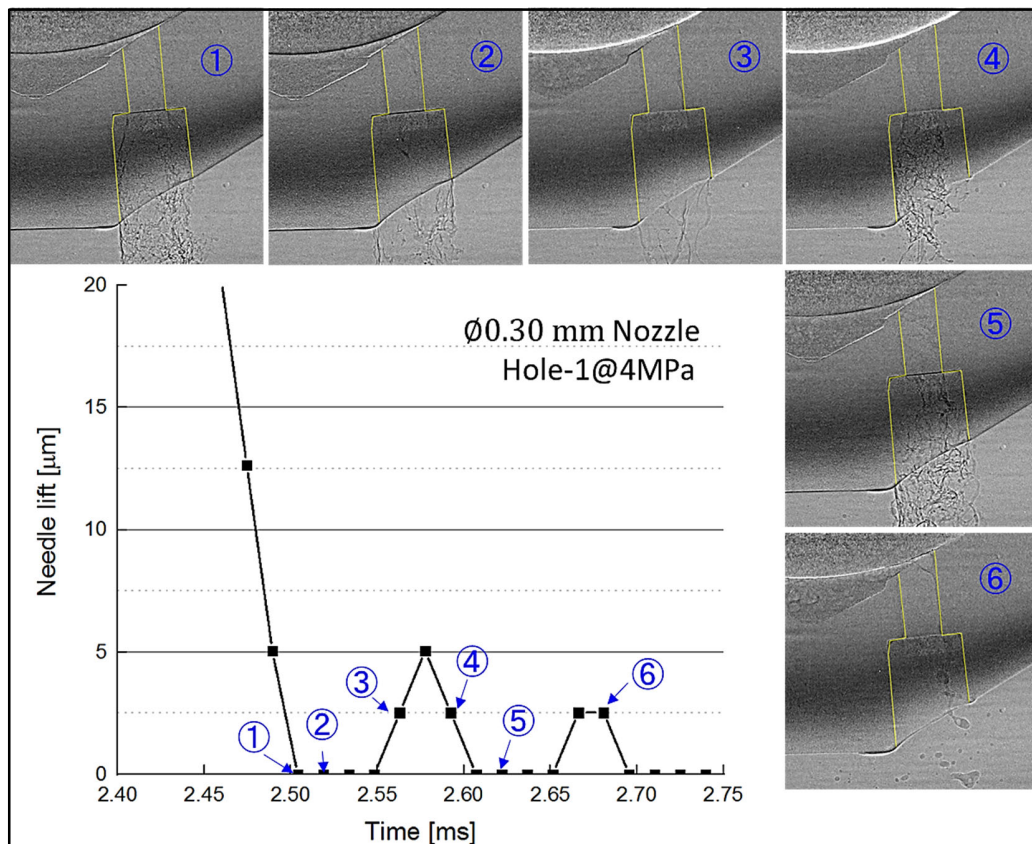
Figure 6. Results of the projected tip-wetting area. Results are compared between the Hole-1 and Hole-2 as (a), while (b) presents the results between different injection pressures from Hole-1. $\emptyset 0.30\text{mm}$ nozzle is used here standing for the identical trend within the nozzles.

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Several observations in Figure 6 deserve further examinations. First of all, it is found that the needle bouncing has the most significant effect on tip wetting. To clarify this influence, Figure 7 reveals the detailed process with the images shown in a time sequence. It is found that after the needle is closed (Timing-1 to-3), fuel still flows down along counterbore walls, resulting in the first increase of the nozzle tip wetting. This can be

236 understood because, at the instance of the initial needle close, the pressure in nozzle sac is still higher than the
 237 ambient outside the nozzle. The pressure differential pushes out the residual fuel in nozzle sac making tip
 238 wetting increase. During the 1st needle bouncing (Timing-3 to-5), the spray-like fuel droppings appear with a
 239 large amount which are quite dispersive so that they touch counterbore walls and increase tip wetting
 240 significantly. During the 2nd needle bouncing (Timing-6 and after), the ligament-like fuel droppings appear,
 241 which incline to the right side in Hole-1, and attach to the counterbore wall in some shots. This can further
 242 increase tip wetting.

243 The needle bouncing effect could be as well responsible for the reduced tip wetting in high injection pressure
 244 conditions. As described earlier, the amplitude of needle bouncing attenuates as an increase in injection
 245 pressures, contributing to a reduced tip wetting. Besides, at high injection-pressures, it is argued that fuel
 246 droppings have relatively larger momentum to overcome the surface-tension effect against counterbore walls.
 247 This may also reduce tip wetting. The injection pressure effect on tip wetting can be explained based on these
 248 considerations.



249 **Figure 7.** Needle-bouncing effect on tip wetting. (Ø0.30mm nozzle under 4 MPa)
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251 It is found that the nozzle tip wetting of Hole-2 is much smaller than that of Hole-1. The reasons for this are
252 considered separately during injection (means the steady-state, 1.0 ms – 2.0 ms) and at the end of injection
253 (means from Timing-1, about 2.5 ms in Figure 7). First of all, during injection, tip wetting occurs in Hole-1
254 only. Figure 8 shows the flow separation profiles derived from the time-average results at the steady-state (1.0
255 ms – 2.0 ms). It is clear that there is a larger and single-sided flow separation in Hole-1 with an asymmetrical
256 spray spreading in its counterbore, while Hole-2, by contrast, has a smaller and both-sided flow separations
257 with the continuous and clear spray boundaries inside the counterbore. Our previous measurements have
258 revealed that the spray velocity is much lower on the flow separation side (left side in the image) compared to
259 that on the opposite side (right side in the image). In addition, *Itaya et al.* have reported in their numerical
260 investigation that the flow separation enables the air entrainment backward to nozzle hole, which enhances flow
261 turbulence resulting in the generation of small-size droplets on the flow separation side [30]. Based on these
262 considerations, it is argued that the larger flow separations in Hole-1 benefit the formation of the small-size and
263 low-velocity droplets that can be easily trapped on the counterbore wall under the surface-tension effect. As a
264 result, the nozzle tip wetting during injection is observed in Hole-1. On the other hand, increasing injection
265 pressures can ensure the sprays and droplets inside the counterbore having larger momentum to overcome the
266 surface tension and to flush the impinging fuel out. Consequently, the tip wetting during injection occurs only
267 in Hole-1 and under the injection pressure of 4 MPa.

268 In addition, it is found that tip wetting decreases when the needle is declining in Figure 4. It has been observed
269 that the flow separations fluctuate more and become smaller during needle declining compared to that at the
270 steady state. This phenomenon should be because of the enhanced flow turbulence when the needle is closing.
271 As explained earlier, the smaller flow separations could reduce the generation of small-size droplets thus
272 making tip wetting become less.

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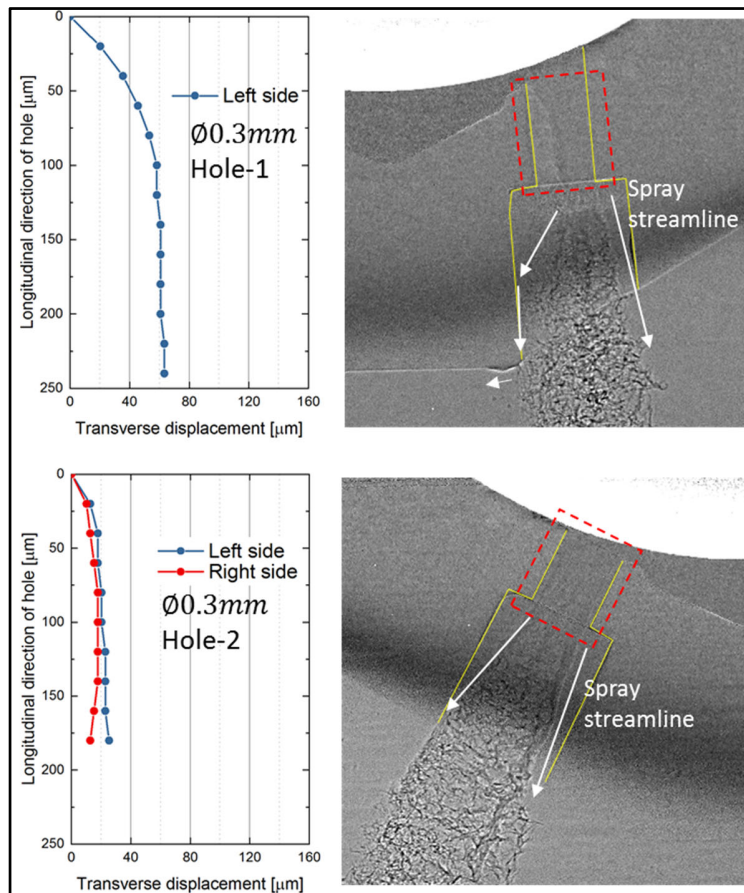


Figure 8. Comparisons of flow separation between Hole-1 and Hole-2. The flow separation results are shown on the left side, while two sampling images are presented on the right side. Results under an injection pressure of 4 MPa are used as a sample.

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279 Furthermore, at the end of injection, tip wetting is found both in Hole-1 and Hole-2, but it increases more
280 sharply in Hole-1. The reason for this can be explained with the images shown in Figure 9 (The ‘Timing-4’ in
281 the figure equals to that shown in Figure 7). It can be seen that during needle bouncing, the spray-like fuel
282 drippings appear both in Hole-1 and Hole-2. However, a larger dispersion of fuel drippings is observed in Hole-
283 1, which increases the tip-wetting amount apparently as seen in Figure 9(b) and (c). The fuel drippings in Hole-
284 2 mostly maintain a continuous boundary and attach less to the counterbore wall. Above differences should
285 result from the different configuration of the nozzle hole inlet passage. At the end of injection (the needle has
286 closed), fuel drippings root from the residual fuel in the nozzle sac. Due to the smoother hole inlet, the fuel
287 entering Hole-2 changes less in its flow direction and thus can maintain more momentum in the axial direction
288 of the fuel streams together with a smaller level of dispersion. This may explain the less tip wetting in Hole-2.

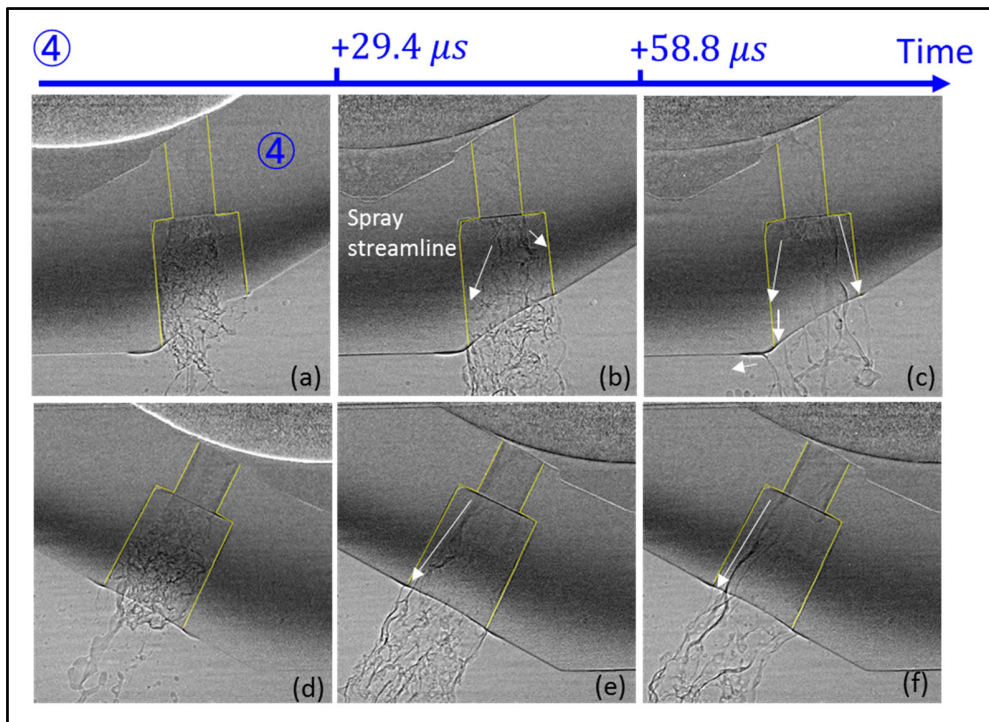


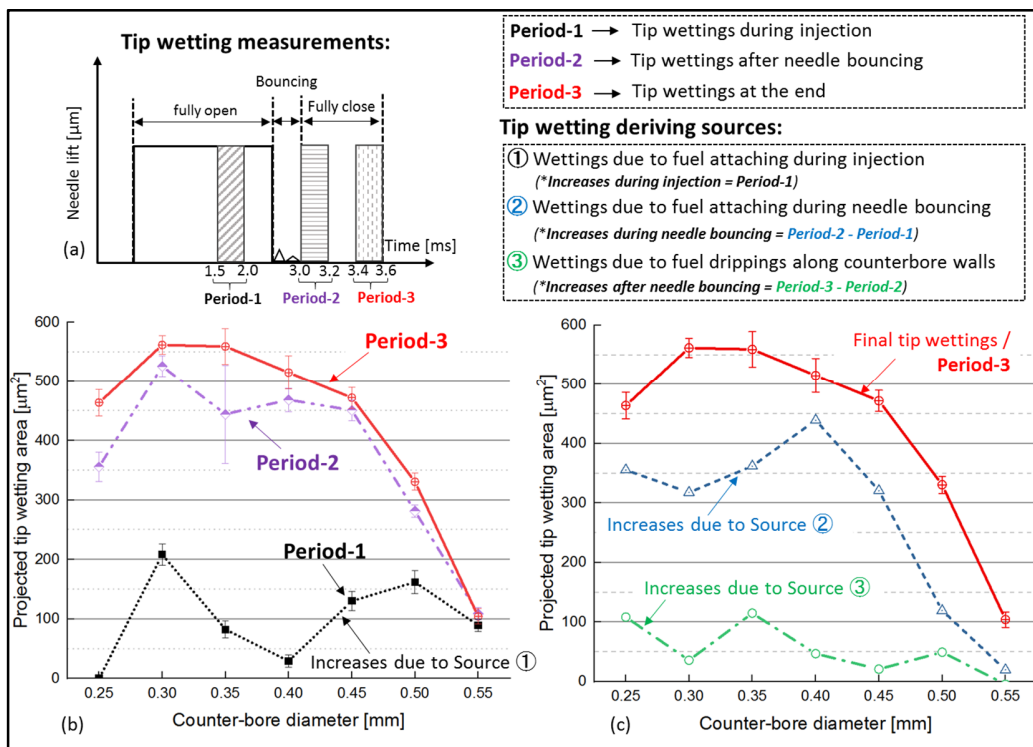
Figure 9. Comparisons of the tip wetting between the Hole-1 (a – c) and Hole-2 (d – f). Results are derived from the $\varnothing 0.30$ mm nozzle and under an injection pressure of 4 MPa.

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3.2.2 Effects of counterbore diameters on the nozzle tip wetting

294 In the section above, it has explained that the needle bouncing, injection pressure, and hole configuration can
295 affect the formation of tip wetting, while the influence from the needle bouncing is the most critical. In this
296 section, the counterbore diameter effect on tip wetting is discussed. Discussions are made in three periods
297 separately, referring to during injection, during needle bouncing, and after the termination of the needle
298 bouncing. First of all, Figure 10(b) shows that the tip wetting during injection (results in Period-1) is quite small
299 and varies without a pattern as an increase in counterbore diameters. Regarding the tip-wetting increase during
300 needle bouncing, it can be estimated by subtracting the results of Period-1 from Period-2, as the blue dash line
301 shown in Figure 10(c). It is of particular interest that the tip wetting during needle bouncing increases slightly
302 as the counterbore diameter increases from 0.25 mm to 0.40 mm and then decreases sharply from 0.40 mm to
303 0.55 mm. After the needle bouncing terminates, it is observed that the residual fuel in nozzle sac still keeps
304 flowing down along the counterbore wall (regarded as ‘the drippings along counterbore walls’). This potential
305 source to tip wetting can be estimated by subtracting the amount in Period-2 from Period-3, as the green dash-
306 dot line shown in Figure 10(c). It is clear that the nozzle tip wetting due to the drippings along counterbore

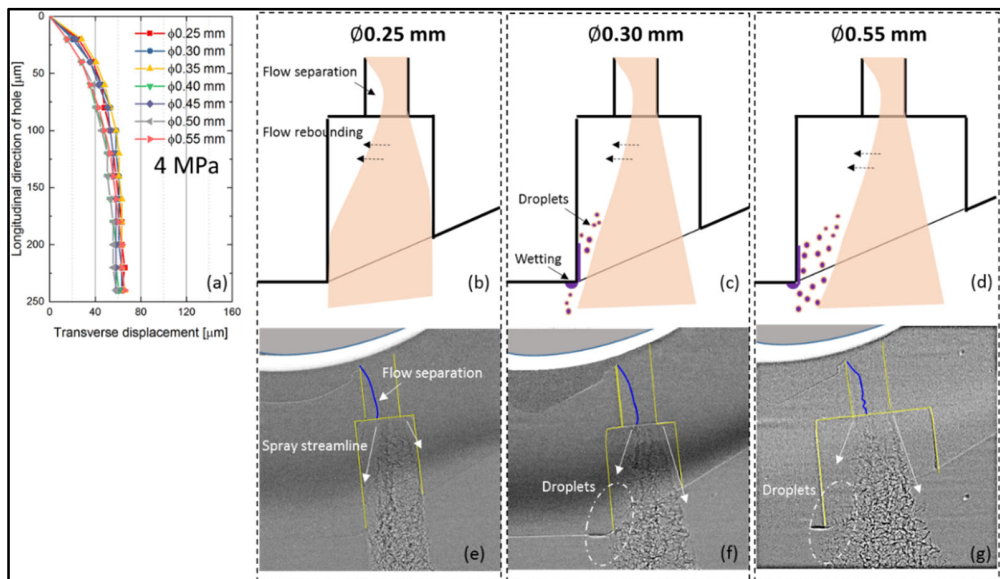
307 walls decrease gradually as an increase of counterbore diameters. The final nozzle tip wetting of each nozzle
 308 depends on the combination of the processes described above, which results can be seen as the red solid line in
 309 Figure 10(c) (measured at Period-3). The final nozzle tip wetting reaches the maximum when the counterbore
 310 diameters of 0.30 mm and 0.35 mm are used. It is clear that the final nozzle tip wetting is dominated by the fuel
 311 drippings during needle bouncing because they have the most resembling tendency. An improved nozzle design
 312 to control the drippings flow against the counterbore and reduced needle bouncing could contribute to a less
 313 formation of tip wetting.



314 **Figure 10.** (a) Tip wetting measurements for a comparison of the counterbore-diameter effect. (b) Results of
 315 projected tip wetting areas measured at different periods. (c) An examination of the tip wetting deriving
 316 sources by using the results of (b).
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 318

319 Further discussions can be made on the features observed in Figure 10. First of all, during injection, the
 320 amount of tip wetting varies among the nozzles without a clear pattern. The potential reasons for this trend are
 321 considered below. First, the effect of counterbore diameters on the flow separation in the nozzle hole is
 322 examined. Figure 11(a) reveals that the flow separations are identical among the nozzles, indicating a similar
 323 flow dynamics in the nozzle hole and counterbore regardless of counterbore diameters. Figure 11(b-g) show
 324 further details that in the nozzle with a counterbore diameter of 0.25 mm, sprays hit the counterbore wall shortly

325 after leaving the nozzle hole, and no tip wetting is observed. It is considered that because the sprays spread little
 326 when the counterbore diameter is small, they hit the counterbore wall with strong momentum so that tip wetting
 327 can be flushed out. From the nozzle with a counterbore diameter of 0.30 mm, as shown in Figure 11(c) and (f),
 328 a clearance between the spray mainstream and the counterbore wall is observed, which might initiate the air
 329 entrainment into the counterbore so that the small-size droplets with low velocity can form tip wetting
 330 substantially. As the diameter further increases to 0.55 mm, as shown in Figure 11(d) and (g), the clearance
 331 between the spray and the counterbore wall further increases. However, individual droplets detached from the
 332 spray mainstream are observed and they still hit on the counterbore wall with the formation of tip wetting.



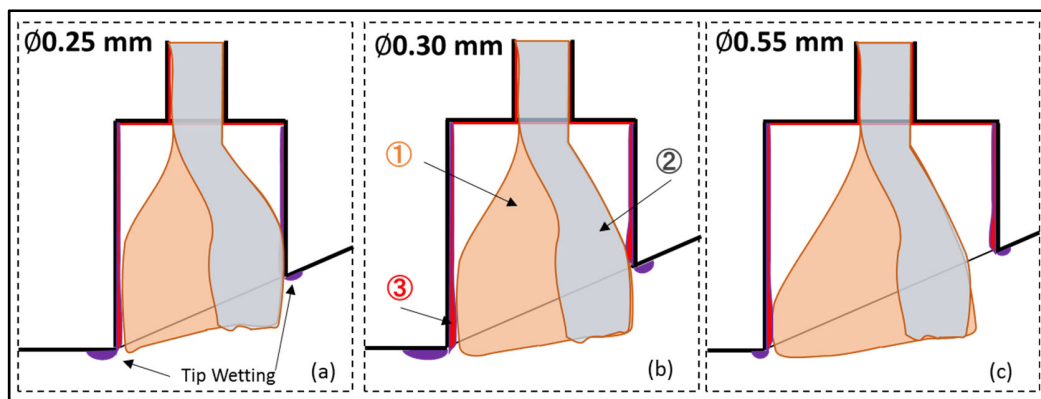
333
 334 **Figure 11.** Counterbore diameter effect on the nozzle tip wetting during the injection: (a) flow separation
 335 profiles; (b-d) sketches of the flow pattern in small counterbore diameter and large counterbore diameter; (e-
 336 g) samples images (Results of the Hole-1 under the injection pressure of 4 MPa)
 337

338 During needle bouncing, it is found in Figure 10(b) that tip wetting increases slightly from the counterbore
 339 diameter of 0.25mm to 0.40 mm, and then decreases sharply afterward from the counterbore diameter of 0.45
 340 mm to 0.55 mm. The reasons are considered below. First of all, Figure 4 has shown that the nozzles have an
 341 identical amplitude of needle bouncing at the end of injection. Thus, it is excluded that the variance in nozzle
 342 tip wetting results from the different needle bouncing. Second, Figure 12(a-c) reveals that during the 1st and 2nd
 343 needle bouncing, the spray-like (left side) and ligament-like (right side) drippings occur, respectively. Based on
 344 our observations, two effects are coupled to decide the fuel drippings turning to tip wetting. When the
 345 counterbore diameter is small, drippings hit the counterbore wall quickly with relatively larger momentum due

346 to the fewer spreadings so that the wetting fuel can be flushed out, as illustrated in Figure 12(a). As the
 347 counterbore diameter increases, the momentum of fuel drippings at the instant of wall impingement attenuates
 348 due to the wider spreading of fuel drippings. As a result, fuel drippings can be more easily trapped on
 349 counterbore walls under the surface-tension effect. With a further increase in the counterbore diameter from 0.4
 350 mm, tip wetting decreases sharply due to the increasingly fewer drippings hitting on counterbore walls, as
 351 depicted in Figure 12(c).

352 After the needle bouncing terminates fully, fuel still flows down along counterbore walls. However, tip
 353 wetting increases less at larger counterbore diameters, as shown in Figure 10(b). It is considered that this is
 354 because the larger counterbore diameter can provide the more wetting surface. Therefore, fuel is trapped on
 355 counterbore walls more due to the surface tension effect, resulting in the less tip wetting formed. In addition, it
 356 is noticed that *Oh et al.*^[19] reported a suppressed tip wetting as the counterbore surface area decreases
 357 (equivalent to decreasing counterbore diameter in this study). Their conclusions slightly deviate from those of
 358 the current study, which shows a non-linear effect of counterbore diameter on tip wetting. The reason for this
 359 inconsistency is considered because in *Oh et al.*'s investigation only two nozzles with small counterbore
 360 diameter have been tested. The counter-diameter effect might have been revealed in a limited range.

361 Furthermore, it is noteworthy that the final nozzle tip wetting of each nozzle is determined by the nozzle tip
 362 wetting during the whole injection event, including the wetting during injection, the drippings during needle
 363 bouncing and the drippings after needle bouncing. It deserves further investigations to cover more influencing
 364 parameters and to achieve better understandings of the phenomenon, which benefit to propose the control
 365 strategies of tip wetting.



366 **Figure 12.** Counterbore diameter effect on the nozzle tip wetting at the end of injection: (a-c) sketches of the
 367 flow patterns of the nozzles. ① and ② represent spray-like drippings and ligament-like drippings,
 368 respectively. ③ represents the drippings along counterball walls.
 369

370 4. Conclusions

371 The nozzle tip wetting has been considered an important contributor to the particle emissions in GDI engines.
372 In this study, tip wetting was examined in the real-size aluminum nozzles under the practical injection conditions
373 of GDI engines. By taking advantages of the X-ray phase-contrast imaging technique, the needle motion and
374 the transient flow characteristics inside nozzle holes and counterbores were visualized and quantified by
375 parameters. A detailed analysis was conducted to understand the formation mechanism and influencing factors
376 of tip wetting. The key conclusions of the current study can be summarized as below.

- 377 1. The needle bouncing, injection pressure, and hole configuration can affect the formation of tip wetting,
378 while the influence from the needle bouncing is the most critical.
- 379 2. During needle bouncing, spray-like fuel drippings can be observed with a large amount. Once the
380 drippings attach to counterbore walls, a significant increase in tip wetting occurs.
- 381 3. Increasing injection pressures can attenuate the needle bouncing and enhance the dripping momentum,
382 resulting in a reduced tip wetting. In addition, the comparisons between Hole-1 and Hole-2 show that
383 the suppressed flow separation in nozzle hole due to the different hole configuration can contribute to
384 the concentrated drippings, leading to a less tip wetting formed.
- 385 4. Tip wetting slightly increases when the counterbore diameter changes from 0.25 mm to 0.35 mm and
386 then decreases sharply afterward from 0.40 mm to 0.55 mm. This non-linear trend depends largely on
387 the interaction between fuel drippings and counterbore walls during needle bouncing. It is discussed
388 that two effects are coupled and responsible for the phenomenon, referring to the momentum of fuel
389 drippings at the instant of wall impingement and the possibility for fuel drippings to interact with
390 counterbore walls.

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